

A Four Pillar Improvement Strategy to Qualification of Embedded Software Systems

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DM18-0609

Outline

► Challenges in Safety-critical Software-intensive Systems

Four Pillar Improvement Strategy

Pillar One

Pillar Two

Pillar Three

Pillar Four

We Rely on Software for Safe Aircraft Operation

Quantas Airbus A330-300 Forced to make Emergency Landing - 36 Injured

Written by htbw on Oct-7-08 1:48pm
From: soyawannaknow.blogspot.com



Thirty-six passengers were injured in a mid-air decompression emergency landing Tuesday.

The terrifying incident saw the Airbus A330-300 issue a mayday call when it suddenly changed altitude during a flight from Singapore to Perth, Qantas said.

Embedded software systems introduce a new class of problems not addressed by traditional system safety analysis

Oct. 15 (Bloomberg) -- **Airbus SAS** issued an alert to airlines after Australian investigators said a computer fault on a **Boeing 787** flight switched off the autopilot and generated false jet to nosedive.

The Airbus A330-300 was cruising at 37,000 feet (11,277 meters) when a computer fed incorrect information to the flight control system, the **Australian Transport Safety Bureau** said yesterday. The plane descended 650 feet within seconds, slamming passengers and crew against the ceiling, before the pilots regained control.

"This appears to be a unique event," the bureau said, adding that Airbus, the Toulouse, France-based Airbus, the world's largest maker of aircraft, issued a telex late yesterday to airlines that fly A330-300s fitted with the same air-data computer. The advisory is aimed at minimizing the risk in the unlikely event of a similar occurrence.

FAA says software problem with Boeing 787s could be catastrophic

By **Dan Catchpole**
[@dcatchpole](https://twitter.com/dcatchpole)

The Federal Aviation Administration says a software problem with Boeing 787 Dreamliners could lead to one of the most advanced jetliners losing electrical power in flight, which could lead to loss of control.

- The Buzz:** Hipster's dilemma
- Boeing & aerospace news
- Aerospace blog

The FAA notified operators of the airplane Friday that if a 787 is powered continuously for 248 days, the plane will automatically shut down its alternating current (AC) electrical power.



Software Problems not just in Aircraft



Expert • Independent • Nonprofit
ConsumerReports.org[®]



This article appeared in [May 2010 Consumer Reports Magazine](#). But it turned out to be a problem for the Kenmore 4027 front-loader, which scored near the bottom in our [February 2010 report](#).

May 7, 2010

Lexus GX 460 passes retest; Consumer Reports lifts "Don't Buy" label

Consumer Reports is lifting the [Don't Buy: Safety Risk](#) designation from the [2010 Lexus GX 460 SUV](#) after recall work corrected the problem it displayed in one of our emergency handling tests. (See the original report and video: "[Don't Buy: Safety Risk--2010 Lexus GX 460](#).")



We originally experienced the problem in a test that we use to evaluate what's called lift-off oversteer. In this test, as the vehicle is driven through a turn, the driver quickly lifts his foot off the accelerator pedal to see how the vehicle reacts. When we did this with our GX 460, its rear end slid out until the vehicle was almost sideways.

Although the GX 460 has [electronic stability control](#), which is designed to prevent a vehicle from sliding, the system wasn't intervening quickly

enough to stop the slide. We consider this a safety risk because in a real-world situation this could cause a rear tire to strike a curb or slide off of the pavement, possibly causing the vehicle to roll over. Tall vehicles with a high center of gravity, such as the GX 460, heighten our concern. We are not aware, however, of any reports of injury related to this problem.

Lexus recently duplicated the problem on its own test track and developed a [software upgrade](#) for the vehicle's ESC system that would prevent the problem from happening. [Dealers received the software fix](#) last week and began notifying GX 460 owners to bring their vehicles in for repair.

We contacted the Lexus dealership from which we had anonymously bought the vehicle and made an appointment to have the recall work performed. The work took about an hour and a half.

Following that, we again put the SUV through our full series of emergency handling tests. This time, the ESC system intervened earlier and its rear did not slide out in the lift-off oversteer test. Instead, the vehicle understeered—or plowed—when it exceeded its limits of traction, which is a more common result and makes the vehicle more predictable and less likely to roll over. Overall, we did not experience any safety concerns with the corrected GX 460 in our handling tests.

Many appliances now rely on electronic controls and operating software. But it turned out to be a problem for the Kenmore 4027 front-loader, which scored near the bottom in our February 2010 report.

Our tests found that the rinse cycles on some models worked improperly, resulting in an unimpressive cleaning.

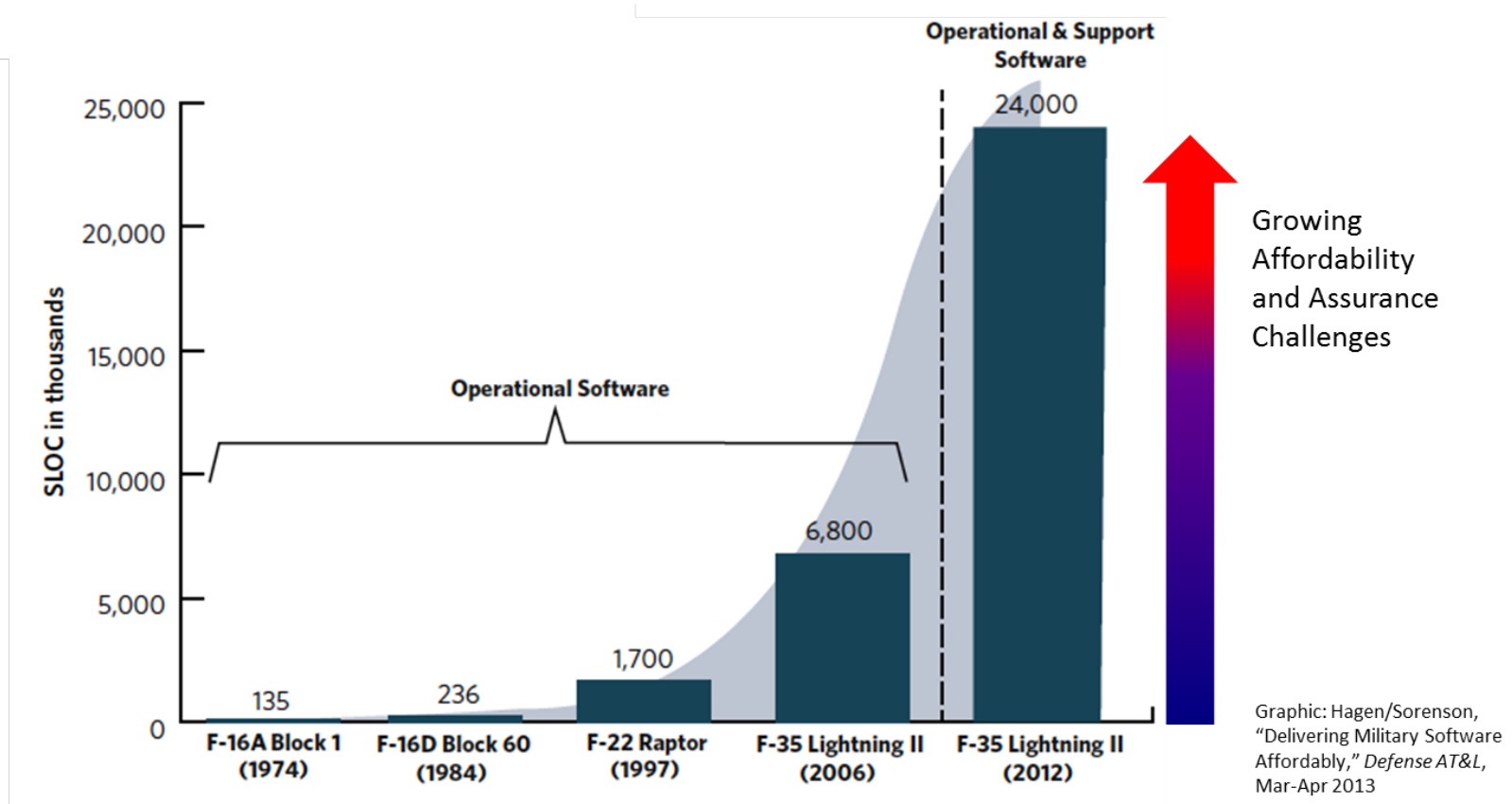
When Sears, which sells the washer, saw our [February 2010 Ratings](#) (available to subscribers), it worked with LG, which makes the washer, to figure out what was wrong. They quickly determined that a software problem was causing short or missing rinse and wash cycles, affecting wash performance. Sears and LG say they have reprogrammed the software on the models in their warehouses and on about 65 percent of the washers already sold, including the ones we had purchased.

Our retests of the reprogrammed Kenmore 4027 found that the cycles now worked properly, and the machine excelled. It now tops our [Ratings](#) (available to subscribers) of more than 50 front-loaders and we've made it a CR Best Buy.

If you own the washer, or a related model such as the Kenmore 4044 or Kenmore Elite 4051 or 4219, you should get a letter from Sears for a free service call. Or you can call 800-733-2299.

How do you upgrade washing machine software?
Internet of Things!

