

UNCLASSIFIED

AD NUMBER
AD216542
NEW LIMITATION CHANGE
TO Approved for public release, distribution unlimited
FROM
AUTHORITY
DSWA ltr dtd 28 Apr 1997

THIS PAGE IS UNCLASSIFIED

UNCLASSIFIED
AD

216542

FOR
MICRO-CARD
CONTROL ONLY

1 OF 1

Reproduced by

Armed Services Technical Information Agency

ARLINGTON HALL STATION; ARLINGTON 12 VIRGINIA

UNCLASSIFIED

"NOTICE: When Government or other drawings, specifications or other data are used for any purpose other than in connection with a definitely related Government procurement operation, the U.S. Government thereby incurs no responsibility, nor any obligation whatsoever, and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications or other data is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use or sell any patented invention that may in any way be related thereto.

UNCLASSIFIED

AFSWP-1117

FC

Copy No. 1

AD 110 216 542

ASTIA FILE COPY

THEORETICAL ANALYSIS OF RADIOMETER PERFORMANCE

Research and Development Report AFOSR-77-0141-1

U.S. NAVAL RADIOLOGICAL DEFENSE LABORATORY

SAN FRANCISCO 24 CALIFORNIA

FILE COPY

ROUTED TO

ASTIA

ARLINGTON HALL STATION

ARLINGTON 12, VIRGINIA

ATTN: TISS

ASTIA
RECEIVED
JAN 24 1978
TICK E

UNCLASSIFIED

U N C L A S S I F I E D

THEORETICAL ANALYSIS OF RADIOMETER PERFORMANCE

Research and Development Technical Report USNRDL-TR-311

NS-681-001

13 January 1959

by

H. G. Ferris

Physics and Mathematics

Technical Objective

AW-7

Thermal Radiation Branch

W. B. Plum, Head

Nucleonics Division

A. Guthrie, Head

**Requests by military activities or contractors
for additional copies of this report should be
submitted to ASITA**

**Scientific Director
P. C. Tompkins**

**Commanding Officer and Director
Captain J. H. McQuinn, USN**

**U. S. NAVAL RADIOLOGICAL DEFENSE LABORATORY
San Francisco 24, California**

U N C L A S S I F I E D

ABSTRACT

A theoretical analysis of the thermal behavior of water-cooled thin foil radiometers is carried out. The expression for the heat balance in the circular foil of the radiometer is:

$$\frac{1}{k} \frac{\partial}{\partial r} \left(r \frac{\partial T}{\partial r} \right) + \frac{1}{r} \frac{\partial T}{\partial r} + \frac{q}{kd} = \frac{\partial T}{\partial t}$$

The steady state solution, $T/t = 0$, subject to certain boundary conditions, yields the thermal sensitivity: $\Delta T/q = R^2/4kd \left(\frac{0.5}{\text{Cal./cm}^2 \cdot \text{sec}} \right)$.

This relationship when applied to the three radiometers in common use at NRDL indicates that they operate linearly over their entire ranges (0 to 10, 0 to 50, and 0 to 100 Cal./cm²) with a maximum error of 2 percent when maximum irradiance is incident on the foil. Considering k constant, the heat balance equation becomes:

$$\frac{\partial^2 T}{\partial r^2} + \frac{1}{r} \frac{\partial T}{\partial r} + \frac{q}{kd} = \frac{1}{\tau} \frac{\partial T}{\partial t}$$

Solving, subject to the same boundary conditions, by means of the Laplace Transformation, an expression for reaction time results.

$\tau = \frac{0.416 R^2}{k}$. Applying this relationship to the three NRDL instruments yields reaction times of 0.010, 0.004, and 0.004 sec. respectively.

SUMMARY

The Problem

High intensity beams of thermal radiation as a function of time have, in recent years, been measured through the use of thin foil radiometers. One of the principles of operation of these instruments has been that a heat sink in the instrument remains at a constant temperature throughout the measurement of a thermal flux. The thin foil changes in temperature and hence measures the flux. However, it is believed that in actual field tests involving high intensity thermal radiation, the heat sink does not remain at a constant temperature.

Findings

The present study is a theoretical analysis of the thermal behavior of a water-cooled radiometer with particular consideration given to (1) the steady state sensitivity of the thin foil and (2) the reaction time of the foil (time to come to 90 percent of steady state temperature above its heat sink after onset of irradiance). The three radiometers in common use at NRDL (1) operate linearly over their entire ranges (0 to 10, 0 to 50, or 0 to 100 Cal/cm²) with a maximum error of less than 2 percent when maximum irradiance is incident on the foil; and (2) have reaction times of 0.016, 0.004 and 0.004 sec., respectively.

ADMINISTRATIVE INFORMATION

I. BACKGROUND OF WORK

During FY 1951, the U. S. Naval Radiological Defense Laboratory was given the responsibility by the Armed Forces Special Weapons Project to develop instrumentation for measuring the irradiances and thermal exposures for nuclear detonations. The field radiometer that was developed under this program was found inadequate for laboratory measurements. The need for a reliable instrument for measuring the spot distribution of thermal irradiance in the focal plane was mandatory for the successful prosecution of laboratory-type programs.

The water-cooled radiometer was then conceived and developed by the Thermal Radiation Studies (now Thermal Radiation Branch), Nucleonics Division, Scientific Department, in approximately October 1951 for the primary purpose of continually monitoring the NRL 36" thermal source. Subsequent developments of the water-cooled radiometer led to fabrication of improved water-cooled radiometers designed for determining the spot distribution of the NRL 36" and modified Mitchell thermal sources capable of simulating thermal pulses produced by a nuclear detonation. The 36" and Mitchell thermal sources are capable of producing a range of energy levels from 3/8" spot diameter at approximately 100 cal/cm²/sec and a range from 1/4" to 1 1/2" diameter spot at intensities from 28 cal/cm²/sec to 15 cal/cm²/sec. The water-cooled radiometer has been utilized as a secondary standard in the calibration of the NRL thermal sources.

