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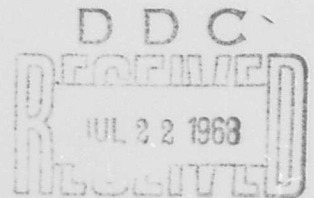
TWO DIMENSIONAL
TRANSIENT HEATING AND
SURFACE THERMOCHEMISTRY
COMPUTER PROGRAM

Second Interim Technical Report
Contract F04611-67-C-0047

Prepared by
Aerotherm Corporation
Under Subcontract 2007

to

Atlantic Research Corporation
A Division of The Susquehanna Corporation
Shirley Highway at Edsall Road
Alexandria, Virginia 22314



July 5, 1968

ATLANTIC  RESEARCH

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**TWO DIMENSIONAL
TRANSIENT HEATING AND
SURFACE THERMOCHEMISTRY
COMPUTER PROGRAM**

**Second Interim Technical Report
Contract F04611-67-C-0047**

**Prepared by
Carl B. Moyer
Aerotherm Corporation**

July 5, 1968

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FOREWORD

This report was prepared by Aerotherm Corporation under subcontract to Atlantic Research Corporation under Contract F04611-67-C-0047. The work was under the direction of the Air Force Rocket Propulsion Laboratory with Mr. Robert Schoner and Lt. David Zorich as project officers.

Mr. Carl Moyer of Aerotherm prepared the report with assistance from Mr. Mitchell R. Wool and Dr. Robert M. Kendall. Mr. Eugene Olcott served as subcontract monitor for Atlantic Research.

The computer programs described in this report were used in the subsequent study of the reactions of combustion products of solid propellant rockets with pyrolytic graphite. These studies are described in Volume II, a classified report, of AFRPL-TR-68-116.

Publication of this report does not constitute Air Force approval of the report's findings or conclusions. It is published only for the exchange and stimulation of ideas.

CHARLES R. COOKE

Charles R. Cooke
Chief, Solid Rocket Division

ABSTRACT

The report describes and gives user's instructions for a computer program (ASTHMA) for predicting the in-depth temperature history and the surface recession (ablation) history of a two-dimensional, axi-symmetric, non-charring material. The in-depth solution procedure is of the conventional explicit finite-difference type. It will account for anisotropic heat conduction in the main material and allows several back-up materials. The heated surface boundary condition can have any one of three forms at each surface location at each instant:

1. Very general film-coefficient-based simultaneous heat and mass transfer type, including any number of equilibrium reactions for an environment and any ablating material, and four specific kinetically controlled reactions for carbon surfaces
2. Simplified radiation energy balance type, no ablation ("cooldown" option)
3. Specified temperature and surface recession rate.

The program was designed specifically for rocket nozzle use but has sufficient generality that it can be employed for many axi-symmetric shapes.

A User's Manual Brief Description is also presented for a computer program (ARCACE) for calculating surface thermochemical response of materials, including kinetically controlled surface reactions. This program provides for surface thermochemical boundary condition information input to the ASTHMA program. The program was designed specifically for graphitic material and solid propellant combustion products environment but can be employed for many materials and chemical systems.

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PART II

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LIST OF SYMBOLS

A	area of side of nodal box	ft ²
a,b,c	nodal box side and center line lengths, Figure 7, Equation (14)	in
B'	defined as $(B'_c - \sum \dot{m}_{r_l} / \rho_e u_e C_M)$ thermochemical ablation parameter	---
B'_c	defined as $\dot{m}_c / \rho_e u_e C_M$, total ablation parameter	---
B'_f	defined as $\sum \dot{m}_{r_l} / \rho_e u_e C_M$	---
B'_g	defined as $\dot{m}_g / \rho_e u_e C_M$	---
B'_{tc}	same as B'	---
b	see a,b,c	
C	nodal capacity, Equation (16)	Btu/ ^o F
C _H	Stanton number for heat transfer (corrected for "blowing", if necessary)	lb/ft ² sec
C _{H0}	Stanton number for heat transfer not corrected for blowing	lb/ft ² sec
C _{k_i}	number of k atoms in molecule i	---
C _M	Stanton number for mass transfer	lb/ft ² sec
c	specific heat	Btu/lb ^o F
c	see a,b,c	in
\bar{D}	constant defined by Equation (36)	ft ² /sec
D _{ij}	binary diffusion coefficient	ft ² /sec
F	radiation view factor	---
F _i , F _j	empirical factors appearing in Equation (36)	---

LIST OF SYMBOLS (continued)

f	denotes general functional relationship	---
H_r	recovery enthalpy	Btu/lb
h	enthalpy	Btu/lb
h_c	enthalpy of ablating material at wall temperature	Btu/lb
h_w	enthalpy of gases adjacent to the wall	Btu/lb
I	total number of identifiable species	---
I	number of iteration cycles in problem	---
i, j	chemical species indices	---
K	total number of elements in system	---
K_i	mass fraction of species i	---
\tilde{K}_k	mass fraction of element k (regardless of molecular configuration)	---
k	thermal conductivity	Btu/ft-sec ⁰ R
k_f	forward rate constant for kinetically controlled reaction	various
L	path lengths in a nodal box, Figure 5	ft
\bar{m}	system molecular weight $\sum x_i m_i$	lb/lb mole
m_i	molecular weight of species i	lb/lb mole
m	node corner and center row number	---
\dot{m}	mass flow rate per unit area from the surface, thermochemical effects only	lb/ft ² sec
\dot{m}_c	total mass flow rate of "char" or main ablating material per unit surface area, all effects (thermochemical plus condensed phase mechanical removal)	lb/ft ² sec
\dot{m}_g	mass flow rate of pyrolysis gas out a unit area of surface	lb/ft ² sec

LIST OF SYMBOLS (continued)

$\dot{m}_{r,l}$	flow rate of condensed phase l mechanically removed from surface	lb/ft ² sec
N	total number of nodes	---
n	node corner and center column number	---
P	total pressure	lb/ft ²
P_i	partial pressure of species i	lb/ft ²
q_{chem}	chemical energy flux term defined by Equation (36)	Btu/ft ² sec
q_{cond}	rate of energy conduction into solid material at surface	Btu/ft ² sec
q_{diff}	rate of energy input to solid surface by diffusional processes in the boundary layer	Btu/ft ² sec
$q_{rad in}$	rate of energy input to the surface by radiation from the boundary layer or from outside the boundary layer, same as q_{rad}	Btu/ft ² sec
$q_{rad out}$	rate of energy radiated away from surface	Btu/ft ² sec
q_{sen}	$\Delta \rho_e u_e C_H (H_r - h_{e_w})$	Btu/ft ² sec
R	thermal resistance, see Equations (17), (18)	sec ^o F/Btu
R*	contact thermal resistance	ft ² sec ^o F/Btu
r	radius	in
S	see ΔS	^o R
T	computing time	hrs
T	temperature	^o R
T_w	wall (surface) temperature, general term for $T_{s,n}$	^o R

LIST OF SYMBOLS (continued)

u_e	velocity of gases at edge of boundary layer	ft/sec
V	nodal volume	ft ³
v	gas velocity (see ρv)	ft/sec
x_i	mole fraction of species i	---
Z	axial coordinate	ft
Z_i^*	diffusion driving force, see Equations (33) and (34)	---

GREEK

α	heated surface absorbtivity (taken equal to ϵ_w), thermal diffusivity	---
γ	constant, empirically chosen = 2/3	---
Δ	denotes change	---
ΔS	change in surface point location during $\Delta\theta$	ft
$\Delta\theta$	time step in finite difference solution	sec
ϵ	emissivity or emittance	---
η	input multiplicative safety factor in time step calculation, Equation (20)	-----
θ	time	sec
λ	constant in Equation (39)	---
μ_2	dimensionless factor defined by Equation (35)	---
ρ	density	lb/ft ³
$(\rho v)_w$	$\dot{m} + \dot{m}_g$	lb/ft ² sec
$\rho_e u_e$	mass flow at boundary layer edge	lb/ft ² sec

LIST OF SYMBOLS (continued)

$\rho_e^u C_H$	heat transfer convective film coefficient	lb/ft ² sec
$\rho_e^u C_M$	mass transfer convective film coefficient	lb/ft ² sec
σ	Stefan-Boltzmann constant	Btu/ft ² sec ⁰ R ⁴
ϕ	parameter defined by Equation (39)	---

SUBSCRIPTS

A,B,C,D	indices for path lengths L (Fig. 5), nodal side areas A (Fig. 6) and thermal resistances R (Fig. 10)
B	see A,B,C,D
C	see A,B,C,D
CL	denotes nodal column center line
c	denotes "char" or ablating material; see \dot{m}_c
D	see A,B,C,D
e	denotes boundary layer outer edge, or boundary layer edge gas
g	denotes pyrolysis gas
H	see C _H
i,j	denotes any identifiable species: atom, ion, molecule
k	denotes element
l	index of condensed phase species mechanically removed from surface
M	see C _M
m	node corner and center row number index
N	denotes node center

LIST OF SYMBOLS (concluded)

- n node corner and center column number index
- p denotes virgin plastic
- r see \dot{m}_r
- s denotes heated surface
- w denotes wall, i.e., heated surface
- O see C_{H_0}
- 1,2 denote "earlier" and "later"
- θ, θ' at times θ and θ' , respectively

SUPERSCRIPTS

- '(prime) see $B', B'_c, B'_f, B'_g, B'_{tc}$
- '(prime) denotes at new time $\theta' = \theta + \Delta\theta$
- T_w enthalpy datum temperature

SPECIAL SYMBOLS

- * see Z^*_i

GLOSSARY

Some of the terms used in the text have very particular meanings. These are collected here for ready reference.

back wall	one of the four sides of the nodal mesh layout, located by convention at the bottoms of the columns; may be convectively and radiatively cooled but does not ablate
box	see "nodal box"
column	refers to one direction of the nodal mesh, heated surface is at the top of the columns
heated surface	the one surface of the four sides of the nodal mesh layout which is exposed to the hyperthermal, chemically reactive environment, located by convention at the top of the columns in the mesh; opposite side of mesh network is the back wall
mesh	assemblage of quadrilateral nodal boxes arranged in rows and columns, each row and each column having the same number of nodal boxes; encompasses all the material of interest and divides it into nodes for finite difference heat conduction analysis
nodal box	one quadrilateral zone in the mesh
nodal center, nodal point	a point within the nodal box at which all the material in the box is presumed to be lumped for the finite difference calculation, not necessarily in the center of the node
node	used for nodal box, nodal center, nodal point
null node	a nodal box in the mesh which contains no material, used for bookkeeping operations only, not involved in finite difference solution
row	refers to one direction in nodal mesh, "parallel" to heated surface
surface node	the top node in each column (excepting null nodes), that node next to (adjacent to, at) the heated surface; note that the surface nodal box is next to the heated surface and one of its four sides is exposed to the hyperthermal environment, the nodal point or nodal center however is on the side of the node farthest from the heated surface; there is a special point on the surface called the surface point which must not be confused with the surface nodal "center"

GLOSSARY (concluded)

surface point

the center of the heated surface side of the top nodal box in each column (not counting null nodes); this point is distinct from the nodal point for that nodal box; the temperature of this point is determined by the surface energy balance in Options 1 and 3 or by assignment in Option 2

INTRODUCTION
AXI-SYMMETRIC TRANSIENT HEATING AND MATERIAL ABLATION
PROGRAM (ASTHMA) DESCRIPTION AND USER'S MANUAL

This report presents an analysis technique and describes an associated computer program (ASTHMA) for the prediction of the thermal response and ablation of two-dimensional axi-symmetric non-charring materials. Chief applications for the computer program are rocket nozzles and entry vehicle nose tips. The in-depth temperature prediction is of the familiar explicit finite-difference type. It allows a completely general finite difference mesh layout relative to the physical r-z axes, and accounts for anisotropic heat conduction effects. The heated surface boundary condition is an unusually general thermochemical type. It accounts for three specific kinetically controlled carbon oxidation reactions, one specific kinetically controlled carbon reduction reaction, one kinetically controlled water gas shift reaction, and any number of equilibrium reactions for any combination of ablating materials and environments.

The present report has two parts. The final part is a complete, but condensed and succinct, user's manual giving explicit instructions for the preparation of input to the program and the interpretation of program output. Part I of the report presents some of the underlying analysis foundation upon which the computer program is built. Since much of the pertinent analysis work has previously been published elsewhere, these particular parts of the analysis are treated only briefly here and the reader is referenced to the previous reports for more information. The following sketch serves to clarify various physical aspects of the ablation problem treated. The

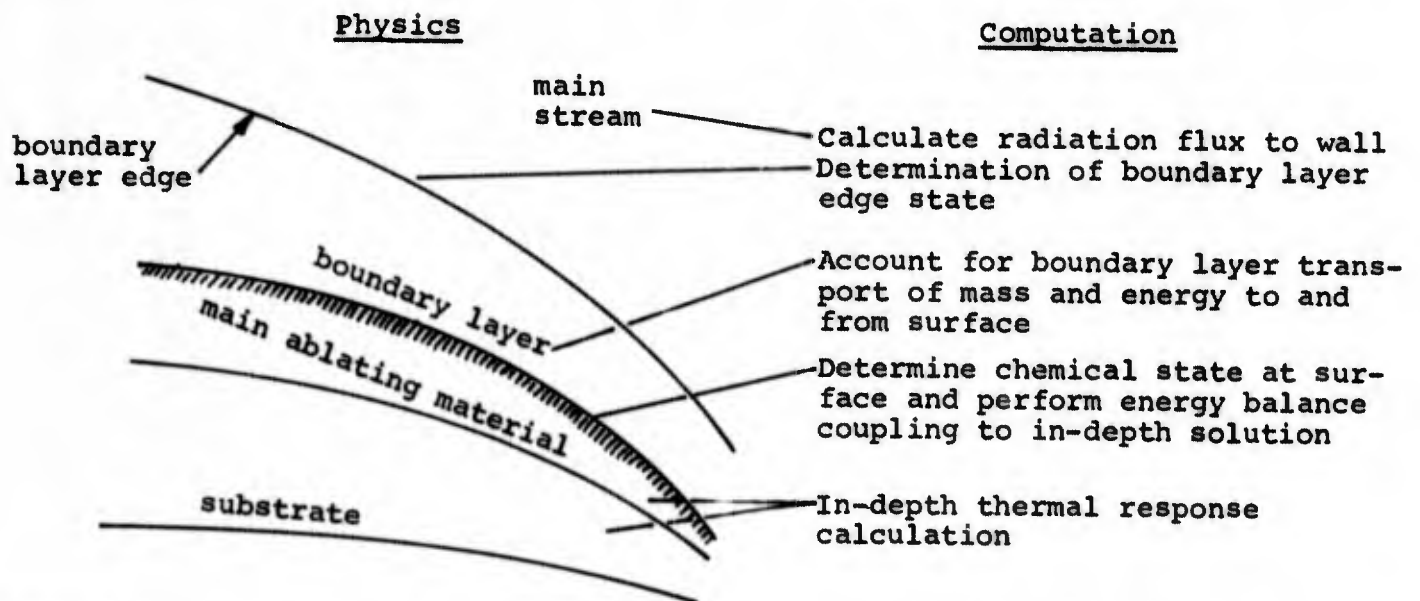


table below cites the aspects and indicates which descriptions of the analysis are to be found in the present report and which are to be found elsewhere. Note that the ASTHMA Program described in this report is usually used in

Task II - Determination of the kinetic rate constants for the reactions of H_2O , H_2 , and CO_2 with pyrolytic graphite using an arc plasma generator and two-dimensional nozzle configuration

Task III - Prediction of pyrolytic coating performance in a typical nozzle configuration for several propellants and firing conditions

The results of Tasks II and III are presented in a separate report.

The author would like to express his appreciation to Mr. Mitchell R. Wool and Dr. Robert M. Kendall who prepared Appendices C and E through G.

SUMMARY OF COMPUTATIONAL TASKS

<u>Task</u>	<u>Program</u>	<u>Where Described</u>
1. Determine radiation flux from free stream to wall	Not computed, but accounted for by ASTHMA. User must provide as input	--
2. Determine boundary layer edge state	ACE	Ref. 9
3. Account for boundary layer transport of		
a. mass	ACE*	Refs. 1-9
b. energy	ASTHMA, but user must provide transport coefficients as input	Briefly below, also Refs. 1-6
4. Surface thermochemical state	ACE*	Refs. 6-9
5. In-depth temperature	ASTHMA	Below

conjunction with another program (ACE) which handles the chemical state computations required in the heated surface boundary condition definition. The ACE program is not described in detail here; however, a user's manual, sample problems and program listing are included as appendices. The reader is referred to Reference 9 for a detailed description of the ACE program.

A complete ablation problem may be solved with the ACE and ASTHMA programs in conjunction in the following steps:

- (1) Lay out nodal geometry, provide this and material properties data to ASTHMA
- (2) Determine radiation flux history and $\rho_e u_e C_H$ history; provide to ASTHMA
- (3) Determine chemical nature of environment and generate surface thermochemical ablation tables with ACE for input to ASTHMA
- (4) Run ASTHMA, obtain temperature and surface recession histories..

The work presented herein represents Task I under Atlantic Research Corporation Subcontract 2007 (Air Force Rocket Propulsion Laboratory Prime Contract F04611-67-C-0047). The overall Aerotherm program consists of the following tasks:

- Task I - Development of a two-dimensional heat conduction program appropriate to pyrolytic graphite coatings on substrates

*The KCG program, Ref. 6, may also be used for these calculations but it has now been superseded by the ACE program.

PART I

DESCRIPTION OF ANALYSIS FOUNDATIONS FOR THE AXI-SYMMETRIC TRANSIENT HEATING AND MATERIAL ABLATION PROGRAM (ASTHMA)

This section describes much of the analysis work upon which the ASTHMA program is built. Section 1 below defines the problem treated, and Section 2 gives some description of the in-depth solution procedure. Section 3 briefly describes the heated surface boundary condition treatment.

SECTION 1

PROBLEM TREATED AND GENERAL METHOD OF SOLUTION

1.1 PROBLEM DESCRIPTION

The basic problem is to predict the surface and in-depth temperature history and surface location history of a two-dimensional, non-charring, anisotropic (but orthotropic) insulating material exposed to a chemically reactive hyperthermal environment. The chief practical examples are rocket nozzles and nose tips; the general problem is depicted schematically in Figure 1-1. Below the heated surface, the material response is characterized by

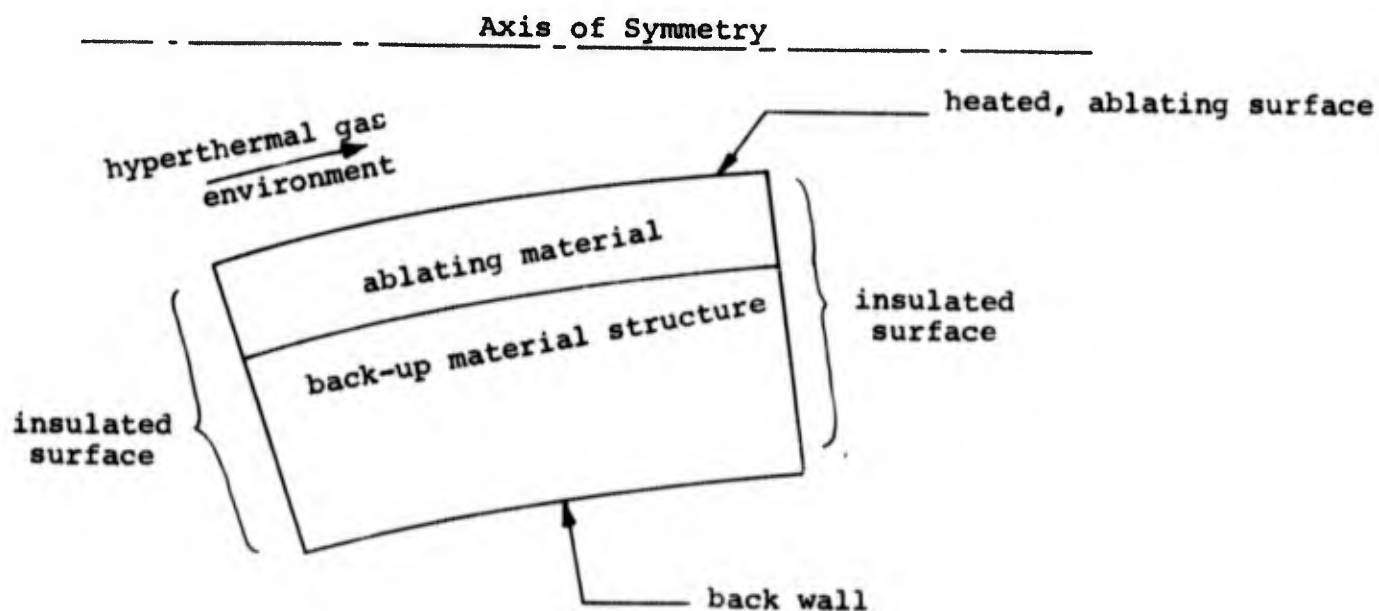


Figure 1-1. Sketch of Ablating System Considered

anisotropic heat conduction with temperature dependent thermal conductivity and specific heat. At the heated surface there is a complex boundary condition involving thermal convection from the hot gases adjacent to the surface, diffusive mass transport, and chemical corrosion. The surface chemical reactions may be in equilibrium or may be kinetically controlled in certain respects (e.g., the oxidation of carbon surfaces).

1.2 SUMMARY OF SOLUTION PROCEDURE

The solution procedure chosen for this problem is complex and is described in more detail in Sections 2 and 3 below. As a quick summary, the in-depth procedure is an explicit two-dimensional finite-difference technique. The surface response procedure solves a general energy balance and mass balance based on a film coefficient model. The ablation events are computed through consideration of the complete thermochemical response including up to four specific kinetically controlled reactions with a carbonaceous surface (all other possible chemical reactions are considered to occur in equilibrium). This approach is in contrast to more usual correlation schemes or "heat of ablation" approaches. There is no direct feedback coupling, however, between body shape change and boundary condition history; that is, the program does not include routines to compute pressure distributions and convective transfer coefficient distributions as functions of body shape (these are input a priori as functions of body location and time).

The basic computer program is called ASTHMA, for Axi-Symmetric Transient Heating and Material Ablation." It relies upon an additional program (ACE) for the calculation of thermochemical aspects of the ablation process.

1.3 COUPLING TO STRESS CALCULATIONS

The program does not include thermal stress calculations but does provide for punched card output which may be used directly as input to a separate stress program (not described here).

1.4 HISTORICAL DEVELOPMENT

The ASTHMA computer code described here constitutes a combination of parts of existing programs. Consequently there exists a large amount of background literature on certain parts of the computer code. This literature will be cited below, and the present report will abbreviate descriptions of aspects already presented elsewhere. Consequently the reader may want to consult other publications for more detailed expositions of those aspects, particularly the surface energy balance equations (Refs. 1-6) and the surface thermochemical state solution procedure (Refs. 6-9).

SECTION 2

ASPECTS OF IN-DEPTH SOLUTION

2.1 BACKGROUND REMARKS

It was the intent of the program development effort reported here not to write a new in-depth heat conduction routine since it seemed that either of two existing routines might be modified to form the basis of the new ablation program. The first such predecessor program is the One Moving Boundary Program (LMBA) reported in Reference 6. This program was originally developed for rocket nozzles and features a rather general thermochemical boundary condition. It does not allow consideration of backup materials, the effects of unequal diffusion coefficients, or mechanical removal. The chief defect of the LMBA program for the requirements herein is the absence of a back-up material capability. A second candidate program was the Aerotherm Axi-symmetric Transient Temperature Program, AATT (Ref. 10). This program did not include ablation and therefore did not allow for receding surfaces.

It was originally intended to modify the LMBA program for the present work since it already featured an ablation boundary condition. However further study indicated that reworking the existing back wall boundary condition treatment of the LMBA program would involve a disproportionate amount of work.* Therefore, consideration was given to adding an ablation "package" to the existing AATT program. Such a development would avoid back-wall complexities and could incorporate without essential difficulty the latest developments in surface ablation computation (as reported in Refs. 3, 5, and 7). However it would involve a frontal attack on a hitherto unsolved computational problem: how to construct a front node dropping scheme which would accommodate an energy-balance-determined surface temperature and recession rate without introducing oscillations in temperature and recession rate. In the front node dropping approach, the nodal network for heat conduction remains fixed with time except in the immediate vicinity of the surface. As the surface recedes due to ablation, the nodal zone adjacent to the surface shrinks. When "sufficiently" small, this node is either "dropped" from the original network or "absorbed" into the adjacent node down and computation continues. It had proven in the past to be a difficult task to design a computation scheme of the front node

* This is the classic difficulty in any scheme like the LMBA routine which has a nodal network tied to the receding surface, so that "nodes" are dropped at the back wall of the ablating material as the heated surface moves. Such a scheme is simple in one space dimension, but two-dimensional body shape changes involve a coordinate bending as well as recession greatly complicating the back wall treatment.

ing type which is free from oscillations in surface temperature or surface heat flux or both. Consequently front node dropping schemes had gone in favor and had been superseded by shrinking nodal network schemes and wall node dropping schemes.

In an attempt to develop a front node dropping scheme which could be added to the existing AATT program, the whole problem of oscillations was treated with a special purpose one-dimensional heat conduction program. Appendix A reports the details of the study; the conclusion was that a smooth solution did exist although it required some sacrifice in accuracy. This scheme was coupled into the existing AATT program to form the two dimensional axisymmetric ablation and heat transfer program (the ASTHMA program) described in the sections below.

2 NODAL LAYOUT AND GEOMETRY

2 General Pattern

The geometric shape considered, illustrated in Figure 2-1, is imagined to be divided up into a grid pattern in the customary manner for finite difference computation schemes. The area within each grid "box" will be termed a nodal box; this is in contrast to terminology which calls each corner of the network a node. The thermal capacity of each node is imagined to be concentrated at a single point within the box; this point will be termed the nodal center. It will be convenient to assume for the moment that the nodal center is located anywhere within the nodal box. Strategies for optimizing the location of the nodal centers will be discussed later.

For convenience, the nodes are imagined to be quadrilaterals, so that the entire nodal network is an assemblage of quadrilaterals. For bookkeeping

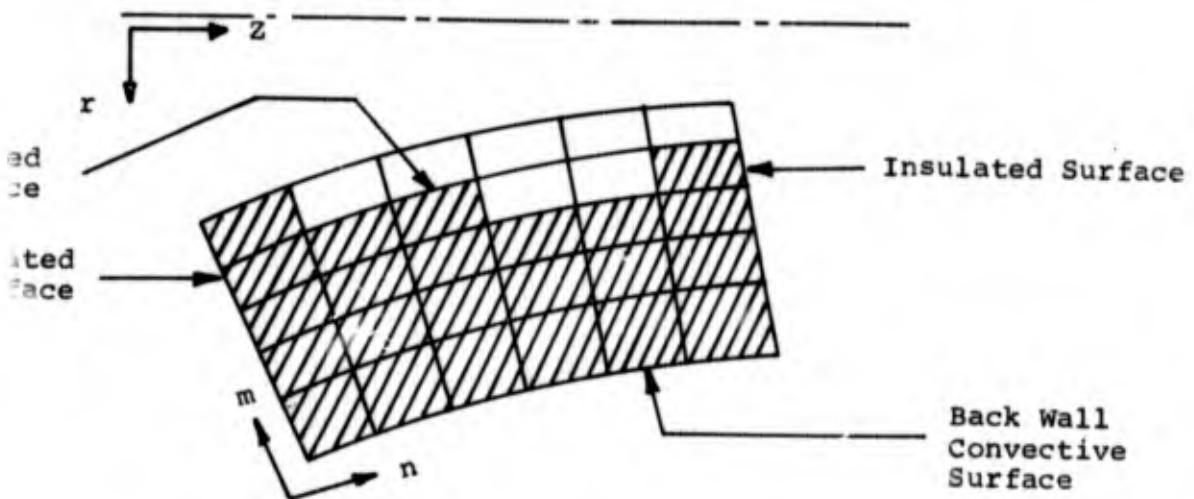


Figure 2-1. Sketch of Typical Nodal Layout

convenience, the network is imagined to be "complete", even though the shape of the material may dictate that some of the mesh boxes are empty. Both nodes (boxes or centers) and mesh corners are numbered in a row and column system which will be labeled as an m-n system, where m denotes a row and n denotes a column. The m-n mesh scheme may be oriented arbitrarily with respect to the physical r-Z coordinate scheme,* so that in general the row-column description might seem to be merely a descriptive artifice; however, in exploiting the simplicities associated with rocket nozzle geometries it has been assumed in constructing the program that the heated surface is at the top of the columns, as indicated in Figure 2-1. Thus as surface recession occurs, the boxes at the top of each column shrink; the other boxes remain fixed. This limitation that the heated surface be located at the top of each column is merely a convenience procedure exploited for the special case of the low curvature geometry of rocket nozzles. For high curvature bodies this restriction becomes inconvenient and would have to be relaxed.

The side walls of the nodal network are presumed insulated and the back wall communicates through a simple heat transfer coefficient law (plus radiation) to a "reservoir" at T_{res} . Thus the boundary condition at this face does not involve thermochemistry.

Three other "convenience limitations" have been applied to the layout of the nodal grid. First, it is assumed for the purposes of computing thermal conductances between nodal centers that the mesh scheme is nearly orthogonal, so that conductance may be taken as conductivity times side-area divided by length.** Secondly, it is presumed that only one material ablates, and that only one material is exposed to the heated surface. (These bookkeeping simplicities can easily be removed in future versions of the program.) Thirdly, it is presumed that the principal directions of thermal anisotropy are aligned with the nodal mesh. This simplifies computations and reduces input requirements. For those applications in which the heated surface intersects the principal directions of anisotropy at "difficult" angles this restriction could become a major inconvenience and more general schemes would have to be devised for those problems.

* That is, the nodal mesh scheme may be above or below the Z-axis line, may be oriented in any general direction, and may be "bent" or shaped in any manner convenient to the user (subject to the limitations described in the paragraphs following below).

** A suitable more general conductance calculating scheme could remove this restriction without any essential change to the program.

2.2.2 Geometry

2.2.2.1 Location of Nodal Centers

The study summarized in Appendix A showed that to obtain a smooth node dropping scheme it was necessary to avoid associating thermal capacity with the surface temperature. In terms of the conventional thermal RC network this means that a one-dimensional scheme would look like Figure 2-2. All the capacity

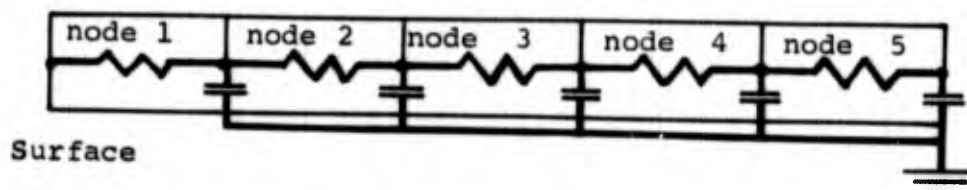


Figure 2-2. Sketch of One-Dimensional Thermal RC Network

of the surface node is located at the back of the first nodal zone, and all of the thermal resistance of this zone is interposed between the surface temperature and the temperature of the first capacity lump. Note that for m nodes or boxes the system has $m+1$ temperature points, in contrast to most schemes.*

In two dimensions, the nodal center is therefore located at the center of the "back wall" of the nodal box, that is, in the center of that side parallel to the heated surface and farthest from it. Thus for an $m \times n$ nodal network scheme, there will be $m \times n$ nodal temperatures (associated with thermal capacity) plus n surface temperatures (not associated with thermal capacity).

Denoting the nodal center coordinates as r_N and z_N , the coordinates for node m , with corner coordinates m,n as shown in the sketch may be represented

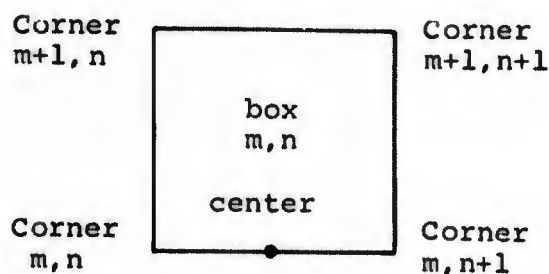


Figure 2-3. Sketch of Nodal Center Location For Ablating Material

*This scheme of course appears to be, and is, somewhat less accurate than more conventional schemes; more nodes must be taken to get accuracy comparable to that of "centered" schemes. The price paid here is compensated for by stability and smoothness of the solution at the surface.

by the following:

$$r_{N,m,n} = \frac{r_{c,m,n} + r_{c,m,n+1}}{2} \quad (1)$$

$$z_{N,m,n} = \frac{z_{c,m,n} + z_{c,m,n+1}}{2} \quad (2)$$

For back-up nodes it is most desirable and common to put the nodal center in the arithmetic center of the box:

$$r_{N,m,n} = \frac{r_{c,m,n} + r_{c,m+1,n} + r_{c,m+1,n+1} + r_{c,m,n+1}}{4} \quad (3)$$

$$z_{N,m,n} = \frac{z_{c,m,n} + z_{c,m+1,n} + z_{c,m+1,n+1} + z_{c,m,n+1}}{4} \quad (4)$$

(The ASTHMA user has the option to put back-up material nodal centers anywhere in the nodal box.)

2.2.2.2 Path Lengths for Conduction

In this and the following sections, the nodal center will have a general location $r_{N,m,n}$, $z_{N,m,n}$ for illustrative purposes. For computing thermal conductances, it will be necessary to have thermal path lengths between nodes. Since material properties are associated with each nodal box, first to consider path segments inside each box. In general there are four path segments of interest, as shown in Figure 2-4. The program computes the lengths $L_{m,n,B}$

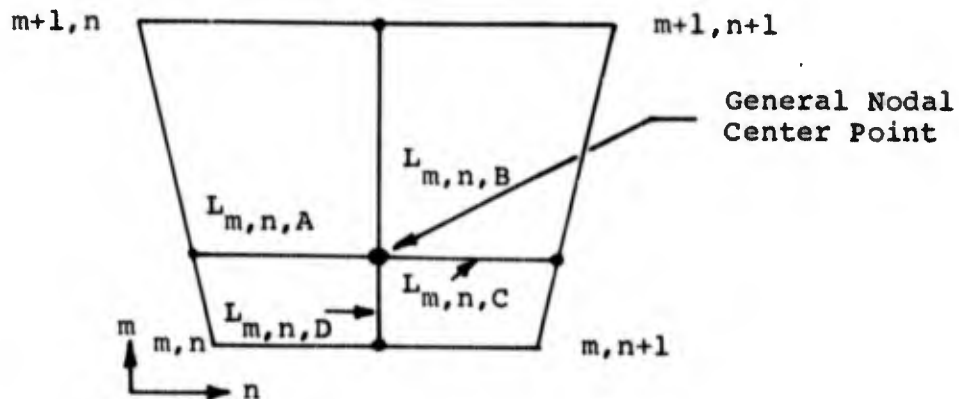


Figure 2-4. Illustration of Path Lengths

and $L_{m,n,D}$, in the m direction, as the distances between the nodal center and the centers of the $m+1$ and m faces of the box. (This is because the nodal center is usually on the line joining the centers of these two faces.) Thus the paths B and D have the lengths

$$L_{m,n,B} = \left[\left(\frac{r_{c,m+1,n} + r_{c,m+1,n+1}}{2} - r_{N,m,n} \right)^2 + \left(\frac{z_{c,m+1,n} + z_{c,m+1,n+1}}{2} - z_{N,m,n} \right)^2 \right]^{\frac{1}{2}} \quad (5)$$

$$L_{m,n,D} = \left[\left(\frac{r_{c,m,n+1} + r_{c,m,n}}{2} - r_{N,m,n} \right)^2 + \left(\frac{z_{c,m,n+1} + z_{c,m,n}}{2} - z_{N,m,n} \right)^2 \right]^{\frac{1}{2}} \quad (6)$$

Since as noted above the nodal center is usually on the m face, however, the paths between the nodal center and the n and $n+1$ faces (paths A and C in the sketch) are assumed to end at points on those faces located as far down the face, proportionally, as the nodal center is located in the box. The two usual cases are the back-shifted node center on the face m , for which the paths A and C end at the corners m,n and $m,n+1$ respectively, and the arithmetically centered node, for which the paths A and C end at the centers of faces m and $m+1$ respectively. For the first case (back-shifted), we have

$$L_{m,n,A} = \left[(r_{c,m,n} - r_{N,m,n})^2 + (z_{c,m,n} - z_{N,m,n})^2 \right]^{\frac{1}{2}} \quad (7)$$

$$L_{m,n,C} = \left[(r_{c,m,n+1} - r_{N,m,n})^2 + (z_{c,m,n+1} - z_{N,m,n})^2 \right]^{\frac{1}{2}} \quad (8)$$

and for the second (centered) we have

$$L_{m,n,A} = \left[\left(\frac{r_{c,m,n} + r_{c,m+1,n}}{2} - r_{N,m,n} \right)^2 + \left(\frac{z_{c,m,n} + z_{c,m+1,n}}{2} - z_{N,m,n} \right)^2 \right]^{\frac{1}{2}} \quad (9)$$

$$L_{m,n,C} = \left[\left(\frac{r_{c,m+1,n+1} + r_{c,m,n+1}}{2} - r_{N,m,n} \right)^2 + \left(\frac{z_{c,m+1,n+1} + z_{c,m,n+1}}{2} - z_{N,m,n} \right)^2 \right]^{\frac{1}{2}} \quad (10)$$

2.2.2.3 Side areas

Areas of the sides of each box are also required for thermal conductance calculations. For the sides lettered as shown in Figure 2-5, the areas are

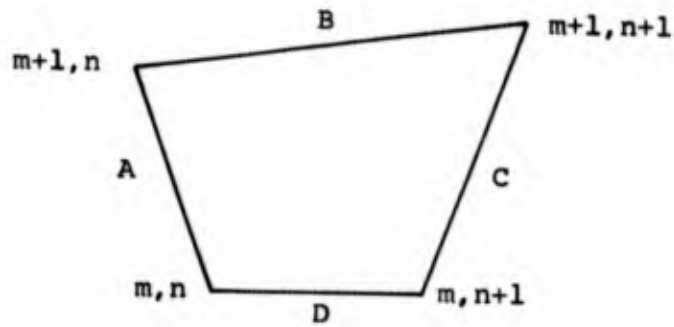


Figure 2-5. Nomenclature of Nodal Sides

given by elementary geometry (First Theorem of Pappus)

$$A_{m,n,A} = \pi(r_{c,m,n} + r_{c,m+1,n}) \left[(r_{c,m+1,n} - r_{c,m,n})^2 + (z_{c,m+1,n} - z_{c,m,n})^2 \right]^{\frac{1}{2}} \quad (11)$$

$$A_{m,n,B} = \pi(r_{c,m,n} + r_{c,m,n+1}) \left[(r_{c,m,n+1} - r_{c,m,n})^2 + (z_{c,m,n+1} - z_{c,m,n})^2 \right]^{\frac{1}{2}} \quad (12)$$

Only these two areas need to be computed for each box, since the areas on the other sides of the nodal box are identical with the areas A and B of adjacent nodes.

2.2.2.4 Volume

The volume of the elemental box is by the Second Theorem of Pappus,

$$\begin{aligned}
 V_{m,n} = & \left[\pi/3 \left\{ z_{c,m,n} \left[r_{c,m,n} (r_{c,m,n+1} - r_{c,m+1,n}) \right. \right. \right. \\
 & + r_{c,m,n+1}^2 - r_{c,m+1,n}^2 \left. \right] + z_{c,m+1,n} \left[r_{c,m+1,n} \cdot \right. \\
 & \left. (r_{c,m,n} - r_{c,m+1,n+1}) + r_{c,m,n}^2 - r_{c,m+1,n+1}^2 \right] \\
 & + z_{c,m+1,n+1} \left[r_{c,m+1,n+1} (r_{c,m+1,n} - r_{c,m,n+1}) \right. \\
 & + r_{c,m+1,n}^2 - r_{c,m,n+1}^2 \left. \right] + z_{c,m,n+1} \left[r_{c,m,n+1} \cdot \right. \\
 & \left. (r_{c,m+1,n+1} - r_{c,m,n}) + r_{c,m+1,n+1}^2 - r_{c,m,n}^2 \right] \left. \right\} \left. \right] \quad (13)
 \end{aligned}$$

2.2.2.5 Geometric Effects of Surface Recession

For nodes adjacent to the heated surface, side B may move due to surface recession (ablation). This recession reduces the thermal resistance between the surface point and the adjacent nodal point, increases transverse thermal resistances (as discussed below), and reduces the thermal capacity associated with the nodal center of the nodal box adjacent to the surface. The program assumes that the surface recession occurs so as to maintain the ratios

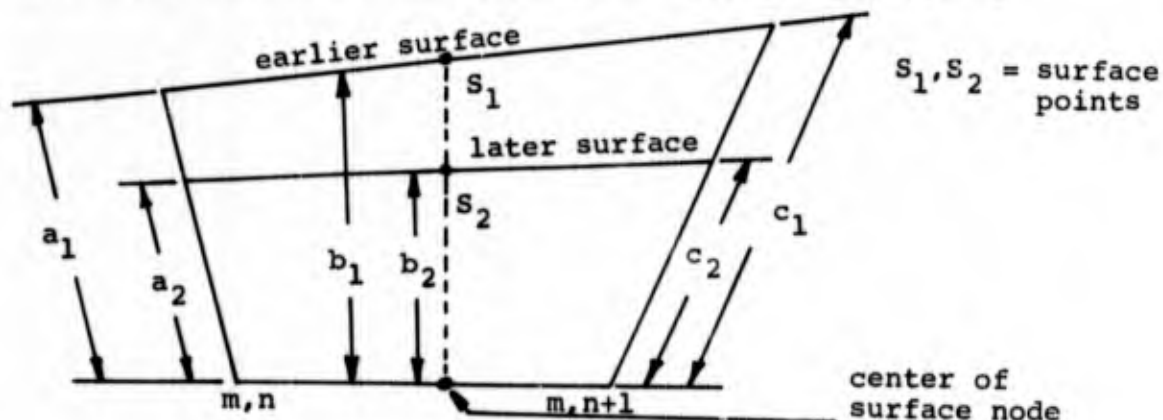


Figure 2-6. Sketch of Surface Nodal Box Undergoing Recession

$$\frac{a_2}{a_1} = \frac{b_2}{b_1} = \frac{c_2}{c_1} \quad (14)$$

as shown in Figure 2-6. The b line joins the centers of the m+1 and m planes and serves to define the location of the surface point S. The moving corners m+1,n and m+1, n+1 are located accordingly, and path lengths, areas and volumes computed as before, except that for these nodes $A_{m,n,A} \neq A_{m,n-1,c}$.

2.2.2.6 Surface Shape

As noted above and illustrated by Figure 2-7, it is most convenient to consider the surface points on the heated surface as being always located on the nodal column center line, that is, on the line joining the center points of the m+1 and m planes ("parallel" to the heated surface), where m is the row index of a nodal box at the surface and m+1 is the local corner index of the heated surface (see Fig. 2-6). The location of the surface points is all the information needed about the surface for those ablation problems with a specified, input surface recession rate as a boundary condition (the various ablation problem boundary condition options are described in Section 3 below), since for those problems, it is most convenient to specify, as input, recession history along the nodal center line. The nodal grid as input thus serves to define the various nodal column center lines, and the recession history for each column then defines the history of the surface points in a perfectly straightforward manner. However, another important heated surface boundary condition option involves not input recessions but various energy and chemistry information sufficient to calculate surface mass loss from energy balance considerations (Option 1, discussed in Section 3 below). This ablation calculation does not produce recession rates directly, of course; instead it produces rates of mass loss from the surface. It will prove convenient, nevertheless, to adhere to the concept of the surface point which moves along the column center line as recession progresses. To define the surface point motion with computed mass loss rates determined from this general energy-balance-determined option requires knowledge of the angle between the local normal to the surface and the column center line, since computed mass loss may be translated directly into recession along a normal to the surface. Recession along the normal may be projected into the nodal column center line once the angle between these two lines is known. The direction of the surface normal may conveniently be determined from the slope of the surface, that is the surface shape, and of course the direction of the column center line is known.

It is obviously not safe to use the slope of the heated (top) surface of the surface nodal box to obtain the surface slope, since in general it is neither possible nor always desirable to lay out a nodal grid which will "conform" to the "real" surface for the entire problem history. Therefore the ASTHMA program includes a special curve-fit subprogram which computes a

surface shape at each time step during one solution after examining the layout of the surface points. Referring to Figure 2-7, the program computes the

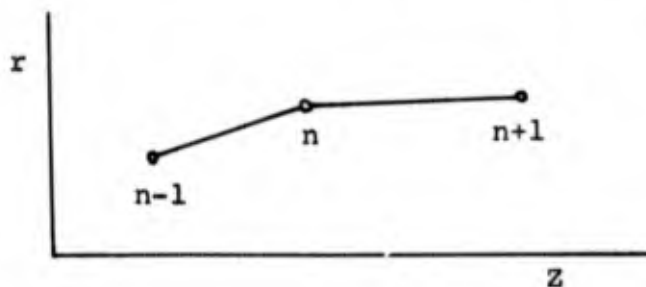


Figure 2-7. Sketch of Surface Points

slope of the surface, dr/dz , at surface point n as the average of the slopes between surface points $n-1$ and n , and n and $n+1$. Thus

$$\left. \frac{dr}{dz} \right|_n = \frac{1}{2} \left(\frac{r_n - r_{n-1}}{z_n - z_{n-1}} + \frac{r_{n+1} - r_n}{z_{n+1} - z_n} \right) \quad (15)$$

Note that $n-1$, n , and $n+1$ are surface points, not nodal corner points. (Problems with only one nodal column have only one surface point. This requires the program to abandon the average slope scheme and to use the nodal box heated surface slope as the surface slope.)

With surface slope determined, the surface movement ΔS computed during the time step may be projected onto the nodal box center line, and then the

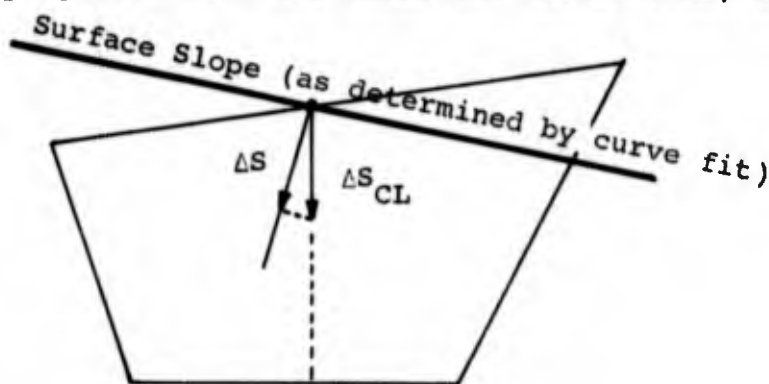


Figure 2-8. Sketch of Surface Geometrical Relationships (exaggerated)

new nodal volume computed, as indicated in the exaggerated sketch of Figure 2-8. Actual surface movement during a time step is limited to a small fraction of the nodal thickness to ensure a good approximation to the conservation of mass.

2.3 INTERNAL CONDUCTION PARAMETERS

The thermal resistances between nodes and the thermal capacity of the nodes are calculated each time interval from the material properties of the node corresponding to its temperature at that time. The material properties (density (ρ), specific heat (c), conductivity (k), and emissivity (ϵ) are input as table look up functions of temperature. Linear interpolation is employed for material property determination at temperatures intermediate to those tabulated. Constant thermal contact resistances may be specified between any or all nodes. The nodal capacities and resistances are calculated as follows:

$$C_{m,n,\theta} = \rho_{m,n,\theta} c_{m,n,\theta} V_{m,n} \quad (16)$$

$$R_{m,n,A,\theta} = \frac{1}{A_{m,n+1,A}} \left(\frac{L_{m,n,C}}{k_{m,n,\theta}} + \frac{L_{m,n+1,A}}{k_{m,n+1,\theta}} + R_{m,n,B}^* \right) \quad (17)$$

$$R_{m,n,B,\theta} = \frac{1}{A_{m+1,n,B}} \left(\frac{L_{m,n,B}}{k'_{m,n,\theta}} + \frac{L_{m+1,n,D}}{k'_{m+1,n,\theta}} + R_{m,n,A}^* \right) \quad (18)$$

Figure 2-9 shows the locations of resistances $R_{m,n,A,\theta}$ and $R_{m,n,B,\theta}$. The

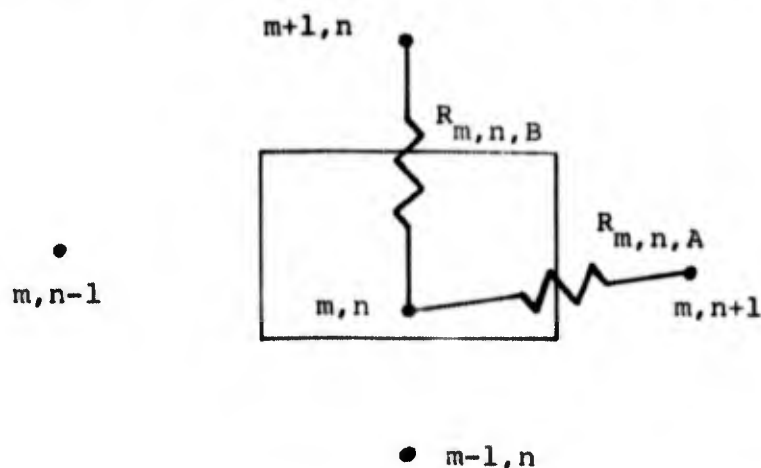


Figure 2-9. Sketch of Thermal Resistance Nomenclature

resistance on the other two sides of the nodal box are calculated when the quantities for the adjacent nodes are calculated. For nodes adjacent to the surface, however, $A_{m,n,C} \neq A_{m,n+1,A}$ generally (refer to Fig. 2-5 for area nomenclature); for these nodes

$$R_{m,n,A} = \frac{L_{m,n,C}}{A_{m,n,C} k_{m,n,\theta}} + \frac{L_{m,n+1,C}}{A_{m,n+1,A} k_{m,n+1,\theta}} + \frac{R_{m,n,B}}{A_{m,n,C}}$$

It should be noted that anisotropic thermal conductivity values k and k' are used in Equations (17) and (18), respectively.

2.4 IN-DEPTH CONDUCTION SOLUTION

The in-depth conduction solution is the explicit finite difference type often employed for transient heat conduction analysis. The temperature of node m, n at time θ' ($T_{m,n,\theta'}$) is obtained by application of the finite difference energy balance and rate equations to the nodal volume.

Solving for $T_{m,n,\theta'}$, one obtains:

$$T_{m,n,\theta'} = \left[\frac{T_{m+1,n,\theta}}{R_{m,n,B,\theta}} + \frac{T_{m,n+1,\theta}}{R_{m,n,A,\theta}} + \frac{T_{m-1,n,\theta}}{R_{m-1,n,B,\theta}} + \frac{T_{m,n-1,\theta}}{R_{m,n-1,A,\theta}} - T_{m,n,\theta} \right] \frac{\Delta\theta}{C_{m,n,\theta}} + T_{m,n,\theta} \quad (19)$$

In the program this equation is used to obtain "new" temperatures for all nodes except those adjacent to the heated surface and for back wall nodes. Nodes adjacent to the heated surface are linked to the surface temperature implicitly and hence a special procedure is used for the temperature of these nodes, as described in the next section.

Back wall nodes include a quantity

$$hA_{m,n,D}(T_{res} - T_{m,n,\theta}) + \sigma\epsilon_{bw}A_{m,n,D}(T_{m,n,\theta}^4 - T_{res}^4)$$

inside the brackets of Equation (19), and have $T_{m-1,n,\theta}$ formally equal to zero.

The explicit relation (19) imposes the familiar stability restriction on the time step size $\Delta\theta = \theta' - \theta$. The ASTHMA program automatically employs a conservative stability equation for interior nodes:

$$\Delta\theta = \eta \left[\frac{C_{m,n}}{\frac{1}{R_{m,n,A}} + \frac{1}{R_{m,n,B}} + \frac{1}{R_{m,n-1,A}} + \frac{1}{R_{m-1,n,B}}} \right] \quad (20)$$

Normally for stability, the input parameter η is less than unity. Surface nodes are not considered for time interval calculation, as will be explained below; back wall nodes include the terms $A_{m,n,D}(h/2 + 4\sigma\epsilon_{bw}T_{m,n,\theta}^3)$ in the denominator.

The automatic stability criterion calculation may be suppressed for any node if the user is sure that the allowed time step for that node will never be the minimum one for the system. This saves some computer time.

Alternatively, the stability criterion calculation can be suppressed entirely. If it is used, then $\Delta\theta$ must be specified. If it is used, then any or all nodes in the conduction network may be specified for time interval determination.

2.5 TEMPERATURES OF SURFACE NODES AND SURFACE POINTS

Temperatures of surface points are determined either by assignment (Option 2) or by the surface energy balance described in the next section (Options 1 and 3). The surface energy balance determines the new surface temperature of the n th column T'_{s_n} with an implicit iteration technique. Stability considerations dictate that the first node temperature also be treated implicitly, and that any transverse heat conduction link (across columns) for surface temperatures must be implicit. This latter requirement is a complex one to meet as columns recede; hence the surface temperature points are not linked transversely. Figure 2-10 shows the implicit and explicit heat conduction paths for two typical columns.

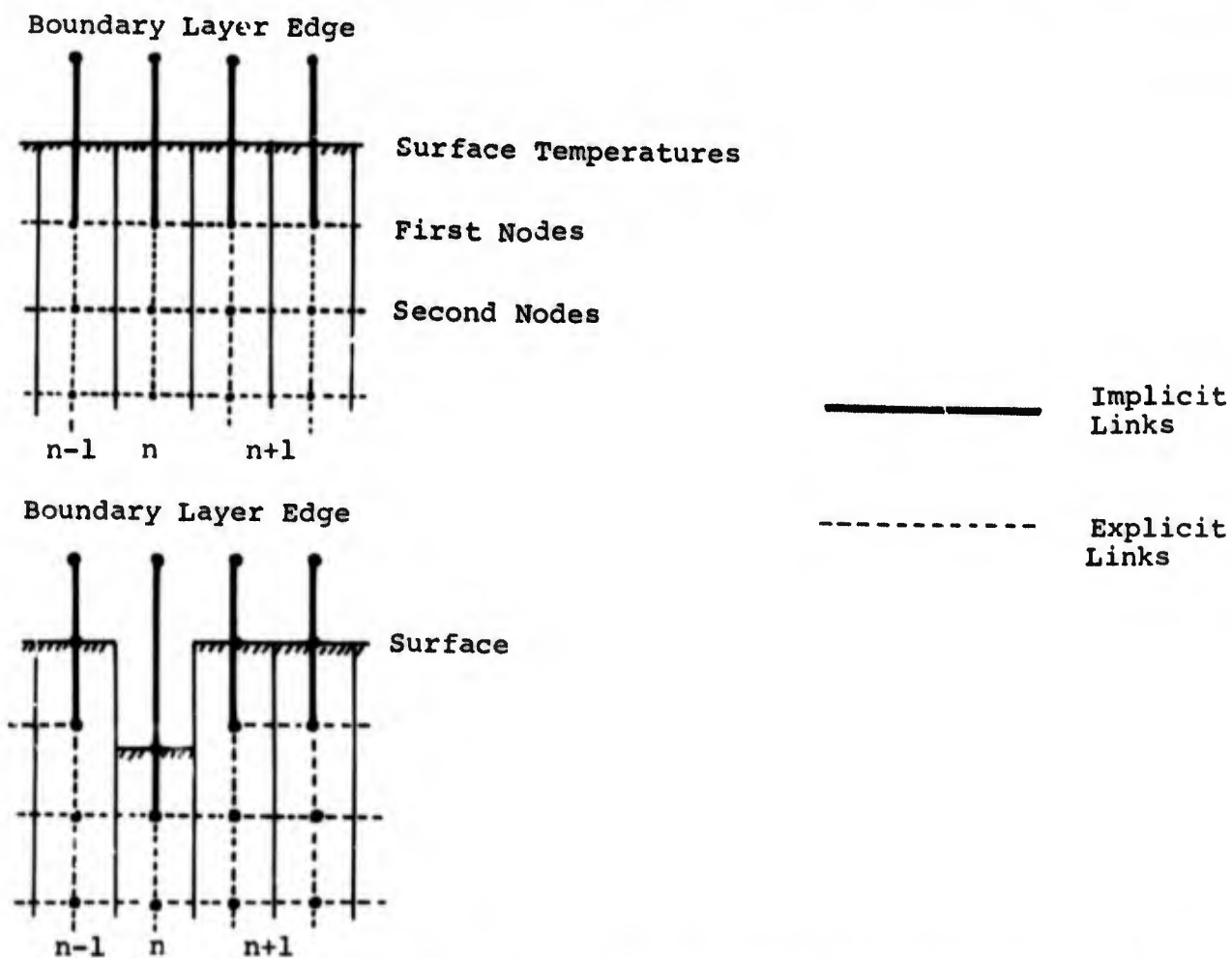


Figure 2-10. Sketch of Implicit and Explicit Temperature Links in Finite Difference Solution, for Two Typical Situations

In general terms, the equation for the new temperature of a surface node (or surface point) is

$$T_{m,n,\theta'} = \left[\frac{T_{m,n+1,\theta} - T_{m,n,\theta}}{R_{m,n,A,\theta}} + \frac{T_{m,n-1,\theta} - T_{m,n,\theta}}{R_{m,n-1,A,\theta}} + \frac{T_{m-1,n,\theta} - T_{m,n,\theta}}{R_{m-1,n,B,\theta}} + \frac{T_{S,n,\theta'} - T_{m,n,\theta'}}{R_{m,n,B,\theta}} \right] \frac{\Delta\theta}{C_{m,n,\theta}} + T_{m,n,\theta} \quad (21)$$

This is linked to the surface energy balance through the surface point temperature $T_{S,n,\theta'}$. Formally we have a relation between the two unknowns as

$$T_{m,n,\theta'} = f(T_{S,n,\theta'}) \quad (22)$$

The surface energy balance has the general form (as described in the next section)

$$\begin{aligned} & \text{convection and chemical energy terms } (T_{S,n,\theta'}) \\ & + \text{radiation to wall - radiation out } (T_{S,n,\theta'}) \\ & = \frac{T_{S,n,\theta'} - T_{m,n,\theta'}}{R_{m,n,B,\theta}} = f_2(T_{m,n,\theta'}, T_{S,n,\theta'}) \end{aligned} \quad (23)$$

where the parentheses denote functional relationship. Relation (22) may be substituted into Equation (23) to give the non-linear surface energy balance equation for $T_{S,n,\theta'}$. When $T_{S,n,\theta'}$ has been found from this equation, the surface node temperature $T_{m,n,\theta'}$ may be found from Equation (22). For the energy balance options, the method for finding $T_{S,n,\theta'}$ is described in section 3 below. In Option 2, $T_{S,n,\theta'}$ is known immediately and Equation (23) then determines $T_{m,n,\theta'}$ at once.

SECTION 3

HEATED SURFACE BOUNDARY CONDITION ASPECTS

The ASTHMA program can account for three different types of heated surface boundary conditions:

1. Surface energy balance determined recession rate, employing general film coefficient model for boundary layer transport of energy and mass, and very general thermochemical relations at the surface (Option 1).
2. Specified surface temperature and recession rate histories (with a different history at each surface point of interest) (Option 2).
3. Simplified energy balance to determine surface temperature including only radiation terms, no recession allowed (Option 3 - "cooldown option")

Option 2 is so simple that it requires no discussion other than that given in the user's manual below. Option 1 is complex and has many interesting aspects, some of which are discussed below. A brief discussion of Option 3 follows.

3.1 GENERAL ASPECTS

The energy balance option of Surface Energy Balance Option (Option 1) of the program performs the energy balance illustrated in the sketch of Figure 3-1. The energy balance is performed for the indicated control volume fixed

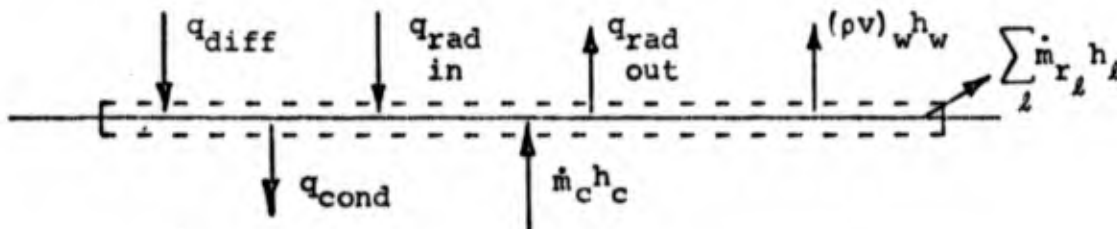


Figure 3-1. Representation of Surface Energy Terms During Ablation

to the receding surface. Energy fluxes leaving the control volume include conduction into the material, radiation away from the surface, energy in any

flow of condensed phase material such as mechanical removal,* and gross blowing at the surface. Energy inputs to the control volume include radiation in from the boundary layer and enthalpy flux due to the convection of material to the surface associated with surface recession. The final input in the sketch is denoted q_{diff} . It includes all diffusive energy fluxes from the gas phase boundary layer and the form of a correlation equation associated with the convective film coefficient model.

The computation of the surface energy balance requires from the in-depth solution a relation between the surface temperature and the rate of energy conduction into the material, q_{cond} . This relation derives naturally from the finite difference energy balance for the node just under the surface. With this information the surface energy balance considerations allow determination of the thermochemical erosion rate \dot{m} and surface temperature T_w . It will be useful to keep in mind that, from this point of view, the purpose of the in-depth solution at any instant is to provide information about $q_{cond}(T_w)$. The surface energy balance equation may be written as

$$q_{diff} + q_{rad} + \dot{m}_c h_c - q_{rad} - (\rho v)_w h_w - \sum_l \dot{m}_{r_l} h_l - q_{cond} = 0 \quad (24)$$

where

$$(\rho v)_w = \dot{m}_c - \sum_l \dot{m}_{r_l} \quad (25)$$

The relation $q_{cond} = f(T_w)$ is delivered by the in-depth solution. Other dependencies of interest are

$$h_c = h_c(T_w) \quad (26)$$

$$q_{rad} = q_{rad}(T_w) \quad (27)$$

*The energy balance procedure has indeed been constructed to include mechanical terms but in the first version of the program failing is not allowed (in order to exploit input conveniences) since it is not usually important for rocket nozzles.

For the other terms, we may write in general

$$T_w, q_{diff}, q_{rad, in}, h_w, \sum_l \dot{m}_{r_l} h_l = \text{functions of boundary-layer-edge enthalpy, pressure, upstream events, laws for conservation of chemical elements, chemical equilibria and/or kinetic relations, } \dot{m}_c, \text{ local boundary layer aerodynamics} \quad (28)$$

Relations of the type of Equation (28) come in many forms, and may even take the form of exact solutions to boundary layer and surface chemical state routines. In the present program, the relations (28) are computed by the separate chemistry program (ACE), for use by Equation (24) in ASTHMA, based on a film coefficient model of the boundary layer transport events. The ACE program is fully described in Reference 9. Since relations (28) are obtained by a film coefficient based program (ACE), the main energy balance (24) used by the ASTHMA program is also of a film coefficient type.

