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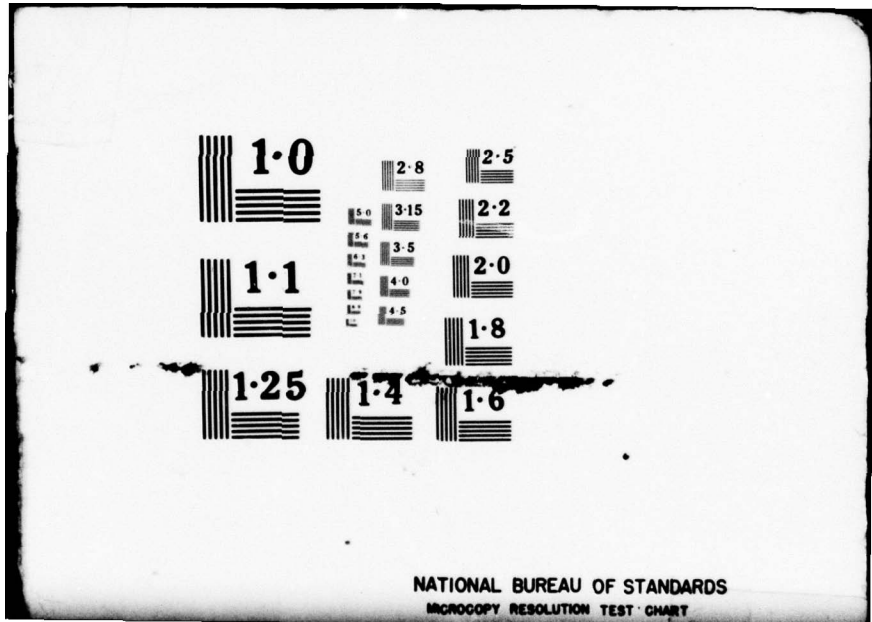
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THESIS

ADAPTATION OF THE IMPROVED ANTI-AIRCRAFT
ARTILLERY SIMULATION COMPUTER PROGRAM (P001)
FOR USE AT THE NAVAL POSTGRADUATE SCHOOL
IN AIRCRAFT COMBAT SURVIVABILITY STUDIES

by

Carl Frederick Swenson

March 1978

Thesis Advisor:

R. E. Ball

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A complete P001/PIP package and user's guide for an aircraft attrition study in the NPS Course AE 3251, Aircraft Combat Survivability, are presented.

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⑥ Adaptation of the Improved
Antiaircraft Artillery Simulation Computer Program (Pp/p1)
for Use at the Naval Postgraduate School
in Aircraft Combat Survivability Studies.

by

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Submitted in partial fulfillment of the
requirements for the degree of

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ABSTRACT

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A complete P001/PIP package and user's guide for an aircraft attrition study in the NPS Course AE 3251, Aircraft Combat Survivability, are presented. ↗

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I. INTRODUCTION

The Air Force Armament Laboratory (AFATL) has developed an antiaircraft artillery (AAA) simulation computer program called P001 which is the present standard program for conducting survivability assessments of aircraft in a hostile AAA environment. P001 is used throughout the aircraft industry and is the aircraft attrition program required by the Department of the Navy MIL-STANDARD-2072(AS), SURVIVABILITY, AIRCRAFT; ESTABLISHMENT AND CONDUCT OF PROGRAMS FOR, August 1977.

Briefly, P001 computes the probability of kill of a target aircraft flying a user-input flight path, as a result of its being fired upon by user-selected antiaircraft artillery located at user-input locations. The technique used by P001 to accomplish this task involves:

- Computation of an aim point with consideration of the errors that can arise therein.
- Simulation of the firing process and the sources of error in the firing process.
- Combination of all the effects of random error into one total projectile trajectory distribution.
- Location of the user-input vulnerable area of the aircraft within the total trajectory distribution.
- Computation of the probability of kill.

P001 has been used in NPS Course AE 3251, Aircraft Combat Survivability, to illustrate the interaction of the various

elements that comprise the aircraft combat survivability problem in a hostile AAA environment. The scenario consists of a typical Naval aircraft on an attack mission. The aircraft's target is the bridge shown in Fig. 1. The student must select a flight path to the target and also the location of the defending AAA. P001 is used to determine the probability of survival of the aircraft.

Use of P001 as an educational tool in aircraft combat survivability studies is very effective since it requires a knowledge of the techniques for calculating aircraft vulnerable areas, as well as the basics of the interaction between the threat, the environment and the target aircraft. Some of the interaction parameters include aircraft vulnerable area, speed, altitude, location and aspect angle with respect to the threat, and aircraft maneuver characteristics; the effect of terrain, target altitude and range on projectile performance; and the antiaircraft artillery threat envelope.

The input to P001 requires many time consuming, tedious computations and a significant amount of keypunching, a use of time that does not profitably contribute to the aircraft combat survivability learning experience. In addition, the realism of the input data has a significant effect on the validity of the result and, up to now, it has not been possible to evaluate input data accuracy. Consequently, a preprocessor computer program that would significantly reduce the time required for a student to prepare the input data, as well as provide an indication as to the realism of the input data, is very desirable.

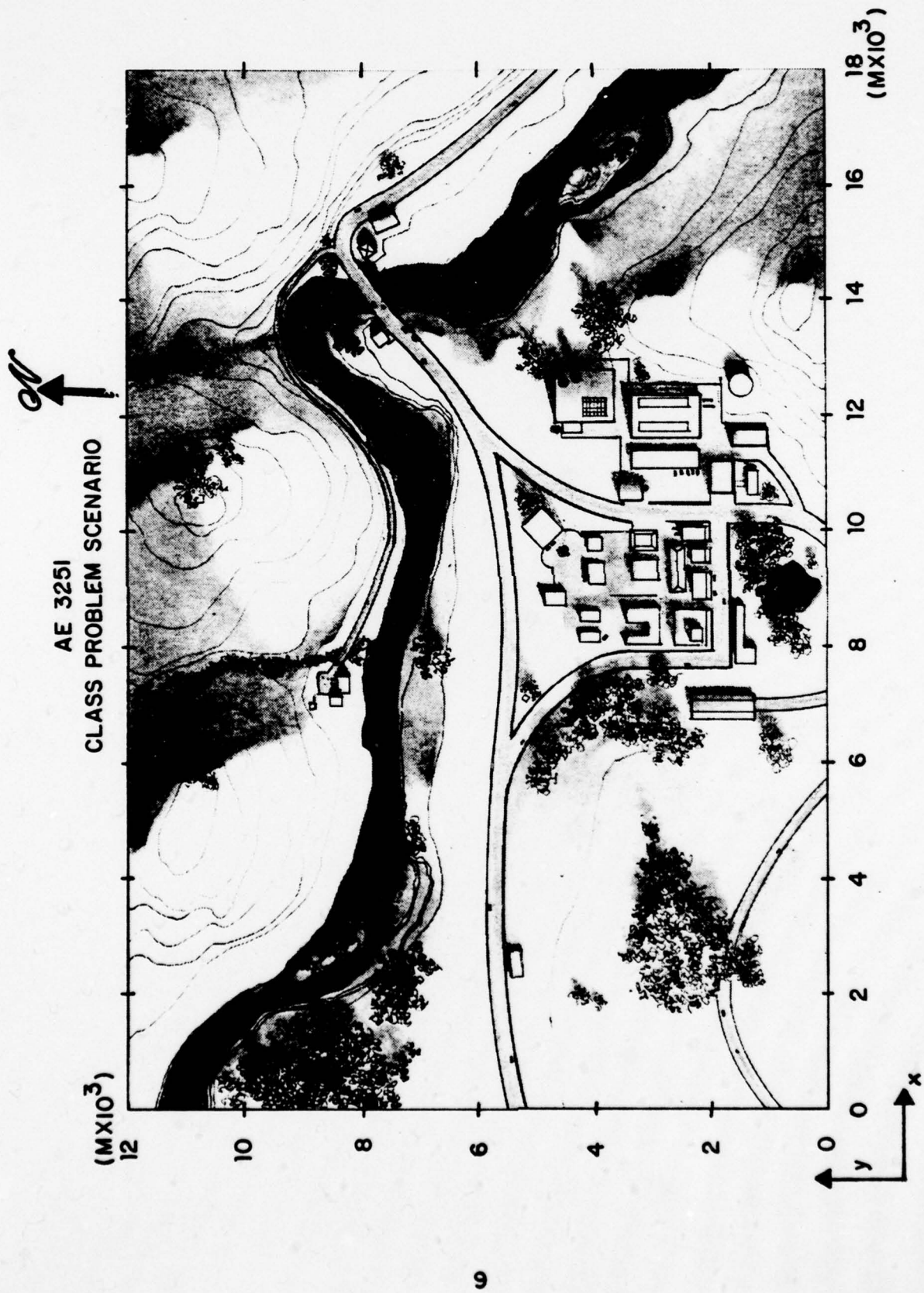


Figure 1

While the preprocessor program was being developed for P001, a version of P001 that was significantly modified by Calspan was obtained from the Air Force Flight Dynamics Laboratory. This improved version of P001 is capable of including the effects of self-contained airborne electronic countermeasures (ECM) on the acquisition/tracking process, of detection system anti-jam capabilities, and of radar beam multipath characteristics. The output of P001 was also expanded. The improved P001 was written for a Control Data Corporation computer system.

The problem solved by this thesis effort is two-fold:

- The adaptation of the improved P001 to the NPS IBM 360/65.
- The development of a P001 input program (PIP) to reduce student time required to prepare the input data required by P001 and to provide an indication of the realism of the input data.

The main body of this thesis describes the adaptation of the improved P001 to the IBM 360/65 and the details of the development of PIP. Appendix A contains a complete package for a problem in aircraft attrition to be used in AE 3251.

The following sources were heavily relied upon for information throughout the entire thesis development process:

- Antiaircraft Artillery Simulation Computer Program - AFATL Program P001 Vol. I User Manual, Air Force Armament Laboratory, Eglin AFB, Florida (Joint Aircraft Attrition Program Advanced Planning Group, September 1973).

- Antiaircraft Artillery Simulation Computer Program -
Program P001 Program Update, Joint Technical Coordinating
Group for Aircraft Survivability (Survivability Assessment
Subgroup, April 1976).

- M. E. Ramaccia, ATS Working Paper No. 9, Calspan Modi-
fication to Antiaircraft Artillery Simulation, AFATL Program
P001 (Calspan Corporation, Buffalo, New York, 11 August 1977).

- G. Gary Maxwell, The Development of Class Problems for
a Course in Aircraft Combat Survivability (Naval Postgraduate
School Master's Thesis, 1978).

II. APPROACH

A. ADAPTATION OF THE IMPROVED P001 TO THE NPS IBM 360/65

The adaptation of P001 from a CDC computer to the NPS IBM computer involved several hundred computer-unique alterations; that is, changes that had to be made due to the inherent differences between the CDC and IBM computer systems. These changes involved:

- Elimination of the FTNBIN, SECOND, TIMREM and DATE subroutines contained in the CDC system, but not available on the IBM 360 system.

