

NOTICE

Reproduction of this document in any form by other than naval activities is not authorized except by special approval of the Secretary of the Navy or the Chief of Naval Operations as appropriate.

The following Espionage notice can be disregarded unless this document is plainly marked CONFIDENTIAL or SECRET.

This document contains information affecting the national defense of the United States within the meaning of the Espionage Laws, Title 18, U.S.C. Sections 793 and 794. The transmission or the revelation of its contents in any manner to an unauthorized person is prohibited by law.

AD 609 355

THIS REPORT MAY BE RELEASED OUTSIDE OF THE MILITARY
DEPARTMENTS AND EXECUTIVE AGENCIES OF THE UNITED
STATES GOVERNMENT ONLY WITH PAGES NONE REMOVED

Qualified requesters may obtain
copies of this report direct
from DDC

NAVAL AIR ENGINEERING CENTER
PHILADELPHIA, PA. 19112

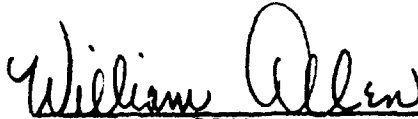
AERONAUTICAL MATERIALS LABORATORY

REPORT NO. NAEC-AML-2083

DATE 30 Nov 1964

INVESTIGATION INTO THE ELECTRICAL CONDUCTIVITY AND
MECHANICAL PROPERTIES OF ALUMINUM ALLOYS SUBJECTED
TO ELEVATED TEMPERATURE EXPOSURE

PROBLEM ASSIGNMENT NO. 10-40 UNDER BUREAU OF NAVAL
WEAPONS WEPTASK RRMA 02 018/200 1/R007 05 01


WILLIAM ALLEN

PROJECT ENGINEER

PREPARED BY:



ROBERT G. MAHORTER
PROJECT ENGINEER

APPROVED BY:



JOHN F. ERTHAL, HEAD
MECHANICAL METALLURGY BRANCH



F. S. WILLIAMS, SUPERINTENDENT
METALLURGICAL DIVISION

TABLE OF CONTENTS

	<u>PAGE</u>
ABSTRACT	3
INTRODUCTION	4
EXPERIMENTAL PROCEDURE	4
RESULTS AND ANALYSIS	5
CONCLUSIONS	5
RECOMMENDATIONS	6

LIST OF TABLES

- 1 - Results of Test
- 2 - Conductivity Data for Various Aluminum Alloys

LIST OF PLATES

- 1 - Electrical Conductivity Measuring Technique
- 2 - Conductivity vs Yield Strength for Aluminum Alloy 6061-T6
- 3 - Conductivity vs Yield Strength for Aluminum Alloy 2024-T81
- 4 - Conductivity vs Yield Strength for Aluminum Alloy 2219-T81
- 5 - Conductivity vs Yield Strength for Aluminum Alloy 7075-T73
- 6 - Conductivity vs Yield Strength for Aluminum Alloy 2020-T6
- 7 - Conductivity vs Yield Strength for Aluminum Alloy 7075-T6
- 8 - Conductivity vs Yield Strength for Aluminum Alloy 7178-T6
- 9 - Conductivity vs Yield Strength for Aluminum Alloy 7002-T6
- 10 - Conductivity vs Yield Strength for Aluminum Alloy 7079-T6
- 11 - Conductivity vs Yield Strength for Aluminum Alloy 2024-T3

ABSTRACT

The relationship between the electrical conductivity (as measured by Magnatester Conductivity Meter, F100 Series) and strength properties of bare aluminum alloys 7075-T6, 7075-T73, 6061-T6, 7178-T6, 7002-T6, 2024-T81, 2024-T3, 7079-T6, 2020-T6, and 2219-T81 was investigated in an attempt to correlate conductivity with heat damage to aircraft structural alloys.

I. INTRODUCTION

The Bureau of Naval Weapons has expressed concern over heat damage to aircraft structures in locations subject to power plant heating. Since this heat damage occurs at temperatures below that which causes changes in the paint surface appearance, it is necessary to use some other parameter to assess the damage. A method which measures the electrical conductivity of the alloy by means of eddy current techniques has been suggested and is in fact being used to assess damage to aluminum alloy 7075-T6 in the A-4 aircraft. It is the purpose of this investigation to obtain a correlation between conductivity as measured by the Magnatester F100 Series and strength properties of other aluminum alloys.

II. EXPERIMENTAL PROCEDURE

The bare aluminum alloys used in this investigation were 7075-T6, 7075-T73, 6061-T6, 7178-T6, 7002-T6, 2024-T81, 2024-T3, 7079-T6, 2020-T6, 2219-T81.

Tensile test specimens of the above alloys were manufactured from available material. Most of the alloys were in the form of bare sheet of various thicknesses. Several alloys, however, were available only as plate from which round tensile specimens were machined. The form of the test specimens used for each alloy is given with the results in Table 1. For each alloy, the test specimens all had the same orientation with relation to the rolling direction.

Conductivity and hardness measurements were made on each material in the as-received condition. The test specimens were then exposed for various times at temperatures of 400°F, 500°F, and 600°F. Conductivity and hardness measurements were then made on the exposed specimens. Following this, the specimens were tensile tested at a strain rate of approximately 0.035 in/in/min. and the 0.2% yield strength and ultimate tensile strength were calculated.

The conductivity measurements were made using a Magnatester Conductivity Meter (F100 Series), Plate 1. The sensitivity of this instrument was $\pm 0.3\%$ IACS and the accuracy is within $\pm 1-1/2\%$ of scale reading. During calibration of the Magnatester, the standard calibration specimens furnished with the instrument were maintained at the same temperature as the specimens whose conductivity was to be measured, thus eliminating a temperature induced error.

III. RESULTS AND ANALYSIS

The results obtained in this investigation are given in Table 1. Since the yield strength is the most useful of the properties investigated, it is plotted as a function of the conductivity for the various alloys in Plates 1 to 10. Table 2 gives the value of the conductivity and amount of change in the conductivity which corresponds to a 20% loss in yield strength. The amount of change is a measure of the ability of the Conductivity Meter to detect a loss in strength.

From the data in Table 2 and Plates 2, 3, 4, 5, and 6, it is apparent that no useful correlation between conductivity and strength properties is possible for alloys 6061-T6, 2024-T81, 2219-81, 7075-T73, and 2020-T6. In each of these cases, the change in conductivity between the original material and that which has been severely reduced in strength properties by heating is so small that it must be considered non-reproducible.

