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ROYAL AIRCRAFT ESTABLISHMENT FARNBOROUGH (ENGLAND)
TABLE OF SPACE VEHICLES: 1973-1977.(U)
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ROYAL AIRCRAFT ESTABLISHMENT

Technical Memorandum Space 256

Received for printing 2 February 1978

TABLE OF SPACE VEHICLES: 1973-1977

by

J. A. Pilkington*

SUMMARY

RAE Technical Report 73006 lists all space vehicles successfully launched before 31 December 1972. This Memorandum, which supersedes Technical Memorandum Space 242, extends the tabulation to the end of 1977, and gives amendments to Technical Report 73006.

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Table 2 (continued)

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(A) Heliocentric orbits (continued)

Name	Launch date (UT)	Mass (kg) and basic shape	Basic size (m)	Perihelion (AU)	Aphelion (AU)	Eccentricity	Inclination to ecliptic (deg)	Period (days)	Remarks
T Pioneer 10	1972-12A	250 hexagon + dish	2.75 dia 1.45 long	0.991 0.98?	5.864 ∞	0.711 1.21	1.31? 1.31?	2318 -	Passed 130400km behind Jupiter on 1973 Dec 4.10
Pioneer 10 rocket	1972 Mar 3.075	66 sphere + nozzle	0.94 dia 1.32 long				Exceeded solar escape velocity*		Up-rated Surveyor Retrorocket
D Venus 8 [†] (capsule)	1972-21A	495 sphere	1.0 dia	0.72?	1.08?	0.20?	2?	313?	Soft-landed on Venus day side 1972 July 22 at 09h 29m
D Venus 8 (compartment)	1972-21E	689 cylinder + 2 Vanes	1.22 dia 2.5 long	0.72?	1.08?	0.20?	2?	313?	Decayed in Venus atmosphere, 1972 July 22 at 08h 38m
Venus 8 rocket	1972-21D	440 cylinder	2.0 dia 2.0 long	-	-	-	-	-	Orbit unknown
T Pioneer 11	1973-19A	259 hexagon + dish	2.75 dia 1.45 long	1.000 0.98?	6.012 ∞	0.715 1.21	1.31? 1.31?	2398 -	Passed 41850km below Jupiter on 1974 Dec 3.224
Pioneer 11 rocket	1973 Apr 6.091	66 sphere + nozzle	0.94 dia 1.32 long				Exceeded solar escape velocity** Orbit probably similar to 1973-19A		Up-rated Surveyor Retrorocket

* Expected final path apex near R.A. 04h 33.0m; Declination 16.42°. Occulted by Jovian satellite Io on 1974 Dec 4.11

** Expected to pass 24000km above Saturn's cloud tops (just outside rings) on 1973 Sep 5; and then past satellite Titan. Renamed 'Pioneer Saturn'.

† Parachute descent took 51min; surface transmissions lasted 50min.

Table 2 (continued)

(A) Heliocentric orbits (continued)