When the boundary layer transport aspects of the problem are modeled by a film coefficient scheme, then both Equations (24) and (28) can be normalized in the mass transfer coefficient in the customary manner. Equation (24) becomes

$$\frac{q_{diff}}{\rho_e u_e C_M} + \frac{q_{rad, in} - q_{rad, out}}{\rho_e u_e C_M} + B'_c h_c - B' h_w - \sum_l \frac{\dot{m}_{r_l}}{\rho_e u_e C_M} h_l - \frac{q_{cond}}{\rho_e u_e C_M} = 0 \quad (29)$$

and Equation (28) takes the general form

$$T_w, \frac{q_{diff}}{\rho_e u_e C_M}, \frac{q_{rad, in}}{\rho_e u_e C_M}, h_w, \sum_l \frac{\dot{m}_{r_l}}{\rho_e u_e C_M} h_l = \text{functions of boundary layer edge enthalpy, pressure, laws for conservation of chemical elements, chemical equilibrium and/or kinetic relations, } B'_c \quad (30)$$

The generation of the Equation (30) relationships is the goal of the surface thermochemistry solution (ACF) as noted above. For the present it may be observed that the energy balance coupling to the in-depth solution (ASTHMA) for each time step proceeds as follows: an initial guess of the dimensionless mass removal rate B'_c is obtained in some manner. With this B'_c , the quantities $q_{diff}/\rho_e u_e C_M$, h_w , $\sum_l \dot{m}_{r_l} h_l / \rho_e u_e C_M$, and T_w are obtained

from the surface thermochemistry solution. The quantities h_c and q_{rad} out are then formulated using the T_w so obtained. The surface energy balance is then computed, the q_{cond} as a function of T_w having been provided by the in-depth solution. In general, however, the sum of the terms will not equal zero but some error. An iteration procedure is then used to select successively better estimates of B'_c which drive the error to zero. Experience shows that Newton's procedure, in which the derivative of the error with respect to B'_c is used to compute the next guess for B'_c , gives good results.

3.2 COMPUTATIONAL APPROACH TO CONVECTIVE ENERGY BALANCE

3.2.1 General Description of Approach

It is evident that each iteration in the search for a surface energy balance, if performed as described above, would require a new surface chemistry solution, generally in the near neighborhood of many such previous solutions. This suggests that a tabular approach in which the surface state solutions are done beforehand for preassigned B'_c values, would offer significant computational economies. The Aerotherm CMA program (a one-dimensional, transient, charring material ablation program) has long exploited this approach with good success (Ref. 5). Since the approach used in the ASTHMA program surface energy calculations has been built on the basic CMA model, it will be useful to first describe the CMA procedures in detail and then to explain the slight changes incorporated in the ASTHMA version. For the CMA program, a complete three-dimensional table is precalculated with an entirely separate computer program* for each ablation problem. The three dimensions are $\dot{m}_c / \rho_e u_e C_M$, $\dot{m}_g / \rho_e u_e C_M = B'_g$, and another variable which effectively is the boundary layer edge state identifier, most conveniently pressure. For the problem of interest here, there is no pyrolysis gas rate \dot{m}_g , and the parameter B'_g instead takes on the significance of a chemical kinetics parameter k'_f , as discussed in Section 3.2.2 below. The CMA description in the following few paragraphs will continue to refer to B'_g , but there is no difference in treatment between B'_g in the CMA program and k'_f in the ASTHMA program. The physical meaning of k'_f is discussed below, in Section 3.2.2. Pressure is necessary for the surface state calculation in all cases. Specification of the boundary layer edge state implies pressure plus, say, edge enthalpy, but since only one physical location on the ablating body is involved in the one-dimensional CMA solution, the boundary layer edge state is really only a function of one variable, time. Pressure can be used instead, for convenience.

The course of a typical CMA solution might be represented as shown in Figure 3-2. As time proceeds, solutions progress through the space of the

*Applicable programs here included the Equilibrium Surface Thermochemistry Program, Version 1 (Ref. 6), and Version 2 (Ref. 8), and the ACE program (Ref. 9).

table independent variables B'_c , B'_g , and P . The dots in the picture represent solutions satisfying the surface energy balance as time progresses. Surrounding these points are a number of other points examined during the iterations to satisfy the surface energy balance. For any iteration, the solution procedure finds itself, so to speak, within a cube formed by the bracketing tabular values of B'_c , B'_g , and P . The dependent quantities T_w , $q_{diff}/\rho_e u_e C_M$, h_w and $\sum \dot{m}_r h_r / \rho_e u_e C_M$ have been precalculated for these tabular points; relevant values of these quantities for the current iteration values of B'_c , B'_g , and P can then be formed by interpolation inside the cube and the surface energy balance equation calculated. If the energy balance is not satisfied to some preselected degree of accuracy, a new value of B'_c can be selected and the process repeated.

The tabular approach was selected in general because the interpolation feature drastically reduces the number of surface state calculations while at the same time allowing sufficient accuracy.* The precalculated table approach** was chosen for the following reasons:

1. In parametric studies, tables once generated are useable for many different problems, yielding even greater economy.
2. For most problems, it is difficult to specify a priori an "adequate" array of independent tabular values B'_c , B'_g , and P . An examination of the precalculated surface tables before execution of the actual ablation-problem-plus-in-depth-solution can reveal if there are any "holes" in the tables in areas where energy terms are varying rapidly. Desirable table points can be added before the in-depth response run.
3. The surface tables are frequently of independent interest in themselves for judging the ablation effects under various conditions.
4. Finally, without the tabular approach, the occasional nonconvergent surface chemistry solution would stop the entire in-depth solution process. With the precalculated table approach, such solutions are automatically weeded out of the tables without damage to the subsequent in-depth solution.

* For example, a typical 2000 time step problem with the usual 5 iterations per time step would require 10,000 surface state calculations, which require 1 to 3 seconds each of machines in the 7094 speed class. A single pressure table, on the other hand, involves only about 600 state solutions (30 char rates times 20 gas rates), a factor of 15 improvement. Multiple pressure tables reduce this advantage, but usually only a few pressure are required.

** As opposed to a direct coupling scheme in which tabular dependent entries were only calculated as needed.

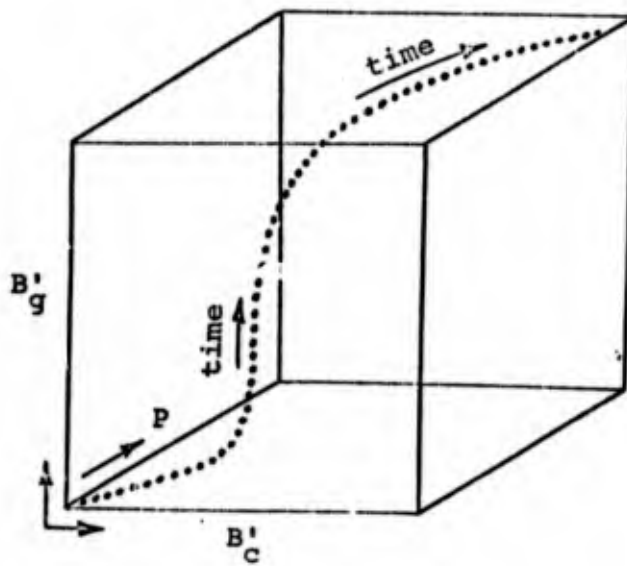


Figure 3-2. Representation of Course of Independent Variables in Surface History

On the other side of the ledger, some disadvantages in the precalculated table approach are evident however:

1. Figure 3-2, which is a realistic schematic view of a typical calculation history, indicates that most of the laboriously calculated and assembled surface state points in the table are never used in the course of a given solution.
2. The "mechanical" linkage between the surface state solution and the in-depth solution, i.e., the transfer of punched card surface state output to input of the in-depth program, leads to computing delays and occasionally to gross input blunders (wrong decks, missing parts of decks, etc.).
3. Significant storage is required by the large precalculated table.

These disadvantages are usually outweighed by the advantages of the precalculated table approach, however. Hence this same well tested approach has been adopted for the ASTHMA program. The char ablation rate parameter B'_c and the pressure (or edge state) parameter P retain their original (CMA) significance, but the gas rate parameter B'_g is dropped and is replaced by a kinetics parameter $k_f/\rho_e u_e C_M = k'_f$. This substitution, however, has brought in a number of new aspects discussed in the following subsection. Further information on this topic of kinetics can be found in Reference 9.

3.2.2 Treatment of Kinetics Parameters

The new surface state program ACE (Ref. 9) differs from earlier Aerotherm surface state programs by allowing certain chemical reactions to be rate controlled, rather than requiring complete equilibrium. In consequence, the ASTHMA program, which uses as input the output of the ACE program, is able to account for chemical kinetics. It does this through a "kinetics parameter" of the form $k_f/\rho_e u_e C_M$ which replaces the pyrolysis gas rate parameter B'_g of the charring material program noted above. This kinetics parameter is explained in detail in Reference 9; it is a measure of the relative importance of chemical surface kinetics for those reactions considered by the state program to be kinetically controlled.

The program user inputs to the ASTHMA program the forward rate coefficient of one particular kinetically controlled reaction used by the chemistry state solution program (ACE) in its solution. The chemistry program will identify this reaction both on its printed output and on certain cards in the punched output.

This odd and somewhat inconvenient kinetics link between the surface state program (ACE) and the in-depth program (ASTHMA) derives from the generality of the ACE program. Reference 9 explains how, in the mass balances

and chemistry relations used in ACE, kinetic rate equations always appear divided by $\rho_e u_e C_M$. Thus, since any forward rate coefficients used in ACE are used in the form $k_f/\rho_e u_e C_M$, the ACE user inputs parametric array values of $k_f/\rho_e u_e C_M$ for all kinetically controlled reactions. The ASTHMA user, however, is forced by the nature of his problem to deal with both the set of k_f 's and $\rho_e u_e C_M$ as separate entities. Therefore he must specify both $\rho_e u_e C_M$ and the specific values of the k_f 's. Regardless of the number of k_f 's used in the ACE solution (i.e., the number of kinetically controlled reactions), after the surface state solution is completed the relationship between these coefficients is fixed and cannot be changed. Thus the ASTHMA user only needs to communicate with ACE through a single "kinetics number," and hence the ACE program punches as its output card one specific value $k_f/\rho_e u_e C_M$ rather than the complete set of values representing all kinetically controlled reactions. The ASTHMA user inputs to ASTHMA values for $\rho_e u_e C_M$ and whatever value of k_f he wishes for the single "flag" reaction used by the ACE program in its output. Unfortunately, a single output card for each surface state point (fixed B'_c , k'_f , and P) has insufficient space even to identify the flag reaction, let alone to identify the complete set of kinetically controlled reactions employed. Therefore the ASTHMA user will have to exercise great caution to ensure that all cards in his surface state "deck" (see Section 1.8 in the User's Manual below) were constructed with the same set of kinetically controlled reactions, the same set of values of kinetic coefficients, and have the same "flag" reaction identification.

3.2.3 Change in Treatment of Pressure

One further change was made from the CMA pattern of solution. Since the ASTHMA program was designed primarily for rocket nozzle applications where the range of pressures is small, the interpolation in the pressure direction was dropped. Instead, each surface column is flagged to one specific pressure (or edge state) in the set of tables. No interpolation is done.

3.3 FORMS OF THE FILM COEFFICIENT MODEL SURFACE ENERGY BALANCE EQUATION USED IN OPTION ..

In making the surface energy balance calculation described in Section 3.2 above, the ASTHMA program uses a variety of surface energy balance equations of the general form of Equation (29). The basic equation, derived in Reference 1 with analogy arguments, is

$$\rho_e u_e C_H (H_r - h_w)_{\text{gas}} + \rho_e u_e C_M \left[\sum_i (K_{i_e} - K_{i_w}) h_i^{T_w} + B'_c h_c + B'_g h_g - B'_w h_w \right] + \alpha_w q_{\text{rad}} - F \sigma \epsilon T_w^4 - q_{\text{cond}} = 0 \quad (31)$$

It should be noted that, like all film coefficient expressions, this equation is not a universally valid one. It is well established for frozen boundary layer, catalytic wall problems with no net mass transfer (Refs. 11, 12). It is presumed accurate for reactive boundary layers as well (see a discussion of similar considerations in Ref. 13). For problems with net mass transfer (ablation), which are of interest here, the equation is less well established, but

1. Has been derived with "respectable" analogy arguments (Ref. 1)
2. Compares well with similar results derived by Lees (Ref. 12)
3. Has given good predictions when compared to some experimental data and to results of "exact" boundary layer solutions.
4. Gives results independent of the choice of enthalpy datum state (an important criterion)
5. Reduces, for $C_M = C_H$, to a widely accepted energy balance law (Refs. 14, 15) for simultaneous heat and mass transfer

Equation (31) has been generalized in Reference 2 to the case of unequal mass diffusion coefficients, again by analogy arguments. The result is

$$\rho_e u_e C_H (H_r - h_w)_{\text{edge gas}} + \rho_e u_e C_M \left[\sum_i (z_{i_e}^* - z_{i_w}^*) h_i^{T_w} + B_c' h_c + B_g' h_g - B' h_w \right] + \alpha_w q_{\text{rad}} - F \sigma \epsilon T_w^4 - q_{\text{cond}} = 0 \quad (32)$$

where

$$z_i^* \triangleq \frac{A_i^Y K_i^{1-Y}}{\sum_j z_j^Y k_j^{1-Y}} \quad (33)$$

$$z_i \triangleq \frac{m_i x_i}{F_i \mu_a} = \frac{m K_j}{F_i \mu_i} \quad \text{since } m_i x_i = m K_i \quad (34)$$

and

$$\mu_a \triangleq \sum_i \frac{m_i x_i}{F_i} = m \sum_i \frac{K_i}{F_i} \quad (35)$$

The factors F_i in these equations derive from the particular relation between the binary diffusion coefficients which must hold if the governing differential equations are to reduce to the forms from which Relation (32) can be inferred. This relation is

$$D_{ij} = \frac{\bar{D}}{F_i F_j} \quad (36)$$

and can be regarded as an accurate correlation of experimental data for the binary diffusion coefficient D_{ij} . The quantity \bar{D} is a constant for a given pressure. The constants F_i depend weakly on temperature.

Equation (32) is somewhat less well founded than the corresponding Equation (31) but does have plausibility. Furthermore, the equation reduces to Equation (31) for equal diffusion coefficients, as it should, and Equation (32) can be shown to be independent of the enthalpy datum state and thus fulfills a basic physical requirement.

The ACE program provides to the ASTHMA program the four dependent quantities $h_{\text{wedge gas}}(T_w, p)$, $T_w(B'_c, p)$, $\sum_i Z_i h_i^{T_w}(B'_c, p)$ and $\sum_i Z_i^* h_i^{T_w}(B'_c, p)$ for a given value of k'_f . With the quantity $h_{\text{wedge gas}}(T_w, p)$ stored in one table with independent variables T_w and p , and the other three dependent quantities stored in a table with independent variables B'_c , k'_f , and p , the ASTHMA program may find surface energy balances at any surface point and time (i.e., given pressure and $\rho_e u_e C_M$) by (1) selecting the correct p table, (2) finding $k'_f = k_f / \rho_e u_e C_M$, (3) finding the dependent quantities in the two tables at k'_f values bracketing the actual k'_f and for some B'_c , (4) interpolating on k'_f , (5) forming the energy balance equation (32) and noting the error or departure from zero, (6) selecting a new B'_c for another try. The Newton-Raphson method is used for selecting the next guess of B'_c in step (6). When that B'_c yielding an energy balance is determined, the new T_w is known as a dependent quantity; thus the new surface temperature is determined.

3.4 Cooldown Energy Balance Option

For representing radiation controlled situations, such as cooldown after exposure to convective heating, it is useful to have a surface energy balance option which has only radiation terms. Such an option is built into the ASTHMA program and is termed "Option 3". The energy balance for this case is simply

$$\alpha_w q_{\text{rad}} - F \sigma \epsilon T_w^4 - q_{\text{cond}} = 0 \quad (37)$$

3.5 INPUT AND CORRECTION OF HEAT TRANSFER COEFFICIENT

To employ the film coefficient formulation just described, the program user must provide the program with values of the heat transfer coefficient $\rho_e u_e C_H$ as functions of time. Two practical problems must be settled in this respect:

1. How is C_M related to C_H ?
2. Can both C_M and C_H be specified as functions of edge conditions (i.e., of time) independent of the subsequent problem solution (i.e., mass transfer rates and body shape)?

In answer to the first question it may be stated that within the present formulation it is adequate to take the ratio C_M/C_H as a constant. The value of this constant is a measure of the ratio of the mean mass transfer aspects of the boundary layer to the mean heat transfer aspects. For equal mass diffusion coefficients, a vast amount of experimental data (as summarized, for example, in Ref. 15) suggest the correlation $C_M/C_H = Le^Y$. It may be hypothesized that for unequal mass diffusion coefficients the same procedure may be employed with the Lewis Number, Le , defined by the procedure set forth in Reference 2 involving \bar{D} . Thus, the input to the program consists of a time table of values for $\rho_e u_e C_H$ and the constant factor C_M/C_H .

In answer to question 2., changes of C_H with body shape are not usually of interest for rocket nozzle problems and hence are not accounted for in the ASTHMA program. A more important problem concerns the dependence of C_H on the actual rate of mass transfer. This problem has been ignored up to now, the implication being that C_H is determined by the boundary layer edge aerodynamics alone. This is known to be incorrect. The value of C_H depends fairly strongly on \dot{m} . If we denote the C_H with $\dot{m} = 0$ as C_{H0} , this dependence is shown by both data and analysis to be accurately represented by

$$\frac{C_H}{C_{H0}} = \frac{\varphi}{e^\varphi - 1} \quad (38)$$

where

$$\varphi \triangleq \frac{2\lambda\dot{m}}{\rho_e u_e C_{H0}} \quad (39)$$

This correction is built into the program.

PART II

USER'S MANUAL
ASTHMA PROGRAM

This section of the report summarizes the input requirements of the program in a form convenient for user of the program and provides the instructions for program operation.

SECTION 1

DESCRIPTION OF INPUT

The input to the Axi-Symmetric Transient Heating and Material Ablation program can conveniently be considered as having eight parts. All eight portions of the deck must be present for each run (with the exception of the steady-state equilibrium data in the case of surface-boundary conditions Options 2, Option 3 in conjunction with Option 3). These individual parts are described in the following subsections.

1. TITLE AND HEADING INFORMATION

The first three cards of the data deck are used to transmit title and heading information to the output. The first 72 columns of each of these cards may be used for the title, the alphanumeric information in columns 61 through 72 of the third card being used as a page heading on all pages after the first.

1. GENERAL PROBLEM CONSTANTS

The first card provides the program with certain general constants.

Code	Format	Data	Units
1	I3	Largest m index in nodal network, number of rows of boxes (not corners)	---
4	I3	Largest n index in nodal network, number of columns of boxes (not corners)	---
7	E6.4	Initial value of problem time	sec
13	E6.4	Final value of problem time	sec
19	E6.4	Output time interval	sec
25	E6.4	Computation time increment. If entered, this interval is always used and stability considerations are suppressed. If blank, stability limited time step will be calculated by program	sec

Column	Format	Data	Units
31-36	E6.4	Safety factor multiplying stability time step limit, see Equation (20)	---
37-42	E6.4	Heat of formation of ablating material at 536°R (298°K)	Btu/lb
43-48	E6.4	Blowing reduction parameter λ of Equation (35); this value may alternatively be inserted as a function of time in the time tables (Section 1.7 below); blank values in time table will be filled with this constant value*	---
49-54	E6.4	Back wall heat transfer coefficient	Btu/ft-sec°R
55-60	E6.4	Back wall emittance	---
61-66	E6.4	Reservoir temperature communicating with back wall	°R
67	I1	KASE, flag for stacked problems; +1 denotes another case to follow, 0 denotes last problem	---
68	I1	KSTRP, calls for punched nodal output compatible with thermal stress programs; 1 calls for punched output at all print times, 2 calls for punched output at special times given on following card	

1.3 SPECIAL PUNCHED TEMPERATURE OUTPUT SPECIFICATION (OPTIONAL)

If KSTRP on the previous card is 2, up to eight special output times are to be provided for the punched output on a single card. See output description for further information.

Column	Format	Data	Units
1-10	8F10.0	Up to eight output times for punched output data	sec
11-20			
21-30...			

1.4 NODAL NET CORNER COORDINATE CARDS

This set of cards provides the nodal box corner coordinates and certain nodal center locations to the program. For m rows and n columns, there will be $(m+1)(n+1)$ corner coordinate cards. It is advisable to run the corner coordinate cards through any convenient plot program to check the nodal layout before any heat transfer calculations are made.

* Recommended values are 0.5 for laminar flow, 0.4 for turbulent.

Corner cards must be entered in order beginning at $m=1, n=1$ and continuing up the first column of corners ($n=1$) for all m 's until column is completed, returning to bottom of next column $m=1, n=2$, and so on.

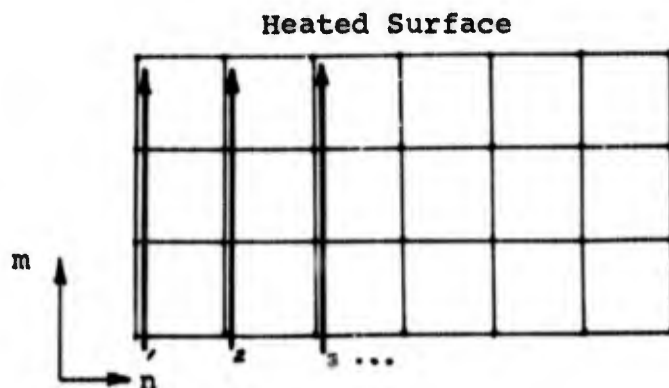


Figure 1-1. Sketch of Nodal Corner Input Order

Corners m, n representing boxes m, n which have no material (null nodes entered to complete the mesh lattice) may be blank entries, but a card must always be included. There must be $(m+1)(n+1)$ corner cards.

Nodal center specifications must be blank for the main material. The program will automatically locate these nodal points. Nodal centers for back up materials may be entered or may be left blank. If blank, the program will compute a centered nodal location for these nodes.

Present dimensions allow 300 nodal boxes, with a maximum of 40 columns.

Column	Format	Data	Units
1-7	E7.5	Corner radius, r_c	inches
8-14	E7.5	Corner axial location, Z_c	inches
15-21	E7.5	Nodal center radius coordinate r_N , must be blank for main ablating material, may be blank for back-ups	inches
22-28	E7.5	Nodal center axial coordinate Z_N , must be blank for main ablating material, may be blank for back-ups	inches

1.5 NODAL DATA CARDS

This set of cards identifies the materials in the nodal boxes, flags the time tables and surface thermochemistry tables to be associated with the nodes, and specifies radiation view factors. For an m by n mesh network mn cards must appear, blank cards being entered for null (no material) nodes. Cards begin at 1,1 and proceed to $m,1$, followed by 1,2 to $m,2$, and so on.

Column	Format	Data	Units
1	I1	Material number of material in this box, refers to numbered material properties tables. Blank for null node	---
2	I1	Enter 1 if used for stability limit calculation of time step. Leave blank if this node can safely be omitted from stability considerations (economy measure)	---
3	I1	Enter 1 if this node is at the heated surface, otherwise blank	---
4	I1	Enter number of surface thermochemistry table which is to be used for energy balance calculations on this node if it is or becomes a surface node (see Section 1.8.3 below)	---
5-6	I2	Enter number of time-function table which is to be used for boundary conditions for this node if it is or becomes a surface node (see Section 1.7 below)	---
7-12	E6.4	Leave blank (reserved for future use)	---
13-18	F6.4	Initial temperature of this node	$^{\circ}\text{R}$
19-24	E6.4	Interface (contact) resistance at top of node (m+1 plane)	$\text{ft}^2 \text{sec}^{\circ}\text{R}/\text{Btu}$
25-30	E6.4	Interface (contact) resistance at right side of node (n+1 plane)	$\text{ft}^2 \text{sec}^{\circ}\text{R}/\text{Btu}$
31-36	E6.4	Option 1 radiation view factor for this node if it is or becomes a surface node	---
37-42	E6.4	Option 3 radiation view factor for this node if it is or becomes a surface node	---

1.6 MATERIALS PROPERTIES TABLES

One table of temperature dependent material properties is input for each material appearing in the nodal network. A lead card gives the total number of materials used. The tables are not numbered and are presumed to be encountered in ascending numerical order, beginning at one and including no gaps or omitted table numbers. The main ablating material must have material number and table number one. There must be at least two temperature entries in each table. Present dimensions allow six material property tables, with 15 temperature entries in each table.

First Card

Column	Format	Data	Units
1-2	I2	Total number of materials	---

Tabular Entry Cards

Column	Format	Data	Units
1-2	I2	Flag. Last entry in table has -1; blank for other cards	---
3-8	E6.4	Temperature	$^{\circ}\text{R}$
9-14	E6.4	Density. Enter in first card only; will not be a function of temperature	lb/ft^3
15-20	E6.4	Specific heat C_p	$\text{Btu}/\text{lb}^{\circ}\text{R}$
21-26	E6.4	Thermal conductivity along rows (n-direction)	$\text{Btu}/\text{ft}\text{-sec}^{\circ}\text{R}$
27-32	E6.4	Emissivity (emittance)	---
33-38	E6.4	Thermal conductivity along columns (m-direction), if blank will be taken as equal to n-direction conductivity	$\text{Btu}/\text{ft}\text{-sec}^{\circ}\text{R}$

1.7 FUNCTIONS-OF-TIME TABLES ("HEATING" TABLES)

A series of tables provides heated surface boundary condition information as functions of time. The various time tables are independent, and each nodal box in the in-depth layout has been given the number of a time-table for use if and when that node becomes a surface node.

Three general types of boundary condition sets are available, referred to here as Options 1, 2, and 3. Section 3 of Part I above provides further description.

The three options are:

- Option 1 - General convection-radiation heating with coupled mass transfer, including the effects of unequal heat and mass transfer coefficients (non-unity Lewis number) and unequal mass diffusion coefficients.
- Option 2 - Specified surface temperature and surface recession rate.
- Option 3 - Specified radiation view factor and incident radiation flux, as functions of time, for a stationary surface ("cooldown" option).

Options 1 and 3 are surface energy balance options; Option 2 does not, of course, use an energy balance.

Each time table may have a sequence of heating options; thus it is convenient to think of each time table as consisting of a number of sub-tables, each sub-table representing one option. The switch from one option to another requires a repeated time entry, the first card representing the last entry of the earlier table and the second card representing the start of the next table.

The total number of time values in each time-table is limited to 25. Each sub-table must have at least 2 entries, hence the number of sub-tables in a table cannot exceed 12. (The most common problem has only 2, representing an Option 1 or Option 2 calculation followed by cooldown, Option 3). Time entries must be in increasing order.

The tables are presumed to be numbered sequentially beginning with table number 1. There may be ten different time tables. The format for the time-tables is as follows:

Columns	Format	Data	Units
1	I2	Flag, nominally blank, punched to indicate the last card of each time table. Punch +1 for last card of last time table, -1 for last card of any preceding time tables	---
3-10	F8.2	Time (independent variable)	sec
11-20	F10.5	Option 1: Recovery enthalpy, relative to the same chemical-base state as used with the heats of formation Option 2: Surface temperature Option 3: Blank	Btu/lb °R
21-30	F10.5	Option 1: Radiant energy flux to the surface Option 2: Surface-recession measured along nodal column center lines Option 3: Radiant energy flux to the surface	Btu/ft ² -sec mil Btu/ft ² -sec
31-40	F10.5	Option 1: Heat transfer coefficient Option 2: Blank Option 3: <u>Must</u> be blank	lb/ft ² -sec
41-50	F10.5	Option 1: Pressure (not used, information only) Option 2: ad lib Option 3: ad lib	atm
51-60	F10.5	Blowing rate parameter λ (see Section 1.2)	---

1.8 SURFACE THERMOCHEMISTRY DATA (OPTION 1 ONLY)

1.8.1 Introduction

Problems involving use of the surface thermochemistry option (Option 1) require the input of an array of surface mass and energy data particular to the option and the material being analyzed. These data include the specification of the ratio of mass transfer coefficient to heat transfer coefficient, the kinetic parameter radiation view factor, and surface thermochemical data. The cards containing this input are described below. Problems not involving Option 1 calculations do not need the surface equilibrium data deck. The program can be instructed to use the surface tables from the preceding problem (if any), in which case additional surface tables are not provided. The "surface tables" then consist of only the "lead card" described below.

1.8.2 Ratio of Mass to Heat Transfer Coefficient and Kinetics Parameter

A single card serves to specify the ratio of the convective mass transfer coefficient to the convective heat transfer coefficient (C_M/C_H) and flag kinetics parameter k_f .

Column	Format	Data	Units
1-10	F10.0	C_M/C_H	---
11-20	F10.0	Kinetics parameter k_f , forward rate coefficient of one kinetically controlled reaction in surface state solution; the particular reaction is identified in the surface state program output (see Section 3.2 in Part I above)	various
21-39	19X	Blank	---
40	I1	One punch means to use surface state tables of preceding job for this job. Blank means to read data on following thermochemistry surface deck.	---

1.8.3 Surface Thermochemistry Table

1.8.3.1 Introduction

This table supplies the necessary input data for the surface energy balance computations in Option 1. (This energy balance is discussed in Section 3 of Part I above.)

Most commonly the deck of cards which make up the surface thermochemistry table is generated by the Aerotherm Chemical Equilibrium Program (ACE). The user's manual for the ACE program describes this table in complete detail (Ref. 9). On occasion the user may desire to construct his own surface

thermochemistry table, and so the following sections include brief descriptions of the organization and format of these tables. The main emphasis however will be on the communication between the ACE program and the ASTHMA program, since this is of the most general interest.

1.8.3.2 Edge Enthalpy Data

Equations (31) and (32) of Section .3, Part I, above, indicate that if diffusion coefficients are not equal or if the ratio C_M/C_H is not unity, then the surface energy balance requires data about the edge gases of the boundary layer. These data are provided in special "edge tables" which precede each pressure section of the surface tables (the various sections of the surface tables are described in Section 1.8.3.3 below). The independent variables for an edge table are pressure and temperature. Dependent variables are h_{ew} and the sum $\sum z_{ie} h_i^{T_w}$.

The edge enthalpy data are entered on the cards as follows:

Column	Format	Data	Units
1-6	E6.4	Pressure	atm
7-26		Blank	---
27-30	F4.2	Unequal diffusion exponent* $1-\gamma$	---
31-37	E7.5	Temperature	$^{\circ}K$ ($^{\circ}R$ if negative, in which case enthalpies below are Btu/lb)
38-43	E6.4	$B'_f = \sum \dot{m}_r / \rho_e u_e C_M$ (not used in present version of ASTHMA program)	---
44-51	E8.5	Summation $\sum z_{ie} h_i^{T_w}$	Cal/gr (Btu/lb if temperature is entered with minus sign)
52-59	E8.5	h_{ew}	Cal/gr (Btu/lb if temperature is entered with minus sign)
60-65	A6	Unused	---
66	I1	0 (flag signifying that this card is part of the edge gas table)	---
67-78	2A6	Problem identification (not read)	---
79-80	I2	Page number on ACE output listing containing the data punched on this card (not read)	---

* This quantity is discussed in the user's manual for the ACE program (Ref. 9) and also in Refs. 1-6.

Note that although the ACE program will provide a data deck using °K and cal/gr, in those rare cases in which a user wishes to supply his own deck and prefers to work in °R and Btu/lb, he may do so simply by introducing a minus sign as a flag in front of the temperature entries.

The table length is limited to 5 pressure sets (it may have only 1 pressure set) with not more than 25 nor less than 3 temperature entries in each set. The series of temperature values may be different for the edge table at each pressure set. The table is organized as a series of sections, each representing one pressure and each preceding the corresponding pressure group of the surface thermochemistry deck as described below. The temperature entries within each section must be ordered, either ascending or descending. Similarly the pressures must be ordered either ascending or descending. (Decks generated by the ACE program will have been automatically ordered properly.)

1.8.3.3 Surface Thermochemistry Tables

This table is comprised of a series of sections. Each section represents one pressure* and one kinetics parameter value. Each section consists of two subsections. The second represents surface temperatures too low for ablation; in this subsection surface temperature is the independent variable. The first subsection contains the ablating cases; here the ablation rate is the third independent variable and the surface temperature is a dependent variable.

Thus one table has three independent variables: pressure, kinetics parameter k_f , and either surface temperature or ablation rate, depending on whether the surface temperature is high enough for ablation.

The table has either two or three dependent variables, according to whether the surface temperature is high enough for ablation. Two dependent variables always present are the summation $\sum z_{iw}^* h_i^T$ and h_w , the enthalpy of the wall gases. The third dependent variable is the surface temperature, but it is dependent only in those cases for which the surface temperature is high enough for ablation. (Otherwise, the surface temperature functions as an independent variable.)

The ACE program generates separate groups for each pressure, one at a time. These groups must be ordered on pressure (either ascending or descending)

*Remember that the word "pressure" is used as a handy edge state parameter. In general, of course, there are two independent edge state variables, which are properly accounted for by the ACE program in its chemistry solution. The ASTHMA program needs only to be given some "flag" to distinguish edge states. Pressure is chosen for this purpose, and hence the word "pressure" is used here as shorthand for "edge state."

by the user to make up the surface thermochemistry deck. Within each pressure group the kinetics parameters k_f' will be ordered in descending order. Within each kinetics section, non-zero ablation rate entries will be grouped ahead of the zero ablation rate entries. The non-zero ablation rate entries will not be ordered in any particular way on the ablation rates; any necessary ordering is made automatically by the ASTHMA Program as it reads in the data. The zero ablation rate entries are ordered with descending temperatures.

(Users providing their own thermochemistry decks must ensure that the pressures and kinetic parameters are ordered, but the ordering may be either ascending or descending in each case. Within each kinetics section, the non-zero ablation rate entries and the zero ablation rate entries may be shuffled together but the zero ablation rate entries must be encountered in order of descending temperature during the reading process. Non-zero (ablating) ablation entries need not be ordered. The ablation rate for non-ablating (zero ablation rate) entries need not actually be zero, but may not exceed an ablation rate in the ablation section. These cards are identified as zero ablation rate cards by a unity flag in column 66, as described in the format specification below.

The number of pressure groups may not exceed 5 (and may be only 1); the number of kinetic parameters in each pressure group may not exceed 8 and may not be less than 2. The sequence of kinetics parameter values need not be the same in the different pressure sections. Within each kinetics section the number of ablation rate entries, including the zero ablation rate (independent surface temperature) cards, may not exceed 24 and may not be less than 2. The series of ablation values B_C' may be unique for each section.

The °R-Btu/lb option described for the edge tables in Section 1.8.3.2 may be used for these tables also.)

The card format for the surface equilibrium data is as follows:

Column	Format	Data	Units
1-6	E6.4	Pressure	atm
7-14	E8.5	Ablation rate $\dot{m}_c / \rho_e u_e C_M = B_C'$	---
15-20	E6.4	Gas rate $\dot{m}_g / \rho_e u_e C_M = B_g'$ (not used by ASTHMA)	---
21-26	E6.4	Kinetics parameter $k_f / \rho_e u_e C_M$	various
27-30	F4.2	Unequal diffusion exponent*	---

*This quantity is discussed in the user's manual for the ACE program (Ref. 9) and also in Refs. 1-6.

Column	Format	Data	Units
31-37	E7.5	Surface temperature	°K (°R if negative in which case enthalpies below are Btu/lb)
38-43	E6.4	$B_f^i = \sum \dot{m}_{r,l} / \rho_e u_e C_M$ (not used in present version of ASTHMA program)	---
44-51	E8.5	Summation $\sum Z_{iw}^* h_i^{T_w}$	Cal/gr (Btu/lb if temperature is entered with minus sign)
52-59	E8.5	Enthalpy of wall gases h_w	Cal/gr (Btu/lb if temperature is entered with minus sign)
60-65	A6	Chemical symbol of surface species. (ACE program prints such symbols arranged alphabetically and truncated from right end if necessary)	
66	I1	1 for assigned-temperature entries in the equilibrium program (no ablation); > 1 for surface thermochemistry with ablation (temperature is dependent)	---
67-78	2A6	Problem identification (not read)	---
79-86	I2	Page number on ACE output listing containing the data punched on this card (not read)	---

1.8.3.4 Termination Card

The surface equilibrium data deck must be terminated by a single blank card. Output decks of the ACE program may not have such a card, in which case the user must supply it.

1.8.3.5 Assembled Thermochemical Deck

Figure 1-2 shows a picture of an assembled thermochemical data deck for several pressures.

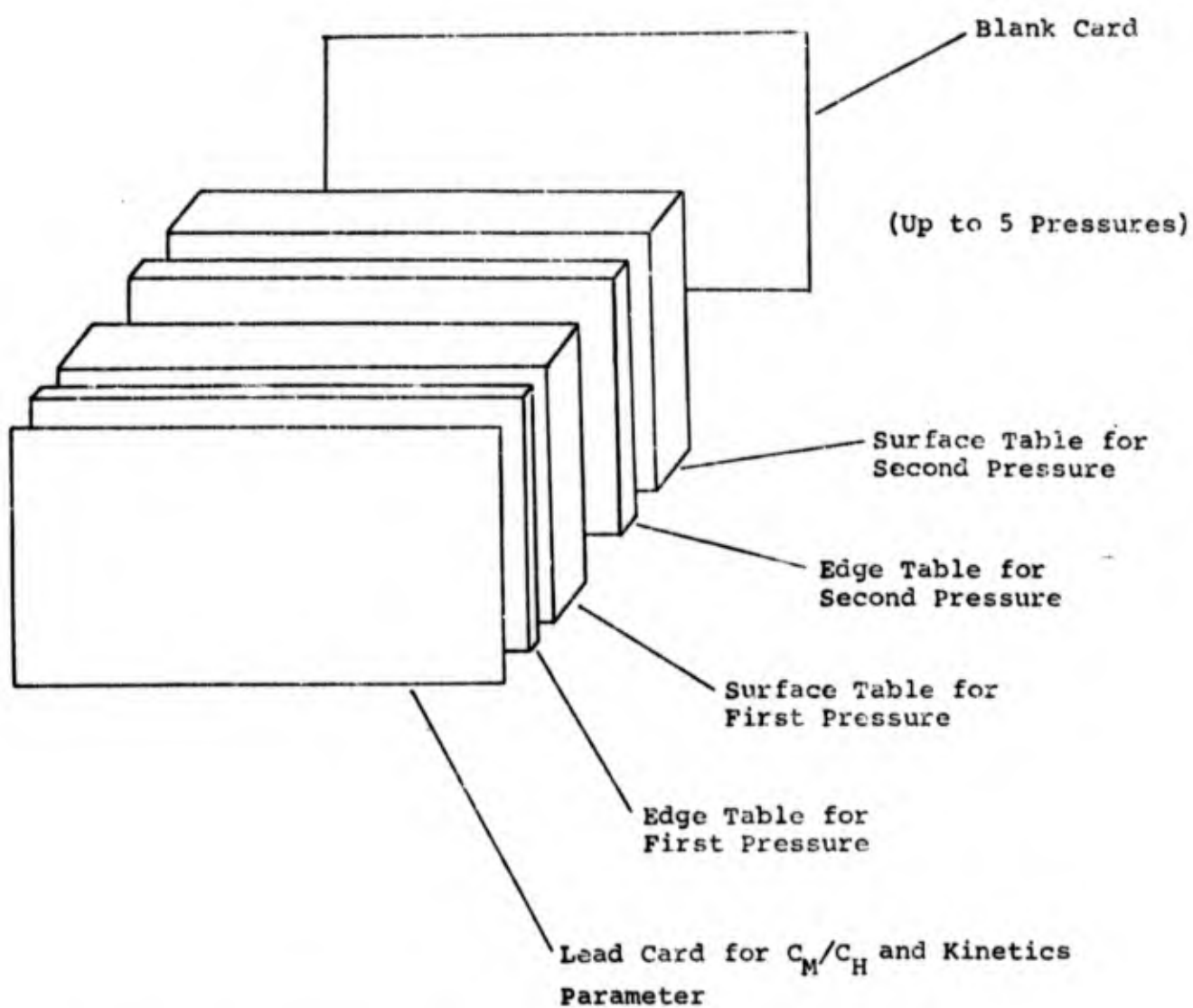


Figure 1-2. Sketch of Surface Thermochemistry Table Make-Up

SECTION 2
PROGRAM OUTPUT

2.1 INPUT DATA (EXCEPT SURFACE THERMOCHEMICAL TABLES)

Program output begins with an output of the input title and heading information, output interval specifications and general program constants, nodal data (including any computed nodal center locations), material property tables, and time dependent boundary conditions table.

All this output is fully labeled and is printed exactly as input by the user.

2.2 SURFACE THERMOCHEMISTRY TABLES

2.2.1 Edge Enthalpy Table

If there is an edge enthalpy table, it is output exactly as input.

2.2.2 Surface Thermochemistry Table

This table is output re-ordered with increasing ablation rates in each section and with a new computed term in place of the input enthalpy terms. For each entry in the surface thermochemistry tables the program computes the quantity

$$\left[\sum_i (z_{ie}^* - z_{iw}^*) h_i^T - B' h_w + \frac{\dot{m}_c h_c}{\rho_e u_e C_M} + \frac{\dot{m}_g h_g}{\rho_e u_e C_M} \right] = \frac{q_{chem}}{\rho_e u_e C_M} = \text{Chem prod} \quad (36)$$

and then outputs it as the dependent variable of interest in the output surface equilibrium table.* This term occurs directly in the surface energy balance (Eqs. (31) and (32) of Section 3.3, Part I above) and is a useful diagnostic quantity. In the output, this quantity is labeled CHEM PROD. It has the units (Btu/ft²-sec)/(lb/ft²-sec) = Btu/lb and can be loosely thought of as the chemical energy release per pound of $\rho_e u_e C_M$, where $\rho_e u_e C_M$ can be thought of as the "scrubbing flux" or "Reynolds flux."

In the output table, the quantity $\rho_e u_e C_M$ is abbreviated to CM.

* If edge tables are omitted, only the last three terms on the left are included. The mechanical removal energy term has been omitted from this equation since it is not presently considered by ASTHMA.

2.3 REGULAR OUTPUT

2.3.1 Introduction

At each output interval, as specified by the user, the program prints out the current values of ablation rates, nodal temperatures, and other supplementary information, as described below.

2.3.2 General Information

The first line of output gives the current problem time. The second line shows the current values of a number of miscellaneous quantities:

<u>Heading</u>	
QTOT, SUR	Total energy input to the surface $\iint q_{\text{cond}} dA d\theta$, Btu
QTOT, INT	Total energy rise of subsurface materials, Btu
CNSV FNER	Conservation of energy check, ratio QTOT,SUR/QTOT,INT
CRNODE	m, n indices of node yielding smallest stability limited time step
ITER	Total number of computational cycles (steps) required to reach the indicated value of output time. Note that the initial time is numbered as 1.
NODE D-TIME	Stability limited time step for critical node, sec
ACT D-TIME	Actual current time step; may be smaller than stability time step due to high surface recession or output matching

2.3.3 Miscellaneous Surface Data

A single line gives a set of data for each point on the heated surface. Each line presents the following data:

<u>Heading</u>	
ROW	Row number of node below this surface point (remember that surface points are distinct from nodal points)
COL	Column number of this surface point
OPTN	Current problem (heating) option at this surface point (1, 2, or 3)

<u>Heading</u>	
SURF ITER	Number of iterations required for the surface energy balance during the previous cycle
SURF TEMP (R)	Surface point temperature in degrees Rankine
H EDGE	The input edge enthalpy as determined by linear interpolation in the functions-of-time table
H WALL	Option 1: Enthalpy of <u>edge</u> gases at the wall temperature, h_{ew} Option 2: Blank Option 3: Blank
B PRIME TOT	Parameter $B'/\rho_e u_e C_M = (B'_f + B'_{tc})/\rho_e u_e C_M$ (current version of ASTHMA requires $B'_f = 0$)
HEAT COEFF	The current value of the convective heat transfer coefficient $\rho_e u_e C_H$ as linearly interpolated in the functions-of-time table and corrected for the effect of transpiration (blowing) according to $C_H/C_{H_0} = \zeta/(e^\zeta - 1)$ where $\zeta = 2\lambda\dot{m}/C_{H_0}$ and C_{H_0} is the heat transfer coefficient before being corrected for blowing
CH/CHO	The ratio C_H/C_{H_0} , indicating the amount of blowing correction
PRESSURE	Current value of pressure for this column is interpolated in functions-of-time tables
RADIUS	Current radial coordinate of this surface point (inches)
Z	Current axial location of this surface point (inches)

2.3.4 Surface Rate quantities

A series of lines, one for each surface point, gives the current recession rates, mass loss (ablation) rates, and surface energy flux rates.

Heading

ROW	Low number of node below this surface point
-----	---

COL	Column number of this surface point
RECESSION RATES- CENTER LINE	Rate of movement of surface point down column center line of surface nodal box (mils/sec)
NORMAL	Rate of movement of surface in direction normal to tangent to surface at this surface point (mils/sec)
MASS RATES MDOT TOTAL	$\dot{m}_{tc} + \sum \dot{m}_r$, total local mass loss rate (ablation), lb/ft ² sec

SURFACE ENERGY FLUX RATES

This group of terms all appear in the surface energy balance, Equation (32), and have the units Btu/ft² sec

CONVECTED IN	$q_{sen} = \rho_e u_e C_H (H_r - h_{e_w})$
RADIATED IN	$q_{rad} = \alpha_w q_{rad}$
RADIATED OUT	$q_{rad} = F \epsilon_w \sigma T_w^4$ out
CHEMICAL GENERATION	$q_{chem} = \rho_e u_e C_M (\text{chem prod})$
CONDUCTION AWAY	q_{cond}

2.3.5 Surface Time Integrated Quantities

These lines correspond exactly to the line of surface rate quantities described above. Here the quantities are integrated over time and the column surface area, so that the units of these quantities are "Btu for this column n" or Btu/col, so to speak. The area integration is done since the surface area of a column varies with time and there is no generally preferred area upon which to base the time integrated flux terms.

2.3.6 In-Depth Data

This block of data gives the current temperatures (degrees Rankine) of the subsurface nodal points. The m,n index coordinates of the points are included for convenience.

2.3.7 Optional Punched Card Output

Optional output called for by the KSTRP flag of Section 1.2 above provides punched card temperature input for various thermal stress codes. These cards are punched for all nodal and surface point temperatures for either special punch times or for all regular output times. The card format is:

Column	Format	Data	Units
1-10	F10.3	Point radius r	in
11-20	F10.3	point axial coordinate Z	in
21-30	F10.3	point temperature	°R
31-38	8H	Units ININDEGR	---
39-41	I3	Point row index m (blank if a surface point)	---
42-45	IH/,I3	Point column index	---
46-48	3H	Title MAT (material)	---
49-50	I2	Material number at this point	---
51-57	F7.2	time	sec
58-59	2H	Sb, units of time	---
60-71	2A6	Problem identification of columns 61-72 of third title card (Section 1.1)	---
72	1X		---
73-75	I3	Card number	---
76-77	2H	OF	---
78-80	I3	Total number of cards punched for this time	---

2.4 DUMPS

2.4.1 Introduction

To prevent the execution of computations that are wasteful or probably erroneous, the program provides certain emergency stops with dumps of diagnostic information.

2.4.2 Too-Small-Time-Step Dump

A dump occurs if the program selects a time step as small as 10^{-6} seconds. Computation ceases and the program prints out the current surface column number. Then the program sets the final problem time equal to the current time, and thus forces an output of all the current values of the standard output quantities, followed by a termination of computation.

2.4.3 Too-Many-Surface-Iterations Dump

The program allows 51 iterations to find an acceptable surface energy balance at any column point. If a balance has not been obtained after 51 iterations, the program writes the diagnostic message "Iteration Stop" and a block of diagnostic data. Since stops of this kind are complex in nature, and generally involve the surface thermochemistry data deck, the dump data must often be communicated to the program authors for analysis.

Iteration stops are almost always due to "unfortunate" variations in the tabular energy quantities making up the surface energy balance. Hence the iteration stop dump contains the 51 sets of paired values of the independent variable (either temperature or $\ln B'_c$) and the surface energy balance error generated during the search for a surface energy balance. The program then dumps the surface energy balance equation error (departure from zero) at each tabular B'_c (or temperature if failure occurred in nonablating section), in the two tables bracketing the current value of kinetics parameter, properly interpolated at the current value of the kinetics parameter, for current values of CH, $q_{\text{rad in}}$, and in-depth conduction data. This information, plus the iteration history, permits a rather complete picture of the surface energy balance error function shape, and this in turn should usually allow an accurate assessment of the particular feature of the function currently hindering convergence.

2.4.4 Unacceptable-Surface-Thermochemistry-Table Stop

A series of checks built into the input routine serve to detect common errors in the make-up of the surface thermochemistry tables. Discovery of an error stops the reading process and the program prints out a single line:

Bad Surface Equilibrium Table of Type ---

Five error types are detected as follows:

Type

- | | |
|---|---|
| 0 | Current section of independent temperature entries (no ablation) is not in descending order in temperature |
| 1 | Mass rate in no-ablation subsection has been set greater than a mass rate in the ablation subsection. Although the "zero-mass-rate" can differ from zero, they must not exceed an actual ablation rate entry. |
| 2 | Edge table has been omitted even though the diffusion coefficients are not equal (unequal diffusion exponent $\neq 0$). |

Type

- | | |
|---|---|
| 3 | Edge table has been omitted even though $C_M/C_H \neq 1$ |
| 4 | Inconsistent unequal diffusion exponent (this must be uniform for all tables) |

2.4.5 Sense Switch Dumps

Sense Switch 1 calls for output every time step, Sense Switch 2 generates extra diagnostic dumps at every output, and Sense Switch 3 dumps some additional surface table information just prior to execution of the main program iteration loop.

SECTION 3
MISCELLANEOUS

3.1 RUNNING TIME

At present, the ASTHMA program has only been run on the Philco 2000-212, for which the execution time is well approximated by

$$T(\text{hrs}) = 5 \times 10^{-7} NI \quad \pm 25\% \quad (41)$$

where N = number of nodes and I = number of iterations, provided that the number of iterations per output page exceeds about 3 (otherwise output time begins to be appreciable). Iterations I may be estimated from

$$I = \frac{\text{problem time}}{\text{time step limit}} \quad (42)$$

and the time step limit may usually be estimated fairly closely from a small number of obvious candidate nodes:

$$\text{time step limit} \approx \frac{l^2}{4\alpha}$$

for square nodes of side l , and

$$\text{time step limit} \approx \frac{l^2}{2\alpha}$$

for "thin" nodes of minimum side l .

Estimates for other machines were made using empirical speed ratios based upon similar Fortran programs for jobs of roughly 10 minutes duration. If we write $T = ANI$, then we have

Machine	A (hrs)	Estimated Error
Philco 2000-212	5×10^{-7}	<u>+25%</u>
CDC 3800	5×10^{-7}	<u>+50%</u>
CDC 6600	3×10^{-7}	<u>+50%</u>
Univac 1108	3×10^{-7}	<u>+50%</u>
IBM 7094	12×10^{-7}	<u>+50%</u>
360/40	50×10^{-7}	<u>+50%</u>
360/44	25×10^{-7}	<u>+50%</u>
360/50	25×10^{-7}	<u>+50%</u>
360/65	5×10^{-7}	<u>+50%</u>

3.2 STORAGE REQUIREMENTS

The ASTHMA program itself requires only about 5200₁₀ word of storage. Associated subroutines require about 1000₁₀ words. Most storage requirement is for dimensioned variables in various commons. With present dimensions, these require about 15,600₁₀ words. Thus the program is a tight fit in a 32,000 word machine, with present dimension values.

3.3 TAPE REQUIREMENTS

The program uses no scratch tapes. It uses logical unit number 5 for input and 6 for printed output, and uses a punch tape through a PUNCH statement.

3.4 FORTRAN DECK MAKE-UP

The ASTHMA program is in Fortran IV and consists of the following units:

- (1) Main Program
- (2) Subroutine SURFB
Surface energy balance procedures
- (3) Subroutine LCOUNT
Counts lines, turns and numbers pages
- (4) Subroutine LOOK
Table look-up with linear interpolation
- (5) Subroutine SLOPQ
Quadratic curve fit and slope finder
- (6) Subroutine OGLE
Table look-up with cubic curve fit
- (7) Subroutine ORDERD
Ordering routine

- (8) Subroutine SEQUA
Orders according to results of ORDERD
- (9) Subroutine SLOPL
Linear curve fit and slope finder
- (10) Subroutine VCOS
Evaluates cosine of angle between surface normal and
nodal center line

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APPENDIX A
STUDY OF FRONT NODE DROPPING SCHEMES

APPENDIX A

STUDY OF ALTERNATIVE FRONT NODE DROPPING SCHEMES FOR TRANSIENT ABLATION CALCULATIONS

A-1 GENERAL REMARKS

Finite difference calculations of the subsurface temperature response and the surface recession history of ablating materials have employed three general types of nodal bookkeeping schemes

1. nodal mesh fixed in the material, with any node dropping required by surface recession occurring at the front (heated) surface
2. nodal mesh fixed to receding surface, with node dropping taking place at back wall
3. nodal mesh fixed at both front and back walls, with all nodes shrinking as recession occurs.

The first scheme, although offering an appealing simplicity, proved to suffer from irritating oscillations in predicted heated surface temperature, recession rate, and heat flux. Experiences such as those reported by Brogan (Refs. A-1, A-2) encouraged investigators to consider back-surface node dropping (type 2 above) and shrinking coordinate systems (type 3 above) with excellent success for problems with one space dimension.

Consequently, interest in front node dropping schemes waned, even though occasional papers on the technique appeared from time to time (e.g., Refs. A-3 and A-4). Ablation problems with two or three dimensions usually do not lend themselves to simple treatments with receding nodal grids or shrinking grids, due to the many tedious bookkeeping aspects of recording the nodal positions in moving grids, and attempts to construct multidimensional computer programs with moving or shrinking nodes have proven discouragingly complex and unreliable (Refs. A-5 to A-8). Therefore, in the course of the present work front node dropping schemes appeared to deserve some reconsideration, and several alternative front node dropping schemes were examined for stability and smoothness of predicted surface temperature using a special-purpose, one-dimensional computer program. Section A-2 below describes the schemes considered.

A-2 METHODS EXAMINED

The schemes considered differ from each other in two aspects:

1. the location of the finite difference thermal capacity lumps relative to the thermal resistance links, and

2. the temperature at which the ablating material is injected into the surface control volume for the surface energy balance calculation.

Under point (1), the three arrangements shown in Figure A-1 were examined.

In Figure A-1 the small arrows indicate which zone thermal capacity is lumped into each capacitor in the R-C network. Note that Scheme (1) never has any capacitance associated with the surface temperature, while Scheme (2) always has capacitance associated with the surface temperature. Scheme (3) has no capacitance at the surface in mode (1), but does have capacitance at the surface in mode (2).

The three capacitance distribution schemes constituted one computational aspect tested; the second aspect concerned the surface energy balance. The surface energy balance equation, Equation (32) in the main text above, simplified to the familiar relation for $Le = 1$ and $C_M/C_H = 1$ is

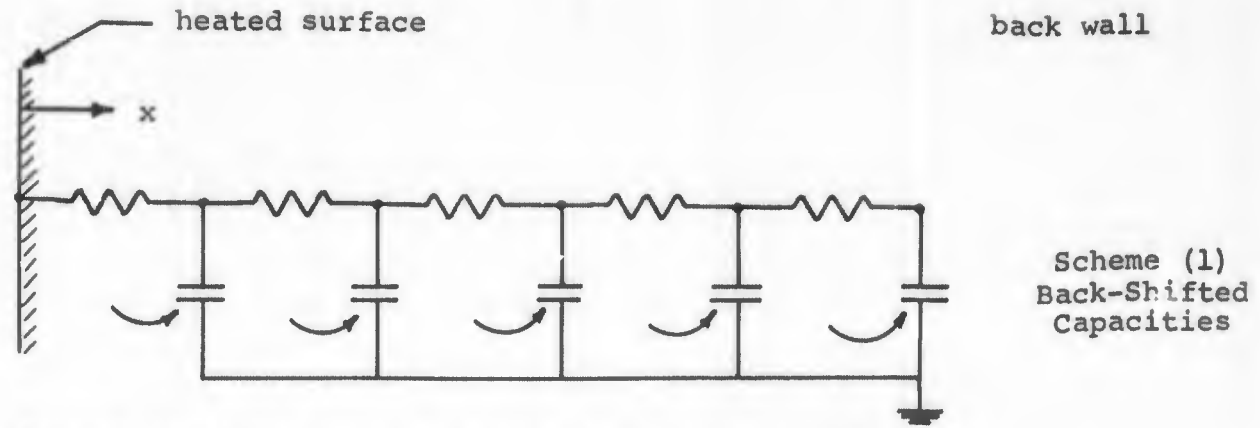
$$\rho_e U_e C_H \left[H_r - (1 + B') h_w \right] + \dot{m}_c h_c - F \sigma \epsilon T_w^4 - q_{\text{cond}} = 0 \quad (\text{A-1})$$

In this equation the term $\dot{m}_c h_c$ represents the energy injected in the surface control volume (see Figure 12 of the main text) with the subsurface material. A study of the finite difference networks of Figure A-1 reveals a choice of enthalpies to use in this term:

1. the enthalpy of the material in the top-most (shrinking) capacitance
2. the enthalpy corresponding to the heated surface temperature.

Choices (1) and (2) lead to different results in both Scheme (1) and in Mode (1) of Scheme (3) since in these cases the shrinking capacitance does not have the temperature of the surface. Enthalpy choice (1) gives an appealing consistency with the in-depth solution, since from the in-depth point of view energy is definitely being convected away at the temperature of the shrinking capacitance. Choice (2) on the other hand seems more accurate from the point of view of the surface energy balance.

A conventional explicit finite difference computer program was written to test numerically the three capacitance distribution schemes with both choices of injection enthalpy, since no amount of refined general analysis could be expected to answer the crucial question of surface temperature smoothness. The test problem was a 0.9-inch slab of graphite, insulated on the back wall, exposed to a step recovery enthalpy in air of 2600 Btu/lb at 65 atm, with $\rho_e U_e C_H = 1.76 \text{ lb/ft}^2 \text{ sec}$. For computation, the slab was divided into a crude network of nine 0.10-inch nodes in order to exaggerate any oscillations.



(Arrows indicate zones associated with each capacitor)

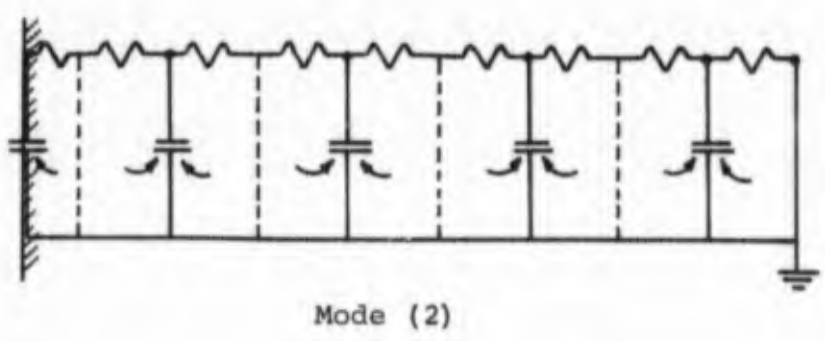
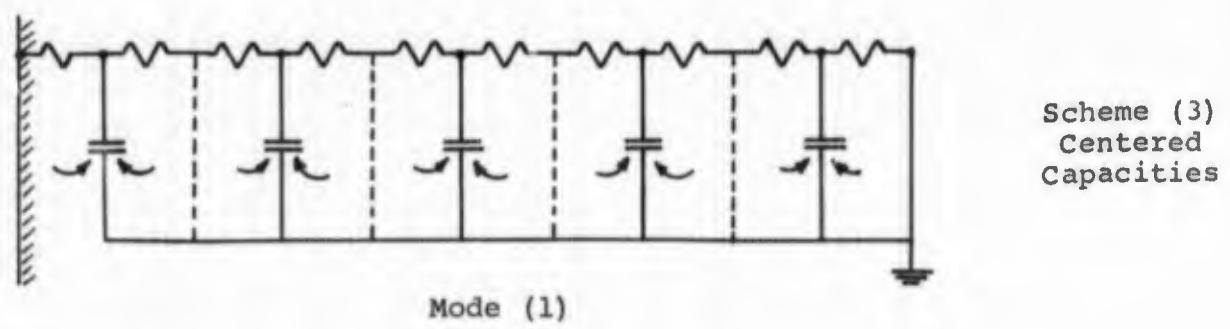
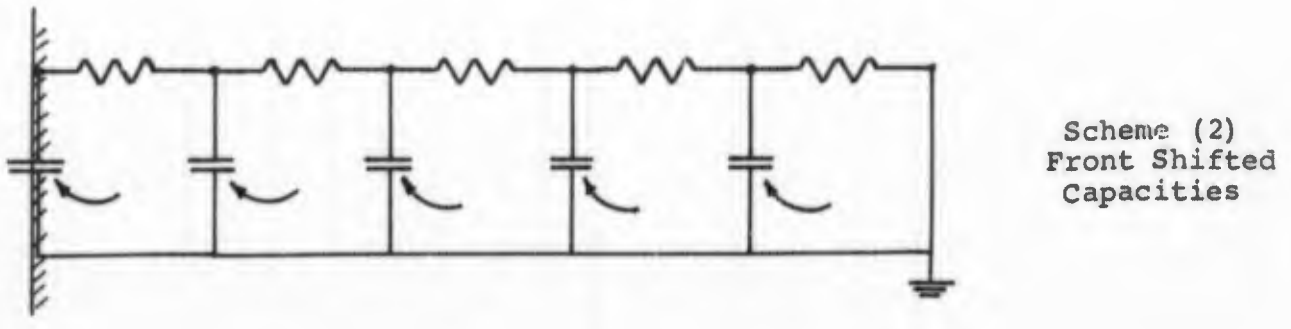


Figure A-1. Thermal R-C Network Representation of Three Alternative Schemes for Front Node Dropping

Figure A-2 shows the results for the three schemes with enthalpy choice (1), top capacitance injection. All three capacitance distributions produce surface temperature predictions which oscillate around the exact result (obtained from the well verified CMA program with small nodes). The back-shifted and centered schemes produce drastic temperature jumps whenever a node is dropped. These jumps are apparently a consequence of the sudden shift in injection temperature which occurs as a node is dropped for these two schemes. (Note that the front shifted scheme, which is inherently a front surface injection scheme, is smoother.)

Figure A-3 confirms this suspicion. It displays results for the same three capacitance arrangements as Figure A-2 but with front surface enthalpy used for the $\dot{m}_c h_c$ injection term. The front shifted scheme is unchanged, of course, and the back shifted and centered capacitance scheme predictions have lost their saw-toothed appearance. The back-shifted scheme, in fact, has produced a very acceptably smooth temperature prediction. The prediction is too high, however, an error due evidently to the obvious failure to account for energy absorption of the injected (ablated) material between the time it figuratively leaves the in-depth scheme at the top capacitance temperature and the time it enters the boundary layer at the higher front surface temperature. This conclusion is confirmed by top-capacitance injection results of Figure A-2, since this energy absorption is accounted for in that injection scheme. The average predicted result in Figure A-2 for the back-shifted capacity scheme is closer to the exact answer than the result shown in Figure A-3 exactly because the absorption is accounted for. The fact that the finite difference scheme applies the absorption correction in a saw-toothed manner (as the top capacitance temperature gradually approaches the heated surface and then abruptly recedes to the next nodal point down as a node is dropped) is what gives the saw-toothed appearance to the temperature prediction of Figure A-2.