Alloys 7075-T6, 7178-T6, 7002-T6, and 7079-T6 give useful correlations between conductivity and yield strength as can be seen from Table 2 and Plates 7, 8, 9, and 10.

While alloy 2024-T3, Plate 11, has a very large change in conductivity for a 20% loss in yield strength, it is improbable that a useful correlation can be obtained since only at a high conductivity is a significant loss in strength observed and this loss occurs over a very small conductivity range. A lower conductivity limit could be chosen to eliminate all damaged structures, but in view of the shape of the curve, it could cause a replacement of a large number of structures whose strength was not significantly reduced.

IV. CONCLUSIONS

1. Heat damage to structures can be correlated to conductivity readings obtained using a Magnatester Conductivity Meter (F100 Series) for aluminum alloys 7075-T6, 7178-T6, 7002-T6, and 7079-T6.

2. A similar correlation cannot be made for aluminum alloys 6061-T6, 2024-T81, 7075-T73, 2219-T81, and 2020-T6 due to the small change in conductivity which occurs due to heat damage.

3. The conductivity of alloy 2024-T3 can be correlated, but such a correlation would lead to rejects of many heated but unweakened structures along with those which had been damaged. Above a conductivity of 38%, a high probability exists that this material has been weakened to below the as-received strength and tests of other types should be made.

V. RECOMMENDATIONS

It is recommended that the Magnatester Conductivity Meter (Series F100) be considered for use in detecting heat damage to aluminum alloys 7178-T6, 7002-T6, and 7079-T6 in addition to 7075-T6 alloy

RESULTS OF TEST

<u>Alloy & Form</u>	<u>Exposure Temperature</u>	<u>Exposure Time</u>	<u>Conductivity Range, % IACS</u>	<u>Hardness</u>	<u>0.2% Yield Strength</u>	<u>Ultimate Strength</u>
7075-T73 0.065" Sheet	As-Received		37.5	87 RB	63,500	75,200
	400°F	1 hr.	38.5	84 RB	58,800	72,600
	400°F	5 hrs.	40.5	68 RB	43,800	59,500
	510°F	10 min.	41.5	61 RB	34,600	53,100
	490°F	1 hr.	41.0	65 RB	37,400	55,800
	600°F	10 min.	39.5	31 RB	19,500	43,900
6061-T6 0.063" Sheet	As-Received		41.5	75 Rf	18,900	41,700
	As-Received		39.0	56 RB	40,900	46,700
	400°F	1 hr.	40.0	54 RB	40,000	46,700
	400°F	5 hrs.	40.5	47 RB	40,800	44,000
	510°F	10 min.	41.0	33 RB	33,000	38,300
	490°F	1 hr.	41.0	85 Rf	35,000	41,200
7178-T6 0.100" Sheet	600°F	10 min.	42.3	--	18,700	28,600
	600°F	1 hr.	42.5	58 Rf	16,700	27,700
	As-Received		29.5	95 RB	84,200	91,800
	400°F	1 hr.	36.0	92 RB	78,800	85,200
	400°F	5 hrs.	41.0	75 RB	48,000	62,800
	510°F	10 min.	40.5	66 RB	37,400	55,900
2020-T6 0.064" Sheet	490°F	1 hr.	41.3	70 RB	41,800	58,700
	600°F	10 min.	40.5	40 RB	21,600	45,500
	600°F	1 hr.	41.8	80 Rf	20,000	42,200
	As-Received		20.5	91 RB	73,500	78,666
	400°F	1 hr.	20.5	90 RB	71,600	76,757
	400°F	5 hrs.	21.0	84 RB	67,968	73,906
	500°F	1 hr.	21.9	74 RB	46,400	58,787
	600°F	1 hr.	23.3	46 RB	28,100	46,099

TABLE 1
PAGE 1 OF 3 PAGES

<u>Alloy & Form</u>	<u>Exposure Temperature</u>	<u>Exposure Time</u>	<u>Conductivity Range, % IACS</u>	<u>Hardness</u>	<u>0.2% Yield Strength</u>	<u>Ultimate Strength</u>
7079-T6 0.088" Sheet	As-Received		31.0	89 RB	70,000	80,000
	400°F	10 min.	33.0	83 RB	62,500	74,300
	400°F	1 hr.	34.0	85 RB	62,500	74,200
	400°F	5 hrs.	36.5	69 RB	44,600	61,500
	510°F	10 min.	35.8	63 RB	34,700	56,600
	490°F	1 hr.	36.5	67 RB	37,900	58,200
	600°F	10 min.	36.0	39 RB	21,200	48,000
	600°F	1 hr.	37.0	74 RF	20,800	45,000
7075-T6 0.100" Sheet	As-Received		32.0	93 RB	79,000	85,500
	400°F	1 hr.	37.0	90 RB	72,400	79,600
	400°F	5 hrs.	41.0	70 RB	45,000	59,700
	510°F	10 min.	41.5	61 RB	35,500	54,300
	490°F	1 hr.	41.0	66 RB	39,200	56,600
	600°F	10 min.	41.5	78 RF	20,000	44,200
	600°F	1 hr.	42.0	76 RF	18,600	41,100
2219-T81 0.062" Sheet	As-Received		32.0	77 RB	52,100	67,290
	400°F	1 hr.	32.0	76 RB	52,100	68,700
	400°F	5 hrs.	32.5	74 RB	34,900	46,500
	400°F	5 hrs.	32.5	73 RB	35,500	46,600
	500°F	1 hr.	33.5	68 RB	40,800	59,200
	600°F	1 hr.	34.6	58 RB	33,200	51,000
7002-T6 0.062" Sheet	As-Received		33.5	82 RB	58,900	69,700
	400°F	10 min.	35.0	76 RB	54,200	65,500
	400°F	1 hr.	36.0	79 RB	54,100	66,800
	510°F	10 min.	37.5	49 RB	29,600	50,200
	490°F	1 hr.	38.5	58 RB	32,600	52,200
	600°F	10 min.	37.0	26 RB	17,300	41,800
	600°F	1 hr.	38.5	73 RF	16,600	41,200