Name	Launch date (UT)	Mass (kg) and basic shape	Basic size (m)	Perihelion (AU)	Aphelion (AU)	Eccentricity	Inclination to ecliptic (deg)	Period (days)	Remarks
Mars 4 * orbiter	1973 Jul 21.813	3440? full cylinder + 2 panels	2.3 dia 2.7 long	1.02?	1.63?	0.23?	2.2?	556?	Passed 2200km behind Mars on 1974 Feb 10
Mars 4 rocket	1973 Jul 21.813	1900? cylinder	3.9 dia 3.9 long?				Orbit probably similar to 1973-47A		
Mars 5 orbiter	1973 Jul 25.789	3440? full cylinder + 2 panels	2.3 dia 2.7 long	1.01?	1.65?	0.24?	2.2?	560?	Entered Areocentric orbit on 1974 Feb 12.66. See page 36
Mars 5 rocket	1973 Jul 25.789	1900? cylinder	3.9 dia 3.9 long?				Orbit probably similar to 1973-49A		
Mars 6 compartment	1973 Aug 5.740	2060? empty cylinder + 2 panels	2.3 dia 2.7 long	1.01?	1.67?	0.24?	2.2?	567?	Passed 1600km behind Mars on 1974 Mar 12.38
D Mars 6 capsule**	1973 Aug 5.740	1200? full 635 empty cone-sphere	2.5 dia 1.1 long	1.01?	1.67?	0.24?	2.2?	567?	Soft landed on Mars 1974 Mar 12 at 24°S, 25°W
Mars 6 rocket	1973 Aug 5.740	1900? cylinder	3.9 dia 3.9 long?				Orbit probably similar to 1973-52A		
Mars 7 compartment	1973 Aug 9.708	2060? empty cylinder + 2 panels	2.3 dia 2.7 long	1.01?	1.69?	0.25?	2.2?	574?	Passed 1300km behind Mars on 1974 Mar 9
Mars 7 † capsule	1973 Aug 9.708	1200? full 635 empty cone-sphere	2.5 dia 1.1 long	1.01?	1.69?	0.25?	2.2?	574?	Passed 1300km behind Mars on 1974 Mar 9
							(Planned landing site 43°S, 42°W)		

* Areocentric orbit insertion engine failed.

** Transmissions ceased on landing, at 09h 11m 05s UT

† Capsule separated from compartment correctly 48000km from Mars, but its small motor failed to put it on a collision course.

Table 2 (continued)

(A) Heliocentric orbits (continued)

Name	Launch date (UT)	Mass (kg) and basic shape	Basic size (m)	Perihelion (AU)	Aphelion (AU)	Eccentricity	Inclination to ecliptic (deg)	Period (days)	Remarks
Mars 7 rocket	1973-53E 1973 Aug 9.708	1900? cylinder	3.9 dia 3.9 long?				Orbit probably similar to 1973-53A		
Mariner 10*	1973-85A 1973 Nov 3.24	526 octagonal cylinder	1.27 dia 2.90 long?	0.70 0.387	1.11 0.839	0.23 0.369	2.6 2.6	317 176	Passed 5750km ahead of Venus 1974 Feb 5.71
Mariner 10 rocket	1973-85B 1973 Nov 3.24	1815 cylinder	3.05 dia 8.14 long				Orbit probably similar to 1973-85A		Centaur. Passed 45000km behind Venus
Helios 1	1974-97A 1974 Dec 10.299	370 double-cone	1.75 to 2.77 dia 2.18 long	0.307	0.985	0.525	0	190	At perihelion on 1975 Mar 15.47
Helios 1 rocket	1974-97C 1974 Dec 10.299	66 sphere + nozzle	0.94 dia 1.32 long				Orbit probably similar to 1974-97A		Upgraded Surveyor Retrorocket
Fragment	1974-97D								
Venus 9 orbiter	1975-50A 1975 Jun 8.109	3376 full cylinder + 2 panels	-	0.70?	1.11?	0.23?	2.3?	316?	Entered orbit round Venus on 1975 Oct 22.17. See Page 36b
Venus 9 lander**	1975-50D 1975 Jun 8.109	1560 sphere-annulus	-	0.70?	1.11?	0.23?	2.3?	316?	Soft landed on Venus day side on 1975 Oct 22 at 05h 13m
Venus 9 rocket	1975-50E 1975 Jun 8.109	1900? cylinder	3.9 dia 3.9 long?				Orbit probably similar to 1975-50A		

* Passed 700km ahead of Mercury (night side) on 1974 Mar 29.87;

passed 48000km below Mercury's South Pole on 1974 Sep 21.874;

passed 330km from Mercury on 1975 Mar 16.94. Transmissions ceased on 1975 Mar 24.

**Parachute descent took 75min; surface transmissions lasted 53min. Location - at 33°N, 293° long. (Inside region Beta).

Helios is German spacecraft launched by US rocket.

