

UNCLASSIFIED

AD NUMBER
AD473711
NEW LIMITATION CHANGE
TO Approved for public release, distribution unlimited
FROM Distribution authorized to DoD and DoD contractors only; Administrative/Operational Use; 17 SEP 1965. Other requests shall be referred to Naval Ordnance Lab., White Oak, MD.
AUTHORITY
NOL ltr, 29 Aug 1974

THIS PAGE IS UNCLASSIFIED

J. Ayres
328

A 473711

NOLTR 65-74

ON THE SCALING OF PRESSURES FROM
NUCLEAR EXPLOSIONS WITH SOME
OBSERVATIONS ON THE VALIDITY OF THE
POINT-SOURCE SOLUTION

NOL

17 SEPTEMBER 1965

UNITED STATES NAVAL ORDNANCE LABORATORY, WHITE OAK, MARYLAND

NOLTR 65-74

20061031472

NOTICE

Requests for additional copies by Agencies of the Department of Defense, their contractors, and other Government agencies should be directed to:

Defense Documentation Center (DDC)
Cameron Station
Alexandria, Virginia

Department of Defense contractors who have established DDC services or have their 'need-to-know' certified by the cognizant military agency of their project or contract should also request copies from DDC.

ON THE SCALING OF PRESSURES FROM NUCLEAR EXPLOSIONS WITH SOME
OBSERVATIONS ON THE VALIDITY OF THE POINT-SOURCE SOLUTION

Prepared by:

M. Lutzky

D. Lehto

ABSTRACT: This report discusses the scaling of pressure-distance curves for a particular model of a nuclear explosion using an ideal-gas equation of state. A family of non-dimensional curves is presented which allows scaling to different ambient conditions for a given energy source. These curves also make possible scaling close-in to the initial fireball. In addition, it is shown that the theoretical point-source solution for an ideal gas agrees with the computed solutions, except at close-in distances and at shock strengths below 50. It is shown that neither of these ideal-gas solutions agrees as well with experiment as does a real-gas solution.

PUBLISHED NOVEMBER 1965

NOLTR 65-74

17 September 1965

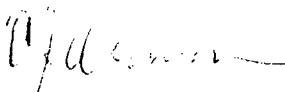
ON THE SCALING OF PRESSURES FROM NUCLEAR EXPLOSIONS WITH SOME
OBSERVATIONS ON THE VALIDITY OF THE POINT-SOURCE SOLUTION

Certain concepts in explosion investigations have been especially valuable because they make generalizations or extensions possible. Two such concepts are Sachs scaling and the point-source solution. Sachs scaling permits extension of one set of explosion results for a given situation to new situations. The point-source explosion model produces an analytic solution (as opposed to the numerical solutions necessary for solution of the usual hydrodynamic equations) which makes examination and generalizations of the behavior of the explosion parameters possible.

The authors of this report examine these two concepts and, by comparison with numerical results of the hydrodynamic equations, suggest the limitations that must be placed upon use of scaling and of point-source solutions for blast from nuclear explosions.

Support for this study was provided by the Defense Atomic Support Agency through NWER Subtask 01.003 (NOL-181).

J. A. DARE
Captain, USN
Commander


C. J. ARONSON
By direction

CONTENTS

	Page
1. INTRODUCTION	1
2. DISCUSSION.....	2
2.1 Nuclear Explosion Model.....	2
2.2 Scaling of Nuclear Explosions.....	2
2.3 Dimensionless Pressure-Distance Curves.....	4
2.4 Comparison with Real-Gas Calculations and Experiment...	6
3. CONCLUDING REMARKS.....	7
4. REFERENCES.....	8
APPENDIX A - Initial Pressure Outside the Fireball.....	A-1

ILLUSTRATIONS

Figure	Title
1	Π Versus λ for Various Values of ζ , High- Π Range
2	Π Versus λ , Low- Π Range
3	Real-Air Versus Ideal-Air Pressure-Distance Curve for 1-Kt at Sea Level
A-1	Correction to Beginning of Π Versus λ Curve

1. INTRODUCTION

Scaling procedures have long been used to extend the range of applicability of both numerical calculations and experimental observations. In particular, "Sachs scaling"¹ has been very useful in the treatment of blast problems, both for high explosives and for nuclear weapons. However, it is well known that Sachs scaling has limitations in the region close to the energy source, since the only property of the source taken into account is the total energy release. This has the consequence that in a rigorous application of Sachs scaling the parameters describing the energy source cannot be varied at will. For example, it will be shown that when an explosion is Sachs scaled to different ambient conditions, it is required that the parameters describing the source be changed correspondingly. This is not the usual experimental situation, where it is generally desired to change the ambient conditions without changing the source.

In this report we shall discuss the scaling of pressure-distance curves for a particular model of a nuclear explosion using an ideal gas equation of state. A family of non-dimensional curves will be presented which allows scaling to different ambient conditions for a given energy source. These curves also make it possible to apply scaling close-in to the initial fireball.

In addition, it will be shown that the theoretical point source solution²⁻⁵ for a shock of infinite strength in an ideal gas agrees with our computed solution (which does not assume a point source) over a wide range of scaled distance. However, neither the theoretical point-source solution nor our computed ideal-gas solution agrees with experimental data as well as do real-gas solutions.

2. DISCUSSION

2.1 Nuclear Explosion Model

A nuclear explosion in air is an extremely complex phenomenon, and many simplifying assumptions are necessary before it is possible to attempt any sort of analytic treatment. One approach is to assume that the explosion deposits an amount of radiative energy within a sphere of air in such a way that the temperature and pressure are increased in this sphere while the density remains at the ambient value. This heated sphere of air, at uniform temperature and pressure, then expands, creating a shock which propagates into the atmosphere.

Some further simplifications allow the numerical computation of the subsequent time evolution by means of a one-dimensional hydrodynamic computer code. This code considers only pure hydrodynamics, using an artificial viscosity to spread the shock over a few spatial zones. Such a numerical calculation has been carried out for a 1-kiloton explosion at sea level, using a real-gas equation of state for air.⁶ The computed results for the airshock agree to within 5 per cent with experimental results,⁶ for nuclear explosions, which shows that the model is useful for sea-level phenomena.

At this stage, it is natural to ask whether this type of calculation may be scaled to different ambient pressures. It turns out that a useful method of scaling does exist if the air is treated as an ideal gas. The method which we describe in the next section is a generalization of Sachs scaling for an ideal gas and allows scaling close-in to the fireball.

2.2 Scaling of Nuclear Explosions

Consider an ideal gas in which an amount of internal energy E is uniformly distributed throughout a sphere of radius a . The density and pressure in the ambient atmosphere (outside the energized sphere) are taken to be ρ_0 and p_0 , respectively. The density within the sphere is taken to be ρ_0 , and the particle velocity is initially zero everywhere. The origin of coordinates is taken at the center of the sphere, and r measures the radial distance from this origin. Then the following quantities are sufficient to describe the phenomena: a , E , p_0 , ρ_0 , r , t , γ , where t = time, and γ = ratio of specific heats for the ideal gas. We may form the set of dimensionless quantities:

$$\lambda = r / (E / p_0)^{1/3}$$

$$\tau = \frac{t p_0^{3/2}}{E^{1/2} \rho_0^{1/2}} \quad (1)$$

$$\zeta = a/(E/p_0)^{1/2}$$

(Note that, because of the ideal gas assumption, we have $\gamma = \frac{p_0}{\rho_0 e_0} + 1$, where e_0 = internal energy per unit mass for ambient atmosphere.) It is not difficult to show that any other dimensionless parameters which may be formed from the available quantities a , E , p_0 , ρ_0 , r , t , γ must be functions of λ , τ , ζ , and γ . We may thus take this set of dimensionless parameters as a fundamental set for the description of the solution to our problem. λ and τ are the dimensionless independent variables, while ζ and γ are dimensionless constants.