Water-cooled radiometers have been developed, fabricated and calibrated by this laboratory for use by virtually all Department of Defense organizations that are engaged in the calibration of thermal sources and solar furnaces.

This report is concerned with a refinement of the theoretical analysis of the radiometer for both the steady and transient states for the measurement of thermal irradiance.

II. AUTHORIZATION AND FUNDING

The work discussed in this report was initially sponsored by the Bureau of Ships, under Project Number NS-051-901, entitled "The Effects of Atomic Warheads and Radiological Shielding," Subtask 3.4, "Department of Laboratory Sources of Thermal Radiation," Technical Objective AW-7. Financial support

support of this type of program has been continuously funded by the Armed Forces Special Weapons Project.

III. DESCRIPTION OF WORK

The work reported herein is concerned with the application of the Laplace Transformation Theory to the solution of the transient Heat Flow Problem. The solutions to the problem give a measure of the sensitivity of the water-cooled radiometer in terms of its physical dimension as well as the response time required to reach 90 percent of the steady state condition.

INTRODUCTION

Thin foil radiometers for the measurement of high intensity beams of thermal radiation as a function of time have come into increasing use in the determination of the flux from solar furnaces and nuclear detonations, as well as from laboratory sources of high intensity thermal radiation. One such type of instrument is shown in Fig. 1. It consists basically of a thin foil mounted over a hole in a heat sink which is maintained at a constant temperature by a circulating stream of water. The temperature differential between the center and the edge of the foil is a measure of the incident flux. This temperature differential is measured by means of a silver-constantan thermocouple which generates an electrical voltage corresponding to the temperature differential developed. Another type of foil radiometer, developed particularly for use in field tests, is provided with a large copper block for a heat sink having no water cooling system. The relatively large thermal capacity of the copper block tends to maintain the temperature of the sink constant with respect to that of the foil when the incident irradiance is not at too high a level. However, it is believed that in actual field tests involving high intensity thermal radiation, e. g., nuclear weapons tests, the temperature of the copper block does not, in fact, remain constant. This provides a source of error which is difficult to estimate.

In the present study a theoretical analysis of the thermal behavior of an instrument of the water-cooled type is undertaken in which particular consideration is given to the steady state sensitivity and to the reaction time, which for present purposes is defined as the time for the foil to come to 90 percent of its full steady state temperature above that of the sink after the onset of irradiance. Any effects on the performance of the instrument due to the reaction of the thermocouple are entirely neglected. However, it is noted that these effects may not be negligible, and hence the actual performance of any instrument of the type discussed may differ somewhat from its performance herein predicted.

Consider a circular foil of thickness, d , and radius, R , which is in contact at its circumference with a large heat sink supposedly maintained at a constant temperature, T_0 . This assumption represents quite accurately the physical situation existing in radiometers of the

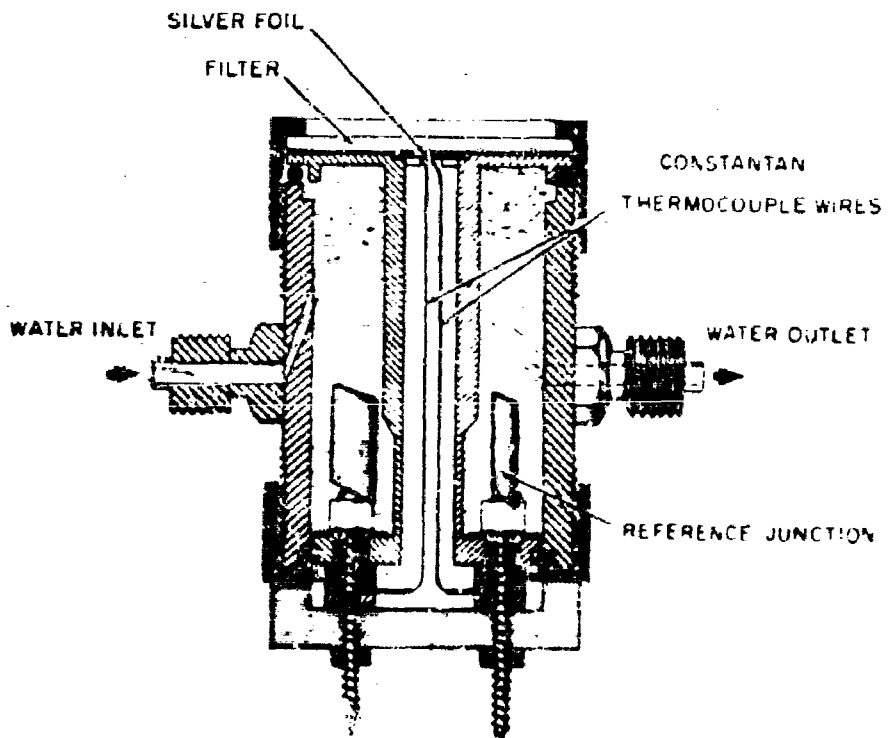


Fig. 1. Thin foil radiometer for measurement of high intensity of thermal radiation as a function of time.