- Substitution of the IBM "REREAD" command for the CDC "DECODE" function.

- Substitution of the IBM "A4" format field descriptor for the CDC "A10" descriptor.

- Substitution of the IBM "" format delimiter for the CDC "*" delimiter.

- Substitution of IBM-NPS job control language (JCL) cards to accomplish the required input, output, tape usage and core size requirements.

- Extensive reconstruction of the P001 main program by creating the subroutine TOOBIG using the input, output and exits sections of the P001 main program. This permitted a scalar map size small enough for the IBM system to compile and thus avoided the "ROLL SIZE EXCEEDED" error given by the NPS compiler.

B. DEVELOPMENT OF A PREPROCESSOR FOR P001 (PIP)

In the early stages of AE 3251 it is not desirable for the student to become heavily involved in the many and various facets and options of the P001 program. For this reason, a simple preprocessor computer program to P001 that will provide punched cards for all of the input in the proper format, requiring only a minimum of student involvement, has been developed. For example, the Calspan-modified P001 program requires an input of 196 cards with over 1,400 entries and 860 calculations for a 50 milestone flight scenario involving 7 ground weapons of 4 threat classes. On the other hand, PIP requires an input of 59 cards with 160 entries and no calculations for the same 50 milestone flight scenario and automatically punches the output that fulfills all P001 input requirements in the proper format and order. This significantly reduces the student involvement in the input process.

PIP requires only that the student provide the X, Y and Z coordinates of each of the aircraft flight path milestones as input data, avoiding the many tedious calculations required by full P001 input. In addition, the preprocessor checks the flight path to determine if it exceeds any aircraft performance limits or scenario guidelines. The aircraft built into PIP has performance characteristics approximating those of the A-7 Corsair. The cruise velocity, climb/dive schedules, acceleration/deceleration schedules, dirty/clean velocity limitations, stall velocity and "G" loading limitation values are only representative quantities and are not intended to

accurately describe the performance characteristics of an actual A-7 since the intent of PIP is only to provide representative flight characteristics that demonstrate the principles of aircraft combat survivability in an AAA scenario.

C. FLIGHT PATH PREPROCESSING

1. FORTRAN Distance Equations

The three cartesian milestone distance components, DX, DY and DZ, are calculated in PIP by taking the difference between the X, Y and Z components of adjacent milestone locations.

Milestone distance component FORTRAN equations:

$$DX = X(I+1) - X(I)$$

$$DY = Y(I+1) - Y(I)$$

$$DZ = Z(I+1) - Z(I)$$

where,

I = ith milestone

X = milestone x-coordinate location

Y = milestone y-coordinate location

Z = milestone z-coordinate location

DX = milestone x-coordinate distance difference

DY = milestone y-coordinate distance difference

DZ = milestone z-coordinate distance difference.

2. Milestone Distance Equations

The distance between successive milestones is the square root of the sum of the squares of the milestone distance components.

Milestone distance FORTRAN equations:

$$DX2 = DX**2$$

$$DY2 = DY**2$$

$$DZ2 = DZ**2$$

$$DIST = SQRT(DX2 + DY2 + DZ2)$$

where,

DIST = milestone distance

3. Heading and Climb Angle Equations

The aircraft heading and climb angle between milestones are calculated in PIP using standard geometrical considerations based on the relative locations of the X, Y and Z coordinates of adjacent flight path milestones.

Aircraft heading FORTRAN equation:

$$HDG(I+1) = ATAN2(DY,DX)$$

where,

HDG = aircraft heading

Aircraft climb angle FORTRAN equation:

$$CA(I) = ATAN2(DZ, SQRT(DX2 + DY2))$$

where,

CA = aircraft climb angle

4. Aircraft Speed Equations

The aircraft is assigned a cruise speed of between 206 and 257 meters per second (400 and 500 knots, respectively) at milestone 1 by the user. Aircraft speed at successive locations is calculated based on the altitude change between milestones, since an increase/decrease in altitude decreases/increases aircraft speed proportionately, and from a schedule

based on the present aircraft speed as compared with the initially assigned cruise speed. If the aircraft speed is found to be less/more than the assigned cruise speed, a slow acceleration/deceleration to the assigned cruise speed is assumed. The X, Y and Z components of the velocity at each milestone are then calculated based on the heading and climb angle at the milestone.

Aircraft velocity FORTRAN equations:

$$\begin{aligned} \text{VEL}(1) &= \text{CVEL} \\ \text{VEL}(I+1) &= \text{VEL}(I) - \text{TAN}(\text{CA}(I)) * \text{DIST} / 100 \\ &\quad + (\text{CVEL} - \text{VEL}(I)) * (\text{DIST} / \text{VEL}(I)) / 30 \\ \text{VAVG} &= (\text{VEL}(I) + \text{VEL}(I+1)) / 2 \end{aligned}$$

where,

CVEL = aircraft cruise speed
VEL = aircraft velocity
VAVG = average aircraft velocity

Aircraft velocity component FORTRAN equations:

$$\begin{aligned} \text{XYVEL} &= \text{VEL}(I) * \text{COS}(\text{CA}(I)) \\ \text{XDOT}(I) &= \text{XYVEL} * \text{COS}(\text{HDG}(I)) \\ \text{YDOT}(I) &= \text{XYVEL} * \text{SIN}(\text{HDG}(I)) \\ \text{ZDOT}(I) &= \text{VEL}(I) * \text{SIN}(\text{CA}(I)) \end{aligned}$$

where,

XYVEL = horizontal velocity component
XDOT = x-coordinate velocity component
YDOT = y-coordinate velocity component
ZDOT = z-coordinate velocity component.

5. Flight Time Equations

The time interval between flight path milestones is calculated by dividing the distance between the milestones by the average velocity between the milestones. The individual milestone time intervals are summed to provide the total time for the scenario.

Flight time FORTRAN equations:

$$T(I+1) = T(I) + \text{DIST}/\text{VAVG}$$

$$\text{DT}(I) = T(I) - T(I-1)$$

where,

T = flight time (cumulative)

DT= flight time between milestones (I) and (I+1)

6. Turn Rate, Roll Angle, "G" Loading

The turn rate required between milestones is calculated from the heading change and the time interval between the milestones. This turn rate and the aircraft speed are used to calculate the "G" loading on the aircraft caused by the turn. The roll angle required for a level turn is calculated from the turn rate and the aircraft speed.

Turn angle FORTRAN equation:

$$\text{TNANG} = \text{HDG}(I+1) - \text{HDG}(I)$$

where,

TNANG = milestone turn angle.

Turn rate FORTRAN equation:

$$\text{TNRT}(I) = \text{TNANG}/\text{DT}(I)$$

where,

TNRT = milestone turn rate.

Roll angle FORTRAN equation:

$$RA(I) = ATAN(TNRT(I)*VAVG/9.81)$$

where,

RA = milestone aircraft roll angle.

"G" loading FORTRAN equations:

$$ABSRT(I) = ABS(TNRT(I))$$

$$G(I) = SQRT(ABSRT(I)**2*VEL(I)**2/9.81**2 + 1)$$

where,

ABSRT = absolute value of turn rate

G = "G" loading

7. P001 Stored Time Increments

The equal time increments between successive "P001 stored" positions along the flight path (not milestones) is equal to the total scenario flight time (T) divided by 1000. This time increment is required in the P001 input data card 02.

8. Probability of Kill Accumulation Periods

The total scenario flight time (T) is divided into ten equal time segments in which the probability of kill will be computed for each segment. These ten increments are required on P001 input card 06.

The values computed above are printed and punched on cards that can be used as part of the P001 input data. All data are punched in the specified order to be input to the P001 program.

D. AIRCRAFT PERFORMANCE LIMITATIONS,
FLIGHT PATH REQUIREMENTS AND ERROR MESSAGES

PIP provides several checks on the performance requirements of the aircraft as it traverses the user-input flight path. It also checks the bombing run portion of the flight path to see if it satisfies the requirements for successful bomb drop. These checks are as follows:

1. Cruise Speed

The aircraft cruise speed is initially input by the user at a value between 206 and 257 meters per second and is changed as dictated by altitude changes and the acceleration/deceleration schedule presented in II.C. If the aircraft is assigned a cruise speed outside of the range from 206 to 257 meters per second, the following cruise speed error message is generated:

Error message: "CRUISE SPEED IS ____ METERS PER SECOND WHICH IS NOT WITHIN THE GIVEN LIMITS OF BETWEEN 206 AND 257 METERS PER SECOND."

2. Stall Speed

If the aircraft speed falls below 90 meters per second (175 knots), the following stall error message is generated, identifying the error, the milestone and the velocity value causing the error message:

Error message: "MILESTONE ____ VELOCITY IS ____ METERS PER SECOND. STALL OCCURS AT 90 METERS PER SECOND. DECREASE THE CLIMB ANGLE PRIOR TO MILESTONE ____."

3. "Red Line"

Prior to the bomb release point, the aircraft is "drag limited" to 260 meters per second (500 knots). After ordnance release, the drag limitation eases to permit a speed of 310 meters per second (600 knots).

4. "G" Loading

If the maximum "G" loading of 6 is exceeded, the following turn rate error message is generated, identifying the error, the milestone, the "G" loading and turn rate that caused the error message and denoting the corrective turn angle which will eliminate the error:

Error message: "MILESTONE ___ TURN RESULTS IN A TURN RATE OF ___ DEGREES PER SECOND WHICH RESULTS IN A G LOADING OF ___ WHICH IS IN EXCESS OF THE 6 G MAX LOADING. DECREASE THE TURN ANGLE AT MILESTONE ___ TO BELOW ___ DEGREES."

5. Minimum Altitude

If the aircraft descends to an altitude less than 61 meters prior to bomb release, the following error message is generated identifying the error, the milestone and the milestone altitude that caused the error:

Error message: "ALTITUDE AT MILESTONE ___ IS ___ METERS WHICH IS BELOW THE MIN ALT OF 61 METERS."

6. Maximum Altitude

If the aircraft attains an altitude greater than 457 meters prior to the "pop-up" maneuver, the following error message is generated identifying the error, the milestone and the milestone altitude that caused the error:

Error message: "ALTITUDE AT MILESTONE ___ IS ___ METERS WHICH IS ABOVE THE MAX ALT PRIOR TO POP UP OF 457 METERS."

7. Overall Maximum Altitude

If, at any time, the aircraft exceeds the overall maximum altitude of 2134 meters, the following error message is generated identifying the error, the milestone and the milestone altitude that caused the error:

Error message: "ALTITUDE AT MILESTONE ___ IS ___ METERS WHICH IS ABOVE THE MAX ALT OF 2134 METERS."

8. Minimum "pop up" Altitude

If, during the "pop up" maneuver, the aircraft fails to attain a minimum altitude of 1219 meters, the following error message is generated identifying the error and the altitude attained during the "pop up" maneuver:

Error message: "MAX ALTITUDE DURING POP UP WAS ___ METERS WHICH IS LESS THAN THE MINIMUM POP UP ALTITUDE OF 1219 METERS."

