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USAAVLABS TECHNICAL REPORT 69-94

**AN EXPERIMENTAL INVESTIGATION OF THE EFFECTS OF ROTOR
HEAD CONFIGURATION AND FUSELAGE YAW ON THE WAKE
CHARACTERISTICS AND ROTOR PERFORMANCE OF
A 1/8TH SCALE HELICOPTER**

AD 869390

By

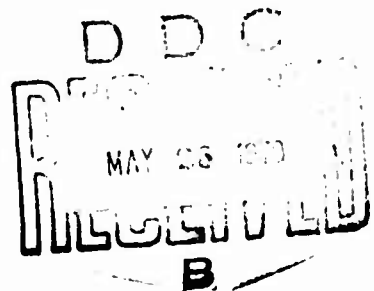
James C. Liville

February 1970

**U. S. ARMY AVIATION MATERIEL LABORATORIES
FORT EUSTIS, VIRGINIA**

**CONTRACT DA 44-177 -AMC-203(T)
UNITED AIRCRAFT CORPORATION
SIKORSKY AIRCRAFT DIVISION
STRATFORD, CONNECTICUT**

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DEPARTMENT OF THE ARMY
US ARMY AVIATION MATERIEL LABORATORIES
FORT EUSTIS, VIRGINIA 23604

This report has been reviewed by the U. S. Army Aviation Materiel Laboratories and is considered to be technically sound. The report is published for the exchange of information and the stimulation of ideas.

Task 1F162204A13903
Contract DA 44-177-AMC-203(T)
USAAVLABS Technical Report 69-94
February 1970

AN EXPERIMENTAL INVESTIGATION OF THE EFFECTS OF ROTOR HEAD
CONFIGURATION AND FUSELAGE YAW ON THE WAKE CHARACTERISTICS
AND ROTOR PERFORMANCE OF A 1/8TH SCALE HELICOPTER

SER-50604

by

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Prepared by

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for

U. S. ARMY AVIATION MATERIEL LABORATORIES
FORT EUSTIS, VIRGINIA

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tories, Fort Eustis, Virginia 23604.

SUMMARY

A wind tunnel test was conducted to investigate the effect of rotor head configuration and fuselage pitch and yaw angle on the rotor performance and wake characteristics of helicopters. The test model had a rotor diameter of 9 feet and was a 1/8th scale representation of a generalized 40,000-pound helicopter. The model rotor was scaled to be dynamically similar to a full-scale rotor when operated at half full-scale forward and tip speeds.

Test configurations included two rotor head fairings in addition to the basic rotor head. One fairing was a "cap" fairing similar to that used on the Sikorsky CH-3C helicopter. The other fairing consisted of an ellipsoidal shell which covered the rotor head, and a streamlined enclosure for the rotor shaft and control system pushrods. This streamlined enclosure incorporated blowing boundary layer control jets which were intended to prevent flow separation. The basic rotor head with and without the ellipsoidal fairing was tested with rotor blades; the cap fairing was not tested with rotor blades.

From results obtained in this test, it has been concluded that helicopter yaw angle has no significant effect upon rotor performance. Wake survey data showed that the wake of the helicopter tends to remain directly downstream of the helicopter rotor head for the pitch and yaw angles investigated. Data obtained for the ellipsoidal rotor head fairing indicated that the boundary layer control jet mass flow during this test was not sufficient to forestall separation around the fairing, and therefore no drag reduction was obtained. However, previous full-scale tests, partially reported in Reference 1, demonstrated that significant reductions in drag and wake size can be obtained with this type of fairing with sufficient boundary layer control. Data obtained for the cap fairing did not indicate the reduction in turbulence or deflection of the wake which is effective in eliminating "tail shake" in full-scale helicopters. These data showed, however, that the cap fairing increased the drag of the model rotor head by about 12 percent.

The wake survey data for the model operating with rotor blades were not regular or consistent, and no discernible trends could be derived from these data. All of the conclusions and results concerning the wake characteristics are based upon data obtained without rotor blades.

FOREWORD

This test program was sponsored by the U. S. Army Aviation Materiel Laboratories, Fort Eustis, Virginia, and was monitored by Mr. Patrick Cancro. The wind tunnel test was conducted in conjunction with the Rotor Transient and Steady State Aeroelastic Characteristics investigation which was conducted under the same contract. The work was authorized by DA Task 1F162204A13903, "Advanced Rotary Wing Research".

Mr. Edmond F. Kiely supervised the modifications to the test article and assisted in the supervision of the wind tunnel tests with Mr. L. J. Bain and Mr. C. F. Niebanck. The wind tunnel test was planned by Messrs L. J. Bain, J. P. Rabbott, and E. A. Fradenburgh.

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LIST OF SYMBOLS

A_{1s}	first harmonic lateral cyclic pitch angle with respect to the rotor shaft, positive for increasing pitch on forward blade, degrees
a_{1s}	first harmonic longitudinal flapping angle with respect to the rotor shaft, positive for forward blade up, degrees
B_{1s}	first harmonic longitudinal cyclic pitch angle with respect to the rotor shaft, positive for pitch down on advancing blade, degrees
b	number of blades
b_{1s}	first harmonic lateral flapping angle with respect to the rotor shaft, positive for advancing blade flapping down, degrees
C_D/σ	rotor drag coefficient - solidity ratio, $D/\pi R^2 \rho (\Omega R)^2 \sigma$
C_L/σ	rotor lift coefficient - solidity ratio, $L/\pi R^2 \rho (\Omega R)^2 \sigma$
C_P	pressure coefficient $(P_T - P_S)/q$
C_{PM}/σ	rotor pitching moment coefficient - solidity ratio, $PM/\pi R^3 \rho (\Omega R)^2 \sigma$
C_{RM}/σ	rotor rolling moment coefficient - solidity ratio, $RM/\pi R^3 \rho (\Omega R)^2 \sigma$
C_Q/σ	rotor torque coefficient - solidity ratio, $Q/\pi R^3 \rho (\Omega R)^2 \sigma$
C_Y/σ	rotor side force coefficient - solidity ratio, $Y/\pi R^2 \rho (\Omega R)^2 \sigma$
c	blade chord, feet
D	drag measured at rotor hub, pounds
f	parasite area, D/q , square feet
L	lift measured at the rotor hub, pounds
$M_{C.30R}$	blade chordwise bending moment measured at 30 percent radius
$M_{C.60R}$	blade chordwise bending moment measured at 60 percent radius

$M_{F.30R}$ blade flapwise bending moment measured at 30 percent radius
 $M_{F.60R}$ blade flapwise bending moment measured at 60 percent radius
 $M_{T.18R}$ blade torsional moment measured at 18 percent radius
 $M_{T.35R}$ blade torsional moment measured at 35 percent radius
 N rotor speed, rpm
 P_J boundary layer control jet total pressure, pounds per square foot
 P_S local static pressure, pounds per square foot
 P_T local total pressure, pounds per square foot
 PM pitching moment measured at the rotor hub, positive for nose up, foot-pounds
 RM rolling moment measured at the rotor hub, positive for right side down, foot-pounds
 Q rotor shaft torque, positive for powered rotor, foot-pounds
 q local dynamic pressure, $\frac{1}{2}\rho V^2$, pounds per square foot
 R rotor radius, feet
 V forward velocity, feet per second or knots, as indicated
 V_J boundary layer control jet velocity, feet per second
 V_S simulated forward speed, knots, ($V_S = 2V$)
 Y rotor hub side force in the body axis system, positive for force to right, pounds
 α_c rotor control axis angle of attack, $\alpha_s - B_{1S}$, degrees
 α_f fuselage angle of attack, called ALPHA in Appendix II tables, positive for nose up, degrees
 α_{TPP} rotor tip path plane angle of attack, $\alpha_f + a_{1S}$, degrees
 α_s rotor shaft angle of attack ($= \alpha_f$ on model), degrees
 δ_3 pitch flap coupling angle, $\tan \delta_3 = -1.0$, degrees
 ϵ upwash angle, degrees

$\bar{\epsilon}$	average upwash angle, formed by averaging the upwash angle measured by all of the flow direction meters except the topmost meter, degrees. (The upwash angle measured by the topmost meter was omitted because of symmetry.)
θ_c	rotor blade collective pitch, positive for leading edge up, degrees
μ	advance ratio, $V/\Omega R$
μ^*	effective advance ratio, the advance ratio which would have been attained if the rotor blades had been attached, $V/\Omega R$
ρ	air density, slugs per cubic foot
σ	rotor solidity, $bc/\pi R$
σ'	sidewash angle, positive for flow from left to right looking aft, degrees
ψ	model yaw angle, called PSI in Appendix II tables, positive for nose to the right, degrees
Ω	rotor angular velocity, radians per second

INTRODUCTION

In typical current helicopters, about 30 percent of the total aircraft drag can be attributed to the rotor head. Advanced high-speed rotorcraft will have improved fuselages, resembling the low-drag configurations commonly used for fixed-wing aircraft. Unless rotor heads for these rotorcraft are faired or are much cleaner in design, rotor head drag will account for much more than 30 percent of the total aircraft drag.

Reduction of rotor head drag is a complicated task. The problem is influenced by a number of variables, including the interference of the rotor, fuselage-pylon-rotor head interference, leakage from the pylon, and fuselage incidence and yaw angles. Further complications arise because both the size and location of the rotor head-pylon wake must be considered with respect to the tail surfaces of the aircraft, for these can have a strong effect on tail rotor loads, rudder and elevator effectiveness, and airframe vibrations.

A large amount of experimental work has been done to investigate the drag characteristics of helicopters and helicopter components. Some of this work is reported in References 2 through 6, which are concerned with body force measurements. No wake surveys were performed.

The objectives of this investigation were to study the effects of rotor head configuration, rotor performance, and fuselage pitch and yaw angle on the performance and wake characteristics of a complete, generalized helicopter. Several rotor head configurations were tested with and without rotor blades at various operating conditions. Fuselage yaw angle was varied for all conditions tested. The flow direction and dynamic pressure at the tail of the model were measured, as were rotor shaft forces and moments. When the model was tested with the rotor, blade vibratory stresses were also measured.

DESCRIPTION OF THE MODEL

The model used in this test was the Sikorsky Aircraft 1/8th scale generalized helicopter model, designed to investigate the aerodynamic characteristics of advanced rotorcraft. For this test, the model consisted of a fuselage which enclosed the rotor drive system and control system and a wake survey rake which was attached to the aft end of the model. A photograph of the model with the wake survey rake and an ellipsoidal rotor head fairing is presented in Figure 1. A more detailed description of the test article follows. The wind tunnel, experimental procedure, data acquisition system, and data accuracy and repeatability are described in Appendix I.

Rotor System

The model rotor was a four-bladed fully articulated system with a radius of 54 inches and a blade chord of 4.24 inches. The resulting solidity was 0.1. The coincident flap-lag hinges were offset 5.6 percent of the rotor radius, and the pitch-flap coupling ratio, $\tan \delta_3$, was -1.0 degree pitch per degree of flap. The rotor blades were fiber glass replicas of typical full-scale construction, similar to those described in Reference 7, and were used previously in the wind tunnel tests described in Reference 8. Blade stiffnesses were scaled so that the model rotor was dynamically similar to a full-scale rotor operating at twice the model forward and rotor tip speeds. Table I lists the model operating conditions, including the actual and simulated forward speeds. A schematic drawing of a rotor blade showing the strain gage locations is presented in Figure 2.

The rotor was powered by a 19-horsepower variable-frequency electric motor, and the control system was hydraulically actuated. All of the rotor system components, including the control actuators, hydraulic pump, rotor gearbox, and the 19-horsepower electric motor were mounted inside the model on a separate frame, which was supported by a six-component force measuring balance. The electric power lines, instrumentation wires, and control system wires were routed so as to minimize interference with the balance readings. The balance was of the floating frame, internal strain gage type and was temperature compensated. The slight interactions between force components were accounted for in the data reduction program.

Fuselage

The fuselage is shown in Figures 1 and 3. The model was constructed around a steel frame and was covered with a smooth fiber glass shell fuselage. The fuselage was 9 feet long and had a projected frontal area of 1.25 square feet, not including the rotor pylon.

Model Configurations

The model was tested without rotor blades in the four configurations shown in Figure 3, and with rotor blades in the two configurations labeled FHB and FHBf. Testing with and without rotor blades was conducted at the

same forward speed-rotor speed combinations which are listed in Table I.

The simplest configuration tested without rotor blades was the fuselage with no rotor head, but with the rotor shaft protruding from the fuselage. The model was also tested without rotor blades with the bare rotor head and with two rotor head fairings.

The simplest fairing was a "cap" fairing, Figure 4, similar to the fairing mounted on Sikorsky H-3 series helicopters. This fairing reduces helicopter "tail shake", which is caused by the interaction of the wake of the rotor head with the tail and possibly the tail rotor of the helicopter. The fairing is a dome-shaped cover that fits over the rotor head.

The second rotor head fairing, Figure 5, consisted of an ellipsoidal fiber glass shell covering the rotor head, a cylinder enclosing the rotor shaft and control pushrods, and a wedge-shaped afterbody which gave the cylinder a sharp trailing edge. The ellipsoidal shell was mounted to the rotor blades outboard of the flap-lag hinges so that it would follow the largest blade motions-coning, average lag, and first harmonic flapping. This minimized the size of the cutout holes necessary to accommodate the motions of the rotor blades. The cylinder enclosing the rotor shaft and pushrods, was mounted to the fuselage shell and was provided with air ejection boundary layer control slots located at the point of maximum thickness. It should be noted that because the cylinder was attached to the fuselage, rotor force measurements taken with this configuration do not include forces on the cylinder. The ellipsoidal shell rotated around the cylinder, and the sliding joint was sealed with felt. A wedge-shaped afterbody was mounted on the aft end of the cylinder and was segmented and spring loaded to stay in contact with the ellipsoidal shell. A felt seal was used between the wedge-shaped afterbody and the ellipsoidal shell. In operation, the boundary layer control jets were intended to prevent flow separation over the wedge-shaped afterbody. This technique has been used successfully on a full-scale model and will be discussed further under "Discussion of Results".

The model was tested with rotor blades attached to the basic rotor head and to the rotor head with ellipsoidal fairing. As mentioned before, the model with the cap fairing was not tested with rotor blades. The test configurations are summarized in Table II.

Wake Survey Rake

The wake survey rake measured total head pressures at 112 stations and flow directions at 9 stations in a 9-square-foot area. Figure 6 shows the wake survey rake with the pressure probe locations. The total head pressure probes were of the shrouded, or Kiel type. The flow direction meters were United Sensor five-holed probes.

Before the test program was conducted, the wake survey rake was calibrated in the 4-by-6-foot Pilot Wind Tunnel at the United Aircraft Research Laboratories. The shrouded total pressure taps were found to be insensitive to incoming flow directions of up to ± 25 degrees.

The pressures to be surveyed were monitored by four 48-port scanning type pressure valves. Each pressure valve was equipped with a 2.5-psi differential pressure transducer. The entire pressure valve module was mounted inside the rear model extension that supported the wake survey rake.

PRESENTATION OF DATA

Wake Survey Data

All of the wake survey data obtained in the test, including the data obtained with rotor blades, are presented in tabular form in Appendix II. These tables list the pressure coefficient measured by each of the 112 total pressure probes and the flow direction measured by each of the 9 flow direction meters. The numbers are arranged so that their locations in the table correspond to the locations of the measuring probes given in Figure 6.

Flow directions are indicated at nine locations by the values within boxes. The topmost number in each box is vertical flow displacement in degrees (positive for upwash), and the bottom number is horizontal flow displacement in degrees (positive for flow from left to right looking aft). These flow directions are measured with respect to the wake survey rake which is rigidly attached to the tail of the model. Therefore, to obtain upwash or sidewash angles with respect to the wind tunnel axis, the fuselage pitch and yaw angles, given in each table, must be removed from the tabulated flow direction:

$$\text{Upwash} = \epsilon = \text{Vertical flow displacement} - \alpha_f.$$

$$\text{Sidewash} = \sigma' = \text{Horizontal flow displacement} + \psi.$$

Total pressures are listed in the table as pressure coefficients, defined as $C_p = (P_T - P_S)/q$. Pressure coefficient is related to fluid speed as shown in Figure 7. The dynamic pressures and velocities given at the top of each table are the actual values obtained in the wind tunnel, not the simulated values.

A selected portion of the wake survey data obtained without rotor blades is presented in graphical form to permit visualization of the wake and to illustrate the effects observed in this test. Figure 8 is an example of this. These wake survey figures show, within the bounds of the wake survey rake, lines of constant pressure coefficient and arrows which indicate upwash and sidewash as seen by an observer looking downstream.

Flow direction arrows are formed by the vector addition of upwash and sidewash arrows. The "tail" of each arrow is located at the measurement point. The scale of these arrows, shown on each figure, is degrees, which for the small angles being dealt with is proportional to velocity. In some cases, the measured angles were unusually large, so that flow direction arrows would go off the figure. In these cases, the upwash and sidewash angles are printed adjacent to the measurement point.

Rotor Performance Data

The rotor performance data obtained in this test are presented in tabular form in Appendix III. Each table in this appendix presents all of the data obtained for a particular configuration. The tables include the

wind tunnel operating velocity, wind tunnel dynamic pressure, model attitude, rotor cyclic pitch, collective pitch, and first harmonic longitudinal and lateral flapping components, as well as rotor shaft force and moment coefficient-solidity ratios. When the rotor blades were not attached to the model, the force and moment coefficient-solidity ratios were computed as if the blades were present. This was done to facilitate comparison of the data. To obtain forces and moments from these data, the following formulae can be used:

$$L = 12.72 (q/\mu^2) C_L/\sigma, \text{ pounds}$$

$$D = 12.72 (q/\mu^2) C_D/\sigma, \text{ pounds}$$

$$Y = 12.72 (q/\mu^2) C_Y/\sigma, \text{ pounds}$$

$$Q = 57.26 (q/\mu^2) C_Q/\sigma, \text{ foot-pounds}$$

$$PM = 57.26 (q/\mu^2) C_{PM}/\sigma, \text{ foot-pounds}$$

$$RM = 57.26 (q/\mu^2) C_{RM}/\sigma, \text{ foot-pounds}$$

Rotor Blade Moment and Flapping Data

The rotor blade moment and flapping data are presented in tabular form in Appendix IV. These data are in the form of maximum and minimum bending moment or flapping values measured during a particular test condition. The blades were dynamically scaled so that a flapwise moment of 60 inch-pounds, a chordwise moment of 125 inch-pounds, and a torsional moment of 85 inch-pounds would each correspond to a 10,000-psi stress in an equivalent aluminum blade. The data are not presented in graphical form because no variation in rotor blade stress or flapping with fuselage yaw angle was observed, and because the data agree with similar data obtained in Reference 8.

DISCUSSION OF RESULTS

The Effect of Fuselage Pitch on the Wake Characteristics

As fuselage pitch angle increases, the position of the wake remains relatively constant with respect to the streamwise projected fuselage profile, i.e., the wake trails directly downstream. This is shown in Figure 8 for the model with the basic rotor head attached but without rotor blades (configuration FH) at a simulated forward speed of 200 knots and at values of α_f between -8 and $+8$ degrees. The wake did, however, move up with respect to the tail of the helicopter. No consistent effects of fuselage pitch angle on the shape of the wake were observed. It is believed that the nonsteady nature of the wake makes an accurate definition of the average wake impossible using the data obtained in this test which represent the instantaneous pressures in the wake.

The Effect of Fuselage Yaw on the Wake Characteristics

Yawing the model in either direction causes the wake to shift with respect to the tail so as to remain behind the streamwise projected profile. This is shown in Figure 9 for the model with the basic rotor head (configuration FH) at a simulated forward speed of 200 knots and at values of ψ between -8 and $+8$ degrees. Examination of the flow directions measured by the upper row of flow direction probes indicates that the downwash is greater on the downstream side of the model than on the undisturbed side. Again, no consistent effects on the shape of the wake were observed.

The Effect of Rotor Head Configuration on the Wake

Adding the basic rotor head to the model increases the size of the wake compared to the bare shaft (configuration F) case. Adding the cap fairing to the basic rotor head (configuration FH) further increases the wake size (this will be discussed in more detail in the next section). However, no change in the shape or the location of the wake was noticed which would indicate why the cap fairing has been effective in flight operation in reducing tail shake. A better understanding of the mechanism of this improvement might be obtained from examination of dynamic measurements of the turbulence in the wake. Adding the ellipsoidal fairing to the configuration FH causes the wake to increase in size slightly. These effects are shown in Figure 10 for the model in an unpitched, unyawed attitude, and in Figure 11 for the model in an attitude which is typical of normal helicopter flight, i.e. nose down and to the left (negative α_f , negative ψ). Data for these figures were taken at a simulated forward speed of 200 knots.

The Effect of Rotor Head Configuration and Fuselage Angle of Attack on Rotor Head Drag

The parasite drag of the rotor head increases slightly with increasing fuselage angle of attack. This is illustrated in Figure 12 for a simulated forward speed of 200 knots. This figure also shows that the addition of

the cap fairing increases the drag of the basic rotor head by about 0.03 square foot (12 percent), and that the addition of the ellipsoidal fairing increases the drag of the basic rotor head by about 0.02 square foot (8 percent).

The drag measurements for the ellipsoidal fairing did not include the drag of the cylinder enclosing the rotor shaft and pushrods (this was discussed in the description of the model configurations), and none of the drag measurements included possible interference drag effects. In order to provide an indication of the total drag, the area enclosed by the $C_p = 0.9$ contour was measured for all of the conditions shown in Figures 10 and 11. Fluid passing through a $C_p = 0.9$ contour has lost momentum; therefore, a large enclosed area connotes a large drag. These areas are presented in Table III in square feet and as a percentage of the area for the basic rotor head. The area within the $C_p = 0.9$ contour is approximately proportional to the total drag of the rotor head and fuselage and includes any interferences between these components. The results show the same trends that are shown in Figure 12: the ellipsoidal fairing (configuration FHf) has less drag than the cap fairing (configuration FHC), and, for zero fuselage angle of attack, both configurations have higher drag than the basic rotor head. The small area enclosed by the $C_p = 0.9$ contour for configuration FHf at -4 degrees fuselage angle of attack is not reflected by a corresponding reduction in the drag measured at the same attitude. This is probably due to a reduction in the interference drag between the pylon and the ellipsoidal fairing, which would not be reflected in the drag measured at the rotor head but which would decrease the momentum loss in the wake and therefore decrease the area enclosed by the $C_p = 0.9$ contour. No attempt has been made to integrate the wake survey data to obtain total drag because of the unknown effects of the model mounting struts, and because the wake is not fully enclosed by the wake survey rake.

The Effect of Boundary Layer Control Jet Pressure Ratio on the Drag and Wake Characteristics of a Full-Scale Ellipsoidal Rotor Head Fairing

As previously mentioned, the model boundary layer control jet pressure ratio was not sufficient to prevent flow separation over the wedge-shaped afterbody. This was determined from observing tufts attached to the cylinder and afterbody. Large pressure losses in the ducts connecting the high-pressure source to the boundary layer control system caused this inadequacy. However, full-scale tests of an ellipsoidal rotor head fairing mounted on a modified SH-3A pylon, Figure 13, were conducted by Sikorsky Aircraft in 1960 in the United Aircraft Corporation 18-foot Main Wind Tunnel. The test model is described in more detail in Reference 1. Wake survey data were taken during this test using a wake survey rake located 38 feet downstream of the center of the rotor head, which is the approximate distance between the rotor head and the tail rotor of the SH-3A. Since this test was full scale, the actual tunnel speed V , and not a simulated forward speed, will be referred to in the discussions. The results presented here were obtained at a forward speed of 105 knots, an effective advance ratio, μ^* , of 0.27, and at zero angle of attack.

This full scale test demonstrated that significant reductions in drag and wake size could be obtained with sufficient boundary layer control jet pressure. Figure 14 presents some of the wake survey data that were obtained in this test. The size of the wake decreased with increasing boundary layer control jet pressure ratio, defined as $(P_J - P_S)/q$, until a jet pressure ratio of about 7.7 was attained. Beyond this point, increasing jet pressure ratio caused no reduction in wake size, indicating that separation had been minimized.

The parasite area, f , of the full-scale model is plotted against boundary layer control jet pressure ratio in Figure 15. Due to jet reaction, the measured drag of the full-scale model decreased as the boundary layer control jet pressure ratio was increased beyond 7.7. This is shown by the lower curve of Figure 15. There was, however, an optimum operating jet pressure ratio where the total equivalent drag, which includes form drag and the drag equivalent of the power required to operate the boundary layer control jets (assuming 75 percent efficient pumps), was a minimum. The upper curve of Figure 15 shows that this optimum point was reached at a jet pressure ratio of about 7.7.

The Effect of Fuselage Angle of Attack on Average Upwash

Increasing fuselage angle of attack has no discrete effect on the average upwash. Figure 16 presents the average upwash versus fuselage angle of attack for each of the four basic configurations tested at various yaw angles. Included in this figure are comparative data obtained from a floating tail flow direction meter which was tested previously on the model, Reference 8. The upwash angle remains essentially constant at minus one degree for all configurations except the ellipsoidal fairing. Upwash increased with angle of attack for this configuration. The rotor head lift data for this configuration do not indicate any significant decrease in lift with angle of attack which might explain this.

The Effect of Fuselage Yaw on Rotor Performance

The effect of fuselage yaw on rotor performance for three forward speed-advance ratio combinations is shown in Figures 17, 18, and 19. Included in Figures 17 and 18 are comparative data obtained from Reference 8 shown as dashed lines. Drag and torque data for the 250-knot simulated forward speed case are not presented in Figure 19 because very little variation in these parameters was observed. Rotor lift, drag, and torque were found to be essentially unaffected by fuselage yaw.

CONCLUSIONS

As a result of the tests described in this report, the following conclusions have been drawn:

1. Helicopter fuselage yaw angle has no significant effect upon rotor lift, drag, or torque characteristics for the configuration tested.
2. The wake survey pressure data obtained for the model with rotor blades was not regular or consistent, and no discernable trends have been observed in these data. The following conclusions are based on wake survey data obtained without rotor blades.
3. The wake of the helicopter tends to remain directly downstream of the helicopter rotor head for the angles of pitch and yaw that were investigated (i.e., angles between +8 and -8 degrees).
4. The ellipsoidal rotor head fairing used in this test resulted in an 8 percent increase in rotor head drag over the basic rotor head and increased the area of the wake slightly. However, this type of fairing, when used with an effective blowing boundary layer control system, may provide significant reductions in drag and wake size.
5. The "cap" rotor head fairing used in this test resulted in a 12 percent increase in rotor head drag over the basic rotor head and increased the size of the wake slightly. The data obtained in this test gave no indication of why this fairing is effective in eliminating tail shake.

LITERATURE CITED

1. Fradenburgh, E. A., HIGH PERFORMANCE SINGLE ROTOR HELICOPTER STUDY, TCREC Technical Report 61-44, U. S. Army Transportation Research Command, Fort Eustis, Virginia, April 1961.
2. Harrington, R. D., REDUCTION OF HELICOPTER PARASITE DRAG, NACA Technical Note 3234, Langley Research Center, Langley Field, Virginia, August 1954.
3. Churchill, G. B., and Harrington, R. D., PARASITE-DRAG MEASUREMENTS OF FIVE HELICOPTER ROTOR HUBS, NASA Memo 1-31-59L, Langley Research Center, Langley Field, Virginia, February 1959.
4. Sweet, George E., and Jenkins, Julian L., WIND TUNNEL INVESTIGATION OF THE DRAG AND STATIC STABILITY CHARACTERISTICS OF FOUR HELICOPTER FUSELAGE MODELS, NASA Technical Note D-1363, Langley Research Center, Langley Field, Virginia, July 1962.
5. Pruyn, R. R., and Miller, N. J., STUDIES OF ROTORCRAFT AERODYNAMIC PROBLEMS AIMED AT REDUCING PARASITE DRAG, ROTOR-AIRFRAME INTERFERENCE EFFECTS, AND IMPROVING AIRFRAME STATIC STABILITY, Kellett Aircraft Corporation, 1960.
6. Moser, H. H., FULL SCALE INVESTIGATION OF HELICOPTER DRAG, Journal of the American Helicopter Society, Vol. 6 No. 1, January 1961, pp 27-33.
7. Fradenburgh, E. A., and Kiely E. F., DEVELOPMENT OF DYNAMIC MODEL ROTOR BLADES FOR HIGH SPEED HELICOPTER RESEARCH, Paper presented at the Symposium on Aeroelastic and Dynamic Modeling Technology, Wright-Patterson Air Force Base, Ohio, Sept. 23-25, 1963.
8. Bain, Lawrence J., and Landgrebe, Anton J., INVESTIGATION OF COMPOUND HELICOPTER AERODYNAMIC INTERFERENCE EFFECTS, Sikorsky Aircraft, Division of United Aircraft Corporation; USAAVLABS Technical Report 67-44, U. S. Army Aviation Materiel Laboratories, Fort Eustis, Virginia, November 1967, AD 665427.
9. Pope, Allan, WIND TUNNEL TESTING, John Wiley and Sons, New York, 1954.
10. Heysen, Harry H., LINEARIZED THEORY OF WIND TUNNEL JET BOUNDARY CORRECTION AND GROUND EFFECT FOR VTOL-STOL AIRCRAFT, NASA Technical Report R-124, National Aeronautics and Space Administration, Langley Field, Virginia, January 1962.

APPENDIX I
DESCRIPTION OF THE WIND TUNNEL, EXPERIMENTAL PROCEDURE,
DATA ACQUISITION SYSTEM, AND DATA ACCURACY AND REPEATABILITY

Wind Tunnel

The 18-foot Main Wind Tunnel at the United Aircraft Research Laboratories is a closed-throat, single-return wind tunnel capable of speeds up to approximately 175 knots. An artist's sketch of this tunnel is shown in Figure 20. The test section is octagonal in cross section. The tunnel stagnation temperature is held approximately constant by variable-opening air exchangers. Stagnation pressure is atmospheric and is constant throughout the test section in the absence of externally caused momentum losses.

All tunnel controls and data acquisition equipment are located in a control room adjacent to the tunnel test section. Windows permit constant observation of the model from the control room.

Experimental Procedure

At the beginning of each run, the rotor shaft speed and the wind tunnel speed were set to one of the shaft speed-tunnel speed combinations listed in Table I. This was done whether or not the model was being tested with blades to include any possible effects due to rotor head rotation. Then the fuselage pitch and yaw angles were varied. When the rotor blades were attached, the rotor controls were set to produce zero first harmonic lateral flapping and the desired first harmonic longitudinal flapping.

The wake survey data and the rotor performance data were obtained simultaneously whenever possible. However, due to data system malfunctions, some of the runs had to be repeated to obtain the rotor performance data. Table IV summarizes the data runs.

Data Acquisition System

The rotor balance data were recorded manually from Baldwin SR-4 Precision Indicators (Type L-5D). The data were then processed and tabulated by a UNIVAC 1108 digital computer. All rotor balance data have been corrected for force variations due to gravity, and corrections have been applied to the data to compensate for wind tunnel blockage and wall effects, using the methods of References 9 and 10. The blockage correction to velocity was less than 2 percent of the wind tunnel velocity; the wall correction to angle of attack was less than 0.5 degree. All rotor forces and moments taken with the rotor blades are presented with the rotor head lift and drag forces removed.

The pressure data from the wake survey rake were transmitted directly to the United Aircraft Research Laboratories' Static Data Acquisition System (STADAS II), which punched the data on paper tape. Model attitude, dynamic pressure, total head pressure, and various other constants were

entered into the STADAS II facility through a manual input board. The paper tape from STADAS II was converted to magnetic tape, which was then reduced utilizing a UNIVAC 1108.

The rotor blade stress and flapping motion data were recorded by an Ampex AR 200 F.M. magnetic tape recorder. These data were transferred to an oscillograph record, from which the maximum and minimum moment and flapping values were obtained. These moment and flapping data are presented in Appendix IV.

Data Accuracy and Repeatability

The repeatability of the rotor force data is indicated in Figure 21, which presents a comparison of data obtained in this test and in the similar tests described in Reference 8. The rotor force data are estimated to be accurate to within 5 percent of the maximum reading.

As mentioned before, the wake survey data were not consistent when the model was equipped with rotor blades. It is believed that either the nonsteady nature of the wake or vibration-induced errors in the pressure transducers caused the lack of consistency in the data. The estimated accuracy of the wake survey data obtained when the model was operating without rotor blades is as follows:

Error in Pressure Coefficient Expressed as a Percentage of Maximum Reading

Simulated Forward Speed, V_s	Error
120 knots	\pm 9.0 pct
200 knots	\pm 3.5 pct
250 knots	\pm 2.0 pct
300 knots	\pm 1.5 pct

Error in the Horizontal and Vertical Flow Displacements

Simulated Forward Speed, V_s	Error
120 knots	\pm 1.5 deg
200 knots	\pm 0.7 deg
250 knots	\pm 0.4 deg
300 knots	\pm 0.3 deg

The rotor blade moment data are estimated to be accurate to within 2 inch-pounds. The rotor blade flapping motion data are estimated to be accurate to within 0.2 degree.

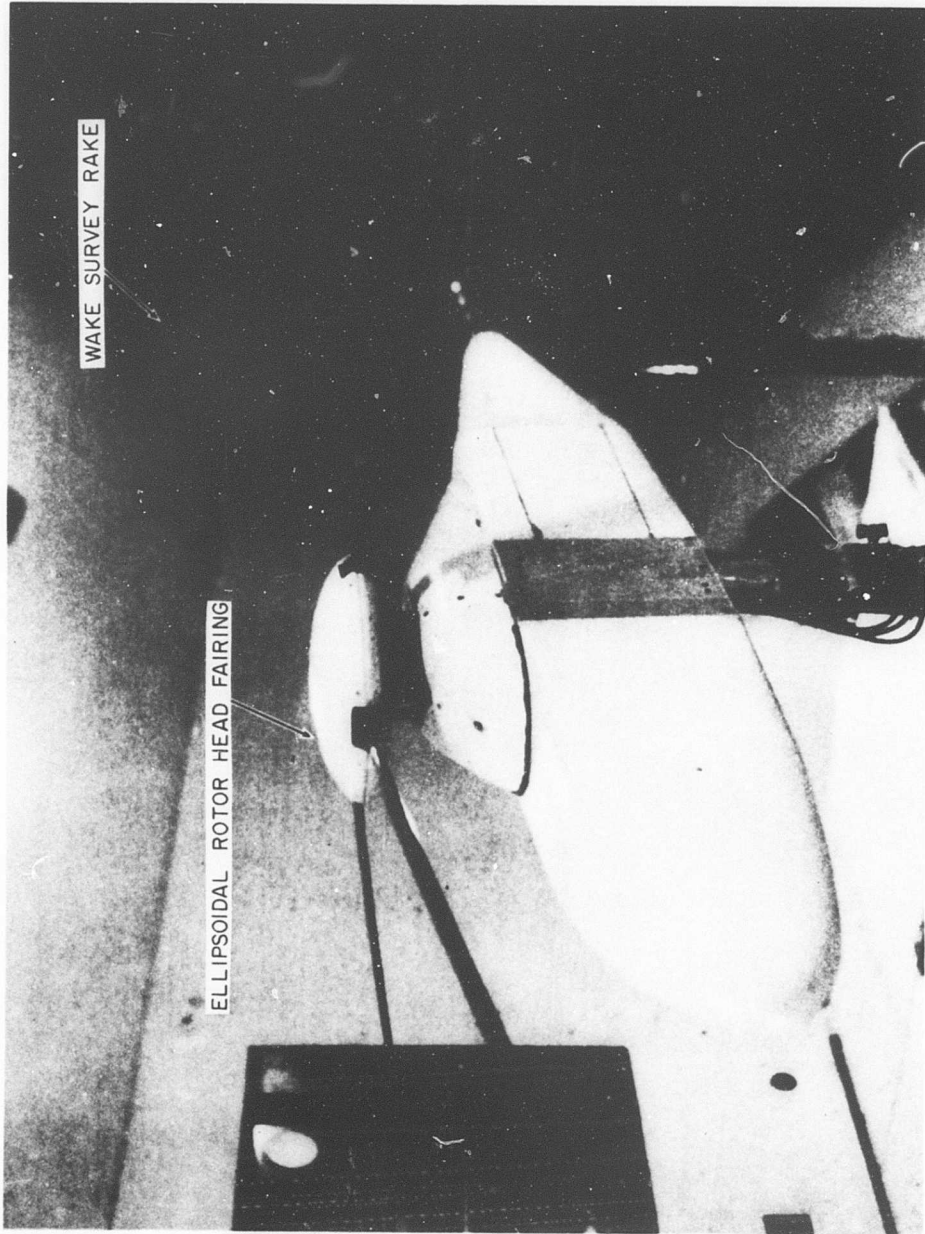


Figure 1. Sikorsky Aircraft Generalized Helicopter Model With Wake Survey Rake Installed.

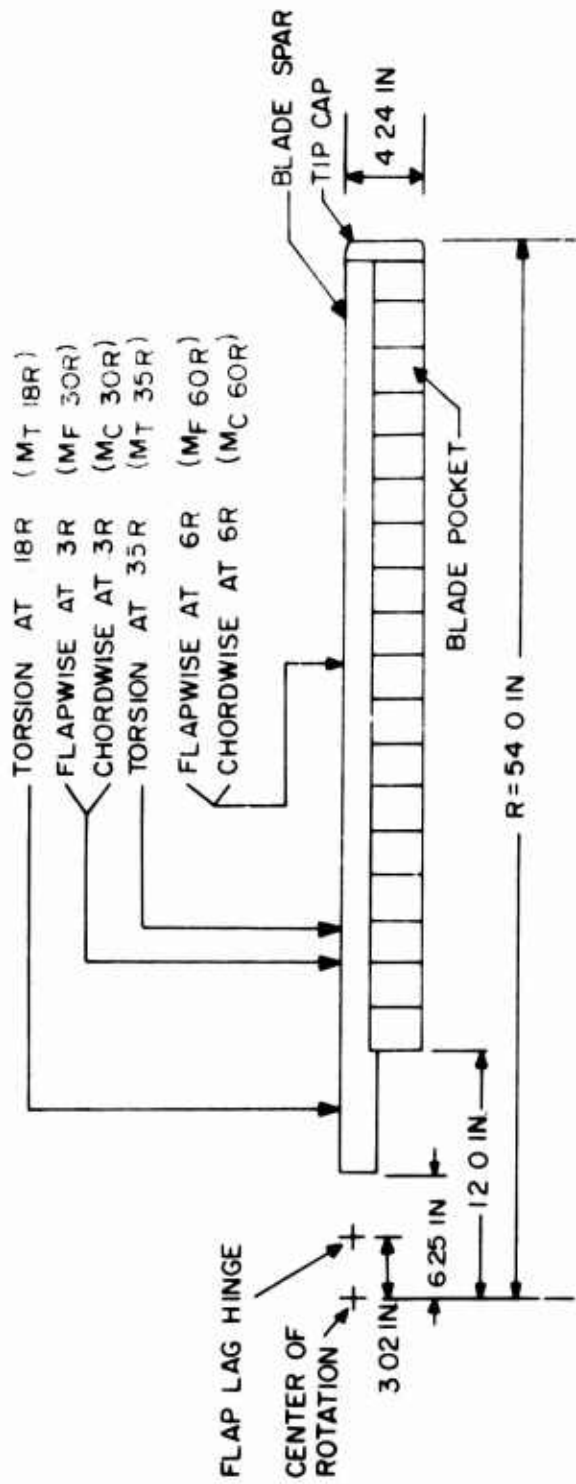


Figure 2. Rotor Blade Strain Gage Locations.

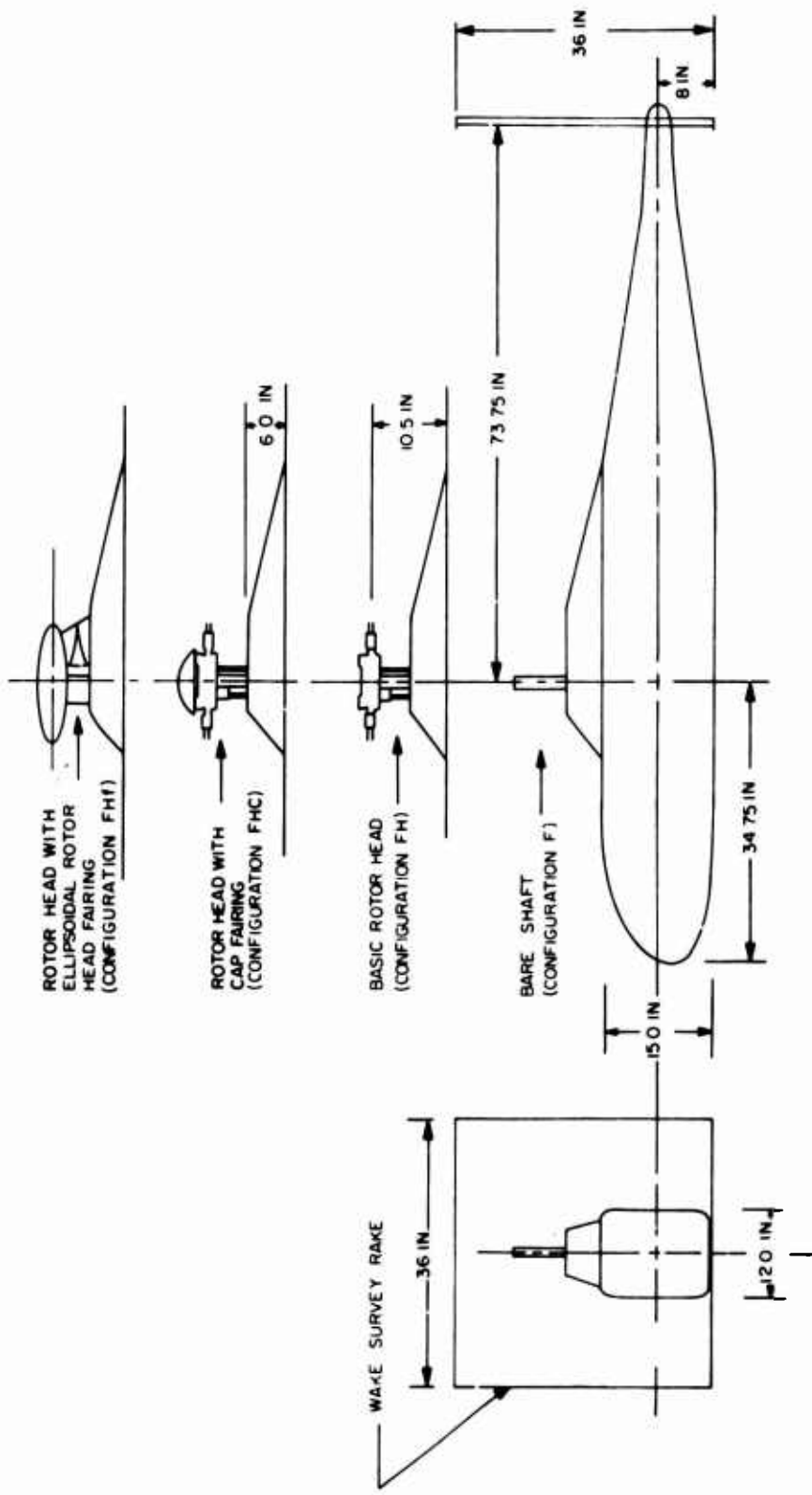


Figure 3. Configurations Tested.

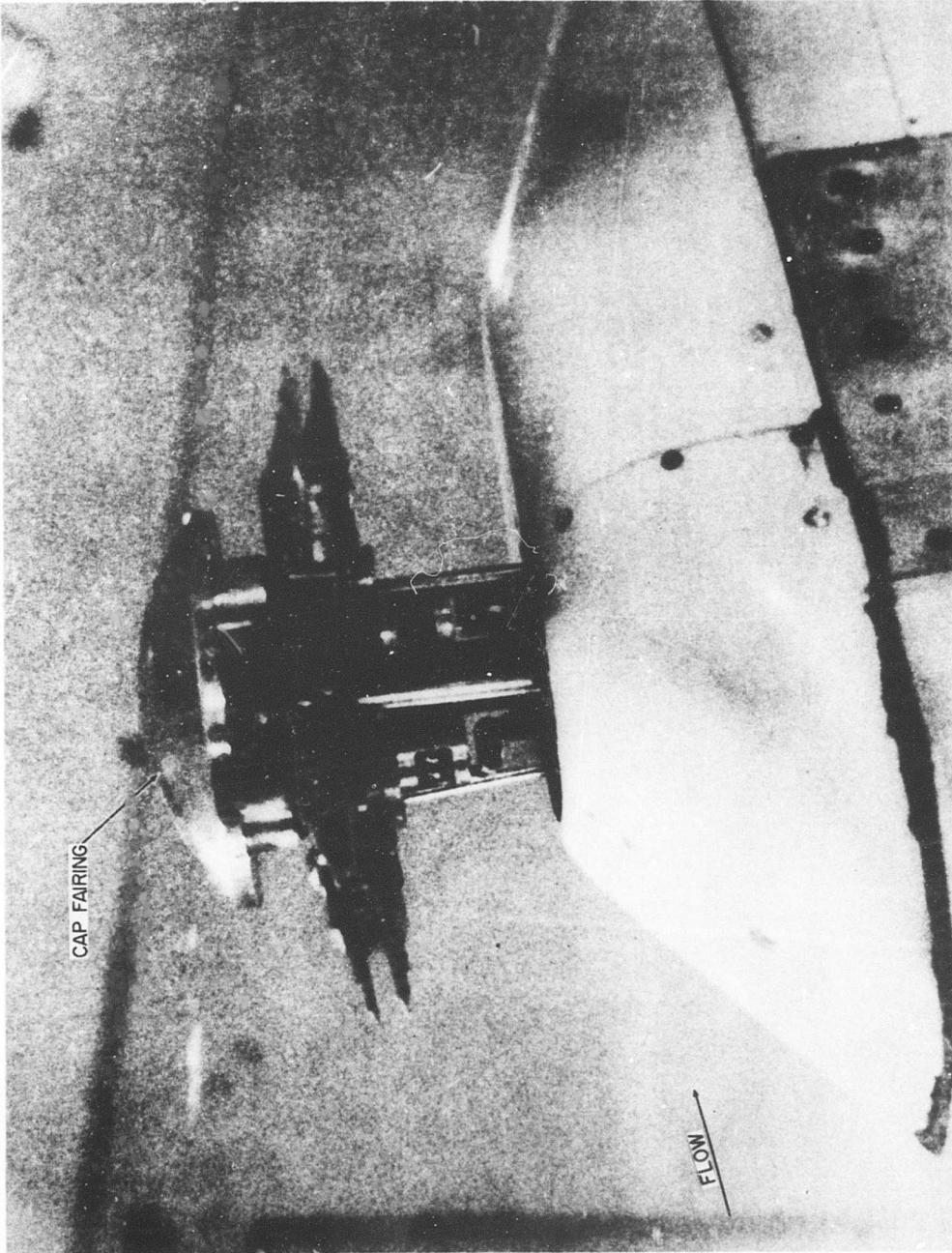


Figure 4. Rotor Head With the Cap Fairing (Configuration FHC).

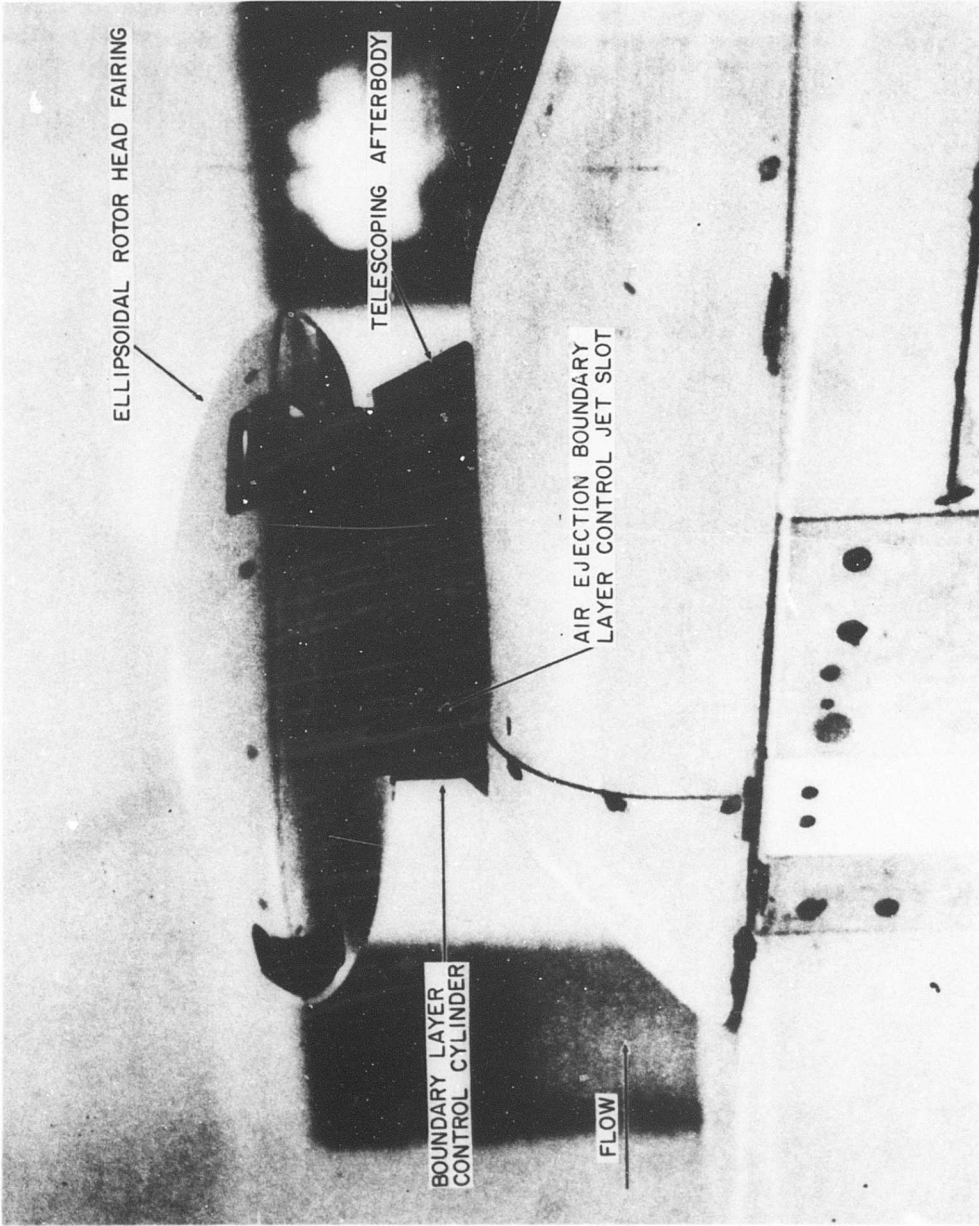


Figure 5. Rotor Head With the Ellipsoidal Fairing (Configuration FHf).

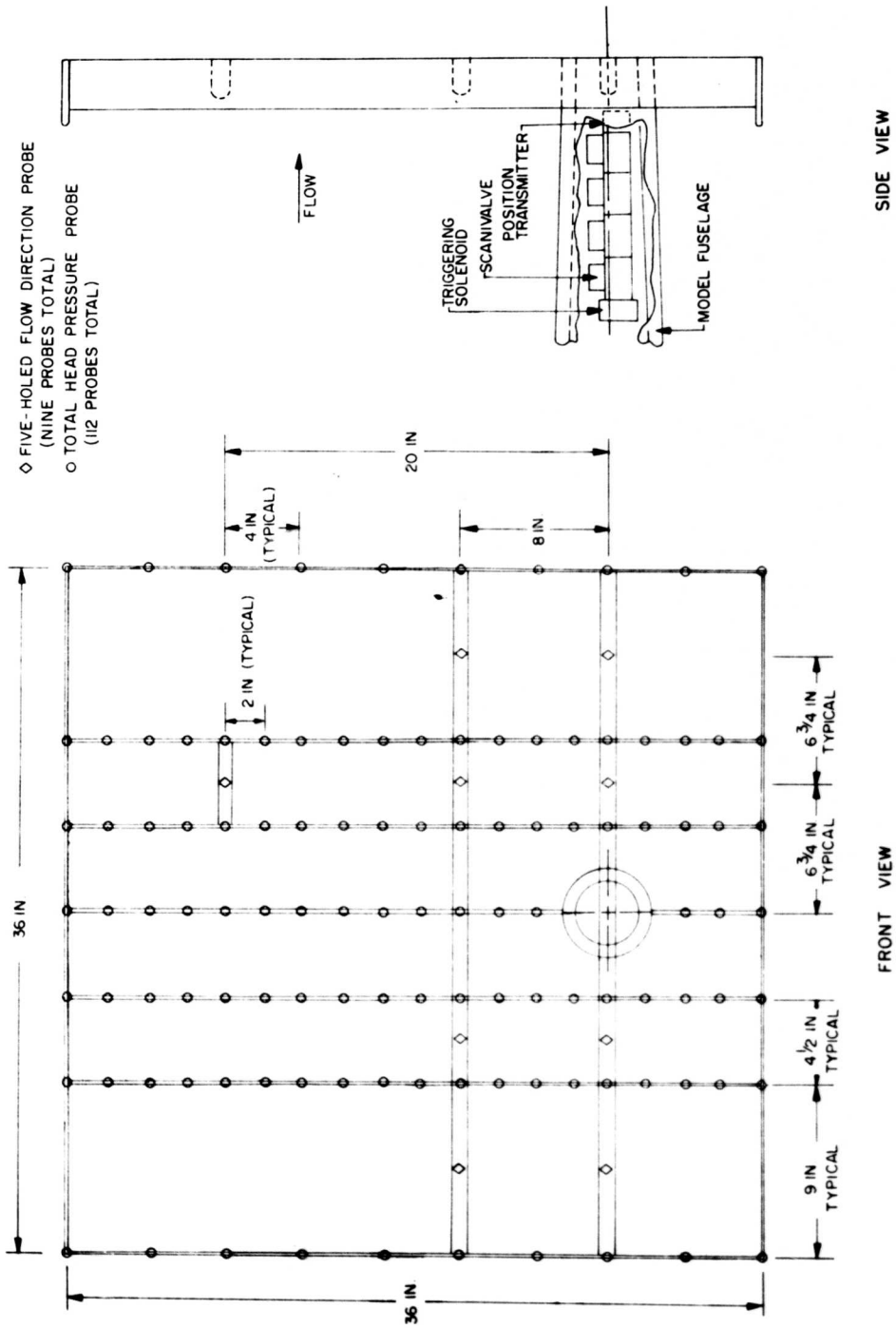


Figure 6. Schematic Drawing of the Wake Survey Rake.

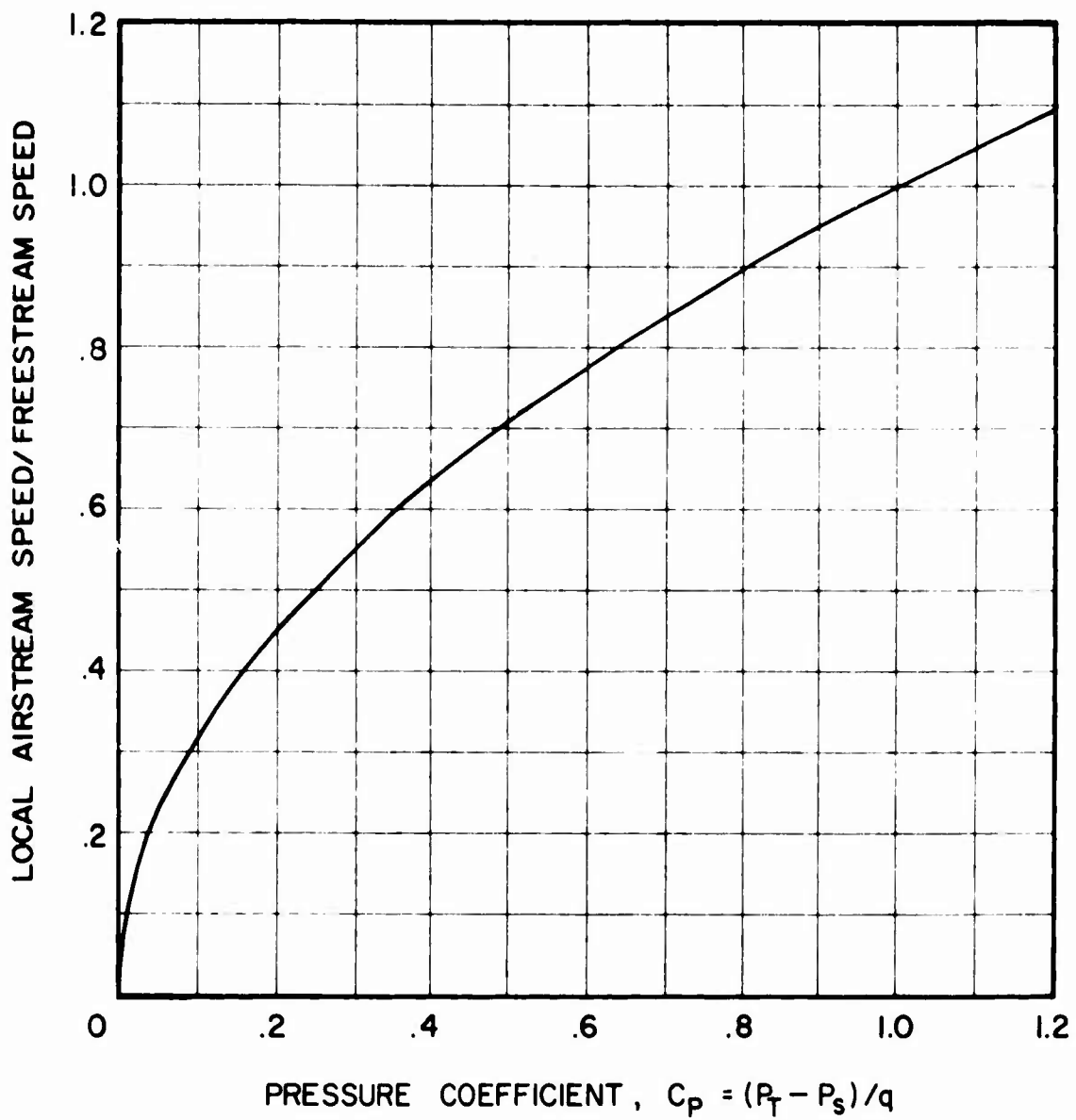
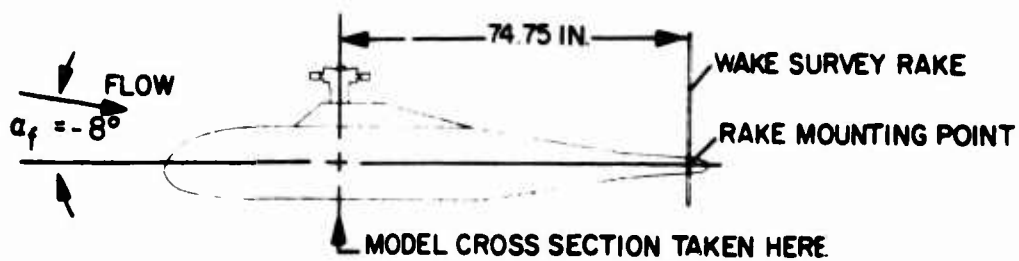
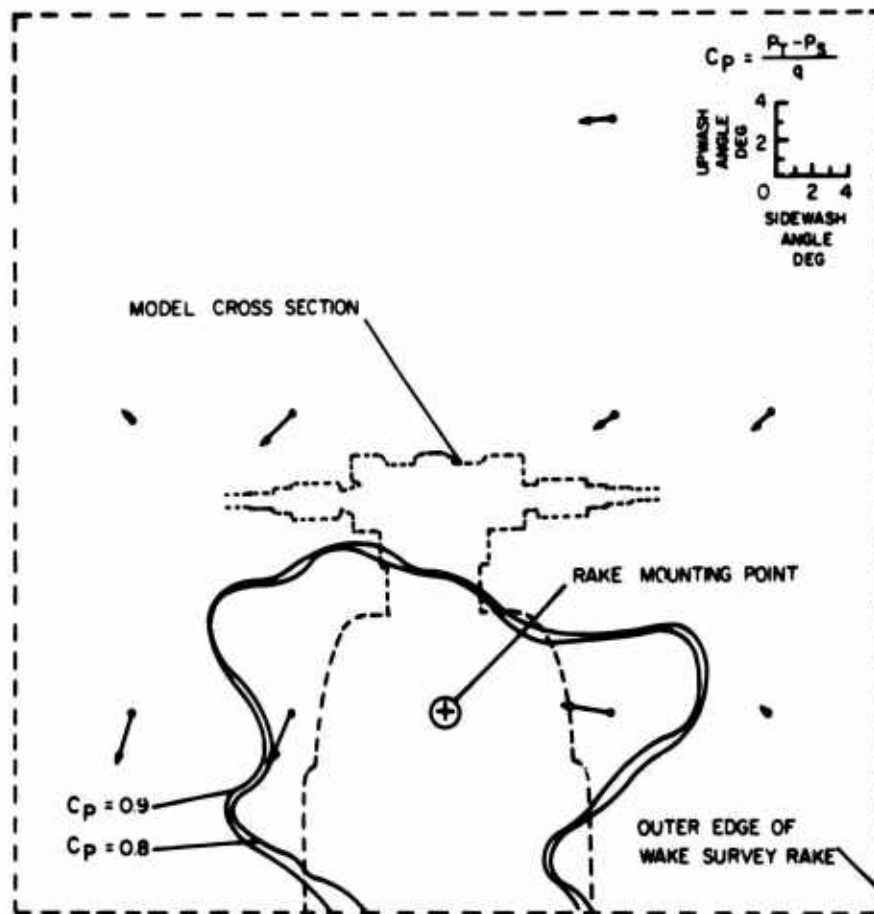
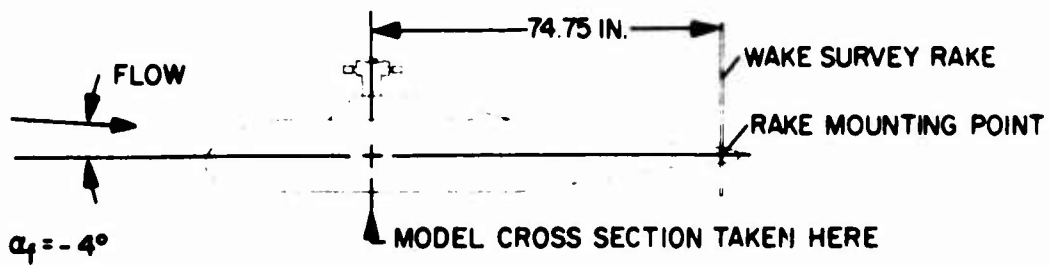
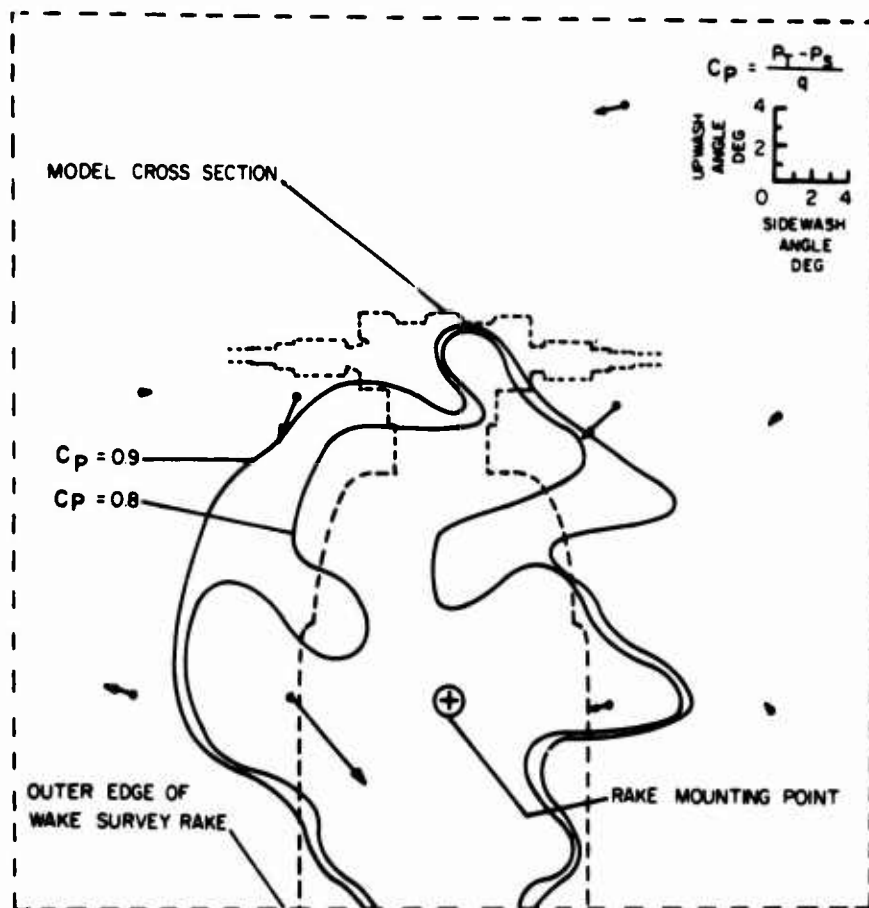


Figure 7. Local Airstream Speed Ratio Versus Pressure Coefficient.



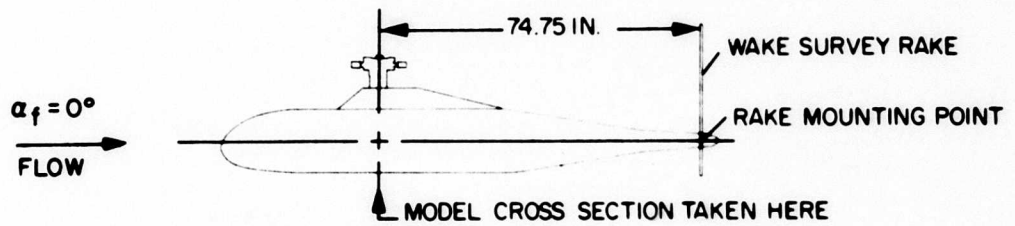
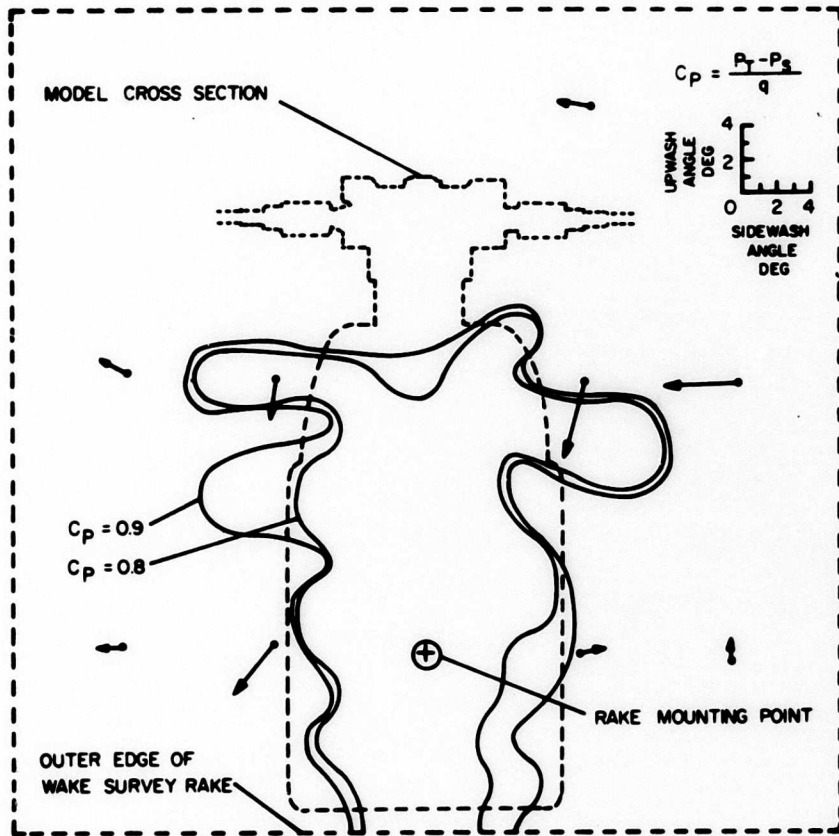
(a) $\alpha_f = -8^\circ$

Figure 8. The Effect of Fuselage Angle of Attack on the Wake of the Helicopter Model, $V_s = 200$ Knots, $\psi = 0^\circ$, $\mu^* = 0.5$, Configuration FH.