Any dependent variable which describes some aspect of the evolution of the physical system can be non-dimensionalized and expressed in terms of our fundamental set. For instance, if p_s is the pressure at the shock front, we may put

$$\Pi = \frac{p_s}{p_0} = F_1(\lambda, \tau, \zeta, \gamma) \quad (2)$$

where F_1 is an unknown function which can only be obtained by solving the partial differential equations describing the problem. Similarly, the shock path equation must be representable in the form

$$F_2(\lambda, \tau, \zeta, \gamma) = 0 \quad (3)$$

We may obtain a dimensionless pressure-distance curve (pressure at the shock front as a function of shock position) by eliminating τ between (2) and (3). The result is an expression of the form

$$\Pi = F_3(\lambda, \zeta, \gamma) \quad (4)$$

If the function $F_3(\lambda, \zeta, \gamma)$ could be somehow determined, a set of dimensionless pressure-distance curves could be generated, each curve being a plot of equation (4) for different values of ζ and γ . Each curve could then be used to predict shock pressure as a function of

shock radius, subject to certain restrictions. That is, a given curve, labeled by $\zeta = \zeta_0$ and $\gamma = \gamma_0$, would not yield pressure versus distance for any arbitrary combination of γ , initial radius, ambient pressure, and explosion energy. It could be used only for values of a , E , p_0 , ρ_0 , ϵ_0 which satisfy $\frac{a}{(E/p_0)^{1/\gamma}} = \zeta_0$ and $\frac{p_0}{\rho_0 \epsilon_0} + 1 = \gamma_0$.

This gives rise to the situation mentioned in the introduction, where it was stated that the scaling process requires that the parameters describing the source be changed when the phenomenon is scaled to different ambient conditions. Thus, suppose that we have available only a single Π versus λ curve, obtained from equation (4) by letting $\gamma = 1.4$ and letting ζ have some specific value, say ζ_0 . Then $\zeta_0 = \left(\frac{E}{p_0}\right)^{-1/\gamma} a$, and if we wish to scale to a new ambient pressure, say p'_0 , then we can use the same curve only if we also change E and/or a in such a way that $\left(\frac{E'}{p'_0}\right)^{-1/\gamma} a'$ is still equal to ζ_0 . This restriction can be avoided if a set of curves is available for which the associated values of ζ cover the desired range.

2.3 Dimensionless Pressure-Distance Curves

Such a set of curves has been generated using a one-dimensional hydrodynamic code⁷ on an IBM-7090. This code solves the partial differential equations of gas dynamics using finite difference methods, and has been applied to the problem described above for various values of initial fireball radius and ambient pressure. From the complete solutions thus obtained, non-dimensional curves for the shock pressure versus shock radius have been constructed. These curves are plotted in Figures 1 and 2, which show the high pressure region ($\Pi > 100$), and low pressure region ($\Pi < 100$), respectively (see Appendix A). In Figure 1, the shock strength, Π , is plotted against λ , while in Figure 2, the overpressure, $\Pi - 1$ is plotted against λ ; this distinction is made for reasons of clarity in plotting the low pressure region. The ambient atmosphere was assumed to be an ideal gas with $\gamma = 1.4$, so that the curves are plots of equation (4), for various values of ζ , and with $\gamma = 1.4$. (In particular, the curve for $\zeta = 0.0123$ was obtained from a calculation of a 1-kt nuclear explosion at sea level. For this case, the initial radius of the fireball was chosen to be $a = 4.251$ meters.)

It can be seen from Figure 1 that each curve joins onto a single main curve at approximately two fireball radii. A result of this fact is that, as far as distances greater than two fireball radii are concerned, only a single curve is necessary to scale any given pressure-distance calculation to some other ambient pressure. In this range, the curve is independent of the dimensionless variable* ζ , and the scaling used is what

* Actually, this is not strictly true, because of the fact that the second and subsequent shocks join the main shock at times which depend on the initial fireball radius. However, these appear only as minor perturbations to the main pressure-distance curve.

is ordinarily known as Sachs scaling. On the other hand, for scaling the close-in pressure-distance curve to other ambient pressures, it is first necessary to locate the proper dimensionless curve by calculating the appropriate value of ζ . This seems reasonable in view of the fact that at small distances the phenomena of energy transfer to the ambient air and the subsequent shock formation are certainly influenced by the initial fireball size. Therefore, we would not expect the scaling to be independent of a in this region.

It is evident from Figure 1 that the main curve is the envelope of the family of different ζ curves. (The complete main curve can be constructed by including the extrapolated portion, shown on the figure as a dashed line.) This main curve can also be considered to be a limiting curve which is approached by the ζ curve as $\zeta \rightarrow 0$. Since $\zeta \rightarrow 0$ may be achieved by letting $a \rightarrow 0$, while keeping E and p_0 constant, it is then clear that the main curve can be regarded as the pressure-distance curve for a point source of energy E in an atmosphere of ambient pressure p_0 .

This interpretation of the main curve can be verified as follows. It can be shown from Figure 1 that, for $\Pi > 100$, the main curve is represented very well by the analytical formula

$$\Pi = 0.158 \lambda^{-3} \quad (5)$$

The problem of a point source explosion for a shock of infinite strength has been solved analytically by similarity methods^{2,3,4}. For the range of values in Figure 1, the shock strengths are quite high ($10^2 < \Pi < 10^5$), so that it seems reasonable to check equation (5) against these analytical results. The similarity theory yields for the pressure-distance curve the expression⁴

$$p = \frac{8}{25(\gamma + 1)\alpha} E/r^3 \quad (6)$$

where α is a dimensionless quantity which depends only on γ . This quantity has been evaluated by Sedov⁵ for $\gamma = 1.4$, and the result is $\alpha = 0.851$. Thus equation (6) takes the form

$$p = 0.1567 E/r^3 \quad (7)$$

If equation (5) is converted to dimensioned variables, p_0 cancels out and the expression becomes

$$p = 0.158 E/r^3 \quad (8)$$

Comparing (7) and (8), we see that the correspondence between the computed and analytic result is excellent. Thus, for $\Pi > 100$, the main curve is identical with the point source, infinite shock strength, pressure-distance curve.

For lower shock strengths, we can expect that the main curve will deviate from the analytic pressure-distance curve given by (7). This is because (7) is derived under the assumption of infinite shock strength, so that the similarity solution is no longer correct when the shock pressure becomes sufficiently small. In fact, referring to Figure 2, we see that the two curves begin to deviate at $\Pi \approx 50$.

2.4 Comparison with Real-Gas Calculations and Experiment

In the previous sections, we have discussed pressure scaling for a particular model of a nuclear explosion. In this model, an ideal-gas equation of state was used for air, with $\gamma = 1.4$. Since it is known⁶ that computations of this model with a real-gas equation of state agree closely (at sea level) with experiment, it is of interest to compare the ideal-gas result with the real-gas result. Figure 3 presents a comparison of these two treatments for the particular case of a 1-kiloton explosion at sea-level conditions. The ideal-gas result, for $\gamma = 1.4$, is obtained from Figures 1 and 2, using $\zeta = 0.0123$. (As we have seen, this curve coincides with the infinite-shock-strength, point-source solution for an ideal gas at distances greater than two fireball radii, and at shock strengths greater than 50.) An ideal-gas result has also been computed for $\gamma = 1.2$, which is a rough limit for γ at high pressures. The comparison then shows that the real-gas calculation agrees with experiment much better than does either of the ideal-gas calculations. In fact, it is clear that the ideal-gas calculation will not correspond with experiment for any value of γ , since the slopes of the ideal-gas curves differ considerably from the experimental and real-gas slopes.

It would therefore be logical to attempt to devise a scaling technique (analogous to the one previously described) for the same model of a nuclear explosion, but with the real-gas equation of state. Unfortunately, more sophisticated equations of state introduce new dimensional quantities which complicate the analysis considerably.

3. CONCLUDING REMARKS

In this report, we have reviewed the use of dimensional analysis to scale pressure-distance curves for a particular model of a nuclear explosion. The following conclusions may be stated:

3.1 It is possible to scale pressure-distance curves for the ideal gas by means of a family of non-dimensional Π versus λ curves, each curve corresponding to a particular value of ζ .

3.2 For distances larger than twice the fireball radius all of the ζ curves coincide, so that in this range only one non-dimensional curve is necessary.

3.3 Within two fireball radii one must first find the proper dimensionless curve to use by determining the relevant value* of the parameter ζ .