9. Bomb Release Heading

If the aircraft heading at the time of bomb release is greater than 5 degrees from the true heading to the target, the following error message is generated identifying the error, the aircraft heading and the true heading to the target at the time of bomb release:

Error message: "THE AIRCRAFT HEADING INTO THE BOMB RELEASE POINT IS _____. THE HEADING TO THE TARGET IS _____."

THE HEADING DIFFERENCE IS _____ WHICH IS GREATER THAN THE 5 DEGREE MAXIMUM DIFFERENCE LIMIT."

10. Target Acquisition Time

If the aircraft does not hold a heading of less than 5 degrees from the true heading to the target for a time period of at least 2 seconds on the leg immediately prior to the bomb release point, the following error message is generated identifying the error and the time duration of the leg that caused the error:

Error message: "THE LENGTH OF THE LEG IMMEDIATELY PRIOR TO THE BOMB RELEASE POINT IS _____ SECONDS IN DURATION WHICH IS LESS THAN THE MINIMUM OF 2 SECONDS."

11. Bomb Release Altitude

If the ordnance is released outside of an altitude envelope of from 305 to 914 meters, the following error message is generated identifying the error and the altitude at bomb release:

Error message: "THE BOMB RELEASE ALTITUDE IS _____ METERS WHICH IS NOT IN THE BOMB RELEASE ALTITUDE RANGE OF BETWEEN 305 TO 914 METERS."

12. Bomb Release Range

If the ordnance is released at a distance greater than 1000 meters from the target, the following error message is generated identifying the error and the distance from the target at the time of bomb release:

Error message: "THE BOMB WAS RELEASED AT A DISTANCE OF _____ METERS FROM THE TARGET WHICH IS IN EXCESS OF THE 1000 METER MAXIMUM BOMB RELEASE RANGE."

13. Gun Location Input Error

If the option is chosen to input the gun locations, but no gun location information is part of the input data or not all of the gun locations are specified, the following error message is generated identifying the error. Program execution terminates after the error message is printed.

Error message: "GUN EMPLACEMENT DATA WAS SPECIFIED AS PART OF THE INPUT DATA; HOWEVER, EITHER NO GUN EMPLACEMENT DATA IS PART OF THE INPUT OR ALL SIX GUN LOCATIONS WERE NOT SPECIFIED. EXECUTION TERMINATES."

14. Anti-jam Error

If the anti-jam option is specified, but no jammer is in operation, the following error message will be generated and the anti-jam function will be "turned off":

Error message: "THE ANTI-JAM FUNCTION IS SPECIFIED; HOWEVER, THE JAM FUNCTION IS NOT SPECIFIED. THE ANTI-JAM FEATURE HAS BEEN TURNED OFF."

15. Type 3 Gun Range Error

If a type 3 weapon is located within 3,000 meters of the center of the bridge, the following error message will be generated, identifying the error, the position of the gun that caused the error and the actual distance from the target of the gun:

Error message: "GUN TYPE 3 LOCATED AT X: ____ Y: ____ IS ____ METERS FROM THE TARGET WHICH IS LESS THAN THE MINIMUM DISTANCE OF 3000 METERS."

16. Zero Power Jammer Error

If the jamming function has been specified, but the jammer has been assigned a power of zero, the jammer function is "turned off" and the following error message is generated, identifying the error and the fact that the jammer has been "turned off":

Error message: "THE JAM FUNCTION IS SPECIFIED, BUT THE JAMMER POWER IS SPECIFIED AS ZERO. THE JAM FUNCTION HAS BEEN TURNED OFF."

17. Maximum Power Jammer Error

If the jammer has been assigned a power greater than 1000 watts, the following error message is generated, identifying the error. The jammer power will then be limited to 1000 watts.

Error message: "THE SPECIFIED JAMMER POWER IS GREATER THAN 1000 WATTS AND HAS BEEN LIMITED TO 1000 WATTS."

E. PROGRAM OPTIONS

PIP provides the following electronic warfare options:

1. Jamming Option

An airborne jammer aboard the target aircraft is utilized to degrade radar acquisition/tracking capabilities.

2. Anti-jam Option

Ground weapons that have an anti-jam capability use it to partially nullify the effects of the airborne jammer.

3. Multipath Option

The performance of all radar units which are susceptible to multipath effects is appropriately degraded.

PIP also provides for any combination of the following input/output options:

4. List the P001 Input Deck

A complete listing of all required cards for input to the P001 program is provided. The green "JOB" card and the orange final "EOF" card are not part of this listing. These two cards are the only cards that must be provided by the student to run the P001 program with the PIP output.

5. Punch the P001 Input Deck

A complete punched input deck in the proper format and order to run P001 is provided by this option. Again, no "JOB" or "EOF" card is provided.

6. Plot the P001 Scenario

A plot of the basic geographical features of the scenario, the aircraft flight path and milestone locations, the bomb release point, the gun emplacement locations and the threat radius for each weapon (coded as to weapon type) are provided by this option. Fig. 2 shows a typical PIP scenario plot.

7. Extended Printout Option

An extended printout of the results of the P001 analysis will be provided as output.

8. Gun Location Option

The locations of six of the seven guns in the scenario may be input to the program or preset gun locations may be utilized, as desired.

AE 3251 P001 SCENARIO

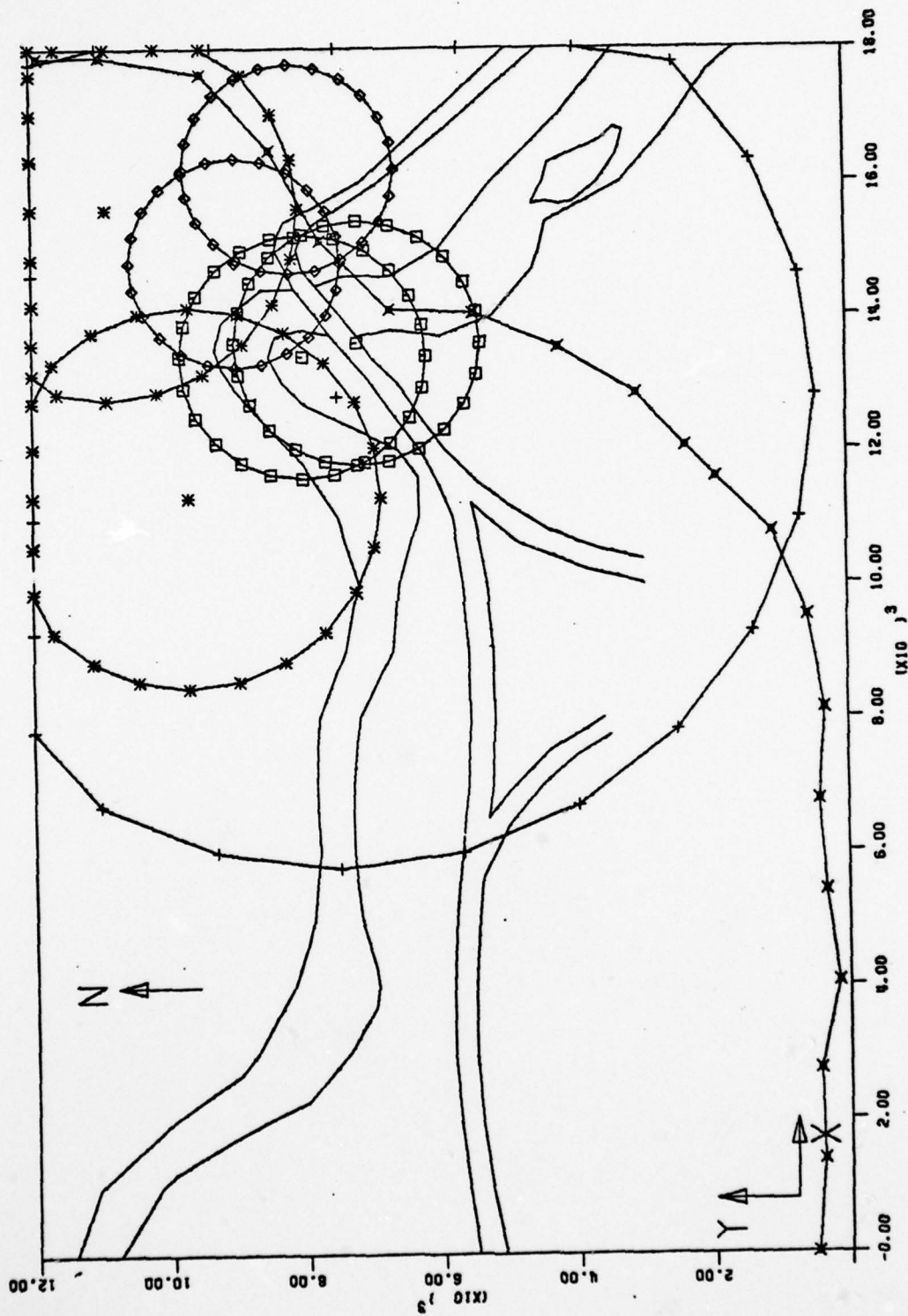


Figure 2 Typical PIP Scenario Plot

In all cases, messages are generated specifying which options were or were not chosen for each execution of PIP.

F. PIP INPUT DATA DEFAULT VALUES

In order to significantly reduce student involvement in the P001 input process, PIP assigns predetermined values to many of the options available under the full P001 input. These default values, over which the student has little or no control, were chosen to provide values that are representative of the typical attack situation simulated by the class problem scenario. The following is a list of these default values as they pertain to each P001 input card:

01 Card: Output Header Information

The output header information is assigned "Aircraft Combat Survivability Scenario" by PIP.

02 Card: Initial Flight Path Data

JMODE = 0: The milestone data are read from cards for one milestone at a time.

KMODE = 12: Flight path stored position data will be printed at every 12th position along the flight path.

TMIN = 0: The time at the beginning of the flight path is assigned a value of zero.

TMAX: The time at the end of the flight path is computed by PIP.

DTFPA: The time increment between successive stored positions along the flight path is calculated by PIP as TMAX/1000.

$XR, YR = 0$: An x, y reference location in the Flight Path Coordinate System. XR and YR are coordinates in the Flight Path Coordinate System of the point located at XT, YT and ZT in the General Reference Coordinate System, as shown in Fig. 3.

$XT, YT = 0$: The x, y coordinates in the General Reference Coordinate System of the point located at XR, YR in the Flight Path Coordinate System, as shown in Fig. 3.¹

$PSI = 0$: The rotational angle required to rotate the Flight Path Coordinate System into the General Reference Coordinate System (positive for counter-clockwise rotation).

$ZT = 0$: Vertical correction factor to be added to each point of the flight path, as shown in Fig. 3.¹

2A Card: Flight Path Milestone Input

All data on Card 2A is calculated by PIP based on the cruise speed and milestone coordinates provided by student input.

03 Card: Ground Weapon Complex Coordinates

If the preset weapon location option is chosen, the seven ground weapons used in the class scenario are assigned in the following locations by PIP:

¹ Setting XR, YR, XT, YT, ZT and PSI equal to zero results in the coincidence of the Flight Path Coordinate System and the General Reference Coordinate System.

