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SMALL-TIME BEHAVIOR
OF UNSTEADY CAVITY FLOWS

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D. P. Wang and T. Yao-tsu Wu

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Hydrodynamics Laboratory
Kármán Laboratory of Fluid Mechanics and Jet Propulsion
California Institute of Technology
Pasadena, California

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SMALL-TIME BEHAVIOR OF UNSTEADY CAVITY FLOWS

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✓ A perturbation theory is applied to investigate the small-time behavior of unsteady cavity flows in which the time-dependent part of the flow may be taken as a small-time expansion superimposed on an established steady cavity flow of an ideal fluid. One purpose of this paper is to study the effect of the initial cavity size on the resulting flow due to a given disturbance. Various existing steady cavity-flow models have been employed for this purpose to evaluate the initial reaction of a cavitated body in an unsteady motion. Furthermore, a physical model is proposed here to give a proper representation of the mechanism by which the cavity volume may be changed with time; the initial hydrodynamic force resulting from such change is calculated based on this model.

△

1. Introduction

It has been noted that the free surface flow problems become very difficult to treat exactly when the flow is time-dependent. Some of the principal difficulties involved in this class of problems can be envisaged as follows. First of all, the problem is generally nonlinear because the boundary conditions on the unsteady free surface are nonlinear and because the location of the free boundary is not known a priori. Secondly, unlike the case of steady flows with free boundary, the surface of constant pressure is no longer a surface of constant speed. Consequently, the powerful hodograph method widely used in the steady free boundary problems loses much of its potential and usefulness for possible

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applications in this case. Probably for this reason, only a relatively small number of unsteady free surface problems (of rather special cases) have been treated. Some comprehensive survey and review of the recent literatures are available elsewhere (see Gilbarg[1], Birkoff and Zarantonello[2] and [3]), and hence will not be recited here.

It is worth noting that there is an essential difference between the unsteady flows with and without a free surface. In the determination of the velocity field of an unsteady flow without a free surface (which includes the vortex sheet shed from an oscillating wing), the time appears only as a parameter. The kinematics of the unsteady motion will therefore not differ basically from its corresponding steady flow. On the other hand, for an unsteady flow with a free surface, the time variation affects the flow explicitly through the boundary conditions on the free surface. Therefore, the flow will depend on all its previous time history, which further complicates the detailed analysis and calculation. However, when an established steady cavitating flow (regarded as the basic flow) is given a sudden acceleration, the problem of the small-time behavior of the flow is relatively simple because there will be only a short history of any time-varying disturbance to be considered in this period. It is therefore clear that some difficult problems in unsteady cavity flows can be best handled in the small-time limit since, for one reason, the analysis is relatively straightforward. We hope that such an investigation may cast light on some of the basic features of the unsteady cavity flow, especially the behavior of the large frequency limit in the case of oscillating cavity flows.

The impact problem of the Kirchhoff-Helmholtz flow with a flat plate normal to the stream has been treated by Curevich[4]. Attempt is made here to generalize the consideration to include the effects of (1) the cavitation number (the cavity size being finite), (2) the angle of attack, and (3) the time variation of the net cavity volume.

The existing mathematical theories for steady cavity flows are seen to be based on some proposed physical models. It is well known that in a positive range of the cavitation number up to moderate values of order unity the agreement between the existing models for steady flows may be

considered very close[5]. No comparison, however, has ever been made (to the authors' knowledge) between these existing models for the case of unsteady cavity flows. In this paper, the re-entrant jet model, Riabouchinsky's model, and a modified wake model are applied to investigate the small-time behavior of the unsteady cavity flow, including the effects of the cavitation number and the incidence angle of the flow.

The problem related to the possibility for the volume of a finite or infinite cavity to vary with time deserves a particular note. For cavity flows of an incompressible liquid surrounding a vapor cavity, it is obvious that, when the cavity volume changes, the conservation of mass and conservation of volume of the entire flow becomes incompatible because of the difference in the densities of these two phases. Consequently, any variation of the cavity volume must come from a source distribution in the flow with its net strength depending on the time-rate of change of the cavity volume. It has been proposed by one of the authors[6] that as a physical model, the change in the cavity volume can be affected by a source or sink located at the point of infinity. A direct consequence of this source with a time-varying strength in a two-dimensional flow of infinite extent is that it generates a pressure field which is logarithmically singular at infinite distances. To create such a flow it requires an infinite amount of energy. Probably for this reason the treatment of the unsteady partial cavity flow by Timman[7], the work of Geurst[8], of Parkin[9], and of Wu[6], have been carried out under the assumption of fixed cavity volume and no source at infinity. In reality, however, the flow is usually finite in extent and never two-dimensional in the large; the pressure singularity at infinity, being outside the flow region, may therefore be regarded as a simplifying idealization, much the same as the representation of a solid body by a distribution of singularities. This physical model for permitting the cavity volume to change has been applied to various cases together with other cavity flow models. It is expected that this variable-cavity-volume model will give a good representation of the flow field near the body-cavity system. The validity of this model of course will have to be verified, directly or indirectly, by future experimental observations.

2. General Formulation

To fix idea, we suppose that for the time $t < 0$, a steady plane flow of an incompressible, inviscid fluid past a solid body with a cavity formation has been established, the solution of which is assumed to be given, or can be determined with the aid of some cavity flow models. Suppose now the solid body to which the cavity is attached is given for $t > 0$ a sudden accelerated motion; the problem is to evaluate the small-time behavior of the resultant unsteady cavity flow.

In general, the motion of the rigid boundary may consist of a translation and a rotation. Let (x_0, y_0) be a point on the rigid surface $S_0(x_0, y_0) = 0$ in the basic steady flow, and let it be displaced in time t to the position (x, y) with translational velocity $(V_1(t), V_2(t))$ and angular velocity $\omega(t)$. In terms of the complex variable $z = x + iy$ and $V(t) = V_1 + i V_2$, the motion of z may be written

$$dz/dt = V(t) + i \omega(t) z . \quad (1)$$

We shall assume that, for small positive t , V and ω may be expanded in power series of t , starting with the linear term so that

$$dz/dt = c_1 t + c_2 t^2 + i (\omega_1 t + \omega_2 t^2) z + O(t^3) \quad (2)$$

where $c_n = a_n + i b_n$, a_n , b_n and ω_n being real constants. It then follows that for small t ,

$$z = z_0 + \frac{1}{2} (c_1 + i \omega_1 z_0) t^2 + \frac{1}{3} (c_2 + i \omega_2 z_0) t^3 + O(t^4) . \quad (3)$$

The displaced surface will be denoted by $S(x, y, t) = 0$. In fact, we have

$$S_0(x_0(x, y, t), y_0(x, y, t)) = S(x, y, t) , \quad (4)$$

regarding (1) as to provide the canonical transformation $x_0 = x_0(x, y, t)$, $y_0 = y_0(x, y, t)$.

From the nature of the body motion it also follows that the complex velocity potential of the flow

$$f(z, t) = \varphi(x, y, t) + i\psi(x, y, t) , \quad (5)$$

will assume for small t the expansion

$$f(z, t) = f_0(z) + t f_1(z) + \frac{1}{2} t^2 f_2(z) + \dots , \quad (6)$$

where $f_n(z) = \varphi_n(x, y) + i\psi_n(x, y)$, $n = 0, 1, 2, \dots$, and $f_0(z)$ is the complex velocity potential of the basic flow. The function $\varphi_1(x, y)$ may be called the initial acceleration potential. While $\psi_0 = \text{const.}$ gives the streamlines of the basic flow, the harmonic functions ψ_n for $n \geq 1$, being the complex conjugate of φ_n , are introduced solely to make $f_n(z)$ analytic functions of z . The velocity components are, as usual

$$u(x, y, t) = \partial\varphi/\partial x , \quad v(x, y, t) = \partial\varphi/\partial y . \quad (7a)$$

We shall introduce the complex velocity $w = u - iv$, and $w_n = df_n/dz$, then from (6)

$$w(z, t) = w_0(z) + t w_1(z) + \frac{1}{2} t^2 w_2(z) + \dots . \quad (7b)$$

Similarly the pressure $p(x, y, t)$ may be assumed to possess the expansion

$$p(x, y, t) = p_0(x, y) + p_1(x, y) + t p_2(x, y) + t^2 p_3(x, y) + \dots , \quad (8)$$

where p_0 denotes the pressure field of the basic flow, p_1 the impulsive pressure due to the sudden acceleration. Then from the Bernoulli equation,

$$p/\rho + \partial\varphi/\partial t + \frac{1}{2} (\nabla\varphi)^2 = p_\infty/\rho + \frac{1}{2} U^2 \quad (9)$$

where p_∞ and U are the pressure and speed of the basic flow at infinity, we obtain, by equating the coefficients of same powers of t , the following relations:

$$\begin{aligned} p_0/\rho + \frac{1}{2} (\nabla \varphi_0)^2 &= p_\infty/\rho + \frac{1}{2} U^2 \\ p_1/\rho &= -\varphi_1, \\ p_2/\rho &= -\varphi_2 - (\nabla \varphi_0) \cdot (\nabla \varphi_1), \text{ and so on.} \end{aligned} \quad (10)$$

The boundary conditions of the problem are as follows:

(i) At the solid surface, the normal component of the flow velocity relative to the moving boundary must vanish. An alternative way of stating this condition is that the fluid particles originally on $S(x, y, t) = 0$, at small time interval apart, will remain on it. That is

$$\frac{DS}{Dt} = \frac{\partial S}{\partial t} + \frac{\partial \varphi}{\partial x} \frac{\partial S}{\partial x} + \frac{\partial \varphi}{\partial y} \frac{\partial S}{\partial y} = 0 \quad \text{on } S(x, y, t) = 0,$$

which becomes, upon using (4),

$$\frac{\partial S_0}{\partial x_0} \left[\frac{\partial \varphi}{\partial x} \frac{\partial x_0}{\partial x} + \frac{\partial \varphi}{\partial y} \frac{\partial x_0}{\partial y} + \frac{\partial x_0}{\partial t} \right] + \frac{\partial S_0}{\partial y_0} \left[\frac{\partial \varphi}{\partial x} \frac{\partial y_0}{\partial x} + \frac{\partial \varphi}{\partial y} \frac{\partial y_0}{\partial y} + \frac{\partial y_0}{\partial t} \right] = 0 \quad (11)$$

on $S(x, y, t) = 0$. Here the functions $x_0(x, y, t)$ and $y_0(x, y, t)$ can be written down immediately from (3) by interchanging z and z_0 , and by changing the signs of a_n , b_n and ω_n , as can readily be seen from (3). Equation (11) is in a form convenient for manipulation since $\partial S_0/\partial x_0$ and $\partial S_0/\partial y_0$ already correspond to the components of the normal to the initial surface $S_0(x_0, y_0) = 0$. Substituting (3) and (6) in (11), expanding the various quantities about (x_0, y_0) and $t = 0$, and equating the coefficients of different powers of t , we obtain the conditions that on $S_0(x_0, y_0) = 0$,

$$\partial \varphi_0 / \partial n_0 = 0, \quad (12a)$$

$$\partial \varphi_1 / \partial n_o = n_{o_1} (a_1 - \omega_1 y_o) + n_{o_2} (b_1 + \omega_1 x_o) , \quad (12b)$$

$$\begin{aligned} \frac{\partial \varphi_2}{\partial n_o} = & 2n_{o_1} (a_2 - \omega_2 y_o) + 2n_{o_2} (b_2 + \omega_2 x_o) - \omega_1 \left[n_{o_1} \frac{\partial \varphi_o}{\partial y_o} - n_{o_2} \frac{\partial \varphi_o}{\partial x_o} \right] \\ & - (a_1 - \omega_1 y_o) \left[n_{o_1} \frac{\partial^2 \varphi_o}{\partial x_o^2} + n_{o_2} \frac{\partial^2 \varphi_o}{\partial x_o \partial y_o} \right] - (b_1 + \omega_1 x_o) \left[n_{o_1} \frac{\partial^2 \varphi_o}{\partial x_o \partial y_o} + n_{o_2} \frac{\partial^2 \varphi_o}{\partial y_o^2} \right] , \end{aligned} \quad (12c)$$

where $\vec{n}_o = (n_{o_1}, n_{o_2})$ is the unit outward normal to the surface $S_o(x_o, y_o) = 0$.

(ii) There are two boundary conditions at the free surface of the cavity. Suppose the free surface may be expressed as

$$F(x, y, t) = y - h(x, t) = 0 ,$$

then the kinematic condition that the fluid particles on the free surface will remain on it requires

$$v = \partial h / \partial t + u \partial h / \partial x \quad \text{on } y = h(x, t) . \quad (13)$$

We assume that for small t , $h(x, t)$ may be expanded as

$$h(x, t) = h_o(x) + t h_1(x) + \frac{1}{2} t^2 h_2(x) + O(t^3) , \quad (14)$$

where $y = h_o(x)$ denotes the cavity boundary of the basic flow on which

$$dh_o/dx = v_o(x, h_o)/u_o(x, h_o) . \quad (15)$$

Substituting (7) and (14) in (13), and expanding u_n and v_n on $y=h(x, t)$ about $y = h_o(x)$ and $t = 0$, we obtain

$$\begin{aligned} h_1 &= v_o(x, h_o) - u_o(x, h_o) dh_o/dx = 0 , \\ h_2 &= v_1(x, h_o) - u_1(x, h_o) dh_o/dx . \end{aligned} \quad (16)$$

Since $h_1 \approx 0$, the free surface will not be displaced in the time of order t , as should be expected.

