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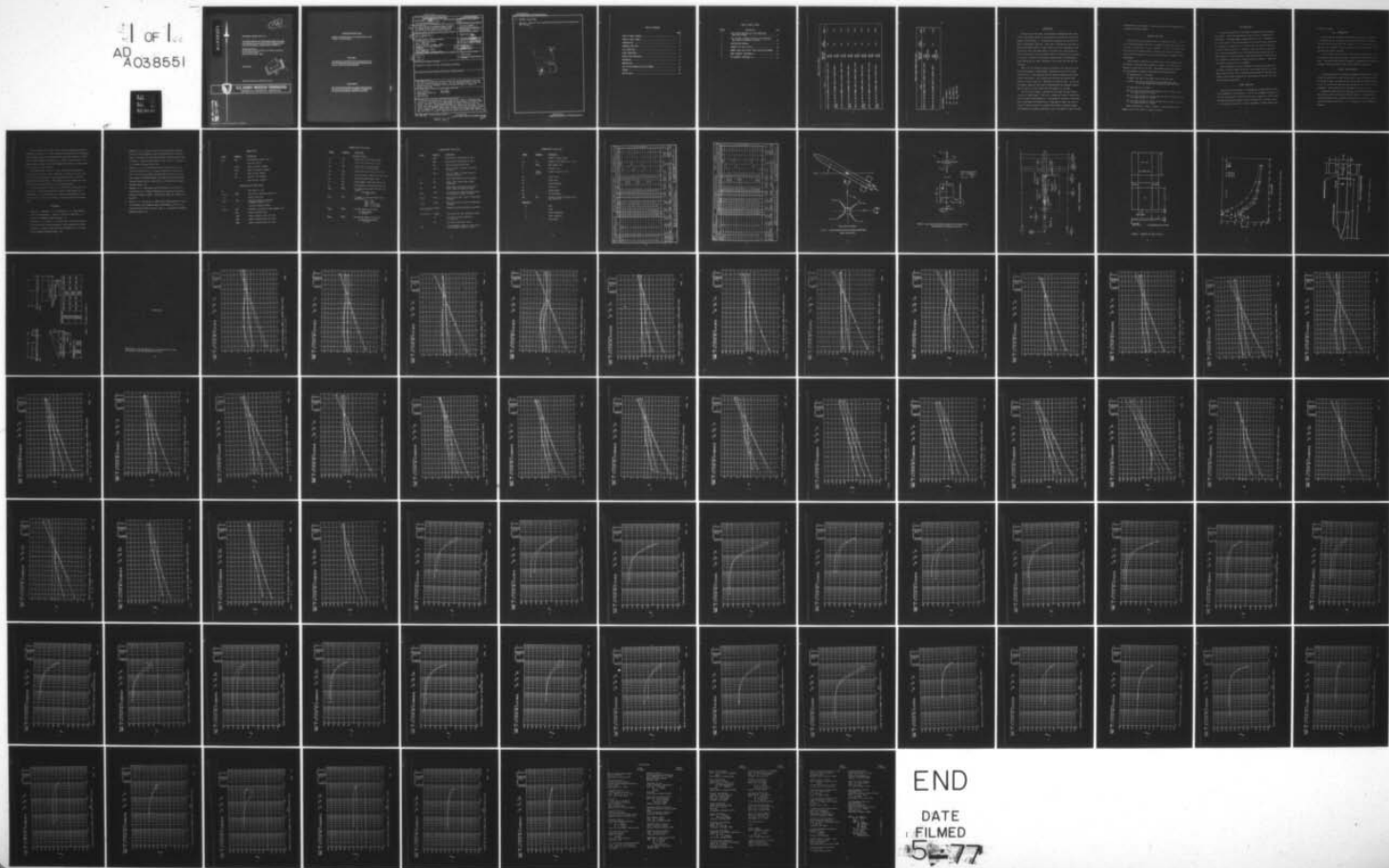
ARMY MISSILE RESEARCH DEVELOPMENT AND ENGINEERING LAB--ETC F/G 16/4
AN INVESTIGATION OF STREAMWISE BODY-FIN GAPS AS A MEANS OF ALLE--ETC(U)
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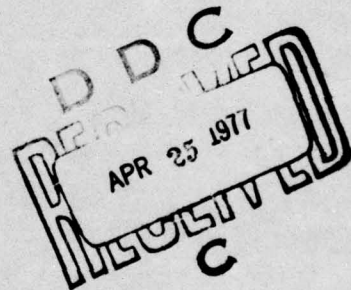
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TECHNICAL REPORT RD-77-13

**AN INVESTIGATION OF STREAMWISE BODY-FIN GAPS
AS A MEANS OF ALLEVIATING THE ADVERSE PLUME
EFFECTS ON MISSILE LONGITUDINAL STABILITY**

Aeroballistics Directorate
US Army Missile Research, Development and Engineering Laboratory
US Army Missile Command
Redstone Arsenal, Alabama 35809

January 1977



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U.S. ARMY MISSILE COMMAND

Redstone Arsenal, Alabama

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| 19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Thrust Effects Jet Plume Longitudinal Stability Gap Effects Plume Effects Base Pressure | | |
| 20. ABSTRACT (Continue on reverse side if necessary and identify by block number) Transonic wind tunnel tests were conducted on a body of revolution with various fin configurations to investigate jet plume effects on missile longitudinal stability. The sting-mounted model utilized a series of cold air normal jets located downstream of the base to simulate the jet plume. Fins of various planform geometry were tested in three longitudinal positions. Present report compares one fin planform with and without a 5 percent body diameter gap. Gap results in a small reduction of adverse plume effects for fins in aft. | | |

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20. ABSTRACT (Concluded)

position. Data for this investigation were obtained from Calspan test CAL T17-123.

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PLOT SCHEDULE:

- (A) C_N VERSUS α
- (B) C_{NF2} VERSUS α
- (C) $C_{N\alpha}$ VERSUS CRT
- (D) $C_{N\alpha F2}$ VERSUS CRT

INTRODUCTION

During the past few years, the Army Missile Command has been interested in the adverse effects of the propulsive jet plume on missile aerodynamics (see reference 1). Of particular importance are the effects on missile longitudinal stability. Wind tunnel investigations have been conducted to determine effect of thrust level on missile longitudinal stability and to assess the effectiveness of locating fins some distance forward of the base to alleviate adverse effects on stability (see references 2, 3, and 4). Results of these tests indicated that fin effectiveness (when affected by the plume) improved as the distance from the base was increased.

One of the fin planforms tested in reference 4 (F2) was also tested with a five percent of body diameter longitudinal gap at the fin-body juncture (F4). It was expected that the gap would reduce the two-dimensional flow effects at this juncture and alleviate the adverse plume effects on missile stability. Reference 4 presents results. In the present report, data for the F2 and F4 configurations are compared directly. Data at angles of attack higher than four degrees are excluded.

Past tests (for example, reference 6) have shown that gaps cause a loss in fin effectiveness. This loss is caused by a partial cancellation of the fin-body interference lift. The purpose of the test of reference 6 was to investigate the feasibility of using gaps to reduce the large increase in stability margin of rockets which occurs at transonic speeds. The reduction in transonic stability is one of the means of reducing rocket

wind sensitivity during boost. Some typical results from reference 6 are compared with present results.

APPARATUS AND TESTS

These tests were described in detail in references 4 and 5. Briefly the sting-mounted model, which had a 6-component strain gage balance, was 52 inches in length and 5 inches in diameter. It had a 3-caliber tangent ogive nose and a cylindrical body. The model was tested in combination with cruciform rectangular fins; each fin panel had a 3-component strain gage balance.

Plume simulator consisted of 24 normal sonic jets arranged circumferentially in two rows with a common air chamber. The simulator was located 2.5 inches aft of the base. The combined exit area of the 24 jets represented six percent of the model base (reference area).

Fin nomenclature is as follows:

F2 (BF2)--fins with 3-inch chord and 2.5-inch semi-span.

F4 (BF4)--fins with 3-inch chord, 2.5-inch semi-span, and a .25-inch gap between the fuselage and the root chord.

Fin positions are as follows:

Aft--fin trailing edge flush with model base (M.S. 52.000, 0 cal.)
--fin hinge line at M.S. 49.750

Mid--fin trailing edge 3.75 inches from base (M.S. 48.250, 0.75 cal.)
--fin hinge line at M.S. 46.000

Fwd--fin trailing edge 7.5 inches from base (M.S. 44.500, 1.5 cal.)
--fin hinge line at M.S. 42.250

Model installation is shown in figure 3. A description of the geometry for fins F2 and F4 is shown in figure 4.

TEST CONDITIONS

The test was conducted in the Calspan Corporation 8-foot Transonic Wind Tunnel. A detailed description of this facility is presented in reference 5. Test Mach numbers ranged from 0.4 to 1.25. The amount of air in the tunnel circuit was equivalent to a wind-off static pressure of 0.5 atmosphere. Plume effect, or value of CR_T , was varied by changing simulator chamber pressure P_C . Normally, angle of attack is varied during a run and configuration, CR_T , and tunnel operating conditions are kept constant. Air was supplied to the plume simulator at flow rates up to 15 pounds per second and at chamber pressures to 600 psi. Angles of attack were varied from -4 to 12 degrees.

Model aerodynamic data were resolved in the body axis system with the origin located at model station 26.5 on the model centerline. Aerodynamic data from the tail fin balances were resolved in individual axis systems; origins were located at the intersection of the tail fin lines and the balance centerlines.

PLUME SIMULATION

Base pressure ratio P_B/P_∞ is a good measure of plume effects on missile aerodynamics (see reference 1). One of the parameters convenient to use with base pressure is thrust coefficient C_T , where C_T is axial thrust non-dimensionalized by dynamic pressure and body cross sectional area S . For the normal jet simulator, a similar parameter is radial thrust coef-

ficient C_{RT} , where

$$C_{RT} = \frac{\text{radial thrust}}{qS}$$

For an axial jet, base pressure appears to be primarily influenced by the portion of the jet plume in the vicinity where the jet boundary interacts with the freestream flow. Where C_T can be considered to represent the axial component of the effective jet, it can be assumed that C_{RT} represents the normal component.

For the plume size of interest in the current investigation, a value of C_T several times the value of C_{RT} is required for matching base pressures. The exact C_T/C_{RT} ratio will depend on a comparison of flight base pressures with base pressure value for the normal jet simulator.

RESULTS AND DISCUSSION

Plotted data consist of comparisons between BF2 and BF4 total normal forces, C_N , and fin no. 2 panel normal forces, C_{NF2} , for various levels of simulated thrust. Matching runs for BF2 and BF4 were made from Mach 0.7 to 1.2 for zero roll position and from Mach 0.9 to 1.2 for a roll of 45 degrees. Panel normal force is presented for fin no. 2 only.

Inspection of the plots of C_N and C_{NF2} versus angle of attack generally shows a small but definite effect of the gap. A better overall view of gap effects may be obtained from plots of $C_{N\alpha}$ and $C_{N\alpha F2}$ versus C_{RT} with no plume effects ($C_{RT} \leq .1$). As expected, a small decrement is apparent.

Values of C_N for fins in the aft position are compared with data of reference 6 in figure 5. In this comparison, body alone C_N was subtracted from the present data since afterbody forces only were measured in reference 6. Model geometry of reference 6 is shown in figures 6 and 7. Although geometry and gap size differ between the two tests, the gap effects for small gaps is about the same.

With plume effects, there is a small but definite improvement in C_{N_α} and $C_{N_{\alpha F2}}$ with fins in the aft position. With fins in the mid position, most of the decrement in C_{N_α} without plume effects is canceled when there are plume effects. There appears to be no change with plume effects due to the gap with fins in the forward position.

For fins in the aft position, it is possible for a larger gap size to result in a significant reduction in adverse plume effects on fin effectiveness. In cases where missile design considerations force the use of aft fins, it may be profitable to investigate gap and fin geometries.

REFERENCES

1. Deep, R. A., Henderson, J. H., and Brazzel, C. E., Thrust Effects on Missile Aerodynamics. Report No. RD-TR-71-9, May 1971, U. S. Army Missile Command, Redstone Arsenal, Ala.
2. Henderson, J. H., Results of a Transonic Wind Tunnel Investigation of Thrust Effects on the Aerodynamics of the Longbow Missile Configuration. Technical Report RD-73-18, November 1973, U. S. Army Missile Command, Redstone Arsenal, Ala.

3. Henderson, J. H., Transonic Wind Tunnel Investigation of Thrust Effects on the Longitudinal Stability Characteristics of Several Body-Fin Configurations (Sting-Mounted Model with Normal-Jet Plume Simulator). Technical Report RD-75-14, Dec. 1974, U. S. Army Missile Command, Redstone Arsenal, Ala.
4. Henderson, J. H., An Investigation of Jet Plume Effects on the Stability Characteristics of a Body of Revolution in Conjunction with Fins of Various Geometry and Longitudinal Positions at Transonic Speeds (Sting-Mounted Model with Normal Jet Plume Simulator). Technical Report RD-75-37, June 1975, U. S. Army Missile Command, Redstone Arsenal, Ala.
5. deKuyper, R. E., Investigation of Plume Effects on the Stability Characteristics of a Body of Revolution with Various Fin Configurations at Transonic Speeds. Calspan Corp. Report No. AA-4017-W-11, May 1974.
6. Dahlke, C. W., and Pettis, W., Normal Force Effectiveness of Several Fin Planforms with Streamwise Gaps at Mach Numbers of 0.8 to 5.0. Technical Report RD-TR-70-8, April 1970, U. S. Army Missile Command, Redstone Arsenal, Ala.

NOMENCLATURE

| <u>SYMBOL</u> | <u>MNEMONIC</u> | <u>DEFINITION</u> |
|---------------|-----------------|-------------------------------------|
| RN/L | RN/L | unit Reynolds number; per ft |
| V | | velocity; ft/sec |
| α | ALPHA | angle of attack, degrees |
| β | BETA | angle of sideslip, degrees |
| ψ | PSI | angle of yaw, degrees |
| ϕ | PHI | angle of roll, degrees |
| ρ | | mass density; slugs/ft ³ |

Reference & C.G. Definitions

| | | |
|--------------------|------|--|
| A_b | | base area; m ² , in ² |
| b, b_{ref} | BREF | wing span or reference span; m, in |
| c.g. | | center of gravity |
| l_{REF}, \bar{c} | LREF | reference length or wing mean aerodynamic chord; m, in |
| P_c | | simulator chamber pressure |
| S, S_{ref} | SREF | reference area based on body diameter, in ² |
| | MRP | moment reference point |
| | XMRP | moment reference point on X axis |
| | YMRP | moment reference point on Y axis |
| | ZMRP | moment reference point on Z axis |

NOMENCLATURE (Continued)

| <u>SYMBOL</u> | <u>MNEMONIC</u> | <u>DEFINITION</u> |
|-----------------------------|-----------------|---|
| | | <u>Body Axis System</u> |
| C _N | CN | normal force coefficient; F_N/qS |
| C _A | CA | axial force coefficient; F_A/qS |
| C _{A_b} | CAB | base axial force coefficient; $(-1) \cdot [(p_b - p_\infty)/q] (A_b/S)$ |
| C _{A_u} | CAU | uncorrected axial force coefficient |
| C _m | CLM | pitching moment coefficient; $M_y/qS \ell_{ref}$ |
| C _y | CY | side force coefficient; F_y/qS |
| C _n | CYN | yawing moment coefficient; $M_z/qS b_{ref}$ |
| C _l | CBL | rolling moment coefficient; $M_x/qS b_{ref}$ |
| C _{N_{TX}} | CNTX | fin normal force coefficient for fin number X $\frac{\text{Fin Normal Force}}{qS}$ |
| C _{N_{FX}} | CNFX | fin number X corrected normal force coefficient, CNF1 = CNT1 CNF2 = CNT2 CNF3 = (-1.0) CNT3 CNF4 = (-1.0) CNT4 |
| C _{H_{MX}} | CLMHX | fin hinge moment coefficient, $\frac{\text{Fin Hinge Moment}}{qS \ell_{ref}}$ |
| C _{B_{MX}} | CLMRX | root bending moment coefficient, $\frac{\text{Fin Root Bending Moment}}{qS \ell_{ref}}$ |

NOMENCLATURE (Continued)

| <u>SYMBOL</u> | <u>MNEMONIC</u> | <u>DEFINITION</u> |
|---|-----------------|---|
| | CPB1 | base pressure coefficient at tap 1 |
| | CPB2 | base pressure coefficient at tap 2 |
| | CPC | cavity pressure coefficient |
| | CPXT1 | tail fin number 1 center of pressure location |
| | CPYT1 | tail fin number 1 lateral center of pressure location |
| C_T | CT | thrust coefficient, axial thrust/qS |
| C_{RT} | CRT | radial thrust coefficient, radial thrust/qS |
| $C_{N\alpha}$ | CNA | normal force coefficient derivative with respect to angle of attack |
| $C_{N\alpha FX}$ | CNAFX | fin normal force coefficient derivative with respect to angle of attack |
| P_{b1}/P_∞ | PB1/P | base pressure tap 1 ratio to freestream pressure |
| P_{b2}/P_∞ | PB2/P | base pressure tap 2 ratio to freestream pressure |
| $\frac{(P_{b1}/P_\infty + P_{b2}/P_\infty)}{2}$ | PB-AVG | average of two base pressure tap ratios |
| Fin Position | FINPOS | fin position on body; defined on page 6 |
| | PL | air supply pressure upstream of orifice, psi |
| | WA | air supply flow rate, lb/sec |
| $C_{m\alpha}$ | | pitching moment coefficient derivative with respect to angle of attack |

NOMENCLATURE (Concluded)

| <u>SYMBOL</u> | <u>MNEMONIC</u> | <u>DEFINITION</u> |
|---------------|------------------|---|
| a | | speed of sound; ft/sec |
| C_p | CP | pressure coefficient; $(p_1 - p_\infty)/q$ |
| M | MACH | Mach number; V/a |
| p | | pressure; psf |
| q | Q(NSM) Q(PSF) | dynamic pressure; $1/2\rho V^2$ |
| F_N | | normal force |
| F_A | | axial force |
| M_y | | pitching moment |
| F_y | | side force |
| M_z | | yawing moment |
| M_x | | rolling moment |
| GAP | GAP | distance between fuselage and fin root chord, in. |

SUBSCRIPTS

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|----------|-------------------|
| b | base |
| l | local |
| s | static conditions |
| t | total conditions |
| ∞ | free stream |

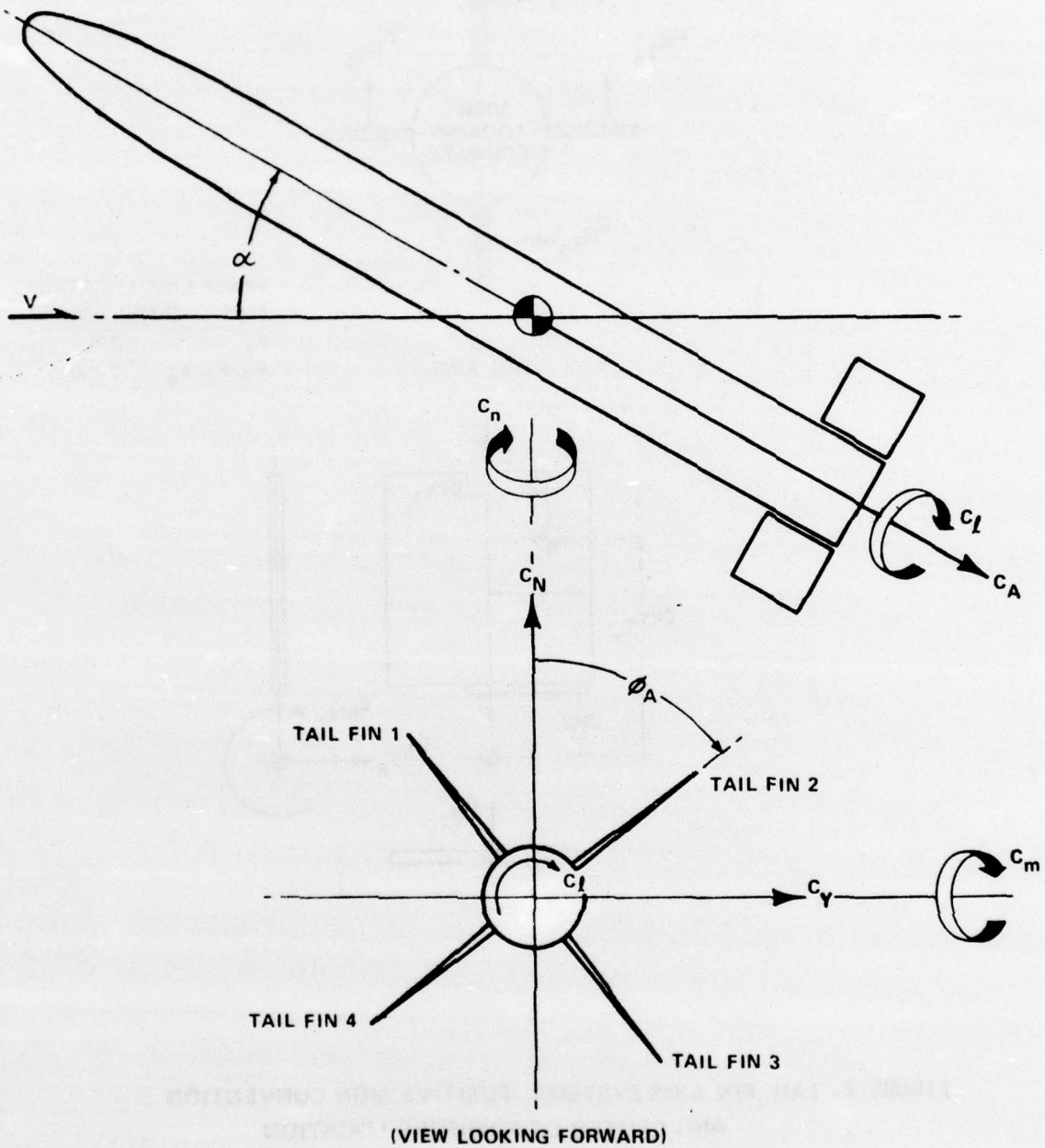
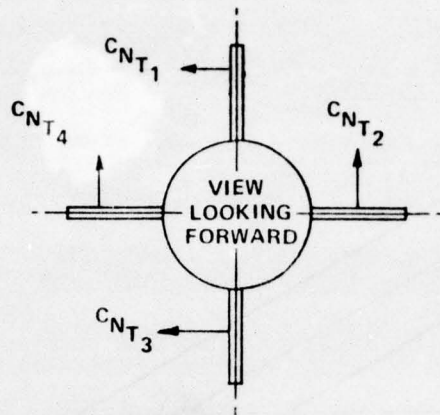


FIGURE 1. AXIS SYSTEMS AND POSITIVE SIGN CONVENTION,
BODY AXIS SYSTEM



TAIL FINS

| HINGE LINE LOCATION | |
|--|---------------|
| FINS | DIMENSION "a" |
| F ₁ | 2.750 |
| F ₂ , F ₃ , F ₄ | 0.750 |

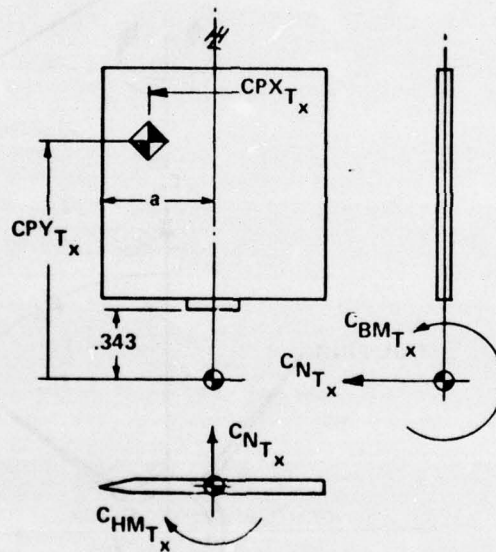


FIGURE 2. TAIL FIN AXIS SYSTEMS , POSITIVE SIGN CONVENTION AND CENTER OF PRESSURE LOCATION

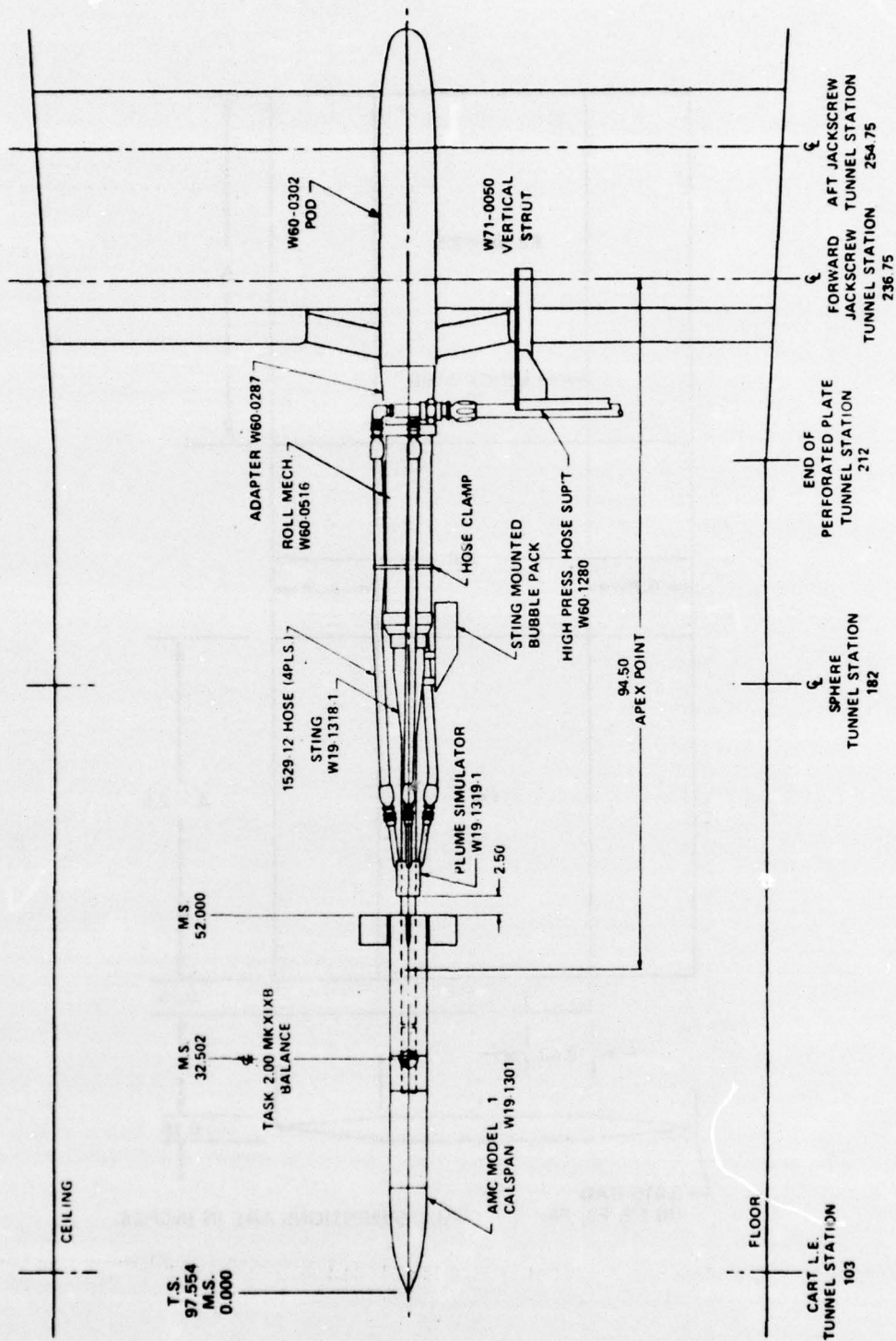


FIGURE 3. INSTALLATION DRAWING

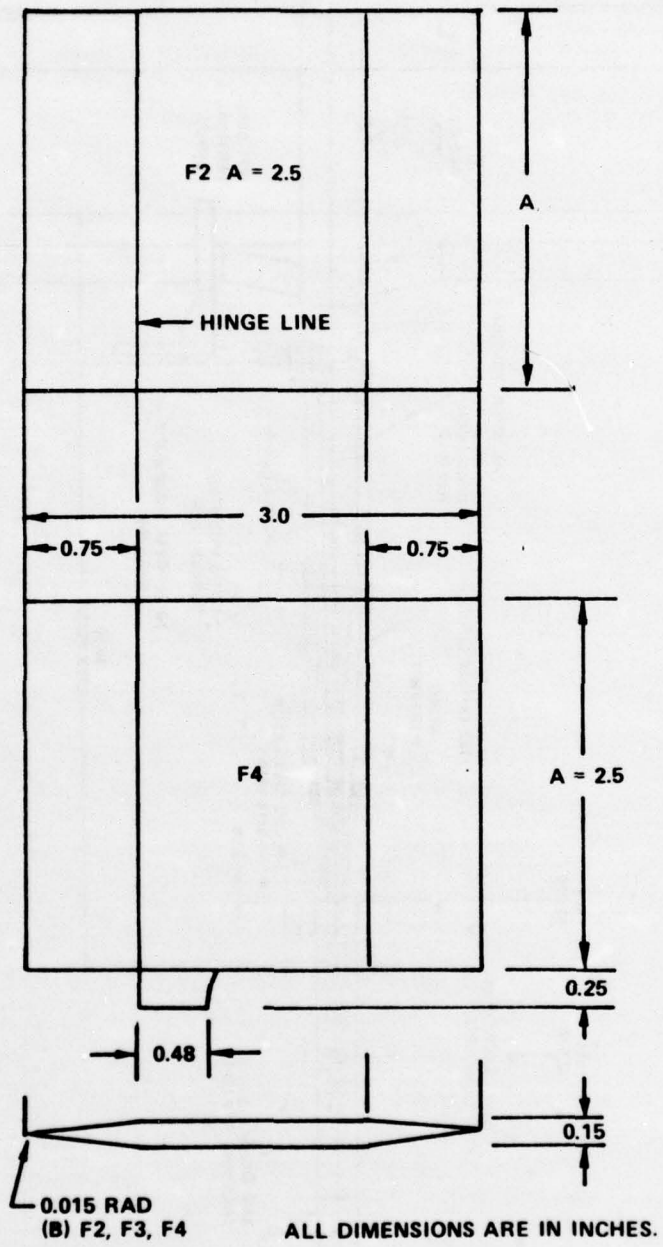


FIGURE 4. GEOMETRY OF FINS F2 AND F4

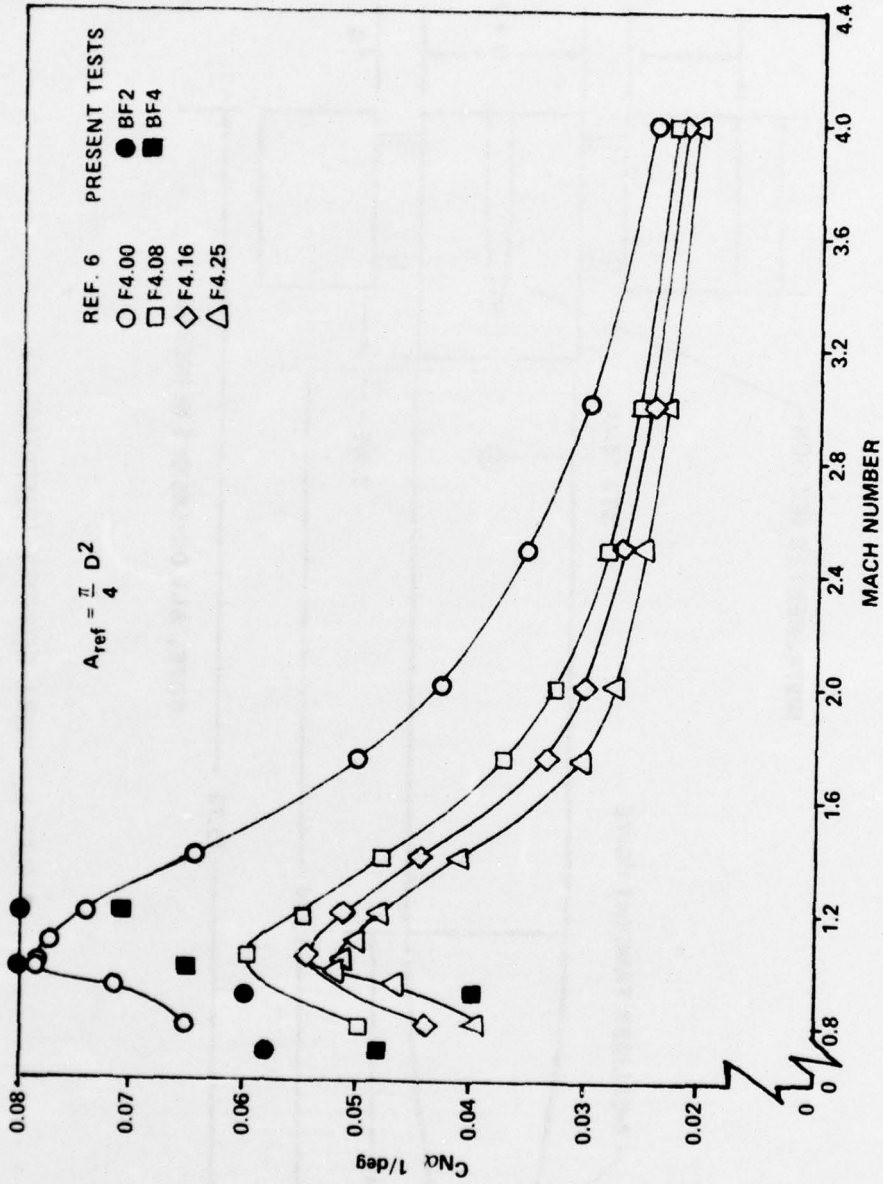


FIGURE 5. NORMAL FORCE COEFFICIENT SLOPE VERSUS MACH NUMBER

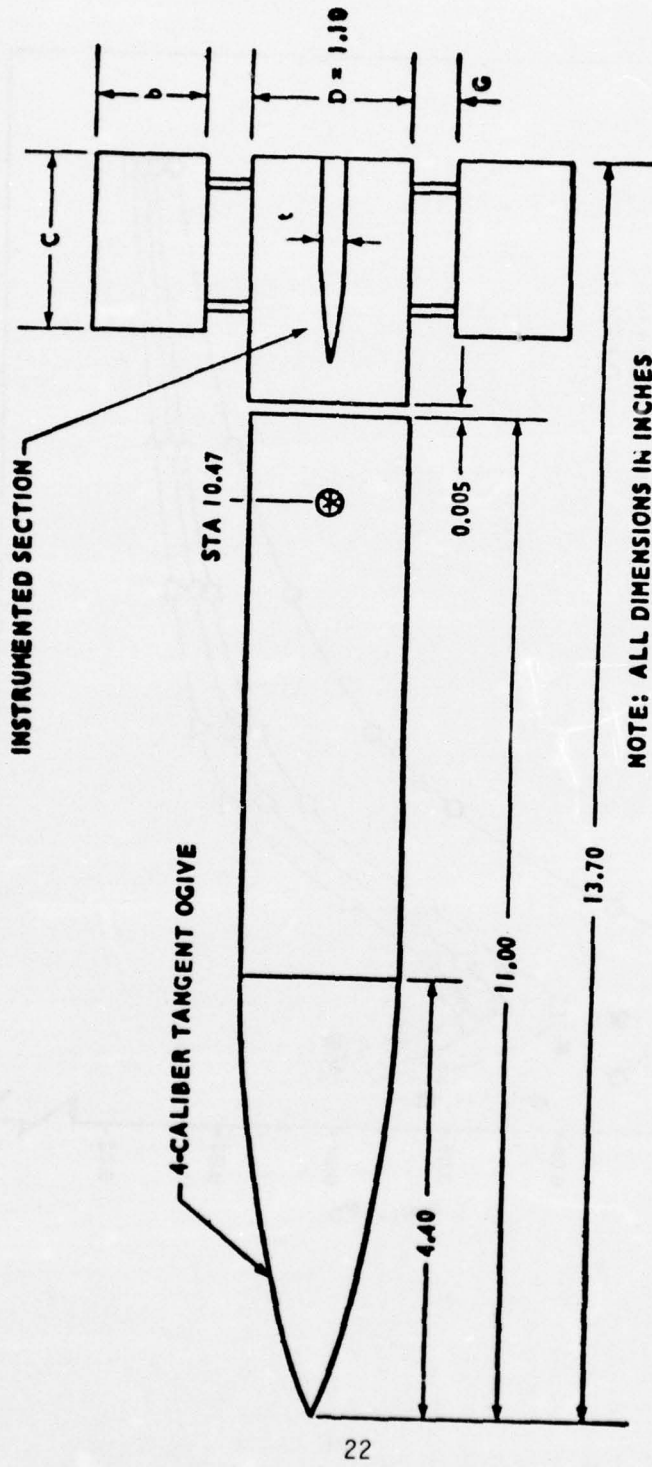


FIGURE 6. MODEL GEOMETRY (REFERENCE 6)