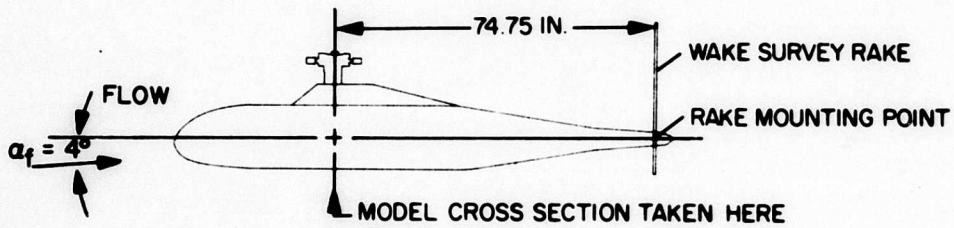
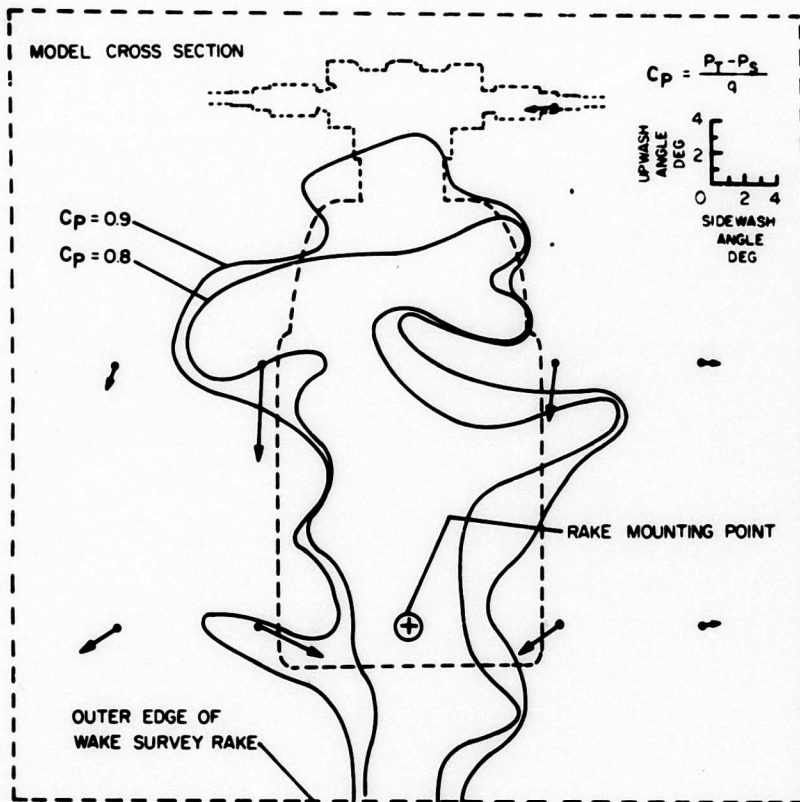
(b) $\alpha_f = -4^\circ$

Figure 8. Continued



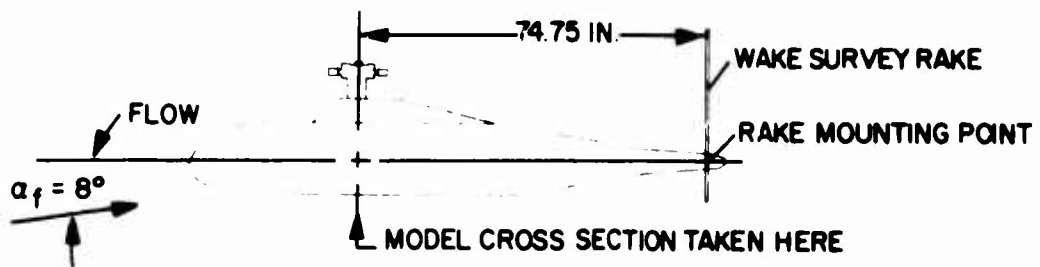
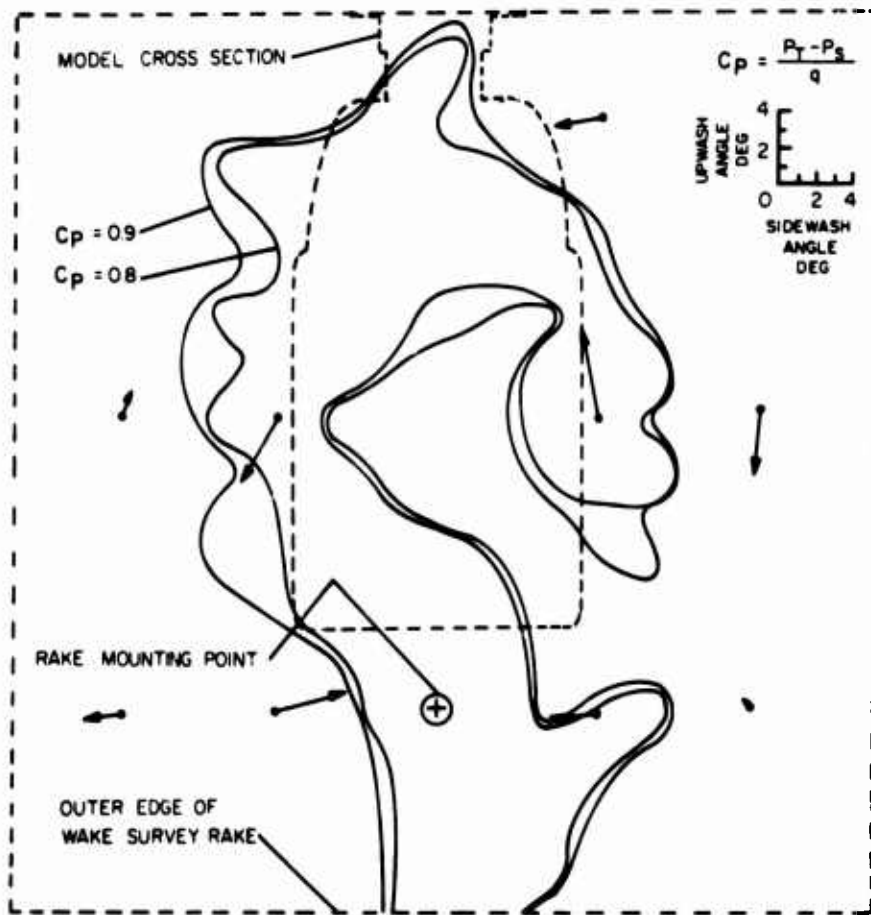
(c) $\alpha_f = 0^\circ$

Figure 8. Continued



(d) $\alpha_f = 4^\circ$

Figure 8 Continued.



(e) $\alpha_f = 8^\circ$

Figure 8. Concluded

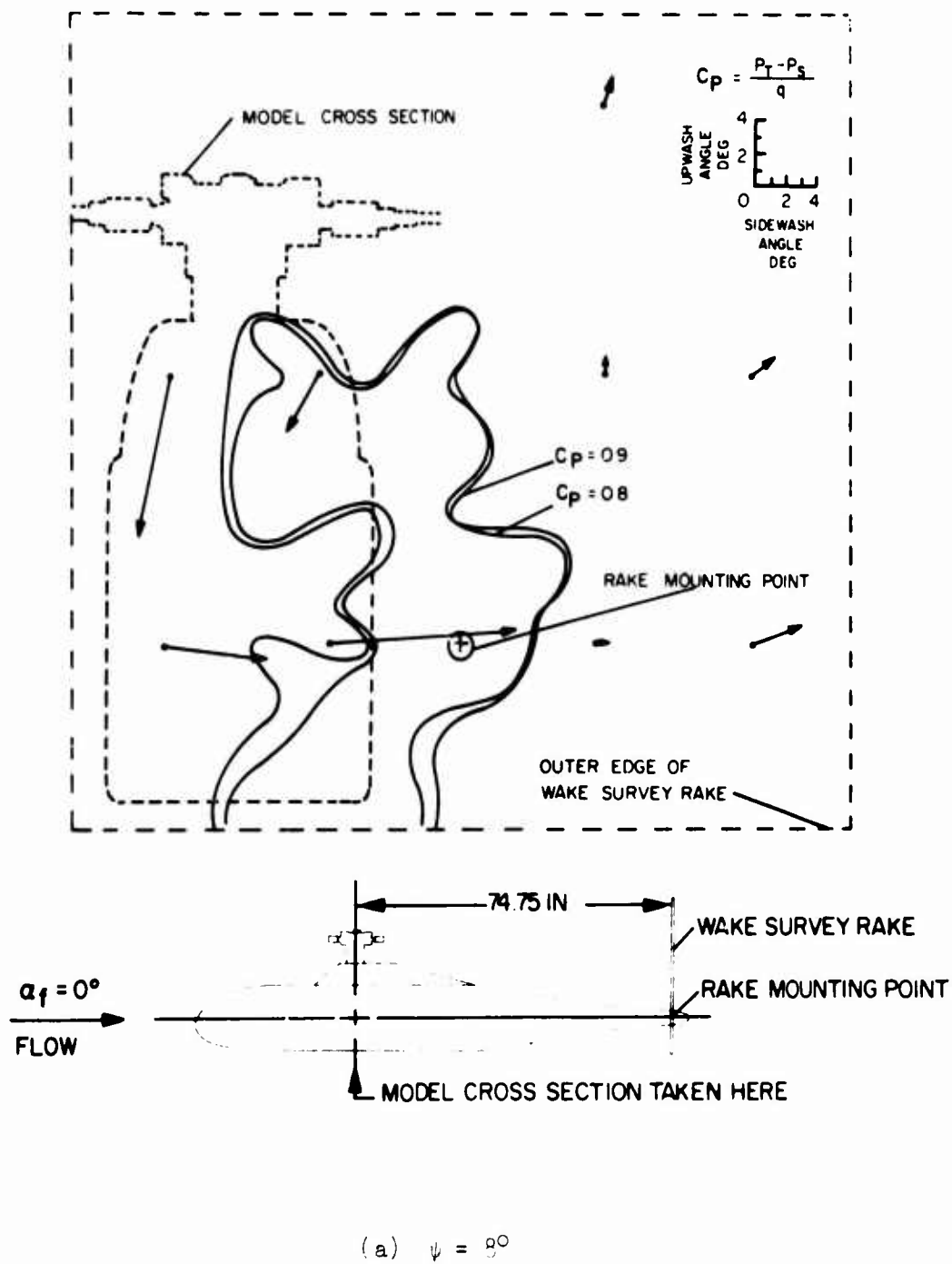
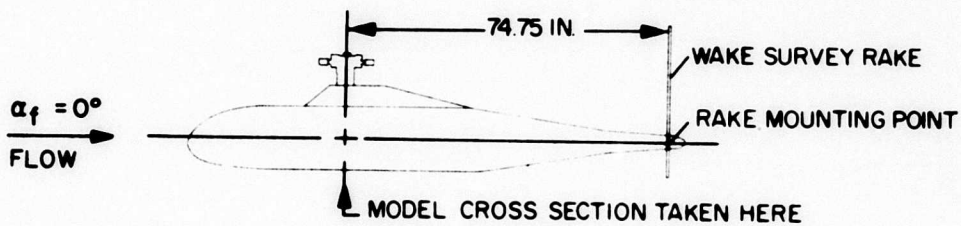
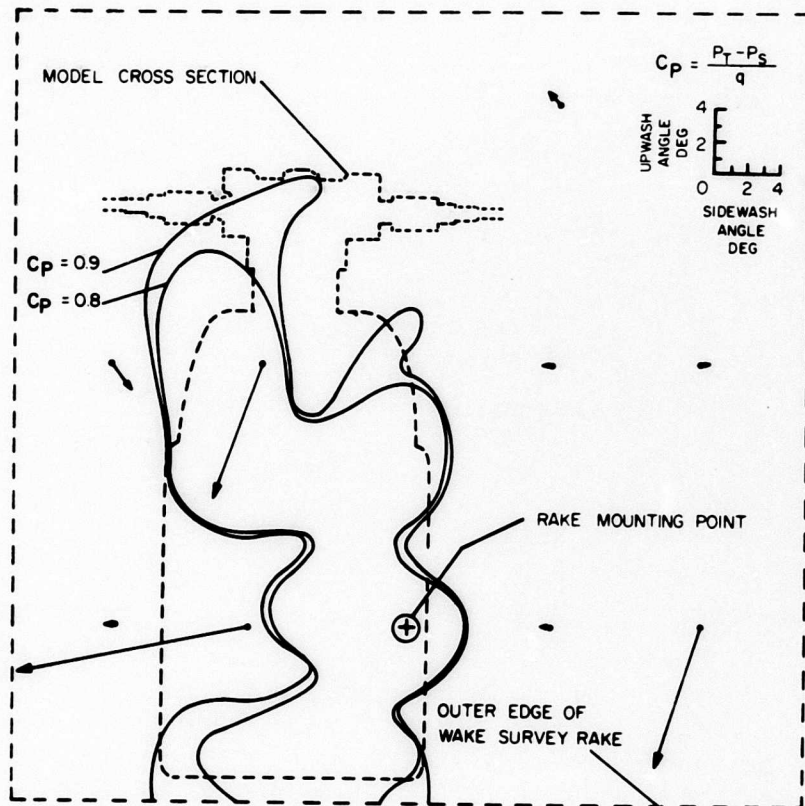
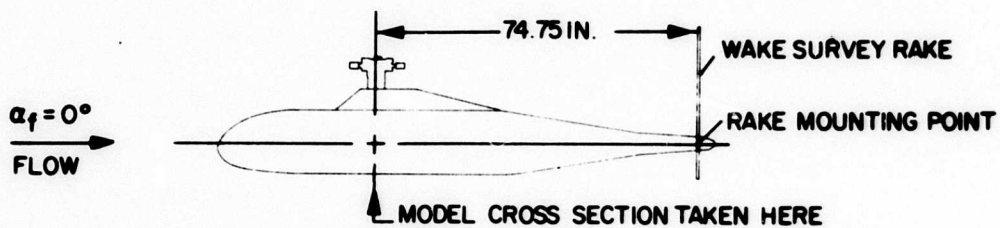
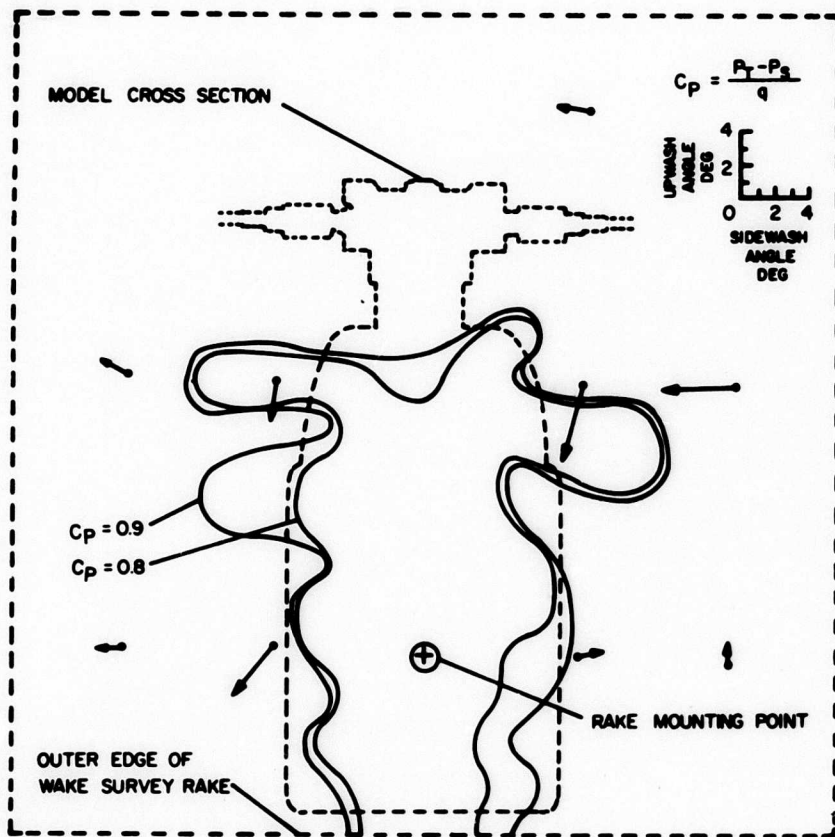


Figure 9. The Effect of Fuselage Yaw Angle on the Wake of the Helicopter Model, $V_S = 200$ Knots, $\alpha_f = 0^\circ$, $\mu^* = 0.5$, Configuration FH.



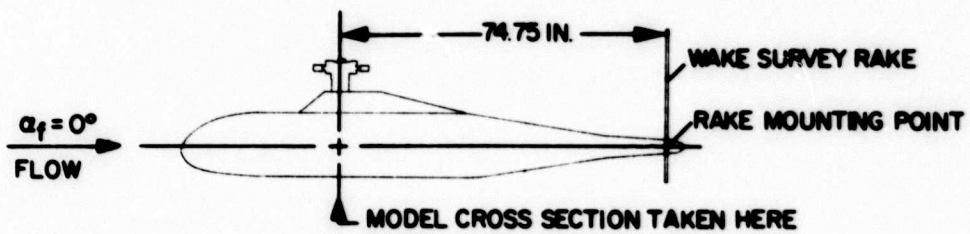
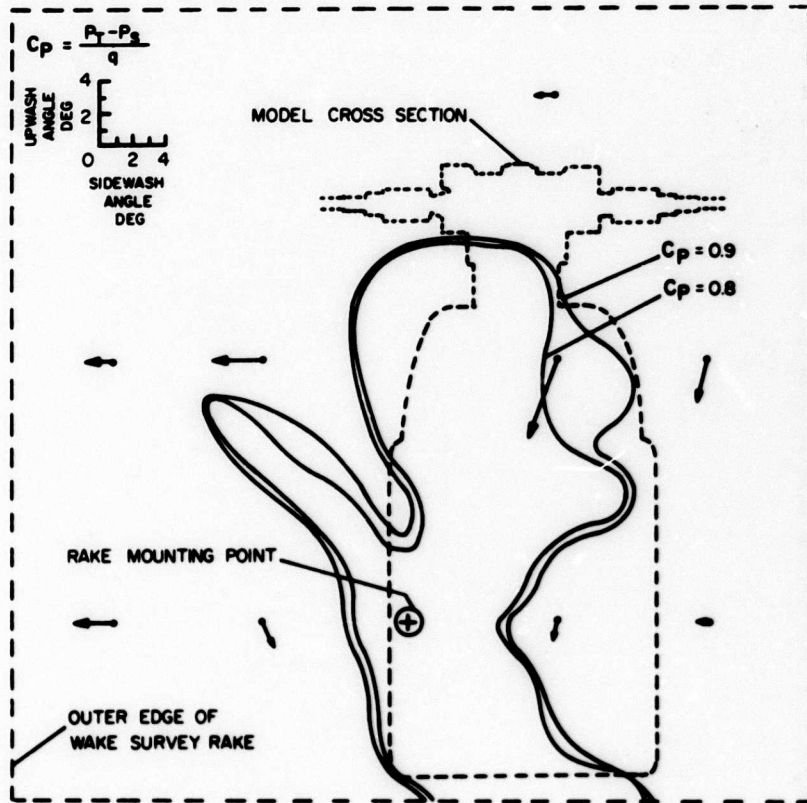
(b) $\psi = 4^\circ$

Figure 9. Continued



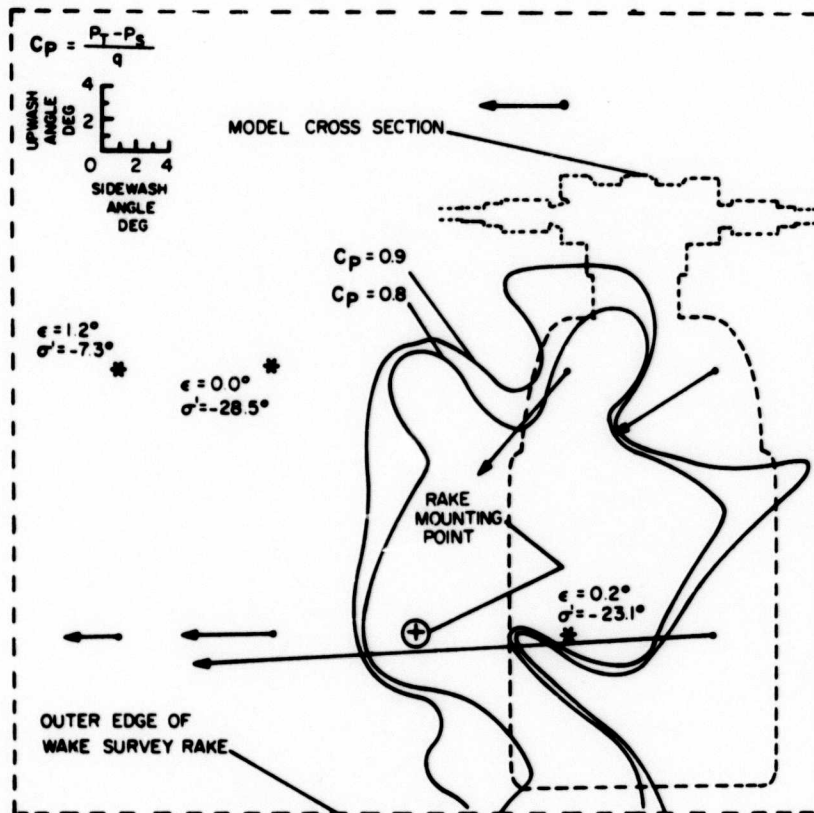
(c) $\psi = 0^\circ$

Figure 9. Continued

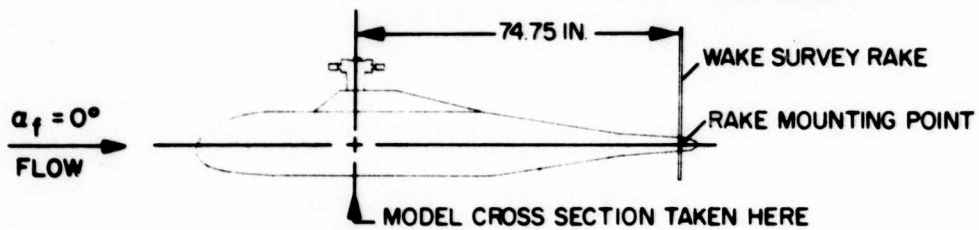


(d) $\psi = -4^\circ$

Figure 9. Continued

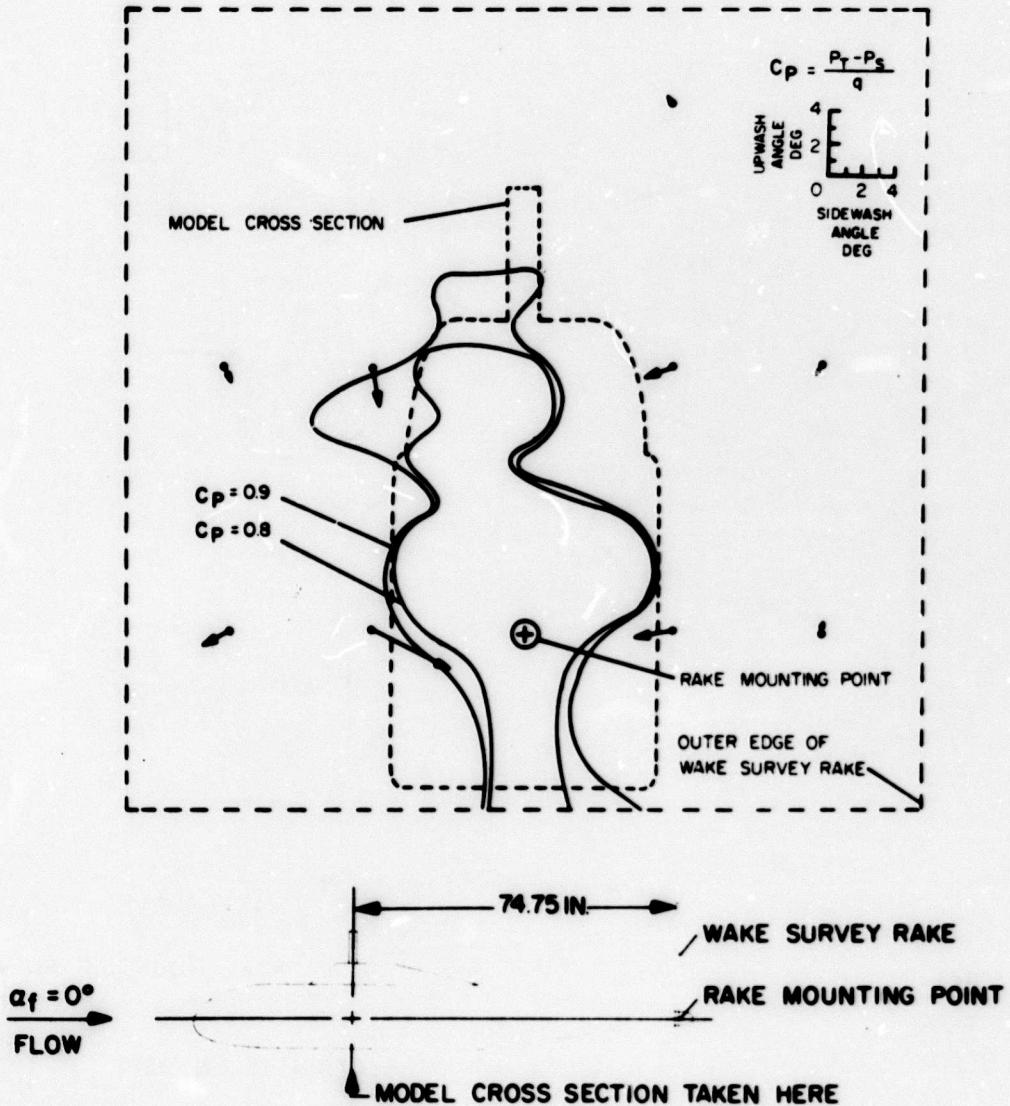


* DATA QUESTIONABLE



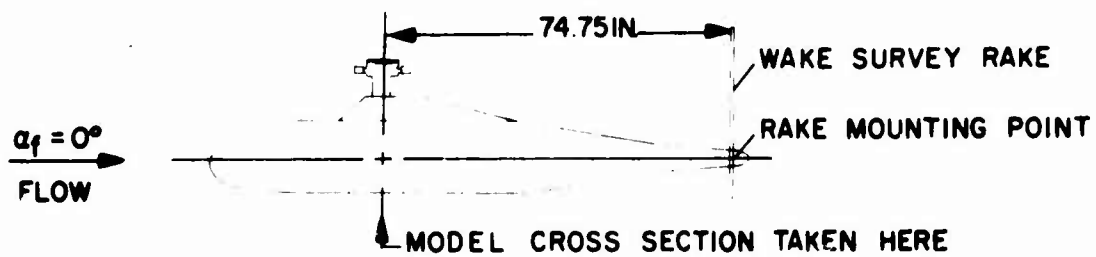
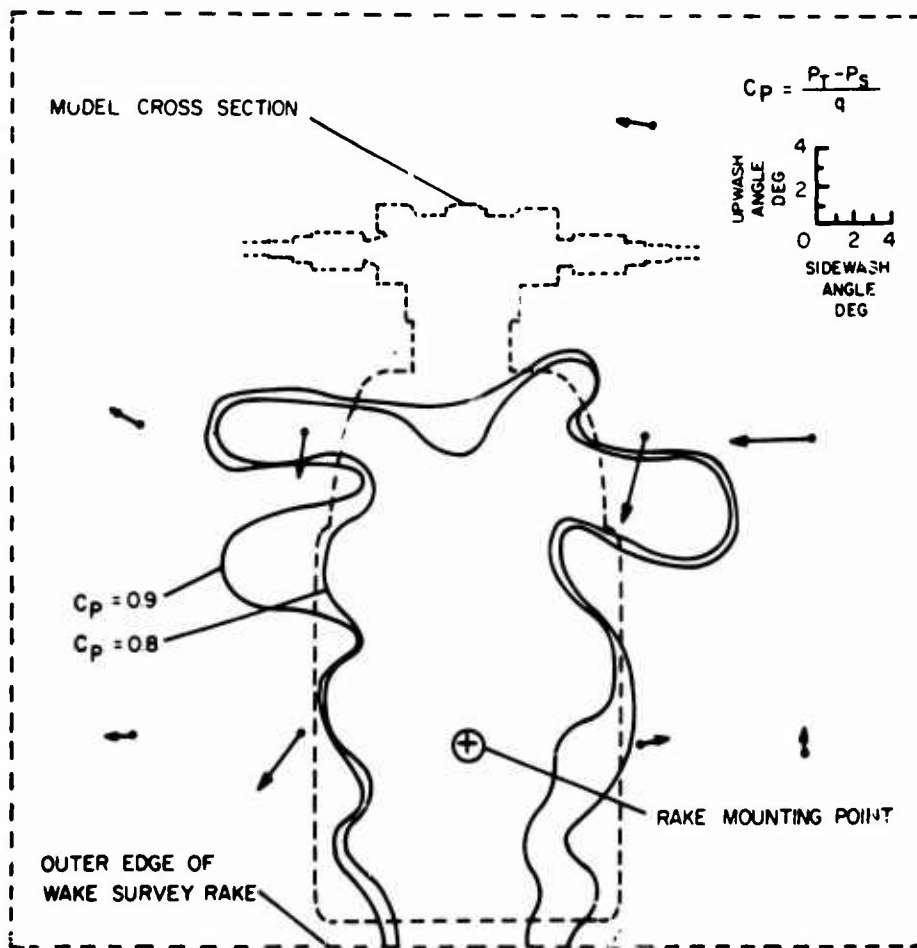
(e) $\psi = -8^\circ$

Figure 9. Concluded



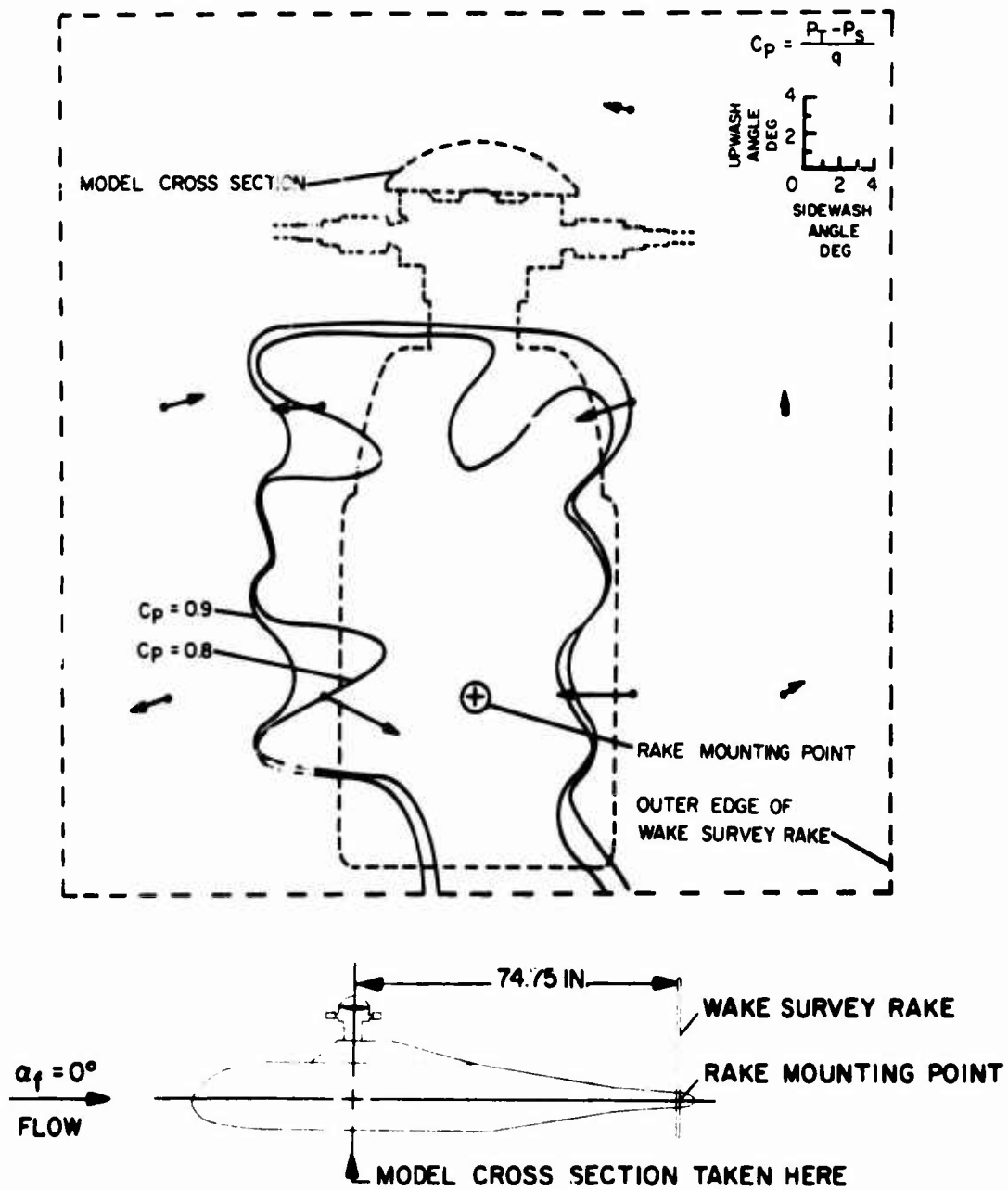
(a) Bare Shaft (Configuration F)

Figure 10. The Effect of Rotor Head Configuration In the Wake of the Helicopter Model, $V_S = 200$ Knots, $\alpha_f = 0^\circ$, $\psi = 0^\circ$, $u^* = 0.5$.



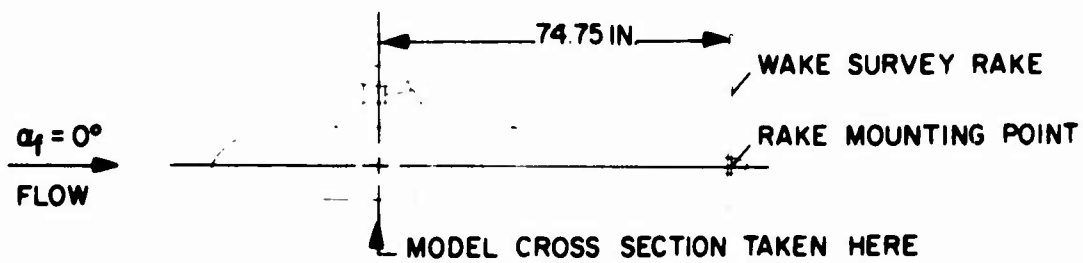
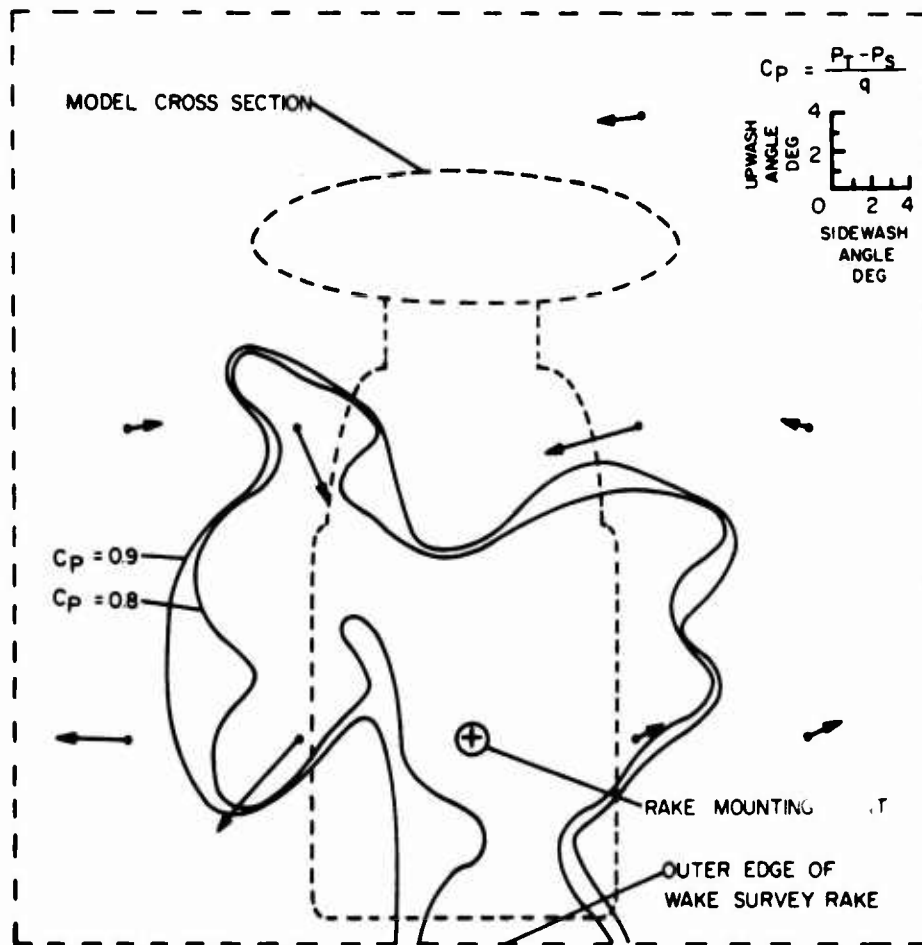
(b) Basic Rotor head (Configuration FH)

Figure 10. Continued



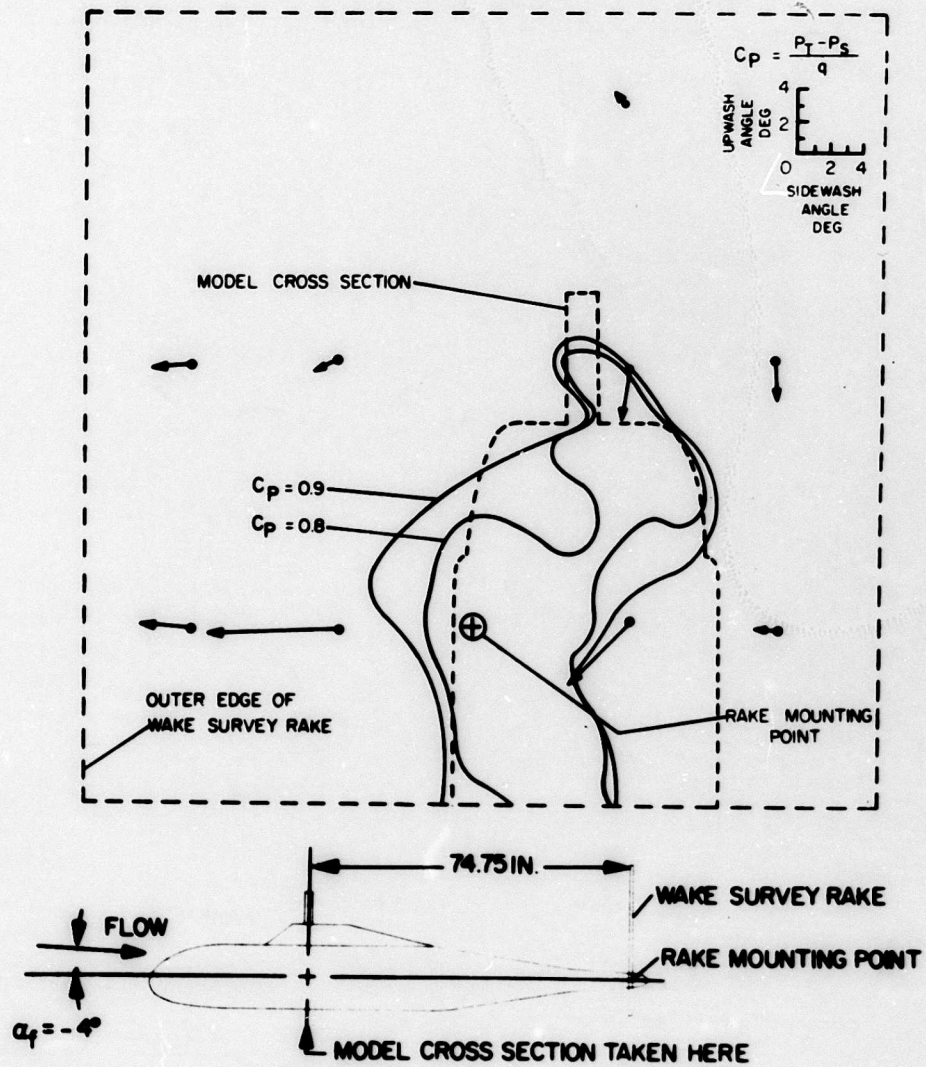
(c) Rotor Head With Cap Fairing (Configuration FHC)

Figure 10. Continued



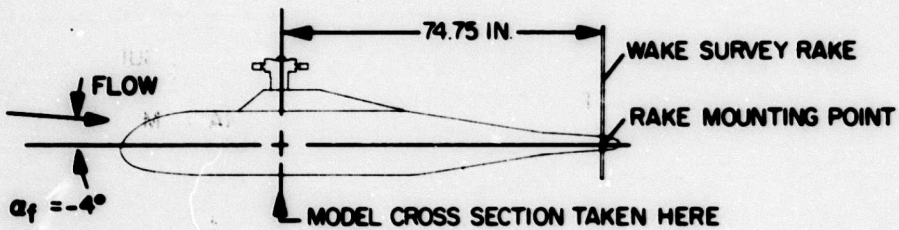
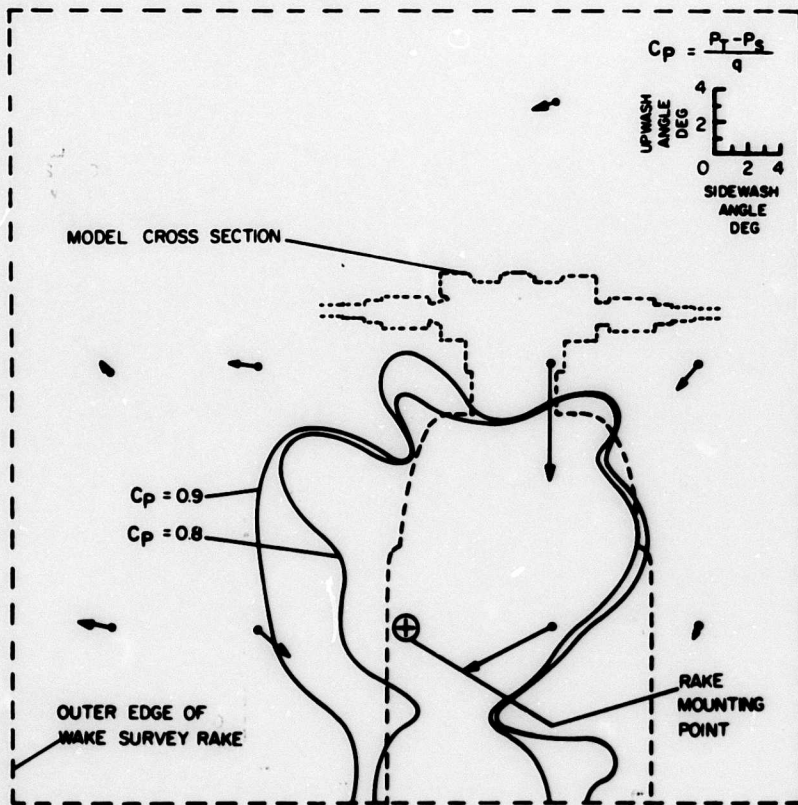
(d) Ellipsoidal Fairing (Configuration FHf)

Figure 10. Concluded



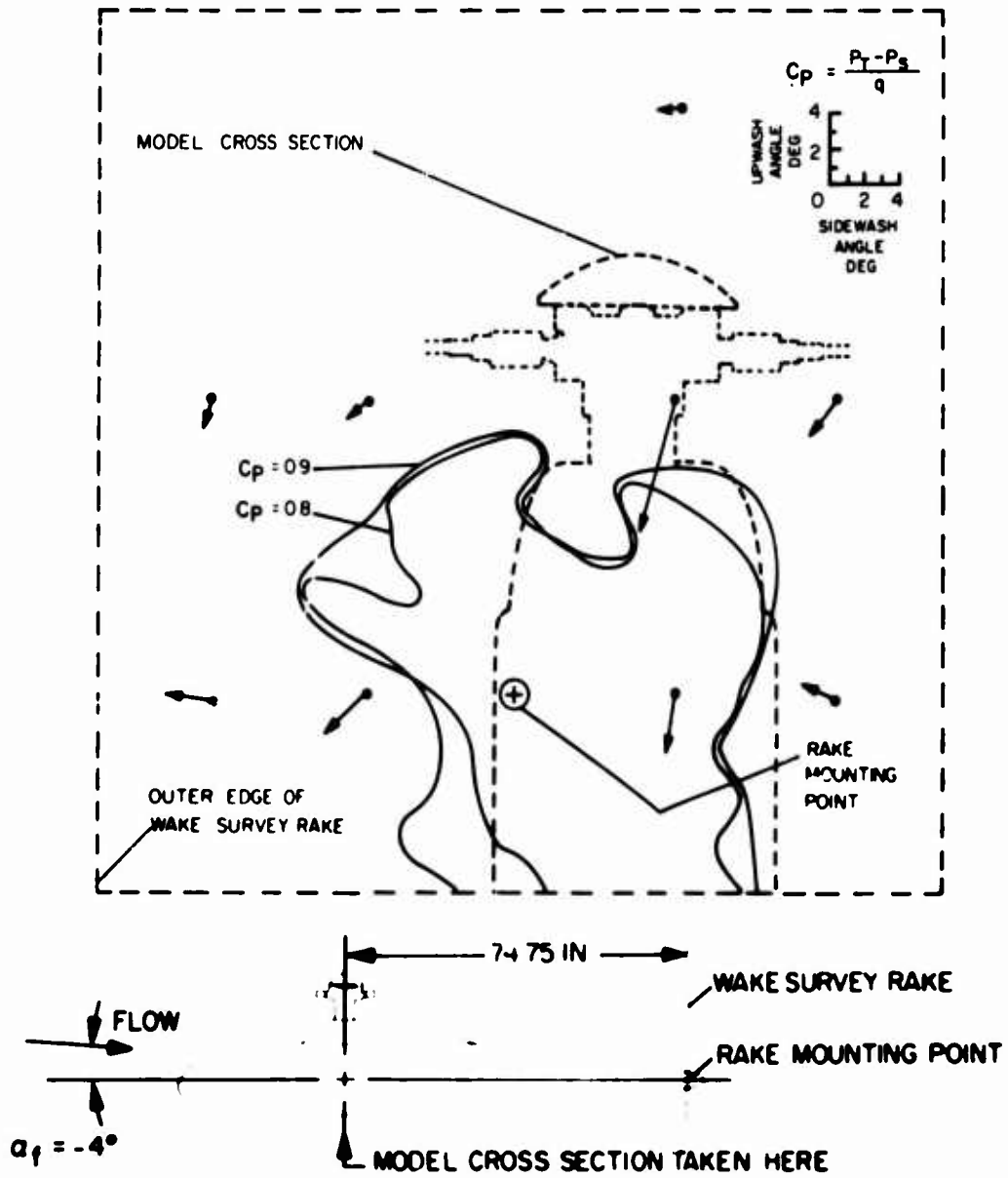
(a) Bare Shaft (Configuration F)

Figure 11. The Effect of Rotor Head Configuration on the Wake of the Helicopter Model, $V_S = 200$ Knots, $\alpha_f = -4^\circ$, $\psi = -4^\circ$, $u^* = 0.5$.



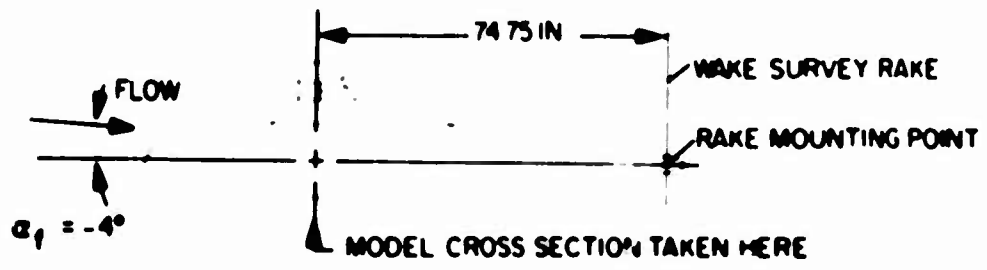
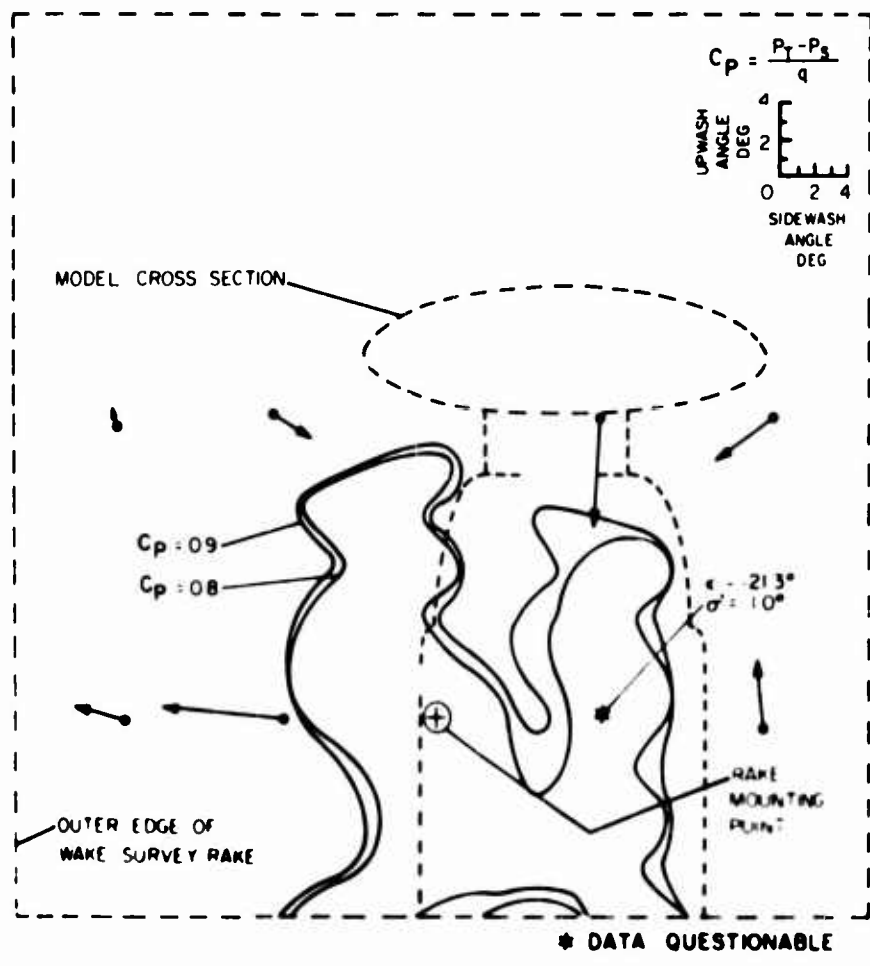
(b) Basic Rotor Head (Configuration FH)

Figure 11. Continued



(c) Rotor Head with Cap Fairing (Configuration FHC)

Figure 11. Continued



(d) Ellipsoidal Fairing (Configuration PHr)

Figure 11. Concluded

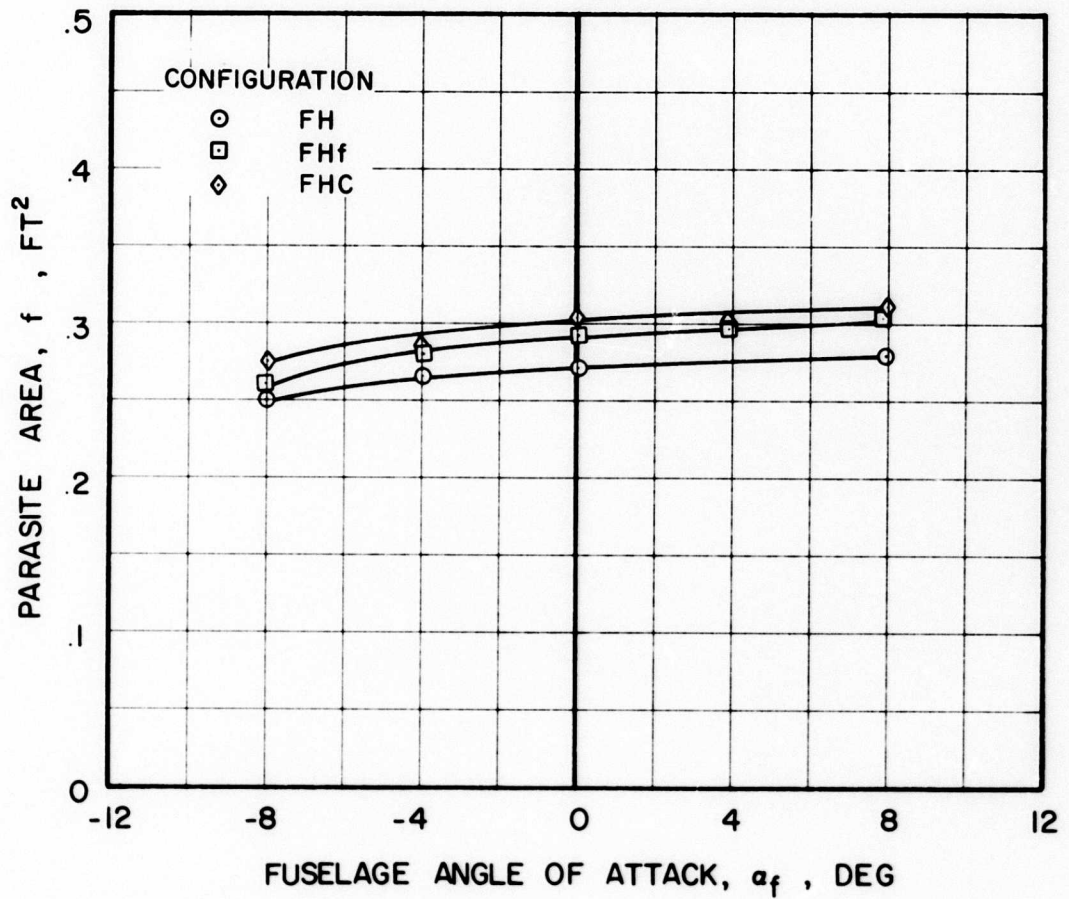
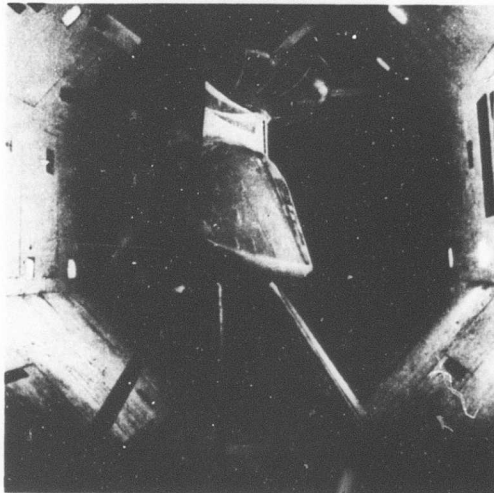
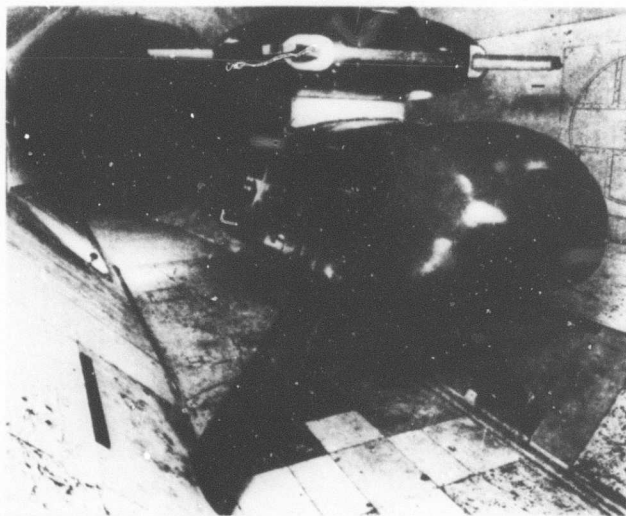


Figure 12. A Comparison of the Drag Characteristics of Three Rotor Head Configurations, $V_s = 200$ Knots, $\psi = 0$, $\mu^* = 0.5$.

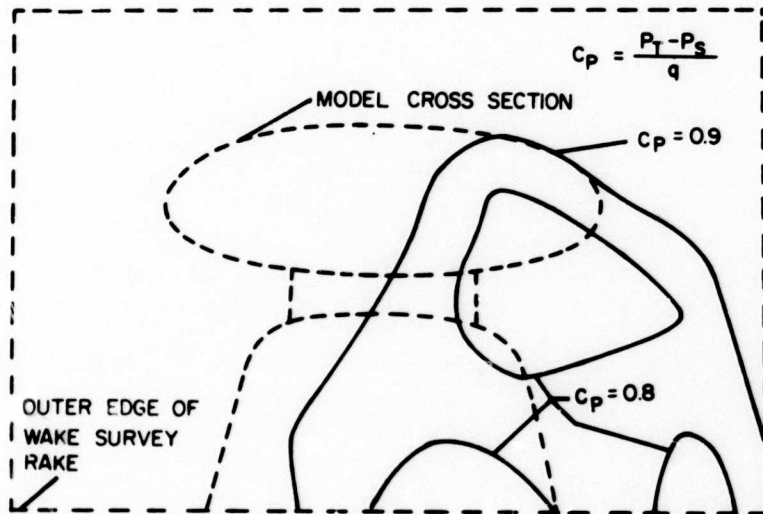


Rear View of Model

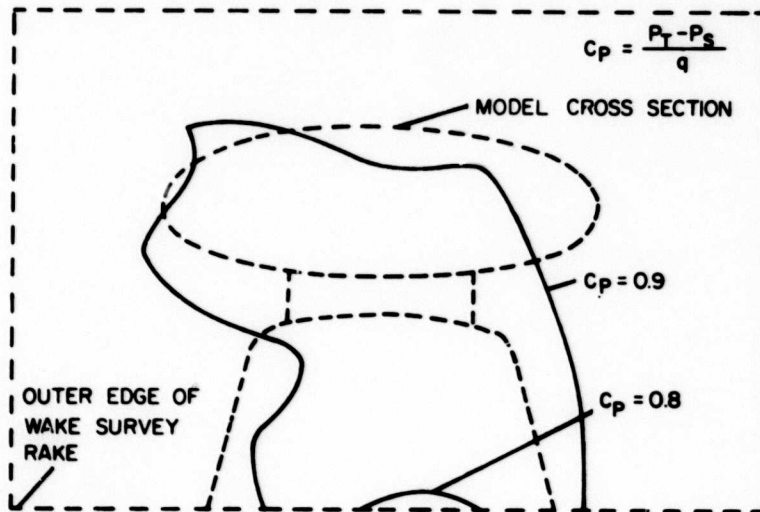


Front View of Model

Figure 13. Full-Scale Pylon-Ellipsoidal Rotor Head Fairing Installed in the United Aircraft Research Laboratories 18-Foot Main Wind Tunnel.

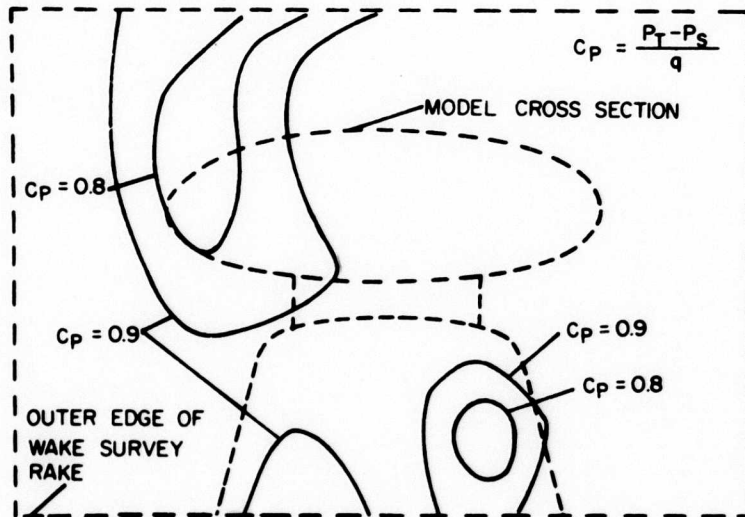


a. $\frac{P_J - P_S}{q} = 0$

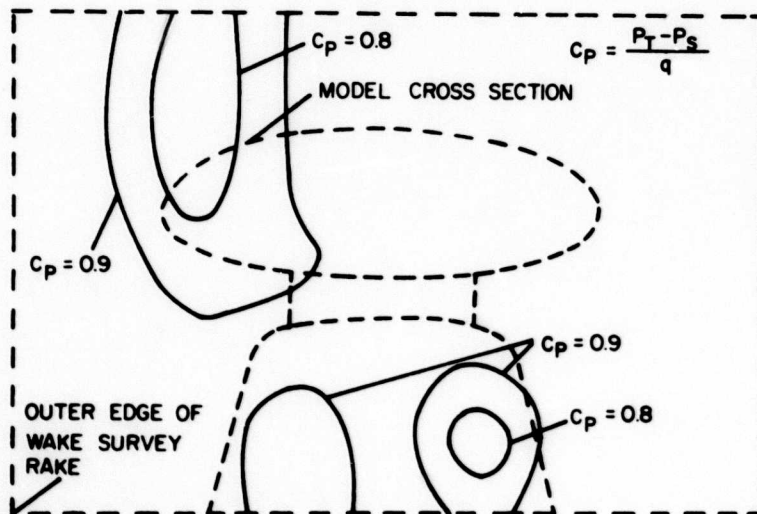


b. $\frac{P_J - P_S}{q} = 5.8$

Figure 14. The Effect of Increasing Boundary Layer Control Jet Pressure Ratio on the Wake 38 Feet Behind the Full-Scale Pylon-Ellipsoidal Rotor Head Fairing Model, $V = 105$ Knots, $\alpha_f = 0^\circ$, $\mu^* = 0.27$.

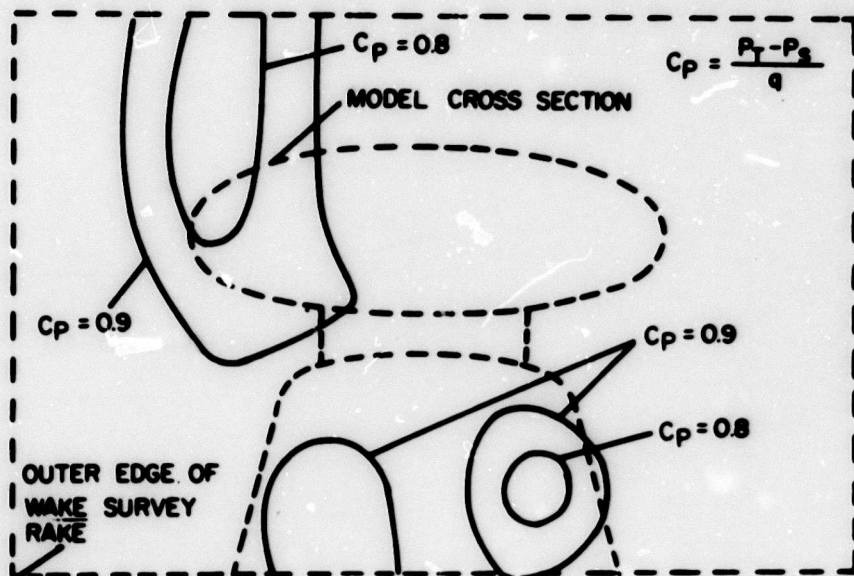


c. $\frac{P_J - P_S}{q} = 7.7$



d. $\frac{P_J - P_S}{q} = 11.6$

Figure 14. Continued



e. $\frac{P_T - P_s}{q} = 10.3$

Figure 14. Continued

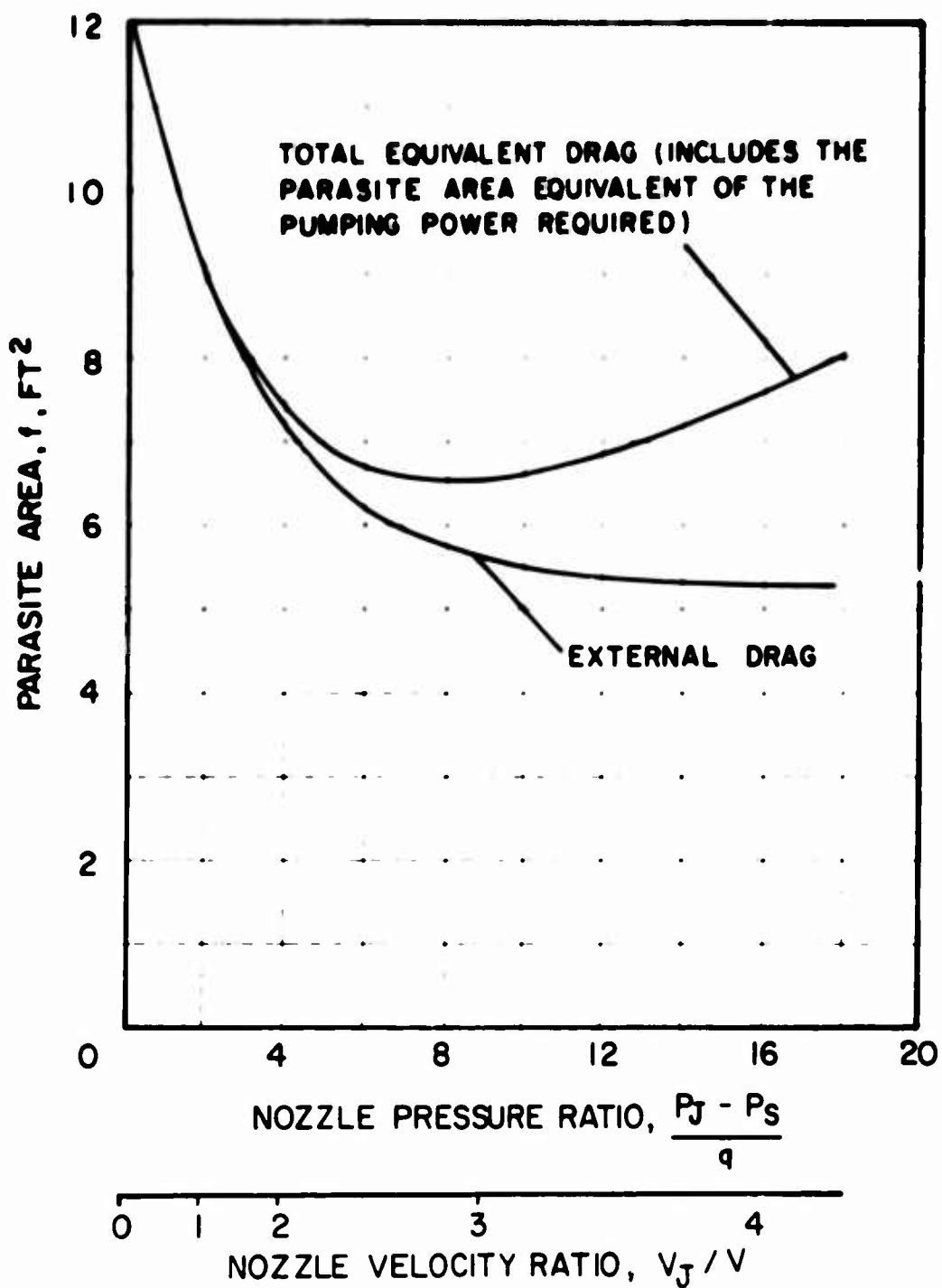


Figure 15. The Effect of Boundary Layer Control Jet Pressure Ratio on the Drag Characteristics of the Full-Scale Pylon-Ellipsoidal Rotor Head Fairing Model.

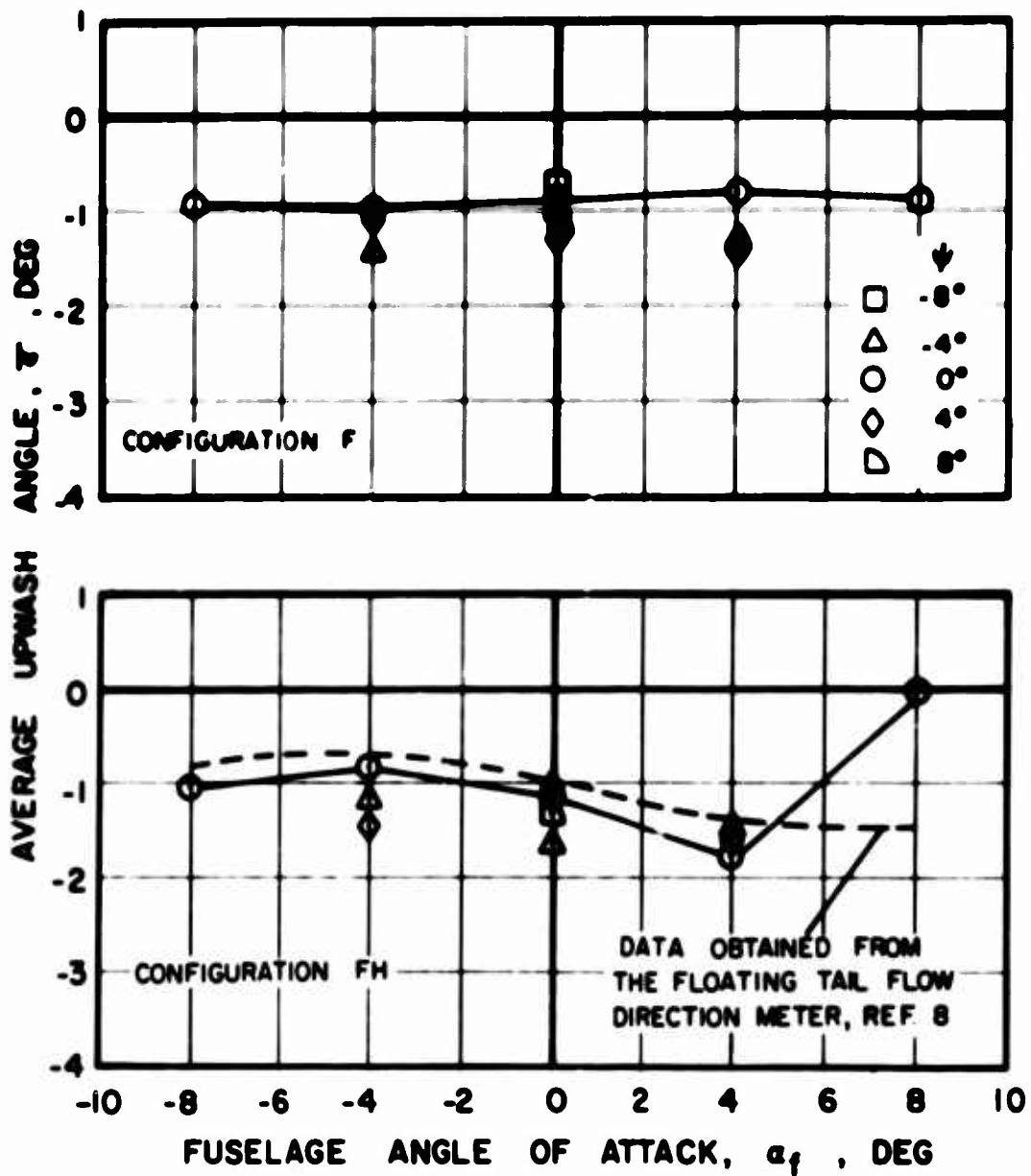


Figure 16. The Effect of Fuselage Angle of Attack on the Average Upwash Measured Behind the Helicopter Model for Various Rotor Head Configurations.

