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**BOTTOM REFLECTION OF UNDERWATER
EXPLOSION SHOCK WAVES, COMPUTER
PROGRAM**

By
James R. Britt
Hans G. Snay

30 JULY 1971

NOL

NAVAL ORDNANCE LABORATORY, WHITE OAK, SILVER SPRING, MARYLAND

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Naval Ordnance Laboratory
White Oak, Silver Spring, Maryland

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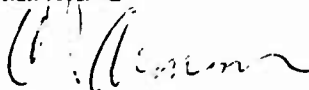
BOTTOM REFLECTION OF UNDERWATER EXPLOSION SHOCK WAVES, COMPUTER PROGRAM

This report is part of a continuing study of the interaction of the underwater explosion shock wave with the ocean bottom. The computer program described in this paper calculates the bottom reflection and generates plots of the pressure history. The calculations of this program are being used in the bottom reflection study to assess the potential danger to ships delivering nuclear underwater weapons posed by various bottom materials.

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ROBERT ENNIS
Captain, USN
Commander



C. J. ARONSON
By direction

CONTENTS

	Page
1. INTRODUCTION	1
2. THEORY USED TO CALCULATE THE BOTTOM REFLECTION	4
2.1 Theory of the Bottom Reflection of a Spherical Wave	4
2.1.1 Critical Angles	6
2.1.2 The Incident Pulse	7
2.1.3 The Surface Reflection	8
2.1.4 The Convolution Integral	9
2.1.5 The Impulse and Energy Flux	15
2.2 The Cagniard-Rosenbaum Method for Calculating the Step Wave Response	16
2.2.1 Non-Rigid Bottom Precursors	17
2.2.2 Rigid Bottom Precursor, Case $c_2 > c_1 > c_4$	18
2.2.3 Rigid Bottom Precursor, Case $c_2 > c_4 > c_1$	19
2.2.4 Step Wave Response at $t = t_c$	20
2.2.5 Non-Rigid Bottom Main Wave, Case $c_2 > c_1$	20
2.2.6 Non-Rigid Bottom Main Wave, Case $c_1 > c_2$	22
2.2.7 Rigid Bottom Main Wave, Case $c_2 > c_4 > c_1$	22
2.2.8 Rigid Bottom Main Wave, Case $c_2 > c_1 > c_4$	24
2.3 The Complex Arithmetic Method for Calculating the Step Wave Response	26
2.4 The Stonley Wave Propagation Velocity	28
2.5 Theory of the Plane Wave Bottom Reflection	29
2.5.1 The Plane Wave Reflection Coefficient and Phase Shift	29
2.5.2 The Plane Wave Bottom Reflection Pressure History	31

NOLTR 71-110

CONTENTS con't	Page
3. THE BOTTOM REFLECTION COMPUTER CODE	34
3.1 FORTRAN Symbols of the Main Program	36
3.2 FORTRAN Symbols of SUBROUTINE STONL	46
3.3 FORTRAN Symbols of SUBROUTINE STPWA	46
3.4 FORTRAN Symbols of SUBROUTINE STPWB	47
3.5 FORTRAN Symbols of FUNCTION ONE	48
3.6 FORTRAN Symbols of FUNCTION ONE1	49
3.7 FORTRAN Symbols in FUNCTION TWO	50
3.8 FORTRAN Symbols in FUNCTION SEVEN	50
3.9 FORTRAN Symbols in SUBROUTINE EXE1	50
3.10 FORTRAN Symbols in SUBROUTINE EXE1	51
3.11 FORTRAN Symbols in FUNCTION EXPO	51
3.12 FORTRAN Symbols in FUNCTION FGI	51
References	53
APPENDIX A--FORTRAN IV LISTING OF THE BOTTOM REFLECTION PROGRAM BOTREF	A-1
APPENDIX B--SAMPLE PROGRAM OUTPUTS	B-1
APPENDIX C--TWO SPECIAL INPUT OPTIONS	C-1

TABLES

Table	Title	Page
B.1	Full Output for a Spherical Wave Bottom Reflection	B-1
B.2	Short Output for a Spherical Wave Bottom Reflection	B-6
B.3	Output for a Plane Wave Bottom Reflection	B-9

ILLUSTRATIONS

Figure	Title	Page
1	Bottom Reflection Geometry	5
2	Sample CALCOMP Plot of Pressure-time History	B-11

BOTTOM REFLECTION OF UNDERWATER
EXPLOSION SHOCK WAVES, COMPUTER PROGRAM

1. INTRODUCTION

The bottom reflection of the underwater explosion shock wave is of interest to the Navy because of the danger it presents for self-damage to ships delivering nuclear ASW weapons. The theory presently being used to describe the reflection is a linear spherical wave theory originally developed by L. Cagniard (1) for the calculation of the reflection at an interface between two elastic solids. On the basis of Cagniard's theory, Rosenbaum (2) derived equations which describe the bottom reflection of underwater explosion shock waves. Britt (3) has greatly extended and generalized Rosenbaum's work. Britt's report should be consulted when using the computer program described here.

This report describes a computer program, BOTREF, written in FORTRAN IV for the NOL CDC 6400 computer. The code calculates the pressure history of the bottom reflection of incident exponential pulses reflected from plane, homogeneous, elastic bottoms using the spherical wave theory. Major portions of this program were written by the second author. The first author later brought this program into its present versatile form and used it successfully in practical applications.

NOLTR 71-110

The program has options for calculating the spherical wave reflection in two ways: (1) using real arithmetic and equations derived using contour integration (referred to as the Cagniard-Rosenbaum method) and (2) using the "complex arithmetic method". The first method is generally faster, but both usually take less than 30 seconds of central processor time on the CDC 6400 for calculating a complete pressure history. Also included in the program is an option for calculating the bottom reflection using the plane wave theory of Arons and Yennie (4). For both the plane wave and the spherical wave, the calculations include corrections for the non-linear changes of the shock wave peak pressure and time constant with the distance from the charge.

The code generates a CALCOMP plot tape of the total pressure history including the incident, bottom reflected, and acoustic surface reflected waves. The print out, in addition to the pressure history, includes information such as the incident angle, the plane wave reflection coefficient and phase shift, critical angles, arrival times, impulses, and energy flux.

The output of the bottom reflection program can be directly transferred to the PTV Program (NOLTR 71-65) which is then used as a subroutine. This program calculates the peak translational velocity (PTV) of a cylindrical target. This velocity can be used as an index for damage.

The equations used in the BOTREF code are described in Section 2 and references are made as to the location in the program where each equation is used. In Section 3 a detailed description is given of the program organization, inputs, outputs, and other important

symbols. The appendices contain a complete FORTRAN listing of the program, sample output, and a CALCOMP plot.

The code contains many comment cards so that most of the inputs and outputs and much of the organization is explained in the program listing.

Comments on Terminology. In the acoustic literature reflectors are called either solids or fluids, depending on whether they have a shear-strength or not. We prefer the terms non-rigid or rigid, because some solids, for instance, sand, have such a low shear strength that the theory for a non-rigid bottom yields sufficiently accurate results, in spite of the fact that the material is a solid. We hope that our terminology will lead to less misunderstandings than the conventional one or the previously used term "liquid bottom".

Rigidity should be understood as the resistance of a body to a change in shape at constant volume. It is equivalent to shear strength and is measured either by the Poisson Ratio or, as in this paper, by the propagation velocity of the shear wave. The shear velocity is zero for a non-rigid material. Compressibility is the resistance to a change in volume at constant shape and can be represented by the propagation velocity of a compression wave, i.e., the sound velocity.

The word rigid often has the connotation of a material having infinite rigidity. We use it in the sense of a material having a finite, non-vanishing rigidity.

2. THEORY USED TO CALCULATE THE BOTTOM REFLECTION

2.1 Theory of the Bottom Reflection of a Spherical Wave

The theory used in the computer program described in this report has been derived by Rosenbaum (2). Britt (3) has reviewed, explained, and greatly extended Rosenbaum's work. A semi-linear theory is used which describes all phenomena of interest with adequate accuracy. The notation used in this section is essentially that of Britt's report. The following exceptions are to be noted. We denote the excess pressure by p instead of P . Britt and Rosenbaum denote the time by τ ; we use t for the time and τ_m for Rosenbaum's reduced time (compare with Equation (2.2.2)). The program calculates the step wave response ${}_n P_m = {}_1 P_1$ which corresponds to one reflection from the bottom. Multiple reflections between the surface and the bottom are not included. (Multiple reflections are of minor importance to underwater explosion phenomena that lead to damage processes. When a strong pressure wave is reflected at the water surface, most of the wave energy is left near the surface and does not propagate down into the water because of cavitation and spray formation.)

We denote ${}_1 P_1$ by P_r , the bottom reflection slant range ${}_1 R_1$ by R_r , the incident or direct wave range by R_i , and the surface reflection range by R_g . We also drop the subscripts n and m except in τ_m and K_m (Equation 2.2.18).

The geometry of the bottom reflection is shown in Figure 1. The water depth is H . The depths of the charge and gauge are d and d_g . The horizontal distance between charge and gauge is r . The incident angle of the bottom reflection is θ . From this figure

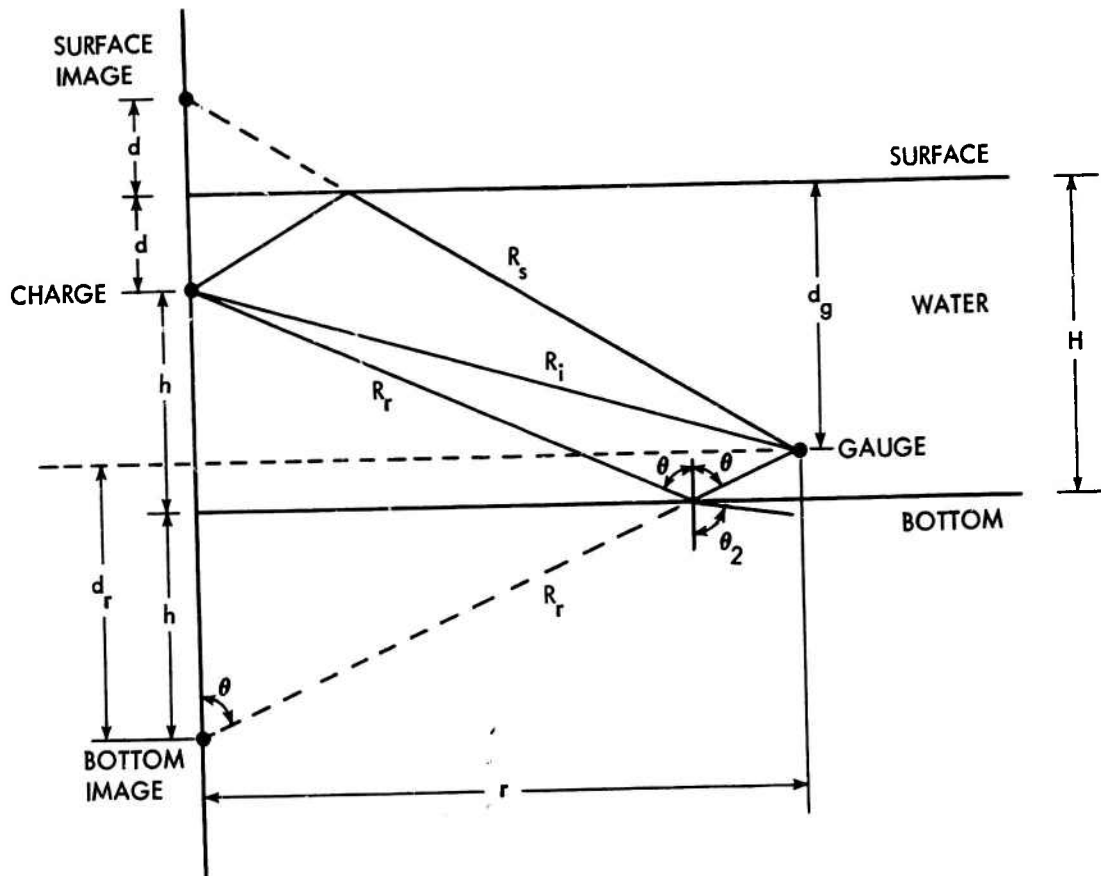


FIG. 1 BOTTOM REFLECTION GEOMETRY

we see that the slant ranges are given by the equations

$$R_i = \left[(d - d_g)^2 + r^2 \right]^{1/2} \quad (\text{slant range of incident wave}) \quad (2.1.1)$$

$$R_r = \left[d_r^2 + r^2 \right]^{1/2} \quad (\text{slant range of wave reflected at bottom}) \quad (2.1.2)$$

and

$$R_s = \left[(d_g + d)^2 + r^2 \right]^{1/2} \quad (\text{slant range of the wave reflected at the water surface}), \quad (2.1.3)$$

where $d_r = 2H - d_g - d$ is the depth of the "image" below the gauge.

Further, we have

$$\cos \theta = d_r / R_r \quad (2.1.4)$$

and
$$\sin \theta = r / R_r. \quad (2.1.5)$$

In the water the sound velocity is denoted by c_1 , and the density by ρ_1 . Similarly, the sound velocity in the bottom material is c_2 , the shear wave propagation velocity is c_4 , and the density is ρ_2 . (Britt denoted the sound velocity and density of a rigid bottom by c_3 and ρ_3 .)

2.1.1 Critical Angles. For an incident angle θ , which is also the reflected angle, the refracted or transmitted ray into the bottom makes an angle θ_2 (see Figure 1) given by Snell's law

$$\sin \theta = \frac{c_1}{c_2} \sin \theta_2. \quad (2.1.6)$$

The angle θ_2 is that angle at which the pressure wave enters the bottom. Similarly, the angle θ_4 of the shear wave in the bottom is defined

$$\sin \theta = \frac{c_1}{c_4} \sin \theta_4. \quad (2.1.7)$$

When $c_2 > c_1$ or $c_4 > c_1$ the angles θ_2 and θ_4 become 90° at incident angles θ_{cr} and θ_{crs} defined by

$$\sin \theta_{cr} = c_1/c_2 \quad (2.1.8)$$

$$\sin \theta_{crs} = c_1/c_4. \quad (2.1.9)$$

θ_{cr} is called the critical angle of the compression wave, and θ_{crs} is called the critical angle of the shear wave. These angles are important for calculating and interpreting the bottom reflection pressure history.

2.1.2 The Incident Pulse. The computer program assumes an exponential incident pulse $p_i(t)$ given by

$$\begin{aligned} p_i(t) &= p_F(R_i) \exp \left[-(t - R_i/c_1)/G \right] && \text{for } t \geq R_i/c_1 \\ p_i(t) &= 0 && \text{for } t < R_i/c_1, \end{aligned} \quad (2.1.10)$$

where G is the time constant (usually denoted by θ) and $p_F = p_F(R_i)$ is the peak pressure of the incident shock wave. A reduced notation is used in the machine program utilizing the incident slant range R_i (Equation 2.1.1) as the characteristic length. The reduced time is $\bar{t} = tc_1/R_i$. (It is denoted by T in the program). The reduced arrival time of the front of the direct wave is thus $\bar{t} = 1$. The incident pulse is then given by

$$\begin{aligned} p_i(\bar{t}) &= p_F(R_i) \exp \left[-(\bar{t} - 1)/\bar{G} \right] && \text{for } \bar{t} \geq 1 \\ p_i(\bar{t}) &= 0 && \text{for } \bar{t} < 1, \end{aligned} \quad (2.1.11)$$

where $\bar{G} = c_1 G/R_i$ is the reduced time constant.

For the time constant G and the peak pressure p_F the relations for the actual underwater explosion shock waves (high amplitude waves) are used which when used together with the wave equation comprise the "semi-linear" theory. The shock wave parameters are obtained from the similitude equations

$$G = C_G W^{1/3} (W^{1/3}/R_i)^{n_G} \quad (2.1.12)$$

$$p_F = C_P (W^{1/3}/R_i)^{n_P} , \quad (2.1.13)$$

where C_G , C_P , n_G , and n_P are constants for a given explosive. W is the charge weight in pounds, or, with appropriate constants, the yield in kilotons. G and p_F are calculated in the main program in Cards BOTR160-167.

Examples of the constants are

Explosive	C_P	n_P	C_G	n_G
TNT	21600	1.13	0.052	-0.23
HBX-1	23800	1.15	0.049	-0.29
Nuclear	$4.291 \cdot 10^6$	1.13	2.242	-0.22
($W = \text{Yield}$ in kt)	$4.380 \cdot 10^6$	1.13	2.274	-0.22

The values for nuclear explosions of the upper row are the most recent ones. Those in the lower row are generally quoted in the literature. The constants C_P and C_G are given in psi and milliseconds.

2.1.3 The Surface Reflection. The surface reflection $p_s(t)$

calculated from the simple acoustic equation is

$$\begin{aligned} p_s(t) &= -p_F(R_s) \exp \left[-(t - R_s/c_1)/G_s \right] & \text{for } t \geq R_s/c_1 \\ p_s(t) &= 0 & \text{for } t < R_s/c_1, \end{aligned} \quad (2.1.14)$$

where $G_s = G(R_s)$. In reduced notation this becomes

$$\begin{aligned}
 p_s(\bar{t}) &= -P_F(R_s) \exp\left[-(\bar{t} - \bar{R}_s)/\bar{G}_s\right] && \text{for } \bar{t} \geq \bar{R}_s \\
 p_s(\bar{t}) &= 0 && \text{for } \bar{t} < \bar{R}_s,
 \end{aligned}
 \tag{2.1.15}$$

where $\bar{G}_s = c_1 G_s / R_i$ and $\bar{R}_s = R_s / R_i$. These equations are coded in Cards BOTR218, 704, and 882.

The surface reflection is a tension wave and its pressure is to be subtracted from the pressure of the incident and the bottom reflected wave.

Equations (2.1.14 and 15) ignore cavitation which in sea water does not let pressures drop substantially below the vapor pressure. In the machine program this is taken into account by a test that makes sure that the total pressure does not fall below zero absolute (Cards BOTR713 and 884).

For a very oblique incidence the acoustic treatment of the surface reflection breaks down and must be replaced by the anomalous surface reflection described in NOLTR 70-31. The machine program described here does not include this mode of the surface reflection. This problem will be treated in another machine program that describes the shock wave propagation in shallow water.

2.1.4 The Convolution Integral. The theory of the bottom reflection yields the reflected wave for an incident step wave. This step wave response, denoted by $P_r(t)$, is the crucial point of the analysis and will be discussed in detail later. It has the dimension of $(\text{Length})^{-1}$. The pressure history of the bottom reflected wave for an exponential incident wave $p_r(t)$ is obtained from the convolution integral:

$$\begin{aligned}
 P_R(t) &= P_F^i(R_R) \left[P_R(t) - 1/G_R \int_{\delta}^t \exp\left[-(t-z)/G_R\right] P_R(z) dz \right] \\
 & \qquad \qquad \qquad \text{for } t \geq \delta \\
 P_R(t) &= 0 \qquad \qquad \qquad \text{for } t < \delta.
 \end{aligned}
 \tag{2.1.16}$$

This equation is explained in Appendix D of Britt's report. The scale factor P_F^i and the time constant G_R are given by

$$P_F^i = R_R P_F(R_R) \tag{2.1.17}$$

$$G_R = G(R_R), \tag{2.1.18}$$

where R_R is the slant range of the reflected wave, Equation (2.1.2). The factor R_R of the reduced pressure scale factor P_F^i stems from the definition of the reduced step wave response $P_R(t)$ which includes R_R^{-1} as a factor.

The reduced form of the convolution integral is readily obtained by introduction of $\bar{t} = tc_1/R_i$, $\bar{\delta}$, \bar{z} , and $\bar{G}_R = c_1 G_R/R_i$.

The symbol δ in Equation (2.1.16) denotes the arrival time of the bottom reflection.

For subcritical incidence, $\theta < \theta_{CR}$, we have

$$\delta = t_c = R_R/c_1 \tag{2.1.19}$$

and the reduced form is

$$\bar{\delta} = c_1 \delta/R_i = R_R/R_i. \tag{2.1.20}$$

In this case δ is the arrival time t_c of the peak of the reflected wave.

For supercritical incidence, $\theta > \theta_{CR}$, the precursor of the bottom reflection arrives before $t = t_c$, namely at

$$\delta = r/c_2 + d_r(c_1^{-2} - c_2^{-2})^{1/2} \quad (2.1.21)$$

or in the dimensionless form

$$\bar{\delta} = rc_1/c_2R_i + d_r \left[1 - (c_1/c_2)^2 \right]^{1/2} / R_i. \quad (2.1.22)$$

The convolution integral is calculated in the BOTREF program Cards BOTR556, 589, 597, 635, 643, and 673 using Simpson's rule for small intervals with three equally spaced points.

For an exponential incident pulse the integral need not be recalculated from $t = \delta$ for each time step because of

$$\exp(t + \Delta t) = \exp(t)\exp(\Delta t).$$

The algorithm used to calculate the integral in Equation (2.1.16), which we call F_I , is as follows:

$$F_I(t) = \exp(-2\Delta t/G_r)F_I(t - 2\Delta t) + \left\{ \left[P_r(t - 2\Delta t)\exp(-\Delta t/G_r) + 4P_r(t - \Delta t) \right] \exp(-\Delta t/G_r) + P_r(t) \right\} \Delta t/3. \quad (2.1.23)$$

This relation permits a convenient step-by-step quadrature of the integral using its value for a time $2\Delta t$ earlier. The expression is readily transformed into a reduced form by the introduction of \bar{t} , $\bar{\Delta t}$, and \bar{G}_r . F_I has the dimension time/length.

For supercritical incidence $P_r(t)$ has a logarithmic singularity at $t = t_c$. Since $P_r(t)$ is a rapidly changing function of t near t_c , a smaller time increment, $\Delta t' \approx \Delta t/8$, is used in the code for the interval $(t_c - \alpha\Delta t, t_c + 4\Delta t)$. The code calculates α so that there are enough points in the bottom reflection pressure-time history (before and after the time increment change) to execute the impulse

and energy flux integrations. The usual range is $2.1 < \alpha < 6.1$. Because these integrations are performed with Simpson's rule on equally spaced points, each integration step is completed on an odd-numbered point.

Further, in the time range $t_c - 2\Delta t' < t < t_c + 2\Delta t'$ we change the integration variable in the convolution integral F_I to

$$\begin{aligned} v^2 &= t_c^2 - z^2 && \text{for } z \leq t_c \\ u^2 &= z^2 - t_c^2 && \text{for } z \geq t_c. \end{aligned}$$

The step wave response $P_I(t)$ behaves near the singularity like

$$\lim_{t \rightarrow t_c} P_I(t) = C \ln(|t_c - t|).$$

The change of variables v and u transforms the last two factors of Equation (2.1.16) as follows:

$$\begin{aligned} P_I(z) dz &= -\frac{v}{z} P_I(z) dv && z \leq t_c \\ &= \frac{u}{z} P_I(z) du && z \geq t_c. \end{aligned}$$

Then we obtain

$$\lim_{z \rightarrow t_c} -\frac{v}{z} P_I(z) = -C \lim_{z \rightarrow t_c} \frac{(t_c^2 - z^2)^{1/s}}{z} \ln(t_c - z) = 0 \quad z \leq t_c$$

$$\lim_{z \rightarrow t_c} \frac{u}{z} P_I(z) = C \lim_{z \rightarrow t_c} \frac{(z^2 - t_c^2)^{1/s}}{z} \ln(z - t_c) = 0 \quad z \geq t_c.$$

This means the integrands vanish at the singularity of P_I , and thus makes numerical integration possible.

Equation (2.1.24) below illustrates the variable change.

$$\begin{aligned}
 P_R(t) = P_F' & \left[P_R(t) - \frac{1}{G_R} \left\{ \int_{\delta}^{t_c - 2\Delta t'} \exp[-(t - z)/G_R] P_R(z) dz \right. \right. \\
 & - \int_{v(t_c - 2\Delta t')}^{v(t_c)} \exp[-(t - z)/G_R] P_R(z) \frac{v}{z} dv \\
 & + \int_{u(t_c)}^{u(t_c + 2\Delta t')} \exp[-(t - z)/G_R] P_R(z) \frac{u}{z} du \\
 & \left. \left. + \int_{t_c + 2\Delta t'}^t \exp[-(t - z)/G_R] P_R(z) dz \right\} \right]. \tag{2.1.24}
 \end{aligned}$$

Up to time $t_c - 2\Delta t'$ and after time $t_c + 2\Delta t'$ the integration variable is z and the algorithm of Equation (2.1.23) is used to perform the quadrature. Around the singularity Simpson's rule on equally spaced intervals of v and u , instead of z or time, is used for the integration.

Using the algorithms described below, $F_I(t)$ and $p_r(t)$ are evaluated in two steps before and after the singularity. When $t_c - 2\Delta t' < t \leq t_c$, the following variables are used:

$$t_1 = t_c - 2\Delta t' \tag{2.1.25}$$

$$v_1 = v(t_1) = (t_c^2 - t_1^2)^{1/2} \tag{2.1.26}$$

$$t_2 = [t_c^2 - (3v_1/4)^2]^{1/2} \tag{2.1.27}$$

$$t_3 = [t_c^2 - (v_1/2)^2]^{1/2} \tag{2.1.28}$$

$$t_4 = [t_c^2 - (v_1/4)^2]^{1/2} \quad (2.1.29)$$

The fifth time used here is t_c . However, $P_r(t_c)$ does not appear in the equations for F_I because the transformed integrand vanishes.

The value of F_I at $t = t_3$ is obtained from

$$F_I(t_3) = F_I(t_1) \exp[-(t_3-t_1)/G_r] + \{P_r(t_1) \exp[-(t_3-t_1)/G_r] v_1/t_1 + 3P_r(t_2) \exp[-(t_3-t_2)/G_r] v_1/t_2 + P_r(t_3) v_1/2t_3\} v_1/12 \quad (2.1.30)$$

This equation is coded in reduced notation in Card BOTR589. For the next step $F_I(t_c)$ is calculated using

$$F_I(t_c) = F_I(t_3) \exp[-(t_c-t_3)/G_r] + \{P_r(t_3) \exp[-(t_c-t_3)/G_r] v_1/2t_3 + P_r(t_4) \exp[-(t_c-t_4)/G_r] v_1/t_4\} v_1/12 \quad (2.1.31)$$

This equation is coded in reduced notation (Card BOTR597).

Similarly, after the singularity we define the following variables:

$$t_5 = t_c + 2\Delta t' \quad (2.1.32)$$

$$u_1 = u(t_5) = (t_5^2 - t_c^2)^{1/2} \quad (2.1.33)$$

$$t_2 = [t_c^2 + (u_1/4)^2]^{1/2} \quad (2.1.34)$$

$$t_3 = [t_c^2 + (u_1/2)^2]^{1/2} \quad (2.1.35)$$

$$t_4 = [t_c^2 + (3u_1/4)^2]^{1/2} \quad (2.1.36)$$

Here t_1 is the time of the singularity t_c , but $P_r(t_c)$ is not needed

since the transformed integrand vanishes. The value of $F_I(t_s)$ is then given by

$$F_I(t_s) = F_I(t_c) \exp[-(t_s - t_c)/G_r] + \left\{ P_r(t_s) u_1 \exp[-(t_s - t_s)/G_r] / t_s + P_r(t_s) u_1 / 2t_s \right\} u_1 / 12 . \quad (2.1.37)$$

This equation is converted to reduced notation and coded in Card BOTR635. Then the last step using the special integration variables is

$$F_I(t_s) = F_I(t_s) \exp[-(t_s - t_s)/G_r] + \left\{ P_r(t_s) u_1 \exp[-(t_s - t_s)/G_r] / 2t_s + 3P_r(t_s) u_1 \exp[-(t_s - t_s)/G_r] / t_s + P_r(t_s) u_1 / t_s \right\} u_1 / 12 . \quad (2.1.38)$$

This equation in reduced form is coded in Card BOTR643.

2.1.5 The Impulse and Energy Flux. The impulse I and energy flux E_F are calculated in the main program Cards BOTR717-766. These calculations are made only if the spherical wave bottom reflection is used. The impulse in psi-sec is evaluated from the equation

$$I = \int_{t_0}^t p(t) dt,$$

where $p(t) = p_i(t) + p_r(t) + p_g(t)$ is the total pressure of the incident, bottom reflected, and surface reflected waves and t_0 is the time of the beginning of the pressure pulse $p(t)$.

The energy flux E_F in in-psi is found from the equation

$$E_F = \left\{ \int_{t_0}^t |p| p dt \right\} / (2.3066 \rho_2 c_1) ,$$

where 2.3066 is a conversion factor necessary for E_F to be in units in-psi when p is in psi, time in seconds, ρ_2 in gm/cm³, and c_1 in ft/sec.

Away from the singularity of $p_r(t)$ of $t = t_c$ and for sub-critical bottom reflections the integrals are determined using Simpson's rule on equally spaced points as a function of time. Near the singularity the change of integration variables is made to v and u as for the convolution integral. This change of variables is made in Cards BOTR738-755.

Also calculated in the same section of the program is the "positive impulse" which is simply the impulse of the positive part of the total pressure $p(t)$. If the full output option is used (see the input Z5 in Section 3.1 and the sample outputs of Appendix B), the magnitudes reduced impulse $I/w^{1/2}$, reduced positive impulse, and reduced energy flux $E_F/w^{1/2}$ are calculated in Cards BOTR793-797.

2.2 The Cagniard-Rosenbaum Method for Calculating the Step Wave Response

In this section the Cagniard-Rosenbaum equations are listed, and forms of these equations similar to the FORTRAN notation are given. This method is faster than the complex arithmetic method which will be discussed in Section 2.3, but it has the disadvantage that separate equations are required for the precursor and the main wave and for each type of bottom (determined by the ordering

of c_1 , c_2 , and c_4). However, in the coding we were able to take advantage of certain common factors and terms and hence reduce the number of statements that would otherwise be required.

2.2.1 Non-Rigid Bottom Precursors. A fast non-rigid bottom ($c_2 > c_1$) for which $\theta > \theta_{cr}$ has a step wave response at times $\delta \leq t < t_c$ expressed by the following equation (Britt (2-1.10)):

$$P_R(t) = \frac{b(\sigma - M)}{R_R} \int_{-1}^1 \frac{\omega(\sigma + \omega)^{1/2} (1 - \sin \pi\psi/2) d\psi}{[(1-b^2)\omega^2 + \sigma^2 b^2] (\omega - N)^{1/2}}, \quad (2.2.1)$$

$$\text{where } \omega = (\sigma + M)/2 + [(\sigma - M)/2] \sin \pi\psi/2, \quad (2.2.2)$$

$$b = \rho_1/\rho_2, \tau_m = t/R_R, \sigma = (c_1^{-2} - c_2^{-2})^{1/2}, M = \tau_m \cos \theta + (c_1^{-2} - \tau_m^2)^{1/2} \sin \theta, \\ N = \tau_m \cos \theta - (c_1^{-2} - \tau_m^2)^{1/2} \sin \theta, \sin \theta = r/R_R, \text{ and } \cos \theta = d_r/R_R.$$

In the program the integration variable $x = \pi\psi/2$ is used. We also set $w = c_1 \omega$. Then after rearranging, Equation (2.2.1) can be put into the form which is coded

$$R_i P_R(t) = \frac{2\sqrt{2} b R_i}{\pi R_R} \int_{-\pi/2}^{\pi/2} \frac{F_x w dx}{w^2 + b^2 (c_1^2 \sigma^2 - w^2)}, \quad (2.2.3)$$

where

$$F_x = (1 - \sin x) \left\{ \left[(c_1 \sigma + w) (c_1 \sigma - c_1 M) \right] / \left[1 + \sin x + 4(1 - c_1^2 \tau_m^2)^{1/2} \sin \theta / (c_1 \sigma - c_1 M) \right] \right\}^{1/2}. \quad (2.2.4)$$

$$= (1 - \sin x) \left\{ [(\cos \alpha + w) P(1)] / [1 + \sin x + P(2)] \right\}^{1/2},$$

$$\text{with } \cos \alpha = c_1 \sigma = [1 - (c_1/c_2)^2]^{1/2},$$

$$P(1) = \cos \alpha - c_1 M,$$

$$P(2) = 4(1 - c_1^2 \tau_m^2)^{1/2} \sin \theta / P(1).$$

The variables $\cos \alpha$, $P(1)$, and $P(2)$ are calculated in Cards BOTR238, STPA022, and 23.