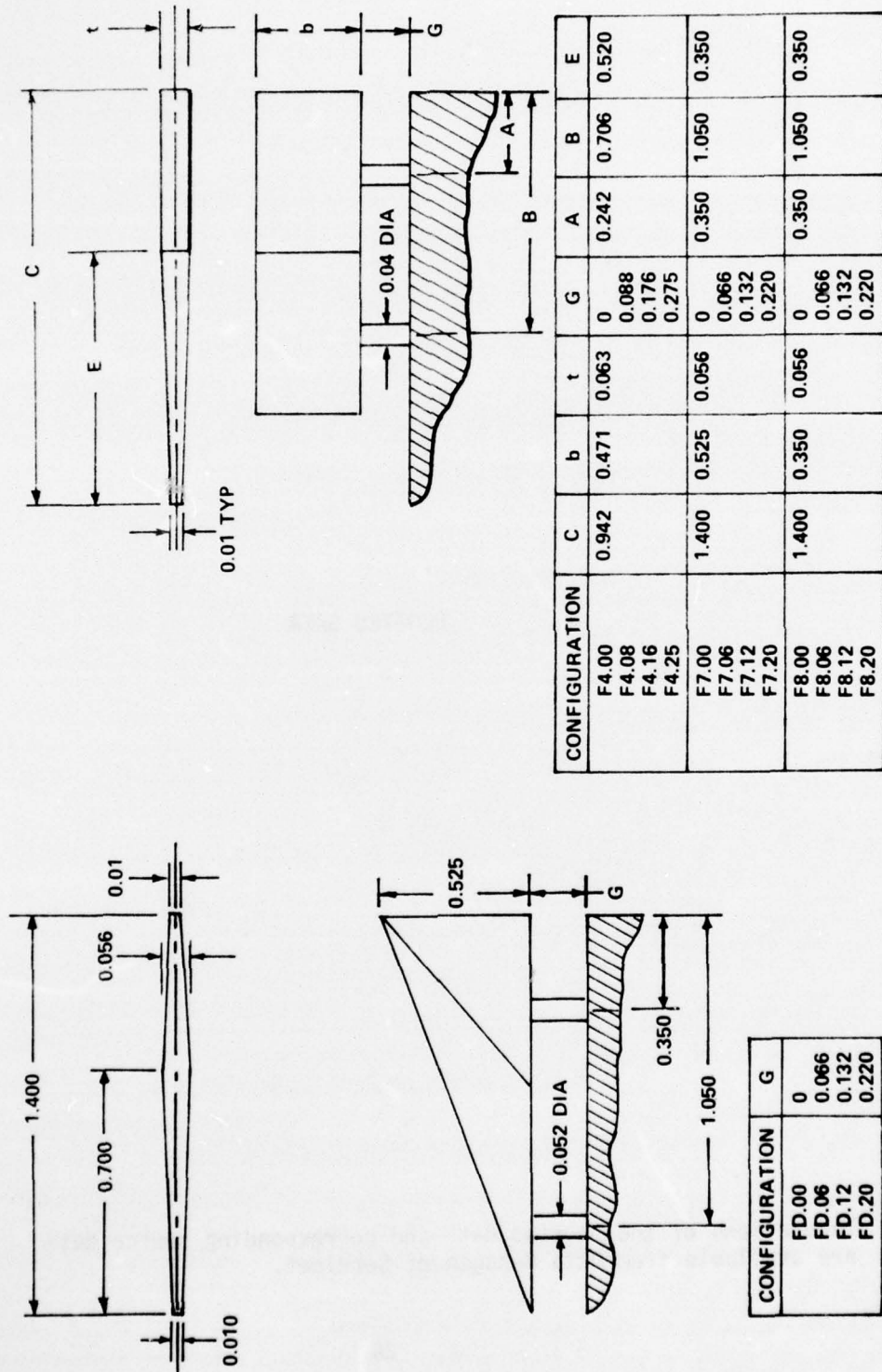


FIGURE 7. FIN GEOMETRY (REFERENCE 6)

PLOTTED DATA

Tabulations of the plotted data and corresponding source data are available from Data Management Services.

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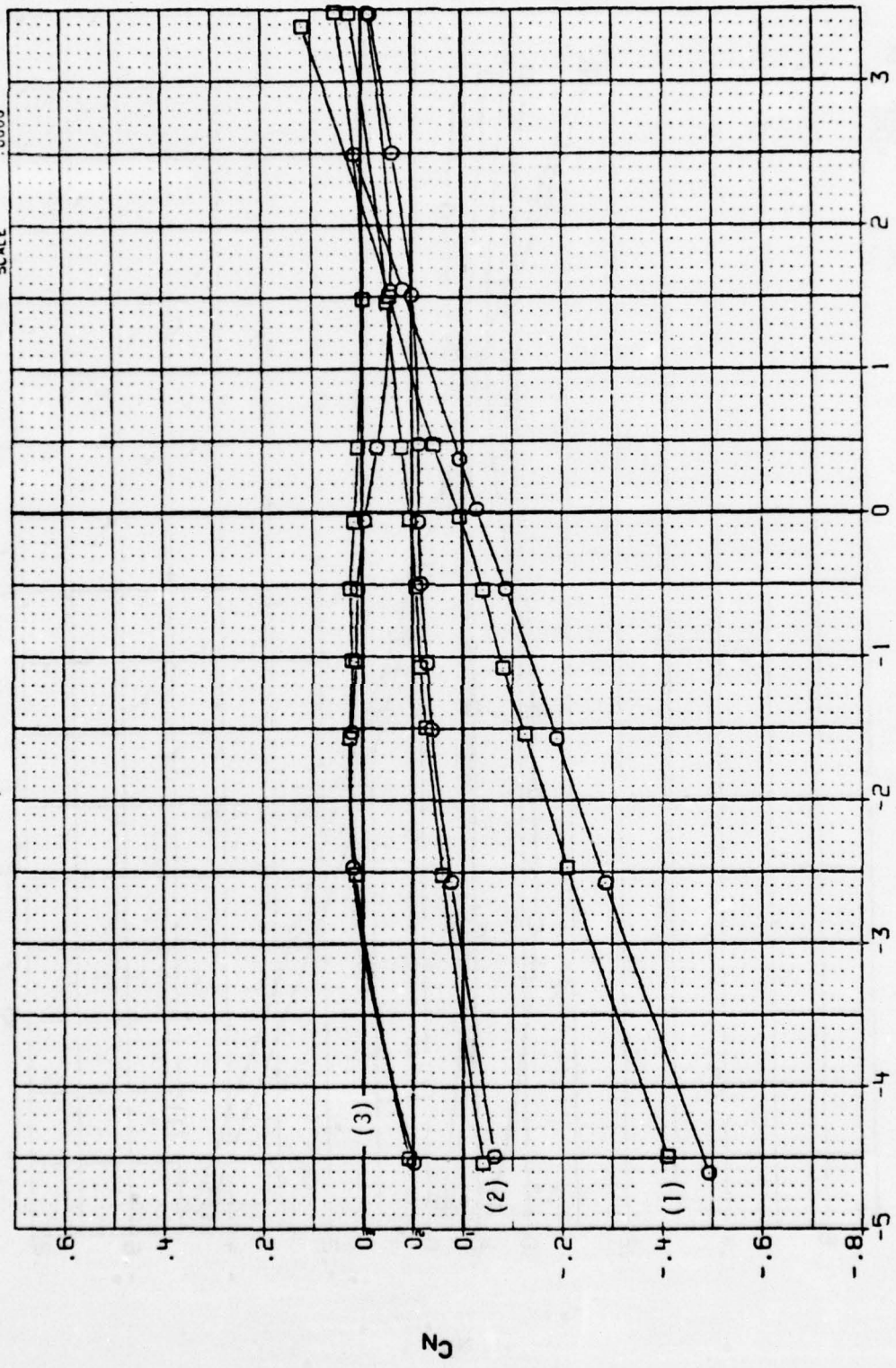
(EXB0671) ○ CAL 117-123 (BF2) BODY, FIN IN AFT POSITION
 (DXB0631) □ CAL 117-123 (BF4) BODY, FIN IN AFT POSITION

GAP MACH PHI

.000 .700 .000
 .250 .000 .000

REFERENCE INFORMATION

SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
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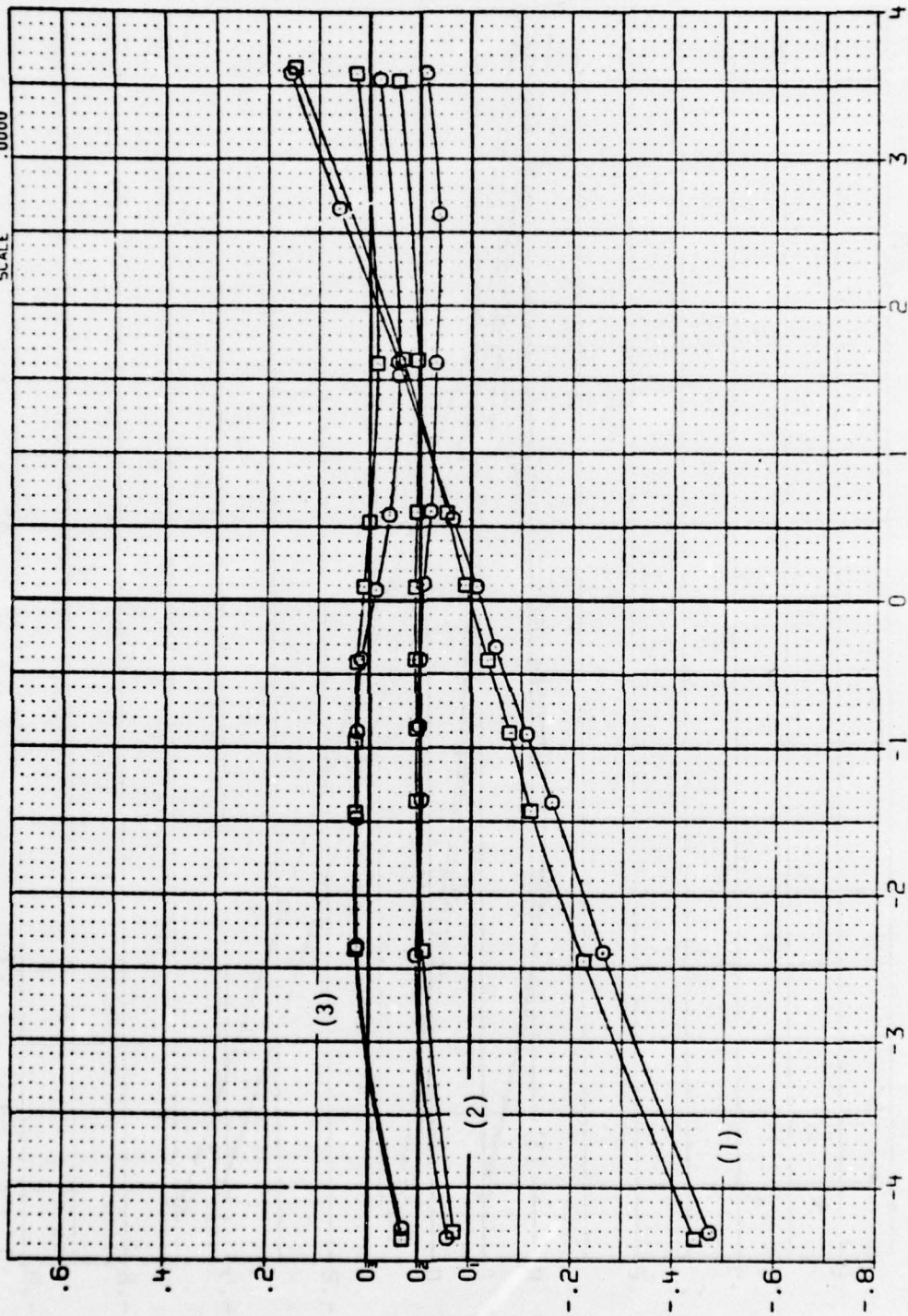
EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .03 (2) 5.79 (3) 10.53

DATA SET SYMBOL CONFIGURATION DESCRIPTION
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 (DX8064) \circ CAL 117-123 (BF4) BODY, FIN IN AFT POSITION

GAP MACH PHI
 .000 .900 .000
 .250 .900 .000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BRREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
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EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .03 (2) 5.72 (3) 8.76

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(EX8069) \square CAL 117-123 (BF2) BODY, FIN IN AFT POSITION

(DX8065) \circ CAL 117-123 (BF4) BODY, FIN IN AFT POSITION

GAP MACH PHI

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.250 1.000 .000

REFERENCE INFORMATION

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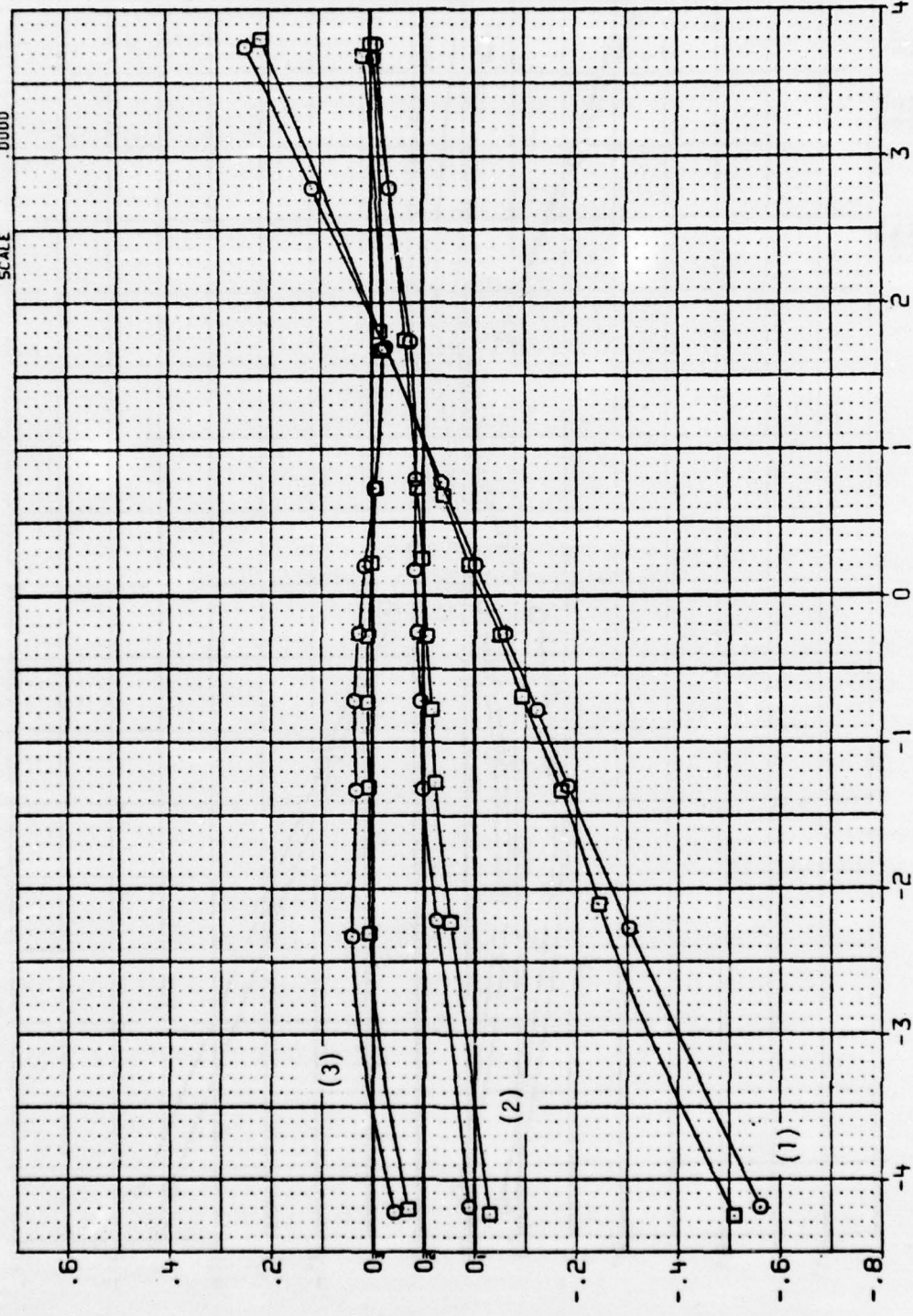
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XMRP 26.5000 IN.

YMRP .0000 IN.

ZMRP .0000 IN.

SCALE .0000



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EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .02 (2) 4.04 (3) 6.06

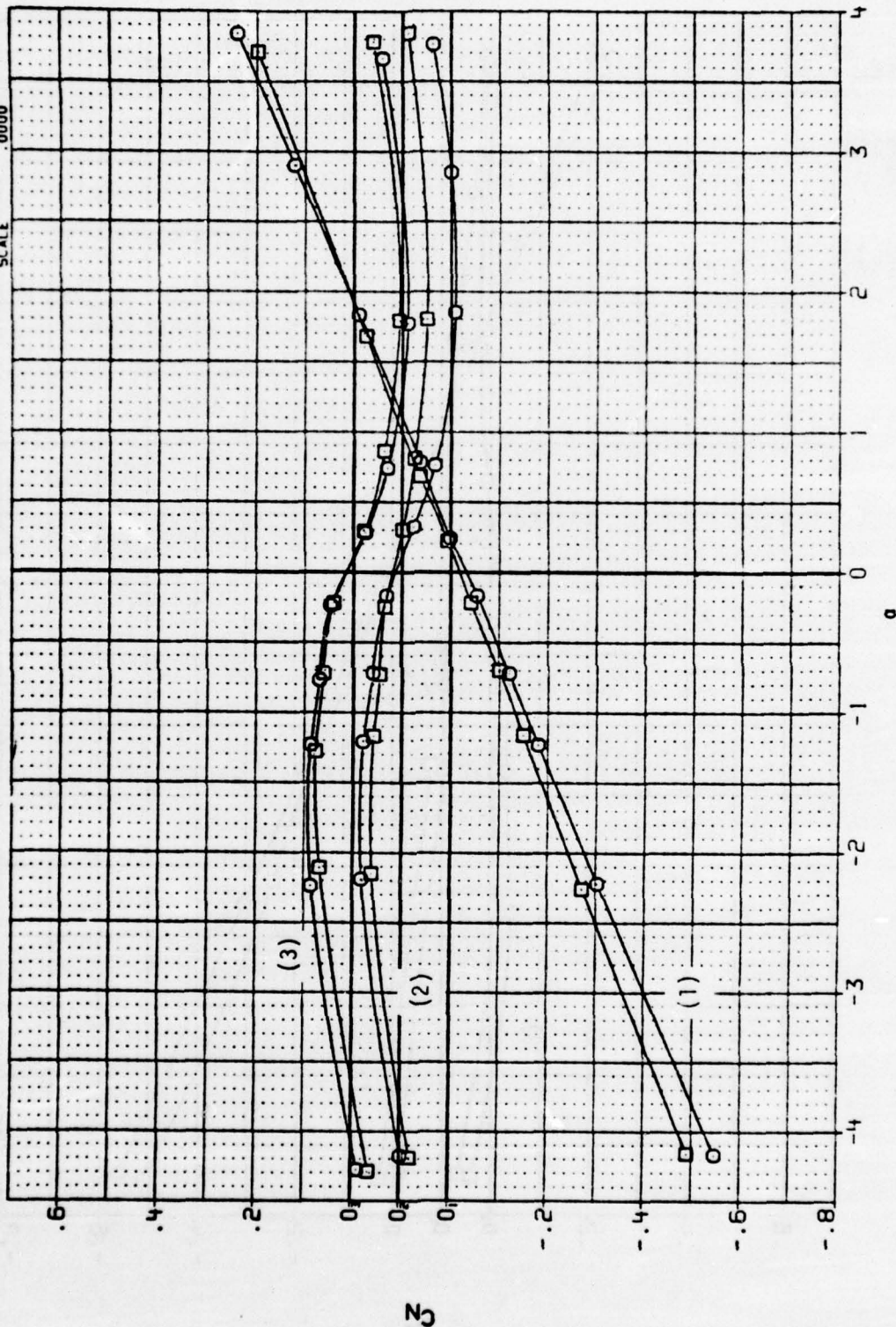
DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (X18070) CAL 117-123 (BF2) BODY, FIN IN AFT POSITION
 (DX18066) CAL 117-123 (BF4) BODY, FIN IN AFT POSITION

GAP .000
 .250

MACH 1.200
 1.200

PHI .000
 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
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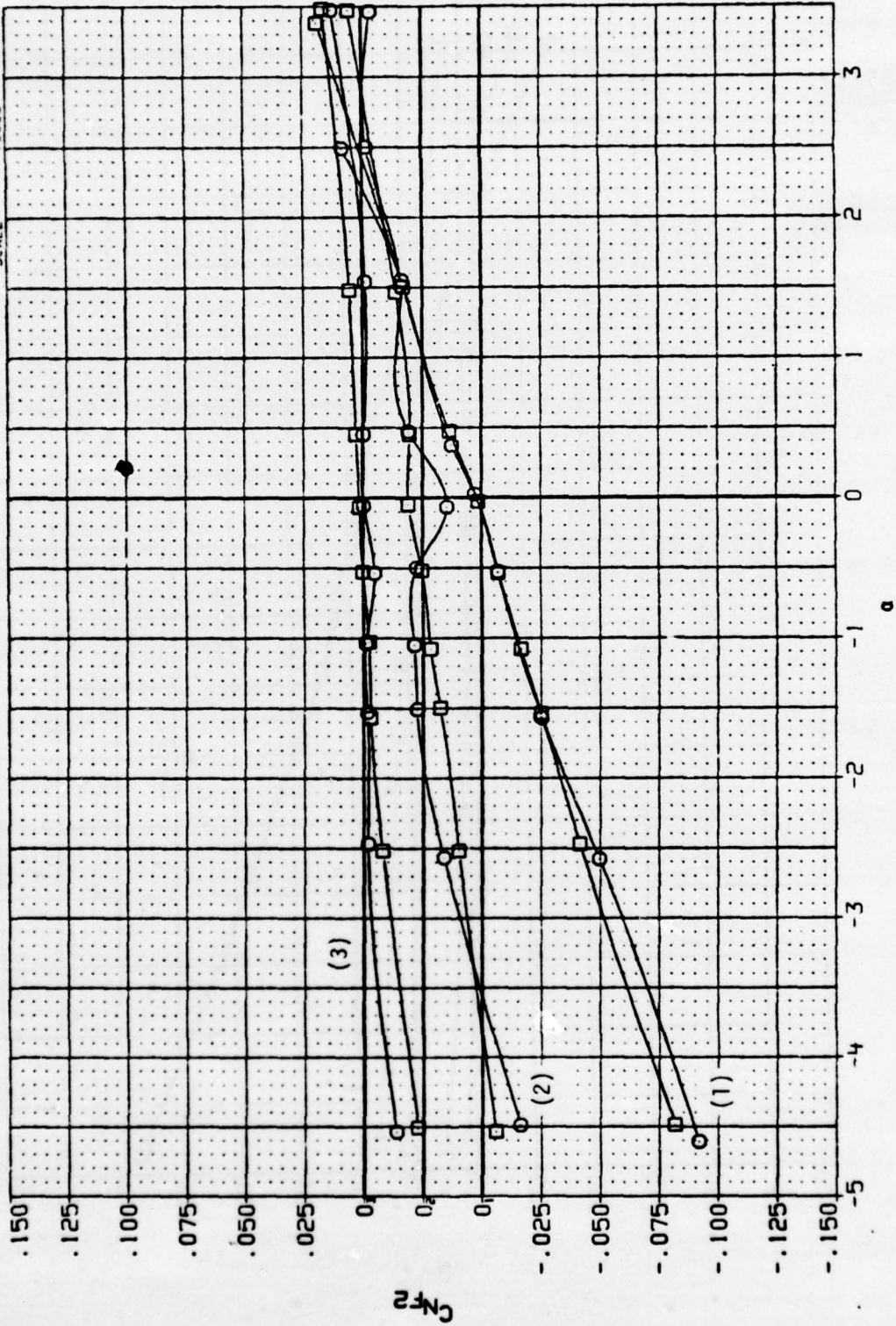
EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .01 (2) 4.18 (3) 5.97

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (EX8087) ○ CAL 117-123 (BF2) BODY, FIN IN AFT POSITION
 (DX8063) □ CAL 117-123 (BF4) BODY, FIN IN AFT POSITION

GAP .000 PHI
 .000 .000
 .250 .700 .700

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .03 (2) 5.79 (3) 10.53

DATA SET SYMBOL (EX0008) (DX0008)

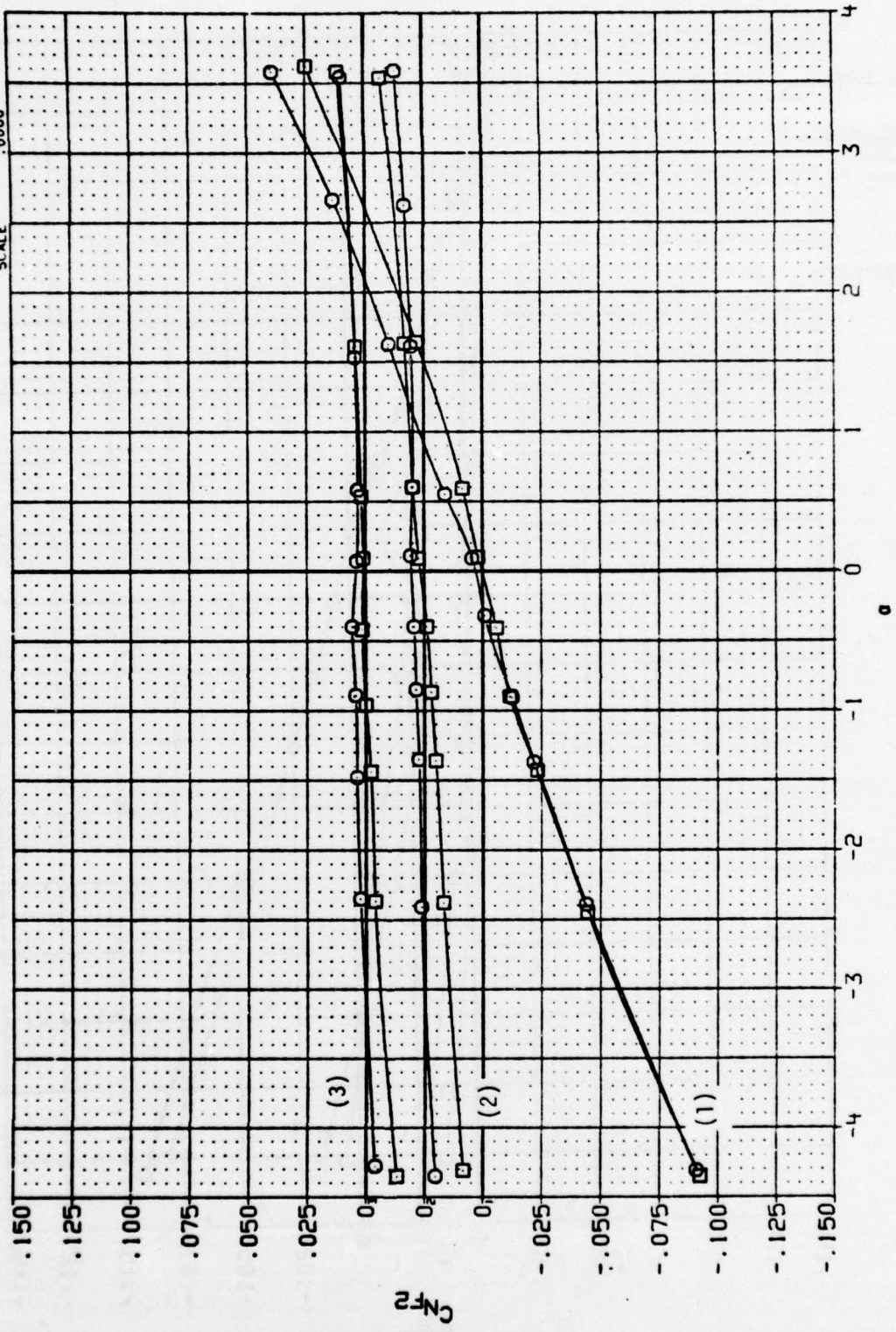
GAP .000 .250

MACH .900 .900

PHI .000 .000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000

CONFIGURATION DESCRIPTION
 CAL 117-123 (BF2) BODY, FIN IN AFT POSITION
 CAL 117-123 (BF4) BODY, FIN IN AFT POSITION



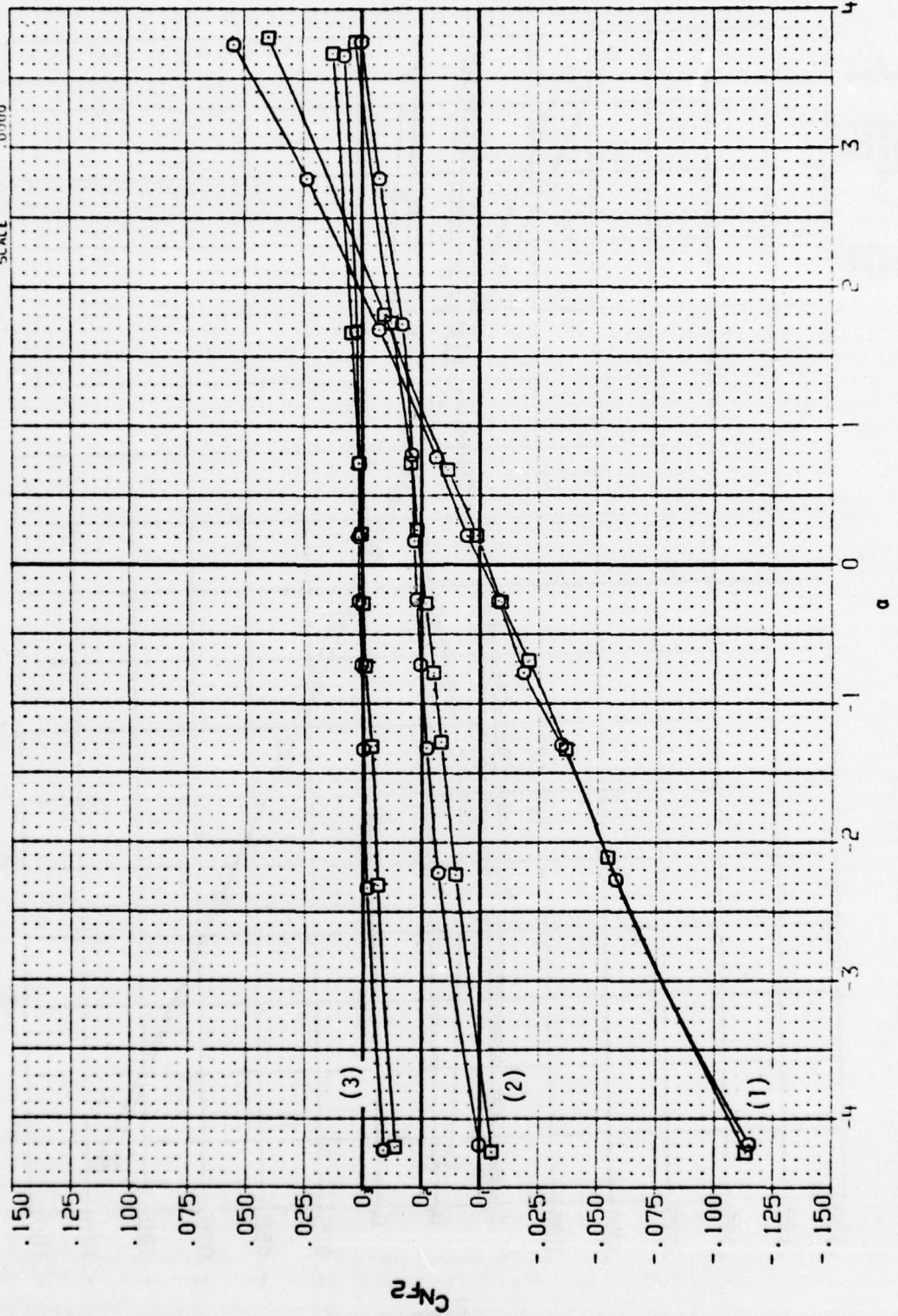
EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .03 (2) 5.72 (3) 8.76

