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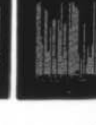
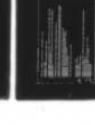
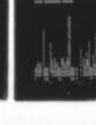
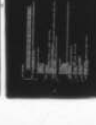
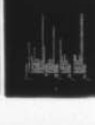
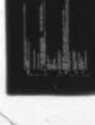
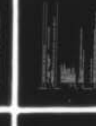
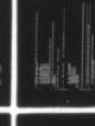
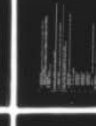
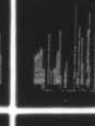
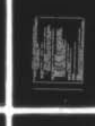
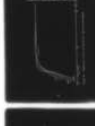
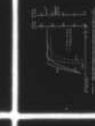
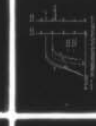
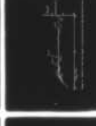
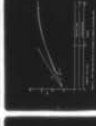
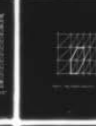
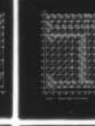
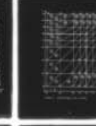
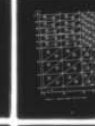
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# NAVAL POSTGRADUATE SCHOOL

## Monterey, California



# THESIS

A COMPARISON OF INTEGRATION METHODS FOR THE  
SOLUTION OF NONLINEAR REACTOR DYNAMICS PROBLEMS  
THROUGH THE USE OF FINITE ELEMENTS

by

Ralph Carroll Sheldrick

December 1976

Thesis Advisor:

David Salinas

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significantly less CPU time and comparable storage to Crank-Nicolson. This was particularly apparent as the degrees of freedom were increased. In addition, the transient solution in all cases was better than that obtained in Crank-Nicolson and compared favorably to that of Gear's method.

The other noteworthy result was in the effect of the error criterion on solution. It was shown that for a range of error from  $10^{-4}$  to 1.0, the steady state solution value remained the same. This results in a significant reduction in computer processing time since the time required decreases substantially as the error conditions imposed are relaxed.

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A Comparison of Integration Methods for the  
Solution of Nonlinear Reactor Dynamics Problems  
Through the Use of Finite Elements

by

Ralph Carroll Sheldrick  
Lieutenant Commander, United States Navy  
B.S., United States Naval Academy, 1967

Submitted in partial fulfillment of the  
requirements for the degree of

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## ABSTRACT

A comparison of numerical methods utilized by the finite element technique for solving a nonlinear nuclear reactor dynamics problem was conducted. Using the Crank-Nicolson, DVOGER (Gear) and Implicit Gear methods, the results showed the Implicit to be the superior method investigated. This is based on the fact that all three methods yielded the same steady state solutions; but, the Implicit Gear method used significantly less CPU time and comparable storage to Crank-Nicolson. This was particularly apparent as the degrees of freedom were increased. In addition, the transient solution in all cases was better than that obtained in Crank-Nicolson and compared favorably to that of Gear's method.

The other noteworthy result was in the effect of the error criterion on solution. It was shown that for a range of error from  $10^{-4}$  to 1.0, the steady state solution value remained the same. This results in a significant reduction in computer processing time since the time required decreases substantially as the error conditions imposed are relaxed.

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## LIST OF SYMBOLS

$A_{ij}$	--	matrix
$a$	--	constant
$B_{ij}$	--	matrix
$b$	--	constant
$C_{ij}$	--	matrix
$c$	--	constant
$D$	--	neutron diffusion coefficient
$H$	--	reactor height
$R$	--	reactor radius
$t$	--	time
$v$	--	neutron velocity
$\alpha$	--	reactivity temperature coefficient
$\epsilon$	--	fission energy
$\nu$	--	number of neutrons per fission
$\Sigma_a$	--	neutron absorption cross-section
$\Sigma_f$	--	neutron fission cross-section
$\psi$	--	neutron dynamic flux
$\omega$	--	constant
$\bar{h}A/V$	--	modified convection heat transfer coefficient

## I. INTRODUCTION

### A. PURPOSE

This research project has been undertaken to compare several numerical methods of solving a nonlinear nuclear reactor dynamics problem. Three methods have been investigated in this thesis, including the Crank-Nicolson, the DVOGER (Gear) and the Implicit Gear methods of solution.

A nuclear reactor dynamics problem with temperature-dependent feedback, when it entails either a non-homogeneous or multi-region reactor, results in a nonlinear field equation in space and time. This problem does not lend itself to solution by analytical means [5, 6]. However, when the physical and neutronic properties of the problem are known, a model can be formulated using the finite element method which will yield the transient and steady state flux solutions. In particular, the partial differential equations investigated were of the form

$$a \frac{\partial \psi}{\partial t} = b \nabla^2 \psi + c \psi - \omega \psi^2 \quad (1)$$

where  $a$ ,  $b$ ,  $c$ , and  $\omega$  are constants and  $\psi(r, z, t)$  is the flux. The finite element method reduces Equation (1) to the system of ordinary differential equations

$$\sum_{j=1}^N A_{ij} \dot{\psi}_j(t) = \sum_{j=1}^N B_{ij} \psi_j(t) \quad i = 1, \dots, N \quad (2)$$

when the nonlinear term is linearized. Nguyen and Salinas [5]

and more recently Olsen [6] discuss the problem and methods of solution.

In this work, a comparison of three numerical integration schemes for Equation (2) was made. The comparisons will be made on several bases: computer storage requirements, computer processing time, rapidity with which solution was obtained and the relative accuracy of the solution.

At steady state, it is expected that all three methods will provide the same solution value. This is due to the fact that at steady state the time derivative of the flux ( $\dot{\psi}$ ), Equation (2), is zero.

In order to explore the relative value of the various methods, the model has been discretized into finite element grids of various degrees of freedom (DOF). The effect on the solution by finer discretizations of the finite element has been investigated to determine if the various methods of solution are similarly influenced to provide a better solution for a finer mesh. To have a method of solution relatively independent of mesh size would greatly reduce computer storage and processing time requirements since a larger grid with fewer elements could be utilized.

In order to test the flexibility of the various equation solvers, an initial disturbance was introduced at different points in the model. This provided both a check of the ability of the method to accept a random disturbance as well as information on how rapidly a solution is obtained with a particular disturbance input. Two nodal points of the system

were considered, a point at the origin and a point on the core-reflector interface. This was done to determine if the tracking ability was consistent throughout the model.

Additional comparisons investigated include the effects of the convergence criterion on the solution for all methods and the effects of the size of the time step on the Crank-Nicolson method.

Modifications have been made to the programs provided in Ref. [6] which was the basis of this research project. In that work, core properties were arbitrarily assigned to the reflector elements at the interface. This occurred because the material and nuclear properties were provided at the nodal points rather than by elements. The properties were introduced in this manner as a means of computer storage reduction. Regardless of the mesh size, there were always fewer nodes than elements. In this project, all properties were provided on an element basis to eliminate this discrepancy while at the same time sacrificing the minor storage savings it represented.

## II. DATA GENERATORS

### A. GENERAL

The data generators formulated in this work are useable only for regular rectangular grids (Figures 1-3). Having selected the number of horizontal and vertical points for the discretization of the model, the data generators will provide the numbering of each nodal point, the horizontal and vertical position of each node and the nodal neighbors of each node.

In addition, the outer boundary nodes, at which the dynamic flux is zero, will be numbered last in order to reduce the number of equations required by the particular method of solution being used. This will substantially decrease the storage requirements. For example, in a one hundred thirty-two node discretization, only one hundred ten equations will be solved.

### B. PROPERTY INPUTS

#### 1. Purpose

A simple data generator has been provided which will produce a data deck containing the physical properties of the reactor core and reflector in the format required by the Crank-Nicolson and DVOGER (Gear) methods. These properties are neutron velocity, neutron diffusion length, neutron absorption cross-section, neutron fission cross-section and reactivity temperature coefficient.

This generator will accommodate any size mesh and will process an indefinite number of grids simultaneously.

The user must provide this routine with a data deck which contains the total number of elements in the grid and the type of each element, core or reflector. This has been simplified by the designation of all core elements by ITYPE=0 and reflector elements by ITYPE=1.

## 2. Programming Notes

a. The property values in the two regions must be provided as an integral part of the program.

b. The type of element, core or reflector, must be provided.

c. The routine will process an unlimited number of models. It, therefore, must be halted when all data has been utilized. This is done by specifying the final value of the number of elements in an IF-STOP statement.

## 3. Parameters

a. NEL - number of elements in the discretized model

b. ITYPE - type of element; ITYPE=0 is a core element and ITYPE=1 is a reflector element.

c. D - vector containing the diffusion length of the core and reflector elements.

d. SGA - vector containing the absorption cross-section of the core and reflector elements.

e. SGF - vector containing the fission cross-section of the core and reflector elements.

f. V - vector containing the neutron velocity.

g. ALPHA - vector containing the reactivity temperature coefficient.

C. NODAL POINT COORDINATES AND ELEMENT  
NODAL POINT CONNECTIVITY

1. Purpose

Nodal point positioning in the model is readily obtained in data deck form from this routine. Once the model to be investigated has been discretized into the finite element grid desired, the user provides the number of vertical and horizontal nodal points and their dimensional position along the axes. By the use of a nested loop, the nodal points are consecutively numbered and assigned the proper coordinate dimensions.

The element connectivity, that is, the nodal points for each element, is also resolved by the use of a nested loop and several counters. Each iteration will yield two elements with their respective nodal point boundaries. This will continue until the nodal points for each element have been computed.

The output of this routine will consist of two data decks. The first will provide the nodal point with its vertical and horizontal coordinates. The second will yield the element nodal point connectivity.

Omitted from the element connectivity data deck, but required by the thesis program, is the type of element, core or reflector. This must be added to the deck individually.

Example of output (See Figure 1):

nodal point coordinates			
nodal point	horizontal position		vertical position
1	0.0		0.0

element connectivity			
element	nodal point connectivity		type element*
1	1	9 10	0

\*Note: must be added manually to data card.

## 2. Programming Notes

a. The maximum number of vertical and horizontal nodal points in the discretized model must be provided.

b. The horizontal and vertical positions of the discretization must be provided.

c. The boundary points of the model will be numbered such that these nodal points are the last in the numerical sequence.

d. The routine is written such that it will process an indefinite number of models. Therefore, it must be halted when all the data has been utilized. This will be accomplished by specifying the maximum number of vertical points in the model in an IF-STOP statement.

## 3. Parameters

a. NH - maximum number of horizontal nodal points in the discretized model.

b. NV - maximum number of vertical nodal points in the discretized model.

c. X - vector containing the horizontal positions of the NH points.

d. Y - vector containing the vertical positions of the NV points.

e. SYSNOD - vector providing the total number of nodal points.

f. R - vector providing the radial position of the nodal points.

g. Z - vector providing the vertical position of the nodal points.

h. ELNOD - array providing the nodal point boundaries for each element.

i. NEL - the total number of elements in the discretized model.

#### D. NODAL NEIGHBOR CONNECTIVITY

##### 1. Purpose

This routine produces a data deck for the Crank-Nicolson and Implicit Gear methods containing each nodal point and the nodal points connected to it by the finite element discretization.

The regular rectangular grids are such that no more than six nodal points contribute to any one; therefore, an Nx7 array can be formulated for nodal neighbor connectivity. For example, nodal point 26 in the 112 element grid would be stored as follows (Figure 4):

26 17 18 27 35 34 25

and nodal point 58

58 49 50 59 57 0 0.

As written, zeros are inputted whenever a nodal point does not have six neighbors to maintain the symmetry of the matrix. This could, however, be compacted further by reading the data as a single vector with a starting point counter to indicate when the next nodal point has been reached in the sequence. This would eliminate the need for the zeros to be supplied.

This routine will provide a data deck for any rectangular grid and for an indefinite number of grids.

## 2. Algorithm

This algorithm provides a dense, compacted matrix which reduces the storage size required by the main program. The use of a finite element method allows a certain amount of compacting, i.e., the banded matrix. This is a function of the finite element size, that is, the greater the number of nodal points, the larger the band. The size of the band is determined by the largest difference between nodal neighbors plus one. For example, in Figure 1, node 11 has a maximum difference of  $20 - 2 = 18$ , while at node 16 the difference is  $43 - 7 = 36$  which is the largest difference in the forty-five node system. As the number of nodal points increases, this maximum band width also increases. In Figure 2, the maximum difference is at node 16 ( $67 - 7 = 60$ ); and, in Figure 3, the maximum difference of  $124 - 10 = 114$  exists at node 22. Therefore, although the size of the system is reduced from  $N \times N$  to  $N \times q$ , the size is not constant and may

not be small. In general, if one used banded storage, the numbering would proceed consecutively in the direction of fewest nodes. However, this would require the identification of those nodes on the boundary since they are of constant value and do not require integration.

In the particular scheme of discretization utilized in this project, no nodal point has more than six nodal point contributors. This allows the matrix to be compacted further from  $N \times Q$  to  $N \times 7$ .

### 3. Programming Notes

a. The total number of nodal points, the maximum number of nodal point contributors and the maximum number of horizontal and vertical nodal points must be provided.

b. The routine will satisfy any rectangular grid. However, it must be instructed when a sufficient number of nodal points have been generated. This is accomplished by using the total number of nodal points desired in an IF-GO TO statement.

c. This routine requires a positive means of stopping. This is accomplished by the use of the maximum number of vertical nodal points in the last data set in an IF-STOP statement.

d. The output of this routine will place the central nodal point first with the contributors following.

### 4. Parameters

a. NV - number of vertical nodal points in the model.

b. NH - number of horizontal nodal points in the model.

c. NVH - total number of nodal points in the model.

$NVH = NV \times NH.$

d. LCON - maximum number of contributing nodal points.

(LCON = 7).

e. MNOD - an array, NVH x LCON, providing the central and contributing nodal points.

### III. CRANK-NICOLSON METHOD OF SOLUTION

#### A. DESCRIPTION

The Crank-Nicolson formulation is a numerical method originally presented by J. Crank and P. Nicolson in 1947. Crank-Nicolson, being an implicit method, does not require the inverse of the matrix to be calculated. Therefore, advantage is taken of the sparse matrix inherently provided by the finite element method. If the inverse of the sparse matrix is formed, it will be full.

The Crank-Nicolson method will solve the set of linear differential equations represented by

$$\sum_{j=1}^N A_{ij} \dot{\psi}_j(t) = \sum_{j=1}^N B_{ij} \psi_j(t) \quad i=1, \dots, N. \quad (3)$$

Crank-Nicolson, after some algebra, yields

$$A \left[ \frac{\psi_t - \psi_{t-\Delta t}}{\Delta t} \right] = C \left[ \frac{\psi_t + \psi_{t-\Delta t}}{2} \right]. \quad (4)$$

The implicit system (4) may be solved by an iterative process; in this case, the Gauss-Seidel method is used to solve the set of simultaneous algebraic equations (4) formulated. Of all the stable numerical methods in the case of single step implicit, Crank-Nicolson has the smallest truncation error [2]. The Gauss-Seidel method of iteration is continuously using the newest solution values allowing convergence to the solution to be the most rapid.

## B. EFFECT OF DEGREES OF FREEDOM

As might be expected, as the grid size became finer, the computer core requirements increase. In addition, more time is required for the problem to reach a steady state solution. At the same time, as shown in Figures 5-7, the solution values converged, as the mesh size became finer, to a more accurate solution.

As shown in Table I and Figure 8, storage and processing time, that is, CPU time to reach a steady state solution, increased markedly between a forty-five node grid and a one hundred thirty-two node grid. This yielded a fifty-nine percent increase in storage and better than a five hundred percent increase in processing time. At the same time there is an improvement in the solution of only four and one-half percent.

Comparison to a seventy-two node grid yielded far more satisfying results. At seventy-two nodes, there is an eighty percent increase in CPU time and a sixteen percent increase in storage requirements while obtaining a two and one-half percent improvement in the solution.

Similar results were obtained in problem time to solution. For one hundred thirty-two nodes, the time more than doubles; while for seventy-two nodes, the time was increased by twenty percent.

## C. EFFECT OF ERROR CRITERION

The error criterion used throughout this thesis informs the integration routine when it has achieved sufficient

agreement between iterates (i.e., when it can stop and go to the next iteration step).

In Crank-Nicolson, the error criterion simply consisted of the difference in successive Gauss-Seidel iterates divided by the current iterative value.

The error criterion was varied from  $10^{-1}$  to  $10^{-4}$  to observe the effects on the solution and computer requirements. For Crank-Nicolson the steady state solution value remained the same regardless of the error criterion. This demonstrated that the extra time required to satisfy a tighter error criterion at each iteration was not necessary to reach a satisfactory steady state solution. As was expected, the computer processing time did increase significantly as the error criterion was made smaller. An unusual result occurred when the error criterion was set to  $10^{-4}$ . This particular value resulted in a processing time less than that required for  $10^{-1}$ . It would appear that this might have been the result of two contributing factors. With the closeness of the error, each time an iteration was performed, it was using a better solution, and each new time step also had a better solution value. Although there were many more time steps required for this error criterion, the steady state value was rapidly approached because fewer iterations were required at each new time. These results are shown in Table II. These results for an error criterion between  $10^{-1}$  and  $10^{-4}$  do not imply that this would be the result obtained for all problems of this type.

#### D. EFFECT OF INCREASE IN TIME STEP SIZE

In this method, the time step can be increased or not depending on the solution. If a solution is not obtained with a particular time step, that same time step is reduced; and the routine attempts to attain solution with the smaller time step. This continues until either a satisfactory solution value is obtained or the built in default value is reached which terminates the routine. When a solution is attained, the routine compares the number of iterations required to reach that value to that of the previous time step and to a specified number of iterations. If the current number of iterations is less than either of those numbers, the program increases the time step by an input constant value.

The change made to the initial time step as the steady state was approached was the critical value if the solution was to converge to a steady state. It appeared that Crank-Nicolson was particularly sensitive to the amount the time step was increased as the solution was approached (Figure 9). The method was unable to recover if, when nearing the knee of the curve, the size of the increase was such that the steady state value was exceeded. Once the solution was passed, the results indicated a diverging oscillation about the steady state value.

Several values of the initial time step were utilized in all grids to attempt to locate the steady state. The results of these trials are provided in Table III. It is noteworthy

that if the increase was greater than 1.2 times the initial step size, the steady state would never be achieved. Whether this would be the case for all grids is not obvious. This is particularly evident in view of the results obtained by Olsen [6], who utilized an increase of 1.5 times the initial step for a thirty-eight node grid. It would, therefore, seem that a trial and error approach would be necessary until an increase in step size resulted in a steady state solution.

#### IV. DVOGER (GEAR) METHOD OF SOLUTION

##### A. DESCRIPTION

The DVOGER (Gear) routine is an IMSL library routine which integrates a system of explicit first order differential equations. Within this library routine is the capability of solving both stiff and nonstiff systems by selection of an indicator. In solving the nonstiff system, the Adams predictor-corrector is utilized. For the stiff system, Gear's predictor-corrector is used. Gear's method computes the Jacobian for the system of ordinary differential equations in order to optimize the time step for each integration.

##### B. EFFECT OF DEGREES OF FREEDOM

Gear's method placed a much greater demand on the computer in terms of processing time and storage than the other methods considered. This was primarily due to three causes: 1) the calculation of the Jacobian, 2) the transformation of Equation (3) to explicit form, and 3) the initial absolute value of the solution. The first two items require a substantial increase in both CPU time and computer storage. Several variances were implemented in this program to determine if better use could be made of the method to make it competitive with Crank-Nicolson or the Implicit Gear methods.

When the system was considered nonstiff, the storage requirements decreased since the Jacobian was not calculated; but, the processing time increased by one order of magnitude

when compared to the stiff system treatment due to the small time steps taken to solution. The routine calls for an initial absolute solution value of one to be supplied. This, however, adversely affects the progress of the problem since this value of one is initially compared to values of  $10^{16}$ . This limits the time step taken by the routine initially and continues to affect the progress until the steady state solution is neared. When an initial value of  $10^{14}$  was utilized, the processing time was decreased by forty percent. This was due to the fact that larger time steps were allowed from the outset, since the previous solution value (in this case the initial value) was comparable to the value obtained in the first calculation.

At best, when utilizing Gear's method, the processing times were two orders of magnitude greater than that required for the Implicit Gear method and twenty times the Crank-Nicolson processing times for the 132 DOF system, as shown in Table I. Core requirements were also increased significantly, particularly at one hundred thirty-two degrees of freedom. For this grid, DVOGER (Gear) was approximately three times Crank-Nicolson and the Implicit Gear core requirements.

Gear's method provided the same steady state solution values as the other two methods, and the transient curves were very similar to those obtained in the Implicit Gear method (Figures 10-15).

### C. EFFECT OF ERROR CRITERION

In Gear's method, the time step size is adjusted so that the single step error estimate divided by the previous maximum solution value is less than the error criterion in the Euclidean norm. The single step error estimate is a multiple of the difference between the predicted and corrected values of the variable.

The error criterion, Table II, was varied for this method as it was for the others. The results were the same; as the criterion was tightened, CPU time increased. In this case, however, the time became excessive very rapidly, more than an hour for  $10^{-3}$  and almost four hours for  $10^{-4}$  for the model with 45 DOF. Similarly, the solution values were the same regardless of the criterion utilized.

## V. IMPLICIT GEAR METHOD OF SOLUTION

### A. DESCRIPTION

This numerical method is particularly useful for the solution of large, sparse systems of implicit, stiff differential equations [4]. In contrast to Gear's method, the Implicit Gear method eliminates the need of an exact  $N \times N$  Jacobian, but rather requires only an exact  $N \times 7$  Jacobian. This is due to the fact that Equations (3) are handled directly, thereby retaining the sparseness of the matrix.

Gear's predictor-corrector method and the compact matrix makes use of storage because the implicit Equations (3) are handled directly. This results in very efficient use of the computer both in terms of storage and processing time requirements.

The user is required to provide a subroutine that evaluates the system of equations being investigated, as well as, for efficiency, a subroutine to evaluate the Jacobian of the system of equations.

This program is a modification by Franke [4] to the routine DFASUB [7]. One of the major changes to the routine is in the treatment of the error. In the Implicit Gear method, instead of using the Euclidean norm, the root mean square norm is utilized. This is no more than the Euclidean norm divided by the square root of the number of components. In addition, the maximum value of the component is updated before the norm of the relative error is computed.

## B. EFFECT OF DEGREES OF FREEDOM

The steady state values obtained in this method were the same as those obtained in Crank-Nicolson and DVOGER (Gear) as shown in Figures 11-18. This result does not imply that these methods will always give identical steady state solutions. Certainly they do give different transient solutions. The major advantages of this method are 1) the computer processing time was reduced as the grid size became finer and 2) the storage requirements are slightly greater than that required in the Crank-Nicolson method due to the computation of the Jacobian and the storage of up to seven past solutions which control the size of the time step. Implicit Gear requires about  $N \times 25$  more storage locations than Crank-Nicolson. This amounts to about only thirteen thousand bytes for the one hundred thirty-two degrees of freedom system (single precision). The magnitudes of the increase in processing time dropped dramatically in this method. For the same initial disturbance, the time increased eighty percent for the one hundred thirty-two node grid and twenty-four percent for the seventy-two degrees of freedom. In addition, Implicit Gear gave a more accurate transient solution. The results of this method are shown in Table I and comparison to the other methods will be made in Chapter VI.

## C. EFFECT OF ERROR CRITERION

In Implicit Gear, the error criterion is defined as in DVOGER (Gear) except that the root mean square of the Euclidean

norm and the updated maximum solution value are used in the computation.

The error criterion was varied substantially to determine what effect it had on computer storage and time requirements, as well as its effect on the accuracy of solution. As shown in Table II and Figures 19-20, a wide range of convergence, 1.0 to  $10^{-4}$ , was investigated with interesting results. Expectedly, the processing time did increase as a tighter error criterion was required. However, the criterion had little effect on the track through the transient solution for values of less than  $10^{-1}$ . When using an error of one-half and one, the transient solution varied substantially. The same steady state solution was obtained, although at a later point in problem time as shown in Figure 19. For the stiffer error criterion requirements, the processing time necessary is doubled by utilizing a criterion of  $10^{-4}$  instead of a value greater than  $10^{-1}$ . This, certainly, did not appear to warrant the closer tolerance level.

## VI. RESULTS

### A. COMPARISON OF TIME AND STORAGE REQUIREMENTS

With all three methods yielding the same steady state solution (Figures 10-18), the comparison of methods became one of time and storage requirements placed on the computer rather than one based on which provided the best solution. In the transient stage, Gear tracked slightly better than Implicit Gear, but both Gear and Implicit Gear tracked better than Crank-Nicolson (See Figures 10-18). For the three systems of equations utilized, the Implicit Gear method performed significantly better than either Crank-Nicolson or DVOGER (Gear) in CPU time and was comparable to Crank-Nicolson and superior to Gear in storage requirements (Table I). This becomes more significant as the number of degrees of freedom increase. The Implicit Gear is less sensitive to the increase in size of the system of equations as shown in Figure 8.

Comparing the forty-five and the one hundred thirty-two degrees of freedom, the core requirements increased slightly and the time doubled for the Implicit Gear method. In contrast, for Crank-Nicolson, the time increased by six times and the core by 100K bytes; and, for DVOGER (Gear), there was a ten times and 500K byte increase in time and storage respectively.

#### B. DEGREE OF FREEDOM EFFECT ON SOLUTION

Varying the degrees of freedom resulted in a better solution value as shown in Figures 5-7 and 21-26. There was a four and one-half percent better solution obtained for the one hundred thirty-two degrees of freedom compared to the forty-five degrees of freedom. This better solution is very costly both in terms of computer core requirements and CPU time (Figure 8).

#### C. ERROR CRITERION

Error criterion was varied for all methods to determine the effect on solution and computer. As might be expected, the computer processing time increased for all methods as more rigid tolerance was imposed (Table II). However, although the criterion was made very close, there was no appreciable effect on the transient solution until the error criterion was greater than  $10^{-1}$  and no effect on the steady state solution. This is significant in that, as stated above, the processing time increases greatly as the criterion is tightened. This shows that for the particular problem considered here, some close error criteria yield no advantage, only the disadvantage of requiring more time in the computer.

#### D. INITIAL DISTURBANCES

The input of various initial disturbances (central, skewed at the core-reflector interface and uniform throughout the core) yielded the same steady state solution value. The transient solution varied substantially due to the input

position of the disturbance. In most cases, due to the magnitude and scope of the initial disturbance, the uniform disturbance required the least processing time to reach the steady state value.

#### E. COMPARISON OF SOLUTION TRACKING

Two nodal points were investigated graphically for each disturbance, integration method and DOF, as shown in Figures 5-7, 10-18 and 21-26. At these test points, the methods performed similarly in all cases in both the transient and steady state solutions.

## VII. TEST PROBLEMS

### A. PHYSICAL MODEL

A cylindrical reactor with the dimensions and properties given in Table IV was used as a model. A radial slice of this model was discretized by various finite element grids as shown in Figures 1-3.

### B. COMPUTER PROCESSING CONSIDERATIONS

The programs presented in this thesis, Appendix A, were written in the FORTRAN IV language and all computer runs, Table V, processed on the IBM 360/67 computer using the FORTRAN 'G' compiler. It is expected that the FORTRAN 'H' compiler would have supplied results similar to those obtained with the FORTRAN 'G' compiler. This was not done because the FORTRAN 'H' compiler requires 350K bytes as a minimum core requirement. The majority of the runs performed in this thesis required significantly less than 300K storage. Single precision (six to seven significant digits) was used throughout this research. As has been shown [6], double precision solutions are at variance by less than 0.01 percent with single precision solutions.

### C. PROBLEM ANALYSIS

The techniques used by Olsen [6], modified as described, and the Implicit Gear method [4] were utilized to solve the problem delineated in Ref. [5]. The problem was approached

using finite element discretizations with forty-five, seventy-two and one hundred thirty-two degrees of freedom and providing an initial disturbance. Three initial disturbances were provided as follows: central at the core center ( $R = 0$  cm.,  $Z = 0$  cm.); uniform throughout the core; and, a skewed disturbance at the core-reflector interface ( $R = 60$  cm.,  $Z = 0$  cm.).

#### D. PROGRAM USAGE

In order to properly utilize the routines presented in this thesis, several points must be considered.

In the Crank-Nicolson routine, there was difficulty in obtaining a steady state solution. The ultimate cause was the size with which the previous time step increases. This requires a trial and error approach for the particular grid size. For this project, an increase of ten or twenty percent yields satisfactory results; but, anything larger results in a divergent oscillation about the steady state value.