Table 2 (continued)

(A) Heliocentric orbits (continued)

Name	Launch date (UT)	Mass (kg) and basic shape	Basic size (m)	Perihelion (AU)	Aphelion (AU)	Eccentricity	Inclination to ecliptic (deg)	Period (days)	Remarks
Venus 10 orbiter	1975-54A 1975 Jun 14.13	3473 full cylinder + 2 panels	-	0.70?	1.11?	0.23?	2.3?	316?	Entered orbit round Venus on 1975 Oct 25.17. See page 36b
D Venus 10 lander*	1975-54D 1975 Jun 14.13	1560 sphere- annulus	-	0.70?	1.11?	0.23?	2.3?	316?	Soft-landed on Venus day side on 1975 Oct 25 at 05h 17m
Venus 10 rocket	1975-54E 1975 Jun 14.13	1900? cylinder	3.9 dia 3.9 long?						Ejected onto impact path 1975 Oct 23
T Viking 1 orbiter	1975-75A 1975 Aug 20.890	2325 full octagonal box + 4 vanes	3.9 dia 3.9 long? 3.3 high 1.8 wide 1.5 deep	1.003	1.672	0.250	4.48	565	Entered areocentric orbit on 1976 Jun 19, then ejected lander
D Viking 1 T lander	1975-75C 1975 Aug 20.890	1090 full 600 empty pyramid	2.1 high** 3.0 wide 2.5 deep	1.003	1.672	0.250	4.48	565	Soft-landed on Mars 1976 Jul 20 at 12h 12m. Site 22.27°N, 47.91°W
Viking 1 rocket	1975-75B 1975 Aug 20.890	1815 cylinder	3.05 dia 8.14 long						Centaur. Passed 80500km from Mars
T Viking 2 orbiter	1975-83A 1975 Sep 9.777	2325 full octagonal box + 4 vanes	3.3 high 1.8 wide 1.5 deep	1.006	1.669	0.248	2.92	565	Entered areocentric orbit on 1976 Aug 7.47 then ejected lander
D Viking 2 T lander	1975-83C 1975 Sep 9.777	1090 full 600 empty pyramid	2.1 high** 3.0 wide 2.5 deep	1.006	1.669	0.248	2.92	565	Soft-landed on Mars 1976 Sep 3 at 22h 58m. Site 47.67°N, 225.71°W.
Viking 2 rocket	1975-83B 1975 Sep 9.777	1815 cylinder	3.05 dia 8.14 long						Centaur. Passed 80500km from Mars.

*Parachute descent took 75min; surface transmissions lasted 65min. Landed 2200km from Venus 9, at 15°N, 295° long.

**including legs.

Viking areocentric orbits - see pages 36 and 36a.

Table 2 (continued)

(A) Heliocentric orbits (concluded)

Page 14d

Name	Launch Date (UT)	Mass (kg) and basic shape	Basic size (m)	Perihelion (AU)	Aphelion (AU)	Eccentricity	Inclination to ecliptic (deg)	Period (days)	Remarks
T Helios 2	1976 Jan 15.23	376 double-cone	1.75 to 2.77 dia 2.18 long	0.28	0.995	0.561	0	186	At perihelion on 1976 Apr 17
Helios 2 second stage	1976 Jan 15.23	1815 cylinder	3.05 dia 8.14 long			Orbit unknown			Centaur
Helios 2 rocket	1976 Jan 15.23	66 sphere + nozzle	0.94 dia 1.32 long			Orbit probably similar to 1976-03A			Up-rated Surveyor Retrorocket
T Voyager 2	1977 Aug 20.603	795 decagon + dish	1.9 and 3.7 dia 1.5 long			To exceed Solar escape velocity			Flyby of four outer planets*
Voyager 2 second stage	1977 Aug 20.603	1815 cylinder	3.05 dia 8.14 long			Unknown heliocentric orbit			Centaur
Voyager 2 rocket	1977 Aug 20.603	66 sphere + nozzle	0.94 dia 1.32 long			Orbit probably similar to 1977-76A			Up-rated Surveyor Retrorocket
T Voyager 1	1977 Sep 5.539	795 decagon + dish	1.9 and 3.7 dia 1.5 long			To exceed Solar escape velocity			Jupiter and Saturn flyby**
Voyager 1 second stage	1977 Sep 5.539	1815 cylinder	3.05 dia 8.14 long			Unknown heliocentric orbit			Centaur
Voyager 1 rocket	1977 Sep 5.539	66 sphere + nozzle	0.94 dia 1.32 long			Orbit probably similar to 1977-84A			Up-rated Surveyor Retrorocket

*To pass 642000km from Jupiter on 1979 Jul 10, Saturn on 1981 Aug 27, Uranus on 1986 Jan 31, and Neptune in 1989 Sep.

