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# ION ACOUSTIC TURBULENCE, ANOMALOUS TRANSPORT, AND SYSTEM DYNAMICS IN HALL EFFECT THRUSTERS

Robert Martin<sup>1</sup>, Jonathan Tran<sup>2</sup>

<sup>1</sup>AIR FORCE RESEARCH LABORATORY,  
<sup>2</sup>ERC INC.,  
EDWARDS AIR FORCE BASE, CA USA



IPAM Mathematics of Turbulence Retreat, June 2017  
UCLA Lake Arrowhead Conference Center, CA

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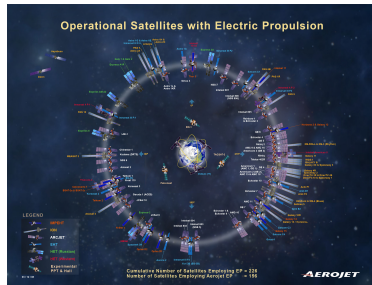


- 1 INTRODUCTION
- 2 TRANSPORT
- 3 DYNAMIC SYSTEM
- 4 SUMMARY AND CONCLUSION



## EP-Devices:

- EP Improves Thrust/Mass (Isp)
- Ion/Elec.-Thermal/HET Flying
- FRC/MPD/Electrospray/etc. in Dev.
- Space notoriously Risk Adverse (i.e. Tech must be “Proven”)



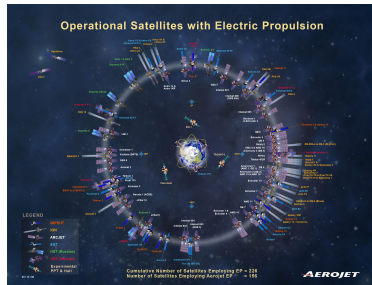
Aerojet Overview of EP Satellites (3/08)



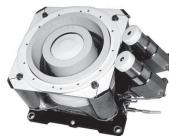
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- Plasma still Threat Thruster/Plume
- Improved Life Estimates Needed
- Space not Replicated in Ground Test

Models needed to Bridge Gap



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Fakel SPT-100



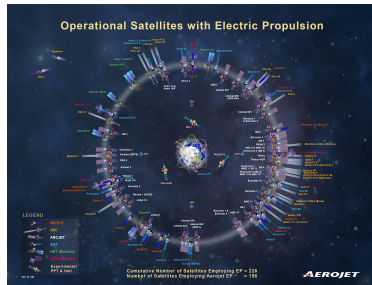
NASA Hermes Thruster



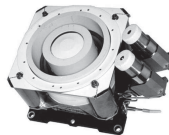
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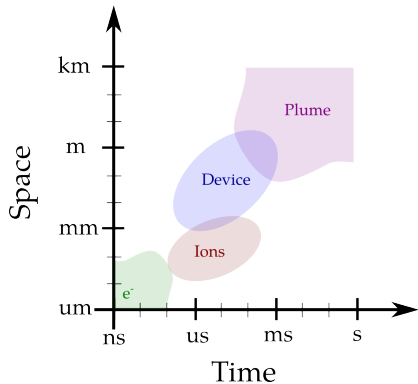


NASA Hermes Thruster



## Inherently Multiscale:

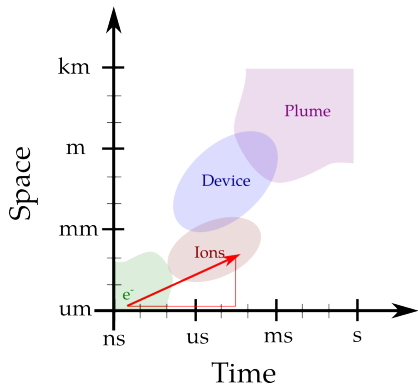
- $\mathcal{O}(10^9)$ -Space and Time
- Naive 3D-Spatial Scales  $\rightarrow$  Cost<sup>3</sup>