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (EX8069) ○ CAL 117-123 (BF2) BODY, FIN IN AFT POSITION
 (DX8065) □ CAL 117-123 (BF4) BODY, FIN IN AFT POSITION

GAP .000 .000
 .250 .000
 MACH 1.000 1.000
 PHI .000 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 YMRP 26.5000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .02 (2) 4.04 (3) 6.06

DATA SET SYMBOL (EX8070) (DX8068)

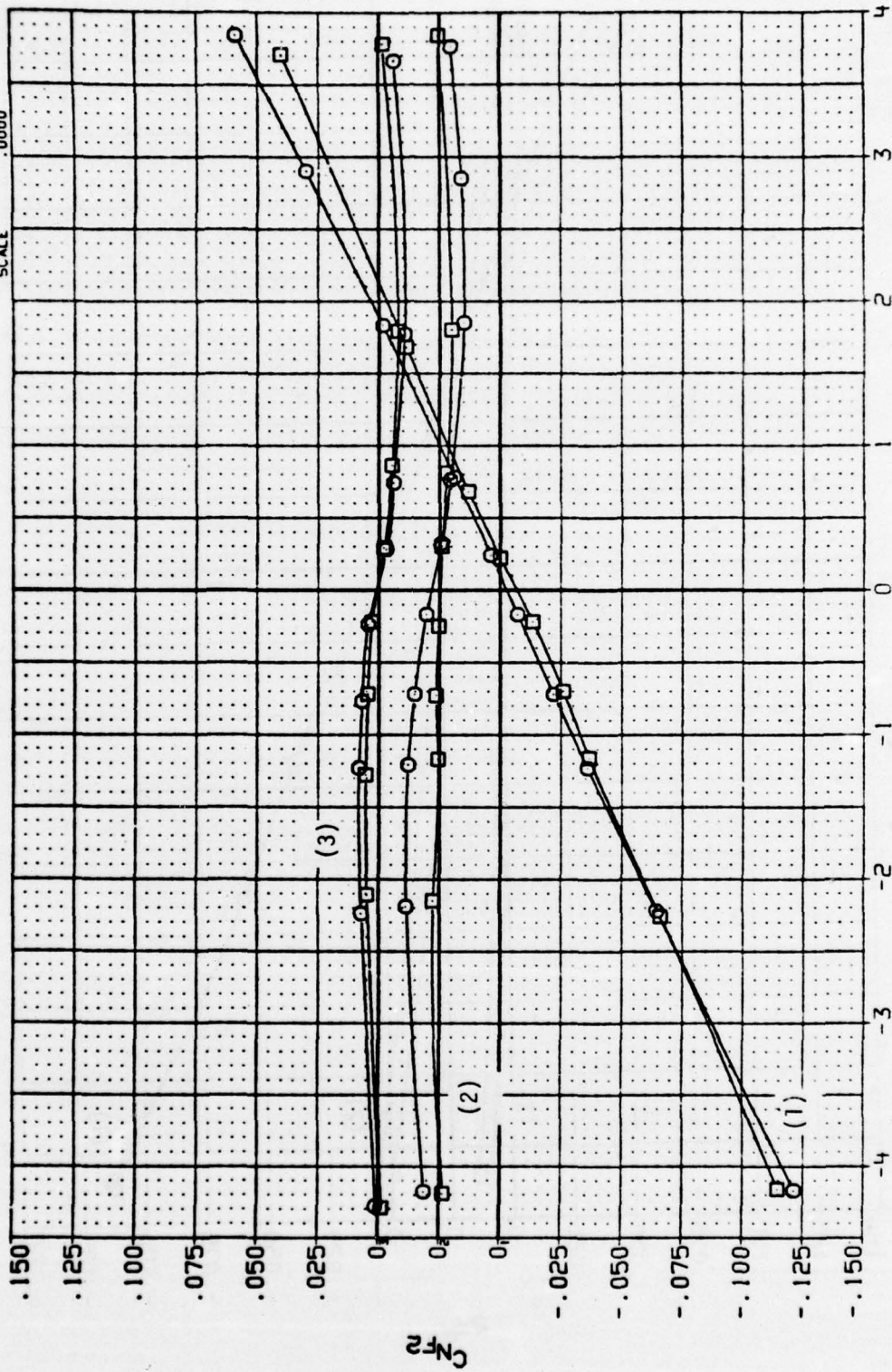
CONFIGURATION DESCRIPTION
 CAL 117-123 (BF2) BODY, FIN IN AFT POSITION
 CAL 117-123 (BF4) BODY, FIN IN AFT POSITION

QAP .000
 .250

MACH 1.200
 1.200

PHI .000
 .000

REFERENCE INFORMATION
 SREF 19.6350 SO. IN.
 LREF 5.0000 IN.
 BRE 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



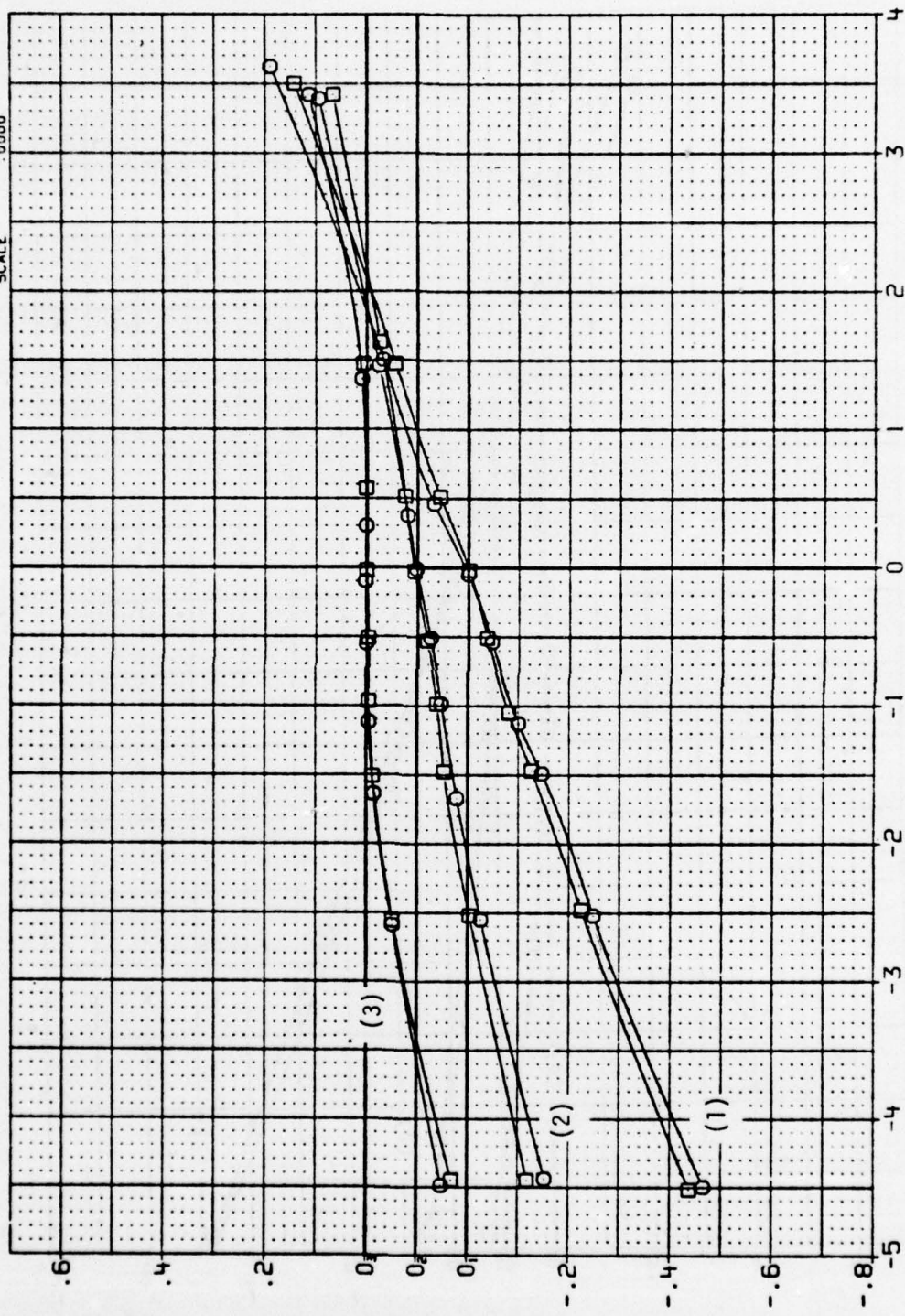
EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .01 (2) 4.18 (3) 5.97

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (DX8045) CAL 117-123 (BF2) BODY, FIN IN MID POSITION
 (DX8049) CAL 117-123 (BF4) BODY, FIN IN MID POSITION

GAP .000 .250
 MACH .700 .700
 PHI .000 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE



EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .08 (2) 5.87 (3) 10.76

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(DX80461) CAL 117-123 (BF2) BODY, FIN IN MID POSITION

(DX80450) CAL 117-123 (BF4) BODY, FIN IN MID POSITION

GAP MACH PHI

.000 .900 .000

.250 .900 .000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.

LREF 5.0000 IN.

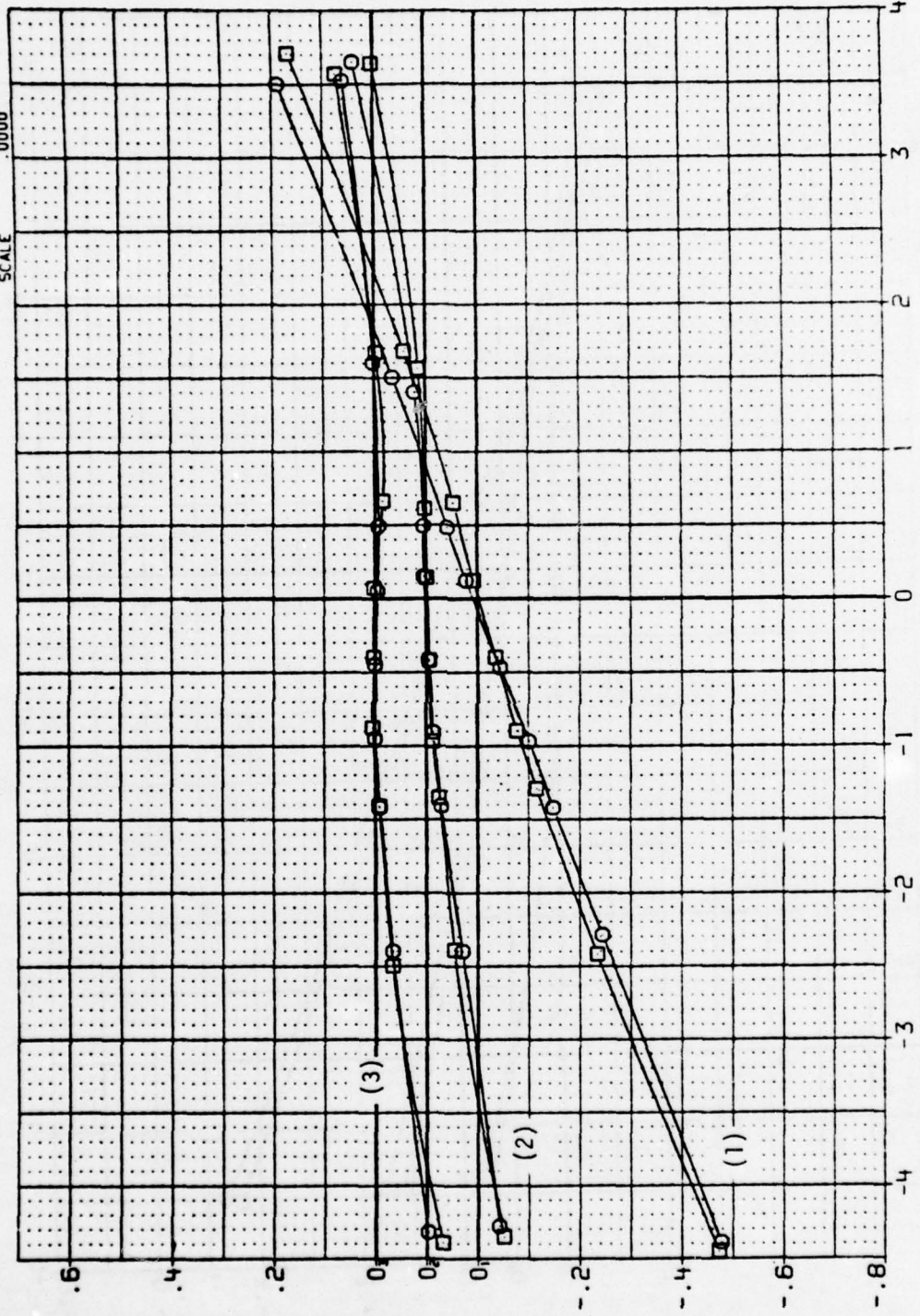
BREF 5.0000 IN.

XMRP 26.5000 IN.

YMRP .0000 IN.

ZMRP .0000 IN.

SCALE .0000



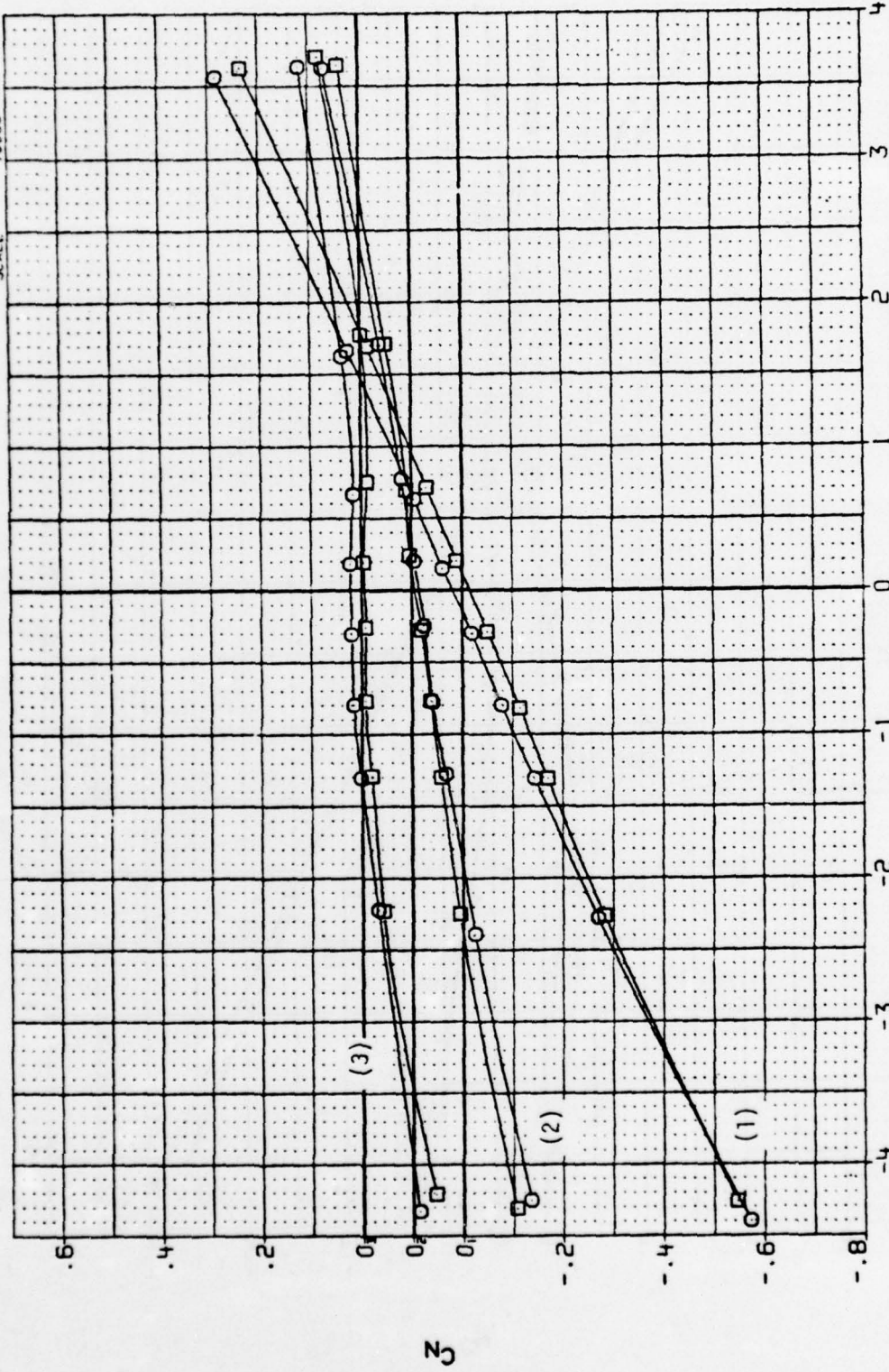
EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .05 (2) 5.65 (3) 8.60

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (DX8047) ○ CAL 117-123 (BF2) BODY, FIN IN MID POSITION
 (DX6031) □ CAL 117-123 (BF4) BODY, FIN IN MID POSITION

GAP MACH PHI
 .000 1.000 .000
 .250 1.000 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMPF 26.5000 IN.
 YMPF .0000 IN.
 ZMPF .0000 IN.
 SCALE .0000



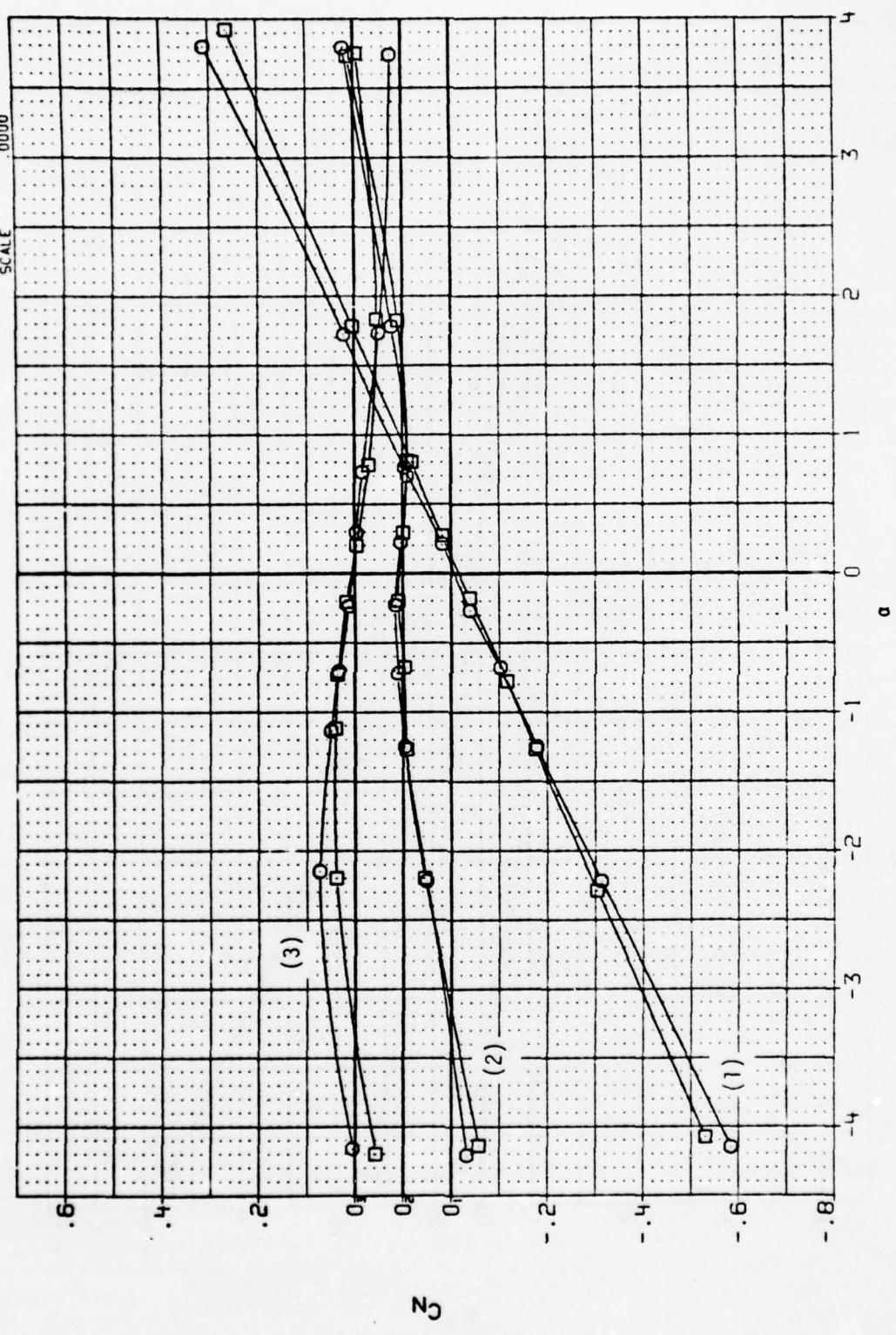
EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .05 (2) 4.13 (3) 5.71

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (DX6048) CAL 117-123 (BF2) BODY, FIN IN MID POSITION
 (DX6052) CAL 117-123 (BF4) BODY, FIN IN MID POSITION

GAP MACH PHI
 .000 1.200 .000
 .250 1.200 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



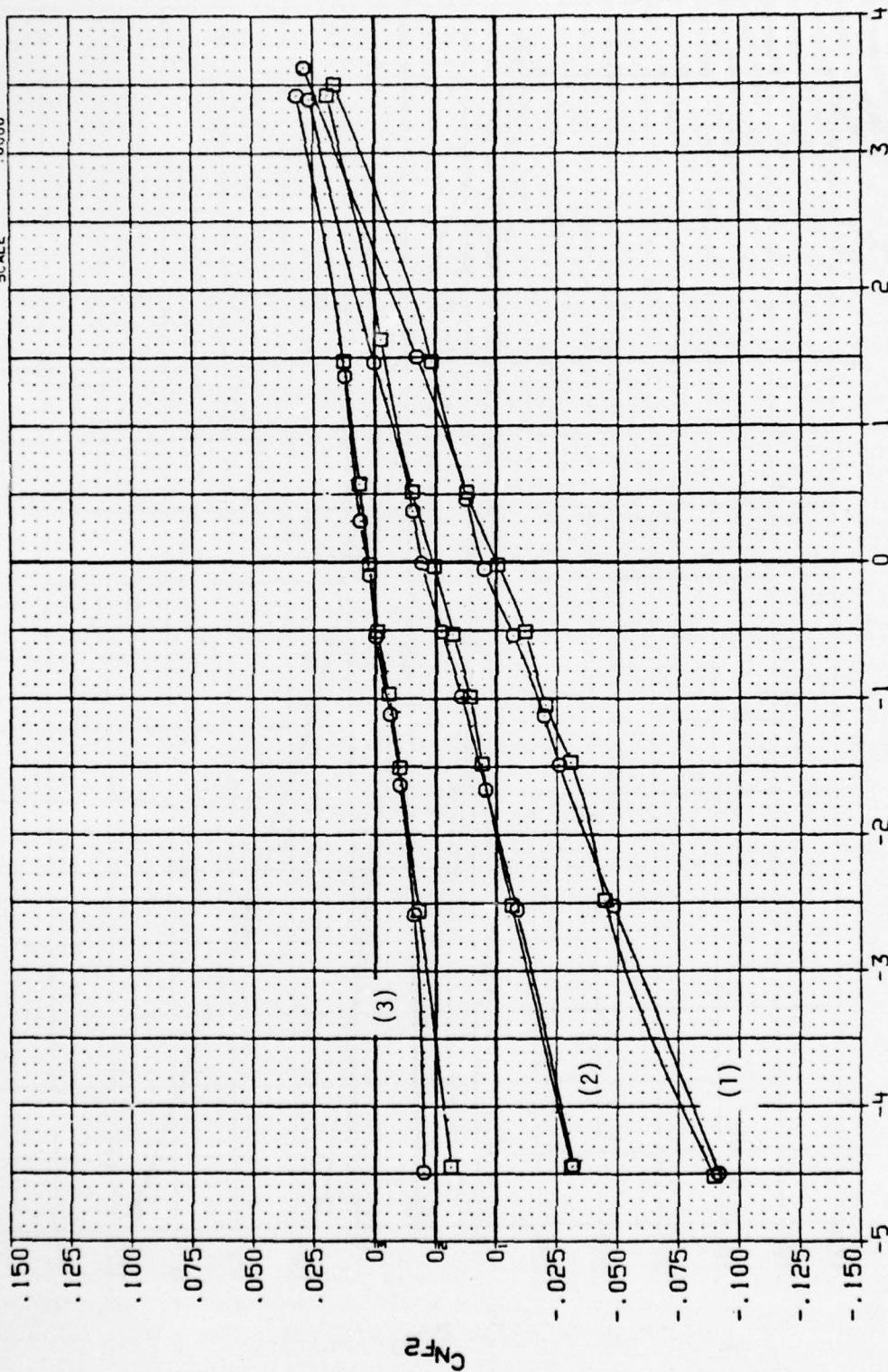
EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .03 (2) 4.24 (3) 5.75

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (DX8045) ○ CAL 117-123 (BF2) BODY, FIN IN MID POSITION
 (DX8049) □ CAL 117-123 (BF4) BODY, FIN IN MID POSITION

GAP MACH PHI
 .000 .700 .000
 .250 .700 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 10. IN.
 BREF 5.0000 10. IN.
 XMRP 26.5000 10. IN.
 YMRP .0000 10. IN.
 ZMRP .0000 10. IN.
 SCALE .0000



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .08 (2) 5.87 (3) 10.76

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(DXB048) ○ CAL 117-123 (B*2) BODY, FIN IN MID POSITION

(DXB050) □ CAL 117-123 (B*4) BODY, FIN IN MID POSITION

GAP MACH PHI

.000 .900 .000

.250 .900 .000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

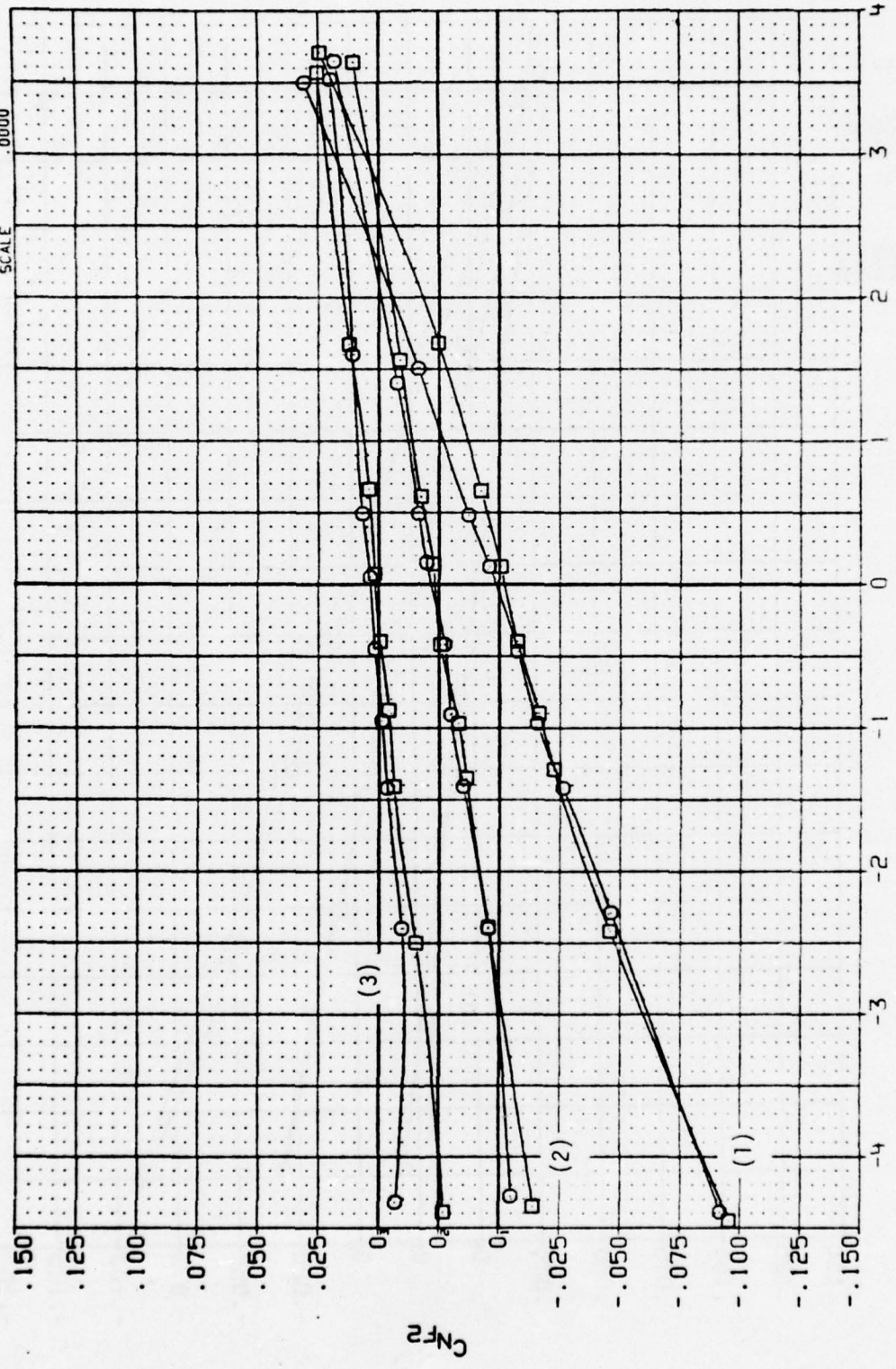
LREF 5.0000 5. IN.

BREF 5.0000 5. IN.

YMRP 26.5000 26. IN.

ZMRP .0000 . IN.

SCALE .0000 . IN.