In all three methods, the dimensioning should be set by the particular problem being solved, i.e., determined by the degrees of freedom. If this is done, most efficient use will be made of the computer, and the user will experience better turn around time on the equipment.

In the use of the Implicit Gear method, the data cards for nodal neighbor connectivity must have the contributing nodes listed consecutively, with the zeros, used for nodes not having six neighbors, being last on the cards. For example, in the one hundred thirty-two degrees of freedom system

(Figure 3), the nodal neighbor connectivity for node 122 would be written as

122	123	11	0	0	0	0.
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The data deck arrangement for all three methods is given in Table VI.

## VIII. CONCLUSIONS AND RECOMMENDATIONS

### A. CONCLUSIONS

It has been shown in this research project that of the three methods investigated in the solution of the test problem that the Implicit Gear performed in a superior manner in CPU time requirements and was comparable to the storage requirements of Crank-Nicolson. In addition, Implicit Gear was the least sensitive to the change in the degrees of freedom.

The method used had no effect on the steady state solution value obtained as all three yielded the same results for the same DOF. In the transient solution, Implicit Gear conformed very closely to DVOGER (Gear). As shown in Figures 10-15, as the DOF increased, the transient solutions moved in the direction of the transient solution of Gear's method.

As the DOF in the discretized finite element was increased, a better solution was obtained. This, however, was counteracted by the fact that for the small increase in accuracy obtained by the finer mesh, the time and storage need of the computer increased significantly.

A very important result of this project was the effect of varying the error requirements of the problem. This yielded essentially no change in the accuracy of the transient solution for a range of  $10^{-1}$  to  $10^{-4}$  and no variance in the steady state solution for a range from 1.0 to  $10^{-4}$ . In using these methods of integration, the error criterion selected would

be a function of the solution desired. If the user is only concerned with the steady state solution, the tolerance could be relaxed to 1.0. However, if the transient solution is needed, an error of  $10^{-1}$  appears to be the least rigid value which will still provide a satisfactory track to steady state.

#### B. RECOMMENDATIONS

It would be of value to investigate further the Implicit Gear method through the use of other problems solving a non-linear system of ordinary differential equations. In addition, larger degrees of freedom should be attempted to evaluate any limitations that may exist in this method of integration.

TABLE I  
COMPARISON OF SOLUTION METHODS

METHOD	CORE (K)	PROCESSING TIME* (min.)		
		Central	Skewed	Uniform
45 Nodes				
CRANKO	158	1.67	1.73	3.57
DVOGER (GEAR)	148	17.8	19.5	12.5
IMPLICIT GEAR	124	1.05	1.00	1.20
72 Nodes				
CRANKO	184	3.07	5.49	2.38
DVOGER (GEAR)	244	89.5	114.1	59.5
IMPLICIT GEAR	124	1.3	1.30	1.25
132 Nodes				
CRANKO	252	9.1	9.1	9.1
DVOGER (GEAR)	610	>400	>400	>400
IMPLICIT GEAR	124	1.9	2.0	2.0

\* Processing time is that required to reach a steady state solution.

TABLE II  
EFFECTS OF SOLUTION ERROR CRITERION

45 Nodes

SOLUTION ERROR CRITERION	PROCESSING TIME (min.)	TIME AT SOLUTION (sec.)
CRANK-NICOLSON		
0.1	1.8	$3.69 \times 10^{-5}$
0.01	5.5	$3.70 \times 10^{-5}$
0.001	14.2	$3.57 \times 10^{-5}$
0.0001	0.83	$3.87 \times 10^{-5}$
IMPLICIT GEAR		
1.0	0.99	$7.56 \times 10^{-4}$
0.5	0.92	$1.48 \times 10^{-2}$
0.1	1.05	$7.06 \times 10^{-5}$
0.01	1.20	$7.64 \times 10^{-5}$
0.001	1.45	$6.36 \times 10^{-5}$
0.0001	1.58	$6.44 \times 10^{-5}$
DVOGER (GEAR)		
0.1	18.6	$5.64 \times 10^{-5}$
0.01	29.5	$5.79 \times 10^{-5}$
0.001	78.8	$5.55 \times 10^{-5}$
0.0001	>300	-

TABLE III  
EFFECTS OF VARYING THE INCREASE OF THE INITIAL TIME STEP  
IN CRANK-NICOLSON METHOD

INCREASE	CORE	PROCESSING TIME (min.)	SOLUTION TIME (sec.)
<u>45 NODES</u>			
1.1	158K	4.99	$3.18 \times 10^{-5}$
1.2	158K	1.65	$2.69 \times 10^{-5}$
1.5	184K	7.42	OSCILLATING
<u>72 NODES</u>			
1.1	252K	3.5	$4.38 \times 10^{-5}$
1.2	184K	3.05	$3.23 \times 10^{-5}$
1.3	252K	3.5	OSCILLATING
1.4	252K	9.1	OSCILLATING
1.5	184K	6.1	OSCILLATING
<u>132 NODES</u>			
1.1	252K	9.1	$5.58 \times 10^{-5}$
1.2	252K	8.4	$8.03 \times 10^{-5}$
1.5	184K	2.6	OSCILLATING

TABLE IV  
PHYSICAL CONSTANTS

SYMBOL	COMPUTER SYMBOL	DEFINITION	VALUE
	C-N, D / IG		
R	R	Total radius	90 cm.
R <sub>C</sub>	R	Core radius	60 cm.
H <sub>C</sub>	Z	Core height	160 cm.
H	Z	Total height	220 cm.
v	V/VELOCT	Neutron velocity	$4.8 \times 10^7$ cm/sec
D <sub>C</sub>	D/DSUBF	Neutron diffusion coefficient (core)	0.913 cm.
D <sub>r</sub>	D/DSUBC	Neutron diffusion coefficient (reflector)	1.20 cm.
$\Sigma_{ac}$	SGA/SGMAAF	Neutron absorption cross-section (core)	$0.01401 \text{ cm}^{-1}$
$\Sigma_{ar}$	SGA/SGMAAC	Neutron absorption cross-section (reflector)	$0.008 \text{ cm}^{-1}$
v	ZNU	Number of neutrons per fission	2.54
$\Sigma_{fc}$	SGF/SGMAF	Neutron fission cross-section (core)	$0.008 \text{ cm}^{-1}$
$\Sigma_{fr}$	SGF/ -	Neutron fission cross-section	$0.0 \text{ cm}^{-1}$
$\epsilon$	FISFAC	Fission energy	$7.652 \times 10^{-12}$ cal/fis
$\bar{h}A/V$	HBAR	Modified convection heat transfer coefficient	$0.0632 \text{ cal/cm}^3\text{-sec-}^\circ\text{C}$
$\alpha$	ALPHA	Reactivity temperature coefficient	$1 \times 10^{-5} / ^\circ\text{C}$

TABLE V  
LIST OF COMPUTER RUNS

RUN	METHOD	NODES	DISTURBANCE	COVERGENCE	TIME STEP CHANGE
1	CRANK-NICOLSON	45	CENTRAL	0.1	1.1
2					1.2
3					1.5
4				0.01	1.2
5				0.001	
6				0.0001	
7			UNIFORM	0.1	
8			SKEWED		
9		72	CENTRAL	0.1	1.0
10					1.1
11					1.2
12					1.3
13					1.4
14					1.5
15			UNIFORM		1.2
16			SKEWED		
17		132	CENTRAL	0.1	1.1
18					1.2
19					1.5
20			UNIFORM		1.2
21			SKEWED		
22	DVOGER MTH=0	45	CENTRAL	0.1	N.A.
23	DVOGER MTH=1 YMAX=1			0.1	

TABLE V (Continued)

RUN	METHOD	NODES	DISTURBANCE	CONVERGENCE	TIME STEP CHANGE
24				0.01	
25				0.001	
26	DVOGER MTH=1 YMAX=1	45	CENTRAL	0.0001	
27			UNIFORM	0.1	
28			SKEWED		
29		72	CENTRAL		
30			UNIFORM		
31			SKEWED		
32		132	CENTRAL		
33			UNIFORM		
34			SKEWED		
35	IMPLICIT GEAR	45	CENTRAL	0.1	
36				0.01	
37				0.001	
38				0.0001	
39			UNIFORM	0.1	
40			SKEWED		
41		72	CENTRAL	0.1	
42			UNIFORM		
43			SKEWED		
44		132	CENTRAL	0.1	
45			UNIFORM		
46			SKEWED		
47		45	CENTRAL	0.5	

TABLE V (Continued)

RUN	METHOD	NODES	DISTURBANCE	CONVERGENCE	TIME STEP CHANGE
48				1.0	
49	DVOGER MTH=1 YMAX=10 <sup>14</sup>	45	CENTRAL	0.1	

TABLE VI  
DATA DECK ARRANGEMENT

CRANK-NICOLSON

Title

NUMEL,NUPBP,NUMSNP,NFULEL

List of outer boundary points

MTH,MAXDER,NCOUNT

ZNU,FISFAC,HBAR,EPSVAL,ERRVAL,AFUEL

TO,H,TF,HMIN,HMAX

V - neutron velocity

D - diffusion length

SGA - absorption cross-section

SGF - fission cross-section

ALPHA - reactivity temperature coefficient

PSIIV - initial disturbance flux

System nodal points with axial and radial position

Element nodal connectivity

Nodal neighbor connectivity

DVOGER

Title

NUMEL,NUPBP,NUMSNP,NFULEL

List of outer boundary points

MTH,MAXDER,NCOUNT

ZNU,FISFAC,HBAR,EPSVAL,ERRVAL,AFUEL

TO,H,TF,HMIN,HMAX

V

TABLE VI (Continued)

D

SGA

SGF

ALPHA

PSIIV

System nodal points with radial and axial position

Element nodal connectivity

IMPLICIT GEAR

NCORE

List of core elements

} with uniform initial disturbance

Title

NUMEL,NBP,NUMSNP,NFULEL,NELROW,MXEVAL,NCOUNT,NCMPK

NDE,NL

List of outer boundary points

VELOCT,DSUBF,DSUBC,SGMAAF,SGMAAC,SGMAF,FISFAC,HBAR

ZNU,ALPHA,RMSEPS,AFUEL,TSTART,TEND,PINITV,SCALE

HMIN,HMAX

MTH,MAXDER,NPROB,NTYPE,JPLOT,NPLOT,IRUN

Nodal neighbor connectivity

System nodal points with radial and axial position

Element nodal connectivity

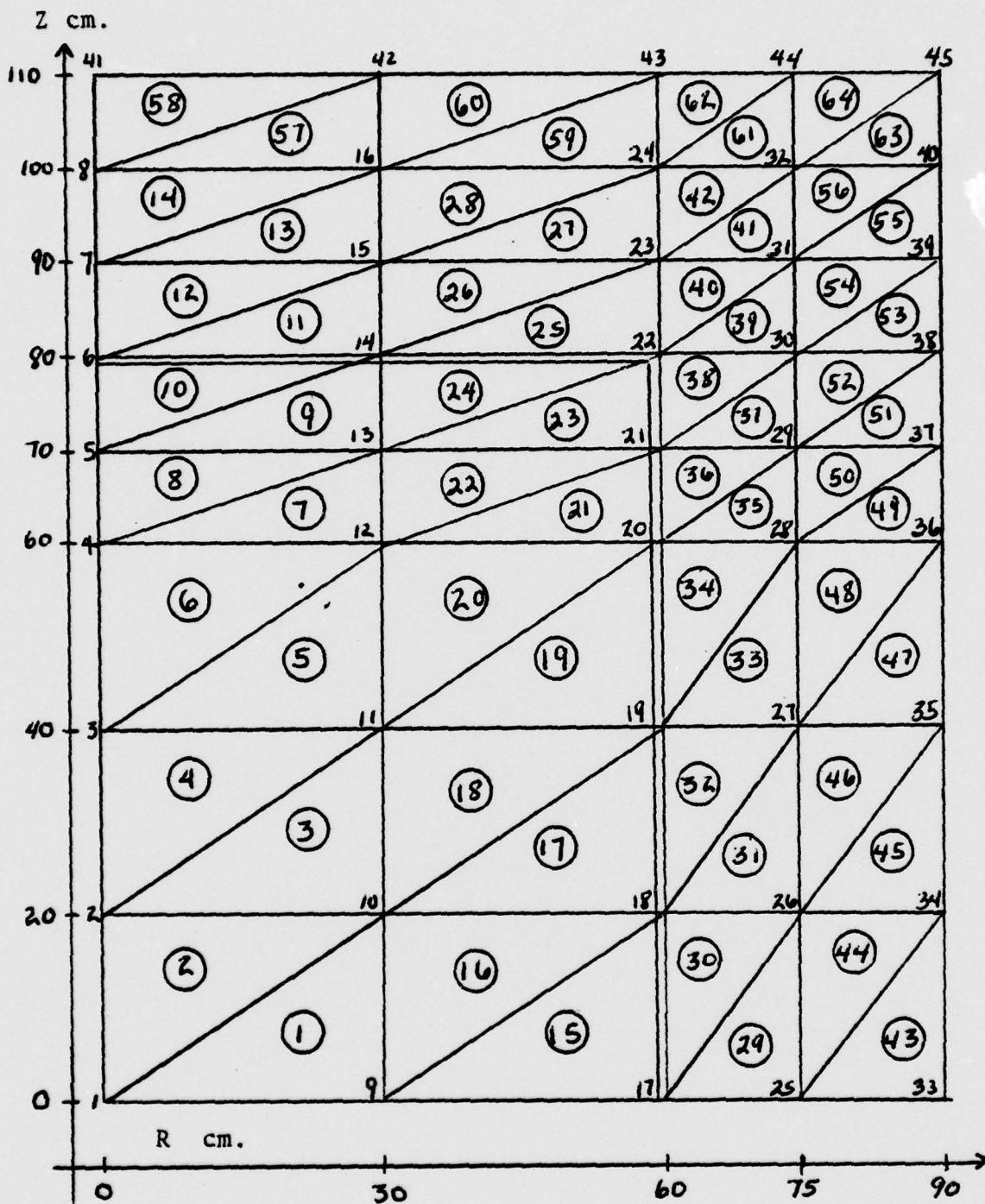


Figure 1. Reactor Model with 45 Nodes

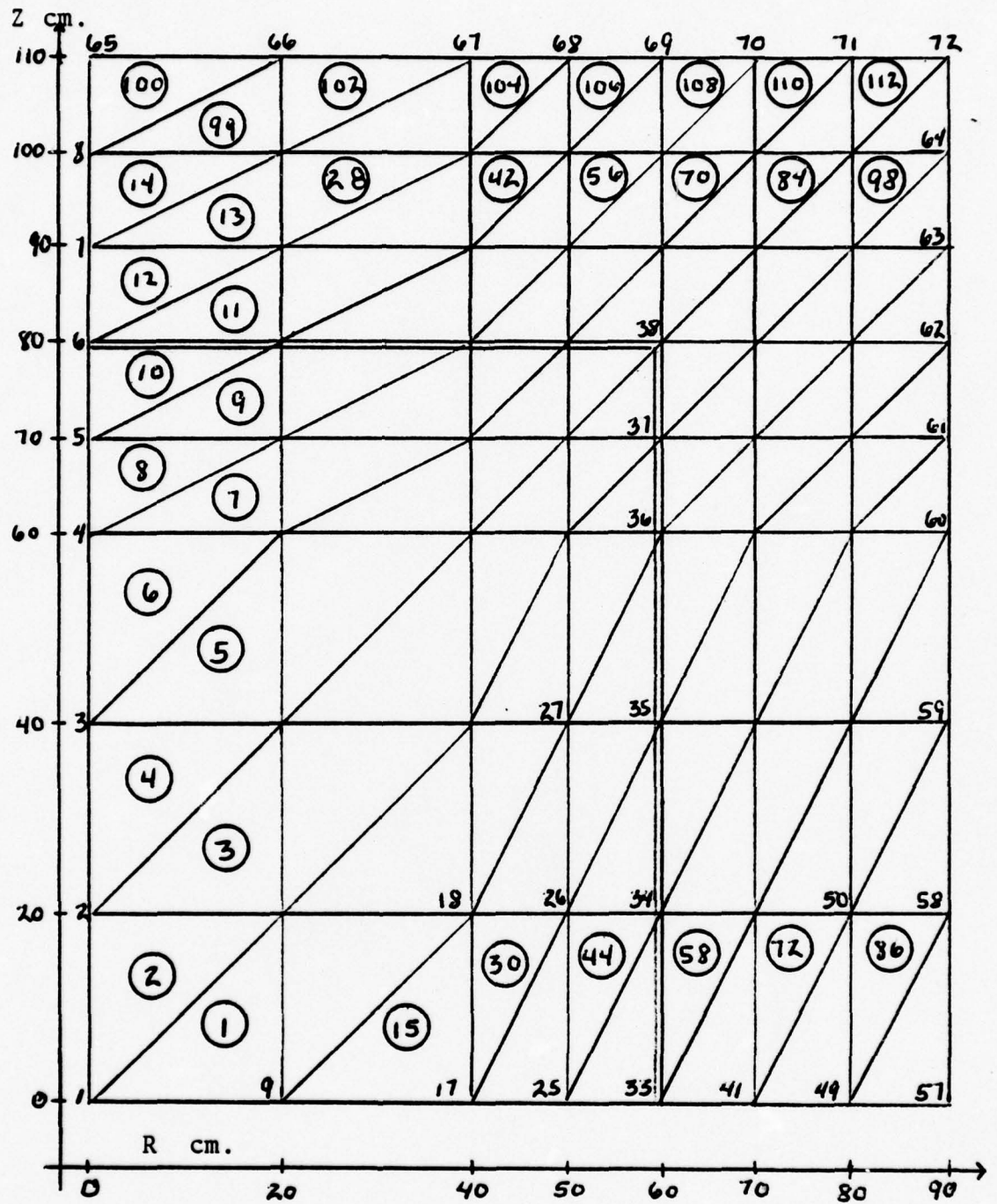


Figure 2. Reactor Model with 72 Nodes

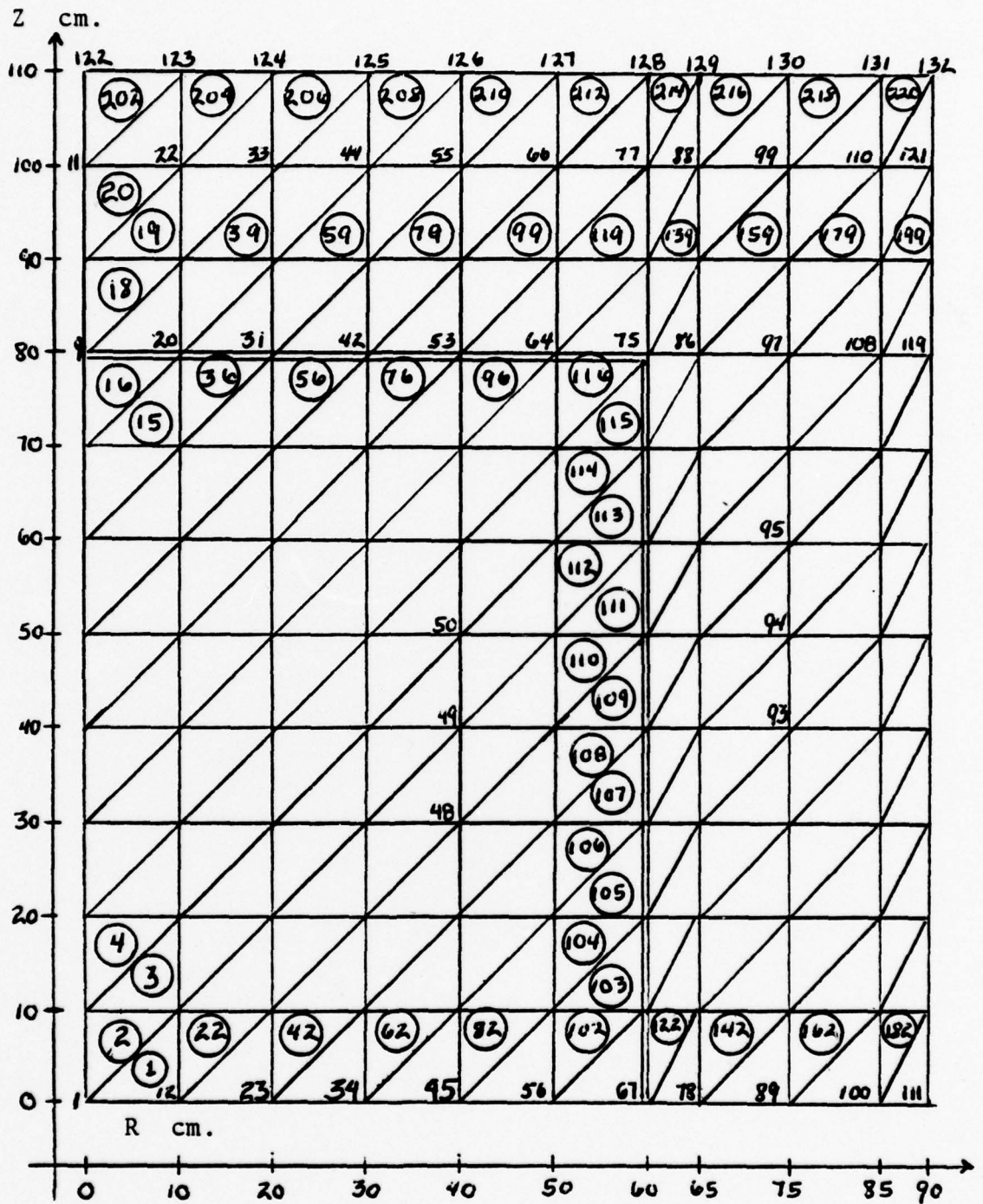


Figure 3. Reactor Model with 132 Nodes

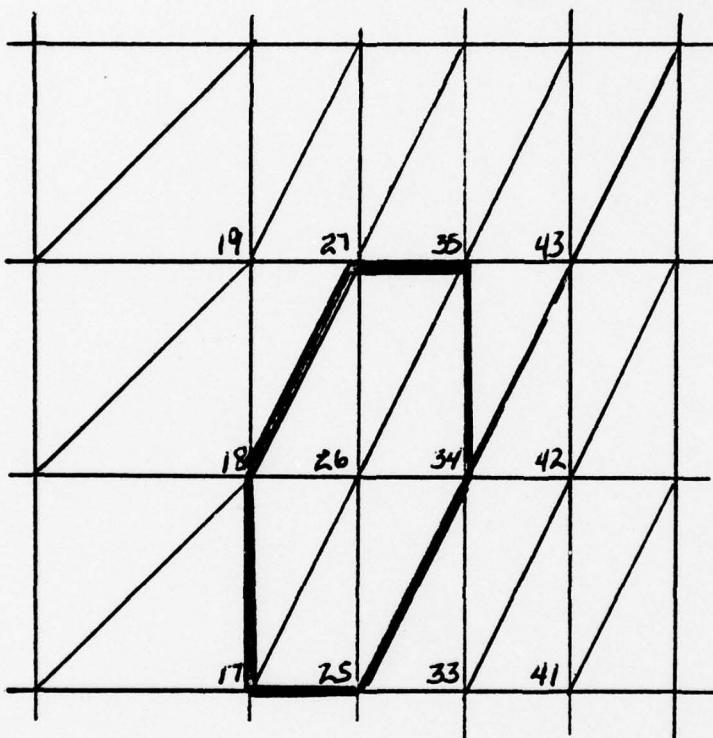


Figure 4. Nodal Neighbor Connectivity

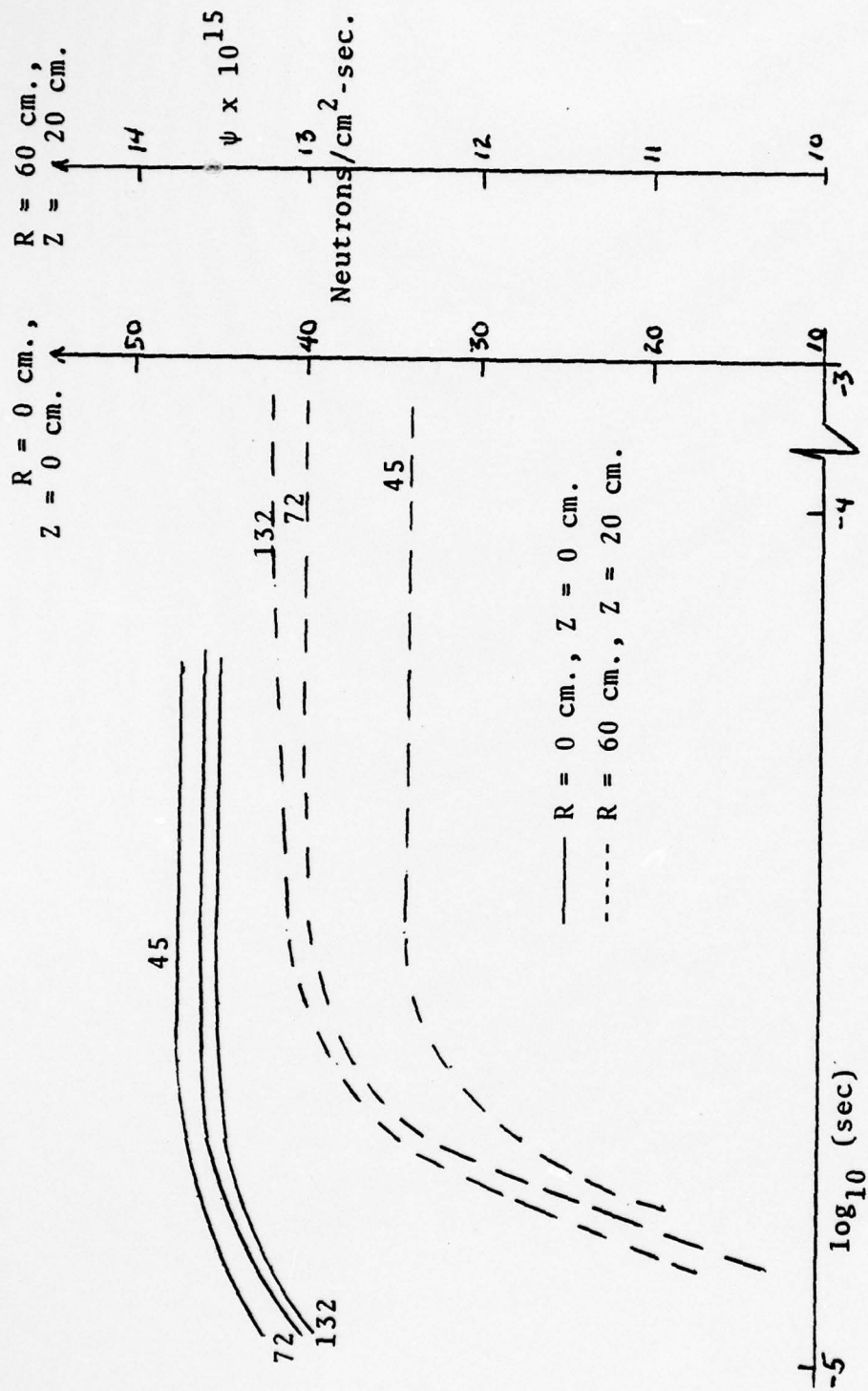


Figure 5. Time Dependent Neutron Flux for All DOF Using Crank-Nicolson Method for a Uniform Disturbance Throughout the Core.

