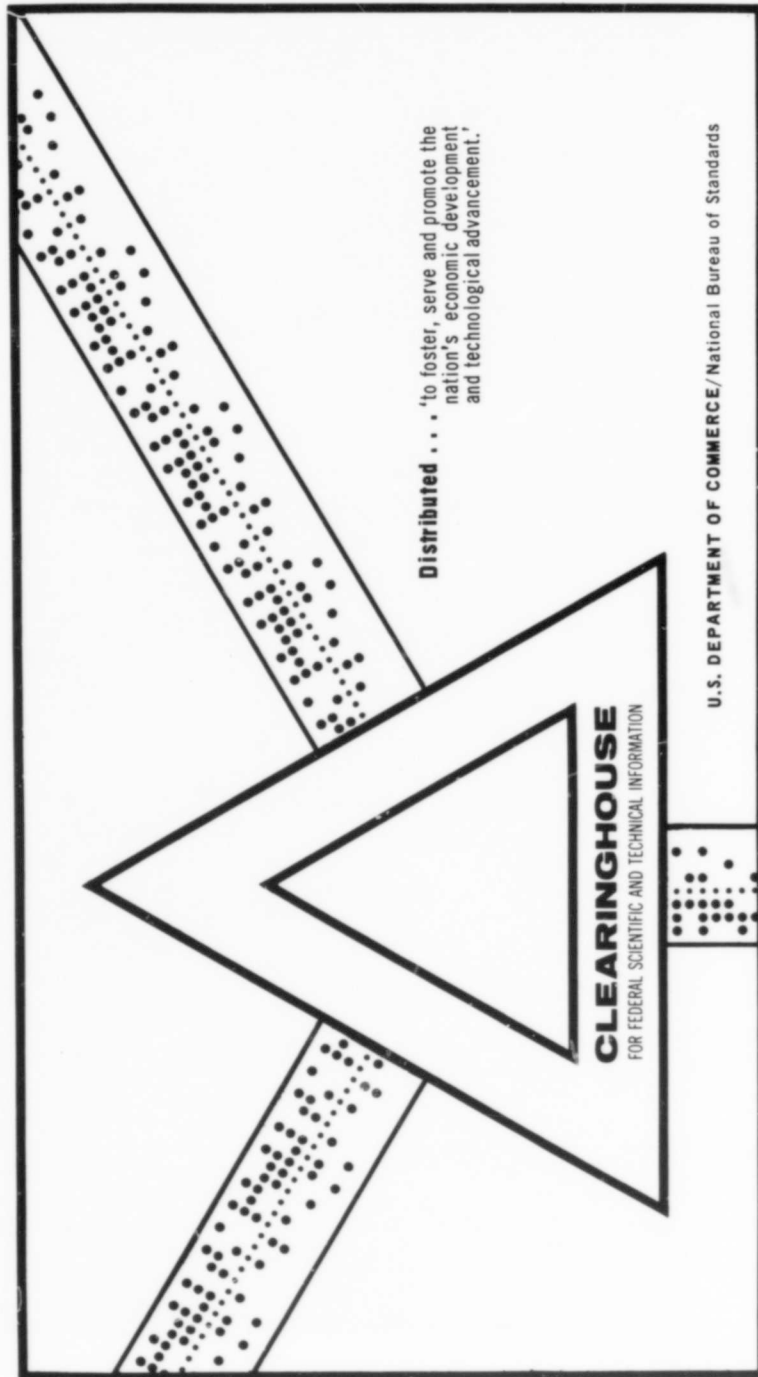


AD 698 530

THERMODYNAMIC PROPERTIES OF THE GASEOUS DISSOCIATING SYSTEM
A12C16 = 2A1C13 (SELECTED ARTICLES)

V. B. Nesterenko, et al
Foreign Technology Division
Wright-Patterson Air Force Base, Ohio

22 August 1969



This document has been approved for public release and sale.

AD 698 530

FTD-HT-23-299-69

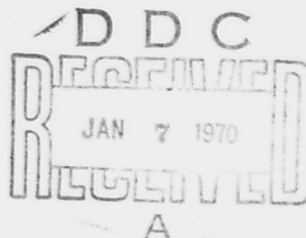
FOREIGN TECHNOLOGY DIVISION



THERMODYNAMIC PROPERTIES OF THE GASEOUS
DISSOCIATING SYSTEM $\text{Al}_2\text{Cl}_6 \rightleftharpoons 2\text{AlCl}_3$
(SELECTED ARTICLES)

By

V. B. Nesterenko, A. V. Zinov'yev
and M. A. Bazhin, et. al.



Distribution of this document is unlimited. It may be released to the Clearinghouse, Department of Commerce, for sale to general public.

Reproduced by the
CLEARINGHOUSE
for Federal Scientific & Technical
Information Springfield Va. 22151

EDITED TRANSLATION

THERMODYNAMIC PROPERTIES OF THE GASEOUS DISSOCIATING
SYSTEM $A_2C_6 \rightleftharpoons 2A_1C_3$ (SELECTED ARTICLES)

By: V. B. Nesterenko, A. V. Zinov'yev, M. A. Bazhin
et. al.

Source: AN BSSR. Vesti. Seriya Fizika-Tekhnichnykh
Navuk, (Academy of Sciences of the Belorussian
SSR. News. Series in the Physical and
Technical Sciences), 1967, No. 2, pp.32-41,
42-49, 50-55

English Pages: 29

Translated Under: 33657-68-D-0866

THIS TRANSLATION IS A RENDITION OF THE ORIGINAL FOREIGN TEXT WITHOUT ANY ANALYTICAL OR EDITORIAL COMMENT. STATEMENTS OR THEORIES ADVOCATED OR IMPLIED ARE THOSE OF THE SOURCE AND DO NOT NECESSARILY REFLECT THE POSITION OR OPINION OF THE FOREIGN TECHNOLOGY DIVISION.

PREPARED BY:

TRANSLATION DIVISION
FOREIGN TECHNOLOGY DIVISION
WP.AFB, OHIO.

DATA HANDLING PAGE

01-ACCESSION NO. 90-DOCUMENT LOC TP9001190		39-TOPIC TAGS aluminum compound, gas dissociation, gas turbine, enthalpy, entropy, specific heat		
09-TITLE THERMODYNAMIC PROPERTIES OF THE GASEOUS DISSOCIATING SYSTEM $A1_2C1_6 \leftrightarrow 2A1C1_3$				
47-SUBJECT AREA 20				
12-AUTHOR/CO-AUTHORS NESTERENKO, V. B. ; 16-ZINOV'YEV, A. V. ; 16-BAZHIN, M. A.				18-DATE OF INFO -----67
43-SOURCE AN BSSR. VESTSI. SERYYA FIZIKA- TEKHNICHNYKH NAVUK (RUSSIAN)				48-DOCUMENT NO. HT-23-299-69 49-PROJECT NO. 60401
63-SECURITY AND DOWNGRADING INFORMATION UNCL, 0		64-CONTROL MARKINGS NONE	97-HEADER CLASH UNCL	
76-REEL FRAME NO. 1889 1863	77-SUPERSEDES	78-CHANGES	66-GEOGRAPHICAL AREA UR	NO OF PAGES 13
CONTRACT NO. F33657-68-D- 0866 P002	X REF ACC. NO. 65-AP7017901	PUBLISHING DATE 94-	TYPE PRODUCT TRANSLATION	REVISION FREQ NONE
STEP NO. 02-UR/0201/67/000/002/0032/0041			ACCESSION NO.	

ABSTRACT 571

(U) This is a continuation of earlier work (Vesti AN BSSR, ser. fiz. tekhnauk no. 1, 1966 and elsewhere) where it was shown that the use of a dissociating gas system as the working medium in a closed gas-turbine cycle increases the gas constant in the turbine and decreases it in the compressor, thus increasing the efficiency, compared with cycles using inert gases, by reducing the irreversible losses in the cycle. The present article deals with the $A1$ (subscript 2) $C1$ (subscript 6) forms and is formed from $2A1C1$ (subscript 3) system, for which there are not many published data. The available specific-heat data were approximated by means of a computer in the form of a fifth degree polynomial and the coefficients of this polynomial for $A1$ (subscript 2) $C1$ (subscript 6) and $A1C1$ (subscript 3) were calculated. From them they calculated the thermodynamic properties of gaseous aluminum chloride under a number of assumptions. Equations are derived on this basis for effective enthalpy and entropy of the dissociating system, for the effective specific heat, at pressures 1-150 atmospheres and 460-2000K. Some of the results are presented in the form of tables and plots. The authors thank S. Zakharova for the computations with the "Minsk-2" computer and Academician AN BSSR A. K. Krasin for interest in the work. Orig. art. has: 3 figures, 27 formulas, and 3 tables.

DATA HANDLING PAGE

1-ACCESSION NO. 18-DOCUMENT LOC P9001191		39-TOPIC TAGS heat transfer coefficient, reynolds number, prandtl number, gas flow, fluid flow, turbulent flow, pipe flow, flow rate		
4-TITLE HEAT TRANSFER IN THE CHEMICALLY-REACTING EQUILIBRIUM SYSTEM $N_2O_4 \rightleftharpoons 2NO_2$				
47-SUBJECT AREA 20				
12-AUTHOR/CO-AUTHORS NESTERENKO, V. B. ; 16-TVERKOVKIN, B. YE. ; 16-VERZHINSKAYA, A. B.		10-DATE OF INFO ----67		
43-SOURCE AN BSSR. VESTSI SERYYA FIZIKA-TEKHNICHNYKH NAVUK (RUSSIAN)		FTD-	48-DOCUMENT NO. HT-23-299-69	
			49-PROJECT NO. 60401	
63-SECURITY AND DOWNGRADING INFORMATION UNCL, O		64-CONTROL MARKINGS NONE		97-HEADER CLASS UNCL
76-REEL FRAME NO. 1889 1864	77-SUPERSEDES	78-CHANGES	46-GEOGRAPHICAL AREA UR	NO OF PAGES 9
CONTRACT NO. F33657-68-D-0866 P002	X REF ACC. NO. 65-AP7017902	PUBLISHING DATE 94-	TYPE PRODUCT TRANSLATION	REVISION FREQ NONE
STEP NO. 02-UR/0201/67/000/002/0042/0049		ACCESSION NO.		

ABSTRACT

(U) (2) Experimental studies of heat transfer in a chemically reactive gas flow have been carried out at the Institute of Nuclear Power Engineering of the BSSR. As suggested by M. A. Mikheyev and S. S. Kutateladze, the Nusselt number was used as a function of the Reynolds and Prandtl criteria in a forced turbulent flow in a tube. The investigated temperature range was 30-300 degrees C, pressure from 1-60 atm and Reynolds numbers from 10^4 to 3×10^5 . The experimental unit consisted of a closed loop using a pump for the circulation of N_2O_4 . The flow rate was controlled by the pump rpm and a by-pass valve. The test section was a U-tube type made of stainless steel. Experimental and calculated values of the heat transfer coefficient showed close correlation. The derived empirical formulas can be recommended for the description of the heat transfer mechanism in a chemically reactive system. The authors express their appreciation to Academician A.K. Krasin of the AN BSSR for his interest in the study, and to A. N. Kremeshnyy, Yu. I. Bakalin, and V. A. Nikolayev who participated in the study. Orig. art. has: 2 figures., 19 formulas, and 1 table.