In the basic flow, the cavity boundary is a surface of constant pressure, and hence also one of constant velocity, say

$$p_0 = p_c, \quad |\nabla \phi_0| = q_c, \quad \text{on } y = h_0(x), \quad (17)$$

so that by the Bernoulli equation (10),

$$q_c = U(1 + \sigma)^{\frac{1}{2}} \quad (18a)$$

where σ is the cavitation number of the basic flow, defined as

$$\sigma = (p_\infty - p_c) / \frac{1}{2} \rho U^2. \quad (18b)$$

We shall assume that the pressure in the cavity of the unsteady flow will be maintained at the same constant value p_c , that is,

$$p(x, y, t) = p_c \quad \text{on } y = h(x, t). \quad (19)$$

By expanding the left side of this dynamic condition about $y = h_0(x)$ and $t = 0$, using (8), (14) and (16), the following conditions result

$$p_1 = p_2 = 0 \quad \text{on } y = h_0(x). \quad (20)$$

Hence from (10),

$$\phi_1 = 0 \quad \text{on } y = h_0(x), \quad (21a)$$

$$\phi_2 = -(\nabla \phi_0) \cdot (\nabla \phi_1) \quad \text{on } y = h_0(x).$$

Upon differentiating (21a) along $y = h_0(x)$ and using (15), it is readily seen that $(\nabla \phi_0) \cdot (\nabla \phi_1) = 0$, and hence

$$\phi_2 = 0 \quad \text{on } y = h_0(x). \quad (21b)$$

From (21a) and (21b) it therefore follows that

$$(\nabla \phi_0) \cdot (\nabla \phi_1) = (\nabla \phi_0) \cdot (\nabla \phi_2) = 0 \quad \text{on } y = h_0(x) \quad (22)$$

which asserts that the perturbation velocity is normal to the original cavity boundary up to the time of order t^2 .

(iii) At the point of infinity we require the perturbation velocity to vanish,

$$|\nabla \varphi_n| \rightarrow 0 \quad \text{as} \quad |z| \rightarrow \infty, \quad \text{for} \quad n = 1, 2, \dots \quad (23)$$

If, in addition to the sudden acceleration of the solid body and the assumption of the constancy of the cavity pressure, a certain amount of fluid is removed at infinity with a source strength $Q_1 t + \frac{1}{2} Q_2 t^2 + \dots$ so that the cavity volume may be changed arbitrarily, then, aside from condition (23), we must impose additional conditions at infinity as

$$\text{Im} \int_{\Gamma} w_n(z) dz = Q_n, \quad n = 1, 2, \dots, \quad (24)$$

where Γ is a contour around the point of infinity. However, we impose no boundedness condition on p at infinity since, if such Q_n can be arbitrarily chosen, p_n ($n \geq 1$) will be logarithmically singular at infinity. It will be seen later that, when the cavity is taken to be infinitely long (Helmholtz flow, corresponding to the cavitation number $\sigma = 0$), the solution exists only when $Q_n = 0$. Consequently the effect of change in cavity volume can be sought only in the case of finite cavities. However, the limit of the hydrodynamic forces in such cases as the cavity becomes infinitely long, with Q_n held fixed, is seen to exist.

On the other hand, from Kelvin's theorem on the conservation of circulation, the circulation around the point of infinity cannot be changed in the unsteady motion for $t < \infty$. By combining this condition with (24), we may write

$$\int_{\Gamma} w_n(z) dz = i Q_n, \quad n = 1, 2, 3, \dots \quad (25)$$

This completes our formulation of the small-time perturbation theory.

3. Solution of the Perturbed Flow.

For the moment we assume that the solution of the basic steady flow is given. We note that in the basic flow the entire boundary of the body-cavity system belongs to a streamline, the form of which is known. Therefore, by the general theory of conformal mapping, it is always possible to find an analytic function

$$\zeta(z) = \xi(x, y) + i\eta(x, y) \quad (26)$$

such that the entire flow region in the z -plane is mapped by (26) into the upper half ζ -plane with the entire boundary of the body-cavity system mapped onto the entire ξ -axis. For simplicity, we shall make the part $|\xi| < 1$ of the real ζ -axis correspond to the wetted solid surface and the part $|\xi| > 1$ to the cavity boundary.

After the transformation to the ζ -plane the boundary value problem formulated in the last section can be stated as a Hilbert problem, as will be shown below, the solution of which is readily attained. It is noted from the last section that the problems of different orders in the perturbation are expressed in a similar form. That is, the normal velocities $\partial\varphi_n/\partial n_0$ are given at the initial solid surface and the potentials φ_n are prescribed at the unperturbed cavity boundary. It is therefore sufficient to treat one as the typical case. To save writing we shall denote $f_n = \varphi_n + i\psi_n$ by $F = \Phi + i\Psi$.

It is convenient to introduce the analytic function

$$G(\zeta) = dF/d\zeta = (dF/dz)(dz/d\zeta) \quad (27)$$

which is defined for $\text{Im } \zeta > 0$. Then on the solid surface, $\eta = 0+$, $|\xi| < 1$,

$$\text{Im } G = -\frac{\partial\Phi}{\partial\eta} = -\frac{\partial\Phi}{\partial n_0} \left| \frac{dz}{d\zeta} \right| = g_1(\xi) \quad \text{say,} \quad (28)$$

which is known, by (12) and (26). Furthermore, on the cavity boundary, $\eta = 0+$, $|\xi| > 1$,

$$\operatorname{Re} G = \partial \Phi / \partial \xi = g_2(\xi) \quad \text{say,} \quad (29)$$

which is also known, by (21). In particular, $g_2 = 0$ for φ_1 and φ_2 . Finally, (25) becomes

$$\int_{\Gamma} \frac{dF}{dz} dz = \int_{\Gamma'} G(\zeta) d\zeta = iQ \quad (30)$$

where Γ' is the image of Γ in the ζ -plane, the subscript n of Q is dropped to show a typical case.

It is possible to transform the above boundary value problems of a mixed type into a Hilbert problem by extending the unknown function $G(\zeta)$ to a sectionally analytic function, defined on the whole ζ -plane (excluding the real ζ -axis if necessary). Since φ_3 and higher terms will not be treated explicitly here, we shall demonstrate the method by taking g_2 of (29) to be zero; the general case of $g_2 \neq 0$ can be carried out similarly. First, the function $G(\zeta)$ may be continued into the lower half ζ -plane by

$$G(\bar{\zeta}) = -\overline{G(\zeta)} . \quad (31)$$

For the case $g_2 = 0$, $G(\bar{\zeta})$ is the analytical continuation of $G(\zeta)$ through the interval $|\xi| > 1$. In the following, $G^{\pm}(\xi)$ will be used to denote the limiting value of $G(\zeta)$ as $\eta \rightarrow \pm 0$. From (28), (29) and (31) it then follows that

$$\begin{aligned} G^+ + G^- &= 2i \operatorname{Im} G^+ = 2i g_1(\xi) \quad \text{for } |\xi| < 1 , \\ G^+ - G^- &= 2 \operatorname{Re} G^+ = 0 \quad \text{for } |\xi| > 1 . \end{aligned} \quad (32)$$

The above Hilbert problem is well-known [10], its general solution can be written

$$G(\zeta) = \frac{i}{\pi} (\zeta^2 - 1)^{-\frac{1}{2}} \int_{-1}^1 \frac{(1 - \xi^2)^{\frac{1}{2}} g_1(\xi)}{\xi - \zeta} d\xi + \frac{iP(\zeta)}{(\zeta^2 - 1)^{\frac{1}{2}}} \quad \text{for } \eta > 0 \quad (33)$$

where the function $(\zeta^2 - 1)^{\frac{1}{2}}$ is defined on the entire ζ -plane with the branch so chosen that $(\zeta^2 - 1)^{\frac{1}{2}} \rightarrow \zeta$ as $|\zeta| \rightarrow \infty$, and $P(\zeta)$ is an arbitrary Laurent's series with real coefficients. The last term in (33) is the general solution to the homogeneous problem (with $g_1 = 0$ also). The real coefficients of $P(\zeta)$, and hence $G(\zeta)$, can be determined uniquely, when (30) is satisfied and the condition that the pressure is integrable over the rigid boundary is observed.

In the following the above perturbation theory will be carried out for several basic steady cavity flows.

4. Inclined Lamina in Kirchhoff Flow

As a simple example we consider the basic flow to be that past a flat plate held at an angle α , with a cavity formation of infinite length. The solution of this problem is known [11], which we simply cite below for the subsequent use. The coordinate system in the z -plane and its conformal mapping planes are shown in figure 1. For simplicity the free stream velocity U and the plate length l , are normalized to unity. The solution $w_0 = w_0(z_0)$ can be written parametrically as

$$w_0 = (\sqrt{1 - \zeta^2} - 1)/\zeta, \quad (34a)$$

$$\begin{aligned} z_0 &= -K \int_1^{\zeta} (\sqrt{1 - \zeta^2} + 1) (\zeta + \sec \alpha)^{-3} d\zeta \\ &= \frac{K}{2} \left\{ \frac{1 + \sqrt{1 - \zeta^2}}{(\zeta + \sec \alpha)^2} - \frac{\cot \alpha}{\sin \alpha} \frac{\sqrt{1 - \zeta^2}}{(\zeta + \sec \alpha)} + 2 \cot^3 \alpha \tan^{-1} \left(\frac{\sqrt{1 - \zeta^2}}{1 + \zeta} \tan \frac{\alpha}{2} \right) - (1 + \sec \alpha)^{-2} \right\} \end{aligned} \quad (34b)$$

where

$$K = 2 \tan^3 \alpha \sin \alpha / (4 + \pi \sin \alpha). \quad (34c)$$

The entire flow region is mapped by (34) into the upper half ζ -plane with the corresponding boundary as prescribed in the last section

(i.e. on $\eta = 0$, $|\xi| < 1$ corresponds to the plate and $|\xi| > 1$ to the

cavity surface). The coefficient of the normal force N_o of the basic flow is

$$C_{N_o} = N_o / \left(\frac{1}{2} \rho U^2 l \right) = 2\pi \sin \alpha / (4 + \pi \sin \alpha) . \quad (35)$$

(a) The first order solution

The unit normal to the plate is now $\vec{n}_o = (0, -1)$. We suppose the rotational motion is referred to the leading edge of the plate. Then the boundary conditions (12b), (21a) and (23) become

$$\partial \varphi_1 / \partial y_o = (b_1 + \omega_1 x_o) \quad \text{on } \eta = 0, \quad |\xi| < 1, \quad (36a)$$

$$\varphi_1 = 0 \quad \text{and hence} \quad \partial \varphi_1 / \partial \xi = 0 \quad \text{on } \eta = 0, \quad |\xi| > 1. \quad (36b)$$

$$|\nabla \varphi_1| \rightarrow 0 \quad \text{as} \quad \xi \rightarrow -\sec \alpha. \quad (36c)$$

It is noted that a_1 , the first term of the x-component of acceleration, drops out from (36a), implying that the acceleration of the plate parallel to itself has no effect on the flow up to time of order t . Furthermore, (25) cannot be satisfied unless all the $Q_n = 0$. This can be seen as follows. If $Q_n \neq 0$, then (25) implies that $w_n = Q_n / 2\pi z + o(|z|^{-1})$, and hence $\varphi_n \sim (Q_n / 2\pi) \log |z|$, as $|z| \rightarrow \infty$. It follows that p_n will be logarithmically singular at $z = \infty$, which contradicts the conditions of p_n , such as (20), on the undisturbed free boundary $y = h_o(x)$ which extends to infinity. This indicates that the Kirchhoff-Helmholtz model with an infinite cavity is not a realistic model for the consideration of change of cavity volume. The problem when $Q_n \neq 0$ will be considered later when other finite-cavity models are adopted.

By making use of (36a), (36b) and (34b), (28) and (29) become

$$\text{Im } G = \frac{\partial \varphi_1}{\partial y_o} \left| \frac{dz}{d\xi} \right| = K(b_1 + \omega_1 x_o) (\sqrt{1 - \xi^2} + 1) (\xi + \sec \alpha)^{-3}, \quad |\xi| < 1$$

$$\text{Re } G = 0, \quad |\xi| > 1,$$

where $x_0(\xi)$ is given by (34b), and K is given by (34c). Finally, by (33), we obtain

$$\frac{df_1}{d\zeta} = \frac{K}{\pi(1-\zeta^2)^{\frac{1}{2}}} \left\{ \int_{-1}^1 \frac{(b_1 + \omega_1 x_0)(1-\xi^2 + \sqrt{1-\xi^2})}{(\xi-\zeta)(\xi+\sec\alpha)^3} d\xi + \sum_{n=-\infty}^{\infty} C_n (\zeta + \sec\alpha)^n \right\} \quad (37)$$

where C_n are real coefficients. The Laurentz series in the last term is expanded about $\zeta = -\sec\alpha$ (or $z = \infty$) for the convenience of application of the boundary conditions at $z = \infty$. The first term in (37) behaves like ζ^{-2} as $|\zeta| \rightarrow \infty$ and is regular everywhere except at $\zeta = \pm 1$. Now by using (34b),

$$w_1 = \frac{df_1}{d\zeta} \frac{d\zeta}{dz} = - \frac{(\zeta + \sec\alpha)^3}{K(\sqrt{1-\zeta^2} + 1)} \frac{df_1}{d\zeta} \quad (38)$$

From (37) and (38) we readily see that $C_n = 0$ except for $n = -1, -2$ in order that $w_1 \rightarrow 0$ as $z \rightarrow \infty$ (or $\zeta \rightarrow -\sec\alpha$) and w_1 is regular as $\zeta \rightarrow \infty$. Furthermore, with $Q_n = 0$, as explained above, (30) requires that

$$z w_1 \rightarrow 0 \quad \text{as} \quad |z| \rightarrow \infty \quad (39)$$

From (34b), we find that as $|z| \rightarrow \infty$,

$$z \sim \frac{K}{2} \frac{1 + i \tan\alpha}{(\zeta + \sec\alpha)^2}.$$

Making use of this result, we readily deduce from (37) and (38) that (39) is satisfied if, and only if, $C_{-1} = C_{-2} = 0$ also. Therefore all the coefficients C_n vanish, thereby the first order solution given by (37) and (38) is uniquely determined.