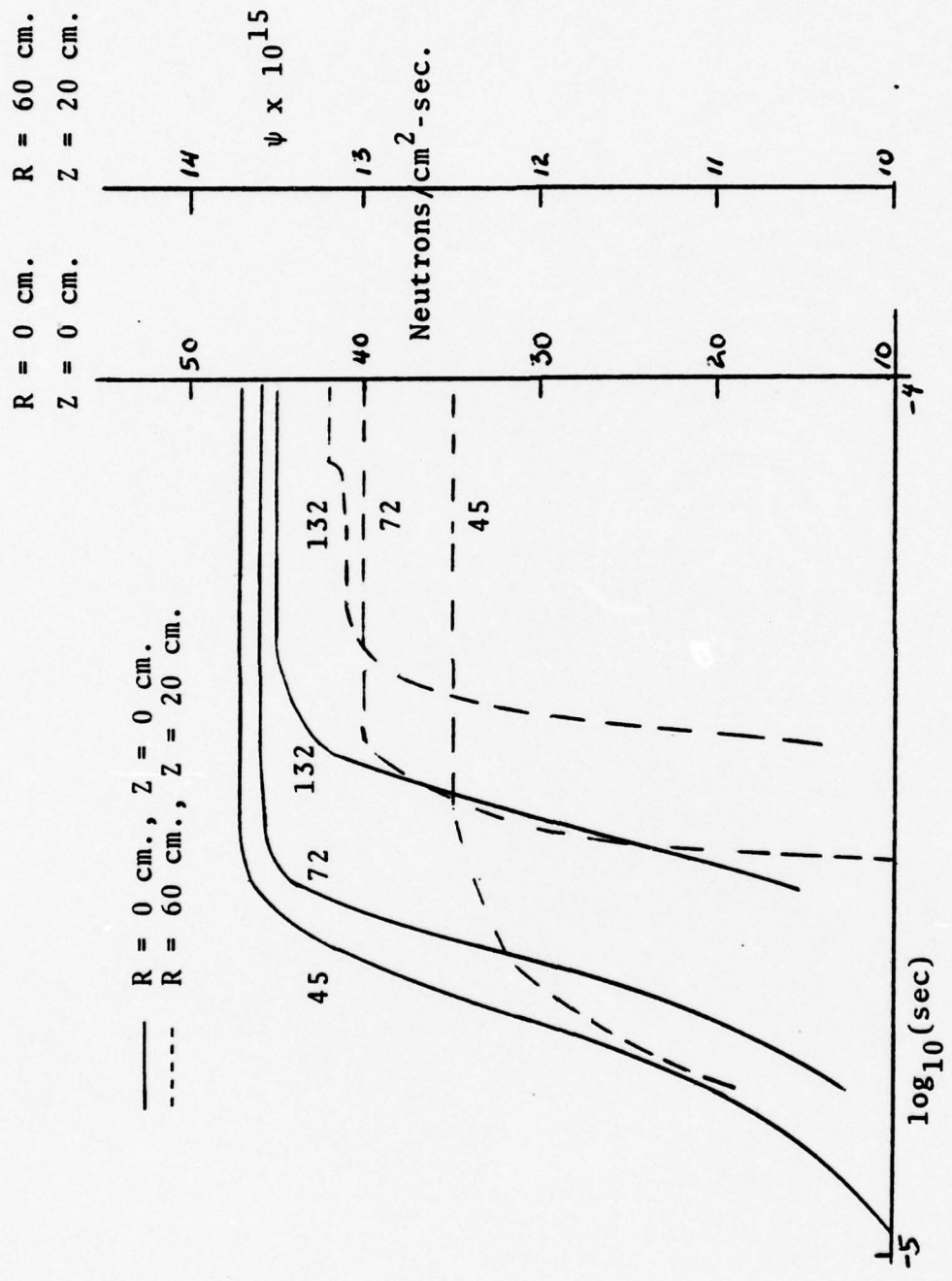


Figure 6. Time Dependent Neutron Flux for All DOF Using Crank-Nicolson Method for a Central Disturbance (R = 0 cm., Z = 0 cm.)

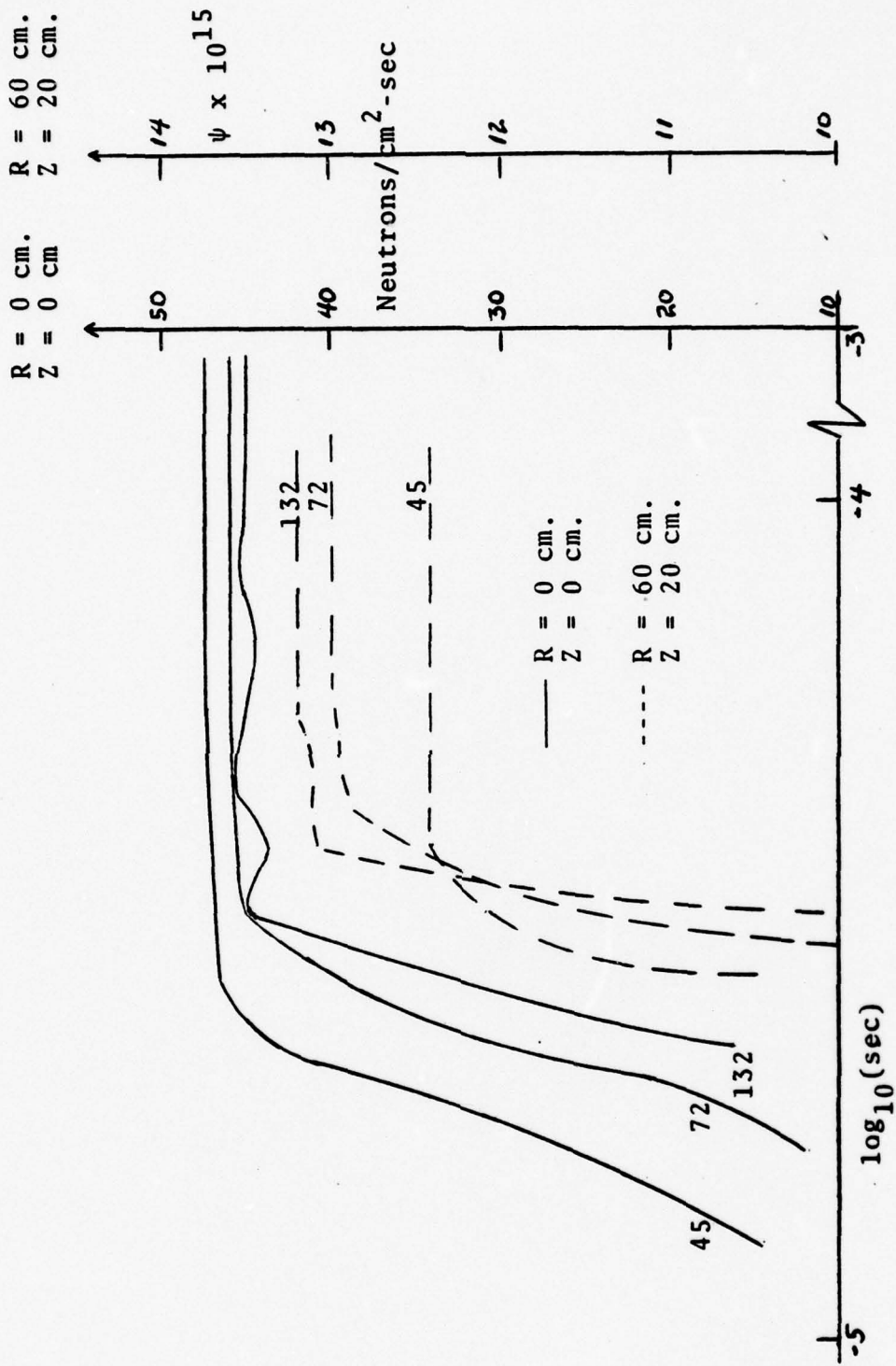


Figure 7. Time Dependent Neutron Flux for All DOF Using Crank-Nicolson Method for a Skewed Disturbance ( $R = 60 \text{ cm.}, Z = 0 \text{ cm.}$ )

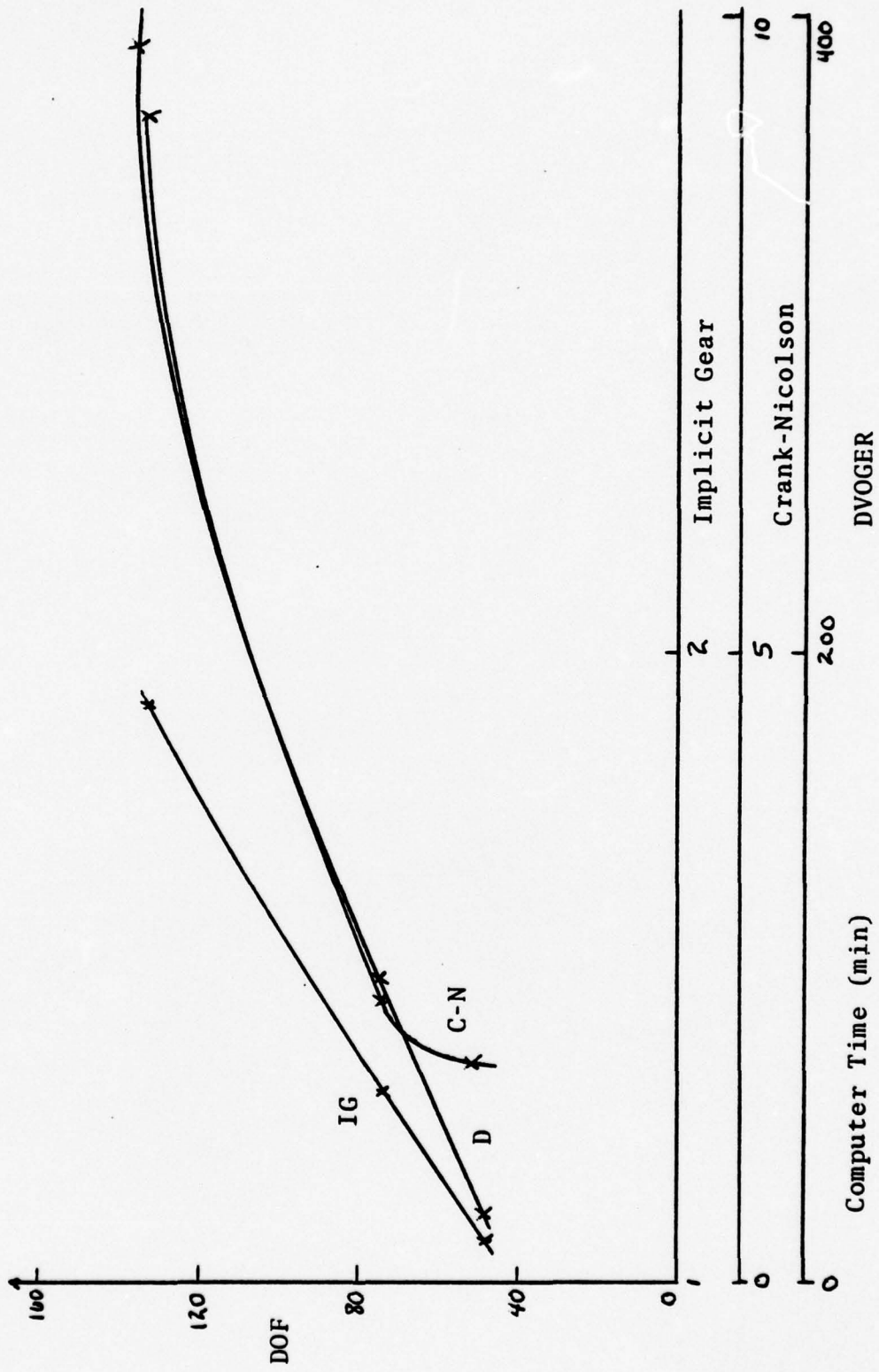


Figure 8. Plot Comparing Degrees of Freedom to Computer Processing Time

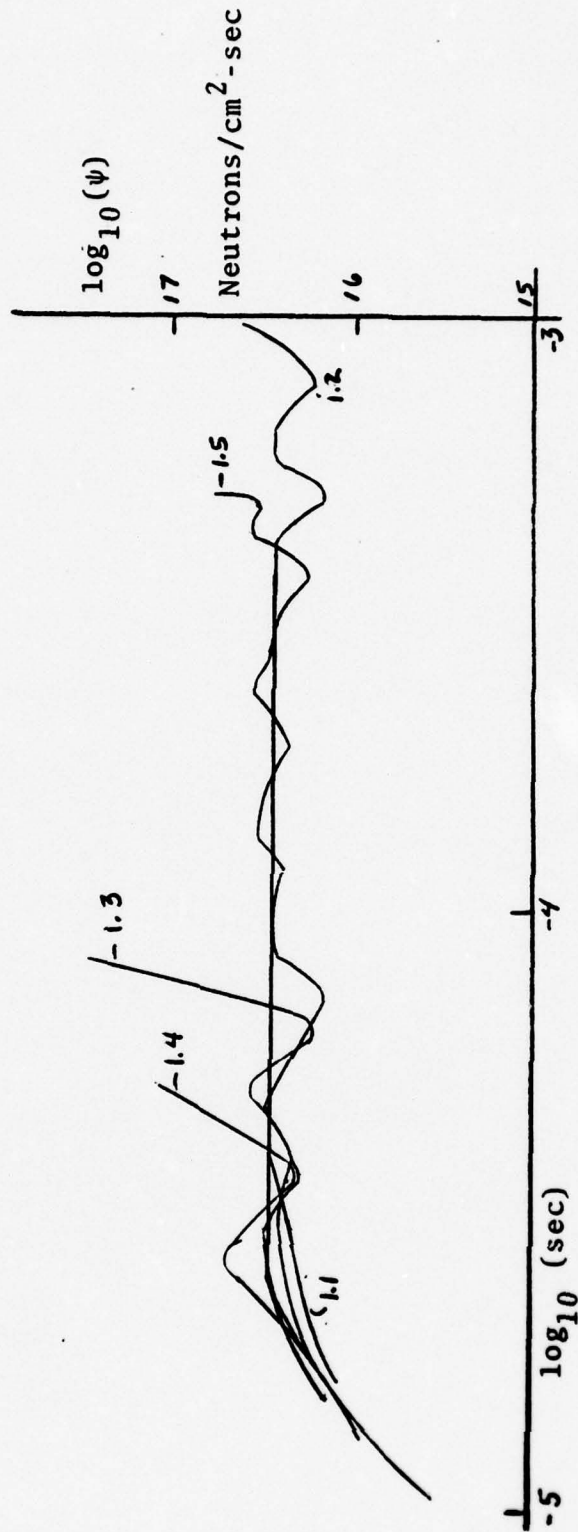


Figure 9. Effect of Time Step Change in Crank-Nicolson on Achieving Steady State.

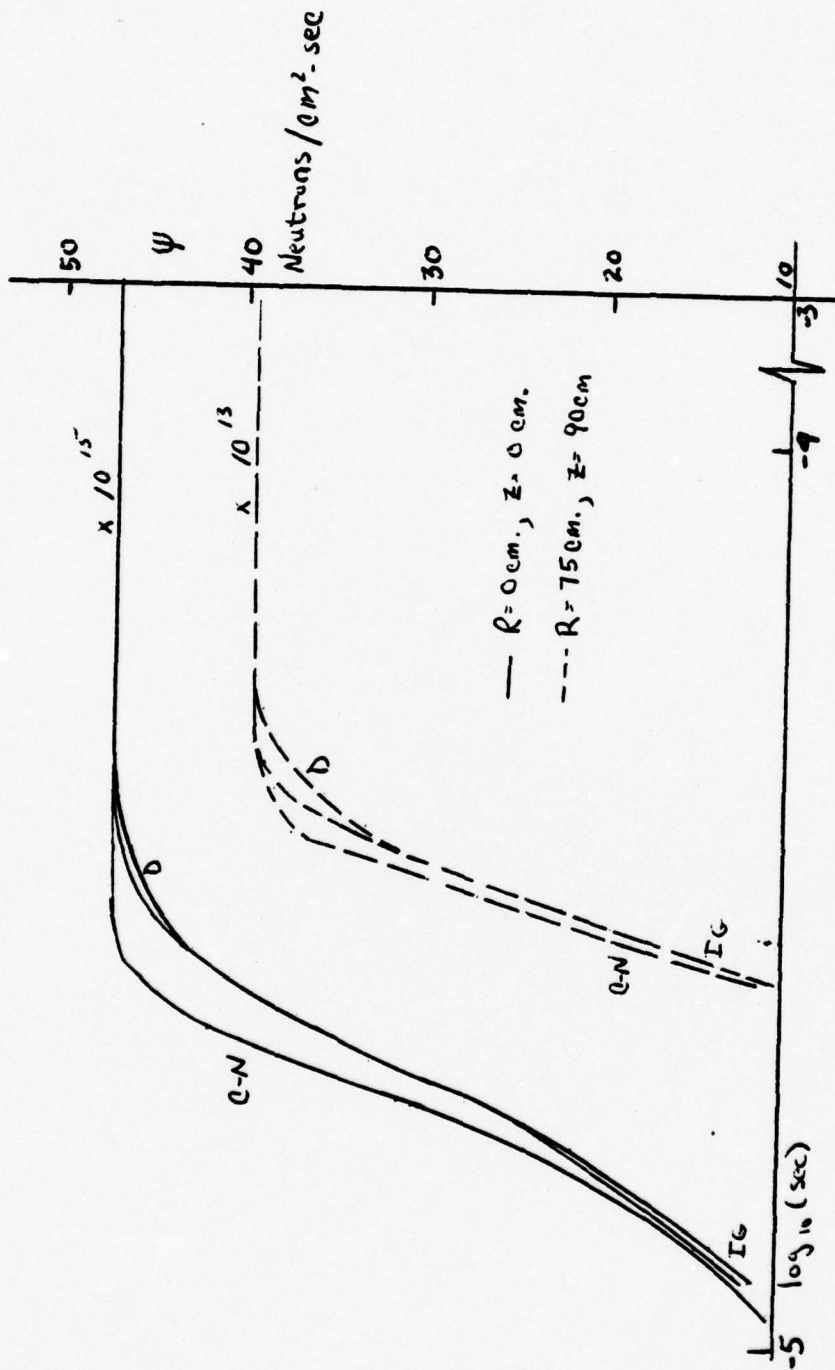


Figure 10. Time Dependent Neutron Flux for All Methods with 45 DOF for a Central Disturbance ( $R = 0\text{ cm.}, z = 0\text{ cm.}$ ).

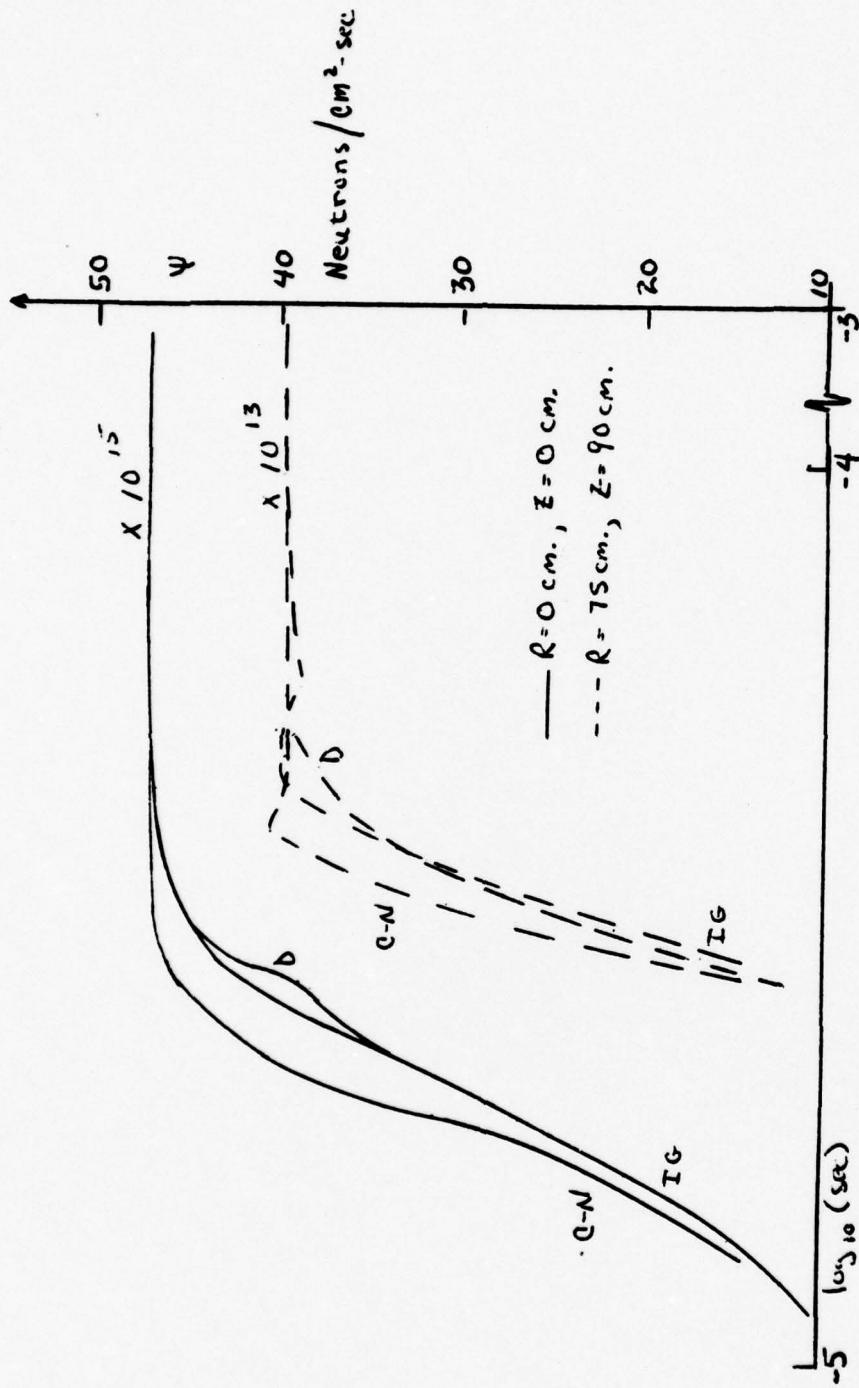


Figure 11. Time Dependent Neutron Flux for All Methods with 45 DOF for a Skewed Disturbance ( $R = 60$  cm.,  $Z = 0$  cm.).

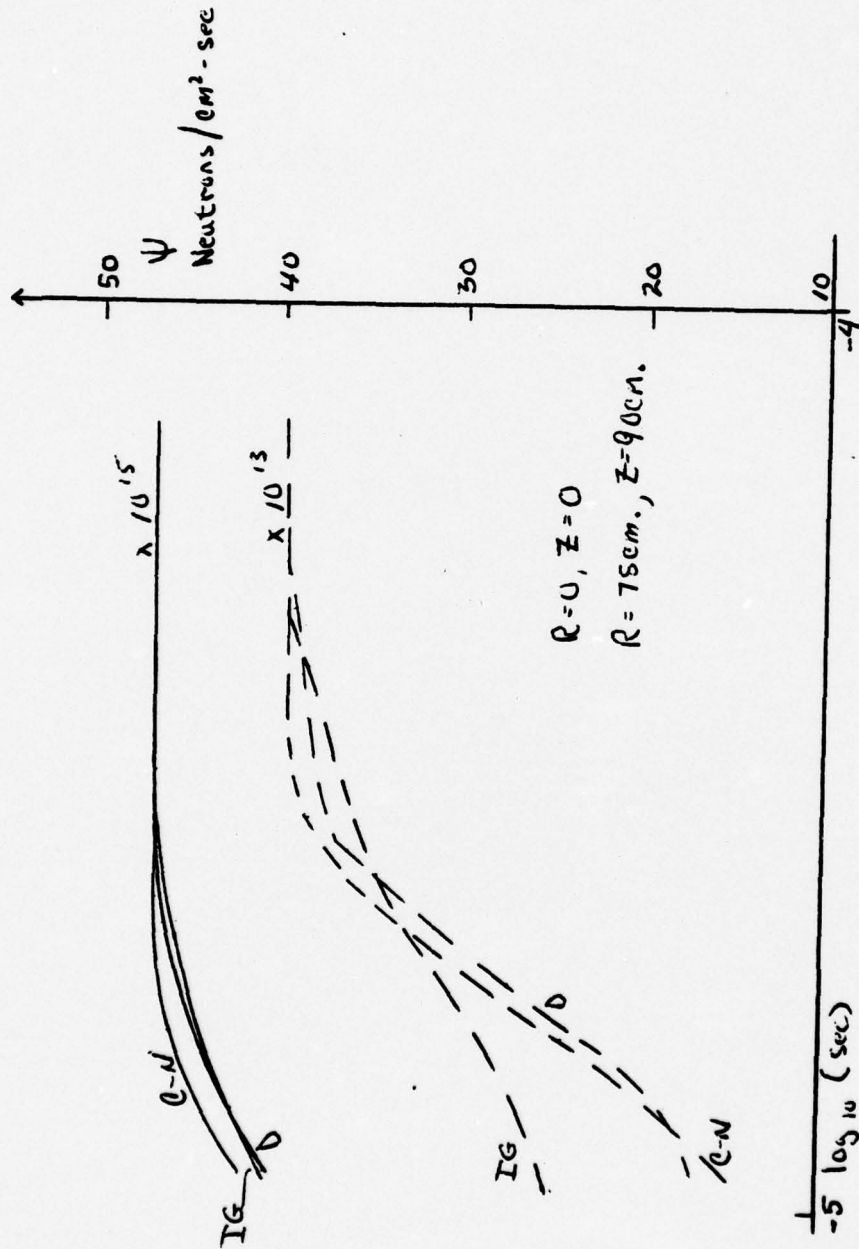


Figure 12. Time Dependent Neutron Flux for All Methods with 45 DOF for a Uniform Disturbance Throughout Core.

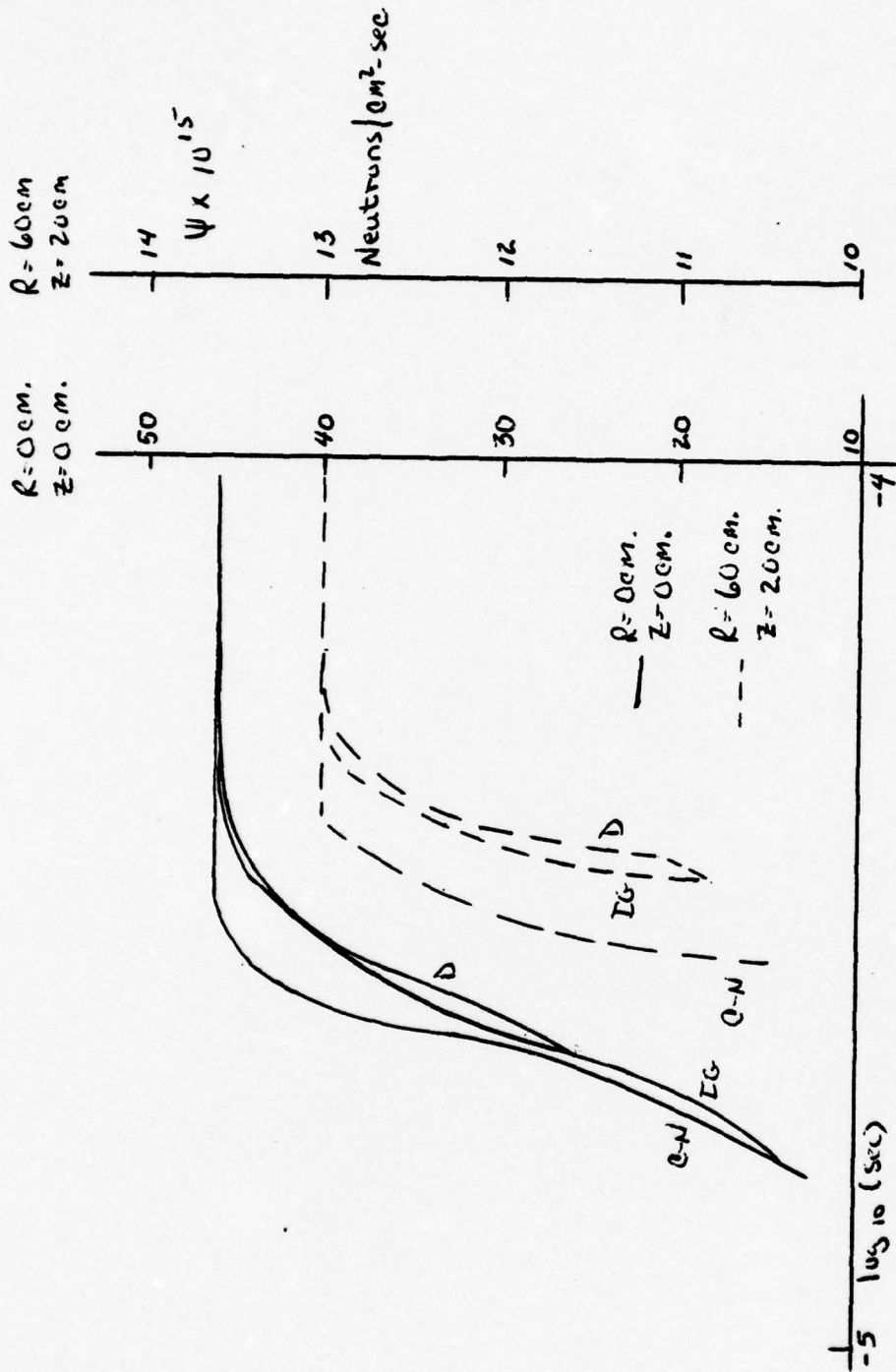


Figure 13. Time Dependent Neutron Flux for All Methods with 72 DOF for a Central Disturbance ( $R = 0 \text{ cm.}, Z = 0 \text{ cm.}$ ).