DATA HANDLING PAGE

61-ACCESSION NO. 98-DOCUMENT LOC TP9001192		39-TOPIC TAGS heat exchange, nitrogen oxide, chemical reaction kinetics, turbulent flow		
69-TITLE HEAT TRANSFER IN THE CHEMICALLY-REACTING SYSTEM $2NO_2 \rightleftharpoons 2NO + O_2$ TAKING THE KINETICS OF THE CHEMICAL REACTION INTO ACCOUNT				
47-SUBJECT AREA 20				
12-AUTHOR/CO-AUTHORS NESTERENKO, V. B. ; 16-TVERKOVKIN, B. E. ; 16-VERZHINSKAYA, A. B.		18-DATE OF INFO -----67		
13-SOURCE AN BSSR. VESTSI. SERIYA FIZIKA-TEKHNICHNYKH NAVUK (RUSSIAN)		68-DOCUMENT NO. FTD-HT-23-299-69		69-PROJECT NO. 60401
63-SECURITY AND DOWNGRADING INFORMATION UNCL, O		64-CONTROL MARKINGS NONE		97-HEADER CLASS UNCL
76-REEL FRAME NO. 1889 1865	77-SUPERSEDES	78-CHANGES	40-GEOGRAPHICAL AREA UR	NO OF PAGES 7
CONTRACT NO. F33657-68-D-0866 P002	X REF ACC. NO. 65-AP7017903	PUBLISHING DATE 94-	TYPE PRODUCT TRANSLATION	REVISION FREQ NONE
STEP NO. 02-UR/0201/67/000/002/0050/0055		ACCESSION NO.		

ABSTRACT

(U) This a companion to the preceding article in the same source (pp. 42-49) dealing with heat exchange in the presence of chemical reaction, but under equilibrium conditions. The present article deals with the modifications that must be made to the calculations when allowance is made for the kinetics of the chemical reaction. It is shown that although the analysis is made much more complicated by the need for taking into account additional internal sources and sinks of the masses of the individual components, it is possible to obtain sufficiently accurate data by using empirical relations for the calculation of the heat transfer coefficient, by making use of a similarity criterion derived by the method of integral analogs. The equation derived for the heat transfer on this basis was compared with experimental data on heat exchange, carried out for forced turbulent flow in a tube at the Institute of Nuclear Power Engineering AN BSSR at gas temperatures of 140-550C, pressures 10-60 atmospheres, and Reynolds numbers 3 times 10^(superscript 4) - 3 times 10^(superscript 5). The experimental setup and procedure were the same as in the companion paper. A criterion empirical relation is derived by evaluating the constants of the theoretical equation to fit the experimental data, with plus or minus 15 percent. The authors thank Academician AN BSSR A. K. Krasin for continuous interest in the work.

Symbol List

Manuscript page	Symbol	Eng. Equivalent
15	cs	wall
15	f'	g
26	A	D

THERMODYNAMIC PROPERTIES OF THE GASEOUS DISSOCIATING SYSTEM $Al_2Cl_6 \rightleftharpoons 2AlCl_3$

V. B. Nesterenko, A. V. Zinov'yev and M. A. Bazhin

The cycle efficiency of gas turbine power plants can be increased by employing (chemically) reactive systems capable of dissociation upon heating. This results in an increase of the number of moles and, consequently, also of the gas "constant"; upon cooling, a recombination takes place, resulting in a decrease of the number of moles and of the gas "constant".

When such systems are used in a closed gas turbine cycle, the gas in the initial state, with a minimal gas "constant", is compressed in the compressor, and is then heated in the regenerator and heater to the maximum temperature of the cycle, this is accompanied by dissociation with heat absorption for the chemical reactions taking place, and an increase of the number of moles and of the gas "constant". After expansion in the turbine, the gas cools in the regenerator and cooler and recombines with evolution of heat of those chemical reactions, a decrease of the mole number and a decrease of the gas "constant" to the minimum cycle value, the cycle is then repeated.

As we noted previously [1], a large gas constant of the working medium in the turbine, as compared to the gas "constant" in the compressor, increases the efficiency and reduces by as much as 30-40% the fraction of the power expended on gas compression in the compressor. The end result is a higher efficiency of the cycle than in inert-gas cycles, due to substantially smaller irreversible losses in the cycle.

There exists a large class of (chemically) reactive systems in which reversible chemical reactions are accompanied by increased number of moles [2, 3].

Thus dissociating aluminum chloride is of great interest as a potential heat carrier and working medium in power plants [1, 4].

Under normal conditions, anhydrous aluminum chloride is a white crystalline substance.

The triple point of Al_2Cl_6 is $192.6^\circ C$ and 2.26 atm. abs. [7, 8].

At atmospheric pressure, aluminum chloride easily sublimates from the crystalline state at $180^\circ C$ [8].

Its critical temperature and pressure are 355 ± 1.5 °C [19, 20] and 26 ± 1 atm. abs., respectively.

The molecular weight of Al_2Cl_6 is 266.70 [13].

The heat of dissociation for the reaction $\text{Al}_2\text{Cl}_6 \rightleftharpoons 2\text{AlCl}_3$ is 108.5 kcal/kg [6, 7].

Unfortunately, experimental data on the thermodynamic properties of gaseous aluminum chloride are very scarce and data on its transport properties are not available at all.

Dimerization of aluminum chloride in the vapor phase with the formation of Al_2Cl_6 molecules was proven by Palmer and Elliot [5] and was confirmed by several other authors. Fischer et al. [6] showed that no noticeable amounts of higher polymers (of the Al_3Cl_6 type) are formed in aluminum chloride vapors.

On the basis of the measured densities of saturated vapors, the authors of [6, 7] have calculated the equilibrium constants for the vapor-phase dissociation of Al_2Cl_6 :



in the temperature ranges of 605–869°K [6] and 669–816°K [7] and the pressure range of 180–475 mm Hg [6].

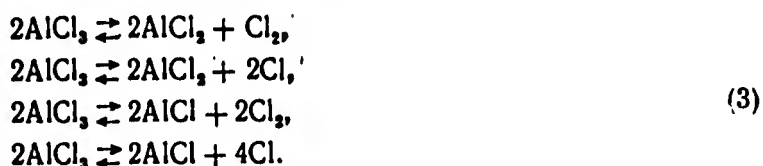
Very few data are available on the second stage of the dissociation of aluminum chloride. Heise and Willand [9] have studied the absorption spectra of AlCl formed in the homogeneous reaction



In the first series of tests they detected absorption lines of the AlCl radical above 1370°K. Repeated tests showed AlCl lines at considerably higher temperature (above 1430°K). As the authors admitted themselves [9], the results of the first test series are doubtful. Tests of the second series also are not very convincing since the AlCl was detected at the very limit of the temperature range accessible to the experimenters and could not be confirmed by a study of the intensity of AlCl lines at higher temperatures.

The possibility that AlCl_2 molecules are formed in aluminum chloride vapors at high temperatures cannot be confirmed by experimental data.

Since the nature of the second stage of the vapor-phase decomposition of Al_2Cl_6 has not been clarified, while almost no data on the equilibrium constant of the second stage are available; we have made a rough calculation of the isobaric-isothermal potential and of the equilibrium constants of the following equilibrium reactions, which are assumed to be possible in the second-stage dissociation of aluminum chloride at high temperatures:



tion α of the corresponding reactions were estimated. This estimate shows that decomposition of the AlCl_3 molecule can be neglected below approximately 1800°K (at 1 atm).

Henceforth, one calculations of thermodynamic properties of gaseous aluminum chloride were carried out by taking into account only the first stage of Al_2Cl_6 dissociation [Reaction (1)].

Insofar as the data on the equilibrium constant given in [6, 7] cover a small range of temperatures and pressures, we have calculated the equilibrium dissociation constants of aluminum chloride on the basis of the van't Hoff law, using the heat capacity values of Al_2Cl_6 and AlCl_3 .

The authors of [10], referring to the work of K. Kelley, give an equation for the heat capacity of gaseous AlCl_3 in between 298 and 1800°K:

$$c_p = 19.8 - 2.64 \cdot 10^5 T^{-2} \text{ cal/mol-deg.}$$

The error incurred by the use of this equation is estimated at 4%.

A similar equation is given in [12], but the temperature range is broader (453.3-2500°K).

In our further calculations we used as a basis the heat capacity data calculated in [11] from spectroscopic data, whereby experimentally-determined frequency characteristics were used in the case of Al_2Cl_6 , while the frequency characteristics for AlCl_3 were obtained by comparison with the AlF_3 molecule. Unfortunately, it is not possible to estimate the accuracy of these data in view of the absence of any experiments on the determination of the heat capacity of aluminum chloride.

The heat capacity values from [11] were recomputed on a "Minsk-2" electronic digital computer, and the following polynomials were obtained:

$$c_p = a + bT + cT^2 + dT^3 + eT^4 + fT^5. \quad (5)$$

The corresponding coefficients for Al_2Cl_6 and AlCl_3 are listed in Table 1.

Table 1. Coefficients of Formula (5)

Molecule	a	b · 10 ⁴	c · 10 ⁸	d · 10 ⁸	e · 10 ¹¹	f · 10 ¹⁶
Al_2Cl_6	24,807929	+7,014366	-10,977460	+8,589349	-3,299125	+4,947072
AlCl_3	10,810480	+3,302241	-5,109850	+3,968190	-1,515227	+2,265114

The values calculated from polynomials (5) differ by less than 0.5% from the results given in [11] and by less than 1% from the results given in [10, 12].

A review of experimental data on the standard heat of formation of gaseous AlCl_3 is given in [13], where the magnitude of the standard entropy for gaseous AlCl_3 is also calculated (Table 2).

Table 2. Standard Thermodynamic Parameters of Aluminum Chloride

1) Величины	2) Al_2Cl_6 (газ)	Литературный 3) источник	2) AlCl_3 (газ)	Литературный 3) источник
4) Теплота образования при 298°K ΔH_{298}° , ккал/моль	$-304600 \pm$ ± 2500	[10]	-142593	[13]
5) Энтропия при стандартной темпе- ратуре S_{298}° , $\frac{\text{ккал}}{\text{моль} \cdot \text{град}}$	113,769	[11]	75,564	[13]

KEY: 1) Parameters; 2) Al_2Cl_6 (gas); 3) Literature sources;
4) Heat of formation at 298°K ΔH_{298}° , cal/mole; 5) Entropy at
Standard temperature S_{298}° , cal/mole-deg.

Among the values for the heat of formation of gaseous Al_2Cl_6 listed in [10, 14, 15], we prefer the value given by Kubaschewski and Evans [10].

The standard entropy of gaseous Al_2Cl_6 used in our computation has been calculated by the authors of [11]. By comparing the values of S_{298}° [18] of octatomic molecules (Fig. 1), we obtained a similar value of 111.0 cal/mole-deg. which confirms the correctness of the computations given in [11].