3.4 The envelope of the family of ζ -curves coincides with the point-source curve except for distances less than two fireball radii and for shock strengths less than 50.

3.5 All of the calculations described above apply for an ideal gas. A computer calculation made for a 1-kt explosion at sea level conditions, using a real-gas equation of state, gives better agreement with experiment than either the point-source solution or the ζ -curve solution.

* Any reasonable physical situation will usually have a value of ζ between zero and 0.05. Even conventional explosives have ζ values in this range: for pentolite at sea level, $\zeta = 0.014$. The particular explosion model discussed in this report, however, does not apply to high explosives, e.g., see reference 8.

REFERENCES

1. R. G. Sachs, "The Dependence of Blast on Ambient Pressure and Temperature," BRL Report No. 466, May 1944, Unclassified
2. J. Lockwood Taylor, "An Exact Solution of the Spherical Blast Wave Problem," Phil. Mag. 46, 317, March 1955
3. R. Latter, "Similarity Solution for a Spherical Shock Wave," J. Applied Physics 26, No. 8, August 1955
4. L. I. Sedov, "Similarity and Dimensional Methods in Mechanics," Academic Press, 1959, p. 217
5. Loc. cit., p. 231
- 6a L. Rudlin, M. Lutzky and D. Lehto, "Hydrodynamic Calculations of the Shockwave from Nuclear Explosions at Sea-Level Altitude of Burst," U. S. Naval Ordnance Laboratory TR 62-152, 14 September 1962, Unclassified
- 6b D. Lehto, "Revised Calculations of the Shockwave from a 1-Kt Nuclear Explosion at Sea-Level Altitude of Burst," U. S. Naval Ordnance Laboratory TR 65-75, to be published, Unclassified
7. D. Lehto and M. Lutzky, "One-Dimensional Hydrodynamic Code for Nuclear Explosion Calculations," U. S. Naval Ordnance Laboratory TR 62-168, 1 March 1965, Unclassified
8. M. Lutzky, "Theoretical vs Experimental Results for Air Blast from 1-lb Spherical TNT and Pentolite Charges at Sea Level Conditions," U. S. Naval Ordnance Laboratory TR 65-57, to be published, Unclassified