TABLE 1
PAGE 2 OF 3 PAGES

<u>Alloy & Form</u>	<u>Exposure Temperature</u>	<u>Exposure Time</u>	<u>Conductivity Range, % IACS</u>	<u>Hardness</u>	<u>0.2% Yield Strength</u>	<u>Ultimate Strength</u>
2024-T81 0.125" Sheet	As-Received		38.0	83 RB	66,400	72,100
	400°F	1 hr.	37.5	84 RB	66,300	72,000
	400°F	5 hrs.	38.7	80 RB	63,800	70,800
	510°F	10 min.	38.8	78 RB	56,600	67,100
	490°F	1 hr.	39.5	78 RB	57,500	67,100
	600°F	10 min.	40.0	64 RB	40,500	57,800
	600°F	1 hr.	41.0	61 RB	36,600	54,400
2024-T3 0.125" Sheet	As-Received		29.0	77 RB	45,400	69,600
	400°F	1 hr.	30.0	75 RB	43,000	67,200
	510°F	10 min.	38.5	76 RB	56,100	65,000
	490°F	1 hr.	38.8	79 RB	60,300	68,400
	600°F	10 min.	40.3	64 RB	41,000	56,500
	600°F	1 hr.	41.5	67 RB	40,800	57,400
	600°F	3 hrs.	42.5	--	26,600	45,400
	700°F	2 hrs.	42.5	--	15,900	39,600

CONDUCTIVITY DATA FOR VARIOUS ALUMINUM ALLOYS

<u>Alloy</u>	<u>Change in Conductivity, % IACS for 20% Loss of Yield Strength</u>	<u>Conductivity at 80% As-Received Yield Strength</u>
2219-T81	1	33.5
7002-T6	3.1	36.6
7178-T6	8.0	37.5
2024-T81	2.0	39.5
7075-T73	2.0	39.5
6061-T6	2.0	41.0
7079-T6	3.5	34.5
2024-T3	12.0	41.0
7075-T6	6.0	38.0
2020-T6	1.0	21.5