The integrand above is evaluated in FUNCTION ONE. The variable F_x is coded in Card ONE023, and the value of the integrand is ONE in Card ONE055. The factor outside the integral is calculated in Card STPA025. The integration for this and all other precursors is controlled by SUBROUTINE STPWA which uses the Gaussian quadrature of FUNCTION FGI to evaluate the integral. The value of $R_i P_r(t)$, called STPW in Card STPA027, is returned to the main program BOTREF where the convolution integral is executed.

2.2.2 Rigid Bottom Precursor, Case $c_2 > c_1 > c_4$. The precursor integrands for a rigid bottom are also evaluated in FUNCTION ONE. For the case $c_2 > c_1 > c_4$ (slow shear) the following equation (Britt (4-1.6)) is used

$$P_r(t) = \frac{b(\sigma-M)}{4R_r c_4^4} \int_{-1}^1 \frac{(\sigma+\omega)^{1/2} A (1-\sin \pi t/2) dt}{(\omega-N)^{1/2} [A^2 + (B+C)^2]} \quad (2.2.5)$$

where

$$A = \omega(c_4^{-2}/2 - c_1^{-2} + \omega^2)^2 \quad (2.2.6)$$

$$B = \omega(c_1^{-2} - \omega^2)(\sigma^2 - \omega^2)^{1/2} \left| \omega^2 + c_4^{-2} - c_1^{-2} \right|^{1/2} \quad (2.2.7)$$

$$C = b c_4^{-4} (\sigma^2 - \omega^2)^{1/2} / 4. \quad (2.2.8)$$

In a manner similar to the non-rigid bottom, Equation (2.2.5) can be rearranged to obtain the program form

$$R_i P_r(t) = \frac{2\sqrt{2} b R_i}{\pi R_r} \int_{-\pi/2}^{\pi/2} \frac{F_x A_x dx}{[A_x^2 + (B_x + C_x)^2]} \quad (2.2.9)$$

where $w = c_1 \omega$, $\cos \alpha = c_1 \sigma$, and

$$A_x = 4c_1 c_4^4 A = w[1-2(c_4/c_1)^2(1-w^2)]^2, \quad (2.2.10)$$

$$B_x = 4c_1 c_4^4 B = 4w(c_4/c_1)^2(1-w^2) [(\cos^2 \alpha - w^2) |(c_4/c_1)^2(w^2-1) + 1|]^{1/2}, \quad (2.2.11)$$

$$C_x = 4c_1 c_4^4 C = b(\cos^2 \alpha - w^2)^{1/2}. \quad (2.2.12)$$

As in the previous case F_x and the factor outside the integral are calculated in Cards ONE023, and STPA025. The variables A_x , B_x , and C_x are coded in Cards ONE043, 044, and 050. The value of the integrand is ONE in Card ONE051, and as before SUBROUTINE STPWA controls the integration.

2.2.3 Rigid Bottom Precursor, Case $c_2 > c_4 > c_1$. The precursor for $c_2 > c_4 > c_1$ (fast shear) is based on the following equation (Britt (4-2.8))

$$P_R(t) = \frac{b(\sigma-M)}{4R_R c_4^4} \int_{\psi_1}^1 \frac{(\sigma+\omega)^{1/2} A (1-\sin \pi\psi/2) d\psi}{(\omega-N)^{1/2} [A^2 + (B+C)^2]} + \frac{b(\sigma-M)}{4R_R c_4^4} \int_{-1}^{\psi_1} \frac{(\sigma+\omega)^{1/2} (A-B) (1-\sin \pi\psi/2) d\psi}{(\omega-N)^{1/2} [(A-B)^2 + C^2]}, \quad (2.2.13)$$

where $\psi_1 = \frac{2}{\pi} \arcsin \left[\frac{2(c_1^{-2} - c_4^{-2})^{1/2} - \sigma - M}{\sigma - M} \right]$.

(In Britt's paper the magnitude B in the second integral is denoted by B_2 , a precaution unnecessary if the definition Equation (2.2.7) is used.) Equation (2.2.13) can then be written in the form used in the program

$$R_i P_r(t) = \frac{2\sqrt{2} b R_i}{\pi R_r} \int_{-\pi/2}^{\pi/2} F_x F_k dx, \quad (2.2.14)$$

where

$$F_k = (A_x - B_x) / [(A_x - B_x)^2 + C_x^2] \text{ for } \omega^2 + c_4^{-2} - c_1^{-2} < 0 \quad (2.2.15)$$

$$F_k = A_x / [A_x^2 + (B_x + C_x)^2] \text{ for } \omega^2 + c_4^{-2} - c_1^{-2} \geq 0. \quad (2.2.16)$$

The variables A_x , B_x , and C_x are defined in Equations (2.2.10), (2.2.11), and (2.2.12). In the first case the integrand is coded in Card ONE047 and the second case in Card ONE051.

2.2.4 Step Wave Response at $t = t_c$. At the peak of the bottom reflection at $t = t_c = R_r/c_1$, the step wave response $P_r(t_c)$ is calculated in the main program BOTREF. For supercritical incidence, $\theta > \theta_{cr}$, $P_r = \pm \infty$ where the sign depends on the phase shift ϕ explained in Section 2.5.1. The treatment of this case is discussed in Section 2.1.4. For subcritical incidence, $\theta < \theta_{cr}$, P_r remains finite and $P_r = K/R_r$ where K is the plane wave reflection coefficient of Section 2.5.1.

2.2.5 Non-Rigid Bottom Main Wave, Case $c_2 > c_1$. A fast non-rigid bottom ($c_2 > c_1$) has a step wave response at times $t > t_c$ given by the equation (Britt (2-2.10))

$$P_r(t) = \frac{1}{R_r} \frac{1-b}{1+b} + \frac{2b}{\pi R_r} \int_0^\sigma \frac{\omega(\sigma^2 - \omega^2)^{1/2}}{[(1-b^2)\omega^2 + \sigma^2 b^2]} \left\{ \left[(\omega - K_m)^2 + L \right]^{-1/2} - \left[(\omega + K_m)^2 + L \right]^{-1/2} \right\} d\omega, \quad (2.2.17)$$

where

$$K_m = \tau_m \cos \theta \quad (2.2.18)$$

$$L = (\tau_m^2 - c_1^{-2}) \sin^2 \theta. \quad (2.2.19)$$

The subscript m has been kept to distinguish it from the reflection coefficient K.

In the code the integration variable is $w = c_1 \omega$, and the form of the equation is similar to that for the precursor:

$$R_i P_r(t) = \frac{(1-b)R_i}{(1+b)R_r} + \frac{2bR_i}{\pi R_r} \int_0^{c_1 \sigma} \frac{F_x w dw}{w^2 + b^2 (c_1^2 \sigma^2 - w^2)}, \quad (2.2.20)$$

where F_x is now

$$F_x = \left[\frac{c_1^2 \sigma^2 - w^2}{c_1^2 L + (w - c_1 K_m)^2} \right]^{1/2} - \left[\frac{c_1^2 \sigma^2 - w^2}{c_1^2 L + (w + c_1 K_m)^2} \right]^{1/2} \quad (2.2.21)$$

$$= \left[\frac{\cos^2 \alpha - w^2}{P(8) + (w - P(7))^2} \right]^{1/2} - \left[\frac{\cos^2 \alpha - w^2}{P(8) + (w + P(7))^2} \right]^{1/2}$$

with $\cos \alpha = c_1 \sigma = [1 - (c_1/c_2)^2]^{1/2}$.

The abbreviations P(7) and P(8) are listed in Cards STPB026 and 27.

The function F_x above is calculated in Card ONE032, and the integrand is ONE in Card ONE055. The factor outside the integral is evaluated in Card STPB032. The first term on the right hand side of Equation (2.2.20) is computed in Card STPB038. The integration for this and all other main wave responses is controlled by SUBROUTINE STPWB.

The value of $R_i P_r(t)$ is calculated in Card STPB047.

2.2.6 Non-Rigid Bottom Main Wave, Case $c_1 > c_2$. The step wave response for a slow non-rigid bottom, one with $c_1 > c_2$, is expressed in the equation (Britt (2-3.14))

$$P_r(t) = \frac{1}{R_r} \frac{1-b}{1+b} - \frac{2\sqrt{2} b}{\pi R_r} \int_0^{\bar{\sigma}} \frac{\bar{\omega}(\bar{\sigma}^2 - \bar{\omega}^2)^{1/2}}{(1-b^2)\bar{\omega}^2 + \bar{\sigma}^2 b^2} \left\{ \frac{[(\bar{\omega}^2 + D)^2 + E]^{1/2} + (\bar{\omega}^2 - F)}{(\bar{\omega}^2 + D)^2 + E} \right\}^{1/2} d\bar{\omega} \quad (2.2.22)$$

where $\bar{\sigma} = (c_2^2 - c_1^2)^{1/2}$, (2.2.23)

$$D = \tau_m^2 \cos 2\theta + c_1^2 \sin^2 \theta, \quad (2.2.24)$$

$$E = 4(\sin^2 \theta \cos^2 \theta) \tau_m^2 (\tau_m^2 - c_1^2), \quad (2.2.25)$$

and $F = \tau_m^2 - c_1^2 \sin^2 \theta$. (2.2.26)

The form used in the program is $\frac{R_i}{c_1 \bar{\sigma}}$

$$R_i P_r(t) = \frac{R_i}{R_r} \frac{1-b}{1+b} - \frac{2\sqrt{2} b R_i}{\pi R_r} \int_0^x \bar{F}_A \bar{F}_B dx, \quad (2.2.27)$$

where $x = c_1 \bar{\omega}$

$$\bar{F}_A = x(c_1^2 \bar{\sigma}^2 - x^2)^{1/2} / [(1-b^2)x^2 + b^2 c_1^2 \bar{\sigma}^2] \quad (2.2.28)$$

$$\bar{F}_B = c_1^{-1} \left\{ \frac{[(\bar{\omega}^2 + D)^2 + E]^{1/2} + (\bar{\omega}^2 - F)}{(\bar{\omega}^2 + D)^2 + E} \right\}^{1/2} \quad (2.2.29)$$

The integrand is evaluated in FUNCTION TWO Card TWO017, \bar{F}_A is coded in Card TWO013, and \bar{F}_B is coded in Cards TWO014 and 015. The terms corresponding to D, E, and F are denoted by P(11), P(12), and P(13) and are evaluated in Cards STPB029-31.

2.2.7 Rigid Bottom Main Wave, Case $c_2 > c_4 > c_1$. The rigid bottom main wave response for the case $c_2 > c_4 > c_1$ (fast shear) is expressed in Britt's equations (4-4.3), (4-3.14), and (4-3.15)

which are as follows:

$$P_r(t) = \frac{1}{R_r} + \Delta \quad (2.2.30)$$

$$+ \frac{b}{2\pi R_r c_4^2} \int_0^{\sigma_2} \frac{(\sigma^2 - \omega^2)^{1/2} (A-B)}{[(A-B)^2 + C^2]} \left\{ \frac{1}{[(\omega - K_m)^2 + L]^{1/2}} \frac{1}{[(\omega + K_m)^2 + L]^{1/2}} \right\} d\omega$$

$$+ \frac{b}{2\pi R_r c_4^2} \int_{\sigma_2}^{\sigma} \frac{A(\sigma^2 - \omega^2)^{1/2}}{[A^2 + (B+C)^2]} \left\{ \frac{1}{[(\omega - K_m)^2 + L]^{1/2}} \frac{1}{[(\omega + K_m)^2 + L]^{1/2}} \right\} d\omega$$

where $\sigma_2 = (|c_4^{-2} - c_1^{-2}|)^{1/2}$ and

$$\Delta = -\frac{\sqrt{2} k}{R_r g_1} \left\{ \frac{(a^2 + f)^{1/2} - a}{a^2 + f} \right\}^{1/2} \Gamma \quad (2.2.31)$$

with

$$\Gamma = \left\{ g_1 \left[\left(\frac{c_4^{-2}}{2} - k^2 \right)^2 - k^2 g_3 g_4 \right] - \frac{b g_3}{4 c_4^4} \right\} / \left\{ \frac{k}{g_1} \left[\left(\frac{c_4^{-2}}{2} - k^2 \right)^2 - k^2 g_3 g_4 \right] - g_1 k \left[4 \left(\frac{c_4^{-2}}{2} - k^2 \right) + 2 g_3 g_4 + k^2 \left(\frac{g_4}{g_3} + \frac{g_3}{g_4} \right) \right] + \frac{b k}{4 g_3 c_4^4} \right\} .$$

Here, c_{st} is the propagation velocity of the Stonley wave, $k = 1/c_{st}$, $g_1 = (k^2 - c_1^{-2})^{1/2}$, $g_3 = (k^2 - c_2^{-2})^{1/2}$, $g_4 = (k^2 - c_4^{-2})^{1/2}$, $a = \tau_m^2 - (k^2 - c_1^{-2} \cos^2 \theta)$, and $f = 4\tau_m^2 g_1^2 \cos^2 \theta$.

The Stonley wave propagation velocity c_{st} is calculated in SUBROUTINE STONL. The equation for c_{st} used in the program is described in Section 2.4.

The above equation is coded in the form

$$R_i P_r(t) = \frac{R_i}{R_r} + R_i \Delta + \frac{2bR_i}{\pi R_r} \int_0^{c_1 \sigma} F_x F_k dw \quad (2.2.32)$$

F_x and F_k have been defined in Equations (2.2.21), (2.2.15), and (2.2.16).

The first two terms of Equation (2.2.32) are calculated in Cards STPB056-71 for all solid bottom main waves, and the result is stored in the variable TERML. The integrand is determined in FUNCTION ONE in Cards ONE047 and 051 in the same way as for the precursor. However, the function F_x and the factor in front of the integral are here calculated in Cards ONE032 and STPB032 as they were for a fast fluid bottom main wave.

2.2.8 Rigid Bottom Main Wave, Case $c_2 > c_1 > c_4$. The main wave response for the rigid bottom case $c_2 > c_1 > c_4$ (slow shear) is given by the following equation (Britt (4-3.13))

$$P_r(t) = \frac{1}{R_r} + \Delta + \frac{b}{2\pi R_r c_4^4} \int_0^{\sigma} \frac{A(\sigma^2 - w^2)^{1/2}}{A^2 + (B+C)^2} \left\{ \frac{1}{[(w-K_m)^2 + L]^{1/2}} - \frac{1}{[(w+K_m)^2 + L]^{1/2}} \right\} dw - \frac{\sqrt{2} b}{2\pi R_r c_4^4} \int_0^{\sigma_2} \frac{(\bar{w}^2 + \sigma^2)^{1/2} \bar{B}}{(\bar{A} + \bar{C})^2 + \bar{B}^2} \left\{ \frac{[(\bar{w}^2 + D)^2 + E]^{1/2} + (\bar{w}^2 - F)^{1/2}}{(\bar{w}^2 + D)^2 + E} \right\} d\bar{w} \quad (2.2.33)$$

where $\bar{A} = \bar{w} [c_4^{-2}/2 - c_1^{-2} - \bar{w}^2]^2$, (2.2.34)

$$\bar{B} = \bar{w} (c_1^{-2} + \bar{w}^2) (\sigma^2 + \bar{w}^2)^{1/2} (\sigma_2^2 - \bar{w}^2)^{1/2}, \quad (2.2.35)$$

$$\bar{C} = \frac{b}{4c_4^4} (\sigma^2 + \bar{w}^2)^{1/2}. \quad (2.2.36)$$

The above equation is coded in the form

$$R_1 P_r(t) = \frac{R_1}{R_r} + R_1 \Delta + \frac{2bR_1}{\pi R_r} \int_0^{c_1 \sigma} F_x F_k dw - \left(\frac{c_1}{c_4} \right)^4 \left\{ \frac{\sqrt{2}}{4} \int_0^{c_1 \sigma_2} \bar{F}_A \bar{F}_B dx \right\} \quad (2.2.37)$$

where $x = c_1 \bar{w}$, F_x and F_k are defined in Equations (2.2.21), (2.2.15), and (2.2.16),

$$\bar{F}_A = \frac{(\bar{w}^2 + \sigma^2)^{1/2} \bar{B}}{c_1^4 [(\bar{A} + \bar{C})^2 + \bar{B}^2]} = \frac{(x^2 + \cos^2 \alpha)^{1/2} \bar{B}_x}{(\bar{A}_x + \bar{C}_x)^2 + \bar{B}_x^2}, \quad (2.2.38)$$

$$\bar{F}_B = c_1^{-1} \left\{ \frac{[(\bar{w}^2 + D)^2 + E]^{1/2} + (\bar{w}^2 - F)}{(\bar{w}^2 + D)^2 + E} \right\}^{1/2}, \quad (2.2.39)$$

$$\cos \alpha = c_1 \sigma = [1 - (c_1/c_2)^2]^{1/2},$$

$$\bar{A}_x = c_1^5 \bar{A} = x [(c_1/c_4)^2 / 2 - 1 - x^2]^2,$$

$$\bar{B}_x = c_1^5 \bar{B} = x(1 + x^2) \{ [\cos^2 \alpha + x^2] [(c_1/c_4)^2 - 1 - x^2] \}^{1/2},$$

$$\bar{C}_x = c_1^5 \bar{C} = b(c_1/c_4)^4 (\cos^2 \alpha + x^2)^{1/2} / 4.$$

The first three terms of Equation (2.2.37) are calculated using the same cards as for the fast shear case. The integrand of the second integral is computed in FUNCTION ONE1. \bar{F}_A and \bar{F}_B are expressed in Cards ONE1019, 20, and 21. \bar{A}_x , \bar{B}_x , and \bar{C}_x are calculated in Cards ONE1015-17. The terms corresponding to D, E, and F are denoted by P(11), P(12), and P(13) and are evaluated in Cards STPB029-31. The value of the integrand is stored in the variable ONE1 in Card ONE1023. The response $STPW = R_1 P_r(t)$ is then determined in Cards STPB079 and 80.

2.3 The Complex Arithmetic Method for Calculating the Step Wave Response

A second option for calculating the step wave response is provided by the complex arithmetic method. This procedure is based on the equation (Britt (5-2.12))

$$P_r(t) = \frac{2}{\pi} \int_{y_1}^{y_2} \operatorname{Re} \left\{ \frac{u}{\alpha_1 y} (K - K_1) \right\} dy + \frac{2}{\pi R_r} \cdot \operatorname{Im} \left\{ K_1 \log \left[\frac{R_r \omega_2 - d_r t / R_r}{f(\omega_1)} \right] \right\} \quad (2.3.1)$$

where $u = x + iy$ and for $t < t_c = R_r/c_1$

$$\begin{aligned} x &= 0 \\ y_1 &= c_2^{-1} \\ y_2 &= R_r^{-2} [tr - d_r (c_1^{-2} R_r^2 - t^2)^{1/2}] \end{aligned} \quad (2.3.2)$$

For times $t > t_c$ these variables are

$$\begin{aligned} x &= R_r^{-2} d_r (t^2 - c_1^{-2} R_r^2)^{1/2} \\ y_1 &= 0 \\ y_2 &= R_r^{-2} tr. \end{aligned} \quad (2.3.3)$$

The reflection coefficient K for a solid bottom is defined

$$K = \frac{\alpha_1 [(2u^2 + c_4^{-2})^2 - 4u^2 \alpha_2 \alpha_4] - b \alpha_2 c_4^{-4}}{\alpha_1 [(2u^2 + c_4^{-2})^2 - 4u^2 \alpha_2 \alpha_4] + b \alpha_2 c_4^{-4}} \quad (2.3.4)$$

where $\alpha_i = (c_i^{-2} + u^2)^{1/2}$ for $i = 1, 2, 4$.

For a fluid bottom $c_1 = 0$, and the equation for K reduces to

$$K = (\alpha_1 - b\alpha_2)/(\alpha_1 + b\alpha_2). \quad (2.3.5)$$

K_1 is the value of K at $u = x + iy_2$. The other variables used above are as follows:

$$\gamma = [u^2 r^2 + (t - d_r \alpha_1)^2]^{1/2} \quad (2.3.6)$$

$$\omega_1 = [c_1^{-2} + (x + iy_1)^2]^{1/2} \quad (2.3.7)$$

$$\omega_2 = [c_1^{-2} + (x + iy_2)^2]^{1/2} \quad (2.3.8)$$

$$f(\omega_1) = [R_r^2 \omega_1^2 - 2d_r t \omega_1 + (t^2 - c_1^{-2} r^2)]^{1/2} + \omega_1 R_r - d_r t / R_r. \quad (2.3.9)$$

The form of Equation (2.3.1) which is coded is

$$R_i P_r(t) = \left\{ A_2 + \int_{c_1 y_1}^{c_1 y_2} \operatorname{Re} [F \cdot (K - K_1)] dz \right\} / \left(\frac{\pi R_r}{2 R_i} \right) \quad (2.3.10)$$

where $z = c_1 y$,

$$A_2 = \operatorname{Im} \left[K_1 \log \left\{ \frac{[c_1 \omega_2 - c_1 \tau_m \cos \theta]}{[(c_1^2 \omega_1^2 - 2c_1 \tau_m \cos \theta (c_1 \omega_1) + (c_1^2 \tau_m^2 - \sin^2 \theta))^{1/2} + c_1 \omega_1 - c_1 \tau_m \cos \theta]} \right\} \right], \quad (2.3.11)$$

and

$$F = \frac{c_1 u}{c_1 \alpha_1 (c_1 \gamma / R_r)} = \frac{u R_r}{c_1 \alpha_1 \gamma}. \quad (2.3.12)$$

As in the Cagniard-Rosenbaum method, the response $STPW = R_i P_r(t)$ is calculated in SUBROUTINE STPWA for the precursor ($t < t_c$) and in SUBROUTINE STPWB for the main wave ($t > t_c$) using the Gaussian quadrature of FGI to evaluate the integral. The last factor in Equation (2.3.10) is calculated in Cards STPA039 and STPB097. The integrand and A_2 are coded in FUNCTION SEVEN, Cards SEVN035 and 045. The value of $A_2(t)$ is obtained from STPWA by a call to SEVEN with

$z = c_1 y_2$. The function $K_1 = K(x+iy_2)$ is evaluated using the same equations as for $K(u)$ in the integral, namely, RCOE in Cards SEVN022 and 029. In FUNCTION SEVEN the variables brought over by COMMON statements are calculated in the main program, and members of the P array are determined in Cards STPA035-38 for the precursor and in Cards STPB093-96 for the main wave.

2.4 The Stonley Wave Propagation Velocity

The Stonley wave propagation velocity c_{st} is defined as the zeroes $u = \pm i/c_{st} = \pm ik$ of the denominator of the solid bottom reflection coefficient expressed in Equation (2.3.4). Thus $u^2 = -c_{st}^{-2}$ is the solution of the equation

$$\alpha_1 \left[(2u^2 + c_4^{-2})^2 - 4u^2 \alpha_2 \alpha_4 \right] + b \alpha_2 c_4^{-4} = 0, \quad (2.4.1)$$

where $\alpha_1 = (c_1^{-2} + u^2)^{1/2},$

$$\alpha_2 = (c_2^{-2} + u^2)^{1/2},$$

and $\alpha_4 = (c_4^{-2} + u^2)^{1/2}.$

To obtain the form of Equation (2.4.1) which is used in the program, first note that the square roots $\alpha_1, \alpha_2,$ and α_4 are imaginary since c_{st} is known to be smaller than $c_1, c_2,$ and c_4 . Next replace u^2 by $-c_{st}^{-2}$, multiply through by $ic_1 c_2 c_4^4 c_{st}^6$, and set $y_2 = c_{st}^2$ to obtain

$$\begin{aligned} (c_1^2 - y_2)^{1/2} \{ c_2 (y_2 - 2c_4^2)^2 - 4c_4^3 [(c_2^2 - y_2)(c_4^2 - y_2)]^{1/2} \} \\ + bc_1 y_2^2 (c_2^2 - y_2)^{1/2} = 0. \end{aligned} \quad (2.4.2)$$

This equation is solved for y_2 in SUBROUTINE STONL by iteration using the secant method. The variable y_2 is denoted by the FORTRAN symbol Y2. Then c_{st} , called CSTON in the code, is the square root of Y_2 .

2.5 Theory of the Plane Wave Bottom Reflection

In cases where the plane wave bottom reflection is adequate for one's needs or when one wishes to compare these results with the spherical wave reflection, the plane wave option of the BOTREF program can be used. The reflection geometry, incident and critical angles, and the incident pulses are the same as for the spherical wave in Section 2.1; and, unless otherwise noted, the notation is the same.

2.5.1 The Plane Wave Reflection Coefficient and Phase Shift.

The plane wave reflection coefficient K and phase shift ϕ for a non-rigid bottom are calculated from the following equations. For subcritical angles of incidence K and ϕ are

$$K = (A_T - 1)/(A_T + 1) \quad (2.5.1)$$

and $\phi = 0$

where
$$A_T = \cos \theta / [b(\sin^2 \theta_{cr} - \sin^2 \theta)^{1/2}]. \quad (2.5.2)$$

At the critical angle θ_{cr} these expressions reduce to $K = 1$ and $\phi = 0$. At supercritical incidence we have

$$|K| = 1$$

and
$$\phi = 2 \arctan [b(\sin^2 \theta - \sin^2 \theta_{cr})^{1/2} / \cos \theta]. \quad (2.5.3)$$

The above equations are coded in the main program Cards BOTR260-274, 307. The FORTRAN variables CR and E2 denote K and ϕ . If K is complex, then $CR = |K|$.

For a rigid bottom K and ϕ are determined from the equations below. At subcritical incidence, $\theta < \theta_{cr} < \theta_{crs}$, we have

$$\begin{aligned} K &= (A_T + B_T - 1)/(A_T + B_T + 1) \\ \phi &= 0 \end{aligned} \quad (2.5.4)$$

where

$$A_T = \cos \theta [1 - 2\sin^2 \theta / \sin^2 \theta_{crs}]^2 / [b(\sin^2 \theta_{cr} - \sin^2 \theta)^{1/2}] \quad (2.5.5)$$

and

$$\begin{aligned} B_T &= 4\cos \theta \sin^2 \theta (\sin^2 \theta_{crs} - \sin^2 \theta) / \\ &[b \sin^4 \theta_{crs} (\sin^2 \theta_{crs} - \sin^2 \theta)^{1/2}]. \end{aligned} \quad (2.5.6)$$

At the critical angle $\theta = \theta_{cr}$ the equations simplify to $K = 1$ and $\phi = 0$. For an incident angle in the range $\theta_{cr} < \theta < \theta_{crs}$ the reflection coefficient is complex. Its modulus is

$$|K| = \left\{ [A_{TA}^2 + (B_T - 1)^2] / [A_{TA}^2 + (B_T + 1)^2] \right\}^{1/2} \quad (2.5.7)$$

and the phase shift is

$$\phi = \arctan[(1-B_T)/A_{TA}] + \arctan[(1+B_T)/A_{TA}], \quad (2.5.8)$$

where

$$A_{TA} = \cos \theta [1 - 2\sin^2 \theta / \sin^2 \theta_{crs}]^2 / [b(\sin^2 \theta_{cr} - \sin^2 \theta)^{1/2}] \quad (2.5.9)$$

At the critical angle of the shear wave θ_{crs} the equations reduce to

$$|K| = 1$$

and $\phi = 2 \arctan(1/A_{TA})$. (2.5.10)

For angles of incidence $\theta > \theta_{crs}$ we have

$$|K| = 1$$

and $\phi = 2 \arctan[1/(A_{TA} + B_{TA})]$ (2.5.11)

where $B_{TA} = 4 \cos \theta \sin^2 \theta (\sin^2 \theta_{crs} - \sin^2 \theta) / [b \sin^4 \theta_{crs} (\sin^2 \theta - \sin^2 \theta_{crs})^{1/2}]$ (2.5.12)

These equations for the solid bottom reflection coefficient and phase shift are coded in Cards BOTR280-307. As for the fluid bottom, K and ϕ are denoted by CR and E2; and if K is complex, CR= |K|.

2.5.2 The Plane Wave Bottom Reflection Pressure History.

The plane wave bottom reflection pressure history $p_r(t)$ is calculated from the following equations:

when $\theta \leq \theta_{cr}$,

$$p_r = 0 \quad \text{for } t < t_c = R_r/c_1$$

$$p_r = p_F(R_r) K \exp[-(t-t_c)/G_r] \quad \text{for } t \geq t_c \quad (2.5.13)$$

when $\theta > \theta_{cr}$,

$$p_r = p_F(R_r) \frac{|K|}{\pi} \exp[-(t-t_c)/G_r] E_1[(t_c-t)/G_r] \sin \phi \quad \text{for } \delta \leq t < t_c$$

$$p_r = \pm \infty \quad \text{with the sign of } \phi \quad \text{for } t = t_c \quad (2.5.14)$$

$$p_r = p_F(R_r) |K| \exp[-(t-t_c)/G_r] \left\{ \cos \phi - \frac{1}{\pi} Ei \left[(t-t_c)/G_r \right] \sin \phi \right\} \quad \text{for } t > t_c \quad (2.5.15)$$

Note that the plane wave theory has been modified to use $p_F(R_r)$ and $G_r = G(R_r)$ which account for non-linear changes of the shock wave peak pressure and time constant with distance. Also the arrival times of the main wave and precursor have been changed to conform to the spherical wave situation. In the strict plane wave theory

the precursor begins at $t = -\infty$, and the incident wave and the reflected peak arrive simultaneously.

The functions $E_1(x)$ and $Ei(x)$ are the exponential integrals defined for $x > 0$ as

$$E_1(x) = \int_x^{\infty} \frac{\exp(-y)}{y} dy \quad (2.5.16)$$

$$\begin{aligned} Ei(x) &= - \int_{-x}^{\infty} \frac{\exp(-y)}{y} dy = -E_1(-x) \\ &= \int_{-\infty}^x \frac{\exp(y)}{y} dy . \end{aligned} \quad (2.5.17)$$

The function $E_1(x)$ is evaluated using the following approximate formula (see for example Abramowitz and Stegun (5))

$$0 \leq x < 1$$

$$E_1(x) \approx a_0 + a_1 x + a_2 x^2 + a_3 x^3 + a_4 x^4 + a_5 x^5 - \log(x) \quad (2.5.18)$$

$$a_0 = -.57721566$$

$$a_3 = .05519968$$

$$a_1 = .99999193$$

$$a_4 = -.00976004$$

$$a_2 = -.24991055$$

$$a_5 = .00107857$$

$$x \geq 1$$

$$x \exp(x) E_1(x) \approx \frac{x^4 + a_1 x^3 + a_2 x^2 + a_3 x + a_4}{x^4 + b_1 x^3 + b_2 x^2 + b_3 x + b_4} \quad (2.5.19)$$

$$a_1 = 8.5733287$$

$$b_1 = 9.5733223$$

$$a_2 = 18.059017$$

$$b_2 = 25.632956$$

NOLTR 71-110

$$a_3 = 8.6347609$$

$$b_3 = 21.0996531$$

$$a_4 = .26777373$$

$$b_4 = 3.9584969$$

The function $Ei(x)$ is evaluated for $x \leq .5$ using the formula (Reference (5))

$$Ei(x) \approx \gamma + \log(x) + \sum_{n=1}^7 \frac{x^n}{nn!} \quad (2.5.20)$$

where $\gamma = .57721566\dots$ is Euler's constant. For $x > .5$, $Ei(x)$ is obtained from

$$\exp(-x)Ei(x) = \exp(-x)Ei(1) + \int_1^x \frac{\exp(y-x)}{y} dy, \quad (2.5.21)$$

where $Ei(1) = 1.8951178$. The integral is then evaluated using the Gaussian quadrature of FUNCTION FGI.

The reflected pressure $p_r = PBOT$ is calculated in the main program in Cards BOTR861-879. The exponential integrals E_1 and Ei are calculated in the subprograms EXE1 and EXEI, and the integrand of Equation (2.5.21) is coded in FUNCTION EXPO.

3. THE BOTTOM REFLECTION COMPUTER CODE

The **Bottom Reflection Code** has been programmed in FORTRAN IV for use on the CDC 6400 computer at NOL. The code is made up of a main program called BOTREF and the following bottom reflection related subprograms: STONL, STPWA, STPWB, ONE, ONE1, TWO, SEVEN, EXE1, EXE1, EXPO, FGI, PLOT1, and SCAL. In addition, the NOL general purpose plotting program CALCML must be included for the generation of a tape to be plotted on CALCOMP incremental plotters. For NOL users CALCML is available on the subroutine library tape. The control cards which are included in the program listing of Appendix A contain the statements necessary for using CALCML from the library tape. For programmers outside of NOL information on the plotting programs may be obtained from the NOL Mathematics Department (Code 330).