water-cooled type. The foil is assumed to absorb thermal radiation of intensity, q , on one face, beginning at a particular instant of time. To obtain the differential equation for the temperature at any point in the foil, it is necessary to consider the heat balance for an elemental ring of the foil material of radius, r , and width, δr , as shown in Fig. 2. Neglecting any temperature gradient between faces of the foil, as well as heat losses from either face, the heat balance for the elemental ring can be expressed by the following equation:

$$\frac{\partial i}{\partial t} \pi r \cdot \delta r \cdot d \cdot \rho = \pi r \cdot \delta r \cdot q - \frac{\partial i}{\partial t} \pi r \cdot \delta r \cdot d \cdot \rho + \left\{ \frac{\partial i}{\partial r} \pi r \cdot \delta r + \frac{\partial i}{\partial r} \pi (r + \delta r) \delta r \right\} 2\pi r \cdot \delta r \cdot d. \quad (1)$$

Here i is foil temperature at radius, r , thickness, d , and time, t , measured from the instant at which irradiation begins, k is the thermal conductivity of the foil material, ρ is its density, and c the specific heat. Equation (1), ultimately simplifies to

$$\frac{1}{k} \frac{\partial}{\partial r} \left(r \frac{\partial i}{\partial r} \right) + \frac{\partial i}{\partial t} = \frac{q}{k d} + \frac{\rho}{k} \frac{\partial i}{\partial t}. \quad (2)$$

and the solution, i , must be subject to the boundary conditions

$$T = T_0, \text{ at } t = 0, \text{ all } r, \quad (3)$$

$$T = T_0, \text{ at } r = R, \text{ for } t > 0. \quad (4)$$

SENSITIVITY

To discuss the sensitivity of a circular foil radiometer, we need consider only the steady state solution to Eq (2), i.e., the solution for which $\partial T / \partial t = 0$ subject to the boundary condition (4). If we allow for variation of thermal conductivity according to the relation,

$$k = k_0 (1 + \alpha T), \quad (5)$$

then with $\partial T / \partial t = 0$, Eq (2) becomes

$$\frac{d}{dr} \left\{ (1 + \alpha T) \frac{dT}{dr} \right\} + \frac{1 + \alpha T}{r} \frac{dT}{dr} + \frac{q}{k_0 d} = 0, \quad (6)$$

or

$$\frac{1}{r} \frac{d}{dr} \left\{ r(1 + \alpha T) \frac{dT}{dr} \right\} + \frac{q}{k_0 d} = 0. \quad (7)$$

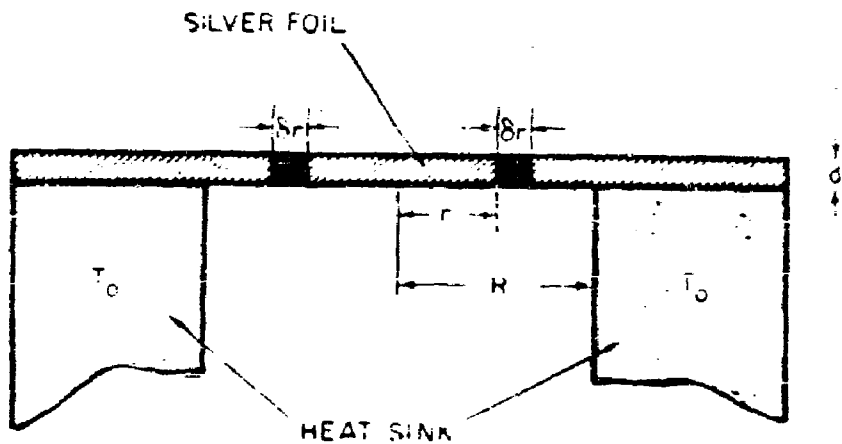


Fig. 2. Schematic diagram of thin foil mounted over a hole in a heat sink. For analysis of heat balance, an elemental ring of foil material of radius r and width δr is considered.

Upon two integrations, the general solution is found to be

$$T + \frac{\alpha T^2}{2} + \frac{qr^2}{4k_{od}} = C_1 \log_e r + C_2, \quad (8)$$

where C_1 and C_2 are arbitrary constants. Since T cannot become infinite at $r = 0$, we must specify that $C_1 = 0$.

The constant C_2 may now be evaluated by imposing the condition that $T = T_0$ at $r = R$, i.e., Eq (4). Thus,

$$C_2 = T_0 + \frac{\alpha T_0^2}{2} + \frac{qR^2}{4k_{od}}, \quad (9)$$

and the required steady state solution is

$$T = T_0 + \frac{\alpha}{2} (1 - T_0) (T - T_0) + \frac{q}{4k_{od}} (R^2 - r^2). \quad (10)$$

Defining Δ to be the difference between the steady-state temperature of the center of the foil and that of its circumference, we then have

$$\Delta = T - T_0, \text{ at } r = 0. \quad (11)$$

and from (10),

$$\Delta \left\{ 1 + \frac{\alpha}{2} (1 + T_0) \right\} = \frac{qR^2}{4k_{od}}. \quad (12)$$

This can also be written as

$$\Delta \left\{ 1 + \alpha T_0 + \frac{\alpha \Delta}{2} \right\} = \frac{qR^2}{4k_{od}}. \quad (13)$$

Also, if we put

$$T_m = \frac{T_0 + T}{2}, \quad (14)$$

then T_m is the arithmetic mean of T_0 and T , the temperature at the center, and Eq (12) can be written as

$$\Delta \left\{ 1 + \alpha T_m \right\} = \frac{qR^2}{4k_{od}}. \quad (15)$$

Equation (12), or the alternative forms (13) or (15), defines the sensitivity of a circular foil radiometer in terms of the temperature difference Δ produced by radiation of intensity q . Thus from (13)

$$\text{Sensitivity} = \frac{\Delta}{q} = \frac{R^2}{4k_{od}} \frac{1}{1 + \alpha T_0 + \frac{\alpha \Delta}{2}} \quad (16)$$

Clearly Δ is not in general a linear function of q unless $a = 0$, in which case there is no variation of thermal conductivity with temperature. Also it is evident that the sensitivity is dependent upon T_0 , the temperature of the sink, as well as upon q unless $a = 0$. This seems to have been overlooked in Gardon's Analysis¹.