This suggests that a preferable accuracy improvement could be made in the smooth prediction of the back-shifted scheme plus surface injection by adding during each time step some smooth correction for the missing absorption quantity instead of the saw-toothed absorption implied by top-capacitance injection. A candidate correction energy absorption rate might be

$$\text{surface energy balance correction term} = \dot{m}_{\text{avg}} c_p (T_w - T_1)_{\text{avg}} \quad (\text{A-2})$$

where "avg" denotes average during the ablation of a surface node and T_1 is the temperature of the first capacitance lump. This would still be inherently saw-toothed but would usually have less drastic jumps than the top capacitance

injection scheme. It is, of course, impractical to use this correction since during a transient problem both \dot{m}_{avg} and $(T_w - T_i)_{avg}$ only become known after the node has ablated away and its entire past history is available. A more workable correction would have the form

$$\text{surface energy balance correction term} = \dot{m}C_p \left(-\frac{\partial t}{\partial x} \right)_s \frac{\delta_{1_0}}{2} \quad (\text{A-3})$$

where δ_{1_0} denotes the initial thickness of the current top node. This correction has all virtues and no defects except possible lack of accuracy. It involves current quantities and, hence, can be calculated at every time step, and since it involves $\left(-\frac{\partial t}{\partial x} \right) = \frac{q_{cond}}{k}$ it can be directly inserted into the surface energy balance, equation (A-1), without difficulty. Thus, the surface energy balance equation becomes

$$\rho_e u_e C_H \left[H_r - (1+B')hw \right] + \dot{m}_c h_{c1} - F\epsilon T_w^4 - q_{cond} \left(1 + \frac{\dot{m}_c C_p}{2k} \delta_{1_0} \right) = 0 \quad (\text{A-4})$$

(for back-shifted capacities, Scheme (1)).

This modification was applied to the back-shifted capacitance scheme and Figure A-4 shows the improvement effected by the convection correction factor $(\dot{m}_c C_p / 2k) \delta_{1_0}$ applied to q_{cond} . The solution rapidly approaches the exact solution. The inaccuracy in early time (< 2 seconds) may be seen to be due to the crude nodal size; reducing the nodes to 0.025 inch instead of 0.10 inch results in the generally improved prediction also shown in Figure A-4.

CONCLUSIONS

The back-shifted capacity scheme with solids injected into the surface energy balance at the surface temperature provides the smoothest temperature prediction of the six methods tested. The accuracy of this method is not good unless some correction is made for the inconsistency resulting from material removal from the in-depth solution at the shrinking capacitor temperature, which is usually different from the surface temperature. A simple, smooth correction (Equation A-3) for this effect provided excellent accuracy. The back-shifted capacity scheme with front surface injection plus the simple correction has been incorporated into the two-dimensional axi-symmetric heat conduction routine (ASTHMA) described in the main text above.

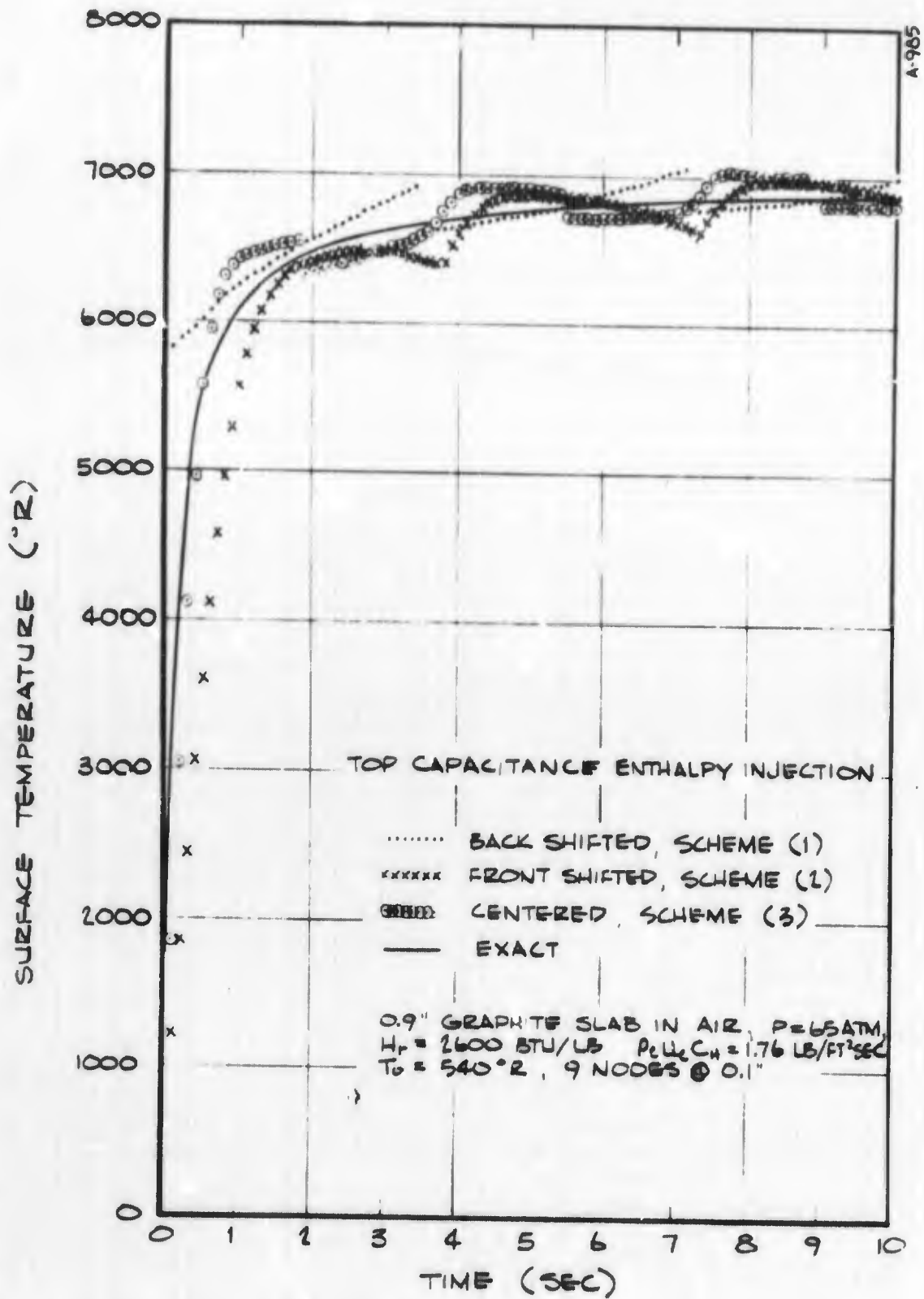


FIGURE A-2 COMPARISON OF RESULTS OF THREE CAPACITANCE DISTRIBUTION SCHEMES.

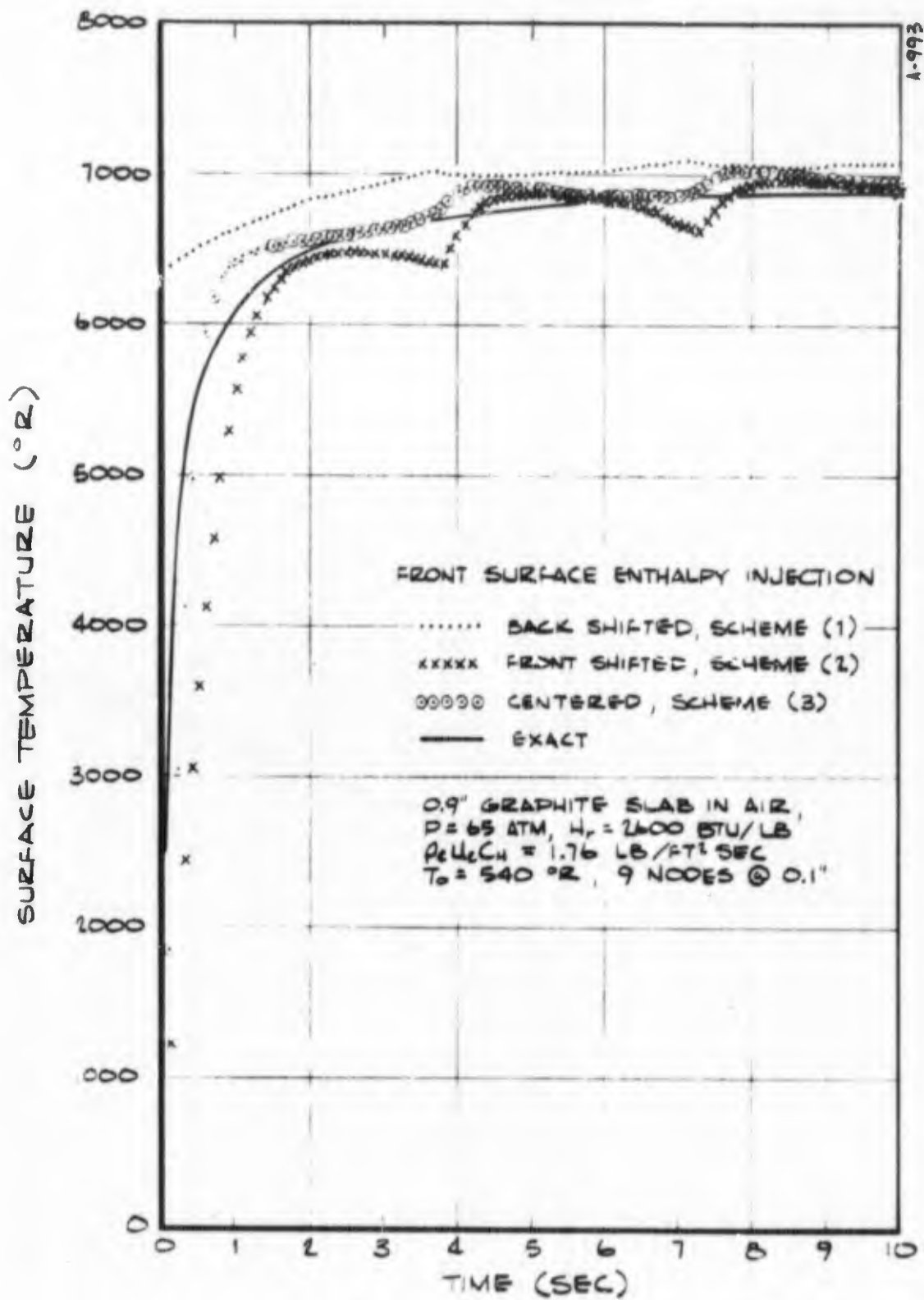


FIGURE A-3 COMPARISON OF RESULTS OF THREE CAPACITANCE DISTRIBUTION SCHEMES.

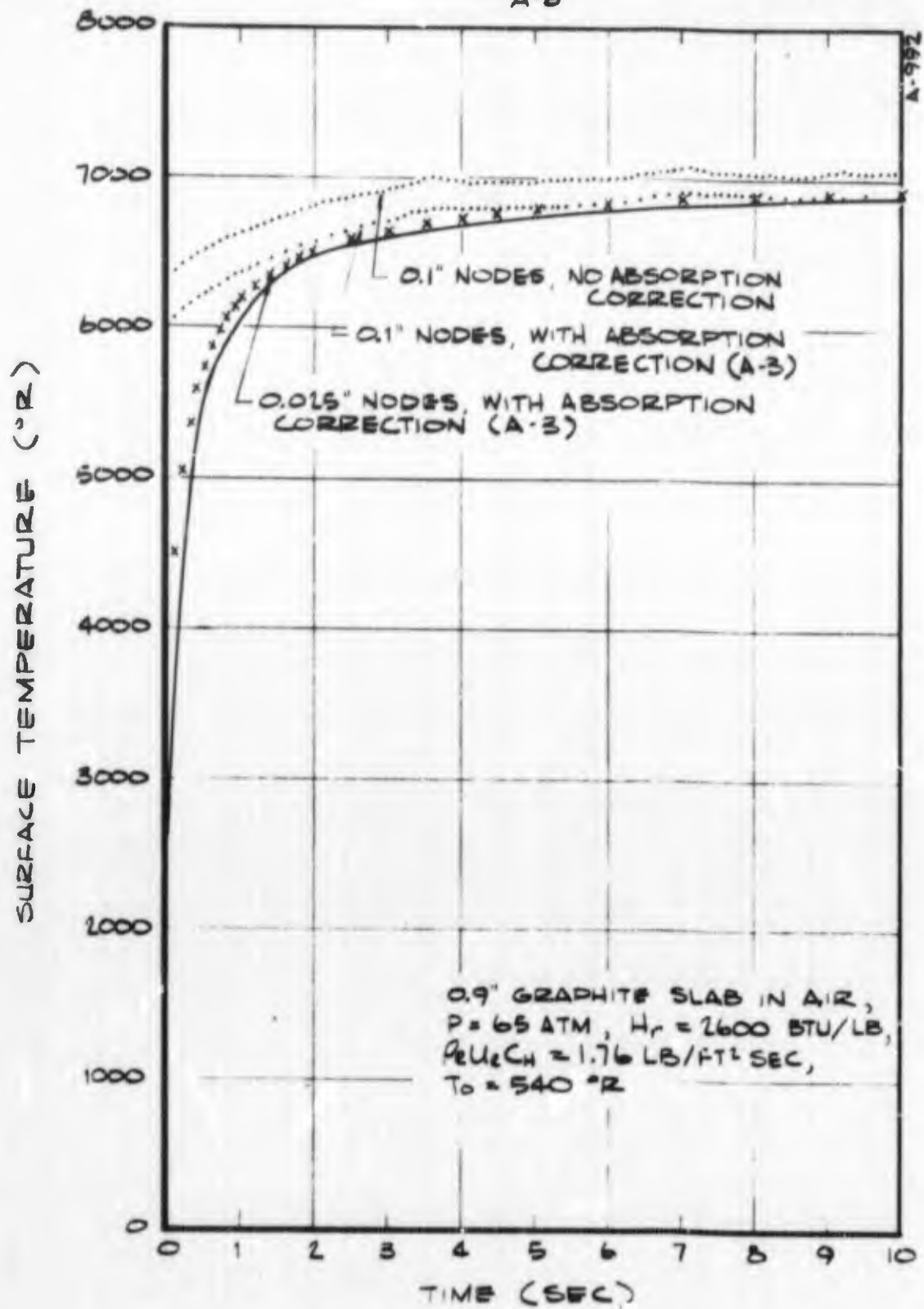


FIGURE A-4 TEMPERATURE PREDICTIONS FOR BACK SHIFTED CAPACITANCE SCHEME PLUS FRONT SURFACE INJECTION, BOTH WITH AND WITHOUT ABSORPTION CORRECTION

APPENDIX A

REFERENCES

- A-1. Brogan, J. J.: A Numerical Method of Solution for Heat Conduction in Composite Slabs with a Receding Surface. Lockheed Missiles and Space Division, Sunnyvale, California, LMSD-288204, Jan. 1960.
- A-2. Brogan, J. J.: A Simple Numerical Solution for Heat Conduction in a Solid with a Receding Surface. Journal of Spacecraft and Rockets, vol. 2, no. 1, Jan.-Feb. 1965, pp. 112-113.
- A-3. Eppes, R., Jr.: Solution of Transient Heat Transfer Problems for Flat Plates, Cylinders, and Spheres by Finite Difference Methods with Application to Surface Recession. U.S. Army Missile Command, Redstone Arsenal, Alabama, RS-TR-66-4, Sept. 1966.
- A-4. Eppes, R., Jr.: A Finite-Difference Heat Conduction Method Applicable during Surface Recession. AIAA Journal, vol. 5, no. 9, Sept. 1967, pp. 1679-1682.
- A-5. Tanzilli, R. A.: Evaluation of Graphite Composites in Reentry Environments. General Electric Co., Philadelphia, Pennsylvania, AFML-TR-65-328 (AD-473 546), October 1965.
- A-6. Persh, J.: Advanced Reentry Programs Semi-Annual Report, Volume II, Materials and AMDT (U). General Electric Co., Philadelphia, Pennsylvania, 64SD787 (AD-351 175), June 1964. SECRET.
- A-7. Nestler, D. E.: Evaluation of Micrometeorite Penetration Effects on Heat Shield Performance. General Electric Co., Philadelphia, Pennsylvania, 66SD2020 (Vols. I-IV and two Addendum Volumes), September 1966 and September 1967.
- A-8. Hurwicz, H., Fifer, S., and Kelly, M.: Multidimensional Ablation and Heat Flow During Re-entry. Journal of Spacecraft and Rockets, Vol. 1, No. 3, May-June 1964, pp. 235-242.

APPENDIX B

RECENT CHANGES IN ASTHMA

APPENDIX B
RECENT CHANGES IN ASTHMA

After the conclusion of the ASTHMA program work presented in the main text, further use of the program demonstrated the desirability of some modifications and additions. These changes have been incorporated in the program deck actually delivered under this subcontract and are presented in this Appendix (written some months after the main report above). Program users may find it helpful to note the new input features described here in the User's Manual portion of the main report above.

B-1 VARIABLE OUTPUT INTERVAL OPTION

Provision has been added to ASTHMA for changing standard output print intervals during a problem (generally similar to the capability of the CMA program described in reference B-1). In the earlier versions of ASTHMA, a single output time interval was provided on the first data card (following the three title cards). In the new version, this scheme has been retained but it has been supplemented with the variable output interval option. This option is called into play by a zero entry for the output increment in the first data card. With this number zero, a new card is introduced, following this first data card, with the following information:

Column	Format	Data	Units
1-10	F10.0	First output time interval, applied at initial time	(sec)
11-20	F10.0	Transition time between first output time interval and second output time interval	(sec)
21-30	F10.0	Second output time interval	(sec)
31-40	F10.0	Transition time between second output time interval and third output time interval	(sec)
41-50	F10.0	Third output time interval	(sec)
51-60	F10.0	Transition time between third output time interval and fourth output time interval	(sec)
61-70	F10.0	Fourth output time interval	
71-80	F10.0	Not used	

It will be noted that up to four output time intervals may be used. Output intervals are changed during the problem at the "transition times." A smaller number may be used by leaving the unneeded output intervals and transition times blank. In general, a blank or zero output interval will stop the problem at the immediately preceding transition time. Blank or zero transition times will be automatically set to the final time. The transition times should be entered in ascending order; no automatic ordering is done.

This optional output interval card precedes the optional card giving special punched output times (see p. 29 above).

B-2 CHANGE IN SURFACE ENERGY BALANCE LINKAGE

The linkage between the surface energy balance and the subsurface solution as described in Section 2.5 above proved to be unstable in certain circumstances.

Consequently, two new linkage procedures were devised, and the program now allows the user a choice of three linkage options. The first option is the procedure described in Section 2.5 above. The second is unconditionally stable but does not conserve energy in the sense that all energy conducted into the surface does not appear in $\rho C_p \partial T / \partial t$ terms. This non-conservation version, although theoretically objectionable, has been tested for a wide range of problems and has worked very well. The third option ties the second node below the surface implicitly to the surface node (in each column) in a consistent manner (i.e. so as to conserve energy) and should be the preferred option as it is both stable and energy-conservative. However, it has received little testing as yet.

The user choice among these options is indicated by a 0 (or blank), 1, or 2 punch in column 69 of the first data card of the data deck (after the three title cards). Refer to page 29 above for a description of this card. The punches designate the options as follows:

Linkage Option No.	Description	Remarks
0	As described in ASTHMA manual, surface) to-first node (of a column) implicit	Inherently unstable if nodes are dropped. Although usually successful, this scheme is not recommended if ablation is sufficient to cause any nodes to be dropped.
1	Surface-to-first-node link implicit, first-to-second node 1/4 implicit	Stable, but theoretically objectionable because it does not conserve energy, Has worked well in practice.
2	Surface-to-first-node link implicit, first-to-second node half implicit	Stable. Conserves energy. Should be best all round procedure but has received little testing.

B-3 SURFACE SHAPE CURVE FIT OPTION

The surface shape curve fit routine described on pages 11 and 12 above has been supplemented with another routine. The user designates which curve fit procedure shall be used during the calculation with a punch in column 70 of the first data card of the data deck (see page 29 of the manual). A zero or blank

calls for the linear slope averaging routine described on pages 11-12 of the manual. A one punch calls for a quadratic curve fit slope averaging routine known as SLØPQ described in Reference B-2. This routine gives better results for relatively smooth and flat (small dr/dz) shapes common to rocket nozzles; it usually performs poorly if surface points are closely spaced in the z direction (large dr/dz) and gives meaningless answers for double valued $r(z)$ shapes. For these latter shapes, the linear fit routine should be used.

B-4 MULTIPLE ABLATING MATERIALS

The ASTHMA program has been modified so that any substrate material may appear in a nodal box at the heated surface. (Previously, as described in the User's Manual above, the ablating material had to be material number one.) Options 2 and 3 boundary conditions may be applied to any surface material without restriction. Option 1 type surface thermochemical boundary conditions may be applied to any number of exposed surface materials simultaneously, but only with a tight restriction; this restriction derives from the program feature that in computing surface energy balance ablation quantities it needs the enthalpy of the substrate material at the wall temperature. In obtaining this it refers only the heat of formation of material number 1 and the specific heat function for material number 1. Therefore, although the ablating substrate materials may carry different material identifications, they must (for Option 1 calculations) have identical enthalpy-temperature relations. This limits the multimaterial capability for Option 1 to materials of different densities, emissivities, and thermal conductivities.

Since in Options 2 and 3 surface thermochemistry tables are not used, the identical enthalpy-temperature limitation does not apply for these options. Any materials with any thermal properties may appear at the surface. (It was really for this application that the multimaterial capability was added.)^{*}

*An additional restriction applies to the pressure-interpolation version of ASTHMA mentioned in Section 6 which is generally not used for rocket nozzle applications. In that version, the program chooses surface thermochemistry tables during the solution only on the basis of pressure. If the pressure range for one exposed material overlaps that of another, an artificial device must be used to distinguish the sets of pressures appropriate to each material. The most convenient device is to repunch false flag pressures in the surface tables appropriate to the additional exposed materials, and to enter appropriate pressure histories in the relevant time (heating) tables to cause the right selection of pressure tables to be made.

This restriction does not apply to the no-interpolation version of ASTHMA described in the main report above, since specific surface tables may be flagged to any node.

B-5 SURFACE THERMOCHEMISTRY TABLE FORMAT OPTION

The surface thermochemistry tables input to the ASTHMA program may be generated by a number of different surface state programs with different punched card output formats. Consequently a format option punch has been added to the ASTHMA program; the user may thus designate the format of the surface thermochemistry tables in the data deck as presented below.

B-5.1 Lead Card

There are four format choices, designated by a 0 (or blank), 1, 2, or 3 punch in column 50 of the lead card for the table (see page 34 above):

Column	Format	Data	Units
1-40		As on p. 34	Various
41-49	9X	Blank	---
50	I1	Format flag; 0 or blank _____ New standard format 1 _____ Format on p. 34 2 _____ "Anderson format" 3 _____ CMA format	

The first three formats vary only in Columns 1-20. Columns 21-80 remain as described on pages 37-38 of the manual. (We describe here only the surface tables; edge tables have the same formats as the surface tables in each case.) The fourth format is the older CMA format used in the Aerotherm Charring Material Thermal Response and Material Ablation Program (Ref. B-1).

B-5.2 New Standard Format (0 Punch)

The format cited on page 37 above was not that originally intended. The ultimate standard card output by the ACE program will follow this plan:

Column	Format	Data	Units
1-6	E6.4	Pressure	atm
7-12	E6.4	Gas rate $\dot{m}_g / \rho_e u_e C_M = B'_g$ (not used by ASTHMA)	---
13-20	E8.5	Ablation rate $\dot{m}_c / \rho_e u_e C_M = B'_c$	---
21-80		As on pp. 37-38, User's Manual	

This is the card output format of the ACE program described in Appendix C.

B-5.3 ASTHMA Manual Format (1 Punch)

This format is cited on pages 37 and 38 above but will probably not become an output option of ACE

B-5.4 "Anderson Format"

This format has been used to date for output by the ACE program and a number of tables have been generated with it. It has, however, an inconvenient format field for the ablation rate (due to a programming slip) and will be abandoned soon. This format is as follows:

Column	Format	Data	Units
1-6	E6.4	Pressure	atm
7-14	E8.4	Gas rate $\dot{m}_g / \rho_e u_e C_M = B'_g$ (not used by ASTHMA)	---
15-20	E8.5	Ablation rate $\dot{m}_c / \rho_e u_e C_M = B'_c$	---
21-80		As on pp. 37-38, User's Manual	

B-5.5 CMA Format

This is an optional ACE output format intended for use with the CMA program. Since so many graphite-in-air surface tables already exist with this format, it is useful to have it as an ASTHMA input format also. In this format the kinetics parameter is assumed by ASTHMA to be in what is the B'_g slot of the CMA input. An ACE problem with kinetics output in the CMA card format will lose the kinetics information, however, since there is no provision for it in the ACE output statement. Hence, this format will probably not be used for problems with kinetic control, and the kinetic parameters entered in the B'_g slot will be fake numbers. This format is as follows:

Column	Format	Data	Units
1-8	F8.5	Pressure	atm
9-16	F8.5	Gas rate $\dot{m}_g / \rho_e u_e C_M$	---
17-24	F8.5	Char rate $\dot{m}_c / \rho_e u_e C_M$	---
25-33	F9.4	Surface temperature	$^{\circ}K$ ($^{\circ}R$ if negative in which case enthalpies below are Btu/lb)

Column	Format	Data	Units
34-38	F5.3	Unequal diffusion exponent	---
39-47	F9.3	Summation $\sum z_{iw}^* h_i^o$	Cal/gr (Btu/lb if temperature is entered with minus sign)
48-56	F9.3	Enthalpy of wall gases h_w	Cal/gr (Btu/lb if temperature is entered with minus sign)
57-58	I2	0 for assigned-temperature entries in the equilibrium program (no ablation); >0 for equilibrium surface thermochemistry with ablation (temperature is dependent)	---
59-60	2X	Blank	---
61-66	A6	Chemical symbol of surface species. (EST* program prints such symbols arranged alphabetically and truncated from right end if necessary.)	---
67-78	2A6	Problem identification (not read)	---
79-80	I2	Page number on EST output listing containing the data punched on this card (not read)	

B-6 CHANGES IN ASTHMA PROGRAM TO INTERPOLATE ON PRESSURE

The ASTHMA program was designed primarily for two dimensional axisymmetric rocket nozzle ablation analysis. Since pressure varies axially in rocket nozzles, but usually does not vary significantly with time, the program exploited the simplifications possible in the direct assignment of particular pressure tables to given surface points. This allowed a kind of coupled time-dependent pressure effect since different pressures could be assigned to different nodes in a column; thus as the surface receded, a time varying pressure would be imposed on a given column of nodes according to the recession history.

This scheme is not very satisfactory for other bodies such as nose-tips, however, since large pressure changes may take place over a period of time, independent of body shape change effects, due to the time varying nature of the flow field. Therefore, appropriate modifications have been made in the surface energy balance routine of the ASTHMA program to allow time dependent pressures, with appropriate interpolation or pressure, for a given nodal column.

* or ACE

Thus there are now two alternative versions of the ASTHMA program, each useful for a given class of problems.

The only physical change needed to convert ASTHMA to the pressure-interpolation version described here is the substitution of the subroutine SURFB. This subroutine is called by ASTHMA and hence has retained its name, but the Fortran decks of the two versions of this subroutine are appropriately distinguished with comment cards.

Since the pressure interpolation version of ASTHMA is of minimal utility for rocket nozzle work, this version is not further described here and is not the version of the program deck provided under this subcontract.

APPENDIX B

References

- B-1 Aerotherm Corporation, Palo Alto, Calif.: User's Manual, Aerotherm Charring Material Ablation Program, Version 2. January 1966.
- B-2 Aerotherm Corporation, Palo Alto, Calif.: Fortran Variable Names, Aerotherm Charring Material Ablation Program, Version 2. February 1966.

APPENDIX C

ASTHMA PROGRAM SAMPLE PROBLEM

MITCHELL R. WOOL

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ASTHMA INPUT DECK

APG 112D 800 PSIA CHAMBER PRESSURE, BASELINE CASE 6032,03, PLUS COOLDOWN
 AFTER 6.0 SECONDS, PYROLYTIC GRAPHITE SURFACE, USING 6 NODAL COLUMNS AND
 6 THERMOCHEMISTRY TABLES ASSIGNED BY NODE.

112,800,COOL
 530. 111 1

12 6 0. 60. 1. .90 0. .4 .001 .85

2.618	0.000
1.998	0.000
1.557	0.000
1.204	0.000
1.059	0.000
0.985	0.000
0.911	0.000
0.850	0.000
0.838	0.000
0.826	0.000
0.814	0.000
0.802	0.000
0.790	0.000
2.618	0.200
1.998	0.200
1.557	0.200
1.204	0.200
1.020	0.193
0.918	0.185
0.825	0.170
0.763	0.159
0.752	0.157
0.741	0.154
0.730	0.152
0.719	0.149
0.708	0.146
2.618	0.419
1.998	0.419
1.557	0.400
1.204	0.400
1.004	0.397
0.868	0.388
0.749	0.370
0.681	0.356
0.669	0.353
0.656	0.350
0.644	0.347
0.631	0.344
0.618	0.341
1.998	0.600
1.557	0.600
1.204	0.600
1.003	0.600
0.845	0.597
0.708	0.592
0.638	0.588
0.623	0.586

0.608 0.585
0.593 0.583
0.578 0.582
0.562 0.580

1.998 0.800
1.557 0.800
1.204 0.800
1.002 0.800
0.841 0.801
0.706 0.806
0.643 0.809
0.627 0.810
0.611 0.811
0.595 0.812
0.580 0.814
0.564 0.818

1.998 1.000
1.557 1.000
1.204 1.000
1.002 1.000
0.850 1.006
0.730 1.020
0.677 1.029
0.664 1.031
0.651 1.034
0.638 1.036
0.625 1.039
0.612 1.042

1.998 1.185
1.557 1.185
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1.010 1.185
0.876 1.185
0.759 1.185
0.708 1.185
0.696 1.185
0.684 1.185
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3 530.
3 530.
2 530.
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11 11

C-3

11 1 1	530.	.97	
11 1 1	530.	.97	
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3	530.		
2	530.		
2	530.		
2	530.		
2	530.		
2	530.		
2	530.		
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11 1 2	530.	.97	
11 1 2	530.	.97	
11 1 2	530.	.97	
1111 2	530.	.97	
3	530.		
2	530.		
2	530.		
2	530.		
2	530.		
2	530.		
11 2 3	530.	.97	.058
11 2 3	530.	.97	.058
11 2 3	530.	.97	.058
11 2 3	530.	.97	.058
1112 3	530.	.97	.058
3	530.		
2	530.		
2	530.		
2	530.		
2	530.		
2	530.		
11 2 4	530.	.97	.17
11 2 4	530.	.97	.17
11 2 4	530.	.97	.17
11 2 4	530.	.97	.17
1112 4	530.	.97	.17
3	530.		
2	530.		
2	530.		
2	530.		
2	530.		
2	530.		
11 2 4	530.	.97	.174
110305	530.	.97	.174
110305	530.	.97	.174
110305	530.	.97	.174
111305	530.	.97	.174

C-6

3140+2	0+0	0+05200+5	010000+4	0+0-23881+3-23881+300000000	EDGE TABLE
3140+2	0+0	0+05200+5	050000+3	0+0-41146+3-41146+300000000	EDGE TABLE
3140+2	0+0-11641-93850+5	0+0000+4	010000+4	0+0-11193+4-11193+4C*	THROAT
3140+2	0+0 24342-63850+5	015000+4	015000+4	0+0-60297+3-60297+3C*	THROAT
3140+2	0+0 32759-33850+5	020000+4	020000+4	0+0-31587+3-31587+3C*	THROAT
3140+2	0+0 37943-23850+5	022500+4	022500+4	0+0-16971+3-16971+3C*	THROAT
3140+2	0+0 41234-13850+5	025000+4	025000+4	0+0 96459+2 96459+2C*	THROAT
3140+2	0+0 12241+03850+5	027500+4	027500+4	0+0 48444+3 48444+3C*	THROAT
3140+2	0+0 16256+03850+5	030000+4	030000+4	0+0 75382+3 75382+3C*	THROAT
3140+2	0+0 23485+03850+5	032500+4	032500+4	0+0 11295+4 11295+4C*	THROAT
3140+2	0+0 38959+03850+5	035000+4	035000+4	0+0 17079+4 17079+4C*	THROAT
3140+2	0+0 12278+13850+5	040000+4	040000+4	0+0 33563+4 33563+4C*	THROAT
3140+2	0+0-11641-9	010000+4	010000+4	0+0-11193+4-11193+4C*	THROAT
3140+2	0+0 24342-6	015000+4	015000+4	0+0-60297+3-60297+3C*	THROAT
3140+2	0+0 32759-3	020000+4	020000+4	0+0-31587+3-31587+3C*	THROAT
3140+2	0+0 37943-2	022500+4	022500+4	0+0-16971+3-16971+3C*	THROAT
3140+2	0+0 41234-1	025000+4	025000+4	0+0 96459+2 96459+2C*	THROAT
3140+2	0+0 12241+0	027500+4	027500+4	0+0 48444+3 48444+3C*	THROAT
3140+2	0+0 16256+0	030000+4	030000+4	0+0 75382+3 75382+3C*	THROAT
3140+2	0+0 23485+0	032500+4	032500+4	0+0 11295+4 11295+4C*	THROAT
3140+2	0+0 38959+0	035000+4	035000+4	0+0 17079+4 17079+4C*	THROAT
3140+2	0+0 12878+1	040000+4	040000+4	0+0 33563+4 33563+4C*	THROAT
1600+2	0+0	0+03850+5	040000+4	0+0 11447+4 11447+400000000	EDGE TABLE
1600+2	0+0	0+03850+5	035000+4	0+0 89673+3 89673+300000000	EDGE TABLE
1600+2	0+0	0+03850+5	030000+4	0+0 65275+3 65275+300000000	EDGE TABLE
1600+2	0+0	0+03850+5	025000+4	0+0 41309+3 41309+300000000	EDGE TABLE
1600+2	0+0	0+03850+5	020000+4	0+0 17996+3 17996+300000000	EDGE TABLE
1600+2	0+0	0+03850+5	015000+4	0+0-45024+2-45024+200000000	EDGE TABLE
1600+2	0+0	0+03850+5	010000+4	0+0-25696+3-25696+300000000	EDGE TABLE
1600+2	0+0	0+03850+5	050000+3	0+0-43140+3-43140+300000000	EDGE TABLE
1600+2	0+0-23283-4820+5	010000+4	010000+4	0+0-10165+4-10165+4C*	EXIT SURF.
1600+2	0+0 27345-64820+5	015000+4	015000+4	0+0-54530+3-54530+3C*	EXIT SURF.
1600+2	0+0 36186-34820+5	020000+4	020000+4	0+0-26271+3-26271+3C*	EXIT SURF.
1600+2	0+0 39897-24820+5	022500+4	022500+4	0+0-11622+3-11622+3C*	EXIT SURF.
1600+2	0+0 33106-14820+5	025000+4	025000+4	0+0 12309+3 12309+3C*	EXIT SURF.
1600+2	0+0 11389+04820+5	027500+4	027500+4	0+0 51930+3 51930+3C*	EXIT SURF.
1600+2	0+0 15822+04820+5	030000+4	030000+4	0+0 81611+3 81611+3C*	EXIT SURF.
1600+2	0+0 24217+04820+5	032500+4	032500+4	0+0 12485+4 12485+4C*	EXIT SURF.
1600+2	0+0 43538+04820+5	035000+4	035000+4	0+0 19317+4 19317+4C*	EXIT SURF.
1600+2	0+0 15255+14820+5	040000+4	040000+4	0+0 36698+4 36698+4C*	EXIT SURF.
1600+2	0+0-23283-9	010000+4	010000+4	0+0-10165+4-10165+4C*	EXIT SURF.
1600+2	0+0 27345-6	015000+4	015000+4	0+0-54530+3-54530+3C*	EXIT SURF.
1600+2	0+0 36186-3	020000+4	020000+4	0+0-26271+3-26271+3C*	EXIT SURF.
1600+2	0+0 39897-2	022500+4	022500+4	0+0-11622+3-11622+3C*	EXIT SURF.
1600+2	0+0 33106-1	025000+4	025000+4	0+0 12309+3 12309+3C*	EXIT SURF.
1600+2	0+0 11389+0	027500+4	027500+4	0+0 51930+3 51930+3C*	EXIT SURF.
1600+2	0+0 15822+0	030000+4	030000+4	0+0 81611+3 81611+3C*	EXIT SURF.
1600+2	0+0 24217+0	032500+4	032500+4	0+0 12485+4 12485+4C*	EXIT SURF.
1600+2	0+0 43538+0	035000+4	035000+4	0+0 19317+4 19317+4C*	EXIT SURF.
1600+2	0+0 15255+1	040000+4	040000+4	0+0 36698+4 36698+4C*	EXIT SURF.

AXI-SYMMETRIC TRANSIENT HEATING AND POTENTIAL FLATION PROGRAM

APR 112F 256 PSIA CHAMBER PRESSURE; RACELINE CASE 4032.03, PLUS COOL DOWN AFTER 4.0 SECONDS, PYROLYTIC GRAPHITE SURFACE. USING A NODAL COLUMNS AND A 7-1/2 CHEMISTRY TABLES ASSIG EC BY NODE.

INPUT DATA

UNITS OF INPUT DATA

TIME SEC TEMPERATURE DEG R LD PER CUHIC FT
SPECIFIC HEAT BTU PER LB DEG R RADIATIVITY BTU PER FT SEC DEG R DIMENSIONLESS
HEAT COEFFICIENT LB PER SQ FT SEC BATHALPY BTU PER LB NODAL COORDINATES INCHES
RESISTANCES SQ FT SEC DEG R PER BTU

PROBLEM CONSTANTS

MAX AMAX INIT TIME FINL TIME PRAT TIME TIME INFR TIME OACT WKTA RESERVOIR TEMPERATURE
12 6 0. 0.5000002 0.1000001 0. 0.0000000 0. 530.00
BACK RAIL CONVECTION PACK RAIL EMISSIVITY
COEF BTU/FTSQ-SEC-DEG R 0.0010 0.850

AEROTERM AXI-SYMMETRIC TRANSIENT HEATING AND MATERIAL ABLATION PROGRAM

0.1330+024	0.7430+002	0.3570+000	0.3380+022	0.9000+000	0.
0.1330+024	0.7430+002	0.4050+000	0.3380+022	0.9000+000	0.
0.1330+024	0.7430+002	0.4450+000	0.3380+022	0.9000+000	0.
0.1330+024	0.7430+002	0.4800+000	0.3380+022	0.9000+000	0.
0.1330+024	0.7430+002	0.4980+000	0.3380+022	0.9000+000	0.
0.1330+024	0.7430+002	0.5200+000	0.3380+022	0.9000+000	0.
0.1330+024	0.7430+002	0.5200+000	0.3380+022	0.9000+000	0.
0.1330+024	0.7430+002	0.5200+000	0.3380+022	0.9000+000	0.

MATERIAL NO. 3

TEMP	DENSITY	SPEC HEAT	CONDUCT	EMISSIV	CONDUCT2
0.5250+023	0.1110+003	0.2310+000	0.5560+024	0.9000+000	0.
0.5250+023	0.1110+003	0.3100+000	0.5560+024	0.9000+000	0.
0.5250+024	0.1110+003	0.5000+000	0.5560+024	0.9000+000	0.
0.5250+024	0.1110+003	0.5370+000	0.5560+024	0.9000+000	0.
0.5250+024	0.1110+003	0.5600+000	0.5560+024	0.9000+000	0.
0.5250+024	0.1110+003	0.6000+000	0.5560+024	0.9000+000	0.
0.5250+024	0.1110+003	0.6100+000	0.5560+024	0.9000+000	0.
0.5250+024	0.1110+003	0.6200+000	0.5560+024	0.9000+000	0.
0.5250+024	0.1110+003	0.6300+000	0.5560+024	0.9000+000	0.
0.5250+024	0.1110+003	0.6400+000	0.5560+024	0.9000+000	0.
0.5250+024	0.1110+003	0.6400+000	0.5560+024	0.9000+000	0.

TIME DEPENDENT BOUNDARY CONDITIONS---

TIME (SEC)	TIME TABLE NUMBER	PROG	OPTN	RECOVERY		RADIATION HEAT RATE (BT/SC FT-SEC/SC)	RADIATION HEAT RATE (BT/SC FT-SEC/SC)	HEAT COEFF (LB/SC FT-SEC/SC)	PRESSURE (ATM)	BLUING PER ACTION PARAMETER
				ENTHALPY (BTU/LB)	VIEW FACTOR					
0.	1	1	1	2020.00		785.00	785.00	0.1750	50.00000	0.400
4.00	1	PROG	OPTN	2020.00		785.00	785.00	0.1750	50.00000	0.400
TIME (SEC)				VIEW FACTOR						
0.	3	3	3	0.		0.	0.			
4.00	3	PROG	OPTN	0.		0.	0.			
TIME (SEC)				RECOVERY ENTHALPY (BTU/LB)						
0.	1	1	1	2006.00		785.00	785.00	1.0530	47.50000	0.400
4.00	1	PROG	OPTN	2006.00		785.00	785.00	1.0530	47.50000	0.400
TIME (SEC)				VIEW FACTOR						
0.	3	3	3	0.		0.	0.			
4.00	3	PROG	OPTN	0.		0.	0.			
TIME (SEC)				VIEW FACTOR						

AXI-SYMMETRIC TRANSIENT HEATING AND MATERIAL ABLATION PROGRAM

TIME TABLE NUMBER 3
 PROB RECOVERY 3
 OPTN ENTHALPY (BTU/LB)
 1 1944.00
 1 1944.00
 PROB VIEW
 OPTN FACTOR
 TIME (SEC)
 0 785.00
 4 785.00
 TIME (SEC)
 0 RADIATION HEAT COEFF PRESSURE BLINDING
 4 (BTU/SC FT- SECONDS) (LB/SC FT- (ATM) PERCENTION
 0 19220 41:00000 0.400
 4 19220 41:00000 0.400

TIME TABLE NUMBER 4
 PROB RECOVERY 4
 OPTN ENTHALPY (BTU/LB)
 1 1957.00
 1 1957.00
 PROB VIEW
 OPTN FACTOR
 TIME (SEC)
 0 785.00
 4 785.00
 TIME (SEC)
 0 RADIATION HEAT COEFF PRESSURE BLINDING
 4 (BTU/SC FT- SECONDS) (LB/SC FT- (ATM) PERCENTION
 0 19310 31:50000 0.400
 4 19310 31:50000 0.400

TIME TABLE NUMBER 5
 PROB RECOVERY 5
 OPTN ENTHALPY (BTU/LB)
 1 1747.00
 1 1747.00
 PROB VIEW
 OPTN FACTOR
 TIME (SEC)
 0 785.00
 4 785.00
 TIME (SEC)
 0 RADIATION HEAT COEFF PRESSURE BLINDING
 4 (BTU/SC FT- SECONDS) (LB/SC FT- (ATM) PERCENTION
 0 19700 17:00000 0.400
 4 19700 17:00000 0.400

TIME TABLE NUMBER 6
 PROB RECOVERY 6
 OPTN ENTHALPY (BTU/LB)
 1 1751.00
 1 1751.00
 PROB VIEW
 OPTN FACTOR
 TIME (SEC)
 0 785.00
 4 785.00
 TIME (SEC)
 0 RADIATION HEAT COEFF PRESSURE BLINDING
 4 (BTU/SC FT- SECONDS) (LB/SC FT- (ATM) PERCENTION
 0 19100 16:00000 0.400
 4 19100 16:00000 0.400

TIME TABLE NUMBER 7
 PROB RECOVERY 7
 OPTN ENTHALPY (BTU/LB)
 1 1751.00
 1 1751.00
 PROB VIEW
 OPTN FACTOR
 TIME (SEC)
 0 785.00
 4 785.00
 TIME (SEC)
 0 RADIATION HEAT COEFF PRESSURE BLINDING
 4 (BTU/SC FT- SECONDS) (LB/SC FT- (ATM) PERCENTION
 0 19100 16:00000 0.400
 4 19100 16:00000 0.400

---SURFACE EMITTIVITY = 1.0---

RATIO OF MASS TO HEAT TRANSFER COEFFICIENTS = 0.1475
 JET EQUIL DIFFUSION EXPOONENT = 0.5
 RADIUS CORRECTION ON C =

P = 18.0000 ATM

TEMP (DEG B)	M-DOT-CHAR/CM	CHEM.PRID (RTU/LB)	SURFACE SPECIES	TEMP (DEG B)	M-DOT-CHAR/CM	CHEM.PRID (RTU/LB)	SURFACE SPECIES
720.00	0.0000	1714.54	C0	405.00	0.1200	204.44	C0
430.00	0.0000	1100.37	C0	540.00	0.1664	43.80	C0
540.00	0.0005	980.04	C0	630.00	0.2327	-399.36	C0
	0.0056	927.61	C0	720.00	0.3684	-1316.22	C0
	0.0712	581.2	C0		1.1560	-4590.67	C0

KINETICS PRM = 0.520005

PRESSURE = 48.0000 ATM

P = 11.4000 ATM

TEMP (DEG B)	M-DOT-CHAR/CM	CHEM.PRID (RTU/LB)	SURFACE SPECIES	TEMP (DEG B)	M-DOT-CHAR/CM	CHEM.PRID (RTU/LB)	SURFACE SPECIES
720.00	0.0000	1714.54	C0	405.00	0.1200	204.44	C0
430.00	0.0000	1100.37	C0	540.00	0.1664	43.80	C0
540.00	0.0005	980.04	C0	630.00	0.2327	-399.36	C0
	0.0056	927.61	C0	720.00	0.3684	-1316.22	C0
	0.0712	581.2	C0		1.1560	-4590.67	C0

P = 11.4000 ATM

TEMP (DEG B)	M-DOT-CHAR/CM	CHEM.PRID (RTU/LB)	SURFACE SPECIES	TEMP (DEG B)	M-DOT-CHAR/CM	CHEM.PRID (RTU/LB)	SURFACE SPECIES
720.00	0.0000	1714.54	C0	405.00	0.1200	204.44	C0
430.00	0.0000	1100.37	C0	540.00	0.1664	43.80	C0
540.00	0.0005	980.04	C0	630.00	0.2327	-399.36	C0
	0.0056	927.61	C0	720.00	0.3684	-1316.22	C0
	0.0712	581.2	C0		1.1560	-4590.67	C0

AXI-SYMMETRIC TRANSIENT HEATING AND MATERIAL DEFORMATION PROGRAM

KINETICS PRM = 0.335+005 PRESSURE = 31.000 ATM

TEMP (DEG F)	M-DOT-CHAR/CM	CHEM.PRD (BTU/LB)	SURFACE SPECIES	TEMP (DEG F)	M-DOT-CHAR/CM	CHEM.PRD (BTU/LB)	SURFACE SPECIES
100.00	-0.0000	1584.80	C*	495.00	0.1224	233.62	C*
270.00	0.0000	1032.30	C*	540.00	0.1424	34.47	C*
300.00	0.0003	916.01	C*	585.00	0.2349	-538.43	C*
450.00	0.0034	864.73	C*	630.00	0.3894	-1415.74	C*
480.00	0.0412	651.32	C*	720.00	1.2872	-7733.27	C*

KINETICS PRM = 0. PRESSURE = 31.000 ATM

TEMP (DEG F)	M-DOT-CHAR/CM	CHEM.PRD (BTU/LB)	SURFACE SPECIES	TEMP (DEG F)	M-DOT-CHAR/CM	CHEM.PRD (BTU/LB)	SURFACE SPECIES
100.00	-0.0000	1584.80	C*	495.00	0.1224	233.62	C*
270.00	0.0000	1032.30	C*	540.00	0.1424	-34.47	C*
300.00	0.0003	916.01	C*	585.00	0.2349	-538.43	C*
450.00	0.0038	864.73	C*	630.00	0.3894	-1415.74	C*
480.00	0.0412	651.32	C*	720.00	1.2872	-7733.27	C*

P = 76.0000 ATM

TEMP (DEG F)	EDGE ENTH AT T-WALL	TEMPERATURE (DEG R)	EDGE ENTH AT T-WALL	TEMPERATURE (DEG F)	EDGE ENTH AT T-WALL
720.00	2060.46	4500.00	743.96	1800.00	-442.53
800.00	1614.11	3400.00	323.63	000.00	-776.59
540.00	1174.95	2700.00	81.04		

KINETICS PRM = 0.482+005 PRESSURE = 14.000 ATM

TEMP (DEG F)	M-DOT-CHAR/CM	CHEM.PRD (BTU/LB)	SURFACE SPECIES	TEMP (DEG F)	M-DOT-CHAR/CM	CHEM.PRD (BTU/LB)	SURFACE SPECIES
100.00	-0.0000	1347.17	C*	495.00	0.1130	139.87	C*
270.00	0.0000	900.51	C*	540.00	0.1582	-179.04	C*
300.00	0.0004	797.41	C*	585.00	0.2422	-810.32	C*
450.00	0.0040	748.14	C*	630.00	0.4354	-2218.69	C*
480.00	0.0331	571.73	C*	720.00	1.5255	-9850.38	C*

KINETICS PRM = 0. PRESSURE = 14.000 ATM

TEMP (DEG F)	M-DOT-CHAR/CM	CHEM.PRD (BTU/LB)	SURFACE SPECIES	TEMP (DEG F)	M-DOT-CHAR/CM	CHEM.PRD (BTU/LB)	SURFACE SPECIES
100.00	-0.0000	1367.17	C*	495.00	0.1130	149.07	C*
270.00	0.0000	900.51	C*	540.00	0.1582	-179.04	C*
300.00	0.0004	797.41	C*	585.00	0.2422	-810.32	C*
450.00	0.0040	748.14	C*	630.00	0.4354	-2218.69	C*
480.00	0.0331	571.73	C*	720.00	1.5255	-9850.38	C*

ALMOST-EXACT AXI-SYMMETRIC TRANSCIENT HEATING AND MATERIAL RELATION PROGRAM

... COMPUTATES

	HC(IN)	ZU(IN)	PA(T)	ZN(JN)
1	1.26180+001	0.	1.2414+001	0.10000+000
2	1.19980+001	0.	1.1998+001	0.10000+000
3	1.15570+001	0.	1.1527+001	0.10000+000
4	1.12040+001	0.	1.1204+001	0.10000+000
5	1.10590+001	0.	1.10395+001	0.06500+001
6	1.98550+000	0.	1.9515+000	1.92500+001
7	1.91190+000	0.	1.8680+000	0.85000+001
8	0.85000+000	0.	1.8065+000	0.79500+001
9	1.83800+000	0.	1.7950+000	0.78500+001
10	1.82500+000	0.	1.7835+000	0.77000+001
11	1.81400+000	0.	1.7720+000	0.76000+001
12	1.80200+000	0.	1.7605+000	0.74500+001
13	1.79000+000	0.		
14	1.26180+001	0.20000+000	1.2614+001	0.30950+000
15	1.19980+001	0.20000+000	1.1998+001	0.30950+000
16	1.15570+001	0.20000+000	1.1557+001	0.30000+000
17	1.12040+001	0.20000+000	1.1204+001	0.30000+000
18	1.10200+001	0.19500+000	1.1012+001	0.29500+000
19	1.91400+000	0.19500+000	1.8930+000	0.28450+000
20	1.82500+000	0.17000+000	1.7870+000	0.27000+000
21	1.76330+000	0.15900+000	1.7105+000	0.25500+000
22	1.75200+000	0.15700+000	1.6985+000	0.25200+000
23	1.74100+000	0.15400+000	1.6870+000	0.24950+000
24	1.73000+000	0.15200+000	1.6750+000	0.24650+000
25	1.71900+000	0.14600+000		
26	1.70800+000	0.14600+000	1.3090+001	0.20950+000
27	1.69800+001	0.41900+000	1.1998+001	0.50950+000
28	1.55570+001	0.40000+000	1.1527+001	0.50000+000
29	1.12040+001	0.40000+000	1.1204+001	0.50000+000
30	1.10400+001	0.39700+000	1.10035+001	0.49850+000
31	1.86800+000	0.38000+000	1.8565+000	0.49250+000
32	1.74900+000	0.37000+000	1.7285+000	0.48100+000
33	1.68100+000	0.36000+000	1.6595+000	0.47200+000
34	1.66900+000	0.35300+000	1.6460+000	0.46950+000
35	1.65800+000	0.35000+000		
36	1.64800+000	0.34700+000	1.6320+000	0.46750+000
37	1.63800+000	0.34400+000	1.6185+000	0.46500+000
38	1.63300+000	0.34400+000	1.6045+000	0.46300+000
39	1.61400+000	0.34100+000		
40	1.9980+001	0.60000+000	1.1998+001	0.70000+000
41	1.5570+001	0.60000+000	1.1557+001	0.70000+000
42	1.1204+001	0.60000+000	1.1204+001	0.70000+000
43	1.10030+001	0.60000+000	1.10030+001	0.70000+000
44	1.84500+000	0.59700+000	1.8430+000	0.69900+000
45	1.70800+000	0.59200+000	1.7070+000	0.69900+000
46	1.63400+000	0.58000+000	1.6345+000	0.69850+000
47	1.62100+000	0.58000+000	1.6220+000	0.69800+000

AXI-SYMMETRIC TRANSIENT HEATING AND MATERIAL ABILATION PROGRAM

10	0.6000+000	0.5000+000	0.6000+000	0.6000+000
11	0.5930+000	0.5330+000	0.5940+000	0.6975+000

12	0.5700+000	0.5720+000	0.5790+000	0.6980+000
13	0.5620+000	0.5000+000	0.5990+000	0.9000+000
14	0.5550+000	0.4000+000	0.6000+000	0.9000+000
15	0.5480+000	0.3000+000	0.6000+000	0.9000+000
16	0.5410+000	0.2000+000	0.6000+000	0.9000+000
17	0.5340+000	0.1000+000	0.6000+000	0.9000+000
18	0.5270+000	0.0000+000	0.6000+000	0.9000+000

19	0.5200+000	0.8000+000	0.6000+000	0.9000+000
20	0.5130+000	0.8100+000	0.6000+000	0.9000+000
21	0.5060+000	0.8200+000	0.6000+000	0.9000+000
22	0.4990+000	0.8300+000	0.6000+000	0.9000+000
23	0.4920+000	0.8400+000	0.6000+000	0.9000+000
24	0.4850+000	0.8500+000	0.6000+000	0.9000+000
25	0.4780+000	0.8600+000	0.6000+000	0.9000+000
26	0.4710+000	0.8700+000	0.6000+000	0.9000+000
27	0.4640+000	0.8800+000	0.6000+000	0.9000+000
28	0.4570+000	0.8900+000	0.6000+000	0.9000+000
29	0.4500+000	0.9000+000	0.6000+000	0.9000+000
30	0.4430+000	0.9100+000	0.6000+000	0.9000+000
31	0.4360+000	0.9200+000	0.6000+000	0.9000+000
32	0.4290+000	0.9300+000	0.6000+000	0.9000+000
33	0.4220+000	0.9400+000	0.6000+000	0.9000+000
34	0.4150+000	0.9500+000	0.6000+000	0.9000+000
35	0.4080+000	0.9600+000	0.6000+000	0.9000+000
36	0.4010+000	0.9700+000	0.6000+000	0.9000+000
37	0.3940+000	0.9800+000	0.6000+000	0.9000+000
38	0.3870+000	0.9900+000	0.6000+000	0.9000+000
39	0.3800+000	1.0000+000	0.6000+000	0.9000+000
40	0.3730+000	1.0100+000	0.6000+000	0.9000+000
41	0.3660+000	1.0200+000	0.6000+000	0.9000+000
42	0.3590+000	1.0300+000	0.6000+000	0.9000+000
43	0.3520+000	1.0400+000	0.6000+000	0.9000+000
44	0.3450+000	1.0500+000	0.6000+000	0.9000+000
45	0.3380+000	1.0600+000	0.6000+000	0.9000+000
46	0.3310+000	1.0700+000	0.6000+000	0.9000+000
47	0.3240+000	1.0800+000	0.6000+000	0.9000+000
48	0.3170+000	1.0900+000	0.6000+000	0.9000+000
49	0.3100+000	1.1000+000	0.6000+000	0.9000+000
50	0.3030+000	1.1100+000	0.6000+000	0.9000+000
51	0.2960+000	1.1200+000	0.6000+000	0.9000+000
52	0.2890+000	1.1300+000	0.6000+000	0.9000+000
53	0.2820+000	1.1400+000	0.6000+000	0.9000+000
54	0.2750+000	1.1500+000	0.6000+000	0.9000+000
55	0.2680+000	1.1600+000	0.6000+000	0.9000+000
56	0.2610+000	1.1700+000	0.6000+000	0.9000+000
57	0.2540+000	1.1800+000	0.6000+000	0.9000+000
58	0.2470+000	1.1900+000	0.6000+000	0.9000+000
59	0.2400+000	1.2000+000	0.6000+000	0.9000+000
60	0.2330+000	1.2100+000	0.6000+000	0.9000+000
61	0.2260+000	1.2200+000	0.6000+000	0.9000+000
62	0.2190+000	1.2300+000	0.6000+000	0.9000+000
63	0.2120+000	1.2400+000	0.6000+000	0.9000+000
64	0.2050+000	1.2500+000	0.6000+000	0.9000+000
65	0.1980+000	1.2600+000	0.6000+000	0.9000+000
66	0.1910+000	1.2700+000	0.6000+000	0.9000+000
67	0.1840+000	1.2800+000	0.6000+000	0.9000+000
68	0.1770+000	1.2900+000	0.6000+000	0.9000+000
69	0.1700+000	1.3000+000	0.6000+000	0.9000+000
70	0.1630+000	1.3100+000	0.6000+000	0.9000+000
71	0.1560+000	1.3200+000	0.6000+000	0.9000+000
72	0.1490+000	1.3300+000	0.6000+000	0.9000+000
73	0.1420+000	1.3400+000	0.6000+000	0.9000+000
74	0.1350+000	1.3500+000	0.6000+000	0.9000+000
75	0.1280+000	1.3600+000	0.6000+000	0.9000+000
76	0.1210+000	1.3700+000	0.6000+000	0.9000+000
77	0.1140+000	1.3800+000	0.6000+000	0.9000+000
78	0.1070+000	1.3900+000	0.6000+000	0.9000+000
79	0.1000+000	1.4000+000	0.6000+000	0.9000+000
80	0.0930+000	1.4100+000	0.6000+000	0.9000+000
81	0.0860+000	1.4200+000	0.6000+000	0.9000+000
82	0.0790+000	1.4300+000	0.6000+000	0.9000+000
83	0.0720+000	1.4400+000	0.6000+000	0.9000+000
84	0.0650+000	1.4500+000	0.6000+000	0.9000+000
85	0.0580+000	1.4600+000	0.6000+000	0.9000+000
86	0.0510+000	1.4700+000	0.6000+000	0.9000+000
87	0.0440+000	1.4800+000	0.6000+000	0.9000+000
88	0.0370+000	1.4900+000	0.6000+000	0.9000+000
89	0.0300+000	1.5000+000	0.6000+000	0.9000+000
90	0.0230+000	1.5100+000	0.6000+000	0.9000+000
91	0.0160+000	1.5200+000	0.6000+000	0.9000+000
92	0.0090+000	1.5300+000	0.6000+000	0.9000+000
93	0.0020+000	1.5400+000	0.6000+000	0.9000+000
94	0.0000+000	1.5500+000	0.6000+000	0.9000+000

ALMOT-ERU AXI-SYMMETRIC TRANSIENT HEATING AND METALLIC ABLATION PROGRAM

SECTION 1 - ABLATION CALCULATION

..... P.000 S.E.P.O.N.C.....