The basic organization of the bottom reflection code is as follows. The main program BOTREF handles all of the input and output and calculates the shock wave peak pressure and time constant and other time independent magnitudes. It performs the time incrementation of the pressure-time histories and calculates the convolution integral, impulse, and energy flux for the spherical wave bottom reflection.

The spherical wave step wave response $P_r(t)$ is obtained by calls from BOTREF to STPWA for the precursor and to STPWB for the main wave. These subroutines in turn set up the integration for $P_r(t)$ using the Gaussian quadrature in FGI. The various integrands described in Sections 2.2 and 2.3 are calculated in subprograms ONE, ONE1, TWO, and SEVEN. The Stonley wave propagation velocity

c_{st} for rigid bottoms is computed in SUBROUTINE STONL on a call from the main program.

The plane wave bottom reflection is also calculated in the main program. Calls to SUBROUTINES EXE1 and EXE2 obtain the exponential integrals E_1 and E_2 which are used to determine the bottom reflection in Equations (2.5.13), (2.5.14), and (2.5.15).

SUBROUTINES PLOT1 and SCAL set up the CALCOMP plots of the pressure-time history. PLOT1 calls SCAL to scale the plot, calls CALCM1 for plotting the axes and the pressure-time curves, and then calls SUBROUTINES SYMBL4 and NUMBR, which are part of the CALCM1 program, to write additional information on the plots.

The Bottom Reflection Program also has an option for calculating the peak translational velocity (PTV) induced in submerged or floating targets by the bottom reflected pulse. Either of the spherical or plane wave reflection theories may be used. The targets are approximated by an infinitely long cylinder of a specified radius, and the PTV Program described in Reference (6), is used to calculate the peak translational velocity. This program uses the additional subroutines PTV, FV, F1, XMAX, VTAB, and PTAB. The PTV is calculated by calling SUBROUTINE PTV (Cards BOTR813L and 813M).

The cards in the main program which are necessary for PTV calculations are denoted by card numbers followed by letters A, B, C, etc. If the bottom reflection program is not to be used for PTV calculations, these cards and the subroutines of the PTV Program may be omitted.

In the following paragraphs the most important FORTRAN symbols of each subprogram are described, and the locations in the program are given where each symbol is calculated.

3.1 FORTTRAN Symbols of the Main Program

Program Input

The input data is read in Statements 3 and 4, Cards BOTR041, 42, and 89, and in Card BOTR101I using the format 8F10.5. These inputs are explained in comment Cards BOTR011-39, 72-87, and 101B-101G. The inputs and their units are as follows:

First Data Card, Statement 3

- WCH charge weight W in pounds or KT
- CWATER sound velocity c_1 of water in ft/sec
- CBOT sound velocity c_2 of the bottom material in ft/sec
- CSHEAR a double purpose input expressing the rigidity of the bottom. If CSHEAR $> .5$, it is the shear wave propagation velocity c_4 of the bottom in ft/sec. If CSHEAR $\leq .5$, it is the dimensionless Poisson ratio from which the shear velocity c_4 is calculated in Card BOTR062. Values of $c_4 \leq .5$ can be neglected.
- RHOWAT density ρ_1 of water in gm/cm³
- RHOBOT density ρ_2 of the bottom material in gm/cm³
- PRECOE coefficient C_p of the pressure similitude equation in psi. PRECOE depends on whether W is in pounds or KT.
- Z5 a control parameter. Z5 greater than zero results in a shorter print out for the spherical wave reflection. See Appendix B to compare the short and long print out.

Second Data Card, Statement 3

- PREEXP exponent n_p of the pressure similitude equation
- THECOE coefficient C_G of the time constant similitude equation in seconds. This variable also depends on the units of W .

NOLTR 71-110

- THEEXP exponent n_G of the time constant similitude equation
- STEPS number of points in the pressure-time history for one time constant G . STEPS = 20. is usually sufficient to obtain a smooth, detailed pressure history. In many cases, STEPS = 10. or 5.0 is adequate.
- DURAT duration of pressure-time history in multiples of the time constant G . If negative, its absolute value is the duration after the arrival of the bottom reflection peak at $t = t_c$. If positive, it is the duration after the direct wave arrival. DURAT = -3.0 is generally sufficient for calculating the significant parts of the bottom reflection.
- X1 CALCOMP plot scaling parameter for the Y-axis in psi per inch of graph. The X-axis is always drawn three inches above the bottom of the graph. The length of the Y-axis is nine inches. Thus the maximum pressure plotted is $6 * X1$, and the minimum is $-3 * X1$. Pressures outside of this range are plotted at the maximum or minimum, whichever is applicable.
- X2 scaling parameter for the X-axis in microseconds per inch of graph. If $X2 \leq 0.$, SUBROUTINE SCAL calculates an appropriate value of X2.
- SLOPE slope of the bottom from charge to gauge in degrees. If the slope is not zero, the internal computing geometry is changed in Cards BOTR170-183. SLOPE must be zero if the geometry changing options of Z2 and THOVAL are used.

Third Data Card, Statement 4

- BIGH water depth H at the charge in feet. BIGH is also used as a control parameter. After completion of each bottom

reflection pressure-history, the program control returns to Statement 4 to read a new set of data. If $BIGH = 0.$, the program stops. If positive, computation continues with the new geometry. If negative, program control transfers to Statement 3 where a new set of charge, physical constants, etc., are read.

D depth d of the charge below the water surface in feet

DGAU depth d_g of the gauge in feet

SMALLR horizontal range r between charge and gauge in feet

THOVAL desired ratio between the bottom reflection incident angle θ and the critical angle θ_{cr} . The variables D and DGAU are changed in Cards BOTR137-142 to obtain this ratio.

SMALLR is not changed. If $THOVAL \leq 0$ the geometry is not changed. See Appendix C for a discussion of this option.

Z1 parameter which selects the theory. When $Z1 = 0.$ the spherical wave Cagniard-Rosenbaum method is used. When $Z1 = 1.0$, the Arons-Yennie plane wave theory is used. And for $Z1 = 3.0$, the complex arithmetic method is used to calculate the spherical wave bottom reflection. Cards BOTR389-443 make the theory selection and write out the appropriate headings.

Z2 arrival time difference between the bottom reflection peak (at $t = t_c$) and the direct wave in microseconds. If $Z2 \leq 0.$, the geometry is not changed. When the geometry is changed, D and DGAU are varied to obtain the desired arrival time difference. SMALLR is not changed, and the change in D is the negative of the change in DGAU so that the incident

angle θ is also unchanged. This geometry change is performed in Cards BOTR121-127. See Appendix C for a discussion of this option.

Z3 plot control parameter. A CALCOMP plot tape is generated if $Z3 = 0$.

Fourth Data Card (BOTR101I), For PTV Calculation

RADIUS cylinder radius in feet. This is the draft or cross-sectional radius of the target vessel. If $RADIUS \leq 0$., the PTV is not calculated.

APRINT controls printing in SUBROUTINE PTV. If $APRINT \leq 0$., the translational velocities calculated in the iteration for the PTV are printed. An example of this printout is given in Table B.1 following the pressure-time history. If $APRINT \geq 0$., the variables TIME1, PTV1, and PTV2 described below are printed from the main program (Card BOTR813N).

Program Output

Appendix B contains examples of the full print out and the shorter print out for the spherical wave reflection and a print out for a plane wave reflection. Most of the variables in the output are self-explanatory; others which are not so well defined are described below.

SMALLH height $h = H - d$ of the charge above the bottom
 DEZERO height $d - d_g$ between the charge and gauge depths
 D2 reduced height d_r/R_i from image charge to gauge
 COSAL $c_1 \sigma$
 COSTH $\cos \theta$
 SINTH $\sin \theta$

DT	increment $\Delta \bar{t}$ of the reduced time \bar{t}
EDT	$\exp(-\Delta \bar{t}/\bar{G}_R)$
T	reduced time \bar{t}
STPW	$R_i P_R(t)$
FI/THETA	$R_i F_I(t)/G_R$
PD	incident pressure p_i in psi
TIME	time in seconds relative to the direct wave arrival time
PBOT	bottom reflected pressure p_r in psi
PS	surface reflected pressure p_s in psi
P	total pressure $p = p_i + p_r + p_s$ in psi. Negative pressures are cut off so that $P + \text{hydrostatic} \geq 0$.
FIMP	total impulse I in psi-sec calculated from the equation $I = \int_{t_0}^t p dt, \text{ where } t_0 \text{ is the minimum of } \delta \text{ and } R_i/c_1$
EFLUX	energy flux E_F in in-psi defined by the equation $E_F = \left(\int_{t_0}^t p p dt \right) / (2.3066 \rho_1 c_1)$
VMID	value of STPW at $t - \Delta t$, $R_i P_R(t - \Delta t)$
PRE	value of STPW at $t - 2\Delta t$
RESID	$R_i \Delta$
RFIMP	reduced impulse $I/w^{1/3}$
REFLUX	reduced energy flux $E_F/w^{1/3}$
POSIMP	impulse of the positive part of the total pressure pulse $p(t)$
RPOSIM	reduced positive impulse, $\text{POSIMP}/w^{1/3}$
TIMEL	time in seconds of the PTV, where time is taken to be zero at the beginning of the bottom reflection
PTV1	the PTV in ft/sec induced by the bottom reflection in a submerged target

PTV2 the PTV in ft/sec induced by the bottom reflection in a target at the surface

Time Independent FORTRAN Symbols

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
B	$b = \rho_1 / \rho_2$	BOTR050
POISR	Poisson ratio $\bar{\sigma} = (.5c_2^2 - c_4^2) / (c_2^2 - c_4^2)$	59
CSHEAR	shear velocity c_4 calculated from $\bar{\sigma}$	62
CSTON	Stonley wave velocity c_{st}	69
SMALLH	h for zero slope	147
RACTU	R_i	151
PH	negative of the hydrostatic pressure at depth d_g	154
RS	reduced surface reflection arrival time	158
W13R	$W^{1/3} / R_i$	162
REDR	$R_i / W^{1/3}$	163
THETA	\bar{G}	164
PACT	$p_F(R_i)$	165
TACT	characteristic time R_i / c_1	166
THET	G in milliseconds	167
A	bottom slope in radians	172
D2ACTU	d_r	187
R2ACTU	R_r	188
CTWO	$\sin \theta_{cr} = c_1 / c_2$	213
R2	reduced bottom reflection slant range	
	R_r / R_i	214
THETAR	$\bar{G}_r = \bar{G}(R_r)$	215
THETR	G_r in milliseconds	216

NOLTR 71-110

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
PACTC	$(R_r/R_i)^{n_G-1}$	BOTR217
R1	$(R_s/R_i)^{n_G}$	218
THETSR	\bar{G}_s	219
SINTH	$\sin \theta$	221
COSTH	$\cos \theta$	222
D2R2	$c_1 \delta/R_i$ for supercritical reflection	226
	$c_1 \delta/R_i$ for subcritical reflection	228
SINAL	$\sin \theta_{cr}$	235
COSAL	$\cos \theta_{cr} = c_1 \sigma$	238
SINBE	$\sin \theta_{crs}$	243
THE	incident angle θ in degrees	245
CR	plane wave reflection coefficient K	261-304
E2	phase shift ϕ in radians	260-307
EE	phase shift ϕ in degrees	308
ANGA	angle of shear wave in bottom in degrees	312, 315
THONE	angle of compression wave in bottom in degrees	319, 321
ALPHA	θ_{cr} in degrees	352
BETHA	θ_{crs} in degrees	358
SHD2R2	reduced arrival time of critically refracted shear wave	363-365
C2	c_1^2	446
CBOT2	c_2^2	447
CSHR2	c_4^2	448
SINTH2	$\sin^2 \theta$	449
CBSH	$-4c_4^3/c_2$	450

NOLTR 71-110

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
C2SHR2	$2c_4^2$	BOTR451
C4CB	$c_1^4 b/c_2$	452
<u>Spherical Wave Pressure-Time Calculations</u>		463-816

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
DT	increment $\Delta \bar{t}$ of reduced time increment $\Delta t' \approx \Delta \bar{t}/8$	BOTR476,500,652 566,609
DT1	original value of $\Delta \bar{t}$	478
DTACT	$2\Delta t/3$	479,501
EDT	$\exp(-\Delta \bar{t}/\bar{G}_r)$	481,503
N	control parameter for pressure history	721,804
VMID	$R_i P_r(\bar{t} - \Delta \bar{t})$	542-662
STPW	$R_i P_r(\bar{t})$	520-690
PRE	$R_i P_r(\bar{t} - 2\Delta \bar{t})$	557-691
FI	convolution integral F_I	556-673
NP	number of subintervals to be used in the Gaussian quadrature integration for $P_r(t)$	196,552-693
V	$v(t)/t_c$ for $t \approx t_c - 2\Delta t'$ where $\Delta t'$ is approximately $\Delta \bar{t}/8$	573
T1	$\bar{t}(V)$	578
T2	$\bar{t}(.75 V)$	579
T3	$\bar{t}(.5 V)$	580
T4	$\bar{t}(.25 V)$	581
U	$u(t_c + 2\Delta t')/t_c$ for $\Delta t' \approx \Delta \bar{t}/8$	616
T2	$\bar{t}(.25 U)$	623
T3	$\bar{t}(.5 U)$	624
T4	$\bar{t}(.75 U)$	625
T5	$\bar{t}(U)$	626

NOLTR 71-110

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
PD	incident pressure $p_i(t)$	BOTR702
PS	surface reflected pressure $p_s(t)$	704
PBOT	bottom reflected pressure $p_r(t)$	707
P	total pressure $p = p_i + p_r + p_s$. Negative values of p are cut off at $p + \text{hydrostatic} \geq 0$.	713

Impulse and Energy Flux Calculations 717-767

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
XP	maximum of pressure p and zero	BOTR720
PMID	pressure p of even numbered time $t - \Delta t$	724
XPMID	maximum of PMID and zero	725
PPRE	pressure p at odd numbered time $t - 2\Delta t$	763
XPPRE	maximum of PPRE and zero	764
PEND	pressure $p(t)$ at odd numbered time	757
XPEND	maximum of PEND and zero	758

Variables Used in the PTV Calculation and in Plotting

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
XX	storage array for time in microseconds for CALCOMP plot. Here time is zero at the arrival of the direct wave.	BOTR800,889
YY	storage array for the total pressure p for plot	801,890
IPMAX	number of plot points stored in XX and YY arrays	807
QX	the array in which the time in seconds is stored for PTV calculations. This time is zero at the beginning of bottom reflection.	802E,813H, 891E

NOLTR 71-110

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
QY	the array in which the bottom reflection pressure $p_r(t)$ is stored for PTV calculations. If $p_r(t)$ is negative, the value stored in QY is calculated so that $p_r(t) + \text{hydrostatic} \geq 0$	BOTR802F, 813I and 891F
TIMER2	arrival time of the peak or singularity of the bottom reflection. Time in this case is measured from the beginning of the bottom reflection.	813C
XT3	signals the approach of the bottom reflection singularity. The value $\text{TIMER2} - 2\Delta t$ is used. The symbol T3 is used for this variable in SUBROUTINE PTV.	813D
XT4	The earliest time at which the translational velocity is to be calculated. The symbol T4 is used instead of XT4 in SUBROUTINE PTV.	813E
XT5	the largest value of time at which the translational velocity is to be calculated. The symbol T5 is used instead of XT5 in SUBROUTINE PTV.	813F
COSA	cosine of the angle which the bottom reflection ray makes with the water surface or a line parallel to the surface if the gauge position is below the surface	813J

NOLTR 71-110

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
PTS	the number of times at which the translational velocity is calculated in the initial search for the PTV. In the call to SUBROUTINE PTV the value PTS = 30. is used.	BOTR813L

Plane Wave Bottom Reflection Variables BOTR819-905

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
SW	direct wave response $p_i(t)/p_F$	BOTR852
PRFL	bottom reflection response $p_r(t)/p_F^i$	857-877
TBTH	$(t - t_c)/G_r$	863
XE1	$\exp(-TBTH) E_1(-TBTH)$	865
XE1	$\exp(-TBTH) Ei(TBTH)$	874

3.2 FORTTRAN Symbols of SUBROUTINE STONL

<u>Symbol</u>	<u>Definition</u>
Y2	$y_2 = c_{st}^2$
FY	Equation (2.4.2) which defines y_2
CK	increment of $y_2 = y_2/1000$
CSTON	Stonley wave velocity c_{st}

3.3 FORTTRAN Symbols of SUBROUTINE STPWA

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
TR	$c_1 \tau_m$	STPA013
V	$c_1 (c_1^{-2} - \tau_m^{-2})^{1/2}$	14
Cagniard-Rosenbaum Method, CONTR = 0.		
P(9)	0. for precursor	20
XM	$c_1 M$	21

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
P(1)	$c_1 (\sigma - M)$	STPA022
P(2)	$4(c_1^2 - \tau_m^2)^{1/2} \sin \theta / (\sigma - M)$	23
P(5)	$c_1 (\sigma + M)$	24
FACTOR	$2 \sqrt{2} b R_i / \pi R_r$	25
STPW	$R_i P_r(t)$	27,43
Complex Arithmetic Method, CONTR = 3.0		
P(1)	$c_1 x = 0.$	35
P(2)	$c_1 \tau_m$	36
P(3)	$c_1 / c_2 = c_1 y_1$	37
P(4)	$c_1 y_2$	38
FACTOR	$\pi R_r / 2R_i$	39
ANS2	A_2	41

3.4 FORTTRAN Symbols of SUBROUTINE STPW

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
TR	$c_1 \tau_m$	STPB014
Cagniard-Rosenbaum Method, CONTR = 0.		
P(9)	1.0 for main wave	20
XK	$c_1 K_m$	23
XL	$c_1^2 L$	24
P(7)	XK	26
P(8)	XL	27
P(11)	$c_1^2 D$	29
P(12)	$c_1^4 E$	30
P(13)	$c_1^2 F$	31
FACTOR	$2bR_i / \pi R_r$	32

NOLTR 71-110

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
TERM1	$(R_i/R_r) (1 - b)/(1 + b)$	STPB038
SIGM	$c_1 \bar{\sigma}$	42
STPW	$R_i P_r(t)$	43,47,79,85,100
XG1	$c_1 g_1$	60
XG3	$c_1 g_3$	61
XG4	$c_1 g_4$	62
XSA	$c_1^2 a/R_i^2$	63
XSF	$c_1^4 f/R_i^4$	64
XNUM	numerator of Γ	65
XDEN	c_1^{-2} times the denominator of Γ	66-67
RESID	$R_i \Delta$	68-69
TERM1	$R_i (1/R_r + \Delta)$	71
SIG2	$c_1 (c_4^{-2} - c_1^{-2})^{1/2}$	78
Complex Arithmetic Method, CONTR = 3.0		
P(1)	$c_1 x = c_1 \cos \theta (\tau_m^2 - c_1^{-2})^{1/2}$	93
P(2)	$c_1 \tau_m$	94
P(3)	$c_1 y_1 = 0$	95
P(4)	$c_1 y_2 = c_1 \tau_m \sin \theta$	96
FACTOR	$\pi R_r / 2R_i$	97
ANS2	A_2	99

3.5 FORTTRAN Symbols of FUNCTION ONE

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
Precursor Variables		
X	integration variable	ONE 008
W	$c_1 w$	20

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
XC2	$c_1^2 (\sigma^2 - \omega^2)$	ONE 021,28
FX	F_x defined by Equation (2.2.4)	23
Main Reflection Variables		
W	integration variable $w = c_1 \omega$	27
FX	F_x defined by Equation (2.2.21)	32
Variables for Rigid Bottom Precursors and Main Waves		
FRCS	$c_4^2 (\omega^2 + c_4^{-2} - c_1^{-2})$	41
XA	A_x	43
XB	B_x	44
XC	C_x	50
ONE	rigid bottom integrands defined in Equations (2.2.9), (2.2.14), and (2.2.32) and the integrand of the first integral in Equation (2.2.37).	47,51
ONE	fast non-rigid bottom integrands of Equations (2.2.3) and (2.2.21)	55

3.6 FORTTRAN Symbols of FUNCTION ONE1

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
X	integration variable $x = c_1 \bar{\omega}$	ONE1008
XAB	$c_1^5 \bar{A} = \bar{A}_x$	15
XBB	$c_1^5 \bar{B} = \bar{B}_x$	16
XCB	$c_1^5 \bar{C} = \bar{C}_x$	17
FAB	\bar{F}_A defined by Equation (2.2.38)	19
FBB	\bar{F}_B defined by Equation (2.2.39)	20,21
ONE1	integrand $\bar{F}_A \bar{F}_B$ of the second integral in Equation (2.2.37)	23

3.7 FORTRAN Symbols in FUNCTION TWO

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
X	integration variable $x = c_1 \bar{w}$	TWO 007
FAB	\bar{F}_A defined by Equation (2.2.28)	13
FBB	\bar{F}_B defined by Equation (2.2.29)	14,15
TWO	integrand in Equation (2.2.27)	17

3.8 FORTRAN Symbols in FUNCTION SEVEN

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
Z	integration variable $z = c_1 y$	SEVN007
V	$c_1 u$	14
RCOE	non-rigid bottom K defined by Equation (2.3.5)	22
	rigid bottom K defined by Equation (2.3.4)	29
F	F defined in Equation (2.3.12)	34
SEVEN	integrand of Equation (2.3.10)	35
	A_2 defined by Equation (2.3.11)	
RT5	K_1	39
U1	$x + iy_1$	40
U2	$c_1^2 \omega_1^2$	41
U3	$c_1 \omega_1$	42
XB	$-c_1 \tau_m \cos \theta$	43

3.9 FORTRAN Symbols in SUBROUTINE EXE1

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
A	array a_i in Equation (2.5.19)	EXE1008
B	array b_i in Equation (2.5.19)	9
C	array a_i in Equation (2.5.13)	10,11
X	x	12

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
ANS	$\exp(x) E_1(x)$ for $x \geq 1$	EXE1014,15
	$\exp(x) E_1(x)$ for $0 \leq x < 1$	17,18

3.10 FORTTRAN Symbols in SUBROUTINE EXEI

<u>Symbol</u>	<u>Definition</u>	<u>Card Number</u>
Y	x	EXEI006
A	array of $1/nn!$ for $n = 2,3,\dots,7$ in Equation (2.5.20)	9
U	sum of the series in Equation (2.5.20)	11
ANS	$\exp(-x)E_i(x)$ using Equation (2.5.20)	12
ANS1	integral in Equation (2.5.21) evaluated using the Gaussian quadrature of FUNCTION FGI	15
ANS	$\exp(-x)E_i(x)$ using Equation (2.5.21)	16

3.11 FORTTRAN Symbols in FUNCTION EXPO

<u>Symbol</u>	<u>Definition</u>
X	integration variable y in Equation (2.5.21)
P(1)	$x = (t - R_r/c_1)/G_r$
EXPO	integrand $\exp(y - x)/y$ in Equation (2.5.21)

3.12 FORTTRAN Symbols in FUNCTION FGI

<u>Symbol</u>	<u>Definition</u>
A	lower limit of integration
B	upper limit of integration

NOLTR 71-110

Symbol

Definition

K	number of subintervals into which the integration interval (A,B) is divided. The integral in each subinterval is evaluated using a 4 point Gaussian quadrature.
F	integrand of the integral to be evaluated
P	array used to transfer parameters to the function F
FGI	value of the integral of F between A and B

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APPENDIX A

FORTRAN IV LISTING OF THE BOTTOM

REFLECTION PROGRAM BOT REF

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BDCROT, T600, CM70000, 52400311, 047, BRITT,
ATTACH(ABC, NOLBIN)
COPYN(0, DEF, ABC)
FTN(L)
LOAD(LGO)
REQUEST, TAPE99, LO, (CALCOMP/RING)
OEF,
' RECORD SEPARATOR =(7-8-9) PUNCH IN COLUMN 1
REWIND(ABC)
CALCM1, 13, ABC
' RECORD SEPARATOR =(7-8-9) PUNCH IN COLUMN 1

PROGRAM BOTREF (INPUT, OUTPUT, TAPE5=INPUT, TAPE6=OUTPUT, TAPE99)
C BOTR001
C BOTR002
C BOTTOM REFLECTION PROGRAM (COC 6400 COMPUTER) BOTR003
C DIMENSION XX(1000), YY(1000) BOTR004
C COMMON /QXY/QX(1000), QY(1000) BOTR004A
C COMMON B, COSAL, COSTH, R2, SINBE, SINTH, CWATER, CBOT, CSHEAR, CSTON, RESID BOTR005
C COMMON C2, CBOT2, CSHR2, CBSH, C2SHR2, C4CB, SINTH2 BOTR006
C ADATF = DATE(0) BOTR007
C ICASF=1 BOTR008
C PI=3.1415926 BOTR009
C BOTR010
C READ INPUT DATA (FORMAT -- 8F10.5) BOTR011
C BOTR012
C WCH----EXPLOSIVE CHARGE WEIGHT (LBS OR KT) BOTR013
C CWATER---SOUND VELOCITY OF WATER (FT/SEC) BOTR014
C CBOT----SOUND VELOCITY OF THE BOTTOM MATERIAL (FT/SEC) BOTR015
C CSHEAR---IF CSHEAR GT 0.5, IT IS THE SHEAR WAVE PROPAGATION BOTR016
C VELOCITY OF THE BOTTOM (FT/SEC). IF CSHEAR LE 0.5, IT IS BOTR017
C THE DIMENSIONLESS POISSON RATIO. BOTR018
C RHOVAT---DENSITY OF WATER (GM/CC) BOTR019
C RHOBOT---DENSITY OF BOTTOM MATERIAL (GM/CC) BOTR020
C PRECOE---COEFFICIENT OF PRESSURE SIMILITUDE EQUATION (PSI) BOTR021
C Z5-----CONTROL PARAMETER. Z5 GREATER THAN ZERO RESULTS IN A BOTR022
C SHORTER PRINT OUT. BOTR023
C PREFXP---EXPONENT OF PRESSURE SIMILITUDE EQUATION BOTR024
C THECOE---COEFFICIENT OF TIME CONSTANT SIMILITUDE EQUATION (SEC) BOTR025
C THEFXP---EXPONENT OF TIME CONSTANT SIMILITUDE EQUATION BOTR026
C STEPS---NUMBER OF POINTS IN P-T CURVE FOR ONE TIME CONSTANT BOTR027
C DURAT---DURATION OF PRESSURE TIME HISTORY IN MULTIPLES OF THE BOTR028
C TIME CONSTANT. (IF NEGATIVE, ITS ABSOLUTE VALUE IS THE BOTR029
C DURATION AFTER THE ARRIVAL OF THE BOTTOM REFLECTION BOTR030
C PEAK. IF POSITIVE, IT IS THE DURATION AFTER THE BOTR031
C DIRECT WAVE ARRIVAL.) BOTR032
C X1-----CALCOMP PLOT SCALING PARAMETER FOR THE Y-AXIS (PSI PER BOTR033
C INCH OF GRAPH) BOTR034
C X2-----SCALING PARAMETER FOR THE X-AXIS (MICROSECONOS PER BOTR035
C INCH OF GRAPH) BOTR036
C SLOPE---SLOPE OF BOTTOM FROM CHARGE TO GAUGE (DEGREES) BOTR037
C BOTR038
C ADDITIONAL DATA IS READ IN STATEMENT 4 (CARD BOTR089) BOTR039

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C
C 3 READ(5,554)WCH,CWATER,CBOT,CSHEAR,RHOWAT,RHOBOT,PRECOE,Z5
C READ (5,554) PREEXP,THECOE,THEEXP,STEPS,DURAT,X1,X2,SLOPE
C FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS BOTR907-1042
C
C DURAT IN PRINT OUT IS THE DURATION AFTER THE DIRECT ARRIVAL
C STORE ORIGINAL DURAT
C
C XDURAT=DURAT
C
C B=RHOWAT/RHOBOT
C
C POISSON RATIO
C
C IF CSHEAR IS 0.5 FT/SEC OR LESS, THE POISSON RATIO POISR IS SET
C EQUAL TO CSHEAR AND THE SHEAR VELOCITY IS CALCULATED.
C
C IF (CSHEAR.LE.0.) GO TO 39
C IF (CSHEAR.LE.0.5) GO TO 42
44 POISR=(0.5*CBOT**2-CSHEAR**2)/(CBOT**2-CSHEAR**2)
C GO TO 41
42 POISR=CSHEAR
C CSHEAR=CBOT*SQRT((0.5-POISR)/(1.-POISR))
C GO TO 41
39 POISR=0.5
C
C
C STONEY WAVE PROPAGATION VELOCITY
C
C 41 CALL STONL
C
C THE GEOMETRY IS NOW READ IN (FORMAT -- 8F10.5)
C
C BIGH----WATER DEPTH AT THE CHARGE (FT). ALSO USED AS A
C CONTROL VARIABLE. SEE CARDS BOTR094-97 BELOW
C D-----DEPTH OF THE CHARGE BELOW THE WATER SURFACE (FT)
C DGAU----DEPTH OF THE GAUGE (FT)
C SMALLR--HORIZONTAL RANGE BETWEEN CHARGE AND GAUGE (FT)
C THOVAL--DESIRED RATIO BETWEEN THE BOTTOM REFLECTION INCIDENT AND
C CRITICAL ANGLES. (IF THOVAL LE 0., THE INPUT GEOMETRY IS
C NOT CHANGED.) SEE APPENDIX C OF NOLTR 71-110.
C Z1-----PARAMETER WHICH SELECTS THEORY. (SEE CARDS BOTR391-394)
C Z2-----ARRIVAL TIME OF THE MAIN BOTTOM REFLECTION AFTER THE
C DIRECT ARRIVAL (MICROSECONDS). GEOMETRY IS UNCHANGED IF
C Z2 LE 0. SEE APPENDIX C OF NOLTR 71-110.
C Z3-----PLOT CONTROL PARAMETER. A CALCOMP PLOT TAPE IS GENERATED
C IF Z3 IS ZERO .
C
C 4 READ(5,554)BIGH,D,DGAU,SMALLR,THOVAL,Z1,Z2,Z3
C FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS BOTR907-1042
C AFTER COMPLETION OF EACH CASE PROGRAM CONTROL RETURNS TO
C STATEMENT 4. DEPENDING ON BIGH, THE CALCULATIONS ARE CONTINUED AS
C FOLLOWS
C IF BIGH = 0 PROGRAM STOPS
C IF BIGH IS POSITIVE COMPUTATION CONTINUES USING THE PRESENT INPUT
C IF BIGH IS NEGATIVE PROGRAM TRANSFERS TO STATEMENT 3 WHERE
C ANOTHER SET OF CHARGE, PHYSICAL CONSTANTS, ETC. ARE READ.
C
C IF (BIGH)3.1000,6
1000 STOP
C 6 WRITE(6,510)ADATE
C FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS BOTR907-1042
C ADDITIONAL DATA IS READ IN FOR PTV CALCULATION (FORMAT -- 8F10.5)

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BOTR040
 BOTR041
 BOTR042
 BOTR043
 BOTR044
 BOTR045
 BOTR046
 BOTR047
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 BOTR101
 BOTR101A
 BOTR101B