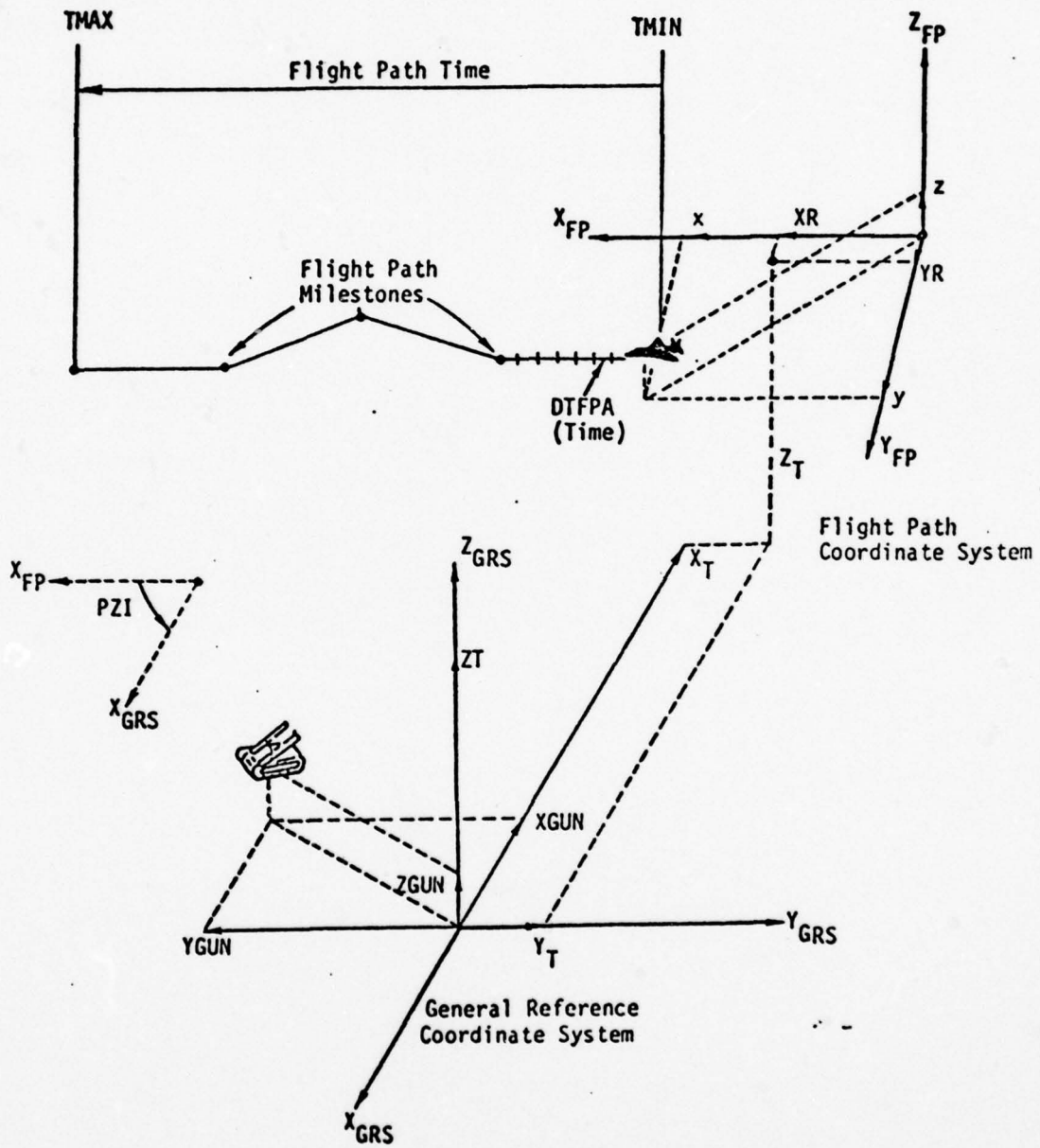


Figure 3 Relationship of the General Reference and Flight Path Coordinate Systems

<u>Gun</u>	<u>XGUN</u>	<u>YGUN</u>	<u>ZGUN</u>
1	14,800	9,000	40
2	16,200	8,200	40
3	13,600	7,200	20
4	13,400	8,000	20
5	11,300	9,700	50
6	15,600	10,900	90
7	12,800	7,500	20

where XGUN, YGUN and ZGUN are as shown in Fig. 3. Fig. 2 shows the weapon locations.

If the preset weapon location option is not chosen, all values on Card 03 are assigned by student input.

04 Card: Ground Weapon Characteristics

Each ground weapon is defined by six parameters. These parameters are:

IGT: Ground weapon type.

IEM: Fire control operation mode.

ICB: Number of barrels of the ground weapon to be fired in a simultaneous manner, where $ICB \times ISB$ is the number of barrels per weapon.

IGL: Number of ground weapons located in the ground weapon complex. ($ICB \times ISB \times IGL$ is the number of barrels at one location.)

CIRCLE: Radius of the circle of the ground weapon complex. If there is only weapon in the ground weapon complex, CIRCLE = 0.0.

The default values selected for the six parameters for each of the seven weapons are as follows:

	<u>IGT</u>	<u>IEM</u>	<u>ICB</u>	<u>ISB</u>	<u>IGL</u>	<u>CIRCLE</u>
Gun 1:	1	1	1	1	1	0.0
Gun 2:	1	1	1	1	1	0.0
Gun 3:	2	1	1	1	1	0.0
Gun 4:	2	1	1	1	1	0.0
Gun 5:	3	4	4	1	1	0.0
Gun 6:	3	3	4	1	1	0.0
Gun 7:	5	3	2	1	1	0.0

05 Card: Ground Weapon Complex Density Factors

IF5 = 0: Ground weapon complex density factors are not printed.

NRHOS = 1: The number of ground weapon density factors equals one.

RHO(1) = 1.0; RHO(2) through RHO(9) = 0.0: Generally, RHO = (number of possible ground weapon complexes in the engagement divided by the number of possible ground weapon complex locations in the scenario).

06 Card: Flight Path P_k Accrual Time Intervals (P_k = probability of kill)

IF6 = 1: Flight path P_k accrual time intervals are printed.

NTINTS = 9: One less than the total number of flight path time intervals to be considered. P001 adds one additional time interval for P_k 's accumulated from NTINTS to infinity.

TINTER(1) through TINTER(9): TINTER values are assigned by PIP. Each TINTER is an increment representing 1/10th of the total flight path time.

07 Card: Aircraft Vulnerable Area Table Title

ICARD = a. "Vulnerable Area Table vs Type 1 and 2 Weapons",
b. "Vulnerable Area Table vs Type 3 Weapons",
c. "Vulnerable Area Table vs Type 5 Weapons"

7A Card: Aircraft Vulnerable Area Tables

The values assigned by PIP to the three aircraft vulnerable area tables representing the three scenario threat classes are given in Tables I, II and III respectively.

08 Card: Ground Weapon Reaction and Tracking Times

Card 08 is omitted by PIP. Values are assigned within P001.

09 Card: Ground Weapon Parameters

Card 09 is omitted by PIP. Values are assigned within P001.

9A Card: Ground Weapon Parameters

Card 9A is omitted by PIP. Values are assigned within P001.

10 Card: Ground Weapon Projectile Parameters

Card 10 is omitted by PIP. Values are assigned within P001.

11 Card: Logical Unit Input Option

Card 11 is omitted by PIP. Logical Unit 5 is assigned for input within P001.

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07	VULNERABLE AREA	TABLE	VS TYPE	1 ANC 2 WEAPONS
0.464	0.464	7.107	7.107	7.107
0.657	0.657	5.574	5.574	5.574
0.697	0.697	7.357	7.357	7.357
0.697	0.697	5.574	5.574	5.574
0.657	0.657	5.574	5.574	5.574
0.697	0.697	7.357	7.357	7.357
0.657	0.657	5.574	5.574	5.574
0.464	0.464	0.743	0.743	0.743
0.657	0.657	2.858	2.858	2.858
0.464	0.464	3.298	3.298	3.298
0.657	0.657	2.858	2.858	2.858
0.657	0.657	0.743	0.743	0.743
0.464	0.464	2.858	2.858	2.858
0.657	0.657	3.298	3.298	3.298
0.657	0.657	2.858	2.858	2.858
0.697	0.697	5.574	5.574	5.574
0.657	0.657	7.357	7.357	7.357
0.697	0.697	5.574	5.574	5.574
0.697	0.697	5.574	5.574	5.574
0.697	0.697	7.357	7.357	7.357
0.464	0.464	7.107	7.107	7.107

TABLE I. Aircraft Vulnerable Area Table (Type 1 and 2 Weapons)

07	VULNERABLE	AREA	TABLE	VS	TYPE	3 WEAPONS	VS	TYPE	3 WEAPONS	VS	TYPE	3 WEAPONS
12.540	12.540	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470
19.853	19.853	10.510	10.510	10.510	10.510	10.510	10.510	10.510	10.510	10.510	10.510	10.510
5.639	9.639	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780
12.643	12.643	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510
9.853	9.853	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780
12.643	12.643	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510
1.394	1.394	11.394	11.394	11.394	11.394	11.394	11.394	11.394	11.394	11.394	11.394	11.394
4.762	4.762	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240
5.342	5.342	7.432	7.432	7.432	7.432	7.432	7.432	7.432	7.432	7.432	7.432	7.432
4.762	4.762	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240
4.762	4.762	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240
4.762	4.762	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240
9.853	9.853	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240
12.643	12.643	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510
9.853	9.853	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780
9.853	9.853	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510
12.643	12.643	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780
19.639	19.639	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510	11.510
12.540	12.540	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470

TABLE II. Aircraft Vulnerable Area Table (Type 3 Weapons)

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07	VULNERABLE AREA	TABLE	VS TYPE	WEAPON	WEAPCN
55	55.370	55.370	55.370	55.370	55.370
43	43.220	43.220	43.220	43.220	43.220
47	47.100	47.100	47.100	47.100	47.100
62	62.530	62.530	62.530	62.530	62.530
47	47.100	47.100	47.100	47.100	47.100
43	43.220	43.220	43.220	43.220	43.220
47	47.100	47.100	47.100	47.100	47.100
62	62.530	62.530	62.530	62.530	62.530
47	47.100	47.100	47.100	47.100	47.100
5	5.761	5.761	5.761	5.761	5.761
27	27.450	27.450	27.450	27.450	27.450
33	33.070	33.070	33.070	33.070	33.070
27	27.450	27.450	27.450	27.450	27.450
5	5.761	5.761	5.761	5.761	5.761
27	27.450	27.450	27.450	27.450	27.450
33	33.070	33.070	33.070	33.070	33.070
27	27.450	27.450	27.450	27.450	27.450
5	5.761	5.761	5.761	5.761	5.761
47	47.100	47.100	47.100	47.100	47.100
62	62.530	62.530	62.530	62.530	62.530
43	43.220	43.220	43.220	43.220	43.220
47	47.100	47.100	47.100	47.100	47.100
62	62.530	62.530	62.530	62.530	62.530
47	47.100	47.100	47.100	47.100	47.100
55	55.370	55.370	55.370	55.370	55.370

TABLE III. Aircraft Vulnerable Area Table (Type 5 Weapon)

13 Card: Radar Multipath Parameters

IMUL: Assigned by the user on the PIP option card. If IMUL = 0, no multipath effects are considered. If IMUL = 1, multipath radar effects are taken into consideration in the P001 analysis.