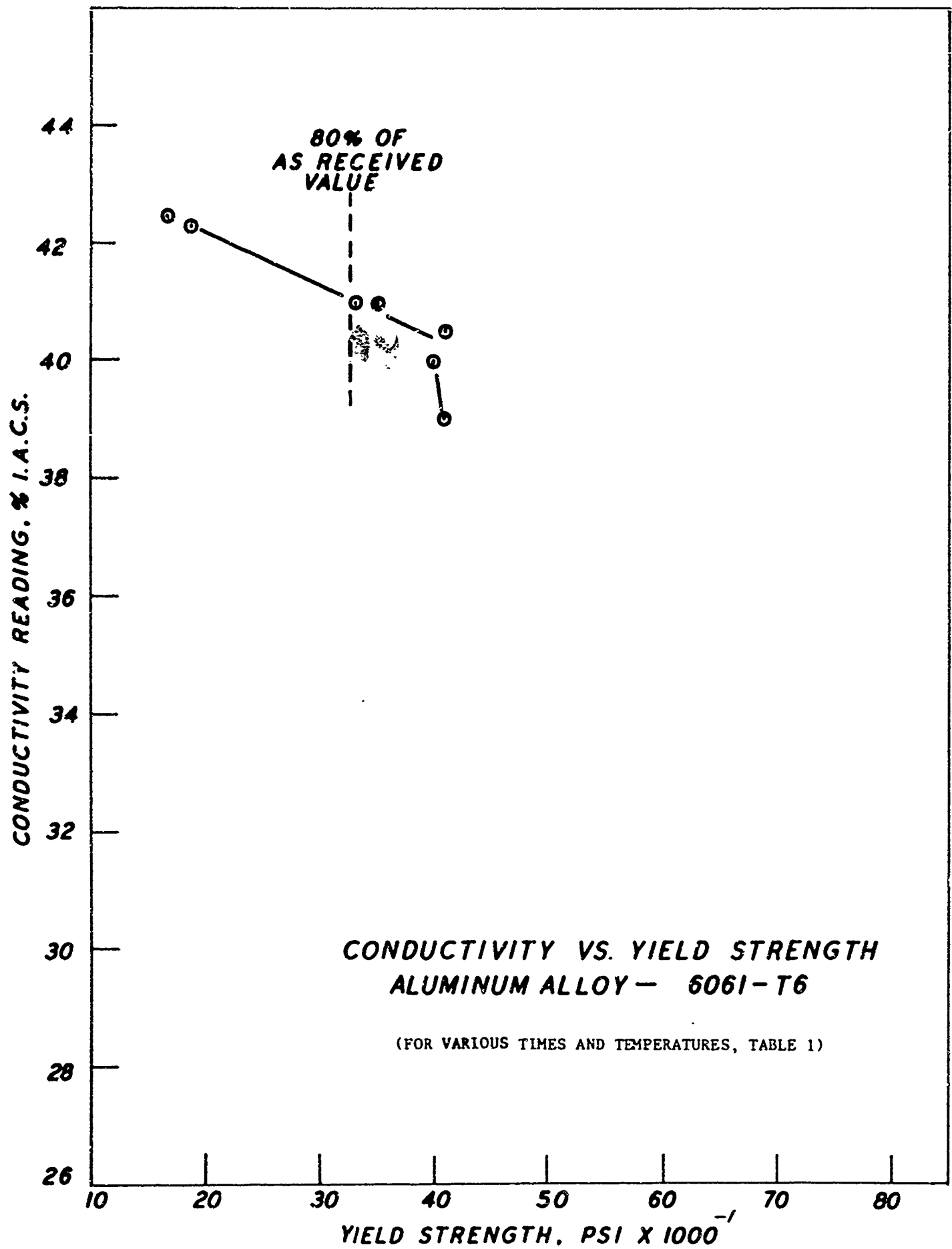
TABLE 2

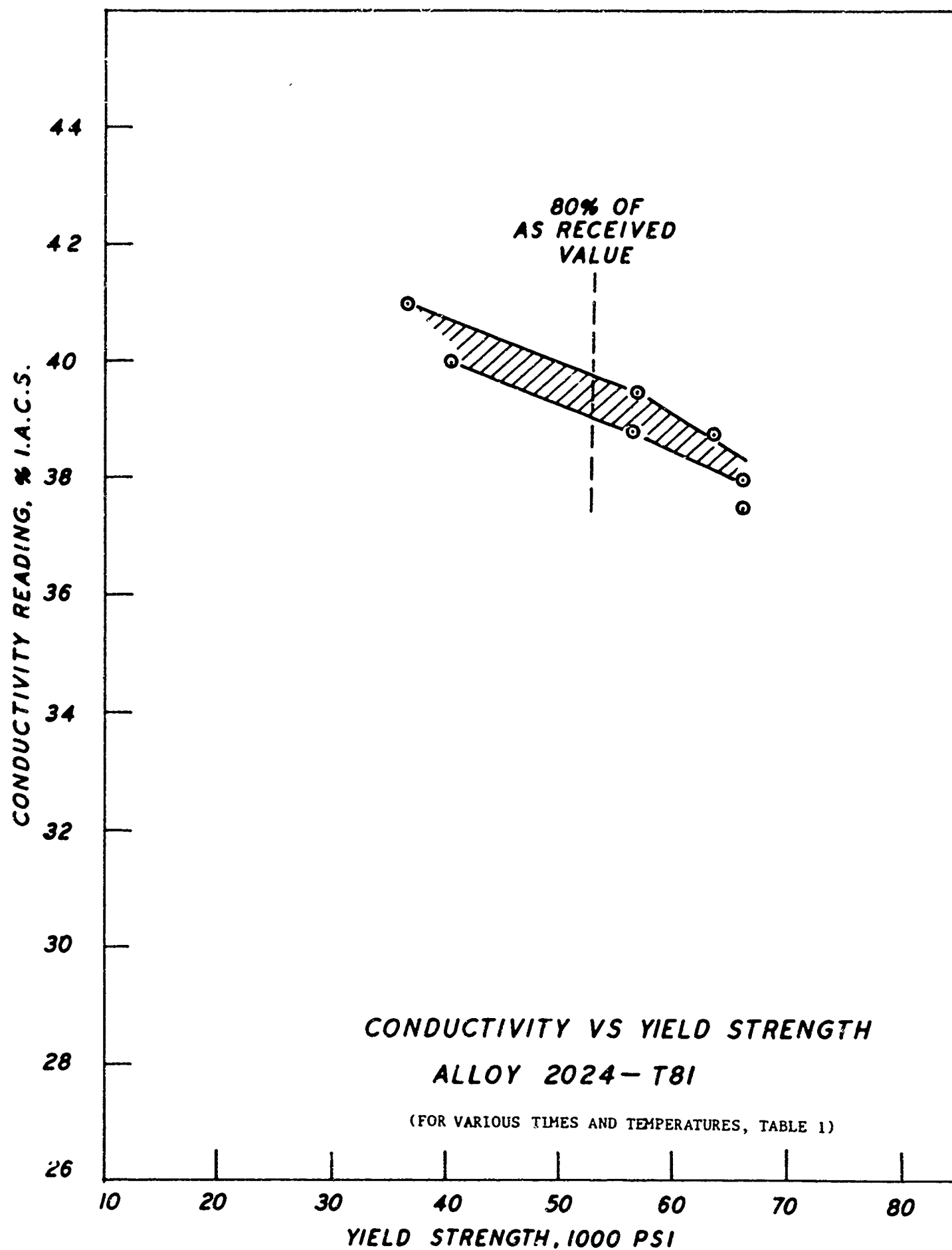


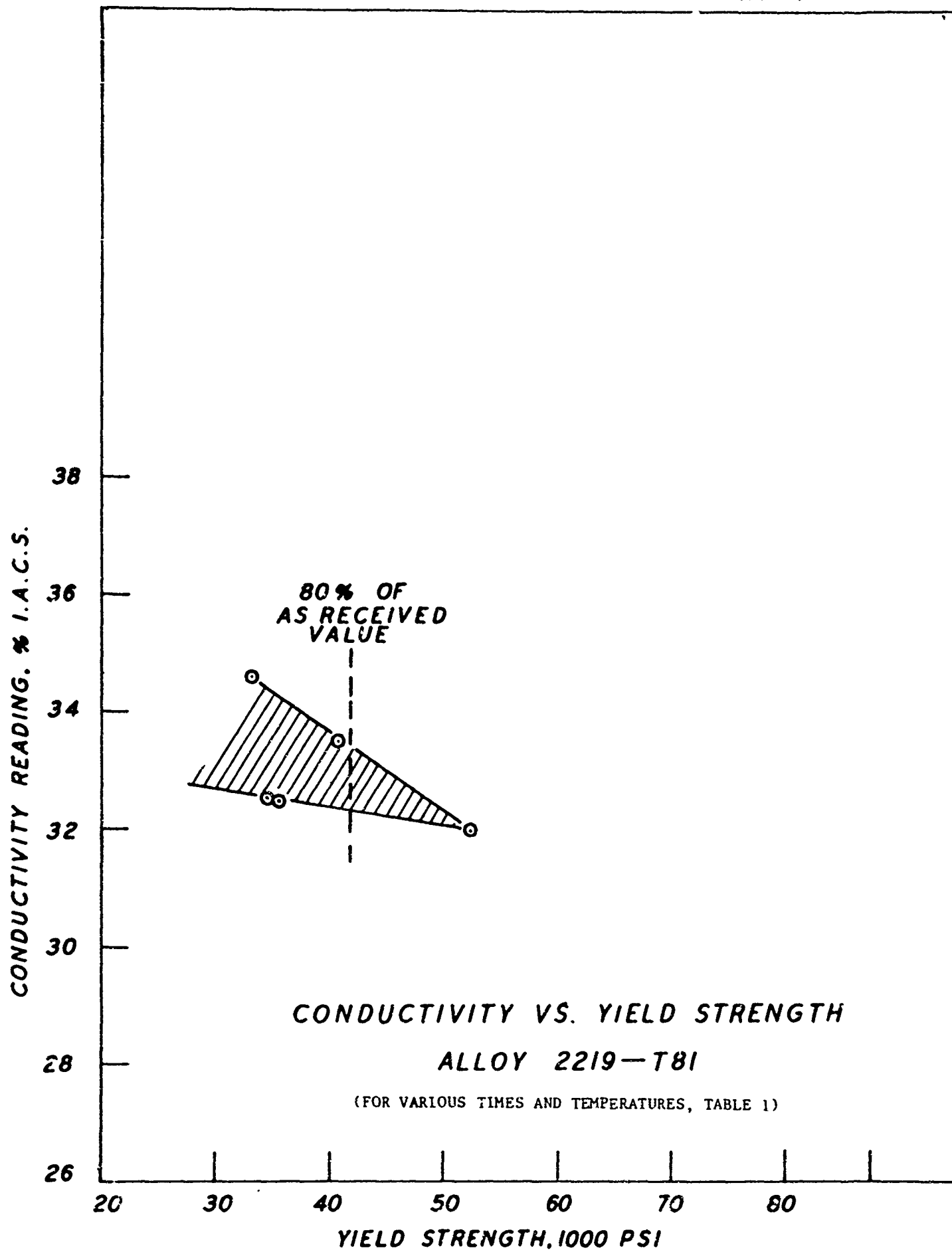
ELECTRICAL CONDUCTIVITY MEASURING TECHNIQUE

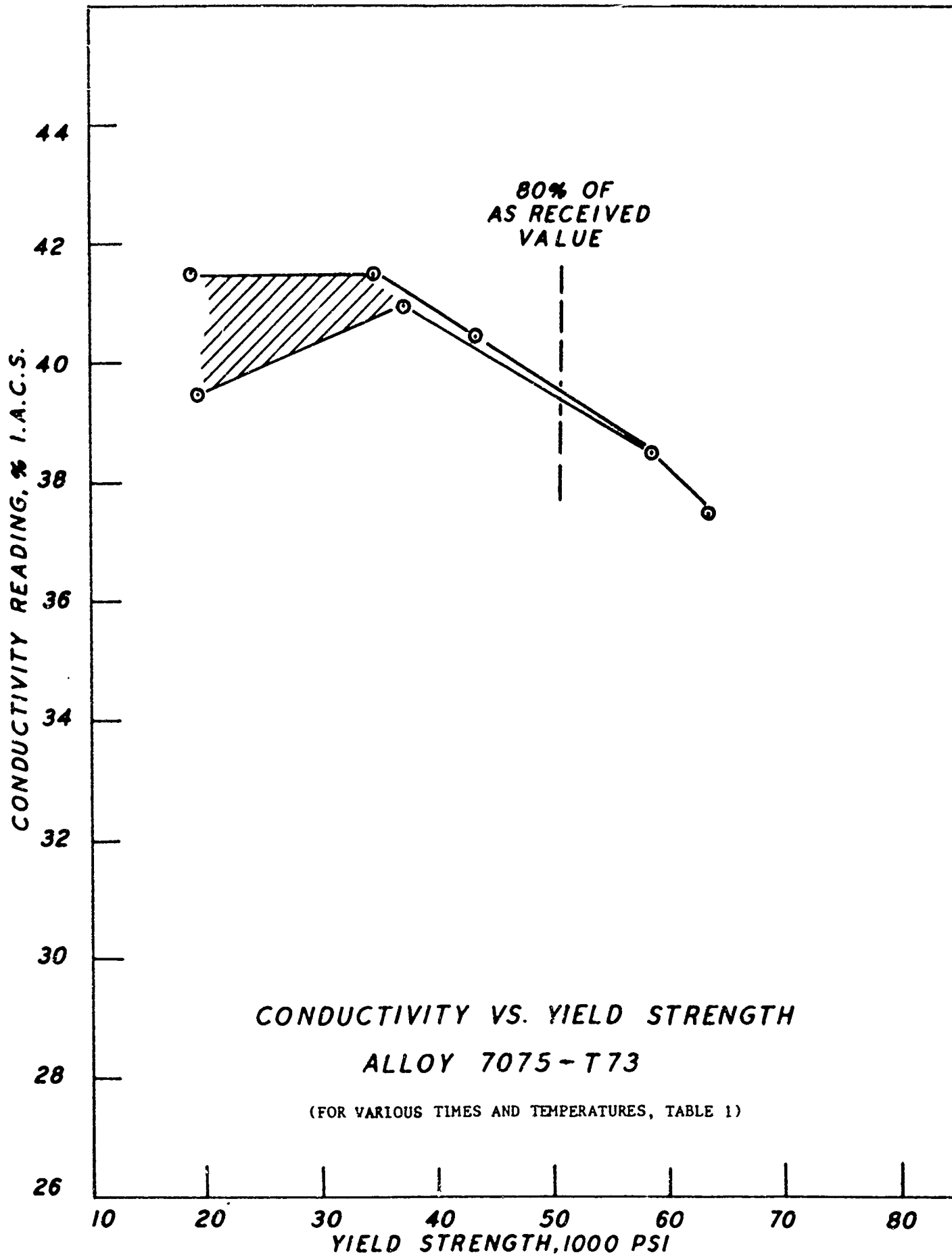
PHOTO NO: CAN-364603(L)-11-64

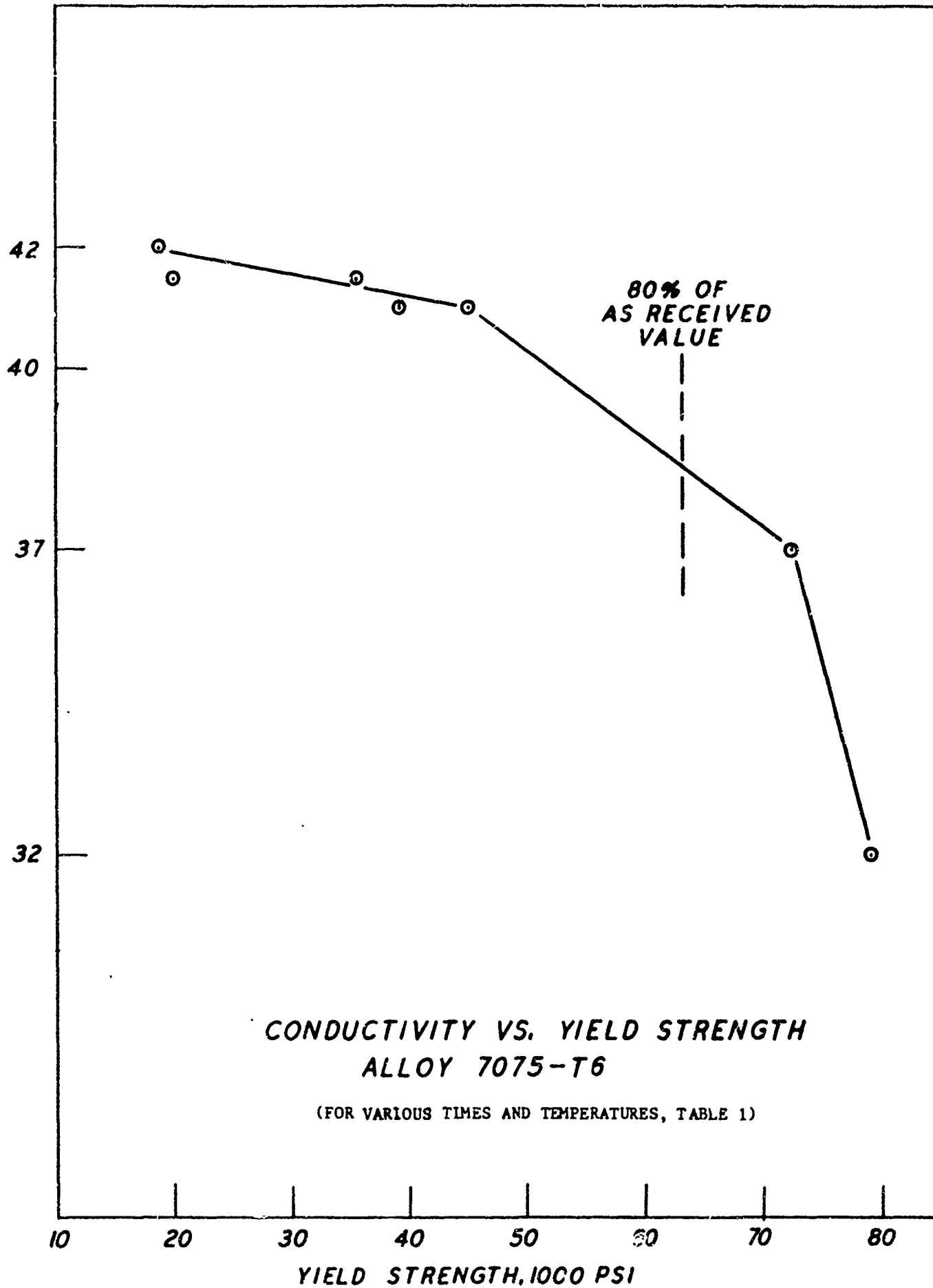
PLATE 1

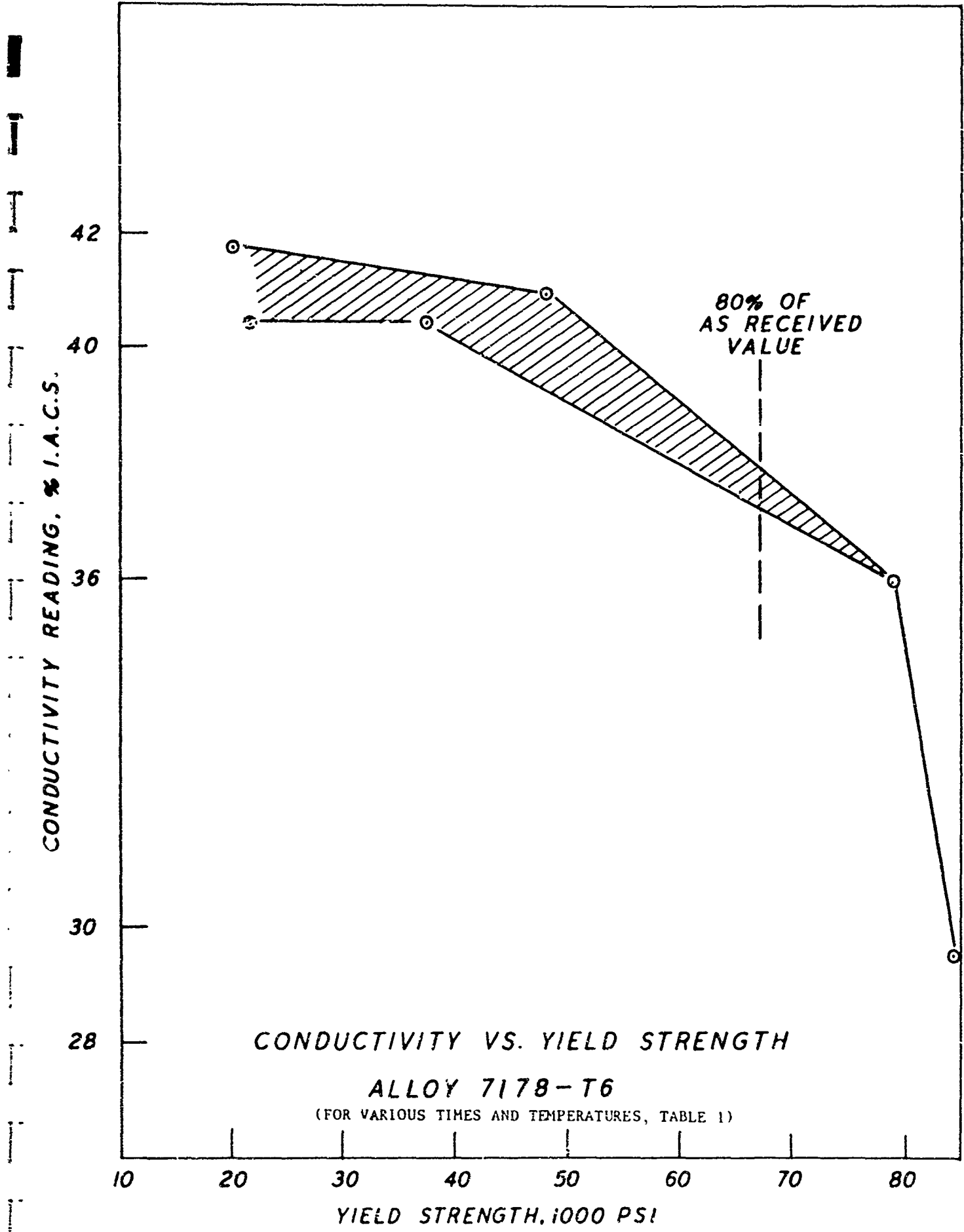


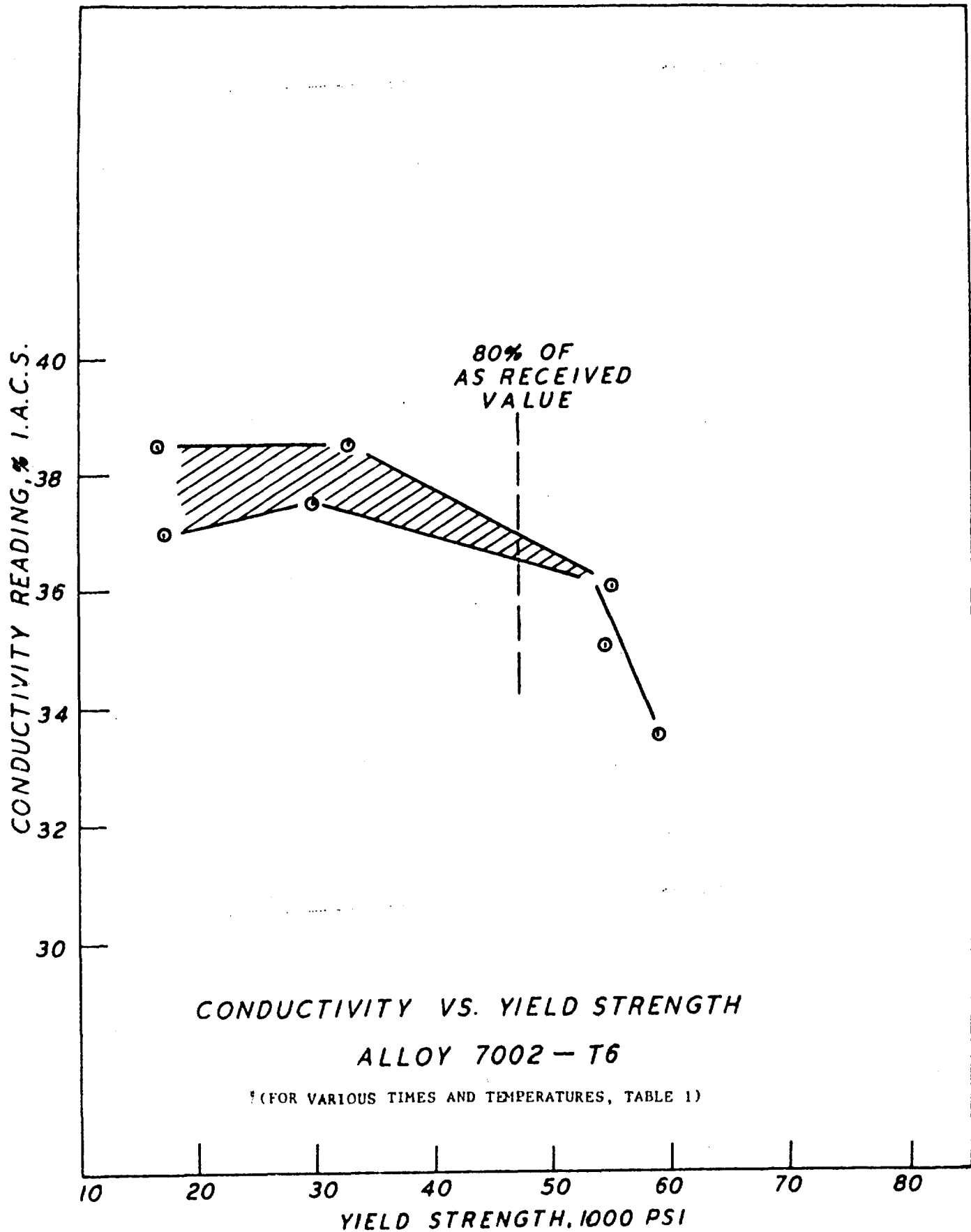


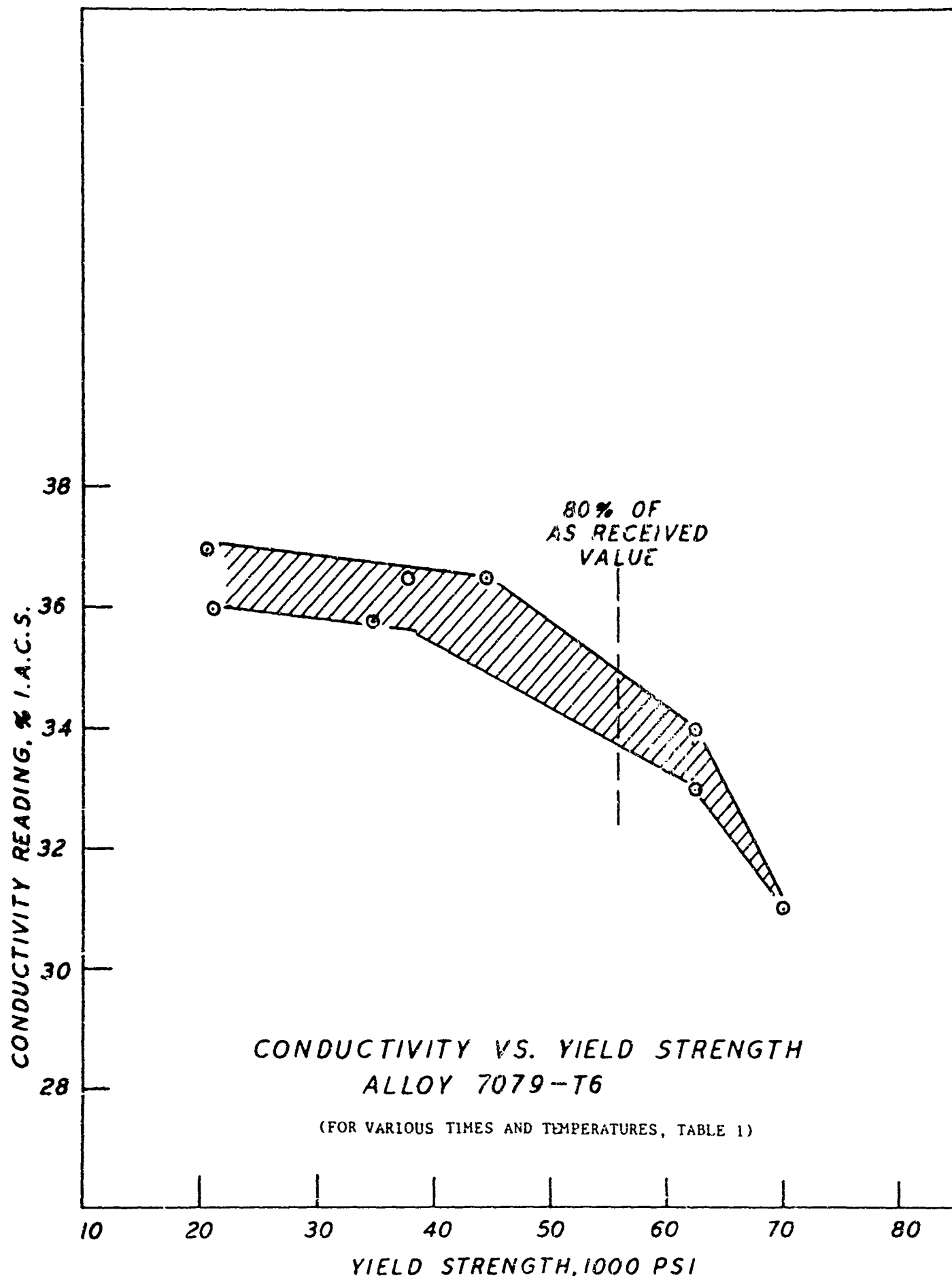


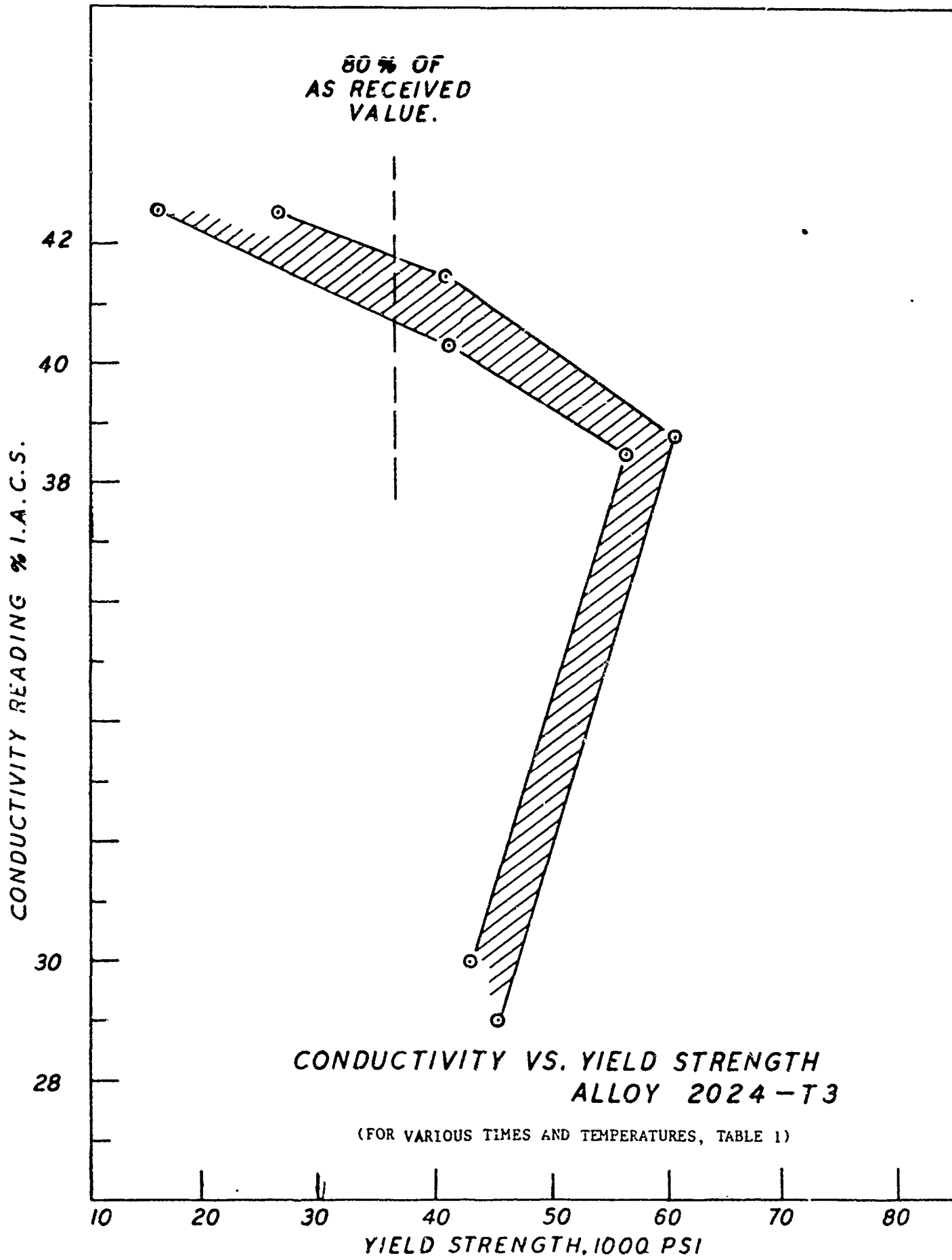












U. S. NAVAL AIR ENGINEERING CENTER, PHILA, PA. AERONAUTICAL MATERIALS LABORATORY Investigation Into the Electrical Conductivity and Mechanical Properties of