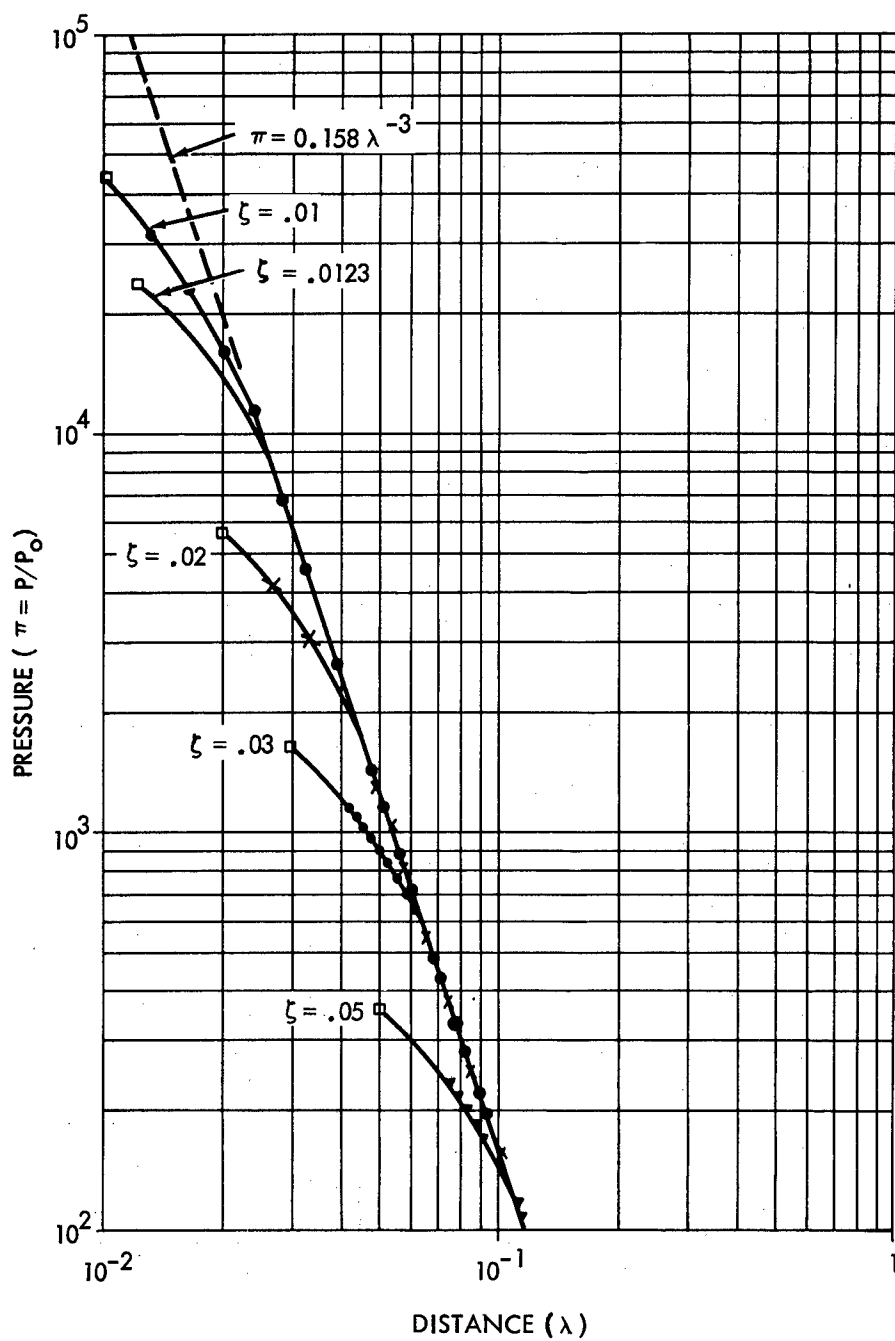


FIG. 1 π VERSUS λ FOR VARIOUS VALUES OF ζ , HIGH - π RANGE

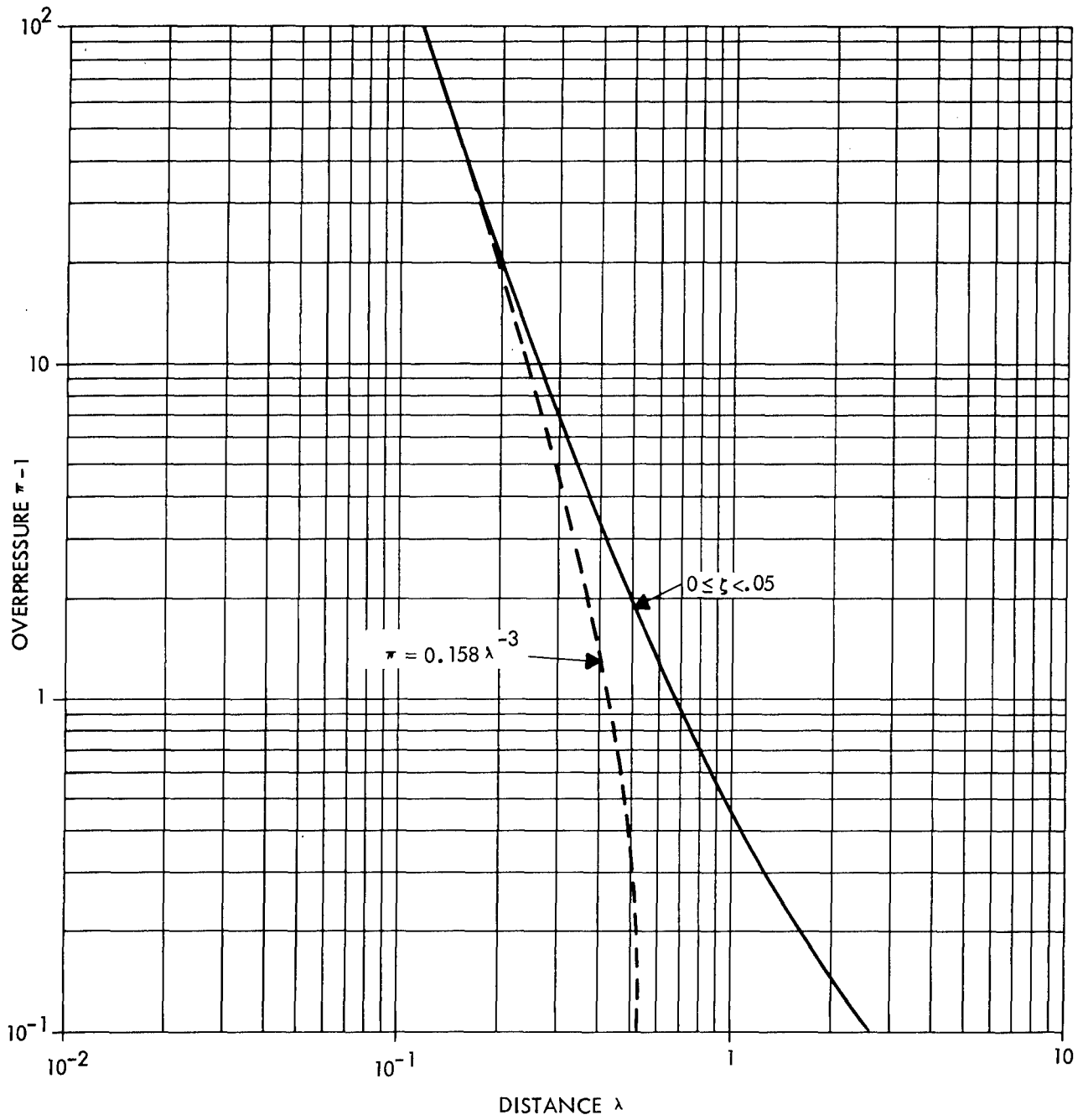


FIG. 2 π VERSUS λ , LOW- π RANGE

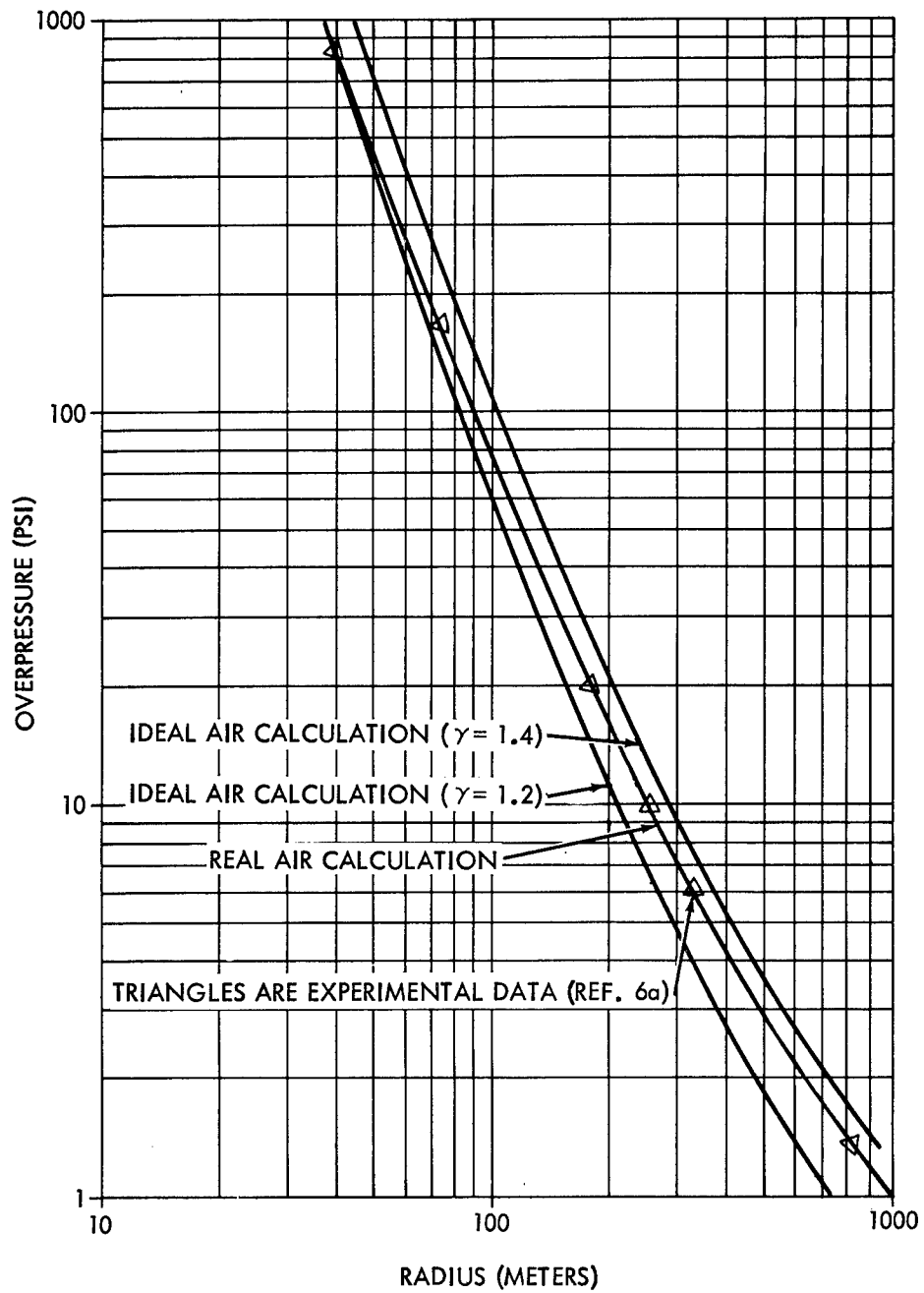


FIG. 3 REAL-AIR VS. IDEAL-AIR PRESSURE-DISTANCE CURVE FOR 1-KT AT SEA LEVEL

APPENDIX A

Initial Pressure Outside the Fireball

Figure A-1 shows some of the computed data points for the case $\zeta = 0.0123$. There is a small gap (labeled "corrected region") between the initial periphery of the fireball and the first meaningful shockfront data because of the finite zone size used in the calculations. This gap can be made smaller by using smaller zones in the problem, but this makes the computer running time excessively long. Instead, we derive theoretically the initial shockfront pressure and extend the curve through the computed data to this derived initial point.

The initial stages of airshock formation at the periphery of the fireball should be very much like those in a linear shock tube, because the interaction region is initially thin compared to the fireball radius. For a linear shock tube of reservoir conditions

P_1 = reservoir (fireball) pressure

c_1 = reservoir sound speed

γ_1 = reservoir specific heat ratio

and similarly defined ambient conditions P_0 , c_0 , γ_0 , a shockfront with pressure P will form. These quantities are related by the equation*

$$\left(\frac{P}{P_0} \frac{P_0}{P_1}\right)^{\frac{\gamma_1 - 1}{2\gamma_1}} = 1 - \frac{c_0}{c_1} (\gamma_1 - 1) \left(\frac{P}{P_0} - 1\right) \left[2\gamma_0 \left\{\gamma_0 - 1 + (\gamma_0 + 1) \frac{P}{P_0}\right\}\right]^{\frac{1}{2}}. \quad (A-1)$$

The nuclear fireball situation differs from the usual shock tube at this point in that the density is equal on both sides of the interface rather than the temperature. For the fireball model being considered here,

$$\gamma_0 = \gamma_1 = 1.4$$

$$\rho_0 = \rho_1$$

$$\frac{c_0}{c_1} = \left(\frac{P_0}{P_1}\right)^{\frac{1}{2}}.$$

Equation (A-1) now becomes

* W. Bleakney and A. H. Taub, "Interaction of Shock Waves," Rev. Mod. Phys. 21, Oct 1949, p. 590.

$$\left(\frac{P}{P_0} \frac{P_0}{P_1}\right)^{\frac{1}{2}} = 1 - \left(\frac{P_0}{P_1}\right)^{\frac{1}{2}} \left(\frac{P}{P_0} - 1\right) \left[7 \left(1 + 6 \frac{P}{P_0}\right)\right]^{-\frac{1}{2}} \quad (\text{A-2})$$

For the high-pressure region where $P/P_0 \gg 1$,

$$\left(\frac{P}{P_1}\right)^{\frac{1}{2}} \approx 1 - \left(\frac{P}{P_1}\right)^{\frac{1}{2}} \frac{1}{\sqrt{42}} \quad (\text{A-3})$$

the solution of which is $P/P_1 = 0.464$. Thus, for large initial shock strengths the initial pressure in the airshock is 0.464 times the initial pressure in the fireball. This is the point to which the calculated data are extended in Figure A-1 to fill in the "corrected region."