**To pass 177000km from Jupiter on 1979 Mar 5, and 209000km from Saturn on 1980 Nov 13.

Table 2 (continued)

(B) Orbits in the Earth-Moon system (continued)

Name	Launch date (UT) and flight time	Mass (kg) and basic shape	Basic size (m)	Maximum distance from Earth (km)	Moon's age at launch (days)	All-burnt velocity km/s	Remarks
D Apollo 17 3M (command module)	1972-96A 1972 Dec 7.23 12.58 days	5826 cone	3.91 dia 3.66 long	400000	1.4	10.90?	Orbited Moon 75 times. Landed on Earth 1972 Dec 19 at 19h 24m
D Apollo 17 (service module)	1972-96E 1972 Dec 7.23 12.57 days	24514 full cylinder	3.91 dia 7.49 long	400000	1.4	10.90?	Attached to CM until decay in Earth's atmosphere on 1972 Dec 19 at 19h 14m
D Apollo 17 rocket	1972-96B 1972 Dec 7.23 87.0 hours?	13930 cylinder	6.6 dia 18.7 long	400000	1.4	10.90?	Saturn I/B. Hit Moon 1972 Dec 10 at 20h 33m? Impact 4.2°S, 12.3°W
D LEM 12 T (AS + DS) (with LRV-3)	1972-96C 1972 Dec 7.23 4.60 days	16440 full box + octagon	4.09 high 3.76 wide 3.13 deep	400000	1.4	10.90?	Soft landed on Moon 1972 Dec 11 at 19h 55m. Site 20.16°N, 30.75°E. AS relaunched from Moon*
D Luna 21 (with Lunokhod 2 **)	1973-01A 1973 Jan 8.288 7.65 days	5600? full pyramid + car	2.3 high 3.2 wide 3.3 deep	400000	3.6	10.90?	Orbited Moon 40 times. Soft landed 1973 Jan 15 at 22h 35m. Position 26.5°N, 30.6°E
Luna 21 rocket	1973-01D 1973 Jan 8.288	1900? cylinder	3.9 dia 3.9 long?	-	3.6	10.90?	Orbit unknown
Explorer 49 third stage	1973-39B 1973 Jun 10.592	66 sphere + nozzle	0.94 dia 1.32 long	390250	9.4	10.90?	Burner 2. In high eccentricity orbit.
Luna 22 rocket	1974-37E 1974 May 29.373	1900? cylinder	3.9 dia 3.9 long?	-	7.5	10.90?	Orbit unknown

* See page 35

** Lunokhod 2 mass 840kg

Table 2 (continued)

(B) Orbits in the Earth-Moon system (concluded)

Name	Launch date (UT) and flight time	Mass (kg) and basic shape	Basic size (m)	Maximum distance from Earth (km)	Moon's age at launch (days)	All-burnt velocity km/s	Remarks
D Luna 23* (AS + DS)	1974-84A 1974 Oct 28.60 8.63 days	5500? full pyramid + cylinder	3.1 high 3.2 wide 3.3 deep	400000	13.1	10.90?	Orbited Moon 53 times? Hard landed 1974 Nov 6 at 05h 37m. Position 13.5°N?; 56.5°E?
Luna 23 rocket	1974-84D 1974 Oct 28.60	1900? cylinder	3.9 dia 3.9 long?	-	13.1	10.90?	Orbit unknown
D Luna 24 (AS + DS)	1976-81A 1976 Aug 9.628 8.65 days	5500? full pyramid + cylinder	3.1 high 3.2 wide 3.3 deep	400000	13.6	10.90?	Orbited Moon 53 times? Soft-landed 1976 Aug 18 at 06h 36m. Position 12.75°N, 62.2°E. AS relaunched from Moon**
Luna 24 rocket	1976-81F 1976 Aug 9.628	1900? cylinder	3.9 dia 3.9 long?	-	13.6	10.90?	Orbit unknown

* Damage on landing prevented re-launch of ascent stage.