Aluminum Alloys Subjected to Elevated Temperature Exposure; W. Allen/R. Mahorter, December 1964, 2 Tables, 11 Plates The relationship between the electrical conductivity (as measured by Magnetester Conductivity Meter, F100 Series) and strength properties of bare aluminum alloys 7075-16, 7075-173, 6061-16, 7178-16, 7002-16, 2024-181, 2024-183, 7079-16, 2020-16, and 2219-181 was investigated in an attempt to correlate conductivity with heat damage to aircraft structural alloy.	1. REPORT NO. NAEC-AML-2083 2. PAN 10-40
U. S. NAVAL AIR ENGINEERING CENTER, PHILA, PA. AERONAUTICAL MATERIALS LABORATORY Investigation Into the Electrical Conductivity and Mechanical Properties of Aluminum Alloys Subjected to Elevated Temperature Exposure; W. Allen/R. Mahorter, December 1964, 2 Tables, 11 Plates The relationship between the electrical conductivity (as measured by Magnetester Conductivity Meter, F100 Series) and strength properties of bare aluminum alloys 7075-16, 7075-173, 6061-16, 7178-16, 7002-16, 2024-181, 2024-183, 7079-16, 2020-16, and 2219-181 was investigated in an attempt to correlate conductivity with heat damage to aircraft structural alloy.	1. REPORT NO. NAEC-AML-2083 2. PAN 10-40