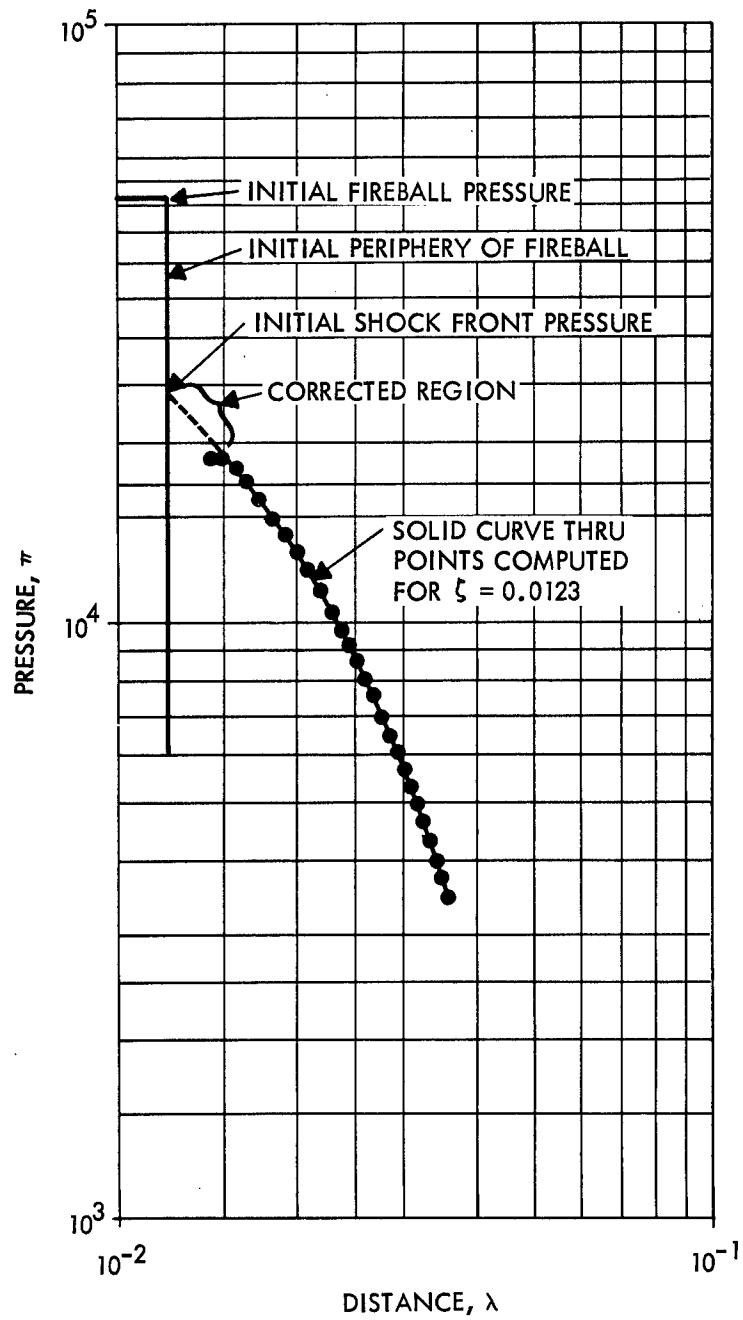


FIG. A-1 CORRECTION TO BEGINNING OF π VERSUS λ CURVE

NAVY

CHIEF OF NAVAL OPERATIONS, ND
ATTN. OP-75
ATTN. OP-03EG
WASHINGTON, D. C. 20350

2
1

DIRECTOR OF NAVAL INTELLIGENCE, ND
ATTN. OP-922V
WASHINGTON, D. C. 20390

1

CHIEF
BUREAU OF NAVAL WEAPONS, ND
ATTN. DLI-3
ATTN. RUME-11, RMMO-611, RRRE-5
ATTN. RRNU
ATTN. RRRE-51
WASHINGTON, D. C. 20360

4
3
1
1

CHIEF
BUREAU OF SHIPS, ND
ATTN. CODE 372, CODE 423
WASHINGTON, D. C. 20390

2

CHIEF
BUREAU OF YARDS AND DOCKS, ND
ATTN. CODE D-400, CODE D-440
WASHINGTON, D. C. 20390

2

CHIEF OF NAVAL RESEARCH, ND
ATTN. CODE 811
ATTN. CODE 493, CODE 418
WASHINGTON, D. C. 20390

2
2

COMMANDER-IN-CHIEF
U. S. PACIFIC FLEET, FPO
SAN FRANCISCO, CALIFORNIA 96601

1

COMMANDER-IN-CHIEF
U. S. ATLANTIC FLEET
U. S. NAVAL BASE
NORFOLK, VIRGINIA 23500

1

COMMANDANT OF THE MARINE CORPS, ND
ATTN. CODE A03H
WASHINGTON, D. C. 20390

4

PRESIDENT U. S. NAVAL WAR COLLEGE NEWPORT, RHODE ISLAND 02840	1
SUPERINTENDENT U. S. NAVAL POSTGRADUATE SCHOOL MONTEREY, CALIFORNIA 93940	1
COMMANDING OFFICER NUCLEAR WEAPONS TRAINING CENTER, ATLANTIC ATTN. NUCLEAR WARFARE DEPARTMENT NAVAL BASE NORFOLK, VIRGINIA 23500	1
COMMANDING OFFICER U. S. NAVAL SCHOOLS COMMAND U. S. NAVAL STATION TREASURE ISLAND SAN FRANCISCO, CALIFORNIA 94100	1
COMMANDING OFFICER NUCLEAR WEAPONS TRAINING CENTER, PACIFIC NAVAL STATION NORTH ISLAND SAN DIEGO, CALIFORNIA 92100	2
COMMANDING OFFICER U. S. NAVAL DAMAGE CONTROL TRAINING CENTER ATTN. ABC DEFENSE COURSE NAVAL BASE PHILADELPHIA, PENNSYLVANIA 19100	1
COMMANDER U. S. NAVAL ORDNANCE TEST STATION CHINA LAKE, CALIFORNIA 96105	1
COMMANDING OFFICER AND DIRECTOR U. S. NAVAL CIVIL ENGINEERING LABORATORY ATTN. CODE L31 PORT HUENEME, CALIFORNIA 93041	1
DIRECTOR U. S. NAVAL RESEARCH LABORATORY WASHINGTON, D. C. 20390	1

COMMANDING OFFICER AND DIRECTOR
NAVY ELECTRONICS LABORATORY
SAN DIEGO, CALIFORNIA 92100

1

COMMANDING OFFICER
U. S. NAVAL RADIOLOGICAL DEFENSE LABORATORY
ATTN. TECH INFO DIV
SAN FRANCISCO, CALIFORNIA 94100

1

COMMANDING OFFICER AND DIRECTOR
DAVID W. TAYLOR MODEL BASIN
ATTN. LIBRARY
WASHINGTON, D. C. 20007

1

UNDERWATER EXPLOSIONS RESEARCH DIVISION
DAVID W. TAYLOR MODEL BASIN
PORTSMOUTH, VIRGINIA 23700

1

DIRECTOR
SPECIAL PROJECTS OFFICE, ND
ATTN. SP-2721
ATTN. SP-2722
WASHINGTON, D. C. 20360

1

1

OFFICER-IN-CHARGE
U. S. NAVAL SCHOOL
CIVIL ENGINEERING CORPS OFFICERS
U. S. NAVAL CONSTRUCTION BATTALION
PORT HUENEME, CALIFORNIA 93041