** See page 35.

Table 2 (continued)

(C) Selenocentric orbits - Earth launch (continued)

Name	Launch date injection date and ejection date	Mass (kg) and basic shape	Basic size (m)	Date of orbital determination (UT)	Inclination to Moon's equator (deg)	Period (min)	Semi major axis (km)	Periselene height (km)	Aposelene height (km)	Orbital eccen- tricity
D Apollo 17 3M (CM + SN)	1972 Dec 7.23 1972 Dec 10.83 1972 Dec 16.98	30340 full cone-cylinder	3.91 dia 11.15 long	1972 Dec 10.83 1972 Dec 11.01 1972 Dec 11.78	159.9 159.9 159.9	128 115 118.8	1944 1807 1849	94 28 96	317 109 126	0.057 0.023 0.008
D LEM 12 (with LRV-3)	1972 Dec 7.23 1972 Dec 10.83 1972 Dec 11.83	4937 empty box + octagon	4.09 high 3.76 wide 3.13 deep	1972 Dec 11.72 1972 Dec 11.79	159.9 159.9	115 114	1807 1800	28 13	109 111	0.023 0.027
D Luna 21 (with Lunokhod 2)	1973 Jan 8.288 1973 Jan 12.60 1973 Jan 15.94	4000? full pyramid + car	2.3 high 3.2 wide 3.3 deep	1973 Jan 12.60 1973 Jan 15.93	60 60	118 114	1838 1801	90 16	110 110	0.005 0.026
Explorer 49 (RAE 2)	1973 Jun 10.592 1973 Jun 15.26	200 cylinder + booms	0.92 dia 0.79 long	1973 Jun 15.31 1973 Jun 20.16 1974 Aug 14 1975 Sep 1.0	38.26 38.71 55.34 76.2	241 221.2 221.9 222.0	2964 2797 2802 2803	1120 1053 1051 1020	1331 1065 1077 1109	0.036 0.002 0.005 0.016
Explorer 49 retrorocket	1973 Jun 10.592 1973 Jun 15.26	134 full cone-cylinder	0.5 dia? 1.0 long?	Orbit similar to 1973-39A						
Fragment Luna 22	1974 May 29.373 1974 Jun 2.687	4000? pyramid	2.3 high 3.2 wide 3.3 deep	1974 Jun 2.7 1974 Jun 9 1974 Jun 13 1974 Nov 11.63 1975 Apr 2.33 1975 Aug 24 1975 Sep 3.4*	19.58 19.58 19.58 19.55 21 21 21	130 121 131 192 192 192 179	1958 1873 1978 2542 2543 2542 2431	219 25 181 171 200 30 100	221 244 299 1437 1409 1578 1286	0.0 0.058 0.030 0.248 0.238 0.304 0.244
D Luna 23 (AS + DS)	1974 Oct 28.60 1974 Nov 1.917 1974 Nov 6.23	4000? full pyramid + cylinder	3.1 high 3.2 wide 3.3 deep	1974 Nov 2 1974 Nov 6.22	138 138	117 114	1837 1799	94 17	104 105	0.003 0.024

See pages 28 and 28a for further details of the above decayed satellites.