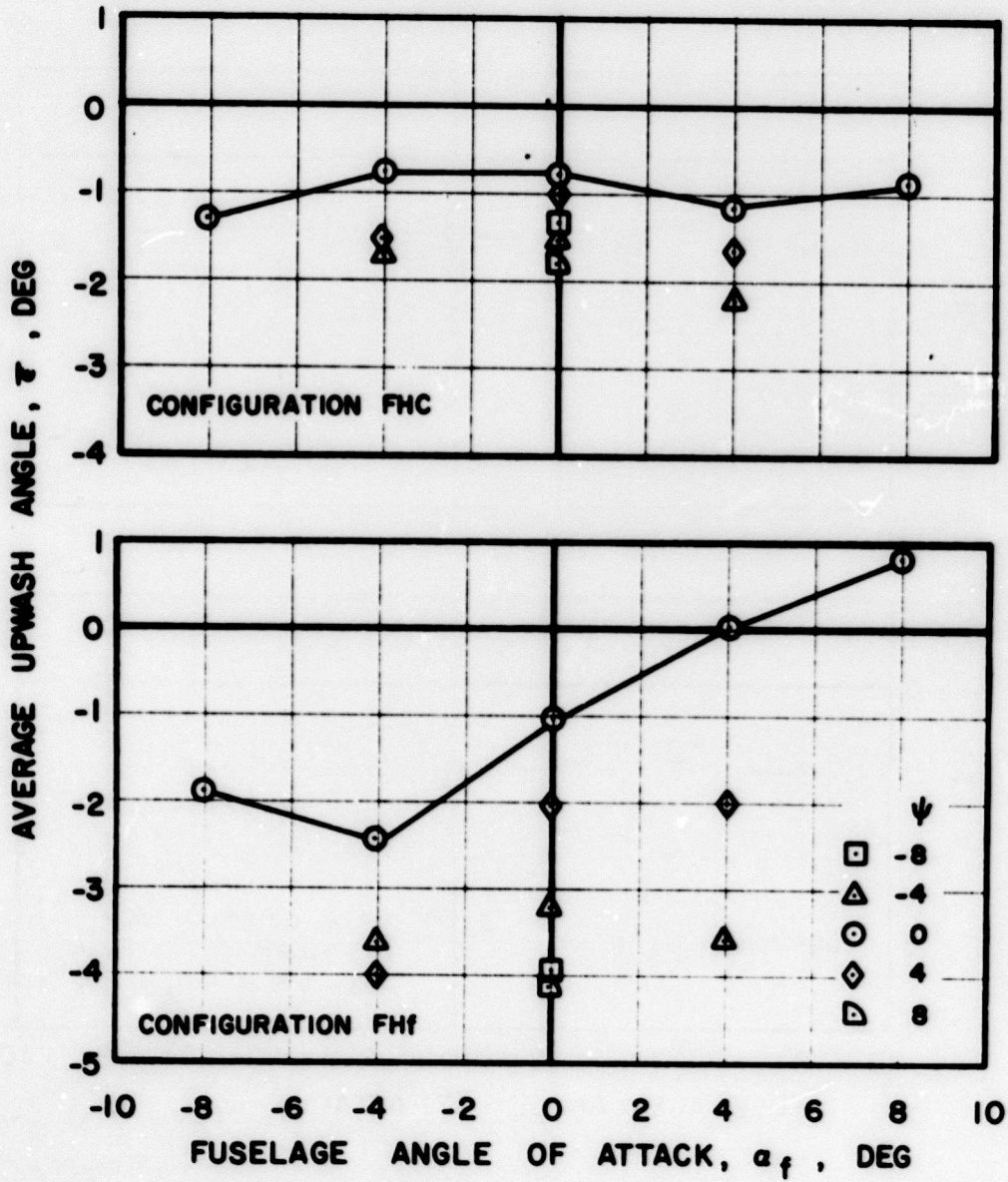


Figure 16. Concluded

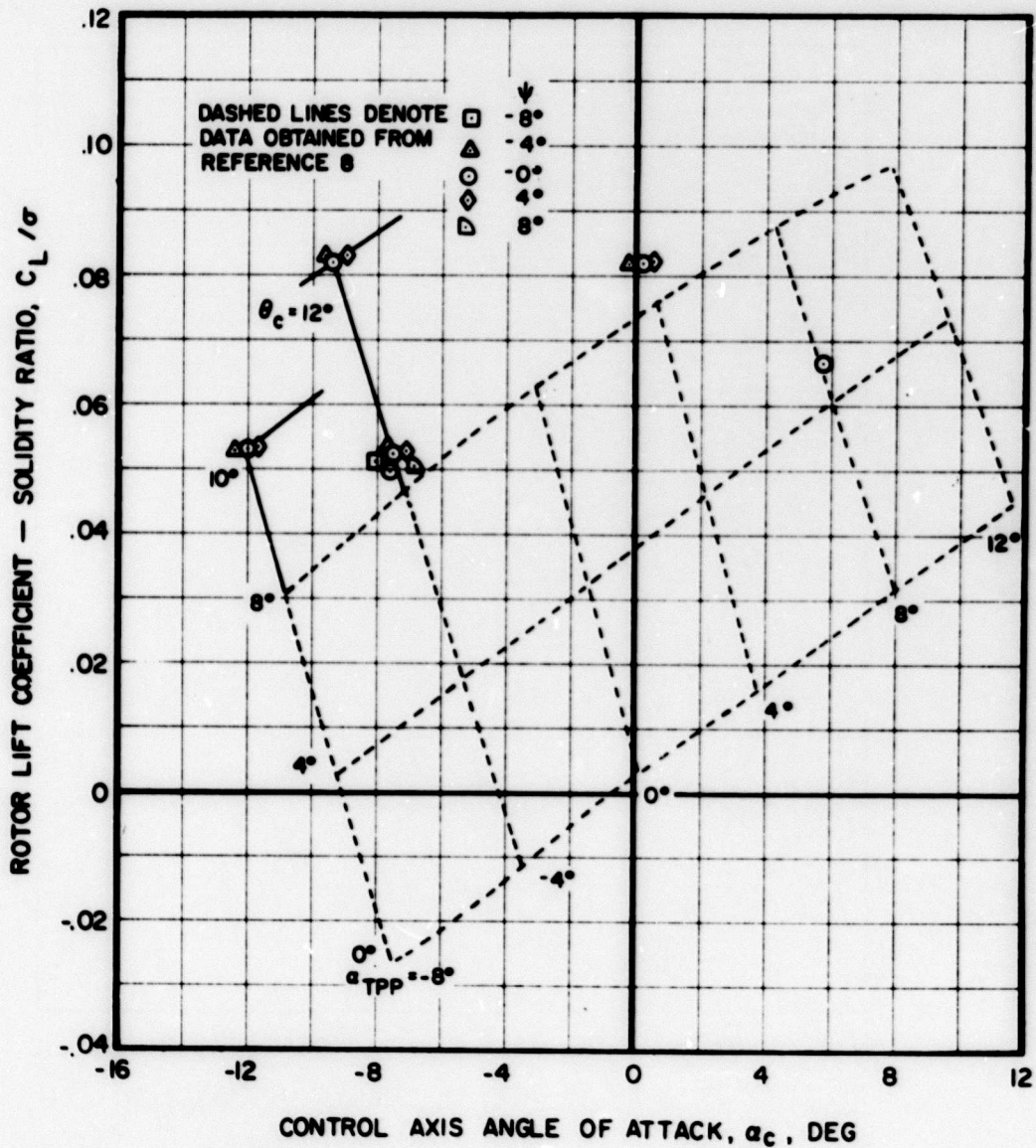


Figure 17. Rotor Performance Data, Including Comparative Data From Reference 8. $V_s = 120$ Knots, $\mu = 0.3$, Configuration FHB.

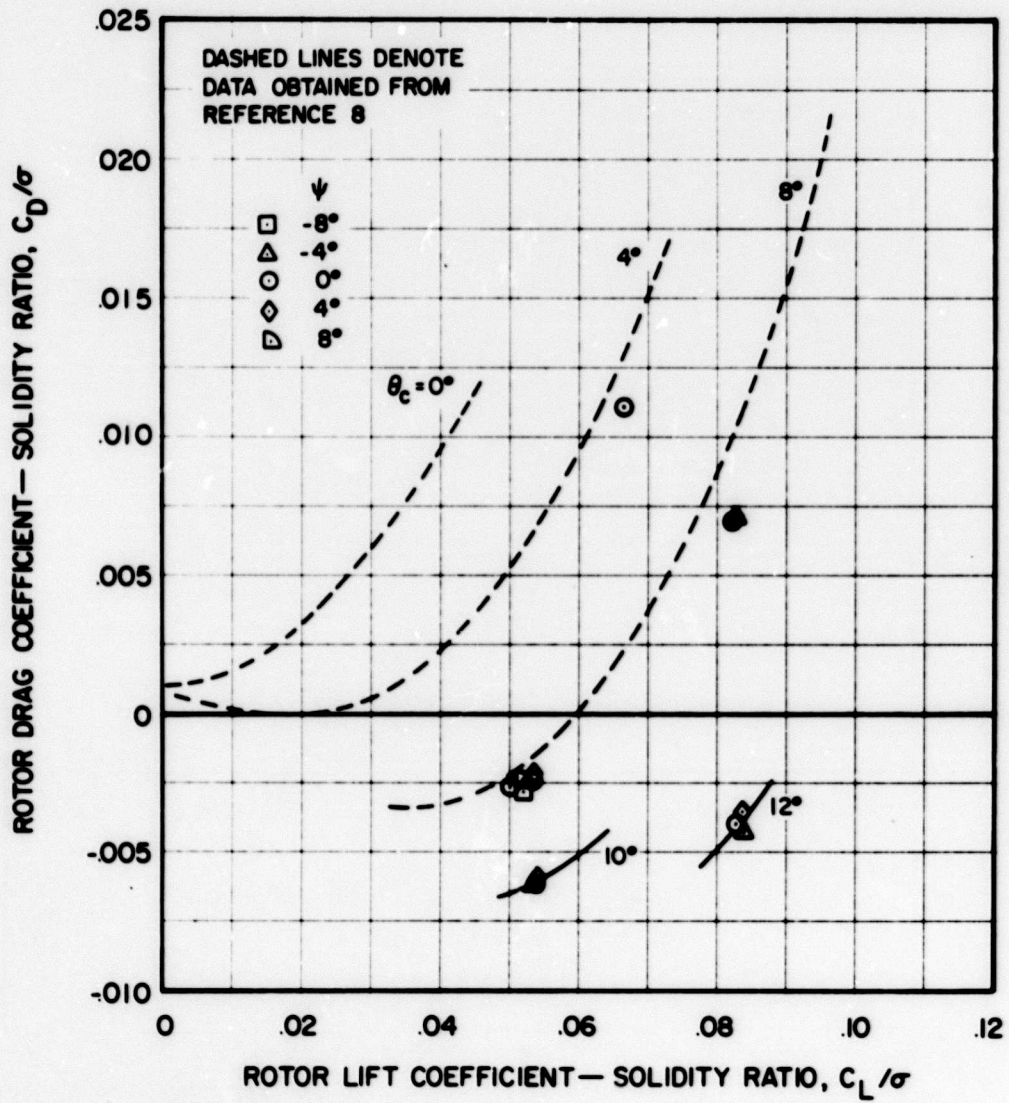


Figure 17. Continued

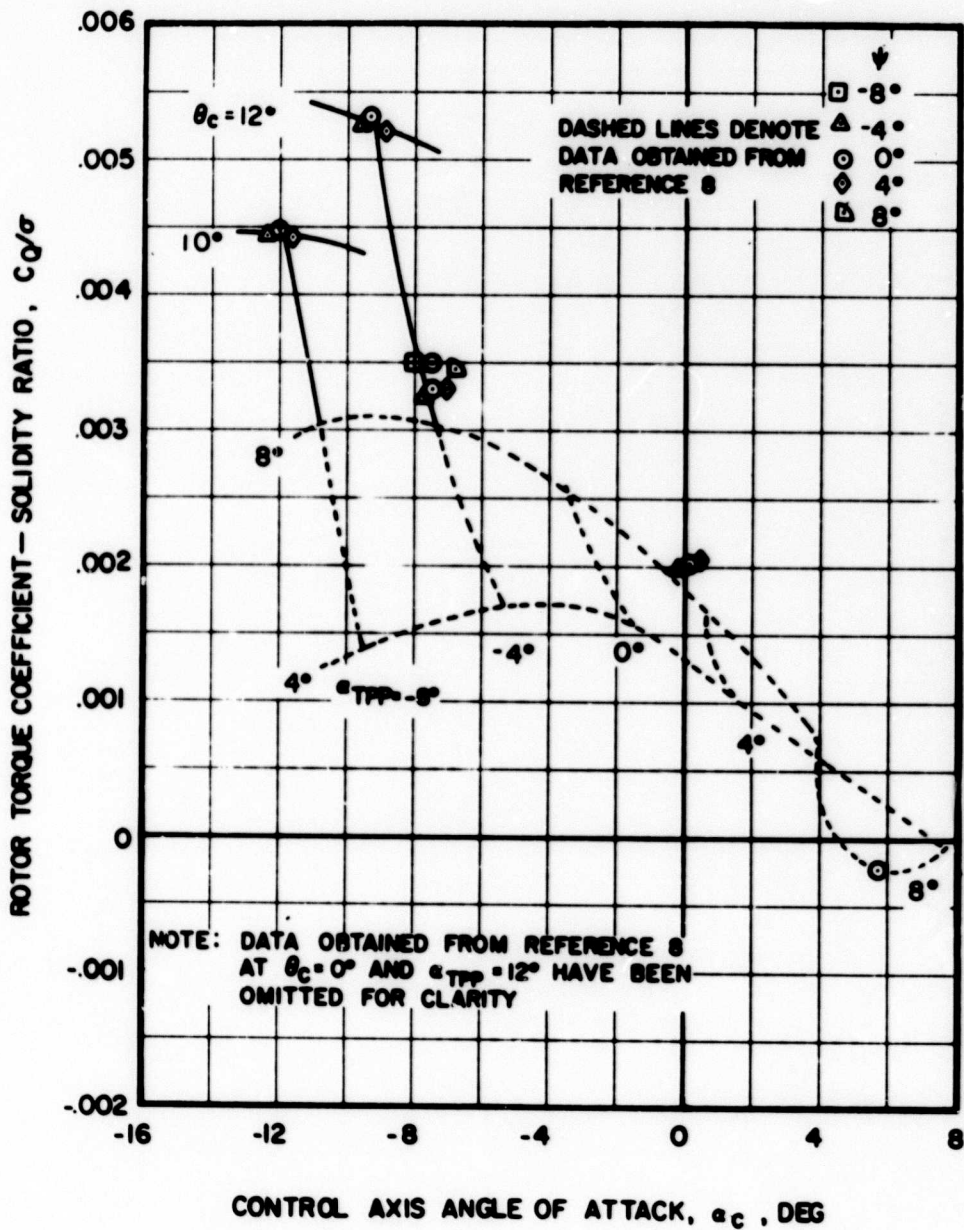


Figure 17. Concluded

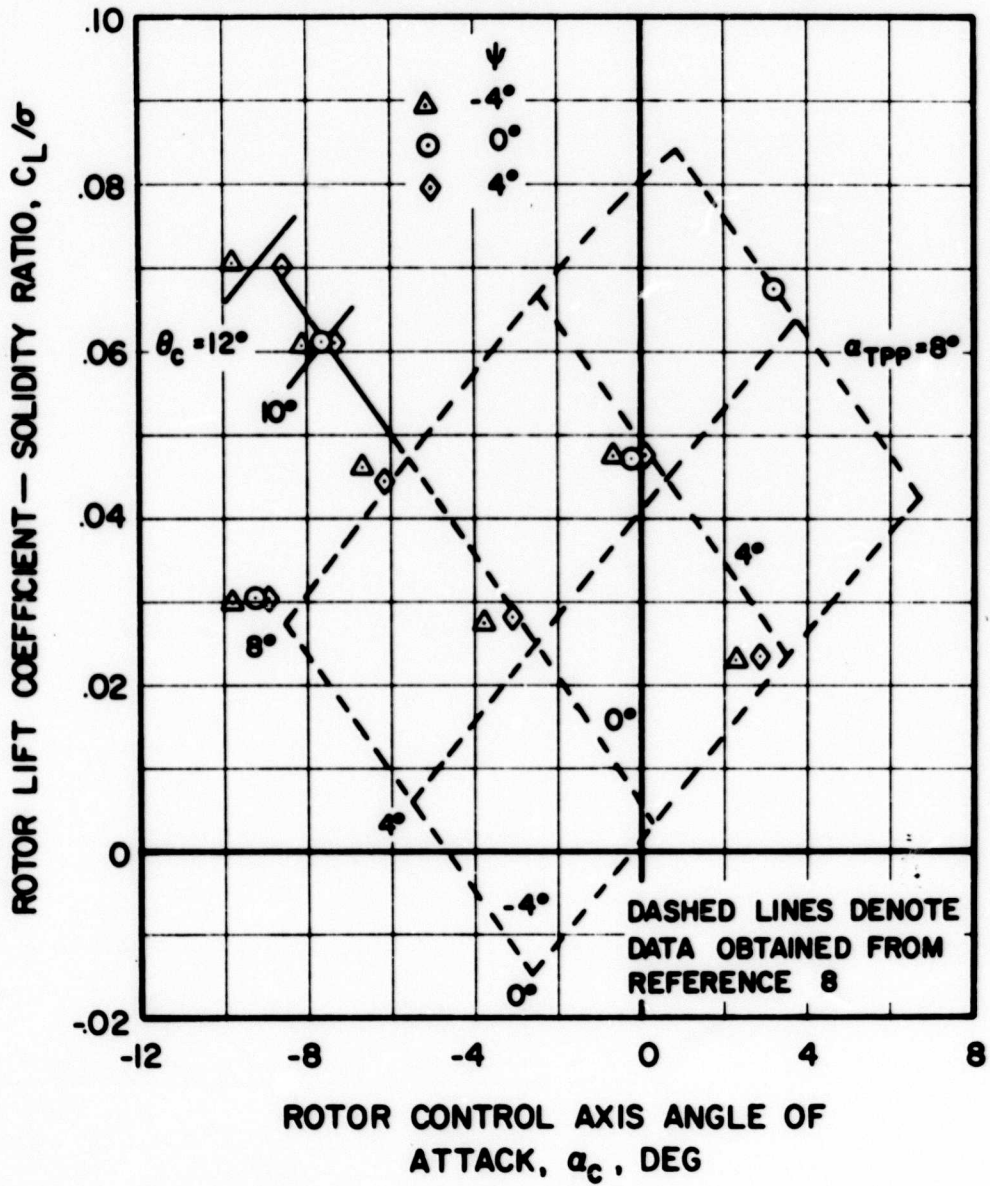


Figure 18. Rotor Performance Data, Including Comparative Data From Reference 8. $V_s = 200$ Knots, $\mu = 0.5$, Configuration FHB.

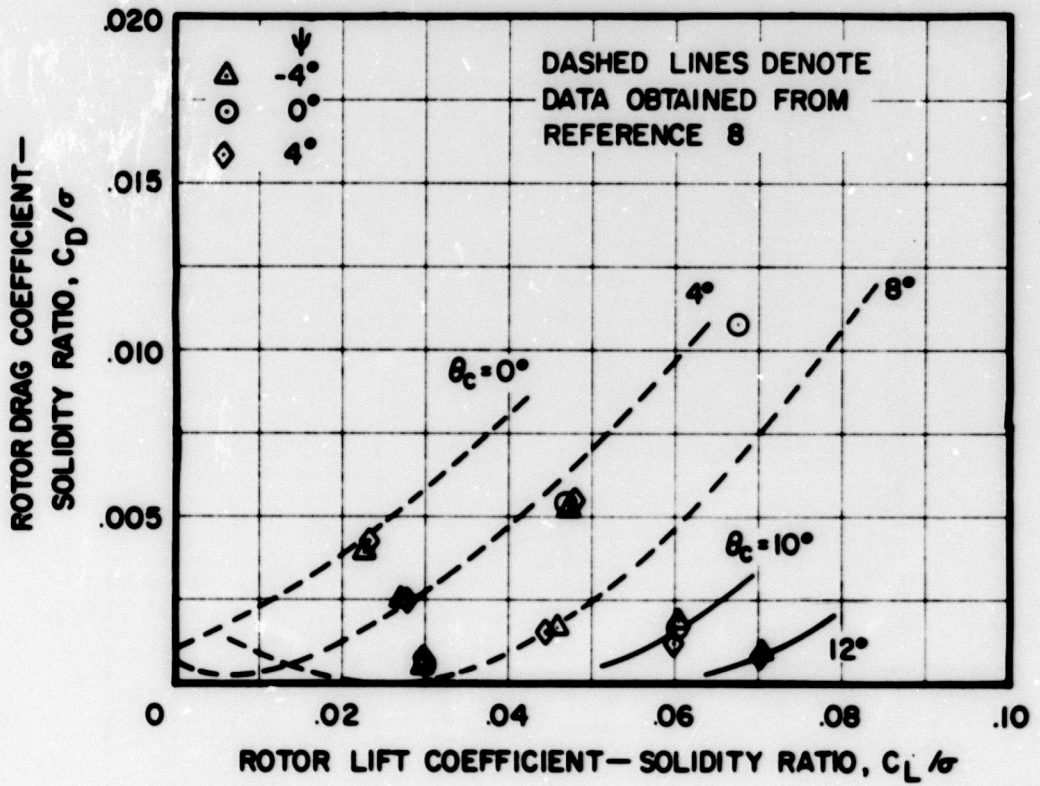


Figure 18. Continued

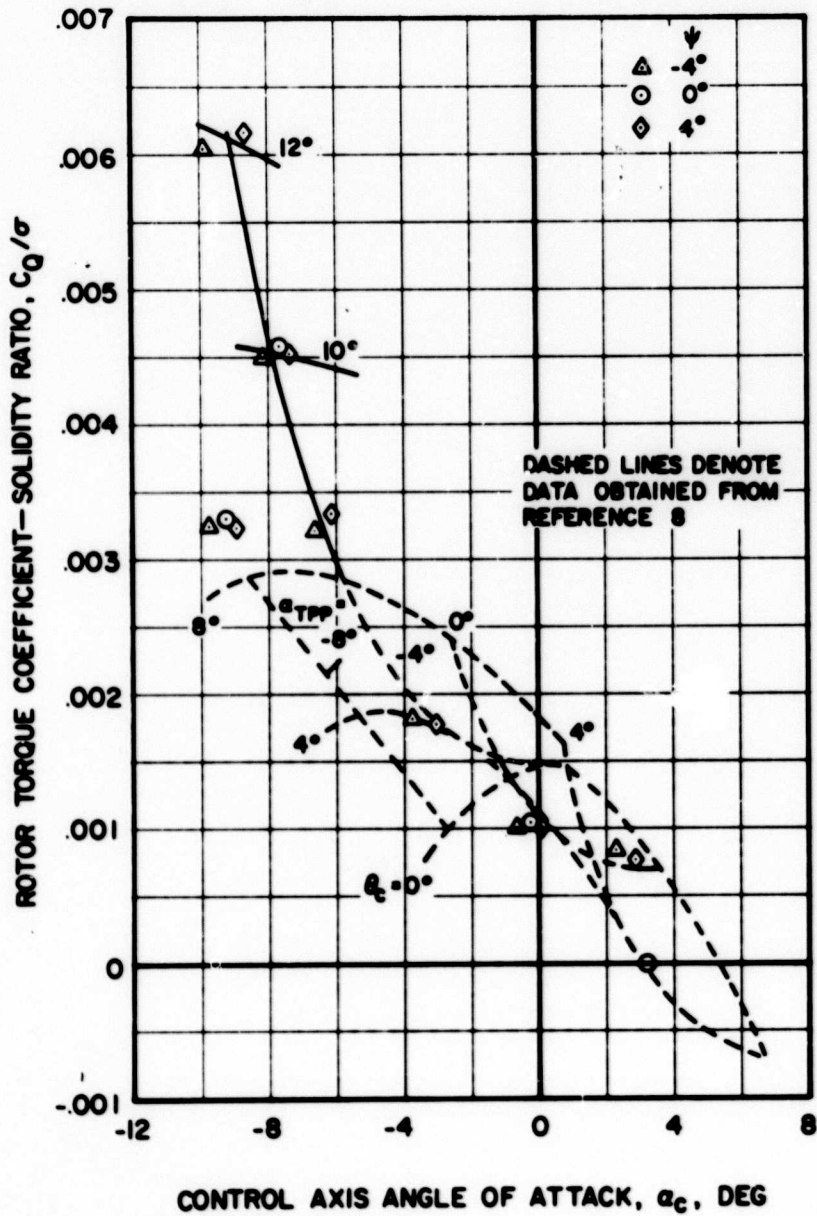


Figure 18. Concluded

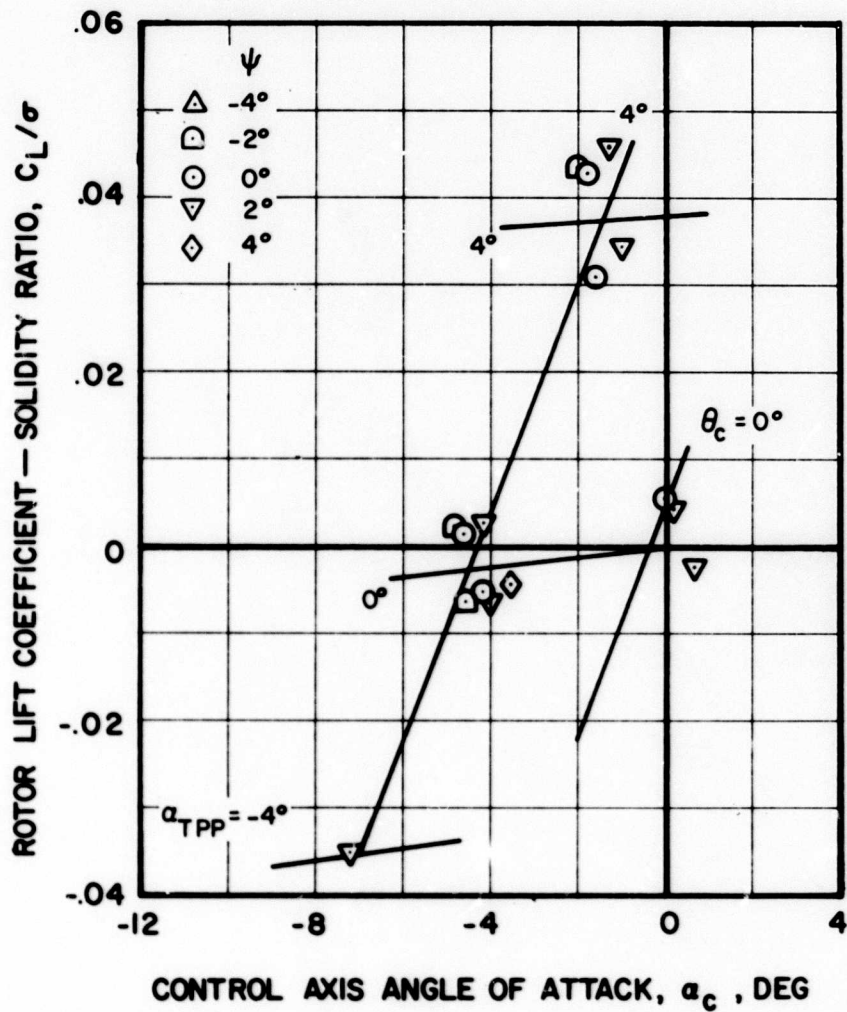


Figure 19. Rotor Lift Data. $V_s = 250$ Knots, $\mu = 1.0$, Configuration FHB.

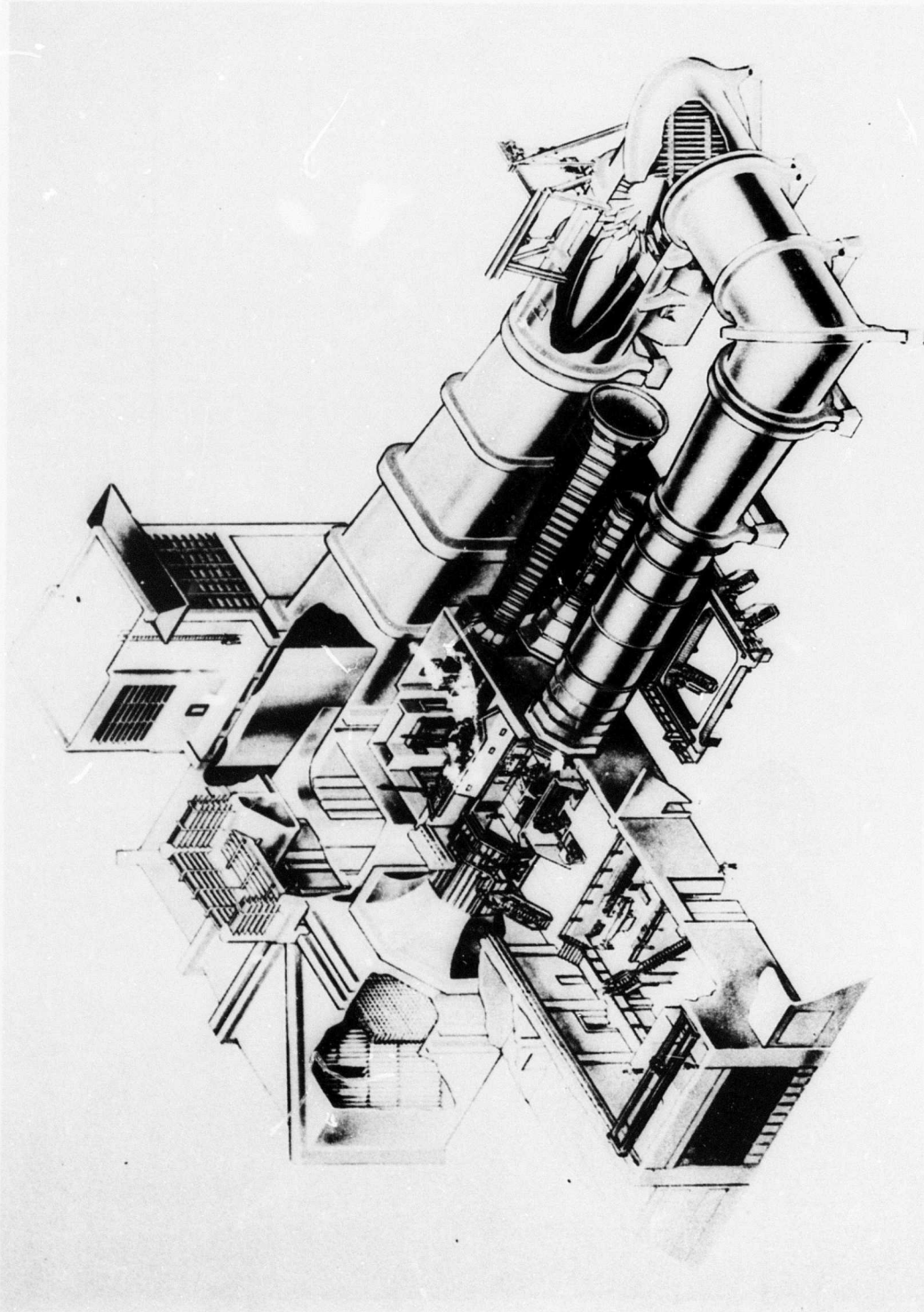


Figure 20. United Aircraft Research Laboratories
18-Foot Main Wind Tunnel.

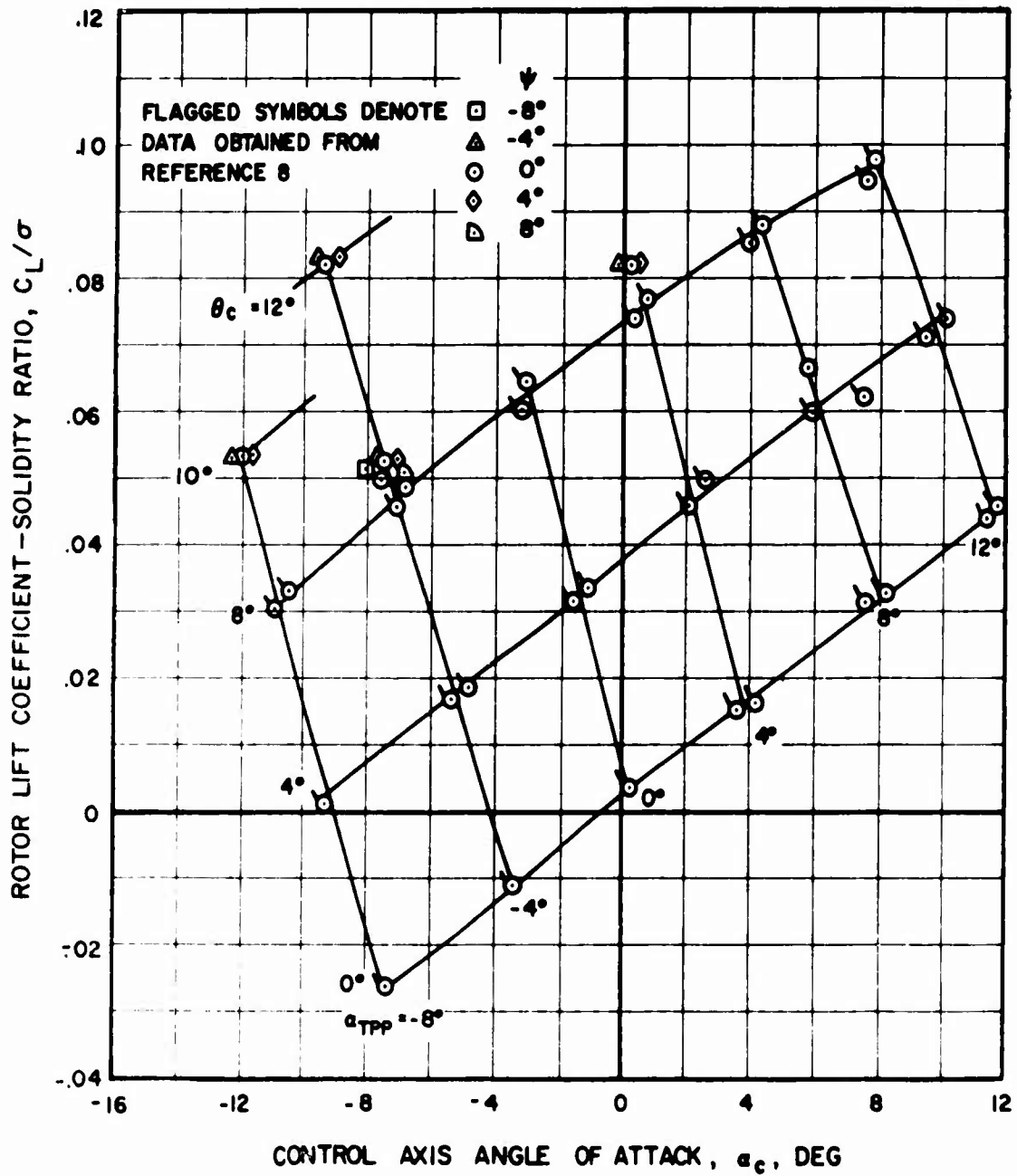


Figure 21. Sample Rotor Performance Data, Including Comparative Performance Data Obtained From Reference 8.
 $V_S = 120$ Knots, $\mu = 0.3$.

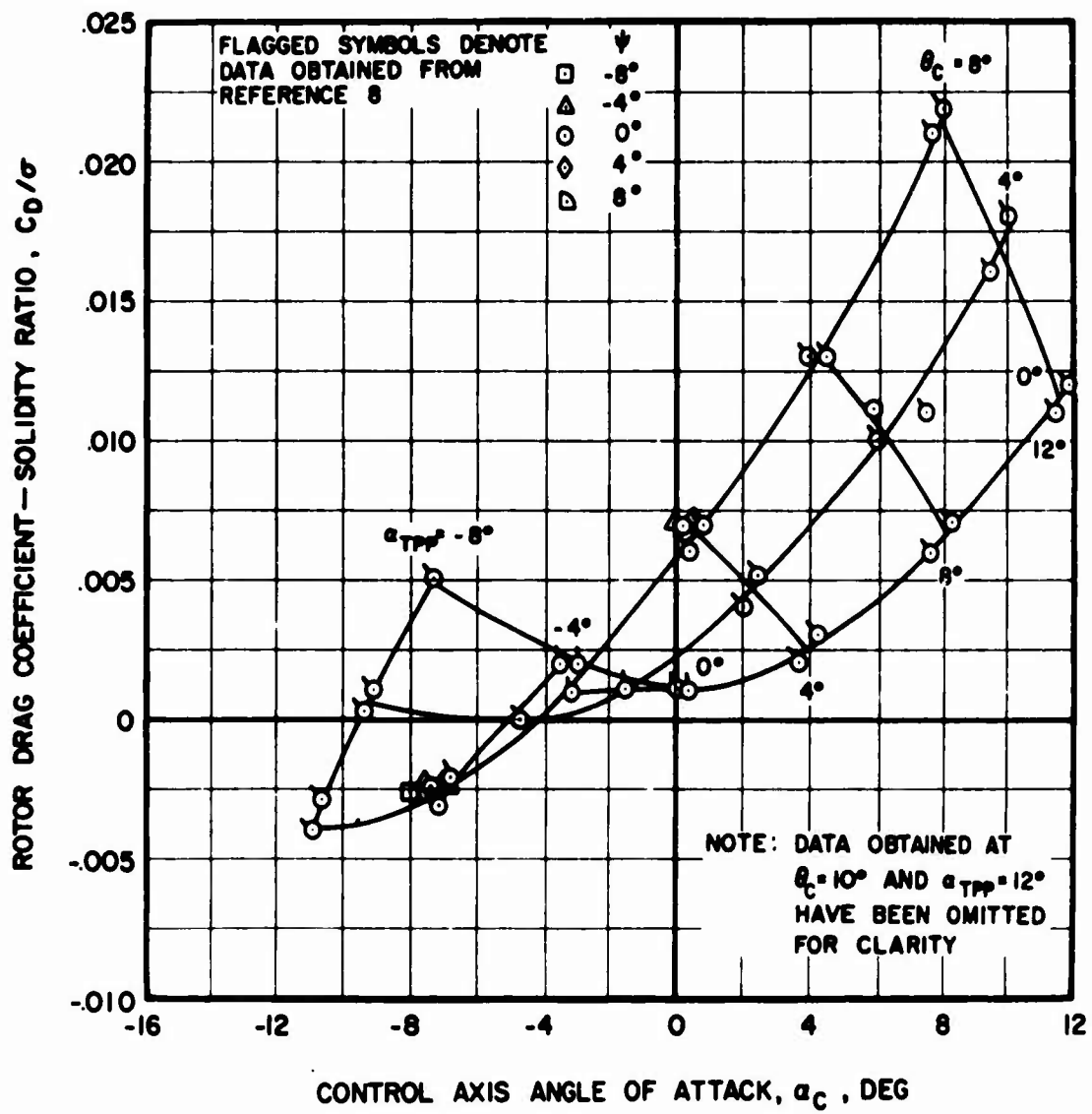


Figure 21. Continued

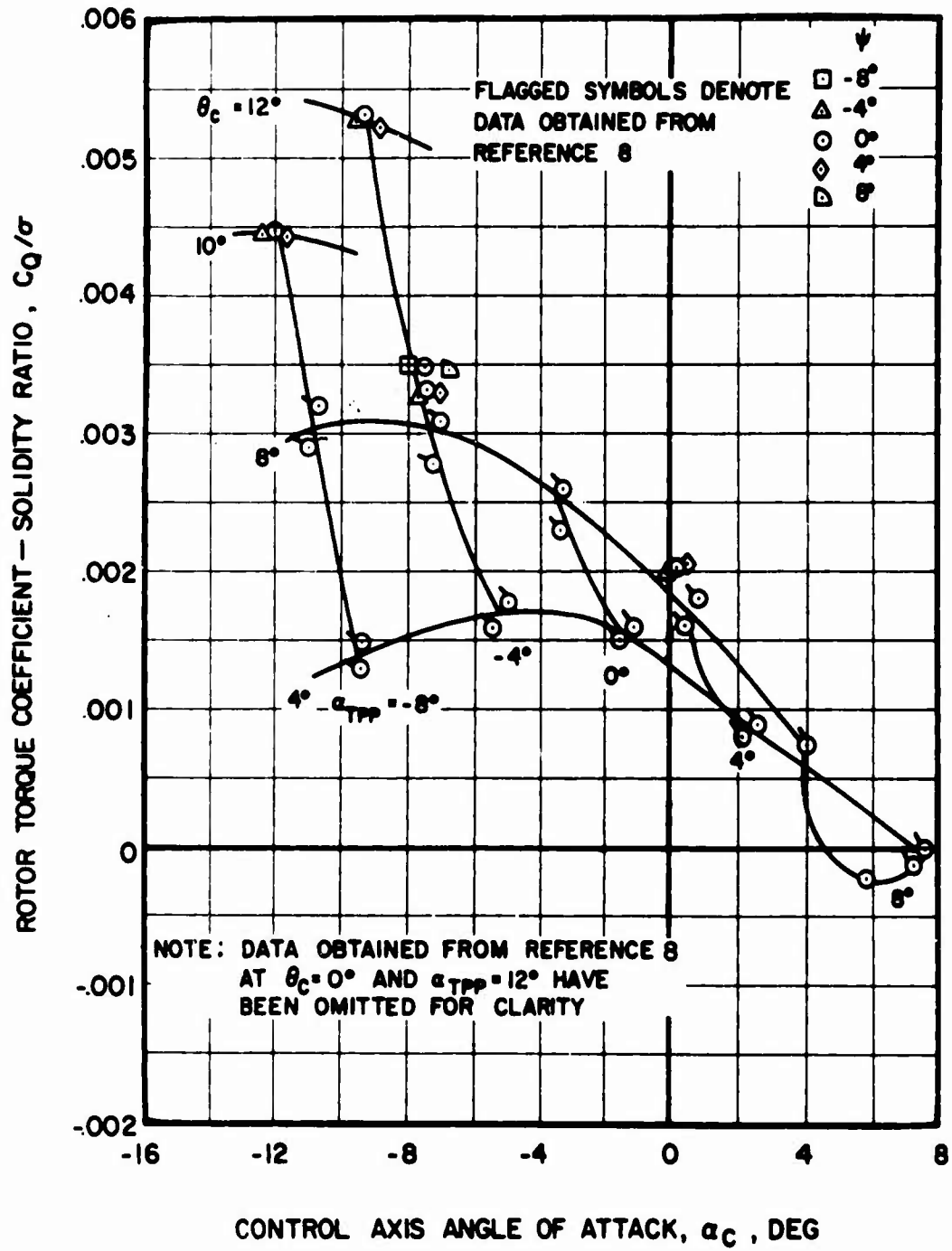


Figure 21. Concluded

TABLE I. SUMMARY OF ROTOR OPERATING CONDITIONS

V, Knots	V _S , Knots	Approximate μ, μ*	N, rpm
60	120	0.3	742
100	200	0.5	716
125	250	1.0	447
150	300	1.0	529
105**	105**	.27**	205**

** These operating conditions are for the full scale data, Reference 1.

TABLE II. SUMMARY OF MODEL CONFIGURATIONS

Designation	Description of Model Configuration
F	Fuselage with no rotor head attached but with the rotor shaft protruding
FH	Fuselage with the rotor head attached
FHC	Fuselage with the rotor head and the "cap" rotor head fairing attached
FHf	Fuselage with the rotor head and the ellipsoidal rotor head fairing attached
FHB	FH with rotor blades
FHBf	FHf with rotor blades

TABLE III. AREAS ENCLOSED BY THE $C_p = 0.9$ CONTOURS FOR VARIOUS ROTOR HEAD CONFIGURATIONS AT TWO FUSELAGE ATTITUDES	
Configuration	Area Enclosed by the $C_p = 0.9$ Contour
	$\alpha_f = 0.0^\circ, \psi = 0.0^\circ$ $\alpha_f = -4.0^\circ, \psi = -4.0^\circ$
F	1.30 ft ² (69% of FH) 1.25 ft ² (71% FH)
FH	1.88 ft ² (100% of FH) 1.76 ft ² (100% of FH)
FHC	2.22 ft ² (118% of FH) 1.90 ft ² (108% of FH)
FHf	2.09 ft ² (111% of FH) 1.50 ft ² (85% of FH)

TABLE IV. DATA RUN SUMMARY

Configuration	Simulated Forward Speed, Knots	Wake Survey Data Run Number	Rotor Performance Data Run Number	Blade Stress Data Run Number
F	120	34	34	-
	200	35	35	-
	300	36	36	-
FH	120	-	87	-
	200	7	87	-
	300	8	87	-
FHC	120	10	88	-
	200	11	88	-
	300	12	88	-
FHf	120	21	89	-
	200	22	89	-
	300	26	89	-
FHB	120	14,15	61	61
	200	16	62	62
	250	41	41	41
FHBf	120	28,29,30	29,30	29,30
	200	31	31	31
	250	33	33	33

APPENDIX II

WAKE SURVEY DATA TABLES

TABLE V. WAKE PRESSURE COEFFICIENTS AND FLOW DIRECTIONS AT THREE FORWARD SPEEDS AND VARIOUS FUSELAGE ATTITUDES, CONFIGURATION F																						
RUN 34 POINT 5 ALPHA=.0 DEG PSI=-.0 DEG V=60 KNOTS q=11.4 PSF. CONFIGURATION F				RUN 34 POINT 6 ALPHA=-8.0 DEG PSI=-.0 DEG V=60 KNOTS q=11.4 PSF. CONFIGURATION F																		
.96	.98	.94	1.03	.93	.95	1.07	.94	.96	.99	.95	.93											
	.94	.98	.96	.96	.92		.91	.98	.96	.98	.91											
.98	.96	.93	.96	1.01	.89	1.06	.95	.97	.96	1.06	.91											
	.95	1.00	.96	1.07	.93		.99	.99	.94	1.07	.94											
.98	.94	.97	.96	1.09	.94	1.00	.97	1.02	.96	1.08	.94											
	.93	.94	.91	1.05	.97		.98	1.01	.98	1.03	.92											
.95	.98	.83	.98	1.04	1.02	.95	.98	.98	1.00	1.02	.94											
	.98	.77	.92	.98	1.08		.99	.95	1.01	.98	.94											
.93	.93	.90	.65	.95	1.07	.88	.97	.93	1.00	.91	.98											
	1.07	.60	.84	.96	1.06		.95	.92	1.01	.90	.90											
.96	.92	.73	.76	.92	1.02	.95	1.02	1.02	1.01	.94	1.02											
	.93	.77	.59	.98	.99		.97	1.00	.74	.95	.95											
1.00	.96	.94	.54	.47	.95	.94	.87	.79	.85	.50	1.08											
	.96	.88	RAKE MOUNTING POINT	.84	.92		1.10	.82	.74	1.04	1.08											
1.13	1.00	.75	RAKE MOUNTING POINT	.84	.91	.98	1.07	.57	1.13	1.12	1.06											
	1.01	.93	RAKE MOUNTING POINT	1.09	.93		1.00	.64	.98	1.07	1.00											
1.05	.99	.88	.36	.96	.99	1.02	.94	.48	.99	1.05	1.07											
	.95	.71	.62	.80	1.09		.97	.64	.88	1.02	1.01											
1.01	.98	.98	.60	.95	1.06	1.07	1.00	.76	.82	.98	1.01											

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\rho q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\rho q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE V - Continued

RUN 34 POINT 7 V=60 KNOTS ALPHA=-4.0 DEG PSI=-.0 DEG q=11.4 PSF. CONFIGURATION F		RUN 34 POINT 8 V=60 KNOTS ALPHA=4.0 DEG PSI=-.0 DEG q=11.4 PSF. CONFIGURATION F	
.98	.95	.94	.99
.99	.96	.96	.96
.97	.97	.97	.97
.96	.94	.98	.96
.97	.92	.97	.94
1.01	.96	.96	.96
1.00	.97	.96	.98
1.00	.91	.92	.95
1.01	.89	.99	.90
.95	.81	.98	.85
.98	.56	.98	1.08
.95	.74	.95	1.06
.99	.58	.93	1.02
1.10	1.05	1.01	1.08
1.08	1.09	1.07	1.12
.99	.95	1.01	1.10
1.08	.90	.97	.74
.94	.93	.96	.84
		.99	1.00
			.94
			.95

RUN 34 POINT 7 V=60 KNOTS ALPHA=-4.0 DEG PSI=-.0 DEG q=11.4 PSF. CONFIGURATION F		RUN 34 POINT 8 V=60 KNOTS ALPHA=4.0 DEG PSI=-.0 DEG q=11.4 PSF. CONFIGURATION F	
.98	.95	.94	.99
.99	.96	.96	.96
.97	.97	.97	.97
.96	.94	.98	.96
.97	.92	.97	.94
1.01	.96	.96	.96
1.00	.97	.96	.98
1.00	.91	.92	.95
1.01	.89	.99	.90
.95	.81	.98	.85
.98	.56	.98	1.08
.95	.74	.95	1.06
.99	.58	.93	1.02
1.10	1.05	1.01	1.08
1.08	1.09	1.07	1.12
.99	.95	1.01	1.10
1.08	.90	.97	.74
.94	.93	.96	.84
		.99	1.00
			.94
			.95

RUN 34 POINT 7 V=60 KNOTS ALPHA=-4.0 DEG PSI=-.0 DEG q=11.4 PSF. CONFIGURATION F		RUN 34 POINT 8 V=60 KNOTS ALPHA=4.0 DEG PSI=-.0 DEG q=11.4 PSF. CONFIGURATION F	
.98	.95	.94	.99
.99	.96	.96	.96
.97	.97	.97	.97
.96	.94	.98	.96
.97	.92	.97	.94
1.01	.96	.96	.96
1.00	.97	.96	.98
1.00	.91	.92	.95
1.01	.89	.99	.90
.95	.81	.98	.85
.98	.56	.98	1.08
.95	.74	.95	1.06
.99	.58	.93	1.02
1.10	1.05	1.01	1.08
1.08	1.09	1.07	1.12
.99	.95	1.01	1.10
1.08	.90	.97	.74
.94	.93	.96	.84
		.99	1.00
			.94
			.95

RUN 34 POINT 7 V=60 KNOTS ALPHA=-4.0 DEG PSI=-.0 DEG q=11.4 PSF. CONFIGURATION F		RUN 34 POINT 8 V=60 KNOTS ALPHA=4.0 DEG PSI=-.0 DEG q=11.4 PSF. CONFIGURATION F	
.98	.95	.94	.99
.99	.96	.96	.96
.97	.97	.97	.97
.96	.94	.98	.96
.97	.92	.97	.94
1.01	.96	.96	.96
1.00	.97	.96	.98
1.00	.91	.92	.95
1.01	.89	.99	.90
.95	.81	.98	.85
.98	.56	.98	1.08
.95	.74	.95	1.06
.99	.58	.93	1.02
1.10	1.05	1.01	1.08
1.08	1.09	1.07	1.12
.99	.95	1.01	1.10
1.08	.90	.97	.74
.94	.93	.96	.84
		.99	1.00
			.94
			.95

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_{\infty}}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FURN MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_{\infty}}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FURN MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_{\infty}}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FURN MODEL NOSE, LOOKING AFT.

TABLE V - Continued

RUN 34 POINT 9 ALPHA= 0.0 DEG PSI= -0 DEG V=60 KNOTS q= 11.4 PSF. CONFIGURATION F										RUN 34 POINT 10 ALPHA= 4.0 DEG PSI=-4.0 DEG V=60 KNOTS q= 11.4 PSF. CONFIGURATION F									
.99	.98	.96	.91	.97	1.00	1.01	.96	.95	.94	1.03	1.05	.96	.98	.95	.94	.97	1.03	1.05	
1.00	1.00	.82	.93	.97	9.0	1.04	.96	.98	.97	5.0	1.09	.96	.98	.95	.97	5.0	1.09		
.94	.94	.84	.79	.93	-1.0	.94	.95	.99	.93	-3.4	1.01	.95	.99	.93	1.07	-3.4	1.01		
.96	.96	.95	.74	1.01	1.11	.94	.99	.90	.96	1.01	.98	.99	.90	.96	1.01	1.13	.98		
1.02	1.02	1.07	.50	.92	1.09	.94	.85	1.00	.97	.96	1.14	.98	.85	1.00	.97	.96	1.14	.98	
.98	.98	1.08	.84	.95	1.14	.94	.80	1.06	.98	1.01	1.13	.80	1.06	.98	1.01	1.13	1.01		
1.04	1.05	.93	.60	.94	1.06	.96	.80	.82	.97	1.00	1.09	.80	.82	.97	1.00	1.09	1.01		
.96	.96	.79	1.03	.94	1.01	.94	.82	.97	.83	1.01	1.04	.82	.97	.83	1.01	1.04	1.04		
1.01	1.03	4.6	.93	.99	7.5	6.4	1.01	.81	.95	4.2	.98	1.01	.81	.95	4.2	.98	4.5		
1.0	1.0	-1.0	1.21	.99	-5.8	1.1	.95	-4.1	.71	-5.3	.98	.95	-4.1	.71	-5.3	.98	-2.0		
1.00	1.00	1.01	.92	1.24	.89	.99	.84	.68	.71	1.13	.98	.84	.68	.71	1.13	.98	.98		
1.06	1.05	.93	.62	1.06	.98	1.02	.84	1.02	.29	1.16	1.04	.84	1.02	.29	1.16	1.04	1.04		
.88	.93	.93	.64	.96	.94	.94	.71	.78	.70	1.17	.99	.71	.78	.70	1.17	.99	.99		
1.02	.97	.43	.60	.84	1.00	1.07	.96	.66	.44	1.18	1.13	.96	.66	.44	1.18	1.07	1.13		
1.03	1.03	.62	RAKE MOUNTING POINT	.78	1.01	.94	.82	1.01	RAKE MOUNTING POINT	1.10	1.12	1.03	1.03	.62	RAKE MOUNTING POINT	1.10	1.12		
1.17	.99	7.1	.69	.69	1.07	1.13	.60	.1	.1	2.5	4.2	.99	.60	.1	2.5	4.2	1.18		
.98	.98	4.7	.69	.69	-2.5	1.1	-4.2	3.2	3.2	-4.9	-1.4	.98	-4.2	3.2	-4.9	-1.4	1.18		
1.05	.96	.68	.57	.76	1.11	1.15	.78	.97	1.04	1.19	1.12	.96	.78	.97	1.04	1.19	1.12		
1.02	1.02	.74	.13	.67	1.11	1.11	.99	1.07	.62	1.00	1.12	1.02	.99	1.07	.62	1.00	1.12		
1.01	.99	.93	.41	1.04	1.06	1.06	.94	.91	.81	.98	1.11	.99	.94	.91	.81	.98	1.11		

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS C_p $\frac{1}{2} \rho V^2$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE V - Continued

RUN 34 POINT 11 ALPHA=-4.0 DEG PSI=-4.0 DEG V=60 KNOTS q = 11.4 PSF. CONFIGURATION F										RUN 34 POINT 12 ALPHA=8.0 DEG PSI=-4.0 DEG V=80 KNOTS q = 11.4 PSF. CONFIGURATION F																			
.97	.90	.98	.95	.99	1.06	.94	.98	.93	.98	.99	.98	.98	.99	.92	.68	.96	.94	.97	.96	.99	1.00	.98	.98	.99	.98	.99	.92	.68	
.96	.94	1.01	.97	.97	1.11	.94	.97	-2.9	1.06	.94	.94	.99	-8.3	.87	.97	.96	1.06	.89	.88	1.01	.98	.99	1.00	.97	-8.3	.87	.87	.46	
.96	.95	.97	.97	.92	1.06	.96	.94	-4.0	1.04	.96	.96	.95	-5.1	.79	.94	.94	1.06	.94	.94	1.02	.94	.94	.94	.88	.81	.81	.79	.96	
.98	.94	.97	.96	.94	1.01	.96	.94	1.01	1.01	.96	.96	.95	.82	.82	.92	.92	1.02	.98	.99	1.02	.96	.96	.95	.82	.82	.82	.96	.96	
.97	.95	.97	.99	1.00	.96	1.00	.99	1.00	.99	.94	.94	.94	.85	.85	.92	.94	.99	.99	.99	1.00	.94	.94	.94	.85	.85	.85	.94	.94	
.97	.89	.91	.98	1.06	.92	.92	.92	-4.3	.94	1.07	.98	.89	-8.9	.89	.93	.91	1.01	1.01	.89	.98	.89	.89	.89	.89	-8.9	.89	.94	.94	
.94	.88	1.00	.97	1.08	.94	.94	.97	-3.9	.96	.96	.95	.88	-4.1	.93	.93	.92	.99	.99	.88	.95	.88	.88	.88	.93	-4.1	.93	.94	.94	
.91	.63	.77	.95	1.11	.96	.96	.95	1.11	.96	1.10	.95	.81	-4.4	.98	.92	.93	1.04	1.04	.81	.99	.81	.81	.81	.98	-4.4	.98	.85	.85	
.94	.81	1.01	.96	1.10	.95	1.10	.96	1.10	.95	1.10	.95	.92	-8.4	.89	.92	.98	.99	.99	.78	1.00	.78	.78	.78	.95	-8.4	.89	.89	.89	
.91	.77	1.01	.84	.98	1.09	1.10	.96	1.06	1.01	1.01	.92	.93	.98	.98	.94	.93	.92	.92	.80	.99	.80	.80	.80	.98	.98	.98	.89	.89	
1.06	.92	.66	.66	.99	1.08	.99	.99	-3.6	1.08	1.06	.84	.94	.98	.85	.94	.94	.98	.98	1.08	1.08	.81	.81	.81	.90	.90	.90	.85	.85	
1.01	.72	.59	.59	.90	1.08	.90	.90	-5.1	1.01	1.06	.84	.94	.98	.85	1.05	.80	.82	.82	RAKE MOUNTING POINT	.85	.85	.85	.85	.87	-7.2	.87	.80	.80	
.95	.69	.66	.66	.95	1.01	.95	.95	-1.5	1.01	.96	.84	.94	.98	.85	.96	.76	.46	.46	RAKE MOUNTING POINT	.96	.96	.96	.96	.79	-4.9	.79	.80	.80	
1.01	1.07	.42	.47	.97	1.02	1.03	.47	.97	1.02	1.03	1.03	1.04	.43	.53	1.04	.64	.43	.43	.53	.99	.99	.99	.99	.82	.82	.82	.83	.83	
.95	1.06	.77	.82	.97	1.02	1.02	.82	.97	1.02	1.02	1.02	.83	.58	.40	.97	.83	.58	.58	.40	.89	.89	.89	.89	.83	.83	.83	.83	.83	
.95	.89	.66	.84	1.01	.92	.96	.84	1.01	.92	.96	.96	.89	.70	.87	.97	.60	.70	.70	.87	.96	.96	.96	.96	.87	.87	.87	.87	.84	.84

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS C_p $\frac{P}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT
 TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

TABLE V - Continued

RUN 34 POINT 15 ALPHA= 4.0 DEG PSI= 4.0 DEG V= 6.0 KNOTS q = 11.4 PSF. CONFIGURATION F										RUN 34 POINT 16 ALPHA= 0 DEG PSI= 8.0 DEG V= 6.0 KNOTS q = 11.4 PSF. CONFIGURATION F										
.96	.99	1.01	.94	.99	1.01	.99	.94	.91	.95	.91	.98	1.00	.99	1.12	.97	.95	1.00	.97	1.02	1.10
.99	.97	1.03	.98	.94	3.3	1.10	.98	1.06	.96	1.01	1.01	.97	1.02	1.10	.96	1.00	1.00	1.00	1.00	1.06
1.02	.99	.98	.88	.94	2.2	1.17	.90	1.17	.94	.99	.99	1.01	.99	1.04	1.01	.99	1.01	1.00	1.02	1.01
1.04	1.03	1.00	.98	.90	1.04	.94	.73	1.05	.97	.98	1.00	1.04	1.02	1.01	1.01	1.03	1.04	1.00	1.03	.95
1.01	1.02	.96	.79	.91	1.05	.99	.91	1.05	.88	.96	.96	.98	1.11	.94	1.04	1.11	.98	.98	1.11	.94
1.02	1.03	.98	1.09	.89	.88	.94	.89	.88	.92	1.00	1.00	1.00	1.09	.95	1.09	1.00	1.00	1.00	1.09	.95
1.01	1.02	1.06	.77	.85	-1.7	.96	.85	1.5	.76	-3.2	1.06	1.00	1.10	.2	1.0	1.00	1.00	1.00	1.10	.94
1.02	1.00	1.08	.76	.87	.1	.87	1.12	2.6	-4.6	-6.2	.91	.87	1.12	-7.7	-6.6	-10.2	.91	.87	1.12	.97
1.02	.99	1.01	.89	1.12	1.08	1.00	1.12	1.08	.71	.78	.78	.58	1.02	1.05	1.09	1.03	.78	.58	1.02	1.05
1.03	.97	.82	.51	.43	1.03	1.00	.43	1.03	.71	.61	.48	.57	1.05	1.11	1.09	.61	.48	.57	1.05	1.11
1.07	.95	.64	1.02	.96	1.01	1.10	.96	1.01	.89	.71	.71	.67	.95	1.14	1.10	.94	.71	.67	.95	1.14
1.17	.98	1.03	.77	.77	1.28	.94	.77	1.28	1.06	.88	.96	RAKE MOUNTING POINT	.91	1.08	.8	-1.6	-2.8	RAKE MOUNTING POINT	.96	.0
1.07	1.01	1.04	.66	.66	1.07	1.15	.66	1.07	1.06	.96	1.00	1.00	.96	1.07	1.07	-3.6	4.7	1.07	.95	-3.8
1.01	.99	.99	.83	.83	1.13	1.07	.83	1.13	.98	.96	.95	.95	.95	1.07	1.02	.96	.95	.95	.95	1.07
	.99	.98	.57	.57	.57	1.14	.57	.57	.98	.16	.16	1.08	1.02	1.01	1.02	1.02	.16	1.08	1.02	1.01
	1.00	1.01	.33	.33	.53	1.04	.33	.53	.71	.66	.66	1.02	1.04	.95	1.02	1.02	.66	1.02	1.04	.95
	.96	.99	.78	.78	1.04	1.07	.78	1.04	.89	.76	.76	.81	1.09	.95	1.00	.62	.76	.81	1.09	.95

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE 15 SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE 16 SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE V - Continued

RUN 34 POINT 17 ALPHA= .0 DEG PSI=4.0 DEG V=60 KNOTS q= 11.4 PSF. CONFIGURATION F				RUN 34 POINT 18 ALPHA= .0 DEG PSI= 4.0 DEG V=60. KNOTS q= 11.4 PSF. CONFIGURATION F			
.96	.89	.98	.93	.99	1.11	.94	.96
	.92	.98	.97	.96	1.12		.98
.97	.96	.99	.94	1.2	1.09	.96	.96
	.94	.98	.96	-3.5	1.08		
.96	.93	.96	.95	.97	1.05	.99	.98
	.96	.93	1.00	1.02	1.01	.97	.97
.93	.97	.99	1.01	1.03	.96	1.01	.99
	.94	.93	.94	1.11	.93	.97	.82
1.00	-1.4	-4.3	.95	.9	-2	-1.7	-2.0
	-4.5	-3.1	.98	-3.9	-2.2	4.5	3.3
.94	.74	.71	.91	1.06	.95	.91	.89
	.65	.90	.86	1.13	1.01	.58	.60
.94	1.01	.72	.43	1.01	1.05	.57	.83
.90	.94	1.03	.21	.99	1.12	1.00	.75
	.94	.60	RAKE MOUNTING POINT	.94	1.13	.86	RAKE MOUNTING POINT
1.06	-3.1	-2.4	.44	-1.1	1.2	.7	1.07
	-4.2	1.9	.70	-5.2	-2.0	-4.3	1.04
1.01	1.11	.41	.97	1.00	1.10	.91	.88
	.77	.66	.90	.99	1.04	1.01	.71
.95	.94	.75	.71	1.03	.96	.99	.80
						1.08	.67

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE V - Continued

RUN 34 POINT 19 ALPHA = 0 DEG PSI = 8.0 DEG V = 60 KNOTS q = 11.4 PSF. CONFIGURATION F					RUN 35 POINT 5 ALPHA = 0 DEG PSI = -0 DEG V = 100 KNOTS q = 30.7 PSF. CONFIGURATION F							
.98	.94	1.01	1.00	.98	1.10	.95	.99	1.00	.98	.96	.98	.99
.99	.92	.99	.99	1.02	1.12	.97	.99	.98	.98	.96	1.00	.99
	.98	1.00	.98	.92	.5	.97	.98	.98	.96	1.00	.6	.99
	.99	1.00	.95	1.00	1.4	1.08	.99	.99	.99	.99	-7	.99
.97	.98	.97	1.01	.97	1.09	.98	1.00	1.00	.98	.95	1.04	.95
	.99	1.01	1.00	.99	1.02	1.02	.99	.99	.96	.97	1.02	.99
.96	1.00	.97	1.03	1.08	.96	.99	.98	.98	.94	.94	1.01	.99
	.99	.90	.97	1.08	.51	.99	.99	.99	.93	.93	1.00	.99
.99	.3	1.1	.98	-2.1	-3.7	1.05	.99	-2.2	.74	1.11	.6	.99
	.98	5.0	.98	.63	.94	1.5	.99	.2	.74	1.00	1.00	.99
	.94	.99	.91	.90	.87	.99	.99	.91	.62	.97	.95	.99
.99	.95	.94	.93	1.06	.46	1.10	.99	1.00	1.01	1.06	.96	.99
	.98	.96	.63	1.00	.88	.99	.99	.99	.79	.97	1.09	.99
.98	1.04	.98	.74	.91	.60	1.10	.99	.99	.77	.67	.95	1.01
	1.00	1.17	RAKE MOUNTING POINT	.54	1.03	.99	1.02	.6U	RAKE MOUNTING POINT	.52	1.07	.99
1.09	.95	.99	RAKE MOUNTING POINT	1.03	1.10	1.12	.99	.6U	RAKE MOUNTING POINT	1.03	.98	1.03
	.99	1.01	RAKE MOUNTING POINT	1.06	1.08	1.06	1.03	1.04	RAKE MOUNTING POINT	.97	1.06	.99
1.02	.97	.97	.92	1.04	.85	1.06	.98	1.05	.51	.96	1.03	1.03
.99	1.01	.98	.18	.33	.91	.99	.98	1.05	.32	.95	1.03	.99
	.98	.96	.85	.39	.80	1.01	.99	1.01	.16	.87	1.03	1.02

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT-SURVEY
TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.



TABLE V - Continued

RUN 35 POINT 6 ALPHA=-8.0 DEG PSI=-0 DEG V=100. KNOTS q=30.7 PSF. CONFIGURATION F		RUN 35 POINT 7 ALPHA=-4.0 DEG PSI=-0 DEG V=100. KNOTS q=30.7 PSF. CONFIGURATION F				
.97	.97	.98	.97	.99	.99	.94
.97	.97	.97	.98	.98	.97	.99
.98	.96	.94	.93	.94	.99	.97
.99	.94	.94	.90	1.00	.95	.98
.96	.91	.97	.92	.99	.96	.98
.96	.98	.96	.91	.90	1.00	.93
.98	.96	.97	.92	.98	1.01	1.01
.96	.98	.97	.93	1.01	1.00	.96
.96	.94	.93	.95	.96	.96	.99
.96	.91	.93	.97	.96	.96	.99
.98	.94	.91	.98	.84	.97	1.00
.98	.94	.85	.98	.85	.76	.96
.95	1.14	.90	.93	.74	.61	.98
.96	.91	.85	.96	.90	.90	.94
1.00	.56	.76	.91	RAKE MOUNTING POINT	.80	.98
1.00	.68	.65	.91	.94	.82	.96
.98	.17	.88	.94	.75	.93	.96
.99	.73	.98	.92	.67	.79	.93
.96	.68	.88	.94	.84	.87	.94
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$		NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$				
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.		NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.				
NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.		NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.				