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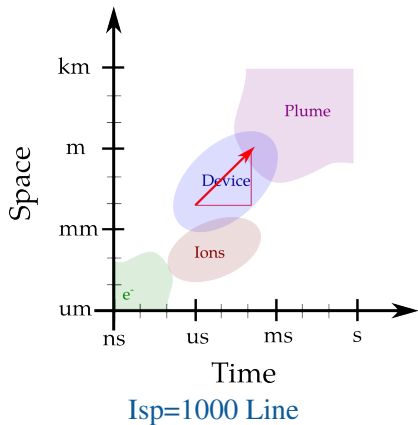
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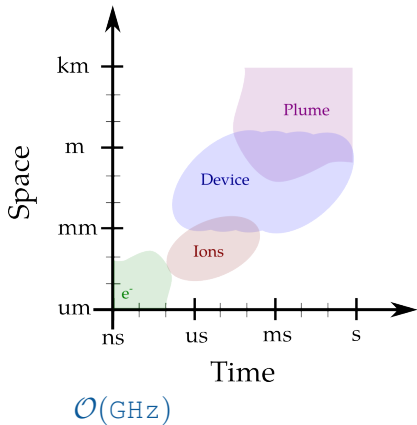
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- Isp & Device Scale  $\rightarrow$  Transit Time





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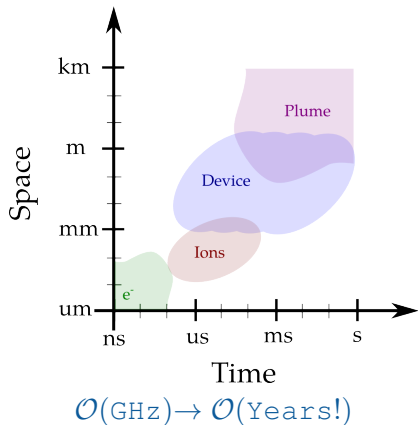
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- Pulsed: Spread Right at Scale & Isp





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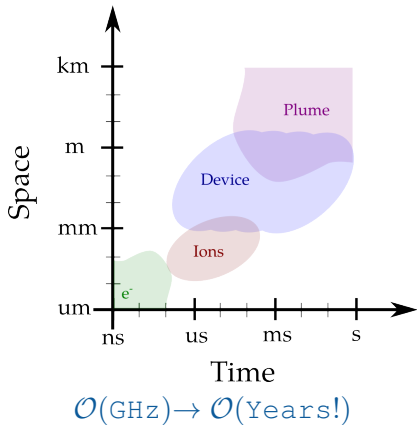




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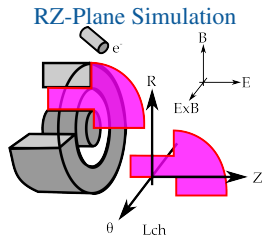
**Modeling Must Exploit Scale Separation**





## Hall Effect Thrusters (HET):

- Traditionally Modeled in R-Z
- Named for Hall Current in  $\theta$
- Uses Quasi-1D Electron Fluid Solve
- Ohm's Law  $\rightarrow$  No  $e^-$ -momentum



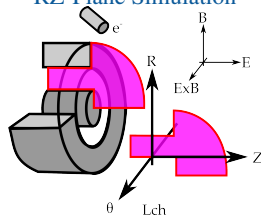


# HALL EFFECT THRUSTERS

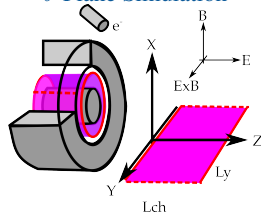
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### RZ-Plane Simulation



### $\theta$ -Plane Simulation

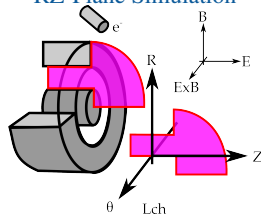




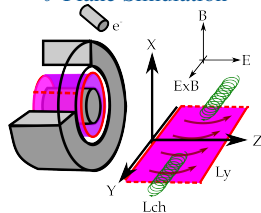
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- Unmagnetized Ions
- Results in Hall Current (Namesake)

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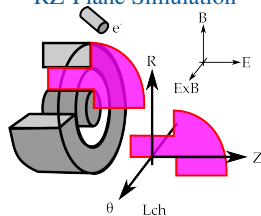




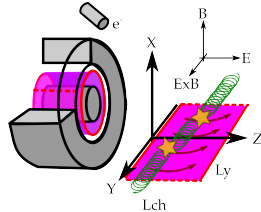
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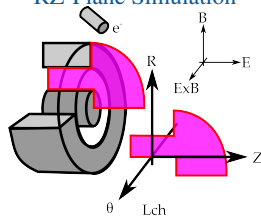