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .05 (2) 5.65 (3) 8.60

DATA SET SYMBOL: (DX8047) (DX8051)

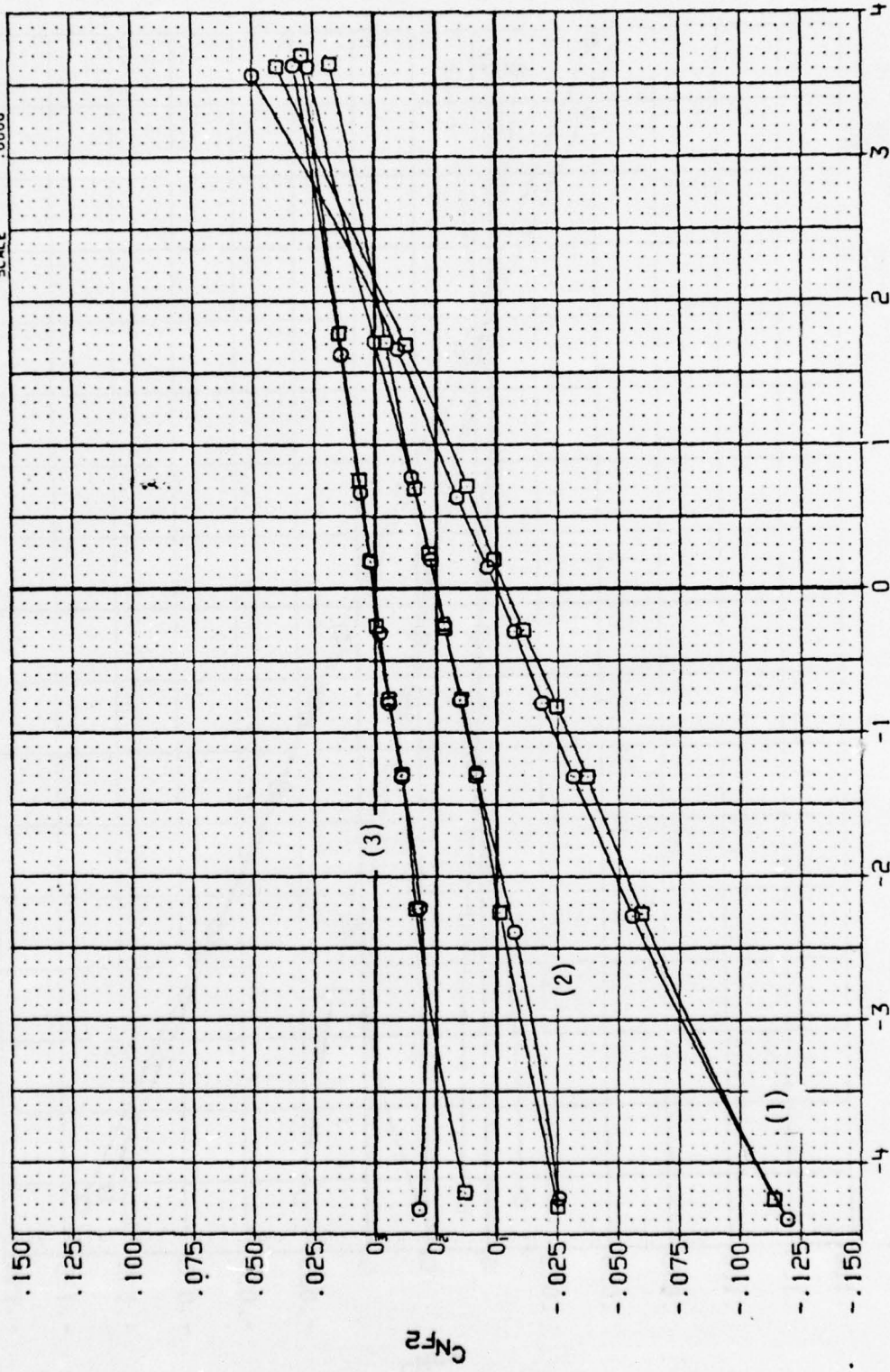
CONFIGURATION DESCRIPTION:
 CAL 117-123 (BF2) BODY, FIN IN MID POSITION
 CAL 117-123 (BF4) BODY, FIN IN MID POSITION

REFERENCE INFORMATION:
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000

GAP: .000, .250

MACH: 1.000, 1.000

PHI: .000, .000



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .05 (2) 4.13 (3) 5.71

DATA SET SYMBOL (DX0048) □ (DX0052) ○

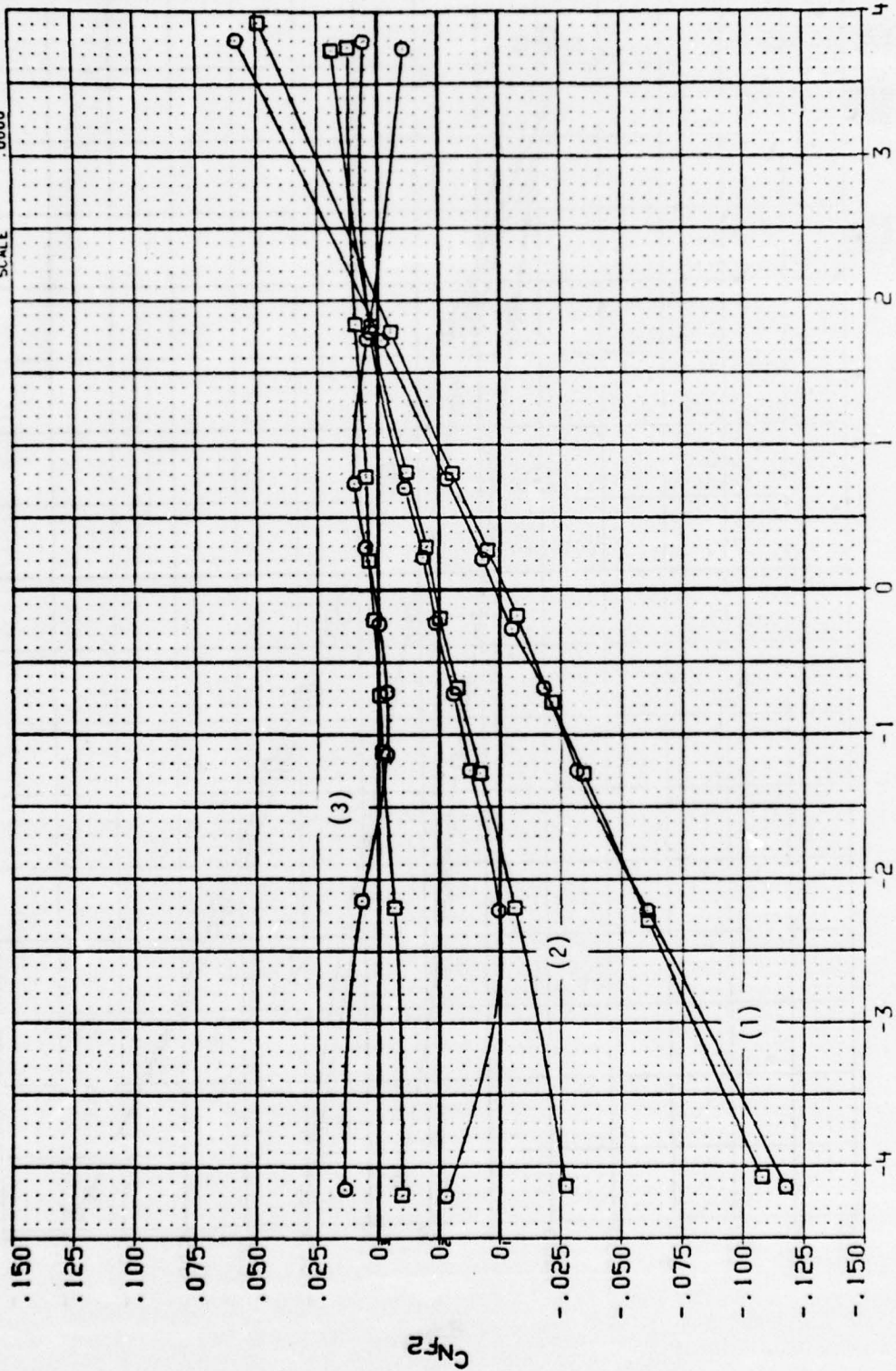
CONFIGURATION DESCRIPTION
 CAL 117-123 (BF2) BODY, FIN IN MID POSITION
 CAL 117-123 (BF4) BODY, FIN IN MID POSITION

GAP .000 .250

MACH 1.200 1.200

PHI .000 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



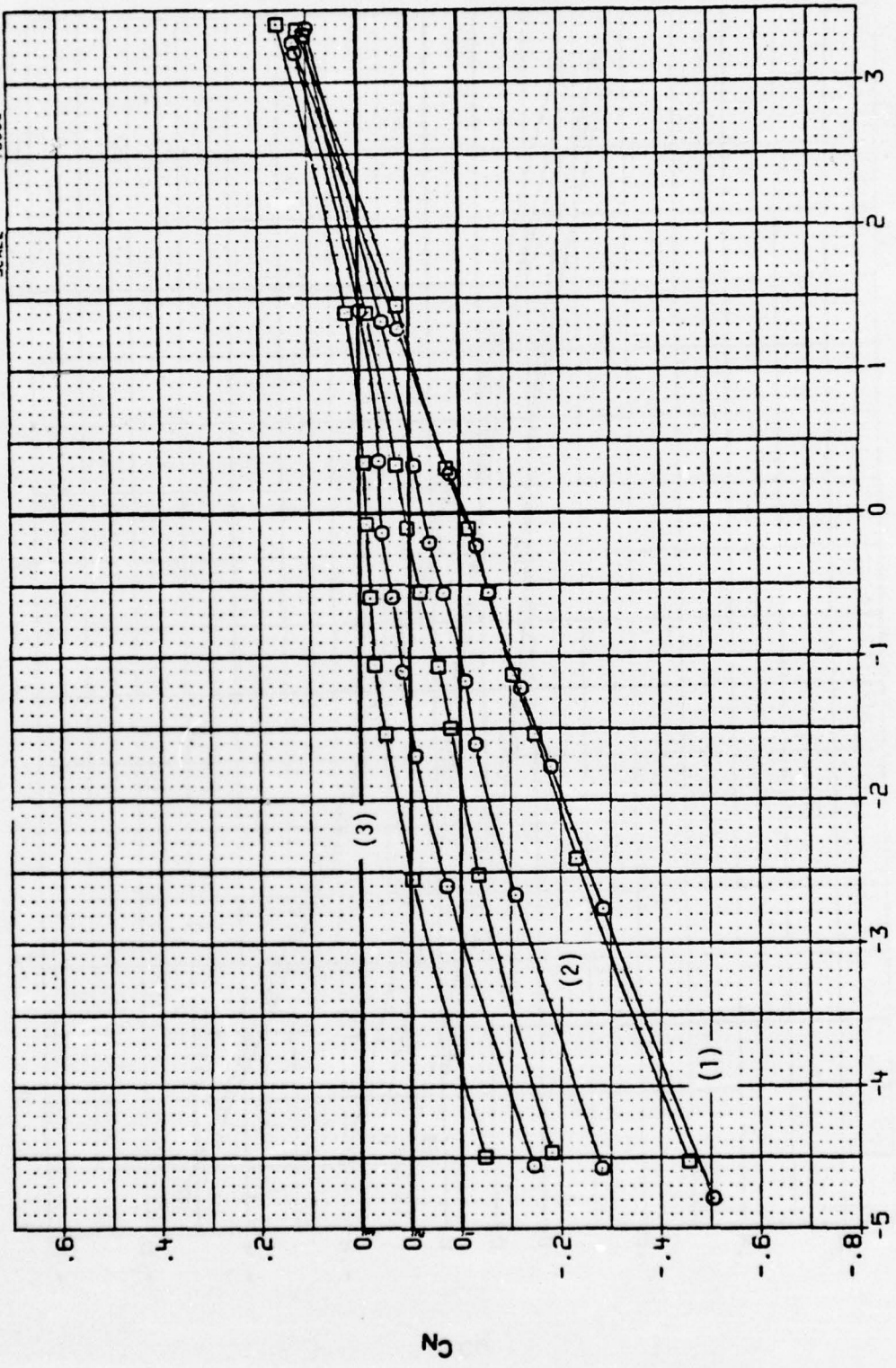
EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .03 (2) 4.24 (3) 5.75

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (EX8013) □ CAL T17-123 (BF2) BODY, FIN IN FORWARD POSITION
 (DX80251) ○ CAL T17-123 (BF4) BODY, FIN IN FORWARD POSITION

GAP MACH PHI
 .000 .700 .000
 .250 .700 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



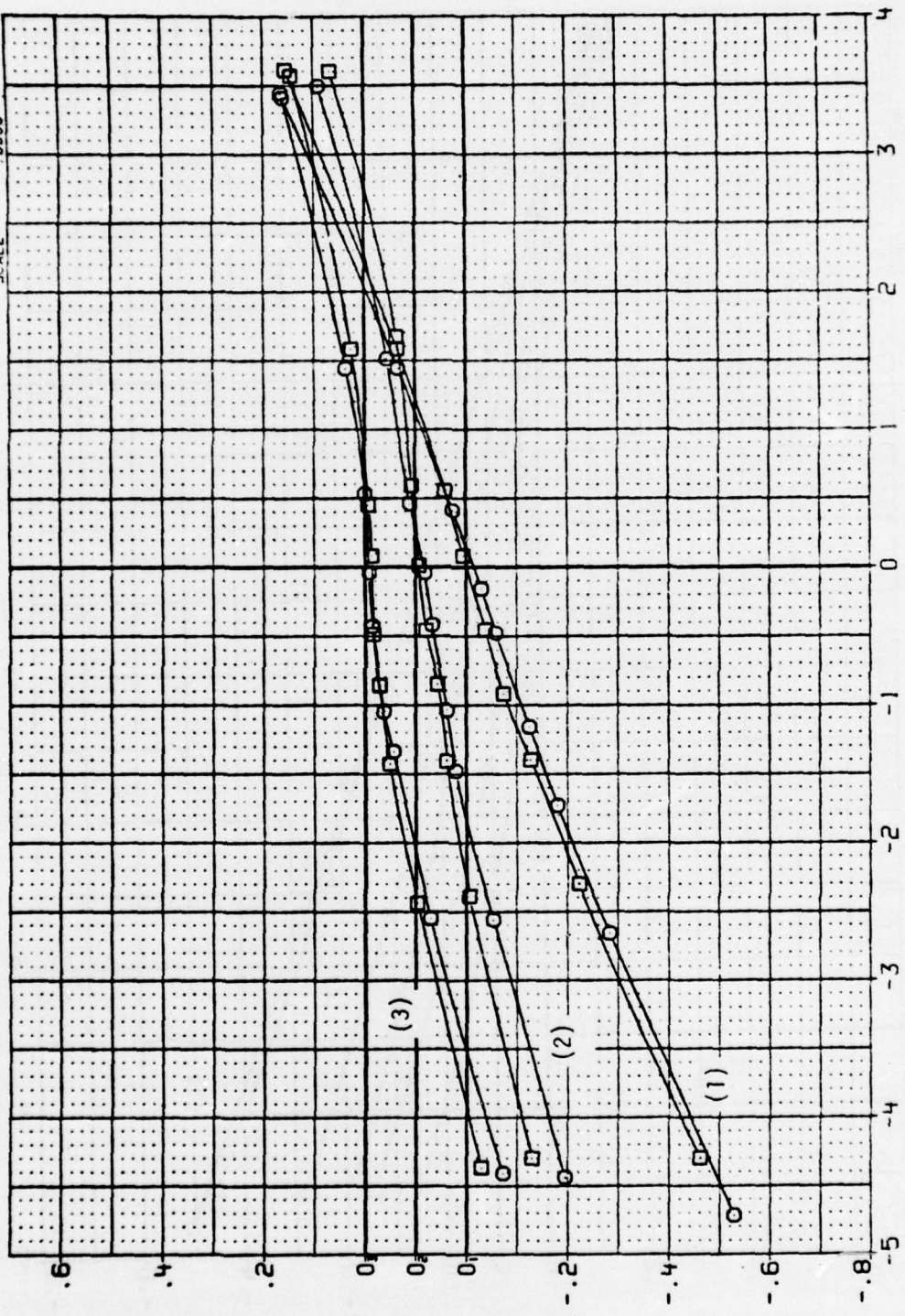
EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .07 (2) 5.91 (3) 10.05

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (EX8014) ○ CAL 117-123 (8F2) BODY, FIN IN FORWARD POSITION
 (DX8026) □ CAL 117-123 (8F4) BODY, FIN IN FORWARD POSITION

GAP MACH PHI
 .000 .900 .000
 .250 .900 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



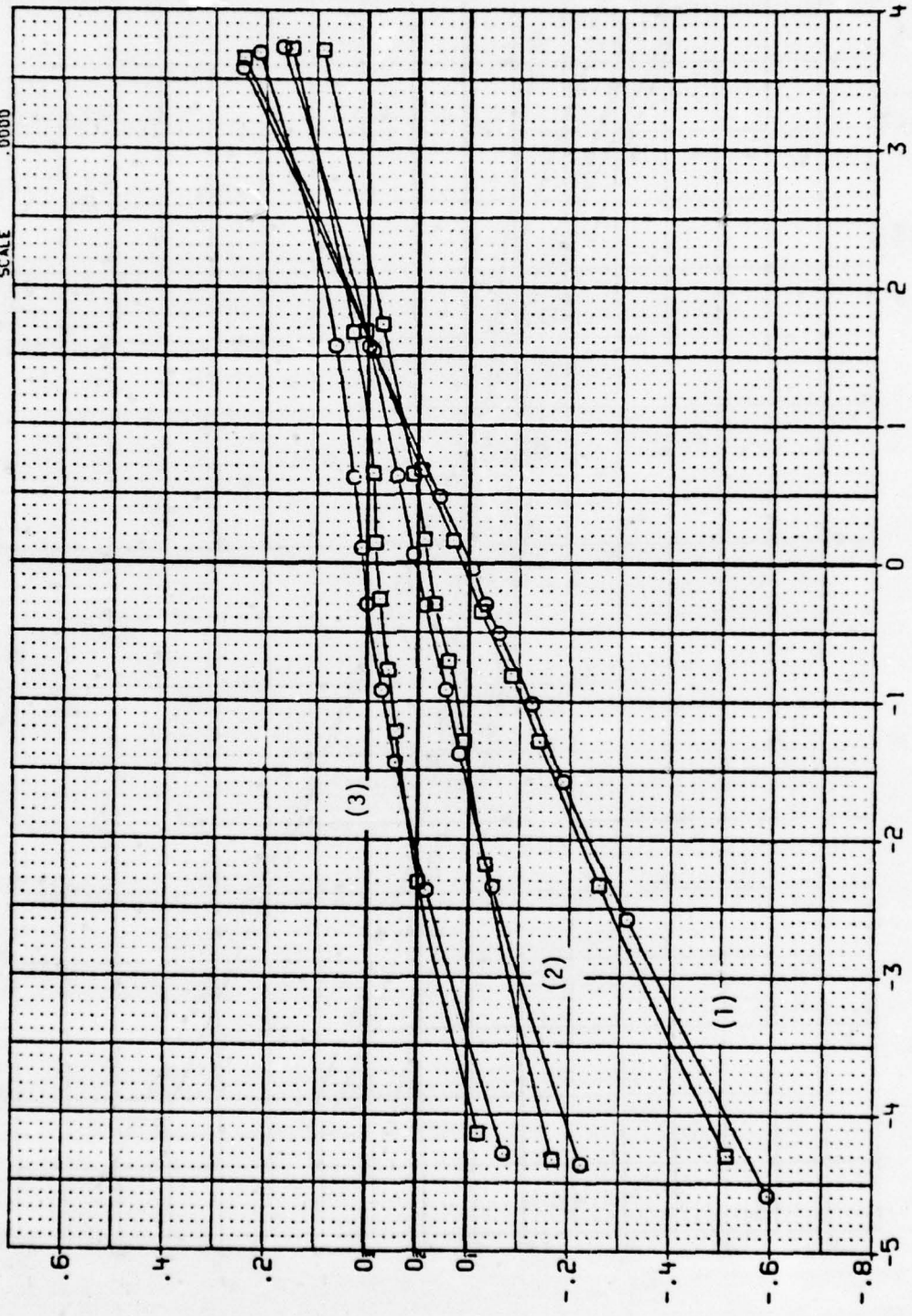
EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .03 (2) 5.83 (3) 8.87

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (EX8019) ○ CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION
 (DX8027) □ CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION

GAP MACH PHI
 .000 1.000 .000
 .250 1.000 .000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .03 (2) 4.18 (3) 5.99

3

DATA SET SYMBOL
 (EX8013) ○
 (DX8023) □

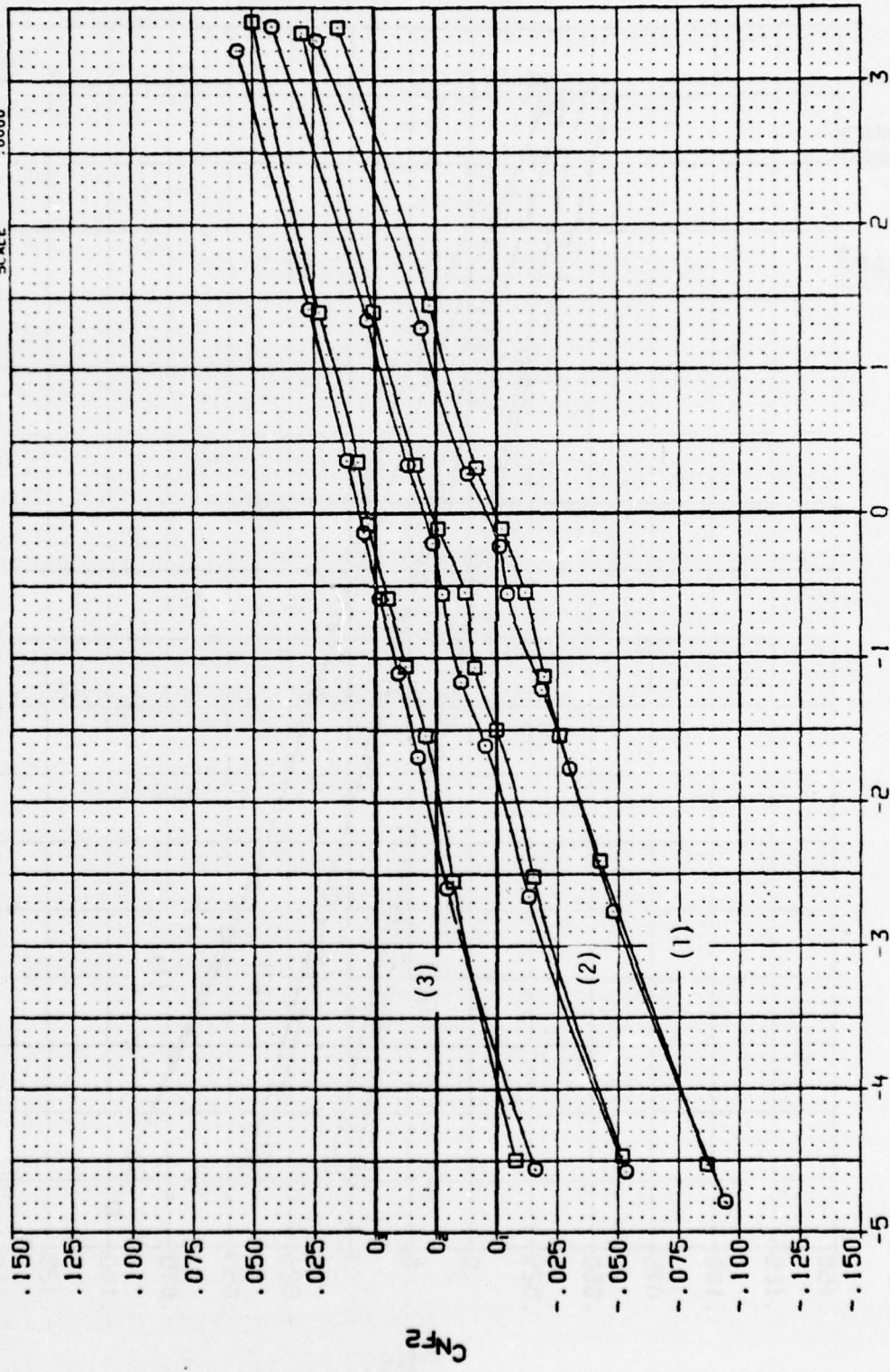
CONFIGURATION DESCRIPTION
 CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION
 CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION

GAP
 .000
 .250

MACH
 .700
 .700

PHI
 .000
 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XHRP 26.5000 IN.
 YHRP .0000 IN.
 ZHRP .0000 IN.
 SCALE .0000



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

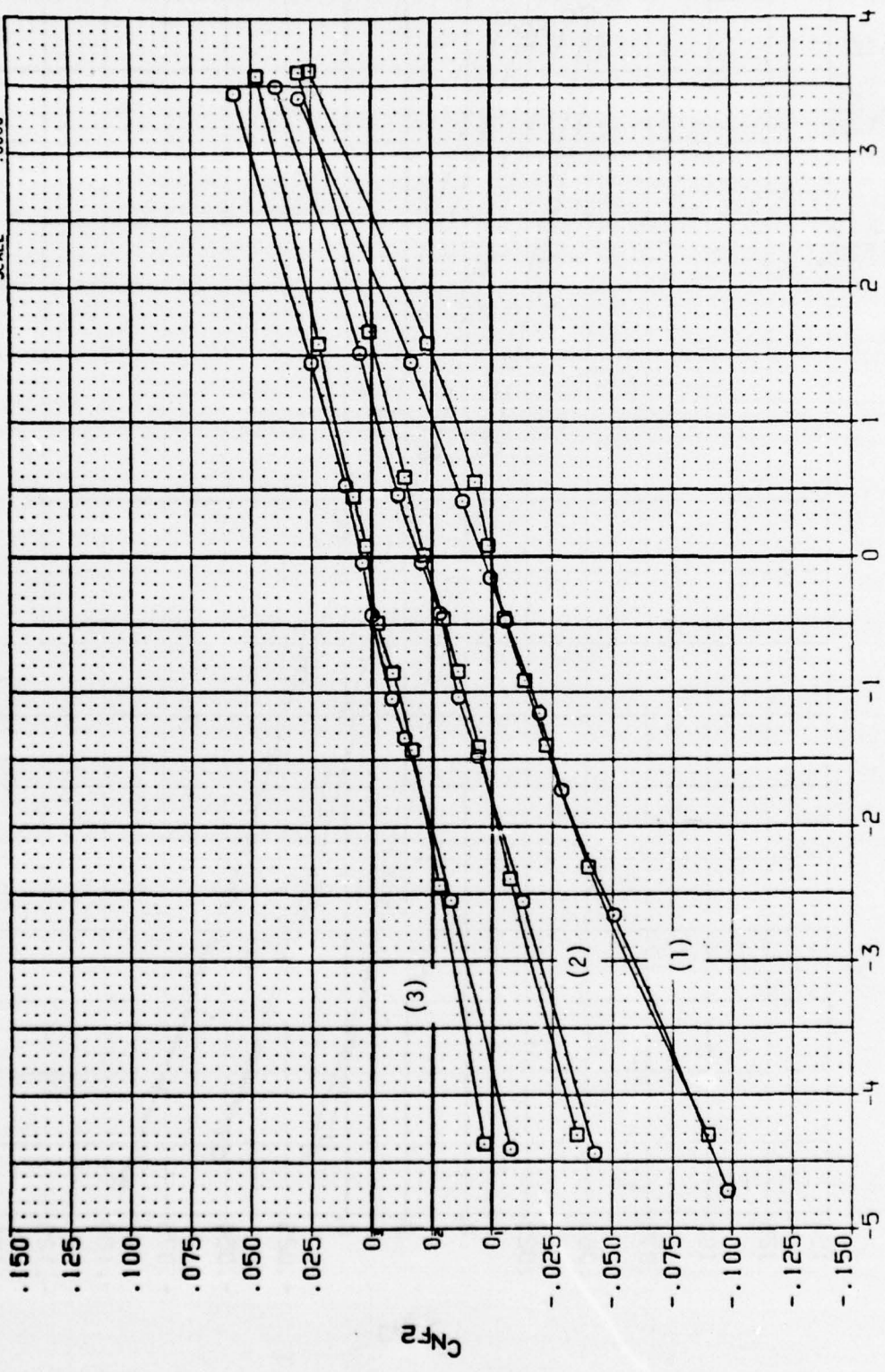
(1) CRT = .07 (2) 5.91 (3) 10.05

DATA SET SYMBOL (EX8014) (DX8026) \square \circ

CONFIGURATION DESCRIPTION
 CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION
 CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION

GAP .000 .250
 MACH .900 .900
 PHI .000 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



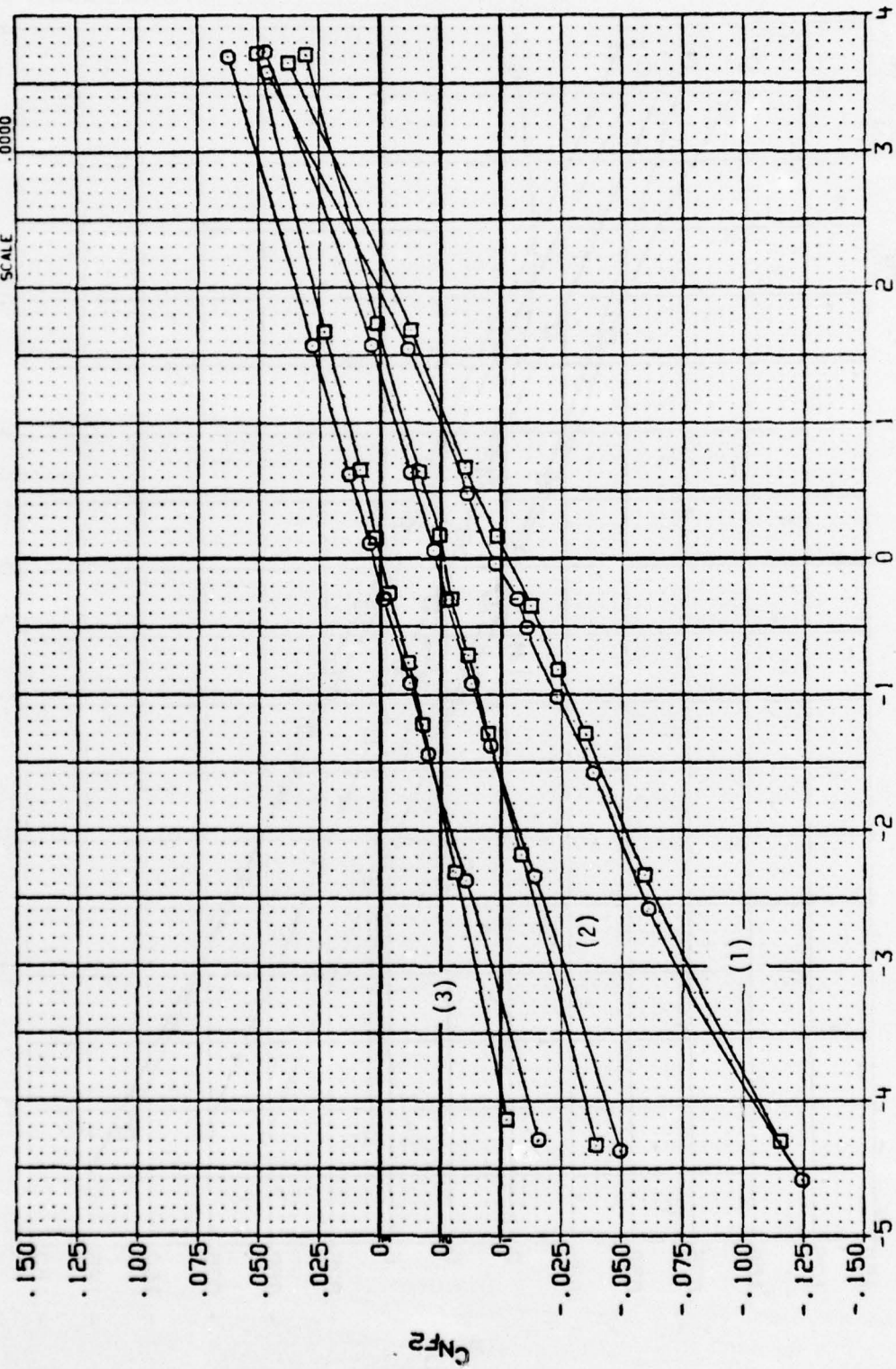
EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .03 (2) 5.83 (3) 8.87

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (EX8015) ○ CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION
 (DX8027) □ CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION

GAP MACH PHI
 .000 1.000 .000
 .250 1.000 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .03 (2) 4.18 (3) 5.99

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(EX80181) O CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION

(0180281) □ CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION

OAP MACH PHI

.000 1.200 .000

.250 1.200 .000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

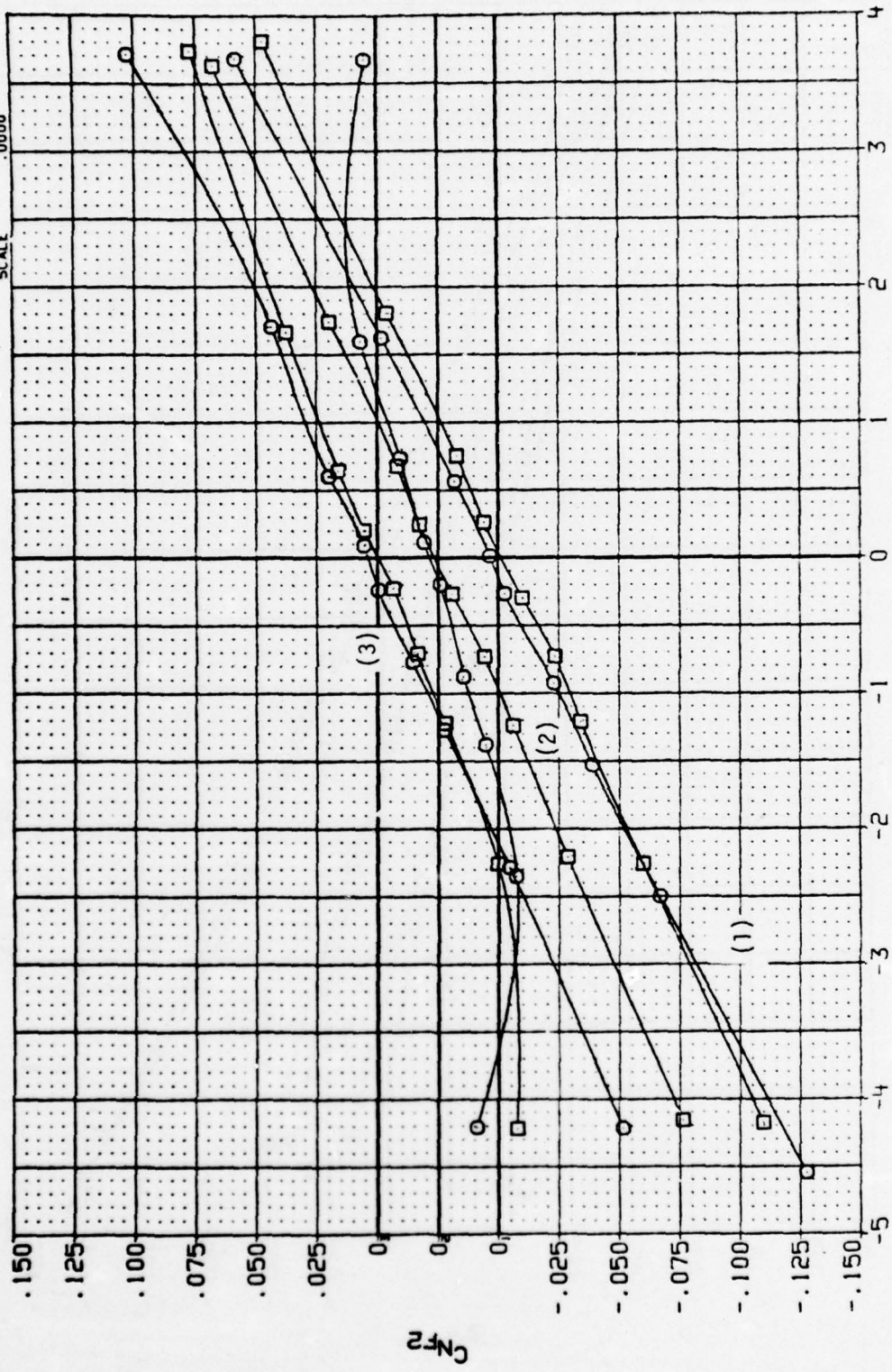
LREF 5.0000 IN.

BREF 5.0000 IN.

XHRP 26.5000 IN.

ZHRP .0000 IN.