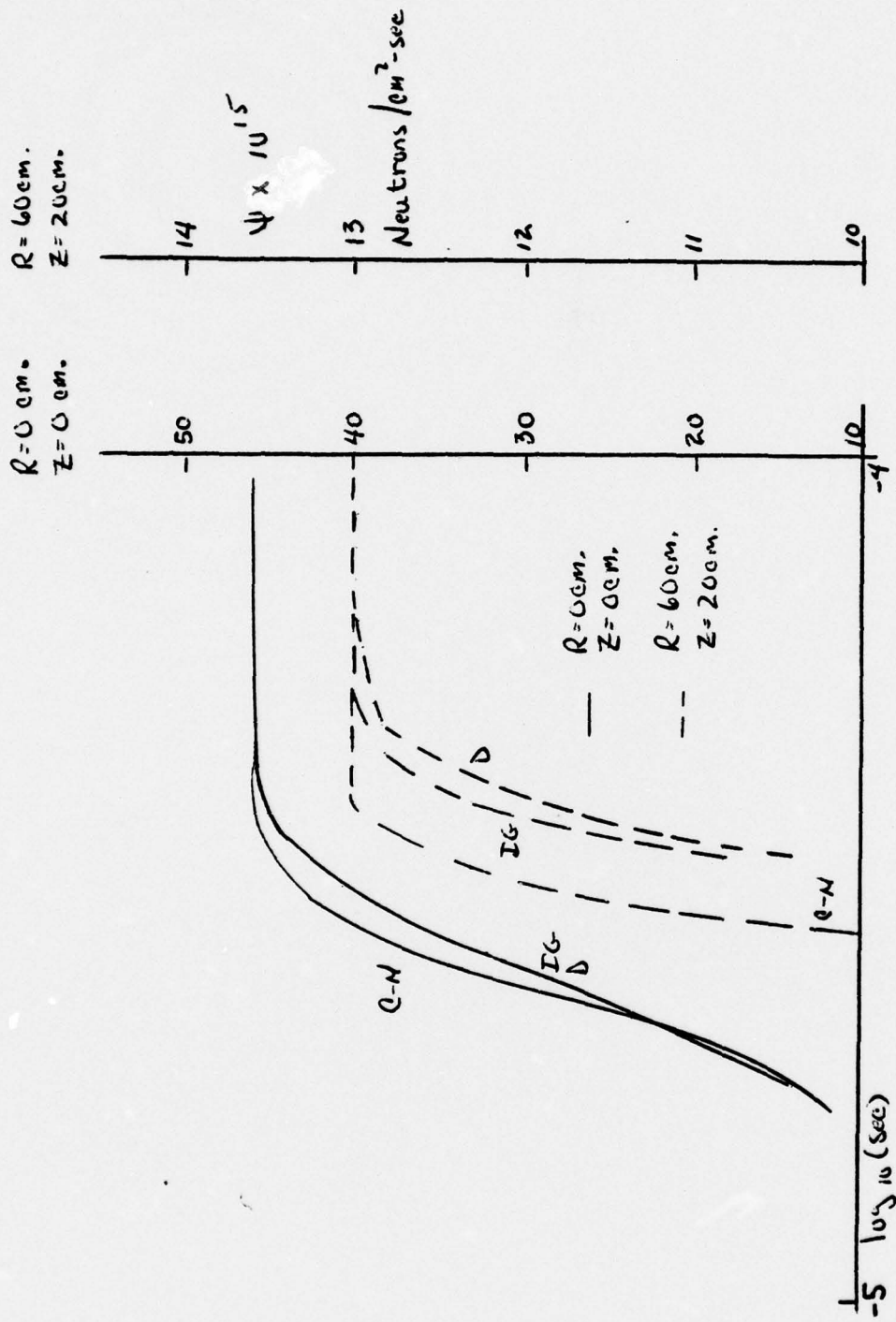


Figure 14. Time Dependent Neutron Flux for All Methods with 72 DOF for a Skewed Disturbance ( $R = 60 \text{ cm.}$ ,  $Z = 0 \text{ cm.}$ ).

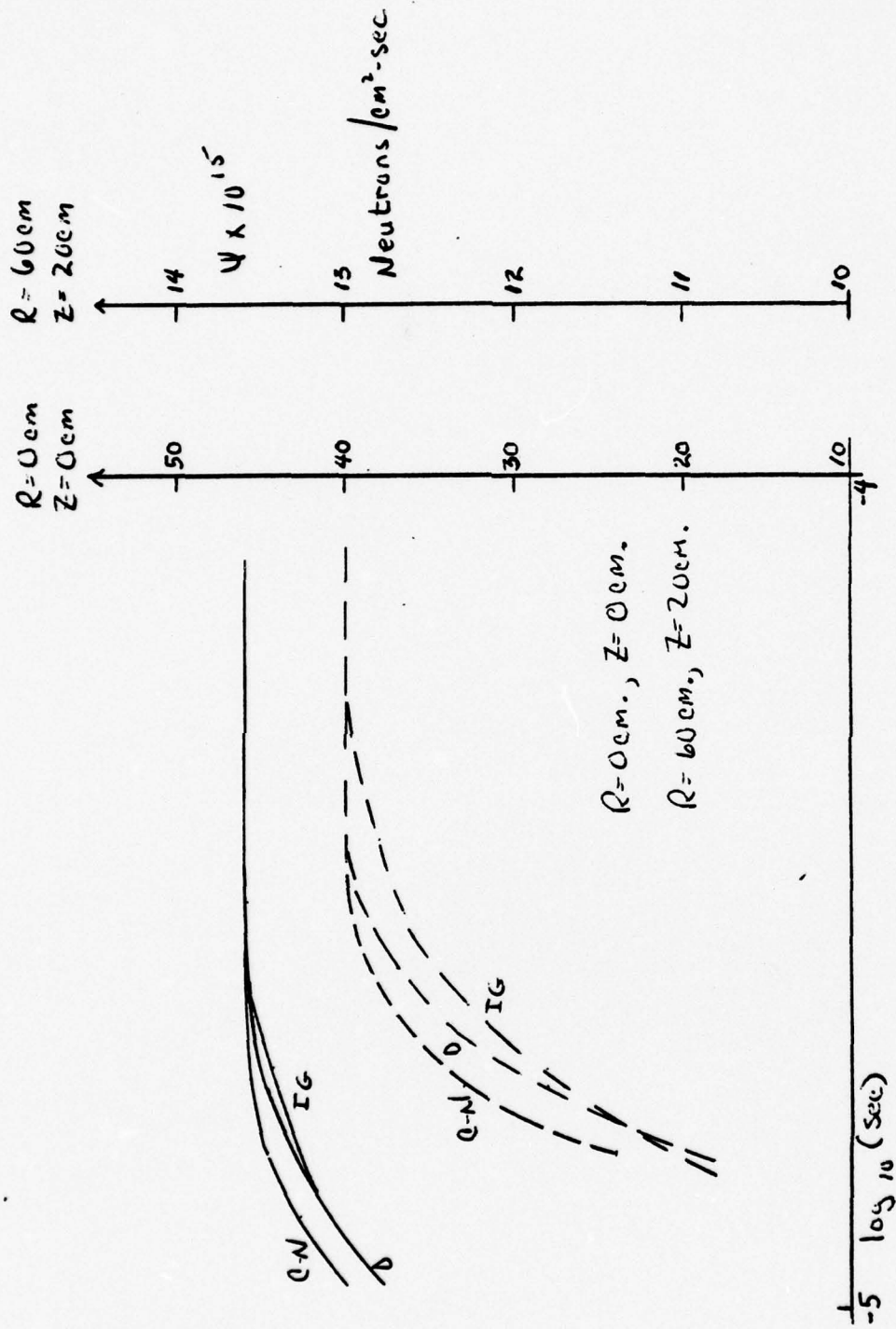


Figure 15. Time Dependent Neutron Flux for All Methods with 72 DOF for a Uniform Disturbance Throughout Core.

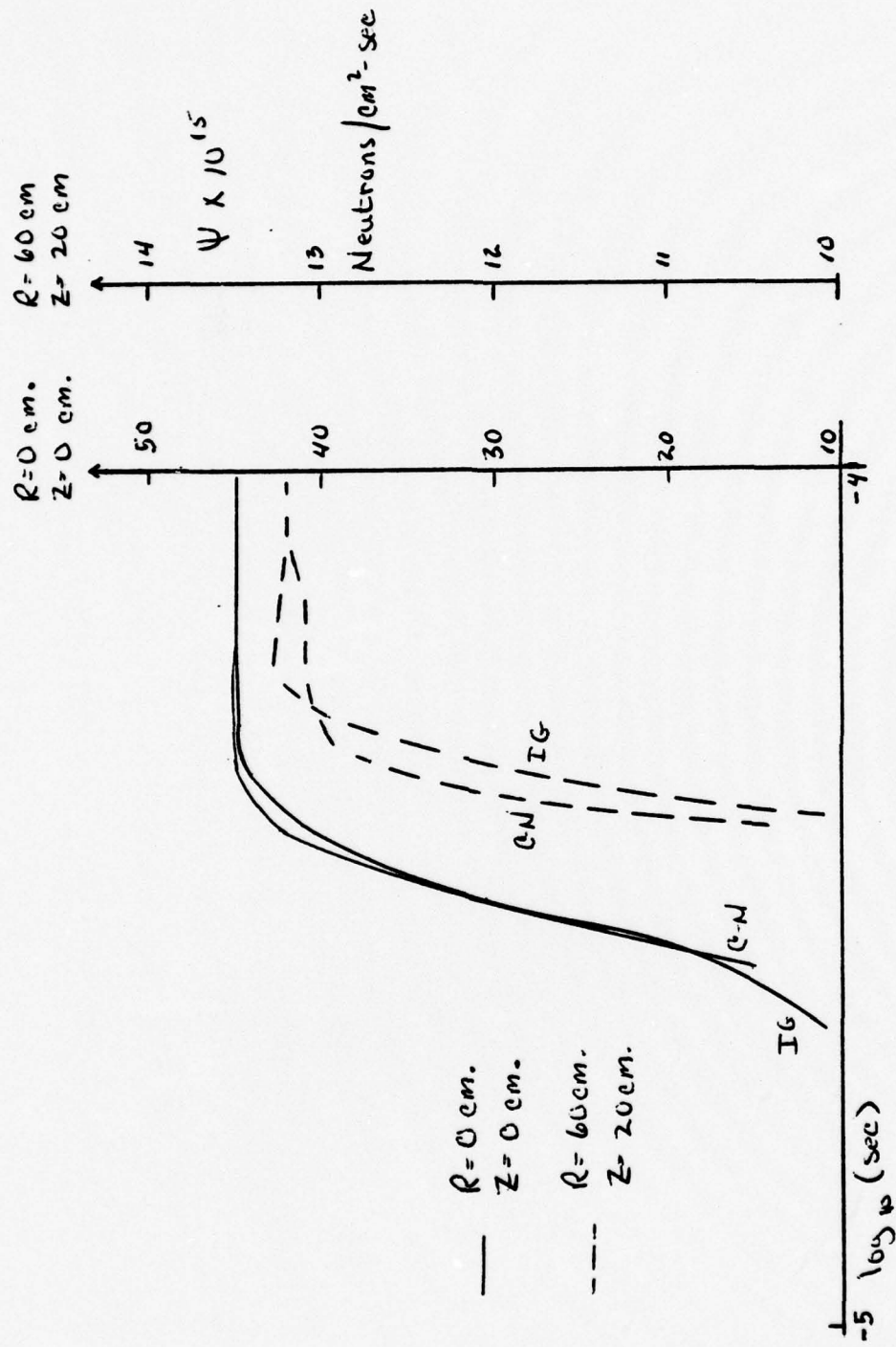


Figure 16. Time Dependent Neutron Flux for Crank-Nicolson and Implicit Gear Methods with 132 DOF for Central Disturbance ( $R = 0 \text{ cm.}$ ,  $Z = 0 \text{ cm.}$ )

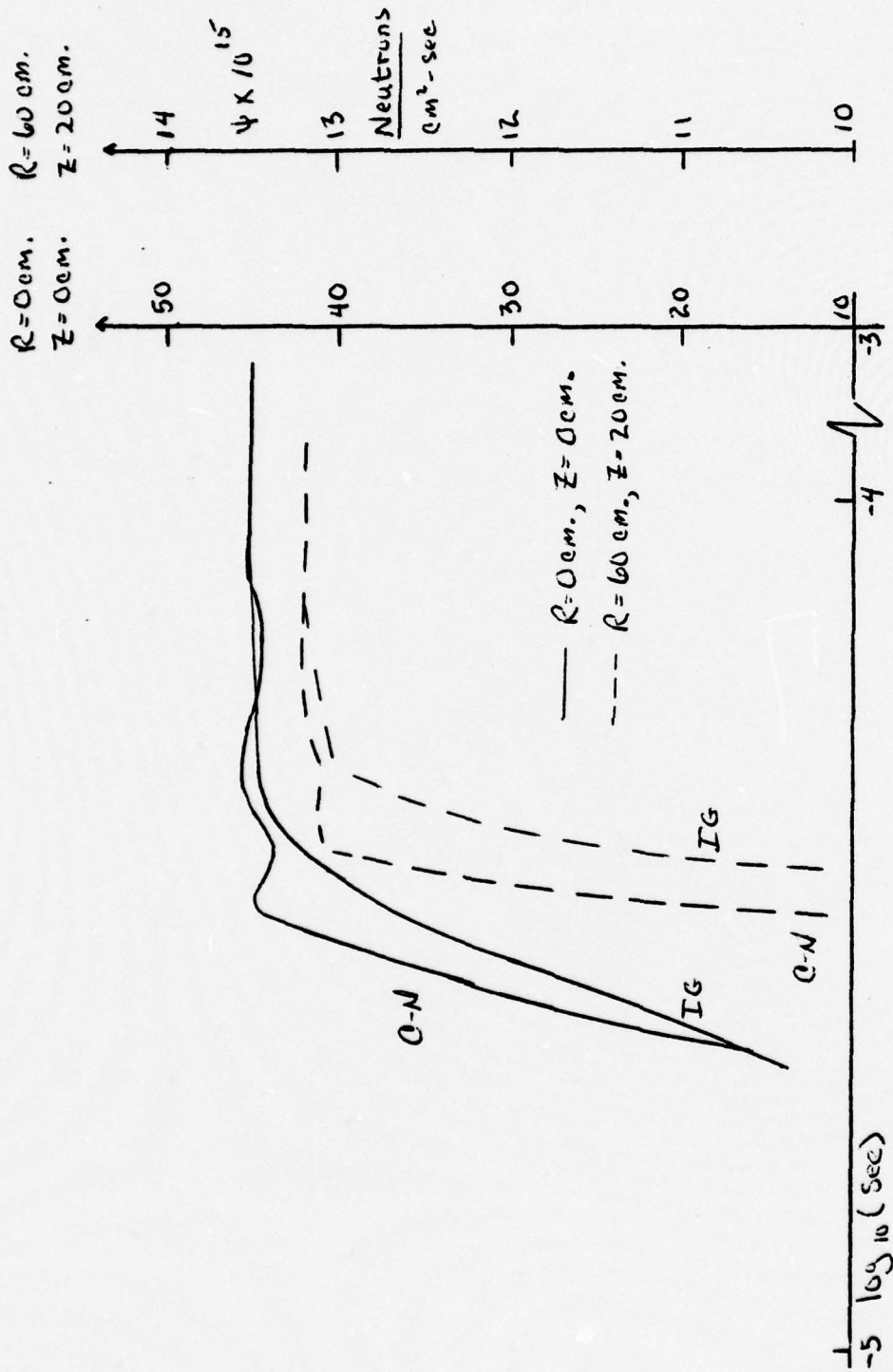


Figure 17. Time Dependent Neutron Flux for Crank-Nicolson and Implicit Gear Methods with 132 DOF for Skewed Disturbance ( $R = 60 \text{ cm.}$ ,  $Z = 0 \text{ cm.}$ )

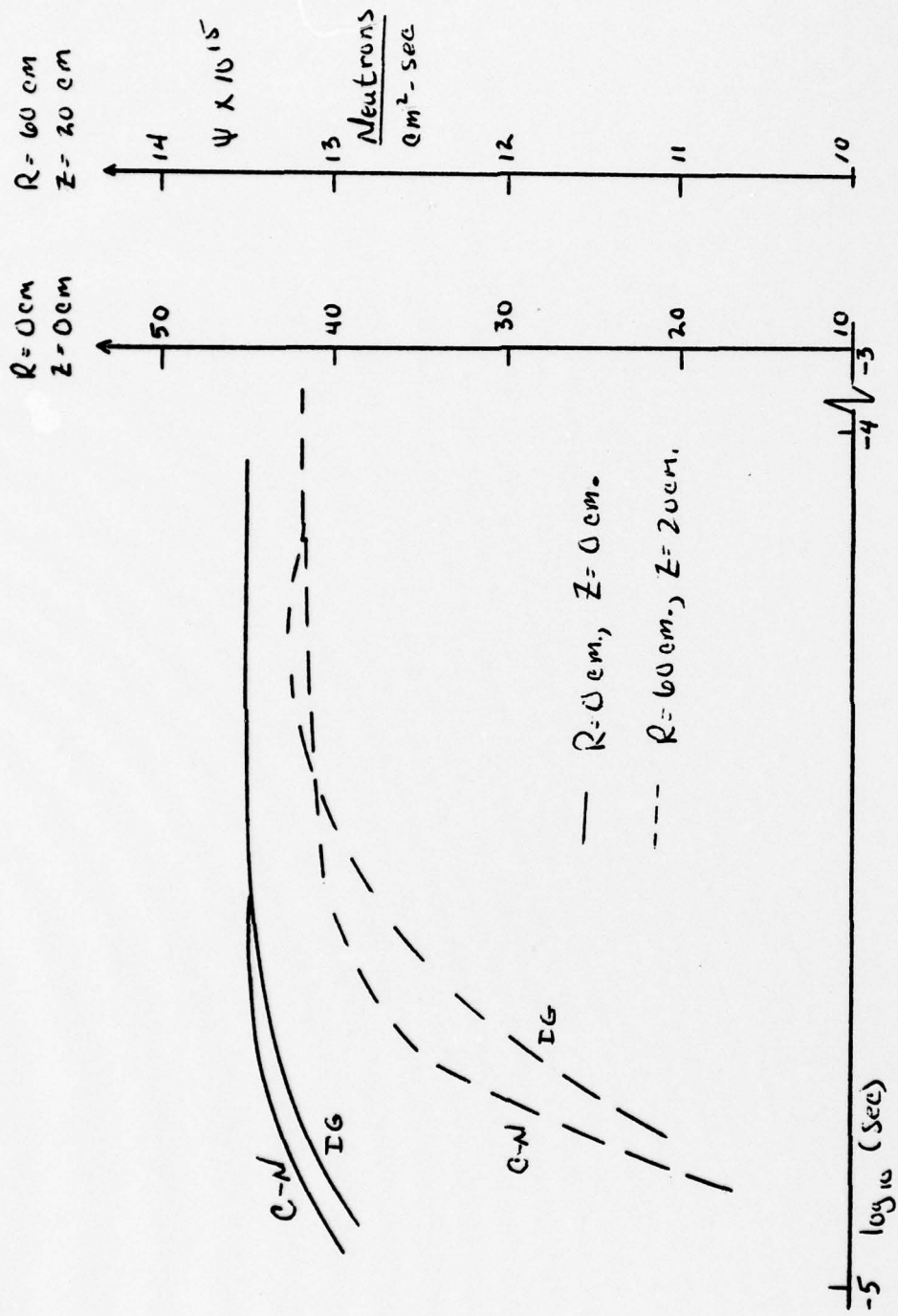


Figure 18. Time Dependent Neutron Flux for Crank-Nicolson and Implicit Gear Methods with 132 DOF for a Uniform Disturbance Throughout the Core.

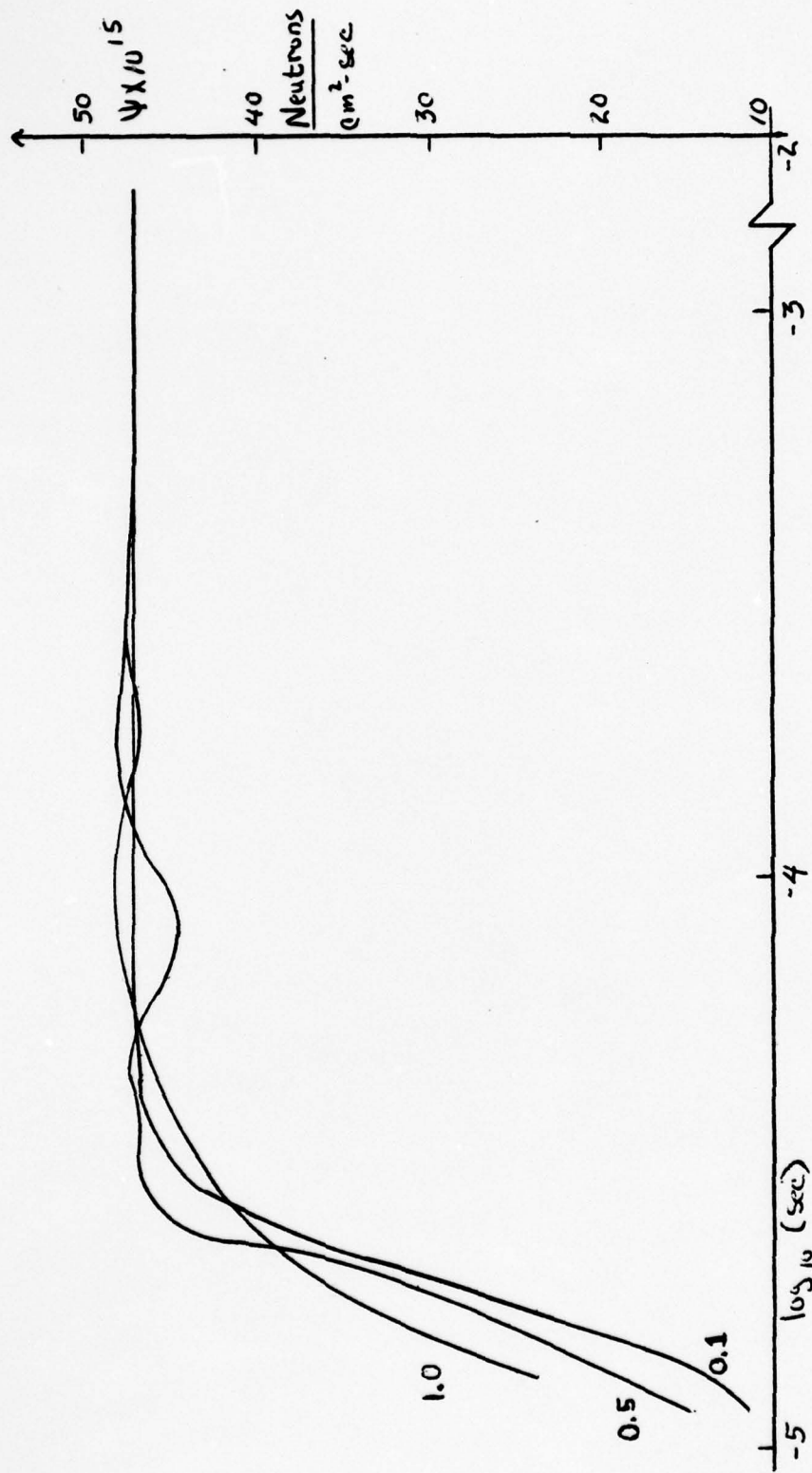


Figure 19. Comparison of Error Criterion for Implicit Gear Method.

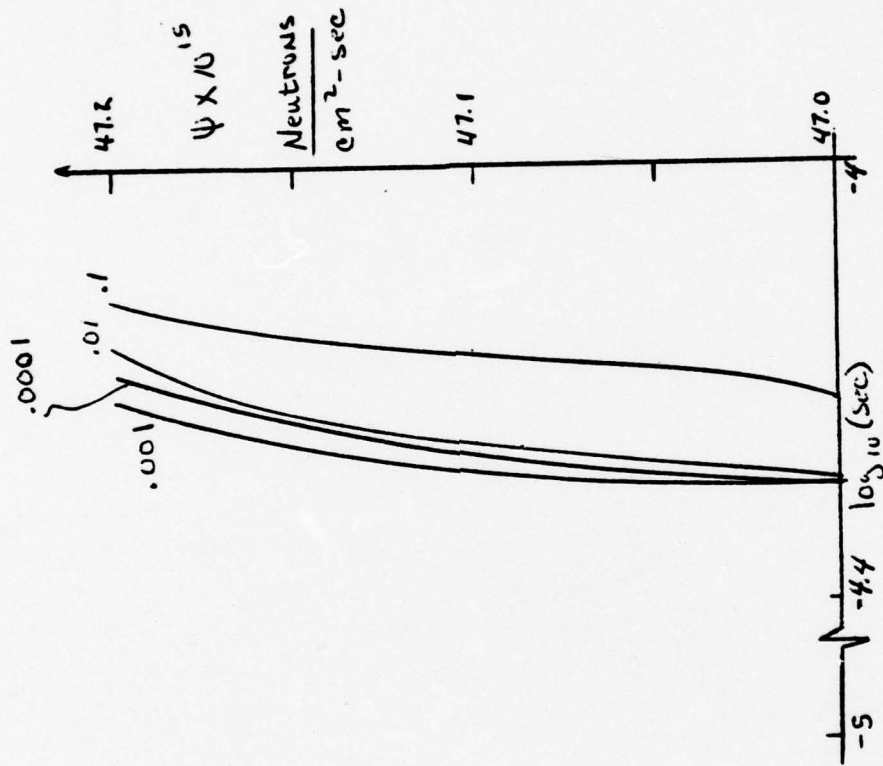


Figure 20. Typical Approach to Steady State Solution Value with Various Error Criterion (Implicit Gear).