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C
C   RAOUS--CYLINDER RADIUS IN FEET. THIS IS THE DRAFT OR CROSS-
C           SECTIONAL RADIUS OF THE TARGET VESSEL.
C   APRINT--CONTROLS PRINTING IN SUBROUTINE PTV. THE ITERATIONS TO
C           OBTAIN THE PTV ARE PRINTED OUT IF APRINT .LE. 0.
C
C   READ(5,554) RADIUS,APRINT
C   IPTV=0
C   A=0.
1006 WRITE(6,550) ICASE
C   WRITE(6,511) BIGH
C   WRITE(6,513) D
C   WRITE(6,512) DGAU
C   FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS ROTR907-1042
C           CHANGE OF DEPTH OF EXPLOSION AND OF GAUGE FOR GIVEN ARRIVAL
C           TIME Z2 OF THE BOTTOM REFLECTION PEAK AFTER THE DIRECT
C           ARRIVAL. THE BOTTOM REFLECTION INCIDENT ANGLE AND SMALLR
C           ARE UNCHANGED. SEE APPENDIX C OF NOLTR 71-110.
C
C   GEOMETRY IS UNCHANGED IF Z2 IS NEGATIVE OR ZERO.
C
C   -D2ACTU- IS THE ORIGINAL TOTAL DISTANCE BETWEEN THE GAGE AND THE
C   IMAGE CHARGE, -R2ACTU- IS THE ORIGINAL TOTAL SLANT DISTANCE FROM
C   THE GAGE TO THE IMAGE CHARGE AND -HG- IS THE NEW VALUE OF THE
C   HEIGHT OF THE GAGE ABOVE THE BOTTOM. NEW VALUES FOR -D- AND
C   -DGAU- ARE CALCULATED.
C
C   IF(Z2.LE.0.) GO TO 2
1  D2ACTU=2.*(BIGH-D)+D-DGAU
C   R2ACTU=SQRT(SMALLR**2+D2ACTU**2)
C   DELR=Z2*(1.E-06)*CWATER
C   CR20D2=DELR*(2.*R2ACTU-DELR)/D2ACTU**2
C   HG=0.5*D2ACTU*(1.-SQRT(1.-CR20D2))
C   DGAU=BIGH-HG
C   D=BIGH-D2ACTU+HG
C   WRITE(6,553) Z2
C   WRITE(6,513) D
C   WRITE(6,512) DGAU
C
C   CHANGE OF GEOMETRY TO OBTAIN THE DESIRED RATIO
C   BETWEEN INCIDENT AND CRITICAL ANGLE=THOVAL
C   GEOMETRY IS UNCHANGED IF THOVAL IS LESS THAN OR EQUAL TO ZERO.
C   SEE APPENDIX C OF NOLTR 71-110.
2  IF(THOVAL.LE.0.) GO TO 5
7  TH=THOVAL*ASIN(CWATER/CBOT)
C   D2ACTU=SMALLR*COS(TH)/SIN(TH)
C   O=2.*BIGH-DGAU-D2ACTU
C   IF(BIGH-D) 8,9,9
8  D = BIGH
C   DGAU = BIGH - D2ACTU
9  WRITE(6,537)
C   WRITE(6,512) DGAU
C   WRITE(6,513) D
C   GEOMETRY
5  SMALLH=BIGH-D
C
C   -RACTU- IS THE SLANT DISTANCE BETWEEN CHARGE AND GAGE.
C
C   RACTU = SQRT((D-DGAU)**2+SMALLR**2)
C
C   CALCULATE HYDROSTATIC PRESSURE -PH-
C   PH=-14.7*DGAU/33.0-14.7
C

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BOTR101C
 BOTR101D
 BOTR101E
 BOTR101F
 BOTR101G
 BOTR101H
 BOTR101I
 BOTR101J
 BOTR101K
 BOTR102
 BOTR103
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 BOTR145
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 BOTR147
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 BOTR149
 BOTR150
 BOTR151
 BOTR152
 BOTR153
 BOTR154
 BOTR155

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C      -RS- IS THE REDUCED ARRIVAL TIME OF ACOUSTIC SURFACE REFLECTION,      BOTR156
C      RS=SQRT(SMALLR**2*(D+DGAU)**2)/RACTU                                  BOTR157
C                                                                              BOTR158
C      EXPONENTIAL PULSE PEAK PRESSURE AND TIME CONSTANT CALCULATED        BOTR159
C                                                                              BOTR160
C      W13R=WCH**(1./3.)/RACTU                                              BOTR161
C      REDR=1./W13R                                                         BOTR162
C      THETA=THECOE*(W13R)**(1.+THEEXP)*CWATER                             BOTR163
C      PACT=PRECOE*(W13R)**PREEXP                                           BOTR164
C      TACT=RACTU/CWATER                                                     BOTR165
C      THET=THETA*TACT*1000.                                                BOTR166
C      IF (SLOPE .EQ. 0.) GO TO 10005                                       BOTR167
C                                                                              BOTR168
C      CHANGE OF GEOMETRY FOR SLOPING BOTTOM                                BOTR169
C                                                                              BOTR170
C      A = SLOPE/57.29578                                                    BOTR171
C      HG = HIGH-DGAU                                                        BOTR172
C      H1 = SMALLH*COS(A)                                                    BOTR173
C      H2 = (HG-SMALLR*TAN(A))*COS(A)                                        BOTR174
C      IF ((H2.LT.0.0).OR.(SMALLR*TAN(A).GT.BIGH)) WRITE(6,555)            BOTR175
C      WRITE (6,514) SMALLR                                                  BOTR176
C      WRITE (6,574)                                                         BOTR177
C      SMALLR=SMALLR*COS(A)+(D-DGAU)*SIN(A)                                  BOTR178
C      SMALLH = H1                                                           BOTR179
C      IF (H2 .GT. H1) D=D+H2-H1                                            BOTR180
C      DGAU = D+H1-H2                                                        BOTR181
C      BIGH = D+H1                                                           BOTR182
C      WRITE (6,511) BIGH                                                    BOTR183
C      WRITE (6,512) DGAU                                                    BOTR184
10005 DEZERO = D-DGAU                                                        BOTR185
C      D2ACTU=2.*SMALLH+DEZERO                                              BOTR186
C      R2ACTU=SQRT(D2ACTU**2+SMALLR**2)                                     BOTR187
C                                                                              BOTR188
C      INITIALIZATIONS                                                       BOTR189
C                                                                              BOTR190
C      1040 FI=0.                                                            BOTR191
C      VMID=0.                                                                BOTR192
C      PRE=0.                                                                  BOTR193
C      IP=1                                                                    BOTR194
C      NP=4                                                                    BOTR195
C      ZZDT=4.                                                                BOTR196
C      RESID=0.0                                                            BOTR197
C      EFLUY=0.                                                              BOTR198
C      FIMP=0.                                                                BOTR199
C      POSINP=0.                                                            BOTR200
C      IPRES=1                                                                BOTR201
C      PPRE=0.                                                                BOTR202
C      XPPRF=0.                                                              BOTR203
C      PMID=0.                                                                BOTR204
C      XPMID=0.                                                              BOTR205
C      PD=0.                                                                  BOTR206
C      PS=0.                                                                  BOTR207
C      PBOT=0.                                                                BOTR208
C                                                                              BOTR209
C      BASIC CONSTANTS OF GROUND WAVE                                        BOTR210
C                                                                              BOTR211
C      CTWO=CWATER/CBOT                                                     BOTR212
C      R2=R2ACTU/RACTU                                                       BOTR213
C      THETAAR=THETA/R2**THEEXP                                             BOTR214
C      THETAR=THET/R2**THEEXP                                              BOTR215
C      PACTC=R2**PREEXP/R2                                                 BOTR216
C      R1=R2**PREEXP                                                         BOTR217
C      THETAR=THETA/R1**THEEXP                                             BOTR218
C                                                                              BOTR219

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	D2=D*ACTH/RACTU	BOTR220
	SINTH=SMALLR/(RACTU*R2)	BOTR221
	COSTH=D2/R2	BOTR222
	COSRM=COSTH/R2	BOTR223
	IF(CTWO.GE.SINTH) GO TO 2000	BOTR224
	GAM=SQRT(1.-CTWO**2)	BOTR225
	D2R2=R2*(CTWO*SINTH+GAM*COSTH)	BOTR226
	GO TO 2005	BOTR227
2000	D2R2=R2	BOTR228
C		BOTR229
C	CALCULATION OF NEW DURATION IF DURAT IS READ IN NEGATIVE.	BOTR230
C	(DURAT IS EXPLAINED IN CARDS BOTR028-32.)	BOTR231
2005	IF(XDURAT.LT.0.) DURAT=(R2-1.)/THETA-XDURAT	BOTR232
	TSTOP=1.+DURAT*THETA	BOTR233
C		BOTR234
	SINAL=CWATER/CR0T	BOTR235
	SIN2AL=SINAL**2	BOTR236
	IF(SIN2AL-1.)811,811,812	BOTR237
811	COSAL=SQRT(1.-SIN2AL)	BOTR238
	GO TO 813	BOTR239
812	COSAL=-0.	BOTR240
813	SIN2TH=SINTH**2	BOTR241
	IF(CSHEAR)15,15,14	BOTR242
14	SINBE=CWATER/CSHEAR	BOTR243
	SIN2BE=SINBE**2	BOTR244
15	THE=57.2958* ASIN(SINTH)	BOTR245
		BOTR246
C		BOTR247
C		BOTR248
C	CALCULATION OF PLANE WAVE REFLECTION COEFFICIENT, PHASE SHIFT,	BOTR249
C	AND ANGLE OF S-WAVE.	BOTR250
C		BOTR251
	IF(CSHEAR)30,30,50	BOTR252
C		BOTR253
C	CALCULATION FOR BOTTOM WITH NO SHEAR STRENGTH (NON-RIGID)	BOTR254
C		BOTR255
30	IF(SIN2TH-SIN2AL)33,32,31	BOTR256
		BOTR257
C		BOTR258
C	SUPERCritical REFLECTION (ANGLE OF INCIDENCE GREATER THAN THE	BOTR259
C	CRITICAL ANGLE)	BOTR260
C		BOTR261
31	E=ATAN(B*SQRT(SIN2TH-SIN2AL)/COSTH)	BOTR262
	CR=1.	BOTR263
	IICA=1	BOTR264
	GO TO 88	BOTR265
32	E2=0.	BOTR266
	CR=1.	BOTR267
	IICA=2	BOTR268
	GO TO 89	BOTR269
C		BOTR270
C	SUBCRITICAL REFLECTION (ANGLE OF INCIDENCE LESS THAN THE	BOTR271
C	CRITICAL ANGLE)	BOTR272
C		BOTR273
33	E2=0.	BOTR274
	AT=COSTH/(SQRT(SIN2AL-SIN2TH)*B)	BOTR275
	CR=(AT-1.)/(AT+1.)	BOTR276
	IICA=3	BOTR277
	GO TO 89	BOTR278
C		BOTR279
C	CALCULATION FOR BOTTOM WITH SHEAR (RIGID)	BOTR280
C		BOTR281
50	CA=COSTH*(1.-2.*SIN2TH/SIN2BE)**2/B	BOTR282
	CB=4.*COSTH*SIN2TH*(SIN2BE-SIN2TH)/B/SIN2BE**2	BOTR283
	IF(SIN2TH-SIN2AL)60,32,51	
51	ATA=CA/SQRT(SIN2TH-SIN2AL)	

	IF (SINTH=SINBE) 52,55,57	
52	BT=CB/SQRT(SIN2BE-SIN2TH)	BOTR284
	CR=SQRT((ATA**2*(BT-1.)**2)/(ATA**2*(BT+1.)**2))	BOTR285
	EA=ATAN((1.-BT)/ATA)	BOTR286
	EB=ATAN((1.+BT)/ATA)	BOTR287
	E2=EA+EB	BOTR288
	IICA=4	BOTR289
	GO TO 89	BOTR290
55	BTA=0.	BOTR291
	GO TO 58	BOTR292
C		BOTR293
	57 BTA=CB/SQRT(SIN2TH-SIN2BE)	BOTR294
	58 E=ATAN(1./(ATA+BTA))	BOTR295
	CR=1.	BOTR296
	IICA=6	BOTR297
	GO TO 88	BOTR298
C		BOTR299
	60 E=0.	BOTR300
	AT=CA/SQRT(SIN2AL-SIN2TH)	BOTR301
	BT=CB/SQRT(SIN2BE-SIN2TH)	BOTR302
	CR=(AT+BT-1.)/(AT+BT+1.)	BOTR303
	IICA=7	BOTR304
C		BOTR305
	88 E2=2.*E	BOTR306
	89 EE=57.2958*E2	BOTR307
	IF(CSHEAR,LE.0.) GO TO 92	BOTR308
	90 IF(SINTH-SINBE) 91,91,92	BOTR309
	91 GAMMA= ASIN(SINTH/SINBE)	BOTR310
	ANGA=57.2958*GAMMA	BOTR311
	GO TO 95	BOTR312
C		BOTR313
	92 ANGA=-0.	BOTR314
C		BOTR315
	ANGLE OF P-WAVE	BOTR316
C		BOTR317
	95 IF(SINTH-SINAL) 293,293,294	BOTR318
	293 THONF=57.2958* ASIN(SINTH/SINAL)	BOTR319
	GO TO 295	BOTR320
	294 THONF=-0.	BOTR321
C		BOTR322
	FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS BOTR907-1042	BOTR323
C		BOTR324
	295 WRITE(6,514)SMALLR	BOTR325
	WRITE(6,504)WCH	BOTR326
	WRITE(6,505)CWATER	BOTR327
	WRITE(6,506)CBOT	BOTR328
	WRITE(6,546)CSHEAR	BOTR329
	WRITE(6,507)RHOWAT	BOTR330
	WRITE(6,508)RHOBOT	BOTR331
	WRITE(6,515)PRECOE	BOTR332
	WRITE(6,503)Z5	BOTR333
	WRITE(6,516)PREEXP	BOTR334
	WRITE(6,517)THECOE	BOTR335
	WRITE(6,518)THEEXP	BOTR336
	WRITE(6,519)STEPS	BOTR337
	WRITE(6,509)DURAT	BOTR338
	WRITE(6,538)THOVAL	BOTR339
	WRITE(6,584) X1	BOTR340
	WRITE(6,585) X2	BOTR341
	WRITE(6,500) SLOPE	BOTR342
	WRITE(6,586) Z1	BOTR343
	WRITE(6,587) Z2	BOTR344
	WRITE(6,588)Z3	BOTR344A
	WRITE(6,568) RADIUS	BOTR344B
	WRITE(6,569) APRINT	BOTR345
	WRITE(6,520)	

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WRITE(6,521)THE
WRITE(6,573)CSTON
WRITE(6,545)POISR
WRITE(6,547)RS
C
IF(1.-SINAL)17,16,16
16 ALPHA=57.2958* ASIN(SINAL)
WRITE(6,522)ALPHA
GO TO 18
17 WRITE(6,541)
18 IF(CSHEAR)49,49,45
45 IF(1.-SINBE)47,46,46
46 BETHA=57.2958* ASIN(SINBE)
WRITE(6,542)BETHA
C
ARRIVAL TIME OF CRITICALLY REFRACTED SHEAR WAVE
C
IF(SINTH.LT.SINBE) .SHD2R2=0.
IF(SINTH.GE.SINBE ) SHD2R2=(SMALLR*SINBE+O2ACTU*SQRT(1.-SIN2BE))
1 /RACTU
WRITE(6,579) SHD2R2
C
GO TO 49
47 WRITE(6,543)
49 WRITE(6,597)THONE
WRITE(6,592)CR
WRITE(6,594)ANGA
WRITE(6,593)EE
WRITE(6,523)D2R2
WRITE(6,533)R2
WRITE(6,525)RACTU
WRITE(6,502) REDR
WRITE(6,526)TACT
WRITE(6,527)PACT
WRITE(6,528)THETA
WRITE(6,539)THET
WRITE(6,548) THETAR
WRITE(6,549) THETR
WRITE(6,535)
WRITE(6,551)SMALLH,OZFR0,O2,COSAL,COSTH,SINTH
WRITE(6,532)
FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS BOTR907-1042
C
SELECTION OF THE THEORY
C
Z1=0. ROSENBAUM METHOD
C
Z1=1. PLANE WAVE APPROXIMATION
C
Z1=2. NOT USED IN THE PRESENT PROGRAM
C
Z1=3. COMPLEX ARITHMETIC METHOD
C
Z1=ARS(Z1)
IF(Z1=1.)800,801,802
802 IF(Z1=3.)803,804,805
C
SPHERICAL WAVE CAGNIARD-ROSENBAUM THEORIES
C
800 WRITE(6,567)
IF(CSHEAR)820,820,821
C
NON-DIGID BOTTOM
C
820 IF(SINAL=1.)830,831,832
C
FAST BOTTOM
C
830 WRITE(6,560)

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BOTR346
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 BOTR409

C	GO TO 11	BOTR410
C	SLOW BOTTOM	BOTR411
C	832 WRITE(6,561)	BOTR412
C	GO TO 11	BOTR413
C	NO REFLECTION	BOTR414
C	831 WRITE(6,599)	BOTR415
C	GO TO 4	BOTR416
C	RIGID BOTTOM	BOTR417
C	821 IF(SINBE-1.)841,841,840	BOTR418
C	FAST SHEARWAVE	BOTR419
C	841 WRITE(6,562)	BOTR420
C	GO TO 11	BOTR421
C	SLOW SHEARWAVE	BOTR422
C	840 WRITE(6,563)	BOTR423
C	GO TO 11	BOTR424
C	PLANE WAVE APPROXIMATION	BOTR425
C	801 WRITE(6,565)	BOTR426
C	GO TO 998	BOTR427
C	Z1 = 2. IS NOT NEEDED FOR THE PRESENT PROGRAM	BOTR428
C	803 WRITE(6,599)	BOTR429
C	GO TO 4	BOTR430
C	COMPLEX ARITHMETIC METHOD	BOTR431
C	804 WRITE(6,566)	BOTR432
C	CONSTANTS FOR SUBROUTINE SEVEN	BOTR433
C	C2=C*WATER**2	BOTR434
C	CBOT=C*CRNT**2	BOTR435
C	CSHR=C*CSHEAR**2	BOTR436
C	SINTH2=SINTH**2	BOTR437
C	CHSH=-4.*CSHEAR**3/CBOT	BOTR438
C	C2SH=2.*CSHEAR**2	BOTR439
C	C4CB=C2**2/CBOT*B	BOTR440
C	GO TO 11	BOTR441
C	Z1=4. IS NOT NEEDED FOR PRESENT PROGRAM	BOTR442
C	805 WRITE(6,599)	BOTR443
C	GO TO 4	BOTR444
C	*****	BOTR445
C	*****	BOTR446
C	*****	BOTR447
C	*****	BOTR448
C	*****	BOTR449
C	*****	BOTR450
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C	*****	BOTR461
C	*****	BOTR462
C	*****	BOTR463
C	*****	BOTR464
C	*****	BOTR465
C	*****	BOTR466
C	*****	BOTR467
C	*****	BOTR468
C	*****	BOTR469
C	*****	BOTR470
C	*****	BOTR471
C	*****	BOTR472
C	*****	BOTR473

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M=4*M+5
C 12 CALCULATE DT, INCREMENT OF REDUCED TIME T
DT=(R2-D2R2)/FLOAT(M)/2.
IF((D2R2-1.0).GT.0.) DT=(R2-1.0)/FLOAT(M)/2.
DT1=DT
DTACT=2.*DT*TACT/3.
DST=DT/3.
EDT=EXP(-DT/THETA)
WRITE(6,536)DT,EDT
WRITE(6,532)
C FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS ROTR907-1042
C -Z5- IS THE PRINTOUT CONTROL PARAMETER. IF -Z5- GREATER THAN
C ZERO A SHORTER PRINTOUT RESULTS, IF -Z5- EQUALS ZERO THE NORMAL,
C LONGER PRINTOUT IS GENERATED.
IF (Z5.GT. 0.0) GO TO 103
WRITE (6,530)
GO TO 100
103 WRITE(6,501)
GO TO 100
C
C ANGLE OF INCIDENCE LESS THAN OR EQUAL TO CRITICAL
C
20 MM=(Z2-1.)*STEPS/THETA
MM=2*MM+4
C CALCULATE DT, INCREMENT OF REDUCED TIME T
DT=(Z2-1.)/FLOAT(MM)
DTACT=2.*DT*TACT/3.
DST=DT/3.
EDT=EXP(-DT/THETA)
WRITE(6,536)DT,EDT
WRITE(6,532)
IF(Z5.GT.0.0) GO TO 104
WRITE(6,530)
GO TO 700
104 WRITE(6,501)
GO TO 700
C
C ANGLE OF INCIDENT WAVE LARGER THAN CRITICAL
C R2 LARGER THAN D2R2
C
100 IF(D2R2=0.9999)101,102,102
C
C PRECURSOR ARRIVES BEFORE DIRECT WAVE
C
101 T=D2R2
STPW=0.
N=10
GO TO 72
C
C PRESSURE CALCULATION IF DIRECT WAVE ARRIVES BEFORE PRECURSOR
C
102 T=1.0
STPW=0.
N=1
GO TO 71
110 N=12
114 T=T+2.*DT
IF(T.LT.D2R2) GO TO 71
117 T=D2R2
WRITE(6,534)
N=11
GO TO 71
C

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BOTR474
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NOLTR 71-110

C	CALCULATION OF THE PRECURSOR	BOTR538
C	150 N=2	BOTR539
	152 T=T+DT	BOTR540
	CALL STPWA(T,VMID,Z1,NP)	BOTR541
C	EVERY FIFTH STEP RECALCULATE STPW WITH TWICE THE INTEGRATION	BOTR542
C	POINTS,NP, TO CHECK THE INTEGRATION ERROR.	BOTR543
	IF(MOD(IP,5).NE.0) GO TO 155	BOTR544
	IF(NP.EQ.16) GO TO 155	BOTR545
	NP2=2*NP	BOTR546
	CALL STPWA(T,VMID2,Z1,NP2)	BOTR547
	ERROR=ABS((VMID-VMID2)/VMID2)	BOTR548
	IF(ERROR.LT.0.005) GO TO 155	BOTR549
C	DOUBLE NUMBER OF INTEGRATION POINTS	BOTR550
	NP=NP2	BOTR551
	VMID=VMID2	BOTR552
	155 T=T+DT	BOTR553
	CALL STPWA(T,STPW,Z1,NP)	BOTR554
	FI=FI*EDT**2+((PRE*EDT+4.*VMID)*EDT+STPW)*DST	BOTR555
	PRE=STPW	BOTR556
	GO TO 70	BOTR557
	159 IF((IPRES.LT.0).OR.(T.LT.(R2-6.1*DT))) GO TO 150	BOTR558
C	CALCULATION OF PRECURSOR NEAR SINGULARITY	BOTR559
C	200 DT=DT/ZZDT	BOTR560
	M=(R2-T)/DT/4.	BOTR561
	M=4*M+5	BOTR562
	DT=(R2-T)/FLOAT(M)/2.	BOTR563
	DTACT=2.*DT*TACT/3.	BOTR564
	DST=DT/3.	BOTR565
	EDT=EXP(-DT/THETAR)	BOTR566
	N=9	BOTR567
	201 IF((IPRES.LT.0).OR.(T.LT.(R2-3.1*DT))) GO TO 152	BOTR568
	TR1=T/R2	BOTR569
	V=SQRT(1.-TR1**2)	BOTR570
	DSV=V/12.*R2	BOTR571
	TR2=SQRT(1.-(0.75*V)**2)	BOTR572
	TR3=SQRT(1.-(0.5*V)**2)	BOTR573
	TR4=SQRT(1.-(0.25*V)**2)	BOTR574
	T1=T	BOTR575
	T2=R2*TR2	BOTR576
	T3=R2*TR3	BOTR577
	T4=R2*TR4	BOTR578
	EDT1=EXP(-(T3-T1)/THETAR)	BOTR579
	EDT2=EXP(-(T3-T2)/THETAR)	BOTR580
	EDT3=EXP(-(R2-T3)/THETAR)	BOTR581
	EDT4=EXP(-(R2-T4)/THETAR)	BOTR582
C	202 CALL STPWA(T2,VMID,Z1,16)	BOTR583
	CALL STPWA(T3,STPW,Z1,16)	BOTR584
	FI=FI*EDT1+(PRE*EDT1*V/TR1+3.*VMID*EDT2*V/TR2+STPW*0.5*V/TR3)*DSV	BOTR585
C	PRE=STPW	BOTR586
	T=T3	BOTR587
	N=3	BOTR588
	GO TO 70	BOTR589
C	210 CALL STPWA(T4,VMID,Z1,16)	BOTR590
	FI=FI*EDT3+(PRE*EDT3*0.5*V/TR3+VMID*EDT4*V/TR4)*DSV	BOTR591
	PRE=0.	BOTR592
	T=R2	BOTR593
	STPW=CR*(1.E+30)*SIGN(1.,ZE)	BOTR594
	WRITE(6,540)	BOTR595
		BOTR596
		BOTR597
		BOTR598
		BOTR599
		BOTR600
		BOTR601

NOLTR 71-110

	N=4	BOTR602
	GO TO 70	BOTR603
C		BOTR604
C		BOTR605
C	CALCULATION OF MAIN BOTTOM REFLECTION NEAR SINGULARITY	BOTR606
C		BOTR607
C	300 IF (DT.GT.(DT1/ZZDT/2.)) GO TO 301	BOTR608
	DT=DT1/ZZDT/2.	BOTR609
	DFACT=2.*DT*FACT/3.	BOTR610
	DST=DT/3.	BOTR611
	EDT=EXP(-DT/THEAR)	BCTR612
	NP=16	BOTR613
C	301 T6=R2+4.*DT1	BOTR614
	DTR=DT/R2	BOTR615
	U=SQRT((1.+2.*DTR)**2-1.)	BOTR616
	DSU=U/12.*R2	BOTR617
C		BOTR618
	TR2=SQRT(1.+(0.25*U)**2)	BOTR619
	TR3=SQRT(1.+(0.50*U)**2)	BOTR620
	TR4=SQRT(1.+(0.75*U)**2)	BOTR621
	TR5=1.+2.*DTR	BOTR622
	T2=TR2*R2	BOTR623
	T3=TR3*R2	BOTR624
	T4=TR4*R2	BOTR625
	T5=TR5*R2	BOTR626
C		BOTR627
	EDT1=EXP(-(T3-R2)/THEAR)	BOTR628
	EDT2=EXP(-(T3-T2)/THEAR)	BOTR629
	EDT3=EXP(-(T5-T3)/THEAR)	BOTR630
	EDT4=EXP(-(T5-T4)/THEAR)	BOTR631
C		BOTR632
C	302 CALL STPWB(T2,VMID,Z1,16)	BCTR633
	CALL STPWB(T3,STPW,Z1,16)	BOTR634
	FI=FI*EDT1+(VMID*U*EDT2 /TR2+STPW*0.5*U/TR3)*DSU	BOTR635
	PRE=STPW	BOTR636
	T=T3	BOTR637
	N=5	BOTR638
	GO TO 71	BOTR639
C		BOTR640
C	305 CALL STPWB(T4,VMID,Z1,16)	BOTR641
	CALL STPWB(T5,STPW,Z1,16)	BOTR642
	FI=FI*EDT3+(0.5*PRE*U*EDT3/TR3+3.*VMID*U*EDT4/TR4+STPW*U/TR5)*DSU	BOTR643
	PRE=STPW	BOTR644
	T=T5	BOTR645
	N=6	BOTR646
	GO TO 71	BOTR647
C		BOTR648
C	308 N=13	BOTR649
	DFACT=2.*FACT*DT/3.	BOTR650
	310 IF ((IPRES.LT.0).OR.(T.LT.T6)) GO TO 410	BOTR651
	320 DT=DT1	BOTR652
	DFACT=2.*DT*FACT/3.	BOTR653
	DST=DT/3.	BOTR654
	EDT=EXP(-DT/THEAR)	BOTR655
C		BOTR656
C		BOTR657
C	CALCULATION OF MAIN BOTTOM REFLECTED WAVE	BOTR658
C		BOTR659
C	400 N=7	BOTR660
	410 T=T+DT	BOTR661
	CALL STPWB(T,VMID,Z1,NP)	BOTR662
C	EVERY TENTH STEP RECALCULATE STPW WITH HALF THE INTEGRATION	BOTR663
C	POINTS NP. IF ERROR IS LESS THAN .005 REDUCE NP BY HALF.	BOTR664
	IF (MOD(IP,10).EQ.0) GO TO 415	BOTR665

NOLTR 71-110

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IF(NP.EQ.4) GO TO 415
NP2=NP/2
CALL STPWB(T,VMID2,Z1,NP2)
ERROR=ABS((VMIL-VMID2)/VMID)
IF(ERROR.LT.0.005) NP=NP2
415 T=T+DT
CALL STPWB(T,STPW,Z1,NP)
FI=FI*EDT**2+((PRE*EDT+4.*VMID)*EDT+STPW)*DST
PRE=STPW
GO TO 71
C
C CALCULATION OF BOTTOM REFLECTION FOR TIMES LESS THAN OR EQUAL TO
C THE ARRIVAL TIME OF THE PEAK IF THE ANGLE OF INCIDENCE IS LESS
C THAN OR EQUAL TO THE CRITICAL ANGLE. THE DIRECT WAVE ARRIVES
C BEFORE OR TOGETHER WITH THE BOTTOM REFLECTION.
700 T=1.0
N=8
707 STPW=0.
GO TO 71
706 N=14
708 T=T+DT*2.0
IF(T+2.0*DT.LE.R2) GO TO 71
710 T=R2
STPW=CR/R2
PRE=0.
WRITE(6,558)
NP=8
N=7
GO TO 71
C
C CALCULATION OF DIRECT WAVE -PD-, BOTTOM REFLECTION -PBOT-,
C SURFACE REFLECTION -PS- AND TOTAL PRESSURE -P-.
70 IF(T.LT.1.0) GO TO 72
71 PD=PACT*EXP((1.0-T)/THETA)
IF(T.LT.RS) GO TO 72
PS=-PACT/R1*EXP((RS-T)/THETSR)
72 PBOT1=PBOT
FTHETA=FI/THETA
PBOT=PACT/PACTC*(STPW-FTHETA)
73 TIME=TACT*(T-1.0)
P=PD + PS + PBOT
C
C TEST TO INSURE THAT THE ABSOLUTE PRESSURE (P+HYDROSTATIC) .GE. 0.
P=AMAX1(P,PH)
IF(T.GT.TSTOP) N=15
C
C CALCULATION OF IMPULSE= FIMP AND ENERGY FLUX= EFLUX
C CALCULATION OF POSITIVE IMPULSE= POSIMP
6004 XP=AMAX1(P,0.)
GO TO (6030,6007,6020,6030,6040,6050,6007,6030,6007,
1 6030,6010,6007,6007,6007,6060),N
6007 IF(IPRES.LT.0) GO TO 6070
PMID=P
XPMID=XP
GO TO 6090
6010 XXDT=TACT/3.*(T-TPRE)
FIMP=3.*(P+PPRE)*XXDT /2.+FIMP
POSIMP=3.*(XP+XPPRE)*XXDT/2.+POSIMP

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BOTR666
 BOTR667
 BOTR668
 BOTR669
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 BOTR674
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 BOTR724
 BOTR725
 BOTR726
 BOTR727
 BOTR728
 BOTR729

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EFLUX=EFLUX*(ABS(P)*P+ABS(PPRF)*PPRE)*XXDT/2./RHOWAT/CWATER*3./
1 2.3066
C THE CONVERSION FACTOR 2.3066 IS NECESSARY FOR EFLUX TO HAVE
C UNITS IN=PSI
PPRE=P
XPPRE=XP
IPRES=1
GO TO 6092
6020 DTACT=2.0*DSV*TACT
PPRE=PPRE*V/TR1
XPPRE=XPPRE*V/TR1
PMID=P*V/2./TR3
XPMID=XP*V/2./TR3
GO TO 6090
6030 PEND=0.
XPEND=0.
GO TO 6072
6040 DTACT=2.0*DSU*TACT
PPRE=0.
XPPRE=0.
PMID=P*U/2./TR3
XPMID=XP*U/2./TR3
GO TO 6090
6050 PEND=P*U/TR5
XPEND=XP*U/TR5
GO TO 6072
6060 IF(IPRES.GT.0) GO TO 6010
6070 PEND=P
XPEND=XP
6072 FIMP=FIMP+(PPRE+4.*PMID*PEND)*DTACT
POSIMP=POSIMP+(XPPRE+4.*XPMID*XPEND)*DTACT
EFLUX=EFLUX*(ABS(PPRE)*PPRE+4.*ABS(PMID)*PMID+ABS(PEND)*PEND)*
1 DTACT/RHOWAT/CWATER/2.3066
PPRE=P
XPPRE=XP
6090 IPRES=-1*IPRES
6092 IF(IPRES.GT.0) TPRE=T
C WHEN D2R2.LT.1.0 THE BOTTOM REFLECTION IS NOT CALCULATED AT
C THE DIRECT WAVE ARRIVAL TIME T=1.0. IN ORDER TO PLOT THE
C INSTANTANEOUS RISE OF THE DIRECT SHOCK AT T=1.0, THE BOTTOM
C REFLECTION IS OBTAINED BY LINEAR INTERPOLATION. PLOT POINTS ARE
C THEN CALCULATED FOR THE TOP AND BOTTOM OF THE SHOCK FRONT.
IF((IP.EQ.1).OR.(IP.GT.1000)) GO TO 7002
IF((TIME.GT.0.).AND.(XX(IP-1).LT.0.)) GO TO 6095
GO TO 7002
6095 XX(IP)=0.
IF(T.NE.R2) PBOTD=PBOT1+XX(IP-1)*(PBOT-PROT1)/(XX(IP-1)-TIME*1.E6)
IF(T.EG.R2) PBOTD=PBOT1
YY(IP)=AMAX1(PBOTD,PH)
XX(IP+1)=0.
YY(IP+1)=AMAX1(PBOTD+PACT,PH)
WRITE(6,559) YY(IP+1)
IP=IP+2
C PRINT ROUTINE
C FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS ROTR907-1042
7002 IF (Z5.GT.0.0) GO TO 7003
WRITE (6,551) T,STPW,FTHETA,PD,TIME,P,FIMP
WRITE(6,598)PROT,PS,VMID,PRE,EFLUX
GO TO 7004
7003 WRITE(6,551) T,PBOT,EFLUX,PD,TIME,P,FIMP
GO TO 7005
C
C REDUCED IMPULSE
7004 RFIMP=FIMP/W13R/RACTU