It may be noted that w_1 has a singularity at the edges of the plate, physically corresponding to a narrow spray sheet. For, with $\zeta = \pm(1+\epsilon)$, $|\epsilon| < 1$, we deduce from (37) and (38) that as $\zeta \rightarrow \pm 1$,

$$w_1 \sim \frac{i}{\pi} \frac{(1 + \sec \alpha)^3}{(2\epsilon)^{\frac{1}{2}}} \int_{-1}^1 \left(\frac{1 + \xi}{1 - \xi} \right)^{\frac{1}{2}} \left[\frac{(b_1 + \omega x_0)(1 + \sqrt{1 - \xi^2})}{(\xi + \sec \alpha)^3} \right] d\xi [1 + O(|\epsilon|^{\frac{1}{2}})]$$

From the behavior of w_1 on the free surface (ϵ positive small) it may be seen that the free surface starts to move into the cavity when the plate accelerates into the fluid (e.g., with $b_1 < 0$, $\omega = 0$), and vice versa. Furthermore, it is noted that w_1 is of order $|z|^{-3/2}$ for large values of $|z|$. The net effect can be seen to be such that there will be no net change in the cavity volume.

Since the spray sheets do not produce any singular force (unlike the leading edge suction on a thin airfoil), the normal force acting on the plate can be obtained by integrating the pressure along the plate so that for small t , in view of (3),

$$\begin{aligned} N &= \int_0^1 p(x_0, 0-, t) dx_0 = \int_0^1 [p_0(x_0, 0) + p_1(x_0, 0) + t p_2(x_0, 0)] dx_0 + O(t^2) \\ &= N_0 + N_1 + t N_2 + O(t^2) \end{aligned} \quad (40a)$$

The first term N_0 is given by (35). Now, from (10)

$$N_1 = \int_0^1 p_1(x_0, 0) dx_0 = -\rho \int_0^1 \varphi_1(x_0, 0) dx_0 = -\rho \int_{-1}^1 x_0(\xi) \frac{\partial \varphi_1(\xi, 0)}{\partial \xi} d\xi \quad (40b)$$

By substituting the real part of (37) for $\partial \varphi_1 / \partial \xi$ in the above integral, the normal force coefficient may be expressed as

$$C_{N_1} = N_1 / \left(\frac{1}{2} \rho U^2 l \right) = - \left(\frac{l b_1}{U^2} \right) \Gamma_{b_1} - \left(\frac{l^2 \omega^2}{U^2} \right) \Gamma_{\omega_1} \quad (41a)$$

where

$$\Gamma_{b_1}(\alpha) = -\frac{2}{\pi} \int_{-1}^1 \frac{x_0(\xi) d\xi}{\sqrt{1 - \xi^2}} \int_{-1}^1 \frac{\sqrt{1 - s^2}}{s - \xi} \frac{\partial x_0(s)}{\partial s} ds \quad (41b)$$

$$\Gamma_{\omega_1}(\alpha) = -\frac{1}{\pi} \int_{-1}^1 \frac{x_0(\xi) d\xi}{\sqrt{1 - \xi^2}} \int_{-1}^1 \frac{\sqrt{1 - s^2}}{s - \xi} \frac{\partial x_0^2(s)}{\partial s} ds \quad (41c)$$

the above integrals being interpreted by their Cauchy principal values. In (41a), b_1 and ω_1 are expressed in the physical units, and the plate length l and free stream velocity U are restored for completeness. The integrals in (41b) and (41c) cannot be expressed in terms of elementary functions of α ; they are integrated numerically and the results are plotted versus α in figure 5. In particular, we have

$$\Gamma_{b_1}(\pi/2) = 0.8448, \quad (42)$$

which is the special case treated by Gurevich [4]. The quantity C_{N_1} represents the jump in the normal force coefficient due to the acceleration.

(b) The second order solution for $\alpha = \pi/2$, $\omega_1 = b_2 = \omega_2 = 0$.

To facilitate investigation of the behavior of the second order solution, let us choose the special case: $\alpha = \pi/2$ (the flat plate being held normal to the stream) and ω_1, b_2, ω_2 all vanish. Then the boundary conditions (12c), (21), and (23) become

$$\partial \varphi_2 / \partial y_0 = -b_1 \partial^2 \varphi / \partial y_0^2, \quad \text{on } \eta = 0, \quad |\xi| < 1, \quad (43a)$$

$$\varphi_2 = 0 \quad \text{on } \eta = 0, \quad |\xi| > 1, \quad (43b)$$

$$|\nabla \varphi_2| \rightarrow 0 \quad \text{as } |z| \rightarrow \infty. \quad (43c)$$

Since the component a_1 of the acceleration parallel to the plate does not appear in the above conditions, the flow will not be affected by it up to the second order terms. Now, in the limit as $\alpha \rightarrow \pi/2$, the zeroth order solution becomes

$$z_0 = [2(1-\xi) - \xi(1-\xi^2)^{\frac{1}{2}} + \cos^{-1} \xi] / (4 + \pi) \quad (44)$$

and w_0 is still given by (34a). Consequently, from (28) and (29)

$$\text{Im } G = b_1 \frac{\partial u_0}{\partial x} \left| \frac{\partial x}{\partial \xi} \right| = b_1 \xi^{-2} [(1-\xi^2)^{-\frac{1}{2}} - 1], \quad \eta = 0, \quad |\xi| < 1,$$

$$\text{Re } G = 0 \quad \eta = 0, \quad |\xi| > 1.$$

Hence, by (33),

$$\frac{df}{d\zeta^2} = \frac{b_1}{\pi \sqrt{1-\zeta^2}} \int_{-1}^1 \frac{1-(1-\xi^2)^{\frac{1}{2}}}{(\xi-\zeta)\xi^2} d\xi \quad (45)$$

in which the complementary solution is zero, as can be shown by the same argument given previously for φ_1 . Carrying out the integration, we find

$$\frac{df}{d\zeta^2} = \frac{b_1}{\pi \zeta^2} (1-\zeta^2)^{-\frac{1}{2}} \left\{ 2\zeta + \log \frac{\zeta-1}{\zeta+1} - i\pi(1-\zeta^2)^{\frac{1}{2}} \right\}, \quad (46a)$$

$$w_2 = \frac{df}{dz} = - \frac{i(4+\pi)}{4\zeta^2} [1-(1-\zeta^2)^{\frac{1}{2}}] \frac{df}{d\zeta^2}. \quad (46b)$$

It is readily verified that w_2 also behaves like $z^{-3/2}$ for large $|z|$.

The second order normal force, by (40a) and (10), can be obtained from

$$\begin{aligned} N_2 &= \int_0^1 p_2(x_0, 0) dx_0 = \rho \int_{-1}^1 [\varphi_2 + u_0 u_1]_{\eta=0} \frac{\partial x_0}{\partial \xi} d\xi \\ &= \rho \int_{-1}^1 \left[-x_0(\xi) \frac{\partial \varphi_2}{\partial \xi^2} + u_0(\xi) u_1(\xi) \frac{\partial x_0}{\partial \xi} \right] d\xi. \end{aligned}$$

After substituting the various terms and evaluating the resulting integral, we find the simple result:

$$N_2 = 0. \quad (47)$$

For this special case we therefore have the normal force coefficient

$$C_N(t) = C_{N_0} + C_{N_1} + C_{N_2} t + O[(Ut/l)^2]$$

$$= 0.8798 + 0.8448 (lb_1/U^2) + O(Ut/l)^2 . \quad (48)$$

Thus for constant acceleration, C_N has, aside from the stepwise change, the following behavior

$$(dC_N/dt)_{t=0+} = 0 .$$

However, N_2 may not vanish when b_2 , ω_1 , and ω_2 are different from zero.

The quantity $m_1 = N_1/b_1$ may be called the initial induced mass of this cavity flow, then by (48)

$$m_1 = \frac{1}{2} \rho U^2 l C_{N_1} b_1^{-1} = 0.4224 \rho l^2 . \quad (49)$$

If the flat plate had undergone an acceleration b_1 normal to the flow without wake formation (a postulated Dirichlet flow), then the induced mass would be

$$m_0 = \pi \rho (l/2)^2 . \quad (50a)$$

Thus

$$m_1/m_0 = 0.5377 . \quad (50b)$$

This ratio is less than unity, as should be expected, since the cavitated side of the plate, being exposed to constant pressure, has no capacity of imparting kinetic energy to the exterior flow.

5. Re-entrant Jet Model

We proceed to consider the effect of the cavitation number σ (defined by (18)) on the unsteady flow when the cavity of the basic flow is finite in size. An additional degree of freedom achieved in this group of flow problems is that the cavity volume can now be changed arbitrary by prescribing a flow source at infinity. In order that the basic steady flow

is tractable to analysis, resort may be made to various cavity-flow models known in literature, such as the Riabouchinsky model [12], the re-entrant jet model [1], the wake model proposed independently by Joukowski [13], Roshko [14] and Eppler [15], and the modified wake model introduced by Wu [16]. In each of these flow models an artifice of some sort is introduced to admit σ as a free parameter, and to replace the actual wake flow of a real fluid by a simplified model within the framework of an equivalent potential flow. It has been found that in a positive range of σ up to moderate values of order unity, the agreement between these models may be considered very close (see Wu [5]). Furthermore, the validity of these models in predicting the hydrodynamic forces acting on the body has been supported by experimental observations. The purpose of the following several sections is to make a comparison between the resultant unsteady flows when different models are used for the basic cavity flow, for such a task should be of fundamental value in the study of unsteady cavity flows.

For simplicity the basic flow is taken to be that past a flat plate held normal to the stream of unperturbed velocity U and pressure p_∞ , forming a finite cavity with a prescribed cavity pressure $p_c (< p_\infty)$. According to the re-entrant jet model, the free streamlines eventually reverse their direction at the rear part of the cavity, forming a re-entrant jet which disappears on another "Riemann Sheet" (see figure 2). Due to the assumed symmetry of the flow, it suffices to consider only the flow in the left half physical z -plane. To save writing, both the half plate length, $l/2$, and the constant speed q_c (see (18)) on the cavity surface will be normalized to unity.

It is convenient to introduce the variable

$$\Omega = \log \frac{dz}{df_0} = \log \frac{1}{w_0} = \log \frac{1}{q_0} + i\theta \quad (51)$$

where f_0 is the complex potential of the basic flow, q_0 is the flow speed, and θ the flow inclination with respect to the x -axis. The part of the flow under consideration in the z, f_0 , and Ω -planes is shown in figure 2. By applying the Schwarz-Christoffel transformation, the flow region can be

mapped conformally into the upper half ζ -plane, with the point of infinity I, front stagnation E, plate edge A, jet infinity B and the rear stagnation C corresponding to $\zeta = \infty, 0, 1, b$, and c respectively. The required transformation is given by

$$\frac{df_o}{d\zeta} = A \frac{\zeta - c}{\zeta - b}, \quad (52)$$

$$\frac{d\Omega}{d\zeta} = \frac{B}{\zeta(\zeta - c) [(\zeta - 1)(\zeta - b)]^{\frac{1}{2}}} \quad (53)$$

where A, B are two coefficients. From the local behavior of Ω at $\zeta = 0$ and c , we find the relations

$$B = -\frac{1}{2} c b^{\frac{1}{2}}, \quad b = c(c-1)(c-\frac{3}{4})^{-1}. \quad (54)$$

Integrating (53), and by making use of (54), we obtain

$$\frac{dz}{df_o} = e^{\Omega} = \frac{[(c-b)(\zeta-1)/\{c-1)(\zeta-b)\}]^{\frac{1}{2}+1}}{[(c-b)(\zeta-1)/\{c-1)(\zeta-b)\}]^{\frac{1}{2}-1}} \left\{ \frac{[(\zeta-b)/b(\zeta-1)]^{\frac{1}{2}+1}}{[(\zeta-b)/b(\zeta-1)]^{\frac{1}{2}-1}} \right\}^{\frac{1}{2}}. \quad (55)$$

As $|\zeta| \rightarrow \infty$, $df_o/dz \rightarrow U = (1+\sigma)^{\frac{1}{2}}$. From this condition and (55) it results

$$(2+\sigma)/\sigma = \frac{1}{2} (c-1)^{\frac{1}{2}} c^{3/2} (c-\frac{3}{4})^{-3/2}. \quad (56)$$

Finally, by integrating (55) and (52) to obtain $z = z(\zeta)$, the coefficient A is determined by the plate length, giving

$$A = (b-1)^{3/2} / [K(1) - K(0)], \quad (57a)$$

where

$$\begin{aligned}
K(\xi) = & (2c-b) [b\xi(1-\xi)]^{\frac{1}{2}} + (2c-1) [\xi(b-\xi)]^{\frac{1}{2}} + (c-\frac{1}{2}b) \sin^{-1} \frac{2\xi-b}{b} \\
& + b^{\frac{1}{2}} [b(b-\frac{3}{2}) + c(2-b)] \sin^{-1}(2\xi-1) + (b-1)^{3/2} (c-b) \sin^{-1} \frac{(2b-1)\xi-b}{b-\xi}
\end{aligned}
\quad (57b)$$

This completes the zeroth order solution for prescribed σ . In particular, on the half plate, $y = 0$, $-1 < x < 0$ (or $\eta = 0+$, $0 < \xi < 1$), we have

$$x(\xi) = - [K(\xi) - K(0)] / [K(1) - K(0)] , \quad (58a)$$

$$\frac{dx}{d\xi} = A \left(\frac{c-\xi}{b-\xi} \right) \frac{[(c-b)(1-\xi)/(c-1)(b-\xi)]^{\frac{1}{2}+1}}{[(c-b)(1-\xi)/(c-1)(b-\xi)]^{\frac{1}{2}-1}} \left\{ \frac{[(b-\xi)/b(1-\xi)]^{\frac{1}{2}+1}}{[(b-\xi)/b(1-\xi)]^{\frac{1}{2}-1}} \right\}^{\frac{1}{2}} .
\quad (58b)$$

This expression is needed for the first order solution.