.....
GENERAL
TIME 0107.5UR 0YCT,TAT 0YCT,ENR CRNODE ITEL 10ME D-TIME ACT D-TIME
0.2000+001 0.5449+002 0.3321+002 0.1039+001 0 0 7 0.54721+001 0.51309-002

HEATED SURFACE

*****MISCELLANEOUS SURFACE DATA*****

RO. C/L	Q/TM	SURF ITEL	SURF TEMP(R)	P EDGE (EVU/LR)	P-R-ALL (HTU/LM)	W PRIME TOT	MASS COEFF (LH/FT2-SEC)	CH/CM	PRESSURE (ATM)	RADIUS (IN)	Z
11	1	1	5907.6	2020.00	1432.04	0.2468	0.49860	0.94489	0.506+002	0.770+000	0.758+001
11	2	1	5917.4	2086.00	1434.71	0.2493	0.50079	0.94462	0.475+002	0.687+000	0.250+000
11	3	1	5892.2	1984.00	1422.74	0.2463	0.71000	0.94483	0.410+002	0.619+000	0.465+000
11	4	1	5881.3	1937.00	1417.48	0.2433	0.74708	0.94576	0.315+002	0.593+000	0.498+000
11	5	1	5717.4	1787.00	1329.27	0.2137	0.61411	0.95140	0.178+002	0.609+000	0.925+000
11	6	1	5671.6	1761.00	1314.87	0.2044	0.58+004	0.95372	0.16+002	0.650+000	0.111+001

*****SURFACE RATE QUANTITIES*****

--RECESSION RATES--		--MASS RATES--		--SURFACE ENERGY FLUX RATES--		CONDUCTION			
RO. C/L	CENTER LINE NORMAL (MILS/SEC)	WDOT TOTAL (LH/FT2-SEC)	WDOT TOTAL (MDOY TPEEM)	CONVERTED TA	CONVERSION GENERATION (HTU/FT2-SEC)	RADIATION ABSORBED	RADIATION EMITTED	CONDUCTION AWAY	
11	1	11.343938	10.778709	0.323+000	0.123+000	0.486+003	0.258+003	0.511+003	0.424+003
11	2	12.977984	13.095356	0.158+000	0.152+000	0.588+003	0.222+003	0.515+003	0.438+003
11	3	15.515767	15.510156	0.177+000	0.177+000	0.669+003	0.440+003	0.506+003	0.410+003
11	4	15.942111	15.918972	0.182+000	0.182+000	0.644+003	0.458+003	0.502+003	0.389+003
11	5	11.502214	11.495942	0.134+000	0.131+000	0.448+003	0.384+003	0.449+003	0.340+003
11	6	10.458619	10.410050	0.110+000	0.119+000	0.437+003	0.324+003	0.435+003	0.384+003

*****SURFACE TIME INTEGRATED QUANTITIES*****

--RECESSION TOTALS--		--MASS ABLATION TOTALS--		--SURFACE ENERGY FLUX TOTALS--		CONDUCTION			
RO. C/L	CENTER LINE NORMAL (MILS)	WDOT TOTAL (LH/FT2-SEC)	WDOT TOTAL (MDOY TPEEM)	CONVERTED TA	CONVERSION GENERATION (HTU/FT2-SEC)	RADIATION ABSORBED	RADIATION EMITTED	CONDUCTION AWAY	
11	1	21.573	21.5173	0.334+002	0.131+002	0.578+004	0.248+004	0.594+004	0.579+004
11	2	25.547	25.0657	0.170+002	0.179+002	0.748+004	0.340+004	0.626+004	0.472+004
11	3	29.5792	29.5742	0.217+002	0.217+002	0.895+004	0.419+004	0.631+004	0.452+004
11	4	30.3612	30.3612	0.223+002	0.203+002	0.791+004	0.474+004	0.572+004	0.577+004
11	5	21.7449	21.7449	0.140+002	0.149+002	0.504+004	0.403+004	0.45+004	0.520+004
11	6	20.0504	20.0854	0.145+002	0.165+002	0.425+004	0.238+002	0.162+002	0.449+004

AXI-SYMMETRIC TRANSIENT HEATING AND NEUTRAL FLAVINITY PROGRAM

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*****INPUT DATA**

NODE	TEMP	NODE	TEMP	NODE	TEMP	NODE	TEMP	NODE	TEMP
1	0.5300+003	1	0.5300+003	1	0.5351+003	4	0.5913+003	5	0.7252+003
6	0.8717+003	7	0.1110+004	8	0.4395+004	9	0.2271+004	10	0.2752+004
11	0.5627+004	12	0.1640+004	1	0.3800+003	2	0.5100+003	3	0.5334+004
4	0.2691+003	5	0.2229+004	6	0.2224+003	7	0.1101+004	8	0.1375+004
9	0.2229+004	10	0.2229+004	11	0.5533+003	12	0.6228+003	1	0.7647+004
2	0.1532+003	3	0.1241+004	4	0.2070+004	5	0.3557+004	6	0.5476+004
7	0.1041+004	8	0.1241+004	9	0.2070+004	10	0.3557+004	11	0.5476+004
12	0.1041+004	1	0.7358+003	2	0.3800+003	3	0.5100+003	4	0.5334+004
5	0.6830+003	6	0.7358+003	7	0.1000+004	8	0.1704+004	9	0.1934+004
10	0.3240+004	11	0.5741+004	12	0.1000+004	1	0.7467+003	2	0.5300+004
3	0.5313+003	4	0.5476+003	5	0.4056+003	6	0.4764+004	7	0.5014+004
8	0.1194+004	9	0.1887+004	10	0.2988+004	11	0.4764+004	12	0.5014+004
1	0.7578+003	2	0.5300+003	3	0.5315+003	4	0.5494+003	5	0.6138+004
6	0.7578+003	7	0.1038+004	8	0.4221+004	9	0.1934+004	10	0.2131+004
11	0.4933+004	12	0.4933+004	1	0.4933+004	2	0.4933+004	3	0.4933+004

*****OPTION 3 - COOLDOWN CALCULATION 5.00 SECONDS*****

TIME 0TOT,SUR 0.5000+001 0.6664+002 0.6529+002 0.1021+001 8 4 144 0.4240+001 0.17000+001
 GENERAL
 CASE EMER CMODE TYPE NONE D-TIME ACT D-TIME

*****HEATED SURFACE**

*****ISFILL'EDS SURFACE DATA*****

RD.	COL	Q-TY	SURF	SURF	F-EDGE	W-ALL	B-PRIME	MASS	COEFF	CM/CMC	PRESSURE	RADIUS
			ITEM	TEMP(R)	(BTU/LB)	(BTU/LB)	TOT	(LP/FT-SEC)	(LP/FT-SEC)		(ATM)	(IN)
9	1	3	4	1779.6	0.	0.	0.	0.	0.	0.	0.100+005	0.793+000
9	2	3	4	1753.0	0.	0.	0.	0.	0.	0.	0.100+005	0.712+000
9	3	3	4	1741.9	0.	0.	0.	0.	0.	0.	0.100+005	0.649+000
9	4	3	4	1776.2	0.	0.	0.	0.	0.	0.	0.100+005	0.624+000
9	5	3	4	2133.6	0.	0.	0.	0.	0.	0.	0.100+005	0.632+000
9	6	3	4	2024.3	0.	0.	0.	0.	0.	0.	0.100+005	0.671+000
9	7	3	4	2024.3	0.	0.	0.	0.	0.	0.	0.100+005	0.671+000
9	8	3	4	2024.3	0.	0.	0.	0.	0.	0.	0.100+005	0.671+000
9	9	3	4	2024.3	0.	0.	0.	0.	0.	0.	0.100+005	0.671+000
9	10	3	4	2024.3	0.	0.	0.	0.	0.	0.	0.100+005	0.671+000
9	11	3	4	2024.3	0.	0.	0.	0.	0.	0.	0.100+005	0.671+000
9	12	3	4	2024.3	0.	0.	0.	0.	0.	0.	0.100+005	0.671+000

ALUMINUM AXI-SYMMETRIC TRANSIENT HEATING AND MATERIAL DEVIATION PROGRAM

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*****SURFACE HEAT QUANTITIES*****

NO.	CENTER LINE	RECESSION RATES (MILS/SEC)	MASS LOSS (LB/FT2-SEC)	NET TOTAL (BTU/FT2-SEC)	CONVECTED T _a	CHEMICAL COMPOSITION	RADIATION ABSORBED (BTU/FT2-SEC)	RADIATION EMITTED	CONDUCTION AWAY
1	0.0	43.8677	0.276002	0.117002	0.117002	0.520000	0.164002	0.116002	0.147000
2	0.0	50.6111	0.347002	0.151002	0.151002	0.700000	0.181002	0.128002	-0.231000
3	0.0	59.6802	0.445002	0.180002	0.180002	0.970000	0.184002	0.129002	-0.733000
4	0.0	61.4642	0.497002	0.212002	0.212002	1.320000	0.168002	0.117002	-0.156000
5	0.0	44.3446	0.332002	0.146002	0.146002	0.860000	0.172002	0.108002	-0.126000
6	0.0	41.4957	0.195002	0.082002	0.082002	0.483000	0.196002	0.075002	0.000000

*****SURFACE VINE INTEGRATED QUANTITIES*****

NO.	CENTER LINE	RECESSION TOTALS (MILS)	MASS ABLATION TOTALS (LB/COL)	NET TOTAL (BTU/COL)	CONVECTED T _a	CHEMICAL COMPOSITION	ENERGY FLUX RADIATION ABSORBED	ENERGY FLUX RADIATION EMITTED	CONDUCTION AWAY
1	43.8677	43.8677	0.276002	0.276002	0.117002	0.520000	0.164002	0.116002	0.112000
2	50.6111	50.6111	0.347002	0.347002	0.151002	0.700000	0.181002	0.128002	0.133000
3	59.6802	59.6802	0.445002	0.445002	0.180002	0.970000	0.184002	0.129002	0.120000
4	61.4642	61.4642	0.497002	0.497002	0.212002	1.320000	0.168002	0.117002	0.111000
5	44.3446	44.3446	0.332002	0.332002	0.146002	0.860000	0.172002	0.108002	0.091000
6	41.4957	41.4957	0.195002	0.195002	0.082002	0.483000	0.196002	0.075002	0.090000

*****DEPTH DATA***

NOTE	NO.	DEPTH	TEMP	NO.	TEMP	NO.	TEMP	NO.	TEMP
1	1	0.530000	0.530000	1	0.423100	1	0.490500	1	0.131900
6	1	0.153600	0.153600	1	0.273900	1	0.178000	1	0.000000
11	1	0.577700	0.577700	2	0.530000	2	0.550100	2	0.605800
9	2	0.577700	0.577700	2	0.451500	2	0.171400	2	0.175300
2	3	0.530100	0.530100	3	0.403700	3	0.111700	3	0.145100
7	3	0.149800	0.149800	3	0.530100	3	0.584300	3	0.761100
12	3	0.173400	0.173400	4	0.530100	4	0.173400	4	0.177600
5	4	0.570600	0.570600	4	0.402000	4	0.158400	4	0.530100
3	5	0.131000	0.131000	5	0.402000	5	0.158400	5	0.171300
4	6	0.157200	0.157200	6	0.478400	6	0.749400	6	0.102800
1	6	0.157200	0.157200	6	0.478400	6	0.749400	6	0.102800
5	6	0.157200	0.157200	6	0.478400	6	0.749400	6	0.102800
11	6	0.157200	0.157200	6	0.478400	6	0.749400	6	0.102800

APPENDIX D

FORTRAN IV LISTING OF ASTHMA COMPUTER PROGRAM

CASHTMA

C-----MODIFICATION OF AATT TO ACCOUNT FOR RECEIVING SURFACES
 C-----DIMENSIONED AS CORNERS
 COMMON CR(4,2),CZ(4,2),AA(4,2),AE(4,2),AC(4,2),AD(4,2)
 C-----DIMENSIONED AS NODES
 COMMON MATL(200),KTH(2,0),KSH(2,0),KAL(2,0),KTU(200),CON(200),
 1TA(200),TB(200),CRA(200),CRB(200),PLA(200),PLB(200),PLC(200),
 2PLD(200),VOL(200),RA(200),RB(200),CAPI(200),ENT(200),ONPI(200),
 3VF1(200),VF3(200)
 C-----DIMENSIONED AS SURFACE NODES (NUMBER OF COLUMNS)
 COMMON KSUR(40),TS(4,0),DSDT(4,0),DSDT5(4,0),DS(4,0),DST(4,0),J(4,0)
 COMMON PLS(4,0),DSTT(4,0),QCONDT(4,0),QCNVT(4,0),QCNVT(4,0),QCHM(4,0),
 1QCHM(4,0),GRAB(4,0),ORABT(4,0),ORAD(4,0),SRAD(4,0),DSDTRN(4,0),DSN(4,0)
 COMMON CVDOT(4,0),HEDG(4,0),CMT(4,0),HAL(4,0),GZ(4,0),PR(4,0)
 COMMON I(4,0),S2(4,0),SR(4,0),EMN(4,0),ITSR(4,0)
 COMMON IAPL(4,0),KDROPI(4,0)
 C-----DIMENSIONED AS PROPERTY TABLES, ENTRIES X MATERIALS
 COMMON I(15,6),RT(15,6),CPT(15,6),CNI(15,6),CNI2(15,6),
 1EPT(15,6),TTMX(6)
 DIMENSION THZ(15)
 C-----DIMENSIONED AS TIME TABLES, ENTRIES X TABLE NO
 COMMON TH(25,10),CHI(25,1,0),REI(25,1,0),TOR(25,1,0),IPI(25,1,0),
 1TBRP(25,1,0)

DIMENSION IOPT(25)
 C-----MISCELLANEOUS QUANTITIES
 COMMON RECORD(36),IMPR(5),MPR(5),NNPR(5)
 COMMON VITER(51),EITER(51),IAB,RO
 COMMON VKLN,CM,CM,CM,CH
 COMMON TH,DTH,PRT,THI,THF,DLTH,ETA,ITS,DTHS
 COMMON FV,FT,PIB,ZRC,KOOP,MM,NN,NAT,NHT,GNTS,QJUM,SIG
 COMMON KRESC,KSLOP
 COMMON/ENERGY/TPR(5),NMG(5),TMG(8,5),NLO(8,5),NHI(8,5),
 1KHI(8,5),TTSEN(25,5),THSEN(25,5),TCPSEN(25,5),
 2TLMC(25,8,5),ISEN(5),TIS(25,8,5),TCHEM(25,8,5),
 3NPR

DIMENSION TASE(25,5),TCLSEN(25,5),TJURF(25),TSEM(25),I2(25)
 COMMON/LK/KOUT,LEX,DEN,VR,HI(4,0),ILO(4,0),IR(4,0)
 DIMENSION TPN(8)
 DIMENSION PRTI(4),TPTCG(4)
 DATA BLANK/64 /
 300 FORMAT(2I3,1,6E6,4,4I1)
 301 FORMAT(4E7,5)
 302 FORMAT(4I1,12,6E6,4)
 303 FORMAT(12,F8,2,5F10,5)
 304 FORMAT(12,6E6,4)
 305 FORMAT(1I,25X,71HAEROTHERM AXI-SYMMETRIC TRANSIENT HEATING AND MA
 1TERRIAL RELATIO: PROGRAM/113X,4HPAGE,13//)
 306 FORMAT(12A6)

307 FORMAT(11H INPUT DATA//27H DIMENSIONS OF INPUT DATA//114H AN 0016
 1ME DENSITY SEC TEMPERATURE DEG R TIAN 0017
 2 STU PER LB DEG R CONDUCTIVITY LB PER CUBIC FT/112H SPECIFIC HEAT AN 0018
 3 SIVITY DIMENSIONLESS/1.5H HEAT COEFFICIENT LB PER SQ FTAY EMISAN 0020
 4SIVITY DIMENSIONLESS/1.5H HEAT COEFFICIENT LB PER SQ FTAY 0021

AN 0012

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5 SEC ENTHALPY BTU PER LB          NODAL COORDINATES AN 0022
6 INCHES/44H RESISTANCES          SQ FT SEC JEG R PER BTU//2JM PRGM AN 0023
7 BLEH CONSTANTS//785H          HMAX IN/AX          INIT TIME FINL TIME FRNT AN 0024
8 TIME TIME INCR TIME CNJT          VKIN//
307 FORMAT//22H          NODAL COORDINATES//5A,1H,5X,1MJ,5X,6HKC(IN),7X,6HZ
1C(IN),7X,6HKC(IN),7X,6ZK(IN)//
308 FORMAT//13H          NODAL DATA//97H          MAIL THTA SIDE ENTB HTTB HEAT CNSAN 0028
1T          INIT TEMP          CONT RES B          VFI          VF3//
309 FORMAT//29H          MATERIAL PROPERTIES TABLES//
310 FORMAT//15H          MATERIAL NO.13//73H          TEMP          DENSITY          SPEC HAN* 0030
1EAT          CONDUCT          EMISSIV          CONDUCT12//
311 FORMAT//25H          HEATING TABLES, OPTION12//
312 FORMAT//17H          HEAT TABLE NO.13//5JH          TIME          HEAT COEFF          RECOVAN 0034
1 ENTH          RAD FACTOR//
313 FORMAT//22H          WALL ENTHALPY TABLE//34H          TEMP          ENTH 1          AN 0036
1 ENTH 2//          AN 0037
314 FORMAT//1X,2I5,5X,6(E11,4,1X)//
315 FORMAT//1X,6E12,4)
316 FORMAT//1X,5I5,2X,6(E11,4,1X)//
317 FORMAT//17H          HEAT TABLE NO.13//37H          TIME          TEMP          RECOVAN 0041
1 ENTH//
318 FORMAT//3X,13,3X,13,1X,4E13,5)
319 FORMAT//12H          OUTPUT DATA//28H          DIMENSIONS OF OUTPUT DATA//118H          AN 0044
1 TIME          SEC          QTOT,SUR,INT          BTU          AN 0045
2          CONVective HEAT COEFF          LB PER SQ FT SEC//115H          TEMPERATAN 0046
3URE          DEG R          QNET AND QCONV          BTU PER SQ FT SEC          AN 0047
4          QTOT          BTU PER SQ FT//          AN 0048
320 FORMAT//216)
321 FORMAT//12,10E7,5)
322 FORMAT//30H          M VS          TEMPERATURE, TABLE NO. 12//34HTEMP9X1HH8X4HTEMP
19X1HH8X4HTEMP9X1HH8X4HTEMP9X1HH8X4HTEMP9X1HH)
323 FORMAT//23H          M VS          TIME, TABLE NO. 12//34HTIME9X1HH8X4HTIME9X1HH8X
14HTIME9X1HH8X4HTIME9X1HH8X4HTIME9X1HH)
324 FORMAT//10E11,3)
325 FORMAT// 111H          NODE          TEMP          NODE          TEMP          AN 0054
1          NODE          TEMP          NODE          TEMP          AN 0055
326 FORMAT//10X,5(15,14,4E12,4))          AN 0056
327 FORMAT//16X,
1          72H          TIME          QTOT,SUR          QTOT,INT          CNSV          ENER          CRNODEAN 0057
2          ITER          NODE          D-TIME,2X,11H          ACT          D-TIME//
328 FORMAT//16X,4(E11,4,1X),2I3,15,2E13,5//
329 FORMAT//11X,2I4,5E12,4)
330 FORMAT (1H //23X40H---TIME DEPENDENT BOUNDARY CONDITIONS---/1H )
331 FORMAT (9X,4HTIME,8X,4HPROB,3X,8HRECOVERY,3X,9HRADIATION,4X,4HMHEAT
15X,8HPRESSURE,3X,7HLOWING/9X,5H(SEC),7X,4HOPTN,3X,8HENTHALPY,3X,
29HHEAT RATE,4X,5HCOEFF,14X,9HREDUCTION/28X,8H(BTU/LB),2X,11H(BTU/S
30 FT-1),1X,10H(LB/50 FT-3),5H(AI),3A,9HPARAMETER /40X,7HSECOND),
44X,7HSECOND))
332 FORMAT (6X,F8,2,8X,12,4X,2(F6,2,3X),F8,4,3X,F8,3,3X,F8,3)
333 FORMAT (1H /9X,69HCH/CHO = PHI/(EXP(PHI))-1.) WHERE PHI = 2.*9RP*M
1DOT/CHO. BRP In TABLE)
334 FORMAT//27X3H---SURFACE EQUILIBRIUM DATA---)
335 FORMAT (9X,4HTIME,5A,4HPROB,3X,7H(SURFACE,4X,7HSURFACE/9X,5H(SEC),

```

17X,4HOPTN,5X,6HTEMP,5X,9HRECESSION/28X,7H(DEG R),5X,6H(MILS))
556 FORMAT (9X,4HTIME,8X,4HPRCB,5X,4HVIEW,5X,9HRADIATION/9X,5H(SEC),
17X,4HOPTN,4X,6HFACTOR,4X,9HHEAT RATE/38X,11H(FTU/5Q FT-740X,
27HSECOND))

5780 FORMAT(E6.4,6X,E8.5,E6.4,F4.2,E7.5,6X,2E8.5,A6,11)
5781 FORMAT(E6.4,E8.5,6X,E6.4,F4.2,E7.5,6X,2E8.5,A6,11)
5782 FORMAT(E6.4,8X,E6.4,E6.4,F4.2,E7.5,6X,2E8.5,A6,11)
5789 FORMAT(/6X,14HKINETICS PRM =E1,3,8X,10HPRESSURE =,F9.4,4H ATM//
17X,2(4HTEMP,5X,26HM-DOT- CHEM,PROD SURFACE,3X)/6X,2(36H(DEG R)
2CHAR/CM (3TU/LE) SPECIES,2X))

5790 FORMAT (6X,26HNO RADIUS CORRECTION ON CH)
5791 FORMAT(3F8.5,F9.4,F5.3,2F9.3,12,A6)
5792 FORMAT(/6X,3HP =,F9.4,4H ATX/6X,3(25HTEMPERATURE EDGE ENTH)/
16X,3(25H (DEG R) AT T-WALL))
5793 FORMAT (/6X,37HBAD SURFACE EQUILIBRIUM TABLE OF TYPE,12)
5794 FORMAT (/6X,74HEQUAL MASS AND HEAT TRANSFER COEFFICIENTS AND EQUA
1L 3 DIFFUSION COEFFICIENTS)
5795 FORMAT(5X,F8.2,2X,F7.4,2X,F9.2,4X,A6,1X,F8.2,2X,F7.4,2X,F8.2,4X,A6
1)

5796 FORMAT(2F10,3,319X,11))
5797 FORMAT(/6X,4SHRATIO OF MASS TO HEAT TRANSFER COEFFICIENTS =,F6.3/
1 6X,28HUNEQUAL DIFFUSION EXPONENT =,F6.3)
5798 FORMAT
22,3X,F9.2,4X,F9.2)

5799 FORMAT (6X,66HHEAT TRANSFER COEFFICIENT MULTIPLIED BY (R INITIAL/R
: CURRENT)**1.8)
581 FORMAT(/34X2-HBACK WALL CONVECTION,10X9HBACK WALL,10X9HRESERVOIR/
132X23HCOEF BTU/FTSQ-SEC-DEG RBXICHEMISIVITYBXIIIITEMPERATURE/
237X10,4,18XF6,3,1VXF10,2)

819 FORMAT(25HOUT OF RANGE OF H TABLES/5X7H TEMP= E9.4,10X6HTIME= E9.
14)
820 FORMAT(54H IS LARGER THAN THE LAST ENTRY IN THE WALL ENTH. TABLE)
821 FORMAT(55H IS SMALLER THAN THE FIRST ENTRY IN THE WALL ENTH. TABLE)
822 FORMAT(24HOTRE TEMPERATURE OF NODEZ14)
823 FORMAT(49HIS LARGER THAN THE LAST ENTRY IN MATL. PROP. TAB.13)
824 FORMAT(51HIS SMALLER THAN THE FIRST ENTRY IN MATL. PROP. TAB.13) AN
CALL DEFIO(0,5,8)
CALL DEFIO(0,6,5)

C-----GENERAL CONSTANTS
FV=5
FT=.083333333
RANK = 459.688
ZRO=0.0
PI3=.021816616
INCH=5
INPUT=5
KOUT=6
SIG=.481E-12
NPG=1

AN 0063
AN 0064
AN 0065
AN 0066

C-----MAIN INPUT BLOCK INCLUDING OUTPUT LISTING OF INPUT
225 WRITE (6,304)
READ (5,305)
WRITE (6,305)

(RECORD(1),1,1,36)
(RECORD(1),1,1,36)

AN 0067
AN 0069

```

WRITE (6,306)
READ (5,300) YR,NA,THI,THF,THP,DLI,ETA,2,ERP,HCONV,EPS,TKES,
1 KSTRP,KRESC,KSLCP
KSLCP=KSLCP+1
KRESC=KRESC+1
IF(ETA) 2253,2253,2254
2253 ETA=.75
2254 CONTINUE
WRITE(6,314) NA,NA,THI,THF,THP,DLI,ETA,VR,K
WRITE(KOUT,581) HCONV,EPS,TKES
IF(THP) 2255,2255,2267
2257 TPTCG(1)=1.E+5~
2255 READ(INPUT,2252) (PRTI(I),TPTCG(I),I=1,4)
TPTCG(4)=THF
DO 2257 I=1,3
IF(TPTCG(I)) 2258,2255,2257
2258 TPTCG(I)=THF
2257 CONTINUE
CALL LCOUNT (5,LCT,NPG,RECORD(35))
WRITE (KOUT,2259)
FORMAT(/,4X,21HOUTPUT TIME INTERVALS)
WRITE(KOUT,226~) PRTI(1),THI,TPTCG(1)
226~ FORMAT(21X,17HOUTPUT INTERVAL =,F7.4,13H SECONDS FROM,F9.4,
114H SECONDS UNTIL,F9.4,8H SECONDS)
IF(TPTCG(1)-THF) 2261,2263,2263
2261 DO 2262 I=2,4
IF(TPTCG(I)-THF) 2266,2264,2264
2264 CALL LCOUNT(1,LCT,NPG,RECORD(35))
WRITE(KOUT,2265) PRTI(1),TPTCG(I-1)
2265 FORMAT(21X,17HOUTPUT INTERVAL =,F7.4,13H SECONDS FROM,
1F9.4,25H SECONDS UNTIL FINAL TIME//)
GO TO 2263
2266 CALL LCOUNT (1,LCT,NPG,RECORD(35))
2262 WRITE(KOUT,226~) PRTI(1),TPTCG(I-1),TPTCG(I)
2263 TPTCG(I)=THF
IF(THP) 2268,2268,2256
2268 THF=T:H
2256 CONTINUE
IF(KSTRP-2) 2250,2251,2251
2251 READ (5,2252) (TPN(I),I=1,6)
2252 FORMAT (8F1.6)
2253 CONTINUE
N=NN+1
M=MM+1
K=J
DO 200 J=1,N
DO 200 I=1,M
K=K+1
READ (5,301) AA(K),A5(K),AC(K),AJ(K)
CR(K)=A5(K)
CZ(K)=A2(K)
200 CALL LCOUNT(2,NA,THI-3,LCT,NPG,RECORD(35))

```

AN 0073
AN 0074
AN 0075
AN 0076
AN 0077
AN 0078

```

WRITE (6,308)
K=0
DO 201 J=1,NM
DO 201 I=1,MM
K=K+1
READ (5,302)
1 TAI(K),CRA(K),CRB(K),VF1(K),VF3(K)
201 WRITE (6,316)
1(K),TAI(K),CRA(K),CRB(K),VF1(K),VF3(K)
CALL LCOUNT(4,LCT,NPG,RECORD(35))
WRITE (6,309)
READ (5,303)
J=0
202 J=J+1
WRITE (6,310) J
I=1
203 READ(5,303) NC,TT(I,J),RT(I,J),CPT(I,J),CNT(I,J),EPT(I,J),CNT2(I,
1J)
RT(I,J)=RT(I,J)
CALL LCOUNT(1,LCT,NPG,RECORD(35))
WRITE (6,315)
1(J),CNT2(I,J)
IF (NC) 205,2-4,205
204 I=I+1
GO TO 203
205 TTX(I)=TT(I,J)
ILO(J)=I
ITI(J)=I
IR(I)=1
IF (J-1) 2053,2050,2-53
2050 THZ(I)=0.
ILO(I)=I
IR(I)=1
DO 2051 L=2,I
2051 THZ(L)=THZ(L-1)+(CPT(L,I)+CPT(L-1,I))/2.*(TT(L,I)-TT(L-1,I))
IZ=536.
CALL LOOK(1,IZ,TT(I,I),THZ(I),0.0,0.0,SH,MA,1)
DO 2052 L=1,I
2052 THZ(L)=THZ(L)-HSH
2053 CONTINUE
IF (J-NM) 202,2-6,2-6
C-----TIME (HEATING) TABLES
206 CALL LCOUNT(4,LCT,NPG,RECORD(35))
WRITE(6,534)
J=0
KNN=J
207 J=J+1
NTH=J
IS=J
KN=J
NOPT=0
371 NTH=NTH+1

```

AN 0080
AN 0081
AN 0082
AN 0083
AN 0084

0087

AN 0089

AN 0091

AN 0092

AN 0093

AN 0094

0097

AN 0099

AN 0100

AN 0101

AN 0102

```

READ(5,303) NC,THT(NTH,J),RET(NTH,J),TUR(NTH,J),CHT(NTH,J),
1T0I(NTH,J),TERP(NTH,J)
IF(TERP(NTH,J)) 374,375,374
375 TRP(NTH,J)=EXP
374 IJ=1
IF(CHT(NTH,J)) 342,342,343
342 IJ=2
IF(RET(NTH,J)-2.) 344,344,343
344 IJ=3
343 IOPT(NTH)=IJ
IF(IJ-IS) 345,346,345
345 NOPT=NOPT+1
IS=IJ
346 IF(NC) 372,371,372
372 ILO(J+20)=1
!MI(J+20)=NTH
IR(J+20)=1
CALL LCOUNT(2+NTH+NOPT*3,LCT,NPG,RECORD(35))
WRITE(6,538) J
5380 FORMAT(15X,18HTIME TABLE NUMBER ,I2/)
IS=J
DO 3476 I=1,NTH
IJ=IOPT(I)
IF(IJ-IS) 347,349,347
347 IS=IJ
GO TO (3471,3472,3473),IJ
3471 KN=1
GO TO 3474
3472 WRITE(KOUT,552)
GO TO 3475
3473 WRITE(KOUT,556)
GO TO 3475
349 GO TO (3474,3475,3475),IJ
3474 WRITE(6,536) THT(I,J),IJ,RET(I,J),TGR(I,J),CHT(I,J),
1TPI(I,J),TRP(I,J)
GO TO 3476
3475 WRITE(6,536) THT(I,J),IJ,RET(I,J),TOR(I,J)
3476 CONTINUE
NHT=J
IF(KN) 3732,3732,373
373 DO 3731 I=1,NTH
3731 TPI(I,J)=ALOG(AMAX1(TPI(I,J),.000001))
KN=1
3732 IF(NC) 207,207,3733
3733 IF(KNN) 139,139,328
328 CALL LCOUNT(2,LCT,NPG,RECORD(35))
WRITE(6,537)
C-----SURFACE THERMOCHEMISTRY TABLES
HEAD(INPUT,5796) CMHS,VFZ,NR,NST,KTCTB
KTCTB=KTCTB+1
VKIN=VFZ
IF(NST) 290,290,29.1

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```

2900 CMH=CMHS
GO TO 2902
2901 IF(KNST-777) 2909,2903,2909
2903 IF(CMH-CMHS) 2907,2905,2907
2905 CALL LCOUNT(4,LCT,NPG,RECORD(35))
WRITE(KOUT,2906)
2906 FORMAT(/10X,5#SURFACE TABLES ARE THE SAME AS IN PREVIOUS PROBLEM
1//)
GO TO 1390
2907 CALL LCOUNT(4,LCT,NPG,RECORD(35))
WRITE(KOUT,2908)
2908 FORMAT(/10X,7#PREVIOUS SURFACE TABLES CALLED FOR BUT CM/CH RATIO
1 HAS CHANGED, QUIT JOB//)
STOP
2909 CALL LCOUNT(4,LCT,NPG,RECORD(35))
WRITE(KOUT,2910)
2910 FORMAT(/10X,7#PREVIOUS SURFACE TABLES CALLED FOR BUT THIS IS FIR
1ST PROBLEM, QUIT JOB//)
STOP
2902 KNST=777
IF (RSV) 3280,3280,3281
3280 NR=0
3281 CONTINUE
NSEN=-1
IP=1
IPN=1
I=1
IN=1
J=0

2800 J=J+1
GO TO (2911,2912,2913,2914),KTCIB
2911 READ(INCH,5780) PSV,TLMC(J,I,IP),DMS,WLQ,TT(S(J,I,IP),TCHEM(J,I,IP)
1,TSEN(J),TSURF(J),JNG
GO TO 2916
2912 READ(INCH,5781) PSV,TLMC(J,I,IP),DMS,WLQ,TT(S(J,I,IP),TCHEM(J,I,IP)
1,TSEN(J),TSURF(J),JNG
GO TO 2916
2913 READ(INCH,5782) PSV,TLMC(J,I,IP),DMS,WLQ,TT(S(J,I,IP),TCHEM(J,I,IP)
1,TSEN(J),TSURF(J),JNG
2916 JNG=JNG-1
GO TO 2915
2914 READ(INCH,5791) PSV,DMS,TLMC(J,I,IP),TT(S(J,I,IP),WLQ,TCHEM(J,I,IP)
1,TSEN(J),JNG,TSURF(J)
2915 CONTINUE
IF(JNG) 2817,2917,2821
2817 TSURF(J)=BLANK
2821 CONTINUE
IF(TTS(J,I,IP)) 2803,2812,2801
TTS(J,I,IP)=TTS(J,I,IP)*1.8
TCHEM(J,I,IP)=TCHEM(J,I,IP)*1.8
TSEN(J)=TSEN(J)*1.8
GO TO 2803

```

```

2803 TTS(J,I,IP)=-TTS(J,I,IP)
2805 IF(WLS) 2809,2807,2807
2807 IX=4
      IF(LS-WLJ) 2824,2811,2824
2809 WLS=WLJ
2811 IF(NSEN) 2802,2828,2828
2802 IF(JNG) 2800,2804,2804
2804 NSEN=J-1
      ISEN(IP)=NSEN
2806 DO 2806 L=1,NSEN
      TTS(L,IP)=TTS(L,IP)
      TZSEN(L,IP)=TCHEM(L,IP)
      THSEN(L,IP)=ISEN(L)
2806 CONTINUE
      IF(NSEN-1) 2820,2820,2808
2808 CALL SLOPQ(NSEN,TTS(L,IP),THSEN(L,IP),TCPSEN(L,IP))
      CALL SLOPQ(NSEN,TTS(L,IP),TZSEN(L,IP),TCZSEN(L,IP))
      LLL=(NSEN-1)/3+1
      IF(IP-1) 2815,2813,2815
2813 CALL LCOUNT(9,LCT,NPG,RECORD(35))
      WRITE (KOUT,538)
      IF (NR) 2816,2816,2816
2816 WRITE (KOUT,5799)
      GO TO 2815
2818 WRITE (KOUT,5790)
2815 CALL LCOUNT(LLL+6
      ,LCT,NPG,RECORD(35))
      IF (LLL-1) 9819,8819,8819
8819 DO 2819 LL=1,LLL
      IF (NSEN-LL) 7819,6819,6819
6819 INT=(NSEN-LL)/LLL
      M2=LL+INT*LLL
      WRITE(KOUT,5798) (TTS(L,IP),THSEN(L,IP),L=LL,M2 ,LLL)
7819 CONTINUE
2819 CONTINUE
9819 CONTINUE
      GO TO 2862
2820 NSEN=0
      IX=3
      IF(CMH-1) 2824,2822,2824
2822 IX=2
      IF(WLQ) 2824,2826,2824
2824 WRITE (KOUT,5793) IX
      STOP
2826 IF(IP-1) 2852,2861,2862
2828 IF(TTS(J,I,IP)) 2829,2832,2829
2829 IF(PSV-TPR(IP)) 2932,2832,2832
2830 IF(DMS-TMG(I,IP)) 2834,2834,2834
2832 IPN=IP+1
      NMG(IP)=I

```

```

IN=U
NSEN=-NSEN
2834 IN=IN+1
      NHI(I,IP)=J-1
      NMC=J-1
      IX=5
      IF(NMC-1) 2824,2824,4852
4852 CONTINUE
      CALL ORDER(NMC,TLMC(I,I,IP),IZ)
      CALL SEQUA(NMC,IZ,TIS(I,I,IP),TCHEM(I,I,IP),ISEN(I),TSURF(I))
      IX=J
      IG=1
      BPG=U
      HGA=U
      NLO(I,IP)=1
      KHI(I,IP)=1
      OP=PG+TLMC(K,I,IP)
      CALL LOOK(I,TIS(K,I,IP),TI(I,I),THZ(I),U,U,U,HCH,CI2,I)
      HCH=HCH+DH2
      IF(NSEN) 2838,2838,2838
2838 TCHEM(K,I,IP)=BPG+HGA+TLMC(K,I,IP)*HCH-EP*ISEN(K)
      GO TO 2840
2838 CALL OGLE(I,TIS(K,I,IP),HZ,ISEN(IP),TISEN(I,IP),TISEN(I,IP),TCZSEN
      1(I,IP))
      CALL OGLE(I,TIS(K,I,IP),HE,ISEN(IP),TISEN(I,IP),THSEN(I,IP),TCPSSEN
      1(I,IP))
      TCHEM(K,I,IP)=BPG+HGA+TLMC(K,I,IP)*HCH-BP*ISEN(K)+HZ-TCHEM(K,I,IP)
      TSEN(K)=HE
2840 IF(TSURF(K)-BLANK) 2844,2842,2844
2842 NLO(I,IP)=K+1
      IF(IG+IX-1) 2846,2846,2824
2844 IX=1
2846 IF(K-IG) 2852,2852,2848
2848 IF(TIS(K,I,IP)-TIS(K-1,I,IP)) 2850,2850,2851
2850 IG=NMC
      GO TO 2852
2851 KHI(I,IP)=K
2852 CONTINUE
      LLL=(NMC-1)/2+1
      CALL LCOUNT(LLL+6 ,LCT,NPG,RECORD(35))
      IF (LLL-1) 6009,6009,6009
6009 WRITE(KOUT,5789) TMG(I,IP),TPR(IP)
      DO 6006 LL=1,LLL
      IF (NMC-LL) 6008,6007,6007
6007 INT=(NMC-LL)/LLL
      M2=LL+INT*LLL
      WRITE(KOUT,5795)
      1(LL,I,IP),TSURF(L),L=LL,M2 ,LLL)
      (TIS(L,I,IP),TLMC(L,I,IP),TCHEM
6008 CONTINUE
6006 CONTINUE
6007 CONTINUE
      IF (NMC-1) 4856,3856,3856

```

```

2856 DO 2856 K=1,NMC
TCHEM(K,I,IP)=CMH*TCHEM(K,I,IP)-TSEN(K)
IF(K-NLO(I,IP)) 2856,2854,2854
2854 VK=K
TLMC(K,I,IP)=ALOG(AMAX1(TLMC(K,I,IP),VK*1.E-10))
2856 CONTINUE
4856 CONTINUE
CALL SWITCH(3,JJ)
GO TO (710,711),JJ
710 CALL LCOUNT(LL+6,LCT,NPG,RECORD(35))
IF (LLL-1) 6015,6014,6014
6014 WRITE(KOUT,5787)
5787 FORMAT(//3X,39H---DUMPED VERSION OF PRECEDING TABLE---//)
DO 6011 LL=1,LLL
IF (NMC-LL) 6013,6012,6012
6012 INT=(NMC-LL)/LLL
M2=LL+INT*LL
WRITE(KOUT,5788)
1(L,I,IP)=ISEN(LL) *L=LL*M2 *LLL)
5788 FORMAT(5X,F8.2,2X,F7.4,2X,F8.2,2X,F8.2,1X,F8.2,2X,F7.4,2X,F8.2,2X,
IF(2)
6013 CONTINUE
6011 CONTINUE
6015 CONTINUE
711 CONTINUE
IF(TTS(J,I,IP)) 2862,2870,2862
2861 CALL LCOUNT(8)
WRITE (KOUT,538)
WRITE(KOUT,5794)
IF (NR) 2863,2863,2864
2864 WRITE (KOUT,5799)
2863 GO TO 2862
2863 WRITE (KOUT,5790)
2862 TPR(IPN)=PSV
TMG(IN,IPN)=DMS
TLMC(I,IN,IPN)=TLMC(J,I,IP)
TTS (I,IN,IPN)=TTS (J,I,IP)
TCHEM(I,IN,IPN)=TCHEM(J,I,IP)
TSURF(I)=TSURF(J)
TSEN(I)=TSEN(J)
J=1
I=IN
IP=IPN
GO TO 2800
2870 NPR=IP
IR(12)=1
ILO(12)=1
IHI(12)=I
DO 2872 I=1,IP
2872 IPR(I)=ALOG(IPR(I))
IR(13)=I
IHI(13)=NPR

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```

ILO(13)=1
CALL SWITCH(3,KSSW)
GO TO (700,139-J),KSSW
700 CALL LCOUNT(3,LCL,NPG,RECORD(35))
    WRITE(KOUT,703)
703 FORMAT(/10X,48HDUMP OF TABLE INDICES ILO(I,J),NMI(I,J),KHI(I,J))
    DO 701 J=1,NPR
CALL LCOUNT(3,LCL,NPG,RECORD(35))
WRITE(KOUT,704) J
704 FORMAT(/15X,6H1PR = ,12/1H )
    L=NMG(J)
CALL LCOUNT(L,LCL,NPG,RECORD(35))
WRITE(KOUT,702) (ILO(I,J),NMI(I,J),KHI(I,J),I=1,L)
701 CONTINUE
702 FORMAT(20X,3(2X,12))
1390 CONTINUE
C-----INITIALIZATIONS
TH=THI
PRT=THI+THP
QNTS=ZRO
QNTI=ZRO
DTHM = 100.0
KK=MM*NN
IAB=J
RO=RT(1,1)
CMD=J.
C-----NOTE TABLE II GIVES K IN M DIRECTION
DO 216 I = 1 , NMT
IF (CNT2(1,I) .NE. J.0) GO TO 216
DO 215 J = 1,15
CNT2(J,I) = CNT(J,I)
215 CONTINUE
216 CONTINUE
DO 825 I=1,NM
DSDI(I)=0.0
DSDTB(I)=0.0
DS(I)=0.0
DST(I)=0.0
QCOND(I)=0.
DSDTSM(I)=0.
DSN(I)=0.
QCNV(I)=0.
QCNT(I)=0.
IABLS(I)=0
QCHMT(I)=0.
QAB(I)=0.
GRAB(I)=0.
GRAD(I)=0.
GRADI(I)=0.
CMCOT(I)=0.
CMT(I)=0.

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AN 0128
AN 0129
AN 0130
AN 0131
AN 0133

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KDROP(I)=0
825 TS(I)=0
C-----SURFACE IDENTIFICATION AND CHECKING
M=0
K=0
DO 828 J=1,NN
L=-1
KT=0
DO 926 I=1,MM
K=K+1
KTS=KT
KI=MATL(K)
IF(KT) 827,827,826
827 IF(KTS) 826,826,829
829 L=L+1
KCAN=K-1
826 CONTINUE
IF(KT) 8291,8261,8260
8260 L=L+1
KCAN=K
8261 IF(L) 8291,8262,8291
8267 KSUR(J)=KCAN
MATL(KCAN)=MATL(KCAN)
IS(J)=TA(KCAN)
KSH(KCAN)=1
GO TO 828
8291 CALL LCOUNT(Z,LCT,NPG,RECORD(35))
WRITE(6,8292) J
8292 FORMAT(10X,33HEXAGONEOUS NODAL LAYOUT IN COLUMN ,I3/)
M=M+1
828 CONTINUE
IF(M) 8281,8281,8280
8280 IH=IHF
GO TO 468
8281 CONTINUE
IFIN = 1
CALL LCOUNT(5,LCT,NPG,RECORD(35))
WRITE (6,307)
C-----GEOMETRY CALCULATIONS
DO 8283 I=1,NN
K=KSUR(I)
J=K+1
L=J+MM+1
SR(I)=(CR(J)+CR(L))/2.
8283 SZ(I)=(CZ(J)+CZ(L))/2.
K=0
L=MM+1
N=L
M=0
DO 1 J=1,NN
DO 2 I=1,MM
K=K+1
L=L+1

```

```

AN 0136
AN 0137
AN 0138
AN 0139
AN 0140
AN 0141
AN 0142

```

M=M+1 IF (AC(K)) 4*3*4 C----- 3 AC(K)=.5*(AA(K)+AA(L)) 4 IF (AD(K)) 6*5*6 5 AD(K)=.5*(AB(K)+AB(L)) 6 CALL LCOUNT(1,LCI,NPG,RECORD(35)) WRITE(6,318) I,J,AA(K),AB(K),AC(K),AD(K) HA=AA(K)-AC(K) HB=AB(K)-AD(K) HC=HA+HA+HB*HB PLA(L)=FT*SQRT (HC) HA=FU*(AA(K+1)+AA(L+1))-AC(K) HB=FU*(AB(K+1)+AB(L+1))-AD(K) HC=HA+HA+HB*HB PLB(M)=FT*SQRT (HC) HA=AA(L)-AC(K) HB=AB(L)-AD(K) HC=HA+HA+HB*HB PLC(M)=FT*SQRT (HC) PLD(M)=ZRO HA=AA(K)*AA(K) HB=AA(K)*AA(K+1) HC=AA(K)*AA(L) HD=AA(K+1)*AA(K+1) HE=AA(K+1)*AA(L+1) HF=AA(L)*AA(L) HG=AA(L)*AA(L+1) HH=AA(L+1)*AA(L+1) HS=ZRO HS=HS+AB(K)*(HC+HB+HF-MD) HS=HS+AB(K+1)*(HE+HE+HA+HH) HS=HS+AB(L)*(HG+HC+HH+HA) HS=HS+AB(L+1)*(HE+HG+HD+HF) VOL(M)=6.060171E-04*ABS (HS) HA=AA(K+1)-AA(K) HB=AB(K+1)-AB(K) HC=HA+HA+HB*HB HD=SQRT (HC) HE=PIB*HD*(AA(K)+AA(K+1)) HA=AA(L)-AA(K) HB=AB(L)-AB(K) HC=HA+HA+HB*HB HD=SQRT (HC) HF=PIB*HD*(AA(K)+AA(L)) AA(M)=HE 2 AB(M)=HF K=K+1 L=L+1 7 HA=AA(L)-AA(K) HB=AB(L)-AB(K) HC=HA+HA+HB*HB HD=SQRT (HC)	AN 0143 AN 0144 AN 0146 AN 0151 AN 0153 AN 0154 AN 0155 AN 0159 AN 0165 AN 0166 AN 0167 AN 0168 AN 0169 AN 0170 AN 0171 AN 0172 AN 0173 AN 0174 AN 0175 AN 0176 AN 0177 AN 0179 AN 0180 AN 0181 AN 0183 AN 0184 AN 0185 AN 0186 AN 0188 AN 0189 AN 0190 AN 0191 AN 0192 AN 0196 AN 0197 AN 0198
---	--

```

AC(M)=PI*HU*(AA(K)+AA(L))
CALL LCOUNT(1,LC,NGP,RECORD(35))
WRITE(6,318) N,J,AA(K),A5(K)

```

AN 0200
AN 0201

```

1 CONTINUE
LU=MM+1
LL=NN+1
DO 12 I=1,LU
K=K+1

```

```

12 CALL LCOUNT(1,LC,NGP,RECORD(35))
M=M-MM
DO 8 I=1,MM
K=K+1
M=M+1

```

AN 0202
AN 0203
AN 0204
AN 0205
AN 0209
AN 0210
AN 0211

```

9 MA=AA(K+1)-AA(K)
HB=AB(K+1)-AB(K)
HC=HA*MA+HB*HB
HD=SQR1 (HC)

```

```

AD(M)=PIB*HD*(AA(K)+AA(K+1))
8 CONTINUE
M=0
L=MM
K=MM-1
N=NN-1
DO 10 J=1,N
DO 11 I=1,K
M=M+1
L=L+1

```

AN 0213
AN 0214
AN 0215
AN 0216
AN 0217
AN 0218
AN 0219
AN 0220
AN 0221
AN 0222
AN 0226
AN 0228
AN 0229
AN 0230
AN 0231
AN 0235
AN 0236

```

13 AL(M)=AB(M+1)
14 AD(M)=AA(L)
11 CONTINUE
M=M+1
L=L+1
18 AD(M)=AA(L)
10 CONTINUE
DO 1000 J=1,NN
K=KSUR(J)
1000 CALL LCOUNT(1,LC,NGP,RECORD(35))
WRITE (6,319)
ITER=0

```

```

C-----MAIN ITERATION LOOP
30 IF (DLTH) 31,31,32
31 DTH=100.
GO TO 33
32 DTH=DLTH
33 ITER=ITER+1
C-----MATERIAL PROPERTIES,MODAL RESISTANCES AND CAPACITIES
I=0
L=MM
DO 39 JJ=1,NN
DO 40 III=1,MM
I=I+1

```

AN 0237
AN 0238
AN 0239
AN 0240
AN 0241
AN 0242
AN 0243
AN 0244
AN 0245
AN 0246
AN 0248

```

L=L+1
IF(MATL(I)) 42,41,42
41 RA(I)=ZRO
RB(I)=ZRO
GO TO 40
42 KT=IABS(MATL(I))
43 IF (ITMX(KT)-TA(I)) 43,43,44
44 HB=TA(I)
GO TO 500
44 IF (TA(I)-TT(I,KT)) 45,46,46
45 HB=TA(I)
GO TO 501
46 IT=1
47 IF (TT(IT,KT)-TA(I)) 48,48,49
48 IT=IT+1
GO TO 47
49 JT=IT-1
HC=(TA(I)-TT(JT,KT))/(TT(IT,KT)-TT(JT,KT))
HD=CN2(JT,KT)+HC*(CNT(IT,KT)-CNT(JT,KT))
HA=CRD(I)+PLC(I)/HD
HM = CNT2(JT,KT) + HC * (CNT2(IT,KT) - CNT2(JT,KT))
HB = CRA(I) + PLD(I) / HM
HE=RT(JT,KT)+HC*(RT(IT,KT)-RT(JT,KT))
HF=CPT(JT,KT)+HC*(CPT(IT,KT)-CPT(JT,KT))
CAP(I)=VOL(I)*HE*HF
35 IF(MATL(I)+1) 50,50,51
50 U(JJ)=AC(I)/HB
RB(I)=0.0
GO TO 61
51 KT=IABS(MATL(I+1))
52 IF(KT) 53,52,53
52 WRITE(6,831)
831 FORMAT (10X,27HERRONEOUS NODAL ARRANGEMENT)
TH=THF
GO TO 468
53 IF (TT(K,KT)-TA(I+1)) 54,54,55
54 HB=TA(I+1)
GO TO 500
55 IF (TA(I+1)-TT(I,KT)) 56,57,57
56 HB=TA(I+1)
GO TO 501
57 IT=1
58 IF (TT(IT,KT)-TA(I+1)) 59,59,60
59 IT=IT+1
GO TO 58
60 JT=IT-1
HC=(TA(I+1)-TT(JT,KT))/(TT(IT,KT)-TT(JT,KT))
HM = CNT2(JT,KT) + HC * (CNT2(IT,KT) - CNT2(JT,KT))
HB = HB + PLD(I+1) / HM
RB(I)=AB(I+1)/HB
61 IF (JJ>NN) 63,62,63
62 RA(I)=ZRO
GO TO 40
AN 0249
AN 0251
AN 0252
AN 0253
AN 0255
AN 0256
AN 0257
AN 0258
AN 0259
AN 0260
AN 0261
AN 0262
AN 0263
AN 0264
AN 0265
AN 0266
AN 0267
AN 0268
AN 0270
AN 0271
AN 0272
AN 0282
AN 0283
AN 0284
AN 0285
AN 0286
AN 0287
AN 0288
AN 0289
AN 0290
AN 0291
AN 0292
AN 0293
AN 0295
AN 0297
AN 0298
AN 0299

```


80 IF (L-KK) 81,81,82	AN 0346
81 HA=HA+TA(L)*RA(I)	AN 0347
HB=HB+RA(I)	AN 0348
82 IF (M) 84,84,83	AN 0349
83 HA=HA+TA(M)*RA(M)	AN 0350
HB=HB+RA(M)	AN 0351
84 IF (CAP(I)) 85,86,85	AN 0352
86 TB(I)=HA/HB	AN 0353
GO TO 76	AN 0354
85 DEN=HB	AN 0355
97 TB(I)=HA-TA(I)*HB	AN 0407
98 IF (DLI(I)) 76,99,76	AN 0408
99 IF (KTM(I)) 76,76,1001	
1001 IF (MATL(I)) 76,76,100	
100 MC=EA*CAP(I)/DEN	AN 0410
IF (MC-DTH) 1,1,76,76	AN 0411
101 DTH=MC	AN 0412
MCRII=III	
MCRII=JJ	AN 0414
76 CONTINUE	AN 0415
75 CONTINUE	AN 0416
C-----GO TO SURFACE ENERGY BALANCE PACKAGE	
CALL SURFB	
C-----NEW TEMPERATURES LOOP	
107 K=0	AN 0422
DO 120 J=1,NN	AN 0423
DO 121 I=1,MM	AN 0424
K=K+1	AN 0425
IF (MATL(K)) 121,121,136	AN 0426
136 IF (CAP(K)) 122,121,123	
122 WRITE(6,127) K	
127 FORMAT(/10X,26#NEGATIVE CAPACITY AT NODE ,I3//)	
THF=TH	
GO TO 468	AN 0429
123 IF (KSM(K)) 124,124,125	
125 WRITE(6,126) K	
126 FORMAT(/10X,34#MODAL BLURDER AT 123, HEATED NODE ,I3,38H HAS MATE	
RIAL NUMBER GREATER THAN ZERO//)	
THF=TH	
GO TO 468	
124 IF (MATL(K+1)) 1240,124,1243	
1240 IF (KRESC-3) 1243,1241,1243	
1241 IF (KDROPIJ) 1242,1242,1243	
1242 TB(K)=(TB(K)+RE(K)+TB(K+1))-TA(K+1))*DTH/CAP(K)+TA(K)	
GO TO 121	
1243 HA=TB(K)*DTH/CAP(K)+TA(K)	AN 0460
TB(K)=HA	AN 0461
121 CONTINUE	AN 0462
120 CONTINUE	AN 0463
KK=MM*NN	
466 DO 467 I=1,KK	AN 0627

D-18

AN 0628
AN 0629

```

ONTI=ONTI+CAP(I)*(TB(I)-TA(I))
467 TA(I)=TB(I)
IF(DDTH.LT.DTH) .AND. (TH.NE.PRT) DTHM = DTH
CALL SWITCH(5,KSS)
GO TO (4671,4672),KSS
4671 IF(ITER=3471) 4672,4672,4672
4673 IF(ITER=3451) 468,468,4672
4672 CONTINUE
CALL SWITCH(1,KSS)
GO TO (468,467),KSS
467J IF(TH-PRT) 4674,468,4674
4674 IF(TH-THF) 472,468,468
C-----OUTPUT BLOCK
468 CEC=ONTI/JNTI

```

AN 0631

```

CALL LCOUNT(17+NN,LCT,NPG,RECORD(35))
WRITE(6,3364) TH
3364 FORMAT(//1X,5-(1H*),F7.2,8H SECONDS,53(1H*)//52X,13H***GENERAL***

```

AN 0632

```

WRITE(6,334)
WRITE(6,335) TH,ONTI,ONTI,CEC,MCRTI,NCRTI,ITER,DTHM,DTHM
WRITE(6,336)
336 FORMAT(//49X,2-H***HEATED SURFACE***//40X,40H-----MISCELLANEOUS
1 SURFACE DATA-----//2X,25HROW COL OPTN SURF SURF,6X,6MM EDGE,
2X,6MM WALL,3X,4UM PRIME MASS COEFF CM/CHD PRESSURE,4X,6HRA
3DIUS,1-X,1H,16X,57HITER TEMP(R) (BTU/LB) (BTU/LB) TOT (
4LB/FT2-SEC),1-X,5H(ATM),6X,4H(IN)
K=0

```

```

DO 949 J=1,NN
DO 949 I=1,MM

```

```

K=K+1
IF(KSHUK) 949,949,950

```

```

950 BP=CMDOT(J)/G(J)

```

```

R=SR(J)
Z=SZ(J)
ITS=ITSR(J)

```

```

WRITE(6,337) I,J,II(J),ITS,TS(J),MEDU(J),H=L(J),EP,G(J),SR,PR(J),
IR,Z

```

```

1370 FORMAT(2X,13,1X,12,3X,11,2X,15,2X,F7.1,2(2X,F9.2),2X,F7.4,3X,
1F9.5,3X,F7.5,3(2X,E11.3))

```

```

949 CONTINUE

```

```

CALL LCOUNT(18+NN,LCT,NPG,RECORD(35))
WRITE(6,3365)

```

```

3365 FORMAT(//40X,39H-----SURFACE RATE QUANTITIES-----//1X,
1 31H-LOCATION-- --RECESSION RATES--8X,14H--MASS RATES--20X,
229H--SURFACE ENERGY FLUX RATES--/18X,10H(MILS/SEC),12X,
312H(LS/FT2-SEC),29X,13H(BTU/FT2-SEC)

```

```

WRITE(6,3361)

```

```

1361 FORMAT(50X,9HCONNECTED,4X,8HCHEMICAL,4X,9HRADIATION,3X,9HRADIATION
1,2X,10HCONDUCTION/2X,25HROW COL CENTER LINE NORMAL MDUT TOT
2AL MDUT TCRE,6X,2HIN,6X,21HGENERATION ASSORBED,5X,THEMITTED,
36X,4HAWAY)

```

```

K=0
DO 951 J=1,NN

```

```

467
4671
4673
4672
467J
4674
468
3364
336
1
2
3
4
K=0
DO 949 J=1,NN
DO 949 I=1,MM
K=K+1
IF(KSHUK) 949,949,950
950
R=SR(J)
Z=SZ(J)
ITS=ITSR(J)
WRITE(6,337) I,J,II(J),ITS,TS(J),MEDU(J),H=L(J),EP,G(J),SR,PR(J),
IR,Z
1370
949
3365
1
2
3
1361
K=0
DO 951 J=1,NN

```



```

9500 CONTINUE
CALL LCOUNT(4,LCT,NPG,RECORD(35))
WRITE(6,9494) PRI,THI,THF,DLI,H,EIA,FV,FT,PIE,ZRO,KDOP,KM,NN,KK,
INMT,MH
9494 FORMAT(/,2X,56HPRI,THI,THF,DLI,H,EIA,FV,FT,PIE,ZRO,KDOP,KM,NN,KK,NM
I,MT,MH)
9497 CONTINUE
IF(KSTRP-1) 849,842,8421
8421 DO 8422 I=1,5
8422 CONTINUE
IF(ARS(TH-TPN(I))-.....01) 842,842,8422
842 DO 849
842 K=0
NS=0
DO 840 J=1,MN
DO 840 I=1,MK
K=K+1
IF(MATL(K)) 8420,840,841
8420 NS=NS+1
841 NS=NS+1
840 CONTINUE
K=0
N=0
LL=0
LR=MM+1
DO 843 J=1,MN
DO 844 I=1,MK
K=K+1
LL=LL+1
LR=LR+1
IF(MATL(K)) 845,844,845
845 N=N+1
Z=(CZ(LL)+CZ(LR))/2.0
R=(CR(LL)+CR(LR))/2.0
PUNCH 846, R,Z,TA(K),I,J,MATL(K),TH,RECORD(35),RECORD(36),H,NS
846 FORMAT (3F10.3,8MININDEGR,I3,IH,I3,3HMAT,I2,F7.2,2HJ,2A6,I3,I3,
12HOF,I3)
IF(MATL(K)) 847,844,844
847 N=N+1
LL=LL+1
LR=LR+1
RAT=PLB(K)/PLB(J)
N=R+RAT*(CR(LL)+CR(LR))/2.0-R
Z=Z+RAT*(CZ(LL)+CZ(LR))/2.0-Z
MOUT=-MATL(K)
PUNCH 848, R,Z,TS(J),J,MOUT,TH,RECORD(35),RECORD(36),N,NS
848 FORMAT (3F10.3,8MININDEGR,I3,IH,I3,3HMAT,I2,F7.2,2HJ,2A6,I3,I3,
12HOF,I3)
844 CONTINUE
843 CONTINUE
849 CONTINUE
CALL LCOUNT(KK/5+6,LCT,NPG,RECORD(35))
WRITE(5,3365)

```

```

3365 FORMAT(/,49X,19H***IN-DEPTH DATA***/)
WRITE (6,332)
KPR=J
K=U
DO 240 J=1,NN
DO 240 I=1,MM
K=K+1
JE (MATLK(I), 952, 951, 952)
951 TMPR(KPR)=TA(K)
GO TO 953
952 TMPR(KPR) = IA(K)
953 MPR(KPR)=I
NNPR(KPR)=J
IF (KPR-5) 240, 241, 240
241 WRITE(6,333) (MPR(L), NNPR(L), TMPR(L), L=1, 5)
KPR=U
240 CONTINUE
IF (KPR) 469, 469, 242
242 WRITE(6,333) (MPR(L), NNPR(L), TMPR(L), L=1, KPR)
469 IF (TH-THF) 472, 470, 470
470 IF (KASE) 471, 471, 225
471 STOP
472 IF (TH-PTH) 30, 472, 472
473 IF (TH-TPICG(I)) 400, 473, 474, 474
473 PRT=AMINI(PRT+THP, TPICG(I))
DTIME=100.
GO TO 30
474 IF (PRT(I)) 476, 476, 475
476 CALL LCOUNT(5, LCT, NPG, RECORD(35))
WRITE (KOUT, 477)
477 FORMAT(/,10X,62H*HAVE ENCOUNTERED A ZERO OUTPUT INTERVAL, AM QUITTI
ING THIS JOB//)
GO TO 470
475 THP=PRT(I)
DO 478 I=1,3
TPICG(I)=TPICG(I+1)
478 PRT(I)=PRT(I+1)
GO TO 473
500 WRITE(6,822) III, JJ
THF=TH
KT
GO TO 468
501 WRITE(6,822) III, JJ
WRITE (6,824)
THF=TH
KT
GO TO 468
505 WRITE (6,819)
THF=TH
TA(I), TH
GO TO 468
END
AN 0643
AN 0644
AN 0645
AN 0646
AN 0647
AN 0648
AN 0652
AN 0654
AN 0655
AN 0656
AN 0664
AN 0665
AN 0668
AN 0669

```

```

SUBROUTINE SURFR
C-----DIMENSIONED AS CORNERS
COMMON C914(2),CZ(4,2),AA(4,2),AB(4,2),AC(4,2),AD(4,2)
C-----DIMENSIONED AS NODES
COMMON NATL(20),KTH(2,0),KSM(2,0),KSE(2,0),KTL(20),CON(20),
1TA(2,0),TR(2,0),CPA(2,0),CRA(2,0),CRR(2,0),PLA(2,0),PLS(2,0),PLC(2,0),
2PLD(2,0),VOL(2,0),RA(2,0),RP(2,0),CAP(2,0),ENT(20),QNP(2,0),
3VF1(2,0),VF3(2,0)
C-----LIVE AS SURFACE NODES (NUMBER OF COLUMNS)
COMMON KSUR(4),TS(4),DST(4,0),DSTB(4,0),DST(4,0),DST(4,0),J(4,0)
COMMON PLS(4),DSTI(4),CONCT(4),QCV(4),QCV(4),QCV(4),GCM(4,0),
1SCHT(4),QRAB(4),QRAPT(4),CRAD(4),QRADT(4),DSDTEN(4),DSN(4,0)
COMMON CVDT(4),MEDG(4),CNT(4),HAL(4),G(4),GZ(4),PR(4,0)
COMMON I1(4),S2(4),SR(4),EMN(4)
COMMON I1SR(4)
COMMON IAPLS(4),KUR(4)
DIMENSION FMS(4),BPSV(4)
C-----DIMENSIONED AS PROPERTY TABLES, ENTRIES X MATERIALS
COMMON TI(15,6),RT(15,6),CPT(15,6),CNT(15,6),CNT2(15,6),
1EPT(15,6),TTX(16)
C-----DIMENSIONED AS TIME TABLES, ENTRIES X TABLE NO
COMMON TH(25,10),CHT(25,1),RCT(25,1),TUR(25,10),TPI(25,10),
1TBRP(25,1)
C-----MISCELLANEOUS QUANTITIES
COMMON RECORD(36),IMPR(5),PPR(5),NNPR(5)
COMMON VITER(5),EITER(5),IAS,RQ
COMMON VKIN,CMM,CV,CH
COMMON TH,DTH,PRI,THI,THF,DUTH,ETA,ITS,DTHS
COMMON FV,FI,PIR,ZRO,KQOP,RK,ANN,NVT,SHI,QNIS,QSUM,SIG
COMMON KRESC,KSLOP
DIMENSION Y2(8)
COMMON/ENERGY/TPR(5),NMG(5),IMG(8,5),NLO(8,5),NHI(8,5),
1KHI(8,5),ITSEN(25,5),ITSEN(25,5),TCPSEA(25,5),
2TLMC(25,8,5),ISFN(5),ITS(25,8,5),TCHEM(25,8,5)
3NPR
COMMON/LK/KOUT,LEX,DEN,VR,THI(4),ILO(4),IR(4)
DATA BLANK/64 /
C-----ADJUSTMENT OF TIME STEP ACCORDING TO LIMITATIONS OF TIME TABLES
DTHS=DTH
TH=TH+DTH
DO 290 J=1,NN
K=KSUR(J)
KT=KTU(K)
IF(KT) 290,290,296
296 NTT=KT+20
I=IR(INT)
291 IF(TH(I+1,KT)-TH+...J01) 294,290,296
294 IF(I+1-TH(INT)) 292,290,296
292 I=I+1
IF(TH(I+1,KT)-TH(I,KT)) 291,293,291
293 TH=TH-DTH
IMD=TH(I,KT)-J
OTM=MAX(1,1,TH(I,KT)-J)

```

```

TH=TH+DTH
GO TO 291
290 IPRINT=1
C-----DEFINE ALLOWABLE TIME STEP
K=C
TH=TH-DTH
DO 2 I=1,NN
DSDI(I)=DSDI(I)
K=KSR(I)
DTH=AMIN(DTH,PLS(I)/(1.-DST(I)+.J000J)),(PLS(I)/5J0.+PLB(K))/
1DSDI(I)+DSDI(I)
IF(DTH-1.0E-12) 106,106,107
106 WRITE(6,108)
108 FORMAT(10X,19HTOO SMALL TIME STEP//)
WRITE(6,109) 1
109 FORMAT(10X,6HCOLUMN,13//)
IHE=IH
RETURN
107 CONTINUE
2 CONTINUE
HA=TH+DTH
IF(HA-PT) 105,105,104
104 DTH=PRI-IH
TH=PT
GO TO 4
105 IHE=TH+DTH
4 CONTINUE
C-----GET SLOPES NOW PENDING REARRANGEMENT OF THIS SECTION
GO TO 2904 CALL SLOP(X,NN,SZ,SR,EMS,EMN)
2904 CALL SLOP(X,NN,SZ,SR,EMS,EMN)
GO TO 2903
2905 CONTINUE
IF(NN-1) 2901,2901,2902
2901 EMN(I)=1.E+30
GO TO 2903
2902 CALL SLOP(NN,SZ,SR,EMN)
DO 2900 J=1,NN
2900 EMN(J)=1./EMN(J)+1.E-30
2903 CONTINUE
C-----MAIN COLUMN LOOP FOR SURFACE OPERATIONS
DO 7 I=1,NN
C
C-----INTERPOLATE IN TIME TABLES AND IDENTIFY OPTION
C
K=KSR(I)
KT=KI(I,K)
IF(KT) 20,20,21
20 HA=TB(K)+DTH/CAP(K)+TA(K)
TB(K)=HA
TS(I)=TQ(K)
DSDT(I)=U*
DSDIBN(I)=Q*

```