TABLE V - Continued

RUN 35 POINT 6 ALPHA= 4.0 DEG PSI= -0.0 DEG V= 100 KNOTS q = 30.7 PSF. CONFIGURATION F					RUN 35 POINT 9 ALPHA= 8.0 DEG PSI= -0.0 DEG V= 100 KNOTS q = 30.7 PSF. CONFIGURATION F							
.98	.97	1.01	.99	.97	1.02	.96	.98	.95	1.01	.99	.99	.95
.94	.94	.98	.98	.99	1.02	.99	.98	.95	.88	.95	1.01	.99
.97	1.00	4.3	.96	4.3	1.03	.99	1.00	1.00	.98	.69	9.7	1.03
.99	.93	-1.0	.94	-1.0	1.01	.98	.81	.81	.96	.97	-1.7	1.02
.98	.95	1.05	.97	1.01	1.01	.98	1.00	.85	.49	1.06	1.00	.95
.99	.70	.83	.96	1.01	1.01	.98	.94	.81	.88	.84	1.01	.98
.95	.80	.93	.96	.96	.96	.98	1.01	.92	.78	.94	.48	.98
.92	.87	1.05	.82	1.05	1.03	.98	1.00	1.01	.86	1.04	1.00	.98
.97	.88	.80	.59	3.3	1.01	3.2	6.3	2.6	5.9	1.01	7.4	1.02
.94	.94	1.01	.91	1.0	.95	-0	.1	.8	-2.0	.92	.5	.92
.98	1.01	.82	1.04	.79	.98	1.02	.97	.98	.48	1.07	.36	1.03
.99	.32	.49	.53	.49	1.05	1.02	1.01	.95	.52	.80	.96	.96
.96	1.14	.34	.48	1.03	1.02	1.04	.95	.98	.74	.45	1.17	1.04
1.03	.98	.57	.67	.91	1.02	4.2	6.2	10.0	.60	RAKE MOUNTING POINT	.68	1.03
1.01	1.01	.94	.91	4.3	1.05	.8	-1.2	.2	.57	9.8	.63	7.9
.98	.95	.80	.99	-3.5	1.04	.6	1.02	.77	1.05	-4.0	.86	2.5
.98	.98	.77	.57	1.02	1.02	1.02	1.01	1.02	.94	.37	1.08	1.02
1.00	1.00	.99	1.19	1.01	1.01	1.01	.99	.94	.44	.97	.99	.99
.99	.70	1.00	.59	1.00	.99	1.01	1.00	.87	.31	.97	.96	.97

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE V - Continued

RUN 35 POINT 10 ALPHA=4.0 DEG PSI=4.0 DEG V=100 KNOTS q = 30.7 PSF. CONFIGURATION F						RUN 35 POINT 11 ALPHA=4.0 DEG PSI=4.0 DEG V=100 KNOTS q = 30.7 PSF. CONFIGURATION F					
.96	.99	.91	.97	.96	1.00	.98	1.00	.98	.98	.97	.94
.97	.97	.96	.98	.96	1.01	.98	.98	.99	.99	.97	.91
.97	.93	.97	.98	1.02	1.04	.98	.96	.97	.98	.98	.94
.98	.94	.96	.95	.98	1.03	.97	.97	.99	.98	.98	.93
.98	1.03	.97	.98	.98	1.04	.98	.98	.99	.98	.98	.98
.99	.88	1.00	.95	.99	1.06	.95	.95	.97	.97	.95	.95
.99	.66	.91	.99	.93	.99	.99	.99	.98	.98	.95	.97
.99	.99	.84	.80	1.05	1.01	.97	.97	.97	.99	.95	.99
.99	.97	.91	1.13	1.05	1.00	.99	.96	.98	.96	.93	.97
.98	.83	.67	.69	1.06	.99	.99	.80	1.01	1.01	.94	.94
.98	.92	.82	.55	.94	1.03	.97	.88	.74	.95	.91	.92
.99	.94	.91	1.04	1.06	.90	.70	.70	.61	.53	.93	.96
1.00	.99	.92	.64	.98	1.01	.98	.66	.61	.69	.86	.93
1.05	.99	.91	RAKE MOUNTING POINT	1.12	.98	RAKE MOUNTING POINT	.64	.71	.97	.97	.91
1.03	.75	.68	3.3 -4.3	3.5 -5.5	1.08	1.04	.97	.53	.98	.98	.94
1.01	.76	.77	.76	1.00	1.05	1.02	1.02	.69	.59	.98	.95
				1.01	1.06	1.00	.98	.73	.70	.94	.98
				1.02	1.04	1.00	.98	.81	.93	.96	.99

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_i - P_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW
MODEL NOSE, LOOKING AFT.

RAKE
MOUNTING
POINT

74.75 IN

TABLE V - Continued

RUN 35 POINT 12 ALPHA=-4.0 DEG PSI= 4.0 DEG V=100. KNOTS q = 30.7 PSF. CONFIGURATION F				RUN 35 POINT 13 ALPHA= 4.0 DEG PSI= 4.0 DEG V= 100. KNOTS q = 30.7 PSF. CONFIGURATION F			
.95	.97	.97	.98	.99	.97	.98	.98
.99	.98	.97	.98	1.02	.98	1.00	.97
.99	.98	.96	.98	1.02	.97	1.01	.95
.99	.98	.96	.96	1.04	.96	.98	.99
.99	.98	.97	.96	1.02	.98	.84	.99
.99	.98	.97	.97	1.01	.92	.79	.80
.99	1.00	.98	.96	.99	.89	.61	.98
.99	1.00	.95	.98	.99	.80	.80	.92
.98	1.00	.99	.79	.96	.87	.93	1.03
.99	.96	.96	.96	.96	.83	.84	.84
.99	.99	.94	.67	.72	.82	1.04	.46
.98	.96	.98	.86	.67	.54	1.01	1.08
.98	.96	.99	.87	.89	.53	.74	1.05
1.03	.99	.82	.71	.95	RAKE MOUNTING POINT	1.01	.97
.98	.97	.96	.51	1.00	1.03	.92	1.00
1.01	.96	1.00	1.02	1.05	1.02	.69	.69
.99	.98	.95	.70	1.02	.39	1.00	.99
.99	.99	.99	.75	1.06	.84	.45	.58
.99	.98	.96	.69	1.00	.97	.64	.75

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

7.475 IN

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

7.475 IN

TABLE V - Continued

RUN 35 POINT 15 ALPHA = 0 DEG PSI = 0.0 DEG V = 100. KNOTS q = 30.7 PSF. CONFIGURATION F		RUN 35 POINT 14 ALPHA = 0 DEG PSI = 0.0 DEG V = 100. KNOTS q = 30.7 PSF. CONFIGURATION F	
.98	.98	.97	.98
.96	.96	.97	.97
.98	.95	.98	1.01
.99	.94	.98	.97
.98	.94	.99	.96
.93	.97	.98	1.02
.85	.98	.99	1.00
.68	.97	.99	1.00
.87	1.00	1.01	1.01
.75	.92	.87	1.02
.48	.96	1.00	1.04
.73	.92	.71	1.03
.69	.99	.87	.99
.64	.96	.94	.98
.94	RAKE MOUNTING POINT	RAKE MOUNTING POINT	1.06
.91	1.01	1.04	.98
.94	.98	1.04	1.05
.80	.91	.85	1.04
.69	.94	.93	1.03
	1.02	.88	1.03
	1.00	.93	1.03

RUN 35 POINT 15 ALPHA = 0 DEG PSI = 0.0 DEG V = 100. KNOTS q = 30.7 PSF. CONFIGURATION F		RUN 35 POINT 14 ALPHA = 0 DEG PSI = 0.0 DEG V = 100. KNOTS q = 30.7 PSF. CONFIGURATION F	
.98	.93	.97	.99
.96	.93	.97	.99
.98	.95	.98	1.03
.99	.94	.98	1.04
.98	.94	.99	1.02
.93	.97	.98	1.02
.85	.98	.99	1.00
.68	.97	.99	1.00
.87	1.00	1.01	1.01
.75	.92	.87	1.02
.48	.96	1.00	1.04
.73	.92	.71	1.03
.69	.99	.87	.99
.64	.96	.94	.98
.94	RAKE MOUNTING POINT	RAKE MOUNTING POINT	1.06
.91	1.01	1.04	.98
.94	.98	1.04	1.05
.80	.91	.85	1.04
.69	.94	.93	1.03
	1.02	.88	1.03
	1.00	.93	1.03

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE V - Continued

RUN 35 POINT 16 ALPHA= .0 DEG PSI= 4.0 DEG VE=100. KNOTS q = 30.7 PSF. CONFIGURATION F				RUN 35 POINT 17 ALPHA= .0 DEG PSI= 8.0 DEG VE=100. KNOTS q = 30.7 PSF. CONFIGURATION F							
.98	1.00	.99	1.00	.99	.94	1.01	.97	.98	.96	.96	1.02
1.00	.98	.97	.93	.97	.93		.98	.98	.95	.96	
.98	.99	.99	.95	.99	.95	.99	.99	.98	1.00	.97	.99
.99	.98	.97	.93	.99	.93		.98	.98	1.02	2.1	1.01
.98	1.01	1.01	.95	.99	.95	.98	.99	.94	1.01	1.02	.97
.99	.98	1.02	.95	1.02	.95		1.01	.98	1.00	1.01	
.98	1.02	1.00	.94	.86	.94	.96	.98	1.01	.99	.99	.95
.99	.99	.97	1.03	.73	1.03		.99	.98	.96	.98	.59
.96	.99	.97	.58	.71	.58	.95	.98	1.01	.92	.83	-4.8
.99	.99	.93	.88	.77	.88		1.1	5.2	4.0	1.5	
.96	.99	1.08	.93	.65	1.01	.95	.98	.96	.61	1.09	.98
.99	.99	.96	.82	.85	.82		.96	.98	1.01	.88	
1.00	1.01	1.12	.89	.50	.95	.95	.96	1.01	.70	1.08	.97
1.01	1.01	.89	.92	.98	.92		.97	.89	.96	.97	
1.05	1.00	.99	.97	.88	.97	.93	1.05	.99	.39	1.02	.99
1.00	1.00	.96	.84	1.03	.84		.97	.97	1.06	1.05	
1.01	.99	1.01	1.04	.91	1.04	.95	1.03	.99	.70	.71	1.05
1.00	1.00	1.03	.85	.32	.85		.97	.93	.90	1.06	
1.00	1.00	1.01	.92	.74	.92	.98	1.00	.95	.82	.37	1.03
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_s}{q}$				NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_s}{q}$							
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.				NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.							
NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.				NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.							

TABLE V - Continued

RUP 36 POINT 5 V=150 KNOTS		ALPHA=0 DEG Q=68.7 PSF. CONFIGURATION F		PSI=-0 DEG		RUP 36 POINT 6 V=150 KNOTS		ALPHA=-4.0 DEG Q=68.7 PSF. CONFIGURATION F		PSI=-0 DEG	
.98	.97	.94	.95	.94	.96	.99	.98	.94	.97	.96	1.00
.98	.97	.95	.97	.97	.96	.98	.98	.97	.97	.97	.97
.98	.98	1.03	.97	.6	.98	.96	.99	.97	-3.5	.96	.98
.99	1.00	1.00	.96	-9	.98	.99	.97	.97	-1.3	.98	.97
.99	.97	.96	.96	.98	.98	.98	.97	.97	.97	.98	.96
.99	.99	.93	.98	.98	.99	.98	1.00	.94	.97	.99	.96
.99	.97	.98	.98	.98	1.01	.98	.97	.97	.97	.99	.96
.96	1.01	.98	.90	.98	.98	.97	.97	.97	.98	1.01	.96
.96	.99	.89	.78	-1.1	.98	.98	.97	.95	-4.9	.98	.96
.97	.99	1.01	.92	-1.3	1.01	.98	.97	.95	-4	.98	.96
.97	1.01	1.07	.70	1.05	.98	.99	.97	.91	1.06	.96	.98
.99	.96	.80	.80	.98	1.01	1.00	1.00	.71	.91	.96	.96
.99	.93	.97	.56	.94	.96	.98	.99	.73	.34	.95	.96
1.01	.97	1.11	RAKE MOUNTING POINT	.69	.97	.99	.99	.63	.99	.99	.99
.99	.97	.97	.97	.97	.97	1.01	.99	.82	-5.4	.99	.99
.99	1.01	1.03	1.02	-1.7	.97	.99	.99	.49	-1.4	.96	.99
.99	.95	.91	1.00	.97	1.01	1.00	.99	.84	.99	.99	.99
.99	.99	.93	.99	1.00	1.00	1.01	1.01	.92	.99	1.01	.98
.98	.97	.99	1.01	1.03	1.00	.98	.98	.85	.97	1.00	.98

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_\infty}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

74.75 IN

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_\infty}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

74.75 IN

TABLE V - Continued

RUN 36 POINT 7 ALPHA=-2.0 DEG PSI=-0 DEG V=150 KNOTS q= 68.7 PSF. CONFIGURATION F		RUN 36 POINT 8 ALPHA= 2.0 DEG PSI= -0 DEG V=150 KNOTS q= 68.7 PSF. CONFIGURATION F	
.96	.97	.99	.98
.98	.99	.97	.98
.99	.98	.95	.99
.98	1.00	.99	1.01
.99	1.00	.97	.95
1.00	.98	.96	.88
.99	.94	.96	.79
1.01	.96	.98	1.01
.99	.91	.98	.93
.99	.90	.99	.58
.99	.83	.92	.99
1.01	.75	.98	.47
.97	1.08	.95	.42
.97	.97	.97	.48
1.00	.92	.98	1.18
1.00	.78	1.01	.91
1.00	.64	1.03	.96
1.00	.70	1.00	1.02
.96	.84	.98	.99
			.96
			1.06
			1.02
			1.03

RUN 36 POINT 7 ALPHA=-2.0 DEG PSI=-0 DEG V=150 KNOTS q= 68.7 PSF. CONFIGURATION F		RUN 36 POINT 8 ALPHA= 2.0 DEG PSI= -0 DEG V=150 KNOTS q= 68.7 PSF. CONFIGURATION F	
.96	.97	.99	.98
.98	.99	.97	.98
.99	.98	.95	.99
.98	1.00	.99	1.01
.99	1.00	.97	.95
1.00	.98	.96	.88
.99	.94	.96	.79
1.01	.96	.98	1.01
.99	.91	.98	.93
.99	.90	.99	.58
.99	.83	.92	.99
1.01	.75	.98	.47
.97	1.08	.95	.42
.97	.97	.97	.48
1.00	.92	.98	1.18
1.00	.78	1.01	.91
1.00	.64	1.03	.96
1.00	.70	1.00	1.02
.96	.84	.98	.99
			.96
			1.06
			1.02
			1.03

RUN 36 POINT 7 ALPHA=-2.0 DEG PSI=-0 DEG V=150 KNOTS q= 68.7 PSF. CONFIGURATION F		RUN 36 POINT 8 ALPHA= 2.0 DEG PSI= -0 DEG V=150 KNOTS q= 68.7 PSF. CONFIGURATION F	
.96	.97	.99	.98
.98	.99	.97	.98
.99	.98	.95	.99
.98	1.00	.99	1.01
.99	1.00	.97	.95
1.00	.98	.96	.88
.99	.94	.96	.79
1.01	.96	.98	1.01
.99	.91	.98	.93
.99	.90	.99	.58
.99	.83	.92	.99
1.01	.75	.98	.47
.97	1.08	.95	.42
.97	.97	.97	.48
1.00	.92	.98	1.18
1.00	.78	1.01	.91
1.00	.64	1.03	.96
1.00	.70	1.00	1.02
.96	.84	.98	.99
			.96
			1.06
			1.02
			1.03

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

74.75 IN

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

74.75 IN

TABLE V - Continued

RUN 36 POINT 9 ALPHA= 4.0 DEG PSI= -2.0 DEG V= 150. KNOTS q = 68.7 PSF. CONFIGURATION F				RUN 36 POINT 10 ALPHA= 4.0 DEG PSI= -2.0 DEG V= 150. KNOTS q = 68.7 PSF. CONFIGURATION F			
.98	.98	.96	.97	1.00	.98	.97	.96
.97	.98	.97	.98	.99	.98	.95	.97
.96	.97	.97	4.6	.99	.95	4.8	.96
.96	.97	.97	-8	.99	.93	-2.8	.98
.98	.98	.96	1.00	.96	.94	1.01	.99
.99	.99	.74	1.00	.96	.70	.97	.98
.98	.99	.81	.99	.97	.88	1.00	.98
.96	.96	.82	.96	.97	.86	.98	.96
.98	.97	1.02	2.9	.99	.81	1.02	.97
.97	.97	1.02	-1.4	.99	1.09	3.8	.94
.98	.97	1.03	.96	.96	.49	-2.9	.99
.96	.96	.92	.98	1.01	.95	.96	.99
.85	.85	.79	.52	.99	.70	.62	.97
.99	.90	1.04	.95	.99	.91	.96	.99
.98	.98	.82	.92	.92	.58	1.00	1.02
.94	.87	.98	3.3	1.02	RAKE MOUNTING POINT	3.4	4.4
.98	.98	.92	-2.6	1.02	2.1	-3.6	.99
.99	1.02	.80	1.02	1.00	4.1	1.01	1.01
.95	.85	.50	.99	1.00	.74	.96	.99
.97	.96	.79	1.00	1.00	.62	.96	.99

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

74.75 IN

RAKE MOUNTING POINT

TABLE V - Continued

RUN 36 POINT 11 ALPHA=2.0 DEG PSI=2.0 DEG VE=150 KNOTS q=68.7 PSF. CONFIGURATION F				RUN 36 POINT 12 ALPHA=2.0 DEG PSI=2.0 DEG VE=150 KNOTS q=68.7 PSF. CONFIGURATION F			
.99	.97	.94	.96	.97	.96	.99	.98
.96	.96	.95	.97	.98	.99	.98	.97
.97	.96	.94	.97	.97	.98	.98	.96
.99	.99	.93	.94	.99	.99	.98	.97
.98	.97	.96	.98	.98	.97	.98	.96
.80	.81	.93	.96	.98	.97	.98	.96
.96	.90	.71	.96	.99	.91	.93	.94
.86	.86	.72	.92	.96	.67	1.00	.96
.96	.86	.88	.97	.97	.79	.99	.94
.97	.93	.88	.97	.92	.70	.74	.97
.99	1.05	.97	.87	.89	.83	.96	.95
.99	.97	.52	.98	.95	.87	1.12	1.01
.95	1.00	.18	.68	.97	.83	.51	.98
1.01	.95	.91	.99	.97	.39	RAKE MOUNTING POINT	.99
.99	.93	.61	.96	1.01	.80	⊕	.99
.98	.89	.81	.95	.97	.72	⊖	.98
				1.00	.50	1.08	.94
				.99	.61	.42	.96

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$
 NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
 TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW
 MODEL NOSE, LOOKING AFT.

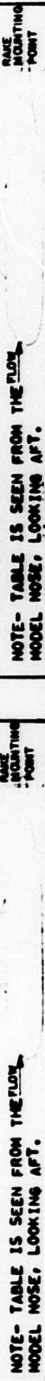


TABLE V - Continued

RUN 36 POINT 13 ALPHA=2.0 DEG PSI=-1.0 DEG VE 150 KNOTS $q = 68.7$ PSF. CONFIGURATION F				RUN 36 POINT 14 ALPHA=-0.0 DEG PSI=-1.0 DEG VE 150 KNOTS $q = 68.7$ PSF. CONFIGURATION F			
.98	.98	.97	.95	.97	.96	.96	1.00
1.00	.98	.97	2.3	.95	.98	.95	.96
.98	.98	.98	.9	1.00	.96	1.8	.96
.98	.97	1.01	.97	.98	.97	-4.6	.95
.99	.96	.48	.98	.97	.97	.98	.95
.99	.99	.85	1.00	.96	.97	1.00	.97
1.01	.84	.68	.91	.91	.98	.97	.96
.99	.99	.66	.49	.74	1.01	.99	.96
.98	1.00	.93	.5	1.02	1.00	.5	.97
.98	.98	.88	.7	.84	.92	-2.1	.99
.99	.98	.88	.92	.84	.98	-3.5	.99
1.02	1.11	.74	.97	.71	.86	.96	.96
.93	.72	.56	.96	.95	.93	1.07	.95
1.01	.94	.59	.91	.94	.77	1.01	.97
1.10	.78	RAKE MOUNTING POINT	1.02	.96	.81	1.05	.97
1.9	.5	.91	1.5	-2.3	-2.7	1.06	.95
.95	.77	.44	-2.6	-4.5	1.7	.96	.97
.99	1.02	.44	.97	1.01	.88	.96	.97
1.00	1.05	.57	.98	.71	.75	.99	.99
.96	.94	.55	.96	.66	.76	.99	.99
.96	1.02	.80	.95	.87	.91	1.01	.98

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

7475M

7475M

TABLE V - Continued

RUN 36 POINT 15 ALPHA= .0 DEG PSI=-2.0 DEG VE=150. KNOTS q = 68.7 PSF. CONFIGURATION F					RUN 36 POINT 16 ALPHA= .0 DEG PSI=-2.0 DEG VE=150. KNOTS q = 68.7 PSF. CONFIGURATION F				
.99	.97	.98	.98	.97	.98	.96	1.01	.99	.98
.98	.99	.97	.99	.97	.98	.97	.98	1.01	.98
.98	.99	.98	.99	1.0	.98	.98	1.00	1.00	.98
.98	.97	.98	.97	-2.8	.96	.96	1.00	1.00	.97
.99	.97	.95	.99	.95	.97	.97	.98	.97	.95
.99	.99	.94	.97	.95	.96	.96	.99	.93	.94
.96	.96	.87	.98	.94	.98	.98	.99	.80	.63
1.00	1.00	.81	1.10	.97	.96	.96	.98	.78	.86
.97	.87	-3.7	.7	-2	.98	-2.0	.98	.93	.46
.97	-1.9	-2.2	-3.3	-1.4	.98	1.7	.94	.95	.94
.96	.92	.74	.57	1.01	.97	.98	.96	.88	.87
.96	.96	.92	.90	.97	.97	1.00	1.00	.74	.76
1.00	.98	.96	.41	1.00	.99	.99	.88	.78	.68
1.00	1.00	.60	.46	.98	.97	.97	1.10	.74	.70
.96	.96	.84	.97	.97	.99	.99	1.09	.50	.50
1.01	-1.2	-1.8	.1	.4	1.00	-1.4	.97	RAKE MOUNTING POINT	.4
1.00	-3.1	2.2	-9.0	-2	.98	1.05	1.13	RAKE MOUNTING POINT	.89
1.00	1.03	.58	.94	.96	.98	.92	.91	.92	.92
.98	.98	.63	.82	.98	1.01	1.01	1.03	.76	.91
.99	1.05	.64	.51	.96	.98	1.00	1.00	.47	.96
					.98	.61	1.12	.76	.98

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_\infty}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER ± VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER ± HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

7475M

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_\infty}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER ± VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER ± HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

7475M

TABLE V - Concluded

RUN 36 POINT 17 ALPHA= .0 DEG PSI= 4.0 DEG
 V= 150. KNOTS q= 68.7 PSF. CONFIGURATION F

.99	.97	.98	.97	.96	.98	.98
	.97	.98	.96	.98	.98	
.99	.97	.99	.98	1.01	.2 3.3	1.00
	.99	.98	.97	.98		1.01
1.01	.98	.98	.97	.96		1.01
	.99	.98	.97	.91		.95
.99	.98	.98	.79	.91		.99
	.98	.99	.98	.73		1.02
1.01	.1 3.0	1.03	.1 2.9	.92	.77	.83
		1.00	.98	.64	.68	.81
			.98	.68	2.5	.89
.98		1.00	1.07	.46	.76	.68
		.98	.70	.71	1.03	.96
.98		.97	.89	.59	1.00	.74
		.98	1.07	.59	.59	.86
.99	-.6 2.0	1.00	1.01	RAKE MOUNTING POINT	.69	.97
		.95	.97		.87	.99
1.00		1.00	1.02		.61	.89
		.99	.98		.56	.89
.98		.99	.99	1.15	.56	.91
						.99

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.



NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

TABLE V. WAKE PRESSURE COEFFICIENTS AND FLOW DIRECTIONS AT TWO FORWARD SPEEDS AND VARIOUS FUSELAGE ATTITUDES, CONFIGURATION FH

RUN 7 POINT 5 V=100 KNOTS		ALPHA=-0 DEG PSI=-0.0 DEG CONFIGURATION FH		RUN 7 POINT 6 V=100 KNOTS		ALPHA=8.0 DEG PSI=-0.0 DEG CONFIGURATION FH	
.98	1.01	1.00	.99	.98	.99	.98	.99
.99	1.00	.96	.98	.97	.97	.98	.97
.99	.98	1.00	.98	.97	.97	.98	.97
.99	.98	1.02	1.00	1.01	1.01	.98	.99
.99	.98	1.02	.98	1.00	.98	.98	.98
.98	.98	.98	.96	.96	.96	.96	.96
.78	1.02	.96	1.09	.93	.98	.96	.96
.99	.99	.95	.94	.60	.99	.99	.96
.99	.22	.71	.87	.90	.97	1.00	.99
.98	1.00	.91	.57	.61	.74	.99	.99
.98	.86	.71	.02	1.16	.52	.99	.99
1.00	1.00	.34	.34	1.03	.99	.96	.99
1.00	1.00	1.01	.47	.85	.98	.97	1.00
1.05	.95	.52	.34	.95	.95	.69	.99
1.03	1.00	.74	.81	.94	.94	.79	1.00
1.03	.98	.71	.83	1.02	1.02	.97	1.00
1.01	1.03	.97	.58	.98	.96	.96	.96

RUN 7 POINT 5 V=100 KNOTS		ALPHA=-0 DEG PSI=-0.0 DEG CONFIGURATION FH		RUN 7 POINT 6 V=100 KNOTS		ALPHA=8.0 DEG PSI=-0.0 DEG CONFIGURATION FH	
.98	1.01	1.00	.99	.98	.99	.98	.99
.99	1.00	.96	.98	.97	.97	.98	.97
.99	.98	1.00	.98	.97	.97	.98	.97
.99	.98	1.02	1.00	1.01	1.01	.98	.99
.99	.98	1.02	.98	1.00	.98	.98	.98
.98	.98	.98	.96	.96	.96	.96	.96
.78	1.02	.96	1.09	.93	.98	.96	.96
.99	.99	.95	.94	.60	.99	.99	.96
.99	.22	.71	.87	.90	.97	1.00	.99
.98	1.00	.91	.57	.61	.74	.99	.99
.98	.86	.71	.02	1.16	.52	.99	.99
1.00	1.00	.34	.34	1.03	.99	.96	.99
1.00	1.00	1.01	.47	.85	.98	.97	1.00
1.05	.95	.52	.34	.95	.95	.69	.99
1.03	1.00	.74	.81	.94	.94	.79	1.00
1.03	.98	.71	.83	1.02	1.02	.97	1.00
1.01	1.03	.97	.58	.98	.96	.96	.96

RUN 7 POINT 5 V=100 KNOTS		ALPHA=-0 DEG PSI=-0.0 DEG CONFIGURATION FH		RUN 7 POINT 6 V=100 KNOTS		ALPHA=8.0 DEG PSI=-0.0 DEG CONFIGURATION FH	
.99	1.01	1.00	.99	.98	.99	.98	.99
.99	.99	.96	.98	.97	.97	.98	.97
.99	.98	1.00	.98	.97	.97	.98	.97
.99	.98	1.02	1.00	1.01	1.01	.98	.99
.99	.98	1.02	.98	1.00	.98	.98	.98
.98	.98	.98	.96	.96	.96	.96	.96
.78	1.02	.96	1.09	.93	.98	.96	.96
.99	.99	.95	.94	.60	.99	.99	.96
.99	.22	.71	.87	.90	.97	1.00	.99
.98	1.00	.91	.57	.61	.74	.99	.99
.98	.86	.71	.02	1.16	.52	.99	.99
1.00	1.00	.34	.34	1.03	.99	.96	.99
1.00	1.00	1.01	.47	.85	.98	.97	1.00
1.05	.95	.52	.34	.95	.95	.69	.99
1.03	1.00	.74	.81	.94	.94	.79	1.00
1.03	.98	.71	.83	1.02	1.02	.97	1.00
1.01	1.03	.97	.58	.98	.96	.96	.96

RUN 7 POINT 5 V=100 KNOTS		ALPHA=-0 DEG PSI=-0.0 DEG CONFIGURATION FH		RUN 7 POINT 6 V=100 KNOTS		ALPHA=8.0 DEG PSI=-0.0 DEG CONFIGURATION FH	
.99	1.01	1.00	.99	.98	.99	.98	.99
.99	.99	.96	.98	.97	.97	.98	.97
.99	.98	1.00	.98	.97	.97	.98	.97
.99	.98	1.02	1.00	1.01	1.01	.98	.99
.99	.98	1.02	.98	1.00	.98	.98	.98
.98	.98	.98	.96	.96	.96	.96	.96
.78	1.02	.96	1.09	.93	.98	.96	.96
.99	.99	.95	.94	.60	.99	.99	.96
.99	.22	.71	.87	.90	.97	1.00	.99
.98	1.00	.91	.57	.61	.74	.99	.99
.98	.86	.71	.02	1.16	.52	.99	.99
1.00	1.00	.34	.34	1.03	.99	.96	.99
1.00	1.00	1.01	.47	.85	.98	.97	1.00
1.05	.95	.52	.34	.95	.95	.69	.99
1.03	1.00	.74	.81	.94	.94	.79	1.00
1.03	.98	.71	.83	1.02	1.02	.97	1.00
1.01	1.03	.97	.58	.98	.96	.96	.96

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_s}{\rho q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_s}{\rho q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

TABLE VI - Continued

RUN 7 POINT 7 ALPHA=-4.0 DEG PSIF=-0.0 DEG V=100. KNOTS q= 32.1 PSF. CONFIGURATION FH				RUN 7 POINT 8 ALPHA= 4.0 DEG PSIF=-0.0 DEG V=100. KNOTS q= 32.1 PSF. CONFIGURATION FH			
.99	1.01	.97	.96	.98	1.00	.98	.99
.99	.99	.98	.99	.98	.98	1.00	1.01
1.00	.98	.99	.97	.97	1.00	.96	.96
.99	.99	.98	.99	.99	1.00	.95	.99
.99	.99	.99	.98	.97	1.00	.95	.96
.77	1.00	1.00	1.02	1.03	1.06	.81	.99
.99	.99	1.00	.99	.99	1.00	.61	.90
.99	.99	.99	1.01	.75	.60	.96	.64
.96	1.01	.84	1.02	.96	.83	.89	.97
1.00	.93	.71	.77	.77	.68	.71	.90
1.00	.89	.78	.77	.86	1.01	.49	.48
1.04	.88	.39	.96	.97	1.25	.41	.83
1.04	.77	.82	.84	.84	.76	.28	.83
1.04	.31	.81	.56	.97	.97	RAKE MOUNTING POINT	.94
1.04	.48	.60	.34	.35	1.03	1.07	1.00
1.04	.74	.56	1.02	1.05	.86	.89	.95
1.02	1.00	.95	1.01	1.03	1.01	.58	1.01
1.02	.95	.61	1.01	1.03	1.01	.98	.99
						.91	.98

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH TO RIGHT LOOKING AFT.

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH TO RIGHT LOOKING AFT.

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

TABLE VI - Continued

RUN 7 POINT 11 ALPHA=4.0 DEG MSI=4.0 DEG V=100. KNOTS q=32.1 PSF. CONFIGURATION FH				RUN 7 POINT 12 ALPHA=4.0 DEG MSI=4.0 DEG V=100. KNOTS q=32.1 PSF. CONFIGURATION FH			
.99	.98	.99	.99	.97	.99	1.00	.98
.97	1.03	1.04	.99	.98	1.00	1.00	.97
.96	.98	.98	5.2	1.03	1.01	.9	.99
.84	.97	.98	-4.1	1.00	.98	2.8	.99
.81	.95	.94	.97	.99	.99	1.03	.94
.80	.81	.81	.97	1.08	1.05	.72	.95
.75	.91	.41	1.01	.99	.93	.66	.82
.56	.47	.92	.79	1.01	.85	.46	1.03
-1.9	-2.5	.80	4.0	4.3	3.4	-3.1	1.2
-5.8	-4.8	.64	.92	3.0	.98	4.6	.81
1.12	.64	.43	1.00	1.05	2.2	.69	.64
.88	.80	.77	.56	1.02	.71	.44	.94
.85	.52	.56	.64	1.03	.81	.71	1.00
.49	.86	.97	.21	1.05	.50	.74	.89
1.03	.58	.90	.90	1.00	.95	RAKE MOUNTING POINT	.93
5.3	2.2	.98	4.0	3.8	2.7	1.0	6.0
-6.2	2.8	.99	-5.3	1.3	-7.7	-2.9	4.2
.51	.81	1.00	1.02	1.04	1.00	.38	.85
.58	.64	.99	.99	1.00	1.01	.68	.42
.72	.90	.53	1.01	1.00	.94	.95	.89
1.04	.93	.95	.99	1.02	.94	.80	1.05

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\rho V^2}$
 NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.
 NOTE - TABLE IS SEEN FROM THE FLOW.
 MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\rho V^2}$
 NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.
 NOTE - TABLE IS SEEN FROM THE FLOW.
 MODEL NOSE, LOOKING AFT.

TABLE VI - Continued

RUN 7 POINT 15 ALPHA= .0 DEG PSI=4.0 DEG V=100. KNOTS ϕ = 32.1 PSF. CONFIGURATION FH									
.99	1.00	.94	1.01	.98	.99	1.01	.97	.96	.98
.97	.98	.98	1.00	.98	1.00	1.00	.94	.94	.98
1.00	.98	.97	1.00	1.00	1.00	1.00	.99	.98	1.00
.99	.99	.98	.99	.99	.98	.99	.94	.97	.98
1.00	.97	.88	.98	.95	.99	.99	.96	.94	.95
.78	.86	1.01	.47	.98	.99	.99	.93	.94	.92
.71	.95	.93	.99	.96	.99	.99	.96	1.01	.76
.59	.94	.94	.81	.90	.98	.98	.96	1.01	.70
1.00	.79	.94	.43	.94	1.00	.1	.94	.93	.50
.62	.62	.98	.02	.96	1.06	-3.5	2.7	1.2	2.1
1.02	.88	.80	.57	1.05	1.00	1.01	.98	.82	.39
.69	.92	.52	.55	.98	1.18	1.00	1.04	.61	.90
.95	.92	.94	.47	1.06	1.04	1.00	.93	.98	.77
1.12	.95	.80	RAKE MOUNTING POINT	1.08	.98	.7	.86	.98	.79
1.02	.87	.73	.94	.97	1.00	1.02	1.01	.95	.75
1.02	.78	.51	.77	.99	.99	.99	.98	1.02	.74
	.81	.56	.82	1.00	1.02	1.00	1.00	1.00	.26

RUN 7 POINT 16 ALPHA= .0 DEG PSI=4.0 DEG V=100. KNOTS ϕ = 32.1 PSF. CONFIGURATION FH									
.97	.96	.98	.99	.98	.98	.98	.97	.98	.98
.97	.94	.94	.97	.97	.97	.98	.97	.98	.98
.96	.94	.98	.97	.97	.98	.98	.97	.98	.98
.96	.96	.94	.97	.97	.98	.98	.97	.98	.98
.77	.93	.94	.95	.92	.97	.92	.95	.97	.98
.98	.96	1.01	.76	.67	.92	.92	.92	.92	.98
.95	.96	1.01	.70	.77	.99	.99	.77	.99	.99
.95	.94	.93	.50	.70	.88	.88	.70	.88	.99
.95	.77	.93	.75	.62	.82	.82	.62	.82	.97
	.98	.82	.39	.56	.98	.98	.56	.98	.97
	1.04	.61	.90	.38	.38	.38	.38	.38	.97
	.93	.98	.91	.77	1.00	1.00	.77	1.00	1.01
1.01	.86	RAKE MOUNTING POINT	.93	.79	.93	.93	.79	.93	.97
1.00	1.03	.97	1.06	1.06	1.19	1.19	1.06	1.19	.97
1.00	.95	.96	.78	.78	.91	.91	.78	.91	.97
	1.01	.95	.65	.75	1.07	1.07	.75	1.07	.99
	.98	1.02	.52	.74	.99	.99	.74	.99	.99
	1.00	1.00	.26	.65	.92	.92	.26	.65	.97

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS C_p . $\frac{P-P_0}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS C_p . $\frac{P-P_0}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

TABLE VI - Continued

RUN 7 POINT 17 ALPHA = -0 DEG PSI = 8.0 DEG V = 100. KNOTS q = 32.1 PSF. CONFIGURATION FH										RUN 8 POINT 5 ALPHA = -0 DEG PSI = -0 DEG V = 150. KNOTS q = 70.5 PSF. CONFIGURATION FH									
.97	.96	.9A	.98	1.00	.99	1.00	.97	.98	.97	.97	.9A	.97	.98	1.01	.98	.97	.97	.98	.97
.99	.9A	1.01	1.01	.99	.96	.99	.96	1.01	.98	.99	.96	.98	.97	.98	.97	.98	.97	.98	.98
.98	.98	1.02	1.03	.96	.99	.96	.99	.98	.99	.98	.99	.98	.99	.98	.97	.96	.98	.98	.98
.98	.99	.9A	1.00	1.04	.99	.99	.98	.98	.99	.94	.9A	.97	.96	.97	.96	.96	.98	.98	.98
.83	.97	.95	1.03	.88	.88	.88	.88	.88	.88	.95	.94	.43	.93	.98	.93	.96	.98	.98	.98
.97	.94	.9A	1.00	1.05	.48	.48	.48	.48	.48	.97	.93	.77	.98	1.00	.98	.96	.98	.98	.98
.97	.97	1.07	.49	1.02	.37	.37	.37	.37	.37	1.02	.74	.84	.65	.65	.91	.90	.90	.90	.90
.98	.97	1.10	.66	.88	.90	.90	.90	.90	.90	.81	1.02	.72	.73	.73	.96	.96	.96	.96	.96
.99	.99	1.0A	.88	.43	.44	.44	.44	.44	.44	.84	.82	.82	.93	.93	.97	.96	.96	.96	.96
.99	.98	.94	.77	.48	.27	.27	.27	.27	.27	.97	.80	.33	.98	.98	1.13	.96	.96	.96	.96
.99	1.00	.97	.53	.66	.18	.18	.18	.18	.18	.56	.59	.24	1.02	1.02	.84	.96	.96	.96	.96
1.02	.90	1.00	RAKE MOUNTING POINT	.53	.36	.36	.36	.36	.36	1.04	.66	RAKE MOUNTING POINT	1.17	1.17	.55	.96	.96	.96	.96
1.00	.99	1.05	.91	.91	.70	.70	.70	.70	.70	1.07	.46	.25	.25	.25	1.01	.99	.99	.99	.99
1.00	1.00	1.04	.52	.52	1.10	1.10	1.10	1.10	1.10	.90	.92	.92	.97	.97	1.15	.96	.96	.96	.96
.97	1.01	1.01	1.06	.73	1.03	1.03	1.03	1.03	1.03	.99	.91	.48	.60	.60	.95	.99	.99	.99	.99
.97	1.02	1.03	.40	.80	.43	.43	.43	.43	.43	.99	.72	.44	.97	.97	.98	.96	.96	.96	.96
.97	1.01	1.01	1.14	.72	.78	.78	.78	.78	.78	.99	1.07	.43	.96	.96	.97	.96	.96	.96	.96

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE DOWN MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE DOWN MODEL NOSE, LOOKING AFT.

TABLE VI - Continued

RUN 8 POINT 8 ALPHA=2.0 DEG PSI=-.0 DEG VE=150. KNOTS q = 70.5 PSF. CONFIGURATION FH				RUN 8 POINT 9 ALPHA=4.0 DEG PSI=-.0 DEG VE=150. KNOTS q = 70.5 PSF. CONFIGURATION FH			
.98	.97	.94	.97	1.00	1.00	1.02	.98
1.00	1.00	.98	1.00	1.00	1.00	.9A	.99
.98	.98	1.00	.97	.99	.98	.98	.99
.97	.97	.99	.98	.98	.90	.85	.85
.97	.9A	1.02	.95	.96	.82	.89	.92
.90	.90	.87	.88	.94	.80	.84	.91
.94	.94	1.08	.97	.90	.61	.80	.97
.95	.95	.94	.76	.87	.86	.42	.79
4.2	1.01	2.9	-1.5	5.9	.66	.8	.74
-8	.70	-8	.1	-8	1.0	-1.0	1.19
.95	.53	.75	.93	.29	.81	.88	.72
.96	.58	.90	.95	1.03	1.10	.45	1.01
.99	.84	.71	.85	.98	.74	.86	1.16
.88	.72	.72	.58	1.00	.82	RAKE MOUNTING POINT	.76
1.2	1.00	.5	1.6	4.0	2.4	3.9	1.09
-3.1	.99	3.9	-2.6	-2.6	5.6	-3.4	.72
1.02	1.15	.90	.87	1.01	.95	.77	1.02
1.05	.85	.23	1.06	.97	.99	.56	.76
.99	1.03	.94	1.01	1.01	1.01	.29	.88

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW.
MODEL NOSE, LOOKING AFT.

TABLE VI - Continued

RUN 8 POINT 10 ALPHA=2.0 DEG PSI=-2.0 DEG V=150. KNOTS q = 70.5 PSF. CONFIGURATION FH										RUN 8 POINT 11 ALPHA=2.0 DEG PSI=-2.0 DEG V=150. KNOTS q = 70.5 PSF. CONFIGURATION FH									
.98	.99	.96	.97	.99	.98	.90	.98	.99	.98	.98	.99	.99	.96	.97	.99	.98	.99	.98	.98
1.00	.99	.98	.99	1.00	.98	.94	.98	.98	.98	.99	.98	.98	.98	.98	.98	.98	.98	.98	.98
.97	.96	.97	.96	.96	.97	.97	.97	.96	.98	.97	.94	.98	.86	.99	.99	.99	.99	.99	.99
.88	.97	.94	.98	.95	.97	.98	.97	.99	.98	.84	.65	.91	.88	1.00	1.00	1.00	1.00	1.00	.98
.99	.87	.92	.76	.95	.99	.98	.99	.99	.98	.97	.91	.76	.75	1.00	.98	2.0	1.00	1.00	.98
.94	.92	.98	.56	.81	.98	.98	.98	.98	.98	.97	.47	.96	.83	.50	.98	3.4	.98	1.4	.98
.99	.96	.81	.35	.44	1.07	.81	.81	.54	.81	.98	.56	.81	.90	.79	.75	-3.4	.75	.96	.95
.99	.61	.54	.24	1.00	1.16	.97	.95	.93	.98	.95	.51	.87	.44	1.10	1.01	1.01	1.01	1.00	1.00
1.01	.51	.35	.72	1.12	.99	.98	.99	1.15	.98	.99	.94	.63	RAKE MOUNTING POINT	1.02	.81	1.0	.81	.99	.99
.98	1.03	.64	.16	1.01	.97	.94	.97	1.02	.98	1.00	.59	.77	.87	1.03	1.05	2.5	1.01	.98	.98
.98	1.00	.70	.82	.95	.98	.98	.98	1.02	.98	.99	.96	.80	.39	.92	1.00	-1.5	.99	.99	.99

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT 74.75 IN

TABLE VI - Continued

RUN 8 POINT 12 ALPHA= 4.0 DEG MSI=-2.0 DEG VE=150. KNOTS q = 70.5 PSF. CONFIGURATION FH										RUN 8 POINT 13 ALPHA= 2.0 DEG MSI= 2.0 DEG VE=150. KNOTS q = 70.5 PSF. CONFIGURATION FH									
.99	.96	.96	.98	.98	.98	.99	.96	.97	.97	1.01	1.01	.97	.97	.97	.98	.98	.98	.98	.97
.99	1.03	.98	1.00	1.00	1.01	1.01	1.01	1.01	1.01	1.01	1.01	.97	1.00	1.00	1.2	.99	1.00	.98	.98
	1.02	.97	1.03	1.10	1.10	1.10	1.10	1.10	1.04	1.04	1.04	.97	1.00	1.00	-.1	.97	.98	.98	.98
	.99	.95	.86	1.00	.97	.97	.97	.97	.97	.97	.97	1.01	.94	1.00	.76	.97	.98	.99	.99
1.00	.84	.93	.49	.88	.97	.97	.94	.94	.86	.86	.96	.96	1.00	.83	.92	.97	.80	.99	.99
	.45	.74	.65	.85	.96	.96	.96	.96	.97	.97	1.04	1.04	.83	.80	.97	.80	.97	.99	.99
.89	.75	.70	.77	.82	.96	.96	.97	.97	.94	.94	1.00	1.00	.89	.80	.97	.43	.90	.90	.90
	.84	.94	.90	.71	.93	.93	.93	.93	.94	.94	.61	.61	.78	.75	.99	.75	.99	.96	.96
1.02	.76	.70	.37	.36	.94	.94	4.3	1.01	1.02	1.02	1.06	.61	.61	.78	1.2	.89	.96	.96	.96
	.50	.81	.96	.88	.92	.92	-1.5	1.01	.95	.95	.28	.59	.59	1.10	1.7	1.04	.94	.94	.94
1.04	.81	.93	.50	.95	.85	.85	1.00	1.00	1.05	1.05	.80	.48	.48	.63	1.00	.94	.94	.94	.94
	.48	.72	.78	1.02	.85	.85	.85	.85	.85	.85	.43	.64	.64	.77	1.12	.94	.94	.94	.94
1.00	.86	.38	.32	1.23	.99	.99	.97	.97	1.01	1.01	1.00	.59	.59	.76	1.05	.90	.90	.90	.90
	.89	.87	RAKE MOUNTING POINT	.92	1.06	1.06	RAKE MOUNTING POINT	.97	1.00	1.00	.52	RAKE MOUNTING POINT	RAKE MOUNTING POINT	.75	1.03	.90	.90	.90	.90
1.03	.95	.78	1.02	1.02	1.05	1.05	.98	.98	1.01	1.01	.62	.62	.62	1.22	.5	.98	.98	.98	.98
	.81	.72	.94	.94	1.01	1.01	.98	.98	1.01	1.01	1.07	1.07	1.07	.60	-3.2	.99	.99	.99	.99
1.00	.75	.74	.99	.99	1.03	1.03	.97	.97	.93	.93	.95	.36	.36	.89	.91	.99	.99	.99	.99
	.77	.58	.53	1.00	.99	.99	.99	.99	.99	.99	.98	.37	.37	.55	1.09	.99	.99	.99	.99
.98	1.01	.74	.42	.98	1.00	1.00	.98	.98	1.01	1.01	.94	.48	.48	.46	.99	.99	.99	.99	.99

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS C_p . θ IN DEGREES.

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH SURVEY

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS C_p . θ IN DEGREES.

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH SURVEY

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE VI - Continued

RUN 8 POINT 14 ALPHA= .0 DEG ψ SI=-4.0 DEG V=190. KNOTS q = 70.5 PSF. CONFIGURATION FH										RUN 8 POINT 15 ALPHA= .0 DEG ψ SI=-2.0 DEG V=190. KNOTS q = 70.5 PSF. CONFIGURATION FH												
.99	.96	.99	.98	.97	.97	.98	.98	.97	.98	.99	.98	.95	.99	.97	.98	.98	.95	.99	.97	.98	.98	1.01
.97	.98	1.00	.95	.98	2.0	.98	.95	.98	.97	.98	.97	.98	1.01	.97	.98	1.2	.98	.98	1.01	.97	.98	1.00
.95	.95	.94	.98	.97	-4.8	.96	.96	.97	.97	.98	.97	.98	.97	.96	-3.5	.98	.98	.98	1.01	.98	.98	.99
.97	.96	1.00	.99	.97	.99	.99	.96	.99	.98	.97	.98	.97	.96	.94	.97	.98	.97	.98	.96	.98	.98	.99
.88	.91	1.01	.96	.97	.99	.96	.95	.99	.97	.98	.97	.98	.97	.89	.98	.98	.97	.98	.96	.98	.98	.98
.87	.88	.72	1.06	.92	.98	.92	.98	.98	.98	.98	.97	.98	.85	.84	.96	.96	.92	.96	.96	.96	.98	.98
.99	.56	.51	1.09	1.02	.9	1.09	1.09	1.03	.9	.98	.98	.98	.83	.80	1.3	.96	.81	.98	.96	.98	.99	.99
.85	.83	.69	1.09	1.01	-4.6	1.09	1.01	1.01	-3.1	.98	.98	.98	.76	.76	-4.1	.96	.76	.96	.96	.98	.98	.99
.99	.75	.68	.76	1.07	.99	.76	1.07	.99	.99	.99	.99	.99	.61	.26	.94	.61	.26	.94	.94	.94	1.00	1.00
.99	.93	.80	.88	1.00	.99	.88	1.00	.99	.99	.99	.97	.99	.54	.40	1.04	.54	.40	1.04	.98	.98	1.00	1.00
.99	.97	.54	.67	.96	1.00	.67	.96	1.00	.97	.97	.97	.99	.27	.92	.98	.27	.92	.98	.98	.98	1.00	1.00
1.02	1.01	.44	.44	.81	1.02	.44	.81	1.02	.99	.99	.99	.99	.96	.93	1.04	.96	.93	1.04	.96	.96	.97	.97
1.01	.91	.75	.75	.99	.99	.99	.99	.99	.99	.99	.99	.99	.96	.93	.96	.96	.93	.96	.96	.96	.97	.97
.98	.64	.93	.52	1.02	.98	.52	1.02	.98	.97	.97	.97	.97	.92	.85	.98	.92	.85	.98	.98	.98	.98	.99
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$										NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$												
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.										NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.												
NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.										NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.												

TABLE VI - Concluded

RUN C POINT 16 ALPHA = .0 DEG PSI = 2.0 DEG V = 150. KNOTS q = 70.5 PSF. CONFIGURATION FH										RUN B POINT 17 ALPHA = .0 DEG PSI = 4.0 DEG V = 150 KNOTS q = 70.5 PSF. CONFIGURATION FH									
.98	.99	.94	.97	.97	.96	.99	1.01	.97	.98	.97	.99	1.01	.97	.98	.97				
.96	.94	.98	.95	.97	.96	1.01	.99	.99	.97	.98	1.01	.99	.97	.98	.97				
.97	.97	1.00	.97	.98	.96	1.00	.99	.97	.95	.97	1.00	.99	.97	.95	.97				
.97	.97	.98	.97	.97	.98	1.00	1.00	.98	.96	.96	1.00	1.00	.98	.96	1.00				
.89	1.01	.92	.98	.97	.98	.94	.99	.95	.96	.96	.94	.99	.95	.96	.96				
.89	.95	.74	.95	.94	.97	.95	1.01	1.00	.87	.83	.95	1.01	1.00	.87	.83				
.86	.86	.45	.72	.91	.98	.93	.90	1.08	.95	.65	.93	.90	1.08	.95	.65				
1.01	1.01	.60	.54	.79	.98	1.00	.94	.86	.84	.69	1.00	.94	.86	.84	.69				
1.00	1.00	1.04	.36	.76	.96	1.00	.83	1.03	.93	.65	1.00	.83	1.03	.93	.65				
.98	.87	.64	.63	.71	.96	.97	.66	.66	.78	.70	.97	.66	.66	.78	.70				
.94	.94	1.04	.48	1.02	.96	.87	.34	.79	.66	.82	.87	.34	.79	.66	.82				
.98	.86	.94	.59	1.01	.92	1.02	.81	.84	.34	.99	1.02	.81	.84	.34	.99				
.94	.94	1.00	.80	1.12	.94	1.01	1.07	RAKE MOUNTING POINT	1.13	.97	1.01	1.07	RAKE MOUNTING POINT	1.13	.97				
1.03	1.03	.44	1.11	1.05	.91	1.01	.75	1.02	.46	1.07	1.01	.75	1.02	.46	1.07				
1.01	1.01	.78	.83	.97	.94	1.04	.95	1.04	.45	.94	1.04	.95	1.04	.45	.94				
1.01	1.05	.94	.79	1.00	.94	1.04	.95	.62	.63	1.07	1.04	.95	.62	.63	1.07				
1.00	1.00	.86	.92	.97	.94	1.00	1.05	1.01	.78	1.08	1.00	1.05	1.01	.78	1.08				
1.00	1.01	.98	.52	.95	.91	.99	1.05	.90	.23	.80	.99	1.05	.90	.23	.80				

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p \cdot \frac{\rho V^2}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT
74.75 M

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p \cdot \frac{\rho V^2}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT
74.75 M

TABLE VII. WAKE PRESSURE COEFFICIENTS AND FLOW DIRECTIONS AT THREE FORWARD SPEEDS AND VARIOUS FUSELAGE ATTITUDES, CONFIGURATION FHC

RUN 10 POINT 5 V=60 KNOTS		ALPHA=0 DEG PSI=-0.0 DEG q = 11.5 PSF. CONFIGURATION FHC		RUN 10 POINT 6 V=60 KNOTS		ALPHA=8.0 DEG PSI=-0.0 DEG q = 11.5 PSF. CONFIGURATION FHC	
1.02	1.00	1.01	.49	1.00	.94	.97	.99
1.03	1.02	.98	.96	.95	.95	.97	.96
1.00	1.02	.97	.47	.96	.96	.92	.94
.98	.98	.97	1.00	.98	.97	.90	.95
.97	.97	1.00	.98	.96	.94	.95	.94
.99	.99	.94	.47	.97	.94	.94	.94
.93	.93	.84	1.00	.97	.98	.96	.96
.73	.73	.71	.80	.89	.98	.95	.95
1.05	1.05	.82	.45	.39	.95	.96	.94
.51	.51	1.00	1.12	.80	1.00	.92	.95
.47	.47	.97	1.07	.89	.98	.99	.86
.88	.88	.30	.49	.73	.87	.91	.80
1.08	1.08	.45	.85	.83	.86	.16	.01
1.00	1.00	.08	RAKE MOUNTING POINT	1.12	.95	.83	.74
.97	.97	.77	.99	.97	.97	.65	.65
.89	.89	.85	.74	.97	.97	.83	.83
1.02	1.02	.80	.25	1.07	1.00	.33	.33
.96	.96	1.03	.43	.80	1.00	.68	.68
.99	.99	.94	.59	.85	.98	.72	.93

RUN 10 POINT 5 V=60 KNOTS		ALPHA=0 DEG PSI=-0.0 DEG q = 11.5 PSF. CONFIGURATION FHC		RUN 10 POINT 6 V=60 KNOTS		ALPHA=8.0 DEG PSI=-0.0 DEG q = 11.5 PSF. CONFIGURATION FHC	
1.02	1.00	1.01	.49	1.00	.94	.97	.99
1.03	1.02	.98	.96	.95	.95	.97	.96
1.00	1.02	.97	.47	.96	.96	.92	.94
.98	.98	.97	1.00	.98	.97	.90	.95
.97	.97	1.00	.98	.96	.94	.95	.94
.99	.99	.94	.47	.97	.94	.94	.94
.93	.93	.84	1.00	.97	.98	.96	.96
.73	.73	.71	.80	.89	.98	.95	.95
1.05	1.05	.82	.45	.39	.95	.96	.94
.51	.51	1.00	1.12	.80	1.00	.92	.95
.47	.47	.97	1.07	.89	.98	.99	.86
.88	.88	.30	.49	.73	.87	.91	.80
1.08	1.08	.45	.85	.83	.86	.16	.01
1.00	1.00	.08	RAKE MOUNTING POINT	1.12	.95	.83	.74
.97	.97	.77	.99	.97	.97	.65	.65
.89	.89	.85	.74	.97	.97	.83	.83
1.02	1.02	.80	.25	1.07	1.00	.33	.33
.96	.96	1.03	.43	.80	1.00	.68	.68
.99	.99	.94	.59	.85	.98	.72	.93

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\rho q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE VII - Continued

RUN 10 POINT 7 V= 60. KNOTS		ALPHA=-4.0 DEG q= 11.5 PSF. CONFIGURATION FHC		PSI=- .0 DEG		RUN 10 POINT 8 V= 60. KNOTS		ALPHA= 4.0 DEG q= 11.5 PSF. CONFIGURATION FHC		PSI= -.0 DEG	
.97	.96	.93	.98	.98	.96	1.01	.97	.99	.96	.95	.97
.94	.95	.97	.98	1.00	.95	1.03	.95	1.00	.98	.96	.96
.97	.94	.97	.95	.95	.94	.97	.97	.97	.94	.97	.94
.97	.99	1.00	.91	.91	.93	1.01	.94	.96	.98	.95	.95
.84	.99	.98	.93	.90	.94	.84	.94	.94	1.00	.99	.94
.84	.98	1.02	.97	.93	.93	.90	.54	.88	1.03	.94	.98
.97	.94	.97	.95	.95	.91	.81	.81	.78	.91	.94	.98
.97	.98	.95	.97	.94	.92	.72	.67	.77	.28	.93	.99
.97	1.08	.97	.93	.93	.94	1.06	.87	.70	.38	.99	.99
.97	1.3	.87	.84	.84	.94	.0	.73	1.00	.63	1.09	.99
1.00	.89	.71	.57	.57	.67	.89	.52	.90	.42	.90	1.00
.95	.82	.75	.80	.80	.92	.92	.37	.77	.66	.97	.98
.95	.54	.65	.51	.51	.98	.90	.64	.58	1.00	.98	.98
1.11	.80	.63	.39	.39	.98	1.02	.91	RAKE MOUNTING POINT	.87	1.00	.98
1.11	.65	.27	.74	.74	.82	1.00	1.05	RAKE MOUNTING POINT	.70	.98	.98
1.03	.85	.32	.88	.88	.80	.98	.70	RAKE MOUNTING POINT	.99	.96	.96
1.03	1.07	.61	.85	.85	.71	1.01	1.03	.44	1.04	1.01	1.00
.98	1.05	.89	.98	.98	.89	1.02	.96	.29	.66	1.01	.95
.98	.93	1.04	.75	.75	.94	.98	.98	.33	.90	1.00	.95

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH SURVEY
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW
 MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH SURVEY
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW
 MODEL NOSE, LOOKING AFT.

TABLE VII - Continued

RUN 10 POINT 9 ALPHA= 0.0 DEG PSI=-0.0 DEG V= 60 KNOTS q= 11.5 PSF. CONFIGURATION FHC				RUN 10 POINT 10 ALPHA= 4.0 DEG PSI=-4.0 DEG V= 60 KNOTS q= 11.5 PSF. CONFIGURATION FHC			
.92	.90	.94	1.00	.99	1.00	.99	1.00
.95	.90	.80	.92	.99	1.01	.99	1.01
	.90	.87	1.00	.99	1.01	.97	1.01
	.83	.90	1.06	.90	1.03	.94	.94
1.00	.88	.84	.74	.83	.96	1.01	.94
	.52	.90	.80	.93	.94	.98	.95
.79	.81	.84	1.19	.94	.82	1.05	.94
	.73	1.08	.73	.90	1.02	.75	.92
.95	.89	.81	.87	.89	.75	.84	.56
	.93	.80	.85	.85	.40	.56	.56
.95	1.01	.84	.84	.76	.78	.43	.56
	1.00	.90	.80	1.06	.76	.96	.83
.98	.98	.80	.80	1.03	1.11	.51	.64
	.97	1.10	.89	1.01	.96	.86	.94
1.06	.97	.54	1.09	1.00	.93	.86	.93
	1.00	.88	.94	.98	.68	.73	.90
1.01	.98	.94	1.15	.94	.75	.77	.95
	1.00	1.02	.90	.94	.75	.80	1.01
.97	.95	1.01	1.04	1.00	.85	.83	.95
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS, C_p .				NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS, C_p .			
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.				NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.			
NOTE - TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.				NOTE - TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.			

TABLE VII - Continued

RUN 10 POINT 11 ALPHA=-4.0 DEG PSI=-4.0 DEG V=60. KNOTS q = 11.5 PSF. CONFIGURATION FHC										RUN 10 POINT 12 ALPHA=-8.0 DEG PSI=-4.0 DEG V=60. KNOTS q = 11.5 PSF. CONFIGURATION FHC														
.99	.96	.99	.99	.99	.86	.96	.94	.93	.98	.98	1.00	.97	1.02	.99	.95	.91	.97	.97	.97	.97	.97	.97	.97	1.02
.99	.99	1.02	1.01	1.00	1.00	.98	.98	.95	.91	.97	.93	.97	1.03	.99	.95	.91	.97	.97	.97	.97	.97	.97	.97	1.03
.97	.97	.96	.98	1.02	1.00	.99	.93	.93	1.00	.98	1.02	1.02	.99	.93	1.00	.98	.98	.98	.98	.98	.98	.98	.99	
.98	.96	.98	1.00	1.00	.94	.98	.93	.92	.97	.91	1.01	1.03	.99	.92	.91	.97	.97	.97	.97	.97	.97	.97	.99	
.79	.96	.99	.98	.97	.99	.96	.78	.87	1.00	.97	1.02	1.01	.97	.87	1.00	.97	.97	.97	.97	.97	.97	.97	.97	
.91	.98	1.00	1.02	1.02	.95	.98	.91	.91	.97	.97	1.03	1.02	.97	.91	.97	.97	.97	.97	.97	.97	.97	.97	.98	
.91	.98	.97	.92	.92	.96	.97	.93	.96	.99	.99	.98	1.00	.98	.96	.99	.99	.98	.98	.98	.98	.98	.98	.98	
.89	.89	.94	1.00	1.00	1.05	.98	.92	.92	1.00	1.00	.96	1.00	.98	.92	1.00	1.00	1.00	1.00	1.00	1.00	1.00	1.00	1.01	
1.05	.92	.77	.88	.88	.92	.96	.88	.93	1.00	1.00	.94	1.00	.98	.88	1.00	.94	.94	.94	.94	.94	.94	.94	1.01	
1.02	.99	.58	.49	.49	.30	.99	.81	.97	.78	.86	1.03	.97	.99	.97	.78	.86	.86	.86	.86	.86	.86	.86	.99	
	.62	.59	.35	.35	.53	.98	.81	.89	.50	.55	.93	.96	.99	.89	.50	.55	.55	.55	.55	.55	.55	.55	.99	
1.09	.77	.99	RAKE MOUNTING POINT	RAKE MOUNTING POINT	1.00	1.00	1.00	.68	.85	RAKE MOUNTING POINT	.97	1.03	1.03	.68	.85	RAKE MOUNTING POINT	.97	1.03	1.03	.68	.85	RAKE MOUNTING POINT	.97	1.03
1.12	.92	.45	.45	.45	1.00	1.01	1.07	.71	.54	.54	.94	1.01	1.01	.71	.54	.54	.94	1.01	1.01	.71	.54	.54	.94	1.01
1.01	.84	.65	.65	.65	1.00	.99	.92	.92	.54	.54	.96	1.00	.99	.92	.54	.54	.96	1.00	.99	.92	.54	.54	.96	1.00
	.87	.56	.33	.33	.96	.97	1.07	.79	.42	.42	.94	1.01	1.01	.79	.42	.42	.94	1.01	1.01	.79	.42	.42	.94	1.01
	.82	.99	.84	.84	.90	.99	.92	.59	.35	.53	.96	1.00	.99	.59	.35	.53	.96	1.00	.99	.59	.35	.53	.96	1.00
	.88	.88	1.04	1.04	.98	.90	.92	.98	.45	.45	.94	1.01	.99	.98	.45	.45	.94	1.01	.99	.98	.45	.45	.94	1.01

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$
 NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.
 NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$
 NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.
 NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE VII - Continued

RUN 10 POINT 13 ALPHA=-8.0 DEG PSI=4.0 DEG V=60. KNOTS q = 11.5 PSF. CONFIGURATION FHC				RUN 10 POINT 14 ALPHA=-4.0 DEG PSI=4.0 DEG V=60. KNOTS q = 11.5 PSF. CONFIGURATION FHC							
.99	.96	1.01	.98	1.02	.94	.98	1.02	.98	.96	.93	.97
1.01	1.00	1.01	1.00	.97	.96	.97	.95	.94	.99	.96	.95
.98	.98	1.00	1.01	.93	7.3	.97	.97	1.01	.89	8.4	.92
.99	.99	1.01	1.03	.96	3.1	.94	.97	.98	.95	2.5	.92
.99	1.01	1.01	1.01	.94	.93	.95	.97	1.01	.98	.97	.98
1.00	1.00	1.00	1.06	.94	.95	.96	.95	.94	.95	.95	.95
.85	.93	1.02	1.01	.97	.96	.96	.95	.97	.97	.95	.98
	.98	1.03	.98	.96	.98	.98	.94	.98	.98	.92	1.00
1.02	.97	1.01	1.00	.99	9.1	.96	.96	.94	.96	8.5	.96
.98	.93	1.00	1.04	1.00	2.8	1.28	.92	4.2	.82	3.4	6.0
.97	1.03	.94	1.12	.83	.77	.95	.96	.99	.84	.85	.93
.97	1.03	.84	.56	.93	.58	.91	1.00	1.03	.88	.68	.96
	1.04	.73	.55	.72	.93	.91	.97	1.02	.81	.95	.96
	1.15	.90	RAKE MOUNTING POINT	.98	.90	.92	.87	.84	.68	.82	.95
1.11	1.10	.52	RAKE MOUNTING POINT	.65	17.1	-10.9	1.09	1.04	.52	-1.6	.95
	1.9	5.6		.76	-3.4	-24.6	1.01	1.00	.50	2.0	.87
1.07	.90	1.01	.70	.56	.88	.95	1.01	.95	.65	.73	.93
.98	.94	.95	.84	.28	.84	.95	1.03	.94	.83	.68	.94
	1.02	1.14	.32	.58	.81	.95	1.02	.94	.49	.90	.94

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

74.75 IN

TABLE VII - Continued

RUN 10 POINT 15 ALPHA=4.0 DEG PSI=4.0 DEG V=60. KNOTS q=11.5 PSF. CONFIGURATION FHC		RUN 10 POINT 16 ALPHA=0 DEG PSI=8.0 DEG V=60. KNOTS q=11.5 PSF. CONFIGURATION FHC	
1.01	1.01	1.03	1.01
1.02	1.01	.94	.97
1.01	1.01	.99	.94
.95	1.00	.98	.94
1.03	.97	1.00	.94
1.03	.98	.99	.94
.84	1.02	.94	.94
1.00	.92	.71	.90
1.04	1.00	.94	.90
.93	1.03	.98	.99
1.02	.95	.78	.94
1.09	.91	.55	.48
.93	.55	.87	.43
.98	.44	.65	.75
1.03	1.03	1.17	1.01
1.00	1.03	1.08	.93
1.00	1.02	.98	.91
1.00	.98	1.03	.61
1.00	.98	.77	.76
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{q}$		NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{q}$	
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.		NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.	
NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.		NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.	

TABLE VII - Continued

RUN 10 POINT 17 ALPHA = .0 DEG PSI = -4.0 DEG V = 60. KNOTS q = 11.5 PSF. CONFIGURATION FHC				RUN 10 POINT 18 ALPHA = .0 DEG PSI = -4.0 DEG V = 60. KNOTS q = 11.5 PSF. CONFIGURATION FHC			
.51	.50	.97	.98	.97	1.01	.99	.96
.49	.47	1.01	.97	.97	1.00	.99	.96
.51	.52	.97	.95	.98	1.01	1.00	.97
.51	.54	1.00	.97	.92	.99	.99	.95
.36	.50	.94	.96	.92	.94	1.02	.94
.55	.51	.97	.96	.94	.97	.97	.85
.54	.42	.94	.93	.98	.94	1.02	.93
.55	.51	.95	.94	.95	1.03	.86	.91
.54	.53	.94	.94	.91	.93	.78	.91
.64	.55	.99	.94	.99	.82	.73	.92
.60	.54	.90	.96	.90	.94	.56	.92
.53	.97	.87	.95	.69	.99	.43	.93
	.95	.95	.95	.98	.63	.97	1.14
	.56	.45	.97	.99	.59	.59	.94
	.73	.35	.95	.93	.85	.79	.94
	.98	.46	1.00	RAKE MOUNTING POINT	RAKE MOUNTING POINT	1.04	.95
	1.02	.63	.94	.91	.97	.76	1.2
	1.01	.94	.94	5.0	.89	.59	5.2
	.68	.94	.94	.97	.44	.89	.97
	.49	.95	.98	.98	1.11	.37	.70
		.95	.98	.97	.94	.95	.69

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS C_p .
 NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE
 TO RIGHT LOOKING AFT.



NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE VII - Continued

RUN 10 POINT 19 V= 60 KNOTS		ALPHA= .0 DEG q= 11.5 PSF. CONFIGURATION FHC		PSI= 8.0 DEG	
.99	1.01	.99	.96	.98	.97
.99	1.00	.99	.98	.95	
1.00	1.00	.97	1.00	1.02	
.99	.98	1.02	1.02	1.00	
.98	.97	1.00	1.03	.99	.48
.96	.97	.98	.98	.98	
.97	.97	1.00	1.06	.97	1.00
.94	.98	.99	.92	.72	
.95	.97	.91	.72	.78	
.95	.95	.99	.87	.75	
.97	1.02	.46	.59	.52	
.98	.95	.58	.54	.57	
.97	1.00	.56	.54	.73	.90
1.01	.95	RAKE MOUNTING POINT	.20	.80	
.97	.95	RAKE MOUNTING POINT	1.32	.90	
1.10	.95	RAKE MOUNTING POINT	1.28	.76	
1.07	.99	.91	1.14	.68	1.13
.99	.98	1.06	.64	.64	
1.02	1.00	1.07	.71	.98	1.01

RUN 11 POINT 5 V= 100 KNOTS		ALPHA= .0 DEG q= 31.0 PSF. CONFIGURATION FHC		PSI= -0.0 DEG	
.97	1.00	.97	.97	.97	.99
.97	.97	.97	.98	.98	
.95	.97	.48	.98	.98	
.99	.91	.99	.96	.98	
.98	.94	.98	.98	.97	.98
.95	.94	.97	.95	1.00	
.96	.91	.92	.96	.98	.99
.74	.70	.72	.85	.93	
1.00	.53	.81	.70	.92	
.92	.87	.83	.20	.94	
.40	.80	.48	.95	.96	.97
.99	.51	.41	.48	.92	
.35	.41	.21	.23	.99	.99
.95	.81	RAKE MOUNTING POINT	.97	.93	
.95	.20	RAKE MOUNTING POINT	1.04	1.01	.98
.62	.28	RAKE MOUNTING POINT	.59	1.00	
1.05	.92	.28	.96	.97	.48
.97	1.01	.16	.86	.97	
.96	.94	.36	.64	.98	.97

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MOUNTING POINT.
 MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MOUNTING POINT.
 MODEL NOSE, LOOKING AFT.