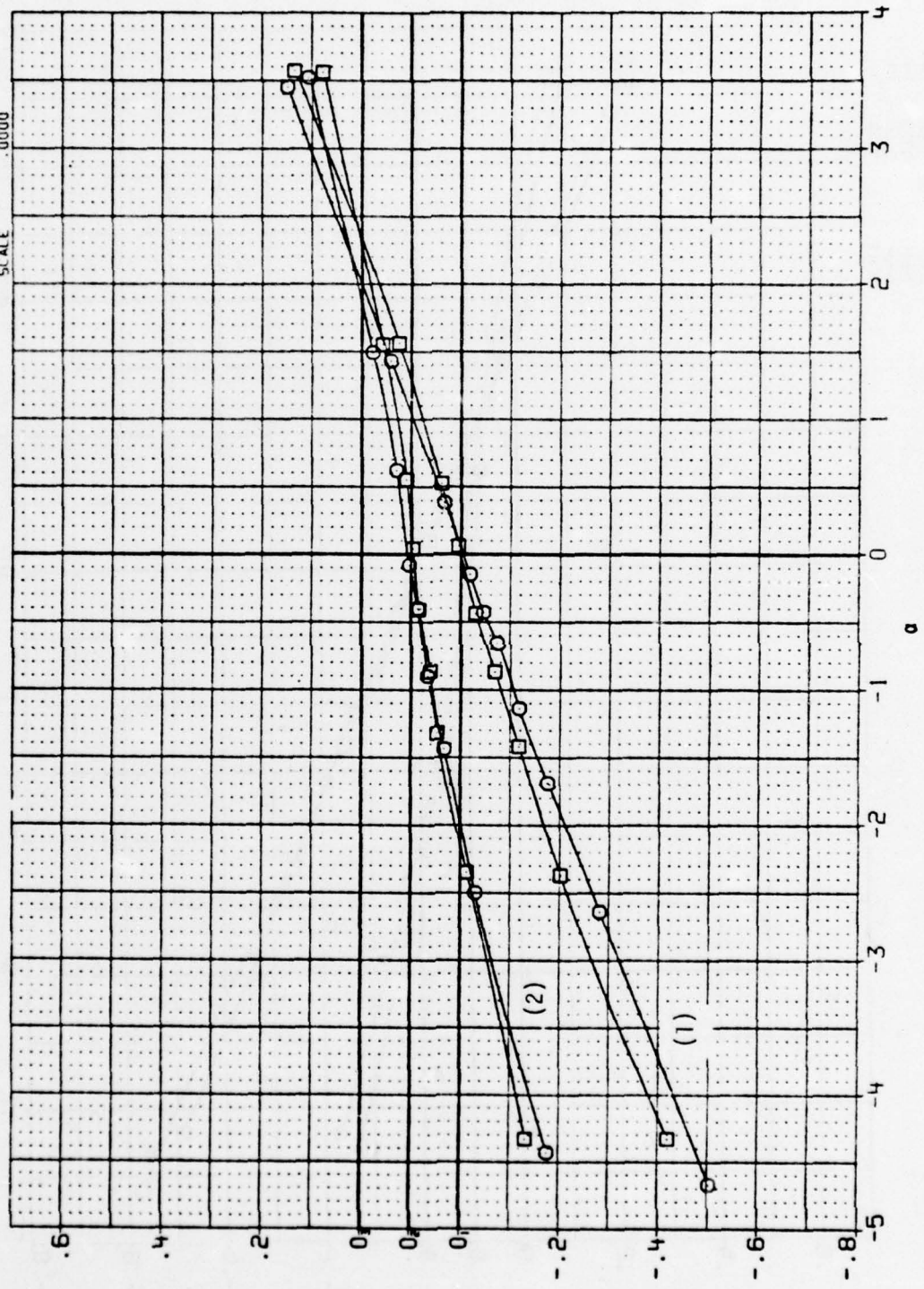
SCALE .0000 IN.



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .02 (2) 4.20 (3) 5.90

| | | | | | |
|------------------------|---|------------|-------------|------------|------------------------------|
| DATA SET SYMBOL | CONFIGURATION DESCRIPTION | GAP | MACH | PHI | REFERENCE INFORMATION |
| (EX8017) | CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION | .000 | .900 | 45.000 | SREF 19.6350 50. IN. |
| (DX8029) | CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION | .250 | .900 | 45.000 | LREF 5.0000 IN. |
| | | | | | BREF 5.0000 IN. |
| | | | | | XMRP 26.5000 IN. |
| | | | | | YMRP .0000 IN. |
| | | | | | ZMRP .0000 IN. |
| | | | | | SCALE .0000 |



EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .02 (2) 5.92

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(DX8018) ○ CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION

(DX8030) □ CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION

GAP .000 .250

MACH 1.000 1.000

PHI 45.000 45.000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

LREF 5.0000 IN.

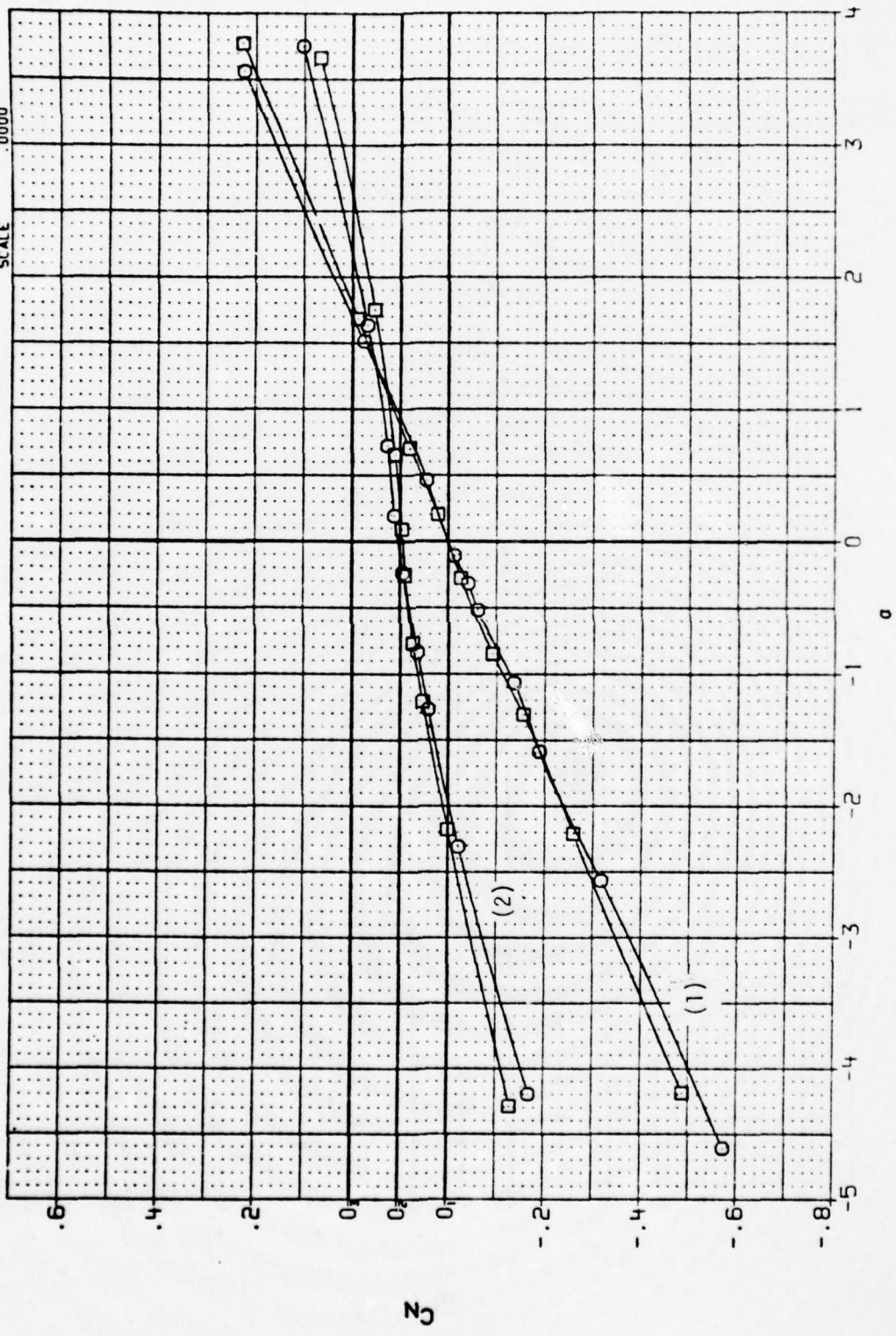
BREF 5.0000 IN.

XMRP 26.5000 IN.

YMRP .0000 IN.

ZMRP .0000 IN.

SCALE .0000



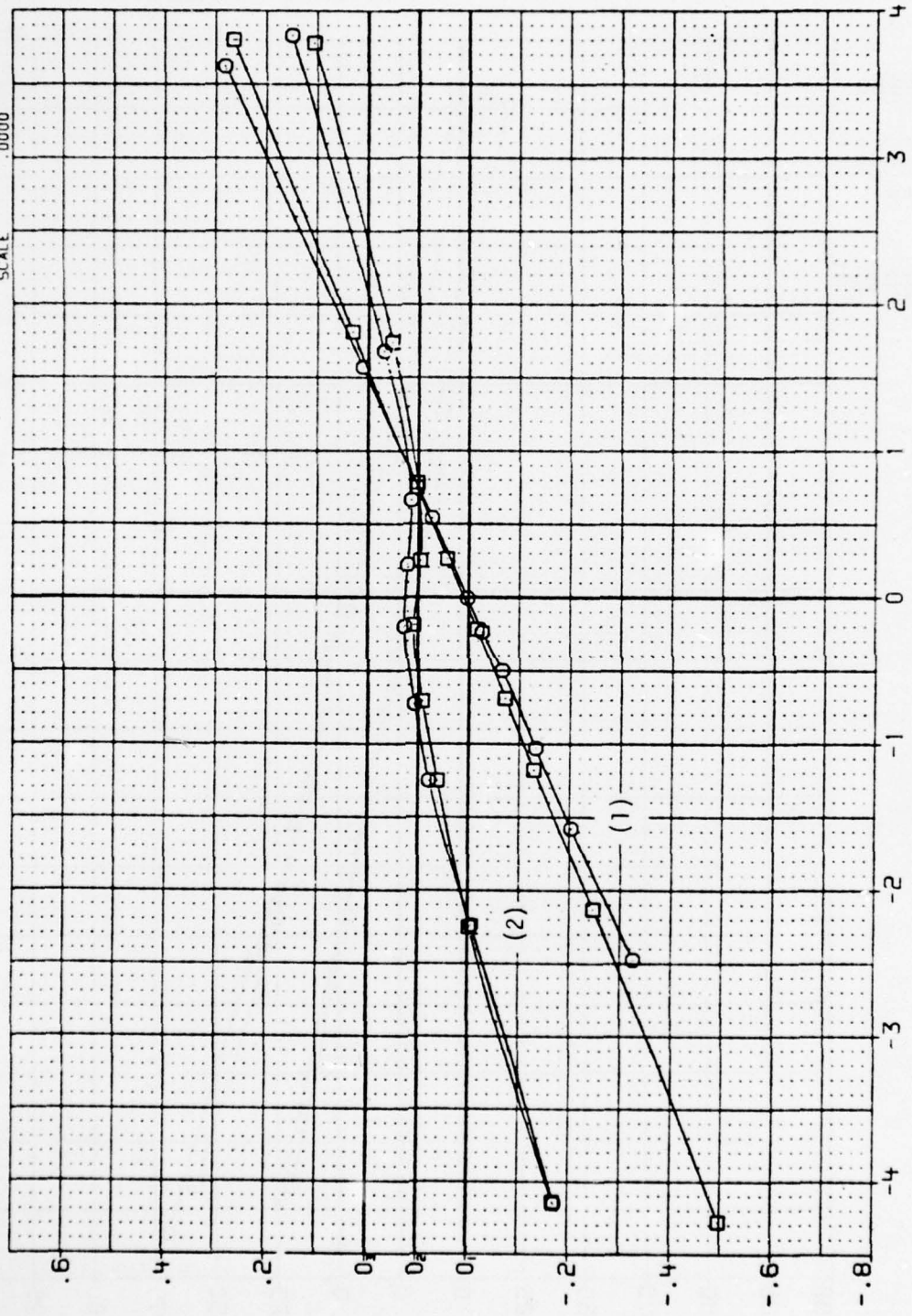
EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .03 (2) 5.93

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (GX8019) ○ CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION
 (DX8031) □ CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION

GAP MACH PHI
 .000 1.200 45.000
 .250 1.200 45.000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XPRP 26.5000 IN.
 YPRP .0000 IN.
 ZPRP .0000 IN.
 SCALE .0000



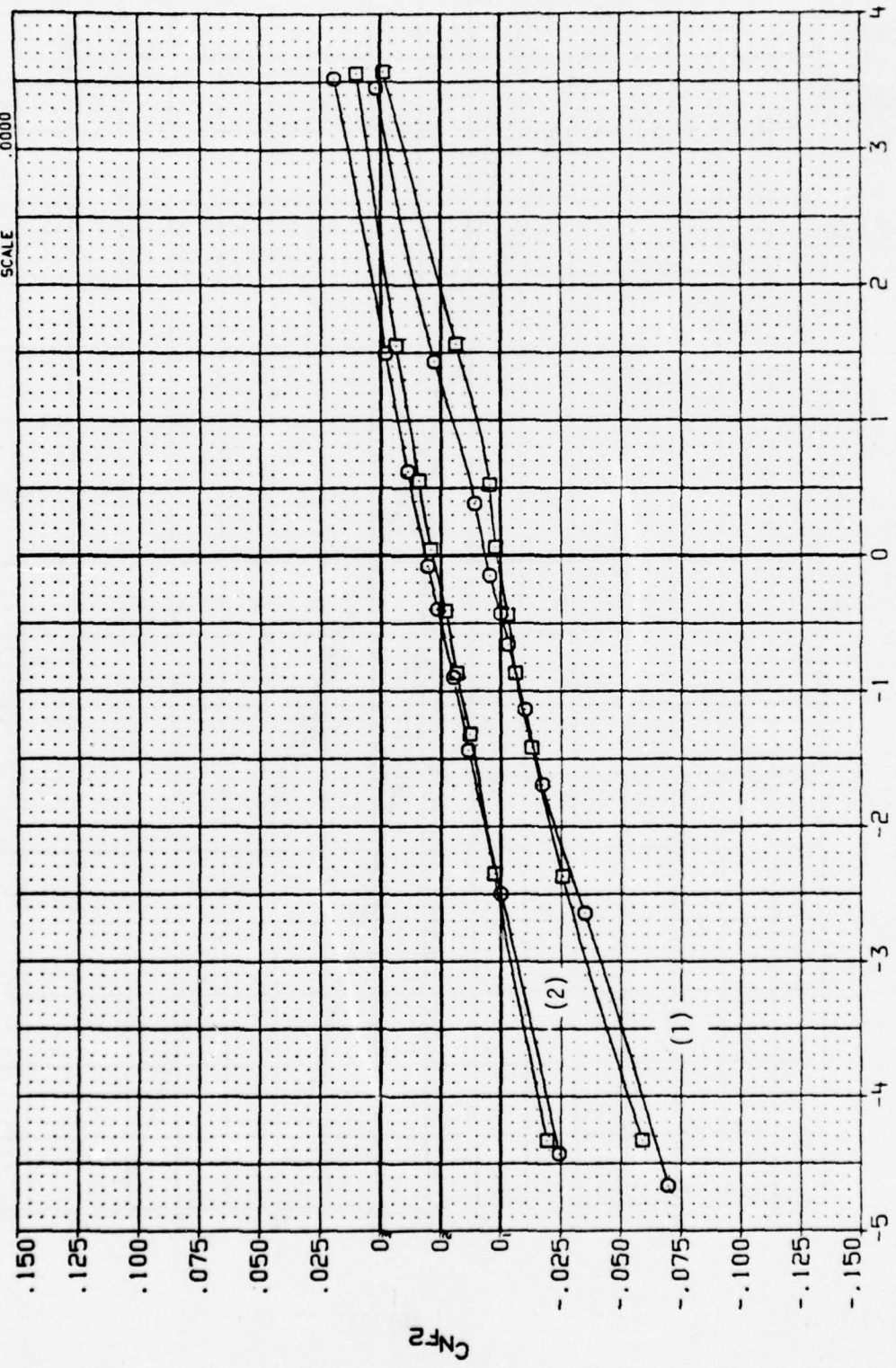
EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .01 (2) 5.96

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (EX8017) ○ CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION
 (DX8029) □ CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION

GAP MACH PHI
 .000 .900 45.000
 .250 .900 45.000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 YMRP 26.5000 IN.
 ZMRP .0000 IN.
 SCALE .0000



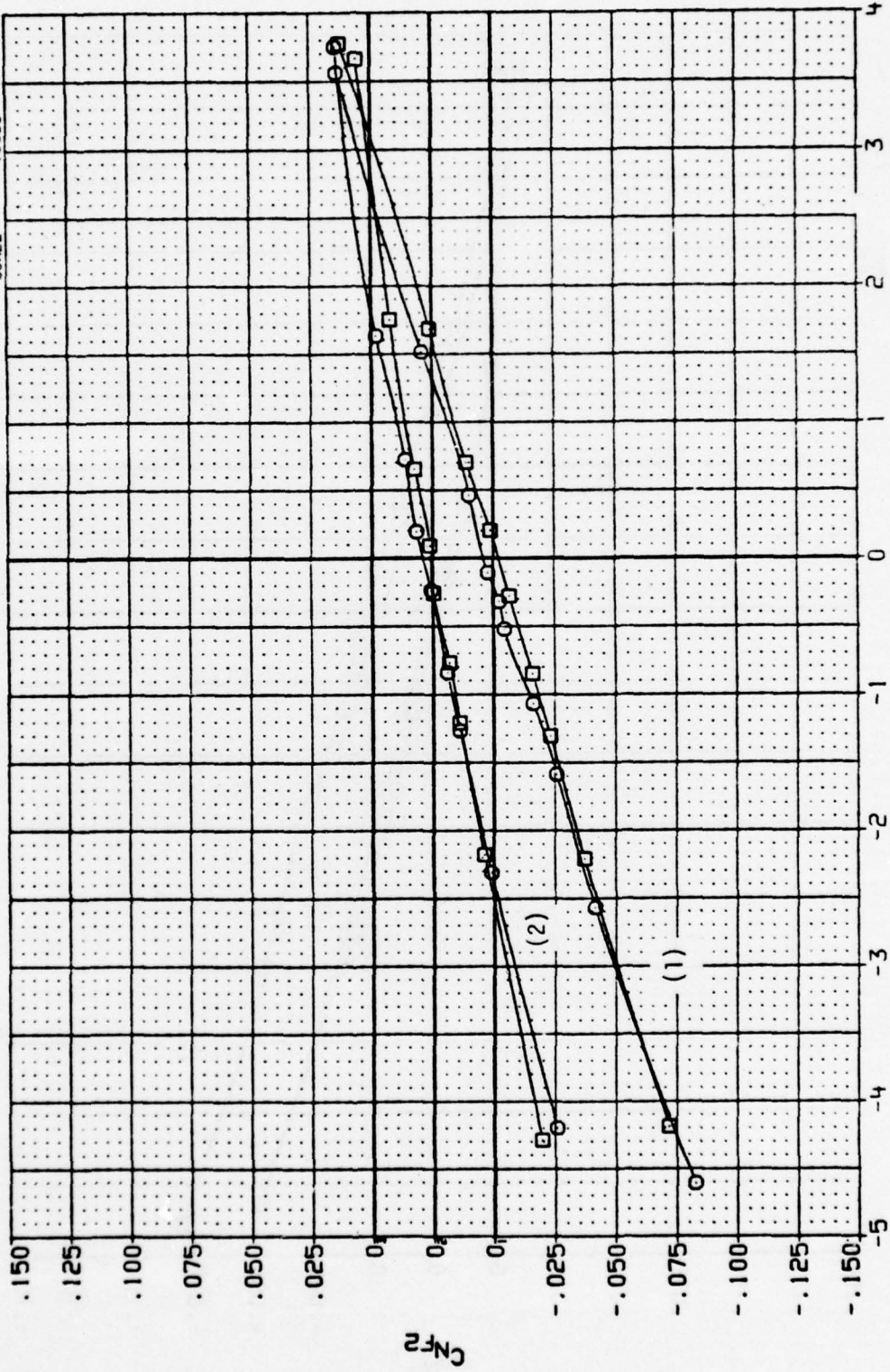
EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .02 (2) 5.92

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (DX8018) ○ CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION
 (DX8030) □ CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION

GAP MACH PHI
 .000 1.000 45.000
 .250 1.000 45.000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BRREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .03 (2) 5.93

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(018018) ○ CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION

(018031) □ CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION

GAP MACH PHI

.000 1.200 45.000

.250 1.200 45.000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

LREF 5.0000 74. IN.

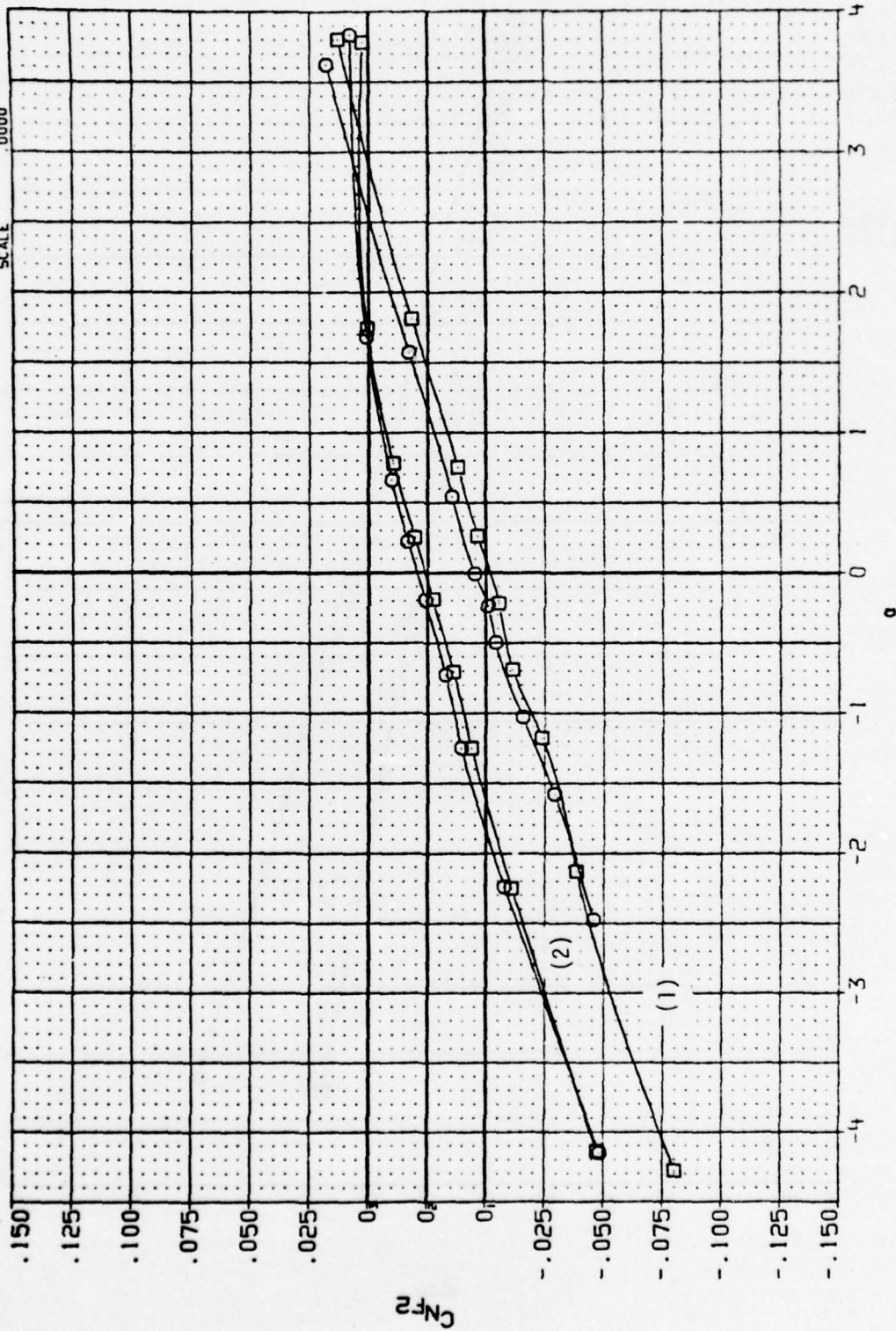
BREF 5.0000 11. IN.

XMRP 26.5000 11. IN.

YMRP .0000 11. IN.

ZMRP .0000 11. IN.

SCALE .0000



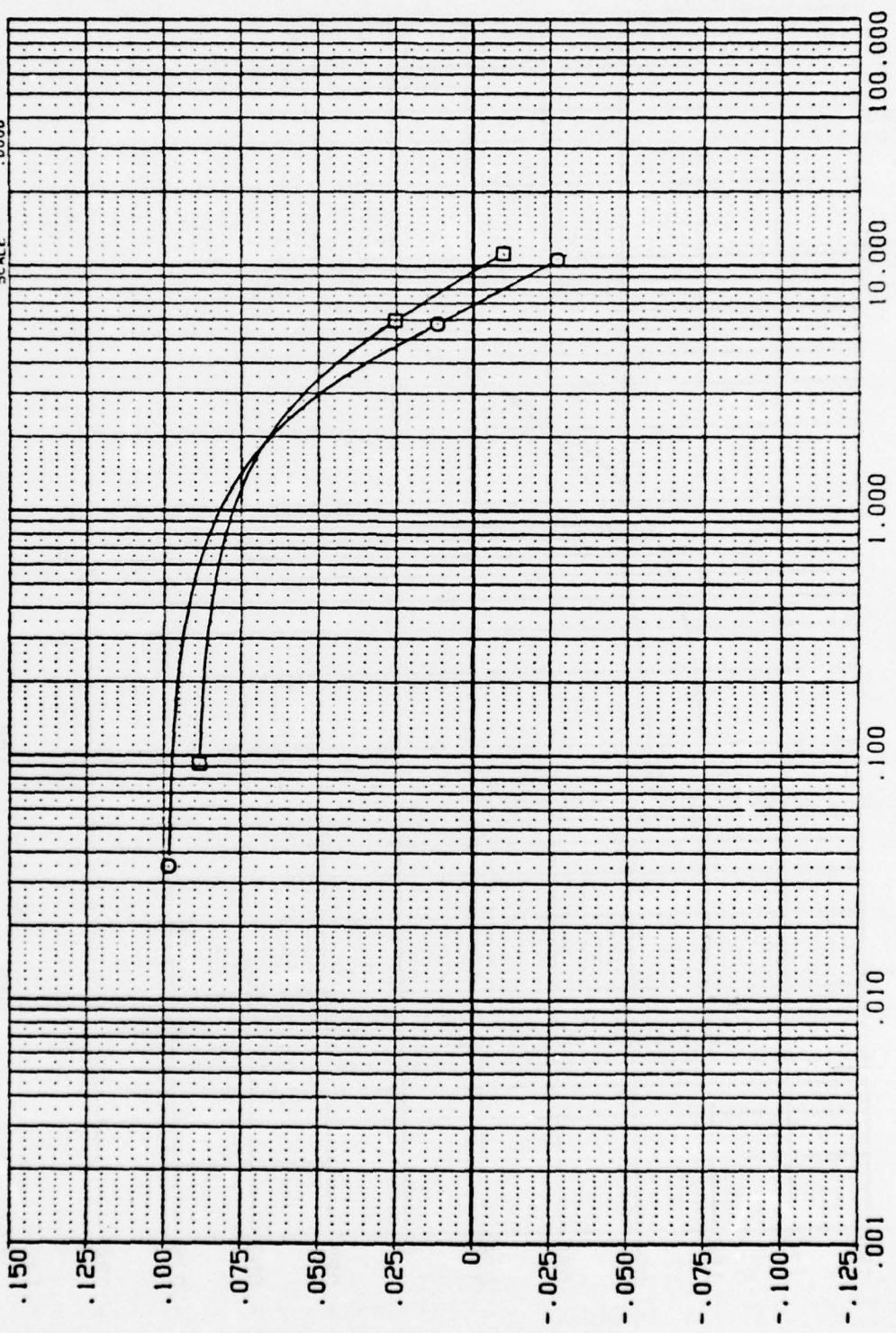
EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

(1) CRT = .01 (2) 5.96

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (EX80671) ○ CAL 117-123 (BF2) BODY, FIN IN AFT POSITION
 (DX80631) □ CAL 117-123 (BF4) BODY, FIN IN AFT POSITION

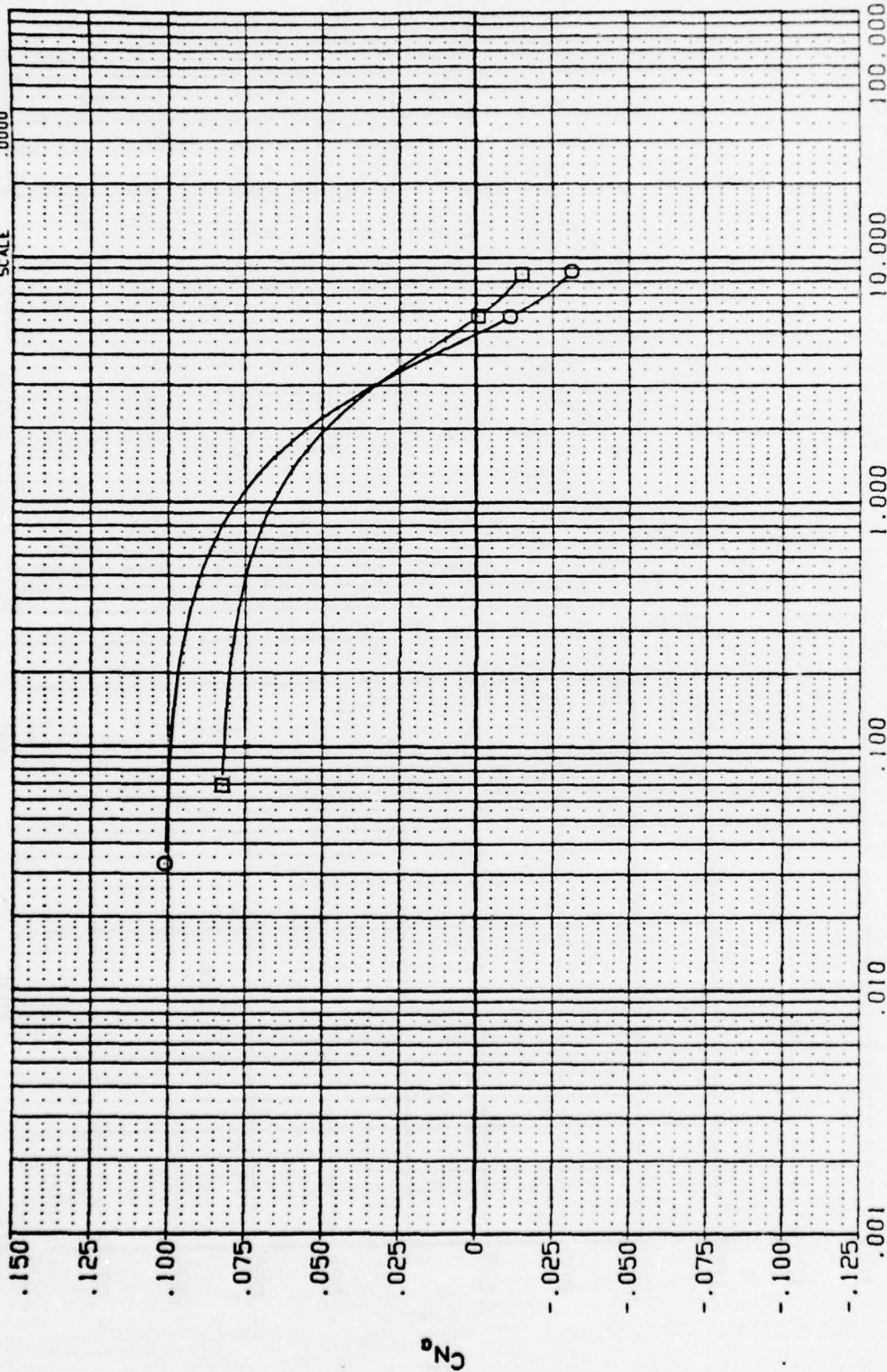
GAP MACH PHI
 .000 .700 .000
 .250 .700 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS
CRT

| | | | | | |
|------------------------|---|------------|-------------|------------|------------------------------|
| DATA SET SYMBOL | CONFIGURATION DESCRIPTION | GAP | MACH | PHI | REFERENCE INFORMATION |
| (EX8058) | CAL 117-123 (BF2) BODY, FIN IN AFT POSITION | .000 | .900 | .000 | SREF 19.6350 IN. 50. IN. |
| (DX8064) | CAL 117-123 (BF4) BODY, FIN IN AFT POSITION | .250 | .900 | .000 | LREF 5.0000 IN. |
| | | | | | BREF 5.0000 IN. |
| | | | | | XMRP 26.5000 IN. |
| | | | | | YMRP .0000 IN. |
| | | | | | ZMRP .0000 IN. |
| | | | | | SCALE .0000 |

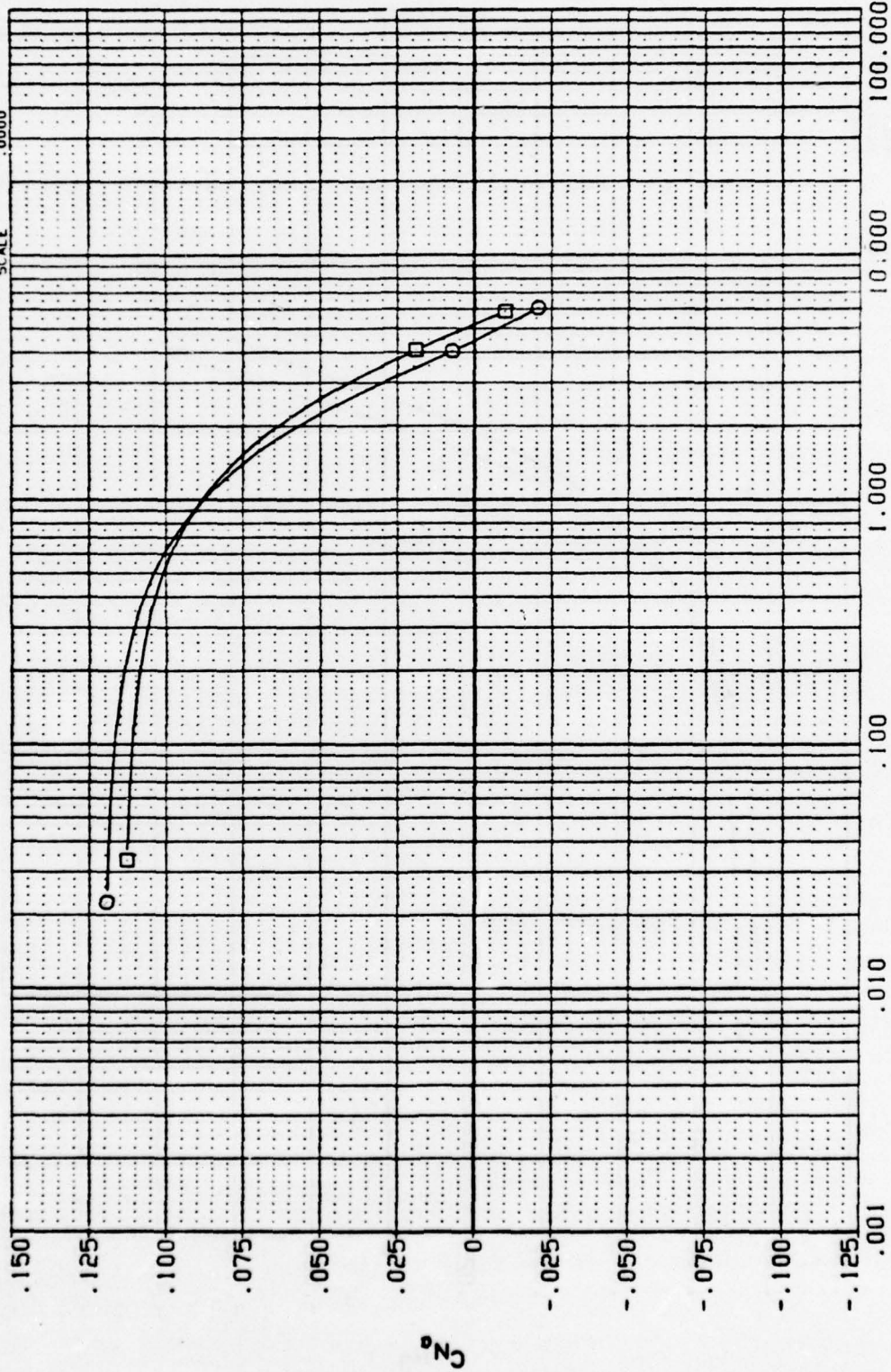


EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS
CRT

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (EX8069) ○ CAL 117-123 (B*2) BODY, FIN IN AFT POSITION
 (DX8065) □ CAL 117-123 (B*4) BODY, FIN IN AFT POSITION