IRMP: Radar type identification assigned by PIP, specifying the tracking radar. The value of IRMP indicates appropriate radar parameters within P001. The value of IRECM, assigned by PIP on Card 14, dictates the value of IRMP. The radar parameters and the relationship between IRMP and IRECM are as follows:

	<u>IRECM</u>	<u>IRMP</u>	<u>Beamwidth (deg)</u>	<u>Squint Angle (deg)</u>	<u>Calibration Constant</u>
Gun 6:	1	1	1.4	0.5	0.759
Gun 7:	2,3	2	1.8	0.6	1.060

Multipath radar effects do not apply to guns 1 through 5.

REFC = 0.35: Reflection coefficient. 0.35 is a typical value for terrain with vegetation.

14 Card: ECM (Jamming) Parameters

IJAM: Assigned by the user on the PIP option card. 0 = no jamming. 1 = jamming effects considered in P001 analysis.

IP = 5: Print every 5th value in J/S printout.

IJ = 0. Therefore, GAINJ is the antenna gain of the jammer.

GAINJ = 1.0: The antenna gain of the jammer is 1.0.

PJW: Jammer power, assigned by the user on the PIP option card.

PLEN = 1.0E-06: The length of the jammer cover pulse is 1 microsecond, a standard value.

IX = 1: A radar cross section table is provided by PIP. (Table IV).

XSEC = 0: XSEC is not used if IX = 1. If used (IX = 0), a constant cross section of XSEC m² is used.

CALX = 1: The radar cross section table is not scaled.

IREFM: The value of IREFM defaulted by PIP depends upon the gun type and mode. IREFM calls up certain radar parameters from a data statement within P001. IREFM values and the relationship with the gun type and mode are as follows:

<u>Gun Type</u> (IGT)	<u>Mode</u> (IEM)	<u>Radar ID</u> (IREFM)	<u>Gain</u> (RGDB)	<u>Power</u> (PRW)	<u>Frequency</u> (FREQ)	<u>SJTMAX</u>
1	1	N/A	N/A	N/A	N/A	N/A
2	1	N/A	N/A	N/A	N/A	N/A
3	3,4	1	40.0 dB	150,000	15.1E9	3.0 dB
5	3	2 (no AJ)	38.5 dB	175,000	9.38E9	1.5 dB
5	3	3 (AJ)	38.5 dB	175,000	9.38E9	17 dB

SJTMAX: Assigned by PIP as indicated above. SJTMAX is the threshold where tracking errors become significant.

RGDB, PRW, FREQ: Assigned within P001.

12 Card: Print Options for Output

IPRINT(1) through IPRINT(7): If the extended output option is chosen by the user on the PIP option card, IPRINT(1) through IPRINT(7) = 1 and an extended printout of the result of the P001 analysis is obtained. If the extended output

option is not chosen, IPRINT(1) through IPRINT(7) = 0 and a summary of the P001 analysis is printed as output from P001.

III. SUMMARY AND CONCLUSIONS

The use of the P001 Input Program (PIP) to provide all required input to P001 greatly reduces the student involvement in the aircraft combat survivability scenario computer input procedure, freeing him from time consuming, tedious computations and keypunching which do not contribute profitably to the aircraft combat survivability learning experience. In addition, PIP provides an indication as to the realism of the input data, thus contributing to the validity of the result of the P001 analysis.

Introduction of the Calspan Improved P001 Computer Program into aircraft combat survivability studies provides the class problem in survivability assessment with ECM (jamming), ECCM (anti-jam) and radar multipath features which are realistic parameters to be considered in any current aircraft combat survivability situation.

As developed, the PIP target aircraft performance parameters are those of a "typical" Navy attack aircraft, having flight characteristics that are realistic, but which can not be used to describe the performance of any specific aircraft. As a future project, specific aircraft flight performance parameters and equations could be added to PIP in the form of a flight path generator program to give the input program the added capability of simulating the flight path of a specific aircraft.

APPENDIX A

AIRCRAFT COMBAT SURVIVABILITY PROBLEM

This Appendix contains a complete package for a class problem in aircraft attrition in a hostile AAA environment for AE 3251, Aircraft Combat Survivability.

AE 3251

AIRCRAFT COMBAT SURVIVABILITY

A STUDY

of

AIRCRAFT ATTRITION

in a

HOSTILE AAA ENVIRONMENT

NAVAL POSTGRADUATE SCHOOL
MONTEREY, CALIFORNIA

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I. INTRODUCTION

This aircraft attrition study is designed to present the student with an opportunity to see first hand how the survivability of an aircraft can be evaluated in a given combat scenario. The methods employed in this study are those used by both industry and government when making crucial decisions in the survivability design of an aircraft weapon system. In this study, a computer program named P001 (AFATL Antiaircraft Artillery Simulation Computer Program) will be used to (1) simulate the flight of a typical Naval attack aircraft through a hostile antiaircraft artillery (AAA) environment and (2) compute the aircraft probability of survival.¹

Section II describes all of the steps necessary to complete this study. Note the flow of the survivability assessment process from a physical description of the aircraft to a determination of its capabilities to withstand certain threat levels (i.e., its vulnerability), to a scenario in which both offensive and defensive strategies must be employed, to the final phase of simulating flight through the hostile environment and computing probabilities of survival using a modern, state-of-the-art computer program.

¹ The Navy specifies the use of P001 in all non-nuclear survivability assessments in MIL-STANDARD-2072(AS), SURVIVABILITY, AIRCRAFT; ESTABLISHMENT AND CONDUCT OF PROGRAMS FOR, August 1977.

The student should develop a good appreciation for the magnitude of the survivability problem by keeping the above survivability assessment process in mind when working each part of the analysis.

II. PROBLEM DEFINITION

- A. You are going to conduct a survivability assessment of a familiar Naval aircraft, shown in Figs. 1 and 2, on a typical attack mission to destroy the bridge shown in Fig. 3.
- B. The class will be divided into groups of four, with two members in each group on the blue team and two members on the red team.
- C. Each team will independently determine the vulnerable areas of the aircraft to the specified threat in the six major views.
- D. Each team will use P001 to determine the survivability of the aircraft in the class problem scenario, as follows:
 - 1. Each team will select a flight path to the bridge according to the rules of the scenario given in Section IV. Keep this path a secret.
 - 2. Each team will also select the locations of six AAA emplacements that will defend the bridge against an air attack. Locate the weapons according to the order of battle given in Section IV. Keep these locations secret, also.
 - 3. Each team will conduct an attack against the other team in the group.
 - 4. The input data cards for the computer run for the blue team attacking the bridge defended by the red team

will consist of the flight path of the blue aircraft flying through the AAA emplacements selected by the red team.

5. The input data cards for the computer run for the red team against the blue team will consist of the flight path of the red aircraft flying through the AAA emplacements selected by the blue team.
- E. May the best team win. A small prize will be awarded to the team whose aircraft has the highest probability of survival against their opponent's weapon distribution.
 - F. Additional runs will be made against a preset AAA distribution to investigate the effects of ECM, ECCM, jinking, etc., on the survivability of the aircraft.