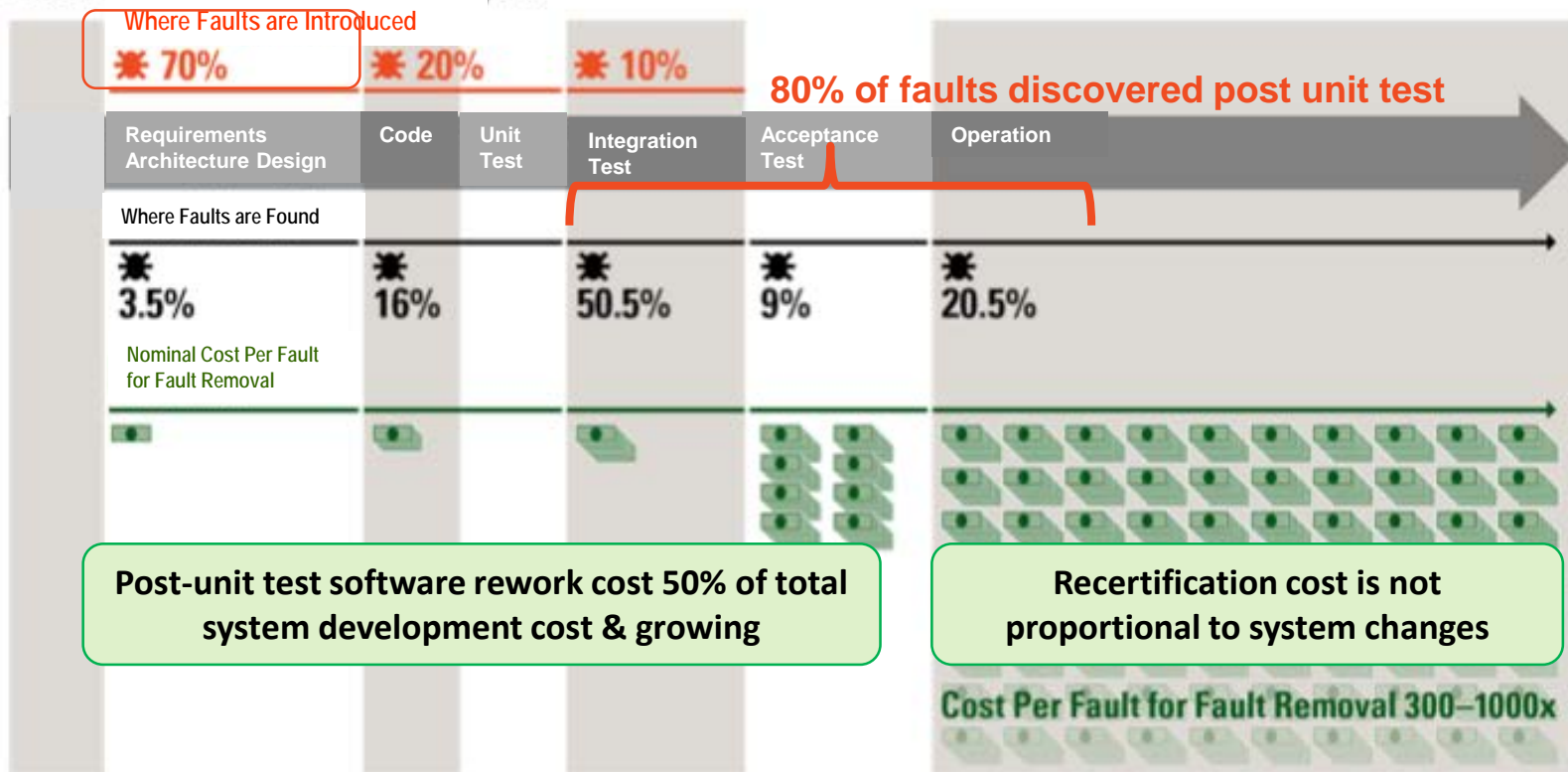
A Growing Reliance on Software



Software as % of total system cost

1997: 45% → 2010: 66% → 2024: 88%

Critical System Assurance Challenges



Sources: Critical Code; NIST, NASA, INCOSE, and Aircraft Industry Studies

Current Industry Practice in DO-178B Compliant Requirements Capture

Industry Survey in 2009 FAA Requirements Engineering Study

Notation

Enter an "x" in every row/column cell that applies

	System Requirements	Data Interconnect {ICD}	High-Level Software Requirements	Low-Level Software Requirements	Hardware Requirements
English Text or Shall Statements	39	27	36	32	29
Tables and Diagrams	31	30	30	19	18
UML Use Cases	1		2	4	
UML Sequence Diagrams			3	6	
UML State Diagrams			1	7	
Executable Models (e.g. Simulink, SCADE Suite, etc.)	7	1	8	8	1
Data Flow Diagrams (e.g. Yourdon)	4		6	9	
Other (Specify)		1			
Operational models or prototypes	1	1			1
UML			1	1	

Tool

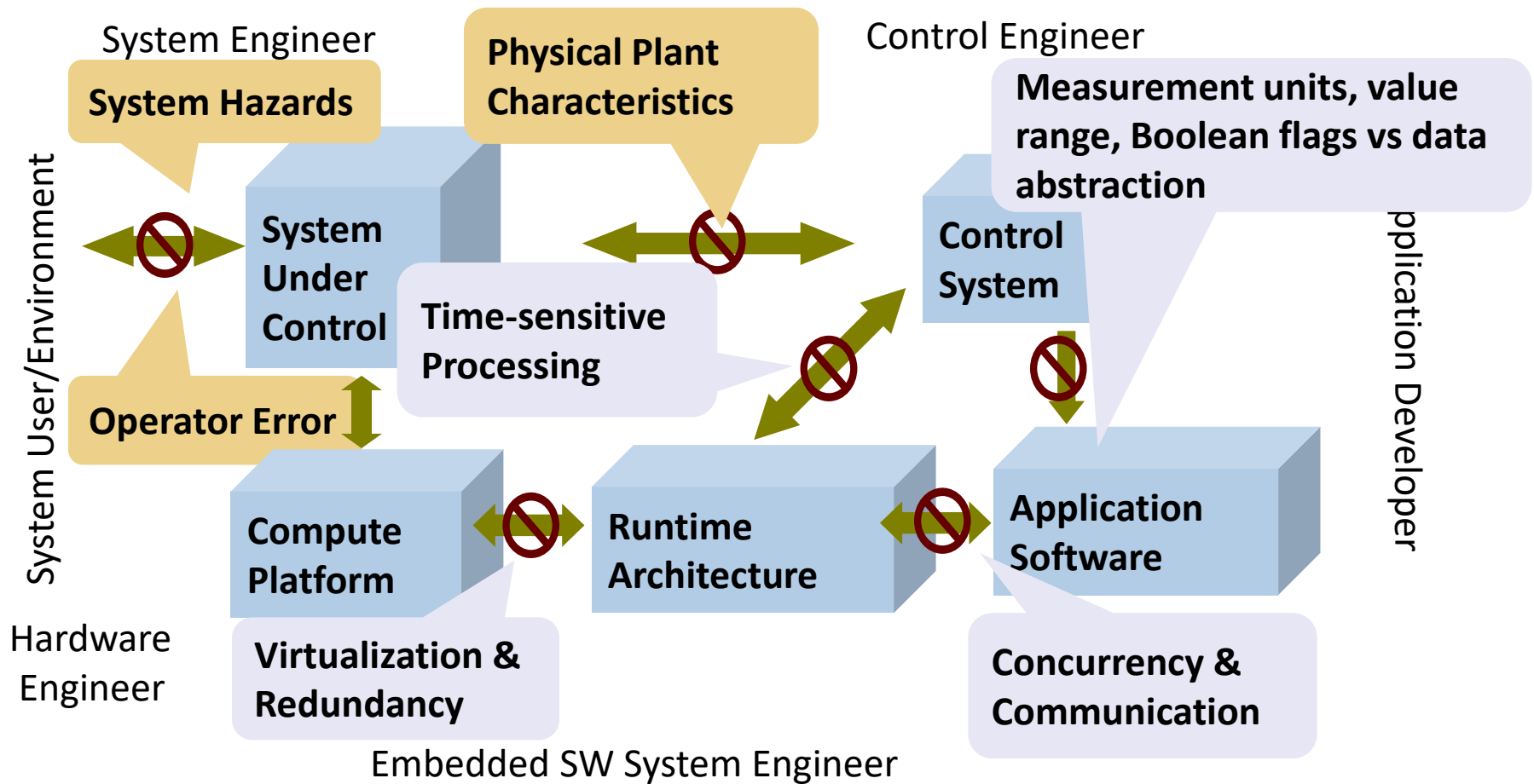
Enter an "x" in every row/column cell that applies

	System Requirements	Data Interconnect {ICD}	High-Level Software Requirements	Low-Level Software Requirements	Hardware Requirements
Database (e.g. Microsoft Access)	3	4	3	3	
DOORS	23	13	22	18	12
Rational ROSE®			1	3	
RDD-100®					
Requisite Pro®	5	3	5	4	4
Rhapsody	1				
SCADE Suite	2		3	1	
Simulink	5	1	5	3	1
Slate	1		1	1	
Spreadsheet (e.g., Microsoft Excel)	5	4	5	4	3
Statamate					
Word Processor (e.g., Microsoft Word)	19	20	18	17	16
VAPSTM		1	3	3	
Designer's Workbench™			1	1	
Proprietary Database, SCADE like pic tool		1	1		
Interleaf	1	1	1	1	1
BEACON	1	1	1	1	
CaliberRM	1	1	1	1	1
XM:		1			
Wiring diagram		1			1

Need analyzable & executable specifications



Mismatched Assumptions in Safety-Critical System Interactions



Embedded software system as major source of hazards

Why do system level failures still occur despite fault tolerance techniques being deployed in systems?