TABLE VII - Continued

RUN 11 POINT 6 ALPHA=-9.0 DEG PSI=-0.0 DEG V=100 KNOTS Q=31.0 PSF. CONFIGURATION FHC				RUN 11 POINT 7 ALPHA=-4.0 DEG PSI=-0.0 DEG V=100 KNOTS Q=31.0 PSF. CONFIGURATION FHC			
.99	1.00	.97	.99	.97	.95	.98	.97
.99	.99	.98	.98	.98	.96	.96	.97
.99	.96	.98	.98	.98	.97	.98	.97
.99	.99	.97	.98	1.00	.96	.97	.97
.99	.99	.99	.98	.98	.97	.98	.97
.97	.97	.98	.98	.97	.98	.98	.97
.83	.94	1.00	.98	.96	1.01	.97	.98
.98	1.00	1.00	.98	.99	.95	.95	.97
.98	.95	.95	.98	1.02	.96	.98	.98
.97	1.05	1.07	.89	.97	.95	.95	.98
.97	1.09	.98	.93	.59	.96	.98	.99
.98	.79	.71	.88	.71	.99	.98	.99
.98	.67	.54	.79	.51	.96	.97	.96
1.05	1.09	.50	RAKE MOUNTING POINT	.41	.62	RAKE MOUNTING POINT	.55
1.03	.38	.32	.94	.94	.99	.99	.98
1.01	1.2	3.0	.76	.76	.95	.95	.98
	1.02	.65	.50	.93	.92	.97	.97
	.87	.41	.59	1.06	.99	.99	.88
	.84	.66	.86	.95	1.03	.99	.95
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_i - P_0}{q}$				NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_i - P_0}{q}$			
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.				NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.			
NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.				NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.			

TABLE VII - Continued

RUN 11 POINT 8 ALPHA= 4.0 DEG PSI= -0 DEG V= 100. KNOTS Q= 31.0 PSF. CONFIGURATION FHC				RUN 11 POINT 9 ALPHA= 8.0 DEG PSI= -0 DEG V= 100. KNOTS Q= 31.0 PSF. CONFIGURATION FHC			
.97	.98	.98	.97	.96	.98	1.00	.99
.97	.97	.99	.96	.82	.90	.91	.97
.99	.98	.99	4.0 -2.1	.96	.74	.86	8.7 -2.2
.99	.96	1.00	1.05	.72	.74	1.02	.89
.97	.90	.94	.96	.62	.94	.74	.69
.98	.87	.89	.97	.83	.50	.76	.82
.82	.59	.78	.93	.88	1.17	.58	.93
.35	.88	.97	.46	.92	.71	.43	.81
4.0	-0	3.1	3.0	6.7	3.5	.57	10.8 5.4
-1.0	.93	.81	.91	-3	-7	.05	-.6 .1
.66	.48	.61	.90	1.05	.72	.05	.46
.97	.72	.48	.93	1.02	1.04	.57	.87
.79	.67	.81	.80	1.03	.98	.56	.98
.99	.72	.60	.93	.99	.93	.34	.90
.97	1.04	.92	1.01	1.04	1.04	.84	1.29
2.7	1.1	4.5	4.0	5.9	9.6	RAKE MOUNTING POINT	8.0 7.2
-2.0	.69	.90	.98	-3.0	2.7	.35	1.03 2.5
1.00	.83	.89	.98	.99	.64	.58	.95
1.00	1.12	.54	.97	.97	.97	.65	.96
1.01	.94	.86	.94	.99	.94	.47	.96
.98	1.00	.92	.97	.98	.96	.48	1.00

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE VII - Continued

RUN 10 POINT 10 ALPHA=4.0 DEG PSI=-4.0 DEG V=100. KNOTS q = 31.0 PSF. CONFIGURATION FHC				RUN 11 POINT 11 ALPHA=-4.0 DEG PSI=-4.0 DEG V=100. KNOTS q = 31.0 PSF. CONFIGURATION FHC			
.99	.98	.97	.97	.99	.98	.97	.97
1.00	1.00	1.00	1.00	1.00	.98	.97	.96
.99	.99	.97	5.5	1.00	.98	3.0	.96
1.00	1.03	1.03	-4.9	.97	.97	-4.4	.97
.98	.90	.81	.96	1.01	.98	.98	.97
.76	.90	.90	1.00	1.00	.98	.98	.98
1.01	1.05	.80	.97	.93	.97	.97	.97
.80	.80	.95	1.01	1.00	.99	.99	.99
1.00	.62	.57	3.0	.96	1.01	-9.4	.98
.91	.83	1.16	-5.1	1.08	.71	-5.0	.95
.96	.62	.51	.96	.84	.77	.84	.98
.45	.94	.94	.35	.75	.93	.93	.95
1.16	.84	.84	.62	1.04	.95	.44	.98
1.18	.41	.41	RAKE MOUNTING POINT	.88	.81	RAKE MOUNTING POINT	1.01
.98	.98	.96	3.2	.33	.64	-7.6	.87
.58	.58	.92	-4.7	.71	.42	.4	.97
.73	.77	.77	1.02	.58	.72	.58	.99
.63	.94	.94	.98	.79	.74	.79	.98
.79	1.04	1.04	.97	.94	.67	.94	.97
1.00	.78	.98	.98	1.02	.73	1.00	.94

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT
 TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW
 MODEL NOSE, LOOKING AFT.

TABLE VII - Continued

RUN 11 POINT 14 ALPHA = 0 DEG PSI = 0.0 DEG V = 100. KNOTS q = 31.0 PSF. CONFIGURATION FHC		RUN 11 POINT 15 ALPHA = 0 DEG PSI = -4.0 DEG V = 100. KNOTS q = 31.0 PSF. CONFIGURATION FHC	
.99	1.00	.98	.97
1.00	1.00	.96	.95
.99	.99	.97	.94
1.00	.97	.99	.96
.96	.97	.94	.96
.99	.97	.97	.98
.81	.95	.90	.97
1.01	.97	1.00	.98
.89	.97	1.00	.80
.56	.98	.60	.75
.63	.99	.62	.92
.55	.96	1.05	.25
.93	.85	.61	.76
.35	.87	.69	.80
.33	.89	.41	.81
.57	.98	.91	.99
.82	1.04	1.07	.94
.86	1.05	.83	.58
.95	1.00	.74	.73
.73	1.03	.82	.75

RUN 11 POINT 14 ALPHA = 0 DEG PSI = 0.0 DEG V = 100. KNOTS q = 31.0 PSF. CONFIGURATION FHC		RUN 11 POINT 15 ALPHA = 0 DEG PSI = -4.0 DEG V = 100. KNOTS q = 31.0 PSF. CONFIGURATION FHC	
1.00	.97	1.00	.98
.96	1.01	.98	.99
1.02	.98	.96	.96
1.00	.99	1.00	.97
1.00	1.00	.95	.99
1.02	1.00	.98	1.00
1.00	1.00	.97	.98
.98	1.01	.94	.97
.97	.98	1.03	1.02
.98	.98	.54	.94
.99	.98	1.04	1.13
.96	1.01	.89	.35
.93	1.02	1.04	1.00
.87	.97	1.04	.94
.89	1.00	.81	.97
.98	1.01	.99	.94
.98	.98	.94	.98
1.04	1.04	.98	1.00
1.05	1.05	1.06	1.01

RUN 11 POINT 14 ALPHA = 0 DEG PSI = 0.0 DEG V = 100. KNOTS q = 31.0 PSF. CONFIGURATION FHC		RUN 11 POINT 15 ALPHA = 0 DEG PSI = -4.0 DEG V = 100. KNOTS q = 31.0 PSF. CONFIGURATION FHC	
1.00	1.00	1.00	1.00
.98	1.01	.98	1.01
.99	.98	.97	1.02
.97	.99	.97	.97
.95	1.00	.94	.98
.97	1.01	.80	1.00
.98	.98	.75	.95
.99	.98	.92	.98
.96	1.01	.25	.99
.93	1.02	.76	.99
.87	.97	.80	.99
.89	1.00	.81	.99
.98	1.01	.99	.99
.98	.98	.94	.98
1.04	1.04	.98	.99
1.05	1.05	1.06	.97

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT
 TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT
 TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE VII - Continued

RUN 11 POINT 10. ALPHA = .0 DEG. PSI = 4.0 DEG. V = 100. KNOTS. q = 31.0 PSF. CONFIGURATION FHC		RUN 11 POINT 17. ALPHA = .0 DEG. PSI = 8.0 DEG. V = 100. KNOTS. q = 31.0 PSF. CONFIGURATION FHC	
.99	.98	.99	.97
.99	.98	.99	.97
.99	.96	.99	.98
.99	1.01	.99	.96
.98	.98	.99	.98
.98	.92	1.00	.98
.93	.96	1.01	.84
.98	.97	.98	.90
.96	.60	.98	.77
1.04	.95	.97	.96
.97	.77	.99	.70
1.00	.29	.87	.51
1.10	.74	.67	.13
1.02	.62	.67	1.11
1.06	1.04	.78	.92
.92	.57	1.09	.94
.96	.23	.76	.69
.98	.38	.59	.76
.98	.70	.50	.86
<p>NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$</p> <p>NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.</p> <p>NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.</p>		<p>NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$</p> <p>NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.</p> <p>NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.</p>	

TABLE VII - Continued

RUN 12 POINT 5 V=150. KNOTS		ALPHA=0 DEG q=69.3 PSF. CONFIGURATION FHC		PSI=-0 DEG		RUN 12 POINT 6 V=150. KNOTS		ALPHA=4.0 DEG q=69.3 PSF. CONFIGURATION FHC		PSI=-0 DEG	
.97	.98	.99	.98	.97	.96	.97	.97	.96	.96	.98	.98
.98	.98	1.00	.96	1.00	.97	.98	.96	.96	.96	.98	.98
.99	.97	.97	.98	.97	.97	.97	.97	.97	.97	.98	.98
1.00	.97	1.00	1.01	.96	.98	.98	.98	.95	.95	.98	.99
.99	.98	.99	.99	.96	.96	.97	.98	.98	.98	.98	.98
.96	.88	.97	.96	.97	.96	.97	.98	.98	.98	.98	.98
.89	.82	.88	.85	1.00	.97	.94	.98	.97	.97	1.00	.98
.99	.76	.92	.97	.97	.97	.95	.77	.46	1.02	.99	.99
1.00	.76	.98	1.11	1.01	.97	.95	.97	.96	1.08	.96	.99
1.00	1.06	.60	.94	.80	.97	.91	.84	.99	1.06	1.04	.98
.56	.62	.47	.49	.95	.97	.98	.42	1.08	.80	1.01	.98
.87	.47	.51	.52	1.01	.96	.69	.58	.85	.42	.79	.99
1.03	.77	.75	.75	.88	.96	.52	.54	.62	.62	.86	.99
1.00	.80	1.26	RAKE MOUNTING POINT	.97	.96	.43	.34	.87	.87	.91	.98
.90	.83	.91	RAKE MOUNTING POINT	1.02	.96	.69	.28	.92	.92	.96	.98
.96	.73	.69	RAKE MOUNTING POINT	.97	.97	.92	.4	1.02	1.02	.97	.98
1.01	.61	.74	.71	1.00	.98	.97	.77	.54	.96	.98	.98
.99	.84	.95	.49	.98	.98	.91	.84	.32	1.09	.99	.99
.96	.92	.95	.35	.97	.96	.99	.77	.42	1.03	1.00	.97

RUN 12 POINT 5 V=150. KNOTS		ALPHA=0 DEG q=69.3 PSF. CONFIGURATION FHC		PSI=-0 DEG		RUN 12 POINT 6 V=150. KNOTS		ALPHA=4.0 DEG q=69.3 PSF. CONFIGURATION FHC		PSI=-0 DEG	
.97	.98	.99	.98	.97	.96	.97	.97	.96	.96	.98	.98
.98	.98	1.00	.96	1.00	.97	.98	.96	.96	.96	.98	.98
.99	.97	.97	.98	.97	.97	.97	.97	.97	.97	.98	.98
1.00	.97	1.00	1.01	.96	.98	.98	.98	.95	.95	.98	.99
.99	.98	.99	.99	.96	.96	.97	.98	.98	.98	.98	.98
.96	.88	.97	.96	.97	.96	.97	.98	.98	.98	.98	.98
.89	.82	.88	.85	1.00	.97	.94	.98	.97	.97	1.00	.98
.99	.76	.92	.97	.97	.97	.95	.77	.46	1.02	.99	.99
1.00	.76	.98	1.11	1.01	.97	.95	.97	.96	1.08	.96	.99
1.00	1.06	.60	.94	.80	.97	.91	.84	.99	1.06	1.04	.98
.56	.62	.47	.49	.95	.97	.98	.42	1.08	.80	1.01	.98
.87	.47	.51	.52	1.01	.96	.69	.58	.85	.42	.79	.99
1.03	.77	.75	.75	.88	.96	.52	.54	.62	.62	.86	.99
1.00	.80	1.26	RAKE MOUNTING POINT	.97	.96	.43	.34	.87	.87	.91	.98
.90	.83	.91	RAKE MOUNTING POINT	1.02	.96	.69	.28	.92	.92	.96	.98
.96	.73	.69	RAKE MOUNTING POINT	.97	.97	.92	.4	1.02	1.02	.97	.98
1.01	.61	.74	.71	1.00	.98	.97	.77	.54	.96	.98	.98
.99	.84	.95	.49	.98	.98	.91	.84	.32	1.09	.99	.99
.96	.92	.95	.35	.97	.96	.99	.77	.42	1.03	1.00	.97

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.



NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.



NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE VII - Continued

RUN 12 POINT 7 ALPHA=-2.7 DEG PSI=-.0 DEG V=150. KNOTS q= 69.3 PSF. CONFIGURATION FHC				RUN 12 POINT 8 ALPHA= 2.0 DEG PSI= -.0 DEG V=150. KNOTS q= 69.3 PSF. CONFIGURATION FHC			
1.00	.99	.97	.97	1.00	.99	1.01	.99
.98	.98	.98	.98	.99	.99	1.00	.98
.98	.98	.97	.97	.97	1.00	.98	.96
1.01	1.01	1.00	.97	.99	1.00	1.01	.99
1.01	1.01	.97	.98	1.00	.74	.93	.95
1.01	1.01	.98	.97	.94	.94	.91	1.03
.49	.95	1.00	.97	.88	.75	.94	.73
.96	.78	.93	.97	.75	.73	.82	.87
1.00	.97	.99	.99	1.00	.36	.88	.88
.90	.80	.93	.95	.92	.58	.45	.57
1.00	.74	.80	.88	1.03	1.02	.73	.65
1.00	.66	.46	.63	1.04	.87	.54	.47
1.04	.58	.43	.86	1.05	.65	.30	.74
1.02	.92	.66	.95	1.03	.64	RAKE MOUNTING POINT	.92
1.01	.97	.90	1.15	1.02	.88	1.00	.84
.99	.98	1.05	1.00	1.00	.98	.59	.71
				1.02	1.00	.29	1.04
				1.00	1.00	.45	.82
				1.00	1.00	.70	.70
							1.01
							.95

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT 74.75 IN

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT 74.75 IN

TABLE VII - Continued

RUN 12 POINT 9 ALPHA=4.0 DEG PSI=-0 DEG V=150. KNOTS q = 69.3 PSF. CONFIGURATION FHC			RUN 12 POINT 10 ALPHA=4.0 DEG PSI=-2.0 DEG V=150. KNOTS q = 69.3 PSF. CONFIGURATION FHC									
.78	.99	1.00	.48	1.01	.98	.79	1.00	.98	.48	.96	.95	.97
.99	.98	1.00	.99	.96	.97	1.00	.97	.98	1.00	.97	5.0	.97
1.01	1.01	1.02	.97	.99	.95	.98	.98	.97	.99	.97	-3.4	.96
1.01	.99	.80	.89	.97	.96	1.00	.96	.98	.99	.97	.95	.97
.55	.89	.60	.88	.92	.96	.88	.88	.95	.92	.94	.97	.97
	1.03	.77	.90	.97	1.02	.82	.82	.97	.85	1.01	.97	.96
	.97	.85	.83	.70	.97	.53	.60	.80	.87	.87	.94	.96
	.74	.83	.88	.86	.93	.63	.63	.63	.91	1.05	.90	.90
1.01	3.9	.6	.73	3.2	3.9	.99	4.5	-3.3	.45	.93	4.6	4.1
1.01	-7	-2.6	.93	-1.3	.1	.99	-3.4	-4.9	.56	.93	-2.7	-1.2
.98	.92	.71	1.03	1.01		1.05	.92	.71	.56	.65	.96	.96
.99	1.05	.82	.74	.87	.77	.81	.54	.70	.81	.67	1.00	1.01
	.89	.70	.78	1.14	.92	.96	.96	.84	.21	.73	1.03	.98
	1.03	.81	.31	.97	.96	1.00	.43	.41	.40	.95	.98	.98
1.02	1.03	.94	RAKE MOUNTING POINT	.82	.97	RAKE MOUNTING POINT	.90	.42	RAKE MOUNTING POINT	.95	1.13	.98
	.99	.91		3.1	4.2	1.02	4.2	2.6	1.02	1.02	2.8	4.0
	.93	.85		-3.1	1.4	1.02	-4.6	-7.1	.96	.96	-3.4	-0
1.01	1.00	.85	.47	.84	.98	1.00	.95	.50	1.08	1.02	.99	.99
	1.01	.92	.76	.91	.94	.95	.91	.67	.95	1.03	.99	.99
1.01	1.03	.75	.87	1.03	.97	.95	.94	.70	.35	.94	.96	.94

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_{\infty}}{\frac{\rho V^2}{2}}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW
MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_{\infty}}{\frac{\rho V^2}{2}}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW
MODEL NOSE, LOOKING AFT.

TABLE VII - Continued

RUN 12 POINT 11 ALPHA=2.0 DEG PSI=-2.0 DEG V=180. KNOTS q=69.3 PSF. CONFIGURATION FHC		RUN 12 POINT 12 ALPHA=-2.0 DEG PSI=-2.0 DEG V=150. KNOTS q=69.3 PSF. CONFIGURATION FHC	
.79	.97	.98	.96
	1.00	.99	.97
.97	.97	.99	.98
	1.00	1.00	1.02
.97	.97	.98	.98
	1.00	.97	.97
.97	.87	.98	.98
	1.00	.97	.99
.55	.90	.98	.94
	.88	1.01	.97
.54	.87	.90	.96
	.87	.90	.96
.97	.61	.86	.94
	.94	.98	.91
.97	.61	.53	1.06
	.91	.58	.62
.97	.76	.61	.63
	.84	.58	.52
.98	.48	.58	.73
	.61	.93	.84
.98	.79	.93	.94
	.61	.93	.94
1.01	.59	.93	.52
	.49	.93	.52
1.01	1.08	.98	.47
	.87	.98	1.0
1.01	.98	.82	.62
	.51	.88	.72
1.01	.56	.93	.82
	.87	.88	.72
1.04	.82	.93	.82
	.82	.93	.82
1.01	1.01	.89	.54
	.76	.89	.54

RUN 12 POINT 11 ALPHA=2.0 DEG PSI=-2.0 DEG V=180. KNOTS q=69.3 PSF. CONFIGURATION FHC		RUN 12 POINT 12 ALPHA=-2.0 DEG PSI=-2.0 DEG V=150. KNOTS q=69.3 PSF. CONFIGURATION FHC	
.79	.97	.98	.96
	1.00	.99	.97
.97	.97	.99	.98
	1.00	1.00	1.02
.97	.98	.98	.98
	1.00	.97	.97
.97	.87	.98	.98
	1.00	.97	.99
.55	.90	.98	.94
	.88	1.01	.97
.54	.87	.90	.96
	.87	.90	.96
.97	.61	.86	.94
	.94	.98	.91
.97	.61	.53	1.06
	.91	.58	.62
.97	.76	.61	.63
	.84	.58	.52
.98	.48	.58	.73
	.61	.93	.84
.98	.79	.93	.94
	.61	.93	.94
1.01	.59	.93	.52
	.49	.93	.52
1.01	1.08	.98	.47
	.87	.98	1.0
1.01	.98	.82	.62
	.51	.88	.72
1.04	.82	.93	.82
	.82	.93	.82
1.01	1.01	.89	.54
	.76	.89	.54

RUN 12 POINT 11 ALPHA=2.0 DEG PSI=-2.0 DEG V=180. KNOTS q=69.3 PSF. CONFIGURATION FHC		RUN 12 POINT 12 ALPHA=-2.0 DEG PSI=-2.0 DEG V=150. KNOTS q=69.3 PSF. CONFIGURATION FHC	
.79	.97	.98	.96
	1.00	.99	.97
.97	.97	.99	.98
	1.00	1.00	1.02
.97	.98	.98	.98
	1.00	.97	.97
.97	.87	.98	.98
	1.00	.97	.99
.55	.90	.98	.94
	.88	1.01	.97
.54	.87	.90	.96
	.87	.90	.96
.97	.61	.86	.94
	.94	.98	.91
.97	.61	.53	1.06
	.91	.58	.62
.97	.76	.61	.63
	.84	.58	.52
.98	.48	.58	.73
	.61	.93	.84
.98	.79	.93	.94
	.61	.93	.94
1.01	.59	.93	.52
	.49	.93	.52
1.01	1.08	.98	.47
	.87	.98	1.0
1.01	.98	.82	.62
	.51	.88	.72
1.04	.82	.93	.82
	.82	.93	.82
1.01	1.01	.89	.54
	.76	.89	.54

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT, NEGATIVE FOR FLOW FROM RIGHT, LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW
 MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT, NEGATIVE FOR FLOW FROM RIGHT, LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW
 MODEL NOSE, LOOKING AFT.

TABLE VII - Continued

RUN 12 POINT 13 ALPHA=2.0 DEG PSI=2.0 DEG V=150 KNOTS Q=69.3 PSF. CONFIGURATION FMC		RUN 12 POINT 14 ALPHA=0 DEG PSI=4.0 DEG V=150 KNOTS Q=69.3 PSF. CONFIGURATION FMC	
.84	.99	.99	.98
1.02	.98	.99	.98
.99	.96	.98	.97
1.00	.98	.96	.97
.97	1.04	.97	.99
.96	.80	.95	.97
.98	.60	.97	.99
.93	.71	.95	.97
1.01	.71	.95	.99
.97	.85	.95	.97
1.02	.60	.95	.99
1.01	.30	.94	.95
.98	.93	.96	.98
	RAKE MOUNTING POINT		RAKE MOUNTING POINT
	74.75 IN		74.75 IN
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$		NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$	
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.		NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.	
NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.		NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.	

TABLE VII - Continued

RUN 12 POINT 15 ALPHA = -0 DEG PSI = -2.0 DEG V = 150. KNOTS q = 69.3 PSF. CONFIGURATION FHC					RUN 12 POINT 16 ALPHA = +0 DEG PSI = +2.0 DEG V = 150. KNOTS q = 69.3 PSF. CONFIGURATION FHC					
.81	.98	.97	.99	.97	.87	.99	.97	.99	.98	1.00
1.00	.98	1.00	.98	.96	.99	.99	.99	.97	.96	.97
.98	.99	.97	.97	.97	.99	1.01	.97	.96	.97	.97
.56	.90	.96	.96	.95	.53	.96	1.03	.96	.96	.97
1.00	.75	.91	.99	.96	.98	.95	.93	.78	.94	.96
.98	.72	.76	.75	.96	.98	1.01	.42	.86	1.01	.96
.95	.72	.78	.57	1.03	.99	1.04	.60	.79	.99	.97
	.44	.58	.47	.95	.98	1.13	.87	1.06	1.12	.97
	.83	.71	.99	.93	.99	1.08	.47	.59	.80	.96
	.67	.68	.20	.97	.99	1.08	.74	.76	.96	.96
	1.06	.57	.88	.94						
1.00	.07	RAKE MOUNTING POINT	.87	1.0	1.01	.92	.51	.26	.41	.97
1.01	1.05	.83	.87	.96	1.02	.90	.97	.54	.59	.97
	.95	.53	.84	.99	1.01	.96	.77	.82	.96	.96
	.92	.73	.96	1.00	1.02	.96	.91	.71	.98	1.00
.98	1.03	.71	.94	.98	1.01	.98	.97	.95	.98	.96
	1.03	.53	.97	1.00	1.01	.98	1.03	.57	1.00	.96

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
 TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW
 MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

74.75 IN

TABLE VII - Concluded

RUN 12 POINT 17 ALPHA= .0 DEG PSI= 4.0 DEG
 V= 150. KNOTS q= 69.3 PSF. CONFIGURATION FMC

.85	.99	1.00	1.01	.98	.96	.98	
	.99	.98	1.03	1.00	.96		
1.01	.99	1.02	.99	.98	-9 3.2	.97	
	1.00	.99	1.02	.97		.95	
.97	.97	1.02	1.04	1.00		.96	
	.99	1.00	.94	.99		.95	
.38	.97	1.07	.92	.95	1.00	.97	
	.98	.95	.54	.81		.87	
.97	.8 3.3	-.5 2.8	.90	.72	-9.2 1.5	.1 1.0	.99
	.95	.94	.81	1.01		.81	
.98	1.05	.92	.95	.74		.85	
	1.03	.70	.59	.69		.59	
.99	.98	.62	.78	.83		.53	
	1.01	1.09		.96		.77	
1.00	.3 1.4	-.8 4.7	1.09	-1.3 .9	-.1 4.7	1.03	
	.91	1.09		.38		.81	
	1.02	1.11		1.10		1.06	
1.01	1.00	.68	.83	.86		.90	
	1.00	.93	.43	.93		.85	
1.01	1.03	1.00	.74	.72		.92	

RAKE MOUNTING POINT

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

← 7.475 IN →

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

TABLE VIII. WAKE PRESSURE COEFFICIENTS AND FLOW DIRECTIONS AT THREE FORWARD SPEEDS AND VARIOUS FUSELAGE ATTITUDES, CONFIGURATION FHf

MUN 21 POINT 5 V=60. KNOTS		MUN 21 POINT 6 V=60. KNOTS		MUN 21 POINT 6 V=60. KNOTS	
ALPHA=0 DEG PSI=-0.0 DEG q=11.5 PSF. CONFIGURATION FHf		ALPHA=0 DEG PSI=-0.0 DEG q=11.5 PSF. CONFIGURATION FHf		ALPHA=8.0 DEG PSI=-0.0 DEG q=11.5 PSF. CONFIGURATION FHf	
1.00	.94	1.00	.97	1.00	.98
.99	.98	.99	.99	.96	.99
.96	.99	.99	.97	.99	.99
.97	1.00	1.01	.92	.98	.99
.98	1.02	1.01	.96	.98	.97
.93	1.02	1.02	.99	.96	.97
.94	1.00	.96	1.00	.97	.95
.99	.85	.99	.96	.99	.96
.97	.77	1.01	.88	1.03	.99
1.14	.67	.94	.97	.95	.94
1.08	.59	.91	.61	.97	.98
1.01	.72	.91	.57	1.01	1.04
.97	.89	.54	.13	1.05	.61
	.77	RAKE MOUNTING POINT		RAKE MOUNTING POINT	
	.55				
	.53				
	.72				
	.85				
	.89				

.98	.93	.99	.99	.98	.98
.99	.96	.92	.92	.96	.98
.96	.99	.91	.96	.99	.99
.96	.99	.98	.92	.99	1.03
.98	.97	.98	.98	.97	.99
.97	.97	.96	.96	.97	.98
.97	.97	.95	.95	.97	1.00
.97	.99	.96	.96	.99	1.01
.97	.92	.99	.99	.92	.96
.96	1.00	.94	.94	1.00	.96
.96	.97	.95	.95	.97	.98
.96	.97	1.01	1.10	1.07	1.04
1.04	1.05	1.10	1.10	1.05	.61
1.13	.79	RAKE MOUNTING POINT	RAKE MOUNTING POINT	.32	.67
1.06	.57			.52	-17.6
1.00	.44			.86	.79
	.94			.33	.99
	.52			.60	1.07
	.75			.49	.90
	.44			.51	.89

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{p - p_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT MOUNTING POINT TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{p - p_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT MOUNTING POINT TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

TABLE VIII - Continued

RUN 21 POINT 9 V=60 KNOTS		ALPHA= 8.0 DEG q= 11.5 PSF.		PSI= -0 DEG CONFIGURATION FH		RUN 21 POINT 10 V=60 KNOTS		ALPHA= 4.0 DEG q= 11.5 PSF.		PSI= -4.0 DEG CONFIGURATION FH	
.07	.97	.94	.98	.95	.95	1.02	.94	.97	.91	.98	.99
1.00	1.02	.99	1.00	.97	6.5	.99	.94	.97	4.9	.99	.95
		.95	.98	.99	-2.4	.96	.94	.92	.97	.99	.95
.99	.98	.97	.99	.99		.99	.94	.94	.98	1.01	.96
	.94	.80	1.15	1.01		.99	.96	.93	.98	.99	.96
.98	.98	.93	.91	.80		.90	.80	.94	.95	.97	.98
	.95	1.03	.92	.95		.95	.84	.95	.96	.96	.98
.96	1.01	.83	1.18	.92	.55	1.01	.95	.93	.94	.95	.96
	.96	1.15	.67	.57	.8	.78	.97	.88	4.6	.99	.96
	1.0	-1.0	.97	.90	.6	-2.3	.42	.60	-5.1	.93	3.6
1.03	.95	.31	.16	.93	7.8	.76	.23	.37	-4.1	.99	-4.1
	.94	.82	.75	.95	2.4	.66	.48	.67		.99	.99
1.02	1.04	.83	.55	.97		.53	1.07	.79		.96	.99
	.94	.95	RAKE MOUNTING POINT	.89		.36	.50	RAKE MOUNTING POINT	.98	.98	.99
1.11	.94	.74	.73	.75	1.02	.61	-10.8	RAKE MOUNTING POINT	4.2	1.03	5.6
	.94	4.2	.79	.79	1.00	.74	.86		-4.1	1.03	-1.9
	-3.5	4.2	.92	.92	-2.5	-9.1	.85	.82		.96	1.00
1.07	1.01	.47	.27	.96	6.9	.85	.75	.82		.96	1.00
	.96	.97	.44	.98	-1.5	.80	.31	.83		.92	.98
1.04	.95	1.11	.34	.95		.81	.72	.11		.94	.99

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

7.475 IN

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

7.475 IN

TABLE VIII - Continued

RUN 21 POINT 13 ALPHA=8.0 DEG PSI=4.0 DEG V=60 KNOTS Q=11.5 PSF. CONFIGURATION PH1				RUN 21 POINT 14 ALPHA=4.0 DEG PSI=4.0 DEG V=60 KNOTS Q=11.5 PSF. CONFIGURATION PH1			
1.01	.98	.97	.94	.97	.98	.96	.94
	1.01	1.03	.96	.94	.94	.93	.93
.98	1.01	.97	.97	.97	.95	.96	.94
	.99	.96	.98	.99	.96	.95	.94
.98	.98	.95	.95	.99	.97	.92	.97
	.96	.96	.95	.96	.95	.91	.95
1.01	.98	.97	.99	.96	.94	.97	.94
	1.03	.96	.98	.96	.94	.95	.97
1.00	-7.5 3.0	-10.7 1.00	1.00	-4.5 1.4	.95	-4.4 4.5	.94
	.98	1.02	1.02	1.00	.92	.99	.93
.98	.97	.95	.98	.93	.96	.96	.78
	.99	1.21	.70	.93	.73	.93	.85
.99	.93	.55	.72	1.00	.89	.93	1.07
	.69	RAKE MOUNTING POINT	.84	.67	.86	.84	.64
1.17	-4.1 1.4	-21.3 2.8	-8.3 -2.7	1.08 1.08	-5.3 3.8	-3.6 1.3	.76 72
	.93	.78	.51	.78	1.02	1.01	.99
1.09	1.05	.73	1.00	.38	.98	.99	.73
	1.00	.81	.39	.48	.97	.96	.50
1.02	.95	.56	.49	.43	1.00	.94	.91

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOKES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
TO RIGHT LOOKING AFT.



NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOKES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
TO RIGHT LOOKING AFT.

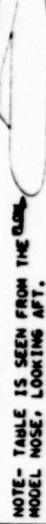


TABLE VIII - Continued

RUN 21 POINT 15 ALPHA = 4.0 DEG PSI = 4.0 DEG V = 60. KNOTS Q = 11.5 PSF. CONFIGURATION FM				RUN 21 POINT 15 ALPHA = 0 DEG PSI = 8.0 DEG V = 60. KNOTS Q = 11.5 PSF. CONFIGURATION FM			
.09	.96	.98	.97	.96	.97	.98	.97
.96	1.00	.97	.98	.97	.97	.94	.98
.98	.98	1.02	.99	1.02	1.00	2.3	1.00
.97	.97	.97	.96	1.00	1.00	1.8	1.00
.97	1.02	.99	1.01	.88	.98	.85	.99
.99	1.03	.96	.99	.98	.93	.93	.98
.99	1.01	1.07	.60	.98	.93	1.05	.99
1.05	4.8	1.00	1.03	.80	.83	4.7	.98
1.02	1.00	1.10	1.00	.88	1.03	7.7	1.04
1.00	1.04	.91	.59	1.05	.50	1.1	.98
.98	.98	.61	.52	1.20	.43	.98	.98
.99	.99	.90	1.09	1.08	.67	.67	.98
.96	.96	.95	.35	.35	1.02	.95	.95
1.15	3.8	1.02	.80	.79	.38	3.4	.94
1.09	1.02	.92	RAKE MOUNTING POINT	.79	-2.4	.50	.98
1.00	2.0	1.01	.90	.68	1.9	.83	.93
.96	1.00	.96	1.14	.78	.94	.94	.98
	1.01	.99	.29	.79	.54	.54	.96
	.96	1.01	.96	.54	.64	.64	.93

RUN 21 POINT 15 ALPHA = 0 DEG PSI = 8.0 DEG V = 60. KNOTS Q = 11.5 PSF. CONFIGURATION FM				RUN 21 POINT 15 ALPHA = 0 DEG PSI = 8.0 DEG V = 60. KNOTS Q = 11.5 PSF. CONFIGURATION FM			
.96	.98	.97	.98	.94	.97	.98	.97
1.00	.94	.97	.94	.97	.97	.98	.95
.98	1.00	1.00	.96	1.00	.96	1.3	.98
.99	.98	.99	.94	.98	.94	-9.0	1.02
.99	1.00	.95	.97	1.00	.95	.94	.98
.97	.97	.94	.94	.97	.94	.94	.98
1.09	.97	.98	1.02	.98	.98	.98	.99
.72	-10.6	.98	.95	.94	.94	1.00	.99
.93	-7.8	.99	.95	.98	.98	.6	.98
.63	.97	.97	.95	.97	.97	-11.2	.98
.92	.89	.89	.12	.99	1.00	1.01	1.01
.92	.93	.93	1.19	.90	.98	.98	.98
.92	.27	.27	.58	.91	.96	.99	.94
.92	1.00	1.00	.52	RAKE MOUNTING POINT	.98	1.01	.97
.97	-2.2	1.02	-0.3	.92	2.3	.99	.96
1.01	-10.5	.79	8.5	.98	-9.6	.96	.94
	.07	.80	.48	.96	.94	.94	.96
	.64	.64	.25	.99	.96	.96	.96
	1.01	.79	1.08	.94	1.00	1.00	.94

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOOR.
MODEL NOSE, LOOKING AFT.

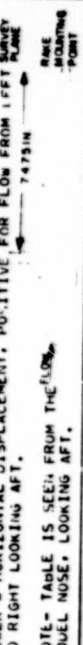


TABLE VIII - Continued

MUN. 21 POINT 17 ALPHA = .0 DEG PSI = 4.0 DEG V = 60 KNOTS q = 11.5 PSF. CONFIGURATION FH1				MUN. 21 POINT 18 ALPHA = .0 DEG PSI = 4.0 DEG V = 60 KNOTS q = 11.5 PSF. CONFIGURATION FH1			
.97	.95	.96	.98	.99	.98	.98	.99
1.00	1.00	.96	1.00	.96	1.00	1.01	.99
.99	.97	.94	1.00	.96	.96	1.00	1.01
.99	.97	.98	1.00	.95	.95	1.00	.98
.99	1.01	.98	.96	.94	.96	1.00	.94
1.01	1.01	1.01	.93	.92	.93	.99	.97
1.02	1.04	1.04	.94	.99	.94	1.00	1.05
1.03	1.04	1.00	.92	.98	.92	.98	.70
.96	1.03	.91	.93	.96	.92	.93	.95
.91	1.04	.91	.94	.94	.95	.98	.76
.96	1.02	.61	.65	.97	.99	.60	.69
.91	.86	.64	1.10	.92	.99	.34	.52
1.14	.79	.79	.74	.73	1.04	.56	.35
1.07	.26	.85	RAKE MOUNTING POINT	.94	.82	.54	.83
.99	1.00	1.28	RAKE MOUNTING POINT	.91	1.16	.90	.96
1.02	.41	.68	RAKE MOUNTING POINT	.95	1.00	.68	.97
1.07	.69	.87	.58	.98	1.00	.41	.98
.99	.96	1.07	.61	.99	.98	.42	.94
.99	1.19	1.03	.70	.98	.92	.91	.99

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

74.75 IN

RAKE MOUNTING POINT

TABLE VIII - Continued

RUN 21 POINT 19 ALPHA= .0 DEG PSI= 8.0 DEG V=60 KNOTS q= 11.5 PSF. CONFIGURATION FH				RUN 21 POINT 20 ALPHA= .0 DEG PSI= -1.0 DEG V=60 KNOTS q= 11.5 PSF. CONFIGURATION FH			
1.01	1.01	1.02	1.00	.99	1.00	.92	1.01
1.01	1.01	1.01	1.00	1.03	1.00	.99	.98
	.96	.99	1.03	1.04	1.00	1.00	.98
.95	.99	1.03	.98	1.03	.99	.99	.98
.97	.99	1.01	.99	1.00	1.01	1.01	.98
	.99	1.01	.97	1.02	.98	.98	.94
.97	1.05	1.02	1.01	1.01	.98	.98	.91
.98	1.04	1.02	.94	1.02	.96	.96	.98
	.3	1.0	-78.3	1.03	-8.1	.80	.97
.98	2.3	4.3	4.3	1.03	5.6	1.03	.96
	1.02	1.02	.86	.72	.86	.79	.94
.99	1.03	1.05	.91	.79	.91	.81	.90
.98	1.04	1.03	.84	.84	1.01	.73	.83
	.99	1.04	.45	.72	.84	.44	.44
1.16	1.02	.99	.33	HAKE MOUNTING POINT	.99	.71	.59
	1.4	.1	-10.8	HAKE MOUNTING POINT	-3.9	.91	.91
	4.1	5.0	.65	HAKE MOUNTING POINT	-5.9	.69	.57
	1.01	1.02	.47	.86	.40	.87	1.00
1.14	1.04	1.03	.67	.79	1.16	.46	.72
	1.04	1.06	.78	.74	1.06	.37	.96
1.04	1.05	1.02	.24	.66	.99	.65	1.00
					1.01	.72	1.03

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT

TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE VIII - Continued

MUN 22 POINT 7 ALPHA=-4.0 DEG PSI=-7.0 DEG V=100 KNOTS q=31.6 PSF. CONFIGURATION FH		MUN 22 POINT 8 ALPHA=4.0 DEG PSI=-7.0 DEG V=100 KNOTS q=31.6 PSF. CONFIGURATION FH	
.06	.99	.98	.96
.07	.97	.97	.96
.08	.95	.96	.96
.09	.98	.97	.98
1.01	.96	.98	.97
.07	.97	.99	.97
.07	.96	.97	.96
.07	.95	.98	.96
.07	.94	.99	.96
.07	.93	.99	.96
.07	.92	.99	.96
.07	.91	.99	.96
.07	.90	.99	.96
.07	.89	.99	.96
.07	.88	.99	.96
.07	.87	.99	.96
.07	.86	.99	.96
.07	.85	.99	.96
.07	.84	.99	.96
.07	.83	.99	.96
.07	.82	.99	.96
.07	.81	.99	.96
.07	.80	.99	.96
.07	.79	.99	.96
.07	.78	.99	.96
.07	.77	.99	.96
.07	.76	.99	.96
.07	.75	.99	.96
.07	.74	.99	.96
.07	.73	.99	.96
.07	.72	.99	.96
.07	.71	.99	.96
.07	.70	.99	.96
.07	.69	.99	.96
.07	.68	.99	.96
.07	.67	.99	.96
.07	.66	.99	.96
.07	.65	.99	.96
.07	.64	.99	.96
.07	.63	.99	.96
.07	.62	.99	.96
.07	.61	.99	.96
.07	.60	.99	.96
.07	.59	.99	.96
.07	.58	.99	.96
.07	.57	.99	.96
.07	.56	.99	.96
.07	.55	.99	.96
.07	.54	.99	.96
.07	.53	.99	.96
.07	.52	.99	.96
.07	.51	.99	.96
.07	.50	.99	.96
.07	.49	.99	.96
.07	.48	.99	.96
.07	.47	.99	.96
.07	.46	.99	.96
.07	.45	.99	.96
.07	.44	.99	.96
.07	.43	.99	.96
.07	.42	.99	.96
.07	.41	.99	.96
.07	.40	.99	.96
.07	.39	.99	.96
.07	.38	.99	.96
.07	.37	.99	.96
.07	.36	.99	.96
.07	.35	.99	.96
.07	.34	.99	.96
.07	.33	.99	.96
.07	.32	.99	.96
.07	.31	.99	.96
.07	.30	.99	.96
.07	.29	.99	.96
.07	.28	.99	.96
.07	.27	.99	.96
.07	.26	.99	.96
.07	.25	.99	.96
.07	.24	.99	.96
.07	.23	.99	.96
.07	.22	.99	.96
.07	.21	.99	.96
.07	.20	.99	.96
.07	.19	.99	.96
.07	.18	.99	.96
.07	.17	.99	.96
.07	.16	.99	.96
.07	.15	.99	.96
.07	.14	.99	.96
.07	.13	.99	.96
.07	.12	.99	.96
.07	.11	.99	.96
.07	.10	.99	.96
.07	.09	.99	.96
.07	.08	.99	.96
.07	.07	.99	.96
.07	.06	.99	.96
.07	.05	.99	.96
.07	.04	.99	.96
.07	.03	.99	.96
.07	.02	.99	.96
.07	.01	.99	.96
.07	.00	.99	.96
.07	-.01	.99	.96
.07	-.02	.99	.96
.07	-.03	.99	.96
.07	-.04	.99	.96
.07	-.05	.99	.96
.07	-.06	.99	.96
.07	-.07	.99	.96
.07	-.08	.99	.96
.07	-.09	.99	.96
.07	-.10	.99	.96
.07	-.11	.99	.96
.07	-.12	.99	.96
.07	-.13	.99	.96
.07	-.14	.99	.96
.07	-.15	.99	.96
.07	-.16	.99	.96
.07	-.17	.99	.96
.07	-.18	.99	.96
.07	-.19	.99	.96
.07	-.20	.99	.96
.07	-.21	.99	.96
.07	-.22	.99	.96
.07	-.23	.99	.96
.07	-.24	.99	.96
.07	-.25	.99	.96
.07	-.26	.99	.96
.07	-.27	.99	.96
.07	-.28	.99	.96
.07	-.29	.99	.96
.07	-.30	.99	.96
.07	-.31	.99	.96
.07	-.32	.99	.96
.07	-.33	.99	.96
.07	-.34	.99	.96
.07	-.35	.99	.96
.07	-.36	.99	.96
.07	-.37	.99	.96
.07	-.38	.99	.96
.07	-.39	.99	.96
.07	-.40	.99	.96
.07	-.41	.99	.96
.07	-.42	.99	.96
.07	-.43	.99	.96
.07	-.44	.99	.96
.07	-.45	.99	.96
.07	-.46	.99	.96
.07	-.47	.99	.96
.07	-.48	.99	.96
.07	-.49	.99	.96
.07	-.50	.99	.96

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_\infty}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
 TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE DOWN
 MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

TABLE VIII - Continued

RUN 22 POINT 9 ALPHA= 8.0 DEG PSI= -0 DEG V= 100 KNOTS q= 31.6 PSF. CONFIGURATION FHY										RUN 22 POINT 10 ALPHA= 4.0 DEG PSI= -4.0 DEG V= 100 KNOTS q= 31.6 PSF. CONFIGURATION FHY									
.05	.93	.97	.98	.97	1.00	1.01	.99	.97	.96	.99	.97	.96	.99	.97	.99	.99	.97	.96	.99
.05	.96	.97	.99	.97	7.7	7.7	.99	.98	4.6	4.6	.98	.96	.99	.97	.98	.97	.96	.99	.98
.04	.94	.95	.98	.95	-2.7	-2.7	.98	.98	-5.2	-5.2	1.00	1.00	.97	.97	1.00	1.00	.97	.97	1.00
.04	.94	.96	.98	.98	.99	.99	.98	.98	1.01	1.01	.95	.98	.97	.97	.97	.99	.98	.98	.98
.04	.95	.86	.95	.94	.96	.96	.95	.94	.97	.97	1.01	.98	.96	.99	.99	.99	.99	1.01	.99
.07	.94	.74	.95	.89	.99	.99	.95	.97	.97	.97	1.06	1.01	.96	.96	.96	.99	1.01	.99	.97
.07	.94	.70	.95	.81	1.01	1.01	.97	.97	.89	.89	1.03	1.01	.96	.96	.96	.99	1.01	.99	.97
.04	.94	1.10	.93	.65	.95	.95	.95	.95	.91	.91	1.03	.94	.96	.96	.96	.99	.94	.99	.99
.04	.86	.45	.19	.68	.91	.91	.99	.99	.96	.96	.78	.85	.78	.78	.78	.99	.85	.99	.98
.04	1.02	.72	.51	1.02	.26	.26	.98	.98	.73	.73	.75	.75	.46	.46	.46	.96	1.07	.96	.98
.05	.98	.38	.63	.64	.95	.95	.98	.98	.95	.95	.34	1.04	.53	.53	.53	.97	1.04	.97	1.00
.06	.67	.69	.83	.51	1.20	1.20	.98	.98	.76	.76	.89	.88	.31	.31	.31	.97	.88	1.00	1.00
.06	.90	.82	1.01	.71	.97	.97	1.01	1.01	.15	.15	.14	1.02	.54	.54	.54	.98	1.02	.98	1.01
1.02	1.01	.63	.86	.63	1.09	1.09	.98	.98	.59	.59	1.06	.98	.84	.84	.84	1.00	.98	1.00	1.01
.09	1.04	1.01	.66	.66	7.3	7.3	.96	.96	1.01	1.01	.59	1.01	.59	1.01	1.01	.98	1.01	.98	1.01
.06	.95	.65	.51	.51	-2.4	-2.4	.95	.95	.75	.75	.84	.97	.84	.84	.84	.97	.97	.97	1.01
.09	.99	.67	.83	.29	1.00	1.00	.99	.99	.90	.90	.82	.71	.71	.71	.71	.98	.98	1.00	.99
.06	.97	1.04	.90	.46	.96	.96	.96	.96	.70	.70	.81	.96	.96	.96	.96	.99	.99	1.01	1.01
.06	.99	.77	.95	.31	1.00	1.00	.96	.96	.75	.75	.81	.67	.67	.67	.94	.94	1.02	.94	.96

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

7.475 IN

TABLE VIII - Continued

NUM 22 POINT 11 ALPHA=-4.0 DEG PSI=-4.0 DEG V=100 KNOTS q= 31.6 PSF. CONFIGURATION FH				NUM 22 POINT 12 ALPHA=-4.0 DEG PSI= 4.0 DEG V=100 KNOTS q= 31.6 PSF. CONFIGURATION FH			
1.01	.99	.98	.97	1.01	.98	.97	.99
1.01	1.00	.98	-3.2	.98	.96	-3.9	.99
.99	1.00	.98	-4.7	1.00	1.00	3.1	.98
.99	.99	1.01	.99	.99	.99	.99	.98
.99	.98	1.02	.99	1.00	1.00	1.00	.98
.99	1.00	.98	.99	.99	.98	1.00	.98
.99	1.02	.98	1.00	.99	.96	.94	.98
.98	1.00	1.05	.99	.97	.95	.95	.98
.98	-0.5	.96	-3.4	.99	.90	-9.5	.96
.98	-2.8	.96	-5.6	.98	.95	3.2	.96
.98	1.01	1.09	.98	1.00	.95	1.03	.96
.98	.97	.92	1.00	1.05	.68	.94	1.03
.98	.87	1.29	.94	.96	1.13	.60	.94
.98	.98	.46	.98	.91	.51	.90	1.01
.93	.72	.58	.98	.95	.60	.83	.44
.93	-4.9	.99	-3.5	1.04	-3.4	-25.3	.44
.93	-3.1	.99	-4.7	1.04	-2.6	.94	.65
.97	.61	.90	1.01	1.01	1.01	.61	.94
.97	.56	.46	1.01	1.04	1.08	.72	.65
.96	.70	.36	1.02	.98	.98	.50	.94
1.04	.89	.36	1.04	.96	.43	1.08	.64
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$				NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$			
NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.				NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.			
NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.				NOTE- TABLE IS SEEN FROM THE BLADE MOUNTING POINT.			

TABLE VIII - Continued

MUI 22 POINT 13 V=100 KNOTS		ALPHA= 4.0 DEG PSI= 4.0 DEG		ALPHA= 31.6 PSF. CONFIGURATION FMH		MUI 22 POINT 14 V=100 KNOTS		ALPHA= .0 DEG PSI=-8.0 DEG		ALPHA= 31.6 PSF. CONFIGURATION FMH	
1.01	.98	.99	.98	.99	1.00	.99	.98	1.00	.97	.99	.99
.99	.98	.99	.99	.99	.98	.97	.99	.98	.95	.99	.96
.99	.97	1.02	.96	.99	.98	.98	1.00	.99	.97	.97	.96
.99	1.02	.97	.99	.99	.98	.97	.99	.99	.99	.99	.97
.99	.99	.98	.92	.99	.98	.97	.97	.99	1.00	.98	.97
.99	.97	.98	1.02	.99	.89	.93	1.01	.97	.98	.98	.97
.99	.94	.95	.83	1.01	.58	1.01	.98	.97	.98	.97	.97
.99	.98	.97	.96	1.15	1.01	1.01	1.02	1.01	.99	.99	.97
.99	.99	.94	.99	.85	-11.6	.88	-5.9	.99	.97	.99	1.8
.99	.98	.96	.66	.32	2.7	.65	-5.4	.96	.97	.98	-7.0
.99	.97	.69	.60	.53	.54	.54	.57	.77	.99	.96	1.00
.99	.99	.51	1.00	.95	.96	.96	.15	.96	.98	.97	.97
.99	1.01	.82	.38	.50	1.30	.98	.28	1.06	1.02	.97	.98
1.04	1.04	.88	.95	.88	.87	.87	.64	RAKE MOUNTING POINT	1.03	.95	.98
1.04	1.00	1.06	.82	.99	-3.2	.99	-13.6	RAKE MOUNTING POINT	.98	1.00	2.2
1.04	1.1	5.8	.96	.76	1.6	.76	21.5	RAKE MOUNTING POINT	.98	.98	-5.8
1.02	.97	1.02	.42	.53	.53	.53	.54	1.00	.96	1.00	1.00
.98	1.03	1.02	.32	.76	.76	.76	.80	.95	1.00	.99	.99
.98	1.02	1.03	.75	.77	.77	.77	.72	.83	.96	.98	.96

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{\rho V^2}{2}}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
 TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE NOSE, LOOKING AFT.

RAKE MOUNTING POINT → 74.75 IN

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{\rho V^2}{2}}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
 TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE NOSE, LOOKING AFT.

RAKE MOUNTING POINT → 74.75 IN

TABLE VIII - Continued

RUN 22 POINT 15 ALPHA= .0 DEG PSI=-4.0 DEG V=100 KNOTS Q= 31.6 PSF. CONFIGURATION FH1				RUN 22 POINT 16 ALPHA= .0 DEG PSI= 4.0 DEG V=100 KNOTS Q= 31.6 PSF. CONFIGURATION FH1			
.09	.98	.99	.97	.98	1.00	.99	1.01
1.00	1.00	.97	1.00	.98	1.01	1.01	1.00
1.05	.97	1.00	.98	.99	1.00	1.01	.96
.98	1.02	1.00	1.01	.98	1.01	1.00	.96
	.99	1.00	.97	.98	1.04	1.01	.96
	.97	.96	1.00	.99	1.05	.98	1.01
	1.01	1.04	.97	.96	1.01	1.07	.99
.97	.76	.96	1.00	.95	.98	1.06	.95
1.01	1.02	1.02	.94	.96	.88	.84	.77
	.92	.90	.94	1.02	.99	1.09	.41
	.64	.67	1.15	1.06	.61	.92	.62
.96	.27	.68	.94	1.01	.92	.80	.26
1.03	.60	.86	.98	.98	.63	1.06	.42
1.02	.75	.73	1.01	1.02	.99	.75	.50
1.01	1.13	.79	1.01	1.01	.92	.74	.90
	.64	1.06	.98	1.02	1.03	.61	.70
	.91	.84	1.02	.94	.94	1.07	.79
	.72	.65	.97	1.01	1.00	.62	.91
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2}\rho V^2}$ NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH. LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.				NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2}\rho V^2}$ NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH. LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.			
NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.				NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.			

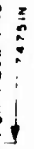
TABLE VIII - Continued

ROW 27 POINT 17 ALPHA = 0 DEG PSI = 8.0 DEG V = 100 KNOTS q = 31.0 PSF. CONFIGURATION FWH		ROW 26 POINT 5 ALPHA = 0 DEG PSI = -0 DEG V = 150 KNOTS q = 69.1 PSF. CONFIGURATION FWH							
.96	1.00	1.01	.99	.97	1.00	1.00	.98	1.00	.98
.97	1.00	1.01	.97	.97	1.01	1.01	.95	.98	.95
.98	1.00	.96	.98	.98	.99	.99	.97	.97	.97
.99	1.00	.97	.96	.97	.99	1.02	.98	.97	.98
.96	.99	.99	.99	.97	1.01	1.03	.99	.97	.99
.97	1.01	.97	.98	.97	.98	.99	.99	.99	.99
.98	.99	1.00	.96	.98	1.01	1.00	.97	.97	.98
.97	.99	.97	.95	1.06	1.00	.97	1.02	1.02	.98
.98	1.00	.96	1.03	.96	1.03	1.04	.97	.97	1.01
.97	1.02	1.03	.71	.95	.94	.77	.79	.79	.95
.98	1.05	.46	.91	.57	.75	.94	.89	.89	.98
.99	.99	.56	.86	.69	.37	1.13	.69	.69	1.11
.96	.99	.27	.95	.42	.82	.40	.53	.53	.94
.97	1.04	.89	.44	.66	.91	RAKE MOUNTING POINT	.72	.72	.97
1.03	1.00	.90	.61	.88	.74	RAKE MOUNTING POINT	.7	.7	1.0
1.02	1.08	.90	.61	.88	.74	RAKE MOUNTING POINT	1.05	1.05	.95
1.01	1.01	.95	.70	1.01	1.00	RAKE MOUNTING POINT	.51	.51	.90
1.01	.99	.95	.96	1.03	.71	RAKE MOUNTING POINT	1.02	1.02	1.01
1.02	.99	.64	1.04	1.02	.99	RAKE MOUNTING POINT	.97	.97	.98
1.01	.96	.96	.23	.98	.91	RAKE MOUNTING POINT	1.05	1.05	.96

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.



RAKE MOUNTING POINT

RAKE MOUNTING POINT

TABLE VIII - Continued

R/JN 26 POINT 6 ALPHA=4.0 DEG PSI=7.0 DEG V=150 KNOTS q=69.1 PSF. CONFIGURATION FMH		R/JN 25 POINT 7 ALPHA=5.1 DEG PSI=7.0 DEG V=150 KNOTS q=69.1 PSF. CONFIGURATION FMH					
.07	.98	.99	.97	.97	.97	.97	.97
.97	.99	1.00	.97	.96	.95	.95	.95
	.97	.98	.98	.98	.96	.96	.97
.96	.98	.98	.97	.98	.96	.96	.97
	.98	.96	.97	.99	.97	.97	.95
1.07	.96	.97	.99	1.00	.96	.96	.96
	.97	.96	.99	1.00	.97	.99	.94
.99	.97	.96	1.00	.99	.91	.98	.95
	1.01	1.03	.98	1.09	1.02	1.01	1.00
	.87	1.03	.95	.99	.64	1.01	.79
.98	1.04	.97	1.01	.82	.90	.88	.50
	.86	.66	.90	.88	.77	1.16	.67
.95	.56	.68	.77	.59	.67	.91	.50
	.71	.89	RAKE MOUNTING POINT	.43	.59	RAKE MOUNTING POINT	.57
1.00	.61	.63	RAKE MOUNTING POINT	.55	.65	RAKE MOUNTING POINT	.70
	1.00	.66	RAKE MOUNTING POINT	.81	.82	.90	.90
.99	.67	.39	.54	.81	.55	1.07	1.07
	1.05	.95	.66	.89	1.08	.54	.98
.99	.94	1.01	.43	1.00	1.01	.53	.80

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.



NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.



TABLE VIII - Continued

RUN 26 POINT 8 ALPHA= 2.0 DEG PSI= -0 DEG V= 150 KNOTS q = 69.1 PSF. CONFIGURATION FM		RUN 26 POINT 9 ALPHA= 4.0 DEG PSI= -0 DEG V= 150 KNOTS q = 69.1 PSF. CONFIGURATION FM	
.09	.97	.98	.96
.97	.97	.97	.99
.97	.98	.97	.96
.98	1.03	.97	.96
.98	.98	.98	.97
.96	.96	.97	.98
.99	.97	.97	.96
.62	1.01	.81	.94
.85	.85	.45	.94
.74	.69	.44	.94
.78	.49	1.01	.63
.52	.51	.86	.59
.63	.38	1.01	.62
.50	.65	.97	1.00
.97	.87	.97	.95
.94	.87	1.17	.96
1.01	.79	.92	1.01
.99	.67	.93	1.01
.97	.94	.16	.99
1.15	.98	.98	.98
.99	.99	.99	.96
.98	.97	.97	.96
.99	1.02	1.02	.90
.98	.94	.94	.97
5.2	.91	.78	.80
1.4	.89	2.1	.57
.77	.45	.45	.87
.83	.59	.81	.86
.83	.32	.82	.54
.70	.80	.80	.77
.93	.60	.60	.44
.86	1.4	1.4	.84
.92	.84	.84	1.13
.96	.53	.53	.96
.98	.89	.89	1.01
.99	.74	.74	.86
1.01	4.0	4.0	3.9
.86	4.3	4.3	1.05
.92	.1	.1	.97
.96	.68	.68	1.00
.98	.68	.68	1.07
.99	.12	.12	.99
1.01	4.0	4.0	3.9
.86	4.3	4.3	1.05
.92	.1	.1	.97
.96	.68	.68	1.00
.98	.68	.68	1.07
.99	.12	.12	.99

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS C_p .
 NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT
 TO RIGHT LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS C_p .
 NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT
 TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW
 MODEL NOSE, LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW
 MODEL NOSE, LOOKING AFT.

TABLE VIII - Continued

RUN 26 POINT 10 ALPHA=4.0 DEG PSI=-2.0 DEG V=150. KNOTS q=69.1 PSF. CONFIGURATION FM		RUN 26 POINT 11 ALPHA=2.0 DEG PSI=-2.0 DEG V=150. KNOTS q=69.1 PSF. CONFIGURATION FM	
.08	.98	.97	.98
1.00	.99	.97	.99
.07	.98	.97	.97
1.01	4.6	.98	2.5
.96	1.00	.98	-3.7
.97	.99	.98	1.00
1.00	.98	.98	.98
.91	1.00	.98	.98
.96	.97	.97	.98
.88	1.01	.92	.96
.79	.94	.91	.71
.51	8.3	.90	3.6
.49	.69	1.02	1.03
.85	-6.4	.86	-8.6
.82	.97	.66	.87
.56	1.01	.88	.92
.60	1.03	.83	.42
.54	1.00	.90	.92
.59	.98	.65	1.00
.86	.97	RAKE MOUNTING POINT	1.04
.35	4.7	.47	2.7
.61	-3.2	.71	-3.2
.44	1.01	.75	1.01
.52	.98	.55	.98
.98	.98	.52	.99
1.00	1.02	.90	.99
.98	.98	.95	1.00
.92	.97	.70	.98
.97	1.02	.90	.99
.91	.98	.90	.99
2.7	-2.3	1.2	1.2
-9.8	5.1	-8.9	-8.9
-2.3	-3.3	1.0	1.0
.91	.98	.83	.83
.98	.97	.91	.91
.54	.98	.95	.95
.47	.98	.70	.70
.44	.98	.52	.52
1.00	1.01	1.00	1.00
1.00	1.02	1.00	1.00
1.00	1.01	.97	.99

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW
NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW
NOSE, LOOKING AFT.

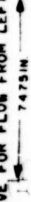


TABLE VIII - Continued

MUN 26 POINT 12 ALPHA=-2.0 DEG PSI=-2.0 DEG V=150 KNOTS q = 69.1 PSF. CONFIGURATION FHJ					MUN 26 POINT 13 ALPHA= 2.0 DEG PSI= 2.0 DEG V=150 KNOTS q = 69.1 PSF. CONFIGURATION FKH				
.99	.98	.97	.96	.95	.98	.97	.96	.95	.94
.97	.99	.98	.96	.95	.99	.98	.96	.95	.94
.97	.96	.97	.96	.95	1.00	.98	.96	.95	.94
.98	1.02	.97	.95	.94	.98	.99	.93	.90	.99
1.00	-2.9	.97	.92	.83	1.01	.96	.95	.91	.51
1.01	.2	.91	.83	.74	1.00	3.8	1.9	.73	-3.1
.99	.87	.86	.81	.74	1.00	2.6	4.2	.85	-3.2
.99	.72	.85	.31	.46	.99	.99	.79	.77	.91
1.02	.58	.93	.86	.91	.99	.96	.63	.67	.63
1.00	.23	.84	.91	.84	1.00	2.1	.51	.51	.31
1.00	.63	1.00	1.09	1.07	.99	2.2	.70	.42	.37
.99	-6.6	-15.7	-3.2	-9	1.00	-1.9	.94	.65	.58
1.00	.82	.50	.54	.99	1.00	1.08	2.2	.92	-3.3
.99	.41	1.02	.80	.99	.99	.77	1.9	.83	5.1
.99	.91	.71	.80	.98	1.00	.95	.99	.83	1.01

MUN 26 POINT 12 ALPHA=-2.0 DEG PSI=-2.0 DEG V=150 KNOTS q = 69.1 PSF. CONFIGURATION FKH					MUN 26 POINT 13 ALPHA= 2.0 DEG PSI= 2.0 DEG V=150 KNOTS q = 69.1 PSF. CONFIGURATION FKH				
.99	.98	.97	.96	.95	.98	.97	.96	.95	.94
.97	.99	.98	.96	.95	.99	.98	.96	.95	.94
.97	.96	.97	.96	.95	1.00	.98	.96	.95	.94
.98	1.02	.97	.95	.94	.98	.99	.93	.90	.99
1.00	-2.9	.97	.92	.83	1.01	.96	.95	.91	.51
1.01	.2	.91	.83	.74	1.00	3.8	1.9	.73	-3.1
.99	.87	.86	.81	.74	1.00	2.6	4.2	.85	-3.2
.99	.72	.85	.31	.46	.99	.99	.79	.77	.91
1.02	.58	.93	.86	.91	.99	.96	.63	.67	.63
1.00	.23	.84	.91	.84	1.00	2.1	.51	.51	.31
1.00	.63	1.00	1.09	1.07	.99	2.2	.70	.42	.37
.99	-6.6	-15.7	-3.2	-9	1.00	-1.9	.94	.65	.58
1.00	.82	.50	.54	.99	1.00	1.08	2.2	.92	-3.3
.99	.41	1.02	.80	.99	.99	.77	1.9	.83	5.1
.99	.91	.71	.80	.98	1.00	.95	.99	.83	1.01

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW

MUDEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

7.475 IN

TABLE VIII - Continued

RUN 26 POINT 14 ALPHA = .0 DEG PSI = -4.0 DEV V = 150 KNOTS q = 69.1 PSF. CONFIGURATION FH1		RUN 26 POINT 15 ALPHA = .0 DEG PSI = -2.0 DEV V = 150 KNOTS q = 69.1 PSF. CONFIGURATION FH1	
.97	.95	.98	.99
.96	.96	.97	.98
.97	.97	.96	.97
.98	.98	.97	.98
.99	.99	.98	.99
.94	.94	.98	.98
1.00	.96	.99	.99
1.00	.94	.98	.99
.97	1.04	.98	1.02
-1.4	-9.0	.89	-6.0
-2.0	-6.4	.95	-2.6
1.00	.79	1.07	1.00
1.01	.94	.98	.92
.74	.74	.97	.85
.73	.78	1.02	.97
.69	.63	.81	.78
.55	.73	.91	.84
.57	.73	.85	.84
.62	.73	.99	.85
.76	.82	.97	.77
.94	.93	.96	.88
.79	.78	.98	.66
.97	.97	.97	.96
.98	.98	.98	.98
.99	.99	.99	.99
1.00	1.00	1.00	1.00
1.01	1.01	1.01	1.01
1.02	1.02	1.02	1.02
1.03	1.03	1.03	1.03
1.04	1.04	1.04	1.04
1.05	1.05	1.05	1.05
1.06	1.06	1.06	1.06
1.07	1.07	1.07	1.07
1.08	1.08	1.08	1.08
1.09	1.09	1.09	1.09
1.10	1.10	1.10	1.10
1.11	1.11	1.11	1.11
1.12	1.12	1.12	1.12
1.13	1.13	1.13	1.13
1.14	1.14	1.14	1.14
1.15	1.15	1.15	1.15
1.16	1.16	1.16	1.16
1.17	1.17	1.17	1.17
1.18	1.18	1.18	1.18
1.19	1.19	1.19	1.19
1.20	1.20	1.20	1.20
1.21	1.21	1.21	1.21
1.22	1.22	1.22	1.22
1.23	1.23	1.23	1.23
1.24	1.24	1.24	1.24
1.25	1.25	1.25	1.25
1.26	1.26	1.26	1.26
1.27	1.27	1.27	1.27
1.28	1.28	1.28	1.28
1.29	1.29	1.29	1.29
1.30	1.30	1.30	1.30
1.31	1.31	1.31	1.31
1.32	1.32	1.32	1.32
1.33	1.33	1.33	1.33
1.34	1.34	1.34	1.34
1.35	1.35	1.35	1.35
1.36	1.36	1.36	1.36
1.37	1.37	1.37	1.37
1.38	1.38	1.38	1.38
1.39	1.39	1.39	1.39
1.40	1.40	1.40	1.40
1.41	1.41	1.41	1.41
1.42	1.42	1.42	1.42
1.43	1.43	1.43	1.43
1.44	1.44	1.44	1.44
1.45	1.45	1.45	1.45
1.46	1.46	1.46	1.46
1.47	1.47	1.47	1.47
1.48	1.48	1.48	1.48
1.49	1.49	1.49	1.49
1.50	1.50	1.50	1.50
1.51	1.51	1.51	1.51
1.52	1.52	1.52	1.52
1.53	1.53	1.53	1.53
1.54	1.54	1.54	1.54
1.55	1.55	1.55	1.55
1.56	1.56	1.56	1.56
1.57	1.57	1.57	1.57
1.58	1.58	1.58	1.58
1.59	1.59	1.59	1.59
1.60	1.60	1.60	1.60
1.61	1.61	1.61	1.61
1.62	1.62	1.62	1.62
1.63	1.63	1.63	1.63
1.64	1.64	1.64	1.64
1.65	1.65	1.65	1.65
1.66	1.66	1.66	1.66
1.67	1.67	1.67	1.67
1.68	1.68	1.68	1.68
1.69	1.69	1.69	1.69
1.70	1.70	1.70	1.70
1.71	1.71	1.71	1.71
1.72	1.72	1.72	1.72
1.73	1.73	1.73	1.73
1.74	1.74	1.74	1.74
1.75	1.75	1.75	1.75
1.76	1.76	1.76	1.76
1.77	1.77	1.77	1.77
1.78	1.78	1.78	1.78
1.79	1.79	1.79	1.79
1.80	1.80	1.80	1.80
1.81	1.81	1.81	1.81
1.82	1.82	1.82	1.82
1.83	1.83	1.83	1.83
1.84	1.84	1.84	1.84
1.85	1.85	1.85	1.85
1.86	1.86	1.86	1.86
1.87	1.87	1.87	1.87
1.88	1.88	1.88	1.88
1.89	1.89	1.89	1.89
1.90	1.90	1.90	1.90
1.91	1.91	1.91	1.91
1.92	1.92	1.92	1.92
1.93	1.93	1.93	1.93
1.94	1.94	1.94	1.94
1.95	1.95	1.95	1.95
1.96	1.96	1.96	1.96
1.97	1.97	1.97	1.97
1.98	1.98	1.98	1.98
1.99	1.99	1.99	1.99
2.00	2.00	2.00	2.00

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT
 TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW
 MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT
 7.475 IN

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT
 TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW
 MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT
 7.475 IN

TABLE VIII - Concluded

RUN 26 POINT 16 ALPHA = .0 DEG PSI = 2.0 DEG V = 150 KNOTS q = 69.1 PSF. CONFIGURATION FH1				RUN 26 POINT 17 ALPHA = .0 DEG PSI = 4.0 DEG V = 150 KNOTS q = 69.1 PSF. CONFIGURATION FH1			
.07	.98	.96	.97	.98	.98	.96	.98
.97	.98	.97	.99	.98	.99	.95	.97
.97	.98	.97	.95	.97	.99	.94	.99
.97	.98	.98	.97	.96	.97	.96	.97
.99	.97	.94	.97	.97	.98	.97	.97
.98	.96	.97	.98	.95	.97	.95	.96
.99	.98	.93	.82	.95	.98	.98	.93
.98	.94	.88	.74	.96	.96	.92	.84
.98	.93	.87	1.16	.95	.96	.99	.94
.98	1.00	.96	1.00	.93	.97	.86	.78
.98	.94	.80	.75	.99	.75	1.02	.69
.97	.83	.67	.68	.98	.74	.58	.62
.97	.83	.09	.26	1.04	.81	.80	.10
1.00	.96	.56	.93	1.01	.67	1.01	.31
1.00	.99	.91	1.12	.7	.64	1.18	.71
1.00	.93	.97	.92	1.2	1.07	.40	.64
1.01	.93	1.06	1.08	.97	1.05	.96	.90
.99	.96	.94	.62	1.00	1.03	.87	.87
.99	.97	.96	.75	1.01	1.01	.82	.82
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$ NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.				NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$ NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.			
NOTE - TABLE IS SEEN FROM THE CLING MODEL NOSE, LOOKING AFT.				NOTE - TABLE IS SEEN FROM THE CLING MODEL NOSE, LOOKING AFT.			