When a sudden acceleration of magnitude b_1 is applied normal to the plate, the boundary conditions corresponding to (32) can be written

$$G^+ + G^- = -2i b_1 dx/d\xi \quad \text{for} \quad 0 < \xi < 1 , \quad (59a)$$

$$G^+ - G^- = 0 \quad \text{for} \quad 1 < \xi < b , \quad (59b)$$

$$G^+ + G^- = 0 \quad \text{for} \quad -\infty < \xi < 0 \text{ and } b < \xi < \infty . \quad (59c)$$

The last condition (59c) expresses the assumption that the perturbed flow preserves the basic flow symmetry about the y -axis, that is, $v_1 = 0$ on $\eta = 0$, for $\xi < 0$ and $\xi > b$. The above conditions are expressed in the form different from (32). The solution, however can be written down directly by applying the same principle. It is obvious that $H(\zeta) = [(\zeta-1)(\zeta-b)]^{\frac{1}{2}}$, defined with branch cuts from $\zeta = -\infty$ to 1 and from b to ∞ so that $H \rightarrow \zeta$ as $|\zeta| \rightarrow \infty$, $0 < \arg \zeta < \pi$, is a homogeneous solution of the present problem. Therefore the general solution can be written

$$G(\xi) = \frac{b_1}{\pi H(\xi)} \int_0^1 \frac{[(1-\xi)(b-\xi)]^{\frac{1}{2}}}{\xi-\xi} \frac{dx}{d\xi} d\xi + \frac{1}{H(\xi)} \left[\sum_{n=-\infty}^{\infty} c_n \xi^n \right] \quad (60)$$

where $dx/d\xi$ is given by (58b) and c_n are arbitrary real constants. To determine c_n , we note first that the pressure must be finite at $\xi = 0$, hence $c_n = 0$ for $n < 0$. Furthermore, we note that the first term in (60) is of order of ξ^{-2} as $|\xi| \rightarrow \infty$, which implies that the behavior of df/dz at large distances is determined by the last term in (60). By applying condition (30) and using the symmetry property of the flow, we obtain $c_0 = Q_1/2\pi$, $c_1 = c_2 = \dots = 0$, where Q_1 is the source strength defined in (24).

The integral in (60) can be integrated explicitly. For the determination of hydrodynamic forces, however, only the real part of G on the plate (where $\eta = 0$, $0 < \xi < 1$) is needed. The final result is

$$\begin{aligned} \text{Re } G = & -\frac{b_1}{\pi} \frac{[(1-\xi)(b-\xi)]^{\frac{1}{2}}}{[K(1)-K(0)]} \left\{ A_1 + B_1(\xi-1) \right. \\ & \left. + \sqrt{b} \frac{(\xi-1)[(2c-b-2)(\xi-1)-c(b-1)]}{[\xi(b-\xi)]^{\frac{1}{2}}} \log \frac{(b-1)\xi+(b-\xi)+2\sqrt{(b-1)\xi(b-\xi)}}{b(1-\xi)} \right\} \\ & - \frac{Q_1}{2\pi} [(1-\xi)(b-\xi)]^{-\frac{1}{2}}. \end{aligned} \quad (61)$$

where

$$A_1 = \pi \left[b(c-1) + \frac{1}{2} \right] + \frac{\sqrt{b}}{2} (b^2 - 2b + 2c) \left(\frac{\pi}{2} + \sin^{-1} \frac{2-b}{b} \right) - \sqrt{b(b-1)(b-2c)},$$

$$B_1 = \sqrt{b} (b-2c) \left(\frac{\pi}{2} + \sin^{-1} \frac{2-b}{b} \right) - \pi(2c-1).$$

By using (40b) and the symmetry property of the flow, the first order normal force N_1 is given by

$$N_1 = -2\rho \int_0^1 x(\xi) \{\text{Re } G\} d\xi, \quad (62a)$$

where $x(\xi)$ and $\text{Re } G$ are given by (58a) and (61). The result can be expressed in terms of nondimensional parameters as

$$C_{N_1} = N_1 / \left(\frac{1}{2} \rho U^2 l \right) = - \left(\frac{l b_1}{U^2} \right) \Gamma_{b_1} - \left(\frac{Q_1}{U^2} \right) \Gamma_{Q_1}, \quad (62b)$$

in which the coefficients Γ_{b_1} and Γ_{Q_1} are functions of the cavitation number σ . Analytic evaluation of the above integral is too tedious to be practical. The numerical computation of these coefficients has been carried out with an IBM 7090, the final result is plotted versus σ in figures 6 and 7 to show the effect of (i) the finite cavity size and (ii) the change in the cavity volume. From this result several salient features of significance may be pointed out.

First, it is noted that as $\sigma \rightarrow 0$, the value of Γ_{b_1} tends to 0.8448, which is the limit of Kirchhoff-Helmholtz case (see equation (42)). From small to moderately large values of σ , Γ_{b_1} increases very slowly with increasing σ as compared with the rate of increase of the zeroth order drag, which increases approximately with the factor $(1+\sigma)$ (see, e.g., Gilbarg [1]).

Another point of interest is that $\Gamma_{Q_1} \rightarrow 0$ rather rapidly as $\sigma \rightarrow 0$, this limit being independent of Q_1 so long as Q_1 is finite. This result shows that the effect on the drag force of removing fluid at infinity is insignificant when the cavity is sufficiently long. Furthermore, it shows that the limit of the solution as $\sigma \rightarrow 0$ is non-uniform with respect to Q_1 since the solution in the Kirchhoff case does not exist unless $Q_1 = 0$.

6. Riabouchinsky's Model

The essential feature of the Riabouchinsky model is the introduction of an appropriate image body downstream of the real body so that the free boundaries of the cavity are connected by this pair of solid boundaries. Let us apply this model to consider the cavity flow past a flat plate set normal to the stream, the flow in the physical z -plane is shown in figure 3. Again, as in the problem stated in the previous section, the unperturbed velocity and pressure are U and p_∞ respectively, the

cavity pressure is p_c , corresponding to the cavitation number σ . Also the half plate length, $l/2$, and the constant speed q_c on the cavity will be normalized to unity. Furthermore, due to the assumed symmetry, only the left half z -plane of the flow need be considered.

For the present case it is convenient to denote by ζ the complex velocity potential. We further introduce an auxiliary complex variable τ defined by

$$\tau = \frac{1}{2} (w_0 + w_0^{-1}) \quad (63)$$

where $w_0 = d\zeta/dz$ is the hodograph plane of the basic flow. The flow field under consideration in the z -plane and conformal mapping planes w_0 , ζ , τ , are shown in figure 3. By the assumed symmetry of the flow, we may choose the potential at the front and rear stagnation points to be $\zeta = -n$ and n respectively, and $\zeta_B = -m$, $\zeta_D = m$. We further define k' and k by

$$k' = \frac{1}{1+\sigma}, \quad k = (1-k'^2)^{\frac{1}{2}} = (2\sigma + \sigma^2)^{\frac{1}{2}}/(1+\sigma). \quad (64)$$

Then, at $z = \infty$, we have

$$w_I = -i\sqrt{k'}, \quad \tau_I = \frac{i}{2} \{k'^{-\frac{1}{2}} - k'^{\frac{1}{2}}\} \quad (65)$$

The upper half τ -plane is mapped into the upper half ζ -plane by the Schwarz-Christoffel transformation

$$\zeta = 2n\sqrt{k'} \tau [4k'\tau^2 + (1-k')^2]^{-\frac{1}{2}}. \quad (66)$$

From the local conformal behavior of $\zeta(\tau)$ at the point D , i.e.

$\zeta_D = m$, $\tau_D = 1$, we find the relationship

$$m = 2n\sqrt{k'}/(1+k'). \quad (67)$$

Let us introduce another auxiliary variable $\omega = \mu + i\nu$ defined by

$$w_0 = \frac{d\zeta}{dz} = -i \frac{\sqrt{k'}}{dn\omega}, \quad (68)$$

where $dn\omega$ is the Jacobian elliptic function, delta amplitude of ω , with modulus k . In the following analysis the conventional notations (see [17]) for the elliptic functions and integrals will be used without specification and the modulus k will always be omitted to save writing. By substituting (68) and (63) into (66) it gives

$$\zeta = i \frac{n}{K} \frac{dn^2 \omega - k'}{\operatorname{sn} \omega \operatorname{cn} \omega} \quad (69)$$

From (68) and (69) we deduce that on both the front and the image half plates $-1 < x < 0$,

$$x = \frac{n}{\sqrt{k'}} \left[k'\mu - E(\mu) + k'(1-k') \frac{\operatorname{sn} \mu \operatorname{cn} \mu}{\operatorname{dn} \mu} \right], \quad 0 < \mu < \frac{K}{2}. \quad (70)$$

Evaluating the above result at $\mu = K/2$, which corresponds to $x = -1$, we obtain

$$n = \frac{2\sqrt{k'}}{E - k'K + \frac{(1-k')^2}{1+k'}} \quad (71)$$

which completes the necessary calculation for the basic flow.

Due to the presence of the image plate, an additional assumption is needed for this model in the study of its small time behavior. It is given that at $t = 0$ a sudden acceleration b_1 is applied on the front plate directed in the positive y -direction. For small $t > 0$ the image plate may be assumed to move in the y -direction with speed

$$v = -\beta b_1 t + O(t^2), \quad (72)$$

where β is an unknown constant. To determine this unknown constant β , we shall assume that on the image plate the jump in the drag due to the suddenly applied acceleration is zero. The physical significance of this assumption may be explained as follows. It has been pointed out by Wu [5] that the image plate in the basic flow may be regarded as a means to represent the energy dissipation in the wake flow of a real fluid. In fact, in a frame of reference at rest with respect to the fluid at infinity, the work done by the moving image plate is negative and numerically equal to the work done by the real plate since the total force on the pair of plates vanishes. This negative work done by the image plate therefore corresponds to the mechanical energy removed from the

system in unit time as there is no other means of dissipating energy in potential flow. Now in the unsteady motion, it is conceivable that the rate of dissipation in the wake (of the actual flow) cannot be affected at small time. This implies that the initial change of momentum at the image plate must vanish. It is this physical meaning that underlies the above assumption. For small $t > 0$, the boundary condition on the part of the ξ -axis ($\eta = 0^+$) corresponding to the image plate can be written by using (72) as

$$\text{Im } G = \beta b_1 \text{Rl} \left(\frac{dz}{d\xi} \right),$$

where $G = df/d\xi$. Or, by applying the continuation (31), we have on the part of the ξ -axis corresponding to the image plate the condition

$$G^+ + G^- = 2i \beta b_1 \text{Rl} \left(\frac{dz}{d\xi} \right). \quad (73)$$

Referring to conditions (32) and (73) and using (68), the boundary conditions of this problem are

$$G^+ + G^- = -2i b_1 \text{Rl} \left(\frac{dz}{d\xi} \right) = 2i \frac{b_1}{\sqrt{k'}} \frac{\text{cn}\mu(\xi)}{\text{sn}\mu(\xi)} \quad \text{for } -n < \xi < -m$$

$$G^+ - G^- = 0 \quad \text{for } -m < \xi < m$$

$$G^+ + G^- = 2i \beta b_1 \text{Rl} \left(\frac{dz}{d\xi} \right) = 2i \frac{\beta b_1}{\sqrt{k'}} \frac{\text{cn}\mu(\xi)}{\text{sn}\mu(\xi)} \quad \text{for } m < \xi < n$$

$$G^+ + G^- = 0 \quad \text{for } -\infty < \xi < -n, \quad n < \xi < \infty;$$

where m and n are given by (67) and (71).