To investigate more precisely the dependence of Δ on both q and T_0 , we must examine the appropriate solution to Eq (13). Since Eq (13) is a quadratic in Δ , there will be two roots of which only one is of physical interest in the present discussion. Writing Eq (13) in the form:

$$\frac{a\Delta^2}{2} + (1 + aT_0)\Delta - q' = 0, \quad (17)$$

where q' stands for $qR^2/4k_0d$, we find for the two roots of (17)

$$\Delta_1, \Delta_2 = \frac{-(1 + aT_0) \pm \sqrt{(1 + aT_0)^2 + 2aq'}}{a} \quad (18)$$

Physical considerations lead us to retain only that root, Δ_1 , in which the positive sign is to be taken before the radical, this being the only root which reduces to zero for $q = 0$.

Introducing the series expansion of the radical and ultimately retaining only terms of the first power in the small quantities aT_0 and aq' , we obtain the result:

$$\Delta_1 = q' (1 - aT_0 - \frac{1}{2} aq'). \quad (19)$$

from which the effects on Δ of variations in T_0 and q can be easily ascertained.

At this point it will be advantageous to discuss a specific numerical example. A common instrument in use by the Thermal Radiation Branch at NRDL is the RW-100 radiometer. This is a water-cooled, silver foil instrument designed to cover the range $0 \leq q \leq 100 \text{ Cal/cm}^2 \cdot \text{sec}$. For this instrument $R = 0.127 \text{ cm}$ and $d = 0.00254 \text{ cm}$. The values of k_0 and a as listed in the International Critical Tables are $k_0 = 1.001 \text{ Cal/cm} \cdot \text{sec} \cdot ^\circ\text{C}$, and $a = -1.7 \times 10^{-4}/^\circ\text{C}$, which are valid in the temperature range $0 - 100^\circ\text{C}$.

¹ Gardon, Rev. Sci. Instr. p 366, May 1953

For this case we then have:

$$-\alpha T_0 = 1.7 \times 10^{-4} \times T_0, \quad (20)$$

$$q' = \frac{R^2 q}{4kd} = \frac{(0.127)^2 q}{(4)(1.001)(1.00254)} = 1.59 q, \quad (21)$$

$$\text{and } -\frac{1}{2} \alpha q = (0.85)(1.59) \times 10^{-4} q = 1.35 \times 10^{-4} q. \quad (22)$$

Thus, according to Eqs (19) and (20) a change in the sink temperature, T_0 , from 0°C to 100°C would only produce a change in Δ_1 of 1.7 percent. However, assuming the value of α to remain about the same for higher temperatures, an increase of T_0 to 100°C could result in as much as 17 percent change in Δ_1 for the same value of q . This is an important fact to keep in mind when using foil radiometers at extremely high ambient temperatures.

Equation (19) indicates that Δ_1 is a quadratic equation in q , when first order terms in q^2 are retained. However, according to Eq (22), the departure from linearity for the RW-100 radiometer is only 1.35 percent when q has its maximum value of $100 \text{ Cal/cm}^2 \cdot \text{sec}$. For lower values of q the quadratic term is of even less importance.

Summarizing, we may say that the RW-100 radiometer operates linearly over its entire range of 0 to $100 \text{ Cal/cm}^2 \cdot \text{sec}$ with a maximum error of less than 2 percent when maximum irradiance is incident on the foil. With regard to the sink temperature, T_0 , it has been shown that variations in T_0 of as much as 100°C in the range of 0 to 100° will produce changes in Δ_1 of less than 2 percent.

The RW-10 and RW-50 are two other radiometers in use by the Thermal Radiation Branch at NRDL designed to cover the ranges $0 \leq q \leq 10$, and $0 \leq q \leq 50 \text{ Cal/cm}^2 \cdot \text{sec}$, respectively. Like the RW-100 they are both water-cooled, silver foil instruments, for which $R = 0.254 \text{ cm}$ and $d = 0.00127 \text{ cm}$ for the RW-10, and $R = 0.127 \text{ cm}$ and $d = 0.00127 \text{ cm}$ for the RW-50. As in the case of the RW-100 it can be shown that both the RW-10 and RW-50 radiometers operate linearly over their entire ranges of irradiance with a maximum error of less than 2 percent when the irradiance is a maximum. The effect of T_0 is, of course, the same as for the RW-100 since this effect is independent of R , d , or q , and depends only on α .

In Table I the linear sensitivity for the three above-mentioned instruments is listed together with the maximum percent error at full scale operation due to the effect of non-linearity.

It should be noted that the present discussion has omitted any discussion of the effects of non-linearity in the thermocouples used for measuring the temperature difference ΔT . Actually, the output of these instruments is a thermocouple voltage, which will reflect the non-linearity of the thermocouple as well as that of the radiometer itself. Also, the sink temperature, T_0 , will affect the operation of the thermocouple to a certain extent. It should therefore not be expected that the measured performance of these instruments will conform exactly to the figures given above and in Table 1.

Table 1

Factor	Sensitivities and Reaction Times of Three Radiometers		
	Instrument		
	RW-10	RW-50	RW-100
R (cm)	0.254	0.127	0.127
d (cm)	0.00127	0.00127	0.00254
k_0 (cgs)	1.001	1.001	1.001
$4/q = R^2/4k_0d \left(\frac{^\circ\text{C}}{\text{cm}^2 \cdot \text{sec}} \right)$	1.59	3.17	12.68
$\epsilon\%$	1.32	1.35	1.35
T (sec)	0.016	0.004	0.004

ϵ : maximum percent error at full scale due to non-linearity.