In view of the fact that the dissociation of Al_2Cl_6 via Reaction (1) has not been investigated to a sufficient extent, and also in view of the absence (according to our calculations) of a high-temperature dissociation of AlCl_3 , the thermodynamic properties of gaseous aluminum chloride were calculated under the following assumptions:

- 1) the equilibrium constant does not depend on the pressure;
- 2) the second stage of the reaction can be neglected;
- 3) the dissociating gas is an ideal gas and its equation of state is

$$Pv = \frac{RT}{M_2} (1 + \alpha); \quad (6)$$

- 4) the reaction follows an equilibrium path.

The thermodynamic properties of dissociating aluminum chloride were determined via a modified method for computing the "effective" properties of a gas undergoing equilibrium dissociation [3].

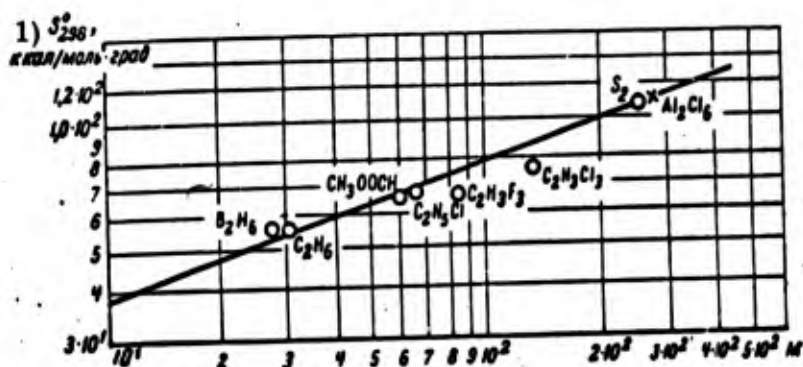


Fig. 1. Standard entropy of gaseous octatomic molecules:

$$\lg S_{298}^0 = 0.3281 \lg M + 1.249; \quad M_{\text{Al}_2\text{Cl}_6} = 266.7;$$

$$\lg S_{298}^0 = 2.045; \quad S_{293}^0 = 111 \text{ kcal/mole-deg}$$

KEY: 1) kcal/mole-deg.

The change in the number of moles during dissociation of Al_2Cl_6 via Reaction (1) can be written as:

$$(1 - \alpha) \rightleftharpoons 2\alpha,$$

whence the total number of moles is $n_\alpha = 1 + \alpha$.

Assuming that the components of the mixture are ideal gases, the equations of state for the components and for the mixture (on the basis of 1 mole) are:

$$\begin{aligned} P_2 v &= (1 - \alpha) RT, \\ P_1 v &= 2\alpha RT, \\ P v &= (1 + \alpha) RT, \end{aligned} \tag{7}$$

whence the partial pressures

$$P_2 = \frac{1 - \alpha}{1 + \alpha} P, \quad P_1 = \frac{2\alpha}{1 + \alpha} P$$

and the equilibrium constant of dissociation of Al_2Cl_6 via Reaction (1)

$$K_p = \frac{P_1^2}{P_2} = \frac{4\alpha^3}{1 - \alpha^2} P. \tag{8}$$

For convenience, all quantities referring to Al_2Cl_6 are henceforth denoted by the subscript 2, and those referring to AlCl_3 by the subscript 1.

By virtue of assumption 3) above, concerning the ideal nature of the gas mixture, we have, according to [16]

$$I_{\text{eff}} = \sum x_j I_j, \quad (9)$$

$$S_{\text{eff}} = \sum x_j S_j^0 - R x_j \ln x_j - R \ln \frac{P}{P_0}. \quad (10)$$

For the j-th component

$$I_j = I_{0j}^0 + \int_{T_0}^T c_p dT, \quad (11)$$

$$S_j = S_{0j}^0 + \int_{T_0}^T c_p \frac{dT}{T}. \quad (12)$$

For the equilibrium system (1) with allowance for dissociation

$$\begin{aligned} I_{\text{eff}} &= \frac{1}{M_2} \left[I_2 \frac{1-\alpha}{1+\alpha} + I_1 \frac{2\alpha}{1+\alpha} \right] = \\ &= \frac{1}{M_2} [I_2(1-\alpha) + I_1 2\alpha], \end{aligned} \quad (13)$$

$$\begin{aligned} S_{\text{eff}} &= \frac{1}{M_2} [S_2(1-\alpha) + S_1 2\alpha] - \\ &- \frac{R}{M_2} [(1-\alpha) \ln(1-\alpha) + 2\alpha \ln 2\alpha - \\ &- (1+\alpha) \ln(1+\alpha) - (1+\alpha) \ln P]. \end{aligned} \quad (14)$$

Here

$$I_2 = \Delta H_{298(2)}^0 + i_2, \quad I_1 = \Delta H_{298(1)}^0 + i_1; \quad (15)$$

$$S_2 = S_{298(2)}^0 + s_2, \quad S_1 = S_{298(1)}^0 + s_1. \quad (16)$$

Using the polynomials (5) obtained by us for the heat capacity of gaseous Al_2Cl_6 and AlCl_3 , we find, after integration

$$i_2 = a_2(T-298) + \frac{b_2}{2}(T^2-298^2) + \frac{c_2}{3}(T^3-298^3) + \frac{d_2}{4}(T^4-298^4) + \frac{e_2}{5}(T^5-298^5) + \frac{f_2}{6}(T^6-298^6), \quad (17)$$

$$s_2 = a_2 \ln \frac{T}{298} + b_2(T-298) + \frac{c_2}{2}(T^2-298^2) + \frac{d_2}{3}(T^3-298^3) + \frac{e_2}{4}(T^4-298^4) + \frac{f_2}{5}(T^5-298^5) \quad (18)$$

and analogous expressions for i_1 and S_1 .

Solving equations (13)-(18) simultaneously, we obtain computational formulas for the "effective" enthalpy and entropy of the dissociating system $\text{Al}_2\text{Cl}_6 \rightleftharpoons 2\text{AlCl}_3$:

$$I_{\text{eff}} = \frac{1}{M_2} [(1-\alpha)(\Delta H_{298(2)}^0 + i_2) + 2\alpha(\Delta H_{298(1)}^0 + i_1)], \quad (19)$$

$$S_{\text{eff}} = \frac{1}{M_2} [(1-\alpha)(S_{298(2)}^0 + s_2) + 2\alpha(S_{298(1)}^0 + s_1)] - \frac{R}{M_2} [(1-\alpha) \ln(1-\alpha) + 2\alpha \ln 2\alpha - (1+\alpha) \ln(1+\alpha) + (1+\alpha) \ln p]. \quad (20)$$

To calculate the degree of dissociation, α , occurring in (19) and (20), the values of the equilibrium constants were determined as follows:

Using the well-known van't Hoff relationship [15]

$$\frac{\partial \ln K_p}{\partial T} = \frac{Q_p}{RT^2}, \quad (21)$$

we obtain, after integration

$$\ln K_p = \frac{1}{R} \int \frac{Q_p}{T^2} dT + \text{const.}, \quad (22)$$

where

$$Q_p = 2I_1 - I_2 = 2\Delta H_{298(1)}^0 - \Delta H_{298(2)}^0 + 2i_1 - i_2. \quad (23)$$

Solving (17) and (22) simultaneously and using the relation

$$\ln K_p = - \frac{\Delta z}{RT} \quad (24)$$

at $T = 298^\circ\text{K}$, we can determine the value of the constant in (22). Then (22) will assume the form

$$\begin{aligned} \ln K_p = & \frac{1}{R} \left\{ 2S_{298(1)}^0 - S_{298(2)}^0 - \frac{1}{T} (2\Delta H_{298(1)}^0 - \Delta H_{298(2)}^0) + \right. \\ & + (2a_2 - a_1) \left[\ln \frac{T}{298} - \left(1 - \frac{298}{T} \right) \right] + \frac{2b_2 - b_1}{2} \frac{(T - 298)^2}{T} + \\ & + \frac{2c_2 - c_1}{3} \left[\frac{T^2 - 298^2}{2} - \left(1 - \frac{298}{T} \right) \right] + \\ & + \frac{2d_2 - d_1}{4} \left[\frac{T^3 - 298^3}{3} - \left(1 - \frac{298}{T} \right) \right] + \\ & + \frac{2e_2 - e_1}{5} \left[\frac{T^4 - 298^4}{4} - \left(1 - \frac{298}{T} \right) \right] + \\ & \left. + \frac{2f_2 - f_1}{6} \left[\frac{T^5 - 298^5}{5} - \left(1 - \frac{298}{T} \right) \right] \right\}. \end{aligned} \quad (25)$$

By differentiation of (19), we can also obtain an expression for calculating the "effective" heat capacity of the dissociating Al_2Cl_6

$$\begin{aligned} c_{p,\text{eff}} = & \left(\frac{\partial I_{\text{eff}}}{\partial T} \right)_p = \frac{1}{M_2} \left[\left(\frac{\partial \alpha}{\partial T} \right)_p (2\Delta H_{298(1)}^0 - \right. \\ & \left. - \Delta H_{298(2)}^0) + 2i_1 - i_2 + c_{p_2}(1 - \alpha) + c_{p_1} 2\alpha \right]. \end{aligned} \quad (26)$$

After differentiating (8) and substitution into (21), we obtain

$$\left(\frac{\partial \alpha}{\partial T} \right)_p = \frac{Q_p}{2RT^2} \alpha(1 - \alpha^2). \quad (27)$$

Using (23) and (27), it is easy to transform (26) into

$$c_{p,\text{eff}} = \frac{1}{M_2} \left[c_{p_2}(1 - \alpha) + c_{p_1} 2\alpha + \frac{\alpha(1 - \alpha^2)}{2RT^2} Q_p^2 \right], \quad (28)$$

where c_{p2} and c_{p1} are calculated from the polynomials (5).

The relations thus found allowed the computation, on the "Minsk-2" electronic digital computer, of the "effective" enthalpy, entropy and heat capacity of dissociating

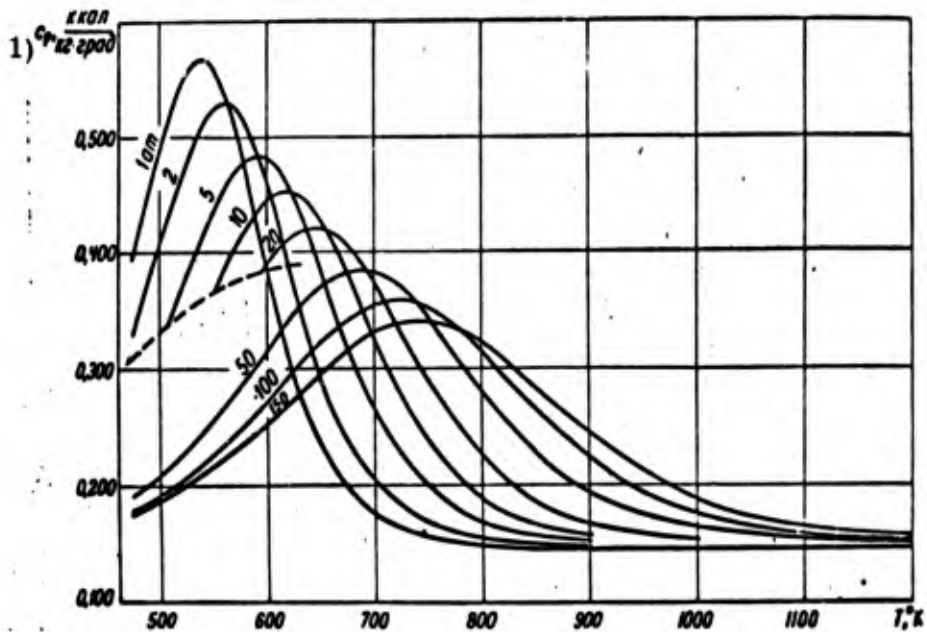


Fig. 2. Effective heat capacity of gaseous, dissociating aluminum chloride.