Outline

Challenges in Safety-critical Software-intensive Systems

▶ Four Pillar Improvement Strategy

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Pillar Two

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Assurance & Qualification Improvement Strategy

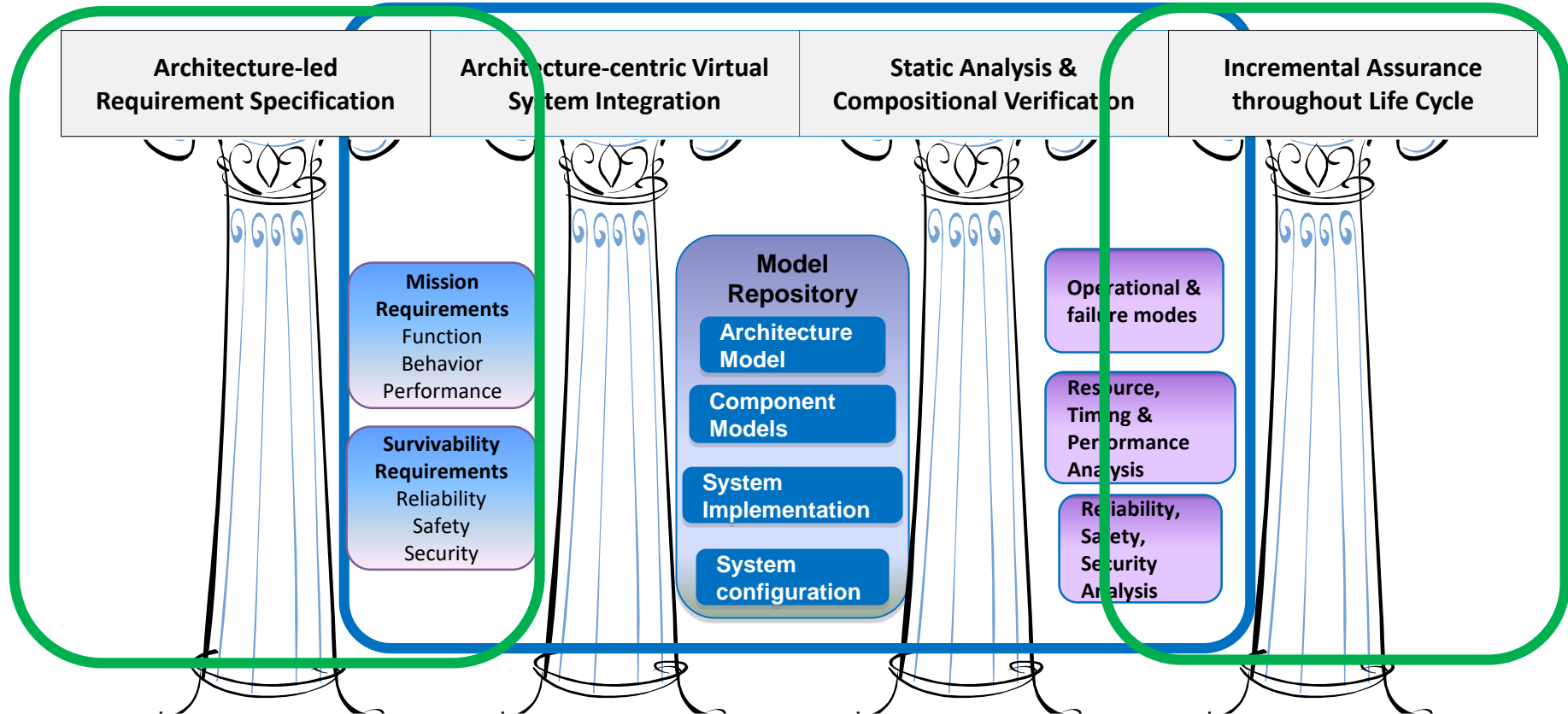


2007 National Research Council Study

Assurance: Sufficient evidence that a system implementation meets system requirements

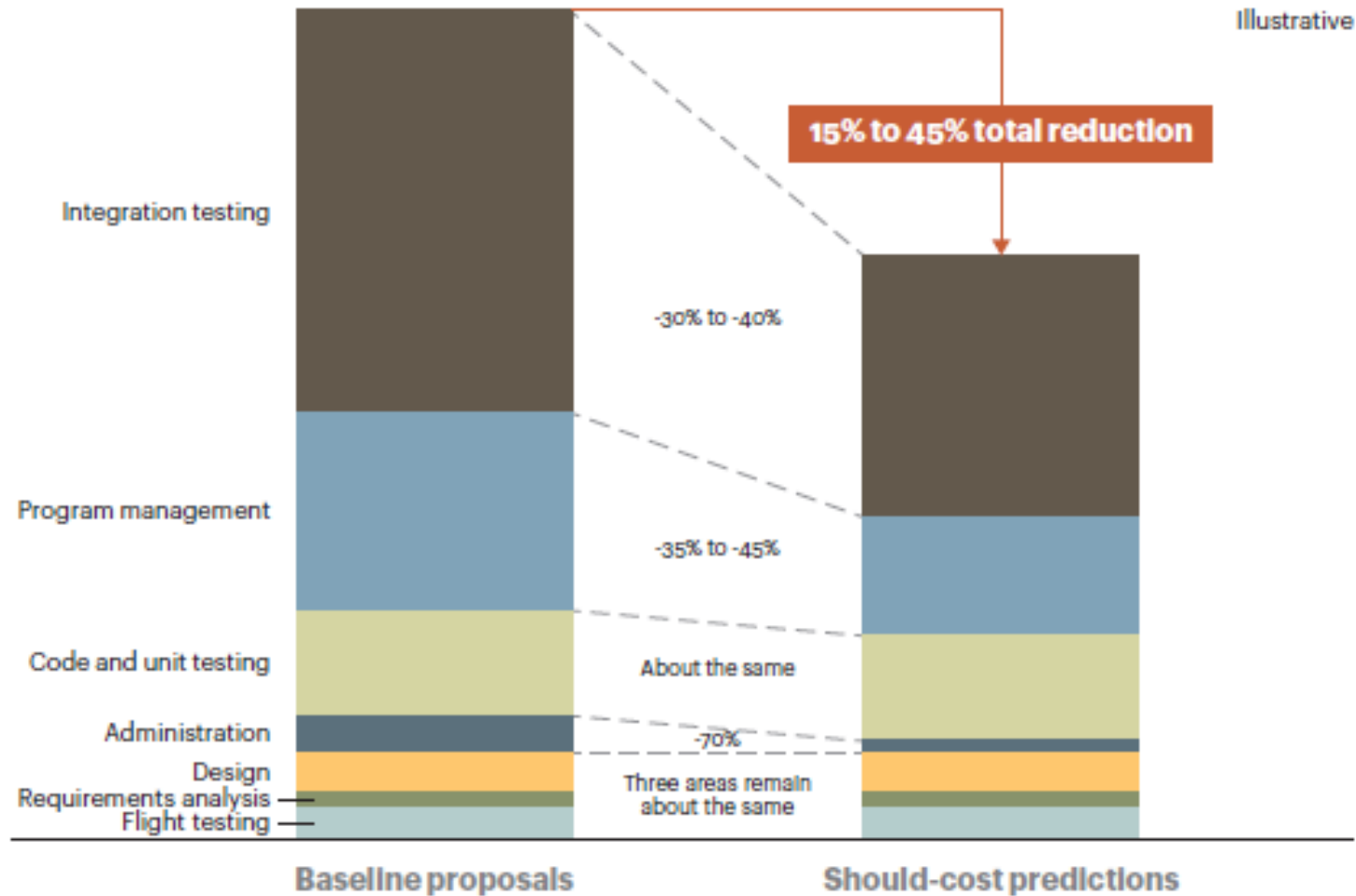


2010 SEI Study for AMRDEC
Aviation Engineering Directorate



Early Problem Discovery through Virtual System Integration & Analysis
Improved Assurance through Better Requirements & Automated Verification

“Should Cost” Predictions For Avionics



ATKearney “Software: The Brains Behind U.S. Defense Systems”



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Architecture-Centric Virtual System Integration

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Value of Requirement Uncertainty Awareness

Textual requirement quality statistics

- Opportunity for high payoff improvement
- Focus on verifiable requirement specification

Requirements error	%
Incomplete	21%
Missing	33%
Incorrect	24%
Ambiguous	6%
Inconsistent	5%

Experience based uncertainty measures

- 80% of requirement changes from development team
- Requirement uncertainty contributors
 - Volatility, Impact, Precedence, Time criticality
- Awareness of requirement uncertainty reduces requirement changes by 50%
 - Focus on uncertainty areas
 - Engineer for inherent variability

Selection	Weight	Precedence
Low Precedence	9	No experience of concept, or environment. Historically volatile
Medium Precedence	3	Some experience in related environments. Some historic volatility
High Precedence	1	Concept already in service. Low historic volatility