```

GNP(K)=0.
GO TO 7
21 CONTINUE
ITL=10
ITS=1
MT=IABS(MATL(K))
NTT=KT+20
J=IRINTT)
DFN=(TH-THT(J,KT))/(THT(J+1,KT)-THT(J,KT))
IF(THT(J+1,KT)-THT(J,KT)) 294,294,295
294 DEN=C.
295 CH=CHT(J,KT)+DEN*(CHT(J+1,KT)-CHT(J,KT))
JRA=TGR(J,KT)+DEN*(TGR(J+1,KT)-TGR(J,KT))
HE=HEI(J,KT)+DEN*(HEI(J+1,KT)-HEI(J,KT))
BRP=TSKP(J,KT)+DEN*(TSKP(J+1,KT)-TSKP(J,KT))
PRES=TPI(J,KT)+DEN*(TPI(J+1,KT)-TPI(J,KT))
FACT=DTH/CAP(K)
KCOOP=1
IF(CH) 297,297,2070
297 KCOOP=2
CH=J+J
IF(HE-2.0) 298,298,299
298 KCOOP=3
HE=C.
GO TO 299
C-----OPTION 2 PREPARATIONS
299 TS(1)=HE
ST=HL
SA=GRA/12000.
DS(1)=CA-DSTT(1)-PLBS(1)+PLB(K)+DST(1)
DSDTB(1)=DS(1)/DTH
HA=VCOSIK*1+NM+CR+CZ+JZ+SR+ENN+PLB+PLBS+DST)
IF(HA-.1) 2991,2991,2992
2991 WRITE(6,2993) 1,KCOOP
2993 FORMAT(/10X,28H=AD SURFACE SHAPE AT COLUMN 12,58H, A= RETURNING
1 TO MAIN PROGRAM WITH THE OPTION IS 12,1H,/)
*RITE(KOUT,2994) (J,ENS(J),ENN(J), J+1,AN)
2994 FORMAT(10X,15H1,ENS(1),ENN(1))/(1-X,12,2E12.3))
THF=TH
RETURN
2992 CONTINUE
DSN(1)=DS(1)*HA
DSDTB(1)=DSDTB(1)*HA
CM2=DSDTR(1)*RO
JCHEM=C.
QCONV=C.
QRA=C.
RAD=C.
CM=C.
CHZ=C.
SR=C.
HHE=C.
HE=C.

```

ITS=U
GO TO 245
C-----OPTION 3 CALCULATIONS

2990 TABC=9000.
DSDIBL11=U.
DS(I)=U.
DSDTBN(I)=0.
DSNLI1=0.
CMD=U.
CM=0.
CHZ=U.

HW=0.
GO TO (2995+2996+2996)*KRESC
2995 A=U(I)/IAC(K)*I1+FACT*U(I)
B=A*(FACT*TB(K)+TA(K))

GO TO 2997
2996 A=U(I)/AC(K)*I1+FACT*RB(K-1)/I1+FACT*(U(I)+RB(K-1))
B=A*(TA(K)+FACT*TB(K)/I1+FACT*RB(K-1))
2997 CONTINUE
EBZ=--B

OCHEM=U.
OCONV=0.
VF=VF1(K)

GO TO 240
C-----OPTION 1 PREPARATIONS
2070 BF=CMOY(I)/CH

VF=VF1(K)
CHZ=CH

PHI=2.88P8BF

IF(PHI<0) 2071+2071+2072
2071 CM=CH*(1.-.5*PHI)
GO TO 2073

2072 CM=CH/(EXP(PHI)-1)*PHI
2073 CM=CH*CMH

GO TO (2076+2077+2077)*KRESC
2076 A=U(I)/IAC(K)*I1+FACT*U(I)
B=A*(FACT*TB(K)+TA(K))

2077 A=U(I)/AC(K)*I1+FACT*RB(K-1)/I1+FACT*(U(I)+RB(K-1))
B=A*(TA(K)+FACT*TB(K)/I1+FACT*RB(K-1))
2078 CONTINUE

CMOL=BPSV(I)
IAB=IABLS(I)
ERFX=CH*HE--B
IPRAX=EK

ILO(I2)=1
IMI(I2)=NMG(IPR)

SPG=YKIN/CM
CALL LOOK(I2*SPG*TMG(I),IPR),YU,0,.Y2(I),Y2(I2),I)
YRMAVR

ILO(I4)=NLO(TM,IPR)
IMI(I4)=NMI(TM,IPR)
ILO(I5)=NLO(IMG+1,IPR)

```

IMI(15)=NMI(IMG+1,IPR)
I1=ILO(14)
I2=ILO(15)
IF(IHI(14)-I1) 240,240,245
203 IF(IHI(15)-I2) 240,240,244
204 TARC=TTSC(1,IMG,IPR)+VRM*(TTSC(1,IMG+1,IPR)-TTSC(1,IMG,IPR))
IF(TS(1)-TARC) 240,240,245
C-----ABLATING SURFACE
205 IF(IAB) 206,206,207
206 CMCL=TLMC(1,IMG,IPR)-VRM*(TLMC(1,IMG,IPR)-TLMC(1,IMG+1,IPR))
CMD=EXP(CMCL)*CN
IAB=1
207 CALL LOOK(14,CMD,TLMC(1,IMG,IPR),TTSC(1,IMG,IPR),
1 TCHEX(1,IMG,IPR),0,0,Y2(1),Y2(3),2)
IRA=IR(14)
CALL LOOK(15,CMD,TLMC(1,IMG+1,IPR),TTSC(1,IMG+1,IPR),
1 TCHEX(1,IMG+1,IPR),0,0,Y2(5),Y2(7),2)
IRB=IR(15)
DO 208 J=1,4
208 Y2(J)=Y2(J)+VRM*(Y2(J+4)-Y2(J))
ST=Y2(1)
CALL LOOK(19,ST,IT(1,PT),EPT(1,PT),0,0,0,EMIV,OVIV,1)
TSSQ=ST*ST
RAD=SIG*FMIV*TSSQ*TSO*VF
ERR=CH*Y2(2)+E*IV*ORA-RAC-A*ST*ERFX
C-----CONVECTION CORRECTION TO BE ADDED
DERR=CH*Y2(4)+((GRA-RAD/EMIV)*OVIV-4.0/ST*RAC-A)*Y2(3)
ERRC=ERR/DERR
VIERR(IT5)=CMDL
EITER(IT5)=ERR
CMDL=CMDL-ERRC
210 IF(ILO(14)-IRA) 210,211,211
211 IF(ILO(15)-IRB) 212,211,211
212 CMNI=AMAX1(TLMC(IRA,IMG,IPR)+TLMC(IRA-1,IMG,IPR),
1 TLMC(IRB,IMG+1,IPR)+TLMC(IRB-1,IMG+1,IPR))/2.
CMDL=AMAX1(CMDL,CMNI)
211 IF(IHI(14)-IRA-1) 220,220,213
213 IF(IHI(15)-IRB-1) 220,220,214
214 CMMA=AMIN1(TLMC(IRA+1,IMG,IPR)+TLMC(IRA+2,IMG,IPR),
1 TLMC(IRB+1,IMG+1,IPR)+TLMC(IRB+2,IMG+1,IPR))/2.
CMDL=AMIN1(CMDL,CMMA)
215 ERRS=ERR
CMDL=CMMA
GO TO 227
216 IF(ERR*ERRS) 218,222,217
217 CMDL=CMMA
GO TO 222
218 ITL=55
219 IF(ERRC) 219,222,222
219 CMDL=CMNI
GO TO 222
220 IF(IT5-ITL) 222,221,222

```

221 CMDL=AMINI(TLMC(I1,IMG,IPR),TLMC(I2,IMG+1,IPR))

222 CHD=EXP(CMDL)*CM

IF(IITS-50) 223,224,224

223 ITS=ITS+1

IF(ABS(ERR1-1.) > .262,262, 2J7

224 WRITE(6,225)(VITER(J),EITER(J),J=1,51)

225 FORMAT(10X,37H SURFACE ENERGY BALANCE ITERATION STOP//12X,26HVARIAB

ILE AND ERROR HISTORY//15X,1VELV,31

2264 WRITE(6,226) TH,DTM,VRM,ERFX,HE,ST,TASC,EMIV,OMIV,

IRAD,ORA,A,B,CM,CM,CM,Y2(1),Y2(2),

Y2(3),Y2(4),SI,SI,I,I,IAB,I,K,I1,I2,ILO(14),

3ILO(15),IMI(14),IMI(15),IRA,IRB,ITS,ITL,IMG,IPR

226 FORMAT(//10X,1,5HTH,DTM,VRM,ERFX,HE,ST,TASC,EMIV,DMIV,RAD,ORA,A,B,

1CM,CM,CM,CM,Y2(1),Y2(2),Y2(3),Y2(4),SI,SI,I,I,IAB,I,K,I1,I2/55H ILO(14

2),ILO(15),IMI(14),IMI(15),IRA,IRB,ITS,ITL,IMG,IPR//10X,9E12,3/10X,

39E12,3/10X,4E12,3,1514//

WRITE(6,2262) VEJCHZ,BRP,PHI,FACT,U(1),AC(K),VRP,ENN(I),PLBS(I),

1PLB(K),CAP(K),VOL(K),T6(K)

2262 FORMAT(//10X,75HV,CHZ,BRP,PHI,FACT,U(1),AC(K),VRP,ENN(I),PLBS(I),

1PLB(K),CAP(K),VOL(K),T6(K)//10X,9E12,3/10X,5E12,3)

L=K-MM

IM=K+MM

WRITE(6,2263) MAIL(I),MAIL(K),MAIL(IM),MAIL(K-1),MA(K),MA(K),MAI

IL),RB(K-1),DST(I),DST(K),AA(K),AS(K),AD(K),AD(L),AA(IH),PLA(K),

2PLD(K),PLC(K),PLC(L),PL9(K-1),PLA(IH)

2263 FORMAT(//10X,1,1H MAIL(I),MAIL(K),MAIL(IM),MAIL(K-1),RA(K),RB(K),

IRAL),RB(K-1),DST(I),DST(K),AA(K),AS(K),AD(K),AD(L)/10X,55HAA(K),

2M),PLA(K),PLD(K),PLC(K),PLC(L),PLA(K+MM)//10X,415,7E12,3/

330X,7E12,3/3, X3E12,3//)

IF(IAB) 2266,2266,2267

2266 RETURN

2267 CONTINUE

L=0

IR(19)=1

IL=ILO(14)

IM=IMI(14)

DO 227 J=IL,IM

L=L+1

CALL LOOK(15,TLMC(J,IMG,IPR),TLMC(I,IMG+1,IPR),TTS(I,IMG+1,IPR),

1TCHEM(I,IMG+1,IPR),VRM,Y2(1),Y2(3),2)

Y2(1)=TTS(J,IMG,IPR)+VRM*Y2(1)-TTS(J,IMG,IPR)

Y2(2)=TCHEM(J,IMG,IPR)+VRM*Y2(2)-TCHEM(J,IMG,IPR)

ST=Y2(1)

ISSQ=ST*ST

CALL LOOK(19,ST,TT(1,MT),EPT(1,MT),J,0,0,EMIV,DMIV,1)

RAD=SIG*EMIV*TSSQ*VF

ERR=CH*Y2(2)+EMIV*GRA-RAU-A*ST+ERFX

EITER(L)=ERR

227 VITER(L)=TLMC(J,IMG,IPR)

WRITE(6,228) IMG

228 FORMAT(//10X,92H COMPLETE SURFACE TABLE FOR ANALYSIS, COMPUTED USIN

1G CURRENT VALUES OF CH,ORA,HE,A,B, AND VR://12X,6H I G = ,12//)

WRITE(6,229) (VITER(J),EITER(J),J=1,51)

```

229 FORMAT(20X,8MLN BRIME,10X,2-RENEASY BALANCE L=JOK/41X,
114M(RTU/SCF-SEC)/(19X,E13.3,14X,E10.3))
L=J
IL=ILO(15)
IM=IMI(15)
DO 230 J=IL,IM
L=L+1
CALL LOOK(14,TLVC(J,IMG+1,IPR),TLMC(1,IMG,IPR),
ITTS(1,IMG,IPR),TCHEM(1,IMG,IPR),J,J,J,Y2(1),Y2(3),2)
Y2(1)=Y2(1)+(ITTS(J,IMG+1,IPR)-Y2(1))*VRM
Y2(2)=Y2(2)+(TCHEM(J,IMG+1,IPR)-Y2(2))*VRM
ST=Y2(1)
TSSQ=ST*ST
RAD=SIG*EMIV*TSSQ*ISSQ*VF
CALL LOOK(19,ST,TT(1,VT),EPT(1,VT),J,J,J,EMIV,DMIV,1)
ERR=CH*Y2(2)+EMIV*ORA-RAD-A*ST+ERFX
EITER(L)=ERR
VITER(L)=TLMC(J,IMG,IPR)
IMG=IMG+1
WRITE(6,231) IMG
231 FORMAT(/12X,5HIMG = ,I2//)
THF=TH
RETURN
C-----NON-ABLATING SURFACE
240 IAB=0
CMD=0
IF(KDOP-3) 240,249L,249L
240 ILO(16)=1
IMI(16)=KMI(IMG,IPR)
ILO(17)=1
IMI(17)=KMI(IMG+1,IPR)
ST=YS(1)
249 CALL LOOK(16,ST,ITTS(1,IMG,IPR),TCHEM(1,IMG,IPR),0,0,0,
Y2(1),Y2(2),1)
CALL LOOK(17,ST,ITTS(1,IMG+1,IPR),TCHEM(1,IMG+1,IPR),
Y2(3),Y2(3),Y2(4),1)
DO 241 J=1,2
241 Y2(J)=Y2(J)+VRM*(Y2(J+2)-Y2(J))
2491 CONTINUE
TSSQ=ST*ST
24J1 CALL LOOK(19,ST,TT(1,VT),EPT(1,VT),J,J,J,EMIV,DMIV,1)
RAD=SIG*EMIV*TSSQ*ISSQ*VF
EPR=CH*Y2(1)+EMIV*ORA-RAD-A*ST+ERFX
ERR=CH*Y2(2)+(JRA-RAU/EMIV)*DMIV-4.7/ST*RAD-A
ERRC=ERR/DEKR
VITER(ITTS)=ST
EITER(ITTS)=ERR
ST=ST-ERRC
IRA=IR(16)
IRB=IR(17)
IF(ILO(16)-IRA) 242,244,244

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```

242 IF(ILO(17)-IRB) 243,244,244
243 ISM1=AMAX1(TTS(J,IR,IMG+1,IPR)+TTS(IRB-1,IMG+1,IPR),
1TTS(IRA,IMG,IPR)+TTS(IRA-1,IMG,IPR))/2.
ST=AMAX1(ST,ISM1)
244 IELH(I(16)-IRA-1) 247,247,245
245 IF(IH(I(17)-IRB-1) 247,247,246
246 TSMA=AMIN1(TTS(IRA+1,IMG+1,IPR)+TTS(IRB+2,IMG+1,IPR),
1TTS(IRA+1,IMG,IPR)+TTS(IRA+2,IMG,IPR))/2.
ST=AMIN1(ST,TSMA)
247 IF(I(17)-50) 248,248,250
248 I(17)=I(17)+1
IF(ABS(ERR)-1.) 260,260, 249
250 WRITE(6,251)(VITER(J),EITER(J),J=1,51)
WRITE(6,251) TH,DTH,VRM,ERFX,HE,ST,TABC,ENIV,DMIV,
1RAD,GRA,A,B,CH,CMH,Y2(1),Y2(2),
2Y2(3),Y2(4),TS(1),DSDT(1),DSDTR(1),
3CMD,PLB(K),PLB(L),L,K,IPR,IMG,
4ILO(16),IH(16),ILO(17),IH(17),IR(16),
5IR(17),IRA,IRB,ITS,IIL,MT
251 FORMAT(10X,1,6HTH,DTH,VRM,ERFX,HE,ST,TABC,ENIV,DMIV,RAD,GRA,A,B,CH
1,CMH,Y2(1),Y2(2),Y2(3),Y2(4),TS(1),DSDT(1),DSDTR(1),10X,95HCMD
2,PLB(K),PLB(L),L,K,IPR,IMG,ILO(16),IH(16),ILO(17),IH(17),IR(16)
3,IR(17),IRA,IRB,ITS,IIL,MT//10X,9E12,3/10X,5E12,3/10X,8E12,3/1513/
4)
L=0
IL=ILO(16)
IH=IH(16)
DO 252 J=IL,IH
L=L+1
CALL LOOK(17,TTS(J,IMG,IPR),TTS(1,IMG+1,IPR),TCHEM(1,IMG+1,IPR),0,
1,0,0,Y2(3),Y2(4),1)
Y2(1)=TCHEM(L,IMG,IPR)+VRM*Y2(3)-TCHEM(J,IMG,IPR)
ST=TTS(J,IMG,IPR)
CALL LOOK(19,ST,IT(1,MT),EPT(1,MT),0,0,0,EMIV,DMIV,1)
ISSQ=ST*ST
RAD=SIG*EMIV*TSO*TSQ*VF
ERR=CH*Y2(1)+EMIV*GRA-RAD-A*ST+ERFX
EITER(L)=ERR
252 VITER(L)=ST
WRITE(6,228) IMG
WRITE(6,253) (VITER(J),EITER(J),J=1,L)
253 FORMAT(20X,9HSURF TEMP,10X,2JHENRGY BALANCE ERROR/
121X,7H(DEG R)+14X,14HBTU/5QFT-SEC)//19X,E10,3,15X,E10,3)
L=L+1
IL=ILO(17)
IH=IH(17)
DO 254 J=IL,IH
L=L+1
CALL LOOK(16,TTS(J,IMG+1,IPR),TTS(1,IMG,IPR),TCHEM(1,IMG,IPR),
1,0,0,0,Y2(3),Y2(4),1)
Y2(1)=Y2(3)+VRM*TCHEM(J,IMG+1,IPR)-Y2(3)
ST=TTS(J,IMG+1,IPR)
CALL LOOK(19,SI,IT(1,MT),EPT(1,MT),0,0,0,EMIV,DMIV,1)

```

```

TSSQ=ST*ST
RAD=SIG*FX*IV*TSU*TSQ*VF
ERR=CH*Y2(1)+EM*IV*GRA-RAD-*.ST+ERFX
EITER(L)=ERR
254 VITER(L)=ST
    IMG=IMG+1
    WRITE(6,231) IMG
    WRITE(6,253) (VITER(J),EITER(J),J=1,L)
    THF=TH
    GO TO 2264
C-----POST ITERATION
C
260 IF(KOOP-2) 261,261,265
261 Y2(2)=Y2(1)
262 QCHEM=Y2(2)
    IF(ISEN(1)) 263,264,263
264 QCONV=Q
    GO TO 266
263 CALL OGLE(1,ST,QCONV,ISFN(IPR),TTSEN(1,IPR),THSEN(1,IPR),
    1TCPSEN(1,IPR))
266 QCHEM=(QCHEM+QCONV)*CH
    HW=QCONV
    QCONV=CH*(HE-QCONV)
    DSDTBN(1)=CMD/RO
    HA=VCOS(K,I,MM,CR,CZ,SZ,SR,EMN,PLB,PLBS,DST)
    IF(HA-.1) 2591,2991,2660
266J CONTINUE
    SP5V(1)=CMDL
    DSDIB(1)=DSDTBN(1)/HA
    DS(1)=DSDTR(1)*DTH
    DSN(1)=DS(1)*HA
265J JS(1)=ST
265 GO TO (2651,2652,2652),KRFS
2651 TR(K)=(FACT*(TB(K)+U(I)*TS(I))+TA(K))/(1.+FACT*U(I))
    GO TO 267
2652 TR(K)=(FACT*(TB(K)+U(I)*ST)+(1.+FACT*RU(K-1))*TA(K))/(1.+FACT*
    1(U(I)+RB(K-1)))
267 PLB(K)=PLB(K)-DS(I)
    ON=(TS(1)-T5(K))*O(I)
    QCOND(1)=QCOND(1)+QN*DTH
    QSUM=QSUM+QN
    QNP(K)=QN/AC(K)
    GCNV(1)=QCONV
    QCNVT(1)=QCONV*AC(K)*DTH+JCNVT(1)
    QCHEM(1)=QCHEM
    JCHMT(1)=JCHMT(1)+QCHEM*DTH*AC(K)
    JRP=EM*IV*JRA
    ORAB(1)=ORP
    JCRABT(1)=JCRABT(1)+JCRP*DTH*AC(K)
    ORAD(1)=RAD
    JCRADT(1)=JCRADT(1)+RAD*DTH*AC(K)
    CVDOT(1)=CMD
    HEDG(1)=HE

```

```

CMT(I)=CMT(I)+CMD*AC(K)*DTM
I(I)=KOOP
ITSR(I)=ITS
IAULS(I)=IAD
HML(I)=HM
G(I)=CM
GZ(I)=CHZ*CMH
PRL(I)=EXPRES)
C-----NODE DROPPING PACKAGE
IF(PLB(K)) 1,13,11
11 IF(K-1)-1*MM-11 13,13,12
12 IF(MATL(K-1)) 15,15,12,0
1200 IF(MATL(K-1)-MT) 13,14,13
14 DS(I)=PLB(K)+DSL(I)
DST(I)=DST(I)+PLBS(I)-DST(I)
PLB(K)=0,v
MAIL(K)=0
MATL(K-1)=MATL(K-1)
KSUR(I)=K-1
KDROPL(I)=1
CAPI(K)=0.
KSH(K)=0
KSH(K-1)=1
PLBS(I)=PLB(K-1)
DST(I)=0.0
QNP(K-1)=QNP(K)
GO TO 1200,12,0,12,0,12,0,1,KRES
1201 TB(K-1)=(TB(K-1)+RB(K-1)*(TS(K)-TA(K)))*DTH/CAPI(K-1)+TA(K-1)
GO TO 1202
1200 TB(K-1)=TB(K-1)*DTH/CAPI(K-1)+TA(K-1)
1202 CONTINUE
GO TO 7
13 WRITE(6,16) I,K
16 FORMAT (//1,X,23HOURN THROUGH IN COLUMN ,13,5H AT NODE ,13//)
THE=TH
GO TO 17
15 WRITE(6,18) I,K
18 FORMAT (//10X,35HIMPROPER NODAL NUMBERING IN COLUMN ,13,9H AT NODE
,13//)
THF=TH
GO TO 17
11 KDROPL(I)=J
IF(PLBS(I)-PLB(K)-DST(I))/PLBS(I)-0,0,1 7,7,11,00
1100 HA=PLBS(I)-PLB(K)-DST(I)
DST(I)=DST(I)+HA
DST(I)=DST(I)+HA
C-----NEW AREAS AND NEW VCLJME
RAT=1.-DST(I)/PLBS(I)
J=K+1-1
CRL=CRI(J)+RAT*(CRI(J+1)-CRI(J))
CZL=CZ(J)+RAT*(CZ(J+1)-CZ(J))
HA=CRL-CRI(J)
HB=CZL-CZ(J)

```

```

HC=HA*HA+HB*HB
HD=SQRT(HC)
HE=PIB*HD*(CR(J)+CRL)
AA(K)=HE
L=J+MM+1
CRR=CR(L)+RAT*(CR(L+1)-CR(L))
CZR=CZ(L)+RAT*(CZ(L+1)-CZ(L))
MA=CRR-CR(L)
MR=CZR-CZ(L)
HC=HA*HA+HB*HB
HD=SQRT(HC)
HF=PIB*HD*(CRR+CR(L))
AD(K)=HE
HA=CRR-CRL
HB=CZR-CZL
HC=MA*MA+MB*MB
HD=SQRT(HC)
AC(K)=PIR*HD*(CRR+CRL)
MA=CR(J)*CR(J)
MB=CR(J)*CRL
MC=CR(J)*CR(L)
MD=CRL*CRL
HE=CRL*CRR
HF=CR(L)*CR(L)
HG=CR(L)*CRR
HH=CRR*CRR
HS=ZRO
HS=HS+CZ(J)*(HC-HP+HF-HD)
HS=HS+CZ(L)*(HG-HC+HB-HA)
HS=HS+CZ(L)*(HB-HE+MA-MM)
HS=HS+CZ(L)*(HE-HG+MC-HF)
VOL(LK)=6.76.171E-04*A35(HS)
SR(I)=(CRL+CRR)/2.
SZ(I)=(CZL+CZR)/2.
7 CONTINUE
QNTS=QNTS+QSUM*DTH
17 RETURN
END

```

D-33

```
SUBROUTINE ORDER (NX,X1,I1)  
DIMENSION X1(1),I1(1)  
DIMENSION LS(2)  
NM=IABS(NX)
```

```
LS(1)=1  
LS(2)=1  
LS(3)=2  
LI=3  
I1(1)=1
```

```
DO 1 N=2,NM  
I1(N)=N  
L=LS(LI)  
LA=L
```

```
LA=L  
J=N  
X1C=X1(J)  
I1C=I1(J)
```

```
J=J-L  
IF(J) 31,31,34  
34 LI=LI+1  
LS(LI)=L+L  
GO TO 29
```

```
33 LA=LA-1  
L=LS(LA)  
IF(L) 3,3,41
```

```
41 J=J-L  
32 IF(J) 31,31,29  
31 LA=LA-1  
L=LS(LA)
```

```
J=J+1  
IF(L) 4,4,32  
30 LA=LA-1  
L=LS(LA)
```

```
IF(L) 4,4,42  
42 J=J+L  
29 IF(NX) 229,129,129
```

```
229 IF(X1C-X1(J)) 30,53,33  
129 IF(X1(J)-X1C) 30,53,33  
53 J=J-1  
GO TO 3
```

```
4 J=J+1  
3 M=N-1  
MM=M
```

```
DO 2 K=J,MM  
X1(M+1)=X1(M)  
I1(M+1)=I1(M)
```

```
2 M=M-1  
I1(J)=I1C  
1 X1(J)=X1C  
RETURN  
END
```

```
SUBROUTINE SEQAIN(L,A,C,D)  
DIMENSION A(1),C(1),D(1),L(1)  
LS=J
```

```

1030 DO 30 I1=1,N
      IF (N-I) 203,1030,1030
      I=I1
21  J=L(I)
      L(I)=I
      IF(J-I) 22,30,22
22  IF(I5) 25,23,25
23  SA=A(I)
      SB=B(I)
      SC=C(I)
      SD=D(I)
      IS=I
26  A(I)=A(J)
      B(I)=B(J)
      C(I)=C(J)
      D(I)=D(J)
      I=J
      GO TO 21
25  IF(I5-J) 26,28,26
28  IS=J
      A(I)=SA
      B(I)=SB
      C(I)=SC
      D(I)=SD
30  CONTINUE
.....2030 CONTINUE
      RETURN
      END
      SUBROUTINE LOOK(I1,XL,XA,AB,C,E,Y,C,ION)
      COMMON/LV/KOUT,IEX,DEN,VR,HI(4J),ILO(40),IR(40)
      DIMENSION A(1),R(1),C(1),E(1)
      I=HI(I1)
      IL=ILO(I1)
      IEX=I
      IF(X(IH)-X(IL)) 30,30,29
30  IEX=1
      IF (XL-X(IH)) 3,2,31
31  IF (XL-X(IL)) 6,5,4
29  IF(XL-X(IH))1,2,3
      1 IF(XL-X(IL))4,5,6
      6 I=IR(I)
      I=MIN(I,IH)
      I=MAX(I,IL)
      IS=I
      IT=I
      GO TO 8
11  I=I+1
      IS=I
      8 IF(IEI) 29,28,38
28  IF(XL-X(I)) 7,10,9
38  IF(XL-X(I)) 9,12,7
      7 I=I-1

```

CMA 1001

CMA 1004
CMA 1005

CMA 1006
CMA 1007
CMA 1008

CMA 1009
CMA 1010
CMA 1011
CMA 1012
CMA 1013

CMA 1014
CMA 1015

D-35

CMA 1016
CMA 1017
CMA 1018
CMA 1019
CMA 1020
CMA 1021
CMA 1022
CMA 1023
CMA 1024
CMA 1025

```

IT=J
IF(IIS+10.10.8
9 IF(IT)10.10.11
3 IEX=3
2 I=IH-1
GO TO 10
4 IEX=2
5 I=IL
10 DEN=X(I+1)-X(I)
IR(I)=I
VR=XL-XII)

```

CMA 1027

```

IF(IDN) 13,13,14
14 GO TO (21,22,23,24),IDN
24 Y(4)=E(I)
23 Y(3)=C(I)
D(4)=E(I+1)-E(I)
D(3)=C(I+1)-C(I)
22 Y(2)=B(I)
D(2)=B(I+1)-B(I)
21 Y(1)=A(I)
D(1)=A(I+1)-A(I)
DO 12 J=1,IDN
24 D(J)=D(J)/DEN
12 Y(J)=Y(J)+D(J)*VR
13 VR=VR/DEN
CALL SWITCH(2,J,J)

```

CMA 102

```

GO TO (200,2-1),JJ
20J IF (IDN-2) 2-2,2-2,204
204 IF(II-1) 201,2-2,201
202 WRITE (KOUT,2-3) II,IL,IM,XL,IR(II),VR,DEN,IEX,IDN
203 FORMAT (3(2X,12),2X,E10.3,2X,12,F10.4,2X,E10.3,2(2X,12))
IF (IDN-1) 2-25,1025,1-25
1025 WRITE(KOUT,2-5)(Y(K),D(K),K=1,IDN)
205 FORMAT(4(2X,E10.3))
2025 CONTINUE
201 RETURN

```

CMA 1035

```

END
SURROUTINE SLOPO(NUMX,X,P,EM)
SLOPE EVALUATION ROUTINE
C

```

01M

```

DIMENSION X(1), P(1), EM(1), Z(1)
3J EM(2) = ( P(2) - P(1) ) / ( X(2) - X(1) )
EM(1) = EM(2)
Z(1) = 0
QC = EM(1)
DO 36 I = 1, NUMX
IPO = I + 1
IPT = I + 2
IT = IPO - NUMX
IF (IT) 33, 31, 32
31 JB=QC
GO TO 41

```

02M
02J

025
02U
02W
02Y
03A
03C

GO TO 41

D.36

035
03U
03W
03Y

04C
04E
04G
04I
04K
04M
04O
04U
04Y

```
32 GO TO 40
33 XOT = X(I) - X(IPO)
   XTT = X(IPO) - X(IPT)
   XTO = X(IPT) - X(I)
   AA = P(I) / ( XOT * XTO )
   XOTT=XOT*XTT
37 AP=P(I+1)/XOTT
   AC = P(IPT) / ( XTT * XTO )
   AAA = AA * ATT
   ABB = AP * XTO
   ACC = AC * XOT
   QA = QC
   QB = EM(I)
   QC = EM(IPO)
   EX(IPO) = AC * (XOT - XTT) + ACC - AAA
   EM(IPT) = AC * (ATT - XTO) + AAA - ABB
   EM(I) = AA * (XTO - XOT) + ABB - ACC
34 QE = EM(I)
   IF(I-2) 36,41,35
35 EM(I) = ( QE + QA ) / 2.
41 EM(I) = (EM(I) + Q5) / 2.
40 XD=X(I)-X(I-1)
36 CONTINUE
   RETURN
END
SUBROUTINE OGLE(N,XAM,PRM,NUMX,X,P,E4)
DIMENSION XA(1),X(1),P(1),E(1),PRM(1),DPDIM(1)
XDIF=X(N:MX)-X(1)
IS=1
1T=1
2 DO 6J0 J=1,N
   XA=XAM(J)
59 10a1
   1T=1
61 IF(XDIF) 72,6J,71
71 IF(XA-X(15)) 52,63,64
72 IF(X(15)-XA) 52,63,64
62 IF(15-1)671,671,68
68 IS=15-1
   1T=2
GO TO (61,66),10
672 IS=NUMX
671 I=15
   H=0.
   DPDI=EM(I)
   GO TO 67
63 PR=P(15)
   DPDI=EM(15)
   GO TO 601
64 IS=15+1
69 10=2
   IF(15-NUMX)69,69 ,672
   GO TO (61,65),1T
65 IS=15-1
```

```

66 I=15
   G=(L*PLI+1)-P(I)/(X(I+1)-X(I))-EM(I)/(X(I+1)-X(I))
   F=((FM(I+1)-FM(I))/(X(I+1)-X(I))-2.*G)/(X(I+1)-X(I))
   H=(F*(XA-X(I+1))+G)*(XA-X(I))
   DPDI=(H+H*EM(I)+F*(XA-X(I)))*(XA-X(I))
67 PR=(H+EM(I))*(XA-X(I))+P(I)
601 CONTINUE
   PRM(J)=PR
600 CONTINUE
60 CONTINUE
4 RETURN
END
SUBROUTINE SLOPL(M,N,X,Y,EMS,FMN)
COMMON CRIA(2),CZ(6-2),AA(4-2),AB(4-2),AC(4-2),AD(4-2)
C-----DIMENSIONED AS NODES
COMMON MATL(2-2),KTH(2-2),KSH(2-2),KWE(2-2),KTU(2-2),CON(2-2),
1IA(2-2),IB(2-2),CRA(2-2),CRB(2-2),PLA(2-2),PLB(2-2),PLC(2-2),
2PLD(2-2),VOL(2-2),RA(2-2),RB(2-2),CAP(2-2),ENT(2-2),QNP(2-2),
3VF1(2-2),VF3(2-2)
C-----DIMENSIONED AS SURFACE NODES (NUMBER OF COLUMNS)
COMMON KSUR(4-2),TS(4-2),DSDI(4-2),DSDTB(4-2),DST(4-2),U(4-2)
COMMON PLB5(4-2)
DIMENSION X(1),Y(1),EMS(1),FMN(1)
100 J=M
   RAI=1-DST(1)/PLAS(1)
   CRL=CR(J)+RAT*(CR(J+1)-CR(J))
   CZL=CZ(J)+RAT*(CZ(J+1)-CZ(J))
   L=J+M+1
   CRR=CR(L)+RAT*(CR(L+1)-CR(L))
   CZR=CZ(L)+RAT*(CZ(L+1)-CZ(L))
   DZ=CZR-CZL
   DR=CRR-CRL
   IF(DZ) 151,152,151
150 S1=J+E*30
   GO TO 152
151 IF(DR) 153,154,153
154 S1=J+E*30
   GO TO 152
153 S1=DR/DZ
152 S2=1./S1
   EMS(1)=S1
   EMN(1)=S2
   RETURN
101 S1=0.
   S2=0.
   NS=0
   DXS=X(2)-X(1)
   K=N-1
   DO 2-2 I=1,K
   DX=X(I+1)-X(I)
   DY=Y(I+1)-Y(I)
   IF(DX) 102,3-3,3-2

```

```

300 IF(DY) 301,32,33,34
320 NS=NS+1
   GO TO 200
301 S2=2.E+10/ABS(DY)*DY
   GO TO 309
302 S2=DY/DX
   IF(DX<DXS) 304,309,303
304 IF(S1+S2) 307,306,307
306 EMS(I)=1.0E+30
   GO TO 305
307 EMS(I)=2.*S1+S2/(S1+S2)
   GO TO 309
303 EMS(I)=(S1+S2)/2.
309 DXS=DX
   IF(NS) 20,32,33,321
321 LL=I-NS
   LU=I-1

```

```

NS=0
DO 323 J=LL,LU
323 EMS(J)=EMS(I)
200 S1=S2
   EMS(N)=2.*EMS(I)
   EMS(N)=S2
   DO 310 I=1,N
   IF(EMS(I)) 311,312,311
312 EMN(I)=1.0E+30
   GO TO 310
311 EMN(I)=-1./EMS(I)
310 CONTINUE
   RETURN
500 FORMAT(2I5)
   END

```

SUBROUTINE LCOUNT (I,LCT,NPG,R)

C-----PHILCO/ASTHMA VERSION
COMMON/LK/KOJT,TEX,DEH,VR,IHI(4J),ILO(40),IR(40)

```

DIMENSION R(2)
551 FORMAT(IHI,25X,7IHAEROTHEM: AXI-SYMMETRIC TRANSIENT HEATING AND MA
552 IERIAL ABLATION PROGRAM/113X,4MPAGE,13/100X,2A6)
553 FORMAT(IHI,25X,7IHAEROTHEM: AXI-SYMMETRIC TRANSIENT HEATING AND MA
554 IERIAL ABLATION PROGRAM/111X6MPAGES,13/100X,8HTHROUGH,13/108X,

```

```

22A6)
1JK=I
IF (1JK) 2,2,3
2 1JK=-1JK
   GO TO 4
3 LCT=LCT-1JK
   IF (LCT) 4,5,5
4 NPG=NPG+1
   LCT=55-1JK
   IF(LCT) 6,7,7
6 NEX=(1JK-1)/55
   LCT=(NEX+1)*55-1JK
   N=NPG

```

D-39