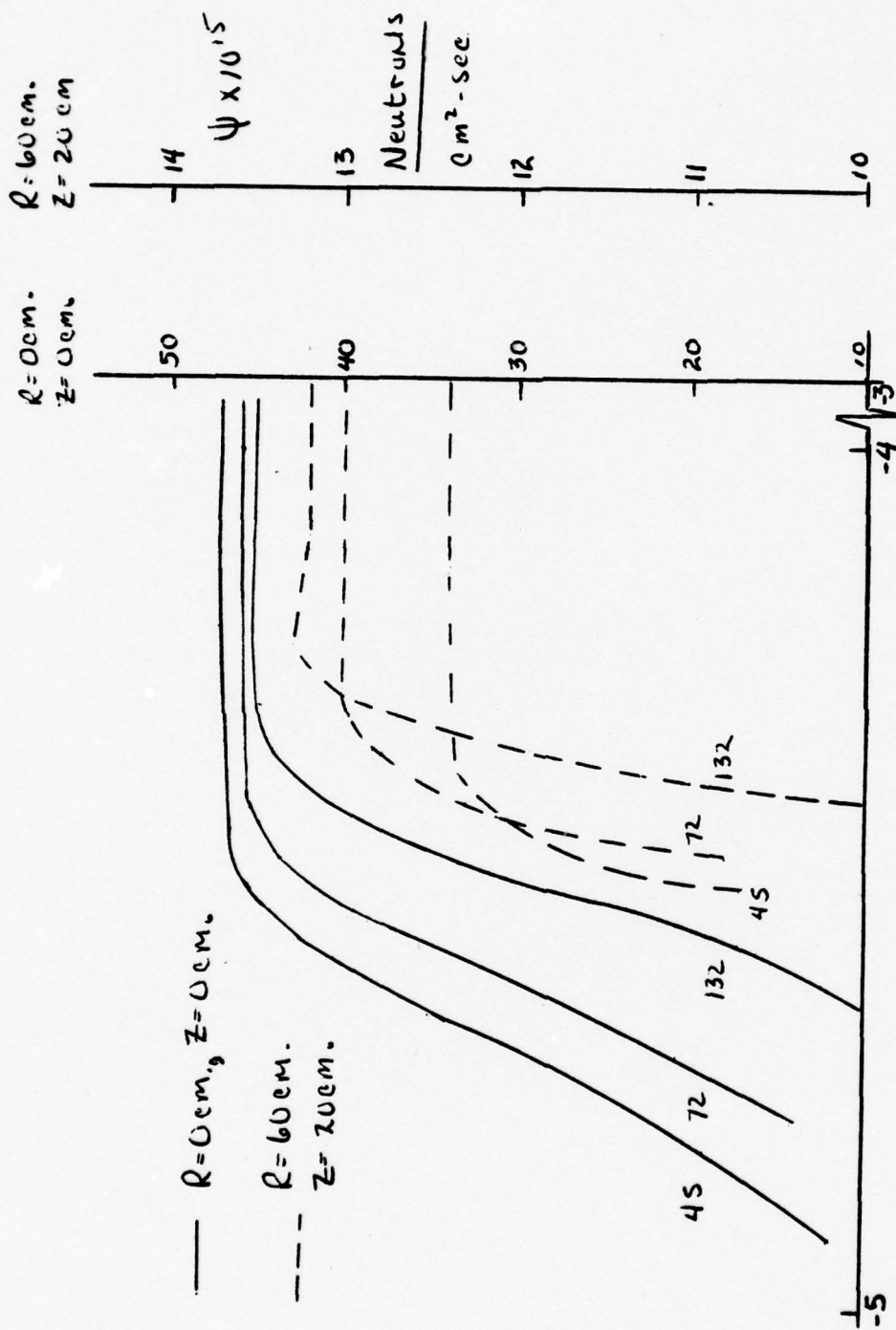


Figure 21. Time Dependent Neutron Flux for All DOF Using Implicit Gear Method for a Central Disturbance ( $R = 0 \text{ cm.}$ ,  $Z = 0 \text{ cm.}$ )

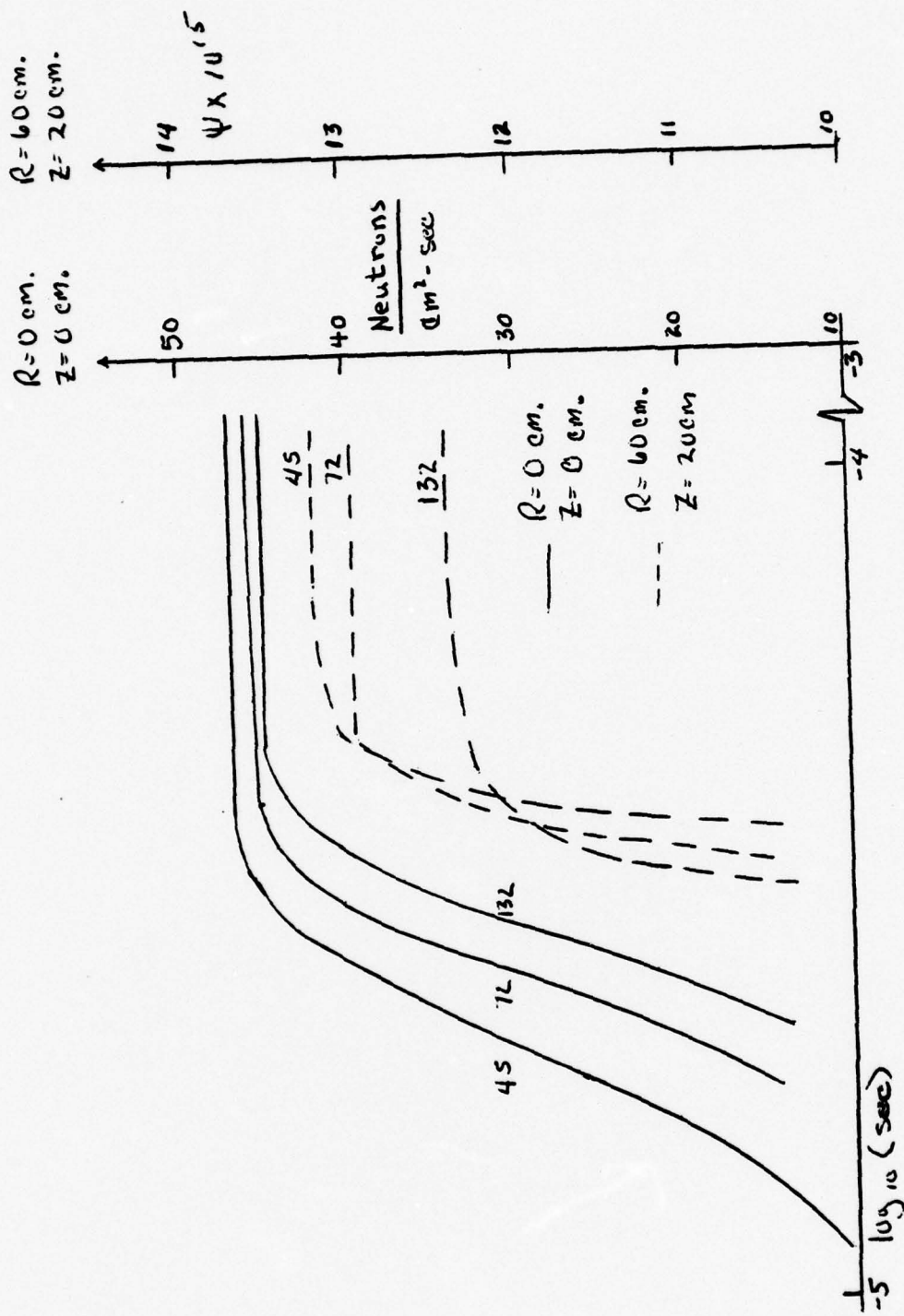


Figure 22. Time Dependent Neutron Flux for All DOF Using Implicit Gear Method for a Skewed Disturbance ( $R = 60 \text{ cm.}$ ,  $Z = 0 \text{ cm.}$ ).

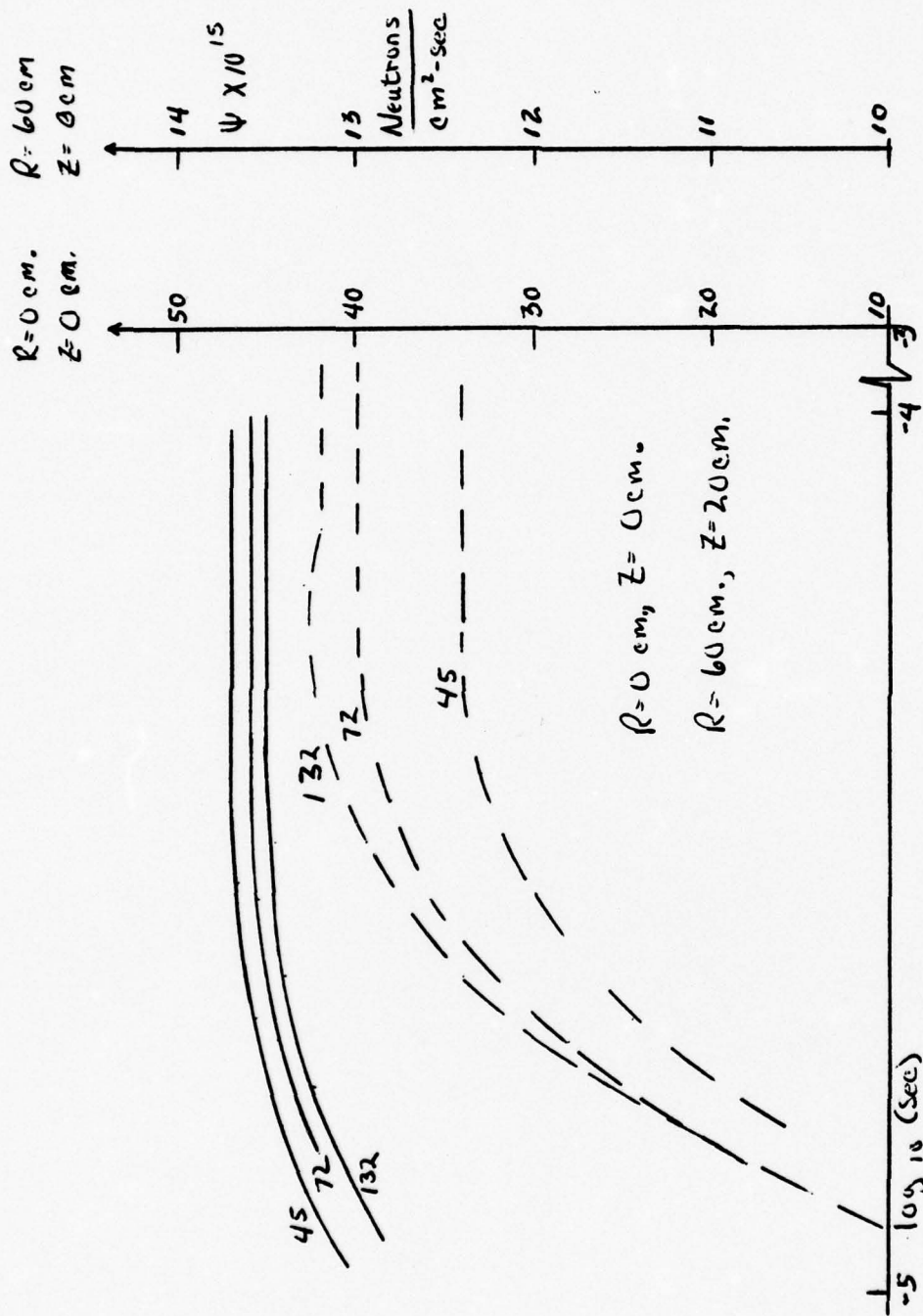


Figure 23. Time Dependent Neutron Flux for All DOF Using Implicit Gear Method for a Uniform Disturbance Throughout the Core.

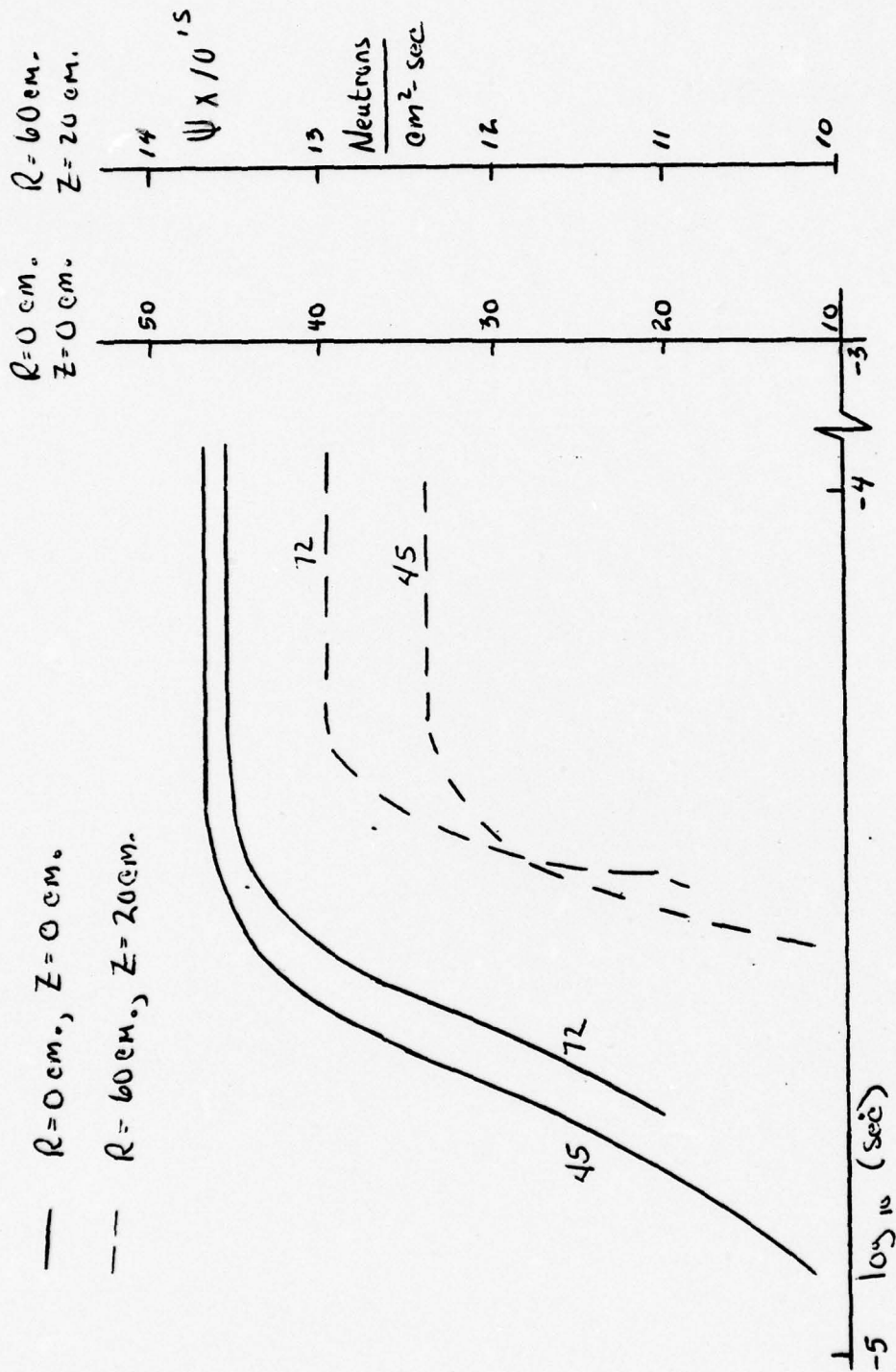


Figure 24. Time Dependent Neutron Flux for 45 and 72 DOF Using the DVOGER (Gear) Method for a Central Disturbance ( $R = 0\text{ cm.}, Z = 0\text{ cm.}$ ).

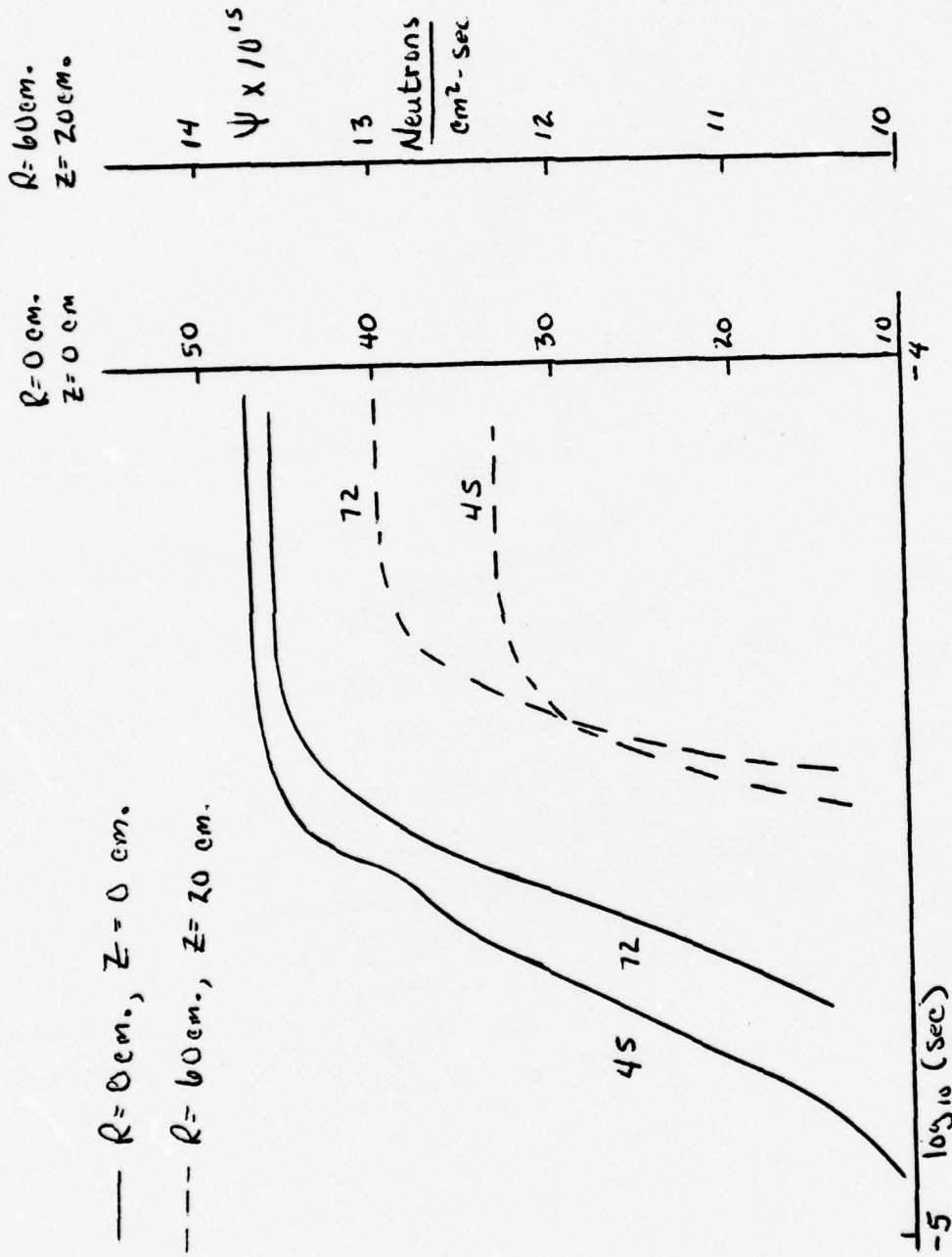


Figure 25. Time Dependent Neutron Flux for 45 and 72 DOF Using the DVOGER (Gear) Method for a Skewed Disturbance ( $R = 60\text{ cm.}, Z = 0\text{ cm.}$ ).

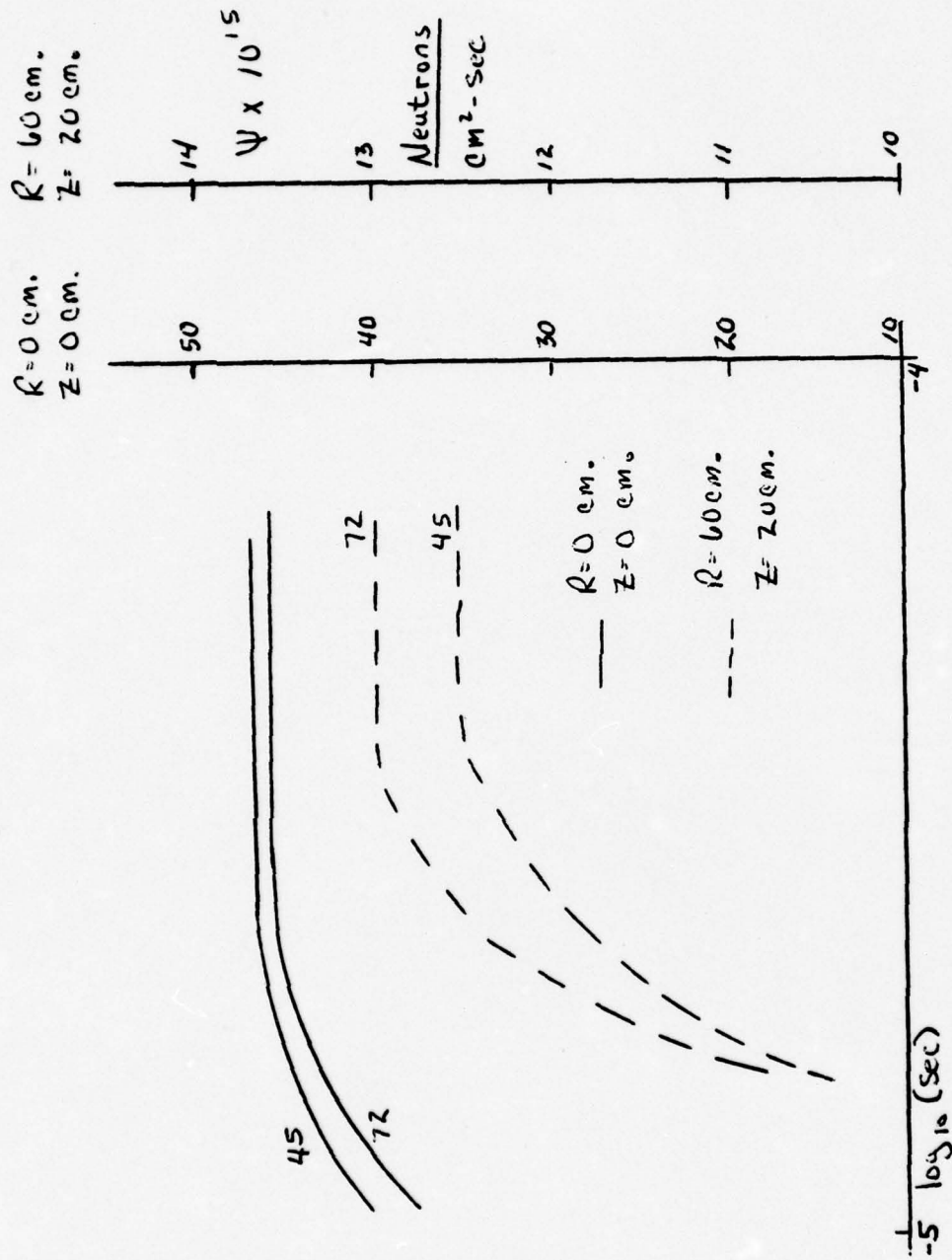


Figure 26. Time Dependent Neutron Flux for 45 and 72 DOF Using the DVOGER (Gear) Method for a Uniform Disturbance Throughout the Core.

```

*****
APPENDIX A: Computer Programming Codes
*****
FINITE ELEMENT SOLUTION OF NONLINEAR REACTOR DYNAMICS
IN TWO DIMENSIONAL SPACE
LINEARIZED VERSION
*****
SOLUTION METHOD -- DVJGER (IMSL)
*****
1- MTH=0 INDICATES A PREDICTOR CORRECTOR
   (ADAMS) METHOD
2- MTH=1 INDICATES A VARIABLE ORDER METHOD
   SUITABLE FOR SYSTEMS OF STIFF DIFFERENTIAL
   EQUATIONS. PROGRAM USES THE JACOBIAN.
*****
INPJT PARAMETERS AND DIMENSIONING REQUIREMENTS.
SINGLE ITEMS
NUMSNP IS THE SYSTEM NODAL POINTS
NUMMEL IS THE NUMBER OF TRIANGULAR ELEMENTS
NUMBP IS THE NUMBER OF OUTER BOUNDARY POINTS
NUMLEL IS THE NUMBER OF ELEMENTS IN THE CORE
EPSVAL IS THE CONVERGENCE CRITERIA
ITYPE=0 DENOTES CORE ELEMENTS
ITYPE=1 DENOTES REFLECTOR ELEMENTS
*****
DIMENSIONING REQUIREMENTS
NUMSNP BY NJMSNP
PART,BIGA,BIGAB,BIGC,BIGCC,WP
NUMSNP
SYSNOD R,Z,YMAX,ERROR,PSIIV,DPSI,PV
*****
NUMEL
ELEMNT,ITYPE,V,VD,SGA,SGF,ALPHA,D,ZLMBDA,VLMBDA,
ZOMEGA,R1,R2,R3,Z1,Z2,Z3,AREA
ELNOD(NUMEL,3),IJPBP(NJPBP),CM(NUMEL,3,3,3),PW(NWK)
*****

```

CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC

```

C
INTEGER*4 SYSNOD,ELEMNT,ELNOD
DIMENSION TITLE(20),ALFA(3,3)
DIMENSION PART(132,132),BIGA(132,132),BIGAB(132,132),BIGC(132,132),
1 BIGCC(132,132),SYSNOD(132),ELEMNT(220),ELNOD(220,3),IUPBP(22),
2 ITYPE(220),V(220),VD(220),D(220),SGA(220),SGF(220),ALPHA(220),
3 ZLMBDA(220),VLMBDA(220),ZOMEGA(220),RHO(220),R(132),Z(132),
4 R1(220),Z1(220),R2(220),Z2(220),R3(220),Z3(220),A(220,3),
5 B(220,3),YMAX(132),ERRCR(132),AREA(220),CM(220,3,3,3),DP$I(132)
6 ,PSIIV(132),PDPSI(132),PW(20000),PSI(8,132),PV(132),S(132),
7 WP(132,132)
READ(5,998) TITLE
998 READ(5,10) NUMEL,NUPBP,NUMSNP,NFULEL
10 FORMAT(20A4)
C * FJR MTH NOT EQUAL TO ZERO NWK = NJMSNP * (NUMSNP + 17)
C NWK=NUMSNP*17
C CALL ERRSET(207,256,0,1,1,209)
C
DO 05 I=1,NUMSNP
ERROR(I)=0.
PDPSI(I)=0.
YMAX(I)=1.
05 CONTINUE
C
NUF=NUMSNP/5
XNP=NUMSNP
XNUF=XNP/5
IF ((XNUF-FLOAT(NUF)) .LT. 1.) NUF=NUF+1
C
C GENERAL PROBLEM DATA
C
READ(5,12) (IJPBP(I),I=1,NUPBP)
12 READ(5,10) MTH,MAXDER,NCOUNT
FORMAT(16I5)
C
WRITE(6,11) NJMEL,NUPBP,NUMSNP,NFULEL
11 FORMAT(/2X,'# OF ELEMENTS=',I5/2X,'# OF UPPER BDRY NODAL POINTS=',
215)
C
READ(5,15) ZNJ,EISFAC,HBAR,EPSVAL,ERRVAL,AFUEL
15 READ(5,19) TO,H,TF,HMIN,HMAX

```





```

120 FORMAT (1X, // 1X, 'ELEMENT', 5X, 'A1', 9X, 'A2', 9X, 'A3', 9X, 'B1', 9X, 'B2',
19X, 'B3', 8X, 'AREA', //)
C
DO 140 I=1, NUMEL
A(I,1)=R3(I)-R2(I)
A(I,2)=R1(I)-R3(I)
A(I,3)=R2(I)-R1(I)
B(I,1)=Z3(I)-Z1(I)
B(I,2)=Z2(I)-Z1(I)
B(I,3)=Z1(I)-Z2(I)
AREA(I)=0.5*(A(I,2)*B(I,1)-A(I,1)*B(I,2))
C
WRITE(6,130) I, A(I,1), A(I,2), A(I,3), B(I,1), B(I,2), B(I,3),
1AREA(I)
140 CONTINUE
C
130 FORMAT (3X, 13, 3X, 7(F12.7, 1X))
C
C----- ZERO THE BIGA, BIGC AND BIGAB MATRICES-----
C
DO 194 KK=1, NUMSNP
DO 193 II=1, NUMSNP
BIGA(KK, II)=0.
PART(KK, II)=0.
BIGAB(KK, II)=0.0
BIGC(KK, II)=0.0
193 CONTINUE
194 CONTINUE
C
C*****
C----- PEPSI CALCULATES THE BIGA MATRIX AND PART OF THE BIGAB
C-----
C
CALL PEPSI (NJMEL, NUMSNP, R1, R2, R3, AREA, ELNOD,
1BIGA, BIGAB, VLMBDA)
C
WRITE(6,195)
195 FORMAT(//5X, 'BIGA MATRIX', //)
WRITE(6,196)
196 FORMAT(//5X, 'BIGAB MATRIX FROM PEPSI, NO DEL**2 CONTRIBUTION', //)

```

```

CCCCC-----
C      CRUISE CALCULATES THE BIGAB MATRIX
C-----
C      CALL CRUISE (NJMEL,NJMSNP,      R1,R2,R3,AREA,ELNOD,
1      BIGAB,
A,B,VD)
C
C      WRITE(6,197)
197  FORMAT('//5X,'BIGAB MATRIX FROM CRUISE, WITH DEL*2 TERM',//)
C-----
CCCCC
C
C      DO 255 I=1,NUPBP
C      II=IUPBP(I)
C      DO 251 J=1,NUMSNP
C      BIGA(J,II)=0
C      BIGAB(I,J,II)=0
C      BIGAB(I,II)=0
C      CONTINUE
251  BIGA(II,II)=1.
C      BIGAB(II,II)=0
C      CONTINUE
255  CONTINUE
C
C      DI=1.
C      CALL LINV3F(BIGA,B,1,NUMSNP,NUMSNP,DI,D2,PART,IER)
C      WRITE(6,250)
250  FORMAT('//5X,'3IGA INVERSE',//)
C-----
CCCCC
C      WRITE OUT NUC-EAR DATA
C
C      WRITE(6,3105) ZNJ,FISFAC,HBAR,EPSVAL,ERRVAL,AFUEL
3105  FORMAT('//5X,'NJCLEAR DATA',/2X,'ZNU=',G12.6/2X,'FISFAC=',G12.6/2X,
1  'HBAR=',G12.6/2X,'EPSVAL=',G12.6/2X,'ERRVAL=',G12.6/2X,'AFUEL=',
2  G12.6)
C      WRITE(6,3110)
3110  FORMAT('IHO,2X,'ELEM',6X,'D',8X,'SGA',9X,'SGF',8X,'ALPHA',9X,'VD',
1  '7X,'ZLMBDA',7X,'VLMBDA',7X,'ZOMEGA',8X,'V',
1  'DO 3115 I=1,NJMEL

```

```

WRITE(6,3120) (I,D(I),SGA(I),SGF(I),ALPHA(I),VD(I),ZLMBDA(I),
1 ZLMBDA(I),ZOMEGA(I),V(I))
3115 CONTINUE
3120 FORMAT(2X,I4,I4,10(1PE12.4))

```

```

C C
C C
C C
NUMEQ=NUMSNP-NJPBP

```

```

C C
EXTERNAL YVETTE

```

```

C CALL YVET (NJMEL, NUMEQ, NUMSNP,
1 ITYPE, ELND, M, BIGA, BIGAB,
2 PDPSI, PV, NCCOUNT, ZOMEGA, WP)

```

```

IUPBP, PSIIV, NUPBP, BIGC,
BIGCC, PART, NWK, PSI, PW, DPSI,

```

```

C----- INITIALIZE ARGUMENTS OF DVOGER -----

```

```

PTIME= 1.E-17
STAR=0.01
NIT=0
T=TO
JSTART=0
EPS= EPSVAL
IER=0

```

```

C C
C 999 WRITE(6,999) IITLE
FORMAT(IH1,20A4)

```

```

WRITE(6,317) I, MTH, MAXDER, JSTART, H, HMIN, HMAX, EPS, NCOUNT
317 FORMAT(7/10X, //)
318 FORMAT(2X, I=, G10.4, 2X, MTH=, I4, 2X, MAXDER=, I4, 2X, JSTART=, I4,
1 2X, H=, G10.4, 2X, HMIN=, G10.4, 2X, HMAX=, G10.4, 2X, EPS=, G10.4, 2X,
2, NCOUNT=, I2, //)

```

```

C C
C 320 DO 320 I=1, NUMSNP
PSI(I, I)=PSIIV(I)
CONTINUE

```

```

C C
C 321 DO 321 I=1, NUPBP
J=IUPBP(I)
PSI(I, J)=0.0
CONTINUE

```

```

C C
NUMEQ=NUMSNP-NJPBP

```

C-----  
C CONSTRUCT THE 3IGC MATRIX ON THE ELEMENT LEVEL  
C-----

DO 200 L=1, NUMEL

DO 583 I=1,3  
DO 582 J=1,3  
DO 581 K=1,3  
CM(L,I,J,K)=0.