KEY: 1) c_p , kcal/kg-deg.

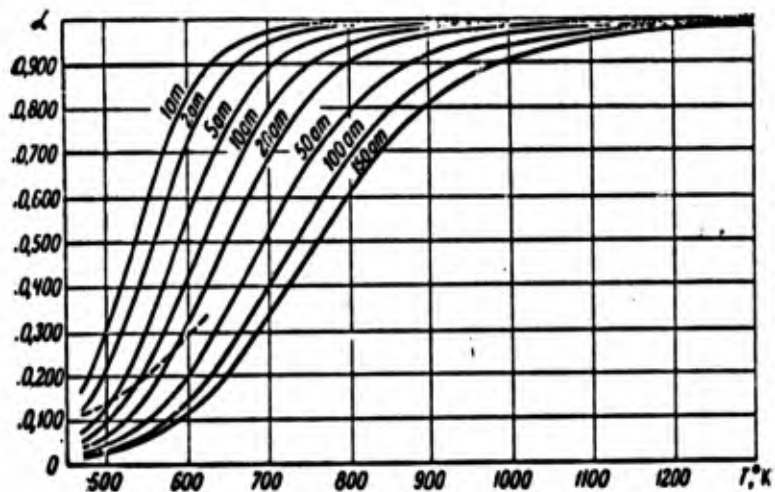


Fig. 3. T-diagram of dissociating gaseous aluminum chloride.

aluminum chloride in the pressure range of 1-150 atm and the temperature range of 460-2000°K (some of the results are given in Table 3).

On the basis of the values so obtained, an I-S diagram was plotted. The plot of $c_p^{\text{eff}} = f(P, T)$ is shown in Fig. 2, and the composition diagram for the system $\text{Al}_2\text{Cl}_6 \rightleftharpoons 2\text{AlCl}_3$ at various temperatures is given in Fig. 3. The critical pressure of the Al_2Cl_6 system was calculated from approximate relations and the saturation curve is shown on the plots by broken lines.

Table 3. "Effective" Enthalpy and Entropy of the System $Al_2Cl_6 \rightleftharpoons 2AlCl_3$

T, °K	P, atm.											
	1		2		4		6		8		10	
	$-I_{eff}$	S_{eff}	$-I_{eff}$	S_{eff}	$-I_{eff}$	S_{eff}	$-I_{eff}$	S_{eff}	$-I_{eff}$	S_{eff}	$-I_{eff}$	S_{eff}
500	1091.31	0.5494	1097.02	0.5315	1101.25	0.5169	1084.65	0.5449	1087.08	0.5377	1067.84	0.5690
550	1063.77	0.5545	1072.79	0.5776	1080.73	0.5560	1061.52	0.5851	1065.15	0.5759	1045.50	0.6048
600	1040.08	0.6432	1047.42	0.6218	1056.22	0.5986	1039.23	0.6208	1042.68	0.6119	1026.78	0.6326
650	1025.34	0.6669	1029.05	0.6513	1034.85	0.6329	1022.51	0.6456	1024.77	0.6385	1013.03	0.6516
700	1015.40	0.6816	1017.03	0.6691	1019.96	0.6550	1010.63	0.6621	1011.86	0.6563	1002.55	0.6651
750	1007.17	0.6930	1007.91	0.6817	1009.31	0.6697	1001.25	0.6742	1001.92	0.6692	993.72	0.6758
800	999.52	0.7029	999.88	0.6921	1000.58	0.6810	993.00	0.6842	993.36	0.6795	985.64	0.6851
850	992.07	0.7119	992.26	0.7014	992.63	0.6906	985.23	0.6931	985.44	0.6886	977.93	0.6934
900	984.70	0.7203	984.80	0.7099	985.02	0.6993	977.67	0.7013	977.80	0.6933	970.38	0.7011
950	977.35	0.7283	977.42	0.7179	977.55	0.7074	970.22	0.7089	970.30	0.7045	955.51	0.7153
1000	970.02	0.7358	970.05	0.7254	970.14	0.7150	955.43	0.7230	955.47	0.7187	940.72	0.7282
1100	955.34	0.7498	955.36	0.7394	955.40	0.7291	940.68	0.7358	940.70	0.7315	925.95	0.7400
1200	940.63	0.7626	940.64	0.7522	940.66	0.7419	925.92	0.7476	925.94	0.7433	911.16	0.7509
1300	925.89	0.7744	925.90	0.7640	925.91	0.7537	911.14	0.7646	911.15	0.7543	896.35	0.7612
1400	911.13	0.7853	911.13	0.7750	911.14	0.7646	896.34	0.7749	896.35	0.7645	881.54	0.7708
1500	896.33	0.7955	896.33	0.7852	896.34	0.7749	881.53	0.7844	881.53	0.7741	866.72	0.7797
1600	881.52	0.8051	881.52	0.7947	881.53	0.7844	866.71	0.7934	866.71	0.7831	851.19	0.7882
1700	866.70	0.8140	866.71	0.8037	866.71	0.7934	851.89	0.8019	851.90	0.7958	837.09	0.7962
1800	851.89	0.8225	851.89	0.8122	851.89	0.8019	837.08	0.8099	837.08	0.7955	822.26	0.8033
1900	837.08	0.8305	837.08	0.8202	837.08	0.8099	822.25	0.8175	822.25	0.8071		
2000	822.25	0.8381	822.25	0.8278	822.25	0.8175						

Table 3 (cont'd)

T, °K	P, atm.											
	20		40		60		80		100		150	
	-I _{eff}	S _{eff}	-I _{eff}	S _{eff}	-I _{eff}	S _{eff}	-I _{eff}	S _{eff}	-I _{eff}	S _{eff}	-I _{eff}	S _{eff}
500	—	—	1108,50	0,4837	1109,12	0,4793	1109,49	0,4763	1109,75	0,4741	1110,14	0,4702
550	—	—	1096,31	0,5069	1097,73	0,5010	1098,58	0,4971	1099,16	0,4913	1100,07	0,4891
600	1075,24	0,5496	1081,03	0,5335	1083,72	0,5253	1085,36	0,5201	1086,49	0,5164	1088,25	0,5099
650	1054,44	0,5829	1062,71	0,5628	1066,89	0,5522	1069,53	0,5454	1071,38	0,5405	1074,35	0,5322
700	1034,34	0,6127	1043,09	0,5918	1048,22	0,5799	1051,68	0,5718	1054,23	0,5659	1058,46	0,5557
750	1018,00	0,6353	1025,05	0,6168	1029,92	0,6052	1033,55	0,5969	1036,38	0,5905	1041,42	0,5792
800	1005,47	0,6514	1010,25	0,6359	1014,03	0,6257	1017,12	0,6181	1019,70	0,6120	1024,66	0,6009
850	995,41	0,6637	998,41	0,6502	1001,01	0,6415	1003,30	0,6349	1005,32	0,6295	1009,50	0,6193
900	986,65	0,6737	988,52	0,6616	990,23	0,6538	991,80	0,6480	993,25	0,6433	996,44	0,6342
950	978,55	0,6825	979,73	0,6711	980,86	0,6640	981,92	0,6587	982,92	0,6545	985,23	0,6463
1000	970,78	0,6905	971,56	0,6795	972,31	0,6728	973,03	0,6678	973,73	0,6639	975,37	0,6565
1100	955,69	0,7049	956,06	0,6842	956,42	0,6879	956,78	0,6833	957,13	0,6797	957,98	0,6730
1200	940,82	0,7178	941,02	0,7073	941,22	0,7011	941,41	0,6967	941,60	0,6933	942,07	0,6869
1300	926,00	0,7297	926,12	0,7192	926,24	0,7131	926,36	0,7088	926,47	0,7054	926,76	0,6991
1400	911,20	0,7406	911,27	0,7303	911,35	0,7242	911,42	0,7198	911,50	0,7165	911,68	0,7103
1500	896,38	0,7508	896,43	0,7405	896,48	0,7344	896,53	0,7301	896,59	0,7268	896,71	0,7206
1600	881,56	0,7604	881,59	0,7501	881,63	0,7440	881,67	0,7397	881,70	0,7364	881,79	0,7303
1700	866,73	0,7694	866,76	0,7591	866,79	0,7530	866,81	0,7487	866,84	0,7454	866,91	0,7393
1800	851,91	0,7779	851,93	0,7675	851,92	0,7615	851,98	0,7572	852,00	0,7539	852,05	0,7478
1900	837,10	0,7859	837,11	0,7755	837,12	0,7695	837,15	0,7652	837,16	0,7619	837,21	0,7553
2000	822,26	0,7935	822,28	0,7832	822,29	0,7772	822,31	0,7728	822,32	0,7695	822,35	0,7634

The authors wish to thank Eng. I. S. Zakharova, who performed calculations on the "Minsk-2" computer, and to Academician A. K. Krasin (Acad. Sci. Beloruss. SSSR) for his concern and interest in our work.

NOTATION

Δ_z - isobaric-isothermal potential of the reaction $\text{Al}_2\text{Cl}_6 \rightleftharpoons 2\text{AlCl}_3$, at temperature T;
P - total pressure of the mixture; v - specific volume of the mixture; R - gas constant;
 α - degree of dissociation of $\text{Al}_2\text{Cl}_6 \rightleftharpoons 2\text{AlCl}_3$, at P and T; M_2 - molecular weight of Al_2Cl_6 ;
 I_{0j}^0 and S_{0j}^0 - enthalpy and entropy of components at standard values P_0 and T_0 ; i_1 ; S_1 ;
 i_2 ; S_2 - relative enthalpy and entropy of the AlCl_3 and Al_2Cl_6 components.