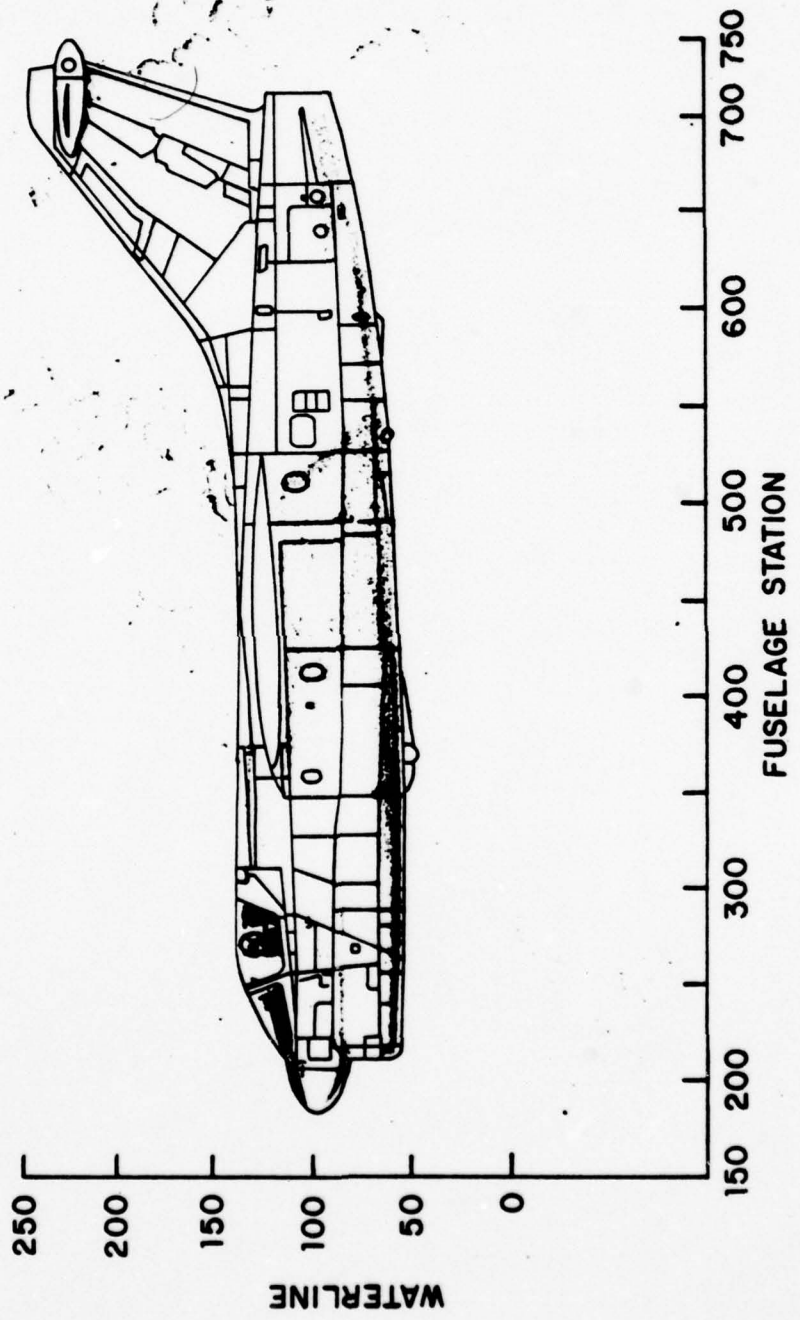


Figure 1

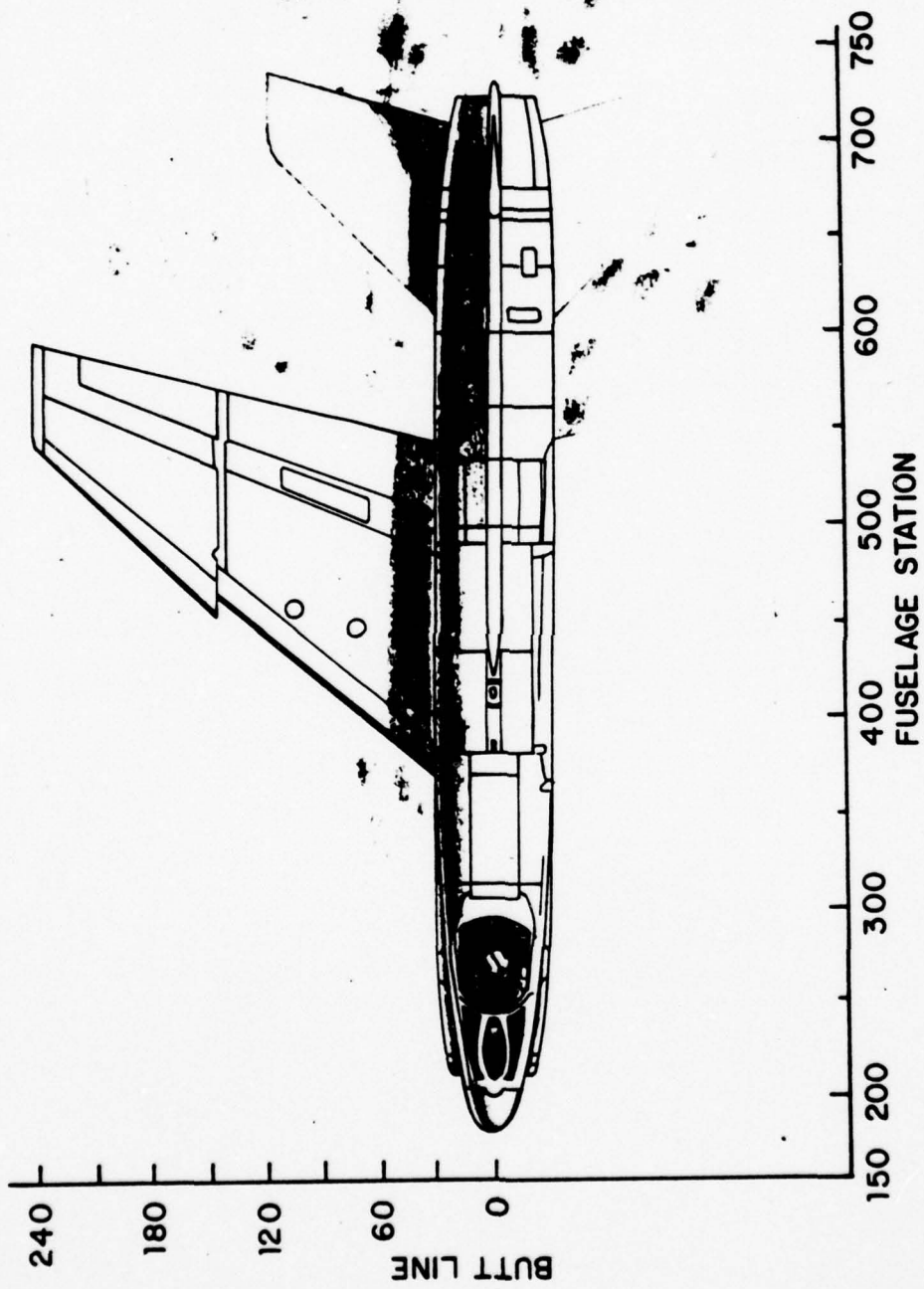


Figure 2

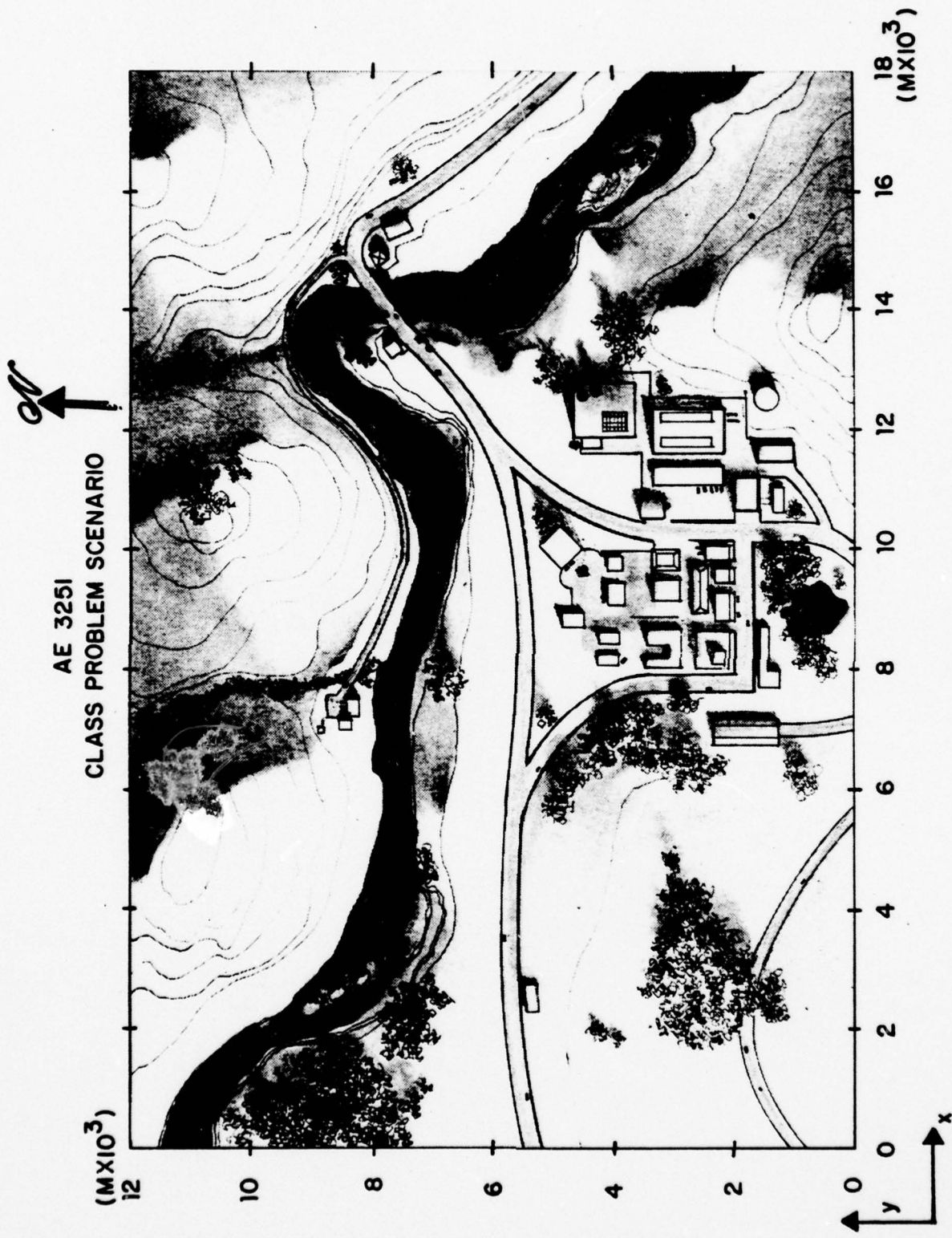


Figure 3

III. VULNERABILITY ASSESSMENT DESCRIPTION

A. GENERAL METHODOLOGY

The six general requirements for a vulnerability assessment are discussed in detail in Chapter IV of the class text. The following data are given in support of this assessment procedure:

1. Kill Category = "A" Kill
2. Technical Description of the Aircraft - Figs. 1 and 2
3. Critical Components - Pilot, engine and fuel tanks.
 - a. Each of these singly vulnerable components will make a contribution to A_{p_i} , the total presented area of the aircraft.
 - b. The total presented area is assumed to be a "shoe box" centered around the aircraft center of gravity.
4. Damage Analysis - Determination of $P_{K/H}$ for each of the components will be discussed in class due to the classified nature of the material.
5. Threat Types - To be discussed in class; Types I, II, III and V threats will be used in the scenario.
6. Determine Vulnerable Area - Use the equation:

$$A_v = \sum_i A_{p_i} \cdot P_{K/H_i}$$

where i = pilot, engine and fuel tanks.

B. SPECIFIC CALCULATIONS

The vulnerability assessment may now be completed in the following manner:

1. Measure the presented area of each critical component of the aircraft shown in Figs. 1 and 2 for the top/bottom, front/rear and left/right aspects and record in Table I.

2. Calculate the A_{V_i} for each component using the given P_{K/H_i} for the appropriate aspect and striking velocity and enter it into Table I.

3. In order to use the information compiled in Table I in P001, a more complete description of the aircraft A_V with changing aspect angle must be tabulated. This is normally done in a 26 view, δ striking velocity vulnerable area table (VAT). Fig. 4 and Table II show how the aircraft is physically divided into these 26 different views. You have tabulated in Table I the total A_V for each of the striking velocities, but only for the six major aspects. The vulnerable area of the other views can be obtained by interpolating between these six aspects. The following is a summary of the six views you have done in Table I and their relationship to the 26 views needed to describe the aircraft:

View	View #	Long (AZ) (degrees)	Lat (Elev) (degrees)
Bottom	1	0	0
	2-9	0-315	45
Tail-on	10	0	90
	11	45	90
STBD Side	12	90	90
	13	135	90
Head-on	14	180	90
	15	215	90
Port Side	16	270	90
	17	315	90
	18-25	0-315	135
TOP	26	0	180

The A_v 's you have calculated will not be used in the P001 analysis. Instead, predetermined VATS for each threat type will be used in order to provide a standardized aircraft for the attrition study.

4. Turn in Table I prior to initiating a computer run for the analysis.

TABLE I.