TABLE IX. WAKE PRESSURE COEFFICIENTS AND FLOW DIRECTIONS AT THREE FORWARD SPEEDS AND VARIOUS FUSELAGE ATTITUDES AND ROTOR OPERATING CONDITIONS, CONFIGURATION FHB

NUM 14 POINT 5 ALPHA=0.0 DEG PSI=-0.0 DEG V=60 KIOTS		ALPHA=8.0 DEG PSI=-0.0 DEG CONFIGURATION FHB		NUM 14 POINT 6 ALPHA=8.0 DEG PSI=-0.0 DEG V=60 KIOTS		CONFIGURATION FHB	
1.02	.99	1.01	1.05	.97	1.02	.96	.95
1.05	1.07	1.06	1.00	.99	.94	1.02	1.00
1.08	1.08	.92	-2.8	1.00	1.04	.96	-10.5
1.05	1.09	.94	-2.0	.98	1.00	.91	-3.7
1.02	1.12	1.01	1.03	.92	.91	1.00	.94
1.05	1.30	1.08	1.11	.91	.88	.96	.85
1.02	.92	.74	.89	.87	.90	1.00	.91
.94	.86	.62	.75	1.04	1.03	.84	1.18
1.02	.83	.71	-3.3	1.15	1.11	.56	-12.0
1.05	1.03	1.07	1.3	1.32	-0.1	1.22	1.13
1.02	.96	.91	.69	1.43	1.01	1.69	.95
1.05	1.04	.71	.98	1.04	1.21	1.07	1.12
1.14	.93	.43	.47	.90	1.34	.53	.67
1.07	.95	.93	.58	1.04	.91	.59	.92
1.01	1.01	.94	-1.5	1.13	.95	.82	1.02
	.94	.77	1.7	1.11	.84	.72	.60
	.94	.94	.67	1.28	1.04	.70	.40
	.94	.94	.97	1.27	1.07	.60	.94
	.94	.94	.95	1.28	1.05	.53	.95
	.94	.94	.95	1.28	1.05	.53	.95

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\rho q}$
 NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UP-ASH SURVEY
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.
 NOTE - TABLE IS SEEN FROM THE FLOW MOUNTING POINT
 MODEL NOSE, LOOKING AFT. 74.75 IN

TABLE IX - Continued

RUN 14 POINT 7 VE = 60 KNOTS		ALPHA = 8.0 DEG PSI = -0 DLG CONFIGURATION PHB		RUN 14 POINT 8 VE = 60 KNOTS		ALPHA = 4.0 DEG PSI = -0 DLG CONFIGURATION PHB	
.94	.98	.90	1.02	.98	1.00	1.05	1.00
.94	.94	.98	.99	1.02	1.00	1.08	1.01
.96	.92	1.00	.97	.98	1.06	.96	1.02
.88	.88	.92	.93	.94	1.06	.92	1.03
.95	.93	.87	.84	1.01	1.07	.88	1.09
1.04	.70	.98	1.02	.95	.88	1.00	.70
.93	1.09	.93	1.16	1.11	1.00	1.13	1.08
1.15	1.14	1.08	1.14	1.10	1.10	1.19	1.22
1.32	1.42	1.03	1.06	1.03	1.20	.91	.86
1.14	1.14	1.15	1.20	.82	.72	.83	.94
1.10	1.00	1.20	1.02	1.01	.50	.83	.99
1.05	.60	.86	1.15	1.13	.87	.37	.93
.91	.92	.94	.87	1.13	.87	.33	.80
.97	.97	.33	1.12	1.15	.91	.56	.22
1.16	.96	.68	.73	1.10	1.09	.08	.76
1.07	1.07	.48	1.01	1.06	1.06	.42	.93
1.09	1.16	.85	.87	1.02	1.05	1.05	.99
1.16	1.16	.80	1.01	1.06	.94	.81	1.01
1.01	1.01	.63	1.02	1.02	1.04	.97	.91
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS, $C_p = \frac{P - P_\infty}{q}$				NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS, $C_p = \frac{P - P_\infty}{q}$			
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UP, ASH SURVEY LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.				NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UP, ASH SURVEY LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.			
NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.				NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.			

TABLE IX - Continued

RUN 14 POINT 9 V= 60. KNOTS		ALPHA= 4.0 DEG q= 11.5 PSF. CONFIGURATION FHB		ALPHA= 8.0 DEG q= 11.5 PSF. CONFIGURATION FHB		RUN 14 POINT 10 V= 60. KNOTS		ALPHA= 8.0 DEG q= 11.5 PSF. CONFIGURATION FHB		
.99	.95	1.04	.83	.82	.93	.65	.80	.70	.59	.77
.95	.95	.94	.79	.89	.78	.75	.70	.60	.69	.80
.95	.98	.84	.84	.93	.78	.84	.58	.61	.74	.80
.29	.99	1.04	.94	.89	.90	.70	.45	.55	.90	1.24
.96	.92	1.12	.91	.94	.90	.57	.47	.77	.51	1.24
.96	.77	.80	1.27	.75	1.09	.61	.62	.97	.68	1.10
.96	.67	.60	.45	.93	1.09	1.02	.90	.93	.83	1.10
.97	.75	.54	1.16	1.08	.99	.92	.87	.78	.97	1.01
.97	-2.7	.40	.84	1.01	.7	1.09	.97	.85	1.14	1.01
.99	-1.1	.42	.62	.99	.93	.99	1.06	.95	1.09	1.0
.99	.91	.94	.76	.91	.93	.94	.82	.96	.94	.99
.99	.98	.76	.81	.87	.96	.72	1.04	1.00	1.00	.94
.99	.95	.42	.76	.96	.96	.88	.93	.74	.92	.94
1.19	.92	.33	.53	1.10	.99	.89	.83	.99	.95	.99
1.07	.6	1.0	.61	.87	.99	.89	.63	.94	1.07	.99
1.02	-2.9	.97	.94	.98	.88	1.00	.90	.90	1.15	.91
	.91	.87	.47	.91	.95	.99	.50	1.00	1.00	.95
	1.01	.94	.98	1.00	.88	1.01	.91	.93	1.00	.95
	1.00	.53	.75	.95	.88	.93	.93	1.01	.90	.91

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE
TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW
MODEL NOSE, LOOKING AFT.



TABLE IX - Continued

RUN 14 POINT 11 ALPHA=0 DEG PSI=0.0 DEG V=60. KNOTS q=11.5 PSF. CONFIGURATION FHB					RUN 14 POINT 12 ALPHA=8.0 DEG PSI=0.0 DEG V=60. KNOTS q=11.5 PSF. CONFIGURATION FHB				
.97	.87	1.07	1.04	.94	.98	.92	.98	.94	.99
.90	.93	1.00	1.06	.82	.98	.92	.98	.91	1.00
.92	1.07	.94	1.13	.7	.76	.91	.98	.92	.93
1.25	1.15	1.20	1.14	-8.3	1.15	.89	.88	.89	.95
.99	1.22	1.22	1.22	1.10	.94	.94	.98	.96	.89
1.17	1.11	1.09	1.26	1.02	1.03	.87	.88	.92	.95
.60	.94	.85	1.16	.96	1.03	1.00	.95	.74	1.11
.66	.97	.97	1.05	1.01	.95	1.08	1.05	1.11	1.21
.82	.97	.81	1.03	-1.6	1.01	.89	1.19	1.13	1.29
.46	.28	.50	.95	-7.6	1.09	-4.3	-9.4	1.44	2.3
.96	.57	.73	.96	1.05	.95	1.65	1.25	1.41	1.47
1.02	1.10	.37	.99	1.09	1.00	1.06	1.13	1.05	1.27
1.06	.32	.15	1.01	.99	1.00	1.06	.71	.92	.92
.95	.65	.93	.96	.98	.98	.78	.64	.55	.92
	-3.4	2.4	RAKE MOUNTING POINT	-1.7	.5	-15.7	-14.1	RAKE MOUNTING POINT	-7.4
	-1.8	3.7		-7.0	-4.6	-3.6	-2.4		-7
	1.07	.70	.96	1.00	.98	1.04	.68	.79	1.03
	1.05	.80	1.01	1.02	1.00	1.03	.51	.94	.69
	.98	.93	.94	.96	1.00	1.03	.57	1.06	.69
	.68	.42	1.02	1.02	.99	.85	.50	1.00	1.06
	.56	.51	1.05	.98	.99	.95	.73	1.05	1.00
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$					NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$				
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.					NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.				
NOTE- TABLE IS SEEN FROM THE DOWN MODEL NOSE, LOOKING AFT.					NOTE- TABLE IS SEEN FROM THE DOWN MODEL NOSE, LOOKING AFT.				

TABLE IX - Continued

RUN 14 POINT 13 ALPHA=8.0 DEG PSI=4.0 DEG V= 60 KNOTS q = 11.5 PSF. CONFIGURATION FHB										RUN 14 POINT 16 ALPHA=4.0 DEG PSI=4.0 DEG V= 60 KNOTS q = 11.5 PSF. CONFIGURATION FHB									
.98	.99	1.00	1.01	1.02	.96	.97	.98	.99	1.00	.98	.99	1.00	1.01	1.02	.96	.97	.98	.99	1.00
.94	.94	1.01	1.01	.97	1.00	1.00	.97	.97	1.00	.92	.92	1.01	1.01	.96	.96	.96	.96	.96	1.01
.98	.98	1.05	.98	.98	-8.4	.99	.98	.98	.98	.74	.83	.83	.89	.88	.88	.88	.88	.88	.88
.91	.91	1.04	.99	.99	-5.9	1.04	.99	.99	.99	.91	.93	.93	.99	.93	.93	.93	.93	.93	.93
1.05	1.05	1.07	.99	.92	.99	.99	.95	.95	.95	.74	.96	1.29	1.40	.89	.89	.89	.89	.89	.89
1.03	1.03	.94	.99	.92	.95	.95	.95	.95	.95	.60	1.21	1.09	1.19	1.18	1.18	1.18	1.18	1.18	1.18
.94	.94	.79	1.13	1.09	.79	1.05	1.05	1.05	1.05	.44	1.27	1.27	1.27	1.24	1.24	1.24	1.24	1.24	1.24
1.43	1.43	1.42	.97	1.27	1.96	1.18	1.18	1.18	1.18	.86	.54	1.05	1.05	1.02	1.02	1.02	1.02	1.02	1.02
1.17	1.17	1.27	1.06	1.11	-11.3	.93	1.18	1.18	1.18	-2.5	-7.5	.43	.47	.78	.78	.78	.78	.78	.78
1.08	1.08	-0.7	1.26	1.02	-5.4	.86	-4.9	-4.9	-4.9	-4.4	-3.3	.33	.81	1.04	1.04	1.04	1.04	1.04	1.04
.97	.97	.96	1.11	.94	.96	.96	.99	.99	.99	1.02	.93	.93	.93	.93	.93	.93	.93	.93	.93
1.01	1.01	.73	.99	1.01	1.16	1.16	1.26	1.26	1.26	1.02	.73	.64	.64	1.04	1.04	1.04	1.04	1.04	1.04
.75	.75	.74	.95	1.00	1.19	1.19	1.26	1.26	1.26	.92	1.05	.57	.57	.56	.56	.56	.56	.56	.56
.64	.64	.85	RAKE MOUNTING POINT	1.04	1.26	1.26	1.26	1.26	1.26	.99	1.04	1.04	1.04	.93	.93	.93	.93	.93	.93
1.15	1.15	.40	RAKE MOUNTING POINT	.59	1.00	1.00	1.15	1.15	1.15	-2.2	-2.7	.85	.85	1.00	1.00	1.00	1.00	1.00	1.00
.64	.64	.52	1.09	.95	-3.3	1.09	1.14	1.14	1.14	-2.7	3.3	.93	.93	-3.7	-3.7	.98	.98	.98	.98
1.17	1.17	.65	.94	.94	1.00	1.00	1.02	1.02	1.02	.66	1.20	1.04	1.04	.98	.98	.97	.97	.97	.97
.78	.78	.14	.59	.59	.98	.98	.98	.98	.98	.92	.72	.69	.69	.95	.95	.94	.94	.94	.94
.98	.98	.79	.99	.95	.98	.98	.96	.96	.96	.81	.83	.83	.83	.97	.97	1.02	1.02	1.02	1.02

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

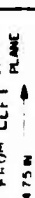


TABLE IX - Continued

NUM 14 POINT 17 ALPHA=4.0 DEG PSI=4.0 DEG V= 60 KNOTS q = 11.5 PSF. CONFIGURATION FHB										NUM 14 POINT 18 ALPHA=4.0 DEG PSI=4.0 DEG V= 60 KNOTS q = 11.5 PSF. CONFIGURATION FHB																
.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96	.94	.97	.94	.98	.96	.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96
.99	.97	.97	.89	1.11	1.39	.94	.94	1.06	1.01	.83	.96	.96	.95	.95	.95	.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96
.82	1.12	1.09	1.09	.94	.94	.96	.96	1.12	.94	1.02	.97	.97	.97	1.02	1.00	.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96
.78	.75	.74	1.12	.75	.85	.89	.84	.79	.94	.85	.84	1.06	1.06	1.16	1.00	.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96
.83	1.13	.85	.79	.94	.80	1.01	.76	.72	.76	.80	1.08	1.08	.98	1.05	1.14	.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96
.98	1.03	.91	1.23	.56	.91	.97	1.01	.83	1.01	.91	.96	1.12	1.16	1.23	1.14	.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96
.98	.91	.87	.74	.82	.91	.87	.77	.82	.77	.91	.93	1.21	.93	1.13	.94	.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96
1.00	.94	.94	1.09	.87	.85	.94	.67	1.9	4.5	1.02	.83	1.07	.74	.79	1.1	.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96
.99	1.02	.80	.48	.89	.64	.89	.64	.89	.64	1.02	.95	1.04	.91	.71	1.05	.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96
1.03	1.03	.50	.55	.58	1.12	.58	1.12	.58	1.12	1.02	.95	.94	.53	.62	.98	.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96
.99	.96	1.01	.83	.85	.85	.85	.85	.85	.85	1.02	1.00	1.02	.77	.92	.98	.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96
1.12	.94	.94	RAKE MOUNTING POINT	1.06	.79	1.06	.79	.72	.80	.98	1.00	.26	RAKE MOUNTING POINT	.65	.98	.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96
1.01	.90	.96	RAKE MOUNTING POINT	.72	.80	.72	.80	.96	.80	.98	.96	.87	RAKE MOUNTING POINT	.63	.97	.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96
.95	.96	.96	.74	.64	.55	.94	.64	.96	.64	.94	.92	.92	.61	1.15	.95	.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96
	1.05	1.05	.72	.91	.56	1.02	.91	.87	.56	.94	.96	.92	.48	1.02	.95	.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96
	1.01	1.01	.74	.68	.55	.94	.68	.68	.55	.94	.96	1.02	.51	1.08	.88	.95	.99	.99	.96	1.04	.99	.93	.97	.94	.98	.96

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT → 74.75 IN

TABLE IX - Continued

RUN 14 POINT 19 ALPHA=-8.0 DEG PSI=4.0 DEG V= 60. KNOTS q= 11.5 PSF. CONFIGURATION FHB				RUN 14 POINT 20 ALPHA=-8.0 DEG PSI=4.0 DEG V= 60. KNOTS q= 11.5 PSF. CONFIGURATION FHB			
.98	.94	1.01	1.05	.98	.95	.98	.95
.97	.97	1.04	1.06	.98	.90	.92	.95
.95	.95	.97	1.03	.85	-10.4	.96	.91
.92	.95	1.00	.97	.93	.7	.94	.94
.72	1.02	.95	.85	.99	.93	.99	.97
	1.05	1.05	.97	1.07	1.04	.99	1.06
	1.09	1.11	1.06	1.14	1.14	.98	1.00
	1.25	1.04	1.14	1.16	.95	.84	1.01
1.23	-16.8	-11.3	1.15	1.49	-13.0	-10.2	-12.8
	-4	-1.1	1.13	1.32	1.16	.78	.41
1.10	1.03	1.38	1.07	1.32	1.8	1.50	1.2
	1.07	1.15	1.13	1.03	.91	1.21	1.18
.93	1.05	.95	.94	.80	.67	1.11	1.17
	1.06	.95	.85	.83	.69	.93	.94
	.99	1.00	RAKE MOUNTING POINT	.61	.59	RAKE MOUNTING POINT	.75
1.18	-10.4	-9.0	1.00	.63	-18.9	-12.5	-11.1
	3.6	-5.1	1.14	.88	4.5	-1.4	.69
1.04	1.18	1.15	.94	.32	.80	.99	.79
	1.02	1.07	.76	.34	.88	.97	.64
.97	1.06	.93	.91	.91	.66	1.10	.48

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.



NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.



TABLE IX - Continued

RUN 16 POINT 21 ALPHA = .0 DEG PSI = 0.0 DEG VE = 60 KNOTS $\alpha = 11.5$ PSF. CONFIGURATION FHB				RUN 15 POINT 5 ALPHA = .0 DEG PSI = -0.0 DEG VE = 60 KNOTS $\alpha = 11.5$ PSF. CONFIGURATION FHB			
1.03	1.03	1.04	.99	1.05	1.04	1.00	.93
	.99	1.01	.96	1.04	.98	1.00	.97
1.00	.92	1.00	.89	1.02	1.01	.98	.90
	.62	.87	.51	.98	.81	.99	.87
1.06	1.24	1.51	1.25	1.47	1.02	1.09	.89
	1.05	1.20	1.02	1.50	.89	.96	.96
.96	.95	1.02	.90	.95	.96	.75	.82
	.92	.87	.99	.79	.94	.02	1.04
.85	.94	1.02	1.06	.64	.60	.40	.81
	.96	.97	1.05	.80	.70	.57	.57
.99	.94	1.00	1.05	1.09	.57	.80	.89
	.98	1.01	1.02	.76	.39	.75	.91
.97	.96	1.00	.49	.76	.68	.01	.85
	.95	.93	.97	.90	.67	RAKE MOUNTING POINT	.15
1.08	1.00	.94	.34	.29	.88	.37	.05
	.95	1.03	.50	.76	.47	1.07	.98
1.00	1.01	1.00	.74	.95	.66	.59	1.02
	1.02	1.00	.59	.76	1.01	.52	.95
1.02	1.03	1.03	.52	.71	.97	.83	.94

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE IX - Continued

RUN 15 POINT 6 ALPHA=-8.0 DEG PSI=-0.0 DEG VE 60 KNOTS q = 11.5 PSF. CONFIGURATION FHB				RUN 15 POINT 7 ALPHA=-8.0 DEG PSI=-0.0 DEG VE 60 KNOTS q = 11.5 PSF. CONFIGURATION FHB			
.94	.96	.97	.95	.97	.94	.97	1.01
.92	.93	.94	.94	.94	.97	.95	1.01
.96	.99	.92	.95	.92	1.00	1.00	1.05
.94	.99	.92	1.04	.92	1.01	1.04	.95
.94	.99	1.02	1.04	1.08	.90	1.00	.87
.94	.85	.97	.98	1.02	.96	1.04	.87
1.07	.97	.84	.84	1.03	1.00	1.09	.89
1.07	1.00	.94	1.05	1.15	1.11	1.06	.89
.96	.87	1.03	1.12	1.04	1.00	.99	1.00
.99	1.07	1.07	.94	1.13	1.07	.83	1.00
.99	1.09	1.00	.79	.91	.71	1.01	1.06
1.11	1.03	.72	.49	.88	.70	.96	1.06
1.06	.81	.72	.52	.75	.80	.84	1.06
.99	.93	.54	.26	.75	1.00	.87	1.06
.92	.85	.42	.01	.4	.50	.84	1.02
.99	.78	.91	.92	1.14	.45	.72	.97
				.98	.50	.44	
				1.04	.70	.72	

RUN 15 POINT 6 ALPHA=-8.0 DEG PSI=-0.0 DEG VE 60 KNOTS q = 11.5 PSF. CONFIGURATION FHB				RUN 15 POINT 7 ALPHA=-8.0 DEG PSI=-0.0 DEG VE 60 KNOTS q = 11.5 PSF. CONFIGURATION FHB			
.94	.96	.97	.95	.97	.94	.97	1.01
.92	.93	.94	.94	.94	.97	.95	1.01
.96	.99	.92	.95	.92	1.00	1.00	1.05
.94	.99	.92	1.04	.92	1.01	1.04	.95
.94	.99	1.02	1.04	1.08	.90	1.00	.87
.94	.85	.97	.98	1.02	.96	1.04	.87
1.07	.97	.84	.84	1.03	1.00	1.09	.89
1.07	1.00	.94	1.05	1.15	1.11	1.06	.89
.96	.87	1.03	1.12	1.04	1.00	.99	1.00
.99	1.07	1.07	.94	1.13	1.07	.83	1.00
.99	1.09	1.00	.79	.91	.71	1.01	1.06
1.11	1.03	.72	.49	.88	.70	.96	1.06
1.06	.81	.72	.52	.75	.80	.84	1.06
.99	.93	.54	.26	.75	1.00	.87	1.06
.92	.85	.42	.01	.4	.50	.84	1.02
.99	.78	.91	.92	1.14	.45	.72	.97
				.98	.50	.44	
				1.04	.70	.72	

RUN 15 POINT 6 ALPHA=-8.0 DEG PSI=-0.0 DEG VE 60 KNOTS q = 11.5 PSF. CONFIGURATION FHB				RUN 15 POINT 7 ALPHA=-8.0 DEG PSI=-0.0 DEG VE 60 KNOTS q = 11.5 PSF. CONFIGURATION FHB			
.94	.96	.97	.95	.97	.94	.97	1.01
.92	.93	.94	.94	.94	.97	.95	1.01
.96	.99	.92	.95	.92	1.00	1.00	1.05
.94	.99	.92	1.04	.92	1.01	1.04	.95
.94	.99	1.02	1.04	1.08	.90	1.00	.87
.94	.85	.97	.98	1.02	.96	1.04	.87
1.07	.97	.84	.84	1.03	1.00	1.09	.89
1.07	1.00	.94	1.05	1.15	1.11	1.06	.89
.96	.87	1.03	1.12	1.04	1.00	.99	1.00
.99	1.07	1.07	.94	1.13	1.07	.83	1.00
.99	1.09	1.00	.79	.91	.71	1.01	1.06
1.11	1.03	.72	.49	.88	.70	.96	1.06
1.06	.81	.72	.52	.75	.80	.84	1.06
.99	.93	.54	.26	.75	1.00	.87	1.06
.92	.85	.42	.01	.4	.50	.84	1.02
.99	.78	.91	.92	1.14	.45	.72	.97
				.98	.50	.44	
				1.04	.70	.72	

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM
TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW
MODEL NOSE, LOOKING AFT.

74.75 IN

74.75 IN

TABLE IX - Continued

RUN 15 POINT 8 V=60 KNOTS		ALPHA=4.0 DEG q=11.5 PSF. CONFIGURATION FHB		PSI= -0 DEG		RUN 15 POINT 9 V=60 KNOTS		ALPHA=4.0 DEG q=11.5 PSF. CONFIGURATION FHB		PSI= -0 DEG	
.92	.94	.47	1.01	.95	.98	1.00	.98	.45	.91	1.00	.87
.97	.94	.97	.97	.97	.94	.95	.94	1.00	.88	.60	.90
.96	.97	1.03	.97	-4.8	.99	1.00	.95	1.10	1.10	-1.4	.98
.94	1.03	1.02	.98	-1.9	.93	.80	.80	.74	.81	-4.1	.72
1.00	.85	1.02	.89	.85	.97	1.02	1.02	.50	.86	1.19	1.20
.99	.90	.97	.90	1.02	.90	1.01	1.01	.82	.96	.65	
1.02	1.00	1.07	1.03	1.08	1.03	.77	.77	.95	.75	.95	1.05
	1.04	.99	1.04	.97	1.03	.75	.75	.50	.94	.64	
.98	1.00	.98	.92	-6.7	1.09	.95	.95	.86	.57	1.02	.94
	1.00	.98	.64	-1.0	1.10	-7	-7	.72	.52	1.1	.92
.98	.99	.58	.82	1.10	1.00	.95	.95	.91	.43	1.05	1.01
	1.03	.61	1.05	.93	1.00	.95	.95	.40	.76	.64	
1.04	1.02	.46	.71	.72	.72	1.04	1.04	.74	1.09	1.06	1.01
.80	.93	.76	.49	.92	.93	.99	.99	.84	.95	1.03	
	.83	RAKE MOUNTING	.27	.95	1.01	1.01	1.01	RAKE MOUNTING	.95	1.03	
	.70	POINT	.92	1.06	.98	.98	.98	POINT	1.00	1.02	1.01
1.06	1.02	.70	.69	-5.7	1.06	1.11	1.11	.60	.41	.94	.99
	1.00	.53	.69	-5.4	1.10	1.02	1.02	.70	.99	.94	
1.05	1.00	.62	.94	1.01	.98	1.01	1.01	.51	.99	.95	.99
	1.00	1.04	1.11	.97	.97	1.02	1.02	.52	.97	1.09	
1.02	1.00	.94	1.06	1.01	.94	1.04	1.04	1.05	.96	1.00	.97

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$
 NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.
 NOTE - TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$
 NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.
 NOTE - TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

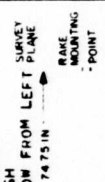


TABLE IX - Continued

RUN 15 POINT 10 ALPHA=8.0 DEG PSI=-0.0 DEG V= 60 KNOTS q= 11.5 PSF. CONFIGURATION FHB		RUN 15 POINT 11 ALPHA= 0.0 DEG PSI=-8.0 DEG V= 60 KNOTS q= 11.5 PSF. CONFIGURATION FHB									
.86	.87	.92	.63	.79	.99	1.00	.91	.97	.93	1.04	1.02
.79	.87	.75	.84	.76	.99	.99	.93	1.04	.80	1.09	1.04
.75	.76	.08	5.7	.74	.88	.88	1.05	1.08	1.10	-0	.92
.78	.74	.67	-5	.87	.87	.87	1.11	1.02	1.07	-7.2	.92
.60	.64	.53	.70	.76	.98	.98	.99	1.25	1.11	1.01	1.11
.89	.92	.65	.89	1.00	.88	.88	1.28	.99	1.05	.96	.97
1.01	.81	.65	.87	1.04	1.02	.87	.52	.92	1.01	.87	.97
.96	.94	.60	.89	1.09	.98	.98	.96	.86	.92	1.03	.94
3.8	1.6	.75	4.7	4.1	-12.1	-4.4	.96	.86	.92	.3	.91
.88	.85	.75	1.01	1.01	-10.4	-6.3	.86	1.02	.97	1.04	.91
.92	1.03	.93	-1.7	.7	.84	.84	.85	1.02	1.02	-5.1	-5.7
.95	.91	.91	.95	1.14	.79	.19	.59	.73	.96	.94	.94
.95	.82	.67	1.13	1.01	.55	.55	.44	.57	.95	.92	.99
.96	.22	.33	1.06	.90	.69	.69	.67	.86	.97	.98	.99
1.09	.59	RAKE MOUNTING POINT	.84	1.04	.82	.82	1.02	RAKE MOUNTING POINT	.96	1.05	.98
4.4	4.5	6.0	.87	5.2	-1.5	-6	.84	.82	.94	-1.1	.5
-1.8	3.7	-3.0	.91	2.3	-2.6	5.3	.84	.82	.94	-7.3	.98
.96	1.13	.91	.97	.98	.83	.76	.76	.82	.93	.98	.98
.98	1.15	.52	1.04	1.02	1.01	.78	.72	.82	.96	.92	.97
1.03	1.07	.95	1.00	.92	1.22	1.22	.91	.92	.94	.97	.97
.97	.91	.57	1.02	.92	.52	.52	.54	1.01	1.01	.99	.93

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\rho q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\rho q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE IX - Continued

MUN 15 POINT 14 ALPHA=4.0 DEG PSI=4.0 DEG V=60 KNOTS q = 11.5 PSF. CONFIGURATION FHB		MUN 15 POINT 15 ALPHA=4.0 DEG PSI=4.0 DEG V=60 KNOTS q = 11.5 PSF. CONFIGURATION FHB	
.92	.91	.97	.93
.91	.96	.92	.86
.93	.97	.93	.71
1.00	.98	.98	1.04
.99	1.05	.96	.92
.79	1.07	.98	.71
.95	1.02	.55	1.04
.99	1.15	.55	1.04
.91	.79	.90	.50
.89	.81	.60	1.11
.91	.83	.95	1.10
.99	.86	.68	.43
.90	.87	.88	1.26
.81	.91	1.07	1.02
.99	.99	.91	RAKE MOUNTING POINT
.89	.96	.74	1.00
.99	.97	1.07	1.02
1.02	.99	.91	1.00
.89	1.16	.74	1.01
	.95	.53	.96
	.97	.63	.97
	.96	.69	1.00
	.96	.96	1.00
	.96	.96	1.00

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.



NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

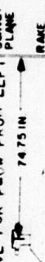


TABLE IX - Continued

RUN 15 POINT 16 ALPHA=4.0 DEG PSI=4.0 DEG V=60 KNOTS q=11.5 PSF. CONFIGURATION FHB										RUN 15 POINT 17 ALPHA=4.0 DEG PSI=4.0 DEG V=60 KNOTS q=11.5 PSF. CONFIGURATION FHB									
1.02	1.05	.94	1.00	1.01	.95	.66	.97	.98	1.04	.97	.97	.98	.98	.98	.98	1.04			
.94	1.00	1.05	.96	.97	1.12	.97	1.00	.99	1.06	1.00	1.00	.97	.97	.97	1.02	.99			
.95	.83	1.00	.96	.98	.88	1.06	.98	.98	.90	.94	.94	1.03	1.03	1.05	.95	.90			
1.00	.84	1.01	.76	1.50	.58	1.26	1.50	.58	1.06	1.17	1.00	1.04	1.04	1.06	1.06	1.02			
1.01	1.09	.98	.57	.58	.47	.83	.58	.47	.97	1.12	1.07	1.07	1.07	1.03	1.18	.96			
1.01	1.03	1.01	.59	.86	1.00	.63	.86	1.00	.97	.83	.90	.92	.92	1.02	.54	.91			
.98	1.02	1.01	.97	.87	.99	.96	.87	.99	.98	.81	.81	.98	.98	.85	.87	.66			
1.05	.96	.18	1.18	.41	.92	1.03	.41	.92	.96	1.04	.98	1.07	1.07	.02	1.03	.86			
1.13	.88	.70	RAKE MOUNTING POINT	.58	1.17	1.06	RAKE MOUNTING POINT	.24	.97	1.04	1.04	.97	.97	.93	.79	.85			
1.10	1.00	.63	RAKE MOUNTING POINT	.48	.90	1.06	RAKE MOUNTING POINT	.91	.85	.85	.93	1.01	1.01	.00	.67	.82			
.99	1.05	.93	.73	.73	.46	1.00	.73	.46	.93	1.10	-5.3	1.00	.24	RAKE MOUNTING POINT	.43	.60			
	1.01	1.02	.60	.60	.55	1.06	.60	.55	.98	1.02	1.02	.94	.94	.38	1.16	.92			
	.97	1.01	.56	.56	.46	1.00	.56	.46	.98	1.01	1.01	1.01	1.01	.30	.36	.77			
	.94	.94	.66	.66	.48	1.00	.66	.48	.95	1.00	1.00	1.01	1.01	.32	1.13	1.02			

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_1 - P_2}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

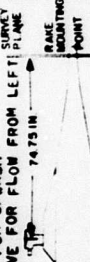


TABLE IX - Continued

RUN 15 POINT 18 ALPHA=8.0 DEG PSI= 4.0 DEG V= 60 KNOTS q= 11.5 PSF. CONFIGURATION FHB		RUN 15 POINT 19 ALPHA=8.0 DEG PSI= 4.0 DEG V= 60 KNOTS q= 11.5 PSF. CONFIGURATION FHB					
1.01	.97	.94	1.00	.94	1.04	.97	.98
	.94	.96	.93	.98	.97	.95	
1.00	1.00	.98	.97	-6.9	.96	-8.7	1.01
	.99	.95	.98	3.4	1.00	2.2	1.00
1.01	.99	.99	1.00	1.04	1.02	1.01	1.02
	.95	.99	.97	.97	.99	.99	
	.99	1.00	1.02	.85	1.04	.89	
.94	.99	1.00	1.04	1.00	.88	.94	.92
	1.03	1.12	1.27	1.10	1.02	1.05	
1.15	-12.1	-11.1	1.06	-10.6	1.04	-13.1	-12.8
	1.21	1.05	1.06	2.2	1.09	2.6	.75
	.94	.93	1.09	1.22	1.07	.92	2.2
.90	.90	.98	.99	1.02	1.14	1.22	.83
	.96	.98	.90	.80	1.14	.49	
.98	1.03	.87	.95	.51	1.00	.66	.78
	.98	.94	.81	.92	1.00	.47	
1.16	-7.6	-9.2	RAKE MOUNTING POINT	-9.5	RAKE MOUNTING POINT	-18.8	-10.3
	1.19	1.02	.65	3.9	1.06	6.0	.92
	1.00	1.02	.66	4.2	2.2	.58	1.3
1.07	1.03	.91	.19	.85	1.04	.68	1.02
	1.01	.94	.47	.69	.98	.73	
.98	.89	.88	.35	1.00	.97	.88	.53
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$				NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$			
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.				NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.			
NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.				NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.			

TABLE IX - Continued

RUN 15 POINT 20 ALPHA= .0 DEG PSI= 8.0 DEG V= 60. KNOTS q= 11.5 PSF. CONFIGURATION FHB		RUN 16 POINT 5 ALPHA=-4.0 DEG PSI= -.0 DEG V= 100. KNOTS q= 31.7 PSF. CONFIGURATION FHB	
.93	1.02	.99	.98
.93	1.08	.98	1.02
1.01	1.02	.97	1.00
.99	.58	1.00	.97
1.11	1.09	1.00	1.03
.92	1.08	1.02	.91
1.02	1.01	.94	1.03
.93	1.05	1.02	.98
1.01	1.08	1.05	1.05
.98	.98	.94	.90
1.00	.95	1.02	1.04
.99	.91	.88	1.00
.97	1.01	.88	.88
.93	.47	.95	1.00
1.06	.89	.86	.99
1.08	1.00	1.03	1.00
1.04	1.02	1.04	.98
	.95	1.01	.98
	.83	.99	1.00
	1.23	.95	.98
		.94	.98

RUN 15 POINT 20 ALPHA= .0 DEG PSI= 8.0 DEG V= 60. KNOTS q= 11.5 PSF. CONFIGURATION FHB		RUN 16 POINT 5 ALPHA=-4.0 DEG PSI= -.0 DEG V= 100. KNOTS q= 31.7 PSF. CONFIGURATION FHB	
1.02	1.01	.99	.98
1.08	.93	.99	1.01
1.00	1.04	.97	.99
.58	1.09	1.00	.98
1.09	1.04	1.02	1.04
1.08	1.05	.94	.89
1.01	1.05	1.02	1.01
1.05	.50	1.05	1.02
1.08	1.00	1.04	.70
.98	.64	.89	.88
.95	.69	.88	.88
.87	.59	.98	.74
.47	1.07	.95	.61
.89	.88	.86	.47
1.00	1.05	1.00	.84
1.02	.59	.99	.81
.99	1.07	.98	.98
1.01	.89	.99	1.03
1.01	.86	.95	.94
1.01	.86	.94	.98
1.01	.86	.97	.98

RUN 15 POINT 20 ALPHA= .0 DEG PSI= 8.0 DEG V= 60. KNOTS q= 11.5 PSF. CONFIGURATION FHB		RUN 16 POINT 5 ALPHA=-4.0 DEG PSI= -.0 DEG V= 100. KNOTS q= 31.7 PSF. CONFIGURATION FHB	
1.02	1.01	.99	.98
1.08	.93	.99	1.01
1.00	1.04	.97	.99
.58	1.09	1.00	.98
1.09	1.04	1.02	1.04
1.08	1.05	.94	.89
1.01	1.05	1.02	1.01
1.05	.50	1.05	1.02
1.08	1.00	1.04	.70
.98	.64	.89	.88
.95	.69	.88	.88
.87	.59	.98	.74
.47	1.07	.95	.61
.89	.88	.86	.47
1.00	1.05	1.00	.84
1.02	.59	.99	.81
.99	1.07	.98	.98
1.01	.89	.99	1.03
1.01	.86	.95	.94
1.01	.86	.94	.98
1.01	.86	.97	.98

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE IX - Continued

RUN ID POINT 6 ALPHA= 0 DEG PSI= -0 DEG V= 100 KNOTS $\phi = 31.7$ PSF. CONFIGURATION FMB		RUN ID POINT 7 ALPHA= 0 DEG PSI= -0 DEG V= 100 KNOTS $\phi = 31.7$ PSF. CONFIGURATION FMB					
.99	1.02	1.04	.98	1.03	.96	.99	.99
.96	1.00	1.01	1.07	1.13	.89	1.06	.92
	.94	.92	1.08	.77	.87	.99	
1.05	1.02	1.02	.84	1.02	.81	1.01	1.14
.87	1.00	1.05	1.01	.89	.60	.97	.94
	.84	.88	.91	1.21	.66	.79	
	.98	1.05	.96	.97	.53	1.19	
1.00	1.00	.81	.87	.99	.87	.96	1.03
	.98	.78	.84	1.24	.67	.78	1.03
	.87	.67	.24	.94	.47	.81	1.6
.97	.86	1.15	.59	.68	.70	.81	.98
.99	1.02	.94	.94	.55	.67	.65	1.05
	.90	.87	.41	.94	.49	.78	1.02
1.04	1.01	.41	1.11	.89	RAKE MOUNTING POINT	1.01	1.00
	.95	.27	.32	.97	2.0	1.8	3.5
	.89	1.05	.66	.86	3.0	-3.1	1.8
1.01	1.06	.90	1.01	1.05	.63	.65	.99
	.93	.94	.58	1.01	.96	.75	.99
1.00	1.00	1.01	.48	1.04	.94	.86	1.00

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT 74.75 IN

TABLE IX - Continued

RUN 10 POINT 8 ALPHA=8.0 DEG PSI=-0.0 DEG V=100 KNOTS q=31.7 PSF. CONFIGURATION FHB		RUN 10 POINT 9 ALPHA=-4.0 DEG PSI=-4.0 DEG V=100 KNOTS q=31.7 PSF. CONFIGURATION FHB	
.92	.83	.84	.95
.90	.86	.91	1.03
.98	.75	5.2	.83
.76	.60	-3.2	.98
.82	.67	.72	.87
1.02	.76	.65	.97
.90	1.13	.88	1.19
.92	.87	.89	.77
4.2	1.1	3.4	5.1
1.08	.57	.78	1.05
-0.8	1.10	.95	.72
1.07	1.10	.94	.99
1.06	1.04	.87	.98
1.00	1.13	.76	1.06
1.00	.86	.90	.73
1.05	.33	.89	.99
5.6	3.0	7.4	8.0
.97	.83	.93	1.05
4.7	.86	-2.0	-1.3
1.02	.94	.58	1.01
1.04	.94	.77	1.10
.97	1.01	.91	.99
1.01	.97	.99	.98

RUN 10 POINT 8 ALPHA=8.0 DEG PSI=-0.0 DEG V=100 KNOTS q=31.7 PSF. CONFIGURATION FHB		RUN 10 POINT 9 ALPHA=-4.0 DEG PSI=-4.0 DEG V=100 KNOTS q=31.7 PSF. CONFIGURATION FHB	
1.00	.99	.97	.96
.98	1.00	.98	1.01
.95	.95	.99	.95
1.01	1.01	.96	1.04
1.00	1.00	.88	.97
.89	.89	.99	.94
.98	.98	.92	.95
7.7	10.1	1.11	1.04
3.3	5.9	.98	.98
.85	.47	.95	1.01
.54	.90	.18	1.11
.88	.86	.71	1.18
.54	.31	.34	.47
-9.8	-12.4	RAKE MOUNTING POINT	.95
-2.7	-1.8	RAKE MOUNTING POINT	.71
1.04	.46	RAKE MOUNTING POINT	.99
1.03	1.06	.43	.96
1.08	1.08	.27	.99
1.01	.93	.96	.96

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_s}{\rho q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT → 74.75 IN

TABLE IX - Continued

RUN 10 POINT 11 ALPHA = -0 DEG PSI = -4.0 DEG V = 100 KNOTS q = 31.7 PSF. CONFIGURATION FMB												
.99	.99	.93	.96	.96	1.05	.97	1.00	.98	.87	1.07	.98	.96
.94	.99	1.05	1.13	1.07	1.07	1.00	.95	.99	1.06	.82	1.13	.99
	.92	.93	1.09	1.07	.93		1.02	.91	.05	1.09	-.2	
1.03	1.05	1.01	.74	.82	.96	1.03	1.05	.99	.96	.84	-4.8	1.10
	.97	1.15	1.06	.74	.98		.85	.86	.95	1.03	.90	.93
1.09	.93	.84	1.03	1.03	1.32	1.06	1.02	1.01	.96	.87	1.24	
	.95	1.01	.75	.84	1.00		.99	.98	.95	1.27	.80	1.21
	.87	.52	1.07	1.07	1.01		1.03	.78	.87	.81	1.12	
.94	.87	.45	.77	.11	.1	.90	.42	.89	1.05	.98	1.4	-.7
	.71	.71	.90	-3.7	-3.2		.65	-6.1	.84	.99	-3.0	.95
.97	.89	.43	.95	.96	1.00	1.04	.85	.71	.52	.33	.99	1.03
	.70	.49	.89	.97			.55	.87	.78	.96	.90	
.98	.64	.43	.27	1.02	1.03	.96	.87	.88	.04	.80	1.06	.97
	.86	.53	.40	1.07			1.04	.88	RAKE MOUNTING POINT	.79	1.05	
1.00	1.02	.71	1.06	-2.0	.3	1.03	.90	1.04	RAKE MOUNTING POINT	.98	.92	1.00
	.91	.78	1.00	-5.6	-1.4		.89	.98		.91	1.00	
1.01	.31	.40	1.03	1.00	.98	1.03	.97	.89	1.09	1.24	.99	1.00
	.80	.65	1.07	1.01			.89	.71	.92	.97	.98	
.98	.68	.72	.99	.98	.96	.99	.82	.83	.25	1.02	.99	.96

RUN 10 POINT 10 ALPHA = 0 DEG PSI = -4.0 DEG V = 100 KNOTS q = 31.7 PSF. CONFIGURATION FMB												
.99	.99	.93	.96	.96	1.05	.97	1.00	.98	.87	1.07	.98	.96
.94	.99	1.05	1.13	1.07	1.07	1.00	.95	.99	1.06	.82	1.13	.99
	.92	.93	1.09	1.07	.93		1.02	.91	.05	1.09	-.2	
1.03	1.05	1.01	.74	.82	.96	1.03	1.05	.99	.96	.84	-4.8	1.10
	.97	1.15	1.06	.74	.98		.85	.86	.95	1.03	.90	.93
1.09	.93	.84	1.03	1.03	1.32	1.06	1.02	1.01	.96	.87	1.24	
	.95	1.01	.75	.84	1.00		.99	.98	.95	1.27	.80	1.21
	.87	.52	1.07	1.07	1.01		1.03	.78	.87	.81	1.12	
.94	.87	.45	.77	.11	.1	.90	.42	.89	1.05	.98	1.4	-.7
	.71	.71	.90	-3.7	-3.2		.65	-6.1	.84	.99	-3.0	.95
.97	.89	.43	.95	.96	1.00	1.04	.85	.71	.52	.33	.99	1.03
	.70	.49	.89	.97			.55	.87	.78	.96	.90	
.98	.64	.43	.27	1.02	1.03	.96	.87	.88	.04	.80	1.06	.97
	.86	.53	.40	1.07			1.04	.88	RAKE MOUNTING POINT	.79	1.05	
1.00	1.02	.71	1.06	-2.0	.3	1.03	.90	1.04	RAKE MOUNTING POINT	.98	.92	1.00
	.91	.78	1.00	-5.6	-1.4		.89	.98		.91	1.00	
1.01	.31	.40	1.03	1.00	.98	1.03	.97	.89	1.09	1.24	.99	1.00
	.80	.65	1.07	1.01			.89	.71	.92	.97	.98	
.98	.68	.72	.99	.98	.96	.99	.82	.83	.25	1.02	.99	.96

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

7475N

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

7475N

TABLE IX - Continued

RUN 16 POINT 13 ALPHA= 4.0 DEG PSI=-4.0 DEG V= 100. KNOTS q= 31.7 PSF. CONFIGURATION FMB											
.95	.98	1.00	1.08	.76	.93	1.02	1.02	1.02	1.02	1.02	1.02
.99	1.00	.86	1.01	1.02	1.06	.94	5.1	.94	1.13	1.08	1.08
.89	1.04	.79	.84	1.28	.97	.89	-5.2	.75	.89	.93	.93
.95	.86	.79	.63	.89	1.13	.99	1.08	.75	1.08	.99	.99
.95	.79	.79	.66	1.18	.93	.94	.94	1.18	.94	.94	.99
.95	.83	.52	.52	.72	.93	1.05	6.8	.93	.90	3.8	1.00
.95	.61	.57	.57	.49	.95	.95	-5.7	.95	.99	-2.9	.95
.95	.56	.78	.78	.83	.34	.99	.99	.34	.99	.99	.95
.88	.84	.35	.35	.88	.69	1.04	1.04	.69	1.04	1.01	1.01
.88	.93	.47	.47	1.12	1.20	1.00	1.00	1.20	1.00	1.00	1.01
.88	1.06	.50	.50	.09	.64	.94	.94	.64	.94	1.01	1.01
1.04	.70	.75	.75	RAKE MOUNTING POINT	.77	1.00	4.3	.77	1.00	4.3	.97
1.00	.91	.51	.51	.98	.99	.99	-9.2	.99	.99	-1.0	.98
1.00	.59	.42	.42	1.06	.97	1.03	1.03	.97	1.03	1.03	.98
.97	.59	.35	.35	.98	.98	1.04	1.04	.98	1.04	.98	.98
.97	.90	.74	.74	.82	.98	.98	.98	.98	.98	.98	.94
.97	.82	.85	.85	.82	.95	1.02	1.02	.95	1.02	1.02	.94

RUN 16 POINT 12 ALPHA= .0 DEG PSI=-4.0 DEG V= 100. KNOTS q= 31.7 PSF. CONFIGURATION FMB											
.98	.97	.96	.99	.98	1.00	.98	.95	.98	.98	.98	.95
1.00	.97	.99	.99	.93	.95	.93	1.2	.93	1.01	.99	.99
.97	.96	.97	.98	1.01	.89	.95	-5.1	1.01	.95	.85	.85
.86	.93	1.02	1.03	.97	1.03	.97	.97	.88	.88	.93	.93
.86	1.02	.91	.80	.88	1.08	.88	1.14	.88	.88	.93	.93
.86	.89	.65	.66	1.14	.85	1.14	.90	1.14	.90	.93	.93
.96	.73	1.06	1.14	.90	.92	.90	.7	.90	.90	.96	.96
.96	.69	.53	.53	.94	.99	.94	-5.8	.94	.94	-2.9	.96
.96	.85	.27	.27	1.02	.34	1.02	.98	1.02	1.02	.96	1.00
.93	.79	.87	.86	.88	1.03	.88	.88	.88	.88	.95	.95
.93	.55	.50	.80	.95	.96	.95	.95	.95	.95	.95	.95
.93	.68	.71	.83	1.03	.10	1.03	.95	1.03	.95	.95	.95
1.02	.48	.80	RAKE MOUNTING POINT	.95	1.09	.95	.95	.95	.95	.95	.94
1.02	1.23	1.06	1.06	.98	.85	.98	-1.0	.98	.98	.98	.98
1.02	1.02	.77	.77	1.01	.99	1.01	-3.7	1.01	1.01	1.00	1.00
1.00	.99	.75	.75	.96	.99	.96	.96	.96	.96	1.00	1.00
1.00	1.01	.49	.49	.98	.95	.98	.98	.98	.98	.98	.94
1.00	.75	.68	.68	.98	.99	.98	.98	.98	.98	.98	.94

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE IX - Continued

RUN 10 POINT 14 ALPHA=4.0 DEG PSI=4.0 DEG V=100. KNOTS q=31.7 PSF. CONFIGURATION FMB		RUN 16 POINT 15 ALPHA=4.0 DEG PSI=4.0 DEG V=100. KNOTS q=31.7 PSF. CONFIGURATION FMB	
.97	.96	.93	.98
.95	1.01	.94	.94
.90	.93	.98	.92
.68	.82	.88	.97
.78	.84	.75	.90
.77	.79	.63	.53
.32	.51	.69	.65
.93	.51	.69	.67
1.07	.88	1.07	.83
1.14	1.14	.80	.87
.99	.97	1.00	.92
.86	.31	.91	.84
1.23	.78	1.02	.87
1.02	.73	1.03	.95
1.02	.62	1.02	.92
.61	.72	1.02	.96
.72	.98	.82	.98
.70	.90	.83	.96
		.83	.97
		.83	.97

RUN 10 POINT 14 ALPHA=4.0 DEG PSI=4.0 DEG V=100. KNOTS q=31.7 PSF. CONFIGURATION FMB		RUN 16 POINT 15 ALPHA=4.0 DEG PSI=4.0 DEG V=100. KNOTS q=31.7 PSF. CONFIGURATION FMB	
1.00	.95	.93	.84
.90	.94	1.07	.95
.44	.98	.98	.99
.82	.87	.88	.80
.89	1.11	.75	1.28
.79	.84	.63	.36
.45	.98	.69	.95
.55	1.07	.69	1.04
.46	.84	.83	.84
.74	.23	.77	1.02
.82	.90	.77	1.14
.35	.38	.94	.39
.75	.95	.91	.92
.78	.50	1.02	.87
.67	1.11	.96	.97
.61	.98	.83	1.02
.72	.94	.75	1.00
.90	1.00	.83	.96
		.83	.96
		.83	.96

RUN 10 POINT 14 ALPHA=4.0 DEG PSI=4.0 DEG V=100. KNOTS q=31.7 PSF. CONFIGURATION FMB		RUN 16 POINT 15 ALPHA=4.0 DEG PSI=4.0 DEG V=100. KNOTS q=31.7 PSF. CONFIGURATION FMB	
.97	.93	.97	.98
.92	.92	.98	.94
.96	.95	.96	.92
.92	.97	.99	.78
.97	.97	.90	1.06
.95	.95	.53	.82
1.01	1.01	.67	.02
1.03	1.03	.57	.09
.48	.48	.82	.50
.99	.99	.42	.06
.96	.96	.92	.42
1.02	1.02	.84	.74
.99	.99	.95	.87
.96	.96	.95	.96
.99	.99	.96	.92
.96	.96	.96	.92
.96	.96	.96	.92

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.



TABLE IX - Continued

RUN 16 POINT 16 ALPHA=4.0 DEG PSI=4.0 DEG V=100 KNOTS q=31.7 PSF. CONFIGURATION FHB										RUN 16 POINT 17 ALPHA=0 DEG PSI=4.0 DEG V=100 KNOTS q=31.7 PSF. CONFIGURATION FHB										
.98	.98	.98	.97	.94	.96	.98	.98	.97	.94	.96	.98	.98	.97	.94	.96	.98	.98	.97	.94	.96
1.01	1.02	1.00	.95	1.01	.98	1.02	1.00	.95	1.01	.98	1.01	1.02	1.00	.95	.98	1.01	1.02	1.00	.95	.98
.95	.98	.97	.97	.95	.90	.98	.97	.95	.95	.90	.68	.88	.87	.87	.76	.95	1.03	.76	.89	.89
1.03	1.00	.98	.99	.99	.90	1.05	.98	.99	.99	.90	1.09	1.09	1.07	.84	.98	.97	.84	.98	.98	.98
.99	.99	1.10	1.01	.91	1.03	1.01	1.01	.91	.91	1.03	1.09	1.09	1.24	.96	.76	1.02	.96	.76	1.03	1.03
.96	.97	.94	.94	1.01	.96	.94	.94	1.01	1.01	.96	.93	.93	.92	.86	.93	.50	.50	.93	.93	.93
1.05	.91	1.00	.75	.95	1.00	.91	.95	.95	.95	1.00	.96	.96	1.01	.07	.82	.89	.89	.82	.82	.82
1.04	1.07	1.17	.66	.42	.96	1.07	1.17	.66	.42	.96	1.04	1.04	.91	.08	.82	.59	.59	.82	.82	.82
1.01	.95	1.03	.75	.97	.96	.95	.97	.75	.97	.96	.91	.91	.91	1.02	.74	.54	.54	.74	.74	.74
1.01	.99	1.08	.72	.74	.95	.99	.74	.72	.74	.95	.96	.87	.87	1.17	.82	.33	.33	.82	.82	.82
1.01	1.04	.15	.29	.38	.95	1.04	.15	.29	.38	.95	.96	.96	.93	.50	.50	.50	.50	.50	.50	.50
1.05	.96	.96	.51	.50	.96	.96	.96	.51	.50	.96	1.03	1.03	.74	.49	.59	.49	.49	.59	.59	.59
1.05	.89	1.10	.42	.99	.96	.89	1.10	.42	.99	.96	1.02	1.02	.94	.08	.89	.08	.08	.89	.89	.89
1.04	1.02	.47	.68	.67	.96	1.02	.47	.68	.67	.96	1.01	1.01	.71	1.03	.88	1.03	1.03	.88	.88	.88
1.05	1.05	.92	.45	.08	1.00	1.05	.92	.45	.08	1.00	.98	.98	1.01	.46	1.04	1.29	1.29	1.04	.94	.94
1.01	1.01	.31	.85	.95	.95	1.01	.31	.85	.95	.95	.95	.95	.97	.77	1.02	.30	.30	1.02	.94	.94
1.00	1.00	.97	.51	.88	.95	1.00	.97	.51	.88	.95	.98	.98	1.02	1.23	.70	.32	.32	.70	.94	.94

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
TO RIGHT LOOKING AFT.



NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE IX - Continued

RUN 16 POINT 16 ALPHA= .0 DEG PSI= 4.0 DEG V= 100. KNOTS ϕ = 31.7 PSF. CONFIGURATION PHB					RUN 16 POINT 19 ALPHA= .0 DEG PSI= 4.0 DEG V= 100. KNOTS ϕ = 31.7 PSF. CONFIGURATION PHB							
.98	.98	1.02	.47	.95	.91	.91	.97	1.00	.96	.99	.98	.95
.94	.97	.93	.44	.97	.92	.92	.99	.91	.48	.93	.95	.95
1.00	1.00	1.00	1.00	1.02	1.00	1.00	1.00	1.00	1.00	1.00	1.00	1.00
1.06	1.06	1.03	1.01	1.02	.92	.92	1.02	.97	.94	.90	1.00	.96
1.00	1.00	1.02	.90	.78	.77	.98	1.01	.92	.98	1.04	.90	.96
.92	.92	.80	.46	1.02	1.11	1.18	1.02	1.01	.42	.93	.87	1.01
.97	.97	1.01	1.05	1.03	.76	1.18	.96	.91	.68	.72	.90	1.01
.89	.89	.73	.83	.57	.85	.85	1.04	.91	.98	.87	1.00	1.01
1.06	1.02	3.2	.93	.80	.79	.93	.94	.93	.69	.71	.86	.93
.96	.98	.98	.79	.64	.77	.93	1.02	.97	.59	.83	.90	.96
.94	.94	.87	.47	.68	.36	.90	1.00	.93	1.01	.59	1.01	.96
.90	.90	.93	.48	.53	.73	.95	1.01	.94	.07	.81	.87	.94
1.01	.98	.94	.76	.72	.56	.95	.98	.91	.41	.79	.96	.94
1.02	1.02	.90	RAKE MOUNTING POINT	.43	.91	.91	1.07	1.00	RAKE MOUNTING POINT	.62	.96	1.00
1.03	.93	.97	.97	.79	.92	.92	1.01	.91	.90	.90	.05	1.00
1.00	1.00	.91	1.17	.64	1.17	.94	.96	.94	.01	.01	.94	.93
1.03	.97	1.01	.71	.80	.71	.98	.97	1.02	.06	.08	.04	.93
1.00	1.00	1.01	.42	.68	.92	.92	.98	.91	.09	.93	.90	.93
1.01	1.01	1.00	.86	.58	.86	.94	.97	1.00	.95	.11	.33	.93

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$
 NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
 TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW
 MODEL NOSE, LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW
 MODEL NOSE, LOOKING AFT.

TABLE IX - Continued

RUN 10 POINT 20 ALPHA= 4.0 DEG PSI= 4.0 DEG VE= 100 KNOTS Q= 31.7 PSF. CONFIGURATION FMB		RUN 10 POINT 21 ALPHA= 4.0 DEG PSI= 4.0 DEG VE= 100 KNOTS Q= 31.7 PSF. CONFIGURATION FMB	
1.00	1.02	1.07	1.07
	1.00	1.02	1.04
.97	.95	1.05	1.04
	1.01	1.08	1.09
.94	.93	1.05	1.00
	.99	1.03	.91
	.99	.87	.85
.98	.99	1.04	.80
	1.01	.76	.86
	1.00	1.11	.83
.97	.96	.46	.81
	.99	.74	.81
.98	1.04	.76	.80
	.85	.46	.71
	1.07	1.03	1.22
1.04	1.00	1.04	.53
	.98	.93	.79
1.01	1.01	1.13	.78
.99	1.01	.64	.88
		.74	1.11
		.95	4.5

RUN 10 POINT 20 ALPHA= 4.0 DEG PSI= 4.0 DEG VE= 100 KNOTS Q= 31.7 PSF. CONFIGURATION FMB		RUN 10 POINT 21 ALPHA= 4.0 DEG PSI= 4.0 DEG VE= 100 KNOTS Q= 31.7 PSF. CONFIGURATION FMB	
1.00	1.02	1.07	1.07
	1.00	1.02	1.04
.97	.95	1.05	1.04
	1.01	1.08	1.09
.94	.93	1.05	1.00
	.99	1.03	.91
	.99	.87	.85
.98	.99	1.04	.80
	1.01	.76	.86
	1.00	1.11	.83
.97	.96	.46	.81
	.99	.74	.81
.98	1.04	.76	.80
	.85	.46	.71
	1.07	1.03	1.22
1.04	1.00	1.04	.53
	.98	.93	.79
1.01	1.01	1.13	.78
.99	1.01	.64	.88
		.74	1.11
		.95	4.5

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

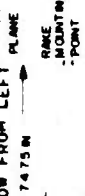


TABLE IX - Continued

RUN 10 POINT 22		ALPHA = 4.0 DEG		PSI = 4.0 DEG		FHB	
V = 100 KNOTS		q = 31.7 PSF.		q = 48.3 PSF.		CONFIGURATION FHB	
.97	1.00	1.01	.95	.91	.95	.98	
.94	.98	.95	.94	.95	.90	.97	1.02
.91	.95	.91	.95	.93	.88	.95	.98
	.91	.77	.78	1.04	.77	.96	1.00
	.80	.88	.56	.98	.93	.92	.99
	1.00	1.04	.72	.65	.57	.98	.96
	1.04	.97	.72	.80	.46	.98	.94
	1.05	1.07	.00	.72	.92	.88	1.00
.97	3.1	1.5	.99	.38	.88	.65	.86
	2.9	1.00	.44	.92	3.7	.92	.36
	1.08	.70	.97	.80	.75	.92	.91
	1.04	.47	.78	1.09	1.08	.75	.87
.97	.97	.93	1.03	.67	.93	1.07	.81
.97	.96	.48	.78	.78	.19	.96	.44
	.99	.85	.56	.56	.78	.88	.53
1.05	4.0	1.05	.65	.65	.71	.90	.26
	.7	-7.0	.40	.40	.55	-1.6	.91
1.05	.98	.87	.50	.50	.77	-2.6	.64
1.00	1.00	.91	.52	.69	.65	1.02	.93
.96	.96	.97	.89	.89	.82	.98	.93

RUN #1 POINT 5		ALPHA = 0 DEG		PSI = -0 DEG		FHB	
V = 125 KNOTS		q = 48.3 PSF.		q = 48.3 PSF.		CONFIGURATION FHB	
.96	.96	.95	1.03	1.02	.98	.96	
.97	.97	.99	1.01	.93	.93	.97	1.02
.95	.95	.99	1.01	1.06	.97	.95	.98
.96	.96	.97	.93	1.10	.99	.96	1.00
.92	.92	.97	.79	.98	.96	.92	.97
.98	.98	.96	.86	.90	.94	.98	.96
.88	.88	.86	.82	.86	1.00	.88	.94
.65	.65	.83	.36	.91	.97	.65	1.02
.92	.92	.87	.87	.94	.95	.92	.86
.75	.75	.81	.81	.44	1.03	.75	.36
1.07	1.07	.80	1.00	.53	.93	1.07	.91
.96	.96	.52	.91	.26	1.02	.96	.97
.88	.88	.89	.52	.64	.93	.88	.95
.90	.90	1.02	.91	.91	1.01	.90	.95
-1.6	-1.6	1.02	RAKE MOUNTING POINT	.91	.91	-1.6	.4
-2.0	-2.0	.76	RAKE MOUNTING POINT	1.04	.96	-2.0	.6
.98	.98	.98	.98	.83	1.05	.98	1.02
1.00	1.00	1.16	.41	1.05	1.08	1.00	1.04
.99	.99	1.19	.62	.84	1.06	.99	1.06
.97	.97	.99	.56	.94	1.04	.97	1.01

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\rho q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT. → 74.75 IN. →

TABLE IX - Continued

RUN #1 POINT 6 V= 125 KNOTS		ALPHA= .0 DEG PSI= -.0 DEG q = 48.3 PSF. CONFIGURATION FHB		RUN #1 POINT 7 V= 125 KNOTS		ALPHA= .0 DEG PSI= -.0 DEG q = 48.3 PSF. CONFIGURATION FHB	
1.01	.98	1.01	1.00	.97	1.02	1.03	.96
.97	.98	1.03	.89	.97	1.02	.97	.86
.95	.99	.95	1.01	.97	.94	.98	.94
.99	.96	.95	1.15	.96	.94	1.05	.92
.99	.94	1.00	.99	1.00	.88	.94	.90
.99	.94	.87	.95	.98	.80	.92	.86
.99	.85	.83	.83	.70	.73	.94	.89
1.08	.85	1.09	.86	1.00	.71	1.05	.68
1.04	.96	.69	.62	.84	.82	.47	.88
.98	.99	.29	.77	.89	.67	.96	.23
.98	.85	.79	.85	.76	1.06	.60	1.00
1.02	.90	.94	1.05	.75	.69	.78	1.00
.96	.74	.76	.57	1.06	.50	.67	.90
1.00	.89	RAKE MOUNTING POINT	1.02	1.02	RAKE MOUNTING POINT	.76	.56
.96	.95	.85	.91	.90	1.04	.94	.84
.97	.94	.91	1.00	.91	.83	.92	1.00
.96	.60	.31	.92	.98	1.01	.91	1.02
.95	.96	.65	.93	.95	.76	1.05	.92
.96	1.06	.21	1.00	.96	.90	.99	.94

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p \cdot \frac{V^2}{2q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW-
MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT → 74.75 IN

TABLE IX - Continued

RUN #1 POINT 8 ALPHA= 4.0 DEG PSI= -0 DEG V=125 KNOTS $\phi = 48.3$ PSF. CONFIGURATION FHB		RUN #1 POINT 9 ALPHA= 4.0 DEG PSI= -0 DEG V=125 KNOTS $\phi = 48.3$ PSF. CONFIGURATION FHB	
.98	.99	1.02	1.03
.96	.99	.98	.98
.91	.98	.99	.94
	.92	.99	.96
	.87	1.12	1.01
.97	1.03	.81	.92
	.92	.87	.88
.97	.96	.78	.56
	1.01	1.04	.96
1.00	.40	.71	1.08
	.96	.73	.84
1.01	1.02	.78	.96
	1.03	.77	.91
.97	1.13	.78	1.03
	1.04	.51	1.06
1.00	.96	.64	.89
	.96	.67	.95
.98	1.00	.83	.98
	1.02	1.02	.95
.97	1.04	.95	.96

RUN #1 POINT 8 ALPHA= 4.0 DEG PSI= -0 DEG V=125 KNOTS $\phi = 48.3$ PSF. CONFIGURATION FHB		RUN #1 POINT 9 ALPHA= 4.0 DEG PSI= -0 DEG V=125 KNOTS $\phi = 48.3$ PSF. CONFIGURATION FHB	
.98	.96	.98	.96
.92	.98	.96	.93
.96	.88	.93	.81
.96	.83	.72	.96
.97	.93	1.09	.91
	1.01	.83	.82
	.97	.60	1.04
1.01	.71	1.04	.92
	.94	.66	.37
.99	.96	1.08	.84
	.88	1.02	.93
.98	1.04	.97	.82
	1.00	.96	.95
1.01	.88	.98	1.04
	1.00	.89	1.02
1.00	.98	.80	.95
.98	.99	.97	.98
	.98	1.01	1.00

RUN #1 POINT 8 ALPHA= 4.0 DEG PSI= -0 DEG V=125 KNOTS $\phi = 48.3$ PSF. CONFIGURATION FHB		RUN #1 POINT 9 ALPHA= 4.0 DEG PSI= -0 DEG V=125 KNOTS $\phi = 48.3$ PSF. CONFIGURATION FHB	
.85	.98	.85	.96
	.94	.96	.93
	1.01	.92	1.06
	.88	.75	.88
	.74	.56	.99
	.64	.96	1.01
	.98	1.08	.84
	1.01	.96	.96
	1.06	.27	.35
	.55	.91	.98
	.33	1.03	.96
	.85	1.06	.96
	1.21	.89	.96
	.96	.95	.96
	.67	.39	.98
	.83	.35	.95
	1.02	.83	.96
	.95	.45	.96

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT
TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW
MOUFL NOSE, LOOKING AFT.

7475 N

7475 N

TABLE IX - Continued

RUN 41 POINT 10 V=125 KNOTS		ALPHA=4.0 DEG q=48.3 PSF. CONFIGURATION FHB		PSI=-4.0 DEG		RUN 41 POINT 11 V=125 KNOTS		ALPHA=.0 DEG q=48.3 PSF. CONFIGURATION FHB		PSI=-2.0 DEG	
.96	.95	.96	1.01	1.02	1.00	.93	.98	1.01	.99	.96	1.00
.99	.98	1.00	.89	1.00	.98	.96	1.00	.99	.98	.90	.98
.94	.90	.94	.99	.95	5.9	.96	.96	1.00	1.02	.7	.98
.97	.93	1.00	1.00	1.04	-4.6	1.01	1.00	.96	1.12	-3.0	.97
.97	.79	1.09	.80	.80	.98	1.01	1.00	1.04	1.00	.97	.95
.97	.78	.51	.88	.82	1.02	1.00	.92	.91	.94	.94	.94
.97	.60	.60	.91	1.06	.86	1.00	1.00	.72	.92	1.01	.99
.98	.49	.49	.47	1.15	.98	1.02	.86	.65	.65	1.00	.98
.98	.65	.79	.57	.68	7.0	1.02	.79	.76	1.01	1.5	1.0
.98	.52	.7	.79	.81	-8.1	.87	.67	.35	.59	-3.3	-1.3
.98	.92	.76	.92	.86	.91	1.00	.86	.69	.56	.70	.96
.99	.87	.66	.96	.39	1.01	1.00	.97	.92	.17	.94	.96
.99	.81	.85	.71	1.26	1.05	1.00	.98	.38	.88	.91	.98
1.01	.87	.15	.99	RAKE	1.01	1.00	.96	RAKE	.31	.99	.96
1.01	.64	.76	.72	MOUNTING	2.5	.98	1.04	POINT	1.00	-6.0	.88
.98	.52	3.6	.88	POINT	-4.1	.97	1.05	POINT	.58	-3.6	-1.2
1.00	.80	.86	.98	.91	.98	.93	.97	.70	.99	.97	.98
.98	.64	.47	1.00	.48	.92	.98	.97	1.05	.96	.96	.98
.98	.93	.69	.98	.62	.94	.97	.94	.72	1.01	.98	.99

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW-
MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW-
MODEL NOSE, LOOKING AFT.