Consider the function $h(\xi) = \sqrt{\xi^2 - m^2}$ with the branch cuts from $-\infty$ to $-m$ and from m to ∞ , and $h(\xi) \rightarrow \xi$ as $|\xi| \rightarrow \infty$,

$0 < \arg \xi < \pi$. This $h(\xi)$ is obviously a solution of the corresponding homogeneous Hilbert problem and satisfies (31). Therefore, by using (33), the required solution may be written

$$G(\xi) = - \frac{b_1 k'^{-\frac{1}{2}}}{\pi \sqrt{\xi^2 - m^2}} \left\{ \int_{-n}^{-m} \frac{\sqrt{\xi^2 - m^2} \frac{\text{cn}\mu(\xi)}{\text{sn}\mu(\xi)}}{\xi - \zeta} d\xi - \beta \int_m^n \frac{\sqrt{\xi^2 - m^2} \frac{\text{cn}\mu(\xi)}{\text{sn}\mu(\xi)}}{\xi - \zeta} d\xi \right\} + \frac{P(\xi)}{\sqrt{\xi^2 - m^2}} \quad (74)$$

where $P(\zeta) = \sum_{n=-\infty}^{\infty} c_n \zeta^n$, and c_n are arbitrary real constants. Again, since the pressure at $\zeta = 0$ should be finite and the first term in (74) is of order of ζ^{-2} as $|\zeta| \rightarrow \infty$, we have, by applying condition (30) and the symmetric property of the flow, $P(\zeta) = Q_1/2\pi$. By using (69), the variable ξ in (74) can be transformed into the variable μ , giving

$$G(\zeta) = \frac{b_1(1-k')n}{\pi\sqrt{k'}\sqrt{\zeta^2-m^2}} \left\{ \int_0^{K/2} \frac{cn^2\mu dn^2\mu \left(1 - \frac{k'}{dn^2\mu}\right)^2}{dn^2\mu + \frac{1+k'}{n}\zeta dn\mu + k'} d\mu \right. \\ \left. + \beta \int_0^{K/2} \frac{cn^2\mu dn^2\mu \left(1 - \frac{k'}{dn^2\mu}\right)^2}{dn^2\mu - \frac{1+k'}{n}\zeta dn\mu + k'} d\mu \right\} \\ + \frac{Q_1}{2\pi\sqrt{\zeta^2-m^2}} \quad (75)$$

After integrating the above expression and by using (69) we deduce that

$$Re\ G = \mp \frac{b_1}{\pi} \left\{ (1+\beta) \left[A_2 \left(\frac{1+k'}{n} \right) \left(\frac{dn\mu}{dn^2\mu - k'} \right) + \frac{K}{2\sqrt{k'}} \left(\frac{dn^2\mu - k'}{dn\mu} \right) \right. \right. \\ \left. \left. + \frac{1}{\sqrt{k'}} \frac{dn\mu cn^2\mu}{sn^2\mu} \pi^* \left(\sin^{-1} \frac{1}{\sqrt{1+k'}}, \frac{1}{sn^2\mu} \right) - k'^{5/2} \frac{sn^2\mu}{dn\mu cn^2\mu} \right] \right. \\ \left. \cdot \pi \left(\sin^{-1} \frac{1}{\sqrt{1+k'}}, \frac{dn^2\mu}{cn^2\mu} \right) \right] \pm (1-\beta) \left[B_2 \frac{dn^2\mu + k'}{dn^2\mu - k'} + \frac{1}{2} \frac{(dn^2\mu - k') k^{-1/2}}{(1-k') sn\mu cn\mu} \right. \\ \left. \cdot \log \frac{cn\mu + \sqrt{k'} sn\mu}{cn\mu - \sqrt{k'} sn\mu} \right] \left\} \mp \frac{Q_1}{2\pi} \frac{1+k'}{n} \left(\frac{dn\mu}{dn^2\mu - k'} \right), \quad 0 < \mu < \frac{K}{2} \quad (76a)$$

where the upper sign is used for the front plate, the lower sign for the image plate, π^* denotes the function which corresponds to the Cauchy principal value of the integral representation of the function π (for the definition, see [17]), and

$$A_2 = \frac{1-k'}{1+k'} \frac{E+k'K+(1+k')}{E-k'K+\frac{(1-k')^2}{(1+k')}} \quad (76b)$$

$$B_2 = \frac{1+k'}{2\sqrt{k'}} \left[\frac{\pi}{2} \frac{1-k'}{1+k'} + \sin^{-1} \frac{1-k'}{1+k'} \right] \quad (76c)$$

By applying (40b), (70), (76a) and (66), we obtain the normal force N_1 on the front plate as

$$N_1 = 2\rho \{b_1 [(1+\beta)I_1 + (1-\beta)I_2] + Q_1 I_3\} \quad (77)$$

and the normal force D_y (in the positive y -direction) on the image plate as

$$D_y = 2\rho \{b_1 [(1+\beta)I_1 - (1-\beta)I_2] + Q_1 I_3\} \quad (78)$$

where

$$\begin{aligned} I_1 = & \frac{(1-k')n}{\pi\sqrt{k'}} \int_0^{K/2} x(\mu) \frac{\operatorname{sn}\mu \operatorname{cn}\mu}{\operatorname{dn}\mu} \left\{ \frac{(1+k')\sqrt{k'}}{n} A_2 + \frac{K}{2} \frac{(\operatorname{dn}^2\mu - k')^2}{\operatorname{dn}^2\mu} \right. \\ & + (\operatorname{dn}^2\mu - k') \left[\frac{\operatorname{cn}^2\mu}{\operatorname{sn}^2\mu} \pi^* \left(\sin^{-1} \frac{1}{\sqrt{1+k'}} , \frac{1}{\operatorname{sn}^2\mu} \right) \right. \\ & \left. \left. - k'^3 \frac{\operatorname{sn}^2\mu}{\operatorname{dn}^2\mu \operatorname{cn}^2\mu} \pi \left(\sin^{-1} \frac{1}{\sqrt{1+k'}} , \frac{\operatorname{dn}^2\mu}{\operatorname{cn}^2\mu} \right) \right] \right\} d\mu \quad (79a) \end{aligned}$$

$$\begin{aligned} I_2 = & \frac{(1-k')n}{\pi\sqrt{k'}} \int_0^{K/2} x(\mu) \left[\sqrt{k'} B_2 \left(\frac{\operatorname{dn}^2\mu + k'}{\operatorname{dn}^2\mu} \right) \operatorname{sn}\mu \operatorname{cn}\mu + \frac{1}{2} \frac{(\operatorname{dn}^2\mu - k')^2}{(1-k')\operatorname{dn}^2\mu} \right. \\ & \left. \cdot \log \frac{\operatorname{cn}\mu + \sqrt{k'} \operatorname{sn}\mu}{\operatorname{cn}\mu - \sqrt{k'} \operatorname{sn}\mu} \right] d\mu \quad (79b) \end{aligned}$$

$$I_3 = \frac{1-k'^2}{2\pi} \int_0^{K/2} x(\mu) \frac{\operatorname{sn}\mu \operatorname{cn}\mu}{\operatorname{dn}\mu} d\mu \quad (79c)$$

and $x(\mu)$ is given by (70).

Now we apply the condition that on the image plate the change in the drag due to the applied acceleration is zero, giving

$$2\rho b_1 [(1+\beta) I_1 - (1-\beta) I_2] = 0,$$

or

$$\beta = \frac{I_2 - I_1}{I_1 + I_2}. \quad (80)$$

Substituting (80) into (77), we obtain the drag on the front plate as

$$N_1 = 4\rho \left[b_1 \frac{2I_1 I_2}{I_1 + I_2} + \frac{Q}{2} I_3 \right]. \quad (81)$$

This result can also be expressed in the following non-dimensional form:

$$C_{N_1} = N_1 / \left(\frac{1}{2} \rho U^2 l \right) = - \left(\frac{l b_1}{U^2} \right) \Gamma_{b_1} - \left(\frac{Q}{U^2} \right) \Gamma_{Q_1} \quad (82a)$$

where

$$\Gamma_{b_1} = -4 \frac{I_1 I_2}{I_1 + I_2} \quad \Gamma_{Q_1} = -2I_3. \quad (82b)$$

The integrals I_1 , I_2 and I_3 are computed numerically with an IBM 7090 and the final result of Γ_{b_1} and Γ_{Q_1} are shown in figures 6 and 7 as a comparison with the results of other flow models. It is further noted that as $\sigma \rightarrow 0$, Γ_{b_1} reduces to (42) of the Kirchhoff case and $\Gamma_{Q_1} \rightarrow 0$.

7. A Wake Model for an Oblique Plate with a Finite Cavity

Thus far we have considered the accelerating motion of an inclined plate in Kirchhoff flow, and of the finite-cavity flow past a plate broad-wise to the stream. Application of either the re-entrant jet model or the Riabouchinsky model to an oblique plate with a finite cavity formation leads to very complicated analyses. The task is considerably simplified, however, if we adopt a modified wake model recently proposed by Wu[16]

to describe the basic steady flow. The purpose of this section is to determine the effect of acceleration on the hydrodynamic forces with the cavitation number σ and the angle of attack α as two free parameters.

The basic flow is taken to be a uniform stream of infinite extent impinging on a flat plate at an incidence angle α , to which a finite cavity is attached. According to this modified wake model, the incoming stagnation streamline branches off the plate at the leading edge A and the trailing edge B, forming two free streamlines ACI and BC'I which are assumed to become asymptotically parallel to the main flow at downstream infinity (see figure 4). The pressure on the parts AC and BC' of the free streamlines is assumed to take the constant cavity pressure p_c , and the space within the closed curve ACC'BA is regarded to represent approximately the cavity. The space in between the free streamlines CI and C'I represents a crude model of the dissipating wake, along its boundary the uniform stream conditions are eventually restored at downstream infinity. The flow outside this infinite wake strip is assumed to be irrotational. The locations of the points C and C' are determined with two assumptions; the first is that both the velocity potential and the flow inclination at C and C' are equal, and the second assumption is the so-called "hodograph-slit condition" that the free streamlines CI and C'I form a slit of undetermined shape in the hodograph plane. With these two additional assumptions the whole flow field outside the wake is then completely determined. For the convenience of subsequent application, the solution of the basic flow is reproduced briefly in the following. The plate length l and the constant speed q_c on AC and BC' are again normalized to unity.

The flow in the physical z -plane, the complex potential f_o -plane, and the hodograph w_o -plane are shown in figure 4. The subscript of w_o will be omitted for simplicity. We further introduce the parametric ζ -plane defined by

$$\zeta = \frac{1}{2} (w + w^{-1}) \quad (83a)$$

or

$$w = \zeta - (\zeta^2 - 1)^{\frac{1}{2}}, \quad (83b)$$

in which the function $(\zeta^2 - 1)^{\frac{1}{2}}$ is defined with a branch cut made between the points $\zeta = -1$ and 1 so that $(\zeta^2 - 1)^{\frac{1}{2}} \rightarrow \zeta$ as $|\zeta| \rightarrow \infty$. At the point of infinity, the complex velocity takes the value

$$w = W = U e^{-i\alpha}, \quad U = (1 + \sigma)^{-\frac{1}{2}}. \quad (84)$$

The corresponding value of ζ is

$$\zeta_{\infty} = \frac{1}{2} (U^{-1} e^{i\alpha} + U e^{-i\alpha}). \quad (85)$$

Since $\text{Im } f_0 = 0$ on the entire real ζ -axis, the complex potential $f_0(\zeta)$ can be continued analytically into the lower-half ζ -plane by

$$f_0(\bar{\zeta}) = \overline{f_0(\zeta)}. \quad (86)$$

Now from the asymptotic behavior of the streamlines $\psi_0 = \text{const.}$ near the point $\zeta = \zeta_{\infty}$, it is evident that f_0 must have there a simple pole. Furthermore, from the local conformal behavior of f_0 at $f_0 = 0$, it is obvious that $f_0 = O(\zeta^{-2})$ as $|\zeta| \rightarrow \infty$. Therefore the solution must be of the form

$$f_0 = \frac{A}{4} \frac{1}{(\zeta - \zeta_{\infty})(\zeta - \bar{\zeta}_{\infty})} \quad (87a)$$

where A is a real constant. Or, expressing in terms of w by (83),

$$f_0 = \frac{A w^2}{(w - W)(w - \bar{W})(w - W^{-1})(w - \bar{W}^{-1})}. \quad (87b)$$

The z -plane is determined by integration of $dz/df_0 = 1/w$, giving

$$z + a = \frac{f_0(w)}{w} + iB \left\{ \left(\frac{1}{\bar{W}} - W \right) \left[\frac{1}{W} \log(w - W) - W \log\left(w - \frac{1}{W}\right) \right] - \left(\frac{1}{W} - W \right) \left[\frac{1}{W} \log(w - \bar{W}) - W \log\left(w - \frac{1}{\bar{W}}\right) \right] \right\}, \quad (88a)$$

where the constant B is related to A by

$$\frac{A}{B} = 2 \left(\frac{1}{U} - U \right) \sin \alpha \left[\left(\frac{1}{U} + U \right)^2 - (2 \cos \alpha)^2 \right]. \quad (88b)$$

Finally, the constant A is determined by the plate length as

$$A = \left[\left(\frac{1}{U} + U \right)^2 - (2 \cos \alpha)^2 \right] / K, \quad (89a)$$

$$K = 2 \frac{(U^{-1} + U)^2 + (2 \cos \alpha)^2}{(U^{-1} + U)^2 - (2 \cos \alpha)^2} + \frac{\pi(U^{-1} + U)}{2 \sin \alpha} + \frac{(U^{-1} + U)^2 - (2 \cos \alpha)^2}{(U^{-1} - U) \sin \alpha} \tan^{-1} \left(\frac{U^{-1} - U}{2 \sin \alpha} \right). \quad (89b)$$

For the unsteady motion we shall confine ourselves to the simple case of constant acceleration so that for any point z of the plate,

$$dz/dt = (a_1 + i b_1)t. \quad (90)$$

Then the boundary conditions of this problem (see equation 32) become

$$\begin{aligned} G^+ + G^- &= -2i b_1 R t (dz/d\zeta) & \text{for } |\xi| > 1, \\ G^+ - G^- &= 0 & \text{for } |\xi| < 1, \end{aligned} \quad (91)$$

where $G = df/d\zeta$. In the first condition, $dz/d\zeta = w^{-1} df_0/d\zeta$ can be deduced from (83) and (87).