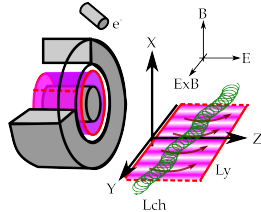
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### $\theta$ -Plane Simulation





## Model Sensitive to Mobility:

- Classical Mobility Insufficient in Near Plume

From LaFleur, Phys Plasmas 23, 053503 (2016)

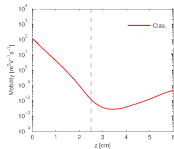


FIG. 3. Electron cross-field mobility as a function of axial position within the PPS<sup>3</sup>1350 thruster. The solid black line is an empirical mobility which is needed in fluid simulations in order to get agreement with experiment,<sup>26</sup> the red line is the mobility based on classical diffusion across a magnetic field, while the open blue triangles show the mobility due to the saturated instability-enhanced electron-ion friction force. The vertical dashed line indicates the thruster exit plane.

(Thruster Simulation will not “Light”)



## Model Sensitive to Mobility:

- Classical Mobility Insufficient in Near Plume
- Enhanced Empirical “Bohm” (1/B) Mobility
- Also Needs Coefficient by “Zone” (i.e. Anode/Channel/Plume Regions)

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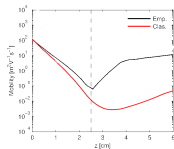
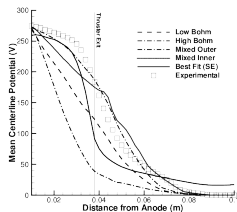


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## Impact of Mobility on Discharge Profile



(Koo, PhD Dissertation)



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- Behavior Sensitive to Plume Coefficient

## Critical if Operating Near Mode Change

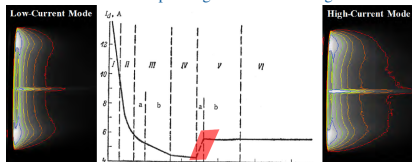
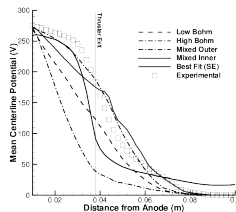


Fig. 1 Discharge current as function of magnetic field with constant discharge voltage showing operational regimes defined by Tillman [2]  
 Sekerak, Gallimore, Brown, Hofer, and Polk, JPP 2016  
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Need Physics Based Model

## Critical if Operating Near Mode Change

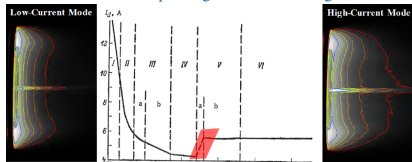
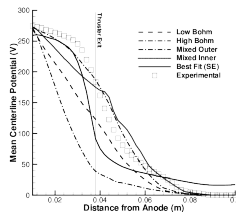


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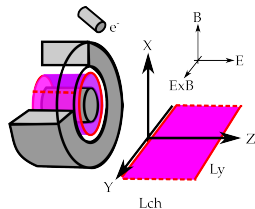


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## Current Driven Instability:

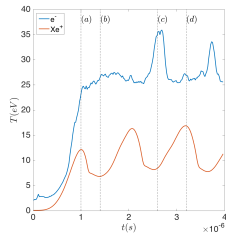
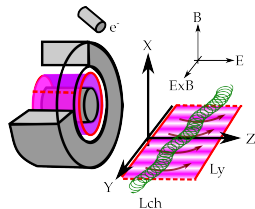
- HET Ion-Acoustic Instability
  - Driven by  $j_{\theta}$  Hall Current
  - Anomalous e-Transport? (LaFleur)
  - Focus of 1D/2D Full-PIC Studies





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Initial Exponential Growth Saturates

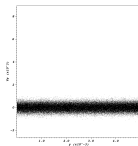




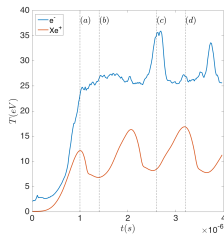
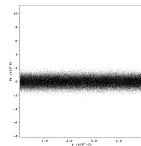
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Xe-VDF