581 CONTINUE  
582 CONTINUE  
583 CONTINUE

IF (LL.NE.0) GO TO 200

CC=PI\*AREA(L)/180.

CM(L,3,3,3)=CC\*(6.0\*R1(L)+4.0\*R2(L)+6.0\*R3(L))  
CM(L,3,3,2)=CC\*(4.0\*R1(L)+2.0\*R2(L)+4.0\*R3(L))  
CM(L,3,3,1)=CC\*(2.0\*R1(L)+0.0\*R2(L)+2.0\*R3(L))  
CM(L,3,2,3)=CC\*(2.0\*R1(L)+4.0\*R2(L)+6.0\*R3(L))  
CM(L,3,2,2)=CC\*(2.0\*R1(L)+2.0\*R2(L)+4.0\*R3(L))  
CM(L,3,2,1)=CC\*(0.0\*R1(L)+0.0\*R2(L)+2.0\*R3(L))  
CM(L,3,1,3)=CC\*(2.0\*R1(L)+6.0\*R2(L)+4.0\*R3(L))  
CM(L,3,1,2)=CC\*(2.0\*R1(L)+4.0\*R2(L)+2.0\*R3(L))  
CM(L,3,1,1)=CC\*(0.0\*R1(L)+2.0\*R2(L)+0.0\*R3(L))  
CM(L,2,3,3)=CC\*(6.0\*R1(L)+4.0\*R2(L)+6.0\*R3(L))  
CM(L,2,3,2)=CC\*(4.0\*R1(L)+2.0\*R2(L)+4.0\*R3(L))  
CM(L,2,3,1)=CC\*(2.0\*R1(L)+0.0\*R2(L)+2.0\*R3(L))  
CM(L,2,2,3)=CC\*(2.0\*R1(L)+6.0\*R2(L)+4.0\*R3(L))  
CM(L,2,2,2)=CC\*(2.0\*R1(L)+4.0\*R2(L)+2.0\*R3(L))  
CM(L,2,2,1)=CC\*(0.0\*R1(L)+2.0\*R2(L)+0.0\*R3(L))  
CM(L,2,1,3)=CC\*(2.0\*R1(L)+6.0\*R2(L)+4.0\*R3(L))  
CM(L,2,1,2)=CC\*(2.0\*R1(L)+4.0\*R2(L)+2.0\*R3(L))  
CM(L,2,1,1)=CC\*(0.0\*R1(L)+2.0\*R2(L)+0.0\*R3(L))  
CM(L,1,3,3)=CC\*(6.0\*R1(L)+4.0\*R2(L)+6.0\*R3(L))  
CM(L,1,3,2)=CC\*(4.0\*R1(L)+2.0\*R2(L)+4.0\*R3(L))  
CM(L,1,3,1)=CC\*(2.0\*R1(L)+0.0\*R2(L)+2.0\*R3(L))  
CM(L,1,2,3)=CC\*(2.0\*R1(L)+6.0\*R2(L)+4.0\*R3(L))  
CM(L,1,2,2)=CC\*(2.0\*R1(L)+4.0\*R2(L)+2.0\*R3(L))  
CM(L,1,2,1)=CC\*(0.0\*R1(L)+2.0\*R2(L)+0.0\*R3(L))  
CM(L,1,1,3)=CC\*(2.0\*R1(L)+6.0\*R2(L)+4.0\*R3(L))  
CM(L,1,1,2)=CC\*(2.0\*R1(L)+4.0\*R2(L)+2.0\*R3(L))  
CM(L,1,1,1)=CC\*(0.0\*R1(L)+2.0\*R2(L)+0.0\*R3(L))

C 200 CONTINUE  
C

```

C 351 CONTINUE
C
C IF (NCOUNT .EQ. 1) GO TO 312
DO 210 L=1, NUMEL
LL=ITYPE(L)
IF (LL .NE. 0) GO TO 210
N1=ELNOD(L,1)
N2=ELNOD(L,2)
N3=ELNOD(L,3)
DO 88 J=1,3
DO 90 N=1,3
ALFA(N,J)=PSIIV(N1)*CM(L,N,J,1)+PSIIV(N2)*CM(L,N,J,2)
+PSIIV(N3)*CM(L,N,J,3)
1 CONTINUE
88 CONTINUE
DO 150 K=1,3
KK=ELNOD(L,K)
DO 180 I=1,3
II=ELNOD(L,I)
BIGC(KK,II)=BIGC(KK,II)+ALFA(K,I)*ZOMEGA(L)
180 CONTINUE
150 CONTINUE
210 CONTINUE

C
C INSERTION OF BOUNDARY POINTS
DO 285 I=1, NUPBP
II=IUPBP(I)
DO 291 J=1, NUMSNP
BIGC(J,II)=0
BIGC(II,J)=0
291 CONTINUE
285 CONTINUE

C
C DO 310 I=1, NUMSNP
DO 310 J=1, NUMSNP
BIGC(I,J)=0
PART(I,J)=0
DO 300 K=1, NUMSNP
BIGC(I,J)=BIGC(I,J)+(BIGAB(K,J)-BIGC(K,J))*BIGA(I,K)
PART(I,J)=PART(I,J)+(BIGAB(K,J)-2.*BIGC(K,J))*BIGA(I,K)
300 CONTINUE
310 CONTINUE

C
C

```

```

312 CONTINUE
C
C
C
C
C
C
C
C
350 1 CALL DVOGER(YVETTE,PSI,T,NUMSNP,MTH,MAXDER,JSTART,H,HMIN,HMAX,EPS,
C      YMAX,ERROR,PW,IER)
C
      IF(IER.EQ.0) GO TO 3540
      JSTART=-1
      H=HMIN*.1
      HMIN=H*.1
      GO TO 350
C
3540 CONTINUE
      JSTART=1
      IF((T-PTIME)/PTIME .LT. STAR) GO TO 352
      PTIME=T
      WRITE(5,3510)I,H,JSTART,IER
      FORMAT(/,2X,'T=',G12.6,2X,'H=',G12.6,2X,'JSTART=',I4,2X,'IER=',I4)
      WRITE(6,354)
      FORMAT(/,5(4X,'NODE',7X,'PSI',4X))
C
      DO 355 I=1,NUF
      I1=I+NUF
      I2=I+2*NUF
      I3=I+3*NUF
      I4=I+4*NUF
      WRITE(6,356) (I,PSI(I,I),I1,PSI(I,I1),I2,PSI(I,I2),I3,PSI(I,I3),I4,
      PSI(I,I4))
      CONTINUE
      355 FORMAT(5(4X,I3,3X,1PE12.4))
C
      NIT=1 + NIT
      IF (NIT.GT. 400) GO TO 377
      377 CONTINUE
C
      DO 350 I=1,NUMSNP
      S(I)=PSI(2,I)/H
      360 CONTINUE
C
      WRITE(6,357)
      FORMAT(/,5(4X,'NODE',7X,'DPSI',3X))
      DO 359 I=1,NUF

```

```

I1=I+NUF
I2=I+2*NUF
I3=I+3*NUF
I4=I+4*NUF
WRITE(6,356)(I,S(I),I1,S(I1),I2,S(I2),I3,S(I3),I4,S(I4))
359 CONTINUE
362 CONTINUE
C
DO 27 MN=1,NUMSNP
PSIV(MN)=PSI(I,MN)
DO 28 I=1,NUMSNP
BIGC(MN,I)=0
28 CONTINUE
27 CONTINUE
C
IF (T.LT.TF) GO TO 351
C
C
STOP
END

```

```

SUBROUTINE PEPSI (NUMEL,NUMSNP,R1,R2,R3,AREA,ELNOD,
1 BIGA,BIGAB,VLMBDA)
C*****
C PEPSI CALCULATES THE BIGA MATRIX AND PART OF THE BIGAB
C*****
C INTEGER*4 ELNDD
C DIMENSION R1(NJMEL),R2(NUMEL),R3(NUMEL),BIGA(NUMSNP,NUMSNP),
1 BIGAB(NUMSNP,NUMSNP),AMATRX(3,3),
2 VLMBDA(NUMEL),AREA(NUMEL),ELNOD(NUMEL,3)
C PI=3.1415927
C CALCULATE THE 3X3 D(I,J) MATRIX FOR ELEMENT L
C DO 200 L=1,NUMEL
C COEFFA=(PI/30.0)*AREA(L)
C AMATRX(1,1)=COEFFA *(6.0*R1(L)+2.0*R2(L)+2.0*R3(L))
C AMATRX(1,2)=COEFFA *(2.0*R1(L)+2.0*R2(L)+R3(L))
C AMATRX(2,1)=AMATRX(1,2)
C AMATRX(2,2)=CJEFFA *(R1(L)+6.0*R2(L)+2.0*R3(L))
C AMATRX(3,1)=AMATRX(1,3)
C AMATRX(3,2)=AMATRX(2,3)
C AMATRX(3,3)=CJEFFA *(2.0*R1(L)+2.0*R2(L)+6.0*R3(L))
C-----
C STORE ELEMENT MATRIX, AMATRX, INTO SYSTEM MATRIX, BIGA
C-----
C DO 20 K=1,3
C KK=ELNOD(L,K)
C DO 10 I=1,3
C II=ELNOD(L,I)
C BIGA(KK,II)=BIGA(KK,II)+AMATRX(K,I)
C BIGAB(KK,II)=BIGAB(KK,II)+VLMBDA(L)*AMATRX(K,I)
C CONTINUE
10 CONTINUE
20 CONTINUE
C 200 CONTINUE
RETURN
END

```

1 SUBROUTINE CRJISE (NUMEL, NUMSNP, R1, R2, R3, AREA, ELNOD, BIGAB, \*  
 A, B, VD)

CRUISE CALCULATES THE BIGAB MATRIX

INTEGER\*4 ELNOD

DIMENSION R1(NUMEL), R2(NUMEL), R3(NUMEL), AREA(NUMEL),  
 1 BIGAB(NUMSNP, NUMSNP), BMATRIX(3, 3),  
 2 ELNOD(NUMEL, 3), A(NUMEL, 3), B(NUMEL, 3), VD(NUMEL)

PI=3.1415927

-----  
 C CALCULATE THE 3X3 B(I, J) MATRIX FOR ELEMENT L

DO 200 L=1, NUMEL

COEFFB=PI\*VD(L)/(6.0\*AREA(L))

BMATRIX(1, 1)=COEFFB \*(-R1(L)\*(2.0\*B(L, 1)\*\*2+2.0\*A(L, 1)\*\*2)-R2(L)\*  
 1(B(L, 1)\*B(L, 2))+B(L, 1)\*\*2+A(L, 1)\*A(L, 2))-R3(L)\*(B(L, 1)\*  
 2B(L, 3)+B(L, 1)\*\*2+A(L, 1)\*A(L, 3))+2.0\*AREA(L)\*B(L, 1)

BMATRIX(1, 2)=COEFFB \*(-R1(L)\*(2.0\*B(L, 1)\*B(L, 2)+2.0\*A(L, 1)\*  
 1A(L, 2)\*B(L, 1))+B(L, 2)\*B(L, 1)\*B(L, 2))+2.0\*A(L, 1)\*A(L, 2))-  
 2R3(L)\*(B(L, 2)\*B(L, 3))+B(L, 1)\*B(L, 2)\*A(L, 3)+A(L, 1)\*A(L, 2))+  
 32.0\*AREA(L)\*B(L, 2)

BMATRIX(1, 3)=COEFFB \*(-R1(L)\*(2.0\*B(L, 1)\*B(L, 3)+2.0\*A(L, 1)\*  
 1A(L, 3)\*B(L, 1))+B(L, 3)\*B(L, 1)\*B(L, 3))+2.0\*A(L, 1)\*A(L, 3))+  
 2A(L, 3))-R3(L)\*(B(L, 3)\*B(L, 3))+2.0\*AREA(L)\*B(L, 3)+  
 32.0\*AREA(L)\*B(L, 3)

BMATRIX(2, 1)=BMATRIX(1, 2)

BMATRIX(2, 2)=COEFFB \*(-R1(L)\*(B(L, 2)\*\*2+B(L, 1)\*B(L, 2)+A(L, 2)\*\*2+  
 1A(L, 1)\*A(L, 2))-R2(L)\*(2.0\*B(L, 2)\*\*2+2.0\*A(L, 2)\*A(L, 3))-R3(L)\*(B(L, 3)\*  
 2B(L, 2)+B(L, 2)\*\*2+A(L, 2)\*A(L, 3))+2.0\*AREA(L)\*B(L, 2)

BMATRIX(2, 3)=COEFFB \*(-R1(L)\*(B(L, 2)\*B(L, 3)+B(L, 1)\*B(L, 3)+A(L, 2)\*  
 1A(L, 3)+A(L, 1)\*A(L, 3))-R2(L)\*(2.0\*B(L, 2)\*B(L, 3)+2.0\*A(L, 2)\*A(L, 3))-  
 2R3(L)\*(B(L, 3)\*B(L, 3))+2.0\*AREA(L)\*B(L, 3)+  
 32.0\*AREA(L)\*B(L, 3)

```

C      BMATRIX(3,1)=BMATRIX(1,3)
C      BMATRIX(3,2)=BMATRIX(2,3)
C      BMATRIX(3,3)=COEFFB *(-R1(L))*(B(L,3)**2+B(L,1)*B(L,3)+A(L,3)**2+
1A(L,1)+A(L,3))-R2(L)*(B(L,3)**2+B(L,2)*B(L,3)+A(L,3)**2+A(L,2)*
2A(L,3))-R3(L)*(2.0*B(L,3)**2+2.0*A(L,3)**2)+2.0*AREA(L)*B(L,3)

```

```

C      STORE ELEMENT MATRIX, AMATRIX, INTO SYSTEM MATRIX, BIGA
C-----

```

```

C      DO 20 K=1,3
C      KK=ELNOD(L,K)
C      DO 10 I=1,3
C      II=ELNOD(L,I)
C      BIGAB(KK,II)=BIGAB(KK,II)+BMATRIX(K,I)
C      10 CONTINUE
C      20 CONTINUE
C      200 CONTINUE
C      RETURN
C      END

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86 CONTINUE
DO 150 K=1,3
KK=ELNDD(L,K)
DO 280 I=1,3
II=ELNDD(L,I)
C
      BIGC(KK,II)=BIGC(KK,II)+ALFA(K,I)*ZOMEGA(L)
280 CONTINUE
150 CONTINUE
210 CONTINUE
C
      INSRRTION OF BOUNDARY POINTS
DO 285 I=1,NUM3P
II=IUPBP(I)
DO 281 J=1,NUMSNP
BIGC(J,II)=0.
BIGC(II,J)=0.
CONTINUE
281 CONTINUE
285 DO 311 I=1,NUM4SNP
DO 310 J=1,NUMSNP
BIGC(I,J)=0.
PART(I,J)=0.
DO 300 K=1,NUMSNP
BIGC(I,J)=BIGC(I,J)+(BIGAB(K,J)-BIGC(K,J))*BIGA(I,K)
PART(I,J)=PART(I,J)+(BIGAB(K,J)-2.*BIGC(K,J))*BIGA(I,K)
300 CONTINUE
310 CONTINUE
311 CONTINUE
C
      05 CONTINUE
C
      IF (IND.EQ.1) GO TO 205
C
DO 81 K=1,NUMSNP
DPSI(K)=0.
IF(K.GT.NUMEQ)GO TO 81
DO 80 I=1,NUMSNP
DPSI(K)=DPSI(K)+BIGCC(K,I)*PV(I)
80 CONTINUE
81 IF(T.GT.0.0) RETURN
WRITE(6,60)T
FORMAT(72X,'FROM YVETTE',5X,'T=',G20.10)
60 WRITE(6,65){DPSI(K),K=1,NUMSNP}
WRITE(6,65){PSI(1,K),K=1,NUMSNP}
65 FORMAT(10(1X,E11.5))

```

```
C      RETURN
C      CONTINUE
C      205
C      C
C      WHEN MTH = 1 COMPUTE THE JACOBIAN
C      DO 900 K=1,NUMSNP
C      DO 880 L=1,NUMSNP
C      MP(K,L)=PART(K,L)
C      CONTINUE
C      CONTINUE
C      880 CONTINUE
C      900 RETURN
C      END
```





```

11 FORMAT(/2X, '# OF ELEMENTS=', I5//2X, '# OF SYSTEM BOUNDARY POINTS=',
1 I5//2X, '# OF SYSTEM NODAL POINTS=', I5//2X, '# OF FUEL ELEMENTS=',
2 I5//2X, '# OF ELEMENTS IN A ROW=', I5//2X, 'MAX. # OF EVALUATIONS=', I
3 I5//2X, 'NCOUNT=', I5//2X, 'NCMPK=', I5)
WRITE(6,10) (I3P(I), I=1, NBP)

C
READ(5, 15) VELOC, DSUBF, DSUBC, SGMAAF, SGMAAC, SGMAF, FISFAC, HBAR
READ(5, 15) ZNJ, ALPHA, RMSEPS, AFUEL, TSTART, TEND, PINITY, SCALE
READ(5, 15) HMIN, HMAX
WRITE(6, 19) HMIN, HMAX
C
19 FORMAT(/2X, 'HMIN=', G12.6/2X, 'HMAX=', G12.6)

C
IF(TSTART.GT.0.) READ(5, 13) (PSTART(I), I=1, NUMSNP)
IF(TSTART.GT.0.) WRITE(6, 13) (PSTART(I), I=1, NUMSNP)
13 FORMAT(5G15.5)

C
READ(5, 10) MTH, MAXDER, NPROB, NTYPE, JPLOT, NPLOT, IRUN
WRITE(6, 12) NPROB, NTYPE, JPLOT, NPLOT, IRJN
12 FORMAT(/2X, 'NPROB=', I5/2X, 'NTYPE=', I5/2X, 'JPLOT=', I5/2X, 'NPLOT=', I
15/2X, 'IRUN=', I5)
IF(JPLOT.EQ.1) READ(5, 13) (TPLOT(I), I=1, NPLOT)
IF(JPLOT.EQ.1) WRITE(6, 13) (TPLOT(I), I=1, NPLOT)
IF(JPLOT.EQ.1) ITPC=1

C
WRITE(6, 16) VELOC, DSUBF, DSUBC, SGMAAF, SGMAAC, SGMAF, FISFAC, HBAR, ZNU,
1 ALPHA, RMSEPS, AFUEL, TSTART, TEND, PINITY, SCALE
16 FORMAT(/5X, 'VELOCLEAR DATA', /2X, 'VELOC=', G12.6/2X, 'DSUBF=', G12.6/2X
1, 'DSUBC=', G12.6/2X, 'SGMAAF=', G12.6/2X, 'SGMAAC=', G12.6/2X, 'SGMAF=',
2 G12.6/2X, 'FISFAC=', G12.6/2X, 'HBAR=', G12.6/2X, 'ZNU=', G12.6/2X,
3 'ALPHA=', G12.6/2X, 'RMSEPS=', G12.6/2X, 'AFUEL=', G12.6/2X,
4 'TSTART=', G12.6/2X, 'TEND=', G12.6/2X, 'PINITY=', G12.6/2X,
5 'SCALE=', E10.5)

C
15 FORMAT(8G10.5)

C
CALCULATE PHYSICAL CONSTANTS FROM NUCLEAR DATA

C
ICOUNT=0
ZLMBDA=ZNU*SGMAF/SGMAAF-1.
VDF=VELOC*DSUBF
VDC=VELOC*DSJBC
VLMBDA=VELOC*ZLMBDA*SGMAAF
ZOMEGA=2.*VELOC*SGMAAF*ALPHA*FISFAC*SGMAF/(HBAR*(AFUEL**2.))*SCAL
1E
IF(NTYPE.EQ.0) ZOMEGA=ZOMEGA*(AFUEL**2)/2.
VSMAC=VELOC*SGMAAC
PI=3.1415927

```

AD-A035 862

NAVAL POSTGRADUATE SCHOOL MONTEREY CALIF  
A COMPARISON OF INTEGRATION METHODS FOR THE SOLUTION OF NONLINE--ETC(U)  
DEC 76 R C SHELDRIK

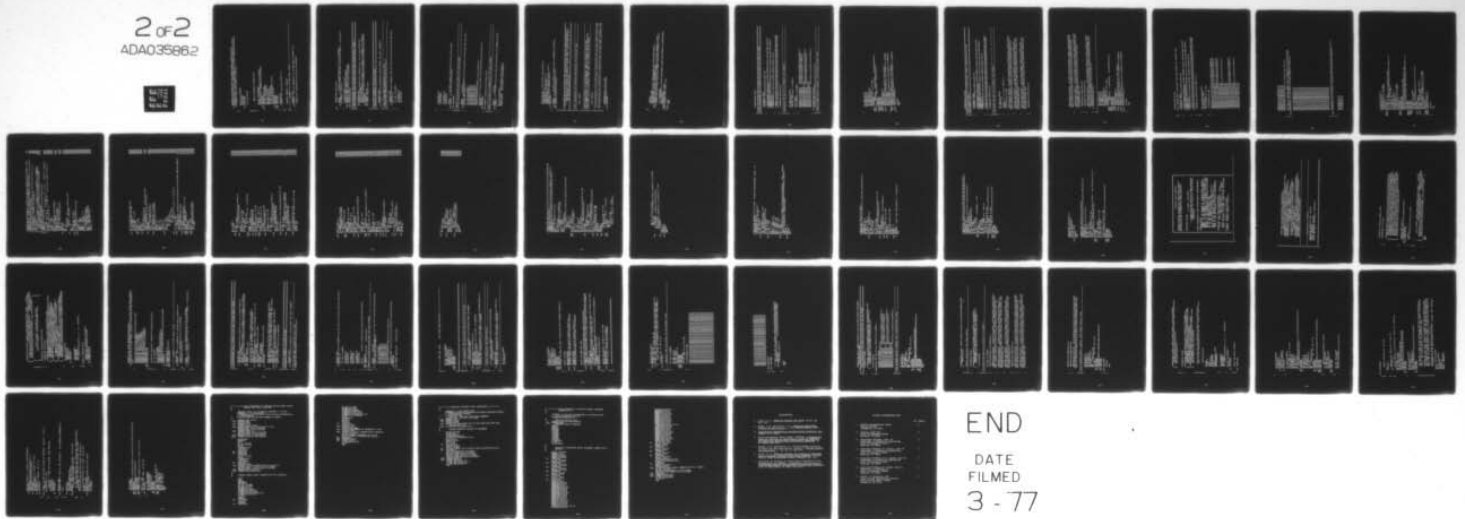
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WRITE(6,17)ZLMBDA,VDF,VDC,VLMBDA,ZOMEGA,VSMAC,VELOCT
17 FORMAT(//5X,'PHYSICAL CONSTANTS',/2X,'ZLMBDA=',G12.6/2X,'VDF=',
1G12.6/2X,'VDC=',G12.6/2X,'VLMBDA=',G12.6/2X,'ZOMEGA=',G12.6/2X,
2,'VSMAC=',G12.6/2X,'VELOCT=',G12.6)

```

C

```

DO 50 I=1,NUMSNP
Y(1,I)=0.
Y(2,I)=0.
50 CONTINUE

```

Y(1,1)=PINITY

C  
C  
C  
C  
C

NODAL NEIGHBOR CONNECTIVITY

```

DO 18 I=1,NUMSNP
READ(5,111)(NAME(I,J),J=1,NCMPK)
WRITE(6,111) (NAME(I,J),J=1,NCMPK)
CONTINUE
111 FORMAT(9(2X,I3))
DO 172 KK=1,NUMSNP
DO 171 II=1,NCMPK
BAGAB(KK,II)=0.
BAGAB(KK,II)=0.
DO 174 J=II,NCMPK
JJ=(II-1)*(2*NCMPK-II)/2+J
SCRAPC(KK,JJ)=0.
174 CONTINUE
171 CONTINUE
172 CONTINUE

```

174  
171  
172

C

```

WRITE(6,8)
8 FORMAT('1','VJCLEAR DATA'///)

```

C

C

NSNPSQ= NUMSNP \* NUMSNP

C

----- GEOMETRY OF SYSTEM NODAL POINTS -----

C

```

WRITE(6,40)
40 FORMAT(//1X,'GEOMETRY OF SYSTEM NODAL POINTS'///)
READ(5,41) (SYSNOD(I),R(I),Z(I),I=1,NUMSNP)

```

C

```

41  FORMAT(5X,15,2F15.5)
C
WRITE (6,60) (SYSNOD(I),R(I),Z(I),I=1,NUMSNP)
60  FORMAT (1X,'SYSTEM NODAL POINT NO.',13,5X,'R COORD.= ',F10.6,5X,'
1Z COORD.= ',F13.6)
C
IF(JPLJ1.EQ.0) GO TO 72
WRITE(7,7000) IRUN,HBAR,NUMSNP
7000 FORMAT(1, RUN NJMBER=,13,F10.5,15)
WRITE(7,7001) (R(I),Z(I),I=1,NUMSNP)
7001 FORMAT(4E15.6)
C
*****
C CORRESPONDENCE TABLE BETWEEN SYSTEM AND ELEMENT NODAL POINTS
*****
C
72  WRITE(6,73)
73  FORMAT(//2X,'CONNECTIVITY MATRIX',//, ' EL #',33X,'TYPE'//)
DO 75 I=1,NUMEL
READ(5,76) ELEMENT(I),ELNOD(I,1),ELNOD(I,2),ELNOD(I,3), ITYPE(I)
WRITE(6,77) ELEMENT(I),ELNOD(I,1),ELNOD(I,2),ELNOD(I,3), ITYPE(I)
75  CONTINUE
76  FORMAT (5I10)
77  FORMAT (2X,13,4I10)
C
*****
C CALCULATE GEOMETRY FOR EACH ELEMENT
*****
C
WRITE(6,110)
110 FORMAT (1X//1X,'GEOMETRY CALCULATIONS FOR EACH ELEMENT')
C
-----
C LOAD SYSTEM NODE COORDINATES (R,Z) INTO
ELEMENT NODE COORDINATES (R1,R2,R3,Z1,Z2,Z3)
-----
C
WRITE(6,95)
95  FORMAT (1X//1X,'ELEMENT',5X,'R1',9X,'Z1',9X,'R2',9X,'Z2',9X,'R3',
19X,'Z3,/')
C
DO 100 I=1,NUMEL
J=ELNOD(I,1)
R1(I)=R(J)
Z1(I)=Z(J)

```