TABLE IX - Continued

RUN #1 POINT 12 ALPHA = 0 DEG PSI = 2.0 DLG V = 125 KNOTS q = 48.3 PSF. CONFIGURATION FHB									
.99	.97	.96	1.02	.99	.98	.97	.97	.99	.97
.96	.96	.98	.98	.96	.89	.96	.99	.96	.94
.97	.97	.92	.98	.99	.1	.99	.99	1.1	.94
.85	.85	.98	.88	1.06	-3.7	.86	1.03	-3.6	1.06
.99	.99	.94	.95	.87	.94	.92	1.02	1.09	.94
.94	.94	.62	.88	.98	.82	.82	.77	.93	1.04
.85	.85	.96	.94	.81	.95	.98	.73	.93	.93
.73	.73	.86	.72	1.04	1.04	.6	.65	1.06	.84
.98	.98	.67	.87	1.10	.97	-1.0	.56	1.8	.98
.70	.70	.68	.39	.95	-2.1	.90	.65	-3.3	.44
.81	.81	.78	.74	1.07	.94	.96	.70	.97	.52
.49	.49	.54	.49	.97	1.04	.96	.30	.62	.62
.99	.99	.60	.62	.89	.92	.93	.75	.38	.95
.98	.98	.70	RAKE MOUNTING POINT	.66	.86	.2	1.01	1.12	.97
.76	.76	.73	.73	1.02	.94	.96	1.4	-.1	.92
.86	.86	1.01	1.09	1.09	-2.3	.95	4.7	-2.6	1.09
.94	.94	.95	.73	.78	.98	.96	.65	1.05	.99
.99	.99	.52	.45	.94	.97	.97	.75	.99	.98
.94	.94	.85	.66	.95	.96	.96	.80	.99	1.00
							.53	.96	.99
									1.00

RUN #1 POINT 13 ALPHA = 0 DEG PSI = 2.0 DEG V = 125 KNOTS q = 48.3 PSF. CONFIGURATION FHB									
.97	.96	.99	1.03	.99	.97	.99	.97	.99	.97
.98	.99	.99	.99	.99	.99	.99	.99	.94	1.00
.98	.98	.98	.96	.94	.94	.94	.96	1.06	1.00
.95	.98	.98	.95	1.03	1.03	.97	.95	-3.6	.97
.99	.99	.99	.85	1.02	1.02	.99	.85	.94	1.00
.83	.83	.83	.77	.91	.77	1.04	.91	.93	.99
.88	.88	.88	.73	.88	.73	.93	.88	.93	.99
.77	.77	.77	.65	.49	.65	.84	.49	1.06	.84
.98	.98	.98	.56	.97	.56	1.8	.97	1.8	.98
.98	.98	.98	.46	.97	.46	-3.3	.97	-3.3	.97
1.02	1.02	1.02	.79	.52	.79	.31	.52	.31	.95
.92	.92	.92	.62	.62	.62	.95	.62	.62	.95
.96	.96	.96	.93	.38	.93	.95	.93	.38	.95
1.00	1.00	1.00	RAKE MOUNTING POINT	1.12	1.12	.97	.97	.97	1.02
.99	.99	.99	1.06	.26	1.06	-.1	.92	-.1	.92
.97	.97	.97	.65	1.05	.65	-2.6	1.09	-2.6	1.09
									.98
									1.00
									.99
									1.00
									.96
									.99
									1.00

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE IX - Continued

RUN #1 POINT 16 ALPHA=4.0 DEG PSI=-2.0 DEG V=125 KNOTS q = 48.3 PSF. CONFIGURATION FHB									
.98	1.00	.97	.98	.97	.91	.97	.94	.95	1.02
	.96	.95	1.05	.95	.98				
	.95	.73	.76	.94	.90				.95
	.84	.93	.96	.92	.97				
1.00	.71	.70	.94	.90	1.03				.99
	.63	.70	.78	.82	1.04				
.97	1.02	.52	.86	.41	.99				.98
	.96	.82	1.01	.24	.96				
1.00	1.08	.85	.95	.96	1.00				1.01
	1.01	.43	.84	.63	1.04				
.99	1.03	.75	.88	.36	.79				1.01
	.98	.77	.13	.64	.95				
1.01	.96	.91	.49	.87	.99				1.01
	.96	.36	RAKE MOUNTING POINT	.91	1.03				
1.01	2.5	1.03	3.5	1.05	1.4	1.07	1.02	1.07	1.01
	1.01	.98	1.04	1.04	1.04	.99	.94	.94	
1.00	.82	.84	.80	.96	.99				1.01
	.92	.60	.72	.93	.99				
.98	.96	.76	.50	1.00	.96				1.01

RUN #1 POINT 17 ALPHA=4.0 DEG PSI=-2.0 DEG V=125 KNOTS q = 48.3 PSF. CONFIGURATION FHB									
1.00	.99	.93	.94	.95	.94	.94	.95	.94	1.02
	.99	.98	.98	.95	.98				
.95	.78	.84	.98	1.01	.92				.95
	.98	.73	.65	1.01	1.02				
.98	.82	.94	.71	.93	.98				.99
	.97	.48	.76	.96	.83				
.97	.86	.79	.93	.88	.97				.98
	.97	.69	.99	.72	.66				
1.03	1.06	.93	.63	.91	.91				1.01
	.89	.95	.38	.95	.94				
1.00	.85	.72	.57	.94	.91				1.01
	.83	.95	.57	.96	.89				
1.02	.75	.82	.51	.99	1.03				1.01
	.95	.75	RAKE MOUNTING POINT	1.02	.57				
1.02	3.1	1.04	3.6	1.07	3.2	1.02	1.07	1.02	1.01
	1.01	.86	.80	.30	1.05	1.02	.94	.93	
1.01	.96	.84	.45	.99	.99	1.04			
.99	1.00	.68	.56	.98	.98	1.02			1.01

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE IX - Continued

RUN 41 POINT 18 ALPHA = 0 DEG PSI = 2.0 DEG V = 125. KNOTS q = 48.3 PSF. CONFIGURATION FHB				RUN 41 POINT 19 ALPHA = 0 DEG PSI = 2.0 DEG V = 125. KNOTS q = 48.3 PSF. CONFIGURATION FHB			
.98	.99	.95	.94	.99	.99	.93	.96
.99	.99	1.00	1.00	.98	.99	1.00	.98
.99	.96	.96	1.03	.98	.97	1.03	.92
.97	.95	1.02	1.02	.99	1.00	.94	.96
.92	.99	1.01	.93	.98	.97	.80	.91
.91	.88	.85	.93	.95	.99	.86	.99
.98	.76	.61	.92	.93	.81	.78	1.05
.98	.98	.80	.76	.93	.71	.83	.96
1.06	1.04	.93	.60	1.02	.83	.73	.57
1.05	.84	.46	.69	1.05	.58	.56	2.8
1.02	.93	.91	.77	1.05	.83	.45	1.2
.69	.97	.49	.87	.91	.94	.51	.66
.97	1.00	.99	.71	.93	.57	.68	.94
1.04	.95	.41	.36	1.09	.68	.73	.93
1.01	.97	1.11	1.10	.91	.47	.88	.96
1.02	.94	.96	.91	1.01	.45	.80	.95
.95	1.01	.25	.64	1.01	.76	.88	.88
1.02	.96	.91	.64	1.00	.92	1.05	1.04
.95	1.02	.94	.42	1.02	1.01	.81	.95
1.02	.95	1.01	.88	1.03	.99	.72	1.04
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$				NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_0}{q}$			
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.				NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.			
NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.				NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.			

TABLE IX - Concluded

RUN #1 POINT 20 ALPHA= 4.0 DEG PSI= 2.0 DEG V= 125 KNOTS q= 48.3 PSF. CONFIGURATION FHB		RUN #1 POINT 21 ALPHA= 4.0 DEG PSI= 2.0 DEG V= 125 KNOTS q= 48.3 PSF. CONFIGURATION FHB	
.97	.97	.96	.98
.97	.95	.95	.96
1.00	1.03	.97	.97
.97	.96	.97	.95
1.01	.80	.95	.75
.93	.83	.89	.64
.99	.74	.92	.68
1.00	.50	.87	.83
.98	.79	.57	.90
1.02	.98	.87	.91
1.04	1.04	1.1	1.2
.84	1.04	.99	.87
.85	.85	.87	.95
1.00	.72	.63	1.20
1.03	.84	.50	.56
.89	.84	.89	.76
1.07	1.10	.96	.70
1.01	.41	.89	.60
1.06	1.13	.93	1.15
1.08	3.6	.96	.63
.93	3.9	.97	.23
1.00	4.6	1.01	.30
.97	.56	.95	.59
1.00	.98	1.00	.98
.99	.17	.97	.99
.93	.69	1.01	.95
1.00	.31	.95	.93
.98	.52	.98	.98
1.00	1.00	.99	1.00

RUN #1 POINT 20 ALPHA= 4.0 DEG PSI= 2.0 DEG V= 125 KNOTS q= 48.3 PSF. CONFIGURATION FHB		RUN #1 POINT 21 ALPHA= 4.0 DEG PSI= 2.0 DEG V= 125 KNOTS q= 48.3 PSF. CONFIGURATION FHB	
2.6	1.1	2.7	1.1
1.2	.8	1.2	.3
1.04	3.0	2.3	.9
1.05	5.6	1.4	4.7
.98	1.08	1.4	1.07
2.8	1.01	2.3	.9
1.5	1.06	1.4	1.07
1.08	1.08	1.4	.96
.97	.93	1.4	.96
.99	1.00	1.4	.98
.98	1.00	1.4	.96
1.00	.98	1.4	.98
.99	.93	1.4	.98
.95	.98	1.4	.98
.93	.95	1.4	.98
.98	.98	1.4	.98
1.00	.98	1.4	.98

RUN #1 POINT 20 ALPHA= 4.0 DEG PSI= 2.0 DEG V= 125 KNOTS q= 48.3 PSF. CONFIGURATION FHB		RUN #1 POINT 21 ALPHA= 4.0 DEG PSI= 2.0 DEG V= 125 KNOTS q= 48.3 PSF. CONFIGURATION FHB	
2.8	1.01	2.7	1.1
1.5	1.06	1.2	.3
1.08	1.08	2.3	.9
.97	.93	1.4	4.7
.99	1.00	1.4	1.07
.98	1.00	1.4	.96
1.00	.98	1.4	.98
.99	.93	1.4	.98
.95	.98	1.4	.98
.93	.95	1.4	.98
.98	.98	1.4	.98
1.00	.98	1.4	.98

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH.
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT TO RIGHT LOOKING AFT.



NOTE - TABLE IS SEEN FROM THE FLOW MOUNTING POINT. LOOKING AFT.

TABLE X. WAKE PRESSURE COEFFICIENTS AND FLOW DIRECTIONS AT THREE FORWARD SPEEDS AND VARIOUS FUSELAGE ATTITUDES AND ROTOR OPERATING CONDITIONS, CONFIGURATION FHBf

RUN 28 POINT 6 ALPHA= .0 DEG PSI= -.0 DEG V= 60 KNOTS q= 11.4 PSF. CONFIGURATION FHBf		RUN 28 POINT 7 ALPHA=8.0 DEG PSI= -.0 DEG V= 60 KNOTS q= 11.4 PSF. CONFIGURATION FHBf	
.87	.78	.88	.96
.77	1.07	.95	1.00
.83	1.02	.94	.99
.96	1.09	.98	.97
1.00	1.09	1.11	1.00
1.11	1.02	1.17	.95
1.03	1.26	1.02	1.04
.93	1.21	1.13	1.07
.87	1.21	1.14	1.06
.89	.89	1.07	1.01
.95	.89	.69	1.13
.83	.91	.86	1.11
.87	1.02	.87	1.02
.87	.88	.48	1.06
.95	.73	1.12	1.06
.66	.70	.81	1.17
.81	.48	.86	1.35
.52	.77	.66	1.02
.79	.29	.32	.93
.92	.87	1.15	.87
.87	.82	.49	.97
.95	1.00	.48	.38
		.49	.72
		1.04	.65
		1.11	.79
		1.01	.55
		1.01	.84

RUN 28 POINT 6 ALPHA= .0 DEG PSI= -.0 DEG V= 60 KNOTS q= 11.4 PSF. CONFIGURATION FHBf		RUN 28 POINT 7 ALPHA=8.0 DEG PSI= -.0 DEG V= 60 KNOTS q= 11.4 PSF. CONFIGURATION FHBf	
.87	.88	.96	.89
.77	1.07	.95	1.00
.83	1.02	.94	.99
.96	1.09	.98	.97
1.00	1.09	1.11	1.00
1.11	1.02	1.17	.95
1.03	1.26	1.02	1.04
.93	1.21	1.13	1.07
.87	1.21	1.14	1.06
.89	.89	1.07	1.01
.95	.89	.69	1.13
.83	.91	.86	1.11
.87	1.02	.87	1.02
.87	.88	.48	1.06
.95	.73	1.12	1.06
.66	.70	.81	1.17
.81	.48	.86	1.35
.52	.77	.66	1.02
.79	.29	.32	.93
.92	.87	1.15	.87
.87	.82	.49	.97
.95	1.00	.48	.38
		.49	.72
		1.04	.65
		1.11	.79
		1.01	.55
		1.01	.84

RUN 28 POINT 6 ALPHA= .0 DEG PSI= -.0 DEG V= 60 KNOTS q= 11.4 PSF. CONFIGURATION FHBf		RUN 28 POINT 7 ALPHA=8.0 DEG PSI= -.0 DEG V= 60 KNOTS q= 11.4 PSF. CONFIGURATION FHBf	
.87	.88	.96	.89
.77	1.07	.95	1.00
.83	1.02	.94	.99
.96	1.09	.98	.97
1.00	1.09	1.11	1.00
1.11	1.02	1.17	.95
1.03	1.26	1.02	1.04
.93	1.21	1.13	1.07
.87	1.21	1.14	1.06
.89	.89	1.07	1.01
.95	.89	.69	1.13
.83	.91	.86	1.11
.87	1.02	.87	1.02
.87	.88	.48	1.06
.95	.73	1.12	1.06
.66	.70	.81	1.17
.81	.48	.86	1.35
.52	.77	.66	1.02
.79	.29	.32	.93
.92	.87	1.15	.87
.87	.82	.49	.97
.95	1.00	.48	.38
		.49	.72
		1.04	.65
		1.11	.79
		1.01	.55
		1.01	.84

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE X - Continued

RUN 28 POINT θ ALPHA=8.0 DEG PSI=-7.0 DEG V=60 KNOTS $q = 11.4$ PSF. CONFIGURATION FHB1				RUN 28 POINT θ ALPHA=8.0 DEG PSI=-7.0 DEG V=60 KNOTS $q = 11.4$ PSF. CONFIGURATION FHB1			
1.04	1.01	.92	.85	1.01	.86	.87	
.98	1.00	.93	.98	.98	9.2	.94	
1.01	1.01	1.08	.89	1.02	1.05	.93	
.96	1.01	.94	.95	.98	1.04		
.99	.95	.96	.85	.95	.99	1.02	
.89	1.02	.96	.95	.89	.86		
.75	1.07	1.04	1.07	.79	.75	.94	
1.21	1.20	1.09	1.09	.73	.81		
1.26	1.32	1.21	1.11	1.09	12.6	14.0	
1.06	1.11	1.06	1.14	1.25	.3	3.0	
1.14	1.34	1.04	1.00	1.30	1.17	1.38	
.87	1.82	1.33	1.05	.78	1.13		
.94	1.19	1.08	.99	1.17	1.09	1.08	
.85	1.04	.95	RAKE MOUNTING POINT	.94	.93	.90	
.87	1.12	.95	15.0	.56	19.7	14.5	
.83	1.7	5.9	.89	.61	3.9	1.1	
.84	1.03	.82	1.01	.61	.37		
.61	.98	.73	.46	.55	.41	.83	
1.00	1.04	.95	.37	.34	.73	.96	

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE X - Continued

RUN 28 POINT 14 ALPHA=8.0 DEG PSI=3.0 DEG V=6.0 KNOTS q=11.4 PSF. CONFIGURATION PH81		RUN 28 POINT 15 ALPHA=8.0 DEG PSI=4.0 DEG V=6.0 KNOTS q=11.4 PSF. CONFIGURATION PH81	
.99	.92	.99	.81
1.01	1.01	.97	.80
1.06	.95	.79	.92
1.02	1.02	.82	.89
.98	.99	.89	1.01
.91	.94	1.04	1.05
.75	.91	1.06	1.14
.90	.78	1.26	1.16
1.13	.27	1.28	.91
1.17	1.30	.95	1.33
1.39	1.26	1.20	1.39
1.27	1.22	1.07	1.11
.94	.97	1.05	1.02
.82	1.13	.64	.98
.99	.53	.82	1.00
.95	.63	1.14	.94
.95	.47	.88	.98
.62	1.12	1.06	.92
.85	.81	.98	.94
1.11	1.11	1.02	1.08
1.04	1.04	.42	.90
.96	.81	.39	.91
1.07	1.07	1.01	1.01
1.08	1.08	.83	1.08
.96	1.06	1.01	.88
1.01	1.02	.77	1.01
1.09	.88	.96	1.06
1.01	.79	1.01	1.18
.84	1.01	.68	1.22
1.08	1.15	1.25	2.22
.90	1.18	1.35	1.31
1.11	1.06	1.06	1.06
1.19	.95	.91	.98
1.58	.87	.82	1.38
1.19	.72	1.18	1.04
1.02	.85	.54	1.18
1.15	.68	.95	1.04
1.06	.64	.86	1.04
1.06	.47	.62	1.00
.97	.66	.92	.82
.82	.81	.61	.52
.88	.80	.87	.85
.97	.97	1.00	1.01
1.01	.98	.96	.89
.99	1.00	1.07	1.01
1.01	.98	1.07	.88
.87	.87	1.07	.80
.97	.97	1.08	.80
1.00	1.00	1.05	.84
1.07	1.07	1.10	.99
1.07	1.07	.98	.84
1.06	1.06	.88	.98
1.18	1.18	.77	1.01
.89	.89	1.22	.89
1.35	1.35	1.12	1.12
1.25	1.25	1.26	1.26
.89	.89	1.08	1.08
.84	.84	.85	.84
.88	.88	.82	.88
.99	.99	1.01	1.02
.69	.69	1.18	1.18
.5	.5	.96	.96

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_\infty}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW.
 MODEL NOSE, LOOKING AFT.

74.75 IN

RAKE MOUNTING POINT

TABLE X - Continued

RUN 28 POINT 16 ALPHA=4.0 DEG PSI=4.0 DEG V= 60. KNOTS q= 11.4 PSF. CONFIGURATION F181										
.92	.93	1.03	.94	.97	1.04	.95	.99	1.00	.84	.94
.91	.91	.91	1.00	.98	1.08	.74	.91	1.28	.77	.74
.94	.98	1.01	1.04	.94	-3.9	1.00	1.15	.60	.71	.84
1.04	1.04	1.02	1.04	.93	-5.8	.96	1.15	.81	.96	.94
1.02	1.15	1.07	.96	.96	1.06	1.02	.51	1.07	.66	.79
.93	.93	.97	.90	1.11	1.19	.83	.76	.89	.87	.83
.99	1.09	1.09	1.15	1.18	1.13	1.10	.99	.97	.70	.81
1.15	1.40	1.30	1.35	1.18	1.22	1.07	.96	.99	.77	.94
1.01	-9.6	-7.7	1.17	-6.5	-7.3	1.07	.90	-6.3	2.4	-3
1.02	-8.1	-6.7	1.05	-4.3	-4.0	1.04	-1.3	-7.2	-4.8	-1.6
.97	.96	1.01	.96	1.09	1.10	1.13	.96	.62	1.18	.94
1.06	.81	.89	.29	1.02	1.12	1.04	.96	.45	.69	.96
1.06	.94	.97	.54	1.02	1.04	1.04	.70	.44	.61	.94
1.06	.77	1.11	RAKE MOUNTING POINT	.98	.94	1.13	.63	.55	.93	.95
1.02	-9.1	-14.2	.56	-6.8	-3.2	1.02	-2	-1.9	1.13	2.1
.96	-1.5	-4.7	.52	-1.7	-2.5	.99	-2.5	-1.2	.97	-2.0
1.02	.75	.41	.58	1.05	1.05	1.02	1.04	.72	.97	.98
.87	.87	.65	.76	1.12	1.10	1.02	.77	.42	.98	1.09
.62	.62	.64	.83	1.09	1.09	.99	.78	.54	1.08	1.03
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_s}{q}$										
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.										
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH										
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.										
NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.										

RUN 28 POINT 17 ALPHA= 4.0 DEG PSI=4.0 DEG V= 60. KNOTS q= 11.4 PSF. CONFIGURATION F181										
.99	.87	1.00	.99	.84	.93	.94	.99	1.00	.84	.94
.91	.91	1.28	.73	.77	.74	.74	.91	1.28	.77	.74
.91	1.15	.60	.56	.71	.84	.79	1.15	.60	.71	.84
1.00	1.15	.81	.99	.96	.94	.85	1.15	.81	.96	.94
.74	.51	1.07	1.03	.66	.79	.85	.51	1.07	.66	.79
.74	.76	.89	1.20	.87	.83	.83	.76	.89	.87	.83
.96	.99	.97	.75	.70	.81	.95	.99	.97	.70	.81
.90	-5.4	.99	1.01	.77	.94	.94	.96	.99	.77	.94
.96	-1.3	.96	.41	.96	1.03	1.01	.90	.96	.96	1.01
.92	.92	.93	.17	.98	1.03	1.08	.92	.93	.98	1.03
.92	.92	.62	.18	1.18	.94	1.08	.92	.62	1.18	.94
.92	.96	.45	.39	.69	.96	1.10	.96	.45	.69	.96
.92	.70	.44	.61	.94	.87	1.10	.92	.44	.61	.94
1.04	-2	.62	RAKE MOUNTING POINT	.93	.95	1.08	.63	.55	.93	.95
1.01	-1.9	.72	1.13	1.03	2.1	1.07	-2	.72	1.13	2.1
.93	-2.5	.98	.97	.97	-4.0	.98	-2.5	.98	.97	-4.0
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_s}{q}$										
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.										
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH										
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.										
NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.										

TABLE X - Continued

RUN 26 POINT 19 ALPHA=4.0 DEG PSI=4.0 DEG V=60 KNOTS q=11.4 PSF. CONFIGURATION FHB1		RUN 26 POINT 20 ALPHA=4.0 DEG PSI=4.0 DEG V=60 KNOTS q=11.4 PSF. CONFIGURATION FHB1										
1.04	.98	.88	.95	.82	.83	1.09	1.01	.99	.97	.93	.94	1.10
1.20	.98	.88	.99	.86	.98		.99	.98	1.01	.98	1.01	
	.95	.92	1.07	.91	-2.6		.92	.95	1.06	1.03	-6.2	1.04
	.89	1.01	1.21	1.08	2.3		.96	1.01	.98	1.10	2.6	1.16
.63	.70	1.14	.96	1.10	1.13	1.10	1.07	1.01	1.11	1.09	1.17	.87
	1.15	.67	.86	.83	.86		1.05	1.05	1.02	.93	1.17	
1.06	1.01	.94	1.05	.64	.75	.84	1.16	1.16	1.01	1.08	1.01	1.32
	1.03	1.04	.78	.60	.76		1.16	1.16	1.19	1.50	.94	
1.00	-2.2	-3.5	.92	.44	-6.2	.3	-9.8	1.28	1.22	1.00	-11.9	.94
	.6	1.8	.91	1.04	2.4	1.5	-1.2	-4.0	1.06	.69	1.06	-11.2
	.92	.91	.91	.74	.74		.95	.95	1.06	.94	1.20	-.1
1.01	.89	.89	1.05	.45	1.34	.88	.91	1.06	.95	.94	.97	.90
	.93	.81	.99	.58	1.29		1.05	.94	.93	.87	1.03	
1.02	.95	.79	.69	.82	.83	.89	1.22	.93	.66	1.12	.81	.87
	.91	.91	RAKE	.92	.93		.82	1.14	RAKE	.69	.94	
1.12	.7	-2	MOUNTING	.47	-1.2	.8	-4.9	-5.6	POINT	.72	-22.5	-5.9
	1.3	5.9	POINT	1.11	.7	3.6	-1.3	6.2	POINT	.65	-5.6	3.7
	.95	.92		.76	.76		.96	1.13		.65	.63	
1.07	.93	.94	.32	.62	.86	.77	1.00	.98	.59	.66	.57	1.03
	.98	.89	.60	.88	.87		1.04	1.00	.74	.73	.69	
1.05	.94	.92	.33	.21	.87	1.08	1.02	.90	.56	.61	1.06	1.09

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW.
MODEL NOSE, LOOKING AFT.



TABLE X - Continued

RUN 20 POINT 21 ALPHA=-8.0 DEG PSI= 4.0 DEG V= 60 KNOTS q= 11.4 PSF. CONFIGURATION FHB1		RUN 20 POINT 22 ALPHA=-8.0 DEG PSI= 4.0 DEG V= 60 KNOTS q= 11.4 PSF. CONFIGURATION FHB1						
.97	1.01	.93	.91	1.02	.89	.99	.80	.88
1.04	.99	.91	.89	1.05	1.02	1.00	-12.1	.84
1.05	1.00	.95	.93	1.06	1.11	.91	1.03	.96
1.06	.99	.98	.92	1.02	1.02	.86	1.07	1.06
1.05	1.02	.93	.97	1.04	1.01	.91	1.08	1.08
1.06	1.04	.82	.98	.99	1.01	.91	1.04	1.06
1.06	1.14	.59	1.14	.98	.92	1.11	.99	.93
1.25	.98	1.25	1.01	1.08	.88	.95	.97	.81
1.25	1.15	1.41	1.12	1.21	.98	1.07	-13.4	-10.4
1.20	1.28	.98	1.23	1.24	-1.3	1.26	1.13	1.44
1.20	1.12	1.03	1.19	1.16	1.28	1.30	1.47	1.27
.99	.92	.89	.86	1.16	1.15	1.17	.88	1.11
.99	.85	1.00	.86	1.19	1.05	.99	.98	.98
1.18	.91	1.10	RAKE MOUNTING POINT	1.02	1.08	RAKE MOUNTING POINT	1.15	.58
1.18	1.00	1.05	1.05	1.04	-11.8	1.02	-14.1	.62
1.06	1.24	1.17	.96	1.11	20.3	.93	5.3	.51
1.06	1.07	1.26	.67	.92	.93	.82	1.33	.85
.91	.98	1.23	.25	.87	1.12	.47	.25	.96
.91	1.28	.94	.64	.96	.99	.22	.57	.88
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\rho V^2}$		NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\rho V^2}$						
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.		NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.						
NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.		NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.						

TABLE X - Continued

RUN 28 POINT 23 ALPHA = 0 DEG PSI = 8.0 DEG V = 60 KNOTS q = 11.4 PSF. CONFIGURATION FHBY				RUN 28 POINT 24 ALPHA = 0 DEG PSI = -0 DEG V = 60 KNOTS q = 11.4 PSF. CONFIGURATION FHBY							
.91	.90	1.07	1.08	1.11	.88	.90	1.00	.96	1.03	1.04	.91
.91	.90	1.06	1.01	1.09	.85	.86	.98	.92	.86	1.02	.89
1.06	.76	.78	-2.5	1.09	1.18	.91	.92	.71	1.05	-3.3	.89
.78	.43	.80	.78	1.00	1.18	1.14	.91	.91	1.08	-1.4	1.23
.92	1.36	1.16	1.40	1.00	.85	1.15	1.57	1.26	.98	1.10	1.05
.95	1.05	1.03	1.18	.96	1.04	1.09	1.08	1.15	1.02	1.26	1.02
.95	.95	.95	1.03	.96	1.04	1.03	.95	.88	1.05	1.23	1.05
.92	1.03	.83	1.15	.61	1.04	.94	.93	.85	1.11	1.16	1.02
.95	.84	1.00	-7.7	-7.5	1.04	1.10	-7.9	.90	1.34	-2.3	1.02
.92	5.3	1.04	6.1	-7.5	.87	.87	.1	1.05	.89	-3.6	1.02
.95	.88	1.07	.92	.63	.98	.95	1.04	.94	1.06	.86	1.07
.92	1.05	.93	.90	.72	.92	.92	.94	.77	.69	.87	1.05
1.10	.97	.99	1.01	.23	.95	.33	.72	.75	.39	.60	1.05
1.03	1.08	1.01	.67	.63	1.15	.73	.77	RAKE MOUNTING POINT	1.21	.91	1.04
.95	-2.6	.30	.77	1.01	1.03	-3.4	-1.7	.42	.12	-0.8	1.02
1.06	1.09	.60	.26	.63	1.03	-3.8	.0	.91	.59	4.3	.86
1.07	.95	1.07	.57	1.01	.98	.96	.91	.42	1.17	.95	.95
1.06	1.06	.94	.75	1.04	1.03	1.05	1.29	.85	1.09	1.02	.95
1.07	1.07	1.02	1.09	1.01	.98	.95	.87	.25	1.11	1.02	.86
1.08	1.08	1.03	.77	1.01	.98	1.02	.92				

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT 7.475 IN

TABLE X - Continued

RUN 29 POINT 5 ALPHA=0.0 DEG PSI=-0.0 DEG V=60 KNOTS q=11.4 PSP. CONFIGURATION FBI/		RUN 29 POINT 6 ALPHA=0.0 DEG PSI=-0.0 DEG V=60 KNOTS q=11.4 PSP. CONFIGURATION FBI/								
.91	.85	.94	.87	1.02	1.07	.96	1.06	1.05	1.18	1.29
.93	.89	.86	.86	1.00	1.00	.99	1.00	1.06	1.17	1.17
.96	1.00	.69	.92	1.00	1.00	1.05	1.03	1.18	-8.6	1.24
1.08	.97	.94	1.12	1.08	.98	1.08	1.02	1.17	-2.1	1.22
1.04	1.04	1.22	1.14	1.07	.95	1.06	.99	1.17	1.17	1.21
1.07	1.19	1.14	1.14	.95	.95	.98	.99	1.18	1.11	1.11
1.11	.87	.77	.77	.93	1.08	.91	.84	1.16	1.12	1.16
1.06	.94	1.09	1.09	1.11	1.11	.68	.98	1.26	1.11	1.11
-5.1	-5.7	1.01	1.00	-8.4	-9.3	.64	1.18	1.40	-9.8	-11.0
1.9	1.02	1.01	1.00	-8	-4.7	.64	1.18	1.40	-1.8	.89
.88	.88	.98	.88	1.08	1.08	1.27	1.24	1.16	2.52	.7
.98	.95	.87	.87	1.50	1.50	1.36	1.28	1.57	1.37	1.52
.84	.89	.83	.83	1.39	1.39	1.33	1.19	1.65	1.42	1.42
.92	.78	.76	.76	1.08	1.08	1.05	.71	1.07	1.17	1.23
.91	.64	RAKE	RAKE	1.00	1.00	.78	RAKE	.93	.96	.96
-2.1	-1.3	MOUNTING	MOUNTING	-11.1	-12.9	.96	POINT	.83	-8.5	-8.3
-2.1	3.8	POINT	POINT	1.8	18.7	.88	POINT	.96	-1.4	-11.0
1.00	.95	.54	.54	1.17	1.17	.88	.74	.74	.64	.64
.98	1.06	.60	.60	1.17	1.17	.78	.74	.83	1.26	1.18
.99	.69	.51	.51	1.02	1.02	1.12	.35	.80	.79	.79
.99	.90	.43	.43	1.06	1.06	1.09	.60	.74	.53	1.18

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

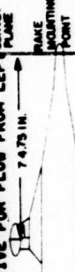


TABLE X - Continued

RUN 29 POINT 7 ALPHA=-8.0 DEG PSI=-.0 DEG V=60. KNOTS q = 11.4 PSF. CONFIGURATION FHB				RUN 29 POINT 8 ALPHA=-4.0 DEG PSI=-.0 DEG V=60. KNOTS q = 11.4 PSF. CONFIGURATION FHB								
1.00	1.02	.94	1.00	.90	1.07	1.13	.98	.99	.99	.93	1.00	1.06
	.98	.92	.99	.93	1.10	1.10		1.00	.95	.94	1.08	
1.01	.98	.97	1.04	1.10	-6.9	1.15	1.00	.98	.98	1.14	-3.6	1.11
	1.03	1.03	1.04	1.03	-2.5	1.04		1.00	.96	1.11	-1.6	1.08
1.01	1.00	1.05	1.07	.92	1.07	1.17	.99	.95	.85	1.09	1.08	1.18
	.93	1.08	.99	.89	1.09		.84	.70	.95	1.02	1.04	
1.06	.61	1.11	.92	1.05	1.18	1.22	.96	1.35	1.15	.86	1.02	1.15
	1.17	1.15	1.55	1.16	1.04	1.16		1.38	1.22	1.32	1.17	
1.21	-16.6	-9.9	1.28	1.38	-12.1	-13.4	1.24	-8.7	-10.0	-10.8	-4.9	1.02
	-4.7	-4.8	1.55	1.19	-2.3	.3		-3.9	-4.2	1.18	-5.0	-1.2
	1.18	1.11	1.10	1.10	1.37			1.01	.98	1.16	1.17	
1.21	1.02	1.00	.96	1.02	1.17	.98	.93	.90	1.10	.72	1.22	.98
	.85	.88	.71	.98	1.18			1.09	.83	.82	1.11	
.90	.90	.95	.75	1.03	.58	1.40	.98	.89	.97	.91	1.02	.88
	1.01	.86	RAKE MOUNTING POINT	.88	1.06	-6.1		.41	.61	RAKE MOUNTING POINT	.39	
1.25	-10.9	-13.5	.84	.93	-14.2	.46	1.08	-7.5	-9.3	.43	-10.7	.93
	2.1	.5	.94	.84	-5.3	.76		-1.2	3.0	.58	1.0	.2
1.14	1.14	.85	.64	.25	.87	1.19	1.15	.94	.57	.46	.85	1.04
	1.08	.63	.71	.29	.75			1.06	.93	.60	.95	
1.01	.82	.80	.40	.37	.94	1.10	1.00	1.16	.74	.85	.91	1.10

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS C_p - PT. 7

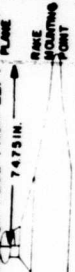
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.



NOTE - TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS C_p - PT. 8

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.



NOTE - TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

TABLE X - Continued

RUN 29 POINT 9 ALPHA= 9.0 DEG PSI= -0 DEG V= 60 KNOTS q = 11.4 PSF. CONFIGURATION FHB/		RUN 29 POINT 10 ALPHA= 8.0 DEG PSI= -0 DEG V= 60 KNOTS q = 11.4 PSF. CONFIGURATION FHB/	
.98	.96	.90	.88
.93	.93	.05	.75
1.05	.92	.84	.80
.97	.81	.87	.86
1.05	.94	.71	.84
.92	1.07	.79	1.32
.65	.98	1.07	.83
.73	.68	.96	.33
.82	.98	.88	.85
.66	.77	.89	.81
.94	.31	.69	.61
.94	1.01	.95	.44
.94	.77	.93	.86
.80	.45	.71	.75
.46	.35	.40	.72
1.03	RAKE	.93	.64
1.25	MOUNTING	4.3	4.2
1.02	POINT	4.6	3.4
.95	1.02	.94	.94
1.03	.76	.91	.83
.89	.89	.92	.80
.89	.89	.91	.89
.95	.89	.91	.93
1.08	1.01	.57	1.03
1.08	1.08	1.08	1.08

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
TO RIGHT LOOKING AFT.



NOTE- TABLE IS SEEN FROM THE FLOW
MODEL NOSE, LOOKING AFT.

TABLE X - Continued

RUN 29 POINT 11 ALPHA=0 DEG PSI=-6.0 DEG V=60. KNOTS q = 11.4 PSF. CONFIGURATION FHBY										RUN 29 POINT 12 ALPHA=0 DEG PSI=-4.0 DEG V=60. KNOTS q = 11.4 PSF. CONFIGURATION FHBY									
.96	.93	.99	.88	.85	.76	.70	.95	.93	.96	1.02	1.00	1.01	.93						
1.09	.98	.99	.96	.82	-1.8	.83	.87	1.02	.99	1.00	-10.5	.93							
1.09	1.08	.99	1.09	.94	-8.6	1.01	.92	.97	.94	.91	-7.4	.85							
1.09	1.18	.93	1.08	.91	1.09	1.06	.91	.97	.88	.92	.89	.95							
1.01	1.13	1.12	1.07	1.06	1.06	1.06	.91	.98	.96	.91	.93	.95							
1.01	1.06	1.21	1.05	1.09	.94	1.01	.96	.96	1.06	.89	.93	.87							
	.96	1.12	.89	1.01	.88	1.01	1.00	1.06	1.08	.85	.86	.87							
.88	-7.3	1.04	.94	.98	.84	.93	1.06	1.14	1.18	.67	1.04	-13.8							
	.96	-5.6	.89	.91	-3.9	.71	1.15	-13.4	1.17	1.14	-11.1	-13.8							
	-7.2	-8.8	.65	.47	-7.7	-5.9	-14.4	-9.1	1.14	1.28	2.8	-5.8							
.86	1.08	.65	.47	.85	.79	.91	.98	1.14	1.22	1.34	1.30	.67							
.86	.98	.21	.57	.79	.82	.91	1.02	1.11	.98	1.17	1.16	1.34							
.82	.68	.21	.49	.78	.89	.81	1.17	1.17	.78	.93	1.08	1.18							
	.72	.73	.94	.93	.85	.81	1.42	1.11	.66	.75	.94	1.18							
.99	.91	.72	RAKE MOUNTING POINT	.86	.93	.72	RAKE MOUNTING POINT	.97	1.08	1.08	1.01	-7.5							
	-1.9	-7.2		.93	-1.2	1.2	-13.7	-17.7	.72	-10.0	1.09	1.07							
	-4.6	9.7		.86	-8.4	-5.7	-2.9	-1.0	.82	-3.5	1.08	.86							
.96	.59	.81	.88	.88	.81	.78	.92	1.25	.71	1.14	1.12	1.00							
.84	.84	.36	.88	.82	.75	.78	1.04	.89	.59	.91	1.04	1.00							
.92	.80	.53	.88	.78	.74	.78	.88	.89	.71	.71	1.01	.86							
							.87	.43	.86	.86	.86	.86							

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
 TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW.
 MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
 TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW.
 MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

TABLE X - Continued

RUN 29 POINT 13 ALPHA=8.0 DEG PSI=-4.0 DEG V= 60 KNOTS q= 11.4 PSF. CONFIGURATION PH87										RUN 29 POINT 14 ALPHA=4.0 DEG PSI=-4.0 DEG V= 60 KNOTS q= 11.4 PSF. CONFIGURATION PH87									
1.00	1.02	1.00	1.03	.93	1.11	1.13	1.08	.93	1.08	1.05	1.02	1.12	1.08	.93	1.08	1.05	1.02	1.12	
1.04	1.00	.97	1.00	.97	1.19	1.13	.99	.99	1.11	1.08	1.02	1.12	.99	.99	1.11	1.08	1.02	1.12	
1.03	.95	1.00	.94	1.04	1.14	1.13	1.03	1.03	1.07	1.12	1.02	1.04	1.03	1.03	1.07	1.12	1.02	1.04	
1.03	.92	1.04	.98	.98	1.05	1.15	1.06	1.06	1.09	1.11	1.00	1.06	1.06	1.06	1.09	1.11	1.00	1.06	
.86	.87	.89	.89	.89	1.02	1.15	.96	.96	.90	1.06	1.06	1.06	.96	.96	.90	1.06	1.06	1.06	
.86	.86	.89	.89	.89	1.13	.99	.88	.88	.75	.96	1.11	1.16	.88	.88	.75	.96	1.11	1.16	
.86	1.05	.52	.85	1.26	.98	1.10	1.40	1.40	1.67	1.49	.89	1.00	1.40	1.40	1.67	1.49	.89	1.00	
1.42	1.19	1.14	1.24	1.07	1.43	1.24	1.20	1.20	1.50	1.42	1.12	1.05	1.20	1.20	1.50	1.42	1.12	1.05	
1.42	1.23	1.23	1.37	1.38	1.51	1.24	.95	.95	1.31	1.00	1.30	1.02	.95	.95	1.31	1.00	1.30	1.02	
.94	1.11	1.26	1.25	1.36	1.23	1.22	1.02	1.02	1.02	1.14	1.26	1.16	1.02	1.02	1.02	1.14	1.26	1.16	
1.03	1.05	1.04	.83	1.17	1.11	1.10	.89	.89	1.02	1.01	1.21	.99	.89	.89	1.02	1.01	1.21	.99	
.94	1.06	1.05	1.14	1.14	1.12	1.17	.93	.93	.97	.84	1.10	1.10	.93	.93	.97	.84	1.10	1.10	
.94	.80	1.12	1.23	1.01	.96	1.22	.89	.89	.93	.81	.95	1.04	.89	.89	.93	.81	.95	1.04	
1.19	.92	.95	RAKE MOUNTING POINT	.71	.72	-6.0	RAKE MOUNTING POINT	.94	.70	.83	.87	1.01	RAKE MOUNTING POINT	.94	.70	.83	.87	1.01	
1.13	.91	.69	.49	.49	.87	1.19	1.12	1.04	.89	.90	.87	1.11	1.12	1.04	.89	.90	.87	1.11	
1.13	.99	.51	.89	.89	1.13	1.17	1.04	.54	.73	.94	.85	1.12	1.04	.54	.73	.94	.85	1.12	
1.03	.89	.82	.31	.91	1.11	1.17	1.04	.81	.63	.39	.82	1.04	1.04	.81	.63	.39	.82	1.04	
1.03	.98	.79	.45	1.08	1.14	1.17	.77	.77	.48	.85	1.01	1.04	.77	.77	.48	.85	1.01	1.04	
1.03	.80	.83	.42	1.14	1.11	1.17	1.04	1.07	.71	.99	1.10	1.04	1.04	1.07	.71	.99	1.10	1.04	

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

NOTE- TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

TABLE X - Continued

RUN 29 POINT 15 ALPHA=4.0 DEG PSI=4.0 DEG V=60 KNOTS q=11.4 PSF. CONFIGURATION FMB1										RUN 29 POINT 16 ALPHA=9.0 DEG PSI=9.0 DEG V=60 KNOTS q=11.4 PSF. CONFIGURATION FMB2																							
.84	.94	1.06	1.43	.93	.83	.91	.94	.99	.94	1.03	1.07	1.07	1.07	1.07	.91	.84	.94	1.06	1.43	.93	.83	.91	.94	.99	.94	1.03	1.07	1.07	1.07	1.07	.91		
.96	1.04	.98	.87	.74	-2.2	.74	1.01	.96	.74	-2.2	.74	1.01	.96	.74	-1.6	.73	.96	1.04	.98	.87	.74	-2.2	.74	1.01	.96	.74	-2.2	.74	1.01	.96	.74	-1.6	.73
1.10	1.13	.81	1.00	.94	-4.9	.97	1.00	.94	.94	-4.9	.97	1.00	.94	.94	2.7	.99	1.10	1.13	.81	1.00	.94	-4.9	.97	1.00	.94	.94	2.7	.99	.94	.94	2.7	.99	
.70	1.04	.91	1.06	.85	.98	1.08	1.14	.74	.74	.93	.93	1.06	1.14	.74	.96	.99	.70	1.04	.91	1.06	.85	.98	1.08	1.14	.74	.74	.93	.93	1.06	1.14	.74	.96	.99
.91	1.39	1.38	1.35	.78	.80	.83	1.35	.78	.78	.80	.80	1.38	1.35	.78	.94	1.00	.91	1.39	1.38	1.35	.78	.80	.83	1.35	.78	.78	.80	.80	1.38	1.35	.78	.94	1.00
1.02	.80	1.22	1.29	.76	.86	.89	1.29	.76	.76	.86	.86	1.22	1.29	.76	.76	.99	1.02	.80	1.22	1.29	.76	.86	.89	1.29	.76	.76	.86	.86	1.22	1.29	.76	.76	.99
.89	.73	.87	.70	.65	-5.3	.90	.51	.71	.85	-2.8	.85	.87	.70	.65	-2.1	1.00	.89	.73	.87	.70	.65	-5.3	.90	.51	.71	.85	-2.8	.85	.87	.70	.65	-2.1	1.00
.85	1.00	1.16	.66	.83	.83	.90	.66	.83	.83	.83	.83	1.16	1.00	.83	.98	1.05	.85	1.00	1.16	.66	.83	.83	.90	.66	.83	.83	.83	1.16	1.00	.83	.98	1.05	1.09
1.05	.90	.56	.56	.98	-2.7	.84	.99	.99	1.5	2.3	.84	.99	.99	1.5	2.3	.84	1.05	.90	.56	.56	.98	-2.7	.84	.99	.99	1.5	2.3	.84	.99	.99	1.5	2.3	.84
1.02	.73	.92	1.03	.87	.87	.84	1.03	.87	.87	.87	.87	1.03	.87	.87	.84	1.08	1.02	.73	.92	1.03	.87	.87	.84	1.03	.87	.87	.87	1.03	.87	.87	.84	1.08	
.94	.89	1.12	.37	.78	.79	.85	.37	.79	.79	.79	.79	.37	.79	.79	.85	1.00	.94	.89	1.12	.37	.78	.79	.85	.37	.79	.79	.79	.37	.79	.79	.85	1.00	

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

TABLE X - Continued

RUN 29 POINT 17 ALPHA=4.0 DEG PSIZ 4.0 DEG V= 60. KNOTS q= 11.4 PSF. CONFIGURATION FMB7									
.98	.91	.93	.97	1.11	.86	1.00			
	.96	.95	1.04	1.05	-6.1	.95			
.91	1.01	1.00	1.11	.91	1.01	1.05			
	.98	1.07	1.07	.86	1.06	.96			
.95	.97	1.23	1.09	.86	1.08	.96			
	1.00	1.23	1.04	.99	1.09				
1.02	1.06	1.22	1.18	1.16	1.02	1.17			
	.98	1.16	1.09	1.16	.94				
1.11	1.15	1.18	1.12	1.07	-12.4	-12.1			
	1.03	1.07	1.14	.80	-1.7	1.20			
1.03	1.11	.92	1.00	.85	1.08	.94			
	1.02	1.32	.87	.73	1.26	.94			
1.01	1.03	1.11	.88	.82	1.26	1.01			
	1.05	1.07	RAKE MOUNTING POINT	.78	-21.8	.82			
1.08	1.07	.76		1.12	.61	-3.5			
	1.11	.67		.53	5.5	3.1			
1.00	1.13	.85	.45	.47	.63	1.12			
	1.11	.92	.71	.89	.93	.96			
1.02	1.00	.93	.75	.50	.83				

RUN 30 POINT 6 ALPHA=3.0 DEG PSIZ 4.0 DEG V= 60. KNOTS q= 11.0 PSF. CONFIGURATION FMB7									
.92	.97	.99	.91	.95	.82	.84			
	.94	.93	.90	.96	-8.3	.80			
.99	.94	.99	.93	.90	2.0	.89			
	.98	.96	.85	1.04	1.00	.82			
.96	1.01	.85	.93	1.00	.95	.82			
	1.05	.81	1.02	1.13	.80	.80			
1.06	1.05	1.10	1.13	.94	1.33	1.39			
	1.02	1.10	1.05	1.07	1.36				
1.00	1.20	1.11	.96	.84	-11.8	-5.7			
	.98	1.00	1.08	.88	-2.5	-1.3			
1.06	.80	.83	.75	.74	.86	.89			
	.90	.94	.91	.79	.78	.89			
.91	1.02	.88	.85	.63	.53	.91			
	1.00	RAKE MOUNTING POINT	1.00	.96	1.06				
1.08	-5.6	-5.2		.87	-21.3	-8			
	-1.1	-2.7		.65	5.7	2.5			
1.04	.95	1.08	.25	.60	1.89	.76			
	.95	.99	.54	.51	.87	.80			
.95	1.00	.96	.71	.33	.86	.80			

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE X - Continued

RUN 30 POINT 7 ALPHA=-0.0 DEG PSIE 4.0 DEG V= 60 KNOTS q = 11.4 PSF. CONFIGURATION FMB7				RUN 30 POINT 8 ALPHA=0.0 DEG PSIE 0.0 DEG V= 60 KNOTS q = 11.4 PSF. CONFIGURATION FMB8			
1.02	.98	1.04	.99	1.00	.93	.99	.99
	.96	.99	.98	.97	1.01	.91	.91
.96	.98	.94	1.02	.98	-8.4	-11.2	1.04
	.91	.89	.97	1.10	1.14	1.07	.99
.96	.96	.87	.94	1.12	1.13	.95	.99
	1.01	.99	.76	1.16	1.03	.87	.84
1.00	1.10	.65	1.04	1.17	.82	.76	.87
	1.04	1.14	1.17	.87	1.60	.81	.93
1.11	1.10	1.23	1.10	1.04	-14.1	-13.0	1.03
	1.28	.99	1.04	1.27	2.0	2.3	1.7
1.04	1.00	1.04	1.13	1.00	.88	1.01	1.10
	1.06	1.03	.94	.96	.88	1.18	.86
.94	1.04	.99	1.02	.88	.79	1.07	1.04
1.20	1.11	1.11	RAKE MOUNTING POINT	.95	.80	RAKE MOUNTING POINT	1.17
	1.08	1.13		.77	-26.0	-15.0	1.03
1.06	1.21	1.05	.24	.54	-31.6	-15.0	.73
	.89	1.12	.85	.16	.40	.35	.34
1.06	.87	1.06	.63	.92	1.05	.27	.86
				.67	.35	.17	.85
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$				NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$			
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SIDE TO RIGHT LOOKING AFT.				NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SIDE TO RIGHT LOOKING AFT.			
NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.				NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.			

TABLE X - Continued

RUN 30 POINT 9 V = 60 KNOTS		ALPHA = 0 DEG PSI = 0.0 DEG		CONFIGURATION FMB7		RUN 30 POINT 10 V = 60 KNOTS		ALPHA = 0 DEG PSI = 0.0 DEG		CONFIGURATION FMB7	
1.03	1.03	.97	.95	1.06	1.07	1.00	.95	.97	.93	.03	.09
1.02	1.02	.99	1.05	1.00	1.07	.95	.92	.92	.99	.05	.09
1.06	1.06	1.04	1.03	-0.6	1.07	.99	.97	1.00	.95	.03	.09
1.13	1.13	1.13	1.19	1.11	1.20	.98	.94	.97	1.03	1.04	.07
1.13	1.13	1.08	.99	1.13	1.29	1.00	1.01	1.03	.95	.04	.07
1.31	1.31	1.14	1.14	1.13		.98	.99	.95	.94	.06	.06
.99	.91	1.07	1.04	.90	.93	.93	.82	.71	.75	.91	1.01
1.06	1.06	1.13	.94	.96		.95	.87	.79	.86	.09	
1.01	1.01	.98	.98	-7.0	.87	.94	-3.5	.92	.81	-1.0	
.80	.80	.84	1.03	6.9	2.7	.80	2.2	.87	.84	-3.8	1.00
.96	.96	.92	1.11	1.16		.94	.56	.56	.52	.75	.09
1.01	1.01	1.00	1.13	1.15	.92	.75	.76	1.03	.12	.68	.09
.96	.96	1.00	.95	1.14	.98	.93	.68	.73	.80	.91	.09
.98	.98	.97	.90	.93		.93	.32	RAKE MOUNTING POINT	1.07	1.02	
1.18	1.18	.93	.70	-9.2	.35	.99	1.2		.95	-2.2	.09
.95	.95	.95	.76	30.8	5.7	1.01	5.5		.93	1.9	1.1
1.09	1.09	.94	.77	.81	1.25	1.09	1.10	.76	.83	.09	.09
.93	.93	.86	.80	.54		1.13	.91	.37	.75	.02	.02
.94	.94	.93	1.22	.76	.89	.97	.68	.68	.87	.68	.09

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SIDE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

TABLE X - Continued

RUN 31 POINT 5 V=100. KNOTS		ALPHA=-4.0 DEG q = 30.8 PSF. CONFIGURATION FHW		ALPHA=-0 DEG q = 30.8 PSF. CONFIGURATION FHW		RUN 31 POINT 6 V=100 KNOTS		ALPHA=-0 DEG q = 30.8 PSF. CONFIGURATION FHW	
1.00	.99	.99	1.00	.96	1.01	.95	1.04	.97	1.03
1.00	1.00	.99	1.01	1.00	1.00	.97	1.00	1.07	.96
1.00	.98	1.03	.99	.97	1.04	.92	.99	.95	1.20
.97	.99	.97	.95	1.03	1.04	1.02	1.01	1.05	.76
1.00	1.00	1.02	1.08	1.04	1.04	.99	.95	.88	1.01
.96	1.01	1.00	1.05	.97	.97	.98	1.04	.96	1.20
1.04	1.01	1.01	1.02	.99	.97	1.03	1.22	.93	.92
.88	1.04	1.39	.89	1.23	1.11	1.09	1.03	1.15	.91
1.04	1.06	1.10	.96	.82	1.06	.75	1.10	.86	1.01
.97	.97	.87	.87	.95	1.06	.96	.99	.97	.94
1.00	1.00	.90	.70	.85	.93	1.03	.83	1.06	1.00
.98	.98	.69	.78	.93	.93	.90	.51	.99	.16
1.01	.96	.56	.50	.68	.98	.71	.95	.55	.25
1.01	.81	.67	.30	.77	.96	1.07	.71	.70	.97
1.02	1.19	.58	1.06	.11	.96	1.07	.97	.74	.97
1.02	.92	.65	.50	.57	.96	.97	.97	.50	.96
1.02	.35	.96	.57	1.19	.98	.97	.96	.73	1.13
1.03	1.02	.74	.97	.57	.98	1.10	.90	.70	.99
1.03	1.01	.42	.89	.99	1.08	.96	1.13	.83	1.06

RUN 31 POINT 5 V=100. KNOTS		ALPHA=-4.0 DEG q = 30.8 PSF. CONFIGURATION FHW		ALPHA=-0 DEG q = 30.8 PSF. CONFIGURATION FHW		RUN 31 POINT 6 V=100 KNOTS		ALPHA=-0 DEG q = 30.8 PSF. CONFIGURATION FHW	
1.00	.99	.99	1.00	.96	1.01	.95	1.04	.97	1.03
1.00	1.00	.99	1.01	1.00	1.00	.97	1.00	1.07	.96
1.00	.98	1.03	.99	.97	1.04	.92	.99	.95	1.20
.97	.99	.97	.95	1.03	1.04	1.02	1.01	1.05	.76
1.00	1.00	1.02	1.08	1.04	1.04	.99	.95	.88	1.01
.96	1.01	1.00	1.05	.97	.97	.98	1.04	.96	1.20
1.04	1.01	1.01	1.02	.99	.97	1.03	1.22	.93	.92
.88	1.04	1.39	.89	1.23	1.11	1.09	1.03	1.15	.91
1.04	1.06	1.10	.96	.82	1.06	.75	1.10	.86	1.01
.97	.97	.87	.87	.95	1.06	.96	.99	.97	.94
1.00	1.00	.90	.70	.85	.93	1.03	.83	1.06	1.00
.98	.98	.69	.78	.93	.93	.90	.51	.99	.16
1.01	.96	.56	.50	.68	.98	.71	.95	.55	.25
1.01	.81	.67	.30	.77	.96	1.07	.71	.70	.97
1.02	1.19	.58	1.06	.11	.96	1.07	.97	.74	.97
1.02	.92	.65	.50	.57	.96	.97	.97	.50	.96
1.02	.35	.96	.57	1.19	.98	.97	.96	.73	1.13
1.03	1.02	.74	.97	.57	.98	1.10	.90	.70	.99
1.03	1.01	.42	.89	.99	1.08	.96	1.13	.83	1.06

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p \cdot \frac{q}{\rho}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SIDE
TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE OLD MODEL NOSE, LOOKING AFT.

TABLE X - Continued

BLM 31 POINT 7 ALPHA 4.0 DEG PSI 0 DEG V=100 KNOTS q = 30.8 PSF. CONFIGURATION FHB		BLM 31 POINT 8 ALPHA 6.0 DEG PSI 0 DEG V=100 KNOTS q = 30.8 PSF. CONFIGURATION FHB												
.95	1.01	.96	1.01	.95	1.03	.96	1.01	.95	.93	.97	.95	.92	1.10	1.17
	.96	1.02	.96	.92	.89	.92	.96	.95	.81	.94	.89	.87	6.3	.93
1.00	.95	.98	1.00	.87	.83	1.80	1.00	.87	.91	.88	.90	.75	1.01	.89
	.96	.98	.88	.86	.92	-2.8	.88	.87	.82	.90	.89	.85	-1.8	.98
.91	.94	.96	.98	.87	.77	1.30	.98	.87	.75	.84	.59	.79	.82	1.16
	.77	.93	.92	.77	1.06	.77	.92	.79	.99	.78	.79	.74	1.15	.86
.96	.94	.67	.96	.87	.76	.97	.96	1.07	.97	1.02	.39	.95	.81	.86
	.95	1.00	.81	.79	.62	.79	.81	.87	.94	1.04	.91	.67	.31	.91
1.00	1.0	-1.9	.74	.2	.85	.85	.62	.93	.97	2.8	.73	.25	5.8	6.4
	1.10	2.1	.66	-6.6	.76	.51	1.08	.81	.97	.8	.81	1.02	.67	1.02
1.00	.75	.64	.63	.65	.65	.99	.63	1.02	1.03	.92	.93	.87	1.25	.98
	.87	.93	.72	.52	.52	.85	.72	.95	.97	.90	.43	.57	.90	.90
1.01	.95	.14	.87	.53	.53	.70	.87	1.01	.79	.91	.54	.87	1.05	1.04
	1.04	1.06	RAKE MOUNTING POINT	.72	.85	.85	RAKE MOUNTING POINT	.91	.96	.91	RAKE MOUNTING POINT	.97	.95	.95
1.03	.95	.71	.93	1.2	.93	1.00	.93	.97	6.5	5.5	.84	.84	5.3	6.5
	.99	1.7	.96	3.2	.96	1.01	.96	.70	-1.7	3.6	.64	.64	-1.9	3.3
1.01	.99	.85	1.02	1.01	1.02	.98	.65	.97	.99	.71	.40	.78	1.00	1.01
	1.02	.64	.66	.94	.50	.94	.66	.94	.99	.59	.31	1.05	1.03	.95
.99	.98	.96	.51	.97	.98	.97	.51	.93	.95	1.04	.54	.86	1.00	.95

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW. RAKE MOUNTING POINT

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW. RAKE MOUNTING POINT

TABLE X - Continued

RUN 31 POINT 9 ALPHA=4.0 DEG PSI=4.0 DEG V=100 KNOTS q = 30.8 PSF. CONFIGURATION FHB1		RUN 31 POINT 10 ALPHA=0 DEG PSI=0 DEG V=100 KNOTS q = 30.8 PSF. CONFIGURATION FHB1	
.96	.97	.94	.92
1.00	.87	.98	.92
.96	.91	.93	.92
1.00	.85	1.01	1.01
.97	.86	.98	.74
.98	.90	1.06	.96
.97	.89	.83	.88
1.02	.87	.83	1.03
.87	.93	1.01	.94
.87	1.06	1.09	1.22
.87	.77	1.03	.83
.87	1.20	.86	1.03
.87	1.16	.94	.97
.87	1.03	.76	1.01
.73	.87	1.03	.74
.70	.81	1.03	.63
.70	.86	.75	.16
.79	.85	.70	.26
.69	.87	.80	.71
.61	.83	.89	.40
.66	.89	.29	.36
.74	.86	.38	.72
.49	.86	.36	.90
.56	.74	1.01	1.03
.77	.49	1.01	.85
.60	.80	.82	.65
.89	.83	.76	.54
.83	.83	1.00	1.00
.83	.83	.97	1.02
.83	.83	1.00	.98

RUN 31 POINT 9 ALPHA=4.0 DEG PSI=4.0 DEG V=100 KNOTS q = 30.8 PSF. CONFIGURATION FHB1		RUN 31 POINT 10 ALPHA=0 DEG PSI=0 DEG V=100 KNOTS q = 30.8 PSF. CONFIGURATION FHB1	
.96	.97	.94	.92
1.00	.87	.98	.92
.96	.91	.93	.92
1.00	.85	1.01	1.01
.97	.86	.98	.74
.98	.90	1.06	.96
.97	.89	.83	.88
1.02	.87	.83	1.03
.87	.93	1.01	.94
.87	1.06	1.09	1.22
.87	.77	1.03	.83
.87	1.16	.94	1.03
.87	1.03	.76	1.01
.73	.87	1.03	.74
.70	.81	1.03	.63
.70	.86	.75	.16
.79	.85	.70	.26
.69	.87	.80	.71
.61	.83	.89	.40
.66	.89	.29	.36
.74	.86	.38	.72
.49	.86	.36	.90
.56	.74	1.01	1.03
.77	.49	1.01	.85
.60	.80	.82	.65
.89	.83	.76	.54
.83	.83	1.00	1.00
.83	.83	.97	1.02
.83	.83	1.00	.98

RUN 31 POINT 9 ALPHA=4.0 DEG PSI=4.0 DEG V=100 KNOTS q = 30.8 PSF. CONFIGURATION FHB1		RUN 31 POINT 10 ALPHA=0 DEG PSI=0 DEG V=100 KNOTS q = 30.8 PSF. CONFIGURATION FHB1	
.96	.97	.94	.92
1.00	.87	.98	.92
.96	.91	.93	.92
1.00	.85	1.01	1.01
.97	.86	.98	.74
.98	.90	1.06	.96
.97	.89	.83	.88
1.02	.87	.83	1.03
.87	.93	1.01	.94
.87	1.06	1.09	1.22
.87	.77	1.03	.83
.87	1.16	.94	1.03
.87	1.03	.76	1.01
.73	.87	1.03	.74
.70	.81	1.03	.63
.70	.86	.75	.16
.79	.85	.70	.26
.69	.87	.80	.71
.61	.83	.89	.40
.66	.89	.29	.36
.74	.86	.38	.72
.49	.86	.36	.90
.56	.74	1.01	1.03
.77	.49	1.01	.85
.60	.80	.82	.65
.89	.83	.76	.54
.83	.83	1.00	1.00
.83	.83	.97	1.02
.83	.83	1.00	.98

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.



NOTE - TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

TABLE X - Continued

RUN 31 POINT 11 ALPHA = 0 DEG PSI = 4.0 DEG V = 100 KNOTS q = 30.8 PSF. CONFIGURATION FBI1				RUN 31 POINT 12 ALPHA = 4.0 DEG PSI = 4.0 DEG V = 100 KNOTS q = 30.8 PSF. CONFIGURATION FBI1			
.97	.98	.96	.99	.91	1.00	.96	.98
.99	.97	.98	.95	1.01	.98	1.01	.94
.98	.95	1.00	1.02	.94	.98	.93	.97
.96	.98	1.02	.88	1.02	1.02	1.00	1.00
.98	1.05	.99	1.04	.79	1.00	.98	.89
.94	.93	1.00	1.10	.98	.83	.97	.89
.86	.95	1.20	.87	.96	.92	1.05	1.11
.96	.88	1.19	.85	.91	.80	1.15	.87
.96	.90	.90	1.07	1.1	.74	.81	1.03
.99	.93	.89	.72	.96	.81	.49	1.04
.99	.62	.95	.28	.93	.60	.12	.96
.64	.64	.65	.70	1.12	.52	.24	.90
.92	.49	.71	.53	.97	.67	.35	.94
1.00	.35	.46	RAKE MOUNTING POINT	.99	1.20	RAKE MOUNTING POINT	.96
1.03	.58	.54	.92	.3	1.4	.86	4.0
1.01	.91	.31	1.03	1.00	.82	1.03	.98
	.96	.81	.92	.97	.96	1.03	1.02
	.68	.84	.91	.95	.55	.51	1.06
	.69	.53	.74	.97	.69	.44	1.02

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW.
MOUJEL NOSE, LOOKING AFT.



TABLE X - Continued

RUN 31 POINT 13 ALPHA=4.0 DEG PSI=-4.0 DEG V=100 KNOTS q=30.8 PSF, CONFIGURATION FHB										RUN 31 POINT 14 ALPHA=4.0 DEG PSI=-4.0 DEG V=100 KNOTS q=30.8 PSF, CONFIGURATION FHB									
.96	.96	.99	1.00	1.04	.96	.99	.95	.95	.95	.96	.96	.85	.97	.94	.96	.96	.94	.96	.96
.96	.96	.97	.93	.96	.96	.97	.86	.86	.86	1.01	1.00	1.02	.92	.81	.92	.81	.92	.92	.92
.99	.99	1.02	1.02	.98	.99	.99	3.9	3.9	.84	.93	.95	.80	.89	5.0	.89	5.0	1.03	.89	.91
.98	.98	.87	.85	.87	.93	.93	-5.3	-5.3	.83	.96	.95	.92	1.10	.68	.92	-5.3	.68	.92	.91
.79	.95	1.09	1.08	.97	1.06	1.06	.88	.93	.83	.96	.86	.78	.87	.96	.87	.96	.96	.87	1.11
.84	.78	.96	.85	.78	.88	.88	.88	.93	.84	1.05	.90	.98	1.09	1.02	.98	1.09	1.02	.98	.93
.98	.83	.94	1.07	.94	1.06	1.06	.93	.93	.94	.73	1.00	.80	.89	.93	.80	.89	.93	.80	.93
.57	.87	1.02	.94	.87	.96	.96	4.2	4.2	.87	.83	.87	.96	1.23	.98	.96	1.23	.98	.96	.93
.91	1.6	.87	.93	.93	.95	.88	5.6	5.6	.70	-7	.70	.59	.92	4.1	.92	4.1	1.07	.92	1.01
.99	-1.1	.74	1.08	1.08	.99	.72	-3.8	-3.8	.85	-9	.85	.26	1.03	-4.3	.26	-4.3	.99	.26	-2.6
.99	.92	.52	.24	.52	.88	.90	.88	1.01	.85	.74	.54	.79	.57	1.08	.79	1.08	.57	.96	.96
1.00	.98	.65	-.05	.65	.92	.95	.92	.99	.86	.87	1.08	.93	.84	.92	.93	.84	.92	.84	.92
	.82	.72	.82	.72	.91	.92	.91	.99	.42	.42	.33	.55	.64	1.01	.55	.64	1.01	.55	1.01
	.70	.52	.52	.52	.96	.99	.96	.95	.63	.63	.92	RAKE	1.06	.90	RAKE	1.06	.90	.92	1.04
1.00	2.1	.76	.62	.62	4.4	1.07	5.1	.95	2.6	2.6	.49	MOUNTING	.98	3.5	.49	3.5	.98	3.5	1.04
1.02	-6.7	.92	.91	.91	-4.9	1.00	-1.7	.94	-5.8	-5.8	.53	POINT	.99	-4.2	.53	-4.2	.99	-4.2	1.00
.98	.74	.66	.85	.66	.91	.89	.91	.93	.70	.70	.73	.96	.96	.99	.73	.96	.99	.96	1.03
	.71	.56	.67	.56	1.01	1.01	1.01	.94	1.01	1.01	.89	.81	1.00	1.05	.89	.81	1.00	.81	1.05
	.82	.85	.89	.82	.97	.99	.97	.93	.80	.80	.63	.42	1.11	1.02	.63	.42	1.11	1.02	1.02
	.74	.66	.89	.66	.91	.89	.91	.93	.75	.75	.63	.42	1.11	1.02	.63	.42	1.11	1.02	1.02

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{\rho V^2}{2}}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY
 TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW.
 MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

74.75 IN

TABLE X - Continued

RUN 31 POINT 15 ALPHA=0.0 DEG PSI=0.0 DEG V=100 KNOTS q=30.8 PSF. CONFIGURATION FHB1									
1.00	1.00	.96	1.00	.96	.92	.90	.97	1.03	1.03
.95	1.01	1.00	.94	.97	4.3	.91	1.04	1.01	1.03
	1.01	.96	.99	.92	3.8	.93	1.01	1.19	1.16
1.02	1.01	.95	.93	.91		.96	1.04	.95	
	.98	1.13	.98	.94		.97	.90	1.03	1.08
1.07	1.04	.97	.99	.97		.99	1.08	.90	
	1.03	1.01	.95	.99	.71	1.05	1.16	1.02	.82
.93	.90	1.05	1.06	1.06	9.8	.98	1.04	.99	.76
	.82	.91	.96	.76	9.4	.97	.97	.89	-10.2
.99	1.21	1.04	.92	.97	9.9	.70	1.3	1.02	1.26
	.95	1.03	.89	.79	.62	1.25	1.04	.92	.81
.99	.86	.59	.57	.83	.74		1.06	1.13	1.07
	1.03	.51	.73	.88	1.29	.83	.90	.98	.94
1.01	.95	.89	RAKE MOUNTING POINT	.78	2.7	.63	.39	.49	.59
	.99	.83	⊕	.84	4.4	.67	1.03	.56	-22.6
1.02	1.06	.91	.88	.88	3.9	.97	.95	.31	9.1
	.95	.98	.43	.56		.92	.67	.96	.92
.99	.96	.87	.44	.52		.99	.98	.78	.77
	1.01	.86	.67	.86	.81	.94	1.01	.82	.58

RUN 31 POINT 16 ALPHA=0.0 DEG PSI=0.0 DEG V=100 KNOTS q=30.8 PSF. CONFIGURATION FHB1									
.97	.96	1.03	.99	.99	.99	1.03	.97	1.03	1.03
.96	.95	.93	1.03	1.03			1.04	1.01	
	.98	1.01	.95	1.00			1.06	1.03	1.08
.95	.91	1.04	1.00	1.06			1.07	.90	
	1.04	.90	1.00	1.00			1.13	1.02	.82
1.15	1.05	1.16	1.01	1.01			.95	.76	-1.4
	.82	1.04	.99	.99			.89	1.26	.97
.95	1.00	.97	.89	1.02			1.09	.73	.92
	.99	.96	1.02	.92			.99	.81	1.09
.96	.97	1.04	.92	.92			.99	1.07	
	.97	1.06	1.13	.99			1.32	.94	.94
1.01	1.01	.90	.98	.98			RAKE MOUNTING POINT	.49	.59
	.95	.39	.39	.49			⊕	.56	-3.1
1.00	-1.1	1.04	1.03	.56			9.1	.72	-3.7
	1.5	.97	.95	.31					
1.02	.98	.67	.67	.96			.46	.92	1.14
.99	1.01	.98	.98	.71			.77	.77	
	.95	1.01	1.03	.82			.58	.58	1.00

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_i - P_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE MODEL NOSE, LOOKING AFT.