* Manoeuvring fuel exhausted on 1975 Sep 2

Table 2 (continued)

(C) Selenocentric orbits - Earth launch (concluded)

Page 33a

Name	Launch date injection date and ejection date	Mass (kg) and basic shape	Basic size (m)	Date of orbital determination (UT)	Inclination to Moon's equator (deg)	Period (min)	Semi major axis (km)	Periselene height (km)	Aposelene height (km)	Orbital eccen- tricity
D Luna 24 (AS + DS)	1976 Aug 9.628 1976 Aug 13.966 1976 Aug 18.275	4000? full pyramid + cylinder	3.1 high 3.2 wide 3.3 deep	1976 Aug 14.0 1976 Aug 17	120 120	119 114	1653 1804	115 12	115 120	0 0.030

See page 28a for further details of the above decayed satellite.

Table 2 (continued)

(D) Selenocentric orbits - Moon Launch (concluded)

Name	Moon launch date and descent date	Mass (kg) and basic shape	Basic size (m)	Date of orbital determination (UT)	Inclination to Moon's equator (deg)	Period (min)	Semi major axis (km)	Periselene height (km)	Aposelene height (km)	Orbital eccentricity
D Luna 20 R ascent stage	1972 Feb 22.96 1972 Feb 25.80	8007 full cylinder + 3 spheres	2.3 high 1.6 wide 0.5 deep	Exceeded Moon escape velocity Landed on Earth 1972 Feb 25.80 with Moon material						
2M LEM 11* ascent stage	1972 Apr 24.06	2134 empty box + 2 tanks	2.52 high 3.76 wide 3.13 deep	1972 Apr 24.07 1972 Apr 24.14	169.3 169.3	114 120	1784 1850	17 98	75 126	0.017 0.008
D LEM 12** ascent stage	1972 Dec 14.96 1972 Dec 15.29	2145 empty box + 2 tanks	2.52 high 3.76 wide 3.13 deep	1972 Dec 14.97 1972 Dec 15.04	159.9 159.9	114 118.8	1784 1849	17 96	75 126	0.017 0.008
D Luna 24 R ascent stage	1976 Aug 19.226 1976 Aug 22.747	8007 full cylinder + 3 spheres	2.3 high 1.6 wide 0.5 deep	Exceeded Moon escape velocity Landed on Earth 1976 Aug 22.75 with Moon material						

* Briefly docked with Apollo 16 for crew transfer on 1972 Apr 24.14. Still in orbit.

**Briefly docked with Apollo 17 for crew transfer on 1972 Dec 15.04. Sent crashing into Moon: 19.95°N, 30.73°E

Table 2 (continued)

(E) Areocentric orbits - Earth launch

Page 36

Name	Launch date injection date and ejection date	Mass (kg) and basic shape	Basic size (m)	Date of orbital determination (UT)	Inclination to Mars equator (deg)	Period (min)	Semi major axis (km)	Periapsis height (km)	Apoapsis height (km)	Orbital eccen- tricity
Mars 2 orbiter	1971-45A 1971 May 19.68 1971 Nov 27.85	3440? full 2060? empty cylinder + 2 panels	2.3 dia 2.7 long	1971 Nov 27.85	48.9	1078	16585	1380	25000	0.712
D Mars 2 lander	1971-45E 1971 May 19.68 1971 Nov 27.85	1200? full 635 empty cone-sphere	2.5 dia 1.1 long	Ejected onto impact path 1971 Nov 27.66 Hard-landed at 44°S, 47°E						
Mars 3 orbiter	1971-49A 1971 May 28.64 1971 Dec 2.57	3440? full 2060? empty cylinder + 2 panels	2.3 dia 2.7 long	1971 Dec 2.57	48.9?	15840	99495	1500	190700	0.951
D Mars 3 lander	1971-49F 1971 May 28.64 1971 Dec 2.57	1200? full 635 empty cone-sphere	2.5 dia 1.1 long	Ejected onto impact path 1971 Dec 2.38 Soft-landed at 45°S, 158°W						
Mariner 9	1971-51A 1971 May 30.93 1971 Nov 14.02	520 empty octagon + 2 tanks	1.38 dia 2.29 long	1971 Nov 14.02 1971 Nov 16.12 1971 Dec 30	64.28 64.36 64.36	754 718 719	13051 12666 12679	1397 1397 1653	17916 17145 16915	0.633 0.622 0.602
Mars 5 orbiter	1973-49A 1973 Jul 25.79 1974 Feb 12.66	3440? full 2060? empty box + 2 vanes	2.3 dia 2.7 long	1974 Feb 12.66	35.0	1493	20525	1760	32500	0.749
Viking 1 orbiter	1975-75A 1975 Aug 20.89 1976 Jun 19	2325 full 950? empty box + 4 vanes	3.2 high 1.8 wide 1.5 deep	1976 Jun 19 1976 Jun 21	37.8? 37.8	2544 1476	29445 20444	1500 1514	50600 32583	0.834 0.760*
D Viking 1 lander	1975-75C 1975 Aug 20.89 1976 Jun 19 1976 Jul 20.51	1090 full 600 empty pyramid	2.1 high 3.0 wide 2.5 deep	1976 Jun 21	37.8	1476	20444	1514	32583	0.760