REFERENCES

1. Nesterenko, V. B. and V. P. Bubnov. Vestsi AN Belarus, SSR, Ser. fiz.-tekhn. navuk; No. 1 (1966).
2. Nesterenko, V. B. and L. V. Mishina. Vestsi AN Belarus, SSSR, Ser. fiz.-tekhn. navuk; No. 2 (1965).
3. Nesterenko, V. B., M. A. Bazhin and V. P. Bubnov. Vesti AN Belarus, SSSR, Ser. fiz.-tekhn. navuk; No. 3 (1966).
4. Garay, P. N. Power, No. 9, p. 73 (1960).
5. Palmer, Elliot. J. Amer. Chem. Soc., No. 60, p. 1852 (1938).
6. Fischer, W., O. Rahlfs and B. Benze. Z. anorg. Chem., Vol. 205, No. 1 (1932).
7. Smits, A. and I. Mejerling. Z. physik. Chem., Vol. B41, No. 98 (1938).
8. Landolt-Börnstein. Physikalisch-chemische Tabellen, Dritter Ergänz, I, 257 (1935).
9. Heise, M. and K. Willand. Helv. Chim. acta, Vol. 34, No. 267, p. 2182 (1951).
10. Kubaschewski, O. and E. Evans. Metallurgische Thermochemie. Berlin (1959).
11. IANAF, Thermochemical Data (1963).
12. Termodinamicheskiye svoystva neorganicheskikh veshchestv. [Thermodynamic properties of inorganic compounds], Edited by A. P. Zefirov, Atomizdat, Moscow (1965).
13. Termodinamicheskiye svoystva individual'nykh veshchestv. [Thermodynamic properties of individual compounds]. Edited by Glushko, Vol. 1, 2. Izd. AN SSSR, Moscow (1962).
14. Rossini, F. D. et al. Selected Values of Chemical Thermodynamic Properties (circular of the National Bureau of Standards, I-III (1952).

15. Britske, E. V., A. F. Kapustinskiy et al. Termicheskiye konstanty neorganicheskikh veshchestv [Thermal constants of inorganic compounds], Moscow-Len'ngard (1949).
16. Dodge, B. F. Khimicheskaya termodinamika [Chemical thermodynamics], ONTI, Moscow-Leningrad (1952).
17. Karapet'yants, M. Kh. Khimicheskaya termodinamika [Chemical thermodynamics], Goskhimizdat, Moscow-Leningrad (1953).
18. Venner, R. Termokhimicheskiye raschety [Thermochemical computations], IL (1950).
19. Rotinjanz and Suchodski. Z. physik. Chem., Vol. 87, p. 635 (1914).
20. Nisel'son, L. A. and T. D. Sokolova. Zh. neorg. khimii; Vol. 10, No. 7, p. 1516 (1965).
21. Denisova, N. D. Dissertation GIREDMET (State Institute of Rare Metals), Moscow (1967).

Institute of Nuclear Power
Engineering, Acad. Sci. Beloruss. SSSR

Received
17 January 1967

HEAT TRANSFER IN THE CHEMICALLY-REACTING EQUILIBRIUM SYSTEM $N_2O_4 \rightleftharpoons 2NO_2$

V. B. Nesterenko, B. Ye. Tverkovkin and A. B. Verzhinskaya

Heat transfer in a chemically-reacting flow has still not yet been investigated to a sufficient extent, although it is a very urgent problem in aircraft and rocket technology, and is a potential means for increasing the rate of heat processes in power- and chemical-engineering [6, 8-10]. The presently available analytic methods for computing heat exchange in a reacting flow system are extremely complex and laborious, while the great majority of studies in this field are concerned with the case of external flow around bodies. For engineering computations of heat-exchange apparatus, sufficient accuracy is achieved by the use of empirical relations. In this case, thermal and fluid flow computations for ducts (channels) are performed on one-dimensional models, where the (effective) coefficients of heat and mass transfer qualitatively allow for the specific nature of the one-dimensional description of real three-dimensional processes [7, 11].

The theory of diffusion and heat transfer shows that processes involving heat transfer via conduction and mass transfer by diffusion are similar. Theoretical and experimental results obtained in the study of molecular conduction can be applied directly to diffusion processes and vice-versa [1, 2].

The basic law of heat transfer via molecular (heat) conduction is Fourier's law

$$q = -\lambda_f \text{grad } T. \quad (1)$$

The corresponding law for diffusion is Fick's law

$$j_k = -D_k \text{grad } c_k. \quad (2)$$

Equations (1) and (2) hold rigorously only in the case of heat transfer in a medium with a constant and identical at all points concentration, and in isothermal diffusion. If both concentration and temperature gradients are present in the system, then one deals with phenomena, known as thermodiffusivity and diffusive heat conduction. In this case, one more term, proportional to the concentration gradient, is introduced into Eq. (1), and a term proportional to the temperature gradient is introduced into Eq. (2). However, the contribution made by cross-effects is usually small, and in most cases such effects can be neglected.

In the computation of processes involving transfer of heat and mass, one uses the heat transfer coefficient α and the mass transfer coefficient β :

$$\alpha_f = \frac{q}{\Delta T}, \quad (3)$$

$$\beta_k = \frac{j_k}{\Delta c_k}, \quad (4)$$

where ΔT , Δc_k are the temperature and concentration differences, respectively.

The heat and mass transfer coefficients are defined with the aid of criteria of similarity:

$$Nu_l = \frac{\alpha_l L}{\lambda_l}, \quad (5)$$

$$Nu_{lk} = \frac{\beta_k L}{D_k}. \quad (6)$$

The similarity theory proves that in forced convection, the Nusselt number is a function of the Reynolds and Prandtl numbers at specific geometric and physical conditions:

$$Nu = f(Re, Pr).$$

The only difference between computation of a heat-transfer process and computation of diffusional process is that, in the first case, one uses the Prandtl thermal number, while in the latter case one uses the Prandtl diffusion number [2]:

$$Pr_l = \frac{\nu}{a}, \quad (7)$$

$$Pr_{lk} = \frac{\nu}{D_k}, \quad (8)$$

where D_k is the kinematic diffusion coefficient (m^2/sec).

To compute heat transfer in a chemically reacting equilibrium flow we can write

$$q = q_l + q_r. \quad (9)$$

where

$$q_l = \alpha_l (T_{ct} - T_r); \quad (10)$$

$$q_r = \sum_{k=1}^n h_k \beta_k (x_{kct} - x_k). \quad (11)$$

The molar concentrations x_{kct} , x_k of each component in an equilibrium mixture are functions of the temperature and pressure and are determined, respectively, from the wall temperature and the mean calorimetric temperature of the flow.

Consequently, by analogy with the determination of thermophysical properties of a chemically-reacting mixture, the heat transfer coefficient can also be represented in the form of two additive components

$$\alpha_e = \alpha_l + \alpha_r, \quad (12)$$

where α_e is the "effective" heat transfer coefficient, defined by the equation

$$\alpha_e = \frac{q}{T_{cr} - T_r}; \quad (13)$$

α_f is the "frozen" component of the heat transfer coefficient, defined by the usual analytic or empirical relations for a "frozen" mixture; α_r is the "reactive" component of the heat transfer coefficient, defined by the equation

$$\alpha_r = \frac{\sum_{k=1}^n h_k \beta_k (x_{kcr} - x_k)}{T_{cr} - T_r}. \quad (14)$$

If μ chemical reactions take place in a multicomponent mixture, then in each j -th chemical reaction it is possible to select a component which does not participate in any other reaction. We shall index this component with the ordinal number of the j -th chemical reaction. Then we can write for the diffusional mass flow of each component

$$\frac{j_k}{m_k} = \sum_{j=1}^n \frac{a_{jk}}{a_{jj}} \frac{j_j}{m_j}. \quad (15)$$

Equation 15) is obtained from the diffusion equation at static conditions.

Using the fact that

$$Q_{pj} = \sum_{k=1}^n a_{jk} m_k h_k,$$

it is not difficult to find that the energy flow due to diffusion is

$$j_r = \sum_{k=1}^n h_k j_k = \sum_{j=1}^n \frac{Q_{pj} j_j}{a_{jj} m_j}. \quad (16)$$

To compute the heat transfer coefficient in a unidimensional, chemically-reacting equilibrium flow with a stoichiometric ratio of molar diffusional flows in the radial direction, we can write

$$\alpha_r = \sum_{j=1}^n \frac{Q_{pj} \beta_j}{a_{jj} m_j} \frac{x_{jcr} - x_j}{T_{cr} - T_r}. \quad (17)$$

The thermophysical properties of the reversibly dissociating nitrogen tetroxide equilibrium system, necessary for the calculation of α_c (the thermal conductivity λ , of the "frozen" mixture, the coefficient of dynamic viscosity μ), were determined from the equations obtained in [3, 4]. The kinematic diffusion coefficient D_k was calculated

from the well-known equation of mass transfer theory

$$D_k = \frac{1-x_k}{\sum_{\substack{j=1 \\ j \neq k}}^n x_j / D_{kj}} \frac{m_k m_j}{m^2} \quad (18)$$

where the generalized diffusion coefficients D_{hj} were determined from the equations obtained in [5].

The method described above for computing the heat transfer coefficient for an equilibrium mixture was used to generalize the experimental data on heat transfer in the system $N_2O_4 \rightleftharpoons 2NO_2$ during heating. In this study, we used the theoretical dependence of the Nusselt numbers on the Reynolds and Prandtl numbers in a forced turbulent pipe flow, postulated by M. A. Mikheyev and S. S. Kutateladze

$$Nu = 0,021 Re^{0,8} Pr^{0,43} \left(\frac{T_{cr}}{T_r} \right)^{-0,55} \quad (19)$$

Description of test units. Experimental studies of heat transfer to a chemically-reacting gas were carried out at the Institute of Nuclear Power Engineering of the Academy of Sciences of the Belorussian SSSR in the temperature range of 30-300°C, pressure range of 1-60 atm. abs, and Reynolds number range of 10^4 - $3 \cdot 10^5$. In the unit used for studying heat transfer to a dissociating gas at low pressure, the tests were carried out in the pressure range of 1-10 atm ($Re = 10^4 - 3 \cdot 10^4$) [6]. A detailed description of this test unit is given in [6]. In the unit used for studying chemically reactive media in the high-pressure region the tests were carried out in the pressure range of 10-60 atm at Re of $3 \cdot 10^4 - 3 \cdot 10^5$. Steady-state conditions prevailed in both units during the study of the heat transfer.

The unit for the experimental study of media which are chemically-reactive at high pressures is a closed loop in which the liquid nitrogen tetroxide is circulated by a pump. The flow rate is regulated by varying the speed of the pump and by passing part of the flow through a by pass valve. The flow rate of the heat carrier in the loop is measured by the calorimetric method in a flow calorimeter [with self-recovery (compensation) of heat losses], placed in a Dewar flask. The temperature of the liquid at the inlet and outlet of the calorimeter is measured with thermocouples whose cold junctions are held at a constant temperature of 0°C. A carefully calibrated four-junction differential thermocouple, yielding twice the emf, is used for measuring the difference of temperatures at the inlet and outlet. Pressure in the system is created by filling it with the necessary amount of product; for this purpose, a special charging system is available. It consists of two preheated containers. The test unit is provided with a surge chamber for damping of pulsations created by the pump and with a mechanical filter for removing from the system any products of corrosion of the structural materials by N_2O_4 . All parts of the unit in contact with N_2O_4 are made from 1Kh18N9T stainless steel.