```

A31(L)=0.5* (R3(L)*Z1(L)-R1(L)*Z3(L))
ZX=Z3(L)-Z2(L)
IF(ZX.EQ.0.) WRITE(6,161) ZX
IF(ZX.EQ.0.) GO TO 170
GAMMA(L)=(R3(L)-R2(L))/(Z3(L)-Z2(L))
DELTA(L)=R3(L)-GAMMA(L)*Z3(L)
170 CONTINUE
C
WRITE (6,160) L,A12(L),A23(L),A31(L),GAMMA(L),DELTA(L)
160 FORMAT (1X,2X,I2,4X,5(F10.4,1X))
161 FORMAT (//2X,2X=,G12.6)
C
*****
CALCULATION OF THE SYSTEM INFLUENCE MATRICES FOR THE FIELD EQUATION IN
PEPSI(CONSTANT TERM), CRUISE(DEL SQUARE TERM), AND PSISQ (OR MOE)
FOR THE NON LINEAR TERM.
*****
-----
PEPSI CALCULATES THE D-MATRIX MULTIPLYING THE D(CY)/DT TERM
-----
CALL PEPSI
-----
CRUISE CALCULATES THE SYSTEM MATRIX, BIGA, ASSOCIATED WITH THE
DEL SQUARE TERM. IT THEN COMBINES BIGA WITH BIGB, THE SYSTEM MATRIX
ASSOCIATED WITH THE CONSTANT TERM, INTO THE SYSTEM MATRIX, BIGAB.
-----
CALL CRUISE
-----
PSISQ CALCULATES THE SYSTEM MATRIX ASSOCIATED WITH THE NON LINEAR PSI*2 TERM
MOE CALCULATES THE SYSTEM MATRIX ASSOCIATED WITH THE NON LINEAR INTEGRAL TERM
-----
IF(NPROB.EQ.1) GO TO 199
IF(INTYPE.EQ.0) CALL PSISQ
199 CONTINUE

```

```

C
  JSKF=0
  WRITE(6,317)
  WRITE(6,318) TSTART, TEND, MAXDER, JSKF, HMIN, HMAX, RMSEPS
317 FORMAT(//5X, 'INITIAL ARGUMENTS: //')
318 FORMAT(2X, 'TSTART=', G12.6/2X, 'TEND=', G12.6/2X, 'MAXDER=', I5/2X,
1, 'JSKF=', I5/2X, 'HMIN=', G12.5/2X, 'HMAX=', G12.6/2X, 'RMSEPS=', G12.6)
C
  H = HMIN*1000.
  CALL SDESOL(Y, YL, TSTART, TEND, NDE, NL, NDE, JSKF, MAXDER, 1, H, HMIN, HMAX,
  RMSEPS, MS)
9998 WRITE(6,9997) JSKF
9997 FORMAT(//2X, 'JSKF=', I5)
C
9999 STOP
      END

```

```

C *****
C SUBROUTINE PEPSI
C *****
C PEPSI CALCULATES THE A-MATRIX MULTIPLYING THE D(CY)/DT TERM
C *****
C
C INTEGER*2 NAME
C INTEGER*4 SYSNJD, ELEMNT, ELNOD, RHO
C
C COMMON/NUMB/ELNOD(220,3), NUMEL, NUMSNP, VFULEL, NELROW, NBP, IBP(30)
C 1, NPROB, NTYPE, ITYPE(220)
C COMMON/CMCL/BAGA(132,7), BAGAB(132,7), SCRAPC(132,28), NDE, NCMPPK,
C 1NAME(132,7)
C COMMON/SOLVE/PI, VDF, VDC, VLMBDA, ZOMEGA, VSMAAC, SCALE
C
C COMMON R1(220), R2(220), R3(220), A(220,3), B(220,3), AREA(220), Z1(220)
C 1, Z2(220), Z3(220), A12(220), A23(220), A31(220), GAMMA(220), DELTA(220)
C
C DIMENSION AMATRIX(3,3)
C
C-----
C CALCULATE THE 3X3 D(I,J) MATRIX FOR ELEMENT L
C-----
C
C DO 200 I=1, NUMEL
C LL=ITYPE(L)
C COEFFA= (PI/30.0) * AREA(L)
C
C CC=-VSMAAC
C IF(LL.EQ.0) CC=VLMBDA
C
C AMATRIX(1,1)=COEFFA *(6.0*R1(L)+2.0*R2(L)+2.0*R3(L))
C AMATRIX(1,2)=COEFFA *(2.0*R1(L)+2.0*R2(L)+R3(L))
C AMATRIX(1,3)=COEFFA *(2.0*R1(L)+R2(L)+2.0*R3(L))
C AMATRIX(2,1)=AMATRIX(1,2)
C AMATRIX(2,2)=COEFFA *(2.0*R1(L)+6.0*R2(L)+2.0*R3(L))
C AMATRIX(2,3)=COEFFA *(R1(L)+2.0*R2(L)+2.0*R3(L))
C AMATRIX(3,1)=AMATRIX(1,3)
C AMATRIX(3,2)=AMATRIX(2,3)
C AMATRIX(3,3)=COEFFA *(2.0*R1(L)+2.0*R2(L)+6.0*R3(L))
C
C 35 FORMAT(2X,3E20.10)
C WRITE(6,35) ((AMATRIX(I,J),J=1,3),I=1,3)
C-----
C STORE ELEMENT MATRIX, AMATRIX, INTO SYSTEM MATRIX, BIGA
C-----
C

```

```

C
DO 7050 K=1,3
KK=ELNOD(L,K)
DO 7040 I=1,3
II=ELNDD(L,I)
DO 7035 M=1,NCMPK
KKM=NAME(KK,M)
IF(KKM.EQ.II) GO TO 7032
CONTINUE
BAGAB(KK,M)=BAGAB(KK,M)+CC*AMATRX(K,I)
7035 BAGAB(KK,M)=BAGAB(KK,M)+AMATRX(K,I)
7032
7040 CONTINUE
7050 CONTINUE
200 CONTINUE
8074 FORMAT(1P3E15.4)
WRITE(6,8071) BAGAB(KK,M)
8071 CONTINUE
8071 WRITE(6,8072) ((BAGA(I,J),J=1,NCMPK),I=1,NUMSNP)
CONTINUE
8071 WRITE(6,8073) ((BAGA(I,J),J=1,NCMPK),I=1,NUMSNP)
8071 WRITE(7,8076) ((BAGA(I,J),J=1,NCMPK),I=1,NUMSNP)
C 8076 FORMAT(1P7E11.4)
8073 FORMAT(//2X,'BAGAB MATRIX FROM PEPSI')
8073 WRITE(6,8072) ((BAGAB(I,J),J=1,NCMPK),I=1,NUMSNP)
C 8072 FORMAT(1P7E15.4)
CONTINUE
RETURN
END
C

```

```

C***** SUBROUTINE CRJISE *****
C CRUISE CALCULATES THE SYSTEM MATRIX, BIGB, ASSOCIATED WITH THE
C DEL SQUARE TERM.
C*****
C
C INTEGER *2 NAME
C INTEGER *4 SYSNJD, ELEMNT, ELNOD, RHO
C
C COMMON/NUMB/ELVDD(220,3), NUMEL, NUMSNP, VFULEL, NELROW, NBP, I8P(30)
C 1, NPROB, NTYPE, ITYPE(220)
C COMMON/CMCL/BASA(132,7), BAGAB(132,7), SCRAPC(132,28), NDE, NCMPK,
C 1 NAME(132,7)
C COMMON/SOLVE/PI, VDF, VDC, VLMBDA, ZOMEGA, VSMAAC, SCALE
C
C COMMON R1(220), R2(220), R3(220), A(220,3), B(220,3), AREA(220), Z1(220)
C 1, Z2(220), Z3(220), A12(220), A23(220), A31(220), GAMMA(220), DELTA(220)
C
C DIMENSION BMATRIX(3,3)
C
C-----
C CALCULATE THE 3X3 B(I,J) MATRIX FOR ELEMENT L
C-----
C
C DO 200 L=1, NUMEL
C LL=ITYPE(L)
C
C COEFFB= PI / (6.0 * AREA(L) )
C CC=VDF
C IF(LL.NE.0) CC=VDC
C
C BMATRIX(1,1)=COEFFB *(-R1(L))*(2.0*B(L,1)**2+2.0*A(L,1)**2)-R2(L)*
C 1(B(L,1)*B(L,2))+B(L,1)**2+A(L,1)*A(L,2)+A(L,1)*B(L,1)*
C 2B(L,3)+B(L,1)**2+A(L,1)*A(L,3)**2+2.0*AREA(L)*B(L,1)
C
C BMATRIX(1,2)=CJEFFB *(-R1(L))*(2.0*B(L,1)*B(L,2)+2.0*A(L,1)*
C 1A(L,2))-R2(L)*(B(L,2)*B(L,1)+B(L,2)+A(L,2)*A(L,1)+A(L,2))
C 2R3(L)*(B(L,2)*B(L,3))+B(L,1)*B(L,2)+A(L,2)*A(L,3)+A(L,2)*
C 32.0*AREA(L)*B(L,2)
C
C BMATRIX(1,3)=CJEFFB *(-R1(L))*(2.0*B(L,1)*B(L,3)+2.0*A(L,1)*
C 1A(L,3))-R2(L)*(B(L,3)+B(L,1)*B(L,3)+A(L,3)+A(L,3))
C 2A(L,3))-R3(L)*(B(L,3)*B(L,3))+2*B(L,1)*B(L,3)+A(L,3)*
C 32.0*AREA(L)*B(L,3)
C
C BMATRIX(2,1)=BMATRIX(1,2)
C

```

```

C
BMATRIX(2,2)=COEFFB *(-R1(L)*(B(L,2)**2+B(L,1)*B(L,2)+A(L,2)**2+
1A(L,1)*A(L,2))-R2(L)*2.0*B(L,2)**2+2.0*A(L,2)*B(L,3)-R3(L)*B(L,3)*
2B(L,2)+B(L,2)**2+A(L,2)*A(L,2)+2.0*AREA(L)*B(L,2))
C
BMATRIX(2,3)=COEFFB *(-R1(L)*(B(L,2)*B(L,3)+B(L,1)*B(L,2)+A(L,2)*
1A(L,3)+A(L,1)*A(L,3))-R2(L)*(2.0*B(L,2)*B(L,3)+2.0*A(L,2)*A(L,3))-
2R3(L)*(B(L,3)+A(L,3)**2+A(L,2)*A(L,3))+
32.0*AREA(L)*B(L,3))
C
BMATRIX(3,1)=BMATRIX(1,3)
C
BMATRIX(3,2)=BMATRIX(2,3)
C
BMATRIX(3,3)=COEFFB *(-R1(L)*(B(L,3)**2+B(L,1)*B(L,3)+A(L,3)**2+
1A(L,1)*A(L,3))-R2(L)*(B(L,3)*B(L,2)+B(L,3)*A(L,3)**2+A(L,2)*
2A(L,3))-R3(L)*(2.0*B(L,3)**2+2.0*A(L,3)*B(L,3))+2.0*AREA(L)*B(L,3))
C
WRITE(6,35) ((BMATRIX(I,J),J=1,3),I=1,3)
C
35 FORMAT(2X,3E20.10)
C

```

```

C
DO 7050 K=1,3
KK=ELNOD(L,K)
DO 7040 I=1,3
II=ELNOD(L,I)
DO 7035 M=1,NCMPK
KKM=NAME(KK,M)
IF(KKM.EQ.II) GO TO 7032
CONTINUE
7035 BAGAB(KK, M)=BAGAB(KK, M)+CC*BMATRIX(K,I)
7032 CONTINUE
7040 CONTINUE
7050 CONTINUE
200 CONTINUE
8074 FORMAT(1P3E15.4)
8073 WRITE(6,8073)
8073 FORMAT(/,2X,'BAGAB MATRIX FROM CRUISE')
8072 WRITE(6,8072)((BAGAB(I,J),J=1,NCMPK),I=1,NUMSNP)
C
8072 WRITE(7,8072) ((BAGAB(I,J),J=1,NCMPK),I=1,NUMSNP)
C
8076 FORMAT(1P7E11.4)
C
RETURN
C
END

```





```

DO 8035 M=1,NCMPK
KKM=NAME(KK,M)
IF(KKM.EQ.0) GO TO 8032
IF(KKM.EQ.1) GO TO 8032
CONTINUE
DO 8030 J=1,3
IF(I.EQ.J) KMN=(M-1)*(2*NCMPK-M)/2+M
IF(I.EQ.J) SCRAPC(KK,KMN)=SCRAPC(KK,KMN)+CM(K,I,J)
IF(I.EQ.J) GO TO 8030
JJ=ELND(L,I,J)
DO 8020 N=1,NCMPK
KKN=NAME(KK,N)
IF(KKN.EQ.JJ) GO TO 8015
IF(KKN.EQ.0) GO TO 8015
CONTINUE
8020 MX=MAXO(M,N)
8015 MN=MINO(M,N)
KMN=(MN-1)*(NCMPK*2-MN)/2+MX
SCRAPC(KK,KMN)=SCRAPC(KK,KMN)+CM(K,I,J)
IF(I.NE.J) SCRAPC(KK,KMN)=SCRAPC(KK,KMN)+CM(K,I,J)
CONTINUE
CONTINUE
CONTINUE
CONTINUE
CONTINUE
DO 8060 I=1,NUMSNP
DO 8060 J=1,NUMSNP
SCRAPC(I,J)=+ZOMEGA*SCRAPC(I,J)*(1.E+14)
CONTINUE
WRITE(6,8074)
FORMAT('SCRAPC FROM PSISQ')
DO 8080 K=1,NUMSNP
WRITE(6,8075) (SCRAPC(K,L),L=1,NCC)
WRITE(7,8075) (SCRAPC(K,L),L=1,NCC)
CONTINUE
FORMAT('IP7E15.4')
8080
8075
8076
C
C
C
RETURN
END

```

```

SUBROUTINE LOASUB(Y, YL, T, TEND, NY, NL, M1, JSTART, KFLAG, MAXOR, IPRT,
1 H, HMIN, HMAX, RMSEPS, SAVE, YLSV, YMAX, ER, ES, FI, DY)
DIMENSION Y(7,1), YL(1), SAVE(7,1), YMAX(1), YLSV(1), FI(1), A(7),
1 PERT(6,3), COF(21), ES(1), DY(1)
DATA PERT/4.,9.,16.,25.,36.,49.,9.,16.,25.,36.,49.,64.,1.,1.,
1 .25,2.7889E-2,1.70569E-3,6.83929E-5/
DATA KZILCH/0/
DATA COF/-1., -1.5, -0.5, -1.833333, -1.833333, -1.666667, -2.083333,
1 -1.458333, -1.666667, -0.4166667, -2.04166667, -2.833333, -1.875, -7.083333,
2 -1.25, -0.08333333, -2.245, -2.255556, -1.208333, -2.430556,
3 -.02916667, -.00138889/
1 FORMAT(2I5,12I10.2,7E14.6/(32X,7E14.6))
2 FORMAT(32X,1P7E14.6)
3 FORMAT(TEND = ,D9.2, , N = ,I3, , NL = ,I3, , RMSEPS = ,1PD9.2,
4 IF(JSTART.GT.0)GO TO 60
N = NY + NL
H = ,D9.2, , H = ,D9.2, , H = ,8X, , T = ,8X, , Y(1,*) AND YL(*)'//)
LCOPYY = 7*NY
LCOPYL = NL
EPS = SQRT(FLOAT(M1))*RMSEPS
MAXDER = MAXOR
IF(MAXDER.GT.5)MAXDER = 6
IF(IPRT.LE.0)GO TO 10
PRINT 3, N,NL,RMSEPS,TEND,H
PRINT 4
10 CONTINUE
NW = 0
DO 20 J=1,NY
YMAX(J) = AMAX1(1.,ABS(Y(1,J)))
Y(2,J) = Y(2,J)*H
NQ = 1
BR = 1.
ASSIGN 100 TO IRET
K = NQ*(NQ - 1)/2
CALL COPYZ(A(2),COF(K+1),NQ)
230 K = NQ + 1
IDDOB = NQ
ENQ1 = .5/NQ
ENQ2 = .5/K
ENQ3 = .5/(NQ + 2)
PEPSH = EPS*2
EUP = PERT(NQ,1)*PEPSH
EDWN = PERT(NQ,2)*PEPSH
BND = (EPS*ENQ3)**2
IMEVAL = 1

```

```

100
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400
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700
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1100
1200

1600
1700
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2800
1900
2000
2100

2300
2400

2900
3000

3100
3200
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3400
3500
3600
3700
3800
4000
4100
4300
4400
4500
4600
4700
4800
4900
5000
5100
5200
5300

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6000
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6200
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7200
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8900
9000
9100
11000
11100
11200
11700
11800
11900
12000
12100
12200
12300
12400
12600

GO TO IRET,(100,250,630,751)
IF(H.EQ.HNEW)GOTO 100
R = H/HNEW
ASSIGN 100 TO IRET
GO TO 800
KFLAG = -2
JSTART = NQ
HNEW = H
RETURN
NS = NS + 1
IF(IPRT.LE.0)GO TO 95
PRINT 1,NS,NW,NQ,H,T (Y(1,I),I=1,NY)
IF(NL.GT.0)PRINT 2,(YL(I),I=1,NL)
95 CONTINUE
IF(KFLAG.LT.0)GO TO 80
IF(T .GE.TEND)GO TO 80
JSTART = 1
CALL COPYZ(SAVE,Y,LCCOPY)
CALL COPYZ(YLSV,YL,LCCOPY)
RACUM = 1.
KFLAG = 1
HOLD = H
NQOLD = NQ
TOLD = T
KZILCH = 1
T = T + H
250 HINV = 1./H
DO 260 J=2,K
J3 = K+J-1
DO 260 J1=J,K
J2 = J3 - J1
DO 260 I=1,NY
Y(J2,I) = Y(J2,I) + Y(J2+1,I)
260 DO 270 I=1,NY
ER(I) = 0.
CALL NXSTP(Y,YL,T,HINV,A(2),EPS,NY,NL,DY,F1,ER,YMAX,BND,BR,IWEVAL,
1 KFLAG,NW,KRET)
GO TO (490,440,780),KRET
440 T = TOLD
IF(IWEVAL)445,455,450
445 IF(H.LE.HMIN#1.00001)GO TO 460
450 RACUM = RACUM#.25
455 CONTINUE
GO TO 750
460 KFLAG = -3
470 CALL COPYZ(Y,SAVE,LCCOPY)
CALL COPYZ(YL,YLSV,LCCOPY)
H = HOLD

```

12700  
12800  
12900  
13000  
13010  
13100  
13200  
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13500  
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13900  
14000  
14100  
14200  
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14900  
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15300  
15400  
15410  
15500  
15600  
15700  
15800  
15900  
16000  
16100  
16200  
16210  
16300  
16400  
16500  
16600  
16700  
16800  
16900  
17000  
17100

```

490 NQ = NQOLD
    GO TO 90
    D = 0
    DO 500 I=1,M1
      YM = AMAX1(ABS(Y(I,I)), YMAX(I))
      D = D + (ER(I)/YM)**2
    IMEVAL = 0
    IF(D.GT.E) GO TO 540
    IF(K.LT.3) GO TO 520
    DO 510 J=3,K
      DO 510 I=1,NY
        Y(J,I) = Y(J,I) + A(J)*ER(I)
    KFLAG = 1
    IDOUB = 1
    IF(IDOUB) 550, 525, 700
    CALL COPYZ(ESV,ER,M1)
    GO TO 700
    IF(JSTART.GT.0) GO TO 548
    DO 544 I=1,NY
      SAVE(2,I) = Y(2,I)
    KFLAG = KFLAG - 2
    IF(H.LE.HMIN) GO TO 740
    T = TOLD
    IF(KFLAG.LE.-5) GO TO 720
    PR2 = (D/E)*ENQ2*1.2
    L = 0
    IF(NQ.LE.1) GO TO 570
    D = 0
    DO 560 J=1,M1
      YM = AMAX1(ABS(Y(I,J)), YMAX(J))
      D = D + (Y(K,J)/YM)**2
    PR1 = (D/EDWN)*ENQ1*1.3
    IF(PR1.GE.PR2) GO TO 570
    PR2 = PR1
    L = -1
    IF(KFLAG.LT.0 .OR. NQ.GE.MAXDER) GO TO 590
    D = 0
    DO 580 J=1,M1
      YM = AMAX1(ABS(Y(I,J)), YMAX(J))
      D = D + ((ER(J) - ESV(J))/YM)**2
    PR1 = (D/EUP)*ENQ3*1.4
    IF(PR1.GE.PR2) GO TO 590
    PR2 = PR1
    L = 1
    R = 1./AMAX1(PR2,1.E-5)
    IF(KFLAG.LT.0 .OR. R.GE.1.1) GO TO 600
    IDOUB = 9
    GO TO 700

```

17200  
17300  
17400  
17500  
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17700  
17800  
17900  
18000  
18100  
18200  
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18500  
18600  
18700  
18800  
18900  
19000  
19100  
19200  
19300  
19400  
19500  
19600  
19700  
19800  
19900  
20000  
20100  
20200  
20300  
20400  
20500  
20600  
20700  
20800  
20900  
21000  
21100  
21200  
21300  
21400  
21500  
21600  
21700  
21720

```

600 NEWQ = NQ + L
    K = NEWQ + 1
    IF(NEWQ.LE.NQ)GO TO 620
    RI = A(NEWQ)/FLOAT(NEWQ)
    DO 610 J=1,NY
        ER(J)*RI
610 CONTINUE
620 IDOUB = NQ
    IF(NEWQ.EQ.NQ)GO TO 630
    NQ = NEWQ
    ASSIGN 630 TO IRET
630 GO TO 170
    IF(KFLAG.GT.0)GO TO 670
    RACUM = RACUM*R
    GO TO 750
670 H = AMAX1(HMAX/H,R),HMIN/H
    IWEVAL = 1
    ASSIGN 700 TO IRET
    GO TO 800
700 DO 710 I=1,M1
710     YMAX(I) = AMAX1(ABS(Y(I,I)),YMAX(I))
720 GO TO 90
    IF(NQ.EQ.1)GO TO 735
    NQ = 1
    IDOUB = 1
    ASSIGN 751 TO IRET
    GO TO 170
735 NQOLD = 1
    KFLAG = -4
740 GO TO 470
    KFLAG = -1
750 H = HOLD*RACUM
751 H = AMAX1(HMIN,AMIN1(H,HMAX))
    RI = 1
    DO 760 J=2,K
        RI = RI*RACUM
        DO 760 I=1,NY
            Y(J,I) = SAVE(J,I)*RI
760 DO 770 I=1,NY
770     Y(I,I) = SAVE(I,I)
        CALL COPY2(YL,YLSV,LCOPYL)
    IWEVAL = 1
    GO TO 250
780 KFLAG = -5
    GO TO 80

```

```

800 R1 = 1.
DO 810 J=2,K
R1 = R1*K
DO 810 I=1,NY
Y(J,I) = Y(J,I)*R1
GO TO JRET,(100,700)
ENTRY REDSUB
IF(KZILCH.EQ.0)RE TURN
DO 910 I=1,NY
Y(1,I) = SAVE(1,I)
Y(2,I) = SAVE(2,I)/HOLD
CALL COPYZ(YL,YLSV,LCOPYL)
T = TOLD
RETURN
END

```

```

21800
21900
22000
22100
22200
22300
22400
22500
22600
22700
22800
22900
23000
23100
23200

```

```

SUBROUTINE NXSTP(Y, YL, T, HINV, A2, EPS, NY, NL, DY, F1, ER, YMAX, BND, BR,
1 IWEVAL, KFLAG, NW, KRET)
DIMENSION Y(7,1), YL(1), DY(1), ER(1), F1(1), YMAX(1), PW(110,7)
COMMON/CMCL/A(132,7), B(132,7), C(132,28), N, NNZ, K(132,7)
INTEGER*2 K
DATA SPD, SPDM1/1.05, .05/
KRET = 1
DO 430 L=1,3
CALL DIFFUN(Y, YL, T, HINV, DY)
CALL IF(IWEVAL.LT.1)GO TO 280
NOIT = N**2
EPS = EPS**2
EPSSA2 = EPSS*.0001
CALL JACMAT(Y, YL, T, HINV, A2, EPS, NY, NL, DY, F1, PW)
KFLAG = 1
IWEVAL = -1
NW = NW + 1
DO 281 I=1, NY
F1(I) = DY(I)/PW(I,1)
DO 287 IT = 1, NOIT
RCH = 0.
CH = 0.
DO 285 I=1, NY
FN = DY(I)
DO 284 J=2, NNZ
IF(K(I, J).LE.0.OR.K(I, J).GT. NY) GO TO 284
FN = FN - PW(I, J)*F1(K(I, J))
CONTINUE
FN = FN/PW(I,1)
FN = FN*SPD - SPDM1*F1(I)
ACH = F1(I) - FN
RCH = RCH + (ACH/VMAX(I))**2
F1(I) = FN
IF(RCH.LT.EPSS)GO TO 288
IF(CH.LE.EPSSA2)GO TO 288
CONTINUE
KRET = 3
RETURN
CONTINUE
IF(NL.LE.J)GO TO 300
DO 290 I=1, NL
YL(I) = YL(I) - F1(I+NY)
CONTINUE
DEL=0.
DO 420 I=1, NY
Y(1, I) = Y(1, I) - F1(I)
Y(2, I) = Y(2, I) + A2*F1(I)

```

```

420 ER(I) = ER(I) + F1(I)
      DEL = DEL + (F1(I)/AMAX1(YMAX(I),ABS(Y(1,I))))**2
      CONTINUE
      IF(L.GE.2)BR = AMAX1(.9*BR,DEL/DEL1)
      DEL1 = DEL
      IF(AMINI(DEL,BR*DEL*2.).LE.BND)GO TO 450
      CONTINUE
      KRET = 2
      RETURN
450 CONTINUE
      RETURN
      END

```

```

SUBROUTINE SDESOL(Y, YL, T, TEND, NY, NL, M, JSKF, MAXDER, IPRT, H,
1 HMIN, HMAX, EPS, W)
DIMENSION Y(7,1), YL(1), W(1)
IF(JSKF.NE.0) GO TO 200
DO 110 I=1, NY
110 Y(2, I) = 0.
CALL DIFFUN(Y, YL, 0., 1., W)
CALL Derval(W) (RETR)
IF(KRETR.NE.0) GO TO 300
DO 120 I=1, NY
120 Y(2, I) = W(I)
NSV = NY + NL
NSVL = 7*NY + 1
NYMAX = NSVL + NL
NER = NYMAX + V
NESV = NER + NY
NFI = NESV + NY
NDY = NFI + N
JS = JSKF
200 CALL LDASUB(Y, YL, T, TEND, NY, NL, M, JS, KF, MAXDER, IPRT, H, HMIN,
1 HMAX, EPS, W, W(NSVL), W(NYMAX), W(NER), W(NESV), W(NFI), W(NDY))
JSKF = ISIGN(JS*10 + IABS(KF), KF)
RETURN = -6
300 JSKF = -6
RETURN
END

```