Rolls Royce case study



Requirements & Architecture Design Specification

Example borrowed from M. Whelan

Requirements for a Patient Therapy System

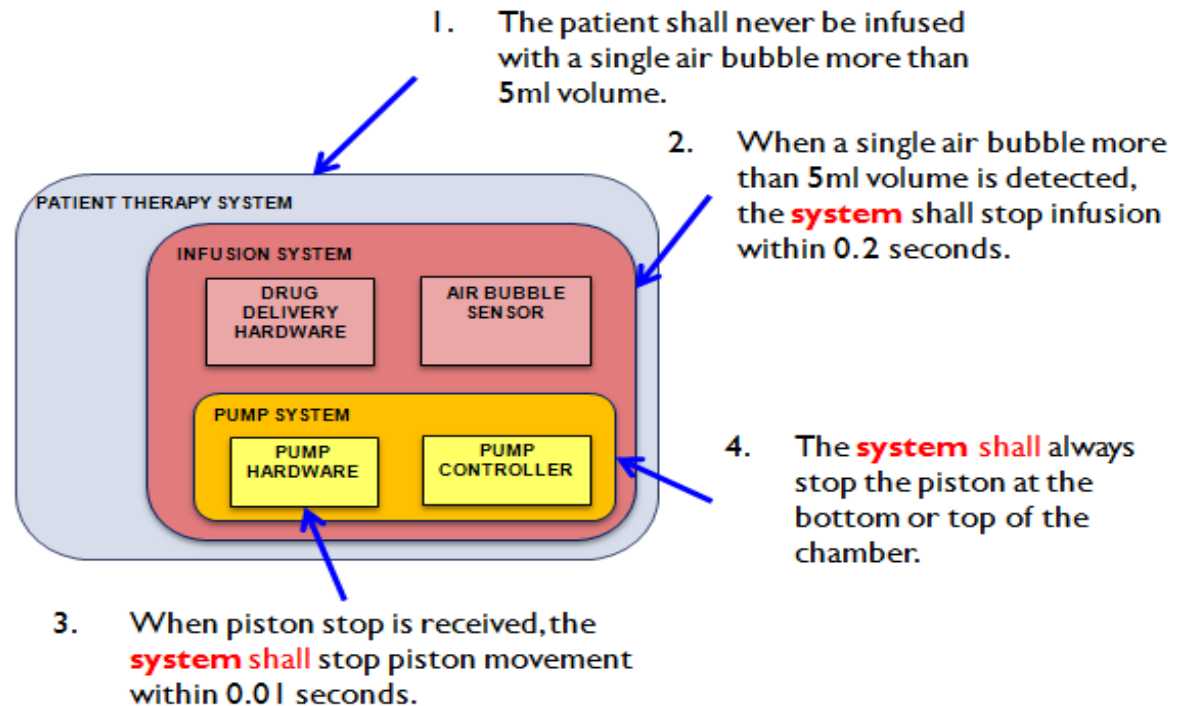
The patient shall never be infused with a single air bubble more than 5ml volume.

When a single air bubble more than 5ml volume is detected, the **system** shall stop infusion within 0.2 seconds.

When piston stop is received, the **system** shall stop piston movement within 0.01 seconds.

The **system** shall always stop the piston at the bottom or top of the chamber.

Requirements and Design Information

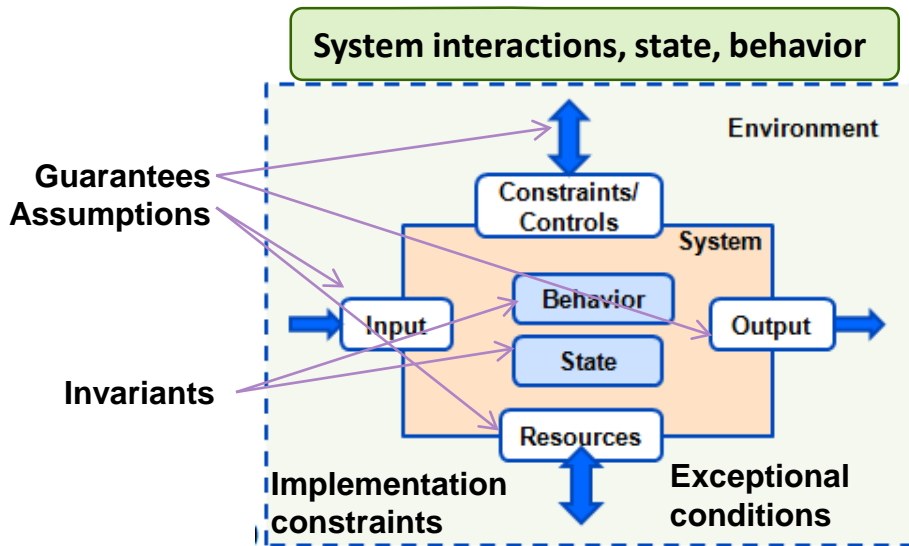


Importance of understanding system boundary
Multiple layers of system boundary

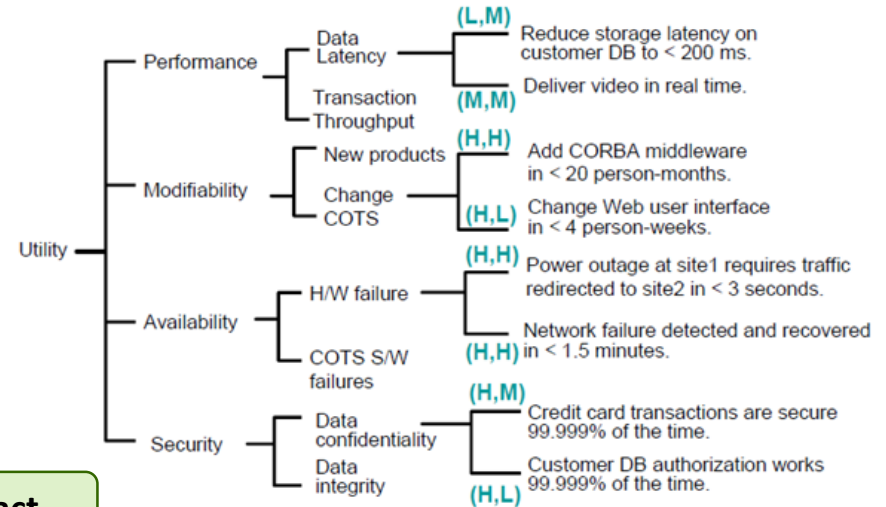
Typical requirement documents span multiple levels of a system architecture
We have effectively specified a partial architecture



Three Dimensions of Requirement Coverage



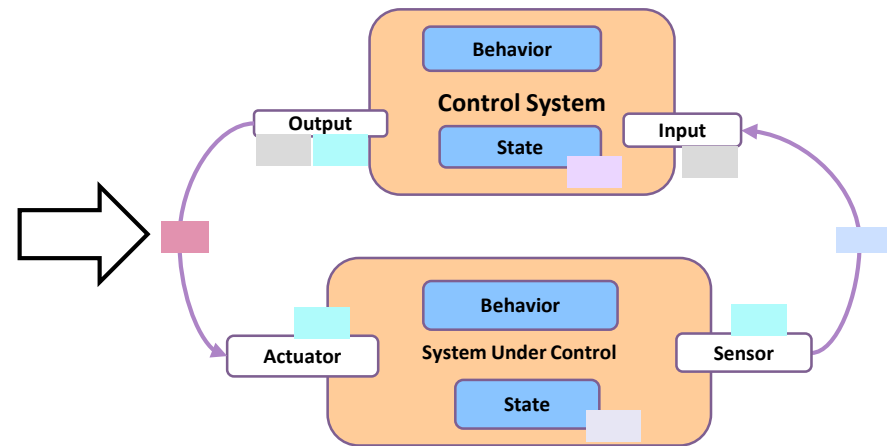
Design & operational quality attributes



Fault contributors & impact

Omission errors	Commission errors
Value errors	Sequence errors
Timing errors	Replication errors
Rate errors	Concurrency errors
Authentication errors	Authorization errors

Fault Propagation Taxonomy



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Verifiable System Requirement Specification

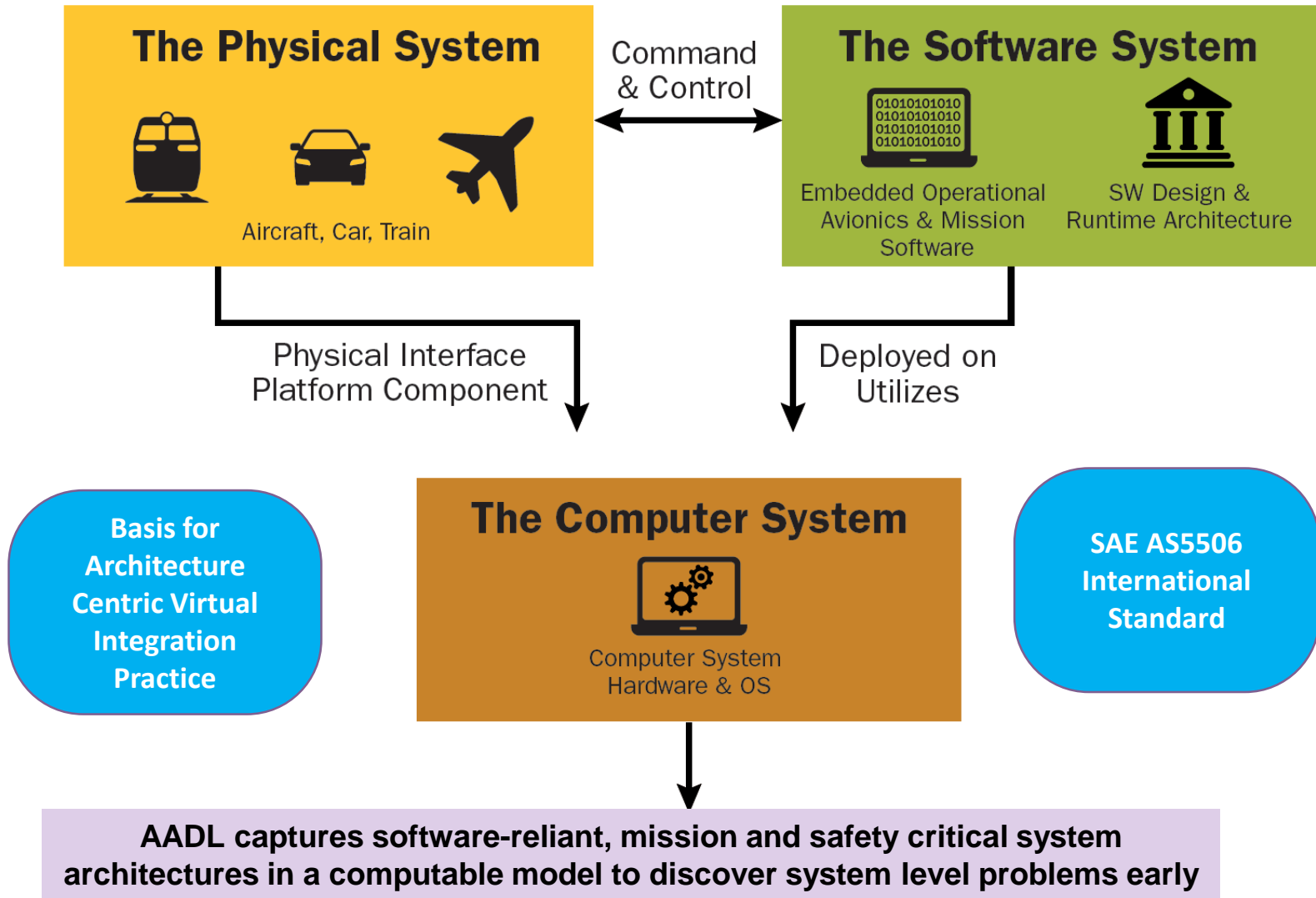
▶ Architecture-Centric Virtual System Integration