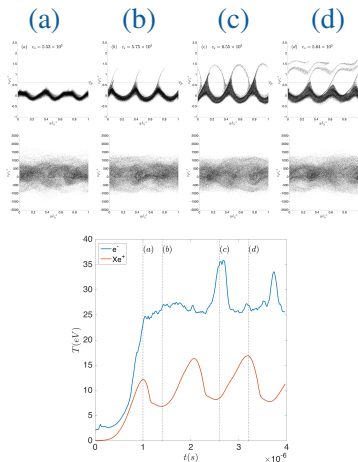
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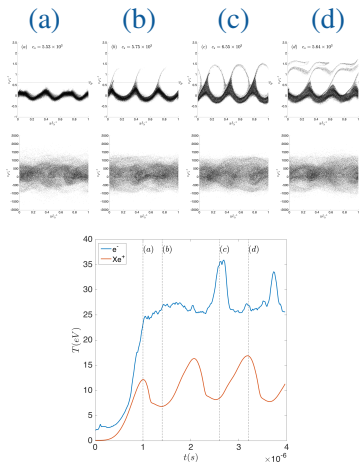
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But only 1D...

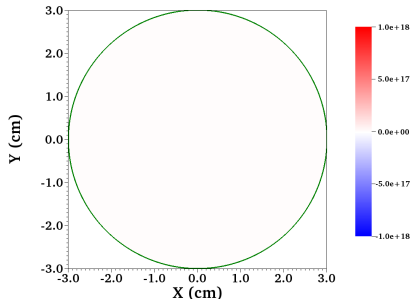




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- $j_\theta \rightarrow$  FRC Spiral Charge Separation?

## Charge Separation

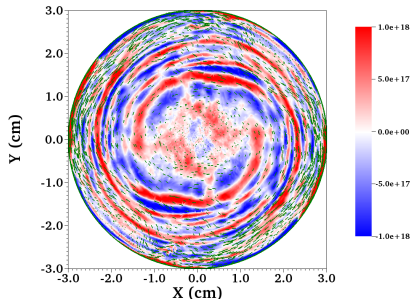




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But only 1D...
- $j_\theta \rightarrow$  FRC Spiral Charge Separation?
- Onset of Plasma Turbulence?

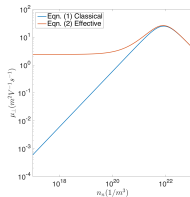
## Charge Separation





## Impact of Ion-Acoustic Instability:

- Instability  $\rightarrow$  Extra Mobility





## Impact of Ion-Acoustic Instability:

- Instability  $\rightarrow$  Extra Mobility
- LaFleur Results Promising

From LaFleur, Phys Plasmas 23, 053503 (2016)

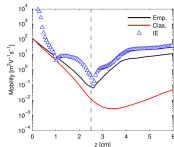
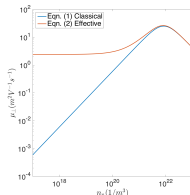


FIG. 3. Electron cross-field mobility as a function of axial position within the PPS<sup>®</sup> 1350 thruster. The solid black line is an empirical mobility which is needed in fluid simulations in order to get agreement with experiment,<sup>18</sup> the red line is the mobility based on classical diffusion across a magnetic field, while the open blue triangles show the mobility due to the ion-sound instability-enhanced electron-ion friction force. The vertical dashed line indicates the thruster exit plane.

(A Posteriori Mobility)

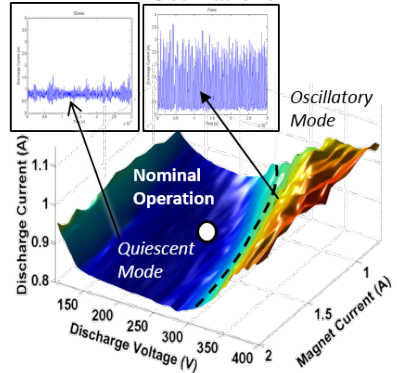




## Impact of Ion-Acoustic Instability:

- Instability  $\rightarrow$  Extra Mobility
- LaFleur Results Promising
- Need High-Dim Validation

## Experimental Discharge Current Oscillations



## I-V-B Plot of Hall Thruster Operation

Brown, EP TEMPEST Program Review, 2015



## Impact of Ion-Acoustic Instability:

- Instability → Extra Mobility
- LaFleur Results Promising
- Need High-Dim Validation
- Model via Lagged Lotka-Volterra?