T : time to reach 90 percent of full value.

SPEED OF RESPONSE

To determine the speed of response of a circular foil radiometer by analytic methods we require the solution to Eq (2) for the time-dependent case. If it be assumed that k is independent of T , i.e., that $a = 0$, it is not hard to find the time-dependent solution to Eq (2) which satisfies (3) and (4). The assumption of constant k should actually make little difference in the determination of the response time.

With the assumption of k constant, Eq (2) reduces to

$$\frac{\partial^2 T}{\partial r^2} + \frac{1}{r} \frac{\partial T}{\partial r} + \frac{q}{kd} = \frac{1}{\kappa} \frac{\partial T}{\partial t} \quad (23)$$

where $\kappa = \frac{k}{\rho c}$.

which is to be solved subject to Eqs (3) and (4).

The solution can conveniently be carried out by means of the Laplace Transformation.

Before proceeding with this, however, it will facilitate matters to introduce the new variable θ , defined as

$$\theta = T - T_0. \quad (24)$$

Thus θ will satisfy Eq (23), as well as the boundary conditions

$$\theta = 0, \text{ at } r = R, t > 0, \quad (25)$$

and $\theta = 0, \text{ at } t = 0, \text{ for all } r. \quad (26)$

Applying the Laplace transformation to the variable θ , and defining

$$\bar{\theta} = \int_0^\infty \theta e^{-pt} dt, \quad (27)$$

the subsidiary equations for $\bar{\theta}$ are

$$\frac{\partial^2 \bar{\theta}}{\partial r^2} + \frac{1}{r} \frac{\partial \bar{\theta}}{\partial r} - s^2 \bar{\theta} = - \frac{q}{p \kappa d} \quad (28)$$

where $s^2 = \frac{p}{\kappa} \quad (29)$

with the condition that

$$\bar{\theta} = 0, \text{ at } r = R. \quad (30)$$

The general solution to Eq (28) is

$$\bar{\theta} = A I_0(sr) + B K_0(sr) + \frac{q}{p \kappa d} \quad (31)$$

where A and B are arbitrary constants, and $I_0(sr)$ and $K_0(sr)$ are modified Bessel functions. Since $K_0(sr) \rightarrow \infty$ as $r \rightarrow 0$, it is clear that we must put $B = 0$. The constant A can then be determined by imposing the condition expressed in Eq (30), i.e.,

$$A I_0(sR) + \frac{q}{p \kappa d} = 0. \quad (32)$$

Hence the desired solution is

$$\begin{aligned}\bar{\theta} &= \frac{q}{p^2 \rho c d} \left\{ 1 - \frac{I_0(sr)}{I_0(sR)} \right\} \\ &= \frac{q}{\rho c d} \left\{ \frac{1}{p^2} - \frac{1}{p^2} \frac{I_0(sr)}{I_0(sR)} \right\}\end{aligned}\quad (33)$$

To find θ , we now take the inverse Laplace Transform (L^{-1}) of $\bar{\theta}$. Thus

$$\theta = \frac{q}{\rho c d} L^{-1} \left\{ \frac{1}{p^2} \right\} - \frac{q}{\rho c d} L^{-1} \left\{ \frac{1}{p^2} \frac{I_0(sr)}{I_0(sR)} \right\} \quad (34)$$

From any table of Laplace transforms

$$L^{-1} \left\{ \frac{1}{p^2} \right\} = t \quad (35)$$

and by the complex inversion theorem

$$L^{-1} \left\{ \frac{1}{p^2} \frac{I_0(sr)}{I_0(sR)} \right\} = \frac{1}{2\pi i} \int_{\gamma-i\infty}^{\gamma+i\infty} \frac{e^{\lambda t}}{\lambda^2} \frac{I_0(\mu r)}{I_0(\mu R)} d\lambda \quad (36)$$

where

$$\mu = \sqrt{\frac{\lambda}{k}} \quad (37)$$

The integrand in Eq (36) is a single valued function of λ over the entire complex λ plane. Therefore, it can be evaluated by finding the residues at the poles. Examination shows that there is a double pole at $\lambda = 0$, and simple poles at the zeros of $I_0(\mu R)$.

To find the residue at the pole $\lambda = 0$, use the power series for the Bessel functions, i.e.,

$$I_\nu(\mu r) = \sum_{m=0}^{\infty} \frac{(1/2 \mu r)^{\nu+2m}}{m! \Gamma(\nu+m+1)}$$

and for $\nu = 0$ this reduces to

$$I_0(\mu r) = \sum_{m=0}^{\infty} \frac{(1/2 \mu r)^{2m}}{m! m!}$$

$$= 1 + (1/2 \mu r)^2 + \frac{(1/2 \mu r)^4}{4} + \frac{(1/2 \mu r)^6}{36} + \dots$$

- Since the zeros of $I_0(sR)$ are all on the negative real axis and $p^2 = k\lambda$, then $\frac{1}{p^2} \frac{I_0(sr)}{I_0(sR)}$ is analytic over any half plane to the right of but not including the origin.