```

SUBROUTINE DERVAL(DY, KERET)
COMMON/CMCL/A(132,7), B(132,7), C(132,28), N, NNZ, K(132,7)
INTEGER*2 K
DIMENSION DY(1)
DO 100 I=1, N
  DY(I+N) = -DY(I)
  DY(I) = -DY(I)/A(I,1)
DO 200 L=1, 500
  DO 180 I=1, N
    DYN = DY(I+N)
    E = 0
    DO 150 J=2, NNZ
      IF(K(I, J).LE.0) CR, K(I, J), GT, N) GO TO 150
      DYN = DYN - A(I, J)*DY(K(I, J))
    CONTINUE
    DYN = DYN/A(I,1)
    EY(I) = AMAX1(ABS((DYN - DY(I))/DYN), E)
    IF(E.LT.1.E-4) GO TO 300
  CONTINUE
  PRINT 1, E
  IF(E.LE.1.E-2) GO TO 300
  KERET = 1
  RETURN = 0
  RETURN
  RETURN = 0
  RETURN
  FORMAT(' AFTER 500 ITERATIONS, E =', 1PE10.3)
  1 END

```

```

SUBROUTINE JACMAT(Y, YL, I, HINV, A2, EPS, NY, NL, DY, F1, PW)
COMMON/CMCL/A(132,7), B(132,7), C(132,28), N, NNZ, K(132,7)
INTEGER*2 K, P
DIMENSION Y(7,1), YL(1), F1(1), DY(1), PW(110,1)
DIMENSION P(8,B)
DATA P/64*0/, INITP/0/
IF(INITP.EQ.NVZ)GO TO 99
INITP = NNZ
DO 98 L=1,NNZ
DO 98 M = L, NVZ
DO 99 P(L,M) = M + (L - 1)*(2*NNZ - L)/2
CONTINUE
AH = -A2*HINV
DO 300 I=1, NY
DO 300 J=1, NNZ
PW(I,J) = AH*A(I,J) - B(I,J)
DO 100 L=1, J
IF(K(I,L).LE.0)GO TO 295
PW(I,J) = PW(I,J) + C(I,P(L,J))*Y(L,K(I,L))
DO 200 M=J, NNZ
IF(K(I,M).LE.0)GO TO 295
PW(I,J) = PW(I,J) + C(I,P(J,M))*Y(L,K(I,M))
CONTINUE
RETURN
END

```

```

SUBROUTINE COPYZ(S,Y,L)
DIMENSION S(1),Y(1)
IF(L.LE.0)RETURN
DO 100 J=1,L
S(J) = Y(J)
RETURN
END
100

SUBROUTINE DIFFUN(YL,I,HINV,DY)
COMMON/CMCL/A(132,7),B(132,7),C(132,28),N,NNZ,K(132,7)
INTEGER*2 K
DIMENSION Y(7,1),YL(1),DY(1)
DO 400 I=1,N
DY(I) = 0
DO 300 J=1,NNZ
IF(K(I,J).LE.0)GO TO 310
DY(I) = DY(I) + Y(2;K(I,J))*A(I,J)*HINV - B(I,J)*Y(1,K(I,J)) +
1 C(I,J)*Y(1,K(I,J))*Y(1,K(I,1))
300 CONTINUE
310 L = NNZ
DO 360 J1=2,NNZ
IF(K(I,J1).LE.0)GO TO 400
DO 350 J2=J1,NNZ
L = L + 1
IF(K(I,J2).LE.0)GO TO 350
DY(I) = DY(I) + C(I,L)*Y(1,K(I,J1))*Y(1,K(I,J2))
350 CONTINUE
360 CONTINUE
400 CONTINUE
END

```



```

THESE ARE THE DIMENSIONING REQUIREMENTS.
PART( NUMSNP, LCON), BIGA( NUMSNP, LCON),
BIGAB( NUMSNP, LCON), BIGC( NUMSNP, LCON), BIGCC( NUMSNP, LCON),
MNOD( NUMSNP, MNOD), V( NUMEL), D( NUMEL), SGA( NUMEL),
SGF( NUMEL), ALPHA( NUMEL), PSIIV( NUMSNP), PDPSI( NUMSNP),
ZLMBDA( NUMEL), VLBDA( NUMEL), ZOMEGA( NUMEL),
ELEMNT( NUMEL), SYSNOD( NUMSNP), R( NUMSNP), Z( NUMSNP),
Z1( NUMEL), R2( NUMEL), Z2( NUMEL), R3( NUMEL), RI( NUMEL),
A( NUMEL, 3), B( NUMEL, 3),
ERROR( NUMSNP), CM( NUMEL, 3, 3, 3), ALFA( 3, 3), PSTEP( NUMSNP),
BIGD( NUMSNP, LCON), BIGE( NUMSNP, LCON), ESTR( NUMSNP), VD( NUMEL),
YMAX( NUMSNP), IUPBP( NUMBP), PSI( 8, NUMSNP), DPSI( NUMSNP),
AREA( NUMEL), PV( NUMSNP), HOLD( NUMSNP)

```

----- DECLARATION STATEMENTS -----

```

NONLINEAR SUPERCRITICAL REACTOR
PROMPT FEEDBACK
FULLY REFLECTED SYSTEM
SPACE-DEPENDENT NEUTRONIC PROPERTIES

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```

C      INTEGER*4  SYSNDD,ELEMNT,ELNOD
C      DIMENSION TITLE(20)
C
C      DIMENSION PART(132,7),BIGA(132,7),BIGAB(132,7),BIGC(132,7),
1     BIGCC(132,7),MNOD(132,7),V(220),D(220),SGA(220),SGF(220),
2     ALPHA(220),PSIIV(132),PDPSI(132),ZLMBDA(220),VLMBDA(220),
3     ZOMEGA(220),RHO(220),SYSNOD(132),R(132),Z(132),ELEMNT(220),
4     ELNOD(220,3),ITYPE(220),RI(220),Z1(220),R2(220),Z2(220),
5     R3(220),Z3(220),A(220,3),PW(2244),BAT(400,4),ERRDR(132),
6     CM(220,3,3,3),PSTEP(132),BIGD(132,7),BIGE(132,7),ESTR(132),
7     VD(220,3),YMAX(132),IUPBP(22),PSI(8,132),AREA(220),PV(132),
8     B(220,3),ASTR(132,7),HOLD(132),WP(132,132)
C
C      READ(5,998) TITLE
C      READ(5,10) NUMEL,NUPBP,NUMSNP,NFULEL
998  FORMAT(20A4)
10  FORMAT(8I10)
C
C      LCON=7
C      CALL ERRSET(207,256,0,1,1,209)
C
C      *      CALL FEDMAN (NJMEL,NUPBP,NUMSNP,NFULEL,LCON, TITLE,
1     ALPHA,PSIIV,PDPSI,ZLMBDA,VLMBDA,ZOMEGA, SYSNOD,R,Z,ELEMNT,
2     ELNOD,ITYPE,RI,Z1,R2,Z2,R3,Z3,A,B,  ERROR,CM,ALFA,
3     PSTEP,BIGD,BIGE,ASTR,ESTR,VD,YMAX,IJPBP,AREA,PV,HOLD)
4     *
C      *      STOP
      END

```





```

*****
C CORRESPONDENCE TABLE BETWEEN SYSTEM AND ELEMENT NODAL POINTS
C *****
C *****
C *****
73 WRITE(6,73)
   FORMAT(/,2X,'CONNECTIVITY MATRIX',/, ' EL #',33X,'TYPE',/)
   DO 75 I=1,NUMEL
   READ(5,76) ELNOD(I,1),ELNOD(I,2),ELNOD(I,3),ITYPE(I)
   WRITE(6,77) ELEMENT(I),ELNOD(I,1),ELNOD(I,2),ELNOD(I,3),ITYPE(I)
75 CONTINUE
76 FORMAT (5I10)
77 FORMAT (2X,I3,4I10)

C
C
C READ AND WRITE THE HEXAGONAL CONNECTIVITY MATRIX.
LCON IS THE MAX NR OF NODAL POINT CONTRIBUTORS.
WRITE(6,1111)
FORMAT(/,2X,'HEXAGONAL CONNECTIVITY MATRIX',/, ' EL #')
1111 DO 776 J=1,NUMSNP
      READ(5,777) (MNOD(J,K),K=1,LCON)
776 CONTINUE
777 FORMAT (9(2X,I3))
      DO 775 J=1,NUMSNP
      WRITE(6,777) (MNOD(J,K),K=1,LCON)
775 CONTINUE

C
C
C
C
C WHEN REQUIRED THE INITIAL DERIVATIVES ARE READ IN HERE,
INSERTED AT THE END OF THE DATA DECK IN FORMAT 771
READ AND WRITE THE INITIAL DERIVATIVES OF THE NODE POINTS.
WRITE(6,2222)
FORMAT(/,2X,'INITIAL DERIVATIVES OF NODE POINTS')
2222 READ(5,771) (PDPSI(J),J=1,NUMSNP)
771 FORMAT (4E15,5)
      WRITE(6,771) (PDPSI(J),J=1,NUMSNP)

C
C
C *****
C CALCULATE GEOMETRY FOR EACH ELEMENT
C *****
C *****
110 WRITE(6,110)
   FORMAT (1X,/,1X,'GEOMETRY CALCULATIONS FOR EACH ELEMENT')

C
C -----
C LOAD SYSTEM NODE COORDINATES (R,Z) INTO
C ELEMENT NODE COORDINATES (R1,R2,R3,Z1,Z2,Z3)

```

```

C      WRITE(6,95)
95     FORMAT(1X,/,ELEMENT,5X,'R1',9X,'Z1',9X,'R2',9X,'Z2',9X,'R3',
C      19X,'Z3,/')
C      DO 100 I=1,NUMEL
C      J=ELNOD(I,1)
C      R1(I)=R(J)
C      Z1(I)=Z(J)
C      K=ELNOD(I,2)
C      R2(I)=R(K)
C      Z2(I)=Z(K)
C      L=ELNOD(I,3)
C      R3(I)=R(L)
C      Z3(I)=Z(L)
C      WRITE      (6,105) I,R1(I),Z1(I),R2(I),Z2(I),R3(I),Z3(I)
C      100 CONTINUE
105     FORMAT (3X,I3,3X, 7(F10.6,1X))
C      COMPUTE A1,A2,A3,B1,B2,B3,AREA FOR EACH ELEMENT
C-----
C      WRITE      (6,120)
120     FORMAT(1X,/,ELEMENT,5X,'A1',11X,'A2',11X,'A3',11X,'B1',11X,
C      1,B2',11X,'B3',10X,'AREA,/')
C      DO 140 I=1,NUMEL
C      A(I,1)=R3(I)-R2(I)
C      A(I,2)=R1(I)-R3(I)
C      A(I,3)=R2(I)-R1(I)
C      B(I,1)=Z2(I)-Z3(I)
C      B(I,2)=Z3(I)-Z1(I)
C      B(I,3)=Z1(I)-Z2(I)
C      AREA(I)=0.5*(A(I,2)*B(I,1)-A(I,1)*B(I,2))
C      WRITE(6,130) I,A(I,1),A(I,2),A(I,3),B(I,1),B(I,2),B(I,3),
C      1,AREA(I)
140     CONTINUE
C      130 FORMAT (3X,I3,3X, 7(F12.7,1X))
C
C

```







```

CM(L,1,3,3)=CC*(4,0*RI(L)+2,0*R2(L)+6,0*R3(L))
CM(L,1,3,2)=CC*(2,0*RI(L)+2,0*R2(L)+2,0*R3(L))
CM(L,1,3,1)=CC*(6,0*RI(L)+2,0*R2(L)+4,0*R3(L))
CM(L,1,2,3)=CC*(2,0*RI(L)+2,0*R2(L)+2,0*R3(L))
CM(L,1,2,2)=CC*(4,0*RI(L)+2,0*R2(L)+2,0*R3(L))
CM(L,1,2,1)=CC*(6,0*RI(L)+2,0*R2(L)+2,0*R3(L))
CM(L,1,1,3)=CC*(6,0*RI(L)+4,0*R2(L)+2,0*R3(L))
CM(L,1,1,2)=CC*(6,0*RI(L)+5,0*R2(L)+2,0*R3(L))
CM(L,1,1,1)=CC*(24,0*RI(L)+5,0*R2(L)+6,0*R3(L))

```

C 200 CONTINUE

C 151 CONTINUE

C

```

IF (MTH.EQ.5) CALL CRANKD (NUMEL,NUMSNP,NUPBP,LCON,H,TO,
ERRVAL,EPSVAL,IUPBP,PSI,IV,ITYPE,ELNOD,CM,MNOD,ZOMEGA,
BIGAB,BIGA,
3 BIGC,BIGCC,HOLD,TF)

```

C

STOP  
END

```

SUBROUTINE PEPSI (NUMEL, NUMSNP, R1, R2, R3, AREA, ELNOD, LCON, MNOD,
1 BIGA, BIGAB, VLMBDA)
C*****
C PEPSI CALCULATES THE BIGA MATRIX AND PART OF THE BIGAB
C*****
INTEGER*4 ELNOD

C DIMENSION R1(NJMEL), R2(NJMEL), R3(NJMEL), BIGA(NUMSNP, LCON),
1 BIGAB(NUMSNP, LCON), MNOD(NUMSNP, LCON), AMATRIX(3,3),
2 VLMBDA(NUMEL), AREA(NUMEL), ELNOD(NUMEL,3)
C
C PI=3.1415927
C
C CALCULATE THE 3X3 D(I,J) MATRIX FOR ELEMENT L
DO 200 L=1, NUMEL
COEFFA=(PI/30.0)*AREA(L)
AMATRIX(1,1)=COEFFA *(6.0*R1(L)+2.0*R2(L)+2.0*R3(L))
AMATRIX(1,2)=COEFFA *(2.0*R1(L)+2.0*R2(L)+R3(L))
AMATRIX(1,3)=COEFFA *(2.0*R1(L)+R2(L)+2.0*R3(L))
AMATRIX(2,1)=AMATRIX(1,2)
AMATRIX(2,2)=COEFFA *(R1(L)+6.0*R2(L)+2.0*R3(L))
AMATRIX(2,3)=COEFFA *(R1(L)+2.0*R2(L)+2.0*R3(L))
AMATRIX(3,1)=AMATRIX(1,3)
AMATRIX(3,2)=AMATRIX(2,3)
AMATRIX(3,3)=COEFFA *(2.0*R1(L)+2.0*R2(L)+6.0*R3(L))
C
C STORE ELEMENT MATRIX, AMATRIX, INTO SYSTEM MATRIX, BIGA
C-----
DO 20 K=1,3
KK=ELNOD(L,K)
DO 10 I=1,3
II=ELNOD(L,I)
DO 91 HX=1,LCJN
NOW =MX
MH =MNOD(KK,MX)
IF (MH .EQ. II) GO TO 92
91 CONTINUE
92 CONTINUE
BIGA(KK, NOW)=BIGA(KK, NOW)+AMATRIX(K, I)
BIGAB(KK, NOW)=BIGAB(KK, NOW)+VLMBDA(L)*AMATRIX(K, I)
CONTINUE
200 CONTINUE
RETURN
END

```



```

C      BMATRIX(3,1)=BMATRIX(1,3)
C      BMATRIX(3,2)=BMATRIX(2,3)
C      BMATRIX(3,3)=CJEFFB *(-R1(L)*(B(L,3)**2+B(L,1)*B(L,3)+A(L,3)**2+
1A(L,1)*A(L,3))-R2(L)*(B(L,3)**2+B(L,2)*B(L,3)+A(L,3)**2+A(L,2)*
2A(L,3))-R3(L)*(2.0*B(L,3)**2+2.0*A(L,3)**2)+2.0*AREA(L)*B(L,3))

```

-----  
C STORE ELEMENT MATRIX, AMATRIX, INTO SYSTEM MATRIX, BIGA

```

C      DO 20 K=1,3
C      KK=ELNOD(L,K)
C      DO 10 I=1,3
C      II=ELNOD(L,I)
C      DO 91 MX=1,LCON
C      NOW =MX
C      MM = MNOD(KK,MX)
C      IF (MM .EQ. II) GO TO 92
C      91 CONTINUE
C      92 CONTINUE
C      BIGAB(KK,NOW)=3IGAB(KK,NOW)+BMATRIX(K,I)
C      10 CONTINUE
C      20 CONTINUE
C      200 CONTINUE
C      RETURN
C      END

```

```

SUBROUTINE CRANKO (NJMEL, NJMSNP, NUPBP, LCON, H, TO, ERRVAL,
1 EPSVAL, IUPBP, PSIIV, ITYPE, ELNOD, CM, MNOD, ZOMEGA, BIGAB, BIGA,
2 PV, PSTEP, BIGD, BIGE, ASTR, ESTR,
3 BIGC, BIGCC, HOLD, TF)

```

CC C

```

INTEGER*4 ELNJD

```

```

DIMENSION IUPBP(NUPBP), PSIIV(NUMSNP), ITYPE(NUMEL),
1 ELNOD(NUMEL, 3), CM(NUMEL, 3, 3), MNOD(NUMSNP, LCON),
2 ZOMEGA(NUMEL), BIGAB(NUMSNP, LCON), BIGA(NUMSNP, LCON),
3 PSTEP(NUMSNP), ALFA(3, 3), PV(NUMSNP)
DIMENSION BIGD(NUMSNP, LCON), BIGE(NUMSNP, LCON),
1 ASTR(NUMSNP, LCON), ESTR(NUMSNP), BIGC(NUMSNP, LCON),
2 BIGCC(NUMSNP, LCON), HOLD(NUMSNP)

```

CCCCCCC

```

DLT IS THE MAX TIME INTERVAL ATTEMPTED FOR ONE STEP.
EPSN IS THE CONVERGENCE CRITERIA FOR SOLN AT TIME T=T+DT

```

```

INITIALIZATION

```

```

DLT=H
PTIME=H
T=TO
LRGE = 1
EPSN=ERRVAL
STAR=EPSVAL
NIT=0
NUMEQ=NUMSNP-NJPBP
MIN=NUMSNP/4

```

CC

```

NUF=NUMSNP/5
XNP=NUMSNP
XNUF=XNP/5
IF ((XNUF-FLOAT(NUF)) .LT. 1.) NUF=NUF+1

```

C

```

DO 05 I=1, NUMSNP
PV(I)=.0
CONTINUE
05

```

CCC

```

BUILD THE BIGCC MATRIX

```

```

351 CONTINUE

```

```

C
DO 07 I=1, NUPBP
J=IUPBP(I)
PSIIV(J)=0.
07 CONTINUE
DO 09 I=1, NUMSNP
PSTEP(I)=PSIIV(I)
09 CONTINUE
C
DO 210 L=1, NUMEL
LL=ITYPE(L)
IF(LL.EQ.0) GO TO 210
N1=ELNOD(L,1)
N2=ELNOD(L,2)
N3=ELNOD(L,3)
DO 88 J=1,3
DO 90 N=1,3
ALFA(N,J)=PSIIV(N1)*CM(L,N,J,1)+PSIIV(N2)*CM(L,N,J,2)
+PSIIV(N3)*CM(L,N,J,3)
1 CONTINUE
90 CONTINUE
88 CONTINUE
DO 150 K=1,3
KK=ELNOD(L,K)
DO 180 I=1,3
II=ELNOD(L,I)
DO 91 MX=1,LCOV
NOW =MX
MM = MNOD(KK,MX)
IF (MM .EQ. II) GO TO 92
91 CONTINUE
92 CONTINUE
180 BIGC(KK,NOW)=BIGC(KK,NOW)+ALFA(K,I)*ZOMEGA(L)
150 CONTINUE
210 CONTINUE
C
INSERTION OF BOUNDARY POINTS
DO 185 I=1, NUPBP
II=IUPBP(I)
DO 191 MX=1,LCOV
BIGC(II,MX)=0
191 CONTINUE
185 CONTINUE
C
DO 310 I=1, NUMSNP
DO 310 J=1,LCOV
BIGCC(I,J)=BIGAB(I,J)-BIGC(I,J)

```

```

310 CONTINUE
CCC
C      BEGIN STEPPING OUT IN TIME.
C      SIP = 4.
C      ITRY=0
C      COMPUTE PSI BASED ON A TRIAL DT.
C      DO 55 JQ=1,12
C          FIRST COMPJTE (2/DT)*A=ASTR, ASTR-C=D, ASTR+D=E
C          DS=2./DLT
C          DO 15 KA=1,NUMSNP
C          DO 13 KB=1,LCON
C          ASTR(KA,KB)=DS*BIGA(KA,KB)
C          BIGD(KA,KB)=ASTR(KA,KB)-BIGCC(KA,KB)
C          BIGE(KA,KB)=ASTR(KA,KB)+BIGCC(KA,KB)
C          CONTINUE
13      CONTINUE
15      CONTINUE
C
C      DO 19 KA=1,NUMSNP
C      ESTR(KA)=0.
C      DO 17 KB=1,LCON
C      NAME = MNOD(KA,KB)
C      ESTR(KA)=ESTR(KA)+BIGE(KA,KB)*PSIIV(NAME)
17      CONTINUE
19      CONTINUE
C
C      THE SYSTEM IS NOW IS THE FAMILIAR FJRM D(I,J)*PV(J)=ESTR(J)
C      AND MAY BE SOLVED USING ANY TECHNIQUE FOR LINEAR SIMULTANEOUS
C      EQNS.  HERE GAUSS-SEIDEL ITERATION IS USED.
C      ACCOMPLISH THE FIRST ITERATION BASED ON THE PV VALUE AT THE
C      LAST SUCCESSFUL TIME POINT.  SECOND AND SUCCESSIVE ITERATIONS
C      ARE BASED ON UPDATED VALUES OF PV.  A TEST FOR CONVERGENCE IS
C      MADE, AND FAILURE RESULTS IN ANOTHER ATTEMPT WITH A SMALLER DT
C      CREATION OF ESTR(I)=BIGE(I,J)*PSIIV(I)
C      PERFORMED ONCE ON EACH ATTEMPT TO FIND A PROPER TIME INTERVAL.
C      DO 45 MTRY=1,NUMEQ
C      KT=0
C      DO 25 K=1,NUMEQ
C      HOLD(K)=0.
C      PV(K)=0.
C      DO 20 L=2,LCON
C      NAME = MNOD(K,L)

```

```

HOLD(K) = HOLD(K) + BIGD(K,L)*PSTEP(NAME)
CONTINUE
20 PV(K) = (ESTR(K)-HOLD(K))/BIGD(K,1)
PDIF = (PV(K)-PSTEP(K))/PV(K)
CONTINUE
24 IF (ABS(PDIF) .LT. EPSN) KT = KT+1
PSTEP(K) = PV(K)
CONTINUE
25 IF (KT .GE. NJMEQ) GO TO 65
45 CONTINUE
C
C 291 FORMAT (//10X,'NO SOLN FOUND IN INTERVAL ',G10.4,' PLUS ',
L G10.4)
WRITE(6,291) T,DLT
C
C PENALTY --- MAKE THE INTERVAL EVEN SMALLER.
DLT = DLT/STP
STP = STP+2.
ITRY = ITRY+1
CONTINUE
55
C
C WRITE(6,292)
FORMAT(//10X,'A SOLN IS NOT POSSIBLE. THE PROGRAM SURRENDERS.')
```

```

I3=I+3*NUF
I4=I+4*NUF
WRITE(6,356) (I,PV(I),I1,PV(I1),I2,PV(I2),I3,PV(I3),I4,PV(I4))
356 CONTINUE(5(4X)I3,3X,IPE12.4))
357 WRITE(6,357)
700 FORMAT(/10X)
C NIT= I + NIT
IF (VIT .GT. 400) GO TO 377
377 CONTINUE
C
DO 27 MN=1,NUMSNP
PS IIV(MN)=PV(MN)
DO 28 I=1,LCON
BIGC(MN,I)=.0
28 CONTINUE
27 CONTINUE
IF (I .LT. TF) GO TO 351
778 FORMAT (2X,8(E14.5))
399 FORMAT (10X,I4)
WRITE (6,399) NIT
WRITE (3) BATCH
WRITE (4) BATCH
STOP
END

```



```

DO 70 I=1,NHH
ELNOD(K,1)=NN
ELNOD(K+1,1)=NN
ELNOD(K,2)=NN+NV-1
ELNOD(K+1,2)=LL+NV
ELNOD(K,3)=ELNOD(K+1,2)
ELNOD(K+1,3)=44+NV
K=K+2
NN=NN+NV-1
LL=LL+1
MM=MM+1
NEL=K-1
70 CONTINUE
WRITE(5,750)
750 FORMAT('1')
WRITE(6,800)NEL
800 FORMAT(5X,'NUMBER OF ELEMENTS =',I4)
WRITE(6,950)
950 FORMAT(/////10X,'CONNECTIVITY MATRIX')
DO 50 I=1,NEL
WRITE(6,900)I,(ELNOD(I,J),J=1,3)
WRITE(7,901) I,(ELNOD(I,J),J=1,3)
50 CONTINUE
900 FORMAT(/4I10)
901 FORMAT(4I10)
IF(NV.EQ.12)STCP
GO TO 1
END

```

C-----NUCLEAR PROPERTY DATA GENERATOR-----

```

C
C
      INTEGER * 4 NEL, ELEM, ITYPE
      DIMENSION V(220), ALPHA(220), SGF(220), SGA(220), D(220),
1      ITYPE(220), ELEM(220)
C
1  READ(5,100) NEL
      IDENTIFY CORE AND REFLECTOR ELEMENTS
C
100  READ(5,200) (ITYPE(I), I=1, NEL)
      FORMAT(2I10)
200  FORMAT(16I5)
      WRITE(6,400)
400  FORMAT('1', 2X, 'ELEM', 6X, 'D', 8X, 'SGA', 9X, 'SGF', 8X,
1      'ALPHA', 8X, 'V')
300  FORMAT(2X, I4, 5(1PE12.4))
C
C
      ASSIGN PROPERTY VALUES TO ELEMENTS
C
      DO 30 I=1, NEL
      ELEM(I)=I
      V(I)=4.8E+07
      ALPHA(I)=1.0E-05
      IF(ITYPE(I).EQ.0)GO TO 10
      D(I)=1.2
      SGA(I)=.008
      SGF(I)=0
      GO TO 20
10   D(I)=0.913
      SGA(I)=0.014
      SGF(I)=0.008
20   WRITE(6,300)(I, D(I), SGA(I), SGF(I), ALPHA(I), V(I))
30   CONTINUE
      WRITE(7,500)(V(I), I=1, NEL)
      WRITE(7,500)(D(I), I=1, NEL)
      WRITE(7,500)(SGA(I), I=1, NEL)
      WRITE(7,500)(SGF(I), I=1, NEL)
      WRITE(7,600)(ALPHA(I), I=1, NEL)
500  FORMAT(8G10.5)
600  FORMAT(5F15.8)
      IF(NEL.EQ.64)GO TO 1
      IF(NEL.EQ.112)GO TO 1
      IF(NEL.EQ.220)STOP
      END

```



```

MNOD(KK,4)=LI+6
MNOD(KK+1,5)=0
MNOD(KK+1,7)=0
MNOD(IJ,2)=LI+6
MNOD(IJ,4)=LI+7
MNOD(IJ+1,5)=0
MNOD(IJ+1,7)=0
MNOD(IK,2)=LI+7
MNOD(IK,4)=LI+8
MNOD(IK+1,5)=0
MNOD(IK+1,7)=0
IF(NVH.EQ.72)GJ TO 15
MNOD(IM,2)=LI+8
MNOD(IM,4)=LI+9
MNOD(IM+1,5)=0
MNOD(IM+1,7)=0
MNOD(LK,2)=LI+10
MNOD(LK,4)=LI+11
MNOD(LK+1,5)=0
MNOD(LK+1,7)=0
MNOD(IN,2)=LI+9
MNOD(IN,4)=LI+10
MNOD(IN+1,5)=0
MNOD(IN+1,7)=0
15 DO 70 I=LJ,NVH
MNOD(I,6)=0
MNOD(I,7)=0
70 CONTINUE
DO 80 I=LL,LI
MNOD(I,3)=0
80 CONTINUE
DO 85 I=LL,LJ
MNOD(I,4)=0
85 CONTINUE
DO 95 I=LJ,NVH
MNOD(I,3)=N
IF(I.EQ.NVH)GJ TO 95
MNOD(I+1,4)=MNOD(I,3)
N=N+NV-1
95 CONTINUE
MNOD(LI,2)=NVH
WRITE(6,300)
300 FORMAT(5X,'HEXAGONAL CONNECTIVITY',/' NODE')
DO 90 I=1,NVH
WRITE(6,100)(MNOD(I,J),J=1,LCON)
WRITE(7,101)(MNOD(I,J),J=1,LCON)
90 CONTINUE
100 FORMAT(9(2X,I3))
101 FORMAT(9(2X,I3))
IF(NV.EQ.12)STOP
GO TO 1
END

```

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