SINGLY VULNERABLE AREA (A_v) SUMMARY FORM

Assessment Date _____		Performing Organization _____		Aircraft Threat _____											
Kill Category _____		Projectile V _s , ft/sec (m/sec)		Aircraft Threat											
Component		500 (152.4)		1,000 (304.8)		1,500 (457.2)		2,000 (609.6)		2,500 (762.0)		3,000 (914.4)		3,500 (1066.8)	
Aspect	Component	A _p	P _k /H	A _v	P _k /H	A _v	P _k /H	A _v	P _k /H	A _v	P _k /H	A _v	P _k /H	A _v	P _k /H
TOP/ BOTTOM	PILOT														
	ENGINE														
	FUEL TANKS														
	FUSELAGE														
	WINGS														
	TOTAL														
FRONT/ REAR	PILOT														
	ENGINE														
	FUEL TANKS														
	FUSELAGE														
	WINGS														
	TOTAL														
LEFT/ RIGHT	PILOT														
	ENGINE														
	FUEL TANKS														
	FUSELAGE														
	WINGS														
	TOTAL														

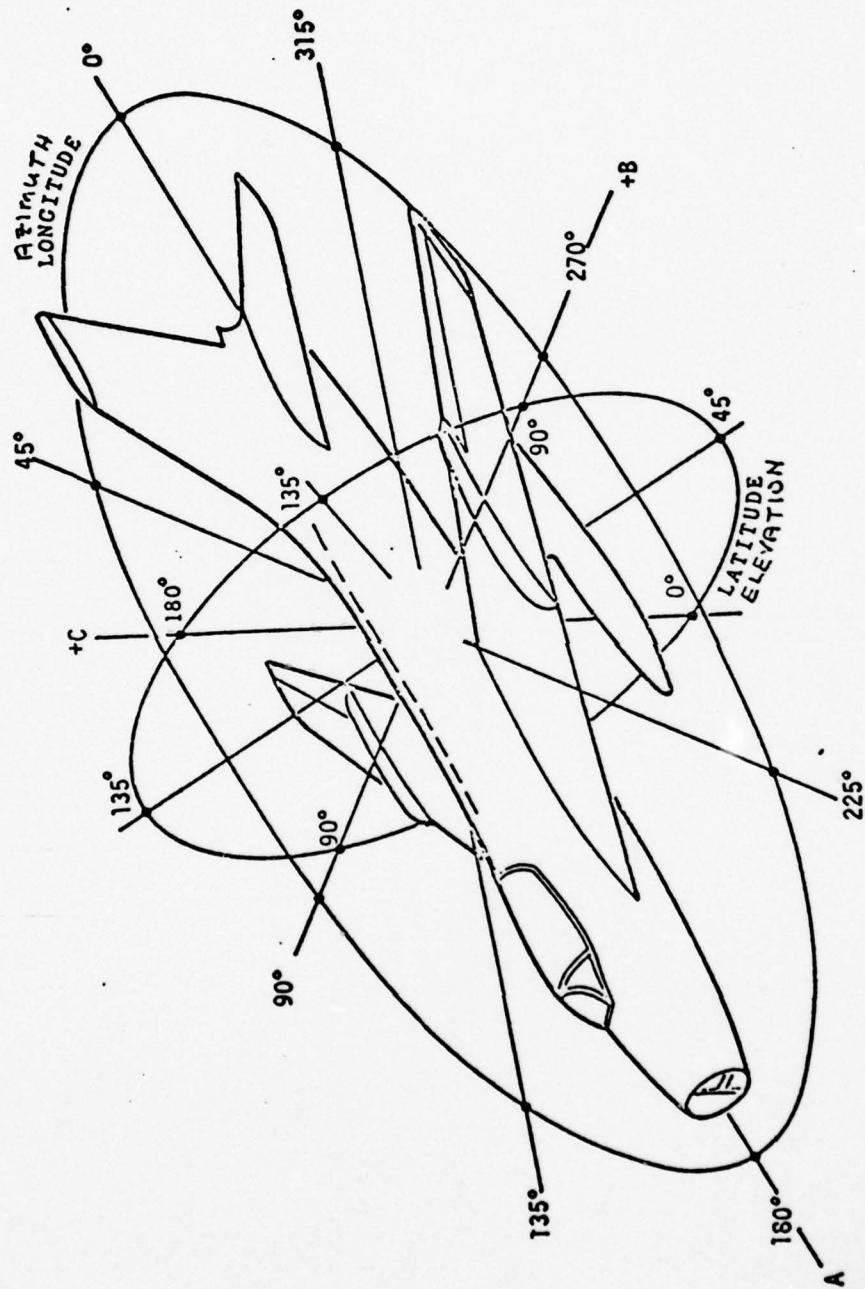


Figure 4 Longitude and Latitude of Aircraft for Vulnerable Area Computation

TABLE II VULNERABLE AREA TABLES

Card Number	I	J	Aircraft View
1	1	1	0° Longitude, 0° Latitude
2	1	2	0° Longitude, 45° Latitude
3	2	2	45° Longitude, 45° Latitude
4	3	2	90° Longitude, 45° Latitude
5	4	2	135° Longitude, 45° Latitude
6	5	2	180° Longitude, 45° Latitude
7	6	2	225° Longitude, 45° Latitude
8	7	2	270° Longitude, 45° Latitude
9	8	2	315° Longitude, 45° Latitude
10	1	3	0° Longitude, 90° Latitude
11	2	3	45° Longitude, 90° Latitude
12	3	3	90° Longitude, 90° Latitude
13	4	3	135° Longitude, 90° Latitude
14	5	3	180° Longitude, 90° Latitude
15	6	3	225° Longitude, 90° Latitude
16	7	3	270° Longitude, 90° Latitude
17	8	3	315° Longitude, 90° Latitude
18	1	4	0° Longitude, 135° Latitude
19	2	4	45° Longitude, 135° Latitude
20	3	4	90° Longitude, 135° Latitude
21	4	4	135° Longitude, 135° Latitude
22	5	4	180° Longitude, 135° Latitude
23	6	4	225° Longitude, 135° Latitude
24	7	4	270° Longitude, 135° Latitude
25	8	4	315° Longitude, 135° Latitude
26	1	5	0° Longitude, 180° Latitude

NOTE: Refer to Figure 4 for definition of longitude and latitude.

IV. SCENARIO DESCRIPTION

A. This scenario is purely for instructional purposes and is not based on any actual or planned combat attack situation. The target site, order of battle, attack heading, and outbound flight path parameter limits have been chosen only to provide guidelines for the class problem. As much realism has been introduced for the players as possible while retaining an unclassified scenario.

B. Your target is the bridge shown in Fig. 3. located at:

x: 14,100 meters

y: 7,900 meters

z: 20 meters

Heavy military supply traffic has been reported in this area. Your mission is to destroy this vital supply link.

C. The following order of battle has been gathered from intelligence reports of the target area:

Interceptor Aircraft - three airfields within striking distance

SAM - six sites within a 125 km radius.

AAA - two type 1 mode 1
two type 2 mode 1
one type 3 mode 4
one type 3 mode 3
one type 5 mode 3

(Note: Gun types and their relationship to AAA will be discussed in class.)

Ground Troops - regular infantry and civilian militia are numerous in the target area.

D. The SAM threat and the presence of enemy aircraft requires that the inbound approach to the target be made from the west at low level. A pop-up maneuver is required to visually identify the target followed by a dive bombing run to weapon delivery. Egress must be made to either the north or south, depending on individual strategy.

E. The following is a list of scenario limitations to be used in the development of your strategy:

1. Flight path milestones - specify at least one milestone for approximately 500 meters of flight path.
2. Aircraft cruise speed - 210 to 250 meters per second.
3. Inbound altitude - 70 to 450 meters.
4. Pop-up maneuver
 - a. Commence maneuver - 4,000 to 6,000 meters from the target.
 - b. Maneuver altitude - minimum 1,220 meters; maximum 2,130 meters.
5. Weapons delivery.
 - a. Alignment - the leg immediately prior to the bomb release point must be 600 meters in length (straight) and must have a heading within 5° of the heading to the target from the bomb release point.
 - b. Bomb release range - 1,000 meters maximum.
 - c. Bomb release altitude - 310 to 910 meters.

(Note: A typical 20° dive commenced from 1,000 meters of altitude at 2,500 meters from the target will release weapons at 400 meters of altitude about 700 meters from the target and will lose 160 meters in the pull-out.)

6. Maneuvering - if any turn along the flight path is greater than 28°, the maximum g loading of 6 will be exceeded.

7. Weapons placement.

a. Two type 1 mode 1, two type 2 mode 1, one type 3 mode 4, and one type 3 mode 3 weapons are available for defense placement.

b. One type 5 mode 3 weapon is placed at x: 12,800 meters, y: 7,500 meters, and z: 20 meters. You do not specify the location of this weapon.

c. Neither of the type 3 weapons may be placed within 3,000 meters of the center of the bridge.

8. Jammer power - if the jamming function is specified, the jammer power you select must be no more than 1,000 watts.

F. Begin the flight path at an entry point of your choosing along the western boundary and end it along the northern or southern boundary. Note the terrain features, anticipate the AAA placement for bridge defense and plan your flight path accordingly.

G. Locate the AAA weapons given in the order of battle to best defend the bridge against your opponent's attacking aircraft.

V. INPUT DATA PREPARATION

A preprocessor for P001 has been developed at NPS that will punch all of the input cards for the execution of P001, with the exception of the green JOB card and the final orange END OF FILE card. This preprocessor is called PIP (P001 Input Program). The inputs to PIP are the x, y and z coordinates of your flight path milestones and your opponent's six AAA emplacement locations.

A. Milestone cards: The x, y and z coordinates of the aircraft (in meters) for up to 199 flight path milestones must be entered into PIP in 3F10.0 format, one milestone per card. (Milestone #1 will have an x coordinate of 0.0).

B. Milestone delimiter card: A card containing 99999., left justified, must be placed after the final milestone card.

C. Option control card: A control card follows the milestone delimiter card and specifies the aircraft cruise speed, the number of the bomb release milestone (count the initial position on the western border as milestone #1), eight input/output/scenario options and the jammer power. The data on the control card must be specified in F10.0,I2,8I1,F10.0 format and contains the following parameters:

(columns 1-10): Aircraft cruise speed in meters per second

(columns 11-12): Number of the bomb release milestone.

- (column 13): EW option - 0 for no jamming, 1 for jamming.
- (column 14): ECCM anti-jam option - 0 for no AJ, 1 for radar AJ.
- (column 15): Radar multipath option - 0 for no multipath effects, 1 for radar degradation caused by multipath effects.
- (column 16): Gun location option - 0 specifies PIP preset AAA locations, 1 requires user input of the six AAA locations.
- (column 17): List option - 0 for no listing of the P001 input deck, 1 for listing of P001 input deck.
- (column 18): Punch option - 0 for no punched P001 input deck, 1 for punched P001 input deck.
- (column 19): Plot option - 0 for no scenario plot, 1 for scenario plot.
- (column 20): Extended output option - 0 for no extended output, 1 for extended printout of P001 analysis results.
- (columns 21-30): Jammer power in watts (0 to 1,000 watts).

D. Gun emplacement location cards: If column 16 on the control card contains a 1, six gun location cards specifying the x, y and z coordinates of each of the gun emplacements (format 3F10.0) specified in the order given in the order of battle are required as input data.

E. Sample PIP input deck:

```

// ( Green JOB Card, TIME=2 )
// EXEC NVTECGO,NAME=PIP,REGION=200K
//STEPLIB DD DSN=F0559.PIP,UNIT=3330,VOL=SER=DISK02,DISP=SHR
//FT06F001 DD SYSOUT=A,DCB=(RECFM=FBA,LRECL=133,BLKSIZE=3325)
//GO.FT07F001 DD SYSOUT=B
//GO.SYSIN DD*
0.          500.          450.
500.        500.          440.
900.        600.          445.
(etc., until all milestones are described)
99999.
232.0      2511111111    500.0
14800.     9000.         20.
16200.     8200.         10.
(etc., until all six AAA locations are described)
/*

```

F. When the P001 input deck punched by PIP is received, put the green JOB card used for PIP on top of the deck and an orange /* (EOF) card on the bottom of the deck and read it through the card reader. The output of this deck will be a combat survivability analysis for the given flight path and AAA emplacement locations.