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ROTR730
 ROTR731
 ROTR732
 ROTR733
 ROTR734
 ROTR735
 ROTR736
 ROTR737
 ROTR738
 ROTR739
 ROTR740
 ROTR741
 ROTR742
 ROTR743
 ROTR744
 ROTR745
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 ROTR792
 ROTR793

<pre> C REDUCED POSITIVE IMPULSE RPOSIM=POSIMP/W13R/RACTU C REDUCED ENERGY FLUX REFLUX=EFLUX/W13R/RACTU WRITE(6,556) RESID,RFIMP,REFLUX,POSIMP,RPOSIM 7005 IF(IP.GT.1000) GO TO 999 XX(IP)=TIME*1.0E6 YY(IP)=P IP=IP+1 C BOTTOM REFLECTION TIME AND PRESSURE STORED IN QX AND QY FOR PTV CALCULATION. C IF(T.LT.D2R2) GO TO 7001 IPTV=IPTV+1 QX(IPTV)=TACT*(T-D2R2) QY(IPTV)=AMAX1(PBOT,PH) C 7001 GO TO (110,159,210,300,305,308,410,706,201,159,114, 1 310,708,999),N C 999 IPMAX=IP-1 C CALCOMP PLOT TAPE GENERATED IF Z3 = 0. C IF(Z3.NE.0.) GO TO 997 CALL PLOT1(XX,YY,IPMAX,X1,X2,ADATE,THE,WCH,CBOT,POISR,Z1) C 997 ICASE=ICASE+1 CALCULATION OF INPUTS FOR SUBROUTINE PTV. IF(RADIUS.LE.0.) GO TO 4 1997 TIMER2=TACT*(R2-D2R2) XT3=TIMER2-2.*DT*TACT XT4=0.8*TIMER2 XT5=QX(IPTV) IPTV=IPTV+1 QX(IPTV)=1.0E20 QY(IPTV)=0. COSA=COSTH*COS(A)+SINTH*SIN(A) C CALCULATION OF PEAK TRANSLATIONAL VELOCITY (PTV). CALL PTV(TIMER2,XT3,XT4,XT5,RADIUS,30.,APRINT,COSA,RHOWAT, 1 CWATER,TIME1,PTV1,PTV2) IF(APRINT.GT.0.) WRITE(6,570) TIME1,PTV1,PTV2 GO TO 4 C ***** C PLANE WAVE APPROXIMATION USING EQUATIONS OF ARONS AND YENNIE C C 998 IF(CSHEAR.GT.0.) GO TO 1005 WRITE(6,590) GO TO 1007 1005 WRITE(6,596) FORMATS ARE LISTED AT THE END OF THE PROGRAM, CARDS ROTR907-1042 C SELECTION OF TIME STEP C 1007 WRITE(6,591) IF(D2R2.GE.R2) GO TO 1020 C CALCULATION OF TIME STEP FOR SUPERCRITICAL REFLECTION C 1010 M=(R2-D2R2)*STEPS/THETA/2.+1.0 M=2*M-1 IF(M.LE.4) GO TO 1020 1012 DT=(R2-D2R2)/FLOAT(M) </pre>	<pre> BOTR794 BOTR795 BOTR796 BOTR797 BOTR798 BOTR799 BOTR800 BOTR801 BOTR802 BOTR802A BOTR802B BOTR802C BOTR802D BOTR802E BOTR802F BOTR803 BOTR804 BOTR805 BOTR806 BOTR807 BOTR808 BOTR809 BOTR810 BOTR811 BOTR812 BOTR813 BOTR813A BOTR813B BOTR813C BOTR813D BOTR813E BOTR813F BOTR813G BOTR813H BOTR813I BOTR813J BOTR813K BOTR813L BOTR813M BOTR813N BOTR814 BOTR815 BOTR816 BOTR817 BOTR818 BOTR819 BOTR820 BOTR821 BOTR822 BOTR823 BOTR824 BOTR825 BOTR826 BOTR827 BOTR828 BOTR829 BOTR830 BOTR831 BOTR832 BOTR833 BOTR834 BOTR835 BOTR836 BOTR837 </pre>
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IF (D2R2-1.)1014,1015,1015	BOTR838
1014 T=D2R2	BOTR839
GO TO 1700	BOTR840
1015 T=1.	BOTR841
GO TO 1700	BOTR842
C	BOTR843
C CALCULATION OF TIME STEP FOR SUBCRITICAL REFLECTION	BOTR844
1020 DT=THETA/STEPS	BOTR845
T=1.	BOTR846
:700 IF (T.GE.1.0) GO TO 1731	BOTR847
1730 SW=0.	BOTR848
GO TO 1732	BOTR849
C	BOTR850
C INCIDENT (DIRECT) WAVE RESPONSE	BOTR851
1731 SW=EXP(-(T-1.0)/THETA)	BOTR852
1732 XE1=0.	BOTR853
XEI=0.	BOTR854
IF (E2.NE.0.) GO TO 1738	BOTR855
1733 IF (T.GE.R2) GO TO 1736	BOTR856
1735 PRFL=0.	BOTR857
GO TO 1745	BOTR858
C	BOTR859
C PRESSURE RESPONSE FOR SUBCRITICAL REFLECTION	BOTR860
1736 PRFL=CR*EXP(-(T-R2)/THETA/R)	BOTR861
GO TO 1745	BOTR862
1738 TBTH=(T-R2)/THETA/R	BOTR863
IF (TBTH) 1741,1742,1743	BOTR864
1741 CALL EXE1(TBTH,XE1)	BOTR865
C	BOTR866
C PRESSURE RESPONSE FOR PRECURSOR OF SUPERCRITICAL REFLECTION	BOTR867
PRFL=CR*SIN(E2)*XE1/PI	BOTR868
GO TO 1745	BOTR869
C	BOTR870
C PRESSURE RESPONSE AT SINGULARITY	BOTR871
1742 PRFL=(1.E+30)*SIGN(1.,EE)	BOTR872
GO TO 1745	BOTR873
1743 CALL EXE1(TBTH,XEI)	BOTR874
C	BOTR875
C PRESSURE RESPONSE FOR MAIN SUPERCRITICAL BOTTOM REFLECTION	BOTR876
PRFL=CR*(EXP(-TBTH)*COS(E2)-XE1*SIN(E2)/PI)	BOTR877
1745 PRFL=PRFL/PACTC/R2	BOTR878
PBOT=PACT*PRFL	BOTR879
1710 TIME=TACT*(T-1.)	BOTR880
PD=PACT*SW	BOTR881
IF (T.GE.RS) PS=-PACT/R1*EXP((RS-T)/THETA/R)	BOTR882
C TOTAL PRESSURE (NEGATIVE VALUE LIMITED TO PH)	BOTR883
P=AMAX1(PD+PBOT+PS,PH)	BOTR884
C OUTPUT ROUTINE	BOTR885
WRITE (/5,551)T,PBOT,PD,PS,TIME,P	BOTR886
C	BOTR887
IF (IP.GT.1000) GO TO 4000	BOTR888
7100 XX(IP)=TIME*1.0E6	BOTR889
YY(IP)=P	BOTR890
IP=IP+1	BOTR891
C BOTTOM REFLECTION TIME AND PRESSURE STORED IN QX AND QY FOR	BOTR891A
PTV CALCULATION.	BOTR891B
IF (T.LT.D2R2) GO TO 7102	BOTR891C
IPTV=IPTV+1	BOTR891D
QX(IPTV)=TACT*(T-D2R2)	BOTR891E
QY(IPTV)=AMAX1(PBOT,PH)	BOTR891F
7102 TDT=T	BOTR892
T=T+DT	BOTR893
IF (ABS(TDT-R2).LT.1.5*DT) T=TDT+0.2*DT	BOTR894
IF ((TDT.LT.R2).AND.(T.GT.R2)) T=R2	BOTR895

NOLTP 71-110

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IF ((TDT,LT,1.0).AND.(T.GT,1.0)) T=1.0
IF (T.LT,TSTOP) GO TO 1700
4000 IPMAX=IP-1
C CALCOMP PLOT TAPE IS GENERATED IF Z3 = 0.
IF (Z3,NE,0.) GO TO 4001
CALL PLOT1 (XX,YY,IPMAX,X1,X2,ADATE,THE,WCH,CBOT,POISR,Z1)
4001 ICASE=ICASE+1
IF (RADIUS,GT,0.) GO TO 1997
GO TO 4
C
C *****
500 FORMAT (10X,26HSLOPE OF BOTTOM IN DEGREES,54X,6HSLOPE=,E16.8 )
501 FORMAT (5X,12HREDUCED TIME,23X,11HENERGY FLUX,26X,4HTIME,11X,
18HPRESSURE,10X,7HIMPULSE/10X,1HT,16X,4HPBOT,31X,2HPD,12X,
27HSECONDS,14X,3HPSI / )
502 FORMAT (10X,39HREDUCED SLANT DISTANCE (RACTU/WCH**1/3),41X,6HREDR=
1,E15.8)
503 FORMAT (10X,60HPRINT OUT CONTROL PARAMETER (Z5,GT,0. FOR SHORTER PR
LINT OUT),20X,3HZ5=,E16.6)
504 FORMAT (10X,40HWEIGHT OF EXPLOSIVE CHARGE IN LB (OR KT),40X,5HWCH=
1 E15.8 )
505 FORMAT (10X,36HVELOCITY OF SOUND IN WATER IN FT/SEC,44X,9HCWATER= ,
1E15.8 )
506 FORMAT (10X,37HVELOCITY OF SOUND IN BOTTOM IN FT/SEC,43X,6HCBOT= ,
1E15.8 )
507 FORMAT (10X,25HDENSITY OF WATER IN GM/CC,55X,8HRHOWAT= ,E15.8 )
508 FORMAT (10X,26HDENSITY OF BOTTOM IN GM/CC,54X,8HRHOBOT= ,E15.8 )
509 FORMAT (10X,51HDURATION AFTER DIRECT ARRIVAL IN MULTIPLES OF THETA,
129X,7HDURAT= 1E15.8 )
510 FORMAT (1H1,52X,17HBOTTOM REFLECTION,10X,4HDATE,2X,1A10 )
511 FORMAT (10X,20HDEPTH OF WATER IN FT,60X,6HBIGH= 1E15.8 )
512 FORMAT (10X,20HDEPTH OF GAUGE IN FT,60X,6HDGAU= 1E15.8 )
513 FORMAT (10X,24HDEPTH OF EXPLOSION IN FT,56X,3HD= 1E15.8 )
514 FORMAT (10X,50HHORIZONTAL DISTANCE BETWEEN CHARGE AND GAUGE IN FT,3
10X,8HSMALLR= 1E15.8 )
515 FORMAT (10X,41HCOEFFICIENT OF SW PRESSURE FORMULA IN PSI,39X,
1 BHPRECOE= ,E15.8 )
516 FORMAT (10X,31HEXPONENT OF SW PRESSURE FORMULA,49X,8HPREEXP= ,1E15.
18 )
517 FORMAT (10X,50HCOEFFICIENT OF SW TIME CONSTANT FORMULA IN SECONDS,
1 30X,8HTHECOE= ,E15.8 )
518 FORMAT (10X,36HEXPONENT OF SW TIME CONSTANT FORMULA,44X,8HTHEEXP= ,
11E15.8 )
519 FORMAT (10X,31HNUMBER OF SUBDIVISIONS OF THETA 49X,7HSTEPS= ,1E15.8
1 )
520 FORMAT (1H0,47X,25HCHARACTERISTIC MAGNITUDES / )
521 FORMAT (10X,33HANGLE OF INCIDENT WAVE IN DEGREES,47X,5HTHE= ,1E15.8
1 )
522 FORMAT (10X,45HCritical ANGLE OF COMPRESSION WAVE IN DEGREES,35X,7H
1ALPHA= 1E15.8 )
523 FORMAT (10X,33HREDUCED TIME OF PRECURSOR ARRIVAL,47X,6HD2R2= ,1E15.
18 )
524 FORMAT (10X,35HREDUCED TIME OF GROUND WAVE ARRIVAL,45X,4HR2= ,1E15.
18/ )
525 FORMAT (10X,68HSLANT DISTANCE BETWEEN CHARGE AND GAUGE=CHARACTERIST
1IC LENGTH IN FT,12X,7HRACTU = ,1E15.8 )
526 FORMAT (10X,44HCHARACTERISTIC TIME=RACTU/CWATER IN SECONDS,36X,6HTAC
1T= ,1E15.8 )
527 FORMAT (10X,58HCHARACTERISTIC PRESSURE=FREE WATER SW PEAK PRESSURE
1IN PSI 22X,6HPACT= ,1E15.8)
528 FORMAT (10X,38HREDUCED TIME CONSTANT OF INCIDENT WAVE,42X,7HTHETA=
1,1E15.8 )
529 FORMAT (1H1)

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530 FORMAT(5X,12HREDUCED TIME,3X,17HSTEPWAVE RESPONSE,3X,12HCONVOLUTI ROTR959
 10N-,25X,4HTIME,11X,8HPRESSURE,10X,7HIMPULSE/ BOTR960
 2 10X,1HT,16X,4HSTPW,11X,8HFI/THETA , 12X,2HPO,12X,7HSECONDS BOTR961
 3 ,14X,3HPSI,7X,11HENERGY FLUX/6X,10HSECOND ROW. BOTR962
 4 11X,4HPROT ,13X,2HPS,15X,4HVMID,14X,3HPRE/ 7X,9HTHIRD ROW,9X, 7H BOTR963
 5RESIDUE,7X,15HREDUCED IMPULSE,3X,13HREDUCED EFLUX,6X,6HPOSIMP, BOTR964
 6 24X,14HREOUCED POSIMP//) BOTR965
 532 FORMAT(1H0) BOTR966
 533 FORMAT(10X,45HREDUCED TIME OF PEAK OF BOTTOM REFLECTED WAVE, BOTR967
 1 35X,4HR2= E15.8) BOTR968
 534 FORMAT(50X,20HARRIVAL OF PRECURSOR /) BOTR969
 535 FORMAT(46X,28HCONSTANTS OF THE CALCULATION/ 9X,6HSMALLH,11X,6HDEZE BOTR970
 1R0,13X,2H02,14X,5HCOSAL,12X,5HCOSTH,12X,5HSINTH/) BOTR971
 536 FORMAT(50X,2HDT,14X,3HEDT/ 40X,2E17.7/) BOTR972
 537 FORMAT(24X,73HINPUT CHANGED SO THAT RATIO BETWEEN INCIDENT AND CR1 BOTR973
 ITICAL ANGLE IS THOVAL/) BOTR974
 538 FORMAT(10X,49HDESIRED RATIO BETWEEN INCIDENT AND CRITICAL ANGLE,31 BOTR975
 1X,8HTHOVAL= 1E15.8) BOTR976
 539 FORMAT(10X,39HACTUAL SW TIME CONSTANT IN MILLISECONDS,40X,6H THET= BOTR977
 1,1E15.8) BOTR978
 540 FORMAT(1H0,39X,10H***** /39X,39HARRIVAL OF GROUNDWAVE PEAK(SI BOTR979
 NGULARITY) / 40X,10H***** /) BOTR980
 541 FORMAT(10X,44HCRITICAL ANGLE OF COMPRESSION WAVE IMAGINARY) BOTR981
 542 FORMAT(10X,38HCRITICAL ANGLE OF SHEARWAVE IN OEGREES,42X,7HBETHA= BOTR982
 11F7.3) BOTR983
 543 FORMAT(10X,37HCRITICAL ANGLE OF SHEARWAVE IMAGINARY) BOTR984
 545 FORMAT(10X,13HPOISSON RATIO,67X,7HPOISR= ,1E15.8) BOTR985
 546 FORMAT(10X,31HVELOCITY OF SHEARWAVE IN FT/SEC,49X,8HCSHEAR= 1E15.8 BOTR986
 1) BOTR987
 547 FORMAT(10X,34HREDUCED TIME OF SURFACE REFLECTION,46X,4HRS= 1E15.8 BOTR988
 1) BOTR989
 548 FORMAT(10X,46HREDUCED TIME CONSTANT OF BOTTOM REFLECTED WAVE ,34X BOTR990
 1 ,8HTHETAR= ,E15.8) BOTR991
 549 FORMAT(10X,51HBOTTOM REFLECTED WAVE TIME CONSTANT IN MILLISECONDS BOTR992
 1 ,29X,7HTHETR= E15.8 /) BOTR993
 550 FORMAT(53X,11HRUN NUMBER ,115/ 57X,5HINPUT//) BOTR994
 551 FORMAT(1X,7E17.7) BOTR995
 553 FORMAT(24X,59HGEOMETRY CHANGED SO THAT ARRIVAL TIME OF GROUNDWAVE BOTR996
 1IS Z2 = ,E15.8/) BOTR997
 554 FORMAT(8F10.5) BOTR998
 555 FORMAT(1H0,10H***** /5X,15H** WARNING ** //5X,35HSLOPE AND BOTR999
 1GEOMETRY ARE INCONSISTENT /5X,52HCOMPUTATION CONTINUES BUT RESULTS BOT1000
 2 MAY BE MEANINGLESS //) BOT1001
 556 FORMAT(18X,4E17.7,17X,E17.7/) BOT1002
 558 FORMAT(39X,26HARRIVAL OF GROUNDWAVE PEAK//) BOT1003
 559 FORMAT(1H0,10X,27HARRIVAL OF DIRECT WAVE P = ,E17.7 //) BOT1004
 560 FORMAT(50X,21HFAST NON-RIGID BOTTOM) BOT1005
 561 FORMAT(50X,21HLOW NON-RIGID BOTTOM) BOT1006
 562 FORMAT(44X,33HRIGID BOTTOM WITH FAST SHEAR WAVE) BOT1007
 563 FORMAT(44X,33HRIGID BOTTOM WITH SLOW SHEAR WAVE) BOT1008
 C BOT1009
 C BOT1010
 565 FORMAT(48X,24HPLANE WAVE APPROXIMATION) BOT1011
 566 FORMAT(45X,25HCOMPLEX ARITHMETIC MTHOD) BOT1012
 567 FORMAT(52X,16HROSENBAUM METHOD) BOT1013
 568 FORMAT(10X,21HCYLINDER RADIUS IN FT ,59X,7HRADIUS= ,E15.8) BOT1013A
 569 FORMAT(10X,43HPRINT CONTROL PARAMETER (FULL PRINT OUT IN , BOT1013B
 1 31HSUBROUTINE PTV IF APRINT.LE.0.) ,6X,8HAPRINT= E15.8) BOT1013C
 570 FORMAT(1H0,18X,5HTIME1,4X,14HPTV(SUBMERGED),2X,12HPTV(SURFACE)/ BOT1013D
 1 19X,5H(SEC),7X,8H(FT/SEC)/11X,3E15.6) BOT1013E
 573 FORMAT(10X,35HVELOCITY OF STONLEY WAVE IN FT/SEC 45X,7HCTSON= BOT1014
 1E15.8) BOT1015
 574 FORMAT(5X,10H***** ,5X,47HCALCULATION GEOMETRY CHANGED FOR SL BOT1016
 1OPING BOTTOM) BOT1017

NOLTR 71-110

579	FORMAT(10X,55HREDUCED ARRIVAL TIME OF CRITICALLY REFRACTED SHEAR WAVE,25X,8HSHD2R2= ,E15.8)	BOT1018
		BOT1019
584	FORMAT(10X,52HSCALING PARAMETER FOR Y-AXIS (PSI PER INCH OF GRAPH) 1 28X,4HX1= ,E15.8)	BOT1020
		BOT1021
585	FORMAT(10X,61HSCALING PARAMETER FOR X-AXIS (MICROSECONDS PER INCH OF GRAPH) 19X,4HX2= ,E15.8)	BOT1022
		BOT1023
586	FORMAT(10X,29HPARAMETER THAT SELECTS THEORY,51X,4HZ1= ,E15.8)	BOT1024
587	FORMAT(10X,43HARRIVAL TIME OF GROUND WAVE IN MICROSECONDS 37X,4HZ2 1= ,E15.8)	BOT1025
		BOT1026
588	FORMAT(10X,55HPLOT CONTROL PARAMETER (Z3 = 0, MEANS PLOTS ARE WANTED) ,25X,4HZ3= ,E15.8)	BOT1027
		BOT1028
590	FORMAT(43X,39HARONS-YENNIE APPROACH NON-RIGID BOTTOMS /)	BOT1029
591	FORMAT(5X,12HREDUCED TIME,3X,17HBOTTOM REFLECTION,4X,9HSHOCKWAVE, 1 4X,18HSURFACE REFLECTION,5X,4HTIME,8X,14HTOTAL PRESSURE/10X, 2 1HT,15X,4HPBOT,14X,2HPD,15X,2HPS,12X,7HSECONDS,9X, 3 7HP (PSI) /)	BOT1030
		BOT1031
		BOT1032
		BOT1033
592	FORMAT(10X,22HREFLECTION COEFFICIENT,58X,4HCR= ,E15.8)	BOT1034
593	FORMAT(10X,30HANGLE OF PHASESHIFT IN DEGREES ,50X,4HEE= ,E15.8)	BOT1035
594	FORMAT(10X,39HANGLE OF SHEARWAVE IN BOTTOM IN DEGREES,41X,6HANGA= 1 ,E15.8)	BOT1036
		BOT1037
596	FORMAT(37X,47HARONS-YENNIE APPROACH EXTENDED TO RIGID BOTTOMS /)	BOT1038
597	FORMAT(10X,43HANGLE OF PRESSURE WAVE IN BOTTOM IN DEGREES,37X,8HTH 1EONE= ,E15.8)	BOT1039
		BOT1040
598	FORMAT(18X,4E17.7,17X,E17.7)	BOT1041
599	FORMAT(10X,42HINPUT INCONSISTENT. COMPUTATION SUPPRESSED/)	BOT1042
		BOT1043
C	END	BOT1044

C*****SUBROUTINE STONL****	STON001
C	STON002
C	STON003
C	STON004
C	STON005
C	STON006
C	STON007
C	STON008
C	STON009
C	STON010
C	STON011
C	STON012
C	STON013
C	STON014
C	STON015
C	STON016
C	STON017
C	STON018
C	STON019
C	STON020
C	STON021
C	STON022
C	STON023
C	STON024
C	STON025
C	STON026
C	STON027
C	STON028
C	STON029
C	STON030
C	STON031
C	STON032
C	STON033
C	STON034

```

CALCULATION OF PROPAGATION VELOCITY OF STONLEY WAVE.

SUBROUTINE STONL
COMMON B,COSAL,COSTH,R2,SINBE,SINTH,CWATER,CBOT,CSHEAR,CSTON,RESID
COMMON C2,CBOT2,CSHR2,CBSH,C2SHR2,C4CB,SINTH2

IF (CSHEAR.LE.0.) GO TO 416
4 C12=CWATER**2
  C32=CBOT**2
  C42=CSHEAR**2

ITERATION PROCESS
IF (CWATER.GT.CSHEAR) GO TO 2
1 Y2=CWATER**2
  GO TO 3
2 Y2=CSHEAR**2
3 CK=Y2/1000.
  FY=SQRT(C12-Y2)*(CBOT*(Y2-2.*C42)**2-4.*CSHEAR*C42*SQRT((C32-Y2)*(
  1C42-Y2)))+B*CWATER*Y2**2*SQRT(C32-Y2)
  Y2=Y2-CK
400 DO 410 IR=1,999

FS STORED
FS=FY

STONLEY WAVE VELOCITY

FY=SQRT(C12-Y2)*(CBOT*(Y2-2.*C42)**2-4.*CSHEAR*C42*SQRT((C32-Y2)*(
1C42-Y2)))+B*CWATER*Y2**2*SQRT(C32-Y2)
    
```

NOLTR 71-110

C		STON035
	IF (FY) 412, 415, 408	STON036
408	Y2=Y2-CK	STON037
410	CONTINUE	STON038
	WRITE (6, 401) CWATER, CBOT, CSHEAR, B, Y2, FS, FY	STON039
	STOP	STON040
C		STON041
C	FY IS NEGATIVE	STON042
C		STON043
C	FALSE POSITION OR SECANT METHOD ITERATION	STON044
412	YS=Y2+CK	STON045
	DO 450 I=1, 50	STON046
	YSS=Y2	STON047
	IF (ABS ((YS-Y2)/YS) .LT. 1.0E-7) GO TO 415	STON048
440	Y2=YS+FS*(Y2-YS)/(FS-FY)	STON049
	FS=FY	STON050
	YS=YSS	STON051
C		STON052
	FY=SQRT(C12-Y2)*(CBOT*(Y2-2.*C42)**2-4.*CSHEAR*C42*SQRT((C32-Y2)*(STON053
	1C42-Y2)))+B*CWATER*Y2**2*SQRT(C32-Y2)	STON054
C		STON055
450	CONTINUE	STON056
	WRITE (6, 402) CWATER, CBOT, CSHEAR, B, Y2, YS, FS, FY	STON057
	STOP	STON058
C		STON059
C	RESULT	STON060
415	CSTON=SQRT(Y2)	STON061
	RETURN	STON062
416	CSTON=-0.	STON063
	RETURN	STON064
C		STON065
401	FORMAT(20X, 42HFIRST ITERATION FOR CSTON DID NOT CONVERGE//	STON066
	1 30H CWATER, CBOT, CSHEAR, B, Y2, FS, FY // 1P7E16.6)	STON067
402	FORMAT(20X, 43HSECOND ITERATION FOR CSTON DID NOT CONVERGE//	STON068
	1 33H CWATER, CBOT, CSHEAR, B, Y2, YS, FS, FY // 1P8E14.6)	STON069
C		STON070
	END	STON071
C	****SUBROUTINE STPWA****	STPA001
C		STPA002
C		STPA003
C	PRECURSOR CALCULATION USING CAGNIARD METHOD.	STPA004
C		STPA005
C		STPA006
	SUBROUTINE STPWA(T, STPW, CONTR, K)	STPA007
	DIMENSION P(30)	STPA008
	COMMON B, COSAL, COSTH, R2, SINBE, SINTH, CWATER, CBOT, CSHEAR, CSTON, RESID	STPA009
	COMMON C2, CBOT2, CSHR2, CBSH, C2SHR2, C4CB, SINTH2	STPA010
	EXTERNAL ONE, SEVEN	STPA011
	DATA AA, BB, SQ2/-1.57079633, 1.57079633, 1.41421356/	STPA012
	TR=T/R2	STPA013
	V=SQRT(1.-TR**2)	STPA014
5	IF (CONTR.EQ.3.) GO TO 100	STPA015
C		STPA016
C		STPA017
C	CALCULATION OF THE PRECURSOR USING CAGNIARD-ROSENBAUM INTEGRALS.	STPA018
C		STPA019
	P(9)=0.	STPA020
	XM=COSTH*TR+SINTH*V	STPA021
	P(1)=COSAL-XM	STPA022
	P(2)=4.*V*SINTH/P(1)	STPA023
	P(5)=COSAL+XM	STPA024

```

C      FACTOR=SQ2*B/(R2*BB)
C      STPW=FACTOR*FGI(AA,BB,K,ONE,P)
C      RETURN
C
C      CALCULATION OF THE PRECURSOR USING COMPLEX ARITHMETIC METHOD
C
100 P(1)=0.
    P(2)=TR
    P(3)=CWATER/CBOT
    P(4)=SINTH*TR-COSTH*V
    FACTOR=R2*BB
C
    ANS2=SEVEN(P(4),P)
C
    STPW=(ANS2+FGI(P(3),P(4),K,SEVEN,P))/FACTOR
C
    RETURN
    END

```

STPA025
STPA026
STPA027
STPA028
STPA029
STPA030
STPA031
STPA032
STPA033
STPA034
STPA035
STPA036
STPA037
STPA038
STPA039
STPA040
STPA041
STPA042
STPA043
STPA044
STPA045
STPA046

```

C*****SUBROUTINE STPWB****
C
C      CALCULATION OF " " IN BOTTOM REFLECTION
C
C      SUBROUTINE STPWB(T,STPW,CONTR,K)
C      DIMENSION P(30)
C      COMMON B,COSAL,COSTH,R2,SINBE,SINTH,CWATER,CBOT,CSHEAR,CSTON,RESID
C      COMMON C2,CBOT2,CSHR2,CBSH,C2SHR2,C4CB,SINTH2
C      EXTERNAL TWO,ONE1,ONE,SEVEN
C      DATA BB,SQ2/1.57079633,1.41421356/
C
C      TR=T/R2
C      5 IF(CONTR.EQ.3.) GO TO 100
C
C      CALCULATION OF THE MAIN REFLECTION USING CAGNIARD-ROSENBAUM
C      INTEGRALS.
C
C      P(9)=.
C
C      MAGNITUDES K AND L
C      XK=COSTH*TR
C      XL=SINTH**2*(TR**2-1.)
C
C      P(7)=XK
C      P(8)=XL
C
C      MAGNITUDES D,E, AND F
C      P(11)=TR**2*(1.-2.*SINTH**2)+SINTH**2
C      P(12)=4.*SINTH**2*COSTH**2*TR**2*(TR**2-1.)
C      P(13)=TR**2-SINTH**2
C      FACTOR=B/BB/R2
C      IF(CSHEAR.GT.0.) GO TO 12
C
C      NON-RIGID BOTTOMS
C
10  TERM1=(1.-R)/(1.+B)/R2
    RESID=0.