1

ARMY

	COPIES
CHIEF OF RESEARCH AND DEVELOPMENT, D/A ATTN. ATOMIC DIVISION WASHINGTON, D. C. 20310	1
CHIEF OF ENGINEERS, D/A ATTN. ENGCW-NE, ENGTE-E, ENGM-C-E WASHINGTON, D. C. 20390	3
COMMANDING GENERAL U. S. ARMY MATERIEL COMMAND ATTN. AMCRD-DE-N WASHINGTON, D. C. 20390	2
COMMANDING GENERAL U. S. CONTINENTAL ARMY COMMAND FT. MONROE, VIRGINIA 23351	1
PRESIDENT U. S. ARMY AIR DEFENSE BOARD FT. BLISS, TEXAS 79906	1
COMMANDANT COMMAND AND GENERAL STAFF COLLEGE ATTN. ARCHIVES FT. LEAVENWORTH, KANSAS 66027	1
COMMANDANT U. S. ARMY AIR DEFENSE SCHOOL ATTN. COMMAND AND STAFF DEPT FT. BLISS, TEXAS 79906	1
DIRECTOR SPECIAL WEAPONS DEVELOPMENT HQ CDC ATTN. CHESTER I. PETERSON FT. BLISS, TEXAS 79906	1
COMMANDING GENERAL ABERDEEN PROVING GROUND ATTN. BRL FOR DIRECTOR, J. J. MESZAROS E. O. BAICY, R. E. SHEAR ABERDEEN, MARYLAND 21005	2 2

COMMANDING GENERAL
THE ENGINEER CENTER
ATTN. ASST COMMANDANT, ENGINEER SCHOOL
FT BELVOIR, VIRGINIA 22060

1

DIRECTOR
U. S. ARMY RESEARCH AND DEVELOPMENT LABORATORY
ATTN. CHIEF, TECH SUPPORT BRANCH
FT BELVOIR, VIRGINIA 22060

1

COMMANDING OFFICER
PICATINNY ARSENAL
ATTN. SMUPA-G,-W,-VL,-VE,-VC,-DD,-DR,-DR-4,-DW,-TX,-TW,-V 12
DOVER, NEW JERSEY 07801

COMMANDING GENERAL
USA MISSILE COMMAND
HUNTSVILLE, ALABAMA 35801

1

DIRECTOR
U. S. ARMY CORPS OF ENGINEERS
WATERWAYS EXPERIMENT STATION
ATTN. LIBRARY
VICKSBURG, MISSISSIPPI 39180

1

COMMANDING GENERAL
WHITE SANDS MISSILE RANGE
ATTN. STEWS-AMTED-E
NEW MEXICO 88002

1

COMMANDANT
ARMY WAR COLLEGE
ATTN. LIBRARY
CARLISLE BARRACKS, PENNSYLVANIA 17013

1

AIR FORCE

HEADQUARTERS USAF
ATTN. (AFRDQPO)
WASHINGTON, D. C. 20330

1

DEPUTY CHIEF OF STAFF
PLANS AND PROGRAMS
HQ USAF
ATTN. WAR PLANS DIVISION
WASHINGTON, D. C. 20390

1

DIRECTOR OF RESEARCH AND DEVELOPMENT
DCS/D, HQ USAF
ATTN. GUIDANCE AND WEAPONS DIVISION
WASHINGTON, D. C. 20390

1

AIR FORCE INTELLIGENCE CENTER
HQ USAF, ACS/I (AFCIN-3K2)
WASHINGTON, D. C. 20390

1

COMMANDER-IN-CHIEF
STRATEGIC AIR COMMAND
ATTN. OAWS
OFFUTT AFB
NEBRASKA 68113

1

COMMANDER
TACTICAL AIR COMMAND
ATTN. DOCUMENT SECURITY BRANCH
LANGLEY AFB
VIRGINIA 23365

1

COMMANDER
AIR FORCE LOGISTICS COMMAND
WRIGHT-PATTERSON AFB
OHIO 45433

2

AFSC
ATTN. RDRWA
ANDREWS AIR FORCE BASE
WASHINGTON, D. C. 20331

1

DIRECTOR
AIR UNIVERSITY LIBRARY
MAXWELL AFB
ALABAMA 36112

2

AFCL
L. G. HANSCOM FIELD
ATTN. CRQST-2, H. P. GAUVIN
BEDFORD, MASSACHUSETTS 01731

2

AFWL
ATTN. CAPT. C.M. GILLESPIE
KIRTLAND AFB
NEW MEXICO 87117

4

COMMANDANT
INSTITUTE OF TECHNOLOGY
ATTN. MCLI-ITRIDL
WRIGHT-PATTERSON AFB
OHIO 45433

1

BSD
NORTON AFB
CALIFORNIA 92409

1

DIRECTOR
USAF PROJECT RAND
VIA U. S. AIR FORCE LIAISON OFFICE
THE RAND CORPORATION
ATTN. DR. H. L. BRODE
1700 MAIN STREET
SANTA MONICA, CALIFORNIA 90400

1

DIRECTOR OF CIVIL ENGINEERING
HQ USAF
ATTN. AFOCE
WASHINGTON, D. C. 20390

1

AFFDL
ATTN. ASRMDS-1, ASRMDS-12, LIBRARY
WRIGHT-PATTERSON AFB
OHIO 45433

3

OTHERS

DIRECTOR OF DEFENSE RESEARCH AND ENGINEERING ATTN. TECH LIBRARY WASHINGTON, D. C. 20390	1
ASSISTANT TO THE SECRETARY OF DEFENSE(ATOMIC ENERGY) WASHINGTON,D.C. 20330	1
U. S. DOCUMENTS OFFICER OFFICE OF THE UNITED STATES NATIONAL MILITARY REPRESENTATIVE, SHAPE APO 55 NEW YORK, N. Y. 10000	1
COMMANDER-IN-CHIEF, PACIFIC FLEET POST OFFICE SAN FRANCISCO, CALIFORNIA 96601	1
DIRECTOR WEAPONS SYSTEMS EVALUATION GROUP, OSD ROOM 1E880 THE PENTAGON WASHINGTON, D. C. 20390	1
COMMANDANT ARMED FORCES STAFF COLLEGE ATTN. LIBRARY NORFOLK, VIRGINIA 23500	1
CHIEF WEAPONS TEST DIVISION DEFENSE ATOMIC SUPPORT AGENCY ATTN. FCWT, FCTG, SANDIA BASE ALBUQUERQUE, NEW MEXICO 87115	1 2
DIRECTOR DEFENSE ATOMIC SUPPORT AGENCY WASHINGTON, D. C. 20301	5

COMMANDANT
NATIONAL WAR COLLEGE
ATTN. CLASS REC. LIBRARY
WASHINGTON, D. C. 20390

1

COMMANDANT
THE INDUSTRIAL COLLEGE OF THE ARMED FORCES
FT MCNAIR
WASHINGTON, D. C. 20390

1

LOS ALAMOS SCIENTIFIC LABORATORY
ATTN. DR. A. C. GRAVES
DR. E. W. BENNETT
DR. CONRAD LONGMIRE
P. O. BOX 1663
LOS ALAMOS, NEW MEXICO 87544

1

1

1

SCIENTIFIC AND TECHNICAL INFORMATION FACILITY
ATTN. NASA REPRESENTATIVE
P. O. BOX 5700
BETHESDA, MARYLAND 20000

1

LANGLEY RESEARCH CENTER, NASA
ATTN. MR. PHILIP DONELY
LANGLEY FIELD
HAMPTON, VIRGINIA 23365

1

AERONUTRONIC
ATTN. DR. R. A. GRANDEY
(CONTRACT NO. 60921-7066)
FORD ROAD
NEWPORT BEACH, CALIFORNIA 92660

1

UNIVERSITY OF CALIFORNIA
E. O. LAWRENCE RADIATION LABORATORY
ATTN. TECH INFO DIV
DR. G. F. BING
P. O. BOX 808
LIVERMORE, CALIFORNIA 94550

2

1

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
MAN-SPACECRAFT CENTER
SPACE TECHNOLOGY DIVISION
ATTN. MR. R. F. FLETCHER
BOX 1537
HOUSTON, TEXAS 77000