GAP MACH PHI
 .000 1.000 .000
 .250 1.000 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



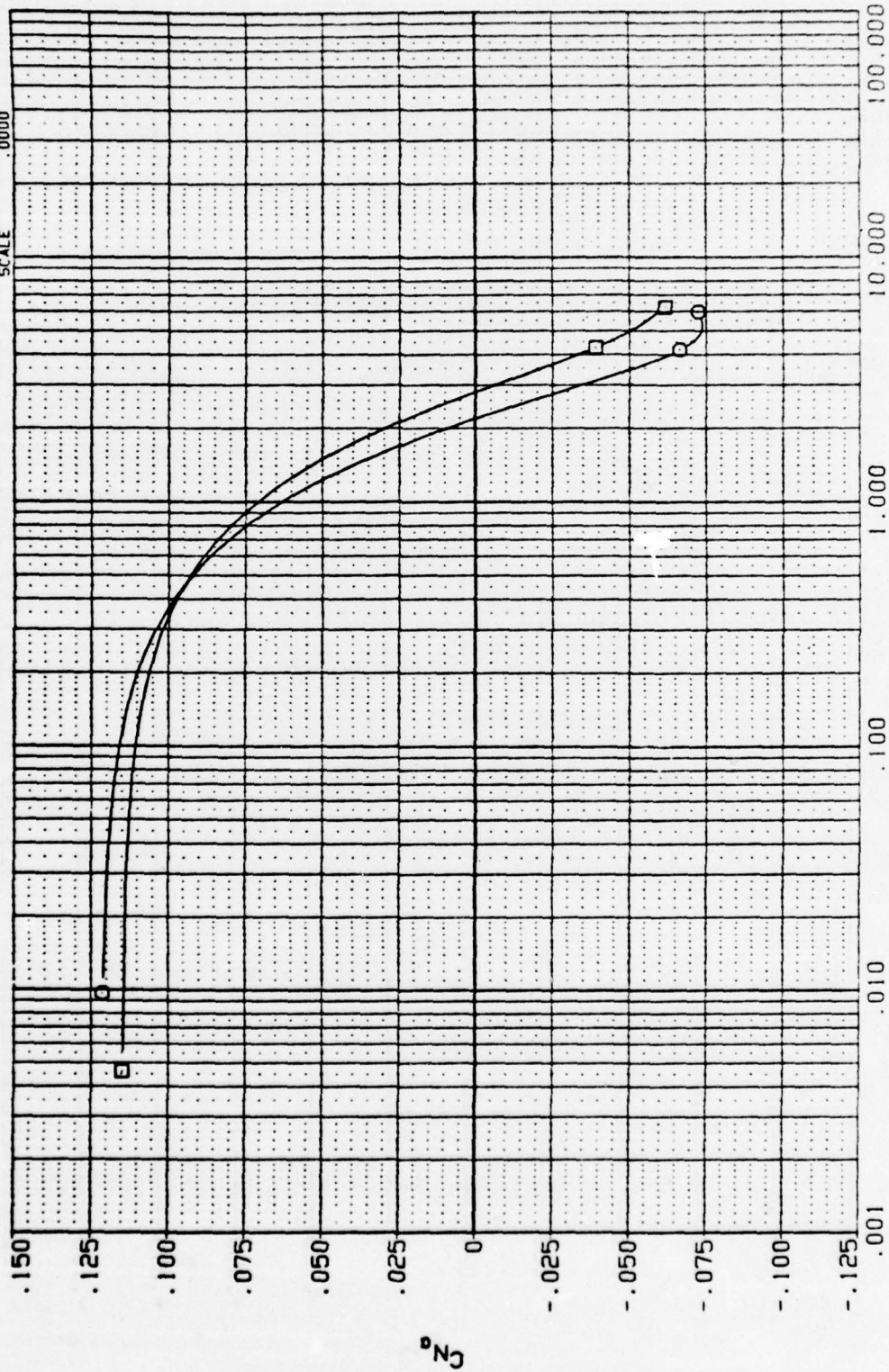
EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS
 CRT

C_n

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (EX8070) CAL 117-123 (BF2) BODY, FIN IN AFT POSITION
 (DX8066) CAL 117-123 (BF4) BODY, FIN IN AFT POSITION

GAP MACH PHI
 .000 1.200 .000
 .250 1.200 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000

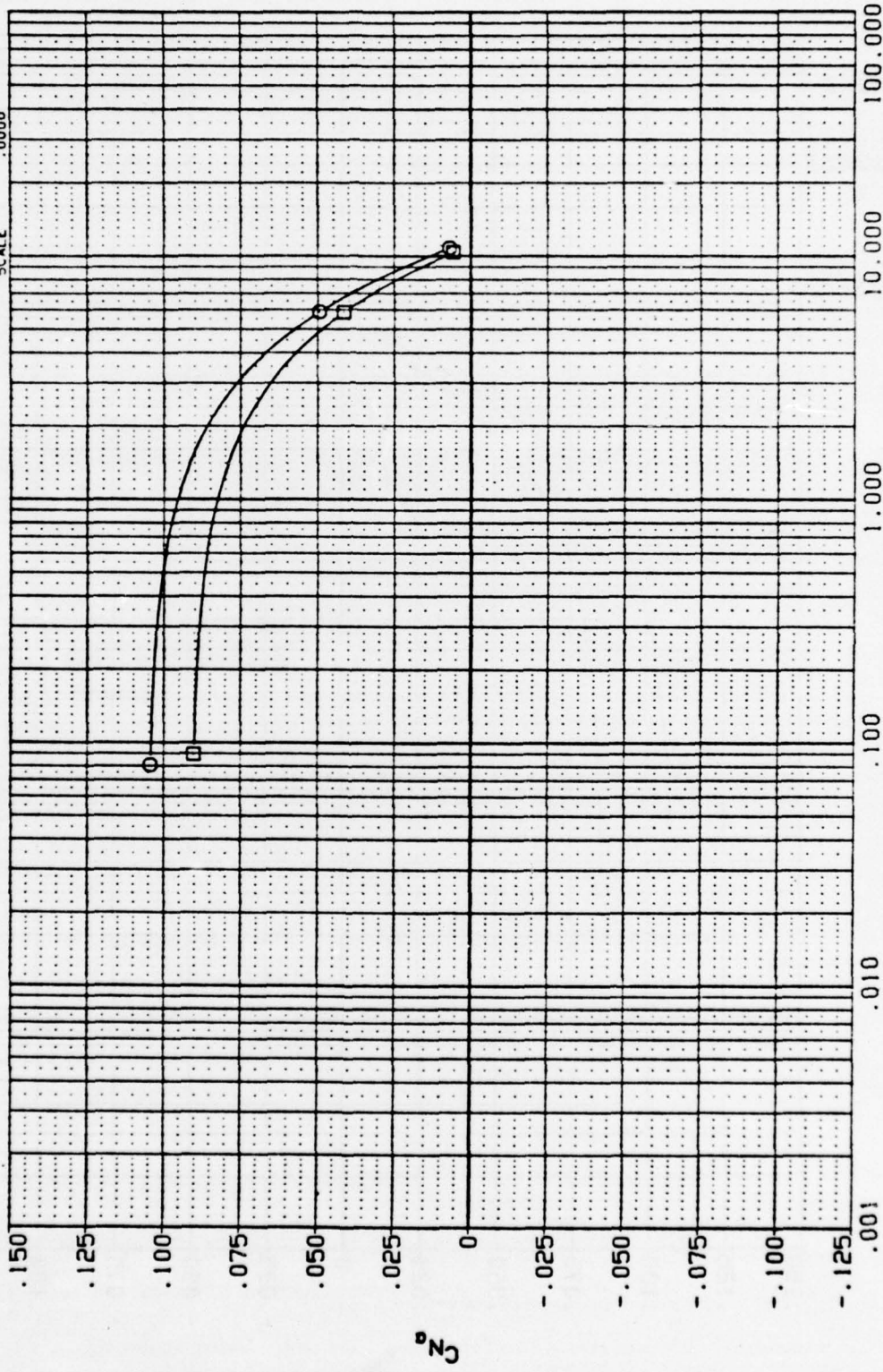


EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

DATA SET SYMBOL (DX8045) \circ CAL 117-123 (BF2) BODY, FIN IN MID POSITION
 (DX8049) \square CAL 117-123 (BF4) BODY, FIN IN MID POSITION

GAP .000 MACH .700 PHI .000
 .250 .700 .000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

Cn

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(DX8046) ○ CAL 117-123 (BF2) BODY, FIN IN MID POSITION

(DX8050) □ CAL 117-123 (BF4) BODY, FIN IN MID POSITION

GAP MACH PHI

.000 .900 .000

.250 .900 .000

REFERENCE INFORMATION

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LREF 5.0000 IN.

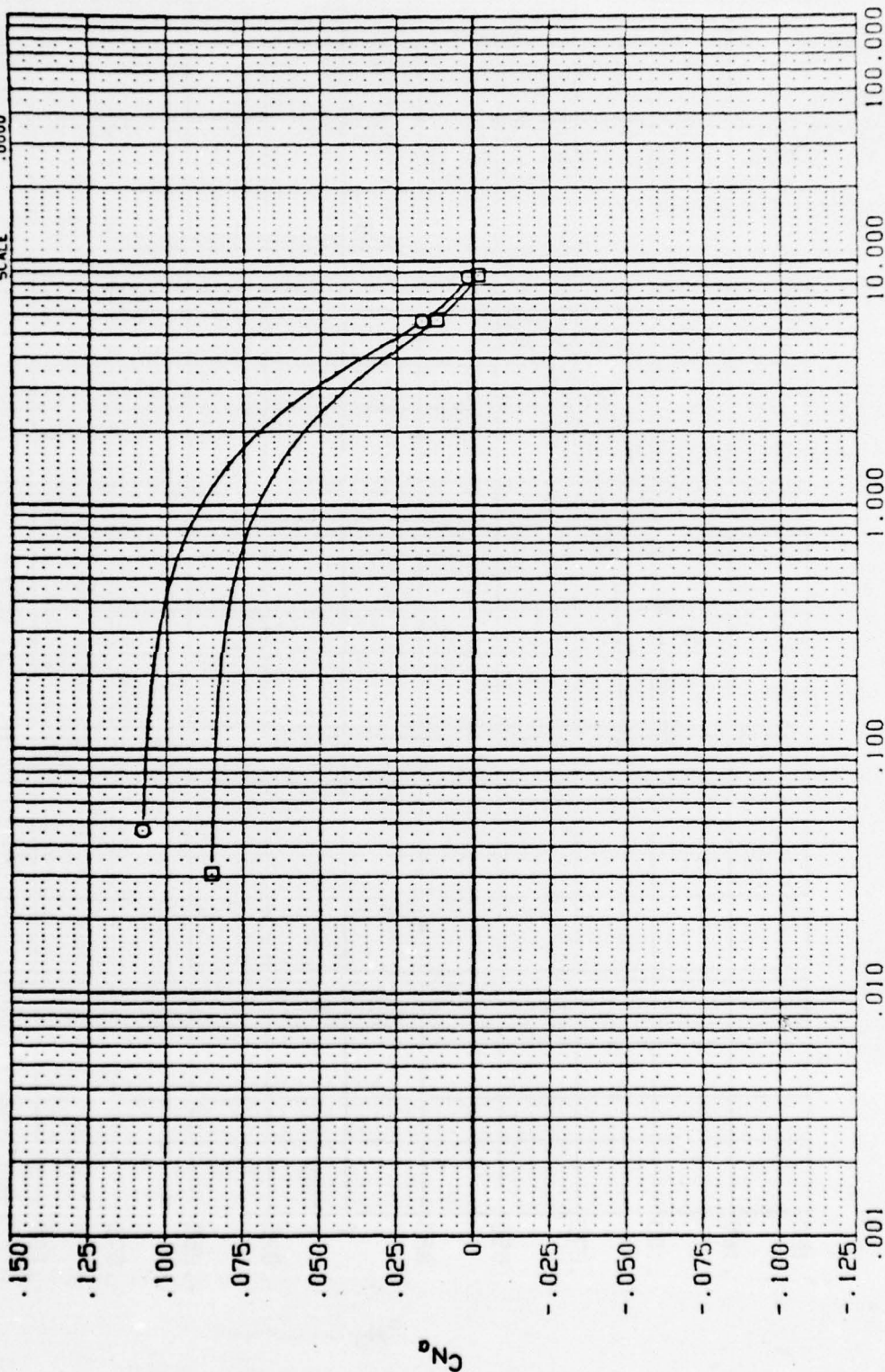
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YMRP .0000 IN.

ZMRP .0000 IN.

SCALE .0000

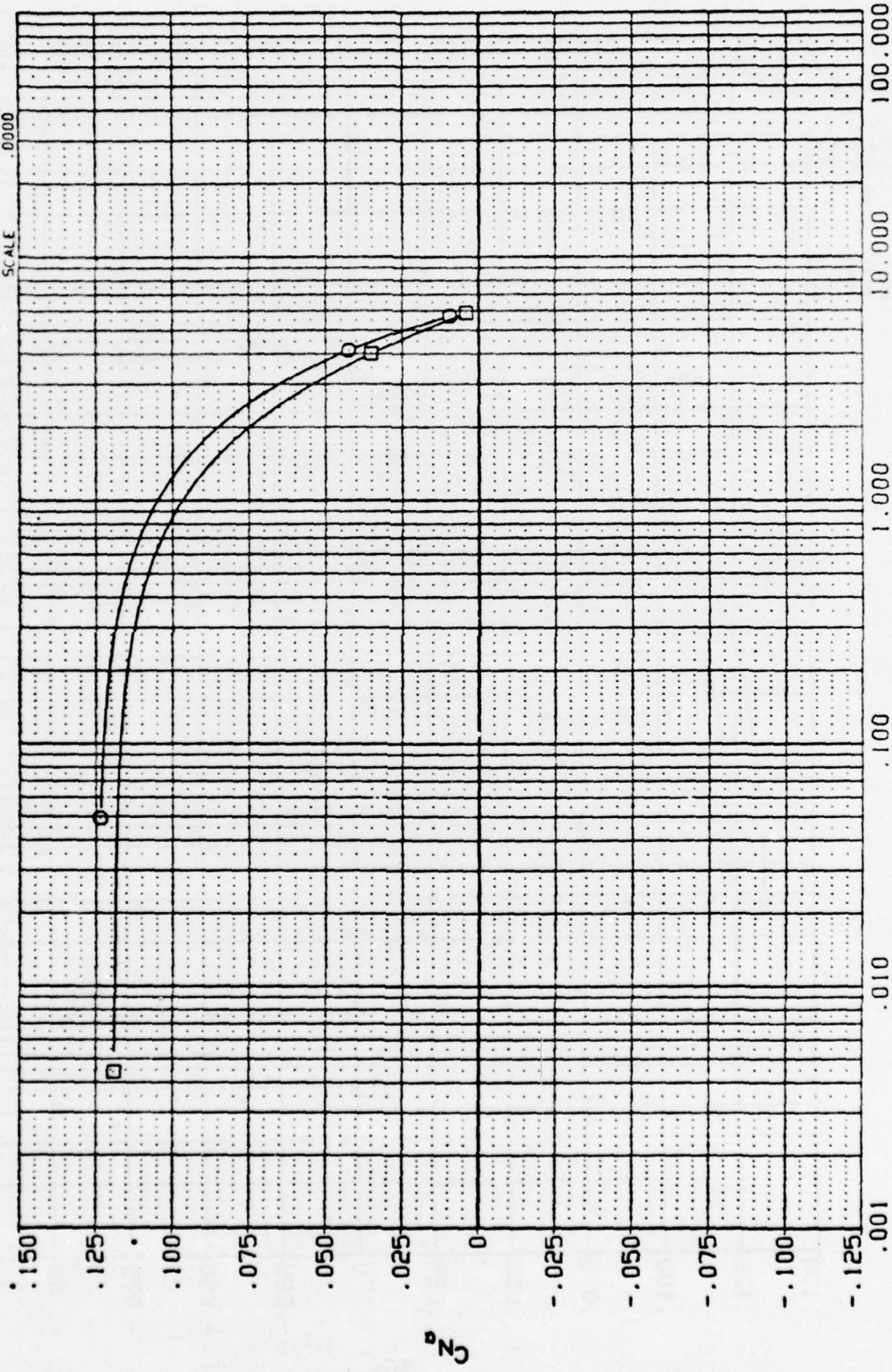


EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (DX8047) ○ CAL 117-123 (BF2) BODY, FIN IN MID POSITION
 (DX8051) □ CAL 117-123 (BF4) BODY, FIN IN MID POSITION

GAP MACH PHI
 .000 1.000 .000
 .250 1.000 .000

REFERENCE INFORMATION
 SPREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BRREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(DX8048) ○ CAL 117-123 (BF2) BODY, FIN IN MID POSITION

(DX8052) □ CAL 117-123 (BF4) BODY, FIN IN MID POSITION

GAP MACH PHI

.000 1.200 .000

.250 1.200 .000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

LREF 5.0000 IN.

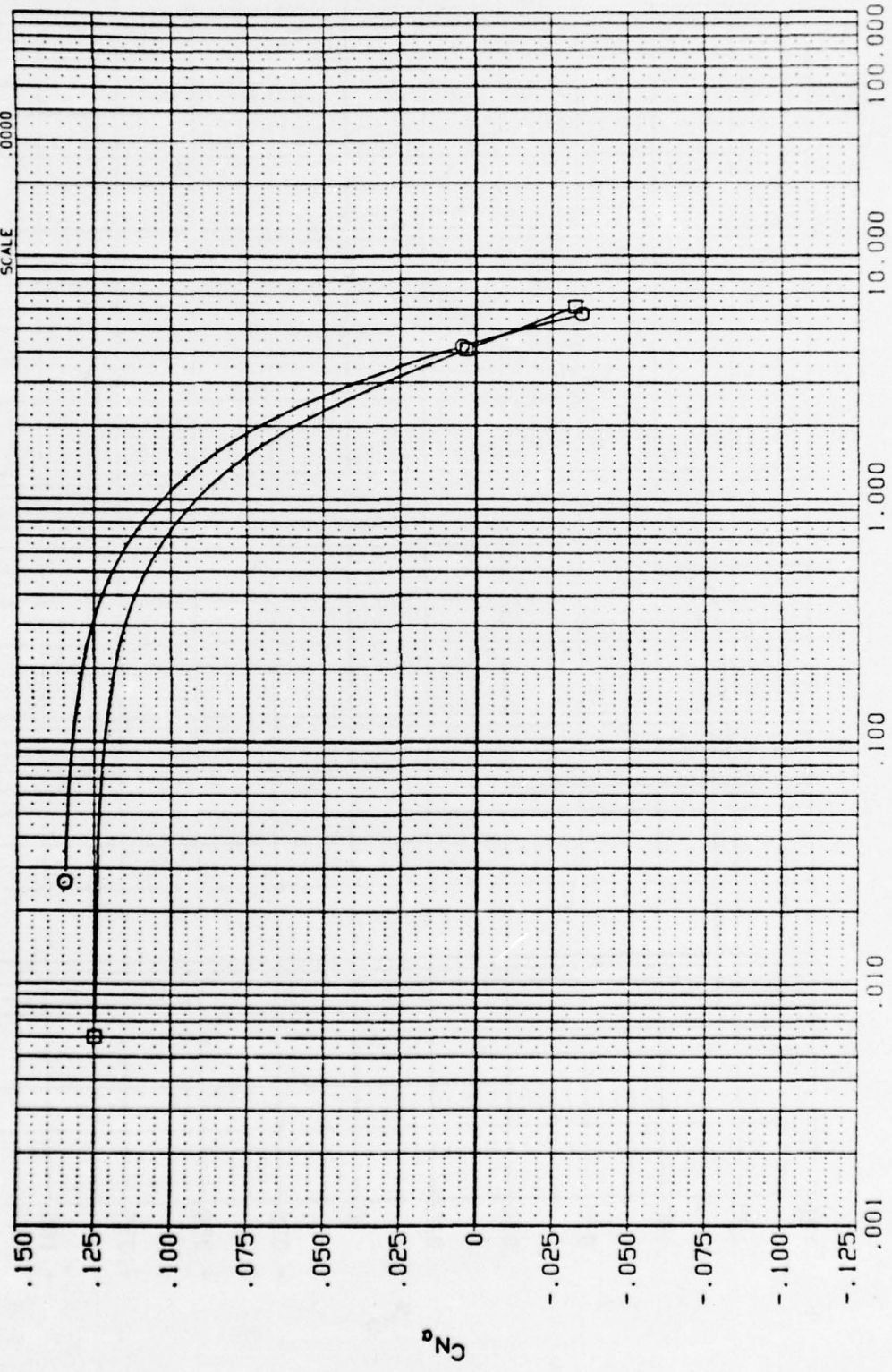
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XMRP 26.5000 IN.

YMRP .0000 IN.

ZMRP .0000 IN.

SCALE .0000



EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

DATA SET SYMBOL
 (EX8014)
 (DX8026)

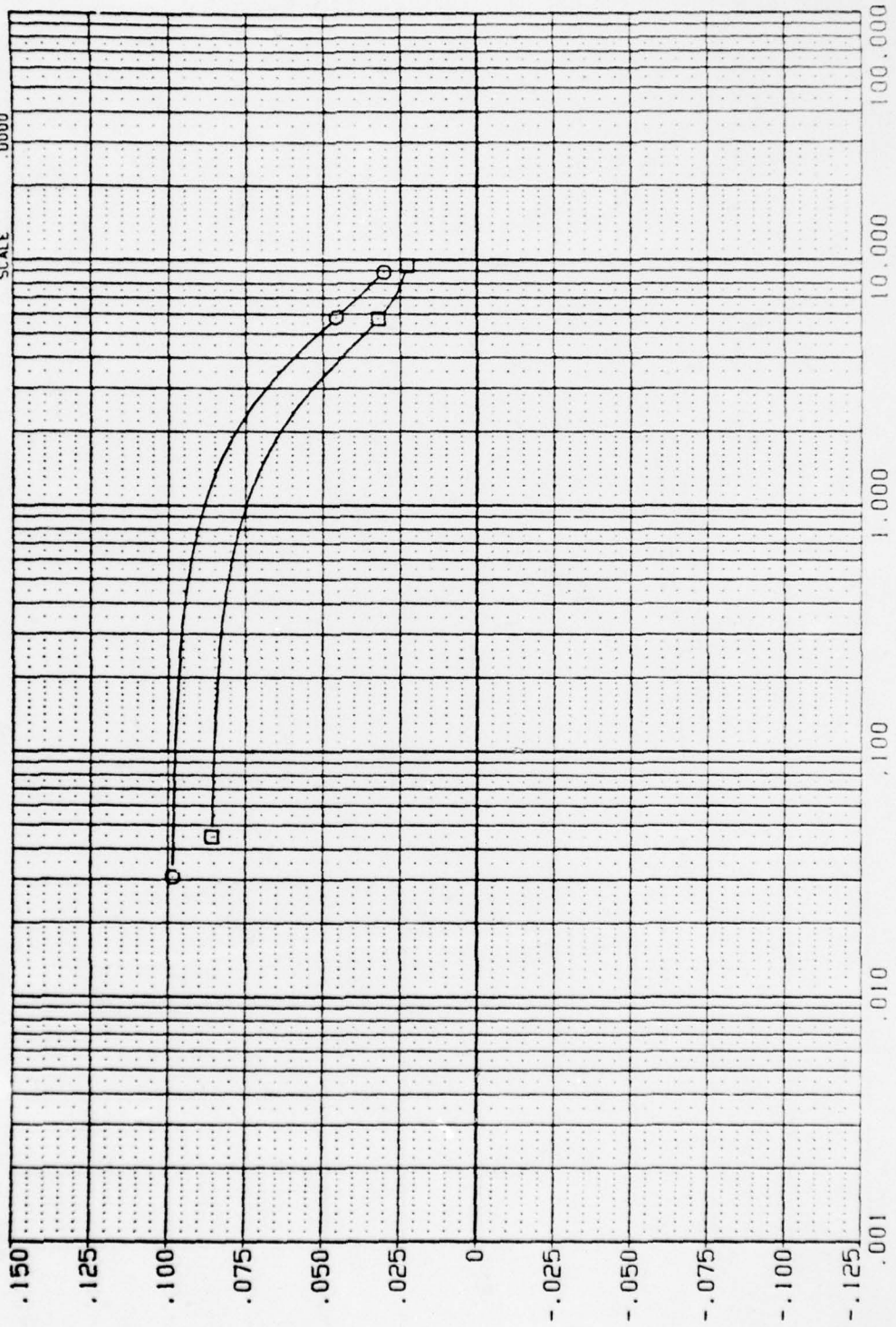
CONFIGURATION DESCRIPTION
 CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION
 CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION

GAP
 .000
 .250

MACH
 .900
 .900

PHI
 .000
 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

DATA SET SYMBOL
 (EXB015)
 (DXB027)

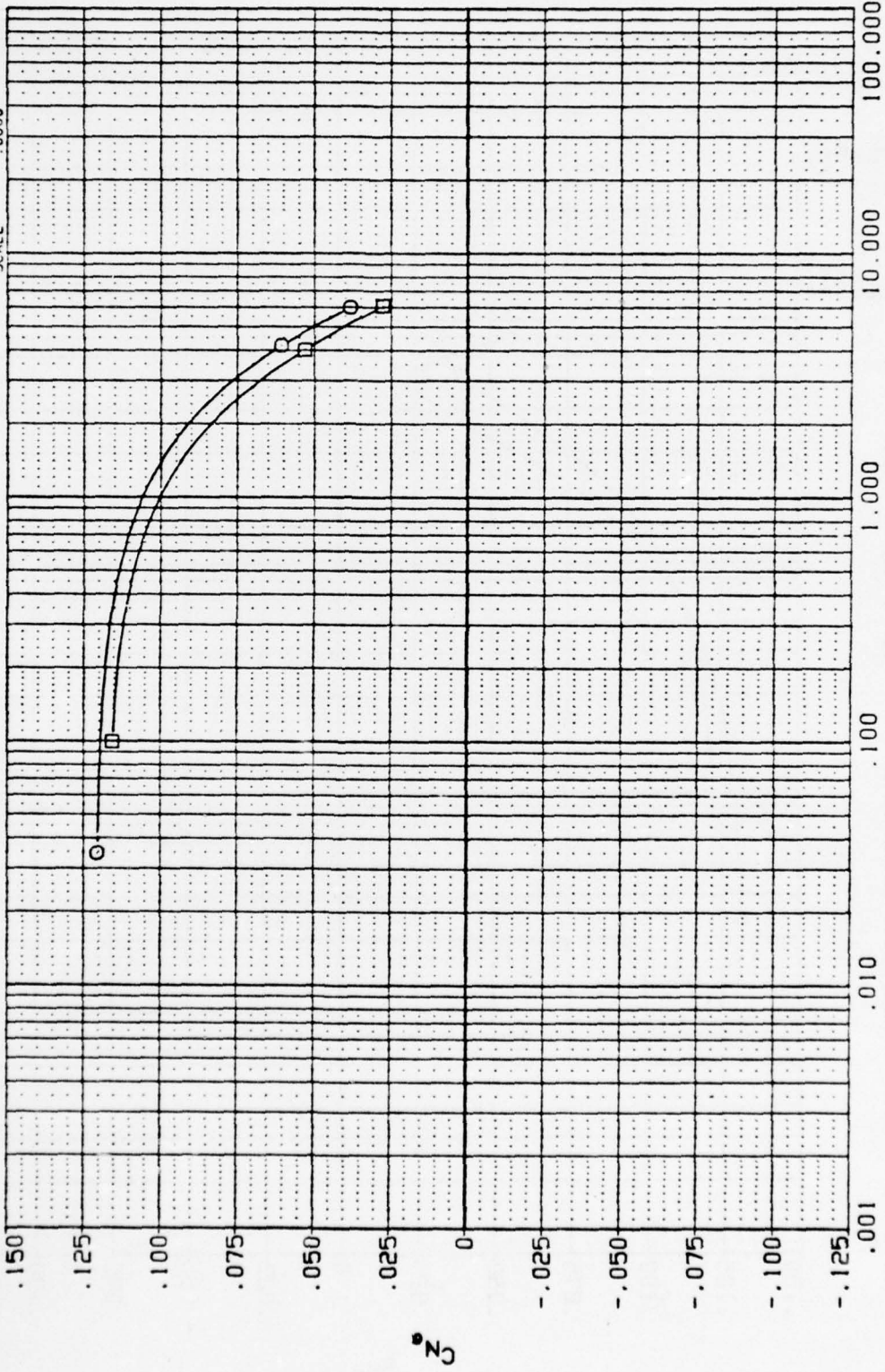
CONFIGURATION DESCRIPTION
 CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION
 CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000

GAP .000
 .250

MACH 1.000
 1.000

PHI .000
 .000

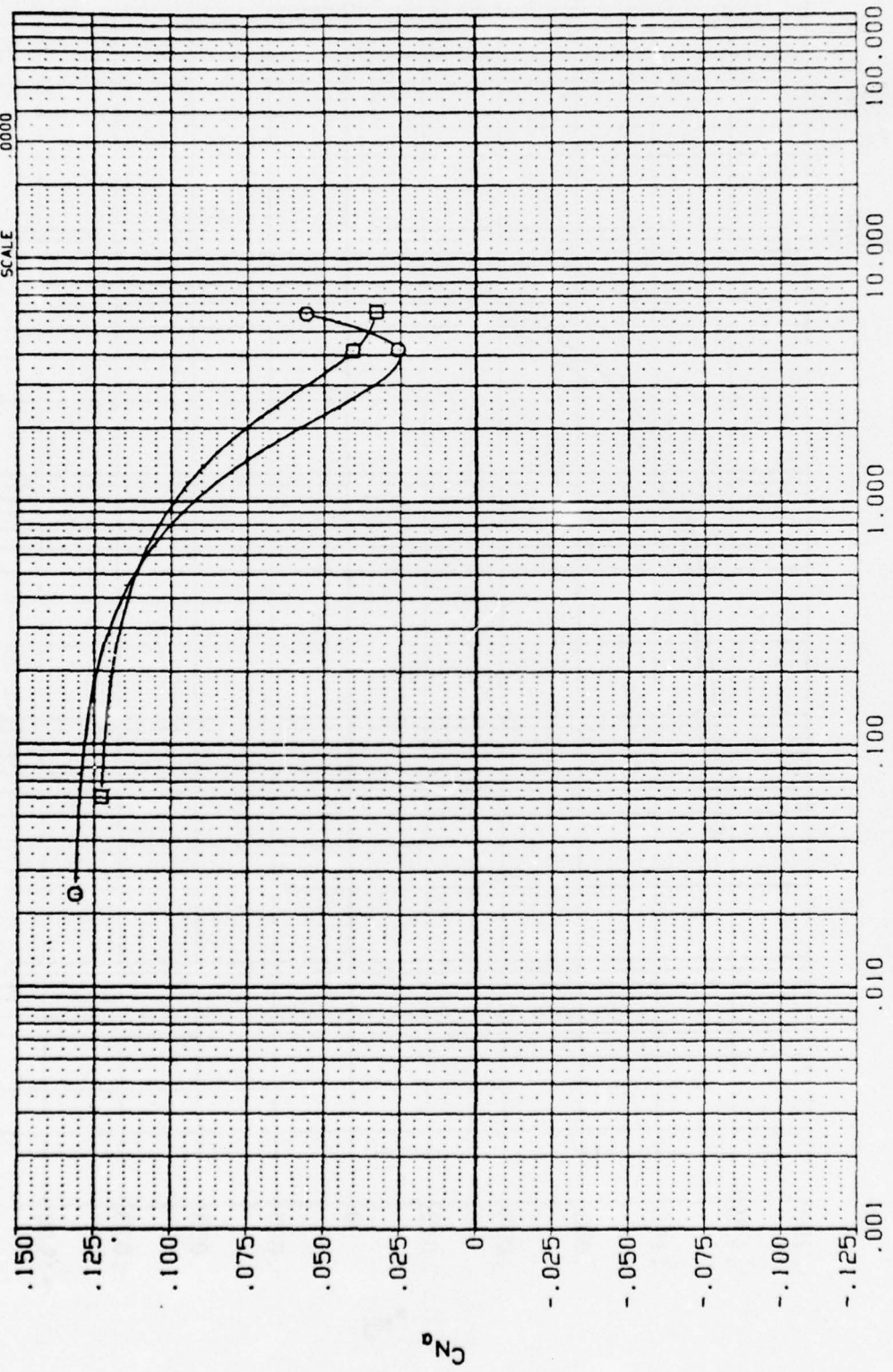


EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (EX8016) ○ CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION
 (DX8028) □ CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION

GAP MACH PHI
 .000 1.200 .000
 .250 1.200 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(EX8017) □ CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION

(DX8029) □ CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION

GAP MACH PHI

.000 .900 45.000

.250 .900 45.000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

LREF 5.0000 IN.