```

STPB001
STPB002
STPB003
STPB004
STPB005
STPB006
STPB007
STPB008
STPB009
STPB010
STPB011
STPB012
STPB013
STPB014
STPB015
STPB016
STPB017
STPB018
STPB019
STPB020
STPB021
STPB022
STPB023
STPB024
STPB025
STPB026
STPB027
STPB028
STPB029
STPB030
STPB031
STPB032
STPB033
STPB034
STPB035
STPB036
STPB037
STPB038
STPB039

```

        IF(CBOT.GT.CWATER) GO TO 11
C      SLOW NON-RIGID BOTTOMS
        SIGM=SQRT((CWATER/CBOT)**2-1.)
        STPW=TERM1-SQ2*FACTOR*FGI(0.,SIGM,K,TWO,P)
        RETURN
C
C      FAST NON-RIGID BOTTOMS
C 11  STPW=TERM1+FACTOR*FGI(0.,COSAL,K,ONE,P)
        RETURN
C
C      RIGID BOTTOMS
C
C      STONELEY POLE RESIDUE
C
C 12  TERM=1./R2
        CWS2=(CWATER/CSHEAR)**2
        SK=CWATER/CSTON
        SK2=SK**2
        XG1=SQRT(ABS(SK2-1.))
        XG3=SQRT(ABS(SK2-(CWATER/CBOT)**2))
        XG4=SQRT(ABS(SK2-CWS2))
        XSA=R2**2*(TR**2-SK2+COSTH**2)
        XSF=(2.*R2**2*TR*COSTH*XG1)**2
        XNUM=XG1*((CWS2/2.-SK2)**2-SK2*XG3*XG4)-B*XG3*CWS2**2/4.
        XDEN=((CWS2/2.-SK2)**2-SK2*XG3*XG4)/XG1-XG1*(2.*CWS2-4.*SK2+2.*XG3
1*XG4+SK2*(XG4/XG3+XG3/XG4))+B*CWS2**2/4./XG3
        RESID=-SQ2 *SQRT(ABS((SQRT(ABS(XSA**2+XSF))-XSA)/(XSA**2+XSF))
1)*XNUM/XDEN/XG1
C
C      TERM1=1./R2+RESID
C
C      IF(CSHEAR.GT.CWATER) GO TO 50
C
C      SLOW SHEAR WAVE
C
C 30  SIG2=SQRT(CWS2-1.)
        STPW=TERM1+FACTOR*(FGI(0.,COSAL,K,ONE,P)-CWS2**2*SQ2/4.*
1 FGI(0.,SIG2,K,ONE,P))
        RETURN
C
C      FAST SHEAR WAVE
C
C 50  STPW=TERM1+FACTOR*FGI(0.,COSAL,K,ONE,P)
C
C 99  RETURN
C
C
C      CALCULATION OF THE MAIN REFLECTION USING COMPLEX ARITHMETIC METHOD
C
C 100 P(1)=COSTH*SQRT(TR**2-1.)
        P(2)=TR
        P(3)=0.
        P(4)=SINTH*TR
        FACTOR=R2*BB
C
C      ANS2=SEVEN(P(4),P)
        STPW=(ANS2+FGI(P(3),P(4),K,SEVEN,P))/FACTOR
C
C      RETURN
        END

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NOLTR 71-110

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C*****FUNCTION ONE****                                ONE 001
C                                                         ONE 002
C                                                         ONE 003
C      INTEGRAND OF THE CAGNIARD-ROSENBAUM INTEGRAL FOR ALL FAST BOTTOMS, ONE 004
C      EXCEPT FOR PART OF THE MAIN REFLECTION OF A BOTTOM WITH SLOW SHEAR ONE 005
C                                                         ONE 006
C                                                         ONE 007
C      FUNCTION ONE(X,P)                                ONE 008
C      DIMENSION P(30)                                  ONE 009
C      COMMON B,COSAL,COSTH,R2,SINBE,SINTH,CWATER,CBOT,CSHEAR,CSTON,RESID ONE 010
C      COMMON C2,CBOT2,CSHR2,CBSH,C2SHR2,C4CB,SINTH2    ONE 011
C                                                         ONE 012
C      P ARRAY CALCULATED IN SUBROUTINES STPWA AND STPW0 ONE 013
C      TEST FOR PRECURSOR PHASE                        ONE 014
C      IF(P(9).GT.0.) GO TO 2                          ONE 015
C                                                         ONE 016
C      PRECURSOR PHASE                                 ONE 017
C                                                         ONE 018
C      SINX=SIN(X)                                     ONE 019
C      W=(SINX*P(1)+P(5))/2.                            ONE 020
C      XC2=COSAL**2-W**2                                ONE 021
C      IF(XC2.LT.0.) XC2=0.                            ONE 022
C      FX=(1.-SINX)*SQRT(((COSAL+W)*P(1))/(1.+SINX*P(2))) ONE 023
C      GO TO 3                                          ONE 024
C                                                         ONE 025
C      MAIN REFLECTED WAVE                             ONE 026
C      2 W=X                                             ONE 027
C      XC2=COSAL**2-W**2                                ONE 028
C      IF(XC2.LT.0.) XC2=0.                            ONE 029
C      RT1=SQRT(XC2/(P(8)+(W-P(7))**2))                ONE 030
C      RT2=SQRT(XC2/(P(8)+(W+P(7))**2))                ONE 031
C      FX=RT1-RT2                                       ONE 032
C                                                         ONE 033
C      RELATIONS FOR PRECURSOR AND MAIN WAVE          ONE 034
C                                                         ONE 035
C      3 IF(CSHEAR.EQ.0.) GO TO 110                   ONE 036
C                                                         ONE 037
C      RIGID BOTTOMS                                   ONE 038
C      CSH=CSHEAR/CWATER                               ONE 039
C      CSW2=CSH**2                                     ONE 040
C      FRCS=CSW2*(W**2-1.)+1.                          ONE 041
C                                                         ONE 042
C      XA=W*(1.-2.*CSW2*(1.-W**2))**2                 ONE 043
C      XB=4.*W*CSW2/CSW*(1.-W**2)*SQRT(XC2*ABS(FRCS)) ONE 044
C                                                         ONE 045
C      IF(FRCS.GE.0.) GO TO 22                        ONE 046
C      ONE=FX*(XA-XB)/((XA-XB)**2+B*B*XC2 )           ONE 047
C      RETURN                                           ONE 048
C                                                         ONE 049
C      22 XC=B*SQRT(XC2)                                ONE 050
C      ONE=FX*XA/(XA**2+(X8+XC)**2)                   ONE 051
C      RETURN                                           ONE 052
C                                                         ONE 053
C      NON-RIGID BOTTOMS                              ONE 054
C      110 ONE=FX*W/(W**2+B*B*XC2 )                   ONE 055
C      RETURN                                           ONE 056
C                                                         ONE 057
C      END                                             ONE 058

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C*****FUNCTION ONE1****                                ONE1001
C                                                         ONE1002
C                                                         ONE1003
C    INTEGRAND OF THE SECOND CAGNIARD-ROSENBAUM INTEGRAL OCCURRING FOR ONE1004
C    THE MAIN REFLECTION OF A BOTTOM WITH A SLOW SHEAR WAVE.    ONE1005
C                                                         ONE1006
C                                                         ONE1007
C    FUNCTION ONE1(X,P)                                       ONE1008
C    DIMENSION P(30)                                         ONE1009
C    COMMON B,COSAL,COSTH,R2,SINBE,SINTH,CWATER,CBOT,CSHEAR,CSTON,RESID ONE1010
C    COMMON C2,CBOT2,CSHR2,CBSH,C2SHR2,C4CB,SINTH2          ONE1011
C                                                         ONE1012
C    1 CWS2=(CWATER/CSHEAR)**2                                ONE1013
C    SIG22=CWS2-1.                                           ONE1014
C    XAB=X*(CWS2/2,-1.-X**2)**2                              ONE1015
C    XBB=X*(1.+X**2)*SQRT((COSAL**2+X**2)*(SIG22-X**2))    ONE1016
C    XCB=B*CWS2**2*SQRT(COSAL**2+X**2)/4.                  ONE1017
C                                                         ONE1018
C    FAB=SQRT(X**2+COSAL**2)*XBB/((XAB+XCB)**2+XBB**2)     ONE1019
C    FBB=SQRT((SQRT((X**2+P(11))**2+P(12))+X**2-P(13))/((X**2+P(11))**2 ONE1020
C    1+P(12)))                                               ONE1021
C    P ARRAY CALCULATED IN SUBROUTINE STPWR                ONE1022
C    ONE1=FAB*FBB                                           ONE1023
C                                                         ONE1024
C    99 RETURN                                              ONE1025
C    END                                                    ONE1026

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C*****FUNCTION TWO****                                TWO 001
C                                                         TWO 002
C                                                         TWO 003
C    CAGNIARD-ROSENBAUM INTEGRAND FOR SLOW NON-RIGID BOTTOMS TWO 004
C                                                         TWO 005
C                                                         TWO 006
C    FUNCTION TWO(X,P)                                       TWO 007
C    DIMENSION P(30)                                         TWO 008
C    COMMON B,COSAL,COSTH,R2,SINBE,SINTH,CWATER,CBOT,CSHEAR,CSTON,RESID TWO 009
C    COMMON C2,CBOT2,CSHR2,CBSH,C2SHR2,C4CB,SINTH2          TWO 010
C                                                         TWO 011
C    SIGM2=(CWATER/CBOT)**2-1.                               TWO 012
C    FAB=X*SQRT(SIGM2-X**2)/((1.-B**2)*X**2+SIGM2*B**2)    TWO 013
C    FBB=SQRT((SQRT((X**2+P(11))**2+P(12))+X**2-P(13))/((X**2+P(11))**2 TWO 014
C    1+P(12)))                                               TWO 015
C    P ARRAY CALCULATED IN SUBROUTINE STPWR                TWO 016
C    TWO=FAB*FBB                                           TWO 017
C                                                         TWO 018
C    RETURN                                              TWO 019
C    END                                                    TWO 020

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C*****FUNCTION SEVEN****                            SEVN001
C                                                         SEVN002
C                                                         SEVN003
C    INTEGRAND OF CAGNIARD INTEGRAL USING COMPLEX ARITHMETIC. SEVN004
C                                                         SEVN005
C                                                         SEVN006
C    FUNCTION SEVEN(Z,P)                                       SEVN007
C    DIMENSION P(30)                                         SEVN008
C    COMMON B,COSAL,COSTH,R2,SINBE,SINTH,CWATER,CBOT,CSHEAR,CSTON,RESID SEVN009
C    COMMON C2,CBOT2,CSHR2,CBSH,C2SHR2,C4CB,SINTH2          SEVN010
C    COMPLEX F,RCOE,Y1,Y3,V,RT1,RT2,RT5,W,U1,U2,U3,U4       SEVN011
C    COMPLEX V2,XY1,XW                                        SEVN012

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NOLTR 71-110

C	P ARRAY CALCULATED IN SUBROUTINES STPWA AND STPWS	SEVN013
	V=CMPLX(P(1),Z)	SEVN014
	V2=V*V	SEVN015
	RT1=CSQRT(1.+V2)	SEVN016
	RT2=CSQRT(V2*CBOT2+C2)	SEVN017
	IF(CSHEAR.GT.0.) GO TO 20	SEVN018
C	NON-RIGID BOTTOMS	SEVN019
C	Y3=B/CBOT*RT2	SEVN020
	RCOE=(RT1-Y3)/(RT1+Y3)	SEVN021
	GO TO 40	SEVN022
C	RIGID BOTTOMS	SEVN023
C	20 XY1=C2SHR2*V2+C2	SEVN024
	Y1=RT1*(XY1*XY1+CBSH*V2*RT2*CSQRT(V2*CSHR2+C2))	SEVN025
	Y3=C4CB*RT2	SEVN026
	RCOF=(Y1-Y3)/(Y1+Y3)	SEVN027
	40 IF(Z.EQ.P(4)) GO TO 50	SEVN028
C	INTEGRAND CALCULATION	SEVN029
	XW=P(2)-COSTH*RT1	SEVN030
	W=V2*SINTH2*XW*XW	SEVN031
	F=V/RT1/CSQRT(W)	SEVN032
	SEVEN=REAL(F*(RCOE-RT5))	SEVN033
	RETURN	SEVN034
C	CALCULATION OF ANS2	SEVN035
C	RT5=RCOE	SEVN036
50	U1=CMPLX(P(1),P(3))	SEVN037
	U2=1.+U1*U1	SEVN038
	U3=CSQRT(U2)	SEVN039
	XB=-P(2)*COSTH	SEVN040
	U4=RT5*CLOG((RT1+XB)/(CSQRT(U2+2.*XR*U3+P(2)**2-SINTH2))+U3+XB))	SEVN041
	SEVEN=AIMAG(U4)	SEVN042
	RETURN	SEVN043
	END	SEVN044
		SEVN045
		SEVN046
		SEVN047

C****	SUBROUTINE EXE1****	EXE1001
C		EXE1002
C		EXE1003
C	CALCULATION OF EXPONENTIAL INTEGRAL E1 TIMES EXP(-Y) , NEGATIVE Y	EXE1004
C		EXE1005
	SUBROUTINE EXE1(Y,ANS)	EXE1006
	DIMENSION A(4),B(4),C(6)	EXE1007
	DATA A/ 8.5733287,18.059017,8.6347609,0.26777373 /	EXE1008
	DATA B/ 9.5733223,25.632956,21.0996531,3.9584969 /	EXE1009
	DATA C/ -0.57721566,0.99999193,-0.24991055,0.05519968,-0.00976004	EXE1010
	1 +0.00107857 /	EXE1011
	X=-Y	EXE1012
	IF(X.LT.1.0) GO TO 10	EXE1013
	ANS=(A(4)+X*(A(3)+X*(A(2)+X*(A(1)+X))))/(B(4)+X*(B(3)+X*(B(2)+	EXE1014
	1 X*(B(1)+X)))/X	EXE1015
	RETURN	EXE1016
10	ANS=EXP(X)*(C(1)+X*(C(2)+X*(C(3)+X*(C(4)+X*(C(5)+X*C(6)))))-	EXE1017
	1 ALOG(X))	EXE1018
	RETURN	EXE1019
	END	EXE1020

```

C*****SUBROUTINE EXFI****                                EXEI001
C                                                         EXEI002
C                                                         EXEI003
C    CALCULATION OF EXPONENTIAL INTEGRAL EI TIMES EXP(-Y) , POSITIVE Y EXEI004
C                                                         EXEI005
C    SUBROUTINE EXEI(Y,ANS)                                EXEI006
C    DIMENSION P(10),A(6)                                EXEI007
C    EXTERNAL EXPO                                        EXEI008
C    DATA A/ .25,.05555556,.01041667,.00166667,.00023148,.00002834 / EXEI009
C    IF(Y.GT.0.5) GO TO 10                               EXEI010
C    U=Y*(1.+Y*(A(1)+Y*(A(2)+Y*(A(3)+Y*(A(4)+Y*(A(5)+Y*(A(6)))))) EXEI011
C    ANS=(0.57721566+ALOG(Y)+U)*EXP(-Y)                 EXEI012
C    RETURN                                              EXEI013
10 P(1)=Y                                               EXEI014
C    ANS1=FGI(1.,Y,4,EXPO,P)                            EXEI015
C    ANS=ANS1+1.8951178 *EXP(-Y)                        EXEI016
C    RETURN                                              EXEI017
C    END                                                EXEI018

```

```

C*****FUNCTION EXPO****                                EXP0001
C                                                         EXP0002
C                                                         EXP0003
C    INTEGRAND OF EXPONENTIAL INTEGRAL                   EXP0004
C                                                         EXP0005
C    FUNCTION EXPO(X,P)                                  EXP0006
C    DIMENSION P(10)                                    EXP0007
C    EXPO=EXP(X-P(1))/X                                  EXP0008
C    P(1) IS THE ARGUMENT OF THE EXPONENTIAL INTEGRAL   EXP0009
C    RETURN                                              EXP0010
C    END                                                EXP0011

```

```

C*****FUNCTION FGI****                                FGI 001
C                                                         FGI 002
C                                                         FGI 003
C    THIS SUBPROGRAM INTEGRATES THE FUNCTION F BETWEEN THE LIMITS FGI 004
C    A AND B USING A FOUR-POINT GAUSSIAN QUADRATURE IN EACH OF THE FGI 005
C    K SUBINTERVALS. P IS AN ARRAY USED TO TRANSFER PARAMETERS TO THE FGI 006
C    FUNCTION F.                                         FGI 007
C                                                         FGI 008
C    FUNCTION FGI(A,B,K,F,P)                             FGI 009
C    DIMENSION V(4),W(2),SUM(4),P(1)                   FGI 010
C    DATA V/ -.861136311594053,-.339981043584856, FGI 011
C    1 .339981043584856,.861136311594053 /             FGI 012
C    DATA W/ .347854845137454,.652145154862546 /     FGI 013
C    SUM(1)=0.0                                         FGI 014
C    SUM(2)=0.0                                         FGI 015
C    SUM(3)=0.0                                         FGI 016
C    SUM(4)=0.0                                         FGI 017
C    H=(B-A)/FLOAT(K)                                  FGI 018
C    H2=H/2.                                            FGI 019
C    AA=A+H2                                            FGI 020
C    DO 20 L=1,K                                        FGI 021
C    DO 10 I=1,4                                       FGI 022
C    X=H2*V(I)+AA                                       FGI 023
10 SUM(I)=SUM(I)+F(X,P)                                FGI 024
C    20 AA=AA+H                                         FGI 025
C    FGI=H2*(W(1)*(SUM(1)+SUM(4))+W(2)*(SUM(2)+SUM(3))) FGI 026
C    RETURN                                              FGI 027
C    END                                                FGI 028

```

```

C*****SUBROUTINE PLOT1****
C
C
C PLOTTING SUBROUTINE WHICH GENERATES A PLOT TAPE FOR
C CALCOMP PLOTTER
C
SUBROUTINE PLOT1 (XX,YY,IPMAX,X1,X2,ADATE,THE,WCH,CROT,POISR,Z1)
DIMENSION XX(1000),YY(1000),BCDX(2),BCDY(2),TITLE0(2),TITLE1(2),
1TITLE2(2),TITLE4(3),TITLES(2),TITLE6(2),TITLE7(2),TITLE8(2),
2TITLE9(2)
DATA RCDX/10HTIME (MICR,10HOSEC) /, /,
1BCDY/10HPRESSURE (,10HPSI) /, /,
2TITLE1/10HBOTTOM REF,10HLECTION /, /,
3TITLE2/10HCAGNIARD-R,10HOSENBAUM /, /,
4TITLE3/10HPLANE WAVE/, /,
5TITLE4/10HPLANE WAVE,10H USING CON,10HV. INT. /, /,
6TITLES/10HCOMPLEX AR,10HITHMETIC /, /,
7TITLE6/10HANGLE OF I,10HNCIDENCE /, /,
8TITLE7/10HCHARGE WEI,10HGHT (LBS) /, /,
9TITLE8/10HCBOT (FT/S,10HEC) /, /,
DATA TITLE9/10HPOISSON RA,10HTIO /, /,
1TITLE0(1)/10HDATE /
TITLE0(2)=ADATE
YMAX=6.*X1
YMIN=-3.*X1
YLIN=(YMAX-YMIN)/X1
XLMAX=90.
XLMIN=1.
IYYMAX=0
DO 4 I=1,IPMAX
IF (YY(I)-YMIN)1,1,2
1 YY(I)=YMIN
GO TO 4
2 IF (YY(I)-YMAX)4,3,3
3 YY(I)=YMAX
IYYMAX=I
4 CONTINUE
XMIN=XX(1)
XMAX=XX(IPMAX)
CALL SCAL (X2,XMIN,XMAX,XLMIN,XLMAX,XLIN)
IF (XMIN)5,10,10
5 IF (XMAX)10,10,6
6 XS=XMIN
XN=1.
7 IF (ABS(XS) -1.E-36)9,9,71
71 IF (XS)8,9,100
8 XS=XMIN+X2*XN
XN=XN+1.
GO TO 7
9 XN=XN-1,5
YN=5.16
CALL CALCM1 (IPMAX,XX,YY,0,XMIN,XMAX,YMIN,YMAX,XLIN,YLIN,TITLE,0,BC
1DX,15,BCDY,0,FLOAT,18)
CALL SYMBL4 (XN,YN,,14,BCDY,90,,14)
GO TO 11
10 CALL CALCM1 (IPMAX,XX,YY,0,XMIN,XMAX,YMIN,YMAX,XLIN,YLIN,TITLE,0,BC
1DX,15,BCDY,14,FLOAT,18)
11 IF (IYYMAX)100,12,13
12 XS=(XMAX-XMIN)/X2-3.
GO TO 14
13 XS=(XX(IYYMAX)-XMIN)/X2+1.
14 CALL SYMBL4 (XS,9,,14,TITLE1,0,,20)
15 IF (Z1-1,)17,18,16
16 IF (Z1-3,)19,20,100

```

```

PLOT001
PLOT002
PLOT003
PLOT004
PLOT005
PLOT006
PLOT007
PLOT008
PLOT009
PLOT010
PLOT011
PLOT012
PLOT013
PLOT014
PLOT015
PLOT016
PLOT017
PLOT018
PLOT019
PLOT020
PLOT021
PLOT022
PLOT023
PLOT024
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PLOT040
PLOT041
PLOT042
PLOT043
PLOT044
PLOT045
PLOT046
PLOT047
PLOT048
PLOT049
PLOT050
PLOT051
PLOT052
PLOT053
PLOT054
PLOT055
PLOT056
PLOT057
PLOT058
PLOT059
PLOT060
PLOT061
PLOT062
PLOT063
PLOT064

```

```

17 CALL SYMBL4 (XS,8.7,.14,TITLE2,0.,20)
   GO TO 21
18 CALL SYMBL4 (XS,8.7,.14,TITLE3,0.,10)
   GO TO 21
19 CALL SYMBL4 (XS,8.7,.14,TITLE4,0.,30)
   GO TO 21
20 CALL SYMBL4 (XS,8.7,.14,TITLE5,0.,20)
21 CALL SYMBL4 (XS,8.4,.14,TITLE6,0.,20)
   XN=XS+.14*18.+2
   CALL NUMBR(XN,8.4,.14,THE,0.,2)
   CALL SYMBL4 (XS,8.1,.14,TITLE7,0.,20)
   XN=XS+.14*19.+2
   CALL NUMBR(XN,8.1,.14,WCH,0.,5)
   CALL SYMBL4 (XS,7.8,.14,TITLE8,0.,20)
   XN=XS+.14*13.+2
   CALL NUMBR(XN,7.8,.14,CBOT,0.,2)
   CALL SYMBL4 (XS,7.5,.14,TITLE9,0.,20)
   XN=XS+.14*13.+2
   CALL NUMBR(XN,7.5,.14,POISR,0.,5)
   CALL SYMBL4 (XS,7.2,.14,TITLE0,0.,20)
   CALL CALCM1(0.0.)
   RETURN
100 WRITE(6,22)
    CALL CALCM1(0.0.)
    STOP
22 FORMAT(1H1,10X,34HPLOTTING ERROR IN SUBROUTINE PLOT1 //)
    END

```

PLOT065
PLOT066
PLOT067
PLOT068
PLOT069
PLOT070
PLOT071
PLOT072
PLOT073
PLJT074
PLOT075
PLOT076
PLOT077
PLOT078
PLOT079
PLOT080
PLOT081
PLOT082
PLOT083
PLOT084
PLOT085
PLOT086
PLOT087
PLOT088
PLOT089
PLOT090
PLOT091

C****SUBROUTINE SCAL****

```

C
C
C      SUBROUTINE FOR SCALING PLOTS
C
C      SUBROUTINE SCAL (XSCALE,XMIN,XMAX,XLMIN,XLMAX,XLIN)
C      DIMENSION X(6)
C      SXMAX=XMAX
C      SXMIN=XMIN
1  I,J=1
2  Ir (XSCALE)3,3,6
C
C      DETERMINATION OF SCALE RANGE
C
C
3  M1=ALOG10 (ABS(XMIN))
   M2=ALOG10 (ABS(XMAX))
   IF (IABS(IABS(M1)-IABS(M2))-1)5,4,4
4  XM=M1
   XM=10.**XM
   GO TO 7
5  XM=M1-1
   XM=10.**XM
   GO TO 7
6  XM=XSCALE
C
C      SCALE FACTORS ALLOWED
C
7  X(1)=1.*XM
   X(2)=2.*XM
   X(3)=2.5*XM
   X(4)=5.*XM
   X(5)=7.5*XM
   X(6)=10.*XM

```

SCAL001
SCAL002
SCAL003
SCAL004
SCAL005
SCAL006
SCAL007
SCAL008
SCAL009
SCAL010
SCAL011
SCAL012
SCAL013
SCAL014
SCAL015
SCAL016
SCAL017
SCAL018
SCAL019
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SCAL025
SCAL026
SCAL027
SCAL028
SCAL029
SCAL030
SCAL031
SCAL032
SCAL033

NOLTR 71-110

	IF(XSCALE)8,8,24	SCAL034
C		SCAL035
C	AUTOMATIC SCALING	SCAL036
C		SCAL037
C		SCAL038
	8 DO 21 I=1,6	SCAL039
C	DETERMINATION OF SCALED MINIMUM	SCAL040
C		SCAL041
	XMIN=SMIN	SCAL042
	XMAX=SMAX	SCAL043
	NSCALE=XMIN/X(I)	SCAL044
	IF(NSCALE)11,9,12	SCAL045
	9 IF(XMIN)11,13,10	SCAL046
	10 XMIN=0.	SCAL047
	GO TO 13	SCAL048
	11 XN=NSCALE-1	SCAL049
	XMIN=XN*X(I)	SCAL050
	GO TO 13	SCAL051
	12 XN=NSCALE	SCAL052
	XMIN=XN*X(I)	SCAL053
C		SCAL054
C	DETERMINATION OF SCALED MAXIMUM	SCAL055
C		SCAL056
	13 NSCALE=XMAX/X(I)+1.	SCAL057
	XN=NSCALE	SCAL058
	XMAX=XN*X(I)	SCAL059
C		SCAL060
C	LENTH OF SCALE AXIS CALCULATED AT THIS POINT	SCAL061
C		SCAL062
	XLIN=(XMAX-XMIN)/X(I)	SCAL063
	IF(XLIN-XLMAX)14,18,17	SCAL064
	14 IF(XLIN-XLMIN)15,18,18	SCAL065
	15 IF(I-1)35,16,16	SCAL066
	16 XM=XM*1.E+01	SCAL067
	IJ=IJ+1	SCAL068
	IF(IJ-4)7,7,23	SCAL069
	17 IF(I-6)21,177,35	SCAL070
	177 XM=XM*1.E-01	SCAL071
	IJ=IJ+1	SCAL072
	IF(IJ-4)7,7,23	SCAL073
	18 IF(XSCALE)19,20,35	SCAL074
	19 IF(ABS(XSCALE)-X(I))22,20,22	SCAL075
	20 XSCALE=X(I)	SCAL076
	GO TO 39	SCAL077
	21 CONTINUE	SCAL078
	22 XSCALE=ABS(XSCALE)	SCAL079
	XMIN=SMIN	SCAL080
	XMAX=SMAX	SCAL081
	GO TO 1	SCAL082
	23 IF(XSCALE)38,36,35	SCAL083
C		SCAL084
C	FIXED SCALING	SCAL085
C		SCAL086
	24 DO 25 I=1,6	SCAL087
	IF(XSCALE-X(I))25,26,25	SCAL088
	25 CONTINUE	SCAL089
	GO TO 37	SCAL090
C		SCAL091
C	DETERMINATION OF SCALE MINIMUM	SCAL092
C		SCAL093
	26 NSCALE=XMIN/XSCALE	SCAL094
	IF(NSCALE)28,27,30	SCAL095
	27 IF(XMIN)28,31,29	SCAL096

NOLTR 71-110

28	XN=NSCALE-1	SCAL097
	XMIN=XN*XSCALE	SCAL098
	GO TO 31	SCAL099
29	XMIN=0.	SCAL100
	GO TO 31	SCAL101
30	XN=NSCALE	SCAL102
	XMIN=XN*XSCALE	SCAL103
C		SCAL104
C	DETERMINATION OF SCALED MAXIMUM	SCAL105
C		SCAL106
31	NSCALE=XMAX/XSCALE+1.	SCAL107
	XN=NSCALE	SCAL108
	XMAX=XN*XSCALE	SCAL109
C		SCAL110
C	LENGTH OF SCALE AXIS	SCAL111
C		SCAL112
	XLIN=(XMAX-XMIN)/XSCALE	SCAL113
	IF (XLIN-XLMAX) 32,39,33	SCAL114
32	IF (XLIN-XLMIN) 34,39,39	SCAL115
33	XSCALE=XSCALE*1.E+01	SCAL116
	IJ=IJ+1	SCAL117
	IF (IJ-4) 2,2,37	SCAL118
34	XSCALE=XSCALE*1.E-01	SCAL119
	IJ=IJ+1	SCAL120
	IF (IJ-4) 2,2,37	SCAL121
35	STOP	SCAL122
36	WRITE (6,200) XMAX,XMIN,XLIN, (X(I),I=1,6)	SCAL123
	GO TO 35	SCAL124
37	WRITE (6,201) XMAX,XMIN,XLIN, (X(I),I=1,6)	SCAL125
	GO TO 35	SCAL126
38	WRITE (6,202) XMAX,XMIN,XLIN, (X(I),I=1,6)	SCAL127
	GO TO 22	SCAL128
200	FORMAT (42X,36HAUTOMATIC SCALING CANNOT BE ACHIEVED/30X,5HXMAX=1E14	SCAL129
	1.5,2X,5HXMIN=1E14.5,2X,5HXLIN=1E14.5//51X,17HSCALE FACTORS ARE//18	SCAL130
	2X,6E14.5//)	SCAL131
201	FORMAT (45X,32HFIXED SCALING CANNOT BE ACHIEVED/30X,5HXMAX=1E14.5,2	SCAL132
	1X,5HXMIN=1E14.5,2X,5HXLIN=1E14.5//51X,17HSCALE FACTORS ARE//18X,6E	SCAL133
	214.5//)	SCAL134
202	FORMAT (9X,76HAUTOMATIC SCALING CANNOT ACHIEVE DESIRED SCALE FACTOR	SCAL135
	1,WILL TRY FIXED SCALING//30X,5HXMAX=1E14.5,2X,5HXMIN=1E14.5,2X,5HX	SCAL136
	2LIN=1E14.5//51X,17HSCALE FACTORS ARE//18X,6E14.5//)	SCAL137
39	RETURN	SCAL138
	END	SCAL139

```

C      ***** PTV PROGRAM *****
C
C      SUBROUTINE PTV(TIMER2,T3,T4,T5,RAD,PTS,OPTION,COSA,RHOW,CWAT,
1 T,V,VS)
C
C      THIS SUBPROGRAM CONTROLS THE ITERATION FOR THE PEAK
C      TRANSLATIONAL VELOCITY, PTV. IT IS THE ONLY SUBROUTINE OF THE
C      PTV PROGRAM WHICH IS CALLED FROM THE MAIN PROGRAM.
C
C      DIMENSION QX(1000),QY(1000),IS(2)
C      DIMENSION G(6)
C      DIMENSION A(50),C(50)
C      COMMON /QXY/QX,QY
C      COMMON /GIS/IS
C
C
C      IF(OPTION.GT.0.) GO TO 10
C      WRITE(6,580)
C      WRITE(6,600) TIMER2,T3,T4,T5,RAD,PTS,OPTION,COSA,RHOW,CWAT
C      WRITE(6,590)
C
C      74.21457 IS A UNITS CONVERSION FACTOR
10 VC=2.*74.21457/RHOW/RAD
C      N=PTS
C      T=T4
C      DT=(T5-T)/FLOAT(N-1)
C      IF(T.LE.0.) N=N-1
C      IF(T.LE.0.) T=DT/2.
C      IS(1)=2
C      IS(2)=1
C      G(2)=CWAT/RAD
C      G(3)=TIMER2
C      G(5)=SQRT(TIMER2-T3)
C      G(6)=SQRT(TIMER2)
C      INITIAL SEARCH FOR MAXIMUM VELOCITY
C      DO 40 I=1,N
C      G(1)=T
C      V=VC*FV(G)
C      A(I)=T
C      C(I)=V
C      VS=2.*COSA*V
C      IF(OPTION.LE.0.) WRITE(6,610) T,V,VS
C      T=T+DT
40 CONTINUE
C      ITERATION FOR PTV
C      DETERMINE THE MAXIMUM VELOCITY FROM C ARRAY
C      CALL XMAX(C,N,M,M1)
C      A2=A(M1)
C      C2=C(M1)
C      A(1)=A(M)
C      C(1)=C(M)
C      A(2)=A2
C      C(2)=C2
C      DA=DT
C      T=A(1)-1.8*DA
C      IF(T.LE.0.) T=DA/5.
C      DT=DA/2.
C      DO 45 I=3,10
C      G(1)=T
C      V=VC*FV(G)
C      A(I)=T
C      C(I)=V
C      VS=2.*COSA*V
C      IF(OPTION.LE.0.) WRITE(6,610) T,V,VS

```

```

T=T+DT
45 CONTINUE
N=10
IF (IABS(M-M1),LT.3) GO TO 55
T=A(2)-0.8*DA
IF (T.LE.0.) T=DA/5.
DT=DA/3.
DO 50 I=11,16
G(1)=T
V=VC*FV(G)
A(I)=T
C(I)=V
VS=2.*COSA*V
IF (OPTION.LE.0.) WRITE(6,610) T,V,VS
T=T+DT
50 CONTINUE
N=16
55 CONTINUE
DO 75 JJ=1,6
CALL XMAX(C,N,M,M1)
IF (JJ.LT.3) GO TO 62
IF (ABS((C(M)-C(M1))/C(M)).LT.0.001) GO TO 110
IF (JJ.EQ.6) GO TO 120
62 N=10
T1=A(M)
T2=A(M1)
V1=C(M)
V2=C(M1)
A(9)=T1
A(10)=T2
C(9)=V1
C(10)=V2
DT=ABS(T1-T2)/5.
II=1
DO 70 I=1,8
T=T1+DT*FLOAT((I-10)/2*II)
IF (T .LE. 0.0) GO TO 64
G(1)=T
V=VC*FV(G)
VS=2.*COSA*V
GO TO 66
C WHEN T IS LESS THAN ZERO SET TO ZERO.
64 T = 0.0
V = 0.0
66 IF (OPTION .LE. 0.0) WRITE (6,610) T,V,VS
A(I)=T
C(I)=V
II=-1*II
70 CONTINUE
75 CONTINUE
110 V=C(M)
T=A(M)
VS=2.*COSA*C(M)
IF (OPTION.LE.0.) WRITE(6,620) A(M),C(M),VS
RETURN
120 V=C(M)
T=A(M)
VS=2.*COSA*C(M)
VS1=2.*COSA*C(M1)
WRITE(6,630) T,V,VS,A(M1),C(M1),VS1
RETURN
C
C
580 FORMAT(1H1,10X,30HTRANSLATIONAL VELOCITY PROGRAM )

```