1

EDGERTON, GERMESHAUSEN, AND GRIER, INC.
ATTN. MR. D. F. HANSEN
160 BROOKLINE AVENUE
BOSTON, MASSACHUSETTS 02100

1

TEMPO
GENERAL ELECTRIC COMPANY
ATTN. DR. W. E. DUDZIAK
(CONTRACT NO. DA-49-146-XZ-109)
735 STATE STREET
P. O. DRAWER Q-Q
SANTA BARBARA, CALIFORNIA 93100

1

DASA DATA CENTER
GENERAL ELECTRIC COMPANY
ATTN. MR. WARREN CHAN
(CONTRACT NO. DA-49-146-XZ-038)
735 STATE STREET
SANTA BARBARA, CALIFORNIA 93100

1

CONVAIR (ASTRONAUTICS) DIVISION
GENERAL DYNAMICS CORPORATION
SPACE SCIENCE LABORATORY
ATTN. DR. D. A. HAMLIN
(CONTRACT NO. DA-49-149-XZ-169)
P. O. BOX 1128
SAN DIEGO, CALIFORNIA 92100

1

BATTELLE MEMORIAL INSTITUTE
ATTN. MR. R. W. KLINGENSMITH
(CONTRACT NO. SD-80)
505 KING AVENUE
COLUMBUS, OHIO 43200
VIA. ADVANCED RESEARCH PROJECTS AGENCY
THE PENTAGON
WASHINGTON, D. C. 20390

1

TECHNICAL OPERATIONS, INC.
ATTN. DR. PAUL I. RICHARDS
(SERVICE PROJ. 5010.21.83027)
BURLINGTON, MASSACHUSETTS 01801
VIA. HEADQUARTERS
ATTN. RD-DE-N
ARMY MATERIEL COMMAND
BLDG T-7
GRAVELLY POINT, VIRGINIA 20390

1

AVCO/RAD
ATTN. MR. R. E. COOPER
201 LOWELL STREET
WILMINGTON, MASSACHUSETTS 01887
VIA. BSD
NORTON AFB, CALIFORNIA 92409

1

SPACE TECHNOLOGY LABORATORIES
SPACE TECHNOLOGY CENTER
ATTN. MR. M. ROSEN
ONE SPACE PARK
REDONDO BEACH, CALIFORNIA 90277
VIA. AFWL
KIRTLAND AFB, NEW MEXICO 87117

1

AEROSPACE CORPORATION
SAN BERNARDINO OPERATIONS
ATTN. MR. DAVID B. SINGER, MR. DALE ANDERSON
1111 EAST MILL STREET
SAN BERNARDINO, CALIFORNIA 92400
VIA. BSD
ATTN. CAPT H.C. TANNER
NORTON AFB, CALIFORNIA 92409

2

LOCKHEED MISSILES AND SPACE CO.
ATTN. DR. R. E. MEYEROTT
DR. EUGENE TERRY
PALO ALTO, CALIFORNIA 94300

1

1

AERONUTRONIC
ATTN. MR. ROBERT C. EVANS
(CONTRACT NO. DA-49-146-XZ-202)
FORD ROAD
NEWPORT BEACH, CALIFORNIA 92660

1

CHIEF
CLASSIFIED TECHNICAL LIBRARY
TECHNICAL INFORMATION SERVICE
ATTN. MRS. JEAN O LEARY
U. S. ATOMIC ENERGY COMMISSION
WASHINGTON, D. C. 20390

1

MANAGER
ALBUQUERQUE OPERATIONS OFFICE
U. S. ATOMIC ENERGY COMMISSION
P. O. BOX 5400
ALBUQUERQUE, NEW MEXICO 87100

1

PRESIDENT
SANDIA CORPORATION
ATTN. DR. M. L. MERRITT, J. W. REED
DR. C. BROYLES, DR. T. COOK
LIBRARY
(CONTRACT-AT-(29-1)-789)
SANDIA BASE
ALBUQUERQUE, NEW MEXICO 87115

2
2
1

URS CORPORATION
ATTN. MR. KENNETH KAPLAN
1811 TROUSDALE DRIVE
BURLINGAME, CALIFORNIA 94010

1

SOUTHWEST RESEARCH INSTITUTE
ATTN. DR. ROBERT C. DEHART
P. O. BOX 28281
SAN ANTONIO, TEXAS 78200

1

SPACE TECHNOLOGY LABORATORIES, INC.
ATTN. DR. LEON
DR. BENJAMIN SUSSHOLZ
5500 WEST EL SEGUNDO BLVD.
LOS ANGELES, CALIFORNIA 90000
VIA. BSD
NORTON AFB, CALIFORNIA 94209

1
1

PRESIDENT
KAMAN NUCLEAR
ATTN. DR. FRANK SHELTON
(CONTRACT N60921-7017)
COLORADO SPRINGS, COLORADO 80900

1

PHYSICS INTERNATIONAL COMPANY
ATTN. DR. W. BIRNBAUM
2229 4TH STREET
BERKELEY, CALIFORNIA 94700
VIA. VULNERABILITY BRANCH, FCDR-4
FIELD COMMAND, DASA
SANDIA BASE
ALBUQUERQUE, NEW MEXICO 87115

1

SANDIA CORPORATION
LIVERMORE LABORATORY
ATTN. CLASSIFIED DOCUMENT DIVISION
DR. L. COBB, 9120
DR. LANE, 8118
DR. MESNARD, 8141
P. O. BOX 969
LIVERMORE, CALIFORNIA 94551

1
1
1
1

LOCKHEED MISSILES AND SPACE COMPANY
ATTN. M. G. WADE. D/81-91
P. O. BOX 504
SUNNYVALE, CALIFORNIA 94086
VIA. BUWEP'S REPRESENTATIVE
(SPECIAL PROJECTS OFFICE)
ATTN. SP L-314
LOCKHEED MISSILES AND SPACE COMPANY
P. O. BOX 504
SUNNYVALE, CALIFORNIA 94086

1

DIRECTOR
DEFENSE INTELLIGENCE AGENCY
ATTN. CAPT. D. ANDERSON
WASHINGTON, D. C. 20301

1

SPACE SCIENCES LABORATORY
GENERAL ELECTRIC COMPANY
ATTN. DR. S. M. SCALA
P.O. BOX 8555
PHILADELPHIA, PENNSYLVANIA 19101

1

DDC
ATTN. TISIA-21
CAMERON STATION
ALEXANDRIA, VIRGINIA 20000

20

UNCLASSIFIED

Security Classification

DOCUMENT CONTROL DATA - R&D		
(Security classification of title, body of abstract and indexing annotation must be entered when the overall report is classified)		
1. ORIGINATING ACTIVITY (Corporate author) U. S. Naval Ordnance Laboratory White Oak, Silver Spring, Maryland		2a. REPORT SECURITY CLASSIFICATION Unclassified
		2b. GROUP
3. REPORT TITLE ON THE SCALING OF PRESSURES FROM NUCLEAR EXPLOSIONS WITH SOME OBSERVATIONS ON THE VALIDITY OF THE POINT-SOURCE SOLUTION		
4. DESCRIPTIVE NOTES (Type of report and inclusive dates)		
5. AUTHOR(S) (Last name, first name, initial) Lutzky, Morton Lehto, Delbert L.		
6. REPORT DATE 17 September 1965	7a. TOTAL NO. OF PAGES 14	7b. NO. OF REFS 8
8a. CONTRACT OR GRANT NO.	9a. ORIGINATOR'S REPORT NUMBER(S) NOLTR 65-74	
b. PROJECT NO.		
c. NOL-181	9b. OTHER REPORT NO(S) (Any other numbers that may be assigned this report)	
d.		
10. AVAILABILITY/LIMITATION NOTICES Qualified requesters may obtain copies of this report from DDC.		
11. SUPPLEMENTARY NOTES		12. SPONSORING MILITARY ACTIVITY Defense Atomic Support Agency
13. ABSTRACT This report discusses the scaling of pressure-distance curves for a particular model of a nuclear explosion using an ideal-gas equation of state. A family of non-dimensional curves is presented which allows scaling to different ambient conditions for a given energy source. These curves also make possible scaling close-in to the initial fireball. In addition, it is shown that the theoretical point-source solution for an ideal gas agrees with the computed solutions, except at close-in distances and at shock strengths below 50. It is shown that neither of these ideal-gas solutions agrees as well with experiment as a real-gas solution.		