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Conclusion

Architecture Analysis & Design Language (AADL) Enables Industry-Wide Virtual Integration and Assurance Approach



SAE AADL Standard Suite (AS-5506 series)

Core AADL language standard (V2.2-Jan 2017, V1-Nov 2004)

- Strongly typed language with well-defined execution and communication semantics
- Textual and graphical notation
- Revision V3 in progress: Interface composition, system configuration, binding, type system unification

Standardized AADL Annex Extensions

Error Model language for safety, reliability, security analysis

ARINC653 extension for partitioned architectures

Behavior Specification Language for modes and interaction behavior

Data Modeling extension for interfacing with data models (UML, ASN.1, ...)

AADL Runtime System & Code Generation

AADL Annexes in Progress

Network Specification Annex

Cyber Security Annex

Requirements Definition and Assurance Annex

Synchronous System Specification Annex

Hybrid System Specification Annex

System Constraint Specification Annex



System Level Fault Root Causes

Addressed by AADL Concepts
with Well Defined Semantics

Violation of Data Stream Assumptions

- Stream miss rates, Mismatched data representation
- Latency jitter & age

End-to-end flows for latency analysis
Sampling & queued ports
Mid-frame & frame-delayed connections
Port connection consistency

Partitions as Isolation Regions

- Space, time, and bandwidth partitioning
- Isolation not guaranteed due to undocumented resource sharing

**Process and virtual
processor to model
partitioned architectures**

Virtualization of Time & Resources

- Logical vs. physical redundancy
- Time stamping of data & asynchronous systems

**Virtual processors &
virtual buses**
Multiple time domains

Inconsistent System States & Interactions

- Modal systems with modal components
- Concurrency & redundancy management

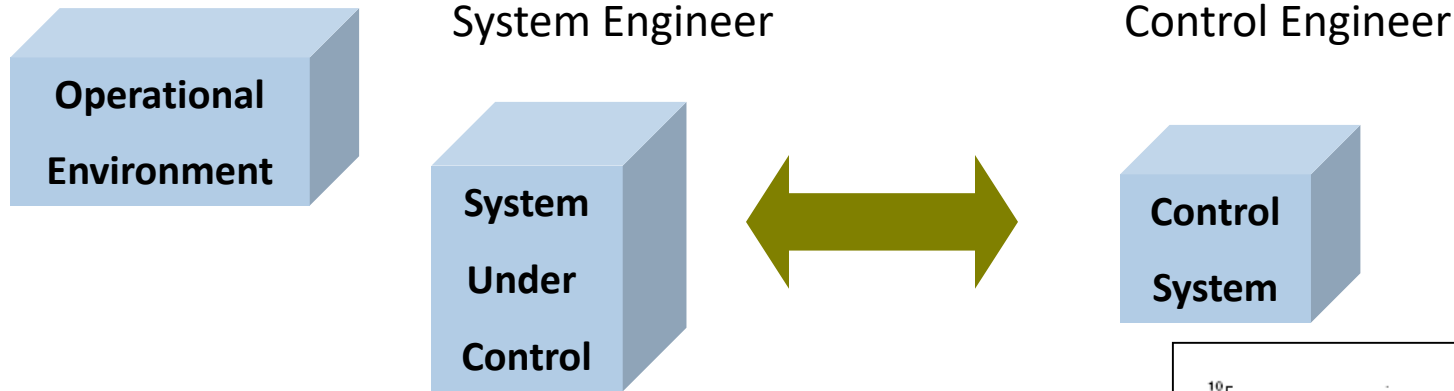
Operational and failure modes
Interaction behavior specification
Dynamic reconfiguration
Fault detection, isolation, recovery

Shared Resource Management

- Processor, memory & network resources
- Unmanaged computer system resources

**Resource allocation &
deployment configurations**
**Resource budget analysis &
scheduling analysis**

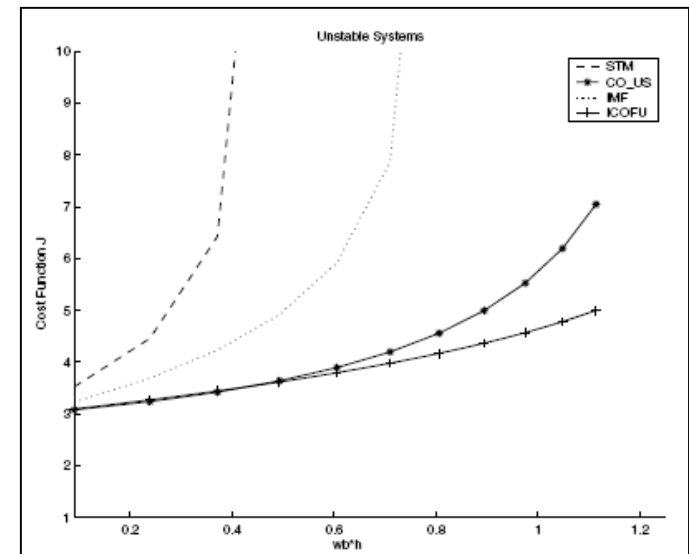
Multi-Fidelity End-to-end Latency in Control Systems



Common latency data from system engineering

- Processing latency
- Sampling latency
- Physical signal latency

Impact of Scheduler Choice on Controller Stability
A. Cervin, Lund U., CCACSD 2006



Software-Based Latency Contributors

Execution time variation: algorithm, use of cache

Processor speed

Resource contention

Preemption

Legacy & shared variable communication

Rate group optimization

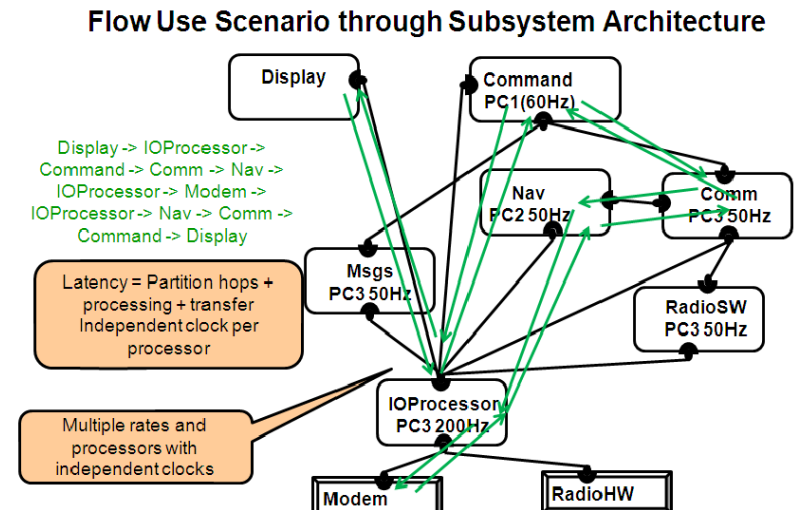
Protocol specific communication delay

Partitioned architecture

Migration of functionality

Fault tolerance strategy

Synchronized time domains



Time-sensitive Sampled Processing Issues

From Customer Design Document

*“The 200 Hz update rate was used because the MUX data needed to be processed at twice the rate of the fastest channel to avoid a race condition. Because channel 3 operates at 100 Hz, the IO processor had to operate at 200 Hz. **The race condition has been fixed by double-buffering data**, but the IO processor execution rate was left at 200 Hz to reduce latency of MUX data.”*

Did double buffering solve the problem?