7.475 IN

7.475 IN

7.475 IN

7.475 IN

TABLE X - Continued

RUN 31 POINT 17 ALPHA = 0 DEG PSI = 4.0 DEG VE = 100. KNOTS q = 30.8 PSF. CONFIGURATION FHBI		RUN 31 POINT 18 ALPHA = 0 DEG PSI = 4.0 DEG VE = 100. KNOTS q = 30.8 PSF. CONFIGURATION FHBI	
1.00	.99	.98	.97
.95	.95	.99	.99
.98	.98	.99	.99
.95	1.03	1.01	1.00
.99	.97	1.03	.97
.99	.98	.88	.87
.99	.97	.86	.81
.99	.98	.87	.82
.99	.98	.88	.82
1.03	.98	.86	.81
1.03	.92	.86	.81
.98	.97	.86	.81

RUN 31 POINT 17 ALPHA = 0 DEG PSI = 4.0 DEG VE = 100. KNOTS q = 30.8 PSF. CONFIGURATION FHBI		RUN 31 POINT 18 ALPHA = 0 DEG PSI = 4.0 DEG VE = 100. KNOTS q = 30.8 PSF. CONFIGURATION FHBI	
1.00	.99	.98	.97
.95	.95	.99	.99
.98	.98	.99	.99
.95	1.03	1.01	1.00
.99	.97	1.03	.97
.99	.98	.88	.87
.99	.97	.86	.81
.99	.98	.87	.82
.99	.98	.88	.82
1.03	.98	.86	.81
1.03	.92	.86	.81
.98	.97	.86	.81

RUN 31 POINT 17 ALPHA = 0 DEG PSI = 4.0 DEG VE = 100. KNOTS q = 30.8 PSF. CONFIGURATION FHBI		RUN 31 POINT 18 ALPHA = 0 DEG PSI = 4.0 DEG VE = 100. KNOTS q = 30.8 PSF. CONFIGURATION FHBI	
1.00	.99	.98	.97
.95	.95	.99	.99
.98	.98	.99	.99
.95	1.03	1.01	1.00
.99	.97	1.03	.97
.99	.98	.88	.87
.99	.97	.86	.81
.99	.98	.87	.82
.99	.98	.88	.82
1.03	.98	.86	.81
1.03	.92	.86	.81
.98	.97	.86	.81

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p \cdot \frac{P - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT → 74.75 IN

TABLE X - Continued

RUN 31 POINT 19 ALPHA= 4.0 DEG PSI= 4.0 DEG V= 100 KNOTS q= 30.8 PSF. CONFIGURATION FHBI										RUN 31 POINT 20 ALPHA= 4.0 DEG PSI= 4.0 DEG V= 100 KNOTS q= 30.8 PSF. CONFIGURATION FHBI															
.97	.95	1.04	.91	.96	.97	.64	.93	.94	.98	.93	.96	.97	.83	.90	.94	.94	.94	.97	1.01	.80	.96	.97	.80		
.95	1.03	1.03	.96	.93	1.0	.87	1.03	1.03	.91	.97	.87	1.06	.85	1.03	1.03	.91	.97	.87	.97	.87	.97	.97	1.06		
.88	1.00	1.04	1.02	.94	1.05	1.06	.94	.86	.86	.91	.91	.92	.85	.80	.80	.86	.91	.91	.91	.91	.91	.91	.92		
1.01	1.04	.95	.98	.80	1.10	1.06	.80	.80	.81	.91	.77	.88	.94	.97	.97	.81	.91	.77	.77	.77	.77	.77	.88		
.97	.94	1.06	.67	.78	.59	.89	.78	.59	.96	.79	1.00	.91	.95	.94	1.00	1.00	.96	.79	1.00	1.00	1.00	1.00	.91	.95	
.96	.98	.85	1.08	.46	.54	.93	.46	.54	.41	.73	.90	.32	.92	.96	1.03	1.03	.41	.73	.90	.90	.90	.90	.90	.32	.92
.99	1.01	1.04	.41	.30	.57	.92	.30	.57	.02	.72	.59	.63	.92	.99	1.00	1.00	.02	.72	.59	.59	.59	.59	.59	.63	.92
1.02	.99	1.05	RAKE MOUNTING POINT	.44	.82	.90	.44	.82	.99	RAKE MOUNTING POINT	.60	.67	.92	1.01	.94	.95	RAKE MOUNTING POINT	.99	.60	.67	.67	.67	.67	.92	
.99	1.04	1.02	.98	.92	.76	.90	.92	.76	.95	.95	.40	.91	.99	1.01	3.4	.8	RAKE MOUNTING POINT	.95	.40	.91	.91	.91	.91	.99	
.98	.98	.98	.52	.70	.65	.99	.84	.51	1.01	.97	.55	.78	1.02	1.01	.8	1.01	1.01	.97	.55	.55	.55	.55	.78	1.02	
.99	1.01	.99	.80	.62	.98	.85	.62	.98	.99	.99	.42	.81	.97	1.00	.99	.99	.99	.99	.42	.42	.42	.42	.42	.81	.97
.98	1.00	.98	.52	.70	.65	.99	.37	.43	1.00	.96	.52	.57	1.02	1.01	1.00	1.00	.96	.64	.52	.52	.52	.52	.57	.99	1.02
									1.01	1.01	.49	.59		1.01	1.01	1.01	.98	.60	.49	.49	.49	.49	.59	1.02	

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

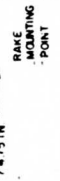


TABLE X - Continued

RUN 31 POINT 21 ALPHA= 4.0 DEG PSI= 4.0 DEG V= 100 KNOTS q= 30.8 PSF. CONFIGURATION FHB1				RUN 31 POINT 22 ALPHA= 0 DEG PSI=-4.0 DEG V= 100 KNOTS q= 30.8 PSF. CONFIGURATION FHB1			
.98	.98	.95	.98	.98	.89	1.01	1.31
.96	.99	1.00	.99	.96	1.03	.78	.83
.94	.93	1.00	1.00	.94	.89	.99	1.19
.85	.98	.94	.94	1.05	.95	.90	.81
.89	.84	.99	.88	.80	.99	1.05	.96
.98	.97	.67	.73	1.03	.86	1.01	1.38
.98	.77	.50	.89	1.02	.95	1.21	.88
.97	.97	.93	.89	.94	.89	.81	1.02
.93	.98	.72	.75	1.01	.67	.79	.87
.97	.97	.65	.89	1.03	.75	.66	.41
.97	1.02	.84	.63	.60	.99	.69	.57
1.04	1.02	.46	.54	.64	.57	1.26	.60
1.03	1.05	.71	.66	.78	.84	.64	.75
.98	.99	.64	.64	.56	.69	RAKE MOUNTING POINT	.49
3.1	4.1	-7.6	.76	-1.6	-18.7	RAKE MOUNTING POINT	.70
1.7	-6.9	-5.3	-5.8	-2.0	1.0	RAKE MOUNTING POINT	-2.8
.96	1.03	.58	1.07	.67	.33	1.06	1.02
.96	.98	.67	.46	.99	1.04	.56	.98
.98	.94	.90	1.04	.82	.62	.80	1.01
.95	1.00	.80	.53	.86	.61	.46	.97

RUN 31 POINT 21 ALPHA= 4.0 DEG PSI= 4.0 DEG V= 100 KNOTS q= 30.8 PSF. CONFIGURATION FHB1				RUN 31 POINT 22 ALPHA= 0 DEG PSI=-4.0 DEG V= 100 KNOTS q= 30.8 PSF. CONFIGURATION FHB1			
.98	.98	.95	.98	.98	.89	1.01	1.31
.96	.99	1.00	.99	.96	1.03	.78	.83
.94	.93	1.00	1.00	.94	.89	.99	1.19
.85	.98	.94	.94	1.05	.95	.90	.81
.89	.84	.99	.88	.80	.99	1.05	.96
.98	.97	.67	.73	1.03	.86	1.01	1.38
.98	.77	.50	.89	1.02	.95	1.21	.88
.97	.97	.93	.89	.94	.89	.81	1.02
.93	.98	.72	.75	1.01	.67	.79	.87
.97	.97	.65	.89	1.03	.75	.66	.41
.97	1.02	.84	.63	.60	.99	.69	.57
1.04	1.02	.46	.54	.64	.57	1.26	.60
1.03	1.05	.71	.66	.78	.84	.64	.75
.98	.99	.64	.64	.56	.69	RAKE MOUNTING POINT	.49
3.1	4.1	-7.6	.76	-1.6	-18.7	RAKE MOUNTING POINT	.70
1.7	-6.9	-5.3	-5.8	-2.0	1.0	RAKE MOUNTING POINT	-2.8
.96	1.03	.58	1.07	.67	.33	1.06	1.02
.96	.98	.67	.46	.99	1.04	.56	.98
.98	.94	.90	1.04	.82	.62	.80	1.01
.95	1.00	.80	.53	.86	.61	.46	.97

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH.

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY POINT TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

74.75 IN

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH.

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY POINT TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

74.75 IN

TABLE X - Continued

RUN 31 POINT 23 ALPHA = 0 DEG PSI = -0 DEG V = 100 KNOTS q = 30.8 PSF. CONFIGURATION FHBI		RUN 33 POINT 4 ALPHA = 0 DEG PSI = -0 DEG V = 125 KNOTS q = 48.1 PSF. CONFIGURATION FHBI							
.94	.96	.97	.97	.95	.97	.94	.96	.97	.99
.98	.97	.97	1.00	.97	.97	.98	.98	1.00	.98
.85	.94	.99	1.01	.97	.95	.97	.98	.97	1.00
1.02	1.00	.95	.69	.96	.96	1.01	.98	1.01	1.02
.94	.95	1.02	.83	.94	.96	.94	.98	.92	.98
.96	.94	.98	.81	.97	.97	.96	1.01	.90	.97
.98	1.03	1.03	.99	.85	.94	.98	.97	1.02	.79
.94	1.12	.98	1.14	.84	.84	.83	.92	.89	.8
.96	-4.0	-7.4	.66	-2.0	-5.8	.88	.87	.76	.8
.98	.86	-2.4	1.11	.67	-3.3	.88	.87	.76	.8
.96	1.03	.88	.51	1.00	.98	.79	.90	.98	.98
.96	.84	.85	.56	.90	.97	.79	.69	1.03	.94
.98	.86	.81	.34	.31	.64	.78	.31	.69	.94
1.03	.85	.79	.30	.20	.56	.92	.54	.57	1.00
1.02	.82	.43	.39	.79	.43	RAKE MOUNTING POINT	.59	.91	1.0
.98	-1.0	-5.9	.90	.39	-3.4	RAKE MOUNTING POINT	.82	.83	1.3
	-3.9	-4.4	1.09	.74	-4.4		.62	1.00	
1.02	.98	.54	.94	.93	.60	.44	.94	.93	1.00
.98	1.11	.96	1.00	.90	.79	.37	1.00	.99	1.02
	.98	.99	.75	.96	.95	.31	.93	.97	1.02

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{\frac{1}{2} \rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.



RAKE MOUNTING POINT

TABLE X - Continued

RUN 33 POINT 5 V=125 KNOTS		ALPHA= 0 DEG q= 48.1 PSF.		PSI= -0 DEG CONFIGURATION FHBF	
.99	.96	.98	.97	.98	1.03
.99	.99	.98	.99	1.02	1.00
.99	.97	1.04	1.00	-.6	1.00
.98	.98	.97	1.00	-2.7	1.01
.95	.95	.97	1.00	.92	.96
.98	.98	.94	.93		
.90	.87	1.02	1.03	1.05	.98
.95	.94	.87	.97	.95	.98
.97	1.06	.87	.89	1.07	.98
.97	1.06	.82	.91	.62	.98
.97	.91	.62	.83	.46	.96
1.02	.47	.82	.74	1.00	.99
	.23	.59	.74	.37	.99
1.01	1.06	.55	.30	.88	.99
1.10	.59	.55	1.05	.65	.99
	.86	.51	.62	5.8	.92
	.95	1.08	.65	.89	1.01
.99	1.02	.84	.80	.67	1.04
	.93	1.07	.33	.81	1.01
					1.05

RUN 33 POINT 6 V=125 KNOTS		ALPHA= 0 DEG q= 48.1 PSF.		PSI= -0 DEG CONFIGURATION FHBF	
.99	.98	.98	1.01	.98	.93
.99	.99	1.00	1.00	.98	.93
.99	1.00	.97	.99	.95	.93
1.00	.99	.99	1.01	.97	.93
.99	1.00	1.03	1.00	.99	.95
.95	.95	1.00	.95	.95	.90
.99	1.02	.95	.91	.87	.98
.99	.71	.84	.92	.89	.99
.99	-.7	-4.3	.90	1.00	.97
	.3	.2	.79	.59	1.04
1.03	.77	.98	.86	.48	.96
.98	.91	.59	.96	.60	.71
	.58	.15	.63	.69	.93
1.04	.42	.77	RAKE MOUNTING POINT	.69	.97
.99	.7	-1.5		-2.4	.75
	-2.6	-1.9		.62	.97
	1.08	.62		1.02	2.1
.99	1.03	1.04	.47	.91	.95
.97	.95	1.02	.47	.88	.87
	.94	.92	.41	.87	.95

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH.

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT → 7.475 IN

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH.

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT → 7.475 IN

TABLE X - Continued

RUN 33 POINT 7 ALPHA= 4.0 DEG PSI= -0 DEG V= 125 KNOTS q= 48.1 PSF. CONFIGURATION FHBI		RUN 33 POINT 8 ALPHA= 4.0 DEG PSI= 7.0 DEG V= 125 KNOTS q= 48.1 PSF. CONFIGURATION FHBI	
.98	.96	.98	.95
1.00	.97	.97	.96
1.01	.94	.93	.94
1.01	.96	.97	.92
1.00	.98	.95	.97
1.00	.97	.97	.96
1.01	.92	1.00	.90
1.01	.95	1.00	.96
1.02	.99	.92	.79
1.00	.96	.96	.71
1.00	.97	.84	.77
1.00	.80	.99	.63
1.00	.85	.43	.81
1.00	.76	.91	.77
1.00	.36	.98	.69
1.00	.53	.57	.42
1.00	.71	2.4	-1.6
1.05	.92	-4.2	-3
1.01	1.00	1.00	.22
1.00	.68	.98	1.01
1.00	.92	1.01	.75
1.00	.67	.95	.88
1.00	.84	.99	.92
	.62	.89	.92
	.84	.94	.94

RUN 33 POINT 7 ALPHA= 4.0 DEG PSI= -0 DEG V= 125 KNOTS q= 48.1 PSF. CONFIGURATION FHBI		RUN 33 POINT 8 ALPHA= 4.0 DEG PSI= 7.0 DEG V= 125 KNOTS q= 48.1 PSF. CONFIGURATION FHBI	
.98	.93	.95	.92
1.00	.95	.94	.92
1.01	.80	.93	1.13
1.01	.93	.94	.92
1.00	1.00	.98	.97
1.00	.99	.89	.96
1.01	.91	.70	.96
1.01	1.02	.79	.96
1.00	.99	.71	.68
1.02	.84	.83	.90
1.00	.83	.94	.65
1.00	.07	.82	.48
1.00	1.08	1.01	.57
1.00	.71	.29	.97
1.05	.89	.33	.86
1.01	1.00	1.5	.93
1.01	.98	-2.2	2.2
1.01	.68	.96	.98
1.01	.92	.93	.96
1.00	.97	.99	.95
1.00	.94	.37	.92

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2}\rho V^2}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.

UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH

LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.

NOTE- TABLE IS SEEN FROM THE FLOW

MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

7.475 IN

TABLE X - Continued

RUN 33 POINT 9 ALPHA=4.0 DEG PSI=-4.0 DEG V=125 KNOTS q=48.1 PSF. CONFIGURATION FHB1				RUN 33 POINT 10 ALPHA=-0 DEG PSI=-4.0 DEG V=125 KNOTS q=48.1 PSF. CONFIGURATION FHB1			
.97	.98	1.00	.97	.98	.97	.98	.96
.97	.97	.93	1.00	.96	.96	.97	.96
.99	.99	.97	.97	.92	.96	.93	.94
.97	.97	1.02	.97	.95	.96	.98	.97
.97	.97	.99	.99	.93	.96	.99	.98
.78	.78	1.00	.86	.92	1.00	.93	.94
.81	.81	1.02	1.02	.98	.98	.99	.96
.84	.84	1.03	.94	1.07	.95	1.02	.97
.98	.98	1.06	.84	.26	.98	.81	.99
.98	.98	1.07	1.33	.67	.94	.51	.96
.95	.95	.19	.54	.73	.98	.44	.99
.64	.64	.53	.26	.77	.94	.73	.89
.53	.53	.99	.99	.98	.98	.66	.83
.98	.98	.73	.73	1.13	.99	.91	.95
.98	.98	.63	.63	1.02	1.01	.96	.99
.69	.69	.89	.89	.95	.99	.96	.94
1.01	1.01	.52	.52	1.16	.98	1.00	.96
.98	.98	1.05	1.05	.83	.97	.80	.99
.98	.98	.71	.71	.53	.94	.22	.96
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2}\rho V^2}$				NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_0}{\frac{1}{2}\rho V^2}$			
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.				NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY TO RIGHT LOOKING AFT.			
NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.				NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.			

TABLE X - Continued

RUN 33 POINT 11 V=125 KNOTS		ALPHA= .0 DEG PSI=-2.0 DEG q= 48.1 PSF. CONFIGURATION FHBY		RUN 33 POINT 12 V=125 KNOTS		ALPHA= .0 DEG PSI=-2.0 DEG q= 48.1 PSF. CONFIGURATION FHBY	
.99	.95	.97	.96	.97	.94	.97	.94
.94	.97	.97	1.00	.95	.98	.98	.98
.96	.98	.98	.94	.98	.95	.95	.95
.94	.97	.97	1.03	.98	.94	.94	.94
.95	1.02	.98	.89	.92	.92	.96	.91
.92	.97	.89	.92	.94	.98	.93	1.00
.92	1.01	.94	.99	.93	.97	1.00	1.08
.84	.92	.90	.90	.91	.92	1.06	.98
.96	.84	.95	.95	1.04	.91	.86	1.3
.75	.89	.69	.47	.99	.57	.52	-5.6
.68	1.13	.49	.69	.74	.82	.52	.67
.76	.90	.34	.65	.87	.34	.20	1.01
.50	1.14	.56	1.00	1.05	.66	1.01	1.00
.73	.58	.48	.48	.93	.76	.77	.98
.72	.48	1.04	1.04	.93	.52	.95	-1.2
.77	.51	.92	.92	.98	.42	1.09	.98
.97	.76	.98	.90	.95	.50	.99	.98
1.03	.81	.39	.94	.94	.59	1.04	1.00
1.04	.49	1.00	.61	1.03	.76	1.01	.97
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$				NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$			
NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.				NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.			
NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.				NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.			

TABLE X - Continued

RUN 33 POINT 13 ALPHA = .0 DEG PSI = -2.0 DEG V = 125 KNOTS q = 48.1 PSF. CONFIGURATION FMBF				RUN 33 POINT 14 ALPHA = 4.0 DEG PSI = -2.0 DEG V = 125 KNOTS q = 48.1 PSF. CONFIGURATION FMBF			
.98	.96	.95	1.04	.94	.98	.97	.94
.99	.98	.95	.98	.95	.99	1.00	.98
.99	.96	.95	.99	.95	.99	.96	.92
.99	.97	.94	.97	.94	1.00	.96	.94
.97	.96	.89	.91	.98	.81	.97	1.01
.94	1.02	.94	.95	.92	.90	1.00	.93
.94	.70	.95	1.00	.92	1.39	.89	1.07
1.00	.98	1.02	1.01	.91	.90	.91	.99
1.00	.96	.95	.99	.89	1.19	.57	.78
1.00	1.16	.94	1.06	.72	.89	.71	.31
.94	.78	.70	.70	.35	.34	.69	.32
1.01	.99	.74	.78	.20	1.07	.40	1.28
1.01	.81	.75	.45	.48	.27	.88	.62
1.04	.11	.72	.94	RAKE MOUNTING POINT	.80	.72	RAKE MOUNTING POINT
1.01	.78	.34	.88	.79	3.8	.4	3.7
.99	1.14	.87	1.01	.50	-6.2	.4	-2.6
	.96	.80	1.05	.43	1.00	.69	.94
					1.00	1.01	.98
					1.01	.43	.93
					1.00	.53	.95

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P - P_\infty}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.

NOTE - TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.



TABLE X - Continued

RUN 33 POINT 15 ALPHA= 4.0 DEG PSI=-2.0 DEG V=125 KNOTS q = 46.1 PSF. CONFIGURATION FHBF		RUN 33 POINT 16 ALPHA= 4.0 DEG PSI=-2.0 DEG V=125 KNOTS q = 48.1 PSF. CONFIGURATION FHBF	
.96	.97	.92	.93
.99	.95	1.01	1.02
.97	.98	.91	.93
.95	.94	.94	.95
.95	.87	.97	.98
.85	.97	1.07	.98
.87	.71	1.05	1.04
.83	.85	.90	.95
.89	1.03	.78	.99
.91	.85	.49	.96
.60	.77	.46	.91
.59	.32	1.12	1.05
1.07	.55	1.09	.73
.93	1.00	MAKE MOUNTING POINT	MAKE MOUNTING POINT
.90	.60	.96	1.03
.92	.94	.93	.97
.98	1.16	1.10	.91
1.01	.52	.92	.96
.97	1.07	.15	1.00
NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_\infty}{q}$		NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P-P_\infty}{q}$	
NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.		NUMBERS IN BOXES INDICATE FLOW DIRECTION, IN DEGREES. UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.	
NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.		NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.	

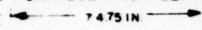
TABLE X - Concluded

RUN 33 POINT 21 ALPHA= .0 DEG PSI= -.0 DEG
 V=125. KNOTS q = 48.1 PSF. CONFIGURATION FHBf

.98	.97	.95	.96	.92	.99	.99
1.00	.99	.96	.97	.99	1.00	.95
	.98	.98	.99	.98	-5	1.00
	.98	.96	.96	1.00	-2.7	.99
.96	.96	.94	.99	.96	1.03	1.00
	1.02	1.00	.97	.95	.93	
.95	.88	.97	.87	.96	.90	1.08
	.92	.94	1.11	.95	.92	
.98	-1.2	-4.0	.90	1.04	-1.6	1.1
	.9	-.0	.78	.71	-5.6	-.8
	.80	1.16	1.26	.37	.92	
.98	.89	.72	.80	.22	1.40	1.04
	.53	.39	.85	.70	.74	
.97	.76	.57	.74	.67	.96	
1.03	.6	-3.7	RAKE MOUNTING POINT	.93	1.03	1.1
	-3.1	-5.7	⊕	-3.4	.67	2.7
	1.13	.40	.66	4.3	.93	.98
1.00	.95	.55	.92	.92	.92	.99
	.91	.75	.98	.94		
	.94	.81	.60	1.01		
	.87	.97				
.99	1.03	1.00				1.02

NUMBERS IN TABLE ARE PRESSURE COEFFICIENTS $C_p = \frac{P_t - P_s}{q}$

NUMBERS IN BOXES INDICATE FLOW DIRECTION IN DEGREES.
 UPPER = VERTICAL DISPLACEMENT, POSITIVE FOR UPWASH
 LOWER = HORIZONTAL DISPLACEMENT, POSITIVE FOR FLOW FROM LEFT SURVEY PLANE TO RIGHT LOOKING AFT.



NOTE- TABLE IS SEEN FROM THE FLOW MODEL NOSE, LOOKING AFT.

RAKE MOUNTING POINT

APPENDIX III

ROTOR PERFORMANCE AND OPERATING CONDITIONAL DATA TABLES

TABLE XI. WIND TUNNEL OPERATING CONDITIONS AND SHAFT FORCE AND MOMENT COEFFICIENT-SOLIDITY RATIOS AT THREE FORWARD SPEEDS AND VARIOUS FUSELAGE ATTITUDES, CONFIGURATION FH

RUN	P	T	CONF	V FT/SEC	q LB/FT ² DEG	γ DEG	α DEG	β _c DEG	β _s DEG	β ₁₀ DEG	C _L /σ	C _D /σ	C _L /σ	C _D /σ	C _L /σ	C _D /σ	C _{PM} /σ	C _{MM} /σ
87-7	FH			103.0	11.9	0	0	0	0	0	.0018	.0007	.00004	.00000	.00007	.00000	.00000	.00001
87-8	FH			103.0	11.9	-8	-8	-8	-8	-8	.0030	.0044	.00000	.00024	.00017	.00017	.00017	.00017
87-9	FH			103.0	11.9	4	4	4	4	4	.0009	.0024	.00003	.00004	.00024	.00004	.00004	.00009
87-10	FH			103.0	11.9	4	4	4	4	4	.0010	.0010	.00006	.00005	.00010	.00005	.00005	.00004
87-11	FH			103.0	11.9	8	8	8	8	8	.0015	.0017	.00007	.00004	.00017	.00007	.00007	.00007
87-12	FH			103.0	11.9	-8	-8	-8	-8	-8	.0029	.0063	.00000	.00004	.00031	.00004	.00031	.00022
87-13	FH			103.0	11.9	4	4	4	4	4	.0017	.0003	.00003	.00012	.00012	.00012	.00012	.00022
87-14	FH			103.0	11.9	-4	-4	-4	-4	-4	.0010	.0003	.00006	.00006	.00011	.00006	.00011	.00011
87-15	FH			103.0	11.9	-8	-8	-8	-8	-8	.0013	.0002	.00000	.00006	.00011	.00006	.00011	.00011
87-16	FH			103.0	11.9	4	4	4	4	4	.0009	.0008	.00002	.00002	.00002	.00002	.00002	.00003
87-17	FH			103.0	11.9	-4	-4	-4	-4	-4	.0010	.0003	.00005	.00005	.00016	.00005	.00016	.00016
87-18	FH			103.0	11.9	0	0	0	0	0	.0007	.0003	.00005	.00000	.00016	.00005	.00016	.00016
87-19	FH			103.0	11.9	0	0	0	0	0	.0007	.0003	.00004	.00001	.00010	.00004	.00010	.00010
87-20	FH			103.0	11.9	0	0	0	0	0	.0008	.0003	.00005	.00001	.00012	.00005	.00012	.00012
87-21	FH			103.0	11.9	0	0	0	0	0	.0008	.0003	.00004	.00001	.00012	.00004	.00012	.00012
87-22	FH			170.2	32.3	0	0	0	0	0	.0054	.0004	.00007	.00000	.00001	.00007	.00001	.00025
87-23	FH			170.2	32.3	8	8	8	8	8	.0019	.0004	.00004	.00000	.00007	.00004	.00007	.00001
87-24	FH			170.2	32.3	-8	-8	-8	-8	-8	.0023	.0021	.00002	.00002	.00007	.00002	.00007	.00007
87-25	FH			170.2	32.3	4	4	4	4	4	.0019	.0003	.00004	.00002	.00007	.00004	.00007	.00005
87-26	FH			170.2	32.3	4	4	4	4	4	.0018	.0011	.00009	.00016	.00011	.00009	.00016	.00005
87-27	FH			170.2	32.3	8	8	8	8	8	.0028	.0036	.00010	.00008	.00015	.00010	.00008	.00015
87-28	FH			170.2	32.3	-8	-8	-8	-8	-8	.0016	.0002	.00001	.00013	.00007	.00001	.00013	.00004
87-29	FH			170.2	32.3	4	4	4	4	4	.0021	.0003	.00009	.00009	.00013	.00009	.00013	.00013
87-30	FH			170.2	32.3	4	4	4	4	4	.0023	.0003	.00008	.00009	.00013	.00008	.00013	.00013
87-31	FH			170.2	32.3	-4	-4	-4	-4	-4	.0020	.0004	.00000	.00022	.00015	.00000	.00022	.00043
87-32	FH			170.2	32.3	0	0	0	0	0	.0019	.0007	.00007	.00004	.00013	.00007	.00013	.00013
87-33	FH			170.2	32.3	0	0	0	0	0	.0019	.0006	.00006	.00008	.00013	.00006	.00008	.00013
87-34	FH			170.2	32.3	0	0	0	0	0	.0021	.0006	.00005	.00004	.00013	.00005	.00004	.00013
87-35	FH			255.8	72.0	0	0	0	0	0	.0021	.0011	.00005	.00003	.00011	.00005	.00011	.00011
87-36	FH			255.8	72.0	-4	-4	-4	-4	-4	.0019	.0007	.00014	.00010	.00011	.00014	.00010	.00011
87-37	FH			255.8	72.0	4	4	4	4	4	.0024	.0008	.00011	.00008	.00011	.00011	.00008	.00011
87-38	FH			255.8	72.0	2	2	2	2	2	.0020	.0005	.00013	.00006	.00010	.00013	.00006	.00010
87-39	FH			255.8	72.0	4	4	4	4	4	.0022	.0000	.00020	.00023	.00020	.00020	.00023	.00005
87-40	FH			255.8	72.0	-2	-2	-2	-2	-2	.0022	.0000	.00020	.00023	.00020	.00020	.00023	.00005
87-41	FH			255.8	72.0	2	2	2	2	2	.0022	.0000	.00020	.00023	.00020	.00020	.00023	.00005
87-42	FH			255.8	72.0	4	4	4	4	4	.0022	.0000	.00020	.00023	.00020	.00020	.00023	.00005
87-43	FH			255.8	72.0	2	2	2	2	2	.0022	.0000	.00020	.00023	.00020	.00020	.00023	.00005
87-44	FH			255.8	72.0	0	0	0	0	0	.0022	.0000	.00020	.00023	.00020	.00020	.00023	.00005
87-45	FH			255.8	72.0	0	0	0	0	0	.0022	.0000	.00020	.00023	.00020	.00020	.00023	.00005
87-46	FH			255.8	72.0	-2	-2	-2	-2	-2	.0022	.0000	.00020	.00023	.00020	.00020	.00023	.00005
87-47	FH			255.8	72.0	0	0	0	0	0	.0022	.0000	.00020	.00023	.00020	.00020	.00023	.00005

TABLE XII. WIND TUNNEL OPERATING CONDITIONS AND SHAFT FORCE AND
MOMENT COEFFICIENT-SOLIDITY RATIOS AT THREE FORWARD
SPEEDS AND VARIOUS FUSELAGE ATTITUDES, CONFIGURATION FHC

U N	P	CONF	V FT/SEC	q LB/FT ²	α DEG	β DEG	μ	ϵ_c DEG	ϵ_s DEG	ϵ_{10} DEG	δ_{10} DEG	A_{10} DEG	B_{10} DEG	C_{10} /r	C_{10} /r	C_{10} /r	C_{10} /r	C_{10} /r	C_{10} /r	C_{10} /r
88-5	FHC	103.0	12.0	0	0	0	.295	0	0	0	0	0	0	.0035	.00203	-.00121	-.00004	.00006	-.00043	
88-6	FHC	103.0	12.0	-8	0	0	.292	-8.0	0	0	0	0	0	.0036	.00161	-.00024	-.00000	.00012	-.00006	
88-7	FHC	103.0	12.0	-4	0	0	.294	-4.0	0	0	0	0	0	.0032	.00194	-.00001	-.00000	.00011	-.00003	
88-8	FHC	103.0	12.0	4	0	0	.294	4.0	0	0	0	0	0	.0034	.00211	-.00012	-.00006	.00004	-.00008	
88-9	FHC	103.0	12.0	8	0	0	.292	8.0	0	0	0	0	0	.0035	.00218	-.00044	-.00006	.00004	-.00021	
88-10	FHC	103.0	12.0	4	4	4	.294	4.0	0	0	0	0	0	.0031	.00219	-.00015	-.00006	.00009	-.00009	
88-11	FHC	103.0	12.0	-4	4	4	.294	-4.0	0	0	0	0	0	.0032	.00198	-.00041	-.00003	.00007	-.00023	
88-12	FHC	103.0	12.0	-8	4	4	.292	-8.0	0	0	0	0	0	.0030	.00165	-.00049	-.00001	.00005	-.00020	
88-13	FHC	103.0	12.0	-8	-4	-4	.292	-8.0	0	0	0	0	0	.0030	.00168	-.00015	-.00000	.00011	-.00017	
88-14	FHC	103.0	12.0	-4	-4	-4	.294	-4.0	0	0	0	0	0	.0026	.00190	-.00022	-.00002	.00007	-.00020	
88-15	FHC	103.0	12.0	4	-4	-4	.294	4.0	0	0	0	0	0	.0030	.00210	-.00049	-.00005	.00002	-.00030	
88-16	FHC	103.0	12.0	0	8	8	.295	0	0	0	0	0	0	.0028	.00222	-.00032	-.00004	.00009	-.00023	
88-17	FHC	103.0	12.0	0	4	4	.295	0	0	0	0	0	0	.0027	.00220	-.00018	-.00003	.00008	-.00011	
88-18	FHC	103.0	12.0	0	-8	-8	.295	0	0	0	0	0	0	.0028	.00221	-.00031	-.00001	.00008	-.00024	
88-19	FHC	170.2	32.4	0	0	0	.504	0	0	0	0	0	0	.0056	.00608	-.00024	-.00005	.00000	-.00016	
88-20	FHC	170.2	32.4	-8	0	0	.499	-8.0	0	0	0	0	0	.0054	.00557	-.00073	-.00005	.00002	-.00024	
88-21	FHC	170.2	32.4	-4	0	0	.503	-4.0	0	0	0	0	0	.0053	.00562	-.00012	-.00001	.00012	-.00000	
88-22	FHC	170.2	32.4	4	0	0	.503	4.0	0	0	0	0	0	.0059	.00598	-.00031	-.00006	-.00013	-.00020	
88-23	FHC	170.2	32.4	8	0	0	.499	8.0	0	0	0	0	0	.0061	.00616	-.00050	-.00006	-.00010	-.00029	
88-24	FHC	170.2	32.4	4	4	4	.503	4.0	0	0	0	0	0	.0056	.00617	-.00028	-.00007	-.00003	-.00024	
88-25	FHC	170.2	32.4	-4	4	4	.503	-4.0	0	0	0	0	0	.0051	.00587	-.00071	-.00002	.00001	-.00046	
88-26	FHC	170.2	32.4	-4	-4	-4	.503	-4.0	0	0	0	0	0	.0050	.00591	-.00048	-.00002	.00001	-.00046	
88-27	FHC	170.2	32.4	0	-4	-4	.503	4.0	0	0	0	0	0	.0058	.00610	-.00099	-.00005	-.00010	-.00067	
88-28	FHC	170.2	32.4	0	8	8	.504	0	0	0	0	0	0	.0051	.00619	-.00105	-.00005	-.00006	-.00047	
88-29	FHC	170.2	32.4	0	4	4	.504	0	0	0	0	0	0	.0050	.00612	-.00047	-.00004	-.00002	-.00033	
88-30	FHC	170.2	32.4	0	-4	-4	.504	0	0	0	0	0	0	.0053	.00601	-.00080	-.00004	-.00004	-.00060	
88-31	FHC	170.2	32.4	0	0	0	.504	0	0	0	0	0	0	.0054	.00609	-.00139	-.00003	-.00003	-.00060	
88-32	FHC	170.2	32.4	0	-8	-8	.504	0	0	0	0	0	0	.0054	.00609	-.00044	-.00003	-.00003	-.00060	
88-33	FHC	255.8	72.4	0	0	1.026	1.026	0	0	0	0	0	0	.0205	.0205	-.00004	-.00008	-.00032	-.00011	
88-34	FHC	255.8	72.4	-4	0	1.026	1.026	-4.0	0	0	0	0	0	.0192	.02516	-.00038	-.00008	-.00005	-.00007	
88-35	FHC	255.8	72.4	-2	0	1.026	1.026	-2.0	0	0	0	0	0	.0197	.02550	-.00025	-.00011	-.00006	-.00001	
88-36	FHC	255.8	72.4	2	0	1.026	1.026	2.0	0	0	0	0	0	.0230	.02533	-.00023	-.00014	-.00002	-.00021	
88-37	FHC	255.8	72.4	4	0	1.026	1.026	4.0	0	0	0	0	0	.0210	.02568	-.00058	-.00016	-.00039	-.00033	
88-38	FHC	255.8	72.4	2	2	1.026	1.026	4.0	0	0	0	0	0	.0209	.02572	-.00079	-.00016	-.00045	-.00064	
88-39	FHC	255.8	72.4	4	2	1.026	1.026	4.0	0	0	0	0	0	.0208	.02563	-.00085	-.00015	-.00032	-.00066	
88-40	FHC	255.8	72.4	-2	2	1.026	1.026	-2.0	0	0	0	0	0	.0196	.02539	-.00144	-.00015	-.00009	-.00065	
88-41	FHC	255.8	72.4	2	-2	1.026	1.026	2.0	0	0	0	0	0	.0204	.02540	-.00137	-.00015	-.00036	-.00109	
88-42	FHC	255.8	72.4	0	4	1.026	1.026	0	0	0	0	0	0	.0195	.02502	-.00253	-.00018	-.00021	-.00177	
88-43	FHC	255.8	72.4	0	-2	1.026	1.026	0	0	0	0	0	0	.0194	.02536	-.00128	-.00017	-.00018	-.00236	
88-44	FHC	255.8	72.4	0	2	1.026	1.026	0	0	0	0	0	0	.0194	.02545	-.00120	-.00015	-.00024	-.00100	
88-45	FHC	255.8	72.4	0	-4	1.026	1.026	0	0	0	0	0	0	.0194	.02560	-.00231	-.00017	-.00022	-.00364	

TABLE XIII. WIND TUNNEL OPERATING CONDITIONS AND SHAFT FORCE AND
MOMENT COEFFICIENT-SOLIDITY RATIOS AT THREE FORWARD
SPEEDS AND VARIOUS FUSELAGE ATTITUDES, CONFIGURATIONS FHf

U N	f	CONF	V FT/SEC	q LB/FT ²	α DEG	β DEG	γ DEG	μ	α _c DEG	α _e DEG	α _h DEG	α _h DEG	α _h DEG	α _h DEG	C _L /σ	C _D /σ	C _y /σ	C _z /σ	C _m /σ	C _m /σ	C _{mm} /σ
89-7		FHf	103.0	11.9	0.	0.	0.	.295	0	0	0	0	0	0	.0034	.00097	-.00020	.00018	-.00033	-.00008	
89-8		FHf	103.0	11.9	-8.	0.	0.	.292	-8.0	0	0	0	0	0	.0020	.00050	.00023	.00013	-.00002	-.00009	
89-9		FHf	103.0	11.9	-4.	0.	0.	.294	-4.0	0	0	0	0	0	.0034	.00074	.00022	.00015	-.00015	-.00009	
89-10		FHf	103.0	11.9	4.	0.	0.	.294	4.0	0	0	0	0	0	.0036	.00121	-.00017	.00016	-.00043	.00008	
89-11		FHf	103.0	11.9	8.	0.	0.	.292	8.0	0	0	0	0	0	.0035	.00134	-.00034	.00017	-.00051	.00020	
89-12		FHf	103.0	11.9	4.	4.	4.	.294	4.0	0	0	0	0	0	.0036	.00122	-.00002	.00018	-.00042	.00000	
89-13		FHf	103.0	11.9	-4.	4.	4.	.294	-4.0	0	0	0	0	0	.0029	.00067	.00033	.00014	-.00022	-.00018	
89-14		FHf	103.0	11.9	-8.	4.	4.	.292	-8.0	0	0	0	0	0	.0036	.00061	.00056	.00014	-.00001	-.00029	
89-15		FHf	103.0	11.9	-8.	-4.	-4.	.292	-8.0	0	0	0	0	0	.0024	.00061	.00056	.00014	-.00001	-.00029	
89-16		FHf	103.0	11.9	-4.	-4.	-4.	.294	-4.0	0	0	0	0	0	.0032	.00044	.00041	.00014	-.00031	.00006	
89-17		FHf	103.0	11.9	4.	-4.	-4.	.294	4.0	0	0	0	0	0	.0041	.00100	-.00004	.00016	-.00056	.00005	
89-18		FHf	103.0	11.9	0.	8.	8.	.295	0	0	0	0	0	0	.0035	.00081	.00023	.00016	-.00042	-.00015	
89-19		FHf	103.0	11.9	0.	4.	4.	.295	0	0	0	0	0	0	.0036	.00075	.00003	.00014	-.00042	-.00005	
89-20		FHf	103.0	11.9	0.	-4.	-4.	.295	0	0	0	0	0	0	.0036	.00071	-.00016	.00016	-.00043	.00012	
89-21		FHf	103.0	11.9	0.	-8.	-8.	.295	0	0	0	0	0	0	.0276	.00478	-.00032	.00014	-.00046	.00027	
89-22		FHf	103.0	11.9	0.	0.	0.	.295	0	0	0	0	0	0	.0278	.00098	-.00004	.00013	-.00047	.00004	
89-23		FHf	170.2	32.4	0.	0.	0.	.504	0	0	0	0	0	0	.0113	.00431	.00018	.00018	-.00087	-.00015	
89-24		FHf	170.2	32.4	-8.	0.	0.	.500	-8.0	0	0	0	0	0	.0066	.00347	.00066	.00015	-.00019	-.00033	
89-25		FHf	170.2	32.4	-4.	0.	0.	.503	-4.0	0	0	0	0	0	.0101	.00380	.00035	.00017	-.00058	-.00020	
89-26		FHf	170.2	32.4	4.	0.	0.	.503	4.0	0	0	0	0	0	.0121	.00450	.00007	.00020	-.00113	-.00009	
89-27		FHf	170.2	32.4	8.	0.	0.	.499	8.0	0	0	0	0	0	.0125	.00487	-.00004	.00021	-.00134	-.00000	
89-28		FHf	170.2	32.4	4.	4.	4.	.503	4.0	0	0	0	0	0	.0117	.00434	.00046	.00021	-.00119	-.00033	
89-29		FHf	170.2	32.4	-4.	4.	4.	.503	-4.0	0	0	0	0	0	.0097	.00394	.00098	.00017	-.00060	.00065	
89-30		FHf	170.2	32.4	-4.	-4.	-4.	.503	-4.0	0	0	0	0	0	.0103	.00350	-.00005	.00016	-.00063	.00009	
89-31		FHf	170.2	32.4	4.	-8.	-8.	.503	4.0	0	0	0	0	0	.0124	.00413	-.00010	.00020	-.00122	.00009	
89-32		FHf	170.2	32.4	0.	4.	4.	.504	0	0	0	0	0	0	.0107	.00390	.00115	.00019	-.00094	-.00084	
89-33		FHf	170.2	32.4	0.	4.	4.	.504	0	0	0	0	0	0	.0111	.00410	.00072	.00019	-.00088	-.00052	
89-34		FHf	170.2	32.4	0.	-4.	-4.	.504	0	0	0	0	0	0	.0119	.02020	.00000	.00019	.00541	.00003	
89-35		FHf	170.2	32.4	0.	-8.	-8.	.504	0	0	0	0	0	0	.0120	.02168	-.00042	.00021	.00606	.00039	
89-36		FHf	170.2	32.4	0.	0.	0.	.504	0	0	0	0	0	0	.0125	.02073	.00025	.00021	.00541	-.00012	
89-37		FHf	213.7	50.8	0.	0.	0.	1.014	0	0	0	0	0	0	.0477	.01832	.00184	.00049	-.00329	.00121	
89-38		FHf	213.7	50.8	4.	0.	0.	1.012	4.0	0	0	0	0	0	.0516	.01997	.00132	.00053	-.00424	-.00089	
89-39		FHf	213.7	50.8	4.	4.	4.	1.012	4.0	0	0	0	0	0	.0502	.01925	.00250	.00053	-.00447	-.00193	
89-40		FHf	213.7	50.8	0.	4.	4.	1.014	0	0	0	0	0	0	.0475	.01808	.00364	.00050	-.00330	.00258	
89-41		FHf	213.7	50.8	0.	2.	2.	1.014	0	0	0	0	0	0	.0480	.01853	.00270	.00047	-.00313	-.00184	
89-42		FHf	213.7	50.8	4.	4.	4.	1.012	4.0	0	0	0	0	0	.0523	.01927	.00219	.00055	-.00452	-.00152	
89-43		FHf	213.7	50.8	0.	-2.	-2.	1.014	0	0	0	0	0	0	.0482	.01770	.00103	.00045	-.00340	-.00059	
89-44		FHf	213.7	50.8	4.	-2.	-2.	1.012	4.0	0	0	0	0	0	.0489	.01823	.00088	.00050	-.00503	-.00053	
89-45		FHf	213.7	50.8	0.	0.	0.	1.014	0	0	0	0	0	0	.0505	.01852	.00147	.00042	-.00360	-.00101	

TABLE XIV. WIND TUNNEL OPERATING CONDITIONS AND SHAFT FORCE AND MOMENT COEFFICIENT-SOLIDITY RATIOS AT THREE FORWARD SPEEDS AND VARIOUS FUSELAGE ATTITUDES AND ROTOR OPERATING CONDITIONS, CONFIGURATION FHB

U N	P T	CONF	V FT/SEC	q LB/FT ²	α DEG	β DEG	δ DEG	ϵ DEG	θ DEG	ϕ DEG	ψ DEG	λ DEG	μ DEG	ν DEG	C_L	C_D	C_Y	C_Z	C_{Mx}	C_{My}	C_{Mz}
61-5	FHB	FHB	103.0	11.9	0	0	0	0	0	0	0	0	0	0	.0507	-.00268	-.00099	.00349	.00397	.00349	.00104
61-6	FHB	FHB	103.0	11.9	-8	0	0	-7.5	8.0	-4.0	0	-7.5	7.7	0	.0828	-.00391	.00103	.00523	.00438	.00523	.00066
61-7	FHB	FHB	103.0	11.9	-8	0	0	-11.9	10.0	4.0	0	2.0	4.1	0	.0538	-.00607	.00209	.00447	.00014	.00447	-.00026
61-8	FHB	FHB	103.0	11.9	-4	0	0	-7.4	8.0	0	0	-2.0	3.7	0	.0532	-.00234	-.00005	.00330	.00014	.00330	.00050
61-9	FHB	FHB	103.0	11.9	4	0	0	7.3	8.0	0	0	-2.9	4.1	0	.0822	.00697	.00155	.00202	.00003	.00202	.00174
61-10	FHB	FHB	103.0	11.9	8	0	0	5.8	4.0	0	0	-1.8	2.6	0	.0666	.01114	.00111	-.00022	.00013	.00022	.00118
61-11	FHB	FHB	103.0	11.9	0	8	0	-6.8	8.0	-4.0	0	8.1	7.1	0	.0516	-.00233	-.00070	.00347	.00400	.00347	.00075
61-12	FHB	FHB	103.0	11.9	-8	4	0	-8.0	12.0	4.0	0	2.0	1.2	0	.0836	-.00355	.00111	.00523	.00427	.00523	.00063
61-13	FHB	FHB	103.0	11.9	-8	4	0	-11.6	10.0	0	0	-2.4	3.9	0	.0541	-.00593	.00225	.00446	.00013	.00446	.00040
61-14	FHB	FHB	103.0	11.9	-4	4	0	-7.0	8.0	0	0	-2.3	3.3	0	.0536	-.00217	.00017	.00329	.00006	.00329	.00041
61-15	FHB	FHB	103.0	11.9	4	4	0	5	8.0	0	0	-3.2	3.9	0	.0822	.00712	-.00126	.00205	.00003	.00205	.00143
61-16	FHB	FHB	103.0	11.9	4	4	0	-7.6	8.0	0	0	-2.4	4.5	0	.0824	.00695	-.00170	.00197	.00005	.00197	.00177
61-17	FHB	FHB	103.0	11.9	-4	4	0	-1.1	8.0	0	0	-1.8	3.8	0	.0535	-.00222	.00049	.00329	.00008	.00329	.00089
61-18	FHB	FHB	103.0	11.9	-8	4	0	-12.3	10.0	0	0	-1.9	4.6	0	.0538	-.00628	.00169	.00445	.00011	.00445	-.00003
61-19	FHB	FHB	103.0	11.9	-8	4	0	-9.5	12.0	4.0	0	2.9	1.9	0	.0836	-.00411	.00070	.00527	.00418	.00527	.00100
61-20	FHB	FHB	103.0	11.9	0	8	0	-8.0	8.0	-4.0	0	-7.1	8.3	0	.0521	-.00258	.00168	.00350	.00019	.00350	.00138
62-5	FHB	FHB	170.2	32.2	0	0	0	-9.0	8.0	0	0	-2.3	5.3	0	.0299	.00062	-.00065	.00330	.00003	.00330	.00062
62-6	FHB	FHB	170.2	32.2	0	0	0	-7.6	10.0	0	0	-3.4	7.7	0	.0612	.00169	.00304	.00459	.00014	.00459	.00247
62-7	FHB	FHB	170.2	32.2	4	0	0	-2.2	4.0	0	0	-2.2	4.2	0	.0471	.00551	-.00167	.00105	.00001	.00105	.00145
62-8	FHB	FHB	170.2	32.2	8	0	0	3.2	4.0	0	0	-2.9	4.9	0	.0674	.01079	-.00299	-.00003	.00003	-.00003	.00258
62-9	FHB	FHB	170.2	32.2	-4	4	0	-9.0	8.0	0	0	-2.5	5.0	0	.0299	.00073	.00014	.00327	.00006	.00327	.00001
62-10	FHB	FHB	170.2	32.2	0	4	0	-8.6	12.0	0	0	-4.9	8.8	0	.0701	.00077	-.00308	.00619	.00009	.00619	.00262
62-11	FHB	FHB	170.2	32.2	0	4	0	-7.3	10.0	0	0	-4.1	7.4	0	.0600	.00147	-.00222	.00453	.00009	.00453	.00184
62-12	FHB	FHB	170.2	32.2	0	4	0	-6.1	4.0	0	0	-1.8	3.0	0	.0270	.00267	.00009	.00179	.00018	.00179	.00002
62-13	FHB	FHB	170.2	32.2	4	4	0	-3.0	8.0	-4.0	0	-8.8	10.2	0	.0444	.00165	.00103	.00334	.00021	.00334	.00084
62-14	FHB	FHB	170.2	32.2	4	4	0	2.9	0	0	0	-1.5	1.1	0	.0235	.00433	.00001	.00076	.00001	.00076	-.00003
62-15	FHB	FHB	170.2	32.2	-4	4	0	-9.7	8.0	0	0	-2.5	4.0	0	.0473	.00557	-.00087	.00101	.00021	.00101	.00071
62-16	FHB	FHB	170.2	32.2	-4	4	0	-9.8	12.0	0	0	-1.6	5.7	0	.0296	.00067	.00132	.00322	.00002	.00322	.00104
62-17	FHB	FHB	170.2	32.2	0	-4	0	-9.8	8.0	0	0	-3.3	9.9	0	.0706	.00084	-.00386	.00605	.00004	.00605	.00307
62-18	FHB	FHB	170.2	32.2	0	-4	0	-8.1	10.0	0	0	-2.6	8.2	0	.0602	.00157	-.00324	.00445	.00001	.00445	.00261
62-19	FHB	FHB	170.2	32.2	0	-4	0	-3.7	4.0	0	0	-1.2	3.7	0	.0274	.00260	-.00139	.00181	.00002	.00181	.00110
62-20	FHB	FHB	170.2	32.2	4	-4	0	-6.6	8.0	-4.0	0	-7.7	10.7	0	.0462	.00163	.00255	.00321	.00405	.00321	.00208
62-21	FHB	FHB	170.2	32.2	4	-4	0	2.6	4.0	0	0	-1.6	4.6	0	.0473	.00540	-.00227	.00179	.00007	.00179	.00179
62-22	FHB	FHB	170.2	32.2	4	-4	0	-1.8	4.0	0	0	-1.0	1.7	0	.0231	.00407	-.00140	.00082	.00013	.00082	.00090
41-5	FHB	FHB	213.2	49.8	0	0	0	-4.6	4.0	0	0	-2.2	4.6	0	.0422	.00572	-.00291	.00183	.00501	.00183	.00331
41-6	FHB	FHB	213.2	49.8	0	0	0	-4.6	4.0	0	0	-2.2	4.6	0	.0414	.00464	.00019	.00218	.00032	.00218	.00034
41-7	FHB	FHB	213.2	49.8	0	0	0	-4.1	4.0	0	0	-1.2	4.0	0	.0494	.00384	.00066	.00221	.00080	.00221	.00029
41-8	FHB	FHB	213.2	49.8	4	0	0	-7.6	4.0	-4.0	0	-7.6	8.1	0	.0057	-.00068	-.00092	.00239	.00000	.00239	.00040
41-9	FHB	FHB	213.2	49.8	4	0	0	-3.6	4.0	0	0	-3.6	5.8	0	.0049	-.00498	.00309	.00217	.00000	.00217	.00256
41-10	FHB	FHB	213.2	49.8	4	0	0	-3.5	4.0	-4.0	0	-3.5	7.5	0	-.0040	.00373	.00224	.00249	.00398	.00249	.00176
41-11	FHB	FHB	213.2	49.8	0	2	0	-4.1	4.0	0	0	-2.5	4.1	0	.0026	.00488	.00150	.00224	.00019	.00224	-.00079

TABLE XV - Concluded

R U N	P Y	CONF	V FT/SEC	Q LB/FT ²	α ₁ DEG	ψ DEG	μ	ε _e DEG	θ _e DEG	θ _m DEG	β ₁₂ DEG	β ₁₃ DEG	β ₁₄ DEG	β ₁₅ DEG	β ₁₆ DEG	β ₁₇ DEG	β ₁₈ DEG	β ₁₉ DEG	β ₂₀ DEG	β ₂₁ DEG	β ₂₂ DEG	c _L /σ	c _D /σ	c _y /σ	c _q /σ	c _{PM} /σ	c _{RM} /σ
33-7		FMBF	213.2	49.9	0.	0.	1.012	-8	4.0	0	0	-1.7	-8	7.7	-8	0.0566	.00008	.00232	-.00323	-.00121							
33-8		FMBF	213.2	49.9	4.	0.	1.010	-3.7	4.0	-4.0	0	-7.3	7.7	7.7	.00823	-.00236	.00270	-.00113	-.00025								
33-9		FMBF	213.2	49.9	4.	0.	1.010	-8	4.0	0	0	-4.3	4.6	4.6	.00858	-.00520	.00263	-.00322	-.00266								
33-10		FMBF	213.2	49.9	4.	4.	1.010	-3.1	4.0	-4.0	0	-8.1	7.1	7.1	.00775	-.00104	.00280	-.00123	-.00183								
33-11		FMBF	213.2	49.9	-0.	4.	1.012	-3.3	4.0	0	0	-2.8	3.3	3.3	.0123	-.00051	.00234	-.00293	-.00094								
33-12		FMBF	213.2	49.9	0.	2.	1.012	-5	4.0	4.0	0	0	7	5	.0636	-.00928	.00223	-.00785	-.00364								
33-13		FMBF	213.2	49.9	-0.	2.	1.012	-3.7	4.0	0	0	-2.7	3.7	3.7	.0114	-.00157	.00245	-.00290	-.00018								
33-14		FMBF	213.2	49.9	0.	2.	1.012	1.0	4.0	0	0	-1.8	1.0	1.0	.0174	-.00708	.00238	-.00333	-.00036								
33-15		FMBF	213.2	49.9	4.	2.	1.010	-3.3	4.0	-4.0	0	-7.7	7.3	7.3	.0113	-.00743	.00279	-.00092	-.00105								
33-16		FMBF	213.2	49.9	4.	2.	1.010	-3	4.0	0	0	-4.7	4.4	4.4	.0571	-.00837	.00265	-.00367	-.00233								
33-17		FMBF	213.2	49.9	4.	2.	1.010	1.2	4.0	-4.0	0	-6.7	2.8	2.8	.0155	-.00636	.00264	-.00097	-.00138								
33-18		FMBF	213.2	49.9	0.	-2.	1.012	-1.1	4.0	4.0	0	1.2	1.1	1.1	.0635	-.00888	.00223	-.00804	-.00565								
33-19		FMBF	213.2	49.9	-0.	-2.	1.012	-4.2	4.0	0	0	-2.1	4.2	4.2	.0119	-.00824	.00240	-.00281	-.00167								
33-20		FMBF	213.2	49.9	4.	-2.	1.010	-3.8	4.0	-4.0	0	-7.3	7.8	7.8	.0111	-.00795	.00269	-.00372	-.00070								
33-21		FMBF	213.2	49.9	4.	-2.	1.010	-1.0	4.0	0	0	-4.0	5.0	5.0	.0559	-.00845	.00255	-.00348	-.00364								
33-22		FMBF	213.2	49.9	-0.	0.	1.012	-3.9	4.0	0	0	-2.4	3.9	3.9	.0130	-.00907	.00250	-.00258	-.00042								

APPENDIX IV

ROTOR BLADE STRESS AND FLAPPING DATA TABLES

TABLE XVI. ROTOR BLADE MOMENT AND FLAPPING DATA, CONFIGURATION FHB

RUN POINT	q, psf	μ	β , DEG		M_{F3R} , IN-LB		M_{F6R} , IN-LB		M_{T18R} , IN-LB		M_{T35R} , IN-LB	
			MAX	MIN	MAX	MIN	MAX	MIN	MAX	MIN	MAX	MIN
61-5	11.9	.30	0.2	-3.3	9.	-3.	12.	-6.	4.	-5.	3.	-6.
61-6	11.9	.29	7.0	-1.0	11.	-3.	17.	-6.	4.	-7.	3.	-7.
61-7	11.9	.29	2.4	1.3	9.	-1.	14.	-7.	2.	-5.	1.	-6.
61-8	11.9	.29	2.3	1.3	7.	-2.	13.	-6.	2.	-5.	3.	-3.
61-9	11.9	.29	3.3	2.5	8.	-2.	12.	-7.	5.	-6.	3.	-5.
61-10	11.9	.29	2.7	1.8	6.	-4.	10.	-7.	5.	-3.	4.	-4.
61-11	11.9	.29	6.3	-3.3	10.	-3.	14.	-7.	4.	-5.	3.	-5.
61-12	11.9	.29	6.3	-3.2	10.	-2.	14.	-6.	4.	-5.	3.	-5.
61-13	11.9	.29	6.9	-1.8	11.	-3.	18.	-8.	2.	-7.	2.	-8.
61-14	11.9	.29	2.4	1.4	8.	-1.	14.	-4.	2.	-6.	3.	-5.
61-15	11.9	.29	2.3	1.3	8.	-1.	15.	-6.	2.	-5.	2.	-5.
61-16	11.9	.29	3.2	2.2	9.	-2.	14.	-7.	6.	-5.	2.	-5.
61-17	11.9	.29	2.3	1.4	7.	-2.	14.	-5.	4.	-5.	2.	-4.
61-18	11.9	.29	2.4	1.3	8.	-1.	16.	-5.	3.	-5.	1.	-6.
61-19	11.9	.29	7.1	-1.9	10.	-3.	18.	-7.	4.	-8.	3.	-7.
61-20	11.9	.29	6.2	-3.2	8.	-3.	14.	-8.	4.	-5.	3.	-5.
62-5	32.2	.50	2.2	0.0	13.	-7.	18.	-9.	5.	-7.	4.	-6.
62-6	32.2	.50	3.6	0.6	19.	-11.	25.	-12.	11.	-19.	11.	-18.
62-7	32.2	.50	2.6	0.5	12.	-7.	16.	-14.	8.	-6.	6.	-8.
62-8	32.2	.50	3.4	1.0	14.	-8.	21.	-19.	8.	-14.	7.	-13.
62-9	32.2	.50	2.0	0.0	14.	-5.	19.	-9.	5.	-7.	6.	-7.
62-10	32.2	.50	4.1	0.8	26.	-11.	25.	-14.	14.	-25.	14.	-24.
62-11	32.2	.50	3.1	0.6	19.	-11.	23.	-12.	13.	-20.	12.	-19.
62-12	32.2	.50	1.5	-0.2	13.	-5.	14.	-10.	7.	-4.	6.	-5.
62-13	32.2	.50	6.5	-4.2	17.	-9.	19.	-11.	12.	-9.	11.	-10.
62-14	32.2	.50	1.4	-0.3	11.	-7.	13.	-13.	9.	-4.	6.	-3.
62-15	32.2	.50	2.6	-0.3	14.	-7.	17.	-13.	10.	-7.	7.	-7.
62-16	32.2	.50	1.7	-0.3	14.	-7.	18.	-11.	5.	-8.	4.	-9.
62-17	32.2	.50	3.9	0.2	21.	-12.	30.	-12.	16.	-25.	14.	-25.
62-18	32.2	.50	3.4	0.4	22.	-9.	25.	-13.	13.	-18.	12.	-18.
62-19	32.2	.50	1.7	-0.2	14.	-5.	17.	-8.	8.	-5.	6.	-6.
62-20	32.2	.50	6.3	-3.9	20.	-8.	18.	-14.	10.	-9.	9.	-10.
62-21	32.2	.50	2.5	0.2	15.	-8.	19.	-15.	11.	-6.	5.	-9.
62-22	32.2	.50	1.3	-0.2	7.	-9.	12.	-12.	8.	-2.	5.	-3.
41-5	49.8	1.01	6.6	-4.8	32.	-30.	37.	-18.	10.	-14.	-	-
41-6	49.8	1.01	2.1	-1.3	25.	-22.	21.	-19.	18.	-8.	60.	0.
41-7	49.8	1.01	1.7	-1.0	22.	-23.	20.	-18.	6.	-7.	52.	5.
41-8	49.8	1.01	4.6	-4.4	21.	-23.	20.	-20.	14.	-8.	55.	-11.
41-9	49.8	1.01	3.9	-1.3	37.	-35.	35.	-20.	10.	-13.	98.	-40.
41-10	49.8	1.01	4.8	-4.5	25.	-28.	23.	-18.	13.	-8.	60.	-10.
41-11	49.8	1.01	2.0	-1.0	20.	-25.	22.	-20.	18.	-7.	65.	-8.
41-12	49.8	1.01	7.2	-5.0	38.	-35.	37.	-20.	9.	-15.	98.	-32.
41-13	49.8	1.01	4.3	-5.3	30.	-35.	30.	-20.	22.	-8.	60.	-15.
41-14	49.8	1.01	1.5	-0.8	27.	-25.	25.	-18.	5.	-7.	54.	0.
41-15	49.8	1.01	4.6	-4.5	22.	-25.	20.	-28.	13.	-8.	61.	-8.
41-16	49.8	1.01	4.5	-1.3	35.	-30.	35.	-19.	10.	-3.	94.	-32.
41-17	49.8	1.01	5.0	-4.5	25.	-25.	27.	-18.	7.	-9.	60.	-7.
41-18	49.8	1.01	7.1	-5.1	40.	-19.	38.	-35.	17.	-16.	98.	-45.
41-19	49.8	1.01	2.0	-1.0	23.	-25.	20.	-19.	13.	-8.	60.	-10.
41-20	49.8	1.01	4.5	-4.3	22.	-25.	21.	-20.	16.	-9.	56.	-9.
41-21	49.8	1.01	4.4	-1.0	45.	-32.	40.	-19.	8.	-14.	95.	-30.
41-22	49.8	1.01	2.0	-1.2	25.	-23.	22.	-18.	13.	-8.	57.	-7.

TABLE XVII. ROTOR BLADE MOMENT AND FLAPPING DATA,
CONFIGURATION FHBf

RUN-POINT	q, psf	μ	β , DEG		$M_{F 3R}$, IN-LB		$M_{F 6R}$, IN-LB		$M_{T 1BR}$, IN-LB		$M_{T 3SR}$, IN-LB	
			MAX	MIN	MAX	MIN	MAX	MIN	MAX	MIN	MAX	MIN
29-5	11.8	.30	0.0	0.0	5	-5	-	-	-	-	2	-2
29-6	11.8	.29	5.5	-3.0	17	-9	-	-	-	-	3	-8
29-7	11.8	.29	6.0	-1.5	18	-12	-	-	-	-	3	-11
29-8	11.8	.29	2.6	1.0	13	-9	-	-	-	-	3	-10
29-9	11.8	.29	2.3	1.0	10	-10	-	-	-	-	4	-8
29-10	11.8	.29	3.5	2.2	15	-13	-	-	-	-	5	-9
29-11	11.8	.29	2.6	1.5	12	-16	-	-	-	-	6	-7
29-12	11.8	.29	5.6	-3.3	17	-14	-	-	-	-	4	-9
29-13	11.8	.29	5.8	-1.5	17	-12	17	-9	8	-10	5	-12
29-14	11.8	.29	2.7	1.1	13	-11	13	-7	7	-7	5	-10
29-15	11.8	.29	2.5	-0.8	12	-11	13	-5	3	-3	5	-8
29-16	11.8	.29	3.4	2.0	15	-13	13	-9	3	-4	8	-9
29-17	11.8	.29	3.5	1.9	16	-14	14	-8	9	-9	5	-10
29-18	11.8	.30	2.5	-0.5	13	-12	12	-6	8	-6	4	-10
30-6	11.8	.29	2.4	0.3	15	-9	13	-6	6	-8	3	-10
30-7	11.8	.29	2.4	1.0	14	-8	14	-5	4	-8	2	-9
30-8	11.8	.29	5.6	-1.6	17	-12	19	-10	8	-12	5	-12
30-9	11.8	.30	5.5	-3.5	17	-10	13	-6	5	-7	4	-8
30-10	11.8	.30	0.5	-0.5	10	-13	8	-7	8	-3	7	-3
30-11	11.8	.30	0.0	0.0	5	-5	-	-	-	-	2	-2
31-5	32.1	.50	2.5	-2.5	15	-15	18	-11	8	-12	8	-15
31-7	32.1	.50	2.8	0.0	12	-18	19	-13	11	-9	10	-15
31-8	32.1	.50	3.5	-1.0	12	-18	19	-15	10	-18	13	-20
31-9	32.1	.50	2.5	0.0	15	-13	18	-11	8	-9	13	-14
31-10	32.1	.50	4.3	0.0	30	-20	20	-12	11	-15	13	-22
31-11	32.1	.50	2.5	-0.5	20	-15	20	-13	12	-10	15	-17
31-12	32.1	.50	6.0	-4.3	25	-23	17	-12	18	-15	14	-20
31-13	32.1	.50	2.0	-1.0	20	-17	17	-14	13	-8	15	-12
31-14	32.1	.50	3.5	-0.5	20	-20	18	-14	11	-12	15	-18
31-15	32.1	.50	2.5	-0.5	18	-17	14	-10	8	-13	10	-18
31-16	32.1	.50	4.5	0.5	30	-25	14	-13	11	-16	13	-23
31-17	32.1	.50	4.0	0.0	28	-25	19	-13	11	-13	14	-23
31-18	32.1	.50	2.3	-0.5	20	-18	19	-12	11	-9	14	-14
31-19	32.1	.50	6.5	-4.0	30	-20	14	-13	15	-14	15	-22
31-20	32.1	.50	3.5	0.0	23	-25	16	-15	12	-11	15	-18
31-21	32.1	.50	2.0	-0.8	18	-20	17	-15	12	-7	15	-13
31-22	32.1	.50	5.0	0.5	30	-25	20	-13	12	-18	15	-25
31-23	32.1	.50	4.3	0.0	30	-25	18	-12	12	-13	14	-22
33-5	49.9	1.01	0.0	-4.4	-	-	35	-55	14	-22	14	-22
33-6	49.9	1.01	3.0	-2.0	-	-	33	-38	20	-8	18	-10
33-7	49.9	1.01	2.8	-1.5	-	-	20	-40	14	-14	10	-14
33-8	49.9	1.01	6.0	-5.3	-	-	35	-38	20	-8	18	-12
33-9	49.9	1.01	5.5	-1.5	-	-	35	-55	13	-20	12	-20
33-10	49.9	1.01	6.2	-5.5	-	-	33	-35	16	-8	17	-10
33-11	49.9	1.01	3.5	-1.6	-	-	33	-37	18	-9	17	-12
33-12	49.9	1.01	3.2	-4.3	-	-	33	-55	15	-22	16	-24
33-13	49.9	1.01	3.1	-1.6	29	-34	35	-38	18	-10	20	-11
33-14	49.9	1.01	2.8	-1.5	24	-31	23	-37	12	-13	10	-14
33-15	49.9	1.01	6.3	-5.5	25	-36	35	-40	20	-10	18	-13
33-16	49.9	1.01	5.7	-1.5	33	-38	35	-60	13	-22	13	-25
33-17	49.9	1.01	6.5	-5.3	-	-	18	-40	13	-15	10	-17
33-18	49.9	1.01	3.0	-4.5	-	-	27	-55	15	-23	17	-23
33-19	49.9	1.01	1.8	-3.0	-	-	30	-40	22	-6	18	-12
33-20	49.9	1.01	6.0	-5.0	-	-	30	-38	22	-6	19	-15
33-21	49.9	1.01	5.5	-1.5	-	-	30	-58	29	-3	10	-21
33-22	49.9	1.01	3.0	-1.6	-	-	27	-38	18	-8	18	-13

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13. ABSTRACT This report presents the results of an experimental investigation of the effects of rotor head configuration and fuselage yaw on the wake characteristics and rotor performance of a generalized 1/8th scale helicopter model. Several rotor head configurations were tested with and without rotor blades at various operating conditions. Fuselage yaw angle was varied for all conditions tested. The flow direction and dynamic pressure at the tail of the model were measured, as were rotor shaft forces and moments. When the model was tested with the rotor, blade vibratory stresses were measured. Wake survey data were obtained with and without rotor blades; however, the data obtained with blades were not consistent.		

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