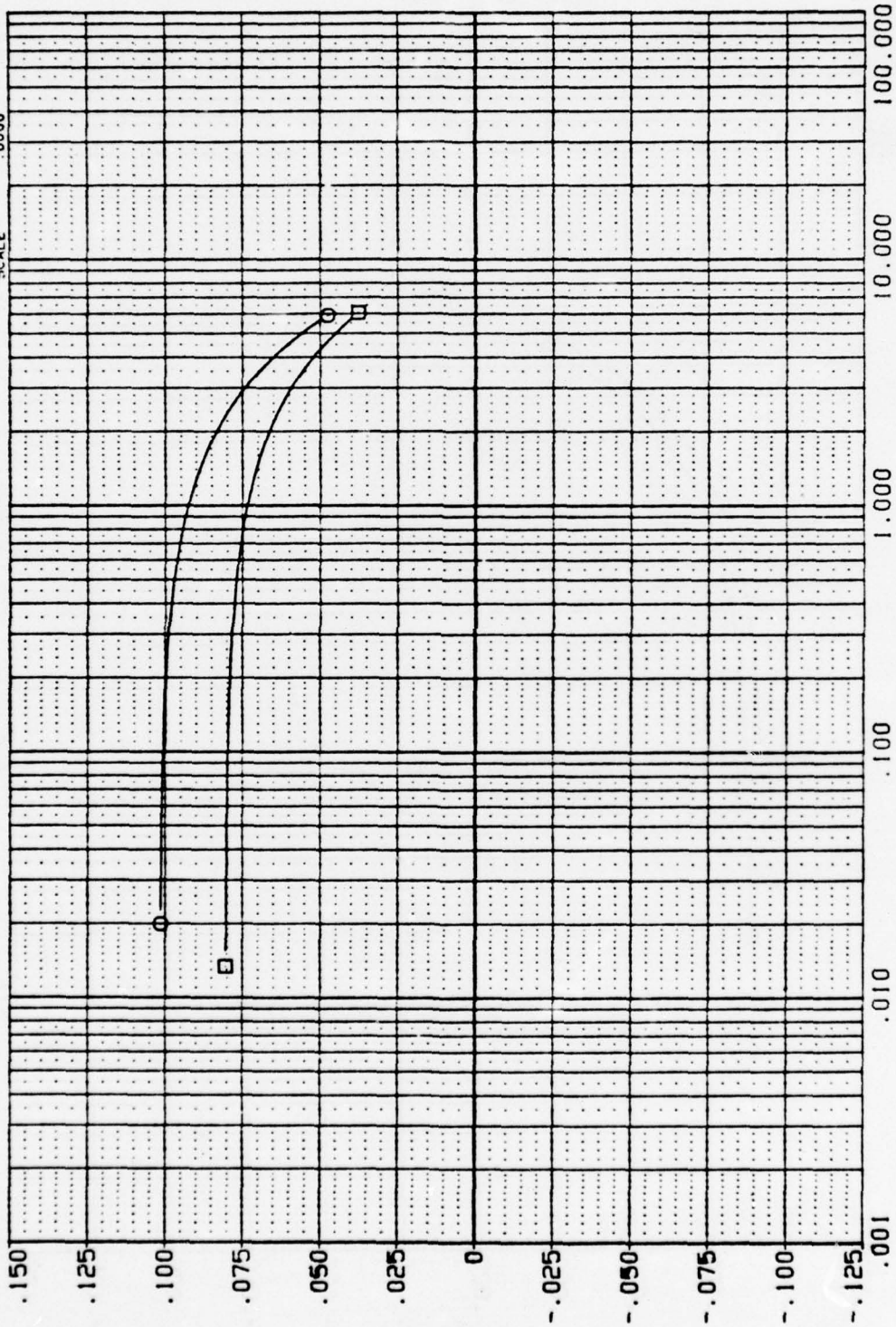
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YMRP .0000 IN.

ZMRP .0000 IN.

SCALE .0000



EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(DX8018) \square CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION

(DX8030) \circ CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION

GAP MACH PHI

.000 1.000 45.000

.250 1.000 45.000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

LREF 5.0000 IN.

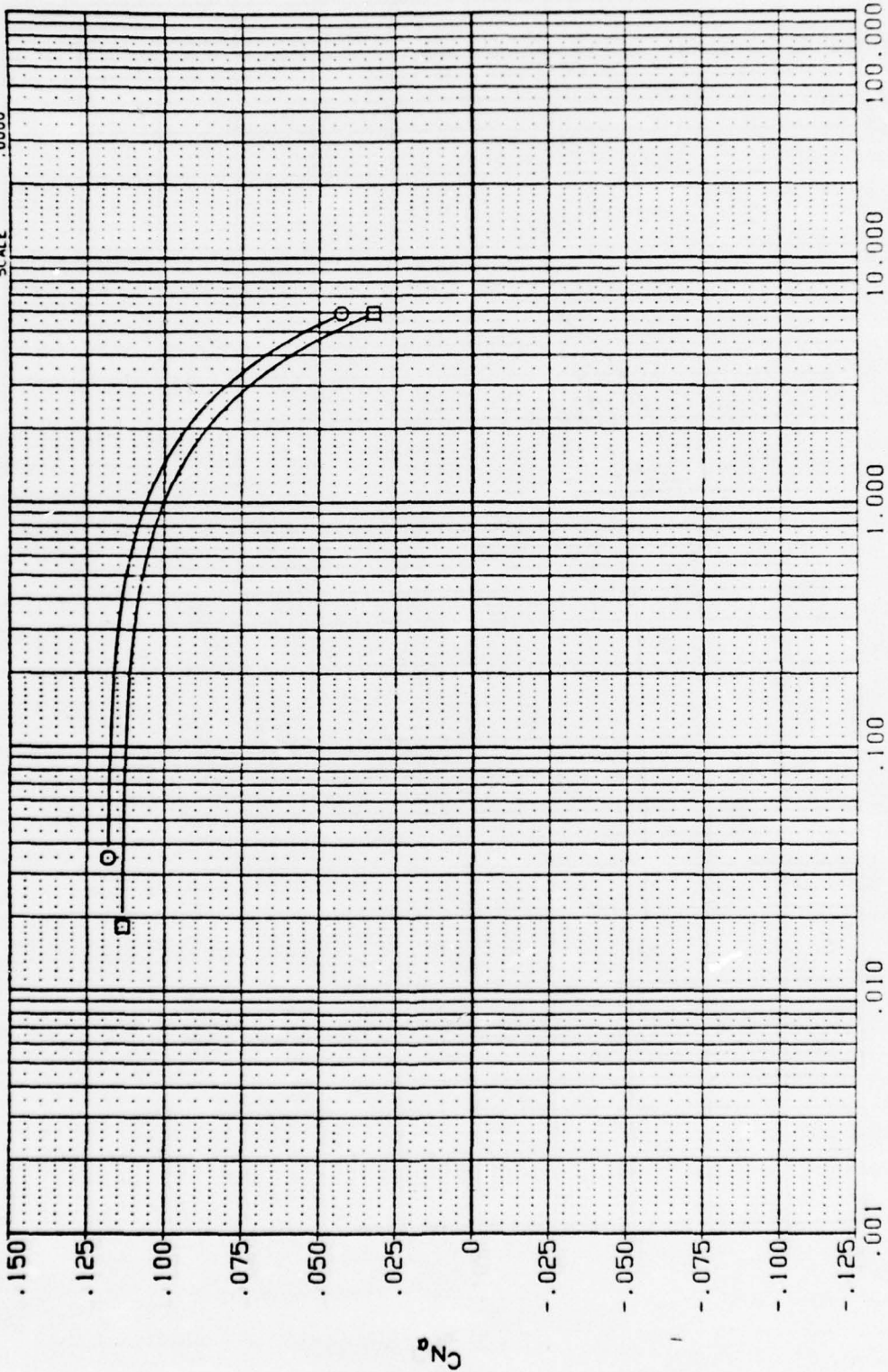
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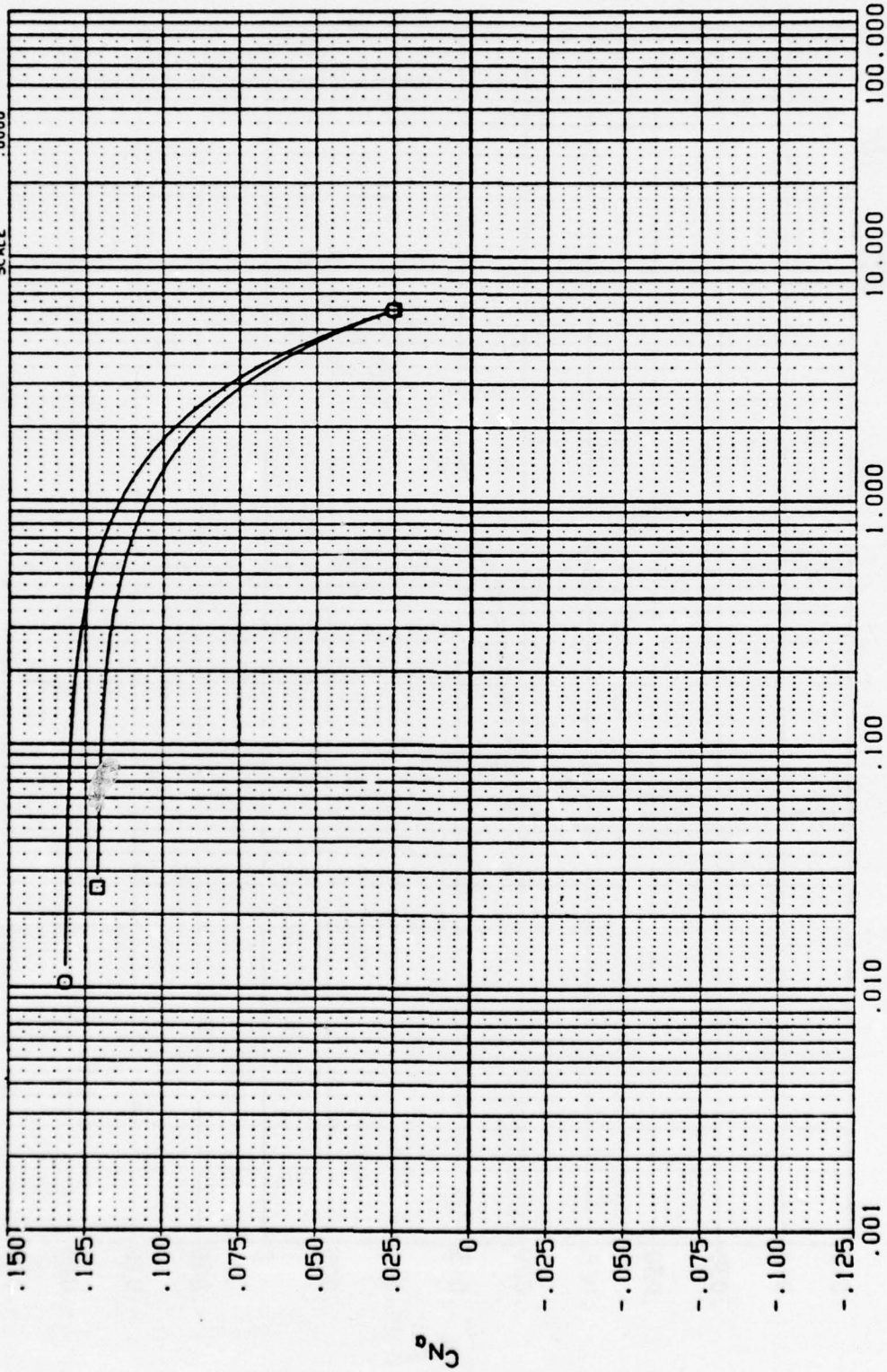
ZMRP .0000 IN.

SCALE .0000



EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

| | | | | | |
|-----------------|---|------|-------|--------|-----------------------|
| DATA SET SYMBOL | CONFIGURATION DESCRIPTION | GAP | MACH | PHI | REFERENCE INFORMATION |
| (GX8019) | CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION | .000 | 1.200 | 45.000 | SREF 19.6350 50. IN. |
| (DX8031) | CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION | .250 | 1.200 | 45.000 | LREF 5.0000 IN. |
| | | | | | BREF 5.0000 IN. |
| | | | | | XMRP 26.5000 IN. |
| | | | | | YMRP .0000 IN. |
| | | | | | ZMRP .0000 IN. |
| | | | | | SCALE .0000 |



EFFECT OF GAP ON NORMAL FORCE AT VARIOUS THRUST LEVELS

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(EX8067) ○ CAL 117-123 (BF2) BODY, FIN IN AFT POSITION

(DX8063) □ CAL 117-123 (BF4) BODY, FIN IN AFT POSITION

GAP MACH PHI

.000 .700 .000

.250 .700 .000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

LREF 5.0000 IN.

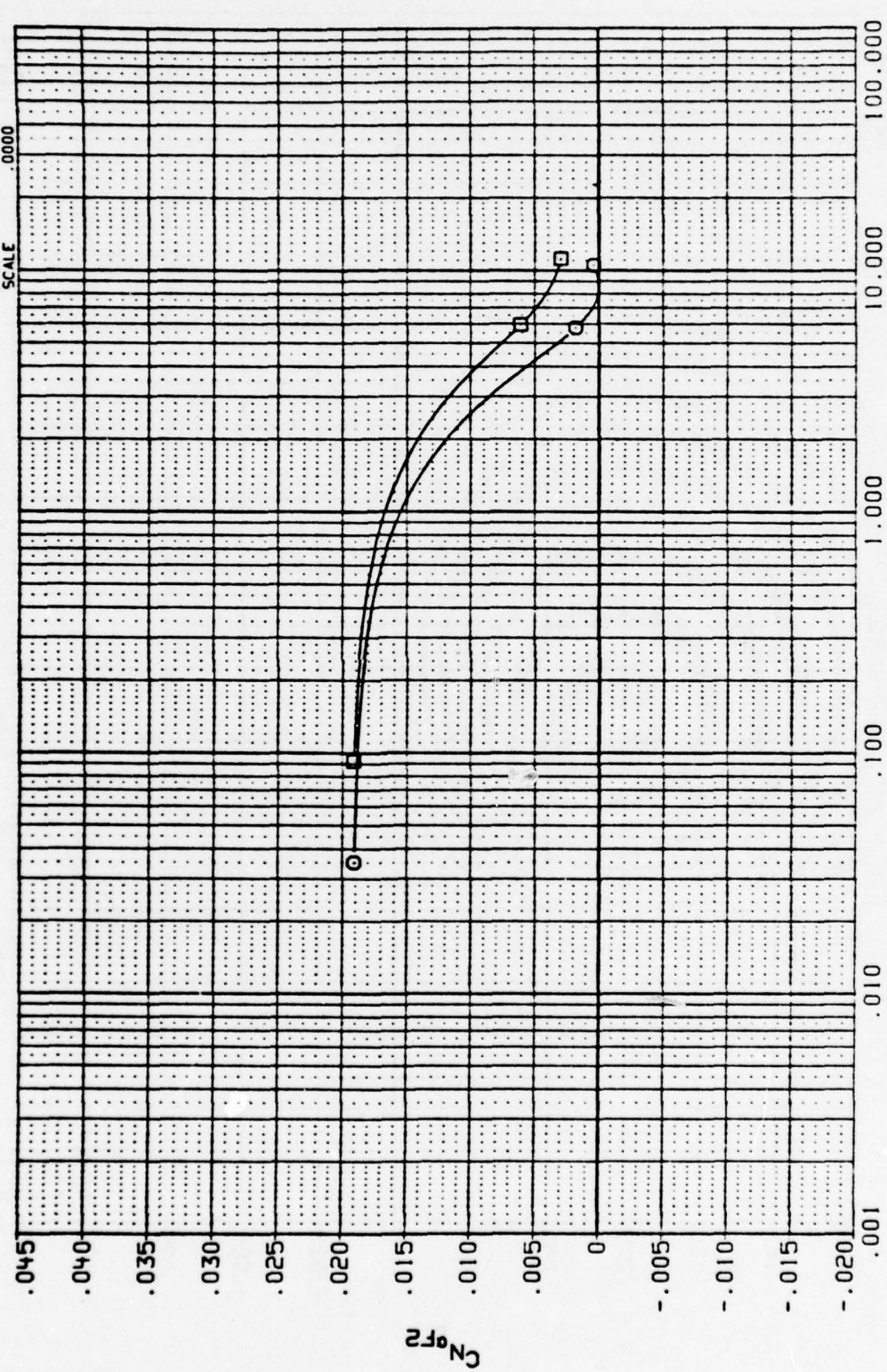
BRF 5.0000 IN.

XMRP 26.5000 IN.

YMRP .0000 IN.

ZMRP .0000 IN.

SCALE .0000



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

Cn^oF₂

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(EX806B) \square CAL 117-123 (BF2) BODY, FIN IN AFT POSITION

(DX806A) \square CAL 117-123 (BF4) BODY, FIN IN AFT POSITION

GAP MACH PHI

.000 .900 .000

.250 .900 .000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

LPREF 5.0000 IN.

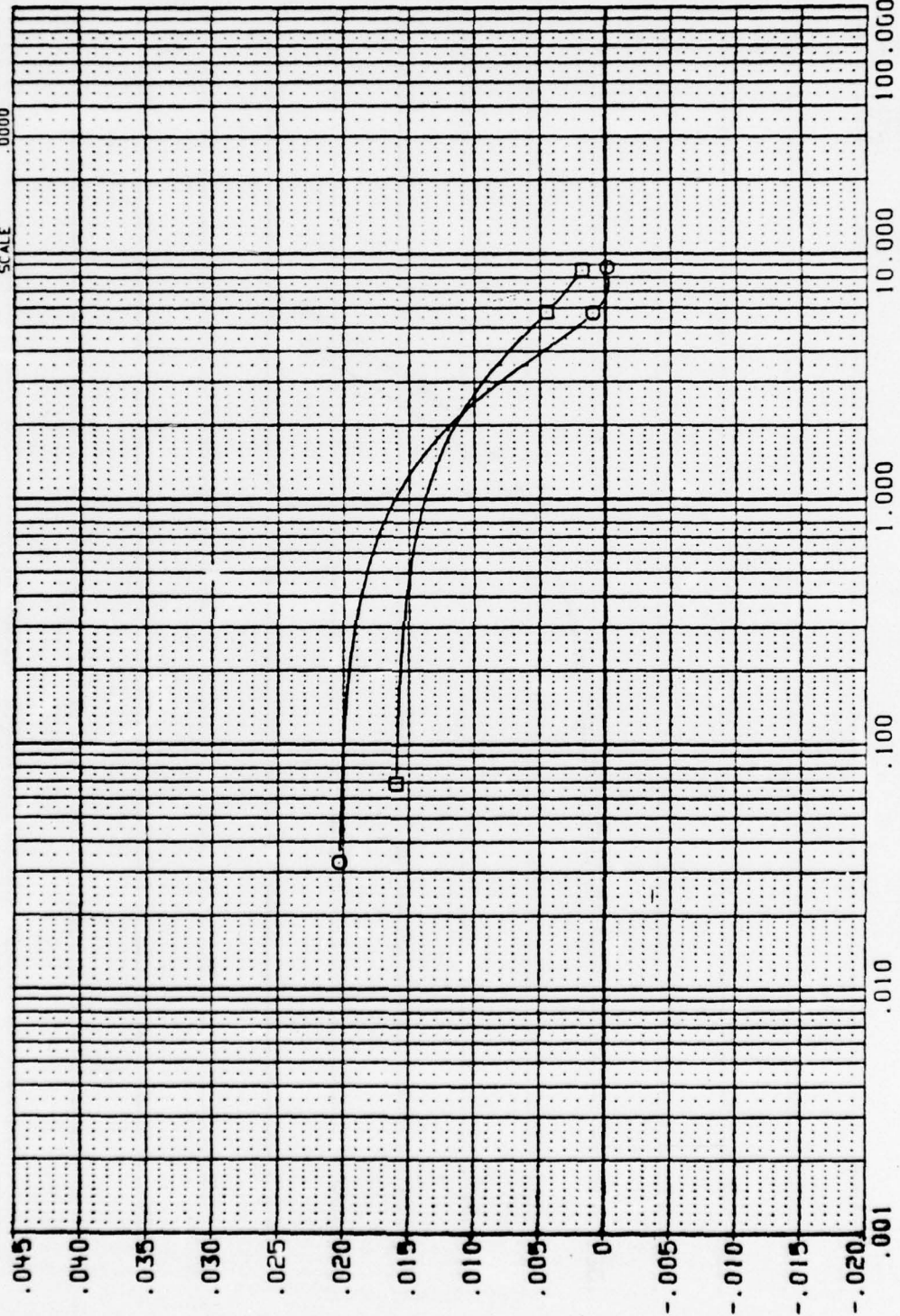
BRFREF 5.0000 IN.

XMRP 26.5000 IN.

YMRP .0000 IN.

ZMRP .0000 IN.

SCALE .0000



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

DATA SET SYMBOL (EX8069) (DX8065)

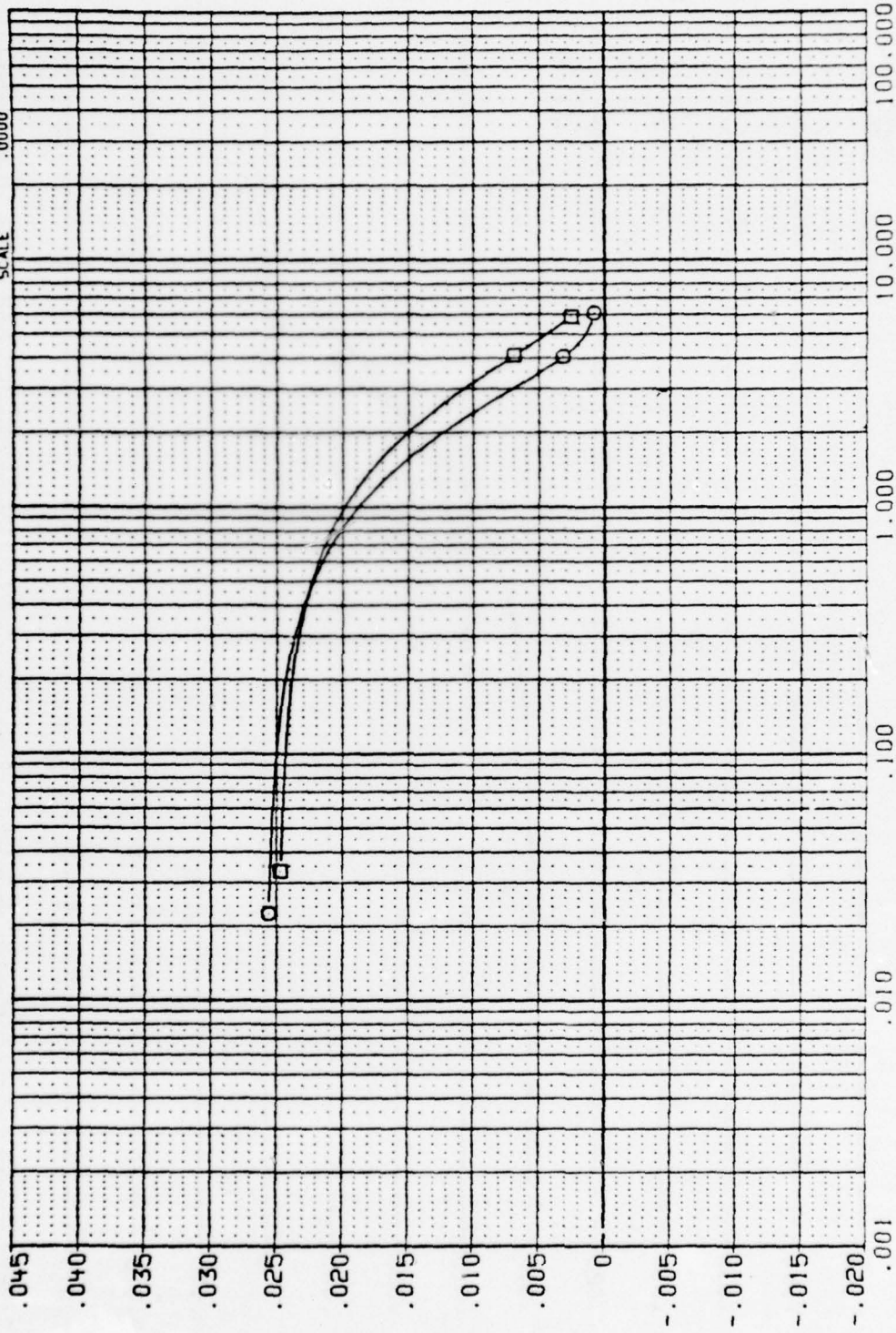
CONFIGURATION DESCRIPTION
 CAL 117-123 (BF2) BODY, FIN IN AFT POSITION
 CAL 117-123 (BF4) BODY, FIN IN AFT POSITION

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BRFP 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000

GAP .000
 .250

MACH 1.000
 1.000

PHI .000
 .000



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS
 CRT

Cn^aF₂

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(EX8070) ○ CAL 117-123 (BF2) BODY, FIN IN AFT POSITION

(DX8066) □ CAL 117-123 (BF4) BODY, FIN IN AFT POSITION

GAP MACH PHI

.000 1.200 .000

.250 1.200 .000

REFERENCE INFORMATION

SREF 19.6350 SQ. IN.

LREF 5.0000 IN.

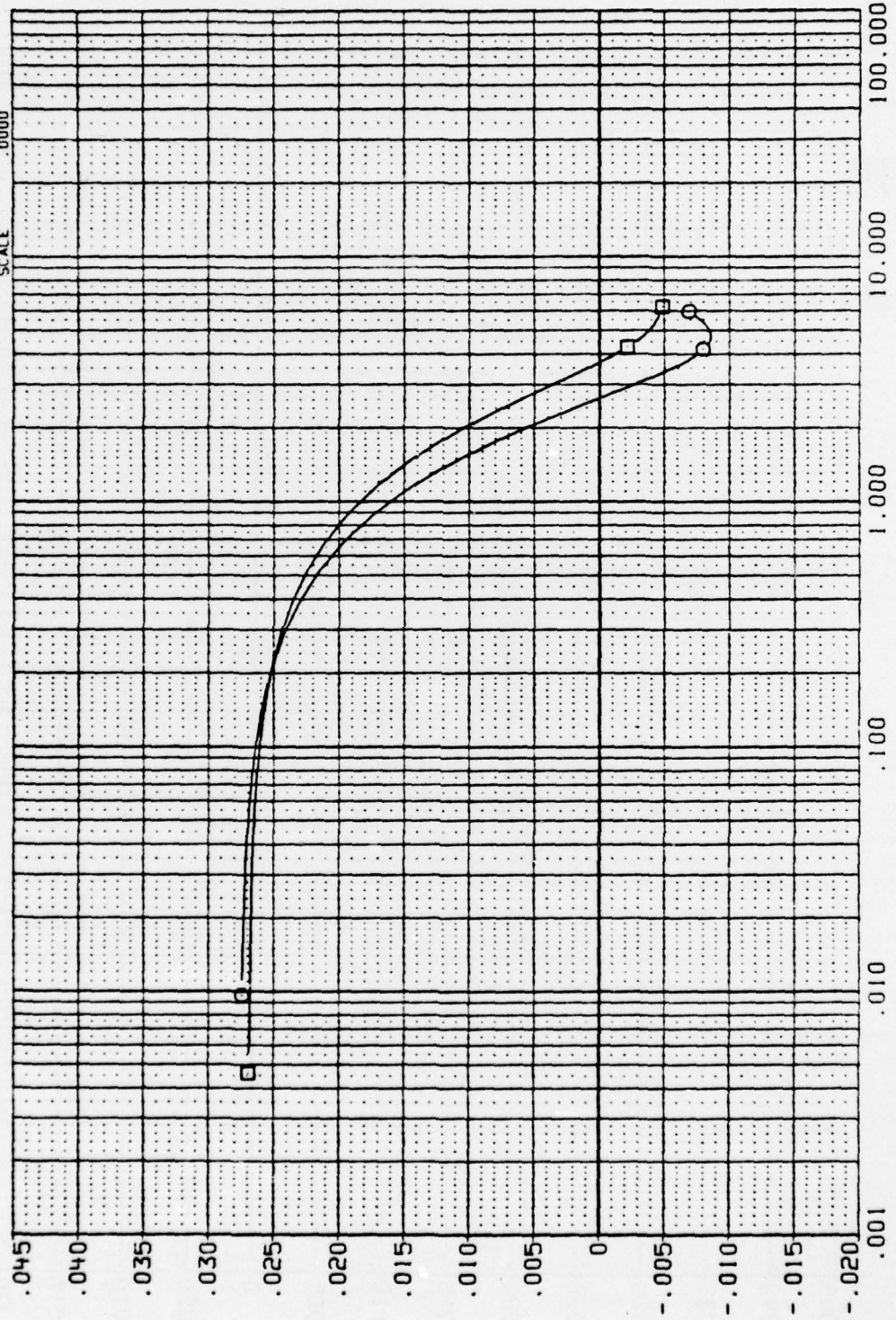
BREF 5.0000 IN.

XMRP 26.5000 IN.

YMRP .0000 IN.

ZMRP .0000 IN.

SCALE .0000



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(DX80N5) ○ CAL 117-123 (BF2) BODY, FIN IN MID POSITION

(DX80N9) □ CAL 117-123 (BF4) BODY, FIN IN MID POSITION

GAP .000
.250

HACH .700
.700

PHI .000
.000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

LREF 5.0000 IN.

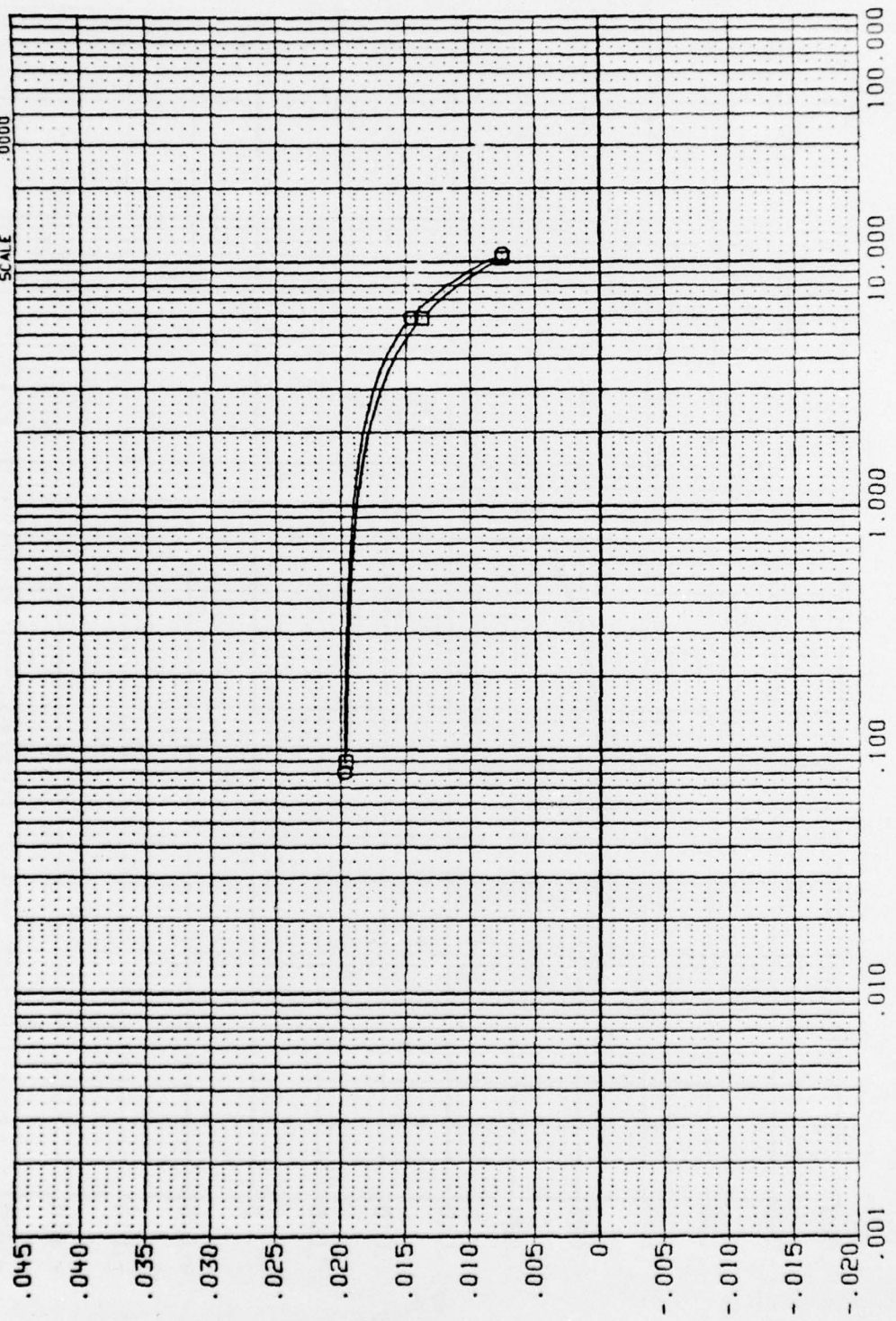
BREF 5.0000 IN.

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YMRP .0000 IN.