Dual channel COM/MON of digital wheel braking system votes itself out due to timing of event sampling



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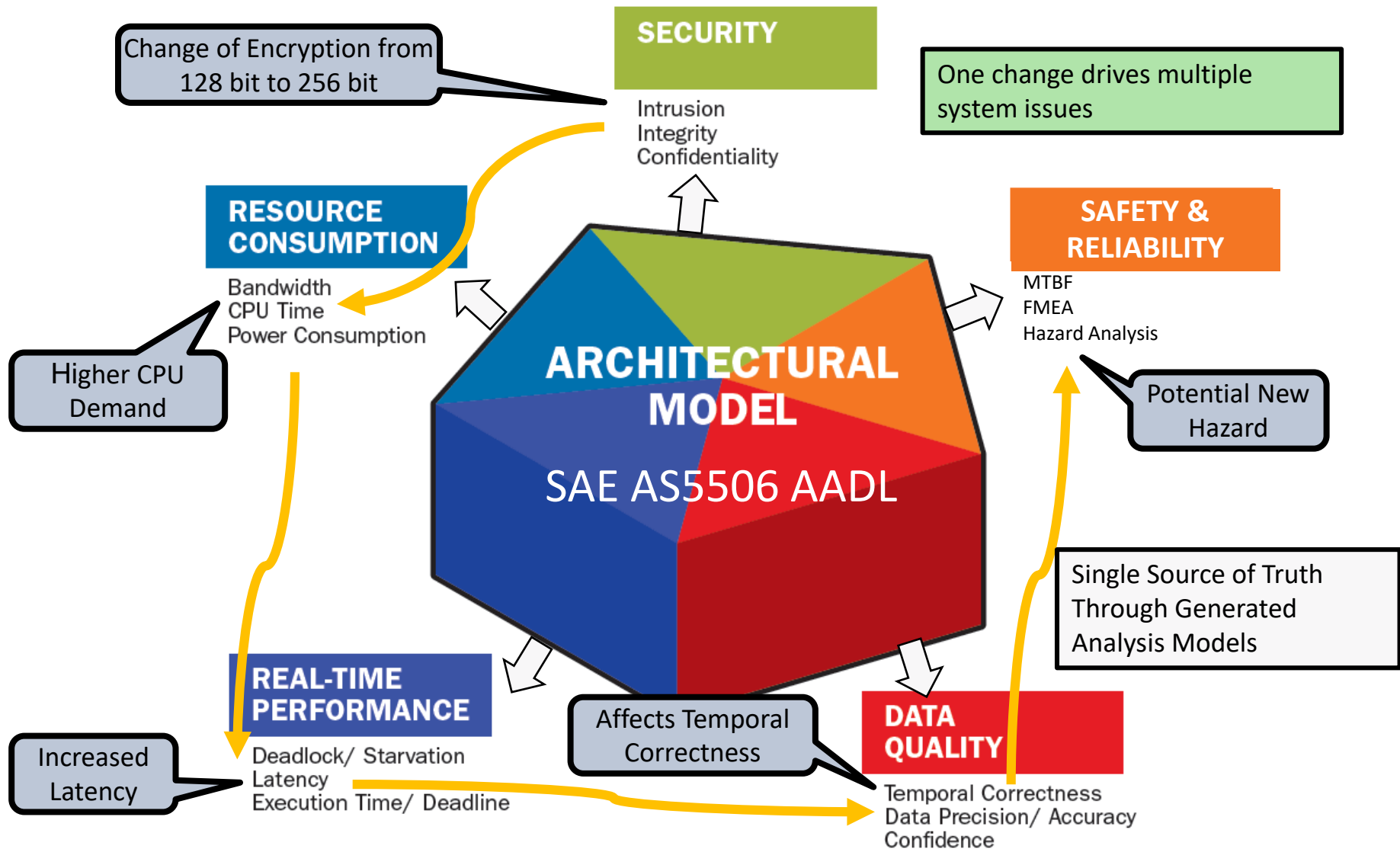
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Incremental System Assurance throughout Development

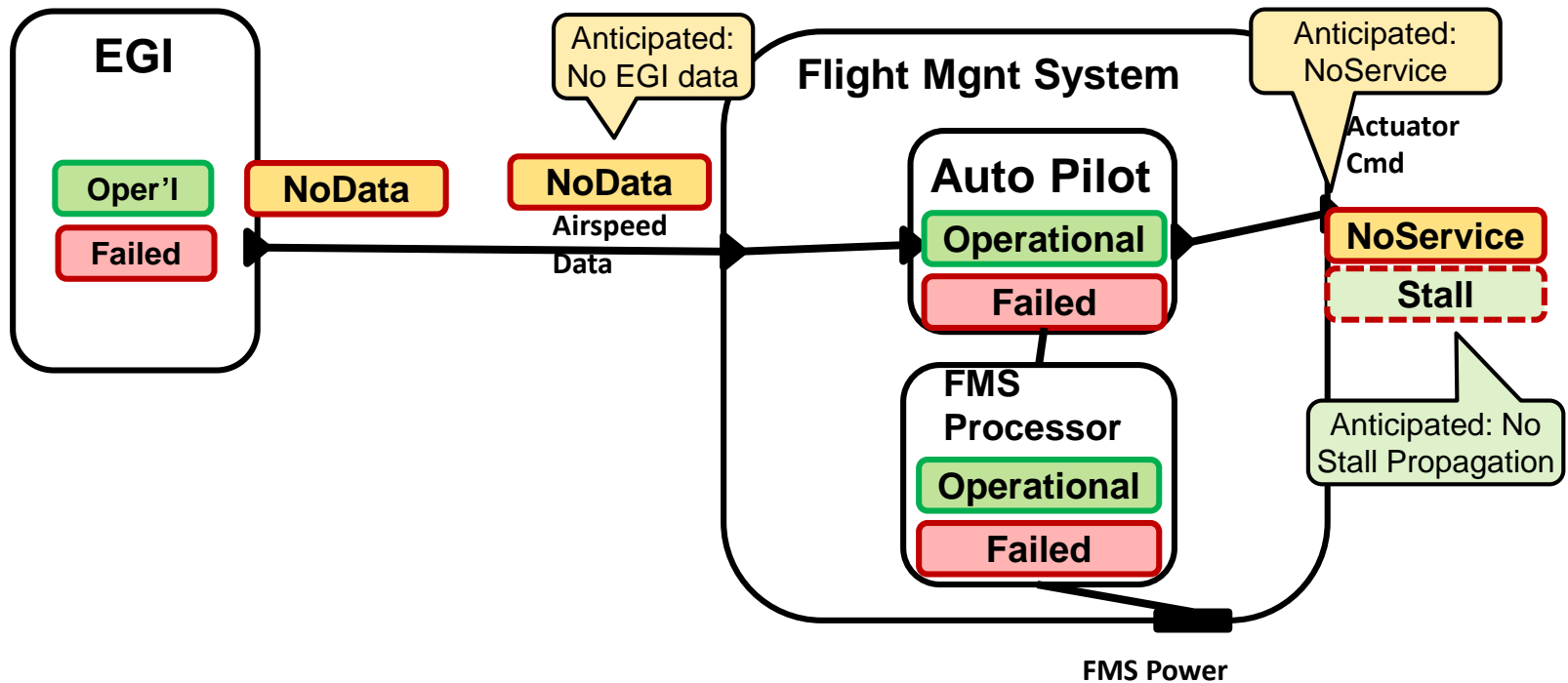
Conclusion



Analyzable Architecture Models Discover System Level Issues Early in Development



Original Preliminary System Safety Analysis (PSSA)



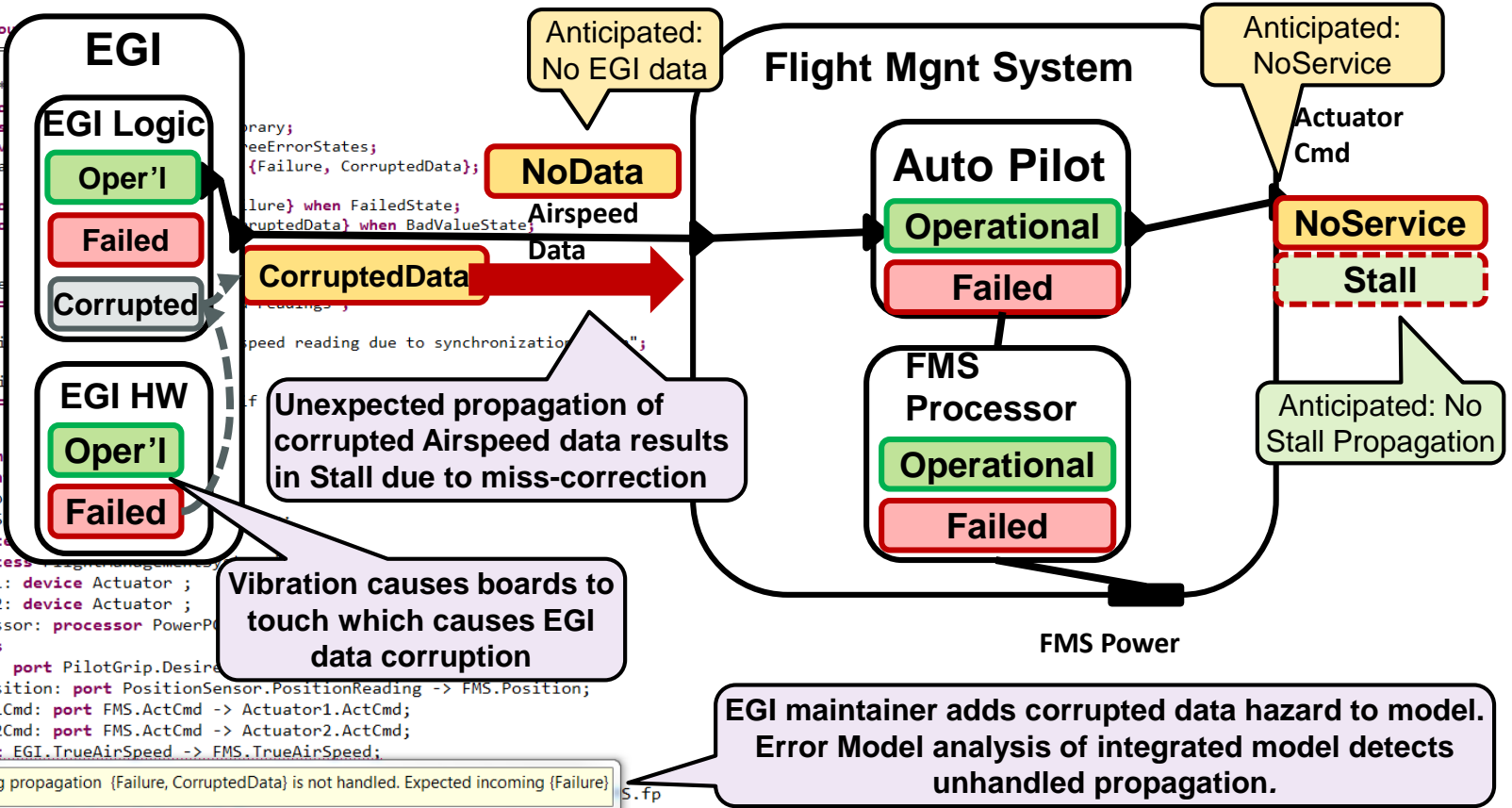
System engineering activity with focus on failing components