*82km flyby of Phobos on 20 Feb 1977; periapsis was lowered to 300km on 11 Mar 1977.

Table 2 (continued)

(E) Areocentric orbits - Earth launch (concluded)

Page 36a

Name	Launch date injection date and ejection date	Mass (kg) and basic shape	Basic size (m)	Date of orbital determination (UT)	Inclination to Mars equator (deg)	Period (min)	Semi major axis (km)	Periapsis Height (km)	Apoapsis height (km)	Orbital eccen- tricity
T Viking 2* orbiter	1975-83A 1975 Sep 9.78 1976 Aug 7.47	2325 full 9507 empty box + 4 vanes	3.3 high 1.8 wide 1.5 deep	1976 Aug 7.47 1976 Aug 25.76 1976 Aug 27 1976 Dec 20	55.6 55.6 55.4 80	1644.6 1438.8 1477.8 1590	22011 20132 20492 21520	1502 1432 1502 787	35728 32042 32692 35463	0.778 0.760 0.761** 0.806
D Viking 2 lander	1975-83C 1975 Sep 9.78 1976 Aug 7.47 1976 Sep 3.96	1090 full 600 empty pyramid	2.1 high 3.0 wide 2.5 deep	1976 Aug 27	55.4	1477.8	20492	1502	32692	0.761

*Passed 23km from Deimos on 1977 Oct 15.

**Orbital inclination changed to 75 deg on 1976 Sep 30, after manoeuvre.

Table 2 (concluded)

(F) Orbits round Venus - Earth launch

Name	Launch date injection date and ejection date	Mass (kg) and basic shape	Basic size (m)	Date of orbital determination (UT)	Inclination to Venus equator (deg)	Period (min)	Semi major axis (km)	Periapsis height (km)	Apoapsis height (km)	Orbital eccen- tricity
Venus 9 orbiter	1975-50A 1975 Jun 8.11 1975 Oct 22.17	3376 full cylinder + 2 panels	-	1975 Oct 22.17	34.17	2898	62910	1510	112200	0.879
Venus 10 orbiter	1975-54A 1975 Jun 14.13 1975 Oct 25.17	3473 full cylinder + 2 panels	-	1975 Oct 25.17	29.50	2963	63810	1620	113900	0.880

The radius of Venus is taken as 6052km.

AMENDMENTS TO RAE TECHNICAL REPORT 73006

Page 3 Add to Census of Space Vehicles table:-

Destination	Launches						Still in orbit on 1 Jan 1978*
	1973	1974	1975	1976	1977	1958-77	
Moon	2	2	0	1	0	64	19
Venus/Mercury	1	0	0	0	0	1	1
Venus	0	0	2	0	0	13	7
Mars	4	0	2	0	0	15	15
Jupiter	0	0	0	0	0	1	1
Jupiter/Saturn	1	0	0	0	2	3	3
Solar orbit	0	1	0	1	0	7	7
Total launches	8	3	4	2	2	104	-
Still in orbit on 1 Jan 1978*	7	2	4	1	2	-	53

*Those launches for which an instrumented payload was in orbit on 1 Jan 1978. (Rocket debris from 95 out of the 104 launches, 9 moon probes being excluded, remained in orbit on 1 Jan 1978.)