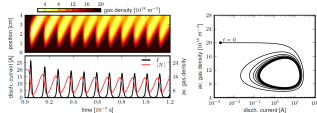


Figure 2. Convergence of the one-dimensional model (ii) to a limit cycle with a low-current, gas-filled startup. The phase portrait shows the coupled evolution of the discharge current and of the average gas density within the discharge. The parameters are  $\tau = 20 \text{ ns}^{-1}$ ,  $Q_0 = 5 \times 10^7 \text{ m}^{-2} \text{ s}^{-1}$ ,  $I = 4 \text{ A}$ ,  $\alpha = 4 \times 10^5 \text{ A}^{-1}$ ,  $\beta_0 = 2 \times 10^5 \text{ m}^{-1}$ . The resonant profiles  $\psi(x)$  and  $\Psi(x)$  are shown on Fig. 1.

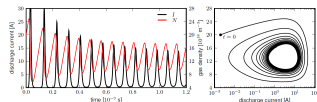


Figure 5. Convergence of the modified positive-ion model (ii) to a limit cycle with a low-current, gas-filled startup. The parameters are defined from those of Fig. 2, assuming that the length of the relaxation region is given approximately by half of the discharge sheath length:  $L = l/2 = 2 \text{ cm}$ ,  $Q_0 = 5 \times 10^7 \text{ m}^{-2} \text{ s}^{-1}$ ,  $I = 4 \text{ A}$ ,  $\gamma = 3.5 \times 10^4 \text{ A}^{-1}$ ,  $\beta_0 = 2 \times 10^5 \text{ m}^{-1}$ ,  $\tau = 2.2 \times 10^{-7} \text{ s}$ .

$$\frac{dI}{dt} = \beta I(N - \bar{N}),$$

$$\frac{dN}{dt} = -\gamma IN + \frac{Q_0}{L} \exp\left[-\gamma \int_{t-\tau}^t I dt\right].$$

Barral & Peradzyński, IEPC-2009-070



## Impact of Ion-Acoustic Instability:

- Instability  $\rightarrow$  Extra Mobility
- LaFleur Results Promising
- Need High-Dim Validation
- Model via Lagged Lotka-Volterra?
- Model Captures a Bifurcation

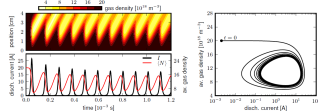


Figure 2. Convergence of the one-dimensional model (ii) in a burst cycle with a low-current, gas-filled startup. The phase portrait shows the coupled evolution of the discharge current and of the average gas density within the discharge. The parameters are  $\tau = 5 \text{ ns}$ ,  $\tau = 20 \text{ ns}^{-1}$ ,  $Q_0 = 5 \times 10^7 \text{ m}^{-2} \text{ s}^{-1}$ ,  $I = 4 \text{ A}$ ,  $\mu = 4 \times 10^5 \text{ A}^{-1}$ ,  $\beta_0 = 2 \times 10^5 \text{ m}^{-3}$ . The resonant profiles  $\psi(x)$  and  $\Psi(x)$  are shown on Fig. 5.

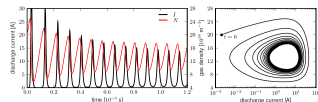


Figure 5. Convergence of the modified proton-gas model (ii) in a burst cycle with a low-current, gas-filled startup. The parameters are defined from those of Fig. 2, assuming that the length of the relaxation region is given approximately half of the discharge volume length:  $L = 1/2 = 2 \text{ cm}$ ,  $Q_0 = 5 \times 10^7 \text{ m}^{-2} \text{ s}^{-1}$ ,  $I = 4 \text{ A}$ ,  $\gamma = 3.5 \times 10^5 \text{ A}^{-1}$ ,  $\beta_0 = 5 \times 10^5 \text{ m}^{-3}$ ,  $\tau = 2.2 \times 10^{-7} \text{ s}$ .