Discovery of Unexpected PSSA Hazard through Repeated Virtual Integration

```

system EGI
features
  trueairspeed: out data port DataDictionary::Velocity;
flows
  f1: flow source
    Latency =
  };
annex EMV2 {
  error propagation
  use types
  use behavior
  true
flows
  ef1: error
  ef2: error
properties
  EMV2::hazard
  [
  crossreference
  failure =
  phase =>
  description
  severity
  criticality
  comment =
system implementation
subcomponents
  PilotGrip
  PositionSensor
  EGI: system
  FMS: processor
  Actuator1: device
  Actuator2: device
  FMSProcessor: processor
connections
  pilotCmd: port PilotGrip.DesiredPosition -> FMS.Position;
  sensedPosition: port PositionSensor.PositionReading -> FMS.Position;
  Actuator1Cmd: port FMS.ActCmd -> Actuator1.ActCmd;
  Actuator2Cmd: port FMS.ActCmd -> Actuator2.ActCmd;
  vtx: port EGI.TrueAirSpeed -> FMS.TrueAirSpeed;
}
}
{
  Latency => 15 ms .. 20 ms;
};

```



⊗ Outgoing propagation (Failure, CorruptedData) is not handled. Expected incoming (Failure)



Automated FMEA Experience

Failure Modes and Effects Analyses are rigorous and comprehensive reliability and safety design evaluations

- Required by industry standards and Government policies
- When performed manually are usually done once due to cost and schedule
- When automated allows for
 - multiple iterations from conceptual to detailed design
 - Tradeoff studies and evaluation of alternatives
 - Early identification of potential problems

ID	Item	Initial State	Initial Failure Mode	1st Level Effect	Transition	2nd Level Effect	Transition	3rd Level Effect	Severity	M
1	Sat_Bus	Working	Failure	Failed		Failed	Recovery	Working		Workin
1	Sat_Payload	Working		Working	Bus failure causes payload transition	Standby		Standby	Bus Recovery Causes Payload Transition	Workin
2	Sat_Bus	Working		Working		Working	5			
2	Sat_Payload	Working	Failure	Failed	Recovery	Working	5			

Largest analysis of satellite to date consists of 26,000 failure modes

- Includes detailed model of satellite bus
- 20 states perform failure mode
- Lonest failure mode sequences have 25 transitions (i.e., 25 effects)

Myron Hecht, Aerospace Corp.
Safety Analysis for JPL, member of DO-178C committee



Integrated Safety and Security Engineering

Safety perspective of safety-critical systems

- Integrated modular avionics: ARINC653 partitions
 - Space and time partitioning of shared resource
- Safety levels and information/control flow
 - Functional analysis: levels of coverage (MCDC for Level A)
- Fault detection, isolation, recovery (FDIR)
- Zero defect assumption not valid for software
 - New focus: analytic redundancy, resilience

Cyber security issues

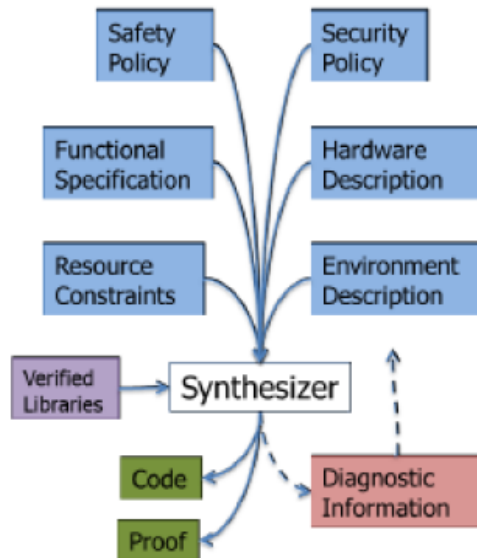
- Malicious external interactions with system
 - Via established interfaces, denial of service
- Unauthorized replacement of system component
 - Need for continuous authentication and isolation within system

Synthesize & Verify High-Assurance Systems

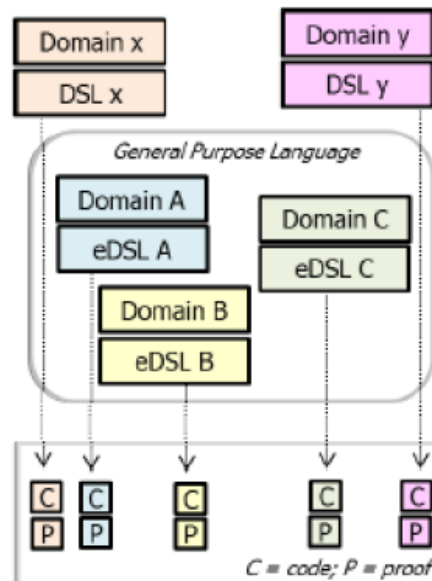
High Assurance Cyber Military Systems

Dr. Lindermann April 2018 Keynote

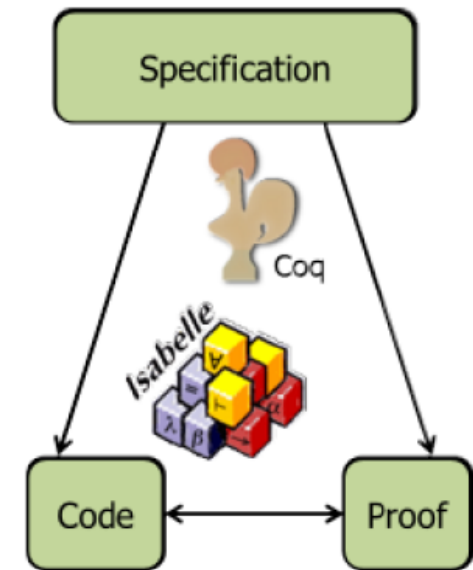
Code Synthesis



Domain Specific Languages (DSLs)



Interactive Theorem Prover as PL



Research Challenges

- Synthesis of attack-resilient control systems
- Synthesis of operating systems code
- Specification languages: function, environment, hardware, resources
- Composition/Proof engineering
- Scaling
- Attack/fault response
- V&V of complete system

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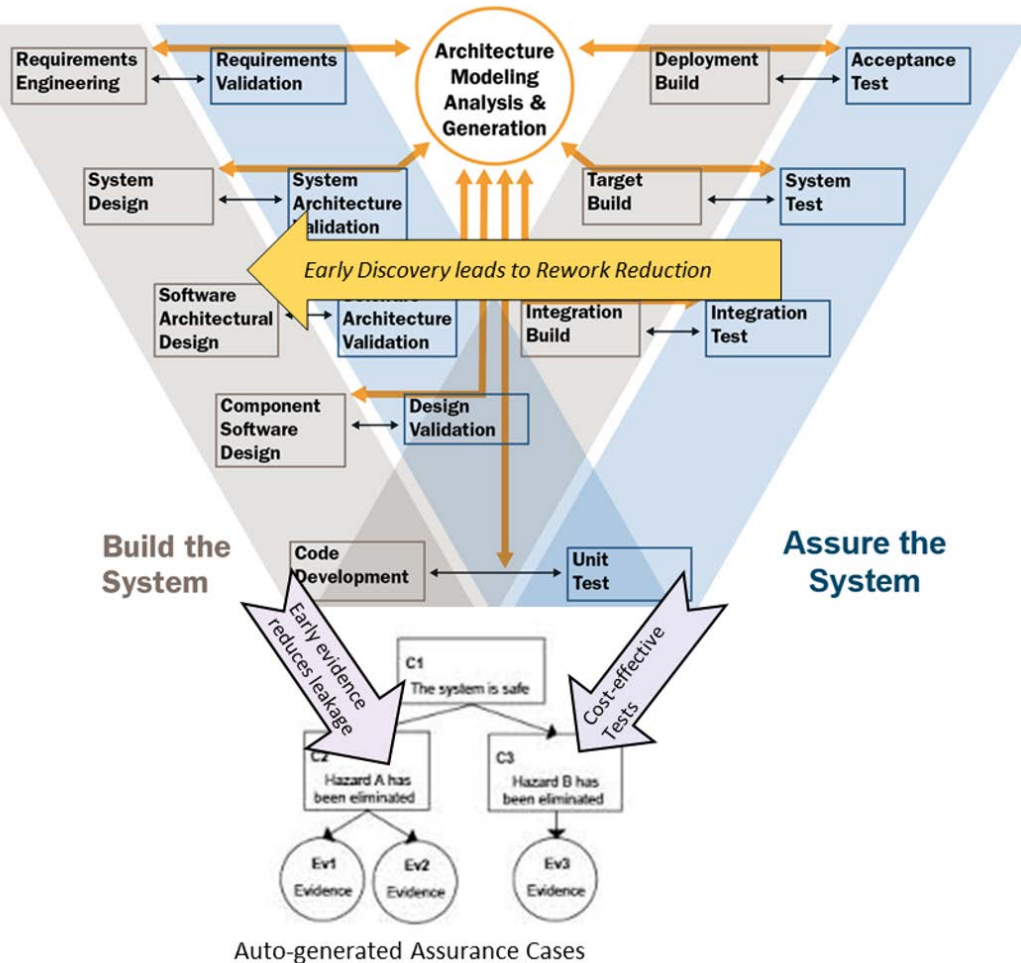
Compositional Verification to Complement Testing