The test section (Fig. 1) consists of a v-shaped tube made of 1Kh18N9T stainless steel; it has an O.D. of 4 mm, wall thickness of 1 mm and total length of 700 mm. The tube itself also functions as an electrical conductor and heater. The hot-line conductor is attached to the middle of the tube, while the ground conductor is

attached to the flanged ends of the tube. Thus, it is not necessary to electrically insulate the test section of the tube from the rest of the system, thus avoiding a problem which is difficult in a system operating under these conditions. The section connecting the tube to the flange is finned to reduce heating of the flange. For protection against heat losses, screens are attached to the flange, while the jacket and flange surfaces are covered with heat-insulating material. A calorimeter is installed to measure the heat losses.

In the test section, we measured the inlet and outlet temperatures of the heat carrier, and also the wall temperatures along the tube (at equal intervals of 100 mm). Special thermometric wells were installed for measuring the inlet and outlet temperatures. The temperature sensors were inserted into internal tubes, over which the gas flowed. This arrangement prevents reaction between the thermocouples and the corrosive heat carrier and allows a simple replacement of the thermocouples. Chromel-alumel thermocouples with an electrode diameter of 0.3 mm were used as temperature sensors. The gas and wall temperatures in the test apparatus were measured with a P-307 potentiometer (accuracy of up to 1 microvolt). The pressure in the system was measured with reference manometers equipped with separators. These manometers were installed ahead and behind the test section and also on the calorimeter-flowmeter. The electric power liberated in the heating tube was measured from the readings of an ammeter and voltmeter. The current was measured with a type D-57 ammeter of 0.2% precision class through a type UTT-6 current transformer. The voltage was measured with voltmeters of type D-523 and E-59 (0.5% precision class). Switching in the voltage-measurement circuit was effected by means of a type PMT-12 switch. All heat transfer experiments were repeated 2-3 times.

Workup of test data. By definition, the local "effective" heat transfer coefficient α_e in steady state is given by equation (13).

The electric power supplied to the test section was calculated from the measured current and voltage data. Then, making allowance for losses to the surrounding medium, the heat supplied to the heat carrier in each section was determined. Knowing the profile of the heat flux along the length of the channel and also the initial conditions from measurements of the temperature and gas pressure at the inlet of the test section, the gas temperature profile along the length of the channel was calculated with the aid of equations obtained in [7]; then the "effective" heat transfer coefficient (α_e) in each section was calculated with the aid of (13). Experimental points located next to the start of the test section and behind the turn (see Fig. 1) and, consequently, lying in the region of hydrodynamic stabilization, were not taken into account in the calculations.

Results and discussion of experimental data. The table below gives calculated and experimental values of the "effective" heat transfer coefficient α_e for the chemically-reactive gaseous system $N_2O_4 \rightleftharpoons 2NO_2$ in turbulent flow in the tube. Comparison of calculated data, obtained from the empirical relations proposed in this study, with experimental data for the equilibrated chemically-reactive system $N_2O_4 \rightleftharpoons 2NO_2$ indicates a satisfactory correlation between these data, as well as the fact that the proposed empirical relations can be used to describe the mechanism of heat transfer in a chemically-reacting system.

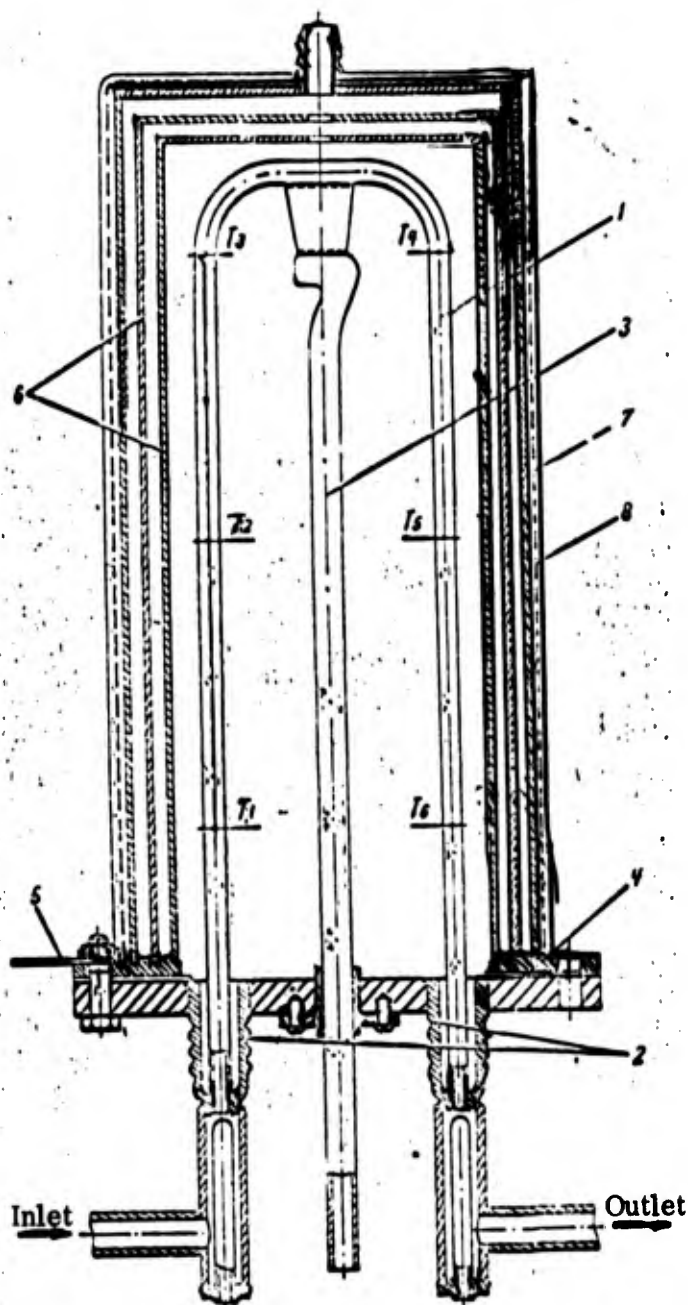


Fig. 1. Schematic of test apparatus:

1—test tube; 2—thermometric wells; 3—hot conductor;
 4—flange; 5—ground conductor; 6—screens; 7—jacket;
 8—calorimeter. T_1 - T_6 —thermocouples in tem-
 perature-measurement circuit.

Analysis of these experimental data on heat transfer shows that, in the case of turbulent flow in a circular tube, concentration gradient-induced diffusional heat transfer in the chemically reactive stream $N_2O_4 \rightleftharpoons 2NO_2$, in the form of chemical enthalpy, is high and a 3-9-fold increase (factor of improvement) of heat transfer coefficients is achieved, as compared to data obtained with an inert gas (CO_2 , N_2 and others).

Table 1. Comparison of Calculated "Effective" Heat Transfer Coefficients α_e with Experimental Results for the System $N_2O_4 \rightleftharpoons 2NO_2$

Наименование 1)	2) Номер опыта											
	3) 48 (P=2,5 атм), участки					4) 49 (P=5 атм), участки						
	1	2	3	4	5	6	7	8	9	10		
8) Температура стенки, °K	358	364	369	378	464	513	372	378	353	391	438	504
9) Температура газа, °K	344	349	355	361	404	423	355	361	366	372	395	419
10) Число Re · 10 ⁻⁴	2,98	2,81	2,73	2,65	2,27	2,09	3,14	3,06	3,65	2,87	2,59	2,32
11) Расчетные значения α_f	370	373	377	376	362	357	368	384	386	405	367	373
11) Расчетные значения α_{fI}	3150	2726	2340	2040	232	66	2290	2250	2030	2176	830	139
11) Расчетные значения α_e	3520	3093	2717	2416	614	422	2658	2634	2416	2575	1197	512
12) Экспериментальные значения α_e	3310	3080	3300	2710	750	496	2630	2630	2630	2330	1060	511

Наименование 1)	2) Номер опыта												
	5) 28 (P=53 атм), участки					6) 32 (P=42 атм), участки				7) 79 (P=22 атм), участки			
	1	2	3	5	6	2	3	5	6	2	3	5	6
8) Температура стенки, °K	468	508	506	590	632	476	586	542	561	477	493	529	543
9) Температура газа, °K	431	441	452	484	503	444	452	476	488	443	450	472	484
10) Число Re · 10 ⁻⁶	2,05	1,97	1,91	1,76	1,69	1,88	1,83	1,72	1,68	1,96	1,92	1,62	1,78
11) Расчетные значения α_f	3840	3740	3820	3740	3730	3720	3740	3700	3730	3810	3810	3850	3890
11) Расчетные значения α_{fI}	8320	4860	4370	4170	6130	5930	4720	1730	1160	4140	2950	1310	910
11) Расчетные значения α_e	12200	8600	8190	4190	4350	9660	8470	5430	4890	7960	6760	5150	4800
12) Экспериментальные значения α_e	12800	7190	8940	4530	3730	10300	9780	4950	4510	7510	5990	4500	4310

KEY: 1) Item; 2) No. of test; 3) 48 (p=2.5 atm) sections; 4) 49 (p=5 atm) sections; 5) 28 (p=53 atm); sections, 6) 32 (p=42 atm) sections; 7) 79 (p = 22 atm) sections; 8) Wall temperature, k^o; 9) Gas temperature, k^o; 10) Rena No. Re · 10⁻⁴; 11) Calculated values; 12) Experimental values α_e .

At pressures of 40-50 atm. abs. and Re of $2 \cdot 10^5$, heat transfer coefficients of 10,000-13,000 kcal/m² - hr-deg were experimentally obtained in the chemically-reactive system.

Figure 2 compares calculated and experimental heat transfer data for in the reactive equilibrated system $N_2O_4 \rightleftharpoons 2NO_2$ in the Re range $10^4 - 2 \cdot 10^5$, P = 1-6 atm. abs. and T = 30-300 °C.

In the range investigated, the error in the experimental heat transfer coefficients is $\pm 15\%$. The accuracy of experimental data can be substantially improved after more reliable experimental data on the heat capacity of liquid nitrogen tetroxide have been obtained, since inaccuracies in the calorimetric determination of the N_2O_4 flow rate constitute the major fraction of the error.

In view of the satisfactory agreement between calculated and experimental data, it is possible to recommend the previously suggested unidimensional model for computing the axial temperature profile in a circular tube when studying heat transfer to a chemically-reactive gas [11], and also to recommend the empirical (numerical) relations for computing heat transfer in the chemically reactive system $N_2O_4 \rightleftharpoons 2NO_2$ in the temperature range of 30-300°C, pressure range of 1-60 atm. abs., and Re range of $10^4 - 3 \cdot 10^5$.