Similar to the general solution of the problem with the boundary conditions given in (32), the general solution of the above boundary value problem, stated in (91), can be written

$$G(\zeta) = \frac{b_1}{\pi(\zeta^2 - 1)^{\frac{1}{2}}} \left\{ \int_{-\infty}^{-1} - \int_1^{\infty} \right\} \frac{(\xi^2 - 1)^{\frac{1}{2}} (dx/d\xi)}{\xi - \zeta} d\xi + (\zeta^2 - 1)^{-\frac{1}{2}} P(\zeta) \quad (92)$$

where the function $(\zeta^2 - 1)^{\frac{1}{2}}$ is defined in the entire ζ -plane with branch cuts from $-\infty$ to -1 and from 1 to ∞ along the ξ -axis so that $(\zeta^2 - 1)^{\frac{1}{2}} \rightarrow \zeta$ as $|\zeta| \rightarrow \infty$, for $0 < \arg \zeta < \pi$ (note that the branch cut

for the function $(\zeta^2 - 1)^{\frac{1}{2}}$ for this perturbation part is different from that for the basic problem as described after (83b)). The arbitrary function $P(\zeta)$ is real on the real ζ -axis, and hence can be expanded into the Laurent's series

$$P(\zeta) = \sum_{n=-\infty}^{\infty} [c_n(\zeta - \zeta_{\infty})^n + \bar{c}_n(\zeta - \bar{\zeta}_{\infty})^n] .$$

Since $z \sim (\zeta - \zeta_{\infty})^{-1}$ as $|z| \rightarrow \infty$, it is necessary to have $c_n = 0$ for $n \leq -2$ in order that the perturbation velocity w_1 vanishes at infinity. Furthermore, we must impose $c_n = 0$ for $n \geq 0$ if we require the pressure at $|\zeta| = \infty$ to be finite. Finally, application of (30) yields

$$P(\zeta) = c_{-1}(\zeta - \zeta_{\infty})^{-1} + \bar{c}_{-1}(\zeta - \bar{\zeta}_{\infty})^{-1} , \quad (93a)$$

$$c_{-1} = - \frac{Q_1}{2\pi} (\zeta_{\infty}^2 - 1)^{\frac{1}{2}} = - \frac{Q_1}{4\pi} (W^{-1} - W) , \quad (93b)$$

where use has been made of (84) and (85).

Transforming the variable ζ in the above solution into w by (83) we obtain

$$G = \frac{2b_1 w^2}{\pi(1-w^2)} \int_{-1}^1 g(w, u) du - \frac{Q_1 w^2}{\pi(1-w^2)} \left[\frac{1-W^2}{(w-W)(w\bar{W}-1)} + \frac{1-\bar{W}^2}{(w-\bar{W})(wW-1)} \right] \quad (94a)$$

where

$$g(w, u) = \frac{1}{u} \frac{df_0}{du} \frac{1-u^2}{(u-w)(uw-1)} . \quad (94b)$$

In particular, on the rigid plate, $-1 < w < 1$, $Re G = \partial \varphi_1 / \partial \xi$, and

$$\frac{\partial \varphi_1}{\partial \xi} = \frac{2w^2}{\pi(1-w^2)} \left\{ b_1 (P.V.) \int_{-1}^1 g(w, u) du - \frac{Q_1}{2} \left[\frac{1-W^2}{(w-W)(w\bar{W}-1)} + \frac{1-\bar{W}^2}{(w-\bar{W})(wW-1)} \right] \right\} \quad (95)$$

where (P.V.) denotes the Cauchy principal value of the integral.

The normal force N_1 acting on the plate due to the acceleration is given by

$$N_1 = \rho \int_{-1}^1 x(w) \frac{\partial \varphi_1}{\partial \xi} \frac{\partial \xi}{\partial w} dw \quad (96)$$

where $\partial \varphi_1 / \partial \xi$ is given by (95), and $\partial \xi / \partial w$ can be obtained from (83). As in the previous cases, we write

$$C_{N_1} = - \left(\frac{lb_1}{U^2} \right) \Gamma_{b_1} - \left(\frac{Q_1}{U^2} \right) \Gamma_{Q_1} \quad (97)$$

then, from (96),

$$\Gamma_{b_1} = \frac{2}{\pi} (\text{P.V.}) \int_{-1}^1 x(w) dw \int_{-1}^1 g(w, u) du \quad (98)$$

$$\Gamma_{Q_1} = - \frac{1}{\pi} \int_{-1}^1 x(w) \left[\frac{1-W^2}{(w-W)(w\bar{W}-1)} + \frac{1-\bar{W}^2}{(w-\bar{W})(wW-1)} \right] dw \quad (99)$$

These integrals have been computed numerically with an IBM 7090, the final result of Γ_{b_1} and Γ_{Q_1} is shown in figures 8 and 9 for $\alpha = 90^\circ$, 75° and 45° . The result of the special case $\alpha = \pi/2$ is also compared with the other flow models in figures 6 and 7. For the special case $\alpha = \pi/2$, we deduce from (98) the following expression for small σ

$$\Gamma_{b_1} = 0.8448 [1 + 0.067 \sigma^2 (1-\sigma) + (\sigma^4)] \quad (100)$$

which is the special case already noted by Wu [6].

8. Conclusion

In the foregoing sections we have treated the accelerating motion of an inclined plate in Kirchhoff flow and in the finite-cavity flow, and that of a pure-drag plate in Riabouchinsky flow and in the re-entrant jet flow. From these results we make conclusions by pointing out the following features of interest.

(i) Effect of the cavitation number σ — For $0 < \sigma < 1$, the force coefficient Γ_{b_1} due to the normal acceleration evaluated by using three different cavity-flow models is nearly identical, tending to 0.8448, the Kirchhoff-Helmholtz limit, as $\sigma \rightarrow 0$, and being very insensitive to σ in this range. As σ increases further from 1, Γ_{b_1} increases very slowly compared with the rate of increase of the steady state drag coefficient, which has the factor $(1 + \sigma)$ approximately. For a fixed σ in this region, the value of Γ_{b_1} by the wake model is higher than that of the Riabouchinsky model, which is in turn higher than that of the re-entrant jet model. This result indicates that in all practical ranges of σ , it is unimportant to consider the effect of σ for the small-time limit or the large frequency limit.

(ii) Effect of incidence α — For moderate and large σ , the force coefficient Γ_{b_1} increases faster with increasing σ , the smaller is the value α . This result should be expected on the physical standpoint since there contains more liquid in the circular cylinder spanned by the plate at smaller α . This effect of α , however, becomes insignificant for σ less than order unity.

(iii) Effect of change in cavity volume — The force coefficient Γ_{Q_1} due to the change in cavity volume displays wider differences between different flow-models. All these models show, however, that $\Gamma_{Q_1} \rightarrow 0$ rather rapidly as $\sigma \rightarrow 0$, saying that the effect on the drag force of removing fluid at infinity is insignificant when the cavity is sufficiently long. This analysis also justifies the assumption that p is bounded at infinity introduced in some of the previous works [4,6,7,8,9,18]

in the case of unsteady flows with infinite cavities. As shown by this analysis, this assumption actually turns out to be a necessary condition for the existence of the solution in the case of $\sigma = 0$.

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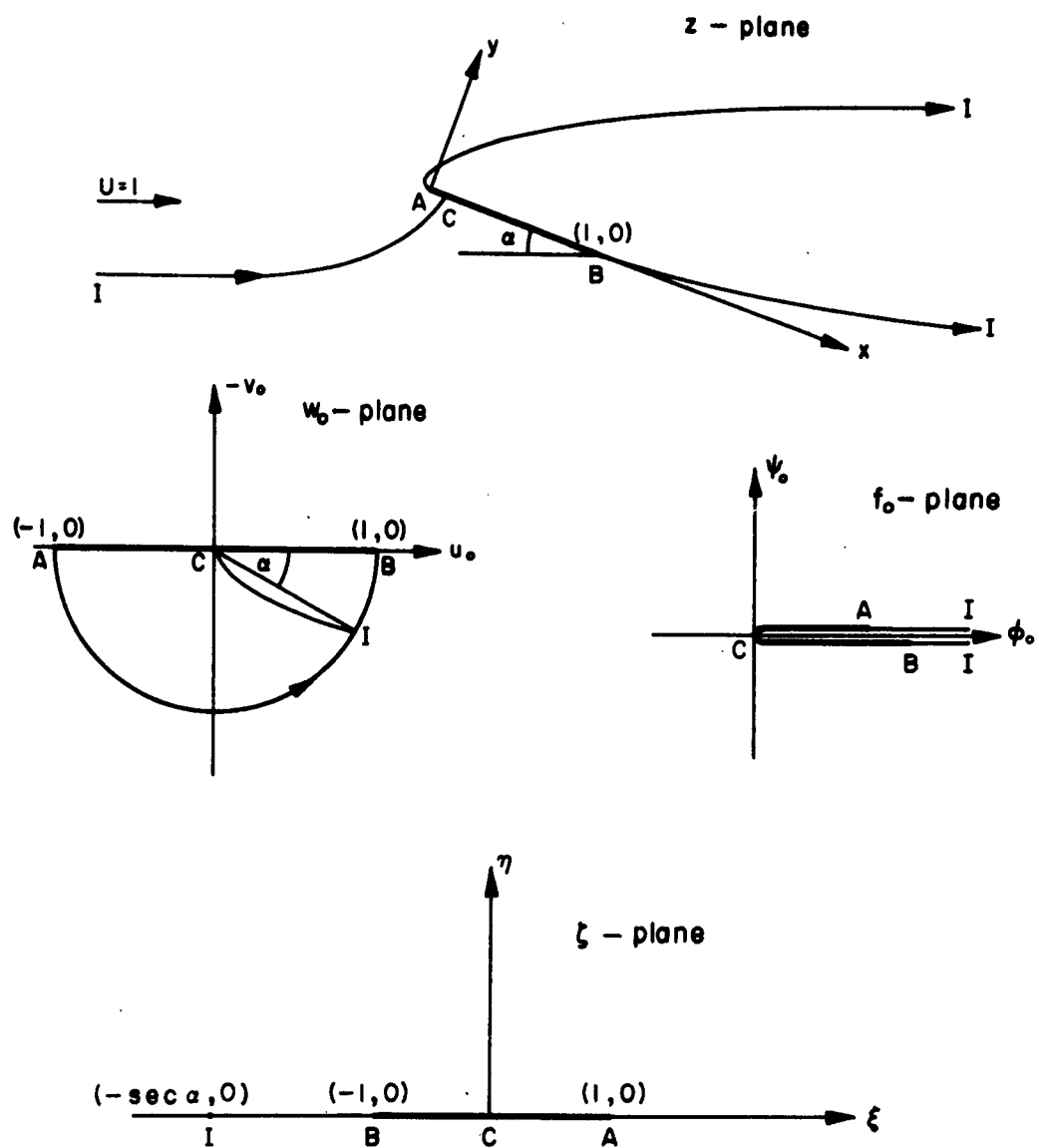


Figure 1. Inclined lamina in Kirchhoff flow and its conformal mapping planes.

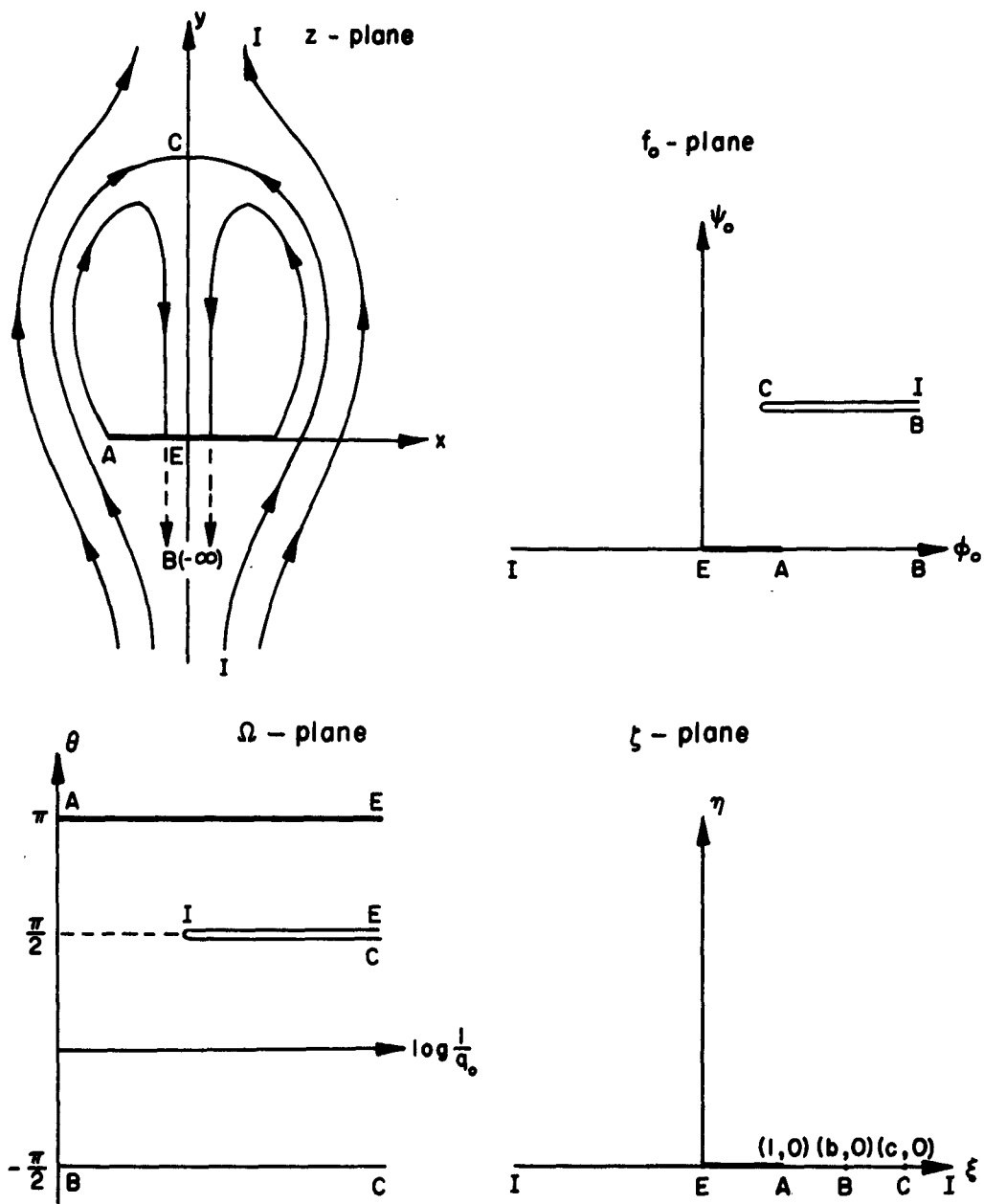


Figure 2. Re-entrant jet flow past a flat plate and its conformal mapping planes.