▶ Incremental System Assurance throughout Development

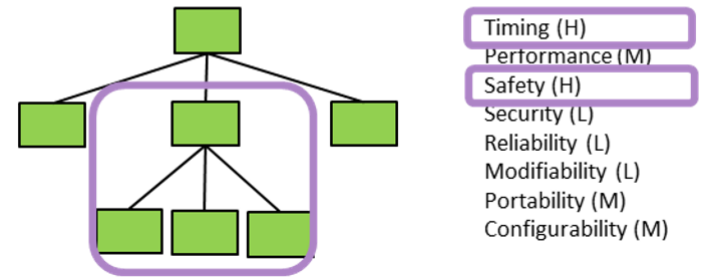
Conclusion

Three Dimensions of Incremental Assurance

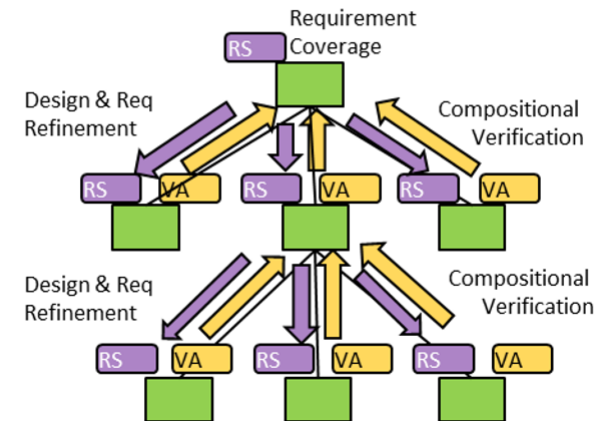
1. Incremental assurance through virtual system integration for early discovery



2. Priority focused architecture design exploration for high payoff



3. Contract-based Compositional Verification



Early Discovery and Incremental V&V through Virtual System Integration (SAVI Proof of Concept in 2009)

Aircraft: (Tier 0)

Aircraft system: (Tier 1)
 Engine, Landing Gear, Cockpit, ...
 Weight, Electrical, Fuel, Hydraulics,...

LRU/IMA System: (Tier 2)
 Hardware platform, software partitions
 Power, MIPS, RAM capacity & budgets
 End-to-end flow latency

System & SW Engineering:
 Mechatronics: Actuator & Wings
 Safety Analysis (FHA, FMEA)
 Reliability Analysis (MTTF)

Subcontracted software subsystem: (Tier 3)
 Tasks, periods, execution time
 Software allocation, schedulability
 Generated executables

OEM & Subcontractor:
 Subsystem proposal validation
 Functional integration consistency
 Data bus protocol mappings

Repeated Virtual Integration Analyses:
 Power/weight
 MIPS/RAM, Scheduling
 End-to-end latency
 Network bandwidth

Proof of Concept Demonstration and Transition by Aerospace industry initiative

- Architecture-centric model-based software and system engineering
- Architecture-centric model-based acquisition and development process
- Multi notation, multi team model repository & standardized model interchange

- Multi-tier system & software architecture (in AADL)
- Incremental end-to-end validation of system properties



Automated Incremental Assurance Workbench

Identify Assurance Hotspots Throughout Lifecycle

Issues with Assurance And/Or logic

High Abstraction

Stakeholder Goals

Abstraction Level

Low Level
Close to Implementation

Tier 0



Model

Ver Plan

Req

Tier 1



Model+1

Ver Plan

Req+1

Tier 2



Model+2

Model+2'

Ver Plan

Req+2

Assurance Case

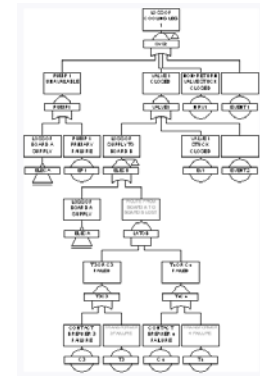
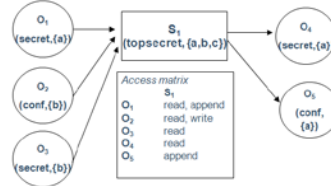
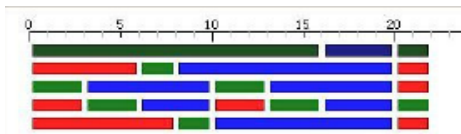


Confidence in Model-based Assurance Evidence



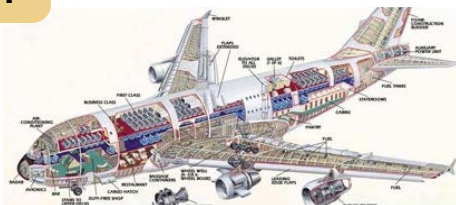
The system

Inconsistency between independently developed analytical models



System models

Confidence that model reflects implementation



System implementation

Need for Single Source of Truth



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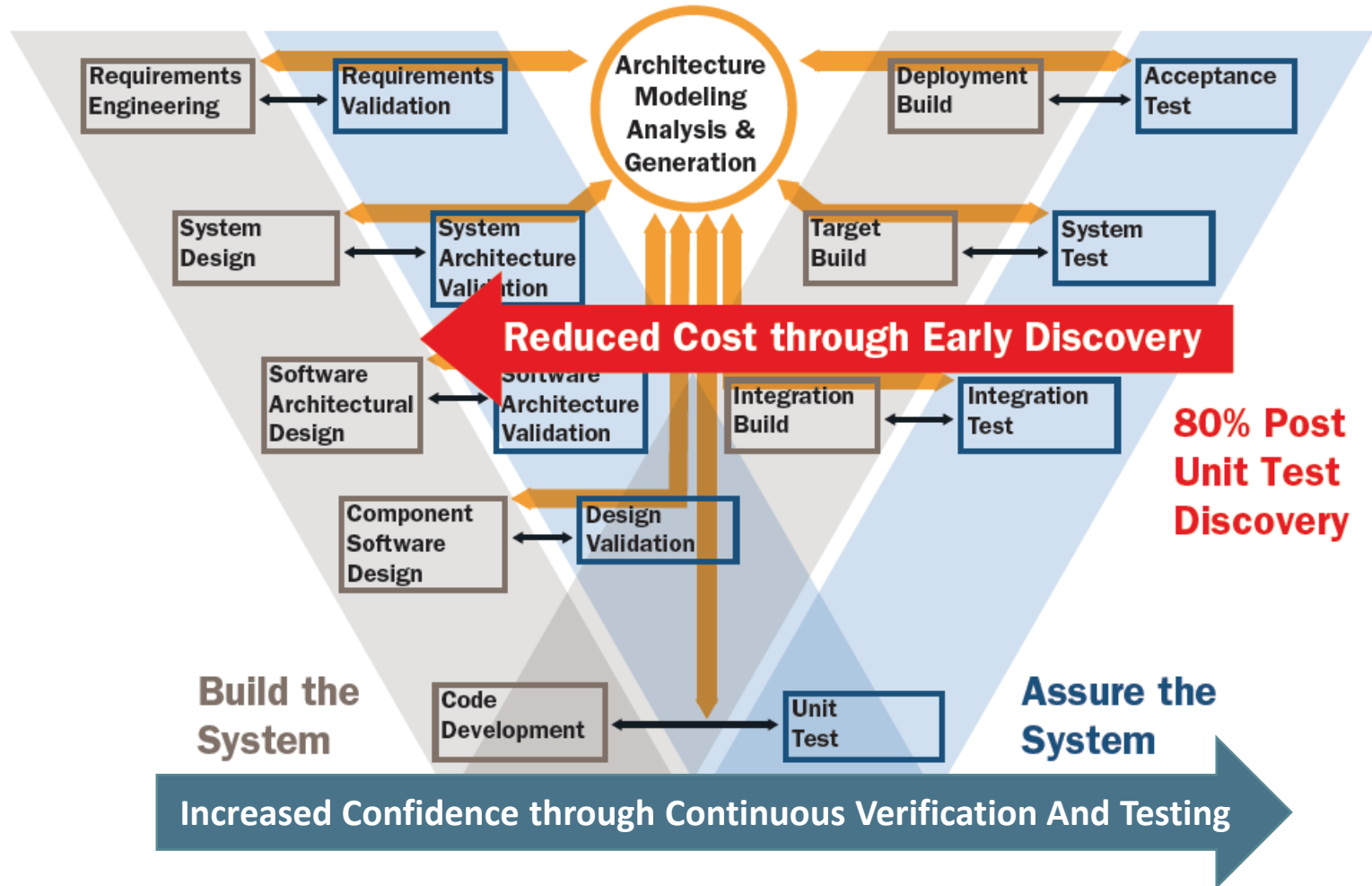
Compositional Verification to Complement Testing

Incremental System Assurance throughout Development

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Benefits of Virtual System Integration & Incremental Lifecycle Assurance



Contact Information

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