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SCALE .0000

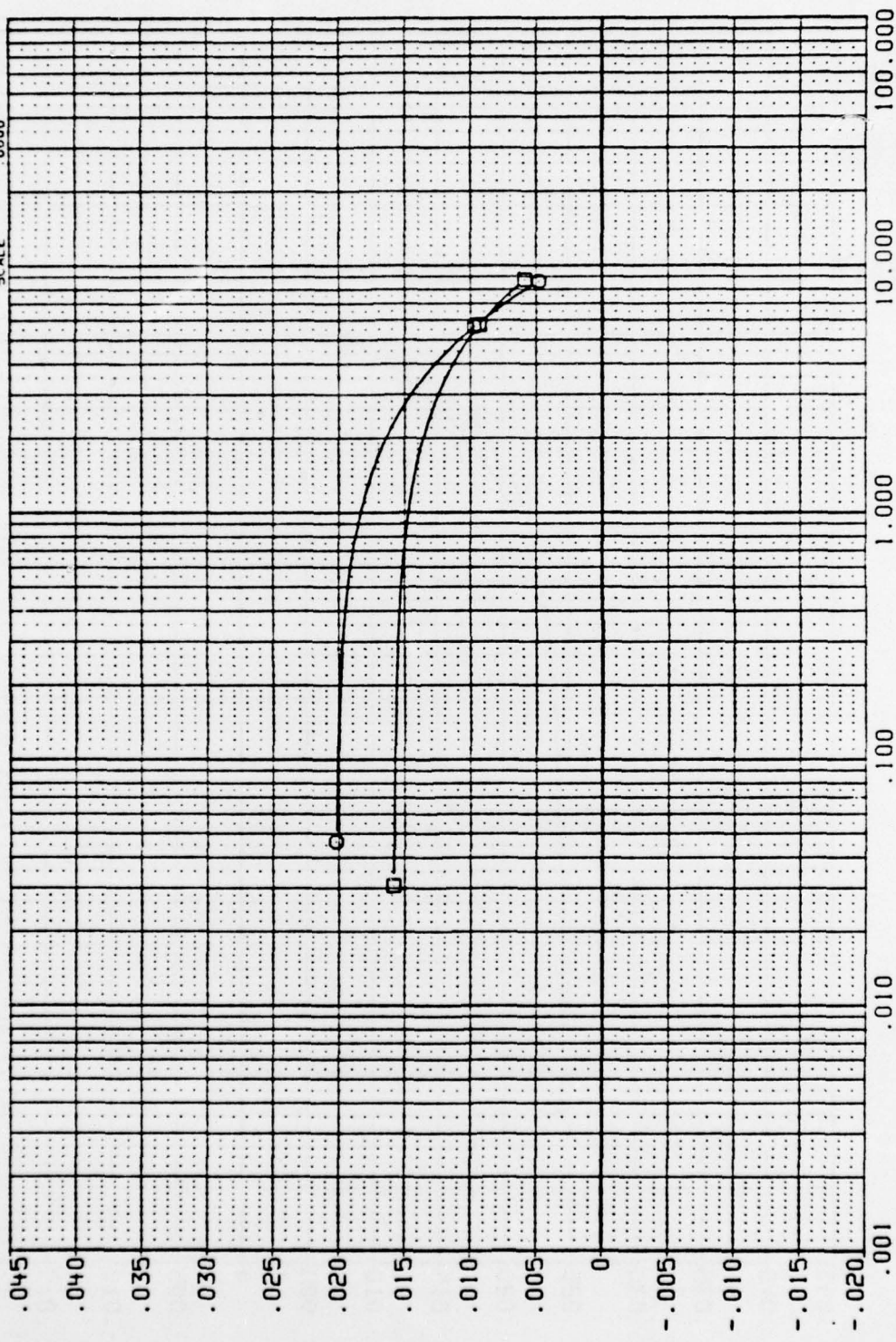


EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (DX80461) ○ CAL 117-123 (BF2) BODY, FIN IN MID POSITION
 (DX8050) □ CAL 117-123 (BF4) BODY, FIN IN MID POSITION

GAP MACH PHI
 .000 .900 .000
 .250 .900 .000

REFERENCE INFORMATION
 SREF 19.6350 50. IN.
 LREF 5.0000 IN.
 BREF 5.0003 IN.
 YMRP 26.5000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS
 CRT

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(DX80N7) □ CAL 117-123 (BF2) BODY, FIN IN MID POSITION

(DX80S1) ○ CAL 117-123 (BF4) BODY, FIN IN MID POSITION

GAP MACH PHI

.000 1.000 .000

.250 1.000 .000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

LREF 5.0000 IN.

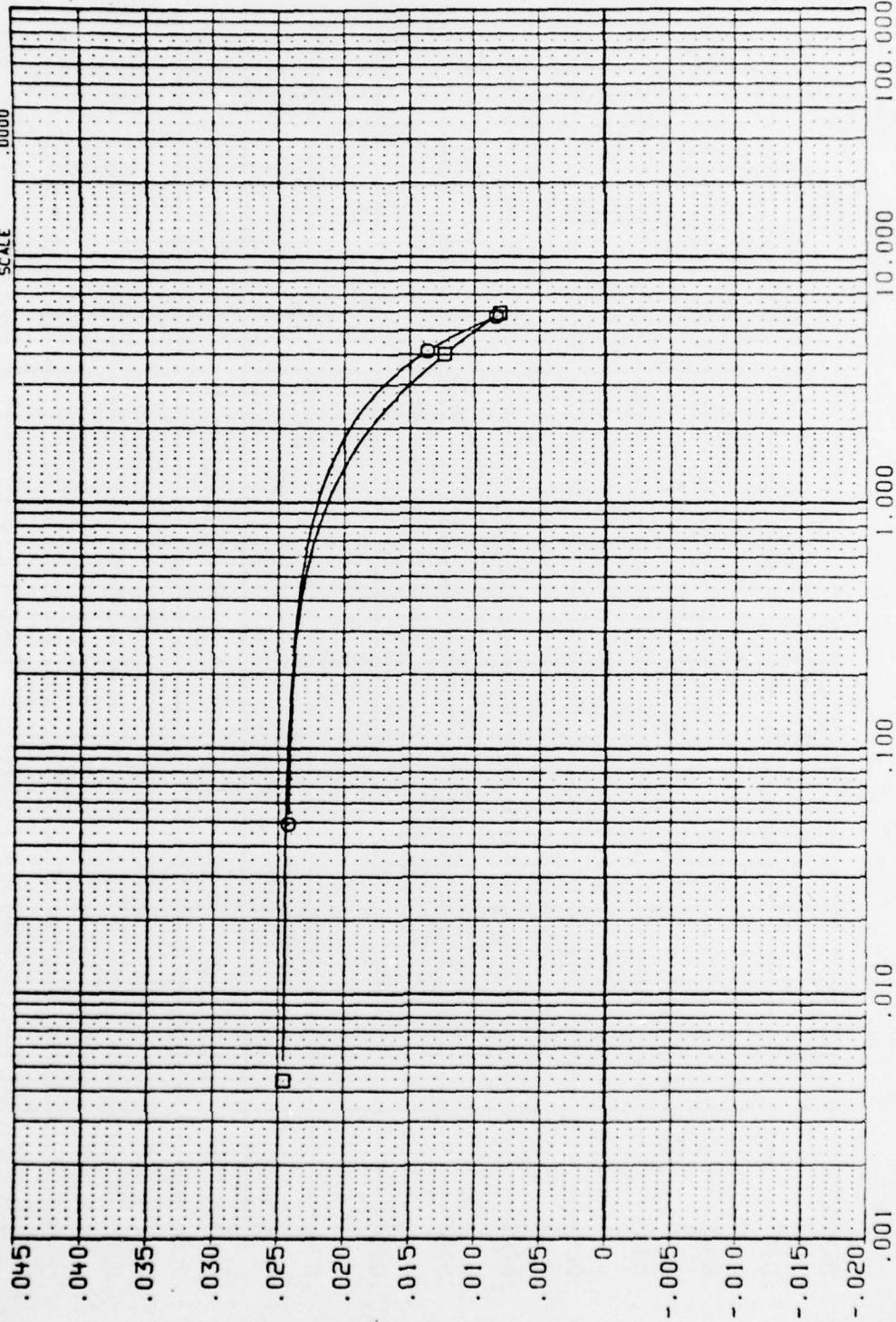
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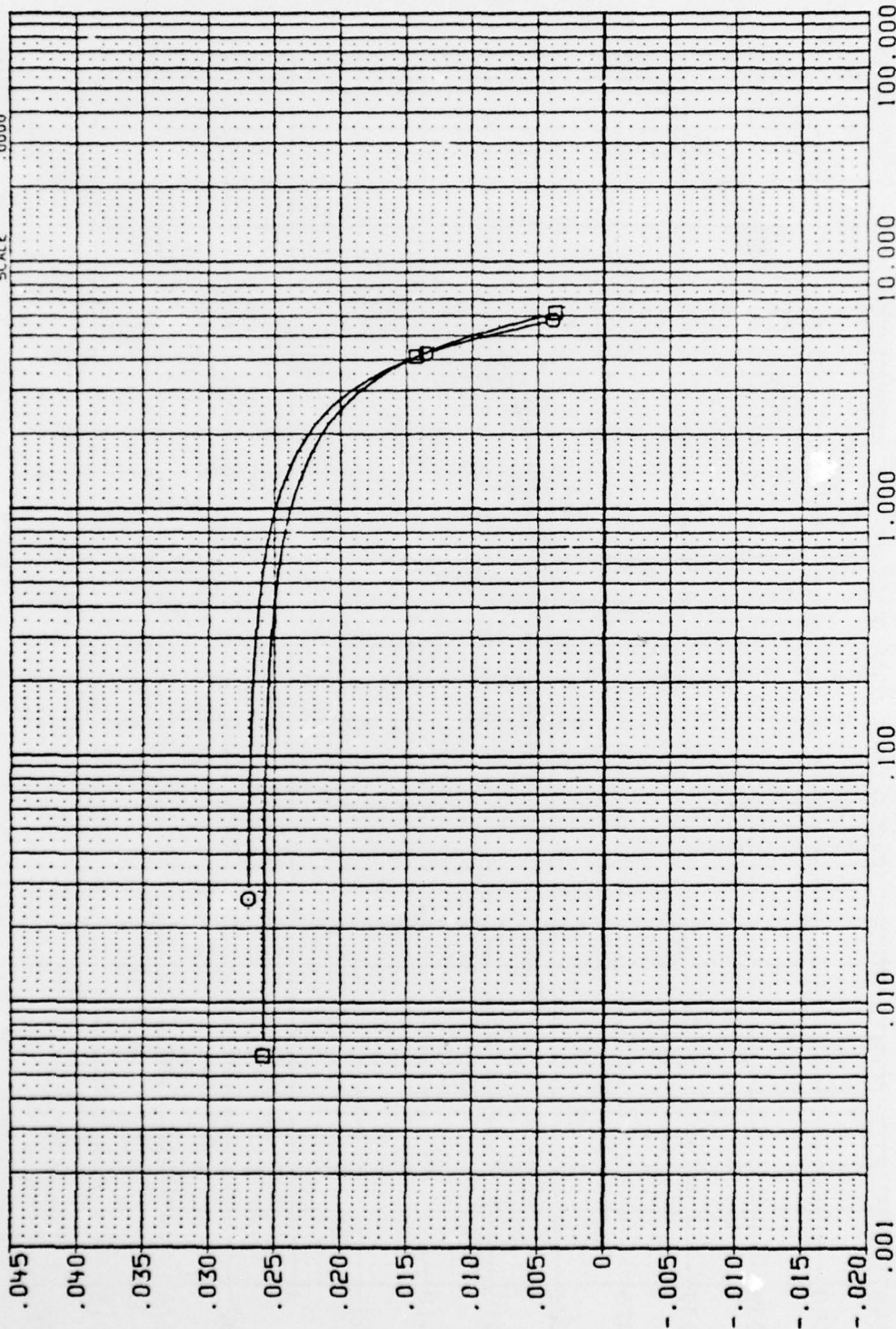


EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (DX8048) ○ CAL 117-123 (BF2) BODY, FIN IN MID POSITION
 (DX8052) □ CAL 117-123 (BF4) BODY, FIN IN MID POSITION

GAP MACH PHI
 .000 1.200 .000
 .250 1.200 .000

REFERENCE INFORMATION
 SREF 19.6350 SQ. IN.
 LREF 5.0000 IN.
 BREF 5.0000 IN.
 XMRP 26.5000 IN.
 YMRP .0000 IN.
 ZMRP .0000 IN.
 SCALE .0000



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS
 CRT

$C_{n_q} F_2$

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(EX8013) \square CAL T17-123 (BF2) BODY, FIN IN FORWARD POSITION

(DX8025) \square CAL T17-123 (BF4) BODY, FIN IN FORWARD POSITION

GAP MACH PHI

.000 .700 .000

.250 .700 .000

REFERENCE INFORMATION

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LREF 5.0000 IN.

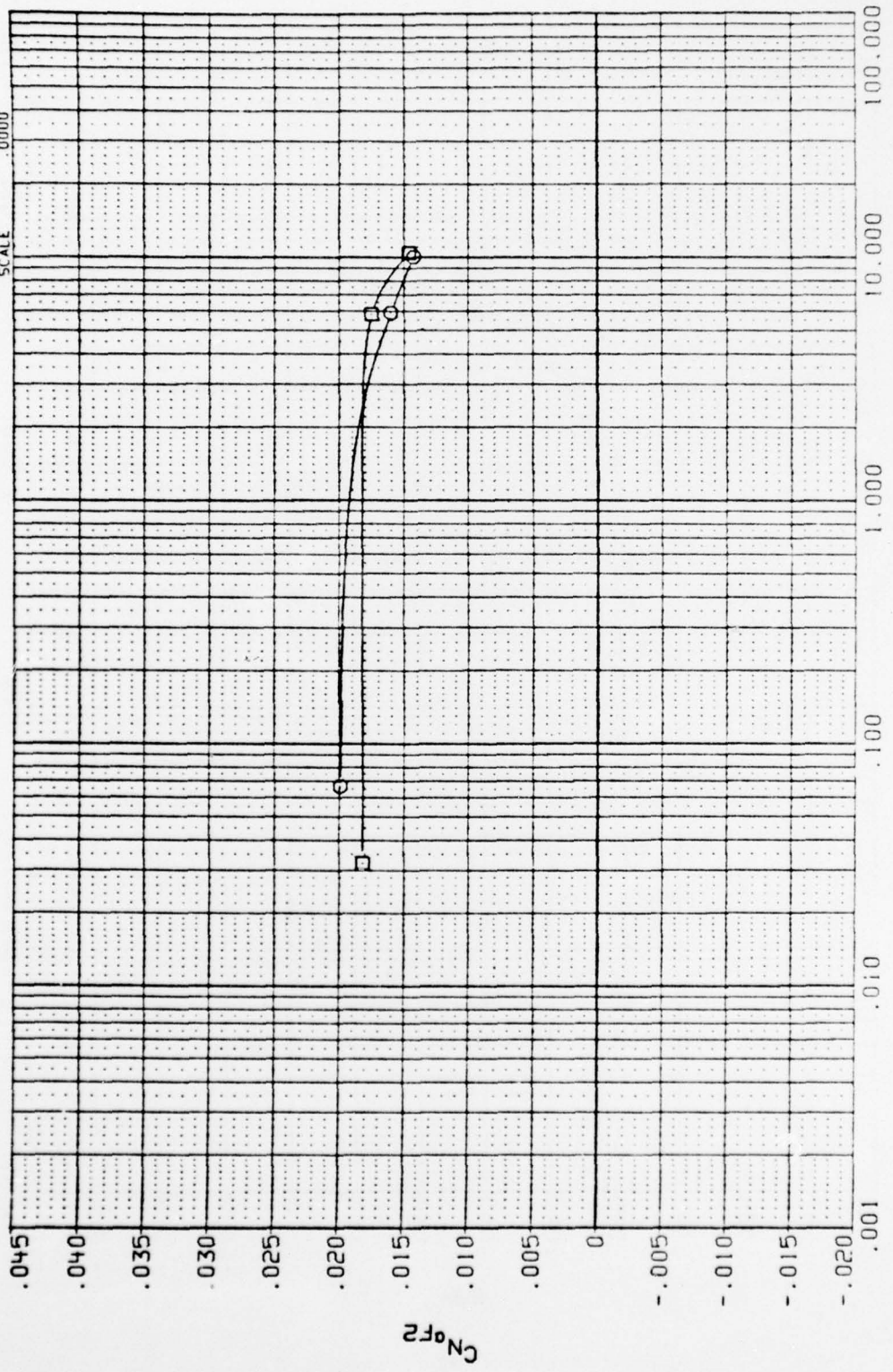
BREF 5.0000 IN.

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YMRP .0000 IN.

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SCALE .0000



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

DATA SET SYMBOL CONFIGURATION DESCRIPTION

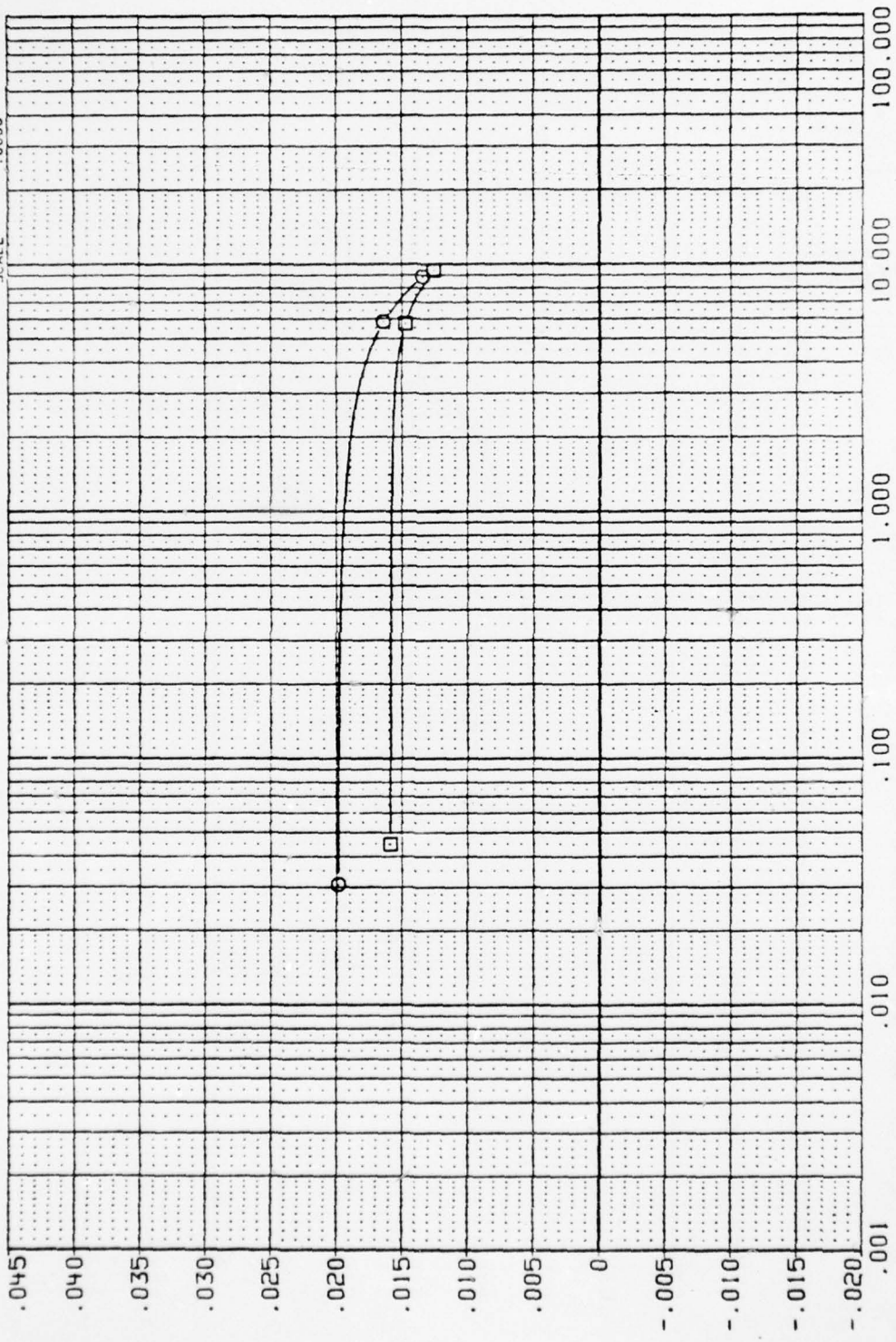
| | | | |
|---------|-------------------|-----------|--------------------|
| (E8014) | CAL 717-123 (BF2) | BODY, FIN | N FORWARD POSITION |
| (D8026) | CAL 717-123 (BF4) | BODY, FIN | N FORWARD POSITION |

GAP MACH PHI

| | | |
|------|------|------|
| .000 | .900 | .000 |
| .250 | .900 | .000 |

REFERENCE INFORMATION

| | | |
|-------|---------|---------|
| SREF | 19.5350 | 50. IN. |
| LREF | 5.0000 | IN. |
| BREF | 5.0000 | IN. |
| XMRP | 26.5000 | IN. |
| YMRP | .0000 | IN. |
| ZMRP | .0000 | IN. |
| SCALE | .0000 | |



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

Cn_q F2

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(EX8015) ○ CAL 117-123 (BF2) BODY, FIN IN FORWARD POSITION

(DX8027) □ CAL 117-123 (BF4) BODY, FIN IN FORWARD POSITION

GAP MACH PHI

.000 1.000 .000

.250 1.000 .000

REFERENCE INFORMATION

SREF 19.6350 50. IN.

LREF 5.0000 IN.

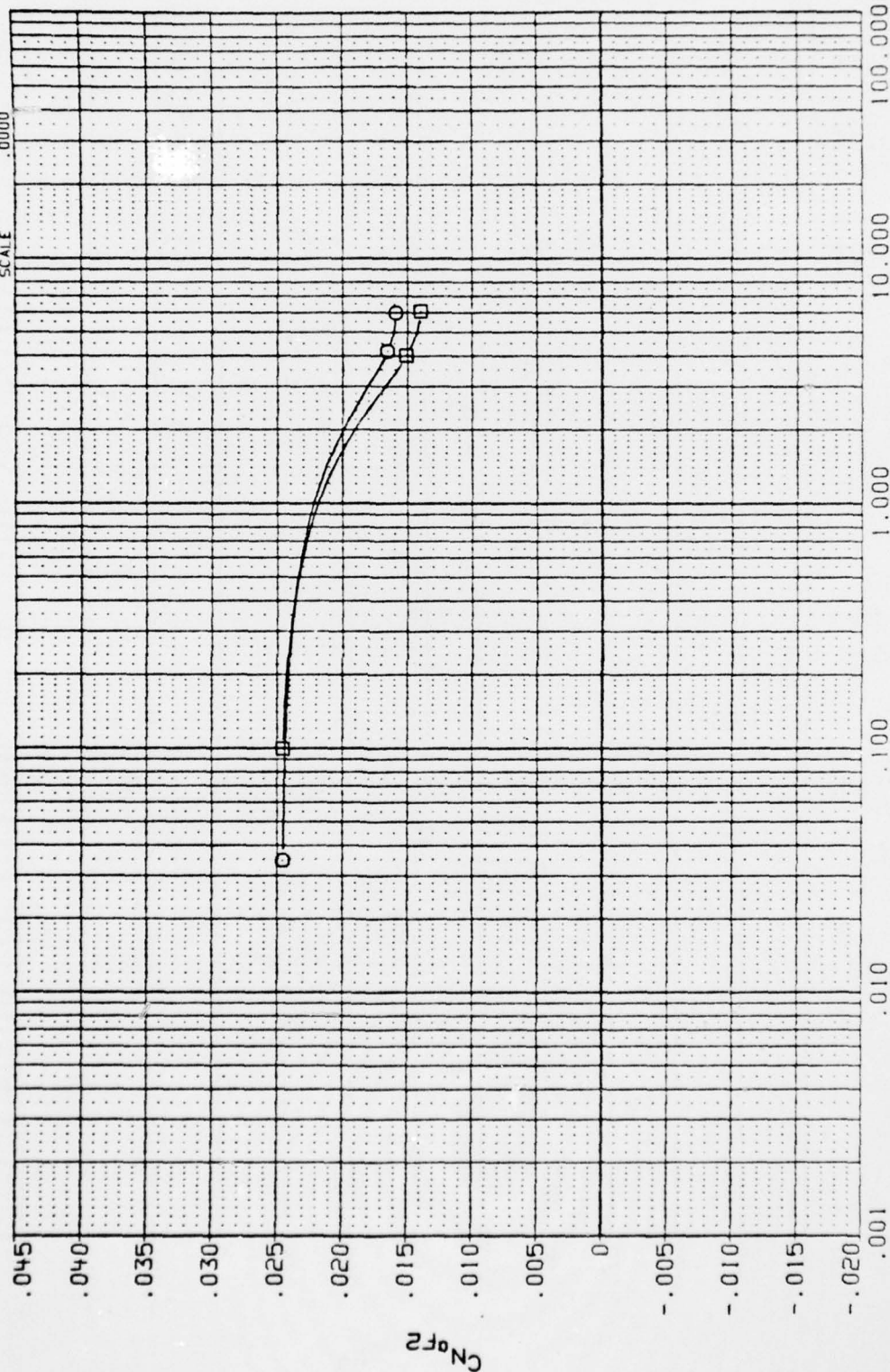
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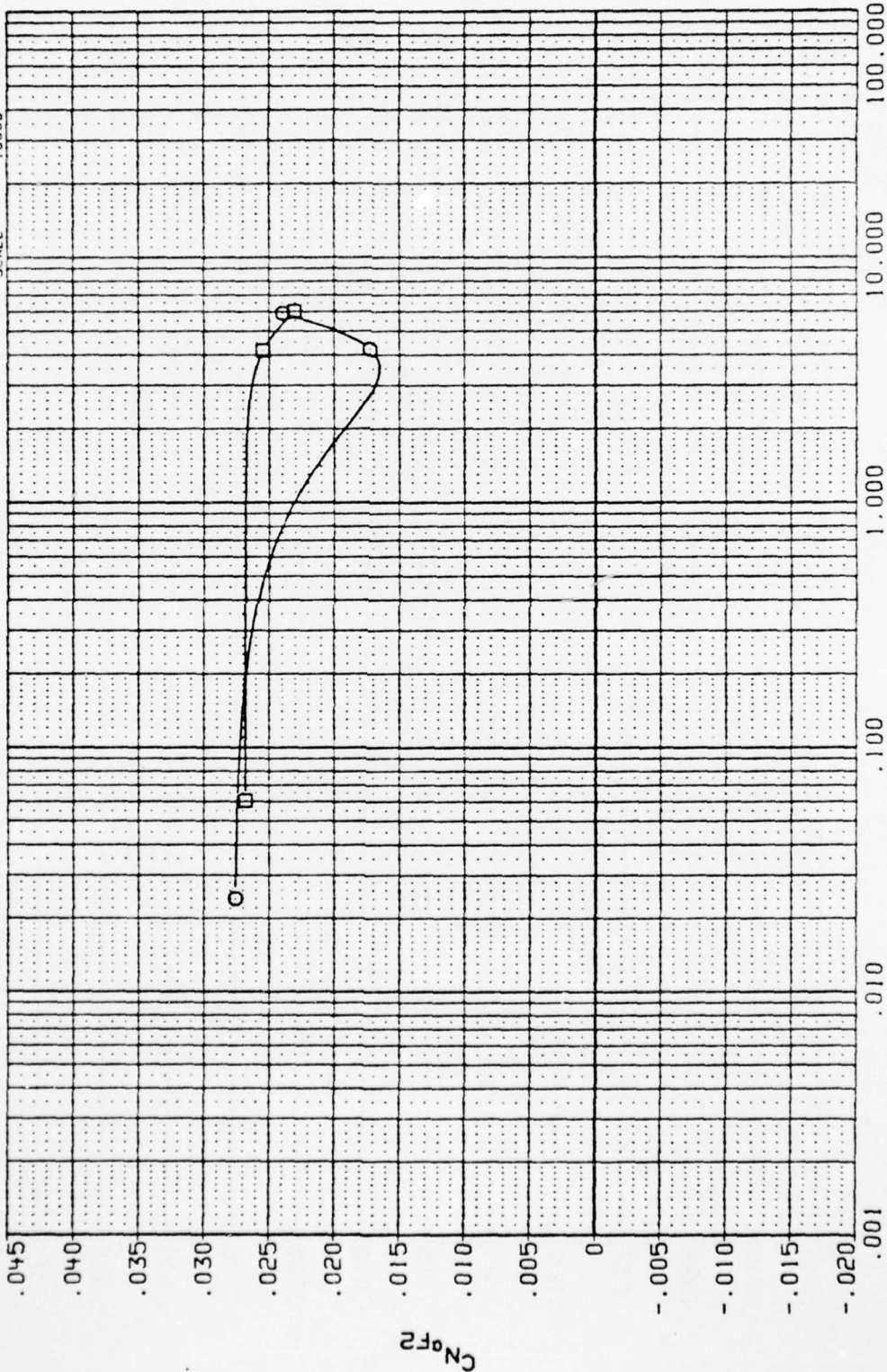
ZMRP .0000 IN.

SCALE .0000



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS
CRT

| | | | | | |
|-----------------|---|------|-------|------|-----------------------|
| DATA SET SYMBOL | CONFIGURATION DESCRIPTION | GAP | MACH | PHI | REFERENCE INFORMATION |
| (EXB016) | CAL T17-123 (BF2) BODY, FIN IN FORWARD POSITION | .000 | 1.200 | .000 | SREF 19.6350 50. IN. |
| (DXE028) | CAL T17-123 (BF4) BODY, FIN IN FORWARD POSITION | .250 | 1.200 | .000 | LREF 5.0000 IN. |
| | | | | | BREF 5.0000 IN. |
| | | | | | XMRP 25.5000 IN. |
| | | | | | YMRP .0000 IN. |
| | | | | | ZMRP .0000 IN. |
| | | | | | SCALE .0000 |



EFFECT OF GAP ON F IN NORMAL FORCE AT VARIOUS THRUST LEVELS
CRT

DATA SET SYMBOL
(DX8018) □
(DX8030) □

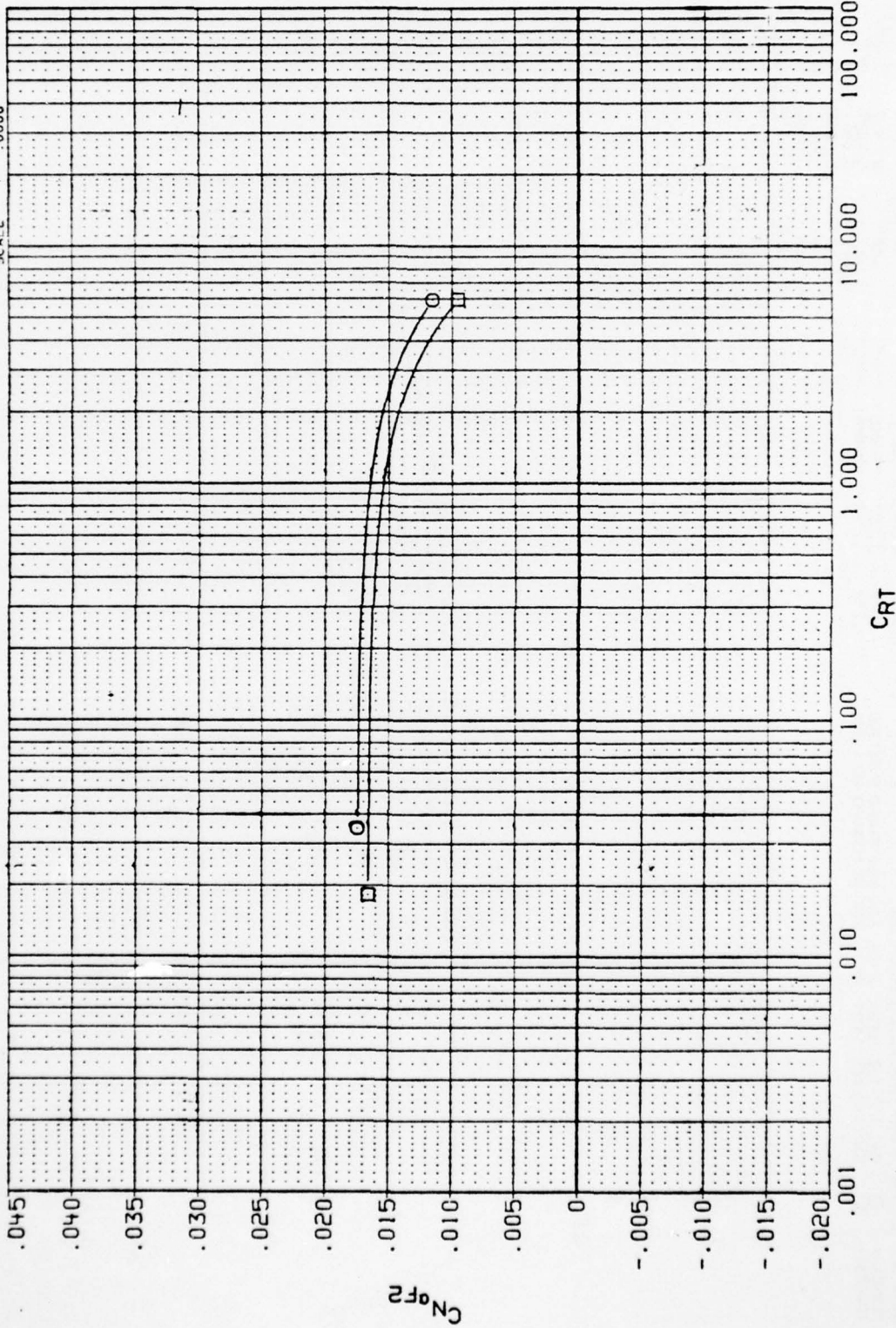
CONFIGURATION DESCRIPTION
CAL 117-12 (BF2) BODY, FIN IN FORWARD POSITION
CAL 117-12 (BF4) BODY, FIN IN FORWARD POSITION

GAP
.000
.250

MACH
1.000
1.000

PHI
45.000
45.000

REFERENCE INFORMATION
SREF 19.6350 50. IN.
LREF 5.0000 IN.
BREF 5.0000 IN.
XMRP 26.5000 IN.
YMRP .0000 IN.
ZMRP .0000 IN.
SCALE .0000



EFFECT OF GAP ON FIN NORMAL FORCE AT VARIOUS THRUST LEVELS

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| Ling-Temco-Vought Aerospace Corp. Vought Aeronautics Division ATTN: C. R. James, Unit 2-53330 Box 5907 Dallas, Texas 75222 | 1 | Data Management Services Department 5807 Chrysler Corporation Huntsville Electronics Division ATTN: Mr. J. E. Vaughn 102 Wynn Drive Huntsville, Alabama 35805 | 1 |
| Lockheed Missiles & Space Company Huntsville R&E Center ATTN: Mr. J. Benefield 4800 Bradford Boulevard, N.W. Huntsville, Alabama 35805 | 1 | DRSMI-R, Dr. McDaniel | 1 |
| Lockheed Aircraft Corporation Missile and Space Division ATTN: Technical Information Center P. O. Box 504 Sunnyvale, California | 1 | -R, Dr. Kobler | 1 |
| The Martin-Marietta Corporation Orlando Division ATTN: D. Tipping L. Gilbert Orlando, Florida 32804 | 1 | -RBD | 3 |
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