```

NPG=NPG+NEK
WRITE(KOUT,552) N,NPG,R
GO TO 5
7 CONTINUE
WRITE (KOUT,551) NPG,R
5 RETURN
END
FUNCTION VCOS(K,I,MM,CR,CZ,SZ,SR,EMN,PLB,PLBS,DSI)
C-----SPECIAL ANDERSON/SCHAEFER/ARC RESTRICTED VERSION
DIMENSION CR(I),CZ(I),SZ(I),SR(I),EMN(I),PLB(I)
DIMENSION PLBS(I),DSI(I)
J=K+I-1
L=J+MM+1
R1=(CR(J)+CR(L))/2.
Z1=(CZ(J)+CZ(L))/2.
DZ=SZ(I)-Z1+1.E-30
DIST=PLBS(I)-DSI(I)
EMNA=ABS(EMN(I))
VCOS=DZ/(DIST *SQRT(1.+EMNA**2))*(EMN(I)/EMNA+(SR(I)-R1)/DZ*EMNA)
1/12.0
VCOS=ABS(VCOS)
RETURN
END
```

APPENDIX E

ARCACE COMPUTER PROGRAM USER'S MANUAL

MITCHELL R. WOOL

ROBERT M. KENDALL

ARCACE COMPUTER PROGRAM USER'S MANUAL

CARD SET 1 - CONTROL CARD (1 CARD)

Field 1 (Columns 1-10, 1011) Control Variable, KR(10).

KR is used to control the various program options.

- Column 1 - State Option
- 0 Assigned temperature
 - 1 Surface equilibrium with kinetics at assigned temperature
 - 2 Assigned enthalpy
 - 3 Constant entropy expansion from previous chamber calculation
 - 8 Stop
- Column 2 - Elemental Composition
- 0 Accept resident values (same as input to a previous solution)
 - ≥1 Input new elemental composition data for the system components (see card set 2)
- Column 3 - Molecular Data
- 0 Accept resident values for molecular data
 - 1,2,3,4 Read from cards JANAF data (card set 3)
 - 3,4 Store all molecular data read from cards on tape
 - 5,6 Read all molecular data from tape as last stored
 - 2, 4, 6 Output JANAF data for selected set of molecules
- Column 4 - Unused
- Column 5 - Unused
- Column 6 - Unused
- Column 7 - Diagnostic Control
- 0 No diagnostic output
 - 1 Single line of diagnostics per iteration
 - j(>1) For 5(j-1) iterations full diagnostics will be output and for subsequent iterations the single line will be output
- Column 8 - Unused
- Column 9 - Kinetics Input Option, If KR(1) = 1, (Col. 1)
- 0 Use previously input kinetic reaction data
 - 1 Input kinetic reaction data (see card set 4)
- Column 10 - Card Output Option
- 0 No card output
 - 1 Output single card per convergent solution of data appropriate to ASTHMA input

Field 2 (Columns 11-20, F10.5) State Variable, Z.
 For non-zero Z, Z is assigned to T (temperature, °K) for KR(1) of 0 or 1, to HIP (enthalpy, cal/gm) for KR(1) of 2, and to SIP (entropy, cal/gm °K) for KR(1) of 3.

A negative temperature entry results in a frozen composition of the system (as previously obtained) and the calculation of properties of this system at the absolute value of the assigned temperature.

If Z is zero resident values of temperature, enthalpy or entropy will be used in the solution specification consistent with the KR(1) assignment.

Field 3 (Columns 21-30, F10.5) Pressure Variable, PR.

For non-zero PR, PR is assigned as system pressure, P (atmospheres). For zero PR, P is unaltered (=1. before first solution).

Field 4 (Columns 71-80, 2A4,A2) Job Name, TILE(3)

This will appear on printed and punched output.

CARD SET 2 - ELEMENTAL COMPOSITION DATA (IS+1 CARDS)

Card 1

Field 1 (Columns 1-3, I3) Number of Elements in the System, IS

Cards 2,3,..., IS+1. One Card for Each Element

Field 1 (Columns 1-3, I3) Atomic Number of an Element, KAT(99 is not allowable).

Field 2 (Columns 4-15, 3A4) Name of Element (in three parts), ATA, ATB and ATC.

For output identification only.

Field 3 (Columns 16-25, F10.5) Atomic Weight of Element, WAT.

Field 4 (Columns 26-55, 3F10.5) Relative Elemental Composition, TK
 Relative amounts of element in components 1, 2 and 3 (edge gas, pyrolysis gas and char, respectively for surface mass balances). Positive values are in relative gram-atomic-weights (or moles) and negative values are in relative gram-mass units. For example, consider component 2 containing 50. mass percent of C and 50. mass percent of SiO₂ or equivalently 60.06 grams of C for each 60.06 grams of SiO₂ (1 mole of SiO₂) or

finally 60.06 gm of carbon for each 1. gram-atomic-weight of Si and 2 gram-atomic-weights of O. For this example it would be most convenient to input in columns 36-45 of the cards for C,O and Si: - 60.06, 2., and 1., respectively.

CARD SET 3 - THERMOCHEMICAL EQUILIBRIUM DATA (3 CARDS PER SPECIES)

Cards 1,4,7, etc. One for each species, whether molecular, atomic, or condensed. This card describes the element composition of the species and establishes its name designation.

Fields 1,3,5,
...,13

(Columns 1-3, 7-9, 13-15, ..., 37-39, each F3.0)
Number of Atoms, ALPT

Fields 2,4,6,
...,14

(Columns 4-6, 10-12, 16-18, ..., 40-42, each I3)
Atomic Number, JAT

The field sets, 1-2, 3-4, 5-6, ..., 13-14, represent the number of atoms, ALPT, of atomic number, JAT, contained in this species. One field set for each element of the species. For example, NO₂ is described in fields 1 through 4 as bblbb7bb2bb8; that is, one atom of Nitrogen (atomic number 7) and two atoms of Oxygen (atomic number 8).

If field 1 is zero this card is presumed to represent the end of the card set.

Field 15

(Columns 43-72, 7A4A2) Thermochemical Data Source Identification, SØRCE.

Used for output only.

Field 16

(Columns 73-80, 2A4) Species Name Designation, AMØA and AMØB.

This variable is used for output.

Cards 2,5,8,...

One for Each Species

Field 1

(Columns 1-9, E9.6) Heat of Formation, RA.

Species heat of formation at 298°K from JANAF base state (elements in natural form at 290°K), cal/mol.

Fields 2,3,4,
5, and 6

(Columns 10-54, 5E9.6) Curve Fit Constants, RB, RC, RD, RE and RF.

Constants appropriate to lower temperature range of thermodynamic data. The thermochemical curve fits are as follows with T in °K.

Heat Capacity

$$C_p = R_C + R_D T + R_E / T^2, \quad \text{cal/mol } ^\circ\text{K}$$

Enthalpy

$$h - h_{298^\circ\text{K}} = R_B + \left[R_C T + \frac{1}{2} R_D T^2 - R_E / T \right]_{3,000^\circ\text{K}}^T, \quad \text{cal/mol}$$

Entropy

$$S = R_F + \left[R_C \ln T + R_D T - \frac{1}{2} R_E / T^2 \right]_{3,000^\circ\text{K}}^T, \quad \text{cal/mol } ^\circ\text{K}$$

Field 7 (Columns 61-66, F6.0) Upper Limit of Lower Temperature Range, TU in °K

Field 8 (Columns 07, 11) Phase Specification for Lower Temperature Range, KPHA.

KPHA equals 1, 2 or 3 for gas, solid or liquid, respectively.

Cards 3,6,9 ... One for Each Molecule. Same as cards 2,5,8 ..., except for upper temperature range of thermochemical data.

The arrangement of these three card sets is, mathematically, unimportant for equilibrium solutions. In fact, if no condensed phase species exist within the system, the convergence path should be identical. Numerically this is not always true. A single species can dominate more than one mass equation and singular arrays may result. Such a situation can be avoided by placing these dominant species at the front of the thermochemical data deck.

The program automatically selects as base species the first set of species satisfying the requirement that 1) all other species may be formed from this base species set and 2) that no balanced reaction can be written

involving only base species. One base species may be considered to represent each element. In the numerics of the program, the base species actually become the "elements" of the mass balances. The means of establishing whether one species may be formed from another are the specifications given in fields 1 to 14 of cards 1,4,7,...., etc. of card set 3. Any reaction which can occur based on these specifications will be considered equilibrated in the system under study. To suppress equilibrium, it is thus required to change names (i.e., atomic numbers) of certain elements and to introduce new "elements."

The last card of this set is blank and serves as the data sentinel for card set 3. Three card data sets have been generated at Aerotherm for over 500 species mostly based on the JANAF thermochemical data. The data format of these cards is the same as that described in NAVWEPS Report 7043.

CARD SET 4 - REACTION RATE INFORMATION

Card 1

Field 1 (Column 1-3, I3) Number of Kinetically Controlled Reactions to be Considered, MT.

Defines the number of 3 card subsets containing reaction rate data, reactant coefficients, and product coefficients, respectively; one subset for each reaction.

Cards 2,5,8,... Forward Rate Data

Field 1 (Columns 1-10, E10.4) Pre-exponential Factor Divided by $\rho_e u_e C_M$, FKF.

The units of the input parameter, FKF, are mols of reaction per unit mass per unit driving potential (in atmosphere units).

Field 2 (Columns 11-20, E10.4) Activation Energy for the Forward Reaction, EAK, cal/gram.

Field 3 (Columns 21-30, E10.4) Reaction Exponent, EXK.
The driving potential as obtained from the reaction stoichiometry is raised to this power in evaluating the reaction rate.

Cards 3,6,9,... Reactant Coefficients

Fields 1,2,3,... (Columns 1-10, 11-20, 21-30, ..., 8E10.4) One for Each Base Species

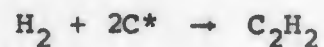
The stoichiometric coefficients on the reactants. In the present formulation only base species may

be used as reactants. Thus some care must be used in establishing the order of card set 3 and thus fixing a priori the set of base species.

Cards 4,7,10,... Product Coefficients

Fields 1,2,3... (Columns 1-10, 11-20, 21-30, ..., 8E10.4) One for Each Base Species

The stoichiometric coefficients on the products or their equilibrium base species equivalents if they are not base species. For example, if the base species are CO, H₂O, H₂, C* for a system where C* is the only isolated (nonequilibrium) species a reaction written as



could equivalently be written



and this is the manner in which the product stoichiometric coefficients would be input.

APPENDIX F
ARCACE PROGRAM SAMPLE PROBLEMS
MITCHELL P. WOOL

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ISENTHROPIC EXPANSION CALCULATION INPUT DECK

APG 112D

011 0 003901. 68.

Species	CO	CO2	H2O	H2	O2	AL	ALCL3	ALN	AL2O3	BE	BECL2	BEF2	BEH2	BEH4	BEH6	BEH8	BEH10	BEH12	
1 HYDROGEN	1.008	3.2219																	
6 CARBON	12.011	0.5649																	
7 NITROGEN	14.008	0.5366																	
8 OXYGEN	16.000	2.2486																	
13 ALUMINUM	26.970	1.0130																	
17 CHLORINE	35.457	0.5369																	
1 6 1 8 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0																			
-264170+5	223570+5	865040+1	117021-3	-898211+6	653700+2	500.	3000.1												
-264170+5	223570+5	115496+2	-424139-3	-131563+8	653700+2	3000.	5000.1												
2 1 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0																			
000000-0	212100+5	711963+1	621950-3	-712694+6	484650+2	500.	3000.1												
000000-0	212100+5	681794+1	589854-3	265106+7	484650+2	3000.	5000.1												
1 6 2 8 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0																			
-940540+5	365350+5	144559+2	210386-3	-182392+7	798480+2	500.	3000.1												
-940540+5	365350+5	156451+2	-381561-4	-602768+7	798480+2	3000.	5000.1												
2 1 1 8 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0																			
-577980+5	302010+5	112254+2	811397-3	-260800+7	684210+2	500.	3000.1												
-577980+5	302010+5	157278+2	-191548-3	-173599+8	684210+2	3000.	5000.1												
1 6 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0																			
000000-0	144120+5	586075+1	953976-4	-766621+6	121290+2	500.	3000.2												
000000-0	144120+5	485134+1	291605-3	307202+7	121290+2	3000.	5000.2												
1 13 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0																			
240000+4	189130+5	696390+1	125221-4	347759+5	251963+2	500.	3000.3												
240000+4	189130+5	744466+1	-758263-4	-210774+7	251963+2	3000.	5000.3												
1 8 1 13 1 17 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0																			
-189599+6	520567+5	190512+2	572817-3	-702975+6	550490+2	500.	1500.2												
-189599+6	514899+5	205116+2	-165920-3	-149571+7	548210+2	1500.	3000.2												
1 13 3 17 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0																			
-168579+6	101100+6	185370+2	114499-1	-169000-1	100183+3	300.	460.2												
-160559+6	720036+5	274254+2	-331897-3	-164863+6	104630+3	460.	3000.3												
1 7 1 13 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0																			
-760000+5	312665+5	122768+2	-178828-4	-673685+6	299013+2	500.	2790.2												
-760000+5	314405+5	408674+2	-925846-4	-230400+9	299601+2	2790.	5000.2												
1 4 4 8 2 13 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0																			
-544999+6	126706+6	349699+2	797503-2	-109209+7	112014+3	500.	2000.2												
-544999+6	126469+6	421930+2	570773-2	-118459+8	111929+3	2000.	3000.2												
3 8 2 13 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0																			
-400400+6	811817+5	319699+2	120451-4	-217514+7	772577+2	500.	2318.2												
-366770+6	759318+5	351153+2	-192777-4	-555942+6	893107+2	2318.	5000.3												
1 4 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0																			
000000-0	193025+5	828125+1	-288958-5	-173046+7	170595+2	500.	1556.2												
288100+4	182708+5	769547+1	-257476-5	-213324+7	184265+2	1556.	5000.3												
1 4 2 17 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0																			
-118599+6	625645+5	160605+2	445380-2	-220604+6	659883+2	300.	682.2												
-117399+6	763159+5	290199+2	0+0	686422-2	804288+2	682.	2000.3												
1 4 2 9 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0																			
-241957+6	546211+5	20746+2	-139307-4	558937+6	583302+2	500.	2000.2												
-241957+6	548613+5	210844+2	256246-5	-533447+6	584292+2	2000.	5000.2												
1 4 1 8 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0																			
-143100+6	337038+5	160771+2	-434631-3	-593740+7	291887+2	500.	2820.2												

SPECIES SET TO BE USED FOR EXPANSION OR EQUILIBRIUM CALCULATIONS ONLY

GENERAL GAS PHASE SPECIES SET PRESENTED IN SURFACE THERMOCHEMISTRY SAMPLE PROBLEM, WHICH FOLLOWS, GDES HERE

CONDENSED SPECIES SET FOR EXPANSION OR EQUILIBRIUM CALCULATIONS ONLY

CONDENSED SPECIES SET
(CONCLUDED)

AT. NO.	ELEMENT	ATOMIC WT	COMPONENT 1	COMPONENT 2	COMPONENT 3
-129562+6	359122+5	153149+2	182960-3	-190747+6	347723+2
2	4	1	6	0	0
-222000+5	519028+5	229814+2	-271974-5	-635112+7	430672+2
-653500+4	526826+5	195224+2	723914-5	-483551+6	499909+2
3	4	2	7	C	0
-137800+6	863645+5	386258+2	-227477-3	-117238+8	776686+2
-110735+6	916598+5	288203+2	-194536-3	779046+8	907280+2
1	1	1	5	2	6
-191869+6	849683+5	943587+1	133656-1	-323428+5	692658+2
-191869+6	573189+5	276870+2	-207617-2	-355870+7	568310+2
1	5	1	7	0	0
-595100+5	291896+5	119161+2	239906-4	-891069+6	257049+2
-595100+5	290739+5	117157+2	257181-5	-240029+6	256618+2
4	1	2	5	4	8
-337099+6	199911+6	287825+2	279798-1	-837893+6	167294+3
-337099+6	164343+6	580007+2	572414-2	-828021+7	151554+3
2	5	3	8	0	0
-300978+6	811530+5	299659+2	700906-3	-130981+7	828910+2
-300978+6	811530+5	288865+2	659184-3	884491+7	828910+2
3	5	3	8	3	9
-586499+6	190799+6	286473+2	258435-1	-572281+6	184615+3
-586499+6	142747+6	679396+2	-414850-2	-105348+8	163343+3
3	1	3	5	3	8
-301699+6	167244+6	311184+2	194520-1	-123978+7	157175+3
-301699+6	139716+6	581678+2	-647494-5	-656656+7	145381+3
4	5	1	6	0	0
-929999+4	829591+5	230566+2	537468-2	-105961+7	681801+2
180999+5	790399+5	325000+2	-160344-8	-237919+2	767634+2
3	6	4	13	0	0
-489999+5	117824+6	411943+2	249972-2	-159504+7	114878+3
-489999+5	116642+6	454353+2	560830-3	-459367+7	114410+3
4	1	1	7	1	17
-753799+5	877759+5	239667+2	525476-2	-392315+6	910481+2
-753799+5	789689+5	300000+2	-152997-9	-247332-0	872322+2
4	1	1	7	4	8
-706900+5	275314+6	145787+2	523599-1	302440+5	224155+3
-706900+5	231364+6	623232+2	200088-1	-150588+8	205225+3
200	0	00	54.4		
300	0	00	48.0		
300	0	00	31.4		
300	0	00	16.0		

ISENTHROPIC EXPANSION CALCULATION OUTPUT

AEROTHERM CHEMICAL EQUILIBRIUM JOB APR 1970 011000000*

RELATIVE ELEMENTAL COMPOSITIONS, ATOMIC WTS/UNTT MASS

AT. NO.	ELEMENT	ATOMIC WT	COMPONENT 1	COMPONENT 2	COMPONENT 3
1	HYDROGEN	1.00800	0.322581	0	0
6	CARBON	12.01100	0.056555	0	0
7	NITROGEN	14.00800	0.053722	0	0
8	OXYGEN	16.00000	0.22512	0	0
13	ALUMINUM	26.97000	0.101417	0	0
17	CHLORINE	35.45300	0.051752	0	0

ELEMENT	HYDROGEN	CARBON	NITROGEN	OXYGEN	ALUMINUM	CHLORINE
PHASE SP	42	CO	N2	H2O	ALCL	CLM

477 X 100 0.000000

AERIAL CHEMICAL EQUILIBRIUM (CASE) SOLUTION

0.42201000 0.122690000 0.77940000 0.46840000 0.11170000

PROPERTY POINTS OUTPUT IN LB-MASS, FT, SEC, BTU/LB AND DEG R
 0.7210004 0.48220004 0.951490004 0.264800004 0.101500004 0.744340000

0.7100000 0.198410002 0.334800002 0.674470004 0.589300000
 ELEMENTAL X AND Z MASS FRACTIONS BY ATOMIC NUMBER : 1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17
 0.1990000 0.113530000 0.752540000 0.360190000 0.273520000 0.190590000
 0.1090000 0.113530000 0.126720000 0.201400000 0.537500000 0.246100000

TEMP = 3901.0000 DEG K PRES = 60.0000 ATM MOLE WT = 53.4870000

SPECIES	MOLE FR.	SPECIES	MOLE FR.	SPECIES	MOLE FR.	SPECIES	MOLE FR.
-2-	0.107490000	CO	0.198209000	-2-	0.329040000	ALCL	0.129100000
-2	0.092690000	CLM	0.024940000	-2	0.355780000	CLM	0.024940000
-22	0.705530000	C	0.977790000	-22	0.295150000	CLM	0.024940000
-3-	0.102040000	C2M	0.011170000	-3-	0.210930000	CLM	0.024940000
-3	0.590320000	C2M2	0.007740000	-3	0.156410000	CLM	0.024940000
-4	0.174300000	NO	0.057210000	-4	0.125540000	CLM	0.024940000
-40	0.173550000	CL2	0.202420000	-40	0.125540000	CLM	0.024940000
-42	0.164490000	ALCL2	0.190510000	-42	0.125540000	CLM	0.024940000
-4L	0.361020000	ALC	0.005770000	-4L	0.125540000	CLM	0.024940000
-4L-	0.538490000	ALN	0.048200000	-4L-	0.125540000	CLM	0.024940000
-4L0	0.485430000	ALO	0.022050000	-4L0	0.125540000	CLM	0.024940000
-4L20	0.129160000	AL202	0.022050000	-4L20	0.125540000	CLM	0.024940000
-4L2	0.454430000	CCL2	0.110260000	-4L2	0.125540000	CLM	0.024940000
-4L20	0.111220000	CCL3	0.044260000	-4L20	0.125540000	CLM	0.024940000
-4L2	0.372040000	CHCL3	0.073100000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.113150000	CH2CL2	0.090890000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.162290000	CH3CL	0.052210000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.074310000	C2M2	0.046760000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.374990000	C2M4	0.010170000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.267150000	C3M2	0.045250000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.590430000	C3M4	0.006100000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.173610000	CLM2	0.007130000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.186170000	CLM2	0.046735000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.698320000	MNO	0.036812000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.117490000	M02	0.022575000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.169700000	M4M2	0.007640000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.369120000	N20	0.001673000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.820000000	N205	0.003666000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.891710000	C4M	0.007060000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.584090000	C4M4	0.004400000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.000000000	ALCL3	0.000000000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.000000000	AL2P3	0.000000000	-4L2	0.125540000	CLM	0.024940000
-4L2	0.000000000	CLM404	0.000000000	-4L2	0.125540000	CLM	0.024940000

ENTHALPY = 3.462994000 CAL/GM
 DENSITY = 0.644041000 LB/CU FT
 VISC = 0.000000000 FT/SEC
 MASS = 0.102770000 BTU/LB
 PRESSURE = 60.0000 ATM
 TEMPERATURE = 3901.0000 DEG K
 MASS CONDENSED/MASS GAS = 0.73720000
 AREA = 0.000000000
 M-WT FRACTION = 0.000000000

CHAMBERK CONTAINER
 54.4 ATM, H₂ DEFINED FROM
 FLEASOL SOLUTION

PROPERTY ROUTINE INPUT IN LB=MASS; FT, SFC, BTU/LB AND DEG-R
 VISC COND CRAP FR SC
 0.75254+001 0.95570+004 0.30429+005 0.20130+000 0.74309+001
 0.19037+002 0.33340+002 0.68205+004 0.90924+000
 0.11331+000 0.12647+000 0.20130+000 0.54054+001 0.24653+000

TEMP = 386H, 0M09 DEG-K PRESS = 54.4+00 ATM WCL WT = 33.347+749
 WCL WT = 33.347+749

SPECIES	MOLE FR.	SPECIES	MOLE FR.	SPECIES	MOLE FR.	SPECIES	MOLE FR.
W2	0.10417+001	CO	0.148136+000	L2	0.32587+000	ALCL	0.29614+001
L2	0.08491+001	CLM	0.02547+001	ALCL	0.38784+001	FM4	0.13617+007
FM2	0.70333+002	O	0.070272+002	FM4	0.13617+007	L	0.96873+001
FM3	0.60321+013	C2M	0.06775+000	L	0.96873+001	FM4	0.13617+007
FM4	0.50329+007	C2M2	0.042296+000	FM4	0.13617+007	FM4	0.13617+007
FM5	0.18352+001	NO	0.05928+002	FM4	0.13617+007	FM4	0.13617+007
FM6	0.18298+001	CL2	0.05928+002	FM4	0.13617+007	FM4	0.13617+007
FM7	0.15856+006	ALCL2	0.05928+002	FM4	0.13617+007	FM4	0.13617+007
FM8	0.40177+002	ALC	0.049053+001	FM4	0.13617+007	FM4	0.13617+007
FM9	0.51300+003	ALN	0.045865+011	FM4	0.13617+007	FM4	0.13617+007
FM10	0.46435+004	ALO	0.042837+004	FM4	0.13617+007	FM4	0.13617+007
FM11	0.13218+002	AL2O2	0.04950+002	FM4	0.13617+007	FM4	0.13617+007
FM12	0.35253+007	CCLO	0.09892+004	FM4	0.13617+007	FM4	0.13617+007
FM13	0.07955+009	CCLS	0.094559+005	FM4	0.13617+007	FM4	0.13617+007
FM14	0.29343+007	CMCL3	0.04290+014	FM4	0.13617+007	FM4	0.13617+007
FM15	0.00149+000	CMCL2	0.070031+013	FM4	0.13617+007	FM4	0.13617+007
FM16	0.11433+004	CM3CL	0.03132+011	FM4	0.13617+007	FM4	0.13617+007
FM17	0.60403+031	C2M3	0.073972+009	FM4	0.13617+007	FM4	0.13617+007
FM18	0.18882+019	C2M6	0.029622+011	FM4	0.13617+007	FM4	0.13617+007
FM19	0.13375+014	C3M2	0.077394+015	FM4	0.13617+007	FM4	0.13617+007
FM20	0.24559+016	C3M5	0.06744+014	FM4	0.13617+007	FM4	0.13617+007
FM21	0.16135+004	CLM2	0.06427+014	FM4	0.13617+007	FM4	0.13617+007
FM22	0.18457+004	CLM3	0.06713+004	FM4	0.13617+007	FM4	0.13617+007
FM23	0.63563+006	CLM4	0.043137+000	FM4	0.13617+007	FM4	0.13617+007
FM24	0.97614+012	M02	0.03257+005	FM4	0.13617+007	FM4	0.13617+007
FM25	0.13521+004	M4N2	0.019793+005	FM4	0.13617+007	FM4	0.13617+007
FM26	0.32666+012	N2O	0.058224+011	FM4	0.13617+007	FM4	0.13617+007
FM27	0.61549+019	N2O5	0.01443+006	FM4	0.13617+007	FM4	0.13617+007
FM28	0.37936+021	C4M	0.016378+022	FM4	0.13617+007	FM4	0.13617+007
FM29	0.18521+019	C6M4	0.017401+017	FM4	0.13617+007	FM4	0.13617+007
FM30	0.01	ALCLO*	0.01480+020	FM4	0.13617+007	FM4	0.13617+007
FM31	0.01	AL2O3*	0.014874+000	FM4	0.13617+007	FM4	0.13617+007
FM32	0.01	CLM4V04*	0.014874+000	FM4	0.13617+007	FM4	0.13617+007

ENTHALPY = 0.4629940+003 CAL/GM
 DENSITY = 0.356761+000 LB/CU.FT.
 VELOCITY = 0.19398+004 HYU/LR
 MASS FRACTIONS BY ATOMIC NUMBER : 1, 13, 17
 ENTPY = 0.21000+000 CAL/GM DEG K
 MASS COMPENSATED MASS GAS = 0.73440+000 J-PHASE = 0.
 AREA = 0.
 METRIC = 0.
 SCF/LB/SEC

HEMISPHERIC EXPANSIONS FROM AR.ME
QUALITY CONTROL

APPROXIMATE CHEMICAL COMPOSITION FOR GAS, 1-4 300000000

PROFESSOR: CP-LEGUL CLAM/UNIT TLM/UNIT CARMA
0.4280000 0.1262000 0.0019000 0.1665700 0.0116200

PROPERTY ROUTINE: VOLUME IN LB MASS, FT, SPECIFIC GRAVITY, AND DEG-R
WAVE: VISC COND CBAR DU
0.0000000 0.0000000 0.9515000 0.3380000 0.1000000 0.7444000 SC

0.7730000 0.1960000 0.3342000 0.1758000 0.0912000
ELEMENTAL ANALYSIS: MASS FRACTIONS BY ATOMIC NUMBER: 1 17

0.5251400 0.6792900 0.7522600 0.3401900 0.2735200 0.1005900
0.1491000 0.1137700 0.1269800 0.0996500 0.1522300 0.1248200

TEMP = 3897.2134 DEG-K PRESS = 40.0000 ATM MOLE WT = 11.4280000

SPECIES	MOLE FR.	SPECIES	MALE FR.	SPECIES	MOLE FR.
W2	0.10327000	CO	0.14189000	W2	0.32810000
W3	0.08948000	CLM	0.04237000	ALCL	0.28873000
P07	0.06975000	O	0.32846000	P2	0.13580000
P3-	0.04454000	C2H	0.04342000	P4	0.11759000
P	0.04216000	C2H2	0.01000000	P	0.09585000
W0	0.17540000	NO	0.04497000	PW	0.16919000
PL	0.18330000	CL2	0.04805000	PW	0.11142000
W0P	0.13681000	ALCL2	0.03879000	ALW02	0.26623000
AL	0.37410000	ALC	0.00126000	ALCL0	0.15132000
AL-	0.04225000	ALN	0.06189000	ALCL0	0.14618000
ALC	0.41274000	ALO	0.14566000	ALCL3	0.09882000
ALP0	0.11910000	AL2O2	0.09622000	PCL	0.09546000
PCLA	0.31075000	CLC	0.08013000	PCL2	0.52384000
PCL20	0.17370000	PCL3	0.03938000	PCL4	0.74734000
P4	0.24232000	CMCL3	0.07844000	CM0	0.78692000
CM2	0.17318000	CM2CL2	0.05940000	CM20	0.15774000
CM3	0.09762000	CM3CL	0.05876500	PV	0.28643000
P2	0.05867000	C2H3	0.00664000	PW4	0.58645000
P2-40	0.13642000	C2H4	0.12502000	P2-2	0.66775000
P3	0.11125000	C3H5	0.03346000	P3-3	0.06450000
P3-4	0.17248000	C3H6	0.04523000	P3-2	0.43358000
CLW0	0.14879000	CLW	0.14554000	P3-2	0.76945000
PL-	0.16884000	CL02	0.04537000	P3-20	0.18691000
WV	0.57229000	WNO	0.09288000	PW02	0.21463000
W-3	0.75332000	W02	0.04897000	PW	0.25748000
P3-	0.12269000	W4N2	0.01927000	P	0.39270000
P0-	0.25921000	W20	0.04884000	P	0.43101000
P2-4	0.40489000	W205	0.07367000	P	0.16006000
P4	0.25151000	C4H	0.04923000	P4-2	0.57413000
P4-3	0.11814500	C4H4	0.09278000	P4-2	0.02185000
AL-	0.00000000	ALCL0	0.00000000	ALCL1	0.00000000
AL0	0.00000000	AL2O3	0.00000000	ALCL4	0.00000000
P1-4N	0.00000000	CLW4N04	0.00000000	P	0.00000000

ENTHALPY = 0.4916510003 CAL/GM
 DENSITY = 0.1189250000 LB/FT³
 MASS FRACTIONS: MASS FRACTIONS BY ATOMIC NUMBER: 1 17
 MACR = 0.104004 FT/SEC
 MOLECULAR WEIGHT = 11.4280000
 CAL/GM DEG-K
 MASS FRACTIONS BY ATOMIC NUMBER: 1 17
 MASS FRACTIONS BY ATOMIC NUMBER: 1 17

1-5

AFROTHERY CHEMICAL EQUILIBRIUM JOB COLS. 3-4 300000000

CP-FROZEN CP-EQUIL QLEN/MINT CLAM/DLAF GAMMA
0.42489+00 0.12149+00;7429+000 0.4403+00 0.11152+000

PROPERTY ROUTINE OUTPUT IN LBMASS, FT. SEC, BTU, AND DEG F
VISC COND PPAR MU SC
0.4455+00 0.4592+00 0.93696+00 0.48799+00 0.29921+000 0.74442+000

7.7445+000 0.19725+012 0.33700+002 0.64739+004 0.5973+000
EQUILVAL W AN Z MASS FRACTIONS BY ATOMIC NUMBER : 1 13 17

0.3254+001 0.67929+001 0.75254+001 0.34019+000 0.27352+000 0.19059+000
0.17149+000 0.11539+000 0.12064+000 0.28414+000 0.44348+001 0.25389+000

TEMP = 3692.1133 DEG-K PRES = 31.4000 ATM MCI WT = 33.690412A

SPECIES	MOLE FR.	SPECIES	MOLE FR.	SPECIES	MOLE FR.
H2	0.10200+000	CO	0.148367+000	H2	0.33654+000
H2O	0.89875+001	CLM	0.140822+000	ALCL	0.26263+001
H2O2	0.67763+002	O	0.2877+002	AL2	0.27344+001
H2O3	0.16088+013	C2H	0.29667+009	CH4	0.71953+004
H2O4	0.23025+007	C2H2	0.51712+009	C6H6	0.91976+004
CL	0.14915+001	NO	0.11859+002	C6H8	0.12542+003
AL	0.18359+001	CL2	0.16430+004	C6H10	0.78645+005
AL2	0.81656+007	ALCL2	0.27895+001	ALH2O2	0.20540+002
AL3	0.28873+002	ALC	0.16908+011	ALCL0	0.11605+003
AL4	0.31702+003	ALN	0.23748+005	ALCL3	0.13947+003
AL5	0.27054+004	ALO	0.10519+002	AL2CL6	0.61872+001
AL2O	0.82177+003	AL2O2	0.47293+004	CL	0.25071+000
ALCLN	0.20249+007	CLCL	0.62182+005	CCL2	0.26383+001
AL2O2	0.50331+009	CLCL3	0.15354+014	CCL4	0.33885+007
AL2O3	0.12543+007	CMCL3	0.43266+014	CMNO	0.52444+004
AL2O4	0.38751+008	CH2CL2	0.13748+011	CM2O	0.10649+005
AL3	0.57341+007	CM3CL	0.16445+009	CM	0.17115+006
AL4	0.17849+011	CM4	0.11113+017	CM4	0.24808+012
AL5	0.45441+016	CM4H	0.11892+014	CM2	0.30542+001
AL6	0.36999+015	CM4O	0.10649+014	CM3	0.14376+009
AL7	0.50491+017	CM5	0.16345+016	CM2	0.18049+003
AL8	0.12231+004	CLN2	0.10099+007	CLNO2	0.34572+002
AL9	0.12336+004	CLO2	0.18205+009	CL2O	0.1591+009
AL10	0.39760+004	MNO	0.08844+005	CM2	0.11774+007
AL11	0.30694+012	M02	0.11675+005	CM4	0.17852+004
AL12	0.88344+005	M4N2	0.26515+017	CM6	0.27965+004
AL13	0.19881+013	N2O	0.42041+006	CM8	0.13544+004
AL14	0.95001+020	N2O5	0.14074+025	CM10	0.73289+000
AL15	0.52147+021	CM4	0.26072+017	CM12	0.14583+004
AL16	0.25939+023	CM4H	0.18874+021	CM14	0.14185+002
AL17	0.00000+000	ALCL0	0.14469+000	CM16	0.00000+000
AL18	0.00000+000	AL2C3	0.14469+000	CM18	0.00000+000
AL19	0.00000+000	CLM4N04	0.00000+000	CM20	0.00000+000

ENTHALPY = 0.5949944+003 CAL/GM ENTROPY = 0.2100+000 CAL/GM-DEG K
 DENSITY = 0.218016+000 LB/CL-FT. MASS CAL/ENR/MASS GAS = 0.77534+000 ENR/IME = 0.
 WEL = 0.333+004 FT/SEC MACL = 0.14+001 ANA = 0.138-000 SDF/LB/SEC
 MOLAFA = 0.17891+004 BTU/LB HEATNETIF = 0.22158+003 BTU/LE

-140423+6	528160+5	199485+2	-209903-4	-446442+6	119584+3	500.	3000.	1	0..ALCL3
-140423+6	528160+5	204484+2	-109302-3	-256062+7	119584+3	3000.	5000.	1	0..ALCL3
1 1 1	8 1 13	0 0 0	0 0 0	0 0 0	OJANAF 12/60				ALHO
-3447C0+4	367340+5	148073+2	231376-4	-155462+7	803550+2	500.	3000.	1	0..ALHO
-3447C0+4	367340+5	143991+2	973440-4	115747+6	803550+2	3000.	5000.	1	0..ALHO
1 8 1	13 C 0	0 0 0	0 0 0	0 0 0	OJANAF 03/62				ALO
213980+5	238050+5	895587+1	772498-4	-329625+6	718590+2	500.	3000.	1	0..ALO
213980+5	238050+5	916408+1	387904-4	-116511+7	718590+2	3000.	5000.	1	0..ALO
2 13 6	17 C 0	0 0 0	0 0 0	0 0 0	OJANAF 03/64				AL2CL6
-3092C0+6	116290+6	437070+2	153763-5	-644812+6	211446+3	1000.	3000.	1	0..AL2CL6
-3092C0+6	116290+6	437101+2	159950-5	-620287+6	211446+3	3000.	5000.	1	0..AL2CL6
1 2 2	13 C 0	0 0 0	0 0 0	0 0 0	OJANAF 09/61				AL20
-3144C0+5	359700+5	138891+2	485413-5	-663265+6	912000+2	500.	3000.	1	0..AL20
-3144C0+5	359700+5	144656+2	-108570-3	-278920+7	912000+2	3000.	5000.	1	0..AL20
2 2 2	13 C 0	0 0 0	0 0 0	0 0 0	OJANAF 12/61				AL202
-953970+5	504970+5	198593+2	224091-5	-134144+7	105665+3	500.	3000.	1	C..AL202
-953970+5	504970+5	185468+2	738440-2	144995+9	105665+3	3000.	5000.	1	0..AL202
1 4 0	C C C	0 0 0	0 0 0	0 0 0	OJANAF 09/61				BE
782550+5	134380+5	464695+1	112298-3	334374+6	440210+2	500.	3000.	1	0..BE
782550+5	134380+5	105297-0	109147-2	147714+8	440210+2	3000.	5000.	1	0..BE
1 4 1	17 C C	0 0 0	0 0 0	0 0 0	OJANAF 03/64				BECL
300000+4	239830+5	893411+1	908676-4	-222435+6	719090+2	1000.	3000.	1	0..BECL
300000+4	239830+5	894691+1	882523-4	-267069+6	719090+2	3000.	5000.	1	0..BECL
1 4 2	17 C C	0 0 0	0 0 0	0 0 0	OJANAF 09/61				BECL2
-579119+5	388209+5	146996+2	608946-4	-425949+6	899049+2	500.	3000.	1	0..BECL2
-579119+5	388209+5	149007+2	485558-6	-604893+6	899049+2	3000.	5000.	1	0..BECL2
1 4 1	5 C 0	0 0 0	0 0 0	0 0 0	OJANAF 03/63				BEF
-496780+5	234680+5	888699+1	919571-4	-421766+6	684020+2	500.	3000.	1	0..BEF
-496780+5	234680+5	842934+1	174020-3	148137+7	684020+2	3000.	5000.	1	0..BEF
1 4 2	9 C 0	0 0 0	0 0 0	0 0 0	OJANAF 12/63				BEF2
-151300+6	374160+5	147905+2	289774-4	-115672+7	822830+2	500.	3000.	1	0..BEF2
-191300+6	374160+5	138154+2	209528-3	274481+7	822830+2	3000.	5000.	1	0..BEF2
1 1 1	4 C 0	0 0 0	0 0 0	0 0 0	OJANAF 3/63				BEH
767680+5	229360+5	877224+1	184173-3	-924908+6	607720+2	500.	3000.	1	0..BEH
767680+5	229360+5	884739+1	160554-3	-963556+6	607720+2	3000.	5000.	1	0..BEH
1 1 1	4 1 8	0 0 0	0 0 0	0 0 0	OJANAF 09/63				BEHO
-250000+5	321730+5	129854+2	208547-3	-239494+7	785130+2	500.	3000.	1	0..BEHO
-250000+5	321730+5	139033+2	-916197-5	-477784+7	785130+2	3000.	5000.	1	0..BEHO
2 1 1	4 C C	0 0 0	0 0 0	0 0 0	OJANAF 12/60				REH2
300000+5	345780+5	145807+2	757364-4	-277162+7	676530+2	500.	3000.	1	0..REH2
300000+5	345780+5	153556+2	-872682-4	-534470+7	676530+2	3000.	5000.	1	0..REH2
2 1 1	4 2 8	0 0 0	0 0 0	0 0 0	OJANAF 09/63				BEH2U2
-158500+6	593950+5	244524+2	313890-3	-473550+7	103667+3	500.	3000.	1	0..BEH2U2
-158500+6	593950+5	252829+2	957844-4	-632116+7	103667+3	3000.	5000.	1	0..BEH2U2
1 4 1	7 C C	0 0 0	0 0 0	0 0 0	OJANAF 06/63				BEH
101599+6	236599+5	871835+1	168927-3	-253217+6	692429+2	500.	3000.	1	0..BEH
101599+6	236599+5	894126+1	102713-3	-471704+6	692429+2	3000.	5000.	1	0..BEH
1 4 1	8 C C	0 0 0	0 0 0	0 0 0	OJANAF 09/63				BEI
310000+5	231840+5	885450+1	909641-4	-552570+6	661420+2	500.	3000.	1	0..BEI
310000+5	231840+5	894167+1	685229-4	-731204+6	661420+2	3000.	5000.	1	0..BEI
2 4 4	17 C C	0 0 0	0 0 0	0 0 0	OJANAF 12/60				RE2CL4
-178999+6	844529+5	316826+2	340183-4	-456569+6	162027+3	500.	3000.	1	C..RE2CL4

-178559+6	844529+5	317876+2	141294-5	-521197+6	162027+3	3000.	5000.1	0.-RE2CL4
2 4 1	8 C C	0 0 0	0 0 0	0 0 0	0 JANAF	09/63		RE20
-150000+5	375349+5	141882+2	210823-3	-681583+6	829279+2	500.	3000.1	0.-RE20
-150000+5	375349+5	149040+2	-185011-6	-142668+7	829279+2	3000.	5000.1	0.-BE20
2 4 2	8 C C	0 0 0	0 0 0	0 0 0	0 JANAF	09/63		BE202
-580000+5	458360+5	197854+2	208746-4	-163011+7	988560+2	500.	3000.1	0.-RE202
-580000+5	458360+5	192883+2	113721-3	337096+6	988560+2	3000.	5000.1	0.-BE202
3 4 3	8 C C	0 0 0	0 0 0	0 0 0	0 JANAF	09/63		BE303
-252000+6	768800+5	316149+2	406679-4	-413048+7	124702+3	500.	3000.1	0.-BE303
-252000+6	768800+5	320268+2	-453443-4	-551555+7	124702+3	3000.	5000.1	0.-BE303
4 4 4	8 C C	0 0 0	0 0 0	0 0 0	0 JANAF	09/63		BE404
-380000+6	107856+6	436112+2	233046-4	-439314+7	156562+3	500.	3000.1	0.-BE404
-380000+6	107856+6	439409+2	-408308-4	-562941+7	156562+3	3000.	5000.1	0.-RE404
5 4 5	8 C C	0 0 0	0 0 0	0 0 0	0 JANAF	09/63		BE505
-505000+6	137169+6	553874+2	617881-4	-542571+7	184299+3	500.	3000.1	0.-BE505
-505000+6	137169+6	555197+2	226391-4	-555927+7	184299+3	3000.	5000.1	0.-RE505
6 4 6	8 C C	0 0 0	0 0 0	0 0 0	0 JANAF	07/63		BE606
-636000+6	166373+6	672405+2	798827-4	-658038+7	211572+3	500.	3000.1	0.-BE606
-636000+6	166373+6	665239+2	197656-3	-331145+7	211572+3	3000.	5000.1	0.-RE606
1 5 0	C C C	0 0 0	0 0 0	0 0 0	0 JANAF	12/60		B
132618+6	134240+5	506024+1	-307108-4	-995641+5	481210+2	500.	3000.1	0.-H
1 5 2	9 1 17	0 0 0	0 0 0	0 0 0	0 JANAF	12/63		0.-B
-211600+6	503830+5	198311+2	886329-5	-147403+7	106267+3	500.	3000.1	BCLF2
-211600+6	503830+5	195580+2	581861-4	-347534+6	106267+3	3000.	5000.1	0.-BCLF2
1 5 1	9 2 17	0 0 0	0 0 0	0 0 0	0 JANAF	12/63		BCL2F
-154000+6	511180+5	198460+2	578157-5	-110178+7	110423+3	500.	3000.1	0.-BCL2F
-154000+6	511180+5	198057+2	126436-4	-924350+6	110423+3	3000.	5000.1	0.-BCL2F
1 5 1	9 C 0	0 0 0	0 0 0	0 0 0	0 JANAF	12/60		BF
-454690+5	232890+5	887795+1	901824-4	-526109+6	669360+2	500.	3000.1	0.-BF
-454690+5	232890+5	816859+1	210898-3	279334+7	669360+2	3000.	5000.1	0.-BF
1 5 1	8 1 9	0 0 0	0 0 0	0 0 0	0 JANAF	12/63		BFO
-144000+6	374660+5	148017+2	227294-4	-134022+7	843900+2	500.	3000.1	0.-RFO
-144000+6	374660+5	157870+2	-169659-3	-501345+7	843900+2	3000.	5000.1	0.-RFO
1 5 2	9 C 0	0 0 0	0 0 0	0 0 0	0 JANAF	12/62		BF2
-130000+6	352490+5	137981+2	306556-4	-979515+6	874860+2	500.	3000.1	0.-BF2
-130000+6	352490+5	148647+2	-178860-3	-507255+7	874860+2	3000.	5000.1	0.-BF2
1 1 1	5 2 9	0 0 0	0 0 0	0 0 0	0 JANAF	12/65		BF2M
-175399+6	472759+5	175176+2	676573-3	-126325+7	949169+2	500.	3000.1	0.-BF2H
-175399+6	472759+5	198231+2	682930-5	-392997+7	949169+2	3000.	5000.1	0.-BF2H
1 5 1	8 2 9	0 0 0	0 0 0	0 0 0	0 JANAF	12/65		BF20
-249999+6	497319+5	190106+2	255063-3	-106089+7	103271+3	500.	3000.1	0.-BF20
-249999+6	497319+5	198673+2	528861-6	-189838+7	103271+3	3000.	5000.1	0.-RF20
1 5 3	9 C C	0 0 0	0 0 0	0 0 0	0 JANAF	12/63		BF3
-270100+6	496940+5	197643+2	260131-4	-174925+7	100317+3	500.	3000.1	0.-BF3
-270100+6	496940+5	190100+2	163117-3	133775+7	100317+3	3000.	5000.1	0.-BF3
1 1 1	5 C 0	0 0 0	0 0 0	0 0 0	0 JANAF	03/63		BH
105630+6	226370+5	871723+1	193099-3	-113880+7	593240+2	500.	3000.1	0.-BH
105630+6	226370+5	962344+1	127652-4	-442568+7	593240+2	3000.	5000.1	0.-RH
1 1 1	5 1 8	C 0 C	0 0 0	0 0 0	0 JANAF	12/60		BHO
-471270+5	356030+5	142974+2	167620-3	-201136+7	764730+2	500.	3000.1	0.-BHO
-471270+5	356030+5	112392+2	644865-3	134967+8	764730+2	3000.	5000.1	0.-BHO

-5653CC+6	123234+6	472395+2	722094-3-292334+7	179085+3	500.	3000.1	0.83F303
-5653CO+6	123234+6	496561+2	357165-5-527295+7	179085+3	3000.	5000.1	0.83F303
3 1 3 5	3 8 0 0	0 0 0	0 0 0	0 JANAF 03/65			R3H303
-291CCC+6	115496+6	427771+2	198595-2-368114+7	157365+3	500.	3000.1	0.83H303
-251CCC+6	115456+6	495563+2	166186-4-115216+8	157365+3	3000.	5000.1	0.83H303
3 1 3 5	5 2 0 0	0 0 0	0 0 0	0 JANAF 12/64			R3M306
-542559+6	160889+6	594694+2	222751-2-381642+7	208309+3	500.	3000.1	0.83H306
-542559+6	160889+6	672365+2	437510-4-147581+8	208309+3	3000.	5000.1	0.83H306
6 1 3 5	3 7 0 0	0 0 0	0 0 0	0 JANAF 03/65			R3M6N3
-121859+6	15C18C+6	537397+2	387996-2-518944+7	180383+3	500.	3000.1	0.83H6N3
-121859+6	15C18C+6	671331+2	581970-4-225429+8	180383+3	3000.	5000.1	0.83H6N3
1 5 1 6	C C C C	0 0 0	0 0 0	0 JANAF 06/63			CB
158CCC+6	233870+5	887101+1	987600-4-461666+6	689650+2	500.	3000.1	0.83H6N3
158CCC+6	233870+5	864500+1	134780-3 549853+6	689650+2	3000.	5000.1	0.83H6N3
1 6 1 17	C C C C	0 0 0	0 0 0	0 JANAF 12/60			CCL
132CCO+6	237850+5	890513+1	416720-4-154333+6	736510+2	500.	3000.1	0.83H6N3
132CCO+6	237850+5	857088+1	993826-4 129567+7	736510+2	3000.	5000.1	0.83H6N3
1 6 3 9	1 17 0 0	0 0 0	0 0 0	0 JANAF 03/64			CCLF3
-166CCO+6	654320+5	257053+2	334186-4-172444+7	120680+3	1000.	3000.1	0.83H6N3
-166CCO+6	654320+5	258243+2	146187-5-193275+7	120680+3	3000.	5000.1	0.83H6N3
1 6 1 7	1 17 0 0	0 0 0	0 0 0	0 JANAF 12/60			CCLN
322CCO+5	375090+5	415997+3	109631-0-651419+9	868930+2	500.	3000.1	0.83H6N3
322CCO+5	375090+5	134744+2	286947-3 348958+7	868930+2	3000.	5000.1	0.83H6N3
1 6 1 8	1 17 0 0	0 0 0	0 0 0	0 JANAF 12/65			CCLN
-150CCO+5	355119+5	131269+2	228137-3-408663+6	926619+2	500.	3000.1	0.83H6N3
-150CCO+5	355119+5	139003+2	148315-5-124889+7	926619+2	3000.	5000.1	0.83H6N3
1 6 2 17	C C C C	0 0 0	0 0 0	0 JANAF 03/65			CCL2
750CCO+5	363009+5	137335+2	526254-4-381617+6	936489+2	500.	3000.1	0.83H6N3
750CCO+5	363009+5	139114+2	748804-7-560298+6	936489+2	3000.	5000.1	0.83H6N3
1 6 2 9	2 17 0 0	0 0 0	0 0 0	0 JANAF 03/64			CCL2
-1150CO+6	662240+5	257477+2	222257-4-139026+7	125629+3	1000.	3000.1	0.83H6N3
-1150CO+6	662240+5	258288+2	844195-6-154245+7	125629+3	3000.	5000.1	0.83H6N3
1 6 1 8	2 17 0 0	0 0 0	0 0 0	0 JANAF 12/60			CCL2
-526CCO+5	5C5950+5	197348+2	333599-4-133141+7	108899+3	500.	3000.1	0.83H6N3
-526CCO+5	5C5950+5	189745+2	170558-3 180683+7	108899+3	3000.	5000.1	0.83H6N3
1 6 3 17	C C C C	0 0 0	0 0 0	0 JANAF 03/65			CCL20
349959+5	5166C9+5	195967+2	821072-4-603556+6	113242+3	500.	3000.1	0.83H6N3
349959+5	5166C9+5	198625+2	165563-5-823646+6	113242+3	3000.	5000.1	0.83H6N3
1 6 1 9	3 17 0 0	0 0 0	0 0 0	0 JANAF 03/64			CCL3
-680CCO+5	665930+5	257811+2	137845-4-106628+7	128911+3	1000.	3000.1	0.83H6N3
-680CCO+5	665930+5	258333+2	838371-7-116610+7	128911+3	3000.	5000.1	0.83H6N3
-2594CO+5	676540+5	258277+2	144070-5-828977+6	129983+3	500.	3000.1	0.83H6N3
-2594CO+5	676540+5	253023+2	998371-4 124339+7	129983+3	3000.	5000.1	0.83H6N3
1 6 1 9	0 0 0 0	0 0 0	0 0 0	0 JANAF 03/61			CF
744CCO+5	234630+5	889329+1	942918-4-469503+6	701280+2	500.	3000.1	0.83H6N3
744CCO+5	234630+5	810974+1	241697-3 264745+7	701280+2	3000.	5000.1	0.83H6N3
1 6 1 7	1 9 0 0	0 0 0	0 0 0	0 JANAF 06/61			CFN
-300CCO+4	370140+5	146756+2	528937-4-156924+7	836950+2	500.	3000.1	0.83H6N3
-300CCO+4	370140+5	156821+2	153382-3-505794+7	836950+2	3000.	5000.1	0.83H6N3
1 6 1 8	1 9 0 0	0 0 0	0 0 0	0 JANAF 12/65			CFN
-409559+5	347970+5	130071+2	264190-3-618325+6	373199+2	500.	3000.1	0.83H6N3

-4C559+5	347970+5	139862+2	-131517-4	-194234+7	H73199+2	3000.	5000.1	0.CFO
1 6 2	9 C 0	0 C 0	0 0 0	0 0 0	OJANAF 12/63			CF2
-30000+5	349740+5	138552+2	128704-4	-118696+7	876320+2	500.	3000.1	0.CF2
-30000+5	349740+5	142542+2	-646408-4	-248538+7	876320+2	3000.	5000.1	0.CF2
1 6 1	8 2 9	0 C 0	0 0 0	0 0 0	OJANAF 12/63			CF20
-152500+6	490560+5	196967+2	414619-4	-206358+7	100607+3	500.	3000.1	0.CF20
1 6 3	9 C 0	0 0 0	0 0 0	0 0 0	OJANAF 12/63			CF3
-116500+6	458600+5	197865+2	200977-4	-166384+7	102161+3	500.	3000.1	0.CF3
-116500+6	458600+5	196575+2	404917-4	-105310+7	102161+3	3000.	5000.1	0.CF3
1 6 4	9 C 0	0 0 0	0 0 0	0 0 0	OJANAF 12/63			CF4
-222000+6	645420+5	243637+2	395728-3	723010+5	113627+3	500.	3000.1	0.CF4
-222000+6	645420+5	265430+2	-135803-3	-518966+7	113627+3	3000.	5000.1	0.CF4
1 1 1	6 C 0	0 C 0	0 0 0	0 0 0	OJANAF 03/61			CH
142000+6	221300+5	826079+1	302211-3	-100184+7	616120+2	500.	3000.1	0.CH
142000+6	221300+5	707091+1	463281-3	552860+7	616120+2	3000.	5000.1	0.CH
1 1 1	6 3 17	0 0 0	0 0 0	0 0 0	OJANAF 12/60			CHCL3
-250000+5	634760+5	252779+2	125963-3	-287824+7	121245+3	500.	3000.1	0.CHCL3
-250000+5	634760+5	257850+2	237794-5	-410561+7	121245+3	3000.	5000.1	0.CHCL3
1 1 1	6 1 8	1 9 0	0 0 0	0 0 0	OJANAF 06/61			CHFO
-500000+5	462359+5	167821+2	878833-3	-128352+7	944419+2	500.	3000.1	0.CHFO
-500000+5	462359+5	198008+2	951575-5	-498090+7	944419+2	3000.	5000.1	0.CHFO
1 1 1	6 3 9	0 C 0	0 0 0	0 0 0	OJANAF 12/63			CHF3
-165100+6	611030+5	252113+2	1415226-3	-400405+7	109082+3	500.	3000.1	0.CHF3
-165100+6	611030+5	259284+2	-224709-4	-603019+7	109082+3	3000.	5000.1	0.CHF3
1 1 1	6 1 7	1 8 C 0	0 0 0	0 0 0	OJANAF 12/60			CHNO
-279000+5	461930+5	179701+2	469024-3	-700904+7	928990+2	500.	3000.1	0.CHNO
-279000+5	461930+5	210167+2	-229915-3	-105568+8	928990+2	3000.	5000.1	0.CHNO
2 1 1	6 C 0	0 C 0	0 0 0	0 0 0	OJANAF 12/62			CH2
550000+5	329560+5	132894+2	413822-3	-290273+7	684940+2	500.	3000.1	0.CH2
550000+5	329560+5	140728+2	132150-3	-239493+7	684940+2	3000.	5000.1	0.CH2
2 1 1	6 2 17	0 0 0	0 0 0	0 0 0	OJANAF 12/60			CH2CL2
-224000+5	595790+5	247355+2	249585-3	-481774+7	110269+3	500.	3000.1	0.CH2CL2
-224000+5	595790+5	258799+2	-161283-4	-794316+7	110269+3	3000.	5000.1	0.CH2CL2
2 1 1	6 2 9	0 0 0	0 0 0	0 0 0	OJANAF 12/63			CH2F2
-107200+6	579240+5	246913+2	260835-3	-564164+7	102228+3	500.	3000.1	0.CH2F2
-107200+6	579240+5	256947+2	195707-4	-815777+7	102228+3	3000.	5000.1	0.CH2F2
2 1 1	6 1 8	0 0 0	0 0 0	0 0 0	OJANAF 03/61			CH20
-277000+5	437910+5	184022+2	379921-3	-431982+7	848880+2	500.	3000.1	0.CH20
-277000+5	437910+5	189628+2	161963-3	-372770+7	848880+2	3000.	5000.1	0.CH20
3 1 1	6 C 0	0 C 0	0 0 0	0 0 0	OJANAF 12/62			CH3
319400+5	434190+5	182763+2	401025-3	-451203+7	786040+2	500.	3000.1	0.CH3
319400+5	434190+5	204899+2	-108028-3	-114692+8	786040+2	3000.	5000.1	0.CH3
3 1 1	6 1 17	0 0 0	0 0 0	0 0 0	OJANAF 12/60			CH3CL
-206340+5	560300+5	241353+2	388635-3	-654504+7	972200+2	500.	3000.1	0.CH3CL
-206340+5	560300+5	256598+2	197863-4	-103065+8	972200+2	3000.	5000.1	0.CH3CL
3 1 1	6 1 9	0 0 0	0 0 0	0 0 0	OJANAF 12/63			CH3F
-560000+5	552230+5	241434+2	386480-3	-701904+7	935320+2	500.	3000.1	0.CH3F
-560000+5	552230+5	257207+2	985709-5	-110460+8	935320+2	3000.	5000.1	0.CH3F
1 6 1	7 C 0	0 C 0	0 0 0	0 0 0	OJANAF 12/62			CN
109000+6	232490+5	655906+1	115326-2	479517+6	669760+2	500.	3000.1	0.CN
109000+6	232490+5	988013+1	313855-3	-649453+7	669760+2	3000.	5000.1	0.CN

8 OXYGEN 16.000 0.0092299
 13 ALUMINUM 26.970 C.0015542
 17 CHLORINE 35.457 C.C053752
 112 SURF CARBON 12.011 0.0002004
 144 FRCZEN CO2 44.011
 CCC C 01-4000.
 CCC C 01-3500.
 CCC C 01-3000.
 CCC C 01-2500.
 CCC C 01-2000.
 CCC C 01-1500.
 CCC C 01-1000.
 CCC C 01- 500.
 100 C 11 1000.

EDGE TABLE
 EDGE TABLE
 EDGE TABLE
 EDGE TABLE
 EDGE TABLE
 EDGE TABLE
 EDGE TABLE
 THROAT

4
 .385 E05 .820 E05 1.0 -1.0
 1.0 1.0
 .551 E13 .192 E06 1.0 1.0
 2.0
 -2.0 2.0
 .350 E07 .936 E05 1.0 3.0
 1.0 -1.0
 1.0 1.0

1.0
 1.0
 100 1 01 1500.
 100 0 01 2000.
 100 0 01 2250.
 100 0 01 2500.
 100 0 01 2750.
 100 0 01 3000.
 100 0 01 3250.
 100 C 01 3500.
 100 C 01 4000.
 C10 C 00 3487.9 16.0

THROAT
 THROAT
 THROAT
 THROAT
 THROAT
 THROAT
 THROAT
 THROAT
 COLS. 5-6

2
 1 HYDROGEN 1.008 0.0322563
 6 CARBON 12.011 C.0054663
 7 NITROGEN 14.008 C.C053722
 8 OXYGEN 16.000 0.0088130
 13 ALUMINUM 26.970 0.0012613
 17 CHLORINE 35.457 0.0053752
 112 SURF CARBON 12.011
 144 FRCZEN CO2 44.011 C.C001892
 CCC C 01-4000.
 CCC C 01-3500.
 CCC C 01-3000.
 CCC C 01-2500.
 CCC C 01-2000.
 CCC C 01-1500.

EDGE TABLE
 EDGE TABLE
 EDGE TABLE
 EDGE TABLE
 EDGE TABLE

EDGE TABLE
EDGE TABLE
EXIT SURF

CCC C 01-1000.
CCC C 01- 500.
1CC C 11 1000.
4
-482 EC5 .82C EC5 1.0
1.C 1.C -1.0
-691 E13 .152 EC6 1.0
2.C 1.C
-2.C 2.C
-438 EC7 .936 EC5 1.0
1.0 -1.0
1.C 1.0
1.C 1.0
1CC C 01 1500.
1CC C 01 2000.
1CC C 01 2250.
1CC C 01 2500.
1CC C 01 2750.
1CC C 01 3000.
1CC C 01 3250.
1CC C 01 3500.
1CC C 01 4000.
8CC C 00

1.0
3.0
1.0
1.0

EXIT SURF.
EXIT SURF.
EXIT SURF.
EXIT SURF.
EXIT SURF.
EXIT SURF.
EXIT SURF.
EXIT SURF.

CARD COUNT - 650

SURFACE THERMOCHEMISTRY OUTPUT

AFROTHERM CHEMICAL EQUILIBRIUM JOB COI.S. 1-2 0113000000

RELATIVE ELEMENTAL COMPOSITIONS, ATOMIC WTS/UNIT MASS	COMPONENT 1	COMPONENT 2	COMPONENT 3
AT. NO. ELEMENT ATOMIC WT			
1 HYDROGEN	0.0543149	0.	0.
6 CARBON	0.0995102	0.	0.
7 NITROGEN	0.0003791	0.	0.
8 OXYGEN	0.0166260	0.	0.
13 ALUMINUM	0.0000000	0.	0.
17 CALCIUM	0.0000000	0.	0.
112 SURF CARBON	0.0000000	0.0812570	0.
144 FROZEN CO2	0.0000000	0.0000000	0.

ELEMENT HYDROGEN CARBON NITROGEN OXYGEN ALUMINUM CALCIUM
SURF CARBON FROZEN CO2 H2O H2O ALCI CLM
BASE S* 42 CO2 C*

AFLOW-TEMP CHEMICAL EQUILIBRIUM (ACE) SOLUTION

BOUNDARY LAYER FLAME EQUILIBRIUM SOLUTION

CP=FCIL DIM/DLNT PLUM/DLNF GAMMA
0.14433+001 0.51164+000 0.41261+001 0.11793+00

PROPERTY ROUTING OUTPUT IN LB-MASS, FT-SEC, BTU, AND DEG-R
VTSC CCAD ORAP OR SC
0.4899E+004 0.4218E+004 0.03793+004 0.33884+003 0.1218E+000 0.48678+000

MUI MUI WT WTIL CBTIL
0.70735+000 0.19807+003 0.19147+002 0.67485+004 0.99105+000

ELEMENTAL & ANION 7 MASS FRACTIONS BY ATOMIC NUMBER . . . 14
1 7 17 144

0.56755+001 0.11423+000 0.13318+000 0.26602+000 0.80871+001 0.33274+000 0.
0.16906+000 0.11079+000 0.12497+000 0.28076+000 0.52573+001 0.24816+000 0.

TEMP = 3427.2000 DEG-K PRESS = 48.0000 ATM MOL WT = 19.1468277

SPECIES	MOLE FR.	SPECIES	MOLE FR.	SPECIES	MOLE FR.
H2O	0.10535+000	CO	0.49596+003	CO	0.18190+000
H2	0.37822+000	CO2	0.28907+001	NO	0.88944+001
CL4	0.94185+001	ALCL	0.44331+013	C2H2	0.28519+002
O2	0.35947+003	C3H4	0.42327+007	C2H4	0.54238+009
CH4	0.11743+007	H	0.17554+000	NO	0.10039+008
H	0.95844+001	CL	0.13707+004	NO	0.14984+002
CHC	0.14938+003	NO2	0.37458+002	CL2	0.19589+004
CHN	0.11150+004	AL	0.46327+003	ALCL2	0.18892+001
ALF02	0.24709+002	ALW	0.41299+004	ALC	0.50153+011
ALCIC	0.15062+003	ALWC	0.11965+002	ALN	0.36241+005
ALC13	0.14619+003	AL2O	0.31033+007	ALF	0.14601+002
AL2CL6	0.94705+011	CCL4	0.77306+009	AL2O2	0.26649+004
CLL	0.49479+009	CH	0.24207+007	CCLO	0.85979+005
CLL2	0.52291+011	CH2	0.97540+007	CCLS	0.33663+014
CLL4	0.74516+017	CH3	0.49783+011	CHCL4	0.87642+014
CHN0	0.78685+006	C2	0.13622+015	CH2CL2	0.25898+011
CH20	0.15771+005	C3	0.11098+014	CH3CL	0.28717+009
CH	0.29616+006	CL34	0.17193+014	C2H3	0.20021+011
C2F4	0.58513+012	CLW0	0.16892+004	C2H6	0.12472+015
C2A2	0.64651+011	CL	0.57221+004	C3H2	0.33248+014
C3F3	0.44309+015	WV	0.75534+012	C3H5	0.65017+016
C3C2	0.41323+013	WV3	0.25090+012	CLN0	0.14551+006
CLAN02	0.77073+012	N03	0.43630+010	CLN2	0.35648+009
CL20	0.18692+009	N2C4	0.25060+020	WV0	0.29312+003
WVNC2	0.21520+007	C4	0.11747+019	WV2	0.19006+005
WVA	0.39269+004	C4H3	0.61959+020	WV4	0.21914+011
N2C3	0.41220+014	CO	0.00000+000	WV0	0.18861+006
N3	0.14141+009			WV2	0.97819+023
C4F2	0.57244+014			WV4	0.49539+017
C4A2	0.61959+020			C4H4	0.91807+021

ENTHALPY = 0.1058145+004 CAL/GM FVTR0PY = 0.29520+001 CAL/GM-DEG K
 DENSITY = 0.1924480+000 LB/CCU.FT. MASS CONCOND/MASS GAS = 0.
 VFL = . FT/SEC WACL = 0. AREA = 0. SFT/LB/SEC
 W-GAS = 0.19047+004 BTU/LB W-KINETTYC = 0. RTU/LB
 W-POINE = 0.

H-23

A'ROTHERM CHEMICAL FOUTLYRATUM JOB

1000001011

OUTPUT OF KINETIC EQUATION'S AND KINETIC
CONSTANTS - COLUMN 1-2)

*INFLIC REACTANT	1	2	3	4
COEFFICIENTS				
H2O	1.00	0.	0.	1.00
C*	1.00	2.00	1.00	0.
CC	0.	0.	-1.00	0.
M2	-1.00	1.00	0.	1.00
CO2	0.	0.	1.00	0.
N2	0.	0.	0.	0.
CLM	0.	0.	0.	0.
ALCL	0.	0.	0.	0.
PRODUCT				
COEFFICIENTS				
H2O	0.	-2.00	0.	0.
C*	0.	0.	0.	0.
CC	1.00	2.00	1.00	0.
M2	0.	3.00	0.	1.00
CO2	0.	0.	0.	1.00
N2	0.	0.	0.	0.
CLM	0.	0.	0.	0.
ALCL	0.	0.	0.	0.

PRE-EXPONENT	
FACTOR	0.320+005 0.744+013 0.472+007 0.100+001
ACTIVATION	
ENERGY	0.820+005 0.163+004 0.936+005 0.
REACTION	
ORDER	0.100+001 0.100+001 0.100+001 0.100+001

AFROTHERM CHEMICAL EQUILIBRIUM JOB INLET SHIF 1000000001
 TYPICAL SUKI-AGE THERM-HEMICAL
 CALCULATION (COLUMN 1-2)

PROPERTY OUTPUT IN LB-MASS, FT-SEC, BTU, AND DEG-R
 TEMPERATURE 0.495800-004 0.48756-004 0.75408-004 0.19581-003 0.97311-000 0.66975-000 SC
 MOLE WT 0.72510-000 0.20703-000 0.20408-000 0.31164-004 0.95461-000
 FLEMENTAL AND 7 MASS FRACTIONS BY ATOMIC NUMBER
 1 4 7 13 17 112 144
 0.50285-001 0.21922-000 0.11458-000 0.24593-000 0.71406-000 0.76800-000 0.18711-001 0.21600-004
 0.17100-000 0.22131-000 0.17331-000 0.24718-000 0.41449-001 0.20174-000 0.18808-009 0.17318-004

TEMP = 2750.0000 DEG-K PRESS = 48.0000 ATM MOLE WT = 20.4081030

SPECIES	MOLE FR.	SPECIES	MOLE FR.	SPECIFS	MOLE FR.
H2C	0.93906-004	C0	0.10014-004	C0	0.31348-000
H2	0.44078-000	CO2	0.11466-001	A2	0.70349-001
CLF	0.68177-001	ALCL	0.16883-003	C2H	0.35247-008
O2	0.30729-012	C	0.01652-007	C2H2	0.28259-003
CM4	0.78117-003	H0	0.38794-004	N0	0.13396-001
N	0.65230-002	CL	0.71649-003	N0	0.12937-007
CMC	0.10374-003	NO2	0.69111-014	CLP	0.67912-006
CMW	0.28680-001	AL	0.71919-004	ALCL2	0.42685-001
ALF02	0.75125-009	ALM	0.24363-004	ALC	0.16318-011
ALCIC	0.13865-007	ALWC	0.16905-004	ALN	0.16223-007
ALC13	0.13011-002	AL2C	0.23353-004	AL0	0.16347-007
AL2CL6	0.18212-008	CCLA	0.12538-004	AL202	0.39266-011
CCL	0.15144-000	CCL2A	0.15178-009	CCL0	0.14942-005
CCL2	0.56688-010	CM	0.26787-004	CCL3	0.45911-013
CCL4	0.36393-015	CM2	0.10334-005	CHCL4	0.28247-011
CMAN	0.13283-005	CM3	0.41405-003	CH2CL2	0.49831-008
CM20	0.37995-005	C2	0.30301-007	CH3CL	0.33849-005
CN	0.12681-004	C2H4M	0.21339-011	C2H3	0.19545-004
C2F4	0.98136-004	C3	0.16681-004	C2H6	0.14574-006
C2A2	0.17335-004	C3H4	0.47838-005	C3H2	0.34721-004
C3F3	0.31520-004	CLHC	0.32738-009	C3H5	0.49206-004
C3C2	0.1197-009	CLC	0.27889-010	C1N0	0.33531-012
CLAN2	0.68005-022	HN	0.99278-004	CL02	0.15331-019
CL20	0.78177-016	HN03	0.60448-025	HN0	0.26888-010
WNC2	0.44192-016	HN04	0.51359-004	W02	0.15867-014
W2N	0.42662-005	N01	0.19164-024	W4N2	0.75752-012
N	0.91258-007	N204	0.12388-037	N205	0.13776-011
N2C3	0.24181-028	C4	0.30334-000	C40	0.82324-046
O3	0.66876-024	C4H3	0.26541-005	C40	0.52596-004
C4F2	0.61674-004	C0	0.31792-001	C4H4	0.94775-006
C4A2	0.14980-006				
0.4800-002	0.1289-000 0.5200-0050				
4810-2	0.72884-05280-5 0.27807-4 0.0 48988-3 46588-100				

ENTHALPY = 0.4658854-003 CAL/GM ENTHALPY = 0.44437-001 CAL/GM-DEG K
 DENSITY = 0.270068-000 LB/CU.FT. MASS CONDENSATION GAS = 0.
 INLET SURF # 2
 H-PRIVE = 0.12888-000
 0.46589-003 0.46589-0030
 INLET SURF #

* THIS PRINT FORMAT IS KEYSK. UNEXPECTED IN MAIN PROGRAM CHECK

APPENDIX G

FORTRAM IV LISTING OF ARCADE COMPUTER PROGRAM

MITCHELL R. WOOL.

BLANK PAGE

CSPACE

SHAC0050

REALNFIA(10),NFIB(10),FFIN(10),UM(10,10),TAU(10,10)
 DIMENSION TF(150),CIJ(150,1)
 COMPCKIN,KCLT,JAN,MCL,NAB,IOC,N,KKX(5)
 COMPFA FAPCA(150),FAPCB(150),RB(150,2),AC(150,2),
 1 RC(150,2),RE(150,2),RF(150,2),TU(150,2),VN(150),Y(150),WTM(150),
 2 VAL(150,10),FF(150),IFC(150),IP(150),CP(150),H(150),SR(150),
 3 TC(150),VLX(150),E(150),LAMP(150),GAPK(1, 1)
 COMMON A(16,16),H(3),HP,B(16),SCH,RHC,VEL,BL,KR(10),TILE(3),NFF,
 1 XC(50),XG(50)
 COMMON ZKE(10),TLC(10),SPCLC(10),WALJP(10),EB(10),ALP(10),GAMH(10)
 1 ,GAMF(10),TO(10,3),TK(10,3),MAT(10),IR(10),IOAT(10),KAT(10),UM
 2 ,EBL(10)
 COMMONFFA,IS,ISP,P,T,SIP,HIP,EL,FENL,FLIG,NG,IFL,ISPG,KRALP,CPF,IRE
 1,IER,AA,ITS,IN,IL,IT,PCOE,MHELT,SMELT,TPAX,TPIN,MELT,SUMN,SUML,JC,
 2HG,CPG,SVA,SVB,SVC,SVD,SUPC
 COMMONXNC,SAL,VMACH,KRZZ,SPI,GAP,THI,MX,DMCHI,HCH,NCV,TBT,IFRZ,NOA
 IT,SRI,IXG,SVI,KKJ,IXG,IXG,SHI,NCATO,SAZ,FFF,WS,RV,CMF
 COMMONVA,HOS,DUM1,DUM2,EP,IFCJG,ISP2,MTG,VMU2,MTL
 COMMONMT,FKF(10),EAK(10),EXK(10),PMU(10,10),RMU(10,10)
 COMMONIB(11),FNL(10),PNUS(10),SLAM(10),BE(10),RY(10),
 1 IBC(10),JJ(10)
 COMMON SLB(16),SLA(16,16)
 EQUIVALENCE (CC(1),FAPCA(1)),(B(1),X(1)),(GAMK(1),VLAM(1))
 EQUIVALENCE(TU(151),TF(1)),(VMU(1),CIJ(1))
 EQUIVALENCE(TAU(1),A(151)),
 1A(51)),(FFIN(1),A(101))
 5 FORMAT(I3,F8.2,11E10.3/9I3E10.3)
 C 1/O SPECIFICATIONS ALSO IN RERAY

SHAC0320
SHAC0340

KIN=8
 KOUT=5
 JAN=13
 REWIND JAN
 ITS = 0
 10 CALL ZIPIN
 15 IF (KR(2) + KR(3)) 15,15,20
 20 II = C
 25 IF (KR(2)) 30,30,25
 30 CALL IMELP
 35 IF (KR(3)) 40,40,30
 40 IF (KR(3) - 9) 35,40,40
 45 CALL INPLT
 50 CALL BELCH
 55 CALL INPLY
 REWIND JAN
 40 CALL ALPST
 45 IF (KR(7) - 1) 80,80,75
 50 IF (SVA) 60,60,55
 55 SVA = C.
 GCTC 65

SHAC0330
 SHAC0370
 SHAC0380
 SHAC0400
 SHAC0410
 SHAC0420
 SHAC0430
 SHAC0440
 SHAC0450
 SHAC0460
 SHAC0470
 SHAC0480
 SHAC0490
 SHAC0500
 SHAC0510
 SHAC0520

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6C IF(KR(1)-2) 65,70,70
65 MCDE = 2
   JJC = JC
   CALL MAT1
   CALL MAT2
   CALL MAT3
   JC = JJC
7C MCDE=KR(1)
   CALL RERAY(IISPC,A,0,8,0,0,IG)
   CALL PRCP3
75 CALL CLTPT
C***PRINCIPAL ITERATIVE LCOP
80 IF (ITS) 10,90,85
85 IF (MCDE) 95,95,90
9C CALL THERM
   IF (IFRZ) 95,95,150
95 MCDE = KR(1)
   FLIG = 0.
   CALL MAT1
   CALL MAT2
   CALL MAT3
   ITS = ITS + 1
   B1 = B(1)
   IF (KR(8)) 105,105,100
100 CALL KINET
105 MCE = 1
110 IF (KR(7) - 1) 125,120,110
110 IQQ = - 2
115 WRITE(OUT,115)I,I=1,IS)
   FORMAT(/50HITS TEMP PRES*PWT EQUIL ER HASBAL ER SCALE 7(I3,
17H MASBAL/90X3(I3,7H MASBAL))
120 WRITE(OUT,5)ITS,T,AA,EL,ENL,CMF,(E(I),I=1,IS)
125 IF (ITS) 50,75,130
130 IF (ITS - 60) 140,135,140
135 KR(7) = 9
   IQQ = - 2
140 IF (ITS - 70) 155,155,145
145 MCV = 1
   KR(7) = 1
150 ITS = - 1
   GOTC 50
155 CALL SWAP(SLA,SLB,A,8)
   ICT = 10
   DUMI=C.
   IQI = IQQ
160 DO 170 I=1,IS
   IF (IFC(I) - 1) 170,170,165
165 A(I + 2,I + 2) = 1.E + 10
170 CCNTINUE
   ICO = IOI
   CALL RERAY(IN,A(1),IL),0,B(1),1,0,IG)
   ICT = ICT - 1

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SHAC0540
SHAC0550
SHAC0560
SHAC0570
SHAC0580
SHAC0590

SHAC0620
SHAC0630
SHAC0640
SHAC0650
SHAC0660
SHAC0670
SHAC0680
SHAC0690
MAT10280
SHAC0700
SHAC0710
SHAC0720
SHAC0730
SHAC0740
SHAC0750
SHAC0760
SHAC0770
SHAC0780
SHAC0790
SHAC0800
SHAC0810
SHAC0820
SHAC0830
SHAC0840
SHAC0850
SHAC0860
SHAC0870
SHAC0880
SHAC0890
SHAC0900
SHAC0910
SHAC0920

SHAC0940

SHAC0950
SHAC0960
SHAC0970
SHAC0980
SHAC0990
SHAC1000
SHAC1010
SHAC1020

```

```

IF (ICT) 190,175,175
175 IF (IQQ) 190,180,180
180 CALL AFMAT
GOTO(160,90,145,185),NC
185 CALL SCALE(MOE)
CALL CRECT(MOE)
IF (ICM - 1.E - 12) 190,190,205
190 IF (MCI) 195,200,195
195 CALL SWAP(A,B,SLA,SLB)
GOTO 145
200 MCV = 1
GOTO 90
205 IF (KR(7) - 1) 80,80,75
END

```

```

SHAC1030
SHAC1040
SHAC1050
SHAC1060
SHAC1070
SHAC1080
SHAC1090
SHAC1100

SHAC1120
SHAC1130
SHAC1140
SHAC1150
SHAC1160

```

CARD COUNT - 118

AFMA0010
AFMA0030

```

SUBROUTINEAFMAT
REALNFIA(10),NFIB(10),FFIN(10),UM(10,10),TAU(10,10)
DIMENSION TF(150),CIJ(150,1)
DIMENSION X(1),VLAM(150,1),CC(7960)
COMMON FAMD(150),FAMOB(150),RB(150,2),RC(150,2),
1 RD(150,2),RE(150,2),RF(150,2),TU(150,2),VM(150),Y(150),WTM(150),
2 VNL(150,10),FF(150),IFC(150),IP(150),CP(150),M(150),SB(150),
3 TC(150),VLNK(150),E(150),LAMI(150),GANK(1,1)
COMMON A(16,16),M(3),MM,B(16),SCH,RHO,VEL,B(10),KR(10),TILE(3),NEF,
1 XC(150),XG(150)
COMMON ZKE(10),TLO(10),DMCLO(10),WALJP(10),EB(10),AL(10),GAMM(10)
1 ,GAMF(10),TQ(10,3),TK(10,3),MAT(10),IR(10),IOAT(10),KAT(10),UM
2 ,EBL(10)
COMMONFFA,IS,ISP,P,T,SIP,MIP,EL,ENL,FLIG,NQ,IFL,ISPO,KRALP,CPF,IRE
1,IER,AA,ITS,IN,IL,IT,MODE,MMELT,SMELT,TMAX,TMIN,MELT,SUMN,SUML,JC,
ZHG,CPG,SVA,SVB,SYD,SUMC
COMMONXC,SAL,VNACH,KRZZ,SP1,GAM,THI,FX,DMCHI,HCH,HCY,TBT,IFRZ,MOA
1Y,SRI,IXG,SVI,KRJI,IXC,MXG,SHI,NOATO,SAZ,FFF,MS,RV,CMF
COMMONVA,HOS,DUR1,DUR2,EP,IFCJC,ISPZ,HIG,VNUZ,MTL
COMMONRT,FKF(10),EAK(10),EXK(10),PMU(10,10),RMU(10,10)
COMMONIB(11),FNU(10),PNUS(10),SLAM(10),BE(10),BY(10),
1 IBC(10),JJ(10)
COMMON SLB(16),SLA(16,16)
EQUIVALENCE (CC(1),FAMD(1)),(B(1),X(1)),(GAMK(1),VLAM(1))
EQUIVALENCE (I(1),IF(1)),(VNU(1),CIJ(1))
EQUIVALENCE (TAU(1),A(151)),
1A(151)),(FFI(1),A(101))
C***** IF TRYING TO PUSH THROUGH IMIN OR TMAX -- REINVERT AND DI TO ZERO
IF (Y - TMIN) 5,5,10
5 IF (X(1)) 70,20,10
10 IF (I - TMAX) 20,15,15
15 IF (X(1)) 20,20,70
C***** IF NEW CCNDESED HAS NEG CORRECTION, DELETE AFTER REINVERT
20 IF (IER) 30,40,25
25 IF (VNI(IRE)) 30,30,40
30 IF (X(IER) + 1.E - 4) 35,40,40
35 CONTINUE
CALL SWAP(A,B,SLA,SLB)
A(IER,IER) = 1.E20
B(IER) = -1.E15
IF (DUM1) 135,150,150
C***** IF S.E. ERROR AND CORRECTION ON T OF CONFLICTING SIGN, REINVERT
40 IF (MODE - 1) 120,45,125
45 IF (X(1) + B1) 70,50,50
C***** ON S.E. IF DELTA LN Y .GT. .9 REINVERT NO DT IF EL AND EML ARE SMAAFMAD470
50 IF (ABS(X(1)) - 0.9) 125,125,55
55 IF (EML - 0.02) 60,65,65
60 IF (EL - 100.) 125,65,65
65 MODE = 0
C***** REINVERT
70 CALL SWAP(A,B,SLA,SLB)

```

AFMA0410
AFMA0420
AFMA0430
AFMA0440
AFMA0450
AFMA0460
AFMA0470
AFMA0490
AFMA0500
AFMA0510
AFMA0520

```

C*****IF CONVERGED EXCEPT FOR T ON H OR S OPTIONS --- NON CONVERGENT
IF (MODE - 1) 115,80,75
75 IF (EL + 100. * ENL - 1.E - 4) 160,160,115
C*****ON S.E. OPTION RESULTIN IN CONFLICTING ERROR/CORRECTION OR T PUSH AFMA0570
C*****IF OTHER BALANCES RELATIVELY GOOD, SET T TO TMIN/TMAX AS PER ERRORAFMA0580
C*****AND GO TO THERM (IF T ALREADY THERE - NONCONVERGE) ELSE DT TO ZEROAFMA0590
80 IF (ABS(B1) - 100. * (EL + ENL)) 115,85,85
85 IF (B1) 90,90,100
90 IF (T - TMIN) 160,160,95
95 T = AMAX1(TMIN,T - 1000.)
GOTO 155
100 IF (T - TMAX) 105,160,160
105 T = AMIN1(TMAX,T + 1000.)
IF (ABS(T - TF(JC)) - .001) 110,155,155
110 IFC(JC) = 1
VM(JC) = 1.E - 3 / WTM(JC) * MTG
YLJCL = C.
GOTO 155
115 X(1) = 0.
MODE = 0
TMIN = 300.
TMAX = 20000.
IN = IN - 1
IL = 2
GOTO 150
120 X(1) = 0.
125 IF (X(2) + 1.) 130,165,165
130 CALL SWAP(A,B,SLA,SLB)
135 X(2)=X1G
IF (NR(G)) 140,145,145
140 X(2) = X(2) + WTL
145 X(2)=1.E25+XLOG(X(2)/AAJ
DUM1=-1.
A(2,2) = 1.E25
150 NG = 1
GOTO 170
155 NG = 2
GOTO 170
160 NG = 3
GOTO 170
165 NG = 4
170 RETURN
END

```

AFMA0540
 AFMA0550
 AFMA0560
 AFMA0570
 AFMA0580
 AFMA0590
 AFMA0600
 AFMA0610
 AFMA0620
 AFMA0630
 AFMA0640
 AFMA0650
 AFMA0660
 AFMA0670
 AFMA0680
 AFMA0690
 AFMA0700
 AFMA0710
 AFMA0720
 AFMA0730
 AFMA0740
 AFMA0750
 AFMA0760
 AFMA0770
 AFMA0780
 AFMA0790
 AFMA0800

AFMA0830
 AFMA0840

AFMA0860
 AFMA0870
 AFMA0880
 AFMA0890
 AFMA0900
 AFMA0910
 AFMA0920
 AFMA0930
 AFMA0940
 AFMA0950

ZIPI0010
ZIPI0030
ZIPI0040

SUBROUTINEZIPIN

```

REALNFIA(10),NFIB(10),FFIN(10),UM(10,10),TAU(10,10)
DIMENSIONNR(3)
DIMENSIONTF(150),CIJ(150,1)
DIMENSIONK(1),YLAM(150,1),CC(7960)
COMMONKIN,KOUT,JAN,MOL,MAB,IOC,M,KKX(5)
COMMONFAMOA(150),FAMOB(150),RB(150,2),RC(150,2),
1 RD(150,2),RE(150,2),RF(150,2),TU(150,2),VM(150),Y(150),WTM(150),
2 VNU(150,10),FF(150),IFC(150),IP(150),CP(150),M(150),SB(150),
3 TC(150),VLNK(150),E(150),LAMI(150),GAMK(1,1)
COMMONA(16,16),W(3),WM,8(16),SCH,RMO,VEL,B1,KR(10),TILE(3),NFF,
1 XC(50),XG(50)
COMMONZKE(10),ILO(10),DMCLO(10),WALJP(10),EB(10),ALP(10),GAMH(10)
1 ,GAMF(10),IQ(10,3),TK(10,3),MAT(10),IR(10),IOAT(10),KAT(10),UM
2 ,EBL(10)
COMMONFFA,IS,ISP,P,T,SIP,HIP,EL,ENL,FLIQ,NO,IFL,ISPQ,KRALP,CPF,IRE
1,IER,AA,ITS,IN,IL,IT,MODE,HMELT,SMELT,TMAX,TMIN,MELT,SUMM,SUML,JC,
2HG,CPG,SVA,SVB,SYC,SVD,SUMC
COMMONMXC,SAL,VMACH,KRZZ,SP1,GAM,THI,MX,DMCHI,HCH,NCV,TBT,IFRZ,NOA
1T,SRL,IXG,SVL,KKJ,IXC,NXG,SHI,NOATO,SAZ,FFF,WS,RV,CMF
COMMONVA,HOS,DUM1,DUM2,EP,IFCJC,ISP2,WTG,VMU2,MTL
COMMONMT,FKF(10),EAK(10),EXK(10),PMU(10,10),RMU(10,10)
COMMONIC(11),ALPT(10),C(10),SORCE(8),RA(2),IM(10),
1II,KK,J,VINT,YINT,JAT(10)
EQUIVALENCE(CC(1),FAMOA(1)),(B(1),X(1)),(GAMK(1),VLAM(1))
EQUIVALENCE(I(15)),TF(1)),(VMU(1),CIJ(1))
EQUIVALENCE(TAU(1),A(151)),
1A(51)),(FFIN(1),A(101))
5FORMAT(10),6F10.5,2A4,A2)
10FORMAT(213)
15FORMAT(8F10.5)
NFF = 0
Z = 0.0
PR = 0.0
II = 0
IF (ITS + 1) 20,25,20
20 FFA = 0.5
W(1) = 1.0
W(2) = 0.
W(3) = 0.
MX = -1
MCV = 0
NOAT = 0
WM = 20.
T = 3000.
HIP = 0.
SIP = 0.
KR(4) = 0
25 ITS = 0
IDC = 1
KRALP = KR(4)
KKJ = 0

```

ZIPI0290
ZIPI0310
ZIPI0320
ZIPI0330
ZIPI0340

ZIPI0350
ZIPI0360
ZIPI0370
ZIPI0380
ZIPI0390
ZIPI0400
ZIPI0410
ZIPI0420
ZIPI0430
ZIPI0440
ZIPI0450
ZIPI0460

ZIPI0480
ZIPI0490
ZIPI0510

6-7

```
30 MS = 1.0
135 READ(KIN,5)KR,Z,PR,FFR,MR,TILE
   KXX(1) = KR(10)
   WRITE(KCLT,140)TILE,KR
   KR(8) = 0
140 FORMAT (1H19X35HAERTHERM CHEMICAL EQUILIBRIUM JOB 2A4,A2,5X1011) ZIP11040
   KR(5)=4-KR(1)
   KR(6)=-1
   W(3)=0.0
   KR(4)=-1
   IF (KR(1)-1) 155,150,155
150 KR(4)=0
   W(3)=2.0
   KR(6)=0
   KR(8)=1
   KR(1)=0
   KR(5)=0
155 IF (KR(7)) 220,220,156
156 KR(7) = 5 * (KR(7) - 1) + 1
220 MODE = KR(1)
225 IFRZ = 0
   IF (MODE - 8) 275,230,275
230 STOP
275 IF (Z) 280,430,280
280 IF (MODE - 1) 285,285,300
285 Y = Z
290 Y = -Y
   IFRZ = 1
295 CONTINUE
   GOTO 430
300 IF (MODE - 2) 430,415,420
415 HIP = Z
   GOTO 430
420 SIP = Z
   IF (SIP + 9999.) 430,425,430
425 SIP = SCH
430 IF (PR) 440, 40,435
435 P = PR
   AA = P * MM
440 KRALP=KR(4)-KRALP
   RETURN
   END
```

```
ZIP10520
ZIP11020
ZIP11030
ZIP11040
ZIP11110
ZIP11380
ZIP11410
ZIP11430
ZIP11570
ZIP11580
ZIP11590
ZIP11600
ZIP11610
ZIP11620
ZIP11630
ZIP12030
ZIP12040
ZIP12050
ZIP12060
ZIP12070
ZIP12080
ZIP12090
ZIP12100
ZIP12120
```