Country of origin for the 104 launches: USA 53; USSR 49; USA/FRG 2.

Page 7 Luna 1 rocket weight should be 1110kg.

Page 9 Venus 3 orbit should be 0.70 to 1.11 AU? 0.237 4.3 deg? 316 days?

Page 10 Pioneer 7 probably no longer transmits.

Page 10 Venus 4 orbit should be 0.70 to 1.11 AU? 0.237 4.3 deg? 316 days?

Page 10 Mariner 5 probably no longer transmits; add footnote - "Heliocentric orbit after Venus flyby."

Page 11 Pioneer 8: perihelion should be 0.990 AU, and inclination 0.06° . It probably no longer transmits.

Page 11 Pioneer 9 probably no longer transmits.

Page 12 Mariner 6 probably no longer transmits.

Page 12 Mariner 7 probably no longer transmits.

Page 13 Add Venus 7 capsule footnote - "Surface transmissions lasted 23 min."

Page 13 Add to Mars 3 footnote - "Surface transmissions lasted 20 sec."

Page 15 Small booster, 1959 Theta 3, weighs 157kg?

Page 21 Zond 4 weighs 5375kg?

Page 21 Zond 5 capsule weighs 2760kg?

Page 21 Zond 5 compartment weighs 2615kg?

Page 22 Zond 6 capsule weighs 2760kg?

Page 22 Zond 6 compartment weighs 2615kg?

AMENDMENTS (continued)

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- Page 23 LEM 5 (surface experiments) probably no longer transmits.
- Page 23 Zond 7 capsule weighs 2760kg?
- Page 23 Zond 7 compartment weighs 2615kg?
- Page 23 LEM 6 (surface experiments) probably no longer transmits.
- Page 24 Zond 8 capsule weighs 2760kg?
- Page 24 Zond 8 compartment weighs 2615kg?
- Page 25 LEM 8: amend to "I?".
- Page 26 LEM 10: amend to "I?".
- Page 26 Luna 18: add footnote - Damage on landing prevented re-launch of ascent stage.
- Page 27 Luna 20 rocket should be designated 1972-07B.
- Page 27 Apollo 16 command module weighs 5840kg.
- Page 27 Apollo 16 service module weighs 24518kg full.
- Page 27 Apollo 16 rocket weighs 13970kg.
- Page 27 LEM 11 landed on Moon at 8.99° S, 15.51° E.
- Page 27 Particles Subsatellite 2 impacted near 10.16° N, 111.94° E.
- Page 29 Luna 10 retrorocket launch date should be 1966 Mar 31.449
- Page 29 Explorer 35 probably no longer transmits.
- Page 30 Luna 14 basic shape confirmed.
- Page 31 Luna 17 empty weight is 1836kg.
- Page 32 Particles Subsatellite 1 probably no longer transmits. Delete 1973 ejection date.
- Page 32 Luna 19 probably no longer transmits. Basic shape, width and depth confirmed. Height should be 2.3m. Third orbital inclination should be 40.68 deg.
- Page 32 Luna 20 empty weight is 1880kg. (On Moon, including ascent stage).
- Page 32 Delete Luna 20 fragment, 1972-07B.
- Page 34 Luna 16 ascent stage weight should be 800kg? full.

Page 37 Add to Index:-

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AMENDMENTS (concluded)

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Voyager 2	1977-76	14d	-

REPORT DOCUMENTATION PAGE

Overall security classification of this page

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As far as possible this page should contain only unclassified information. If it is necessary to enter classified information, the box above must be marked to indicate the classification, e.g. Restricted, Confidential or Secret.

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16. Descriptors (Keywords) (Descriptors marked * are selected from TEST) Space vehicles. Space vehicle orbits.			
17. Abstract RAE Technical Report 73006 lists all space vehicles successfully launched before 31 December 1972. This Memorandum, which supersedes Technical Memorandum Space 242, extends the tabulation to the end of 1977, and gives amendments to Technical Report 73006.			

ABSTRACT