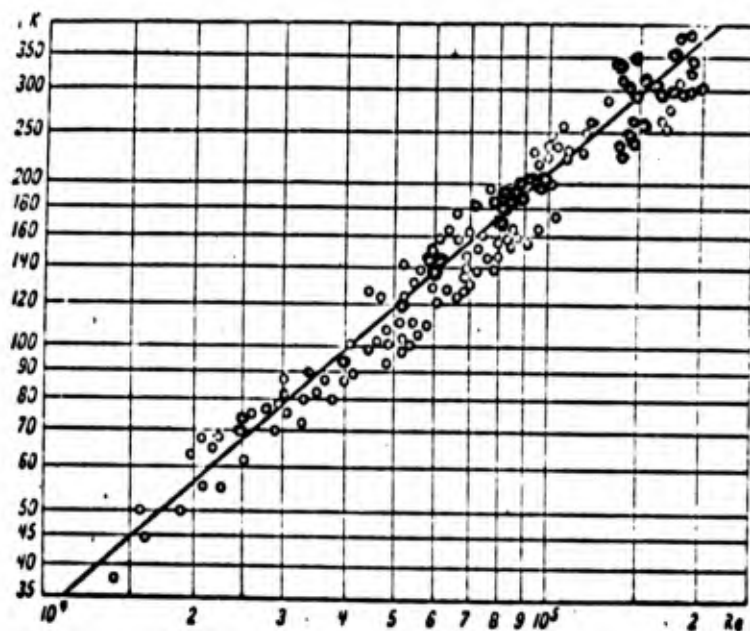


Fig. 2. Heat transfer for the forced turbulent flow of the chemically equilibrated system $N_2O_4 \rightleftharpoons 2NO_2$ in tube.

$$K = Nu_f Pr_f^{-0.43} \left(\frac{T_{ct}}{T_r} \right)^{0.55} = Nu_{jk} Pr_{jk}^{-0.43} \left(\frac{T_{ct}}{T_r} \right)^{0.55}$$

In conclusion, the authors wish to thank A. K. Krasin, member of the Academy of Sciences of the Belorussian SSSR, for his constant concern with their work, and also their department associates A. N. Kremeshnyy, Yu. I. Bakalin and V. A. Nikolayev, who took part in this work or in its discussion.

NOTATION

α_e - "effective" heat transfer coefficient; α_f , α_r - "frozen" and "reactive" components of the heat transfer coefficient; β_k - mass transfer coefficient; Nu_f , Nu_{jk} - Nusselt numbers for heat transfer and diffusional processes; Pr_f , Pr_{jk} - Prandtl numbers for heat transfer and diffusional processes; λ_f - thermal conductivity of "frozen" mixture; ν - kinematic viscosity coefficient; a - thermal diffusivity coefficient; D_k - kinematic diffusion coefficient; $D'_k = \rho D_k$ - dynamic diffusion coefficient; ρ - mixture density; L - characteristic dimension; h_k - specific enthalpy of each component; x_{kct} , x_k - molar concentrations of each component near and away from the wall; m_k - molecular weight of each component; Q_{pj} - heat of the j -th reaction; a_{kj} - stoichiometric factor; D_{kf} - generalized diffusion coefficient; q - specific heat flow per unit surface from wall to heat carrier in a given cross-section; T_{ct} - current (moving) temperature of channel wall; T_r - mean calorimetric flow temperature in given section.

REFERENCES

1. Levich, V.G. Fiziko-khimicheskaya gidrodinamika [Physicochemical Hydrodynamics], Fizmatgiz, Moscow (1959).
2. Frank-Kamenetskiy, D.A. Diffuziya i teploperedacha v khimicheskoy ykinetike [Diffusion and Heat Transfer in Chemical Kinetics], Izd. AN SSSR, Moscow-Leningrad (1947).
3. Maksimov, B.G., L.V. Mishina and B.D. Timofeyev. Vesti AN BSSR, ser. fiz.-tekhn. nauk, No. 1 (1966).
4. Mishina, L.V. and Yu.G. Kotelevskiy. Vesti AN BSSR, ser. fiz.-tekhn. nauk, No. 1 (1966).
5. Hirschfelder, D., C. Curtis and D. Bird. Molecular Theory of Gases and Fluids, Russian transl.
6. Bakalin, Yu.I., V.B. Nesterenko and A.I. Kremeshniy. Vesti AN BSSR, ser. fiz.-tekhn. nauk, No. 3 (1966).
7. Nesterenko, V.B. and B.E. Tverkovkin. Vesti AN BSSR, ser. fiz.-tekhn. nauk, No. 1 (1966).
8. Schotte, W. Ind. Eng. Chem.; Vol. 50, No. 4 (1958).
9. Beer, H. Heat transfer in a dissociating gas. Dissertation Institute of Thermodynamics of Aircraft and Rocket Engines, Stuttgart (1964).
10. Beer, H. Chemie-Ingenieur Technik; Vol. 377, pp. 1047-1054 (1965).
11. Nesterenko, V.B. and B.E. Tverkovkin. Vesti AN BSSR, ser. fiz.-tekhn. nauk, No. 2 (1966).

Institute of Nuclear Power Engineering,
Acad. Sci. Belorussian SSR

Received 17 January 1967

**HEAT TRANSFER IN THE CHEMICALLY-REACTING SYSTEM
 $2\text{NO}_2 \rightleftharpoons 2\text{NO} + \text{O}_2$ TAKING THE KINETICS OF THE
 CHEMICAL REACTION INTO ACCOUNT**

V. B. Nesterenko, B. E. Tverkovkin and A. B. Verzhinskaya

Previous studies of heat transfer in the presence of chemical reactions were concerned with systems in chemical equilibrium [1, 2, 5, 6]. When chemical reactions proceeding at a finite rate are present, the relations obtained for equilibrium flows are no longer satisfactory since they do not take the rate of the chemical process into account. The available analytic methods for computing heat transfer in a reacting flow are extremely complex and laborious even in the case of equilibrium systems. The calculation of heat transfer in chemically reactive media, taking the kinetics of chemical reactions into account, will require even more complex analytic methods, since the differential equations of heat transfer will include internal sources and mass flows of individual components, which exhibit a complex functional relationship.

For practical computations of heat-exchange units, designed for chemically-reactive heat carriers, a sufficient accuracy can be achieved by using empirical relations. For this reason, the aim of the theoretical and experimental study which we have undertaken is to obtain empirical relations for calculating the heat transfer coefficient in the case of a forced turbulent flow $2\text{NO}_2 \rightleftharpoons 2\text{NO} + \text{O}_2$ in a tube, taking the kinetics of chemical reactions into account. In this case, a nonequilibrium chemical process is understood to represent a process in which the Maxwell-Boltzmann distribution of reacting particles is not disturbed to any significant extent, i. e. the system is in a static equilibrium. This is caused by the difference in the rates of the processes in the system, i. e. by the fact that equilibrium with respect to the motion of molecules (thermodynamic equilibrium) sets in much more rapidly than the equilibrium of chemical processes.

In this study, the method of integral analogs [4], widely used in the theory of similarity, has been utilized for the determination of a similarity criterion which takes the nonequilibrium nature of a chemical reaction into account. This method consists in reducing the equations of a physical process to a nondimensional form. If we analyze the analytic relations which describe heat and mass transfer in the presence of chemical reactions, we can easily obtain the following nondimensional complex:

$$N_h = \frac{I_s L}{\rho \omega} \quad (1)$$

which is a measure of the ratio between the velocity of the source or the mass flow rate of each component and the convective transfer rate. The criterion (number) N_h is closely related in meaning to the criterion well-known in two-phase flow theory:

$$\frac{Nu}{K Pe} = \frac{q}{r \rho \omega} \quad (2)$$

which is a measure of the ratio between the phase conversion rate $q/r\rho$ and the flow rate of the given phase ω .

In a flow in chemical equilibrium the gas composition is a single-valued function of the temperature (it is assumed that the pressure p is constant). In this case, the

concentration gradients of individual components and, consequently, also the diffusion energy flows are uniquely dependent on the temperature gradient and can be determined without solving the diffusion equation. If the chemical reaction takes place at a finite rate, then not all molecules have a chance to react in view of the limited residence time in the channel. In this case, the concentration gradients of the components are determined by solving the diffusion equation and depend to a considerable extent on the profile of the axial component of the flow velocity w_z along the radius.

Variation of w_z from zero near the channel wall to specific values at the boundary of the flow core results in a variation, within a very wide range, of the residence time of individual gas particles in a radial direction. For this reason, the nonequilibrium character of the chemical reaction will become more pronounced as one draws closer to the flow core, which the gas composition will vary from equilibrium values near the channel wall to values reflecting a considerable deviation of the gas composition from equilibrium values near the flow core.

The Nusselt number Nu_{jk} for the diffusion process in a chemically-reactive flow, when one nonequilibrium reaction is present, is a function of the following numbers (criteria):

$$Nu_{jk \text{ non eq.}} = f(Re, Pr_{jk}, N_k). \quad (3)$$

The mass transfer coefficient β_k in this case is equal to

$$\beta_k = \frac{Nu_{jk \text{ non eq.}} D_k}{d_s} \quad (4)$$

The heat transfer coefficient α_e in a nonequilibrium flow was calculated from the relations given in [2]:

$$\begin{aligned} \alpha_e &= \alpha_f + \alpha_r, \\ \alpha_f &= \frac{Nu_f \lambda_f}{d_s}, \\ \alpha_r &= \frac{\sum_{k=1}^n h_k \beta_k (x_{k \text{ cr}} - x_k)}{T_{\text{cr}} - T_r}. \end{aligned} \quad (5)$$

of, if the calculation is carried out in terms of the thermal effects of the reaction, from the formula

$$\alpha_r = \sum_{j=1}^n \frac{Q_{rj} \beta_j}{a_{jj} n_j} \frac{x_{j \text{ cr}} - x_j}{T_{\text{cr}} - T_r}$$

The specific heat flow, q , per unit surface in the given section from the wall to the heat carrier is determined by the equations

$$q = \alpha_e (T_{cr} - T_r) \quad (6)$$

or

$$q = q_l + q_r,$$

where

$$q_l = \alpha_l (T_{cr} - T_r); \quad q_r = \sum_{k=1}^n h_k \beta_k (x_{k cr} - x_k) \quad (7)$$

or

$$q_r = \sum_{j=1}^n \frac{Q_{rj} \beta_j}{a_{jj} m_j} (x_{j cr} - x_j).$$

The thermophysical properties of the chemically-reactive system $2NO_2 \rightleftharpoons 2NO + O_2$, necessary for calculating α_e and q , were determined in the same manner as in [2], except that the composition of the mixture was determined with the aid of equations taking the kinetics of the chemical reaction into account.

The method, described above, for calculating the heat transfer coefficient in a chemical reaction was used to generalize experimental data on heat transfer during heating for the system $2NO_2 \rightleftharpoons 2NO + O_2$. Tests aimed at studying heat transfer in the case of a forced turbulent flow in a tube for the system $2NO_2 \rightleftharpoons 2NO + O_2$ were carried out at the Institute of Nuclear Power Engineering of the Academy of Sciences of the Belorussian SSR in the range of gas temperatures of 140-550°C, pressures of 10-60 atm. abs. and Reynolds number $Re = 3 \cdot 10^4 - 3 \cdot 10^5$. The test section and experimental method used in this case were the same as those used in the experimental study of heat transfer in the equilibrium system $N_2O_4 \rightleftharpoons 2NO_2$ [2].

In order to calculate the temperature profiles of the reacting flow along the length of the channel from the equations given in [3], it is necessary to know the initial conditions with respect to the flow temperature $T(z=0)$ and concentration of the component $O_2, \rho_{i0}(z=0)$. Since the concentration of the O_2 component was not determined in this case, tests aimed at studying heat transfer in the system $2NO_2 \rightleftharpoons 2NO + O_2$ were carried out under the condition that a binary mixture $N_2O_4 \rightleftharpoons 2NO_2$ was fed at the inlet of the test section, while the reaction $2NO_2 \rightleftharpoons 2NO + O_2$ took place already far away from the initial section.