CARD COUNT -- 95

IMEL0010
IMEL0030
IMEL0040

```

SUBROUTINE IMELM
REAL NFIA(10), NFIB(10), FFIN(10), UM(10,10), TAU(10,10)
DIMENSION ATA(10), ATB(10), ATC(10)
DIMENSION TF(150), CIJ(150,1)
DIMENSION X(1), VLAM(150,1), CC(7960)
COMMON KIN, KOUT, JAN, MOL, NAB, IDC, N, KX(5)
COMMON FAMOA(150), FAMOB(150), RB(150,2), RC(150,2),
1 RD(150,2), RE(150,2), RF(150,2), TU(150,2), VN(150), Y(150), WTM(150),
2 VNU(150,10), FF(150), IFC(150), IP(150), CPI(30), H(150), S8(150),
3 TC(150), VLNK(150), E(150), LAMJ(150), GAMP(1,1)
COMMON A(16,16), M(3), NH, B(16), SCH, RHO, VEL, BL, KR(10), TILE(3), NFF,
1 XC(50), XG(50)
COMMON ZKE(10), TLO(10), DMCL(10), WALJP(10), EB(10), ALP(10), GAMH(10)
1 GAMP(10), TQ(10,3), TK(10,3), MAT(10), IR(10), IOAT(10), KAT(10), SUM
2 EBL(10)
COMMON FFA, IS, ISP, P, T, SIP, HIP, EL, ENL, FLIG, NO, IFL, ISPQ, KRALP, CPF, IRE
1, IER, AA, ITS, IM, IL, IT, MODE, HMELT, SMELT, THAX, THIN, MELT, SUMN, SUHL, JC,
2HG, CPG, SVA, SVB, SVC, SVD, SUMC
COMMON XC, SAI, VPACH, KRZZ, SPI, GAM, THI, RX, DMCHI, HCH, MCV, TBT, IFRZ, NOA
1T, SRL, IXC, SVL, KKJ, IXC, MXG, SHL, MOATO, SAZ, FFF, MS, RV, CMF
COMMON VA, MOS, DUM1, DUM2, EP, IFCJC, ISP2, MTG, VMU2, MTL
COMMON HT, PKF(10), EAK(10), EXK(10), PMU(10,10), RMU(10,10)
COMMON B(11), EMU(10), PMUS(10), SLAM(10), BE(10), DY(10),
1 TBC(3), JJ(10)
EQUIVALENCE (CC(1), FAMOA(1)), (B(1), X(1)), (GAMP(1), VLAM(1))
EQUIVALENCE (TU(1), TE(1)), (VMU(1), CIJ(1))
EQUIVALENCE (TAU(1), A(151)),
1A(51)), (FFIN(1), A(101))
EQUIVALENCE (TQ(1), ATA(1)), (IQ(1), AYB(1)), (IQ(2), ATC(1))
5 FORMAT (I3)
10 FORNAT(13,3A4,4F10.5)
15 FORNAT(15) $RELATIVE ELEMENTAL COMPOSITIONS, ATOMIC WTS/UNIT MASS
1/6X6HAT, NO. 3X7HELEMENT4X9HATOMIC WT4X30HCOMPONENT 1 COMPONENT 2
1COMPONENT 3 / (7X13,3X3A4,F10.5,3F13.7)
20 READ(KIN,5115)
IX = 3
IF (IS - 10) 25,25,80
25 READ(KIN,10)(KAT(J),ATA(J),ATB(J),ATC(J)),WAT(J),TK(J,1),TK(J,2),TK
1(J,3),J=1,IS)
30 DO 55 K=1,3
VA = 0.
DO 50 J=1,15
IF (KAT(J) - 99) 35,45,35
35 IF (TK(J,K)) 40,50,45
40 VA = VA - TK(J,K)
TK(J,K) = - TK(J,K) / WAT(J)
GOTO 50
45 VA = VA + TK(J,K) * WAT(J)
50 CONTINUE
DO 55 J=1,15
55 TK(J,K) = TK(J,K) / VA
WRITE(KOUT,15)(KAT(J),ATA(J),ATB(J),ATC(J),WAT(J),TK(J,1),TK(J,2),TK(J,3))

```

G-4

```
1TK(J,3),J=1,IS)
ISP = IS + 1
IF (KR(3)) 65,65,60
6C IF (KR(3) - 9) 85,65,85
65 CONTINUE
DO 7C L=1,3
DC 7C I=1,IS
TQ(I,L) = 0.
DO 7C K=1,IS
70 TQ(I,L) = TQ(I,L) + UM(I,K) * TK(K,L)
75 FORMAT (///31H PROGRAM LIMITED TO 10 ELEMENTS)
85 RETURN
80 WRITE(KOUT,75)
END
```

IMEL0530
IMEL0540
IMEL0550
IMEL0560

IMEL0580
IMEL0590
IMEL0600
IMEL0610
IMEL0620
IMEL0630
IMEL0640
IMEL0650
IMEL0670

CARD COUNT - 67

INPU0010
INPU0030

```

SUBROUTINE INPUT
REAL NFA(10), NF(10), FFIN(10), UM(10,10), TAU(10,10)
DIMENSION TF(150), CIJ(150,1)
DIMENSION X(1), VLAM(150,1), CC(7960)
COMMON KIN, KOUT, JAN, FCL, NAB, IDC, N, KX(5)
COMMON FAMOA(150), FAMOB(150), RB(150,2), RC(150,2),
1 RD(150,2), RE(150,2), RF(150,2), TU(150,2), VN(150), Y(150), WTM(150),
2 VNU(150,10), FF(150), IFC(150), IP(150), CP(150), H(150), SB(150),
3 TC(150), VLNK(150), E(150), LAMI(150), GAMK(1,1)
COMMON A(16,16), W(3), WM, B(16), SCH, RHO, VEL, B1, KR(10), TILE(3), NFF,
1 XC(150), XG(150)
COMMON ZKE(10), TLO(10), DMCLO(10), WALJP(10), EB(10), ALP(10), GAMH(10)
1 , GAMF(10), TQ(10,3), TK(10,3), WAT(10), IR(10), IOAT(10), KAT(10), UM
2 , EBL(10)
COMMON FFA, IS, ISP, P, T, SIP, HIP, EL, ENL, FLIQ, NQ, IFL, ISPO, KRALP, CPF, IRE
1 , IER, AA, ITS, IN, IL, IT, PODE, HMELT, SMELT, IMAX, TMIN, MELT, SUMN, SUML, JC,
2 HG, CG, SWA, SVB, SVC, SVD, SUPC
COMMON XNC, SAI, VRACH, KRZZ, SP1, GAM, THI, NX, DMCHI, MCH, NCV, TBT, IFRZ, MOA
1 , SRI, IXG, SVI, KRJ, IXC, NXG, SHI, NOATO, SAZ, FFF, NS, RV, CMF
COMMON VAM, MOS, DUM1, DUM2, EP, IEC, JC, ISP2, MIG, YMI2, WIL
COMMON T, FKF(10), EAK(10), EXK(10), PMU(10,10), AMU(10,10)
COMMON IC(11), ALPT(10), C(10), SORCE(8), RA(2), IM(10),
1 , IK, JJ, YINI, YINT, JAT(10), KPHA(2)
EQUIVALENCE (CC(1), FAND(1)), (B(1), X(1)), (GAMK(1), VLAM(1))
EQUIVALENCE (TU(151), TF(1)), (VMU(1), CIJ(1))
F EQUIVALENCE (TAU(11), A(11)),
1 A(51)), (FFIN(1), A(101))
5 FORMAT (I3)
10 FORMAT (A69.6, A69.6, 0, I11)
15 FORMAT (1X6E12.5, F10.4, I2/1X6E12.5, F10.4, I2, F7.4)
20 FORMAT (2A4, E12.4, 2A4, E12.4, 2A4, E12.4, 2A4, E12.4)
25 FORMAT (7F3.0, 13I4, 2, 9A4)
J = ISP
IF (II) 30, 30, 35
30 CONTINUE
NFF = 0
JC = KR(3) - (KR(3) / 2) * 2
YINI = P * I.E. - A
YINT = ALOG(VINT)
JAT(8) = 0
N = 0
II = ISP
J = 1
IEL = -1
IDC = 1
35 IF (KR(3) - 5) 40, 50, 50
40 READ (BIN, 25) (ALPT(K), JAT(K), K=1, 7), SORCE, AMOA, AMOB
IF (KR(3) - 3) 55, 45, 45
45 WRITE (JAN) ALPT, JAT, SORCE, AMOA, AMOB
60 GO TO 55
50 READ (JAN) ALPT, JAT, SORCE, AMOA, AMOB
55 IF (ALPT(1)) 60, 375, 60

```

INPU0290
INPU0300
INPU0310
INPU0320
INPU0340
INPU0350
INPU0370
INPU0390
INPU0400
INPU0410
INPU0420
INPU0430
INPU0440
INPU0450
INPU0460
INPU0470
INPU0490
INPU0510
INPU0530

```

6C IF (JAT(1)) 65,65,85
65 INF = NFF + I
K = ALPT(1)
NFF = K + NFF
IF (KR(3) - 5) 70,80,80
70 READ(KIN,20)(NFIA(I),NFIB(I),FFIN(I),I=INF,NFF)
IF (KR(3) - 3) 35,75,75
75 WRITE( JAN) (NFIA(I),NFIB(I),FFIN(I),I = INF,NFF)
GOTO 35
80 READ( JAN) (NFIA(I),NFIB(I),FFIN(I),I = INF,NFF)
GOTO 35
85 DO 90 N=I,IS
90 C(K) = C.
DO 11C I=1,7
IF (JAT(I)) 95,110,95
95 DO 100 K=1,IS
IF (JAI(I) - KAT(K)) 100,105,100
100 CONTINUE
KK = II
GOTO 285
105 C(K) = ALPT(I)
110 CONTINUE
115 WT = C.
IF (JC) 125,120,125
120 WRITE(KOUT,25)(ALPT(K),JAT(K),K=1,7),SORCE,AMDA,AMOB
125 L = 1
LAMKK = 0
DO 140 I=1,IS
IF (C(I)) 130,135,130
130 LAMKK = LAMKK + L
135 L = L + L
ALPT(I) = C(I)
140 WT = WT + C(I) * MAT(I)
IF (J - IS) 145,145,215
145 JM = J - 1
IF (JM) 175,175,150
150 DO 170 L=1,JM
IML = IM(L)
UGH = C(IML)
UM(L,J) = 0.
IF (UGH) 155,170,155
155 DO 160 I=1,L
160 UM(I,J) = UM(I,J) - UM(I,L) * UGH
DO 165 I=IML,IS
165 C(I) = C(I) - TAU(I,L) * UGH
170 UM(J,L) = 0.
175 DO 180 I=1,IS
IF (ABS(C(I)) - .001) 180,180,185
180 TAU(I,J) = 0.
GOTO 215
185 IM(J) = I
UM(J,J) = 1.

```

INPU0540
 INPU0550
 INPU0560
 INPU0570
 INPU0580
 INPU0590
 INPU0600

INPU0620
 INPU0640
 INPU0650
 INPU0660
 INPU0670
 INPU0680
 INPU0690
 INPU0700
 INPU0710
 INPU0720
 INPU0730
 INPU0740
 INPU0750
 INPU0760

INPU0790
 INPU0800
 INPU0810
 INPU0820
 INPU0830
 INPU0840
 INPU0850
 INPU0860
 INPU0870
 INPU0880
 INPU0890
 INPU0900
 INPU0910
 INPU0920
 INPU0930
 INPU0940
 INPU0950
 INPU0960
 INPU0970
 INPU0980
 INPU0990
 INPU1000
 INPU1010
 INPU1020
 INPU1030
 INPU1040
 INPU1050

```

00 190 L-1,J
190 UM(L,J) = UM(L,J) / C(I)
DC 195 L-1,IS
195 TAU(L,J) = C(L) / C(I)
205 YC = YINT
DO 210 L-1,IS
210 CIJ(J,L) = ALPT(L)
KK = J
KKK = J
J = J + 1
GOTO 285
215 CONTINUE
250 IF (J - IS) 255,255,270
255 DO 260 I-1,JH
260 VNU(I,I) = - UM(I,J)
DO 265 I-J,IS
265 VNU(I,I) = 0.
GOTO 280
270 DO 275 L-1,IS
VNU(L,L) = 0.
DO 275 I-1,IS
275 VNU(I,L) = VNU(I,I) + C(I) * UN(L,I)
280 KK = I
II = II + 1
YC = 0.
285 IF (KR(3) - 5) 290,300,300
290 READ(KIN,10)(RA(K),RB(KK,K),RC(KK,K),RD(KK,K),RE(KK,K),RF(KK,K),TU
1(KK,K),KPHA(K),K1,2)
IF (KR(3) - 3) 305,295,295
295 WRITE( JAN) (RA(K),RB(KK,K),RC(KK,K),RD(KK,K),RE(KK,K),RF(KK
1(KK,K),TUKK,K),KPHA(K),K = 1,2)
GOTO 305
300 READ( JAN) (RA(K),RB(KK,K),RC(KK,K),RD(KK,K),RE(KK,K),RF(KK,
1(KK,K),TUKK,K),KPHA(K),K = 1,2)
305 IF (KK - II) 310,35,310
310 YF(KK) = 20000.
IF (KPHA(1) - KPHA(2)) 315,330,320
315 IF (KPHA(1) + KPHA(2) - 5) 320,335,320
320 WRITE(KOUT,325)AMOA,AMOB
STOP
325 FORMAT (///25H BAD PHASE NUMBERING FOR 244)
330 IF (KPHA(1) - 1) 345,345,340
335 IU(KK,1) = - TUKK,1)
340 FF(KK) = 1.E + 10
IFC(KKK) = - 1
YH(KK) = 0.
Y(KK) = AMAX1(YC,0.)
GOTO 355
345 FF(KK) = (YI / 20.) * FFA
350 YH(KK) = VINT

```

```

INPU1060
INPU1070
INPU1080
INPU1090
INPU1150
INPU1160
INPU1170
INPU1180
INPU1190
INPU1200
INPU1210
INPU1350
INPU1360
INPU1370
INPU1380
INPU1390
INPU1400
INPU1410
INPU1420
INPU1430
INPU1440
INPU1450
INPU1460
INPU1470
INPU1480
INPU1490
INPU1500
INPU1510
INPU1520
INPU1540
INPU1550
INPU1570
INPU1580
INPU1590
INPU1600
INPU1610
INPU1620
INPU1630
INPU1640
INPU1650
INPU1660
INPU1670
INPU1680
INPU1690
INPU1700
INPU1710
INPU1720
INPU1730
INPU1740

```

```
Y(KK) = YINT
355 IF (JC) 365,360,365
360 WRITE(KOUT,15)(RA(K),RB(KK,K),RC(KK,K),RD(KK,K),RE(KK,K),RF(KK,K)),
1TU(KK,K),KPHA(K),K-1,2),FF(KK)
365 FAMOA(KK) = AMOA
FAMOB(KK) = AMOB
LAMI(KK) = LAMK
WIM(KK) = WT
RB(KK,1) = RB(KK,1) + RA(1)
RB(KK,2) = RB(KK,2) + RA(2)
N = N + 1
IF (KK - IS) 35,370,35
370 CONTINUE
375 CONTINUE
390 RETURN
END
```

```
IMPU1750
IMPU1770
IMPU1780
IMPU1790
IMPU1800
IMPU1810
IMPU1820
IMPU1830
IMPU1840
IMPU1850
IMPU1860
IMPU1920
IMPU1930
```

CARD COUNT - 172

BELC0010
 BELC0030
 BELC0040
 BELC0050
 BELC0060

```

SUBROUTINEBELCH
REALNFIA(10),NF(B(10),FFIN(10),UM(10,10),TAU(10,10)
DIMENSIONFAM(10),FAMB(10)
DIMENSIONATZ(10),ATB(10),ATC(10)
DIMENSIONLIM(10,10)
DIMENSIONTF(150),CIJ(150,1)
DIMENSIONX(1),VLAM(150,1),CC(17960)
COMMONKIN,KOUT,JAN,MOL,MAR,JDC,M,KXX(5)
COMMONFAMOA(150),FAMOB(150),RB(150,2),RC(150,2),
1 RO(150,2),RE(150,2),RF(150,2),TU(150,2),VM(150),Y(150),WTM(150),
2 VNU(150,10),FF(150),IFC(150),IP(150),CPI(150),H(150),SB(150),
3 TC(150),VLNK(150),E(150),LAMI(150),GAMK(1,1)
COMMONA(16,16),M(3),MM,B(16),SCH,RMO,VEL,B1,KR(10),TILE(3),NFF,
1 XC(150),XE(150)
COMMONZKE(10),TLO(10),DMCLO(10),WALJP(10),EB(10),ALP(10),GAMH(10)
1 *GAMF(10),TQ(10,3),TK(10,3),MAT(10),IR(10),IOAT(10),KAT(10),UM
2 *EBL(10)
COMMONFFA,IS,ISP,P,T,SIP,MIP,EL,EML,FLIQ,MQ,IFL,ISPO,KRALP,CPP,IRE
1,IER,AA,ITS,IM,IL,IT,MODE,HMELT,SMELT,TMAX,TMIN,MELT,SUMM,SUML,JC,
2 *E,EC,SVA,SVB,SVC,SVD,SUM
COMMONNXC,SAL,VNACH,KRZZ,SPI,GAM,THI,HX,DMCHI,HCH,NCV,TBT,IFRZ,NOA
1T,SRI,IXG,SVI,KKJ,IXG,XXG,SML,NOATO,SZ,FPF,MS,RV,CMF
COMMONVA,MOS,DUM1,DUM2,EP,IEC,IC,ISP2,NEG,VMU2,MTI
COMMONTT,PKF(10),EAK(10),EMK(10),PMU(10,10),AMU(10,10)
COMMONC(11),ALPT(10),C(10),SORCE(8),RA(2),IM(10),
1 *KX,J,XYHT,XYMT,JAY(10)
EQUIVALENCE(C(11),FAMOA(11)),(B(1),X(1)),(GAMK(1),VLAM(1))
EQUIVALENCE(TU(15),TF(11)),(VNU(1),CIJ(1))
EQUIVALENCE(LAM(1),A(11)),(MFI(11),A(11)),(MFI(11),
1A(51)),(FF(101),A(101))
EQUIVALENCE(TQ(1),ATA(1)),(TQ(11),ATB(1)),(TQ(21),ATC(1))
IF IM = 199 15,15,5
5 MEX = M - 150
WRITE(KOUT,10)MEX
10 FORMAT(//11H THERE ARE 12,15H EXCESS SPECIES)
STOP
15 IF (J - IS) 20,20,35
20 M = J - 1
WRITE(KOUT,25)JM,IS,(FAMOA(I),FAMOB(I),I=1,JM)
WRITE(KOUT,30)(FAMOA(I),FAMOB(I),I=1SP,N)
STOP
25 FORMAT(//11H ONLY FOLLOWING13,23H BASE SPECIES FOUND FOR13,
19H ELEMENTS/12X20A4)
30 FORMAT(10H OTHER SPECIES CONSIDERED WERE/12X20A4)
35 DO 45 I=2,IS
JM = 1SP - I
IMJ = IM/JM + 1
IF (JM - 1) 50,40,40
40 DO 45 K=1,JM
UGH = TAU(IM,J,K)
DO 45 I=1,IS
45 UM(I,K) = UM(I,K) - UGH + UM(I,JM + 1)
    
```

BELC0310
 BELC0320
 BELC0330
 BELC0340
 BELC0350
 BELC0360
 BELC0370
 BELC0380
 BELC0390
 BELC0400
 BELC0410
 BELC0420
 BELC0430
 BELC0440
 BELC0450
 BELC0460
 BELC0470
 BELC0480
 BELC0490
 BELC0500
 BELC0510
 BELC0520

```

50 CONTINUE
DO 70 I=1,IS
55 IMI = IM(I)
IF (IMI - 1) 60,70,60
60 DO 65 K=1,IS
    V = UM(K,IMI)
    UM(K,IMI) = UM(K,I)
65 UM(K,I) = V
IMI = IM(IMI)
IM(IMI) = IMI
GOTO 55
70 CONTINUE
C-----ELEMENT --- BASE GAS CORRESPONDENCE
C-----INITIALIZE ROW AND COLUMN SUMS
    IC = IS
DO 75 I=1,IS
    IR(I) = -1
75 IC(I) = -1
C-----EVALUATE INITIAL SUMS
    LAMD = 1
DO 90 I=1,IS
DO 85 J=1,IS
    LAMT = LAM(I,J) / LAMD
    LIM(I,J) = LAMT - (LAMT / 2) * 2
    IF (LIM(I,J)) 80,85,80
80 IC(J) = IC(J) + 1
    IR(I) = IR(I) + 1
85 CONTINUE
90 LAMD = LAMD + LAMD
C-----CHECK FOR ZEROS
95 IZ = 0
100 DO 105 I=1,IS
    IF (IC(I) - IZ) 140,105,140
105 DO 110 J=1,IS
    IF (LIM(J,I)) 115,110,115
110 CONTINUE
115 YC(I) = - J
    IR(J) = - I
    LIM(J,I) = 0
DO 135 K=1,IS
    IF (LIM(J,K)) 120,125,120
120 IC(K) = IC(K) - 1
    LIM(J,K) = 0
125 IF (LIM(K,I)) 130,135,130
130 LIM(K,I) = 0
    IR(K) = IR(K) - 1
135 CONTINUE
GOTO 190
140 IF (IR(I) - IZ) 185,145,185
145 DO 155 J=1,IS
    IF (LIM(I,J)) 150,155,150
150 IC(J) = - 1

```

BELC0530
 BELC0540
 BELC0550
 BELC0560
 BELC0570
 BELC0580
 BELC0590
 BELC0600
 BELC0610
 BELC0620
 BELC0630
 BELC0640
 BELC0680
 BELC0690
 BELC0700
 BELC0710
 BELC0720
 BELC0730
 BELC0740
 BELC0750
 BELC0760
 BELC0770
 BELC0780
 BELC0790
 BELC0800
 BELC0810
 BELC0820
 BELC0830
 BELC0840
 BELC0850
 BELC0860
 BELC0870
 BELC0880
 BELC0890
 BELC0900
 BELC0910
 BELC0920
 BELC0930
 BELC0940
 BELC0950
 BELC0960
 BELC0970
 BELC0980
 BELC0990
 BELC1000
 BELC1010
 BELC1020
 BELC1030
 BELC1040
 BELC1050
 BELC1060
 BELC1070

G-16

```

IR(I) = - J
LIM(I,J) = 0
GCTC 16C
155 CONTINUE
160 DO 18C K=1,IS
  IF (LIM(K,J)) 165,170,165
165 IR(K) = IR(K) - 1
  LIM(K,J) = 0
170 IF (LIM(I,K)) 175,180,175
175 LIM(I,K) = 0
  IC(K) = IC(K) - 1
180 CONTINUE
  GOTO 190
185 CONTINUE
  IZ = IZ + 1
  GOTO 100
190 IG = IG - 1
  J = IS + 1
  IF (IG) 195,195,95
195 DO 200 I=1,IS
  K = - IR(I)
  FAMA(I) = FAMA(K)
  FAMBI(I) = FAMBI(K)
  WRITE(ROUT,205) (ATA(I),ATB(I),ATC(I),I=1,IS)
  WRITE(ROUT,210) (FAMA(I),FAMBI(I),I=1,IS)
205 FORMAT(5X9HELEMENT 18A4/15X21A4)
210 FORMAT( 5X9HBASE SP 6(4X2A4)/5X7(4X2A4))
  DO 215 L=1,3
  DO 215 I=1,IS
  TQ(I,L) = 0.
  DO 215 K=1,IS
215 TQ(I,L) = TQ(I,L) + UR(I,K) * TK(K,L)
220 RETURN
  END

```

BELC1080
BELC1090
BELC1100
BELC1110
BELC1120
BELC1130
BELC1140
BELC1150
BELC1160
BELC1170
BELC1180
BELC1190
BELC1200
BELC1210
BELC1220
BELC1230
BELC1240
BELC1250
BELC1260
BELC1270
BELC1280
BELC1290
BELC1300
BELC1310
BELC1320
BELC1330
BELC1340
BELC1350
BELC1360
BELC1370
BELC1380
BELC1390
BELC1400
BELC1410

CARD COUNT - 138

ALPSC010

```

SUBROUTINEALPST
REALNFIA(10),NFIB(10),FFIN(10),UH(10,10),TAU(10,10)
DIMENSION X(1),VLAM(150,1),CC(7960)
COMMONKIN,KCUT,JAN,MCL,NAB,IOC,N,KKX(5)
COMMONFAMOA(150),FAMGB(150),RB(150,2),RC(150,2),
1 RD(150,2),RE(150,2),RF(150,2),TU(150,2),VN(150),Y(150),WTM(150),
2 VNU(150,10),FF(150),IFC(150),IP(150),CP(150),H(150),SB(150),
3 YC(150),VLNK(150),E(150),LAMI(150),GAMK(1,1)
COMMONA(16,16),W(3),WM,B(16),SCH,RHC,VEL,B1,KR(10),TILE(3),NFF,
1 XG(50),XG(50)
COMMONZKE(10),TLO(10),DMCLO(10),WALJP(10),EB(10),ALP(10),GAMH(10)
1 ,GAMF(10),TQ(10,3),TK(10,3),MAT(10),IR(10),IOAT(10),KAT(10),UM
2 ,EBL(10)
COMMONFFA,IS,ISP,P,T,SIP,HIP,EL,ENL,FLIQ,NQ,IFL,ISPQ,KRALP,CPF,IRE
1 ,TER,AA,ITS,IN,IL,IT,MCDE,HMELT,SMELT,TMAX,TMIN,MELT,SUMN,SUML,JC,
2HG,CPG,SVA,SVB,SVC,SVD,SUMC
COMMONNXC,SAL,VMACH,KRZZ,SPI,GAM,THI,MX,DMCHI,PCH,NCV,TBT,IFRZ,NOA
1 T,SRI,IXG,SVI,KKJ,IXC,NXG,SHI,MOATO,SAZ,FFF,WS,RV,CMF
COMMONVA,HOS,DUM1,DUM2,EP,IFCJC,ISP2,WTG,VMU2,WTL
COMMONMT,FKF(10),EAK(10),EXK(10),PMU(10,10),RMU(10,10)
COMMONIB(11),FNU(10),PNUS(10),SLAM(10),BE(10),BY(10),
1 IBE(10),JJ(10)
EQUIVALENCE(CC(1),FAMOA(1)),(B(1),X(1)),(GAMK(1),VLAM(1))
EQUIVALENCE(TU(151),TF(1)),(VNU(1),CIJ(1))
EQUIVALENCE(TAU(11),A(11)),(NFIB(1),
1A(51)),(FFIN(1),A(101))
5 FORMAT(13,/,2A4,E12.5,2A4,E12.5,2A4,E12.5,2A4,E12.5)
10 FORMAT(51H UPDATE OF FAIL TEMPERATURES AND DIFFUSION FACTORS/10X,
17HSPECIES16HFAIL TEMPERATURES16HDIFFUSION FACTOR)
15 FORMAT(11X2A4,6XF10.3,2A2)
20 FORMAT(11X2A4,22XF15.5,2A2)
25 FORMAT(13)
30 FORMAT(8E10.4)
125 WS = 1.0
155 MOATO = NOAT
NOAT = 0
DUM1 = 0.
DO 200 K=1,IS
ALPH = (W(1) * TK(K,1) + W(2) * TK(K,2) + W(3) * TK(K,3)) / WS
IF (ALPH) 160,160,200
160 IF (KAT(K) - 99) 165,200,165
165 IF (NR(6)) 195,170,170
170 IF (TR(K,1)) 195,195,200
195 NOAT = NOAT + 1
DUM1 = DUM1 - ALPH
IOAT(MOAT) = K
200 CONTINUE
WS = WS * (1. + DUM1)
DO 205 K=1,IS
205 ALPK(K) = (W(1) * TQ(K,1) + W(2) * TQ(K,2) + W(3) * TQ(K,3)) / WS
TEST = 1.0

```

ALPS0280
ALPS0290
ALPS0300
ALPS0310
ALPS0320
ALPS0330
ALPS0340
ALPS0840
ALPS0920
ALPS0930
ALPS0940
ALPS0950
ALPS0960
ALPS0970
ALPS0980
ALPS0990
ALPS1000
ALPS1070
ALPS1080
ALPS1090
ALPS1100
ALPS1110
ALPS1120
ALPS1130
ALPS1140

```

IF (NR(6)) 280,215,210
21C STOP222
C****SURFACE MASS BALANCE PACKAGE GOES HERE
215 DC 27C K=1,IS
      GAMF(K) = 0.
255 IF (KAT(K) - 99) 255,260,255
260 ALP(K) = 0.
      GOTO 270
265 ALP(K) = ALP(K) + ZKE(K)
270 CONTINUE
275 TEST=1.+W(2)+W(3)
280 DO 285 K=1,IS
285 TEST = TEST - ALP(K) * WIM(K)
      IF (ABS(TEST) - .001) 300,300,290
290 WRITE(KOUT,295)
      STOP
295 FORMAT(///10X25HBAD EDGE TABLE * * * STOP)
300 IF (NR(3)) 305,305,370
305 IF (NOAT) 370,370,310
C REINITIALIZE SPECIES
310 PIN = P / 10.
      PLOG = ALOG(PIN)
      J = 1
315 DO 365 IK=1,M
      IF (IFC(J) + 1) 320,365,330
320 IFC(J) = IFC(J) + 3
      IF (J - IS) 340,340,325
325 YC = 0.
      GOTO 345
330 IF (IFC(J) - 1) 365,365,335
335 IFC(J) = IFC(J) - 3
340 YC = PLOG
345 IF (IFC(J)) 355,360,350
350 IFC(J) = - 1
355 VN(J) = 0.
      Y(J) = YC
      GOTO 365
360 VN(J) = PIN
      Y(J) = PLOG
365 J=J+1
370 IF (NOAT) 425,425,375
375 J = 1
380 DO 405 IK=1,M
385 DO 395 L=1,NOAT
      K = IOAT(L)
      LAMT = LAMT(J) / 2 + * (K - 1)
      LAMT = LAMT - (LAMT / 2) + 2
      IF (LAMT) 390,395,390
390 VN(J) = 0.
      IFC(J) = IFC(J) - 3
      GOTO 405

```

- ALPS1150
- ALPS1160
- ALPS1170
- ALPS1350
- ALPS1360
- ALPS1370
- ALPS1380
- ALPS1390
- ALPS1400
- ALPS1410
- ALPS1420
- ALPS1430
- ALPS1440
- ALPS1450
- ALPS1460
- ALPS1470
- ALPS1480
- ALPS1490
- ALPS1500
- ALPS1510
- ALPS1520
- ALPS1530
- ALPS1540
- ALPS1550
- ALPS1560
- ALPS1570
- ALPS1580
- ALPS1590
- ALPS1600
- ALPS1610
- ALPS1620
- ALPS1630
- ALPS1640
- ALPS1650
- ALPS1660
- ALPS1670
- ALPS1680
- ALPS1690
- ALPS1700
- ALPS1710

- ALPS1730
- ALPS1740
- ALPS1750
- ALPS1770
- ALPS1780
- ALPS1790
- ALPS1800
- ALPS1810
- ALPS1820
- ALPS1830
- ALPS1840

6-19

395 CONTINUE
405 J=J+1
410 DO 415 L=1,NCAT
K = IOAT(L)
J = - IR(K)
415 IFC(J) = IFC(J) + 6
425 RETURN
END

ALPS1850
ALPS1890
ALPS1900
ALPS1910
ALPS1920
ALPS1940
ALPS1950

CARD COUNT - 112

THER0010
THER0030
THER0040

```

SUBROUTINE THERM
REAL NFIA(10), NFIB(10), FFIN(10), UM(10,10), TAU(10,10)
DIMENSION NLNK(10), BC(10)
DIMENSION YF(150), CIJ(150,1)
DIMENSION X(1), VLAP(150,1), CC(7960)
COMMON KIN, KOUT, JAN, MOL, NAB, IDC, N, KX(5)
COMMON FAP0A(150), FAPCB(150), RB(150,2), RC(150,2),
1 RO(150,2), RE(150,2), RF(150,2), TU(150,2), VN(150), Y(150), WTM(150),
2 VNU(150,10), FF(150), IFC(150), IP(150), CP(150), H(150), SB(150),
3 YC(150), VLNK(150), E(150), LAMI(150), GAMK(1,1)
COMMON A(16,16), W(3), WM, B(16), SCH, RHO, VEL, B1, KR(10), TILE(3), NFF,
1 XC(150), XG(150)
COMMON ZKE(10), TLO(10), DMCL0(10), WALJP(10), EB(10), ALP(10), GAMH(10)
1, GAMF(10), TQ(10,3), TK(10,3), MAT(10), IR(10), IOAT(10), KAT(10), UM
2, EBL(10)
COMMON FFA, IS, ISP, P, T, SIP, HIP, EL, ENL, FLIQ, NQ, IFL, ISPG, KRALP, CPF, IRE
1, IER, AA, ITS, IN, IL, II, MODE, HMELT, SMELI, TMAX, TMIN, MELT, SUMN, SUML, JC,
ZMG, CPG, SVA, SVB, SVC, SVD, SUPC
COMMON XNC, SAL, VMACH, KRZZ, SPI, GAM, THI, MX, DMCHI, MCH, MCV, TBT, IFRZ, NOA
1, SRI, IXG, SVI, KKJ, IXC, MXG, SHI, NOATO, SAZ, FFF, MS, RV, CMF
COMMON VA, MOS, DUMI, DUM2, EP, IFC, JC, ISP2, NTG, VMU2, WTL
COMMON MT, FK(10), EAK(10), EXK(10), PMU(10,10), PMU(10,10),
COMMON B(11), FNU(10), PHUS(10), SLAM(10), BE(10), BY(10),
1 IBC(10), JJ(10)
EQUIVALENCE (CC(1), FAND(1)), (B(1), X(1)), (GAMK(1), VLAM(1))
EQUIVALENCE (U(151), IE(11)), (VMU(1), CIJ(1))
EQUIVALENCE (TAU(1), A(151)),
1A(51)), (FFIN(1), A(101))
5 MODE = ER(1)
10 WTC = 0.
MTL = 0.
DUM2 = 0.
SUMN = 0.
DO 15 I=1,15
PMUS(I) = 0.
15 SLAM(I) = 0.
20 HMELT = 0.
FLIQ = 0.
SMELT = 0.
MELT = 1
TMAX = 20000.
TMIN = 500.
TFMAX = 500.
VA = ALOG(T / 3000.)
VB = T - 3000.
VC = (T + 3000.) / 2.
VD = T + 3000.
VE = VC / (VD + VD)
RT = 1.9869 * T
IF (MCV) 35,35,25
25 PIN = P * 10.E - 3

```

THER0300
THER0310
THER0320
THER0330
THER0340
THER0350
THER0360
THER0370
THER0380
THER0390
THER0400
THER0410
THER0420
THER0430
THER0440
THER0450
THER0460
THER0470
THER0480
THER0490
THER0500
THER0510
THER0520
THER0530

6-21

```
MM = 20-
AA = P + MM
IF (MCDE) 35,35,30
3C T = 3000.
35 CONTINUE
I = 1
40 DO 215 IK=1,M
   IF (INCV) 55,55,45
45 IF (IFC(I)) 55,50,55
50 VMI(I) = VN(I) + PIN
   Y(I) = ALOG(VN(I))
55 CONTINUE
   IF (IFC(I) + 1) 130,60,60
60 J = 2
   IF (IFC(I)) 65,70,95
65 IF (MODE - 1) 130,185,130
70 IF (ITS) 130,75,130
75 SUMM = SUMM + VN(I)
   DUM1 = WM(I) + VN(I)
   MIZ = MIZ + DUM1
   DUR2 = DUR2 + DUM1 / FF(I)
   IF (IK - IS) 80,80,85
80 PHUS(I) = VMI(I)
   SLAR(I) = VN(I) / FF(I)
   GOTO 130
85 DO 90 K=1,IS
   DUM1 = VNU(I,K) + VMI(I)
   PHUS(K) = PHUS(K) + DUM1
   SLAR(K) = SLAR(K) + DUM1 / FF(I)
   GOTO 130
95 IF (IFC(I) - 1) 100,100,130
100 IF (ITS) 125,105,125
105 IF (HR(6) - KR(8)) 120,110,110
110 IF (T - TF(I) + .001) 115,115,120
115 IFC(I) = -1
   VMI(I) = 0.
   GOTO 65
120 MTL = MTL + VMI(I) + WM(I)
125 IF (MODE - 1) 130,180,130
180 TMIN = AMAX(TMIN,TF(I))
185 IFRAX = AMAX(IFRAX,IFMAX)
130 IF (T - ABS(TU(I,1))) 135,140,140
135 J = 1
140 CPIII = RC(I,J) + I + RD(I,J) + RE(I,J) / (T * T)
   H(I) = R(I,J) + VB + (RC(I,J) + RD(I,J) + VC + RE(I,J) / VD)
   SB(I) = RF(I,J) + RC(I,J) + VA + VB + (RD(I,J) + RE(I,J) + VE)
145 IF (MODE - 1) 190,150,150
150 IF (IFC(I) - 1) 190,155,190
155 IF (TU(I,1)) 160,190,190
160 IF (T + TU(I,1)) 170,165,175
165 HMELT = - M(I) - HMELT
   SMELT = - SB(I) - SMELT
```

THER0550
THER0560
THER0570
THER0580
THER0590
THER0600
THER0610

THER0630
THER0640
THER0650
THER0660
THER0670
THER0680
THER0690
THER0700
THER0710
THER0720
THER0730
THER0740
THER0750
THER0760
THER0770
THER0780
THER0790
THER0800
THER0810
THER0820
THER0830
THER0840
THER0850
THER0860
THER0870
THER0880
THER0890
THER0900
THER0910
THER0920
THER0930
THER1100
THER1110
THER0940
THER0950
THER0960
THER0970

THER0980
THER0990
THER1000
THER1010
THER1020
THER1030

```

MELT = 1
IF (J - 2) 190,135,190
17C TMAX = APINI(TMAX, - TU(I,1))
GOTC 190
175 TMIN = APAXI(TMIN, - TU(I,1))
190 TC(I) = - H(I) / RT
VLNK(I) = TC(I) * SB(I) / 1.9869
IF (IK - IS) 195,195,200
195 BLNK(I) = VLNK(I)
18C(I) = IPC(I)
18C(I) = TC(I)
GOTO 215
200 DO 210 K=1,IS
IF (IDC(K) * I) 210,205,205
205 VLNK(I) = VLNK(I) - VNU(I,K) * BLNK(K)
TC(I) = TC(I) - VNU(I,K) * BC(K)
210 CONTINUE
215 I=I+1
NEV = -IABS(MCV)
IF (MORR - 1) 230,220,230
220 IF (TPMAX - T) 225,230,230
225 T = TPMAX
GOTO 5
230 IF (ITS) 355,235,355
235 AA = P * WA
NEV = A * B
240 FORMAT(F10.6)
245 FORMAT (2I3)
250 FORMAT(3E10.4)
255 READ(IN,245)HT
DO 260 M=1,RT
READ(KIN,250)FKF(M),EAK(M),EXK(M)
READ(KIN,240)(RNU(I,M),I=1,IS)
260 READ(KIN,240)(PHU(I,M),I=1,IS)
265 FORMAT (/3X,7HKINETIC)
270 FORMAT (3X,11HREACTION---,17,9110)
275 FORMAT (/3X,8HREACTANT)
280 FORMAT (5X,12HCOEFFICIENTS/)
285 FORMAT (8X,2A4,F8.2,9F10.2)
290 FORMAT (/3X,7HPRODUCI)
295 FORMAT (/3X,12HPRE-EXPONENT)
300 FORMAT (5X,6HFACTOR,4X,10E10.3)
305 FORMAT (/3X,11HACTIVATION)
310 FORMAT (5X,6HENERGY,4X,10E10.3)
315 FORMAT (/3X,8HREACTION)
320 FORMAT (5X,5HORDER,5X,10E10.3)
WRITE(KOUT,265)
WRITE(KOUT,270)(M,M=1,MT)
WRITE(KOUT,275)
WRITE(KOUT,280)
DO 325 I=1,IS

```

THER1040
 THER1050
 THER1060
 THER1070
 THER1080
 THER1130
 THER1140
 THER1150
 THER1160
 THER1170
 THER1180
 THER1190
 THER1200
 THER1210
 THER1220
 THER1230
 THER1240

THER1260
 THER1270
 THER1280
 THER1290
 THER1300
 THER1310
 THER1320
 THER1330
 THER1340
 THER1350

THER1370
 THER1380
 THER1390
 THER1400
 THER1410

THER1460

THER1

THER1500

```

325 WRITE(KCLT,285)FAMOA(I),FAMOB(I),(RMU(I,M),M=1,MT)
    WRITE(KCLT,290)
    WRITE(KCUT,280)
    DO 330 I=1,15
330 WRITE(KOLT,285)FAPOA(I),FAMOB(I),(PMU(I,M),M=1,MT)
    WRITE(KOLT,295)
    WRITE(KOUT,300)(FKF(M),M=1,MT)
    WRITE(KOLT,305)
    WRITE(KOLT,310)(EAK(M),M=1,MT)
    WRITE(KOLT,315)
    WRITE(KOLI,320)(EXK(M),M=1,MT)
335 SUMN = SUMN / P
    SUML = ALOG(SUMN)
    FFF = WTC / DUM2
    WTC = WTC / SUMN
    WTL = WTL / SUMN
    DO 340 I=1,15
340 PMS(I) = PMS(I) / SUMN + GAMF(I) * AA
    SLAM(I) = SLAM(I) / SUMN + FFF
    MS=M(2)+M(3)
    SUPA = 1.0
    IF (KRALP + KR(2)) 355,355,345
345 IF (MTL / WTC - MS) 355,355,350
350 SUMC = WTL / (WTC + MS)
    WTL = WTL / SUMC
355 RETURN
    END

```

THER1

THER1620
THER1

THER1710
THER1720
THER1730
THER1740
THER1750
THER1760
THER1770
THER1780
THER1800
THER1810
THER1820
THER1830
THER1840
THER1850
THER1860
THER1870

183

CARD COUNT -

MAT10010
MAT10030

```

SUBROUTINE MAT1
REAL NFI(10), MFI(10), FFIN(10), UM(10,10), TAU(10,10)
DIMENSION TF(150), CIJ(150,1)
DIMENSION X(1), VLAM(150,1), CC(7960)
COMMON KIN, XOUT, JAN, MCL, NAB, IDC, N, XKX(15)
COMMON FAMOA(150), FAMOB(150), RB(150,2), TU(150,2), VM(150), Y(150), WTH(150),
1 RO(150,2), RE(150,2), RF(150,2), RP(150), CP(150), H(150), SB(150),
2 VNU(150,10), SF(150), IFC(150), LAMI(150), GAMK(1,1)
3 TC(150), VLAK(150), E(150), MM, B(16), SCH, RHO, VEL, BI, KR(10), TILE(3), NFF,
COMMON A(16,16), W(3), MM, B(16), SCH, RHO, VEL, BI, KR(10), TILE(3), NFF,
1 XC(150), XG(150)
COMMON Z(10), TLO(10), DMCLD(10), WALJP(10), EB(10), ALP(10), GAMK(10)
1 ,GAMF(10), TQ(10,3), YK(10,3), MAT(10), IR(10), IOAT(10), KAT(10), UM
2 ,EBL(10)
COMMON FFA, IS, ISP, P, T, SIP, HIP, EL, ENL, FLIQ, NG, IFI, ISPQ, KRALP, CPF, IRE
1, IER, AA, ITS, IN, IL, IT, MODE, MMELT, SMELT, TMAX, THIN, MELT, SUMN, SUML, JC,
ZNG, CPG, SVA, SVB, SVC, SVD, SUMC
COMMON XNC, SAI, VRACH, KRZZ, SPI, GAM, TH, NK, DMCHI, MCH, NCV, TBT, IFRZ, MOA
1 T, SRL, IXG, SVI, RKJ, IXC, NXG, SHI, NOATO, SAZ, FFF, WS, RV, CMF
COMMON VA, MDS, DUAL, DUNZ, EP, IFC, JC, ISPZ, NIG, YMUZ, NTL
COMMON MT, FRF(10), EAK(10), EXK(10), PMU(10,10), RMU(10,10)
COMMON B(11), PMU(10), PMUS(10), SLAM(10), BE(10), BY(10),
1 IFC(10), JJJ(10)
EQUIVALENCE (CC(1), FAMOA(1)), (B(1), X(1)), (GAMK(1), VLAM(1))
EQUIVALENCE (TU(151), TF(1)), (VNU(1), CIJ(1))
EQUIVALENCE (TAM(1), A(151))
IA(51), (FFIN(1), A(101))
RV = WS - MTL / WTC
IF (IR(1) - 1) 30,5,10
5 IF (ITS) 30,20,30
10 WRITE(KOUT,15) FFF, MTL, WTC, AA, RV, ALP, PMUS, SLAM
15 FORMAT(13H FFF, MTL, WTC, AA, RV/ALP/PMUS/SLAM/1XSE12.5/(1X10E12.5))
KR(7) = KR(7) - 1
IF (KR(7) - 1) 30,20,30
20 WRITE(KOUT,25) I, I-1, IS
25 FORMAT (50H ITS TEMP PRES=MT EQUIL ER MASBAL ER SCALE 7(13,
17H MASBAL/90X3(13,3H MASBAL))
C INITIALIZE
30 EL = 0.
CPG = 0.
EP = P
CPF = 0.
JC = 0
ISPQ = IS + 2
9(1) = 0.
8(2) = C.
A(1,1) = 0.
A(1,2) = 0.
A(2,1) = 0.
A(2,2) = 0.
C-----INITIALIZE CONTRIBUTION OF MOST SIGNIFICANT SPECIES IN EACH MASS
DO 35 I=1,IS

```

MAT10300
MAT10310
MAT10320
MAT10330
MAT10340
MAT10350
MAT10360
MAT10370
MAT10380
MAT10390
MAT10400
MAT10410
MAT10420
MAT10440
MAT10450
MAT10460
MAT10470
MAT10480
MAT10490
MAT10500
MAT10510
MAT10520
MAT10530
MAT10540
MAT10550

MAT10560
 MAT10570
 MAT10580
 MAT10590
 MAT10600
 MAT10610
 MAT10620
 MAT10630
 MAT10640
 MAT10650
 MAT10660
 MAT10670
 MAT10680
 MAT10690
 MAT10700
 MAT10710
 MAT10720
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 MAT10810
 MAT10820
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 MAT10870
 MAT10880
 MAT10890
 MAT10900
 MAT10910
 MAT10920

 MAT10940
 MAT10950
 MAT10960
 MAT10970
 MAT10980
 MAT10990
 MAT11000
 MAT11010
 MAT11020
 MAT11030
 MAT11040
 MAT11050
 MAT11060
 MAT11070
 MAT11080

```

EBL1) = 0.
35 E11) = AA * ALP(1)
ISP2 = ISPO
C -- -- MAIN BASE SPECIES LOOP
40 DO 25C I=3,ISPQ
C ZERO MATRIX
45 DO 5C K=1,ISPQ
50 A(K,1) = 0.
IF (I75) 70,55,70
C NORMALIZE ON PRESSURE ON FIRST PASS
55 VN(I - 2) = VN(I - 2) / SUMN
EBL(I - 2) = 0.
IF (IFC(I - 2)) 70,60,65
60 Y(I - 2) = Y(I - 2) - SUML
GOTO 70
65 VN(I - 2) = VN(I - 2) / SUMC
C INITIALIZE SOME MORE
70 B(I) = 0.
A(I,1) = 0.
IPI(I - 2) = 0.
C SET FLAG INDICATING SIGNIFICANCE OF SPECIES IN MASS
C BALANCE(S) AND INCREMENT COUNT ON SIGNIFICANT SPECIES
75 IF (VN(I - 2) - EBL(I - 2)) 85,85,80
80 IPI(I - 2) = 1
C TREAT BASE SPECIES CONTAINING BUT NOT REPRESENTING NON-PRESENT
C ELEMENTS IN SAME MANNER AS NON-PRESENT CONDENSED SPECIES
85 IF (IFC(I - 2) + 1) 290,90,90
90 IF (IFC(I - 2)) 140,170,95
95 A(I,1) = 1.
VA = VN(I - 2)
IF (IFC(I - 2) - 1) 100,100,240
100 IF (EBL(I - 2) - ABS(VA)) 105,110,110
105 ER(I - 2) = ABS(VA)
110 E(I - 2) = 1 - 2
IF (NR(6)) 115,120,120
115 IF (MODE - 1) 240,130,240
120 DO 125 K=1,15
125 A(K + 2,1) = - WTH(I - 2) / WTG * PNUS(K)
A(I,1) = A(I,1) + 1.
130 IMIN = AMAXI(TMIN,TF(I - 2))
IF (Y - TF(I - 2) + .001) 135,135,140
135 A(I,1) = 1.0E + 10
E(I - 2) = - VN(I - 2) * 1.001E + 10
MODE = 0
140 IF (MODE - 1) 290,145,290
145 IF (Y(I - 2) + .001 - Y) 290,150,150
150 IF (Y(I - 2) + .001 - Y) 165,165,155
155 IF (Y(I - 2) - B(I)) 290,290,160
160 A(I,JC + 2) = 0.
165 B(I) = Y(I - 2)
YMAX = Y(I - 2)

```

```

JC = I - 2
IFCJC = IFC(I - 2)
A(1,1) = - 1.
GOTO 290
C-----GAS PHASE
170 VA = VN(I - 2)
CPG = CPG + VA * CP(I - 2)
A(1,1) = VA
A(2,1) = VA
EP = EP - VA
IF (KAT(I - 2) - 99) 175,235,175
175 IF (KR(6)) 235,210,180
180 DO 185 K=1,15
185 A(K + 2,1) = GAMK(I - 2,K) + GAMH(K) * H(I - 2)
A(1,1) = A(1,1) + VA * RV
190 DUM2 = MTM(I - 2) / WTG * WTL / WTG * VA
DC 205 K=1,15
IF (EB(K) - ABS(A(K + 2,1))) 195,200,200
195 EB(K) = ABS(A(K + 2,1))
IB(K) = I - 2
200 E(K) = E(K) - A(K + 2,1)
205 A(K + 2,1) = A(K + 2,1) + DUM2 * PNUS(K)
GOTO 240
210 DUM1 = MTM(I - 2) / WTG * VA
DUM2 = WTL / WTG * DUM1
IF (KR(4)) 215,215,220
215 DUM1 = 0.
VA = (RV + 1.) * VA
GOTO 225
220 DUM1 = DUM1 * (1. - FFF / FF(I - 2))
VA = (RV + FFF / FF(I - 2)) * VA
225 DO 230 K=1,15
230 A(K + 2,1) = DUM1 * SLAM(K) + DUM2 * PNUS(K)
A(1,1) = A(1,1) + VA
235 EB(I - 2) = ABS(VA)
IB(I - 2) = I - 2
E(I - 2) = E(I - 2) - VA
240 IF (MODE - 2) 285,260,245
245 IF (IFC(I - 2)) 290,255,250
250 HOS = SB(I - 2)
GOTO 270
255 HOS = SB(I - 2) - 1.9869 * Y(I - 2) - 1.9869
GOTO 275
260 HOS = H(I - 2)
265 IF (IFC(I - 2)) 290,275,270
270 A(1,1) = HOS
GOTO 280
275 A(1,1) = HOS * VN(I - 2)
280 A(1,2) = A(1,2) - HOS * VN(I - 2)
285 CPF = CPG + CPEI - 2) * VN(I - 2)
290 CONTINUE
DO 295 I=1,15

```

```

MAT11090
MAT11100
MAT11110
MAT11120
MAT11130
MAT11140
MAT11150
MAT11160
MAT11170
MAT11180
MAT11190
MAT11200
MAT11210
MAT11220
MAT11230
MAT11240
MAT11250
MAT11260
MAT11270
MAT11280
MAT11290
MAT11300
MAT11310
MAT11320
MAT11330
MAT11340
MAT11350
MAT11360
MAT11370
MAT11380
MAT11390
MAT11400
MAT11410
MAT11420
MAT11430
MAT11440
MAT11450
MAT11460
MAT11470
MAT11480
MAT11490
MAT11500
MAT11510
MAT11520
MAT11530
MAT11540
MAT11550
MAT11560
MAT11570
MAT11580
MAT11590
MAT11600

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6-27

BE(I) = E(I)
BY(I) = Y(I)
295 IBC(I) = IFC(I)
RETURN
END

MAT11610
MAT11620
MAT11630
MAT11640
MAT11650

CARD COUNT - 161

MAT20010
MAT20030

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SUBROUTINE MATZ
  REAL NFIA(10), NFIB(10), FFIN(10), UM(10,10), TAU(10,10)
  DIMENSION TF(150), CIJ(150,1)
  DIMENSION X(1), VLAM(150,1), CC(7960)
  COMMON /KIN/, JAN, PCL, MAB, IDC, N, KKX(5)
  COMMON /FMOA/, FAMC(150), RB(150,2), RC(150,2),
  1 RD(150,2), RE(150,2), RF(150,2), TU(150,2), VNU(150), Y(150), WTM(150),
  2 VNL(150,10), FF(150), IFC(150), IP(150), CP(150), M(150), SB(150),
  3 TC(150), VLNK(150), E(150), LAM(150), GAMK(1,1)
  COMMON /A1616/, M(3), MP, B(16), SCH, RHO, VEL, B1, KR(10), TILE(3), NFF,
  1 XC(150), XG(150)
  COMMON /ZKE/, TLO(10), DMCLN(10), WALJP(10), EB(10), ALP(10), GAMH(10)
  1 GAMF(10), TQ(10,3), TK(10,3), MAT(10), IR(10), IOAT(10), KAT(10), UM
  2 EBL(10)
  COMMON /FFA/, IS, ISP, F, T, SIP, HIP, EL, ENL, FLIQ, NG, IFL, ISPO, KRALP, CPF, IRE
  1 IER, AA, ITS, IN, IL, IT, PODE, HMELT, SMELT, TMAX, TMIN, MELT, SUMN, SUML, JC,
  2HG, CPG, SVA, SVB, SVC, SVD, SUMC
  COMMON /NXC/, SAI, VRACH, KRZZ, SPL, GAM, THI, MX, DMCHI, HCH, NCV, TBT, IFRZ, NOA
  1 T, SRI, IXG, SVI, KKJ, IXC, NXG, SHI, NOATO, SAZ, FFF, WS, RV, CMF
  COMMON /MVA/, M0S, DUN1, DUN2, EP, IFCJC, ISP2, MTG, VMU2, MTL
  COMMON /FKF/, FKF(10), EAK(10), EXK(10), PMU(10,10), RMU(10,10)
  COMMON /IB/, FNU(10), PNU(10), SLAM(10), BE(10), BY(10),
  1 IBC(10), JJ(10)
  EQUIVALENCE (CC(1), FAMOA(1)), (B(1), X(1)), (GAMK(1), VLAM(1))
  EQUIVALENCE (TU(151), TF(1)), (VNU(1), CIJ(1))
  EQUIVALENCE (TAU(1), A(151)),
  1A(51)), (FFIH(1), A(101)),
  ISPQ = IS + 2
  ISP2 = ISPQ
  IER = 0
  IRE = 0
  EER = 0
  C -- -- MAIN NON-BASE SPECIES LOOP
  5 J = ISP
  10 DO 345 IK=ISP,N
    IF (IFC(J) + 1) 345,15,15
  15 IF (ITS) 35,20,35
  20 VN(J) = VN(J) / SUMN
    IF (IFC(J)) 35,30,25
  25 VNI(J) = VNI(J) / SUMC
    GOTO 35
  30 YE(J) = Y(J) - SUML
  35 E(J) = VLNK(J) - Y(J)
    DO 50 I=1,IS
      IF (IBC(I)) 45,45,40
  40 FNU(I) = 0.
    GOTO 50
  45 FNU(I) = VNU(J,I)
    E(J) = E(J) + FNU(I) + BY(I) * BY(I)
  50 CONTINUE
    EAB = ABS(E(J))

```

MAT20300
MAT20310
MAT20340
MAT20350
MAT20360
MAT20370
MAT20380
MAT20390
MAT20400
MAT20410
MAT20420
MAT20430
MAT20440
MAT20450
MAT20460
MAT20470
MAT20480
MAT20490
MAT20500
MAT20510
MAT20520
MAT20530
MAT20540
MAT20550
MAT20560

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IF (IFG(J)) 55,215,85
C++++CONDENSED SPECIES
55 EAB = E(J)
60 IF (KR(6) - KR(8)) 65,60,60
65 IF (T - TF(J) + .001) 180,65,65
70 EER = E(J)
75 ISPQ = ISPQ + 1
IER = ISPQ
80 IE = IER
IRE = IK
GOTO 90
85 ISPQ = ISPQ + 1
IE = ISPQ
90 WTR = 0.
IF (KR(6)) 100,95,95
95 TMIN = AMAX1(TF(J), TMIN)
WTR = WTR(J) / WTR
100 DO 105 I=1,ISPQ
A(I,IE) = 0.
105 A(IE,I) = 0.
DO 105 K=1,IS
DUMI = VMU(J,K)
IF (T - TF(J) + .001) 110,120,120
110 IF (KR(6)) 120,120,115
115 DUMI = TQ(K,3) + WTR(J)
JC=J
120 IF (IBCKI) - 11,125,125,145
125 VA = DUMI + VN(J)
130 AK + 2,IE) = DUMI - WTR + PMUS(K)
135 BEIK = RE(K) - YA
IF (ABS(VA) - EB(K)) 145,145,140
140 EB(K) = ABS(VA)
IBLK) = IK
145 CONTINUE
K = IE - ISP2
JJ(K) = J
A(IE,1) = TC(J)
B(IE) = E(J)
IF (T + .001 - IF(J)) 150,165,165
150 IF (MODE - 1) 155,160,195
155 IF (KR(6) - KR(8)) 165,160,160
160 ALIE,IE) = 1,IE + 10
B(IE) = - VN(J) + 1.001E + 10
MODE = 0
165 IE (MODE - 2) 170,200,205
170 IF (MODE - 1) 335,175,335
175 IF (T - TF(J) - .001) 180,180,335
180 IF (JC) 190,190,185
185 IF (E(J) - B(1)) 345,345,190
190 B(1) = E(J)
MAT20570
MAT20580
MAT20590
MAT20600
MAT20610
MAT20620
MAT20630
MAT20640
MAT20650
MAT20660
MAT20670
MAT20680
MAT20690
MAT20700
MAT20710
MAT20720
MAT20730
MAT20740
MAT20750
MAT20760
MAT20770
MAT20780
MAT20790
MAT20800
MAT20810
MAT20820
MAT20830
MAT20840
MAT20850
MAT20860
MAT20870
MAT20880
MAT20890
MAT20900
MAT20910
MAT20920
MAT20930
MAT20940
MAT20950
MAT20960
MAT20970
MAT20980
MAT20990
MAT21000
MAT21010
MAT21020
MAT21030
MAT21040
MAT21050
MAT21060
MAT21070

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```

JC = IK
IFCJC = IFC(JC)
TMAX = TF(IJ)
00 195 K=1,IS
195 A(1,K + 2) = - FNU(K)
A(,1) = TC(J)
GOTO 345
200 HOS = H(IJ)
GOTO 210
205 HOS = SB(IJ)
210 A(1,IEJ) = HOS
A(1,2) = A(1,2) - VN(J) * HOS
CPF = CPF + CP(J) * VN(J)
GOTO 335
C*****GAS PHASE SPECIES
215 IP(J) = 0
220 IQ = 0
CPC = CPC + VN(J) * CP(J)
IF (XBL(I) 745,225,230)
225 FFJ = FFF / FF(J)
230 DUM1 = VTM(J) / VTG * VN(J)
DUM2 = DUM1 / MIG * MTL
IF (KR(4)) 235,235,240
235 DUM1 = 0.
FFJ = 1.0
GOTO 245
240 DUM1 = DUM1 * (1. - FFJ)
245 DO 300 I=3,ISP2
VA = VNU(J,I - 2) * VN(J)
IF (KR(6)) 265,255,250
250 VA = VA * RV + VN(J) * (VLAN(J,I - 2) + GAMH(I - 2) * H(J))
BE(I - 2) = BE(I - 2) - VA
ABSVA = ABS(VA)
VA = VA + PNUS(I - 2) * DUM2
GOTO 270
255 IF (KAT(I - 2) - 99) 260,265,260
260 VA = VA + VA * FFJ
VA = RV + VA + V8
BE(I - 2) = BE(I - 2) - VA
ABSVA = ABS(VA)
VA = VA + SLAM(I - 2) * DUM1 + DUM2 * PMUS(I - 2)
GOTO 270
265 BE(I - 2) = BE(I - 2) - VA
ABSVA = ABS(VA)
270 IF (ABSVA - EBL(I - 2)) 280,280,275
275 IQ = I
IF (ABSVA - EB(I - 2)) 290,290,285
280 IF (ABS(VA) - EBL(I - 2)) 300,300,290
285 EB(I - 2) = ABSVA
IB(I - 2) = IK
290 DO 295 K=3,ISP2

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MAT21080
MAT21090
MAT21100
MAT21110
MAT21120
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MAT21140
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MAT21190
MAT21200
MAT21210
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MAT21280
MAT21290
MAT21300
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MAT21400
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MAT21500
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MAT21570
MAT21580
MAT21590