U. S. NAVAL AIR ENGINEERING CENTER, PHILA, PA. AERONAUTICAL MATERIALS LABORATORY Investigation Into the Electrical Conductivity and Mechanical Properties of Aluminum Alloys Subjected to Elevated Temperature Exposure; W. Allen/R. Mahorter, December 1964, 2 Tables, 11 Plates The relationship between the electrical conductivity (as measured by Magnetester Conductivity Meter, F100 Series) and strength properties of bare aluminum alloys 7075-16, 7075-173, 6061-16, 7178-16, 7002-16, 2024-181, 2024-183, 7079-16, 2020-16, and 2219-181 was investigated in an attempt to correlate conductivity with heat damage to aircraft structural alloys.	1. REPORT NO. NAEC-AML-2083 2. PAN 10-40
U. S. NAVAL AIR ENGINEERING CENTER, PHILA, PA. AERONAUTICAL MATERIALS LABORATORY Investigation Into the Electrical Conductivity and Mechanical Properties of Aluminum Alloys Subjected to Elevated Temperature Exposure; W. Allen/R. Mahorter, December 1964, 2 Tables, 11 Plates The relationship between the electrical conductivity (as measured by Magnetester Conductivity Meter, F100 Series) and strength properties of bare aluminum alloys 7075-16, 7075-173, 6061-16, 7178-16, 7002-16, 2024-181, 2024-183, 7079-16, 2020-16, and 2219-181 was investigated in an attempt to correlate conductivity with heat damage to aircraft structural alloys.	1. REPORT NO. NAEC-AML-2083 2. PAN 10-40