DD FORM 1473
1 JAN 64

UNCLASSIFIED

Security Classification

UNCLASSIFIED

Security Classification

14. KEY WORDS	LINK A		LINK B		LINK C	
	ROLE	WT	ROLE	WT	ROLE	WT
Explosions, Nuclear Pressure-Distance Curve Scaling Point-Source Solution Similarity Solution Ideal Gas						

INSTRUCTIONS

1. **ORIGINATING ACTIVITY:** Enter the name and address of the contractor, subcontractor, grantee, Department of Defense activity or other organization (*corporate author*) issuing the report.

2a. **REPORT SECURITY CLASSIFICATION:** Enter the overall security classification of the report. Indicate whether "Restricted Data" is included. Marking is to be in accordance with appropriate security regulations.

2b. **GROUP:** Automatic downgrading is specified in DoD Directive 5200.10 and Armed Forces Industrial Manual. Enter the group number. Also, when applicable, show that optional markings have been used for Group 3 and Group 4 as authorized.

3. **REPORT TITLE:** Enter the complete report title in all capital letters. Titles in all cases should be unclassified. If a meaningful title cannot be selected without classification, show title classification in all capitals in parenthesis immediately following the title.

4. **DESCRIPTIVE NOTES:** If appropriate, enter the type of report, e.g., interim, progress, summary, annual, or final. Give the inclusive dates when a specific reporting period is covered.

5. **AUTHOR(S):** Enter the name(s) of author(s) as shown on or in the report. Enter last name, first name, middle initial. If military, show rank and branch of service. The name of the principal author is an absolute minimum requirement.

6. **REPORT DATE:** Enter the date of the report as day, month, year; or month, year. If more than one date appears on the report, use date of publication.

7a. **TOTAL NUMBER OF PAGES:** The total page count should follow normal pagination procedures, i.e., enter the number of pages containing information.

7b. **NUMBER OF REFERENCES:** Enter the total number of references cited in the report.

8a. **CONTRACT OR GRANT NUMBER:** If appropriate, enter the applicable number of the contract or grant under which the report was written.

8b, 8c, & 8d. **PROJECT NUMBER:** Enter the appropriate military department identification, such as project number, subproject number, system numbers, task number, etc.

9a. **ORIGINATOR'S REPORT NUMBER(S):** Enter the official report number by which the document will be identified and controlled by the originating activity. This number must be unique to this report.

9b. **OTHER REPORT NUMBER(S):** If the report has been assigned any other report numbers (*either by the originator or by the sponsor*), also enter this number(s).

10. **AVAILABILITY/LIMITATION NOTICES:** Enter any limitations on further dissemination of the report, other than those imposed by security classification, using standard statements such as:

- (1) "Qualified requesters may obtain copies of this report from DDC."
- (2) "Foreign announcement and dissemination of this report by DDC is not authorized."
- (3) "U. S. Government agencies may obtain copies of this report directly from DDC. Other qualified DDC users shall request through _____."
- (4) "U. S. military agencies may obtain copies of this report directly from DDC. Other qualified users shall request through _____."
- (5) "All distribution of this report is controlled. Qualified DDC users shall request through _____."

If the report has been furnished to the Office of Technical Services, Department of Commerce, for sale to the public, indicate this fact and enter the price, if known.

11. **SUPPLEMENTARY NOTES:** Use for additional explanatory notes.

12. **SPONSORING MILITARY ACTIVITY:** Enter the name of the departmental project office or laboratory sponsoring (*paying for*) the research and development. Include address.

13. **ABSTRACT:** Enter an abstract giving a brief and factual summary of the document indicative of the report, even though it may also appear elsewhere in the body of the technical report. If additional space is required, a continuation sheet shall be attached.

It is highly desirable that the abstract of classified reports be unclassified. Each paragraph of the abstract shall end with an indication of the military security classification of the information in the paragraph, represented as (TS), (S), (C), or (U).

There is no limitation on the length of the abstract. However, the suggested length is from 150 to 225 words.

14. **KEY WORDS:** Key words are technically meaningful terms or short phrases that characterize a report and may be used as index entries for cataloging the report. Key words must be selected so that no security classification is required. Identifiers, such as equipment model designation, trade name, military project code name, geographic location, may be used as key words but will be followed by an indication of technical context. The assignment of links, roles, and weights is optional.

UNCLASSIFIED

Security Classification

Naval Ordnance Laboratory, White Oak, Md. (NOL technical report 65-74) ON THE SCALING OF PRESSURES FROM NUCLEAR EXPLOSIONS WITH SOME OBSERVATIONS ON THE VALIDITY OF THE POINT-SOURCE SOLUTION, by M. Lutzky and D. Lehte. 17 Sept. 1965. 8p. charts, tables. NOL task 181. UNCLASSIFIED A family of non-dimensional curves is presented which allows scaling to different ambient conditions for a given energy source. These curves also make possible scaling close-in to the initial fireball. In addition, it is shown that the theoretical point-source solution for an ideal gas agrees with the computed solutions, except at close-in distances and at shock strengths below 50. It is shown that neither of these idea-gas solutions agrees as well with experiment as a real-gas solution.	1. Explosions, Nuclear - Scaling Title I. Lutzky, II. Merten III. Lehte, Delbert J., jt. author IV. Project Abstract card is unclassified.
Naval Ordnance Laboratory, White Oak, Md. (NOL technical report 65-74) ON THE SCALING OF PRESSURES FROM NUCLEAR EXPLOSIONS WITH SOME OBSERVATIONS ON THE VALIDITY OF THE POINT-SOURCE SOLUTION, by M. Lutzky and D. Lehte. 17 Sept. 1965. 8p. charts, tables. NOL task 181. UNCLASSIFIED A family of non-dimensional curves is presented which allows scaling to different ambient conditions for a given energy source. These curves also make possible scaling close-in to the initial fireball. In addition, it is shown that the theoretical point-source solution for an ideal gas agrees with the computed solutions, except at close-in distances and at shock strengths below 50. It is shown that neither of these idea-gas solutions agrees as well with experiment as a real-gas solution.	1. Explosions, Nuclear - Scaling Title I. Lutzky, II. Merten III. Lehte, Delbert J., jt. author IV. Project Abstract card is unclassified.
Naval Ordnance Laboratory, White Oak, Md. (NOL technical report 65-74) ON THE SCALING OF PRESSURES FROM NUCLEAR EXPLOSIONS WITH SOME OBSERVATIONS ON THE VALIDITY OF THE POINT-SOURCE SOLUTION, by M. Lutzky and D. Lehte. 17 Sept. 1965. 8p. charts, tables. NOL task 181. UNCLASSIFIED A family of non-dimensional curves is presented which allows scaling to different ambient conditions for a given energy source. These curves also make possible scaling close-in to the initial fireball. In addition, it is shown that the theoretical point-source solution for an ideal gas agrees with the computed solutions, except at close-in distances and at shock strengths below 50. It is shown that neither of these idea-gas solutions agrees as well with experiment as a real-gas solution.	1. Explosions, Nuclear - Scaling Title I. Lutzky, II. Merten III. Lehte, Delbert J., jt. author IV. Project Abstract card is unclassified.
Naval Ordnance Laboratory, White Oak, Md. (NOL technical report 65-74) ON THE SCALING OF PRESSURES FROM NUCLEAR EXPLOSIONS WITH SOME OBSERVATIONS ON THE VALIDITY OF THE POINT-SOURCE SOLUTION, by M. Lutzky and D. Lehte. 17 Sept. 1965. 8p. charts, tables. NOL task 181. UNCLASSIFIED A family of non-dimensional curves is presented which allows scaling to different ambient conditions for a given energy source. These curves also make possible scaling close-in to the initial fireball. In addition, it is shown that the theoretical point-source solution for an ideal gas agrees with the computed solutions, except at close-in distances and at shock strengths below 50. It is shown that neither of these idea-gas solutions agrees as well with experiment as a real-gas solution.	1. Explosions, Nuclear - Scaling Title I. Lutzky, II. Merten III. Lehte, Delbert J., jt. author IV. Project Abstract card is unclassified.