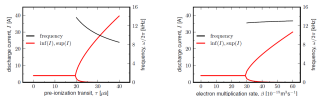


Figure 6. Bifurcation diagrams for the modified proton-gas model (ii). Parameters  $\tau$  and  $\beta$  are varied around the resonant point investigated in Fig. 5.

$$\frac{dI}{dt} = \beta I(N - \bar{N}),$$

$$\frac{dN}{dt} = -\gamma IN + \frac{Q_0}{L} \exp \left[ -\gamma \int_{t-\tau}^t I dt \right].$$

Barral & Peradzyński, IEPC-2009-070



## Impact of Ion-Acoustic Instability:

- Instability → Extra Mobility
- LaFleur Results Promising
- Need High-Dim Validation
- Model via Lagged Lotka-Volterra?
- Model Captures a Bifurcation

How can we Compare  
Model, Experiment, and Simulation?

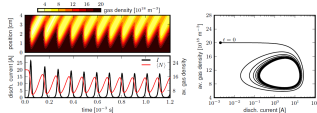


Figure 2. Convergence of the one-dimensional model (ii) in a limit cycle with a low-current, gas-filled startup. The phase portrait shows the coupled evolution of the discharge current and of the average gas density within the discharge. The parameters are  $\tau = 5 \text{ ns}$ ,  $\tau' = 20 \text{ ns}^{-1}$ ,  $Q_0 = 5 \times 10^7 \text{ m}^{-3} \text{ s}^{-1}$ ,  $I = 4 \text{ A}$ ,  $\mu = 4 \times 10^4 \text{ m}^2 \text{ V}^{-1} \text{ s}^{-1}$ ,  $\eta_0 = 2 \times 10^{-3} \text{ m}^{-3} \text{ s}^{-1}$ . The assumed profiles  $\psi(x)$  and  $\Psi(x)$  are shown on Fig. 1.

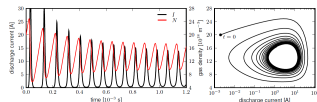


Figure 3. Convergence of the modified protonic-gas model (ii) in a limit cycle with a low-current, gas-filled startup. The parameters are defined from those of Fig. 2, assuming that the length of the discharge region is given approximately half of the discharge volume length:  $L = 1/2 \text{ cm}$ ,  $Q_0 = 5 \times 10^7 \text{ m}^{-3} \text{ s}^{-1}$ ,  $I = 4 \text{ A}$ ,  $\gamma = 3.5 \times 10^4 \text{ s}^{-1}$ ,  $\beta = 5 \text{ W}$ ,  $\beta_0 = 3.5 \times 10^{10} \text{ m}^{-3} \text{ s}^{-1}$ ,  $\tau = 2.2 \times 10^{-7} \text{ s}$ .

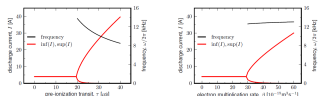


Figure 4. Bifurcation diagrams for the modified protonic-gas model (ii). Parameters  $\tau$  and  $\beta$  are varied around the assumed point investigated in Fig. 5.

$$\frac{dI}{dt} = \beta I(N - \bar{N}),$$

$$\frac{dN}{dt} = -\gamma N + \frac{Q_0}{L} \exp\left[-\gamma \int_{t-\tau}^t I dt\right].$$

Barral & Peradzyński, IEPC-2009-070



## Impact of Ion-Acoustic Instability:

- Instability  $\rightarrow$  Extra Mobility
- LaFleur Results Promising
- Need High-Dim Validation
- Model via Lagged Lotka-Volterra?
- Model Captures a Bifurcation

How can we Compare  
Model, Experiment, and Simulation?  
With only  $I(t)$  Accessible Experimentally?

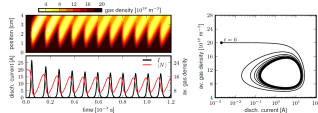


Figure 2. Convergence of the one-dimensional model (ii) to a limit cycle with a low-current, gas-filled startup. The phase portrait shows the coupled evolution of the discharge current and of the average gas density within the diode. The parameters are  $\beta = 0.06$ ,  $\tau = 20 \text{ ns}^{-1}$ ,  $Q_0 = 5 \times 10^7 \text{ m}^{-2} \text{ s}^{-1}$ ,  $I = 4 \text{ A}$ ,  $\gamma = 4 \times 10^4 \text{ s}^{-1} \text{ A}^{-1}$ ,  $\beta_0 = 2 \times 10^4 \text{ s}^{-1}$ . The assumed profiles  $\psi(x)$  and  $\Psi(x)$  are shown on Fig. 1.