Thus, near $\lambda = 0$, assuming $R > r$, the integrand looks like

$$\begin{aligned} & \frac{1}{\lambda^2} \frac{(1 + \lambda t + \frac{\lambda^2 t^2}{2!} + \frac{\lambda^3 t^3}{3!} + \dots) (1 + \frac{\mu^2 r^2}{4} + \frac{\mu^4 r^4}{64} + \dots)}{1 + \frac{\mu^2 R^2}{4} + \frac{\mu^4 R^4}{64} + \dots} \\ &= \frac{1}{\lambda^2} (1 + \lambda t + \frac{\lambda^2 t^2}{2!} + \frac{\lambda^3 t^3}{3!} + \dots) (1 - \frac{\mu^2 [R^2 - r^2]}{4} + O(\mu^4) + \dots) \\ &= \frac{1}{\lambda^2} - \frac{1}{\lambda} \left(\frac{R^2 - r^2}{4\kappa} - t \right) + \text{higher terms in } \lambda. \end{aligned}$$

Thus, the residue at $\lambda = 0$ is

$$t - \frac{R^2 - r^2}{4\kappa}.$$

To this must be added the contributions from the remaining poles at the zeros of $J_0(\mu R)$, i.e., at $\lambda = -\kappa \beta_n^2$ where β_n are the roots (all real) of

$$J_0(\beta R) = 0.$$

Since these are all simple poles, the residues will be given by

$$\begin{aligned} & \sum_{n=1}^{\infty} \frac{e^{\lambda t} I_0(\mu r)}{(d/d\lambda) [\lambda^2 I_0(\mu R)]} \bigg|_{\lambda = -\kappa \beta_n^2} \\ &= \frac{2}{R} \sum_{n=1}^{\infty} \frac{I_0(\beta_n r) i \kappa \beta_n}{\kappa^3 \beta_n^3 J_1(\beta_n R)} e^{-\kappa \beta_n^2 t}. \end{aligned}$$

Using the fact that

$$I_0(i\beta_n r) = J_0(\beta_n r),$$

$$I_1(i\beta_n r) = iJ_1(\beta_n r),$$

this becomes

$$\frac{2}{R\kappa} \sum_{n=1}^{\infty} \frac{J_0(\beta_n r)}{\beta_n^3 J_1(\beta_n R)} e^{-\kappa \beta_n^2 t}.$$

where J_0 and J_1 are ordinary Bessel functions.

We then have the result that

$$\begin{aligned} L^{-1} \left\{ \frac{1}{p^2} - \frac{1}{p^2} \frac{I_0(sr)}{I_0(sR)} \right\} \\ = t - \left\{ t - \frac{R^2 - r^2}{4\kappa} + \frac{2}{R\kappa} \sum_{n=1}^{\infty} \frac{J_0(\beta_n r)}{\beta_n^3 J_1(\beta_n R)} e^{-\kappa \beta_n^2 t} \right\} \\ = \frac{R^2 - r^2}{4\kappa} - \frac{2}{R\kappa} \sum_{n=1}^{\infty} \frac{J_0(\beta_n r)}{\beta_n^3 J_1(\beta_n R)} e^{-\kappa \beta_n^2 t} \end{aligned}$$

Thus, from Equation (34)

$$\theta = \frac{q}{kd} \left\{ \frac{R^2 - r^2}{4} - \frac{2}{R} \sum_{n=1}^{\infty} \frac{J_0(\beta_n r)}{\beta_n^3 J_1(\beta_n R)} e^{-\kappa \beta_n^2 t} \right\} \quad (38)$$

For very large t the exponential terms may be disregarded, and θ reduces to the steady state value

$$\theta_s(r) = \frac{q}{kd} \frac{R^2 - r^2}{4} \quad \text{as } t \longrightarrow \infty \quad (39)$$

This agrees with Eq (10) when $\alpha = 0$ and θ is put equal to $T - T_0$.

For large t , at $r = 0$, Equation (38) reduces to

$$\begin{aligned} \theta &= \frac{qR^2}{4kd} - \frac{2q}{kdR} \frac{e^{-\kappa \beta_1^2 t}}{\beta_1^3 J_1(\beta_1 R)} \\ &= \frac{qR^2}{4kd} - \left\{ \frac{qR^2}{4kd} \cdot \frac{8}{(\beta_1 R)^3 J_1(\beta_1 R)} \right\} e^{-\kappa \beta_1^2 t} \end{aligned}$$

or

$$\theta = \theta_s(0) - \theta_{ex(0)} \left\{ \frac{8}{(\beta_1 R)^3 J_1(\beta_1 R)} \right\} e^{-\kappa \beta_1^2 t} \quad (40)$$

$$\text{Thus, } \frac{\theta_s(0) - \theta}{\theta_{ex(0)}} = \frac{8}{(\beta_1 R)^3 J_1(\beta_1 R)} e^{-\kappa \beta_1^2 t} \quad (41)$$

* since $J(0) = 1$ and since all the terms in the summation may be neglected compared to the first for very large t .

Using tables of Bessel functions, we find

$$\beta_1 R = 2.405$$

$$J_1(\beta_1 R) = 0.519$$

and, therefore,

$$\frac{\theta_s - \theta}{\theta_s} = 1.11 e^{-(2.405)^2 k t / R^2} \quad (42)$$

To find the time T for the center of the circular foil radiometer to reach a temperature within 10 percent of its steady state temperature, put $(\theta_s - \theta)/\theta_s$ equal to 10^{-1} in Eq (42), i.e.,

$$10^{-1} = 1.11 e^{-(2.405)^2 k T / R^2}$$

and taking $\log_{10}$ of both sides,

$$-1 = \log_{10}(1.11) - \frac{(2.405)^2 k T}{R^2} \log_{10} e,$$

or

$$\left. \begin{aligned} \frac{k T}{R^2} &= \frac{\log_{10}(1.11) + 1}{(2.405)^2 \log_{10} e} \\ &= \frac{1.0453}{(5.784)(0.4343)} = 0.416 \end{aligned} \right\} \quad (43)$$

Thus,

$$T = \frac{(0.416) R^2}{k} \quad (44)$$

In Gardon's analysis, previously referred to, an approximate exponential solution for the time-dependent problem is obtained which yields a value of

$$t = \frac{(0.25) R^2}{k}$$

corresponding to a value of

$$\frac{\theta_s - \theta}{\theta_s} = \frac{1}{e} = 0.368$$

From this it follows that a value of

$$\tau = \frac{(0.575)R^2}{\kappa}$$

is required for θ to be within 10 percent of its steady state value, i.e., for

$$\frac{\theta_s - \theta}{\theta_s} = 0.10.$$

However, the more rigorous solution derived above indicates a somewhat faster approach to the steady state since it has been shown that when Eq (44) is satisfied, θ is very nearly within 10 percent of its steady state value.