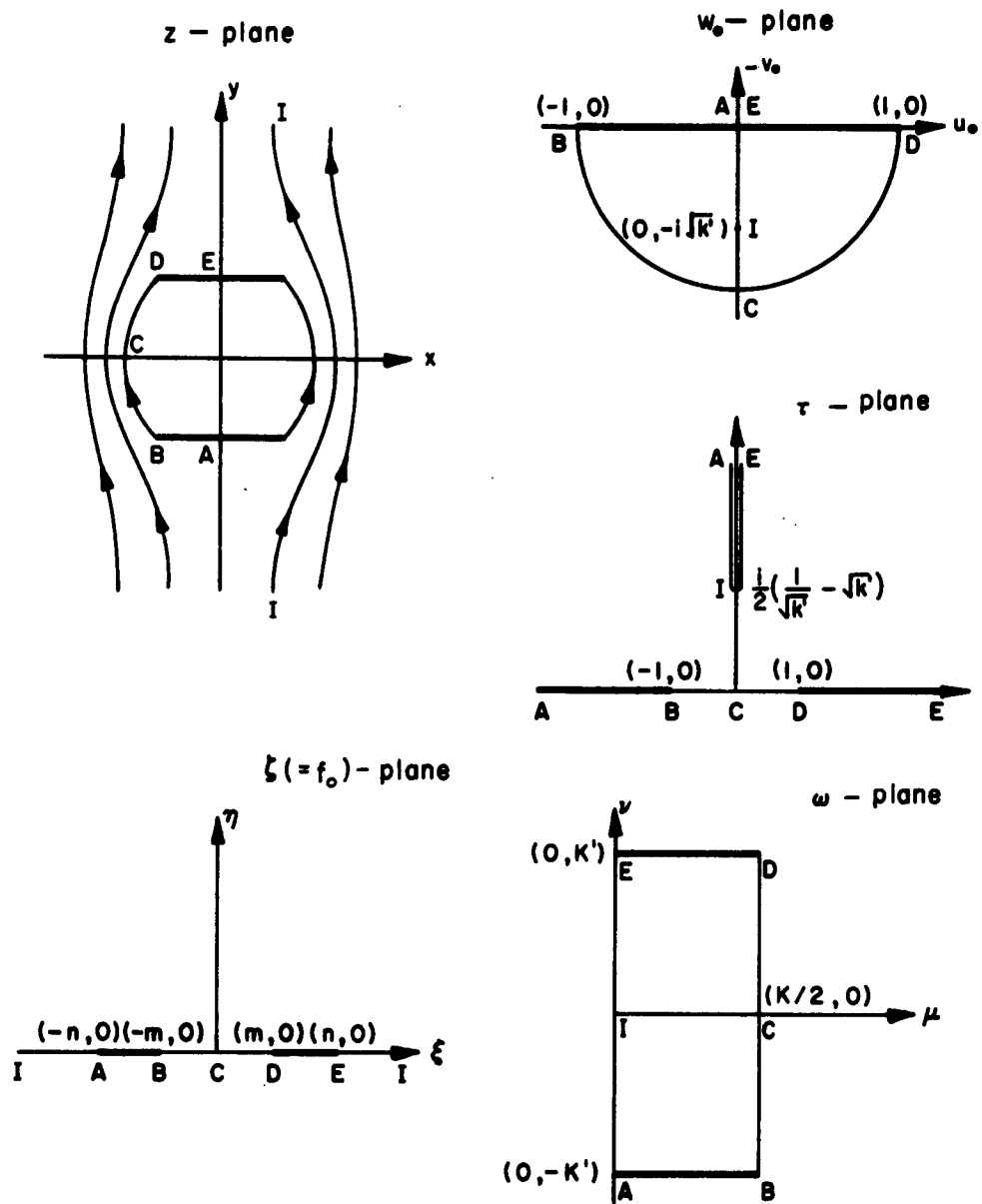


Figure 3. Riabouchinsky's model for a flat plate and its conformal mapping planes.

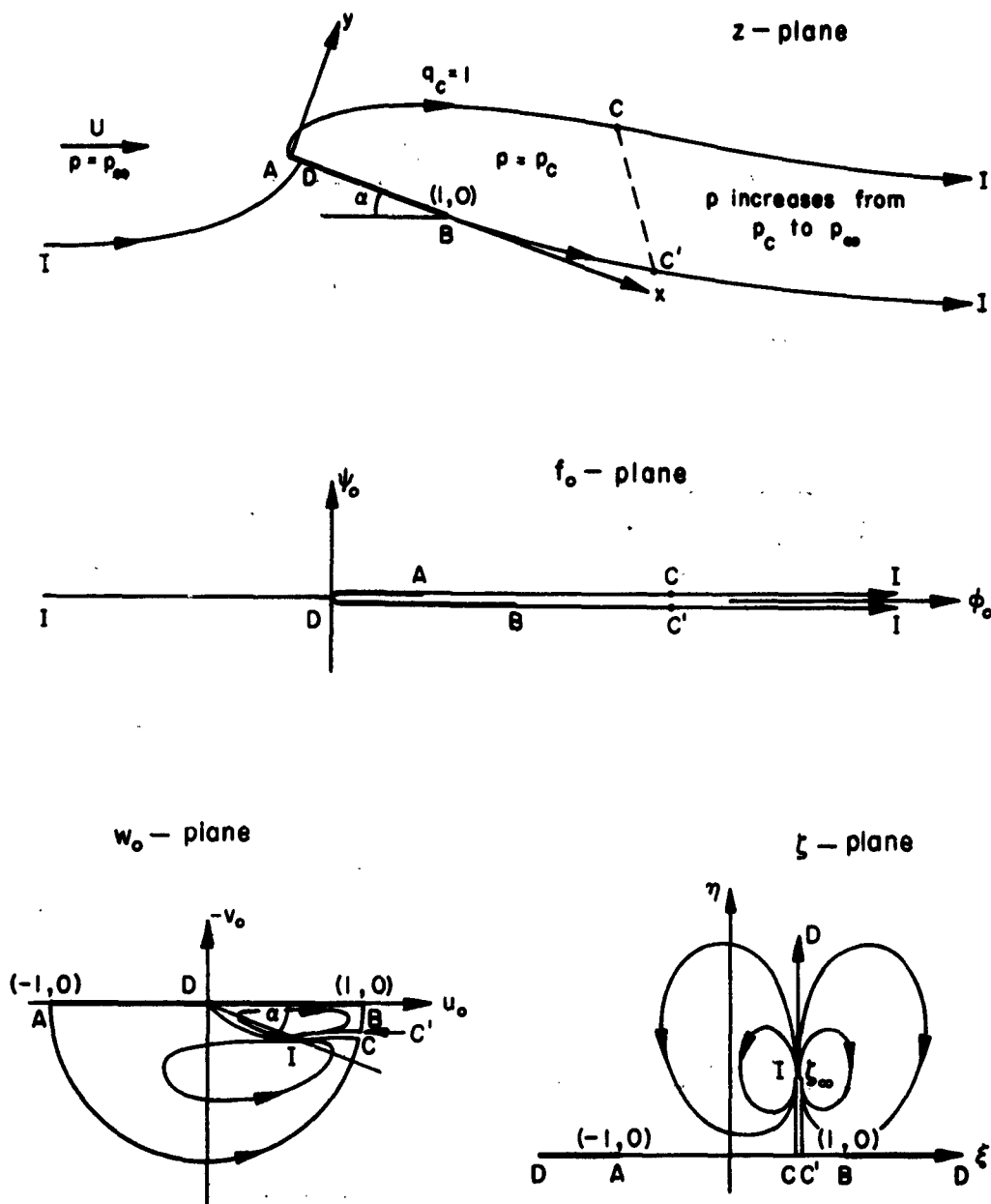


Figure 4. A wake model for an oblique plate with a finite cavity and its conformal mapping planes.

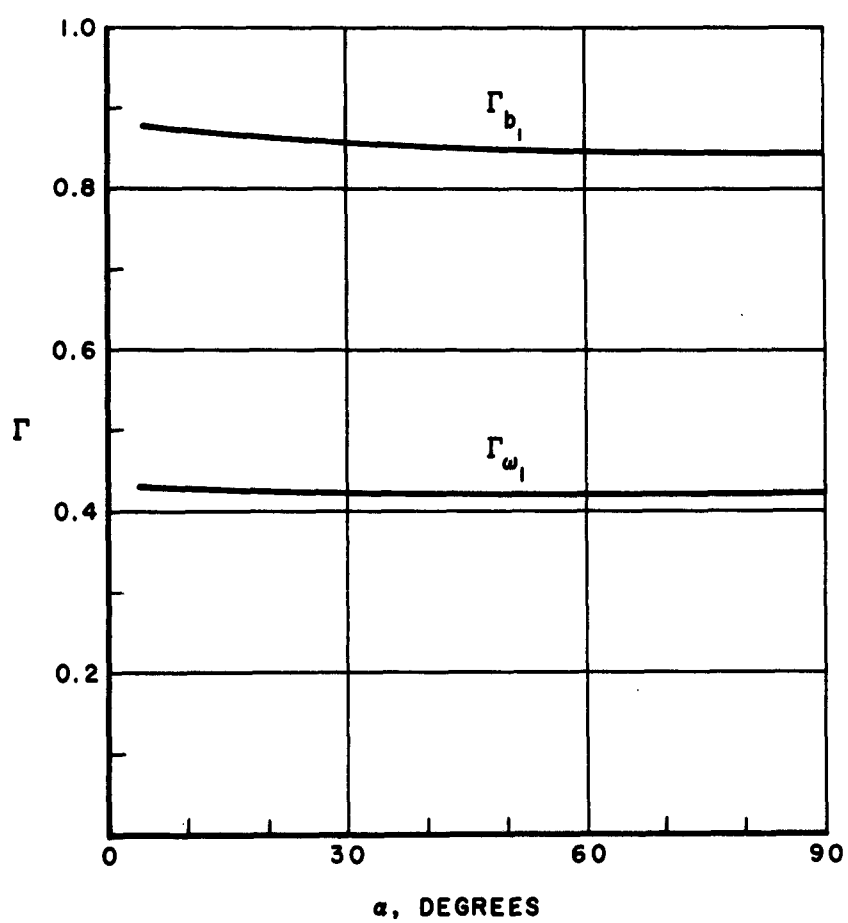


Figure 5. Normal force coefficients for inclined lamina in Kirchhoff flow.

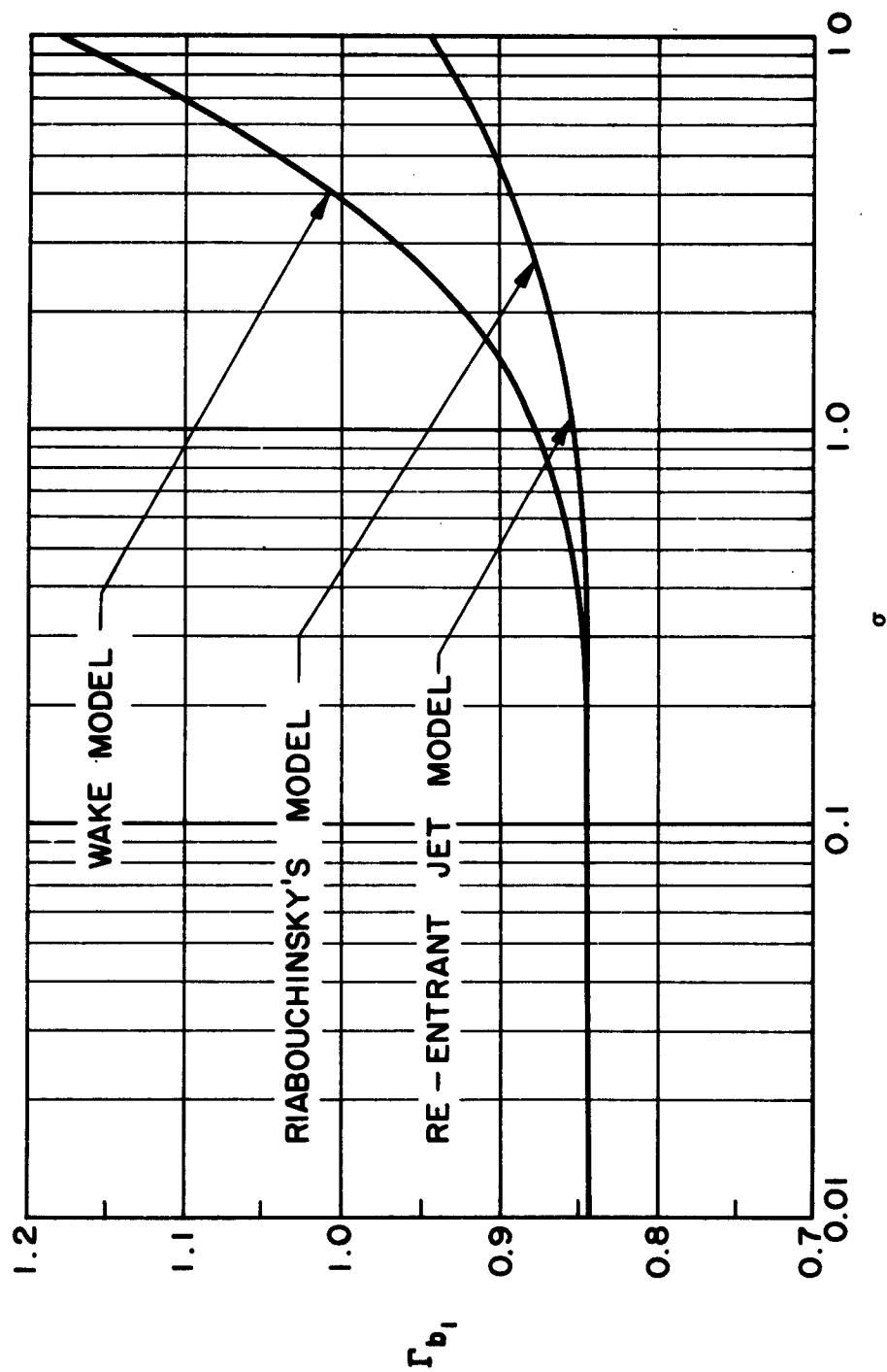


Figure 6. Normal force coefficients for the sudden acceleration of a flat plate placed normally to the incoming stream.