From the analysis of experimental heat transfer data in the case of forced turbulent flow in a tube we obtained the following criterial dependence for the determining component:

$$Nu_{j4 \text{ non eq.}} = Nu_{j4} N_{k4}, \quad (8)$$

where Nu_{j4} is the Nusselt number defined in terms of the criterial dependence proposed by

M. A. Mikheyev

$$Nu_{jk} = 0,021 Re^{0,6} Pr_{jk}^{0,43} e_T; \quad (9)$$

e_T is a coefficient characterizing the variability of the physical parameters.

As recommended by S. S. Kutateladze

$$e_T = \left(\frac{T_{ct}}{T_r} \right)^{-0,55} \quad (10)$$

The criterion of chemical nonequilibrium was defined as

$$N_A = \frac{l_A z}{\rho w}, \quad (11)$$

where z is the moving coordinate;

$$l_A = \frac{m_1}{m_2^2} K_{CTIA} (\rho \rho_{20})^2; \quad 0 \leq N_A \leq 1.$$

In case $N_A > 1$, chemical equilibrium takes place and the "reaction" component α_r is calculated by the method described in [2] for equilibrium systems.

The composition of the mixture in the flow was determined with the aid of equations obtained in [3], taking the kinetics of the chemical reaction into account. The molar concentrations of components at the wall, x_{hct} , were determined from the wall temperature in the assumption that chemical equilibrium is established here ($w_{ct} = 0$). This assumption, which is valid for equilibrium systems, may not be fulfilled when the rate of the chemical reaction is finite. The decisive factor is the ratio of the chemical relaxation and diffusion times. However, even in this case, the assumption of a chemical equilibrium at the wall will not result in a large error, since under conditions close to a "frozen" flow the contribution made by the "reaction" component is infinitely small: $\alpha_r \approx 0$ since $N_A \approx 0$. The contribution made by α_r becomes significant when the number N_A increases. In this case, the rate of the chemical reaction is comparable with the rate of convective transfer. The chemical reaction rate will play a decisive role in regard to diffusion transfer in the layer located near the wall and also in the region where temperatures near the channel wall are higher than in the flow during heating, and the composition of the mixture will be close to the equilibrium composition.

Calculated and experimental values of the "effective" heat transfer coefficient α_e are given in the table.

The experimental study of heat transfer in the second stage of the reaction was carried out with residence times of the gas in the heated channel of 0.01-0.03 sec.

Comparison of calculated and experimental data shows a satisfactory agreement between these data and also that the proposed empirical relations can be used in describing

Table 1. Comparison of Calculated "Effective" Heat Transfer Coefficient α_e With Experimental Results for the $2NO_2 \rightleftharpoons 2NO + O_2$ System Taking the Kinetics of the Chemical Reactions Into Account

Наименование 1)	2) Номер опыта						
	3) 71 (P = 23 атм), участки				4) 46 (P = 33,5 атм), участки		
	2	3	5	6	3	5	6
8) Температура стенки, °K	632	710	802	802	891	921	942
9) Температура газа, °K	475	520	659	701	671	760	781
10) Число $Re \cdot 10^{-4}$	9,47	8,72	7,21	6,84	6,46	5,80	5,06
11) Число N_{O_2}	0	0	0,012	0,045	0,023	0,193	0,262
12) Расчетные значения α_f	2087	2146	2539	2705	2270	2490	2516
13) Расчетные значения α_{fI}	309	75	0	0	0	0	0
14) Расчетные значения α_{fII}	0	0	65	620	156	1453	1737
15) Расчетные значения α_e	2396	2222	2604	3325	2426	3943	4253
16) Экспериментальные значения α_e	2306	1906	2523	3684	2688	3948	3939

Наименование 1)	2) Номер опыта								
	5) 30 (P = 55,5 атм), участки			6) 31 (P = 57,5 атм), участки			7) 35 (P = 42 атм), участки		
	3	5	6	3	5	6	3	5	6
8) Температура стенки, °K	804	874	878	928	950	945	684	802	812
9) Температура газа, °K	557	709	726	646	743	766	512	642	656
10) Число $Re \cdot 10^{-4}$	1,45	1,20	1,17	1,27	1,13	1,10	1,56	1,30	1,23
11) Число N_{O_2}	0,0007	0,126	0,180	0,015	0,182	0,278	0	0,014	0,0535
12) Расчетные значения α_f	3446	4142	4213	3590	3970	4090	3438	3937	4167
13) Расчетные значения α_{fI}	86	0	0	12	0	0	290	13	0
14) Расчетные значения α_{fII}	5	1563	2118	120	1705	2560	0	155	815
15) Расчетные значения α_e	3537	5705	6331	3722	5675	6650	3728	4105	4982
16) Экспериментальные значения α_e	3984	6001	6521	3910	5310	6140	3842	4144	5276

KEY: 1) Item; 2) No. of test; 3) 71 (P = 23 atm. abs.), sectors; 4) 46 (P = 33.5 atm. abs.), sectors; 5) 50 (P = 55.5 atm. abs.), sectors; 6) 31 (P = 57.5 atm. abs.), sectors; 7) 35 (P = 42 atm. abs.), sectors; 8) Wall temperature, °K; 9) Gas temperature, °K; 10) Rena No. $Re \cdot 10^4$; 11) No. N_{O_2} ; 12) Calculated values α_f ; 13) Calculated values α_{fI} ; 14) Calculated values α_{fII} ; 15) Calculated Values α_e ; 16) Experimental values α_e .

the mechanism of heat transfer in a chemically reacting system with allowance for the actual finite rates of chemical reactions.

In the above figure, a comparison is given in criterial form of calculated and experimental heat transfer data in the system $2NO_2 \rightleftharpoons 2NO + O_2$ for the Re number range of $3 \cdot 10^4 - 2 \cdot 10^5$. Analysis of the results obtained in a study of heat transfer in the case of forced turbulent flow in a circular tube shows that, even at short residence times of the gas in the heated channel, heat transfer in the form of chemical enthalpy due to concentration diffusion is high and that a 1.5-2-fold increase of the heat transfer coefficient is achieved as compared to heat transfer data for an inert gas (N_2 , CO_2 and others).

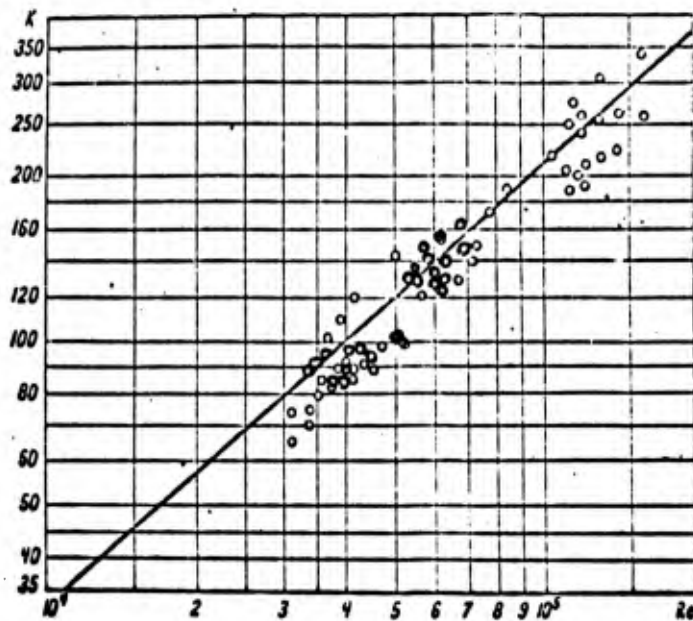


Fig. 1. Heat transfer during forced flow of the chemically nonequilibrium flow ($2\text{NO}_2 \rightleftharpoons 2\text{NO} + \text{O}_2$) in a tube.

$$K = \text{Nu}_f \text{Pr}_f^{-0.43} \left(\frac{T_{cr}}{T_r} \right)^{0.55} = \text{Nu}_{jk} \text{Pr}_{jk}^{-0.43} N_k \left(\frac{T_{cr}}{T_r} \right)^{0.55}$$

The error in the experimental data on heat transfer coefficients in the studied temperature and pressure range amounts to $\pm 15\%$. The experimental accuracy can be substantially improved after more reliable experimental data on the heat capacity of liquid nitrogen tetroxide have been obtained, since errors in the calorimetric determination of the flow rate of N_2O_4 constitute the major portion of the experimental error.

On the basis of the experimental and theoretical studies which have been carried out, it is possible to recommend the criterial relations proposed here for the computation of heat transfer in the chemically-reacting system $2\text{NO}_2 \rightleftharpoons 2\text{NO} + \text{O}_2$, with allowance for the kinetics of chemical reactions, in the temperature range of $140\text{--}700^\circ\text{C}$, pressure range of $1\text{--}60$ atm. abs., and Reynolds number range of $10^4 - 3 \cdot 10^5$.

In conclusion, the authors wish to thank Academician A. K. Krasin (Acad. Sci. Belorussian SSR) for his constant concern and interest in this work, and also their divisional colleagues for active participation in the tests, calculations on an electronic digital computer, and discussion of the results of this work.

NOTATION

α_e - "effective" heat transfer coefficient, $\text{kcal}/\text{m}^2\text{-hr-}^\circ\text{K}$; α_f , α_r - "frozen" and "reaction" component of heat transfer coefficient; Nu_f , Nu_{jk} - Nu numbers for the heat transfer and diffusion process, respectively; Pr_{jk} - Pr no. for the diffusion process;

λ_f - thermal conductivity for the "frozen" mixture; D_k - dynamic diffusion coefficient; L - characteristic length; h_k - specific enthalpy for each component; x_{kct} , x_k - molar concentration of each component at the wall and from the wall; m_k - molecular weight of each component; Q_{pj} - thermal effect of the j -th reaction; a_{jk} - stoichiometric coefficient; p - mixture density.

REFERENCES

1. Petukhov, B. S. and V. N. Popov. Teplofizika vysokikh temperatur [High-Temperature Thermophysics]; No. 4 (1964).
2. Nesterenko, V. B., B. E. Tverkovkin and A. B. Verzhinskaya. Vestsi Akad. Navuk Belarus. SSR, Ser. fiz.-tekhn. navuk; No. 2, pp. 42-49 (1967).
3. Nesterenko, V. B. and B. E. Tverkovkin. Vestsi Akad. Navuk Belarus. SSR, Ser. fiz.-tekhn. navuk; No. 1 (1966).
4. Venikov, V. A. Teoriya podobiya i modelirovanie [Theory of Similarity and Simulation], Izd. "Vysshaya Shkola", Moscow (1966).
5. Beer, H. Heat transfer in a dissociating gas; Dissertation, Institute of Thermodynamics of Aircraft and Rocket Engines, Stuttgart (1964).
6. Beer, H. Chem.-Ing. Technik; Vol. 377, pp. 1047-1054 (1965).

Institute of Nuclear Power Engineering
Acad. Sci. Belorussian SSR

Received 17 January 1967