Actually, the rigorous solution to the time-dependent problem, given in Eq (38) shows that, in general, θ does not have a simple exponential time dependence, but only approaches an exponential behavior for large values of the time. It can be shown that the large time approximation is valid, with negligible error, for computing the reaction time τ as we have done.

Using Eq (44), the values of the reaction time τ required to reach 90 percent of full value have been calculated for each of the three instruments already discussed. These values are included in Table I. The value of the diffusivity, κ , of silver was computed from the basic definition $\kappa = k/\rho c$, where $k = 1.001$ Cal/cm. sec. $^{\circ}\text{C}$, $\rho = 10.45$ gm/cc, and $c = 0.057$ Cal/gm $^{\circ}\text{C}$.

Approved by:

A. Guthrie

A. GUTHRIE
Head, Nucleonics Division

For the Scientific Director

DISTRIBUTION

NAVY

10 Chief, Bureau of Ships
 10 Chief, Bureau of Medicine and Surgery
 11 Chief, Bureau of Aeronautics (Code AA40)
 12 Chief, Bureau of Supplies and Accounts (Code SS)
 13-14 Chief, Bureau of Yards and Docks (Code YD40)
 15 Chief of Naval Operations (Op-30)
 16-17 Commander, New York Naval Shipyard (Material Lab.)
 18-20 Director, Naval Research Laboratory (Code 2021)
 21 CO, Naval Electronics Laboratory
 22 Chief of Naval Research (Code S11)
 23 Office of Naval Research (Code 4.2)
 24-30 Office of Naval Research, NYC, New York
 31 Office of Naval Research Branch Office, SF
 32 CO, Naval Unit, Army Chemical Center
 33 U.S. Naval Civil Engineering (Res. and Eval. Lab.)
 34 U.S. Naval School (CIC Officers)
 35 Commander, Naval Air Materiel Center, Philadelphia
 36 Commander, Naval Air Development Center, Johnsville
 37 Naval Medical Research Institute
 38 CO, Naval School Island, Treasure Island
 39 CO, Naval Damage Control Training Center, Philadelphia
 40 U.S. Naval Postgraduate School, Monterey
 41 CO, Nuclear Weapons Training Center, Atlantic
 42 CO, Nuclear Weapons Training Center, Pacific
 43-44 Commander, Naval Ordnance Laboratory, Silver Springs
 45 CIO, Naval Supply and Facility
 46 Office of Patent Counsel, ONR, DIO
 47 Commander Naval Air Force, Atlantic Fleet (Code 104)

ARMY

56-57 Chief of Research and Development (Atomic Div.)
 58-59 Office of Assistant Chief of Staff, G-2
 60 Chief of Engineers (ENGB, Rhein)
 61 Chief of Engineers (ENGB)
 62 Chief of Transportation (TC Technical Committee)
 63 Chief of Ordnance (ORDTN-RE)
 64 Ballistic Research Laboratories
 65 The Quartermaster General

66 CG, Chemical Corps Res. and Dev. Command
 67 Hq., Chemical Corps Materiel Command
 68 President, Chemical Corps Board
 69 CG, Chemical Corps Training Command (Library)
 70-71 CG, Chemical Warfare Laboratories
 72-74 CG, Aberdeen Proving Ground
 75 Office of Chief Signal Officer (SIGRD-8B)
 76 CG, Army Medical Research Laboratory
 77 Director, Walter Reed Army Medical Center
 78 CG, Continental Army Command (ATCSV-1)
 79-80 CG, Quartermaster Res. and Eng. Command
 81 Director, Operations Research Office (Librarian)
 82 CG, Dugway Proving Ground
 83 U.S. Army Electronic Proving Ground
 84-86 The Surgeon General (MEDNE)
 87 Office of Quartermaster General (RAD Div.)
 88 Office of Ordnance Research
 89 CG, U.S. Army Signal R&D Laboratory
 90 CG, Engineer Res. and Dev. Laboratory (Library)
 91 CG, Engineer Res. and Dev. Laboratory (Tech. Support Br.)
 92 CG, Transportation Res. and Dev. Command
 93 MLC, ONARS, Fort Monroe
 94 Director, Office of Special Weapons Development
 95 Director, Waterways Experiment Station
 96 CG, Ordnance-Tank Automotive Command
 97 CG, Ordnance Materials Research Office, Watertown
 98 CG, Watertown Arsenal
 99 CG, Frankford Arsenal
 100 Diamond Ordnance Fuse Laboratories
 101 Jet Propulsion Laboratory
 102 Army Ballistic Missile Agency

AIR FORCE

103 Assistant Chief of Staff, Intelligence
 104 Commander, Air Materiel Command (AMCMT)
 105-109 Commander, Wright Air Development Center (WACT)
 110 Commander, Air Res. and Dev. Command (ADTC)
 111 Directorate of Installations (AFDIE-ES)
 112-113 Director, USAF Project RAND
 114-115 Commandant, School of Aviation Medicine, Randolph AFB
 116 CG, Strategic Air Command (Operations Analysis Office)
 117-121 Commander, Special Weapons Center, Kirtland AFB
 122 Director, Air University Library, Maxwell AFB
 123-124 Commander, Technical Training Wing, 3415th TFG
 125 CG, Cambridge Research Center (CRZT)
 126-127 CG, Cambridge Research Center (CRGST-2)