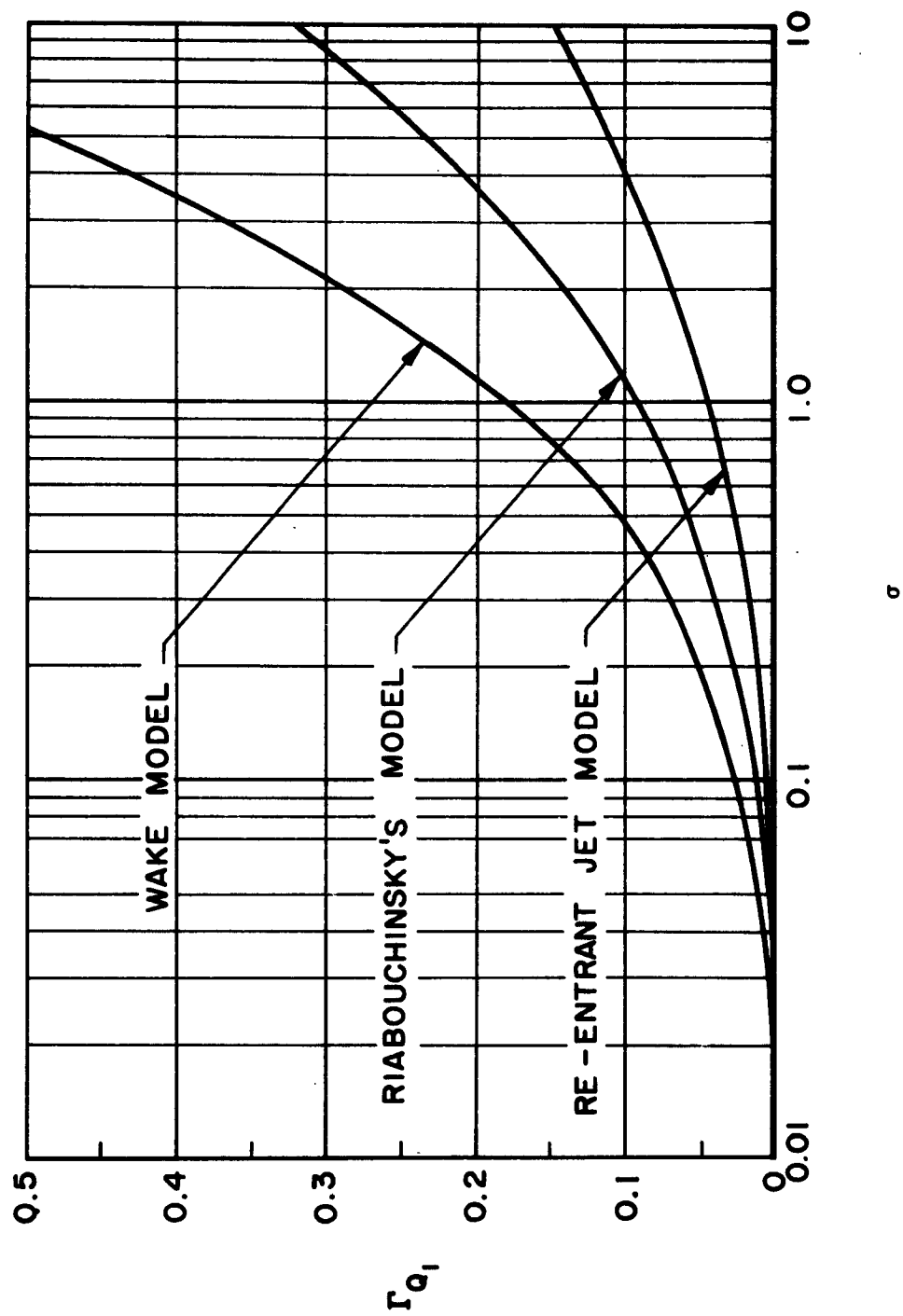


Figure 7. Normal force coefficients due to a source at the point of infinity for a flat plate placed normally to the incoming stream.

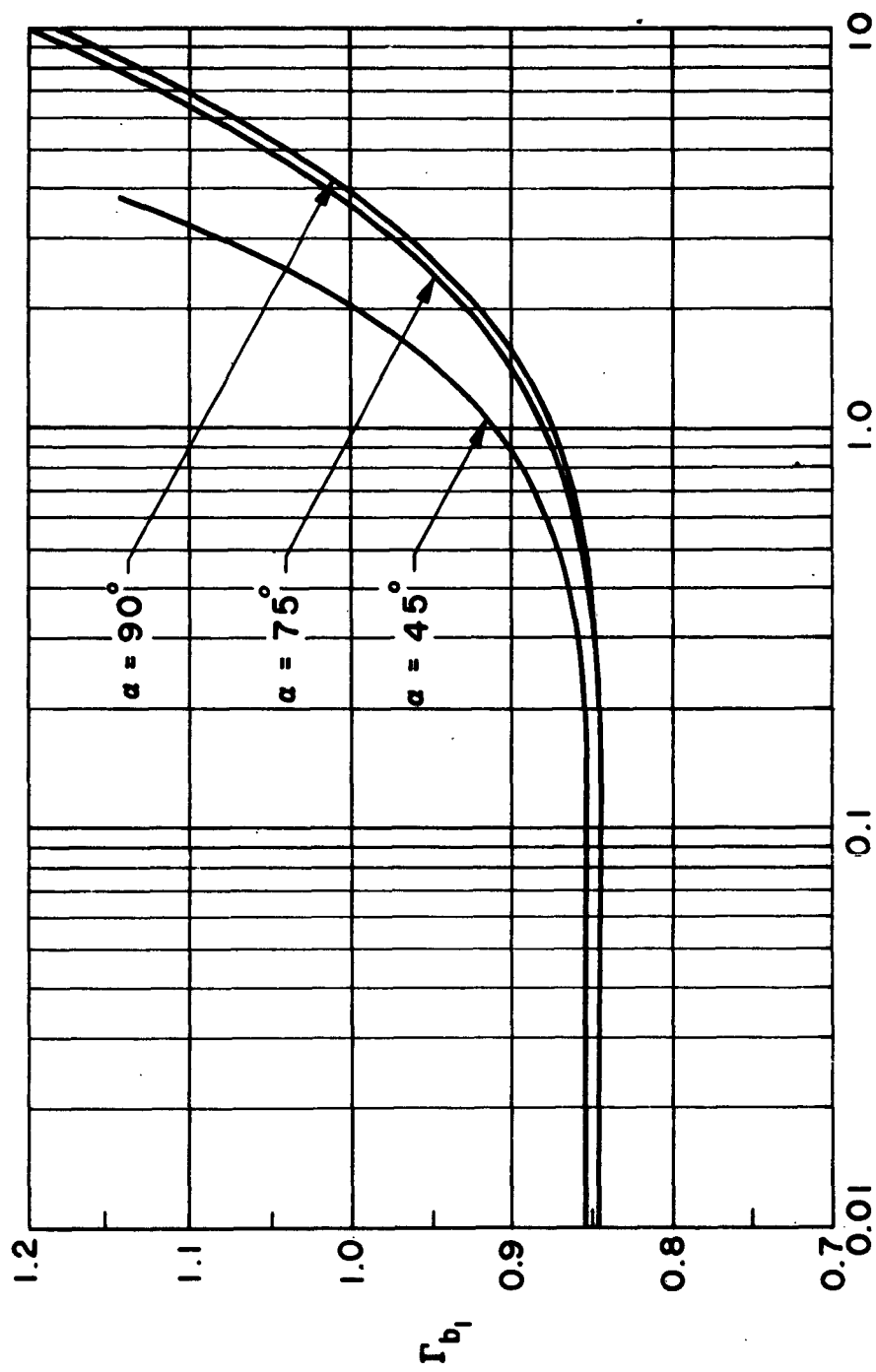


Figure 8. Normal force coefficients for the sudden acceleration of a flat plate, at an angle of attack α , in the wake model.

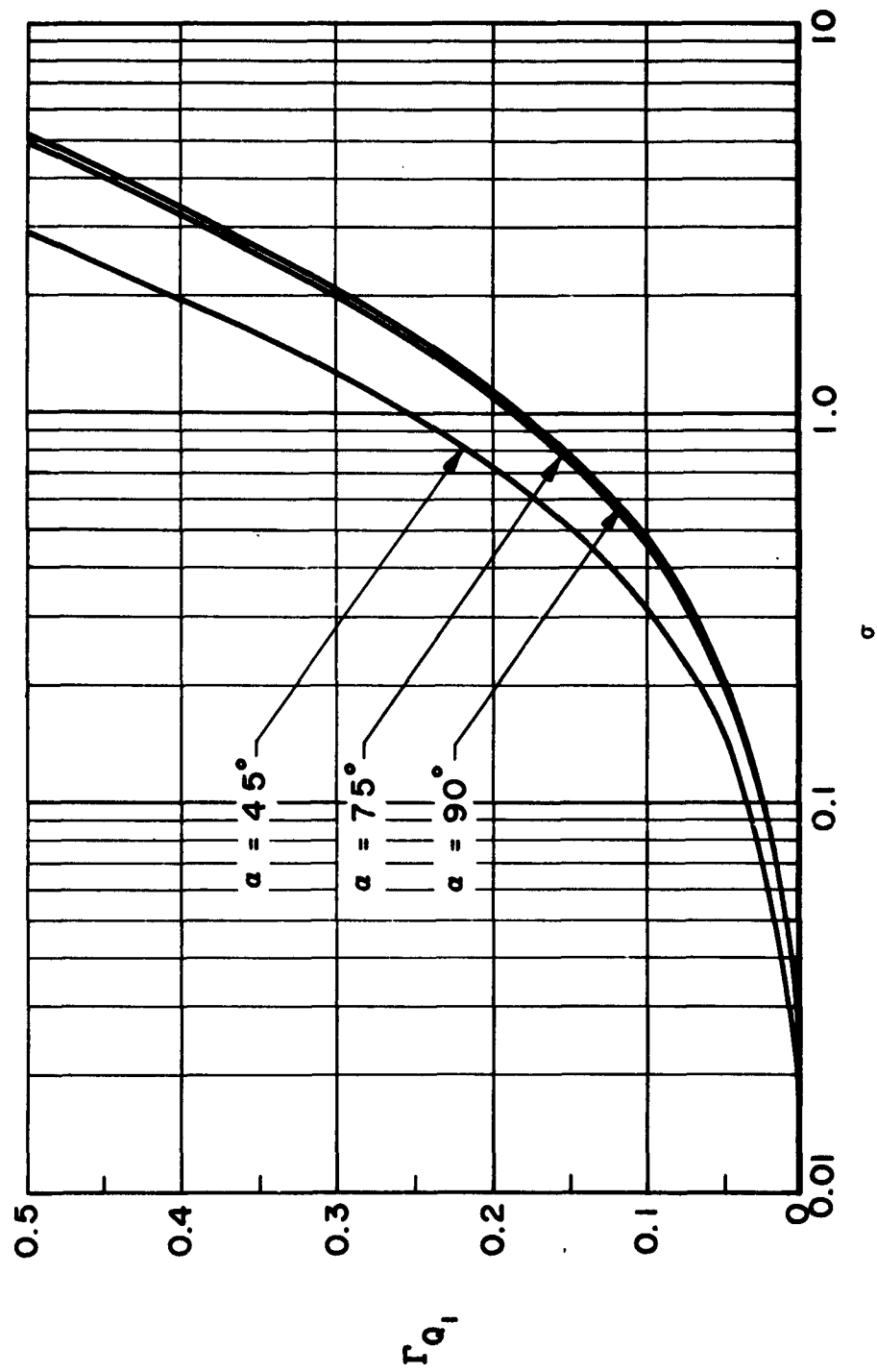


Figure 9. Normal force coefficient due to a source at the point of infinity for a flat plate, at an angle of attack α , in the wake model.

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