```

590 FORMAT(1H0,5X,45HITERATION FOR PEAK TRANSLATIONAL VELOCITY PTV //
  1 12X,9HTIME (SEC),8X,16HVELOCITY (FT/SEC) ,3X,25HVERTICAL VELOCITY (F
  2T/SEC) /29X,16HTARGET SUBMERGED,7X,17HTARGET AT SURFACE )
600 FORMAT(1H0,5X,23HINPUT TO SUBROUTINE PTV // 10X,
  1 45HTIMER2,T3,T4,T5,RAD,PTS,OPTION,COSA,RHOW,CWAT //1P5E14.5/
  2 1P5E14.5 )
610 FORMAT(1P3E22.6)
620 FORMAT(1H0,6X,20H***** ,9X,3HPTV,19X,3HPTV//
  1 1P3E22.6)
630 FORMAT(1H0,42H*** WARNING ITERATION DID NOT CONVERGE *** ,5X,
  1 35HMAXIMUM AND NEAREST VALUE ARE GIVEN //
  1 12X,9HTIME (SEC),8X,16HVELOCITY (FT/SEC) ,3X,25HVERTICAL VELOCITY (F
  2T/SEC) /29X,16HTARGET SUBMERGED,7X,17HTARGET AT SURFACE /
  3 (1P3E22.6))
C
  END

```

```

FUNCTION FV(G)
C
C THIS SUBPROGRAM SETS UP THE INTEGRATION FOR
C THE TRANSLATIONAL VELOCITY V
C
  DIMENSION G(6)
  EXTERNAL F1
  DATA N/18/
C
  NN=FLOAT(N)*G(1)*G(2)/8.
  NN=MAX0(NN,8)
  NN=MIN0(NN,N)
  X=G(1)-8./G(2)
  IF(X.GT.G(3)) GO TO 43
  Z1=G(6)
  IF(X.GT.0.) Z1=SQRT(G(3)-X)
  IF(G(1).GT.G(3)) GO TO 40
  G(4)=-1.0
  Z2=SQRT(G(3)-G(1))
C
  INTEGRATION FOR T .LE. TIMER2
  FV=-FGI(Z1,Z2,NN,F1,G)
  RETURN
  40 Z2=0.
  Z3=SQRT(G(1)-G(3))
  IF(G(3).EQ.0.) GO TO 45
  G(4)=-1.0
  N1=Z1/(Z1+Z3)*FLOAT(NN)+2.0
  NNN=Z3/(Z1+Z3)*FLOAT(NN)+2.0
C
  INTEGRATION FOR INTERVAL WHICH INCLUDES TIMER2
  V1=-FGI(Z1,Z2,N1,F1,G)
  G(4)=1.0
  V2=FGI(Z2,Z3,NNN,F1,G)
  FV=V1+V2
  RETURN
  43 Z2=SQRT(X-G(3))
  Z3=SQRT(G(1)-G(3))
  45 G(4)=1.0
C
  INTEGRATION FOR T LARGER THAN TIMER2 BUT THE
  C INTERVAL DOES NOT INCLUDE TIMER2.
  FV=FGI(Z2,Z3,NN,F1,G)
  RETURN
  END

```

```

FUNCTION F1(Z,G)
C
C THIS SUBPROGRAM CALCULATES THE PRODUCT INCIDENT PRESSURE *
C REDUCED STEP WAVE ACCELERATION BY CALLING THE INTERPOLATION
C PROGRAMS VTAB AND PTAB.
C
DIMENSION QX(1000),QY(1000),IS(2)
DIMENSION G(6),QQX(120),QQY(120)
COMMON /QXY/QX,QY
COMMON /QIS/IS
C
C REDUCED STEP WAVE ACCELERATION OF A CYLINDER
C
DATA (QQX(I),I=1,106) / 0.,.0125,.025,.0375,.050,.075,.100,
1 .125,.150,.175,.200,.225,.250,.275,.300,.325,.350,.375,
2 .4000,.425,.450,.475,.500,.525,.550,.575,.600,.625,.650,
3 .675,.700,.725,.750,.775,.800,.825,.850,.875,.900,.925,.950,
4 .975,1.00,1.05,1.10,1.15,1.20,1.25,1.30,1.35,1.40,1.45,
5 1.50,1.55,1.60,1.65,1.70,1.75,1.80,1.85,1.90,1.95,2.00,
6 2.05,2.10,2.15,2.20,2.25,2.30,2.35,2.40,2.45,2.50,2.55,
7 2.60,2.65,2.70,2.75,2.80,2.85,2.90,3.00,3.10,3.2,3.3,3.4,
8 3.5,3.6,3.7,3.8,3.9,4.0,4.2,4.4,4.6,4.8,5.0,5.25,5.50,
9 5.75,6.00,6.25,6.5,7.0,7.5,8.0 /
DATA (QQY(I),I=1,60) / 0.0, .198193,.275935,.332694,.378180,
1 .448836,.502189,.544000,.577342,.604111,.625589,.642701,
2 .656143,.666457,.674079,.679365,.682612,.684070,.683955,
3 .682452,.679721,.675904,.671127,.665499,.659120,.652078,
4 .644453,.636315,.627730,.618755,.609444,.599844,.589999,
5 .579949,.569730,.559374,.548913,.538372,.527777,.517151,
6 .506515,.495887,.485284,.464215,.443417,.422977,.402968,
7 .383447,.364460,.346042,.328218,.311008,.294424,.278471,
8 .263152,.248465,.234404,.220960,.208124,.195881 /
DATA (QQY(I),I=61,106) / .184219,.173122,.162573,.152555,
1 .143051,.134041,.125509,.117435,.109801,.102590,.095782,
2 .089361,.083308,.077608,.072242,.067196,.062453,.057999,
3 .053818,.049897,.046221,.039556,.033725,.028637,.024209,
4 .020368,.017044,.014177,.011712,.009599,.007795,.006260,
5 .003863,.002172,.001009,.000230,-0.000267,-0.000619,
6 -0.000774,-0.000804,-0.000767,-0.000696,-0.000606,
7 -0.000430,-0.000297,-0.000206 /
C
IF(G(4).GT.0.) GO TO 20
X=G(3)-Z*Z
GO TO 30
20 X=G(3)+Z*Z
30 XD=(G(1)-X)*G(2)
IF(Z.GT.G(5)) GO TO 35
P=PTAB(X,QX,QY,IS(2))
GO TO 40
35 P=VTAB(X,QX,QY,IS(2))
40 F1=Z*P*VTAB(XD,QQX,QQY,IS(1))
RETURN
END

```

NOLTR 71-110

```

SUBROUTINE XMAX(B,N,M,M1)
C
C
C THIS SUBPROGRAM DETERMINES THE LOCATIONS OF THE TWO LARGEST
C ABSOLUTE VALUES OF MEMBERS OF THE B ARRAY.
C
C
C DIMENSION R(50)
C X=ABS(B(1))
C M=1
C DO 10 I=2,N
C IF(ABS(B(I)).LT.X) GO TO 10
C M=I
C X=ABS(B(M))
10 CONTINUE
C M1=1
C IF(M.EQ.1) M1=2
C X=ABS(B(M1))
C DO 20 I=2,N
C IF(ABS(B(I)).LT.X) GO TO 20
C IF(I.EQ.M) GO TO 20
C M1=I
C X=ABS(B(M1))
20 CONTINUE
C RETURN
C END

```

```

FUNCTION VTAB(X,Y,Z,K)
C
C THIS SUBPROGRAM PERFORMS A SECOND ORDER LAGRANGIAN INTERPOLATION
C
C THE INDEPENDENT VARIABLE IS STORED IN THE Y ARRAY IN INCREASING
C ORDER. THE DEPENDENT VARIABLE IS STORED IN THE Z ARRAY.
C X IS THE POINT AT WHICH THE FUNCTION IS TO BE EVALUATED.
C K IS THE NUMBER OF THE ELEMENT IN THE Y ARRAY WHICH IS FIRST
C COMPARED WITH X.
C
C
C DIMENSION Y(1000),Z(1000)
C IF(X.LE.0.) GO TO 50
C DO 10 I=K,1000
C J=I
C IF(Y(I).GT.X) GO TO 20
10 CONTINUE
20 J=MAX0(3,J-1)
C DO 30 I=1,1000
C IF(Y(J).LT.X) GO TO 40
C J=J-1
C IF(J.LT.3) GO TO 40
30 CONTINUE
40 K=J+1
C IF(Z(J).EQ.Z(K)) GO TO 60
C L=J-1
C A=(X-Y(K))/(Y(J)-Y(L))
C C=(X-Y(L))/(Y(K)-Y(J))
C IF((A.LT.-5.0).OR.(C.GT.5.0)) GO TO 60
C B=(X-Y(J))/(Y(K)-Y(L))
C VTAB=C*(B*Z(K)-A*Z(J))+A*B*Z(L)
C RETURN
50 VTAB=0.
C RETURN
60 VTAB=Z(J)+(X-Y(J))*(Z(K)-Z(J))/(Y(K)-Y(J))
C RETURN
C END

```

```
FUNCTION PTAB(X,Y,Z,K)
```

```
C
C
C
C
C
C
```

```
THIS SUBPROGRAM PERFORMS A SECOND ORDER LAGRANGIAN INTERPOLATION
WITH PROVISIONS FOR HANDLING A SINGULARITY.
FUNCTION ARGUMENTS ARE THE SAME AS IN VTAB .
```

```
DIMENSION Y(1000),Z(1000)
IF(X.LE.0.) GO TO 50
DO 10 I=K,1000
  J=I
  IF(Y(I).GT.X) GO TO 20
10 CONTINUE
20 J=MAX(3,J-1)
  DO 30 I=1,1000
  IF(Y(J).LT.X) GO TO 40
  J=J-1
  IF(J.LT.3) GO TO 40
30 CONTINUE
40 J=J+1
  JJ=J
```