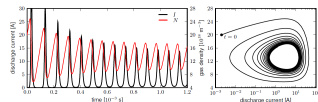


Figure 3. Convergence of the modified protonic-gas model (ii) to a limit cycle with a low-current, gas-filled startup. The parameters are defined from those of Fig. 2, assuming that the length of the reduction cycle is now approximately half of the discharge relaxation time:  $L = 1/2$ ,  $Q_0 = 5 \times 10^7 \text{ m}^{-2} \text{ s}^{-1}$ ,  $I = 4 \text{ A}$ ,  $\gamma = 3.5 \times 10^4 \text{ s}^{-1} \text{ A}^{-1}$ ,  $\beta_0 = 5 \times 10^4 \text{ s}^{-1}$ ,  $\tau = 2.2 \times 10^{-7} \text{ s}$ .

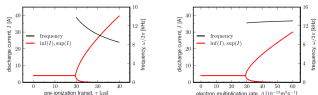


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Barral & Peradzyński, IEPC-2009-070



## State Space Reconstruction: Time Series and Dynamic Systems

A supplemental simulation and animation for  
"Detecting Causality in Complex Ecosystems"

George Sugihara, Robert May, Hao Ye, Chih-hao Hsieh,  
Ethan Deyle, Mike Fogarty, and Stephan Munch

animation by: Peter Sugihara, Hao Ye, and George Sugihara

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Same Could be Done for Hall Thruster Simulation



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Same Could be Done for Hall Thruster Simulation  
But Experiment has Limited Access?



## State Space Reconstruction: Takens' Theorem and Shadow Manifolds

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### Shadow Maifold from only $I(t)$ ?



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Shadow Maifold from only  $I(t)$ ?  
Useful Tuning Model/Simulation Parameters?



## State Space Reconstruction: Convergent Cross Mapping

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Useful Detecting Causal Relationships.



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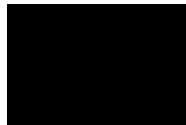
© August 2012

Useful Detecting Causal Relationships.  
Potential Tool for Self-Similarity/Causality in Turbulence?



## DMD → Turbulence:

- Low Re Vortex Shedding



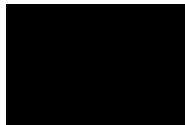
Von Karman Vortex Street

[http://commons.wikimedia.org/wiki/File:Karman\\_Vortex\\_Street\\_Off\\_Cylinder.org](http://commons.wikimedia.org/wiki/File:Karman_Vortex_Street_Off_Cylinder.org)



## DMD → Turbulence:

- Low Re Vortex Shedding
- Sparse Dynamic System via DMD



Von Karman Vortex Street

[http://commons.wikimedia.org/wiki/File:Karman\\_Vortex\\_Street\\_Off\\_Cylinder.org](http://commons.wikimedia.org/wiki/File:Karman_Vortex_Street_Off_Cylinder.org)

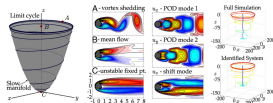


Fig. 2. Example of high-dimensional dynamical system from fluid dynamics. The vortex shedding past a cylinder is a prototypical example that is used for flow control, with relevance to many applications, including drag reduction behind vehicles. The vortex shedding is the result of a local bifurcation. However, because the Navier-Stokes equations have quadratic nonlinearity, it is necessary to use a mean-field model with a separation of timescales, where a fast mean-field deflection is slow to the slow vortex shedding dynamics. The parabolic slow manifold is shown (left), with the unstable fixed point (C), mean flow (B), and vortex shedding (A). A POD basis and shift mode are used to reduce the dimension of the problem (middle right). The identified dynamics closely match the true trajectory in POD coordinates, and most importantly, they capture the quadratic nonlinearity and timescales associated with the mean-field model.

Brunton, Proctor, and Kutz, 3932-3937, PNAS, 4/12/16, v.113, no. 15



## DMD $\rightarrow$ Turbulence:

- Low Re Vortex Shedding
- Sparse Dynamic System via DMD
- What if Flow is Non-Sparse? (Turbulent)

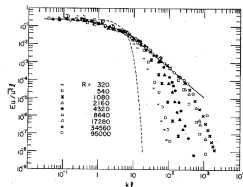


Fig. 6. Dependence of the frequency spectra  $E_u / \overline{u^2} l$  on Reynolds number in the center of the wake.