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6-31

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295 A(I,K) = A(I,K) + VA * FNU(K - 2)
   B(I) = B(I) - VA * E(J)
   A(1,1) = A(1,1) - VA * TC(J)
   A(2,1) = A(2,1) + VN(J) * FNU(1 - 2)
300 CONTINUE
   IF (IQ) 335,335,305
305 EP = EP - VN(J)
   IP(J) = - 1
   B(2) = B(2) - VN(J) * E(J)
   A(2,1) = A(2,1) - VN(J) * TC(J)
   IF (MODE - 2) 330,310,315
310 HOS = H(J) / VN(J)
   GOTO 320
315 HOS = VN(J) * (SB(J) - 1.9869 * Y(J) - 1.9869)
320 DO 325 I=3,15P2
325 A(1,I) = HOS * FNU(I - 2) + A(1,I)
   A(1,2) = A(1,2) - HOS
   A(1,1) = A(1,1) - HOS * TC(J)
   B(1) = B(1) - HOS * E(J)
330 CPF = CPF + VN(J) * CPE(J)
335 IF (EL - EAB) 340,345,345
340 EL = EAB
345 J=J+1
350 CONTINUE
   RETURN
   END

```

MAT21600
MAT21610
MAT21620
MAT21630
MAT21640
MAT21650
MAT21660
MAT21680
MAT21690
MAT21700
MAT21710
MAT21720
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MAT21740
MAT21750
MAT21760
MAT21770
MAT21780
MAT21790
MAT21800
MAT21810
MAT21820

MAT21840
MAT21850
MAT21860

CARD COUNT - 182

SUBROUTINE MAT3
 REAL NFIA(10), NFIB(10), FFIN(10), UM(10,10), TAU(10,10)
 DIMENSION NL(10), AR(10)
 DIMENSION TF(150), CIJ(150,1)
 DIMENSION X(1), VLAM(150,1), CC(7960)
 COMMON KIN, KOUT, JAN, MOL, MAB, IDC, N, MKX(5)
 COMMON FAMOA(150), FAMOB(150), RB(150,2), RC(150,2)
 1 RC(150,2), RE(150,2), RF(150,2), TU(150,2), VN(150), Y(150), WTH(150),
 2 VNL(150,10), FF(150), IFC(150), IP(150), CP(150), H(150), SB(150),
 3 TC(150), VLNK(150), E(150), LAMI(150), GAMK(1,1)
 COMMON A(16,16), W(3), WM, B(16), SCH, RHO, VEL, BL, KR(10), TILE(3), NFF,
 1 XC(50), XG(50)
 COMMON ZKE(10), TLO(10), DMCLO(10), WALJP(10), EB(10), ALP(10), GAMH(10)
 1 GAMF(10), TQ(10,3), TK(10,3), WAT(10), IR(10), IQAT(10), KAT(10), UM
 2 EBL(10)
 COMMON FFA, IS, ISP, P, I, SIP, HIP, EL, ENL, FLIQ, NQ, IFL, ISPO, KRALP, CPF, IRE
 1 IER, AA, ITS, IN, IL, IT, PODE, HMELT, SMELT, TMAX, TMIN, MELT, SUMN, SUML, JC,
 2 HG, CPG, SVA, SVB, SVC, SVD, SUNC
 COMMON XNC, SAI, VMACH, KRZZ, SPL, GAM, THI, MX, DMCHI, HCH, NCV, TBT, IFRZ, NOA
 1 LSRI, IXG, SYL, KKJ, IXC, NXG, SHI, NCATO, SA2, FFF, HS, RV, CMF
 COMMON VA, HOS, DUM1, DUM2, EP, IFCJC, ISP2, WTG, VMU2, WTL
 COMMON MT, FRF(10), EAK(10), EXK(10), PMU(10,10), RMU(10,10)
 COMMON IBL, FNU(10), PNU(10), SLAM(10), BE(10), BY(10),
 1 IBC(10), JJ(10)
 EQUIVALENCE (CC(1), FAMOA(1)), (B(1), X(1)), (GAMK(1), VLAM(1))
 EQUIVALENCE (TQ(1,1), TF(1)), (VMU(1), CIJ(1))
 1A(51), (FFIN(1), A(151)),
 1A(51), (FFIN(1), A(101))
 LSPE = IS + 3
 IF (MODE - 2) 50, 5, 25
 5 CPA = CPF + T
 SHMLT = HMELT + VN(MELT)
 EHS = AA + HIP + A(1,2)
 IF (SVA) 15, 15, 10
 10 DUM1 = SVA / AA + T + I
 EHS = AA + SVC - DUM1 + A(1,2)
 HIP = - A(1,2) / AA
 A(1,2) = - AA + SVC - DUM1
 CPA = (CPF + 2. * DUM1 / T) + T
 DUM2 = SVB / AA + T
 P = P - EP
 EP = - P + SVD - DUM2
 A(2,1) = A(2,1) + DUM2
 A(2,2) = - DUM2
 15 IF (KR(2) - MODE) 30, 20, 30
 20 A(1,2) = - AA + HIP
 GOTO 30
 25 CPA = CPF
 SHMLT = SMELT + VN(MELT)
 EHS = AA + SIP + A(1,2) - 1.9869 * (P - EP)
 A(1,2) = - AA + SIP
 30 B(1) = B(1) + EHS

MAT30010
 MAT30030
 MAT30060

MAT30310
 MAT30320
 MAT30330
 MAT30340
 MAT30350
 MAT30360
 MAT30370
 MAT30380
 MAT30390
 MAT30400
 MAT30410
 MAT30420
 MAT30430
 MAT30440
 MAT30450
 MAT30460
 MAT30470
 MAT30480
 MAT30490
 MAT30500
 MAT30510
 MAT30520
 MAT30530
 MAT30540

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A(I,1) = A(I,1) + CPA
IF (SHMLT) 50,50,35
35 IF (EHS) 50,50,40
40 EHS = EHS - SHMLT
B(I) = B(I) - SHMLT
IF (EHS) 45,50,50
45 FLIQ = 1. + EHS / SHMLT
MODE = 0
TMIN = 500.
TMAX = 5000.
50 ENL = ABS(EP) / P
DO 75 I=3,ISP2
L(I) - 2) = 1 - 2
E(I) - 2) = BE(I) - 2)
EBL(I-2) = EB(I-2) * L.E-7
A(I,2) = -AA * ALP(I - 2)
IF (IFC(I - 2) - 1) 60,60,55
55 NFM = NFM + 1
GOTO 75
60 IF (KR(6)) 70,70,65
65 A(I,1) = A(I,1) + GAMH(I - 2) * T + CPG
DUM1 = GAMF(I - 2) * RV * AA
A(I,2) = A(I,2) - DUM1
E(I) - 2) = E(I) - 2) - DUM1
70 ER = E(I - 2)
ABER = ABS(ER) / (EB(I - 2) + 1.E - 20)
B(I) = B(I) + ER
75 CONTINUE
80 IV = 0
IF (ISP2 == ISP) 80,255,255
JZ = 0
C ADD CONDENSED NONBASE SPECIES TO ARRAY
85 DO 90 IE=ISP3,ISPQ
J = IE - ISP2
J = JJ(J)
DO 90 K=1,FS
90 A(IE,K + 2) = - VNU(J,K)
IF (IV) 95,95,215
C CHECK FOR SINGULAR CONDENSED SET
95 DO 140 IE=ISP3,ISPQ
IV = IV + 1
IF (IV - 1) 120,120,100
C REDUCE NEW ROW BY PREVIOUS ROWS
100 DO 115 J=2,IV
K = L(J - 1)
DIV = - A(IE,K + 2)
IF (DIV) 105,115,105
105 A(IE,K + 2) = 0.
JM = J + ISP
DO 110 M=3,ISP2
110 A(IE,M) = A(IE,M) + DIV * A(JM,M)

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MAT30550
MAT30560
MAT30570
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MAT30590
MAT30600
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MAT30700
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MAT30800
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MAT30900
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MAT30970
MAT30980
MAT30990
MAT31000
MAT31010
MAT31020
MAT31030
MAT31040
MAT31050
MAT31060

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115 CONTINUE
C   SEEK VALID NCN-ZERO ELEMENT
12C DD 13C J=IV,IS
M = L(J)
IF (ABS(A(IE,M + 2)) - .01) 130,130,125
125 IF (IFC(M)) 135,135,130
13C CONTINUE
C   SINGULAR CONDENSED SET
GOTO 145
135 L(J) = L(IV)
L(IV) = P
DIV = A(IE,M + 2)
A(IE,M + 2) = 1.
NORMALIZE NEW ROW
DO 14C J=3,ISP2
140 A(IE,J) = A(IE,J) / DIV
C   COMPLETES MAJOR LOOP=NCN SINGULAR
GOTO 85
C   EXTRACT COEFFICIENTS ON SINGULAR SET-NONBASE
145 DD 150 J=2,IV
K = L(J - 1)
150 AR(J - 1) = A(IE,K + 2)
AR(IV) = 1.
C   EXTRACT COEFFICIENTS ON SINGULAR SET - BASE
DO 165 K=1,IS
IF (IFC(K)) 165,165,155
155 IF (ABS(A(IE,K + 2)) - .01) 165,165,160
160 IV = IV + 1
AR(IV) = - A(IE,K + 2)
JJ(IV) = K
165 CONTINUE
C   ISOLATE NEW CONDENSED CANDIDATE
DO 175 J=1,IV
K = JJ(J)
IF (VNI(K)) 170,170,175
170 JZ = J
KZ = K
GOTO 185
175 CONTINUE
180 STOP
C   WHO MUST GO TO ALLOW ENTRY OF NEW CANDIDATE
185 TEST = - 1.0E10
JX = 0
DO 200 J=1,IV
IF (ABS(AR(J)) - .01) 200,200,186
186 DUM1=AR(JZ)/AR(J)
IF (DUM1) 190,200,200
190 K = JJ(J)
DUM1 = DUM1 * VN(K)
IF (DUM1 - TEST) 200,200,195
195 TEST = DUM1
JX = J

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MAT31070
MAT31080
MAT31090
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MAT31120
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MAT31190
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MAT31460
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MAT31480
MAT31490
MAT31500

MAT31530
MAT31540
MAT31550
MAT31560
MAT31570

6-35

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KX = K
200 CONTINUE
TEST = VN(KX) / AR(JX)
C DOES NEW CANDIDATE ENTER
IF (E(KZ) - E(KX)) 85,85,205
C YES
205 DO 210 J=1,IV
K = JJ(J)
210 VN(K) = VN(K) - TEST * AR(J)
KR(7) = KR(7) + 1
VN(KX) = 0.
KZ = KX
JZ = JX
GOTO 85
C ELIMINATE TERMS CORRESPONDING TO PRESENT BASE CONDENSED
215 DO 230 K=1,IS
IF (IFC(K)) 230,230,220
220 DO 225 IE=ISP3,ISPQ
225 A(IE,K + 2) = 0.
230 CONTINUE
C IF (JZ) 250,250,235
FIX MATRIX TO YIELD SMALL NEG CORRECTION ON ELIMINATED SPECIES
235 IF (IX - IS) 245,245,240
240 KZ = JZ + IS
245 A(KZ + 2, KZ + 2) = 1.E + 15
B(KZ + 2, 2) = - 1.E + 10
250 CONTINUE
255 B(2) = EP + B(2)
IF (MODE = 1) 290,270,260
260 IF (ABS(B(1) / A(1,1)) - .0001) 290,290,300
265 IF (ABS(B(1)) - 1.E - 4) 290,290,300
270 IF (IFC(JC)) 275,265,285
275 IF (JC - IRE) 280,285,280
280 GOTO 265
285 MODE = 0
TMIN = 500.
TMAX = 5000.
GOTO 265
290 IF (EL - 1.E - 4) 295,295,300
295 IF (EML - 1.E - 5) 315,315,300
300 IN = ISPQ
IL = 1
IF (MODE) 305,305,310
305 IN = ISPQ - 1
IL = 2
X(1) = 0.
310 RETURN
315 ITS = - 2
GOTO 310
END

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MAT31580
MAT31590
MAT31600
MAT31610
MAT31620
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MAT31670
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MAT31870
MAT31880
MAT31890
MAT31900
MAT31910
MAT31920
MAT31930
MAT31940
MAT31950
MAT31960
MAT31970
MAT31980
MAT31990
MAT32000
MAT32010
MAT32020
MAT32030
MAT32040
MAT32050
MAT32060
MAT32070

SCAL0010
SCAL0030

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SUBROUTINE SCALE(MCE)
REAL NFA(10), NFIB(10), FFIN(10), UM(10,10), TAU(10,10)
DIMENSION CMFF(150)
DIMENSION DY(150), DYA(150,1)
DIMENSION TF(150), CIJ(150,1)
DIMENSION X(1), VLAM(150,1), CC(17960)
COMMON KIN, KOUT, JAN, MOL, NAB, IDC, N, KXX(5)
COMMON FAMOA(150), FAMOB(150), RB(150,2), RC(150,2),
1 RD(150,2), RE(150,2), RF(150,2), TU(150,2), VN(150), Y(150), WTM(150),
2 VNU(150,10), FF(150), IFC(150), IP(150), CP(150), H(150), SB(150),
3 TC(150), VLNK(150), E(150), LAMI(150), GAMK(1,1)
COMMON A(16,16), W(3), WM(8(16)), SCH, RHO, VEL, B1, KR(10), TILE(3), NFF,
1 XC(150), XG(150)
COMMON ZKE(10), TLO(10), DMCL(10), WALJP(10), EB(10), ALP(10), GAMH(10)
1 , GAMF(10), TQ(10,3), TK(10,3), MAT(10), IR(10), IOAT(10), KAT(10), UM
2 , EBL(10)
COMMON FFA, IS, ISP, P, T, SIP, HIP, EL, ENL, FLIQ, NG, IFL, ISPO, KRALP, CPF, IRE
1 , IER, AA, ITS, IM, IL, IT, HODE, HMELT, SMELT, TMAX, TMIN, MELT, SUMN, SUML, JC,
2HG, CPG, SVA, SVB, SVC, SWD, SUMC
COMMON XNC, SAL, VMACH, KRZZ, SPI, GAM, THI, MX, DMCHI, MCH, NCV, TBI, IFRZ, NOA
1 , SRI, IXG, SVI, KRJ, IXC, MXG, SHI, NOATO, SAZ, FFF, WS, RV, CMF
COMMON VA, MOS, DUM1, DUM2, EP, IFCJC, ISP2, WTG, VMU2, WTL
COMMON RT, FK(10), EAK(10), EK(10), PMU(10,10), RMU(10,10)
COMMON IB(11), FNU(10), PMS(10), SLAM(10), BE(10), BY(10),
1 IBC(10), JJ(10)
EQUIVALENCE (CC(1), FAMOA(11), IB(1), X(1)), (GAMK(1), VLAM(1))
EQUIVALENCE (TU(151), TF(1)), (VNU(1), CIJ(1))
EQUIVALENCE (TAU(1), A(151)),
1A(151)), (FFIN(1), A(10))
EQUIVALENCE (A(1), DY(1), DYA(1))
CLIM = AMAX1(1, W(2) + W(3)) * 0.2 * WTG
DNTL = 0.
DMTG = 0.
DUMP = P * 1.E - 7
BUMP = P * 1.E - 4
BULP = ALOG(BUMP)
15 CMF = 1.
20 K = 0
DO 35 J=2,15
IF (IB(J) - IB(J - 1)) 25,30,35
25 JA = IB(J)
IB(J) = IB(J - 1)
IB(J - 1) = JA
K = 1
GOYO 35
30 IB(J) = 1000
35 CONTINUE
IF (K) 40,40,20
40 IB(15 + 1) = 1000
M = IB(1)
M1 = 1
L = 15 + 2

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SCAL0300
SCAL0350
SCAL0360
SCAL0370
SCAL0380
SCAL0390
SCAL0400
SCAL0410
SCAL0420
SCAL0430
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SCAL0450
SCAL0460
SCAL0470
SCAL0480
SCAL0490
SCAL0500
SCAL0510
SCAL0520
SCAL0530
SCAL0540
SCAL0550
SCAL0560

```

I = C
LL = 1
45 DO 320 IK=1,M
I = I + 1
65 IF (IK - IS) 70,70,85
70 SLAM(I) = 0.
IBC(I) = IFC(I)
PNUS(I) = 0.
IF (IFC(I) - 1) 75,75,115
75 OYI = X(I + 2)
IF (IFC(I) + 1) 315,80,80
80 IF (IFC(I)) 245,135,270
85 IF (IFC(I) + 1) 115,90,90
90 IF (IFC(I)) 110,95,125
95 VA = E(I) - TC(I) + X(I)
DO 105 J=1,IS
IF (IBC(J)) 100,100,105
100 VA = VA + VMU(I,J) + X(J + 2)
105 CONTINUE
OYI = VA
GOTO 135
110 IF (IK - IRE) 115,120,115
115 OYL = 0.
GOTO 315
120 IFC(I) = 1
OYL = X(IER)
GOTO 270
125 L = L + 1
IF (L - IER) 130,125,130
130 OYI = X(L)
GOTO 270
135 DWTG = DWTG + VN(I) + OYI + WTM(I)
IF (IP(I)) 140,235,140
140 IF (IK - M) 145,185,145
145 IF (VN(I) - DUMP) 150,180,180
150 IF (MOE) 235,235,155
155 IF (OYI) 160,315,165
160 IF (VN(I) / BUMP - .9999995 - CMF + OYI) 315,315,170
165 IF (BUMP / VN(I) - 1. - CMF + OYI) 175,315,315
170 CMF = (VN(I) / BUMP - .9999995) / OYI
GOTO 315
175 CMF = (BUMP / VN(I) - 1.) / OYI
GOTO 315
180 IF (MOE) 215,215,190
185 M1 = M1 + 1
M = IR(M1)
190 IF (OYI) 195,315,200
195 IF (OYI + CMF + .999) 205,315,315
200 IF (OYI + CMF - 9.) 315,315,210
205 CMF = -.999 / OYI
GOTO 315
210 CMF = 9. / OYI

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SCAL0570
SCAL0600
SCAL0610
SCAL0620
SCAL069C
SCAL0700
SCAL0710
SCAL0720
SCAL0730
SCAL0740
SCAL0750
SCAL0760
SCAL0770
SCAL0780
SCAL0790
SCAL0800
SCAL0810
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SCAL0840
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SCAL0870
SCAL0880
SCAL0890
SCAL0900
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SCAL0920
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SCAL0940
SCAL0950
SCAL0960
SCAL0970
SCAL0980
SCAL0990
SCAL1000
SCAL1010
SCAL1020
SCAL1030
SCAL1040
SCAL1050
SCAL1060
SCAL1070
SCAL1080
SCAL1090
SCAL1100
SCAL1110
SCAL1120
SCAL1130
SCAL1140
SCAL1150

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GOTO 315
215 IF (DYI * CMF - 2.303) 220,230,230
220 IF (DYI * CMF + 6.909) 225,315,315
225 CMF = - 6.909 / DYI
GOTO 315
230 CMF = 2.303 / DYI
GOTO 315
235 IF (Y(I) - BULP + ABS(DYI) * CMF) 315,240,240
240 CMF = - (Y(I) - BULP) / ABS(DYI)
GOTO 315
NON-PRESENTI BASE
245 IF (KR(6)) 255,250,250
250 IF (I - TF(I) + .001) 315,255,255
255 IF (Y(I) * CMF + DYI - 0.1) 315,260,260
260 DUM1 = (1 - Y(I)) / DYI
IF (DUM1 - .001) 315,315,265
265 CMF = DUM1
GOTO 315
270 DMTL = DMTL + DYI * MTM(I)
IF (DYI) 275,315,290
275 IF (VN(I)) 290,290,280
280 IF (VN(I) + DYI * CMF) 285,290,290
285 CMF = - VN(I) / DYI * 1.00001
290 IF (KR(6)) 305,295,295
295 CLIP = CLIM / MTM(I)
IF (ABS(CMF * DYI) - CLIP) 315,315,300
300 CMF = CLIP / ABS(DYI)
GOTO 315
305 IF (ABS(CMF * DYI) - P) 315,315,310
310 CMF = P / ABS(DYI)
315 CMFF(I) = CMF
320 DY(IK) = DYI
IF (KR(6)) 322,321,321
321 RVL=AMAX1(-1,RV/2.)
CMF = AMIN1(CMF,MTG / ABS(ABS(DMTL - DMTG / MTG * MTL) / RVL - DMTSCAL1490
16))
322 IF (KR(7)-1) 355,355,330
325 FORMAT (1X2(8X2MVN9X1HY8X2HDY7X5HSCALE7X1ME4X6HFC IP)/(1XA4,5E
110.3,13,12,1X,A4,5E10.3,13,12))
330 NQ = I
335 WRITE(KOUT,325)(FANCA(J),VN(J),Y(J),DYA(J,LL),CMFF(J),E(J),IFC(J),
11P(J),J=1,NQ)
340 WRITE(KOUT,350)(E(I),I=1,IS)
WRITE(KOUT,350)(X(I),I=1,ISPQ)
WRITE(KOUT,345)(I(I),I=1,IS)
345 FORMAT(10I5)
350 FORMAT(8E12.4)
355 CONTINUE
IF (X(I)) 370,415,370
370 X1 = X(I) * CMF
ABX = ABS(X(I))
IF (ABS(X1) - .5) 380,380,375

```

```

SCAL1160
SCAL1170
SCAL1180
SCAL1190
SCAL1200
SCAL1210
SCAL1220
SCAL1230
SCAL1240
SCAL1250
SCAL1260
SCAL1270
SCAL1280
SCAL1290
SCAL1300
SCAL1310
SCAL1320
SCAL1330
SCAL1340
SCAL1350
SCAL1360
SCAL1370
SCAL1380
SCAL1390
SCAL1400
SCAL1410
SCAL1420
SCAL1430
SCAL1440
SCAL1450
SCAL1460
SCAL1470

```

```

SCAL1490
SCAL1500
SCAL1520
SCAL1530
SCAL1540
SCAL1550
SCAL1560
SCAL1580
SCAL1590
SCAL1600
SCAL1610
SCAL1620
SCAL1630
SCAL1690
SCAL1700
SCAL1710
SCAL1720

```

G - 39

```
375 CMF = .5 / ABX
    X1 = CMF * X(1)
380 IF (X1) 390,415,385
385 TM = TMAX
    X1 = AMIN(.2,X1)
    GOTO 395
390 TM = TMIN
    X1 = AMAX(.2,X1)
395 DTM = (TM - 1) / (TM * X1)
    IF (DTM - 1.) 400,410,410
400 CMF = DTM * CMF
405 T = TM
    GOTO 415
410 T = T / (1. - X1)
415 AA = AA * EXP(CMF * X(2))
    RETURN
    END
```

```
SCAL1730
SCAL1740
SCAL1750
SCAL1760
SCAL1770
SCAL1780
SCAL1790
SCAL1800
SCAL1810
SCAL1820
SCAL1830
SCAL1840
SCAL1850
SCAL1860
SCAL1870
SCAL1880
SCAL1890
```

CARD COUNT - 173

CREC0010
 CREC0030
 CREC0060

```

SUBROUTINE RECT(PCE)
  REAL NFIA(10), NFIB(10), FFIN(10), UM(10,10), TAU(10,10)
  DIMENSION DY(15), DYA(15,1)
  DIMENSION IF(150), CIJ(150,1)
  DIMENSION X(1), VLAM(150,1), CC(7960)
  COMMON KIN, KCUT, JAN, PCL, NAB, IDC, N, KRX(5)
  COMMON FAMA(150), FAPB(150), RB(150,2), RC(150,2)
  1 RD(150,2), RE(150,2), RF(150,2), TU(150,2), VN(150), Y(150), WM(150),
  2 VNU(150,10), FF(150), IFC(150), IP(150), CP(150), H(150), SB(150),
  3 TC(150), VLNK(150), E(150), LAMI(150), GAMK(1,1)
  COMMON A(16,16), W(3), WM, B(16), SCH, RHG, VEL, BL, KR(10), TILE(3), NFF,
  1 XC(50), XG(50)
  COMMON ZKE(10), TLO(10), DMCL(10), WALJP(10), EB(10), ALP(10), GAMH(10)
  1 GAMF(10), TQ(10,3), TK(10,3), WAT(10), IR(10), IOAT(10), KAT(10), UM
  2 EBL(10)
  COMMON FFA, IS, ISP, P, T, SIP, HIP, EL, ENL, FLIC, NO, IFL, ISPO, KRALP, CPF, IRE
  1 IER, AA, ITS, IN, IL, IT, MODE, HMELT, SMELT, TMAX, TMIN, MELT, SUMN, SUML, JC,
  ZHG, CPG, SVA, SVB, SVC, SVD, SUMC
  COMMON X, SAI, VMACH, KRZZ, SPI, GAM, THI, MX, DMCHI, HCH, NCV, TBT, IFRZ, NDA
  1 SRL, IXG, SVI, KKJ, IXC, NXG, SHI, NOATO, SAZ, FFF, MS, RV, CMF
  COMMON VA, HOS, DUM1, DUM2, EP, IFCJC, ISP2, WTG, VMU2, WTL
  COMMON MT, FKF(10), EAK(10), EXK(10), PMU(10,10), RMU(10,10)
  COMMON CIB(11), FNU(10), PNU(10), SLAM(10), BE(10), BY(10)
  1 IBC(10), JJ(10)
  EQUIVALENCE (CCI), FAMOA(1), (B(1), X(1)), (GAMK(1), VLAM(1))
  EQUIVALENCE (TU(151), IE(1)), (VNU(1), CIJ(1))
  EQUIVALENCE (TAU(1), A(151)),
  1A(51)), (FFIN(1), A(101))
  EQUIVALENCE (A(1), DY(1)), DYA(1))

```

CREC0290
 CREC0300
 CREC0310
 CREC0320
 CREC0330
 CREC0340
 CREC0350
 CREC0360
 CREC0370
 CREC0380
 CREC0390
 CREC0400
 CREC0410
 CREC0420
 CREC0430
 CREC0440
 CREC0450
 CREC0460
 CREC0470
 CREC0480
 CREC0490
 CREC0500
 CREC0510
 CREC0520

```

  M1 = 1
  M = 10(1)
  M1L = 0.
  MTG = 0.
  DUM2 = 0.
  I = 1
  DO 115 IK=1,N
  5 DYI = CMF * DY(IK)
  10 IF (DYI) 10,115,10
  15 IF (IFC(I)) 75,15,100
  20 IF (M - IK) 20,50,20
  25 IF (MOE) 50,50,25
  30 M1 = M1 + 1
  35 VN(I) = VN(I) * (1. + DYI)
  40 Y(I) = - 100.
  45 Y(I) = ALOG(VN(I))
  50 Y(I) = Y(I) + DYI
  VN(I) = EXP(Y(I))

```

```

55 VA = WTM(I) * VN(I)
   WTG = WTG + VA
   DUM2 = DUM2 + VA / FF(I)
   IF (IK -- IS) 60,60,65
60 PNUS(I) = VN(I)
   SLAM(I) = VN(I) / FF(I)
   GOTO 115
65 DO 7C K=1,15
   VA = VNU(I,K) * VN(I)
   PNUS(K) = PNUS(K) + VA
   SLAM(K) = SLAM(K) + VA / FF(I)
   GOTO 115
C   NON-PRESENT BASE CORRECTIONS AND TESTS
75 Y(I) = Y(I) + DYI
   IF (Y(I)) 115,80,80
80 IF (IFC(I) + 1) 115,85,85
85 IF (KR(6)) 95,90,90
90 IF (Y - YF(I) + .001) 115,95,95
95 Y(I) = 0.
   IFC(I) = + 1
   GOTO 115
100 VN(I) = VN(I) + DYI
   IF (.VN(I)) 110,110,105
105 WTL = WTL + VN(I) * WTM(I)
   GOTO 115
110 VN(I) = 0.
   IFC(I) = - 1
115 I=I+1
   EFF = WTG / DUM2
   DO 120 I=1,15
   PNUS(I) = PNUS(I) + AA * GAMF(I)
120 SLAM(I) = SLAM(I) * FFF
   RETURN
   END

```

```

CREC0530
CREC0540
CREC0550
CREC0560
CREC0570
CREC0580
CREC0590
CREC0600
CREC0610
CREC0620
CREC0630
CREC0640
CREC0650
CREC0660
CREC0670
CREC0680
CREC0690
CREC0700
CREC0710
CREC0720
CREC0730
CREC0740
CREC0750
CREC0760
CREC0770
CREC0780
CREC0790

CREC0810
CREC0820
CREC0830
CREC0840
CREC0950
CREC0860

```

OUTP0010
OUTP0030

SUBROUTINEOUTPT
REALNFIA(10),NFIB(10),FFIN(10),UM(10,10),TAU(10,10)
DIMENSION TF(15C),CIJ(150,1),IO(25)
DIMENSION X(1),VLAM(150,1),CC(7960)
COMMONKIN,KCUT,JAN,MCL,NAB,IOC,N,KKX(5)
COMMON FAMOA(150),FAMCB(150),RB(150,2),RC(150,2),
1 RD(150,2),RE(150,2),RF(150,2),TU(150,2),VN(150,2),VM(150),WTM(150),
2 VNU(150,10),FF(150),IFC(150),IP(150),CP(150),H(150),SB(150),
3 TC(150),VLNK(150),E(150),LAMI(150),GAMK(1,1)
COMMON A(16,16),W(3),WM,B(16),RHO,VEL,B1,KR(10),FILE(3),NFF,
1 XC(50),XG(50)

COMMON ZKE(10),TLC(10),DMCLO(10),WALJP(10),EB(10),ALP(10),GAMH(10)
1 ,GAMF(10),TQ(10,3),TK(10,3),MAT(10),IR(10),IOAT(10),KAT(10),UM
2 ,EBL(10)
COMMONFFA,IS,ISP,P,T,SIP,HIP,EL,ENL,FLIQ,NO,IFL,ISPG,KRALP,CPF,IRE
1,IER,AA,ITS,IN,IL,IT,MODE,HMELT,SMELT,TMAX,TMIN,MELT,SUMN,SUML,JC,
2HG,CPG,SVA,SVB,SVC,SVD,SUMC
COMMONXG,SAL,VMACH,KRZZ,SPI,GAM,THI,MX,DMCHI,HCH,NCV,TBT,IFRZ,NOA
1T,SRI,IXG,SVI,KKJ,IXC,NXG,SHI,NOATO,SAZ,FFF,WS,RV,CHF
COMMONVA,HOS,DUM1,DUM2,EP,IFCJC,ISP2,WTG,VMU2,MTL
COMMONHT,FKF(10),EAK(10),EXK(10),PMU(10,10),RMU(10,10)
COMMONIB(11),FNU(10),PNUS(10),SLAM(10),BE(10),BY(10),
1 IBC(10),JJ(10)

EQUIVALENCE(CC(1),FAMOA(1)),(B(1),X(1)),(GAMK(1),VLAM(1))
EQUIVALENCE(TU(151),TF(1)),(VNU(1),CIJ(1))
EQUIVALENCE(TAU(1),A(151)),
1A(51)),(FFIN(1),A(101))
5 FORMAT(9X6HTEMP =F10.4,17H DEG-K PRES =F8.4,17H ATM MOL WOUTP0280
1T =E11.7) OUTP0290

10 FORMAT(/87H SPECIES MOL-FRAC. D-LN-KP/D-LN-T LOG-PP LOGOUTP0300
16-KP FLAG ERROR CP / (1X2A4,E13.5,15,2E13.5)) LOUTP0310
15 FORMAT(10X10HENTHALPY =E14.7,20H CAL/GM ENTROPY =E12.5,13H CAL/OUTP0320
10H-DEG K/10X 9H DENSITY =E13.6,10H LB/CU-FT.
15X25H MASS CONDENSED/MASS GAS =E12.5,11H B-PRIME =,E12.5) OUTP0340
20 FORMAT(//) OUTP0350
25 FORMAT(5X SHVEL =E10.3,16H FT/SEC MACH =E10.3, 9H AREA =E10.30UTP0360
1,12H SQFT/LB/SEC./9H H-GAS =,E12.5,21H BTU/LB H-KINETIC =E12.5,
27H BTU/LB)

30 FORMAT(10X17HFRACTION LIQUID =F8.5)
YAN(Q) = SIN(Q) / COS(Q)
HEPS = 0.01
C FIND VKIN AS FIRST NONZERO FKF(M)
IF(MT) 702,702,699
699 DO 704,M=1,RT
704 CONTINUE
701 VKIN=FKF(M)
702 VKIN=0.0
703 CONTINUE
35 MM = AA / P
WRITE(KOLT,5)T,P,MM

OUTP0380
OUTP0390
OUTP0400

OUTP0410
OUTP0420

```

SHIP = HIP
SSIP = SIP
HG = C.
HL = C.
HIP = 0.
WTLS=MTL
SIP = 0.
PMU2 = 0.
HTIL = 0.
BPRM=M(3)
60 MINI = - 1
65 IKMM=N
70 DO 100 J=1,IKMM
HTIL = HTIL + VN(J) / FF(J) * H(IJ)
PRU2 = PMU2 + VN(J) / FF(J) * WTM(J)
VA = VN(J) * H(IJ)
IF (IFC(J)) 80,75,80
75 HG = HG + VA
GOTO 95
80 HL = HL + VA
IF (KR(8)) 95,95,85
85 IF (TF(J) - .001 - T) 95,95,90
90 BPRM = BPRM - VN(J) * WTM(J) / WTG
HL = HL - VA
WTLS=WTLS-VN(J) * WTM(J)
95 SIP = SIP + VN(I) * [SB(J) - 1.9869 * Y(J)]
100 CONTINUE
IF (ITS) 105,135,135
105 CO 110 I=1,IKMM
110 VN(I) = VN(I) / P
IF (MCV) 115,115,120
115 WRITE(KOUT,270)(FAMQA(I),FAMOB(I),VN(I),I=1,IKMM)
GOTO 125
120 WRITE(KOUT,10)(FAMQA(I),FAMOB(I),VN(I),TC(I),Y(I),VLNK(I),IFC(I),
1(I),CP(I),I=1,IKMM)
125 DO 130 I=1,IKMM
130 VN(I) = VN(I) * P
135 CONTINUE
HTIL = HTIL / PMU2
WTR = WTLS/WTG
HIP = HG + HL
HIP = (HIP + VN(MELT) * FLIQ * HMELT) / AA
SIP = (SIP + VN(MELT) * FLIQ * SMELT) / AA
KRI = KR(1) - IFRZ + KR(8)
150 HTIL = HG / AA
FAF = 0.
IF (KR(4)) 160,155,155
155 HIP = (HG + (HL - WTR * HG) / MS) / AA
160 IF (KR(10)) 165,180,165
165 IF (MCV + ITS) 166,180,180
166 IF (KRX(1)) 180,180,168
168 CONTINUE

```

```

OUTP0450
OUTP0460
OUTP0470
OUTP0480
OUTP0490

OUTP0500
OUTP0510
OUTP0520

OUTP0630

OUTP0650
OUTP0660
OUTP0670
OUTP0680
OUTP0690
OUTP0700
OUTP0710
OUTP0720
OUTP0730
OUTP0740
OUTP0750
OUTP0760

OUTP0780
OUTP0790
OUTP0800
OUTP0810
OUTP0820
OUTP0830
OUTP0840
OUTP0850
OUTP0860
OUTP0870
OUTP0880
OUTP0890

OUTP0950

OUTP0970
OUTP0980
OUTP0990
OUTP1000
OUTP1030
OUTP1040

OUTP1050
OUTP1060
OUTP1070

```

```

*****THE SQUEE CALLS AND FORMAT 167 YIELD A COMPRESSED OUTPUT TO BE
*****READ AS (E6.4,E8.5,2E6.4,F4.2,E7.5,E6.4,2E8.5,.....)
CALL SQUEE (P,IO(1),4)
CALL SQUEE (M(2),IO(4),4)
CALL SQUEE (BPRM,IO(7),5)
CALL SQUEE (VKIN,IO(10),4)
IO(13) = FAF*100.
CALL SQUEE (T,IO(14),5)
CALL SQUEE (WTR,IO(17),4)
CALL SQUEE (HTIL,IO(20),5)
CALL SQUEE (HIP,IO(23),5)
KRI=KRI+1
WRITE (KOUT,169) P,M(2),BPRM,VKIN,FAF,
1 T,WTR,HTIL,HIP,FAMOA(JC), FAMOB(JC),KRI,TILE
WRITE (KOUT,1675) IO,FAMCA(JC),FAMOB(JC),KRI,TILE
PUNCH 167, IO,FAMOA(JC), FAMOB(JC), KRI,TILE
KRI=KRI-1
167 FORMAT (I4,A1,I1,I4,A1,I1,I6,A1,I1,I4,A1,I1,
1 14,I5,A1,I1,I4,A1,I1,I6,A1,I1,I6,A1,I1,I6,A1,I1,A4,A2,I1,4X,2A4,A2)
1675 FORMAT(I1X,I4,A1,I1,I4,A1,I1,I6,A1,I1,I4,A1,I1,I4,I5,A1,I1,
1 14,A1,I1,I6,A1,I1,I6,A1,I1,A4,A2,I1,4X,2A4,A2)
169 FORMAT(I1X,E11.4,E12.5,2E11.4,F4.2,E12.5,E11.4,2E12.5,A4,A2,I1,
1 2A4,A2)
180 RHO = P / T * MM / 1.3146
WRITE(KOUT,15)HIP,SIP,RHO,WTR,BPRM
IF (FLIQ) 185,190,185
185 WRITE(KOUT,30)FLIQ
190 IF (ITS) 245,260,260
245 IF (KR(5) - 1) 260,255,250
250 MCH = HIP
SCH = SIP
255 VELSQ = (MCH - HIP) * 2.
VMACH = SORT(VELSQ / GAM * MM / (1.9869 * T))
VEL = SORT(VELSQ * 45054.)
AREA = 1. / (VEL * RHO)
MG=MG/WTG*1.8
MKIN=(MCH-HIP)*1.6
WRITE(KOUT,25)VEL,VMACH,AREA,MG,MKIN
260 CONTINUE
265 FORMAT(22H SURFACE SPECIES IS 2A4)
270 FORMAT(/3(5X7HSPECIES3XBHOLE FR.2X)/(5X2A4,E12.5,5X2A4,E12.5,5X2A4,
14,E12.5))
IF (MODE - 1) 280,275,280
275 WRITE(KOUT,265)FAMOA(JC),FAMOB(JC)
280 WRITE(KOUT,20)
IF (ITS) 285,315,315
285 RETURN
315 HIP = SHIP
SIP = SSIP
RETURN
END

```

OUTP1100
 OUTP1110
 OUTP1120
 OUTP1130

OUTP1420
 OUTP1430
 OUTP1440
 OUTP1450
 OUTP1460
 OUTP1470
 OUTP1480

OUTP1490
 OUTP1500
 OUTP1510
 OUTP1520
 OUTP1530
 OUTP1540
 OUTP1550
 OUTP1560
 OUTP1570

OUTP1700
 OUTP1710
 OUTP1720
 OUTP1730

CARD COUNT - 155

```

SUBROUTINEPROPS
REALNFA(10),MFI(10),FFIN(10),UM(10,10),TAU(10,10)
DIMENSIONTT(1),RHO(1),CPGAR(1),SC(1),PR(1),VK(10)
DIMENSIONZK(10)
DIMENSIONZI(10)
DIMENSIONVI(10)
DIMENSIONCAPC(1)
DIMENSIONTF(150),CIJ(150,1)
DIMENSIONX(1),VLAM(150,1),CC(7960)
COMMONKIN,KOUT,JAN,MOL,NAB,IOC,N,KKX(5)
COMMONFAPDA(150),FAMOB(150),R(150,2),TU(150,2),VM(150),Y(150),WTM(150),
1 RD(150,2),RE(150,2),RF(150,2),IFC(150),IP(150),CP(150),H(150),SB(150),
2 VMU(150,10),FF(150),IFC(150),LAMI(150),GAMK(1,1)
3 TC(150),VLNK(150),E(150),MM,B(16),SCH,RHO,VEL,B1,KR(10),TILE(3),NFF,
COMMONA(16,16),M(3),MM,B(16),SCH,RHO,VEL,B1,KR(10),TILE(3),NFF,
1 XC(50),XG(50)
COMMONZKE(10),TLC(10),DMCLO(10),WALJP(10),EB(10),ALP(10),GAMH(10)
1 *GAMF(10),TQ(10,3),TK(10,3),WAT(10),TR(10),IOAT(10),KAT(10),UM
2 *EBL(10)
COMMONFFA,IS,ISP,P,Y,SIP,HIP,EL,ENL,FLIQ,NQ,IFL,ISPQ,KRALP,CPF,IRE
1 IER,AA,ITS,IN,IL,IT,MODE,MMELT,SHELT,TMAX,TMIN,MELT,SUMN,SUML,JC,
2HG,CPG,SWA,SVB,SVC,SVO,SUMC
COMMONNXC,SAL,VMACH,KRZZ,SPI,GAM,THI,MX,DMCHI,HCH,NCV,TBT,IFRZ,NOA
1T,SRI,FXG,SVI,KKJ,IXC,NXG,SHI,NOATO,SAZ,FFF,MS,RV,CMF
COMMONVA,MOS,DUM1,DUM2,EP,IFCJC,ISP2,WTG,VMU2,MTL
COMMONMT,FKF(10),EAK(10),EXK(10),PMU(10,10),RMU(10,10)
COMMONB(11),FNU(10),PNUS(10),SLAM(10),BE(10),BY(10),
1 IBC(10),JJ(10)
EQUIVALENCE(CC(1),FAPDA(1)),(B(1),X(1)),(GAMK(1),VLAM(1))
EQUIVALENCE(TU(151),TF(1)),(VMU(1),CIJ(1))
EQUIVALENCE(TAU(1),A(151)),
IA(51)),(FFIN(1),A(101))
INFA(1),A(1)),(NFIB(1),
5 FORMAT(10X60H CP-FROZEN CP-EQUIL DLNM/DLNT DLNM/DLNP
IAMPA /10X5E12.5)
10 FORMAT(1H15X4SHAETHERM CHEMICAL EQUILIBRIUM (ACE) SOLUTION//)
15 FORMAT(4H1-----50H-----FOLLOWING OUTPUT NON-CONVERGENT-----
1-//)
IF (MCV) 25,25,20
20 WRITE(KOUT,15)
GOTO 35
25 IF (KR(10) / 2 + KR(7) + KR(2) + KR(3)) 35,35,30
30 WRITE(KOUT,10)
35 CPF = CPF / AA
ALF = A(2,1) / A(1,1)
CSP = 1. / (A(1,1) * AA)
IF (MODE - 3) 40,45,40
40 CSP = CSP / T
45 BETA = P * (A(2,2) - A(1,2) / A(1,1) * A(2,1)) - 1.
GAM = 1. - ALF
GAM = 1. / (1. + BETA - 1.9869 / AA * GAM / CSP * GAM * P)
WRITE(KOUT,5)CPF,CSP,ALF,BETA,GAM
ITS = - 1

```

PROP0010
PROP0030
PROP0040
PROP0050
PROP0060
PROP0070

GPROP0320
PROP0330
PROP0340
PROP0350
PROP0360
PROP0370
PROP0380
PROP0390
PROP0400
PRC0410
PROP0420
PROP0430
PROP0440
PROP0450
PROP0460
PROP0470
PROP0480
PROP0490
PROP0500
PROP0510

CAPC(I) = 1.
 I1 = 1
 NM = AA / P
 ISM = 15 - 1
 IT(I) = T * 1.8
 RHO(I) = AA / (1.3146 * T)
 CPBAR(I) = CPF
 AISTR = 1.13
 FORM NECESSARY SUMMATIONS

PMU2 = 0.
 CPTIL = 0.
 HTIL = 0.
 PMU1 = 0.
 TMU3 = 0.
 I = 1

50 DO 60 IK=1,N
 IF (IFC(I)) 60,55,60
 55 PMU1 = PMU1 + VN(I) * FF(I)
 60 I=I+1
 VMU1 = PMU1 / P

DO 66 K=1,ISM
 VK(K)=VN(K)
 IF (MTRIK)-1.1 66,65,65
 65 ZR(K) = VN(K) / FF(K)
 66 CONTINUE
 AMU3 = 0.
 PMU6 = 0.
 MD2 = 1.385
 MD7 = MD2 / PMU1

MD2 = 1.2 * AISTR / (PMU1 * MN)
 MD4 = 0.284 * MD2
 MD5 = .32 * AISTR / (PMU1 * MN)
 MD8 = MD4 / PMU1
 I = 1

DO 90 IK=1,N
 IF (MTM(I) - 1.1 90,70,70
 70 VA = VN(I) / FF(I)
 VB = VA * MTR(I)
 VC = VN(I) * FF(I)
 IF (IK - 15) 85,85,75

IF (IK - 15) 85,85,75
 75 DO 80 K=1,ISM
 VK(K)=VK(K)+VN(I)*VMU(I,K)
 80 ZR(K) = ZR(K) + VA * VMU(I,K)
 85 PMU2 = PMU2 + VB
 TMU3 = TMU3 + VA
 CPTIL = VA * CP(I) + CPTIL
 HTIL = HTIL + VA * H(I)
 AMU5 = AMU5 + VB / (MD2 + VC * (MTM(I) * MD2 - MD7))
 PMU6 = PMU6 + VA / (MD4 + (MD5 * MTM(I) - WDR) * VC)
 90 I=I+1
 VMU5 = AMU5 / AA
 VMU6 = (PMU6 + CPTIL / 1.9869 - 2.5 * TMU3) / P

PROP0520
 PROP0530
 PROP0540
 PROP0550
 PROP0560
 PROP0570
 PROP0580
 PROP0590
 PROP0600
 PROP0610
 PROP0620
 PROP0630
 PROP0640
 PROP0650
 PROP0660
 PROP0670
 PROP0680
 PROP0690
 PROP0710

PROP0730

PROP0740
 PROP0750
 PROP0760
 PROP0770
 PROP0780
 PROP0790
 PROP0800
 PROP0810
 PROP0820
 PROP0830
 PROP0840
 PROP0850
 PROP0860
 PROP0870
 PROP0880
 PROP0890

PROP0900
 PROP0910
 PROP0920
 PROP0930
 PROP0940
 PROP0950
 PROP0960

PROP0980
 PROP0990

G-47

```

VMU2 = PPU2 / P
VMU3 = TPU3 / PMU2
ZKS = 1.0
VKS=1.0
DO 95 K=1,ISM
VK(K)=VK(K)/AA*WTM(K)
ZK(K) = ZK(K) / PMU2 * WTM(K)
VKS=VKS-VK(K)
95 ZKS = ZKS - ZK(K)
DO 100 I=1,IS
ZI(I) = ZKS * CIJ(IS,I) / WTM(IS) * WAT(I)
VI(I)=VKS*CIJ(IS,I)/WTM(IS)*WAT(I)
DO 100 K=1,ISM
VI(I) = VI(I) * CIJ(K,I) * VK(K) / WTM(K) * WAT(I)
ZI(I) = ZI(I) * CIJ(K,I) * ZK(K) / WTM(K) * WAT(I)
CPTIL = CPTIL / PMU2
HTIL = HTIL / PMU2 * 1.0
OMEGA = 1.07 / (T / 106.7) * 0.159
DBAR = 4.16E - 8 * T / P * SORT(T) / OMEGA
VMU = RHO(I) * DBAR * VMUS / VMU1
105 CONTINUE
TCOMD = RHO(I) * DBAR / WM * VMU6 / VMU1 * 1.9869
SC(II) = VMUS / VMU2 * WM
PA(II) = CPF / VMU6 * VMUS / 1.9869 * WM
VMU12 = VMU1 * VMU2
WRITE(KOUT,110)T1,VMU,TCOMD,DBAR,PR,SC,VMU1,VMU2,WM,HTIL,CPTIL
110 FORMAT(/58H PROPERTY ROUTINE OUTPUT IN LB-MASS,FT,SEC,BTU,AND
1-R /67H TEMP VISC MUI MUZ MOL.WT
1 SC/6E12-5/57H MUI MUZ
1 CPTIL/5E12-5)
115 FORMAT(1X10E12-5)
120 FORMAT (58H ELEMENTAL K AND Z MASS FRACTIONS BY ATOMIC NUMBER . .
1. -/1X10(4X13,5X))
WRITE(KOUT,120)(KAT(J),J=1,IS)
WRITE(KOUT,115)(VI(J),J=1,IS)
WRITE(KOUT,115)(ZI(J),J=1,IS)
RETURN
END

```

```

PROPI000
PROPI010
PROPI020
PROPI030
PROPI050
PROPI060
PROPI070
PROPI080
PROPI090
PROPI100
PROPI110
PROPI120
PROPI130
PROPI140
PROPI150
PROPI160
PROPI170
PROPI180
PROPI190
PROPI200
DEGPROP1230
PR PROP1240
HTILPROP1250
PROPI270
PROPI300
PROPI310
PROPI320
PROPI330

```

CARD COUNT - 142

```

C
SUBROUTINE RERAY(N,C,MN,D,MN,LS,IS)
DIRECT INVERSION PROCEDURE -- C IS REPLACED BY C*-1
DIMENSION D(16,1),S(16),C(16,1),L(16),S(16),LL(16),LL(16),LS(1)
M1=N+1
KOUT=5
NP=M*NN
DO 11 I=1,MP
  LLL(I)=1
  IF(LS(I)) 113,113,112
112 L(I)=LS(I)
  GO TO 11
113 L(I)=I
  11 CONTINUE
  IX=1
  IF(IX+2) 111,109,111
106 FORMAT(1H L(I)=1,13,5X (30I3))
107 FORMAT(15H ((C(I,J),J=1,13,12H), (D(J),J=1,13, 6H), I=1,13,15H) DE
  1FORE RERAY)
108 FORMAT(2X 11E10.3/112X 10E10.3)
109 WRITE(KOUT,107) NP,MN,N
  WRITE(KOUT,106)NP,(L(I),I=1,MP)
  IX=0
DO 110 I=1,M
110 WRITE(KOUT,108)(C(I,J),J=1,MP),(D(I,J),J=1,MN)
111 IS=1
C
  TRIANGULATE MATRIX
  DO 15 I=1,M
  DO 160 M=1,MP
160 S(M)=ABS(C(I,M))
  18 IS=0
  S(II)=1.
  GO TO 12
C
  REDUCE ROW I BY PRECEDING ROWS
  K=L(J-1)
  DIV=C(I,K)
  IF(DIV) 161,17,161
161 C(I,K)=0.
  DO 162 M=1,MP
  DIV=C(I,M)
  DIVC=DIV*(J-1,M)
  S(M)=AMAX1(S(M),ABS(DIVC))
162 C(I,M)=C(I,M)+DIVC
  IF(MN) 17,17,163
163 DO 164 M=1,MN
164 D(I,M)=D(I,M)+DIV*(J-1,M)
  17 CONTINUE
C
  SEEK MAXIMUM PIVOT
  12 DIV=0.
  DO 13 JJ=1,M
  M=L(JJ)
  IF(ABS (C(I,M))-DIV)13,13,121

```

```

121 DIV=ABS (C(I,M))
    K=M
    J=J
13 CONTINUE
    SD(I)=DIV/S(K)
    L(J)=L(I)
    L(I)=R
    IF(SD(I)-1.E-8) 131,131,14
SINGULAR MATRIX RETURN
C 131 IS=-1
    WRITE(KOUT,151) (II,L(II),SD(II),II=1,I)
    RETURN
14 DIV=C(I,K)
    C(I,K)=1.0
    K=LL(J)
    LLL(J)=LL(I)
    LLL(II)=K
    LL(K)=I
C NORMALIZE ROW
    IF(NNN) 143,143,141
141 DO 142 J=1,NNN
142 O(I,J)=O(I,J)/DIV
143 DO 15 J=1,NP
15 C(I,J)=C(I,J)/DIV
    IF(IX) 152,150,152
151 FORMAT(24H PIVOT ROW/COL/RES.RATIO 5(14,1M/13,1M/E9.2,1M,))
150 WRITE(KCUT,151) (I,L(II),SD(II),I=1,NP)
C DIAGONALIZE MATRIX
152 MN=N-1
    DO 20 I=1,NM
    K=L(I+1)
    DO 20 J=1,I
    DIV=-C(J,K)
    IF(DIV) 19,20,19
19 C(I,J,K)=C.
    IF(NNN) 191,191,192
192 DO 193 M=1,NNN
193 D(I,J,M)=D(I,J,M)+DIV*D(I+1,M)
191 DO 201 M=1,NP
201 C(I,J,M)=C(I,J,M)+DIV*C(I+1,M)
20 CONTINUE
C INTERCHANGE COLUMNS
    DO 30 II=1,NP
    I=II
21 J=L(II)
    L(II)=I
    IF(J-1) 22,30,22
22 IF(15) 25,23,25
23 DO 24 M=1,N
    S(I)=C(I,M,I)
24 C(I,M,I)=C(I,M,J)
    IS=I

```

```

I=J
60 TC 21
25 IF(I5-J)26,28,26
26 DO 27 M=1,N
27 C(M,I)=C(M,J)
I=J
60 TO 21
28 DO 29 M=1,N
29 C(M,I)=S(M)
IS=0
30 CONTINUE
INTERCHANGE ROWS
DO 40 I1=1,N
I=I1
31 J=LL(I)
LL(I)=I
IF(J-I)32,40,32
32 IF(I5)35,33,35
33 DO 34 M=1,MP
S(M)=C(I,M)
34 C(I,M)=C(J,M)
IF(MNN) 343,343,341
341 DO 342 M=1,NNN
SD(M)=D(I,M)
342 D(I,M)=D(J,M)
343 IS=1
I=J
60 TO 31
35 IF(I5-J)36,38,36
36 DO 37 M=1,MP
37 C(I,M)=C(J,M)
IF(MNN) 373,373,371
371 DO 372 M=1,NNN
372 D(I,M)=D(J,M)
373 I=J
60 TO 31
38 DO 39 M=1,MP
39 C(I,M)=S(M)
IF(MNN) 393,393,391
391 DO 392 M=1,NNN
392 D(I,M)=SD(M)
393 IS=0
40 CONTINUE
IF(I5) 411,409,411
407 FORMAT(15H '(C(I,J),J=1,13,12H), (D(J),J=1,13, 6H), I=1,13,15H) AF
ITER RERAY )
409 WRITE(KOUT,407) NP,NNN,N
DO 410 I=1,N
410 WRITE(KOUT,108)(C(I,J),J=1,MP), (D(I,J),J=1,NNN)
411 RETURN
END

```



```

25 CONTINUE
VK2 = AA * FK(F) * (ABS(VK1)) * (EXK(M) - 1.) * EXP1 - EAK(M) /KINE0580
IRT) KINE0590
VK3 = VK2 KINE0600
IF (EXK(M) - 1.) 35,30,30 KINE0610
30 VK3 = VK2 * EXK(M) KINE0620
PM TIMES FORWARD RATE OF REACTION I (PM=AA) KINE0630
35 PMR(M) = VK2 * VK1 KINE0640
PKP(M) = PKP(M) * VK3 KINE0650
PKR(M) = PKR(M) * VK3 KINE0660
RAI(M)=AMAX1(PKP(M),PKR(M))
IF (KR(7)-1) 40,40,36
36 F(M-1) 37,37,39
37 M=1;E(KO,I,38)
38 FORMAT(2X1M7X3HLKP6X8HDLKP/DLT6X4HPMRR8X4HPMRP9X3HPNR9X3HRAT)
39 WRITE(KOUT,41) M,SUMK,DKPT(M),PKR(M),PKP(M),PMR(M),RAT(M)
40 MACH) = M
41 FORMAT(13,2X,6E12.5)
45 FORMAT(1X26H(I,J),B(1),I=1,8,J=1,8 IN)
50 FORMAT(1X12E10.3)
55 FORMAT(1X27H(I,J),B(1),I=1,8,J=1,8 OUT)
IF (KR(7) - 1) 80,80,65
45 CONTINUE
WRITE(KOUT,50)PMU
WRITE(KOUT,215)
WRITE(KOUT,50)IER(I),I=1,IS)
7- WRITE(KOUT,45)
DO 75 I=1,ISP2
75 WRITE(KOUT,50)IA(I,J),J=1,ISP2),B(1)
80 CONTINUE
C-----ORDER REACTIONS
85 IF IRT - 1) 105,105,90
90 K = 0
DO 100 M=2,MT
IF (RAT(M) - RAI(M) - 1) 100,100,95
95 K = RA(M)
RAI(M) = RA(M - 1)
MACH - 1) = K
DUM1 = RAT(M)
RAT(M) = RAT(M - 1)
RAI(M - 1) = DUM1
100 CONTINUE
IF (K) 105,105,85
C-----START SECOND MAJOR LOOP ON REACTIONS
105 DO 200 MM=1,MT
M = RA(M)
C-----IS IT A CONTROLLING REACTION
LL(M) = 0
DO 125 L=1,15
IF (PRM(L,M)) 110,125,110
110 DO 115 K=1,MM
IF (L - LL(K)) 115,125,115

```

KINE0680
KINE0690
KINE0700
KINE0710
KINE0730

KINE0810
KINE0820
KINE0830
KINE0840
KINE0850
KINE0860
KINE0870
KINE0880
KINE0890
KINE0900
KINE0910
KINE0920
KINE0930
KINE0940
KINE0950
KINE0960
KINE0970
KINE0980
KINE0990
KINE1000
KINE1010

KINE1030
KINE1040
KINE1050
KINE1060
KINE1070
KINE1080
KINE1090

```

115 CONTINUE
IF (RAT(M)) - EB(L) 125,125,120
C* * * YES, IT IS FOR MASS BALANCE L
120 LL(MP) = L
    GOTC 130
125 CONTINUE
C* * * NO, IT IS NOT, ADD INTO ALL MASS BALANCES
    I1 = 1
    I2 = 15
    GOTC 170
C*****REARRANGE ACCORDING TO CONTROLLING REACTION
130 DUM1 = PRMU(L,M)
    PRMU(L,M) = 0.
    DO 145 I=1,15
    IF (PRMU(I,M)) 135,165,135
135 DUM2 = PRMU(I,M) / DUM1
    MP = MM + 1
    IF (MT - MP) 155,140,140
140 DO 150 K=MP,MT
    MI = MAX(K)
    PRMU(I,MI) = PRMU(I,MI) - DUM2 * PRMU(L,MI)
    IF (ABS(PRMU(I,MI)) - .001) 145,150,150
145 PRMU(L,MI) = 0.
150 CONTINUE
155 DO 160 K=1,15PQ
160 A(L + 2,K) = A(L + 2,K) - DUM2 * A(L + 2,K)
    B(L + 2) = B(L + 2) - DUM2 * B(L + 2)
    E(L) = E(L) - DUM2 * E(L)
    DUM2 = ABS(DUM2)
    EB(L) = AMAX1(EB(L), DUM2 * EB(L))
165 CONTINUE
    PRMU(L,MI) = DUM1
C*****ADD CONTROLLING REACTION INTO ITS MASS BALANCE
    I1 = L
    I2 = L
170 DO 175 J=1,15
    SUMD = RRU(J,M) * PKR(M) - PRMU(J,M) * PKP(M)
    DO 175 I=1,12
175 A(I + 2,J + 2) = A(I + 2,J + 2) - SUMD * PRMU(I,M)
    SUMD = - PKP(M) * DKPT(M) - EAK(M) / RT * PMR(M)
    DO 180 I=1,12
    DUM1 = PMR(M) * PRMU(I,M)
    A(I + 2,2) = A(I + 2,2) - DUM1
    A(L + 2,1) = A(L + 2,1) + SUMD * PRMU(I,M)
    E(L) = E(L) + DUM1
    B(L + 2) = B(L + 2) + DUM1
180 EB(L) = AMAX1(EB(L), ABS(PRMU(I,M) * RAT(M)))
    IF (KR(7) - 1) 200,200,185
185 WRITE(KOUT,215)
190 WRITE(KOUT,55)
    DO 195 I=1,15P2

```

KINE1100
KINE1110
KINE1120
KINE1130
KINE1140
KINE1150
KINE1160
KINE1170
KINE1180
KINE1190
KINE1200
KINE1210
KINE1220
KINE1230
KINE1240
KINE1250
KINE1260
KINE1270
KINE1280
KINE1290
KINE1300
KINE1310
KINE1320
KINE1330
KINE1340
KINE1350
KINE1360

KINE1370
KINE1380
KINE1390
KINE1400
KINE1410
KINE1420
KINE1430
KINE1440
KINE1450
KINE1460
KINE1470
KINE1480
KINE1490
KINE1500

KINE1510

KINE1520
KINE1530
KINE1540
KINE1550
KINE1560
KINE1570
KINE1580

KINE1590
KINE1600
KINE1610
KINE1620
KINE1630

```

195 WRITE(KOLT,50)(A(I),J),J=1,ISP2),8(1)
    CONTINUE
WRITE(KOLT,5)M,I1,I2,L,LL,MM,MA
WRITE(KOLT,5)PRM
200 CONTINUE
DO 206 MM=1,MT
  L=LL(MH)
  IF(L) 201,206,201
  201 M=MA(MH)
  AR=1-
  EXEL=(PKR(M)+1.E-36)/(PKP(M)+1.E-36)
  IF(ABS(EXEL-1.)-.1) 204,204,202
  202 ELK=ALOG(EXEL)
  AR=EXEL/EXEL-1.)-1./ELK
  204 DO 205 J=1,IS
  205 A(L+2,J+2)=A(L+2,J+2)+E(L)*(PMU(J,M)*(1.-AR)+AR*RMU(J,M))
  A(L+2,1)=A(L+2,1)-E(L)*DKPT(M)*(1.-AR)
  206 CONTINUE
  EML = 0.
  DO 210 I=3,ISP2
  210 ENL = AMAX1(ABS(E(I-2)) / EB(I-2) ,EML)
  215 FORMAT(1X12HEB(M),M=1,MT)
  IF(ENL-1.E-5) 216,216,225
  216 IF(ENL-1.E-4) 220,225,225
  220 ITS = - 1
  225 RETURN
END

```

KINE1650
KINE1660
KINE1680

KINE1700
KINE1710
KINE1720

CARD COUNT = 183

G-55

```
SUBROUTINE SQUEE (A,I,IDEE)
DIMENSION I(1)
DATA LUS,INUS/1H+,1H-/
I(1)=C
I(3)=C
B=ABS(A)*1.000001
IF(ABS(A)-1.E-10) 7,7,1
1 IF(B-1.C) 3,2,2
2 B=B/10.
I(3) = I(3)+1
GO TO 1
3 IF(B-0.1) 4,5,5
4 B=B*10.
I(3) = I(3)-1
GO TO 3
5 I(1) = A*10.** (IDEE-I(3))*1.000001
IF(I(3)) 6,7,7
6 I(2) = INUS
GO TO 8
7 I(2) = LUS
8 I(3) = IABS(I(3))
RETURN
END
```

CARD COUNT -

23

G-56

```
SUBROUTINE SWAP(A,B,SLA,SLB)
DIMENSION A(16,1),B(1),SLA(16,1),SLB(1)
DO 5 I=1,16
  B(I)=SLB(I)
DO 5 J=1,16
  5 A(I,J)=SLA(I,J)
RETURN
END
```

CARD COUNT - 8

UNCLASSIFIED

Security Classification

DOCUMENT CONTROL DATA - R&D

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11. SUPPLEMENTARY NOTES		12. SPONSORING MILITARY ACTIVITY Air Force Rocket Propulsion Laboratory	
13. ABSTRACT <p>The report describes and gives user's instructions for a computer program (ASTHMA) for predicting the in-depth temperature history and the surface recession (ablation) history of a two-dimensional, axi-symmetric, non-charring material. The in-depth solution procedure is of the conventional explicit finite-difference type. It will account for anisotropic heat conduction in the main material and allows several back-up materials. The heated surface boundary condition can have any one of three forms at each surface location at each instant:</p> <ol style="list-style-type: none">1. Very general film-coefficient-based simultaneous heat and mass transfer type, including any number of equilibrium reactions for an environment and any ablating material, and four specific kinetically controlled reactions for carbon surfaces.2. Simplified radiation energy balance type, no ablation ("cooldown" option).3. Specified temperature and surface recession rate. <p>The program was designed specifically for rocket nozzle use but has sufficient generality that it can be employed for many axi-symmetric shapes.</p> <p>A User's Manual Brief Description is also presented for a computer program (ARCACE) for calculating surface thermochemical response of materials, including kinetically controlled surface reactions. This program provides for surface thermochemical boundary condition information input to the ASTHMA program. The program was designed specifically for graphitic material and solid propellant combustion products environment but can be employed for many materials and chemical systems.</p>			

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		ROLE	WT	ROLE	WT	ROLE	WT

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