```
C
C
C
C
```

```
THE FOLLOWING THREE STATEMENTS PROVIDE FOR EXTRAPOLATION
AROUND A SINGULARITY.
```

```
IF(ABS(Z(J)).GT.1.0E20)JJ=J-2
IF(ABS(Z(J-1)).GT.1.0E20)JJ=J+1
IF((JJ.EQ.J).AND.(ABS(Z(J-2)).LT.1.0E20)) JJ=J-1
```

```
C
```

```
J=JJ
K=J+1
IF(Z(J).EQ.Z(K)) GO TO 60
L=J-1
A=(X-Y(K))/(Y(J)-Y(L))
C=(X-Y(L))/(Y(K)-Y(J))
IF((A.LT.-5.0).OR.(C.GT.5.0)) GO TO 60
B=(X-Y(J))/(Y(K)-Y(L))
PTAB=C*(B*Z(K)-A*Z(J))+A*B*Z(L)
RETURN
50 PTAB=0.
RETURN
60 PTAB=Z(J)+(X-Y(J))*(Z(K)-Z(J))/(Y(K)-Y(J))
RETURN
END
```

APPENDIX B
SAMPLE PROGRAM OUTPUTS

TABLE B.1 FULL OUTPUT FOR A SPHERICAL WAVE BOTTOM REFLECTION

HOTTOM REFLECTION DATE 06/29/71
RUN NUMBER I
INPUT

DEPTH OF WATER IN FT
DEPTH OF EXPLOSION IN FT
DEPTH OF GAUGE IN FT
HORIZONTAL DISTANCE BETWEEN CHARGE AND GAUGE IN FT
WEIGHT OF EXPLOSIVE CHARGE IN LB (OR KI)
VELOCITY OF SOUND IN WATER IN FT/SEC
VELOCITY OF SOUND IN HOTTOM IN FT/SEC
VELOCITY OF SHEARWAVE IN FT/SEC
VELOCITY OF WATER IN GW/CC
DENSITY OF HOTTOM IN GW/CC
COEFFICIENT OF SW PRESSURE FORMULA IN PSI
PRINT OUT CONTROL PARAMETER (Z3=0, GT=0, FOR SHORTER PRINT OUT)
EXPONENT OF SW PRESSURE FORMULA
COEFFICIENT OF SW TIME CONSTANT FORMULA
EXPONENT OF SW TIME CONSTANT FORMULA
NUMBER OF SUBDIVISIONS OF THETA
DURATION AFTER DIRECT ARRIVAL IN MULTIPLES OF THETA
DESIRED RATIO BETWEEN INCIDENT AND CRITICAL ANGLE
SCALING PARAMETER FOR Y-AXIS (PSI PER INCH OF GRAPH)
SCALING PARAMETER FOR X-AXIS (MICROSECONDS PER INCH OF GRAPH)
SLOPE OF HOTTOM IN DEGREES
PARAMETER THAT SELECTS THEORY
ARRIVAL TIME OF GROUND WAVE IN MICROSECONDS
PLOT CONTROL PARAMETER (Z3 = 0, MEANS PLOTS ARE WANTED)
CYLINDER RADIUS IN FT
PRINT CONTROL PARAMETER (FULL PRINT OUT IN SUBROUTINE PTV IF APRINT.LE.0.)

BIGH= .20000000E+04
D= .70000000E+03
DGAU= .80000000E+03
SMALLR= .70000000E+04
WCH= .10000000E+02
CWATER= .49000000E+04
CRAT= .52000000E+04
CSHEAR= 0.
PCWAT= .10300000E+01
RHOROT= .16600000E+01
PRECOF= .43800000E+07
Z5= -0.
PREEXP= .11300000E+01
THECOE= .22740000E-02
THEEXP= -.22000000E+00
STEPS= .40000000E+01
DURAT= .50400981E+01
THOVAL= -0.
X1= .10000000E+03
X2= .10000000E+05
SLOPE= -0.
Z1= -0.
Z2= -0.
Z3= .10000000E+01
RADIUS= .22000000E+02
APRINT= 0.

CHARACTERISTIC MAGNITUDES

ANGLE OF INCIDENT WAVE IN DEGREES
VELOCITY OF STONLEY WAVE IN FT/SEC
POISSON RATIO
REDUCED TIME OF SURFACE REFLECTION
CRITICAL ANGLE OF COMPRESSION WAVE IN DEGREES
ANGLE OF PRESSURE WAVE IN HOTTOM IN DEGREES
REFLECTION COEFFICIENT
ANGLE OF SHEARWAVE IN HOTTOM IN DEGREES
ANGLE OF PHASESHIFT IN DEGREES
REDUCED TIME OF PRESSURE ARRIVAL
REDUCED TIME OF PEAK OF BOTTOM REFLECTED WAVE
SLANT DISTANCE BETWEEN CHARGE AND GAUGE=CHARACTERISTIC LENGTH IN FT.
REDUCED SLANT DISTANCE (RACTU/WCH**1/3)
CHARACTERISTIC TIME=RACTU/CWATER IN SECONDS.
CHARACTERISTIC PRESSURE=PRFF WATER SW PEAK PRESSURE IN PSI
REDUCED TIME CONSTANT OF INCIDENT WAVE
ACTUAL SW TIME CONSTANT IN MILLISECONDS
REDUCED TIME CONSTANT OF ROTTON REFLECTED WAVE
HOTTOM REFLECTED WAVE TIME CONSTANT IN MILLISECONDS

THE= .70346201E+02
CTSON= -0.
POISR= .50000000E+00
RS= .10225972E+01
ALPHA= .70442811E+02
THEONE= .88014453E+02
CR= .88638002E+00
ANGA= -0.
EE= 0.
D2R2= .10617537E+01
R2= .10917537E+01
RACTU= .70007142E+04
REDR= .32494437E+04
TACT= .14287172E+01
PACT= .47111824E+03
THETA= .20313067E-01
THETAR= .29021626E+02
THETR= .29406749E+02

SMALLH DELZRO D2 COSTH COSAL SINTH
.1300000E+04 -.1000000E+03 .3571064E+00 .3347480E+00 .3363364E+00
.9417419E+00

TABLE B.1 CONTINUED

ROSENBAUM METHOD
FAST NON-RIGID BOTTOM

DT
.2205490E-02
EDT
.09H38A7E+00

REDUCED TIME	STEPWAVE RESPONSE	CONVOLUTION=	REDUCED IMPULSE	REDUCED EFFLUX	TIME	PRESSURE	IMPULSE
SECONDR	STPW	F1/THETA	PS	VMTO	SECONDS	PSI	ENERGY FLUX
TWIRD	RESIDU	PS	REDUCED	REDUCED	PKE		REDUCED PUSIMP
					PCSTMP		
.1000000E+01	0.	0.	.4711182E+03	0.	0.	.4711182E+03	0.
	0.	0.	0.	0.	0.	0.	0.
	0.	0.	0.	0.	0.	0.	0.
.1000111E+01	0.	0.	.3791604E+03	0.	.6302043E-02	.3791604E+03	0.
	0.	0.	0.	0.	0.	0.	0.
	0.	0.	0.	0.	0.	0.	0.
.1008822E+01	0.	0.	.3051514E+03	0.	.1260409E-01	.3051518E+03	.4816675E+01
	0.	0.	0.	0.	0.	0.	.1606213E+00
	0.	0.	.2235703E+01	0.	.4816675E+01	.2235703E+01	.2235703E+01
.1012333E+01	0.	0.	.2455690E+03	0.	.1890613E-01	.2455690E+03	.4816675E+01
	0.	0.	0.	0.	0.	0.	.1606213E+00
	0.	0.	.2235703E+01	0.	.4816675E+01	.2235703E+01	.2235703E+01
.1017666E+01	0.	0.	.1976523E+03	0.	.2520817E-01	.1976523E+03	.7935523E+01
	0.	0.	0.	0.	0.	0.	.280082E+00
	0.	0.	.3683804E+01	0.	.7935523E+01	.3683804E+01	.3683804E+01
.1022055E+01	0.	0.	.1590725E+03	0.	.3151021E-01	.1590725E+03	.7935523E+01
	0.	0.	0.	0.	0.	0.	.280082E+00
	0.	0.	.3683808E+01	0.	.7935523E+01	.3683808E+01	.3683808E+01
.1026466E+01	0.	0.	.1280230E+03	0.	.3781224E-01	-.2520864E+03	.9158816E+01
	0.	0.	0.	0.	0.	0.	.2418550E+00
	0.	0.	.3801094E+03	0.	.9688369E+01	.4496943E+01	.4496943E+01
	0.	0.	.4251146E+01	0.	.1122591E+00	-.2032497E+03	.9158816E+01
.1030877E+01	0.	0.	.1030341E+03	0.	.4411430E-01	-.2032497E+03	.9158816E+01
	0.	0.	0.	0.	0.	0.	.2418550E+00
	0.	0.	.3062838E+03	0.	.9688369E+01	.4496943E+01	.4496943E+01
	0.	0.	.4251146E+01	0.	.1122591E+00	-.1638740E+03	.6577165E+01
.1035288E+01	0.	0.	.8292281E+02	0.	.5041634E-01	-.1638740E+03	.6577165E+01
	0.	0.	0.	0.	0.	0.	.1957243E+00
	0.	0.	.2467968E+03	0.	.9688369E+01	.4496943E+01	.4496943E+01
	0.	0.	.3052850E+01	0.	.9084715E-01	-.1321264E+03	.6577165E+01
.1039699E+01	0.	0.	.6673705E+02	0.	.5571838E-01	-.1321264E+03	.6577165E+01
	0.	0.	0.	0.	0.	0.	.1957243E+00
	0.	0.	.1988634E+03	0.	.9688369E+01	.4496943E+01	.4496943E+01
	0.	0.	.3052850E+01	0.	.9084715E-01	-.1065292E+03	.4898913E+01
.1044110E+01	0.	0.	.5371061E+02	0.	.6302043E-01	-.1065292E+03	.4898913E+01
	0.	0.	0.	0.	0.	0.	.1762299E+00
	0.	0.	.1602398E+03	0.	.9688369E+01	.4496943E+01	.4496943E+01
	0.	0.	.2273874E+01	0.	.8179866E-01	-.8589092E+02	.4898913E+01
.1048521E+01	0.	0.	.4322680E+02	0.	.6592247E-01	-.8589092E+02	.4898913E+01
	0.	0.	0.	0.	0.	0.	.1762299E+00
	0.	0.	.1291177E+03	0.	.9688369E+01	.4496943E+01	.4496943E+01
	0.	0.	.2273874E+01	0.	.8179866E-01	-.8589092E+02	.4898913E+01

TABLE B.1 CONTINUED

.1052932E+01	0.	0.	.3478933E+02	.7562451E-01	-.6925090E+02	.3807938E+01	
.1057343E+01	0.	-.1040402E+03	0.	0.	0.	.1679918E+00	
.1061754E+01	0.	.1767488E+01	.7797489E-01	.9688369E+01	.7562451E-01	.4496943E+01	
.1066165E+01	0.	0.	.2799878E+02	.8192655E-01	-.5583455E+02	.3807938E+01	
.1070074E+01	0.	-.8383333E+02	0.	0.	0.	.1679918E+00	
.1074087E+01	0.	.1767488E+01	.779749E-01	.9688369E+01	.7562451E-01	.4496943E+01	
.1077999E+01	0.	ARRIVAL OF GROUNDWAVE PEAK					
.1082022E+01	0.	.8344264E+00	.2253368E+02	.8822860E-01	.3452327E+03	.3918525E+01	
.1086045E+01	0.	.3902501E+03	0.	0.	0.	.1863832E+00	
.1090068E+01	0.	.1818818E+01	.8651140E-01	.1041359E+02		.4833562E+01	
.1094091E+01	0.	.9159359E-01	.1813532E+02	.9453064E-01		.3918525E+01	
.1098114E+01	0.	-.5443117E+02	.5704848E+00	.5128732E+00	.1406366E+03	.1863832E+00	
.1102137E+01	0.	.1818818E+01	.8651140E-01	.1041359E+02		.4833562E+01	
.1106160E+01	0.	.4508035E+00	.1459547E+02	.1008427E+00	.1038651E+03	.6211721E+01	
.1110183E+01	0.	.1331290E+03	.4768147E+00	.4508035E+00		.2284621E+00	
.1114206E+01	0.	.2883225E+01	.1060427E+00	.1270679E+02		.5897969E+01	
.1118229E+01	0.	.2170511E+00	.1171657E+02	.1071347E+00	.6458357E+02	.6211721E+01	
.1122252E+01	0.	-.3534097E+02	.4306137E+00	.4142390E+00		.2284621E+00	
.1126275E+01	0.	.2883225E+01	.1060427E+00	.1270679E+02		.5897969E+01	
.1130298E+01	0.	.2524272E+00	.9453715E+01	.1134368E+00	.4476635E+02	.7100237E+01	
.1134321E+01	0.	-.2847698E+02	.4005603E+00	.3888664E+00		.2341655E+00	
.1138344E+01	0.	.3294638E+01	.1086900E+00	.1359531E+02		.6310382E+01	
.1142367E+01	0.	.2767704E+00	.7408467E+01	.1197388E+00	.2817912E+02	.7100237E+01	
.1146390E+01	0.	-.2246134E+02	.3787600E+00	.3698623E+00		.2341655E+00	
.1150413E+01	0.	.3294638E+01	.1086900E+00	.1359531E+02		.6310382E+01	
.1154436E+01	0.	.2911791E+00	.6123345E+01	.1264049E+00	.1647967E+02	.7465677E+01	
.1158459E+01	0.	-.1844948E+02	.3819620E+00	.3548863E+00		.2351493E+00	
.1162482E+01	0.	.3445263E+01	.1091466E+00	.1396074E+02		.6480004E+01	
.1166505E+01	0.	.3038280E+00	.4928142E+01	.1323429E+00	.8704032E+01	.7465677E+01	
.1170528E+01	0.	-.1484948E+02	.3445263E+01	.3427056E+00		.2351493E+00	
.1174551E+01	0.	.3445263E+01	.1091466E+00	.1396074E+02		.6480004E+01	
.1178574E+01	0.	.3102863E+00	.3966215E+01	.1386449E+00	.2374752E+01	.7574220E+01	
.1182597E+01	0.	-.1260828E+02	.3374157E+00	.3325626E+00		.2352479E+00	
.1186620E+01	0.	.3515641E+01	.1091924E+00	.1406929E+02		.6530385E+01	
.1190643E+01	0.	.3137022E+00	.3142047E+01	.1449470E+00	.1686862E+01	.7574220E+01	
.1194666E+01	0.	-.9673228E+01	.3280921E+00	.3239583E+00		.2352479E+00	
.1198689E+01	0.	.3515641E+01	.1091924E+00	.1406929E+02		.6530385E+01	
.1202712E+01	0.	.3149234E+00	.2568989E+01	.1512490E+00	.4463819E+01	.7555657E+01	
.1206735E+01	0.	-.7794471E+01	.3201229E+00	.3165532E+00		.2352479E+00	
.1210758E+01	0.	.3507025E+01	.1091924E+00	.1407428E+02		.6532701E+01	

NOT REPRODUCIBLE

TABLE B.1 CONTINUED

TRANSLATIONAL VELOCITY PROGRAM

INPUT TO SUBROUTINE PTY

TIMER2,T3,T4,T5,RAD,PTS,OPTION,COSA,RHOW,CWAT
 0. -6.30204E-03 0. 6.30204E-02 2.20000E+01
 3.00000E+01 0. 3.36336E-01 1.03000E+00 4.90000E+03

ITERATION FOR PEAK TRANSLATIONAL VELOCITY PTY

TIME (SEC)	VELOCITY (FT/SEC) TARGET SUBMERGED	VELOCITY (FT/SEC) TARGET AT SURFACE
1.086559E-03	6.525452E-01	4.389494E-01
3.259677E-03	2.147045E+00	1.444259E+00
5.432795E-03	2.922026E+00	1.965567E+00
7.605914E-03	3.073522E+00	2.067475E+00
9.779032E-03	2.887512E+00	1.942351E+00
1.195215E-02	2.592360E+00	1.743810E+00
1.412527E-02	2.320794E+00	1.561135E+00
1.629838E-02	2.068588E+00	1.391456E+00
1.847150E-02	1.831783E+00	1.232191E+00
2.064462E-02	1.620060E+00	1.089770E+00
2.281774E-02	1.429563E+00	9.616284E-01
2.499086E-02	1.259587E+00	8.472898E-01
2.716398E-02	1.109541E+00	7.463583E-01
2.933709E-02	9.757967E-01	6.563919E-01
3.151021E-02	8.567997E-01	5.763459E-01
3.368333E-02	7.511928E-01	5.053070E-01
3.585645E-02	6.566918E-01	4.417387E-01
3.802957E-02	5.723684E-01	3.850148E-01
4.020269E-02	4.972796E-01	3.345044E-01
4.237580E-02	4.297510E-01	2.890818E-01
4.454892E-02	3.693076E-01	2.484232E-01
4.672204E-02	3.152316E-01	2.120477E-01
4.889516E-02	2.666219E-01	1.793493E-01
5.106828E-02	2.231784E-01	1.501260E-01
5.324139E-02	1.842856E-01	1.239639E-01
5.541451E-02	1.493711E-01	1.004779E-01
5.758763E-02	1.182308E-01	7.953042E-02
5.976075E-02	9.038491E-02	6.079947E-02
6.193387E-02	6.546200E-02	4.403451E-02
6.410699E-02	4.363885E+00	1.590121E+00
6.628011E-02	2.767551E+00	1.866165E+00
6.845323E-02	2.992797E+00	2.313173E+00
7.062635E-02	3.075552E+00	2.068840E+00
7.280047E-02	3.054987E+00	2.055006E+00
7.497359E-02	2.863759E+00	1.993640E+00
7.714671E-02	2.631359E+00	1.904578E+00
7.932083E-02	2.682011E+00	1.804116E+00
8.149395E-02	3.050738E+00	2.052148E+00
8.366707E-02	3.076577E+00	2.069530E+00
8.584019E-02	3.059309E+00	2.057914E+00
8.801331E-02	3.078365E+00	2.070733E+00
9.018643E-02	3.066342E+00	2.062645E+00
9.235955E-02	3.078826E+00	2.071043E+00
9.453267E-02	3.071737E+00	2.066274E+00
9.670579E-02	3.077909E+00	2.070426E+00
9.887891E-02	3.078206E+00	2.070625E+00
1.005103E-01	3.078566E+00	2.070847E+00
1.022315E-01	3.078445E+00	2.070787E+00
1.039527E-01	3.078712E+00	2.070966E+00

TABLE B.1 CONTINUED

7.162597E-03	3.078628E+00	2.070909E+00
7.266907E-03	3.078805E+00	2.071028E+00
7.188675E-03	3.078755E+00	2.070995E+00
7.240830E-03	3.078843E+00	2.071054E+00
*****	PTV	PTV
7.240830E-03	3.078843E+00	2.071054E+00

TABLE B.2 SHORT OUTPUT FOR A SPHERICAL WAVE BOTTOM REFLECTION

BOTTOM REFLECTION
RUN NUMBER 2
INPUT
DATE 06/29/71

DEPTH OF WATER IN FT
DEPTH OF EXPLOSION IN FT
DEPTH OF GAUGE IN FT
HORIZONTAL DISTANCE BETWEEN CHARGE AND GAUGE IN FT
***** CALCULATION GEOMETRY CHANGED FOR SLOPING BOTTOM
DEPTH OF WATER IN FT
DEPTH OF GAUGE IN FT
HORIZONTAL DISTANCE BETWEEN CHARGE AND GAUGE IN FT
WEIGHT OF EXPLOSIVE CHARGE IN LB (OR KT)
VELOCITY OF SOUND IN WATER IN FT/SEC
VELOCITY OF SHEARWAVE IN FT/SEC
DENSITY OF WATER IN GM/CC
DENSITY OF BOTTOM IN GM/CC
COEFFICIENT OF SW PRESSURE FORMULA IN PSI
PRINT OUT CONTROL PARAMETER (Z5=GT.0, FOR SHORTER PRINT OUT)
EXPONENT OF SW PRESSURE FORMULA
COEFFICIENT OF SW TIME CONSTANT FORMULA IN SECONDS
EXPONENT OF SW TIME CONSTANT FORMULA
NUMBER OF SUBDIVISIONS OF THETA
DURATION AFTER DIRECT ARRIVAL IN MULTIPLES OF THETA
DESIRED RATIO BETWEEN INCIDENT AND CRITICAL ANGLE
SCALING PARAMETER FOR Y-AXIS (PSI PER INCH OF GRAPH)
SCALING PARAMETER FOR X-AXIS (MICROSECONDS PER INCH OF GRAPH)
SLOPE OF BOTTOM IN DEGREES
PARAMETER THAT SELECTS THEORY
ARRIVAL TIME OF GROUND WAVE IN MICROSECONDS
PLOT CONTROL PARAMETER (Z3 = 0. MEANS PLOTS ARE WANTED)
CYLINDER RADIUS IN FT
PRINT CONTROL PARAMETER (FULL PRINT OUT IN SUBROUTINE PTY IF APRINT.LE.0.)

BIGH= .54064000E+01
D= .74166000E+00
DGAU= .13000000E+01
SMALLR= .39416600E+01

BIGH= .40401293F+01
DGAU= .39236405E+01
SMALLR= .23923685E+01
WCH= .12500000E-02
CHATER= .49000000E+04
CBOT= .18500000E+05
CSHEAR= .10200000E+05
RHOWAT= .99800000E+00
RHOBOT= .27203000E+01
PRECEDE= .15400000E+05
Z5= .10000000E+01
PREEXP= .11300000E+01
THECOE= .75000000E-04
THEEXP= -.22000000E+00
STEPS= .10000000E+02
DURAT= .46528584E+01
THOVAL= 0.
X1= .10000000E+03
X2= .10000000E+02
SLOPE= .45000000E+02
Z1= -0.
Z2= -0.
Z3= .10000000E+01
RADIUS=-0.
APRINT= -0.

CHARACTERISTIC MAGNITUDES

ANGLE OF INCIDENT WAVE IN DEGREES
VELOCITY OF STONLEY WAVE IN FT/SEC
POISSON RATIO
REDUCED TIME OF SURFACE REFLECTION
CRITICAL ANGLE OF COMPRESSION WAVE IN DEGREES
CRITICAL ANGLE OF SHEARWAVE IN DEGREES
REDUCED ARRIVAL TIME OF CRITICALLY REFRACTED SHEAR WAVE
ANGLE OF PRESSURE WAVE IN BOTTOM IN DEGREES
REFLECTION COEFFICIENT
ANGLE OF SHEARWAVE IN BOTTOM IN DEGREES
ANGLE OF PHASESHIFT IN DEGREES
REDUCED TIME OF PRECURSOR ARRIVAL
REDUCED TIME OF PLAK OF BOTTOM REFLECTED WAVE
SLANT DISTANCE BETWEEN CHARGE AND GAUGE=CHARACTERISTIC LENGTH IN FT,
REDUCED SLANT DISTANCE (RACTU/WCH**1/3)
CHARACTERISTIC TIME=RACTU/CHATER IN SECONDS,
CHARACTERISTIC PRESSURE=FREE WATER SW PEAK PRESSURE IN PSI
REDUCED TIME CONSTANT OF INCIDENT WAVE
ACTUAL SW TIME CONSTANT IN MILLISECONDS
REDUCED TIME CONSTANT OF BOTTOM REFLECTED WAVE
BDTOM REFLECTED WAVE TIME CONSTANT IN MILLISECONDS

THE= .35013398E+02
CTYSON= .48865514E+04
POISR= .28162126E+00
RS= .11150537E+01
ALPHA= .15358929E+02
BETHA= 28.711
SHD2R2= .10410370E+01
THEONE= -0.
CR= .10000000E+01
ANGA= -0.
EE= -.47801586E+02
D2R2= .98634537E+00
R2= .10473669E+01
RACTU= .39810083E+01
REDR= .36956407E+02
TACT= .81245067E-03
PACT= .26063414E+03
THETA= .22001857E-01
THETAR= .17875424E-01
THETRA= .2227012E-01
THETR= .18058351E-01

SMALLH .3298469E+01
DEZERO -.3181987E+01
CONSTANTS OF THE CALCULATION
COSAL .9642855E+00
COSTH .8190180E+00
SINTH .5737678E+00

TABLE B.2 CONTINUED

ROSENBAUM METHOD
 RIGID BOTTOM WITH FAST SHEAR WAVE
 DT
 .152095E-02 .9537687E+00

REDUCED TIME T	PROT	ENERGY FLUX	PD	TIME SECONDS	PRESSURE PSI	IMPULSE
.9863454E+00	0.	0.	0.	-.1109371E-04	0.	0.
.9884496E+00	.2659025E+01	0.	0.	-.9384164E-05	.2659025E+01	0.
.9905537E+00	.3420566E+01	.2019874E-08	0.	-.7674413E-05	.3420566E+01	.8010188E-05
.9926679E+00	.3393037E+01	.2019874E-08	0.	-.5965062E-05	.3393037E+01	.8010188E-05
.9947821E+00	.3035006E+01	.5402798E-08	0.	-.4255512E-05	.3035006E+01	.1942298E-04
.9968963E+00	.2553975E+01	.5402798E-08	0.	-.2545961E-05	.2553975E+01	.1842298E-04
.9989705E+00	.2049203E+01	.7398414E-08	0.	-.8364102E-06	.2049203E+01	.2814175E-04
ARRIVAL OF DIRECT WAVE P = .7624485E+03						
1.001175E+01	.1569388E+01	.7398414E-08	.2482092E+03	.8731405E-06	.2497786E+03	.214175E-04
1.003175E+01	.1138247E+01	.1521177E-04	.2555710E+03	.2582691E-05	.2567093E+03	.7278453E-03
1.005282E+01	.766901E+00	.1521177E-04	.2449975E+03	.4592242E-05	.2057645E+03	.7278453E-03
1.007387E+01	.4601681E+00	.2812633E-04	.1863005E+03	.6001793E-05	.1867607E+03	.1432481E-02
1.009491E+01	.2194434E+00	.2812633E-04	.1493088E+03	.7711343E-05	.1695262E+03	.1432481E-02
1.011594E+01	.4533097E-01	.3689249E-04	.1538648E+03	.9202094E-05	.1539121E+03	.2013036E-02
1.013700E+01	-.4153364E-01	.3689249E-04	.1798332E+03	.1113044E-04	.1797717E+03	.2013036E-02
1.015804E+01	-.3948816E-01	.4285707E-04	.1707746E+03	.1284000E-04	.1769801E+03	.2491698E-02
1.017908E+01	-.6501643E-01	.4285707E-04	.1154891E+03	.1454955E-04	.1154241E+03	.2491698E-02
1.020012E+01	.420993F-01	.4691527E-04	.144958E+03	.1625910E-04	.1050040E+03	.2886992E-02
1.022117E+01	.2506342F+00	.4691527E-04	.9538320E+02	.1796855E-04	.9563384E+02	.2886992E-02
1.024221E+01	.588821E+00	.4970565E-04	.868837E+02	.1767820E-04	.8724250E+02	.3214532E-02
1.026325E+01	.9887322E+00	.4970565E-04	.7877759E+02	.2138773E-04	.9776733E+02	.3214532E-02
1.028429E+01	.1411407E+01	.5164694E-04	.7159240E+02	.2309730E-04	.7320409E+02	.3487805E-02
1.030533E+01	.2464019E+01	.5164694E-04	.6506292E+02	.2480648E-04	.6752669E+02	.3487805E-02
1.032637E+01	.3470633E+01	.5303409E-04	.5912878E+02	.2651640E-04	.6799942E+02	.3719227E-02
1.034741E+01	.4480911E+01	.5303409E-04	.5373588E+02	.2822595E-04	.5917679E+02	.3719227E-02
1.036845E+01	.4203004E+01	.5410966E-04	.4883444E+02	.2993550E-04	.5703784E+02	.3922404E-02
1.038949E+01	.1300474E+02	.5410966E-04	.4438040E+02	.3164505E-04	.5738554E+02	.3922404E-02
1.041054E+01	.2434542E+02	.5515082F-04	.4033300E+02	.3335460E-04	.6667842E+02	.4122569E-02
1.043158E+01	.2266901E+02	.5515082F-04	.3933594E+02	.3533145E-04	.6820585E+02	.4122569E-02
1.045262E+01	.3261254E+02	.5529715E-04	.3544278E+02	.370830E-04	.7215538E+02	.4146718E-02
1.047366E+01	.4663359E+02	.5529715E-04	.3315349E+02	.388515E-04	.7578708E+02	.4146718E-02
1.049470E+01	.410145E+02	.5547801E-04	.3478044E+02	.3400200E-04	.8016949E+02	.4173568E-02
1.051574E+01	.4659329E+02	.5547801E-04	.3838638E+02	.3423885E-04	.8697967E+02	.4173568E-02
1.053678E+01	.5152452F+02	.5570456E-04	.3800848E+02	.3441570E-04	.8663300E+02	.4203617E-02
1.055782E+01	.4437441E+02	.5592666E-04	.3763430E+02	.3459255E-04	.9586691E+02	.4203617E-02
1.057886E+01	.4437441E+02	.5592666E-04	.3726340E+02	.3476940E-04	.1016399E+02	.4237497E-02
1.059990E+01	.7075732E+02	.5599266E-04	.3489695E+02	.3494625E-04	.1076502E+03	.4237497E-02
1.062094E+01	.7075732E+02	.5599266E-04	.3553371E+02	.3512310E-04	.1149337E+03	.4275648E-02
1.064198E+01	.4617748E+02	.5635744E-04	.3617444E+02	.3529995E-04	.1223317E+03	.4275648E-02
1.066302E+01	.4362152F+02	.568744E-04	.3581792E+02	.3547480E-04	.1294394E+03	.4318904E-02
1.068406E+01	.1011125F+03	.568744E-04	.3546531E+02	.3565365E-04	.136578E+03	.4318904E-02
1.070510E+01	.1077041F+03	.5741149E-04	.3511616E+02	.3583650E-04	.1428202E+03	.4367159E-02
1.072614E+01	.1127944F+03	.5741149E-04	.3477045F+02	.3600735E-04	.1475601E+03	.4367159E-02
1.074718E+01	.1142913E+03	.580894E-04	.3442815E+02	.3618420E-04	.1487195E+03	.4419140E-02
1.076822E+01	.1104209F+03	.580894E-04	.3408921E+02	.3636105E-04	.1449190E+03	.4419140E-02

NOT REPRODUCIBLE

TABLE B.2 CONTINUED

104472E+01	10714609E+03	5873911E+04	3375362E+02	3653790E-04	1352025E+03	4470049E-02
104519E+01	567702E+02	5873911E+04	3342132E+02	3671476E-04	1202003E+03	4470049E-02
104520E+01	547979E+02	5934674E+04	3309230E+02	3689161E-04	9788308E+02	4512133E-02
104525E+01	4808719E+02	5914674E+04	3276651E+02	3706866E-04	7085361E+02	4512133E-02
104543E+01	4214427E+01	5935067E+04	3244394E+02	3724531E-04	4126247E+02	4537043E-02
104611E+01	1195623E+02	5935067E+04	3212454E+02	3722216E-04	1256216E+02	4537043E-02
104627E+01	5907737E+02	5936165E+04	3180828E+02	3759901E-04	1527909E+02	4541537E-02
1046694E+01	7420246E+02	5936165E+04	3149514E+02	3777586E-04	1527909E+02	4541537E-02
1046714E+01	1097522E+03	5935433E+04	3118508E+02	379571E-04	1527909E+02	4536133E-02
1046132E+01	1446154E+03	5935433E+04	3087807E+02	3812056E-04	1527909E+02	4536133E-02
104746E+01	1194472E+03	5937401E+04	3057409E+02	3830641E-04	1527909E+02	4530728E-02
1047312E+01	1797102E+03	5934606E+02	3034606E+02	3843905E-04	1527909E+02	4530728E-02

ARRIVAL OF GROUNDWAVE PEAK (SINGULARITY)						

104746E+01	1047309E+02	5934674E+04	3014274E+02	3848326E-04	1527909E+02	4528024E-02
104743E+01	299133E+02	5934674E+04	299133E+02	3853668E-04	1527909E+02	4528024E-02
104749E+01	295374E+02	5934674E+04	295374E+02	3869695E-04	8739761E+01	4526472E-02
104790E+01	292066E+02	5942943E+04	292066E+02	3891064E-04	4637140E+02	4526472E-02
104815E+01	288595E+02	5942943E+04	288595E+02	3912434E-04	6570371E+02	4545058E-02
104844E+01	285164E+02	5947050E+04	285164E+02	3933803E-04	8004714E+02	4545058E-02
104804E+01	281772E+02	5967050E+04	281772E+02	3955172E-04	8961180E+02	4579000E-02
104920E+01	278424E+02	6002424E+04	278424E+02	3974542E-04	9882052E+02	4579000E-02
104947E+01	275120E+02	6002424E+04	275120E+02	3997911E-04	1024286E+02	4620266E-02
104934E+01	271850E+02	6045614E+04	271850E+02	400650E-04	106883E+03	4620266E-02
104997E+01	268420E+02	6045614E+04	268420E+02	4042019E-04	1104751E+03	4665885E-02
105026E+01	265428E+02	6094289E+04	265428E+02	4062019E-04	1133957E+03	4665885E-02
105023E+01	262273E+02	6094289E+04	262273E+02	4083389E-04	1157840E+03	471431E-02
105149E+01	259157E+02	6146747E+04	259157E+02	4104788E-04	1177410E+03	471431E-02
105131E+01	256074E+02	6146747E+04	256074E+02	4126128E-04	1193427E+03	4744607E-02
105175E+01	253034E+02	6201463E+04	253034E+02	4147497E-04	1206479E+03	4764607E-02
105138E+01	250274E+02	6201463E+04	250274E+02	4168865E-04	1237029E+03	4816153E-02
105138E+01	247056E+02	6254734E+04	247056E+02	4190235E-04	1254355E+03	4816153E-02
105334E+01	224523E+02	6254734E+04	224523E+02	4211605E-04	1292009E+03	4868513E-02
105647E+01	204453E+02	6720299E+04	204453E+02	4232560E-04	1240883E+03	4868513E-02
105815E+01	185435E+02	6720299E+04	185435E+02	4253515E-04	1204023E+03	5290405E-02
106055E+01	168522E+02	7125532E+04	168522E+02	4274470E-04	1157127E+03	5290405E-02
106235E+01	153152E+02	7125532E+04	153152E+02	4295425E-04	1098258E+03	5685584E-02
106443E+01	139183E+02	7451497E+04	139183E+02	4325630E-04	1036425E+03	5685584E-02
106656E+01	126489E+02	7451497E+04	126489E+02	4353515E-04	9743462E+02	6039935E-02
106872E+01	114952E+02	7705043E+04	114952E+02	4382560E-04	9135665E+02	6039935E-02
107074E+01	104468E+02	7705043E+04	104468E+02	4419023E+02	8549732E+02	6352417E-02
107284E+01	944014E+01	7899123E+04	944014E+01	4459211E-04	7990644E+02	6352417E-02
107748E+01	804444E+02	804444E+04	7831147E+01	4492111E-04	7400798E+02	6625793E-02
107993E+01	684648E+02	804444E+04	7126004E+01	4521156E-04	6941785E+02	6625793E-02
108197E+01	615795E+02	815795E+04	6443402E+01	4549402E-04	6493134E+02	6843997E-02
108360E+01	588542E+02	815795E+04	6263666E+04	4575931E-04	6054480E+02	6843997E-02
108505E+01	538467E+02	8242194E+04	538467E+01	4604976E-04	5644792E+02	7071171E-02
108741E+01	409044E+02	8242194E+04	4417465E+01	4646866E-04	5262907E+02	7071171E-02
108914E+01	382803E+02	8305904E+04	4014565E+01	4688000E+01	4907438E+02	7251265E-02
109114E+01	358347E+02	8305904E+04	3644413E+01	4728879E-04	4574942E+02	7251265E-02
109322E+01	335435E+02	8354200E+04	3315655E+01	4759751E-04	4269950E+02	7407899E-02
109624E+01	314402E+02	8354200E+04	3031324E+01	4780166E-04	3985003E+02	7407899E-02
1098130E+01	2947105E+02	8390934E+04	2738421E+01	4813357E-04	3720677E+02	7544258E-02
1100435E+01	2766102E+02	8390934E+04	2488660E+01	4843575E-04	3475593E+02	7544258E-02
1102345E+01	2594072E+02	8414999E+04	2284421E+01	4874452E-04	3248434E+02	7663194E-02
110443E+01	2436129E+02	8430494E+04	2261679E+01	4885482E-04	3037944E+02	7663194E-02
						7767153E-02
						7814210E-02

TABLE B.3 OUTPUT FOR A PLANE WAVE BOTTOM REFLECTION

BOTTOM REFLECTION DATE 06/29/71
RUN NUMBER 3
INPUT

DEPTH OF WATER IN FT BIGH= .26042000E+01
DEPTH OF EXPLOSION IN FT D= .14350000E+01
DEPTH OF GAUGE IN FT DGAU= .24900000E+01
GEOMETRY CHANGED SO THAT ARRIVAL TIME OF GROUNDWAVE IS Z2 = .13500000E+02

DEPTH OF EXPLOSION IN FT D= .14370658E+01
DEPTH OF GAUGE IN FT OGAU= .24879342E+01
HORIZONTAL DISTANCE BETWEEN CHARGE AND GAUGE IN FT SMALLP= .39580000E+01
WEIGHT OF EXPLOSIVE CHARGE IN LB (OR KT) WCH= .12500000E-02
VELOCITY OF SOUND IN WATER IN FT/SEC CWATER= .48700000E+04
VELOCITY OF SOUND IN BOTTOM IN FT/SEC CBOT= .58700000E+04
VELOCITY OF SHEARWAVE IN FT/SEC CSHEAR= 0.
DENSITY OF WATER IN GM/CC RHOWAT= .99800000E+00
DENSITY OF BOTTOM IN GM/CC RHOBOT= .18600000E+01
COEFFICIENT OF SW PRESSURE FORMULA IN PSI PRECOE= .15400000E+05
PRINT OUT CONTROL PARAMETER (Z5=0.0 FOR SHORTER PRINT OUT) Z5= .100000E+01
EXPONENT OF SW PRESSURE FORMULA PREEXP= .11300000E+01
COEFFICIENT OF SW TIME CONSTANT FORMULA THECOE= .95000000E-04
EXPONENT OF SW TIME CONSTANT FORMULA THEXP= -.22000000E+00
NUMBER OF SUBDIVISIONS OF THETA STEPS= .10000000E+02
DURATION AFTER DIRECT ARRIVAL IN MULTIPLES OF THETA DURAT= .25925359E+01
DESIRE HATIO BETWEEN INCIDENT AND CRITICAL ANGLE THOVAL= -.10000000E+01
SCALING PARAMETER FOR X-AXIS (PSI PER INCH OF GRAPH) X1= .10000000E+03
SCALING PARAMETER FOR Y-AXIS (MICROSECONDS PER INCH OF GRAPH) X2= .10000000E+02
SLOPE OF BOTTOM IN DEGREES SLOPE= -0.
PARAMETER THAT SELECTS THEORY Z1= .10000000E+01
APPVAL TIME OF GROUND WAVE IN MICROSECONDS Z2= .13500000E+02
PLOT CONTROL PARAMETER (Z3 = 0. MEANS PLOTS ARE WANTED) Z3= .10000000E+01
CYLINDER RADIUS IN FT RADIUS= -0.
PRINT CONTROL PARAMETER. (FULL PRINT OUT IN SUBROUTINE PTV IF APRINT.LE.0.) APRINT= -0.

CHARACTERISTIC MAGNITUDES

ANGLE OF INCIDENT WAVE IN DEGREES THE= .72034497E+02
VELOCITY OF STONLEY WAVE IN FT/SEC CTSON= -0.
POISSON RATIO POISR= .50000000E+00
REDUCED TIME OF SURFACE REFLECTION RS= .13611709E+01
CRITICAL ANGLE OF COMPRESSION WAVE IN DEGREES ALPHA= .56062026E+02
ANGLE OF PRESSURE WAVE IN BOTTOM IN DEGREES THEONE= -0.
REFLECTION COEFFICIENT CP= .10000000E+01
ANGLE OF SHEARWAVE IN BOTTOM IN DEGREES ANGA= -0.
ANGLE OF PHASESHIFT IN DEGREES EE= .77981406E+02
REDUCED TIME OF PRECURSOR ARRIVAL D2R2= .97682869E+00
REDUCED TIME OF PEAK OF BOTTOM REFLECTED WAVE R2= .10160544E+01
SLANT DISTANCE BETWEEN CHARGE AND GAUGE=CHARACTERISTIC LENGTH IN FT. RACTU= .40951299E+01
REDUCED SLANT DISTANCE (RACTU/WCH**1/3) REOR= .38015818E+02
CHARACTERISTIC TIME=RACTU/CWATER IN SECONDS. TACT= .84088909E-03
CHARACTERISTIC PRESSURE=FREE WATER SW PEAK PRESSURE IN PSI PACT= .2524165E+03
REDUCED TIME CONSTANT OF INCIDENT WAVE THETA= .27094453E-01
ACTUAL SW TIME CONSTANT IN MILLISECONDS THET= .22783430E-01
REDUCED TIME CONSTANT OF BOTTOM REFLECTED WAVE THETAR= .27189556E-01
BOTTOM REFLECTED WAVE TIME CONSTANT IN MILLISECONDS THETR= .22863401E-01

CONSTANTS OF THE CALCULATION

SMALLN DEZERO OZ COSAL SINTH
.1167134E+01 -.1050868E+01 .3133967E+00 .5582744E+00 .3084447E+00 .9512423E+00

TABLE B.3 CONTINUED

PLANE WAVE APPROXIMATION					ARONS-YENNIE APPROACH NON-RIGID BOTTOMS						
REDUCED TIME T	BOTTOM REFLECTION PBOT	SHOCKWAVE PD	SURFACE REFLECTION PS	TIME SECONDS	TOTAL PRESSURE P (PSI)	REDUCED TIME T	BOTTOM REFLECTION PBOT	SHOCKWAVE PD	SURFACE REFLECTION PS	TIME SECONDS	TOTAL PRESSURE P (PSI)
.9768287E+00	.3559761E+02	0.	0.	0.	.3559761E+02	.9768287E+00	.3559761E+02	0.	0.	0.	.3559761E+02
.9794437E+00	.3741375E+02	0.	0.	0.	.3741375E+02	.9794437E+00	.3741375E+02	0.	0.	0.	.3741375E+02
.9820589E+00	.3944024E+02	0.	0.	0.	.3944024E+02	.9820589E+00	.3944024E+02	0.	0.	0.	.3944024E+02
.9846738E+00	.4171822E+02	0.	0.	0.	.4171822E+02	.9846738E+00	.4171822E+02	0.	0.	0.	.4171822E+02
.9872889E+00	.4430085E+02	0.	0.	0.	.4430085E+02	.9872889E+00	.4430085E+02	0.	0.	0.	.4430085E+02
.9899039E+00	.4725831E+02	0.	0.	0.	.4725831E+02	.9899039E+00	.4725831E+02	0.	0.	0.	.4725831E+02
.9925190E+00	.5068544E+02	0.	0.	0.	.5068544E+02	.9925190E+00	.5068544E+02	0.	0.	0.	.5068544E+02
.9951340E+00	.5471418E+02	0.	0.	0.	.5471418E+02	.9951340E+00	.5471418E+02	0.	0.	0.	.5471418E+02
.9977491E+00	.5953483E+02	0.	0.	0.	.5953483E+02	.9977491E+00	.5953483E+02	0.	0.	0.	.5953483E+02
.1000000E+01	.6453371E+02	.2524417E+03	0.	0.	.3169754E+03	.1000000E+01	.6453371E+02	.2524417E+03	0.	0.	.3169754E+03
.1002615E+01	.7171971E+02	.2292158E+03	0.	0.	.3009355E+03	.1002615E+01	.7171971E+02	.2292158E+03	0.	0.	.3009355E+03
.1005230E+01	.8109632E+02	.2081269E+03	0.	0.	.2892232E+03	.1005230E+01	.8109632E+02	.2081269E+03	0.	0.	.2892232E+03
.1007845E+01	.9405649E+02	.1889783E+03	0.	0.	.2830347E+03	.1007845E+01	.9405649E+02	.1889783E+03	0.	0.	.2830347E+03
.1010460E+01	.1137444E+03	.1715914E+03	0.	0.	.2853359E+03	.1010460E+01	.1137444E+03	.1715914E+03	0.	0.	.2853359E+03
.1013075E+01	.1499189E+03	.1558042E+03	0.	0.	.3057231E+03	.1013075E+01	.1499189E+03	.1558042E+03	0.	0.	.3057231E+03
.1015690E+01	.1618262E+03	.1528255E+03	0.	0.	.3146517E+03	.1015690E+01	.1618262E+03	.1528255E+03	0.	0.	.3146517E+03
.1018305E+01	.1770567E+03	.1499037E+03	0.	0.	.3269605E+03	.1018305E+01	.1770567E+03	.1499037E+03	0.	0.	.3269605E+03
.1020920E+01	.1978137E+03	.1470379E+03	0.	0.	.3448515E+03	.1020920E+01	.1978137E+03	.1470379E+03	0.	0.	.3448515E+03
.1023535E+01	.2299034E+03	.1442268E+03	0.	0.	.3737502E+03	.1023535E+01	.2299034E+03	.1442268E+03	0.	0.	.3737502E+03
.1026150E+01	.2933076E+03	.1414694E+03	0.	0.	.434770E+03	.1026150E+01	.2933076E+03	.1414694E+03	0.	0.	.434770E+03
.1028765E+01	.2479390E+03	.1395809E+03	0.	0.	.4415582E+03	.1028765E+01	.2479390E+03	.1395809E+03	0.	0.	.4415582E+03
.1031380E+01	.3046459E+03	.1369124E+03	0.	0.	.445502E+03	.1031380E+01	.3046459E+03	.1369124E+03	0.	0.	.445502E+03
.1033995E+01	.2459064E+03	.1342949E+03	0.	0.	.482013E+03	.1033995E+01	.2459064E+03	.1342949E+03	0.	0.	.482013E+03
.1036610E+01	.2102222E+03	.1317274E+03	0.	0.	.519696E+03	.1036610E+01	.2102222E+03	.1317274E+03	0.	0.	.519696E+03
.1039225E+01	.1842524E+03	.1292090E+03	0.	0.	.5419698E+03	.1039225E+01	.1842524E+03	.1292090E+03	0.	0.	.5419698E+03
.1041840E+01	.1636880E+03	.1267388E+03	0.	0.	.5234615E+03	.1041840E+01	.1636880E+03	.1267388E+03	0.	0.	.5234615E+03
.1044455E+01	.1466344E+03	.1243158E+03	0.	0.	.5290501E+03	.1044455E+01	.1466344E+03	.1243158E+03	0.	0.	.5290501E+03
.1047070E+01	.1320882E+03	.1219391E+03	0.	0.	.52709501E+03	.1047070E+01	.1320882E+03	.1219391E+03	0.	0.	.52709501E+03
.1049685E+01	.1193354E+03	.1196079E+03	0.	0.	.5238943E+03	.1049685E+01	.1193354E+03	.1196079E+03	0.	0.	.5238943E+03
.1052300E+01	.7279964E+02	.1086034E+03	0.	0.	.521731E+04	.1052300E+01	.7279964E+02	.1086034E+03	0.	0.	.521731E+04
.1054915E+01	.6423667E+02	.9861135E+02	0.	0.	.5214628E+04	.1054915E+01	.6423667E+02	.9861135E+02	0.	0.	.5214628E+04
.1057530E+01	.2035509E+02	.8953865E+02	0.	0.	.521525E+04	.1057530E+01	.2035509E+02	.8953865E+02	0.	0.	.521525E+04
.1060145E+01	.6037982E+01	.8130067E+02	0.	0.	.521421E+04	.1060145E+01	.6037982E+01	.8130067E+02	0.	0.	.521421E+04
.1062760E+01	.1840837E+02	.7382063E+02	0.	0.	.5219401E+04	.1062760E+01	.1840837E+02	.7382063E+02	0.	0.	.5219401E+04
.1065375E+01	.2544547E+02	.6702879E+02	0.	0.	.521818E+04	.1065375E+01	.2544547E+02	.6702879E+02	0.	0.	.521818E+04
.1067990E+01	.3120062E+02	.552225E+02	0.	0.	.521215E+04	.1067990E+01	.3120062E+02	.552225E+02	0.	0.	.521215E+04
.1070605E+01	.3526452E+02	.5017786E+02	0.	0.	.52112E+04	.1070605E+01	.3526452E+02	.5017786E+02	0.	0.	.52112E+04
.1073220E+01	.3898950E+02	.4556126E+02	0.	0.	.521008E+04	.1073220E+01	.3898950E+02	.4556126E+02	0.	0.	.521008E+04
.1075835E+01	.4136941E+02	.4136941E+02	0.	0.	.520905E+04	.1075835E+01	.4136941E+02	.4136941E+02	0.	0.	.520905E+04
.1078450E+01	.4330813E+02	.3756323E+02	0.	0.	.520802E+04	.1078450E+01	.4330813E+02	.3756323E+02	0.	0.	.520802E+04
.1081065E+01	.4455562E+02	.3410724E+02	0.	0.	.520700E+04	.1081065E+01	.4455562E+02	.3410724E+02	0.	0.	.520700E+04
.1083680E+01	.4534303E+02	.3096922E+02	0.	0.	.520600E+04	.1083680E+01	.4534303E+02	.3096922E+02	0.	0.	.520600E+04
.1086295E+01	.4575551E+02	.2811990E+02	0.	0.	.520500E+04	.1086295E+01	.4575551E+02	.2811990E+02	0.	0.	.520500E+04
.1088910E+01	.4586286E+02	.2553274E+02	0.	0.	.520400E+04	.1088910E+01	.4586286E+02	.2553274E+02	0.	0.	.520400E+04
.1091525E+01	.457262E+02	.2318361E+02	0.	0.	.520300E+04	.1091525E+01	.457262E+02	.2318361E+02	0.	0.	.520300E+04
.1094140E+01	.4538237E+02	.2105061E+02	0.	0.	.520200E+04	.1094140E+01	.4538237E+02	.2105061E+02	0.	0.	.520200E+04
.1096755E+01	.4488166E+02	.1911385E+02	0.	0.	.520100E+04	.1096755E+01	.4488166E+02	.1911385E+02	0.	0.	.520100E+04

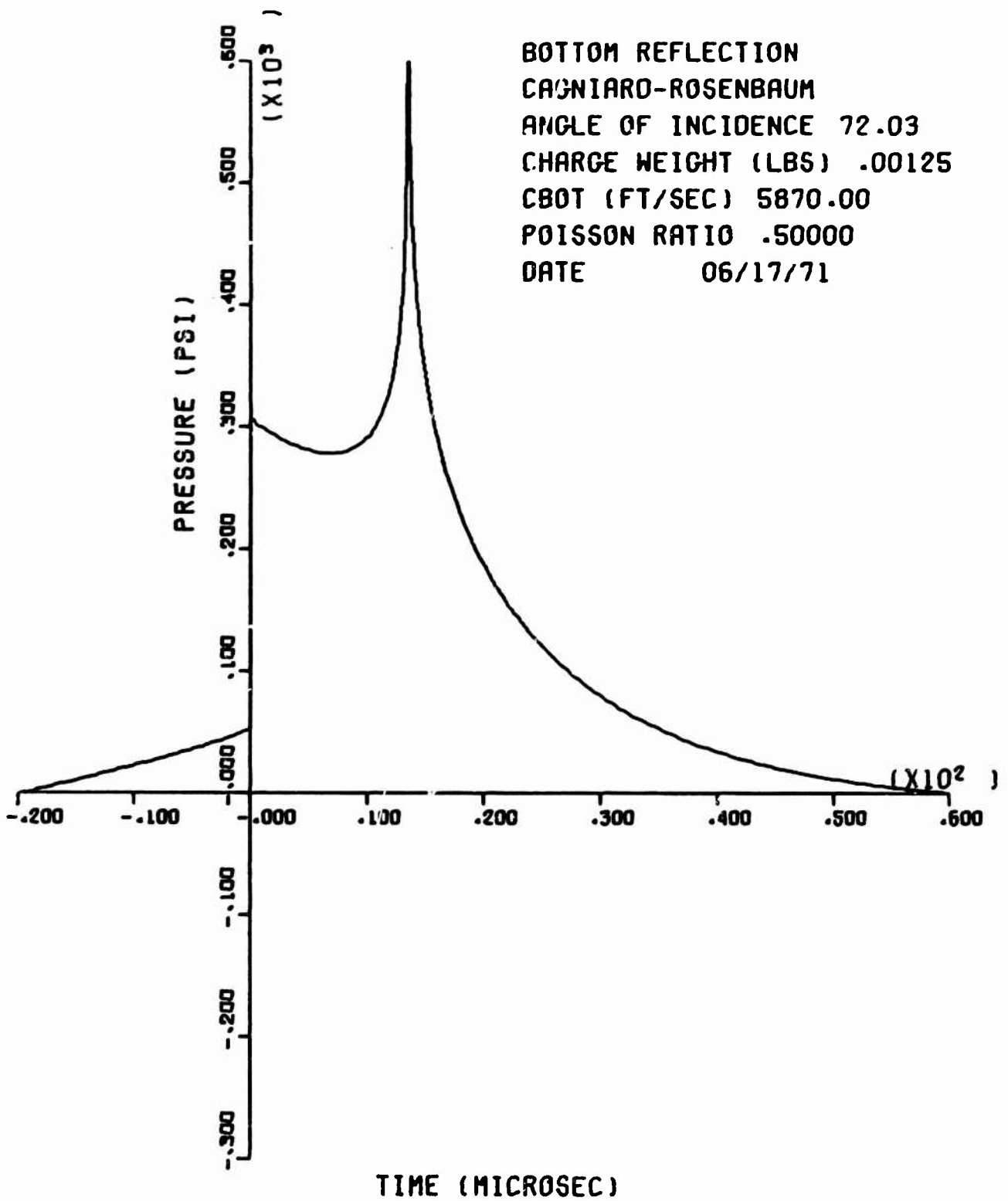


FIG. 2 SAMPLE CALCOMP PLOT OF PRESSURE-TIME HISTORY

APPENDIX C
TWO SPECIAL INPUT OPTIONS

Two special input options for altering the input geometry are provided in the code BOTREF using the variables THOVAL and Z2. The first option allows the programmer to specify an incident angle θ which is expressed as the ratio of the incident angle θ to the critical angle θ_{cr} . This is accomplished by setting $THOVAL = \theta/\theta_{cr} > 0$. If $THOVAL \leq 0$, this option is ignored. The programmer also must supply the water depth H , the horizontal range r , and the gauge depth d_g . If possible, the code calculates the required charge depth d keeping d_g fixed. Otherwise, d is set to H , and a new value of d_g is determined. Thus this option permits the user to calculate bottom reflections for a range of incident angles without first having to determine the exact geometry that is required.

The second option using $Z2 > 0$ provides an alternative means of specifying the reflection geometry. The arrival time $Z2$ (in microseconds) of the bottom reflected wave after the direct wave is exceedingly sensitive to the geometry which often cannot be measured with the necessary accuracy. This time, which can be accurately measured, provides the means which can be used to correct the input geometry. The geometry is changed by altering d and d_g and holding r fixed so that the incident angle θ and the bottom reflected slant range R_r are unchanged. Typically, the direct and reflected pulses are only slightly altered, but the change in their sum may be significant.