Ubeeroi and Freymuth, Phys. Fluids 12, 1359 (1969).



## DMD $\rightarrow$ Turbulence:

- Low Re Vortex Shedding
- Sparse Dynamic System via DMD
- What if Flow is Non-Sparse? (Turbulent)
- Inertial Range Turbulence really Self-Similar?
- Universal if Agnostic to Large and Small  $k$

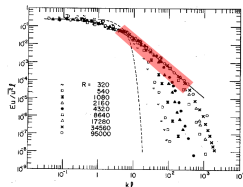


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- Bandpass Invariant up to Similarity?  
(Filter Local Space/Time: Gabor? Wavelets?)

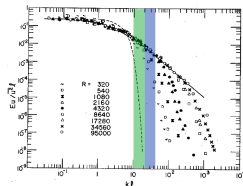
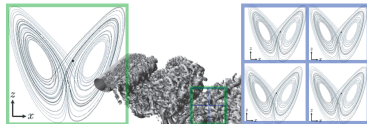


Fig. 6. Dependence of the frequency spectra  $E_w(k)/l$  on Reynolds number in the center of the wake.

Ueberoi and Freymuth, *Phys. Fluids* 12, 1359 (1969).



Piomelli, *Phil. Trans. Royal Soc., A* 372, 2014.

[https://commons.wikimedia.org/wiki/File:A\\_Trajectory\\_Through\\_Phase\\_Space\\_in\\_a\\_Lorenz\\_Attractor.gif](https://commons.wikimedia.org/wiki/File:A_Trajectory_Through_Phase_Space_in_a_Lorenz_Attractor.gif)



## DMD $\rightarrow$ Turbulence:

- Low Re Vortex Shedding
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- Inter-Band Dynamics Causal via CCM?

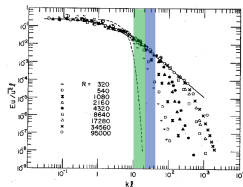
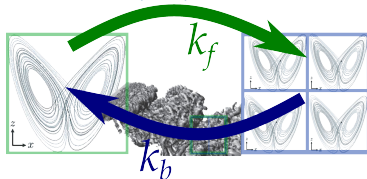


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Piomelli, Phil. Trans. Royal Soc., A 372, 2014.

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(Filter Local Space/Time: Gabor? Wavelets?)
- Inter-Band Dynamics Causal via CCM?
- CCM  $\rightarrow K_j$  in Mori-Zwanzig LES?

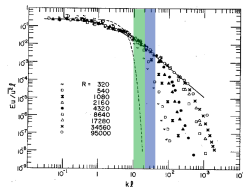
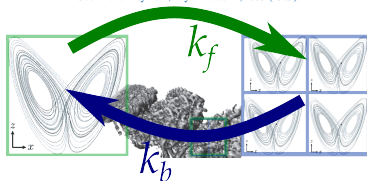


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Piomelli, Phil. Trans. Royal Soc., A 372, 2014.

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$$w_j^{(0)} = \mathcal{P} \int_0^t K_j(\bar{u}(t-s), s) ds$$

Parish and Duraisamy, AIAA Expo, 6/16



## Thank You

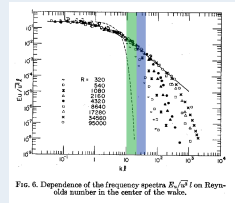
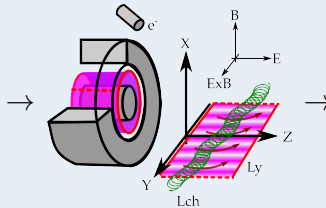
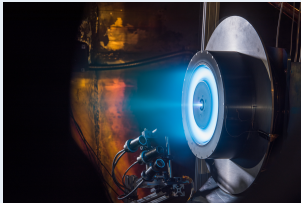


Fig. 6. Dependence of the frequency spectra  $E_0/\sqrt{L}$  on Reynolds number in the center of the wake.

Work Supported through AFOSR Task 17RQCOR465 (PM: Birkan)

## Questions?