OTHER DOD ACTIVITIES

128-137 Chief, Armed Forces Special Weapons Project
 138-142 Commander, FC/AFSWP, Sandia Base (FCWT)

143 Commander, FC/AFSWP, Sandia Base (FCTG Library)
 144 Commander, FC/AFSWP, Sandia Base (FCDV)
 145 OIC, Livermore Branch, FC/AFSWP
 146 Director, Weapons Systems Evaluation Group
 147 Assistant Secretary of Defense (Res. and Dev.)
 148-162 Armed Services Technical Information Agency

ABC ACTIVITIES AND OTHERS

163-164 Office of Civilian and Defense Mobilization
 165 AEC Scientific Representative, Japan
 166 Aerojet-General Corporation
 167 Aerojet-General, San Ramon
 168 Alcoa Products, Inc.
 169-171 Air Project Office, Convair, Fort Worth
 171 Aronne Cancer Research Hospital
 172-181 Aronne National Laboratory
 182 Atomic Bomb Casualty Commission
 183-189 Atomic Energy Commission, Washington
 190-198 Atomics International
 189-192 Babcock and Wilcox Company
 193-194 Battelle Memorial Institute
 195-198 Boeing Plant
 199 Boeing Airplane Company, Seattle
 200-203 Brookhaven National Laboratory
 204 Brush Beryllium Company
 205 Carnegie Institute of Technology
 206 Chicago Patent Group
 207 Columbia University (Havens)
 208 Columbia University (SU-187)
 209-210 Combustion Engineering, Inc.
 211 Convair-General Dynamics Corporation, San Diego
 212 Curtiss-Wright Corporation
 213 Defense Research Member
 214-216 duPont Company, Alcoa
 217 duPont Company, Wilmington
 218 General Atomic Division
 219-220 General Electric Company (AED)
 221-226 General Electric Company, Highland
 227 General Electric Company, San Jose
 228 General Nuclear Engineering Corporation
 229 Gibbs and Cox, Inc.
 230-231 Goodyear Atomic Corporation
 232 Grand Junction Operations Office
 233-234 Iowa State College
 235-237 Knolls Atomic Power Laboratory
 238-239 Lockheed Aircraft Corporation, Marietta
 240-241 Los Alamos Scientific Laboratory
 242 Lovelace Foundation
 243 Maritime Administration
 244 Martin Company
 245-246 Midwestern Universities Research Association

247	Mound Laboratory
248	National Aeronautics and Space Administration
249-250	National Bureau of Standards (Taylor)
251	National Bureau of Standards (Library)
252	National Lead Company of Ohio
253	New Brunswick Area Office
254-255	New York Operations Office
256	New York University
257-258	Nuclear Development Corporation of America
259	Nuclear Metals, Inc.
260	Oak Ridge Institute of Nuclear Studies
261	Patent Branch, Washington
262-263	Pennsylvania State University (Blanchard)
264-267	Phillips Petroleum Company
268	Power Reactor Development Company
269-271	Pratt and Whitney Aircraft Division
272	Princeton University (White)
273-274	Public Health Service, Washington
275	Public Health Service, Savannah
276	Rensselaer Polytechnic Institute
277	Sandia Corporation
278	Stevens Institute of Technology
279	Sylvania Electric Products, Inc.
280	Technical Research Group
281	Tennessee Valley Authority
282	Texas Nuclear Corporation
283-284	Union Carbide Nuclear Company (RGDP)
285-289	Union Carbide Nuclear Company (OHL)
290	Union Carbide Nuclear Company (Paducah Plant)
291	U.S. Geological Survey, Denver
292	U.S. Geological Survey, Menlo Park
293	U.S. Geological Survey, Naval Gun Factory
294	U.S. Geological Survey, Washington
295	U.S. Patent Office
296	UCLA Medical Research Laboratory
297	University of California, San Francisco
298-299	University of California Lawrence Radiation Lab., Berkeley
300-303	University of California Lawrence Radiation Lab., Livermore
304	University of Puerto Rico
305	University of Rochester (Technical Report Unit)
306-307	University of Rochester (Marshak)
308-309	University of Washington (Geballe)
310	University of Washington (Rhode)
311	Vitro Engineering Division
312	Westinghouse Electric Corporation (Schafer)
313	Yale University (Breit)
314	Yale University (Schultz)
315	Yankee Atomic Electric Company
316-340	Technical Information Service, Oak Ridge

USNRDL

341-375 USNRDL, Technical Information Division

UNCLASSIFIED

AD

216542

FOR
MICRO-CARD
CONTROL ONLY

1

OF

1

Reproduced by

Armed Services Technical Information Agency

ARLINGTON HALL STATION; ARLINGTON 12 VIRGINIA

UNCLASSIFIED



Defense Special Weapons Agency
6801 Telegraph Road
Alexandria, Virginia 22310-3398

TRC

28 April 1997

MEMORANDUM FOR DEFENSE TECHNICAL INFORMATION CENTER
ATTENTION: OMI/Mr. William Bush (Security)

SUBJECT: Change of Distribution Statement on AD-216542

The Defense Special Weapons Agency Security Office (OPSSI)
has approved the following report for **public release**:

AD-216542 AFSWP-1119 (USNRDL-TR-311)
Theoretical Analysis of Radiometer Performance,
13 January 1959 by H. G. Ferris.

Distribution statement "A" now applies.

Ardith Jarrett
ARDITH JARRETT
Chief, Technical Resource Center

copy furn: FC/DASIAC
DASIAC