APPENDIX B

PIP INPUT DECK LISTING

THE FOLLOWING IS AN EXAMPLE OF THE INPUT TO THE PIP INPUT PROGRAM (PIP) TO BE RUN FROM A LOAD MODULE (DISK 02). THE INPUT UTILIZES THE PRESET GUN EMPLOYMENT LOCATIONS OPTION AND THE ELECTRONIC WARFARE (JAMMING), FIES, ANTI-JAM AND MULTIPATH SCENARIO OPTIONS. THE CONTROL CARD ALSO SPECIFIES THAT THE OUTPUT BE LISTED PUNCHED AND PLOTTED. THERE ARE 20 MILESTONES, MILESTCNE IS THE BOMB RELEASE MILESTCNE AND THE CRUISE SPEED IS SPECIFIED AS 232.0 METERS PER SECOND. THE EXTENDED OUTPUT OPTION IS SPECIFIED.

```

( GREEN JOB CARD )
// EXEC NVTECGO,NAME=PIP
// STEPLIB DD DSN=F0559.PIP,UNIT=3330,VOL=SER=DISK02,DISP=SHR
//FT06FC01 DD SYSCUT=A,DCB=(RECFM=FBA,LRECL=133,BLKSIZE=3325)
//GO.FTC7F001 DD SYSCUT=B
//GO.SYSIN DD *
0.
1386.
2742.
4054.
5406.
6762.
8129.
9508.
10767.
11588.
12054.
12844.
13525.
14050.
14097.
15127.
16479.
17619.
17881.
17872.
99999.
232.0
1511101111 500.0
/*
457.
423.
434.
374.
421.
436.
380.
412.
448.
455.
1880.
2130.
2120.
1698.
1914.
606.
562.
474.
456.
457.

```

APPENDIX C
PIP OUTPUT LISTING

THE FOLLOWING IS THE OUTPUT FROM THE POOL INPUT PROGRAM (PIP) UTILIZING THE PRESET GUN EMPLACEMENT LOCATIONS AND THE ELECTRONIC WARFARE (JAMMING), ANTI-JAM, MULTIPATH AND EXTENDED OUTPUT OPTIONS:

```
// EXEC PGM=PIEM, REGION=200K
// STEPLIB DD DSN=F0559.PIEM, UNIT=3330, VOL=SER=DISK02, DISP=SHR
// FT06F001 DD SYSOUT=A, DCB=(RECFM=FBA, LRECL=133, BLKSIZE=3236)
// GO.FT04F001 DD UNIT=SYSDA, SPACE=(CYL,(1,1)),
//          LCB=(RECFM=VBS, LRECL=404, BLKSIZE=3236)
//          LCB=(RECFM=VBS, LRECL=404, BLKSIZE=3236)
// GO.FT07F001 DD UNIT=SYSDA, SPACE=(CYL,(1,1)),
//          LCB=(RECFM=VBS, LRECL=404, BLKSIZE=3236)
// GO.FT08F001 DD UNIT=SYSDA, SPACE=(CYL,(1,1)),
//          LCB=(RECFM=VBS, LRECL=404, BLKSIZE=3236)
// GO.FT09F001 DD UNIT=SYSDA, SPACE=(CYL,(1,1)),
//          LCB=(RECFM=VBS, LRECL=404, BLKSIZE=3236)
// FT11F001 DD UNIT=SYSDA, SPACE=(CYL,(1,1)),
//          LCB=(RECFM=VBS, LRECL=404, BLKSIZE=3236)
// GO.FT05F001 DD *
```

01	AIRCRAFT	COMBAT	SURVIVABILITY	SCENARIO
02	0.0	114.73	0.1147	231.37
	6.8	500.0	457.0	231.8
	17.6	399.0	423.0	227.4
	29.3	436.0	374.0	230.0
	35.2	155.0	421.0	231.0
	47.2	338.0	436.0	232.6
	52.4	438.0	380.0	213.4
	65.0	349.0	412.0	67.8
	71.4	589.0	448.0	139.1
	78.1	1122.0	455.0	147.5
	85.4	1934.0	1880.0	173.4
	97.8	2378.0	2130.0	77.3
	104.8	3114.0	2120.0	183.2
	114.7	4236.0	1698.0	205.6
		5488.0	1914.0	175.5
		6922.0	606.0	40.4
		7738.0	562.0	
		8466.0	474.0	
		9480.0	456.0	
		10972.0	457.0	
		11894.0		

```
//GC.FTC5F002 DD * 9000. 40.
03 14800.
```

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04	11111	11.0	34.419	VS	TYPE	45.893	57.366	68.839	80.312	91.785	103.258
05	1	9	473	WEAPONS	1	AND	2				
06	0.464	0.464	0.464	7.107	7.107	7.107	7.107	7.107	7.107	7.107	7.107
07	0.697	0.697	0.697	5.574	5.574	5.574	5.574	5.574	5.574	5.574	5.574
	0.657	0.657	0.657	7.357	7.357	7.357	7.357	7.357	7.357	7.357	7.357
	0.697	0.697	0.697	5.574	5.574	5.574	5.574	5.574	5.574	5.574	5.574
	0.657	0.657	0.657	5.574	5.574	5.574	5.574	5.574	5.574	5.574	5.574
	0.697	0.697	0.697	7.357	7.357	7.357	7.357	7.357	7.357	7.357	7.357
	0.464	0.464	0.464	5.574	5.574	5.574	5.574	5.574	5.574	5.574	5.574
	0.657	0.657	0.657	0.858	0.858	0.858	0.858	0.858	0.858	0.858	0.858
	0.464	0.464	0.464	3.298	3.298	3.298	3.298	3.298	3.298	3.298	3.298
	0.657	0.657	0.657	2.858	2.858	2.858	2.858	2.858	2.858	2.858	2.858
	0.464	0.464	0.464	0.743	0.743	0.743	0.743	0.743	0.743	0.743	0.743
	0.657	0.657	0.657	0.858	0.858	0.858	0.858	0.858	0.858	0.858	0.858
	0.464	0.464	0.464	2.298	2.298	2.298	2.298	2.298	2.298	2.298	2.298
	0.657	0.657	0.657	2.858	2.858	2.858	2.858	2.858	2.858	2.858	2.858
	0.464	0.464	0.464	5.574	5.574	5.574	5.574	5.574	5.574	5.574	5.574
	0.657	0.657	0.657	5.574	5.574	5.574	5.574	5.574	5.574	5.574	5.574
	0.697	0.697	0.697	7.357	7.357	7.357	7.357	7.357	7.357	7.357	7.357
	0.657	0.657	0.657	5.574	5.574	5.574	5.574	5.574	5.574	5.574	5.574
	0.697	0.697	0.697	7.357	7.357	7.357	7.357	7.357	7.357	7.357	7.357
	0.464	0.464	0.464	5.574	5.574	5.574	5.574	5.574	5.574	5.574	5.574
04	121111111	16200.	8200.	40.							
05	121111111	13600.	7200.	20.							
06	21111										
07	121111111	13400.	8000.	20.							
	121111111	11300.	9700.	50.							
	34411										
	12.540	12.540	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470
	9.853	9.853	10.510	10.510	10.510	10.510	10.510	10.510	10.510	10.510	10.510
	12.640	12.640	14.780	14.780	14.780	14.780	14.780	14.780	14.780	14.780	14.780
	9.853	9.853	11.150	11.150	11.150	11.150	11.150	11.150	11.150	11.150	11.150
	9.639	9.639	10.510	10.510	10.510	10.510	10.510	10.510	10.510	10.510	10.510
	9.639	9.639	11.150	11.150	11.150	11.150	11.150	11.150	11.150	11.150	11.150

12.649	14.780	14.780	14.780	14.780	14.780	14.780	14.780	14.780	14.780	1.0	1	3.00
12.639	11.394	11.394	11.394	11.394	11.394	11.394	11.394	11.394	11.394	1000.000	1000.000	1000.000
1.394	1.394	1.394	1.394	1.394	1.394	1.394	1.394	1.394	1.394	1000.000	1000.000	1000.000
4.762	7.432	7.432	7.432	7.432	7.432	7.432	7.432	7.432	7.432	100.000	100.000	100.000
4.762	7.432	7.432	7.432	7.432	7.432	7.432	7.432	7.432	7.432	100.000	100.000	100.000
1.394	1.394	1.394	1.394	1.394	1.394	1.394	1.394	1.394	1.394	0.156	0.063	0.173
4.762	7.432	7.432	7.432	7.432	7.432	7.432	7.432	7.432	7.432	0.073	0.072	0.019
4.762	7.432	7.432	7.432	7.432	7.432	7.432	7.432	7.432	7.432	0.093	0.039	0.068
5.342	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	6.240	0.035	0.032	0.017
4.762	7.432	7.432	7.432	7.432	7.432	7.432	7.432	7.432	7.432	0.156	0.063	0.173
9.853	10.510	10.510	10.510	10.510	10.510	10.510	10.510	10.510	10.510	0.073	0.072	0.019
9.853	10.510	10.510	10.510	10.510	10.510	10.510	10.510	10.510	10.510	100.000	100.000	100.000
12.640	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	100.000	100.000	100.000
9.853	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	100.000	100.000	100.000
9.853	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	0.023	0.023	0.015
12.640	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	11.780	0.015	0.015	0.015
12.540	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	0.023	0.023	0.015
12.540	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	0.015	0.015	0.015
14.1	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	100.000	100.000	100.000
14.1	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	100.000	100.000	100.000
14.1	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	13.470	100.000	100.000	100.000
19	180.0	180.0	180.0	180.0	180.0	180.0	180.0	180.0	180.0	1000.0	1000.0	1000.0
1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000
100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	0.053	0.053	0.053
0.273	0.178	0.178	0.178	0.178	0.178	0.178	0.178	0.178	0.178	0.020	0.020	0.020
0.235	0.110	0.110	0.110	0.110	0.110	0.110	0.110	0.110	0.110	0.454	0.454	0.454
0.028	0.005	0.005	0.005	0.005	0.005	0.005	0.005	0.005	0.005	0.019	0.019	0.019
0.112	0.032	0.032	0.032	0.032	0.032	0.032	0.032	0.032	0.032	0.509	0.509	0.509
0.033	0.102	0.102	0.102	0.102	0.102	0.102	0.102	0.102	0.102	0.053	0.053	0.053
0.273	0.047	0.047	0.047	0.047	0.047	0.047	0.047	0.047	0.047	0.020	0.020	0.020
0.235	0.110	0.110	0.110	0.110	0.110	0.110	0.110	0.110	0.110	0.454	0.454	0.454
100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000
100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000
100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000	100.000
1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000	1000.000
15600.	10900.	10900.	10900.	10900.	10900.	10900.	10900.	10900.	10900.	15600.	10900.	90.
121111111	15600.	10900.	10900.	10900.	10900.	10900.	10900.	10900.	10900.	15600.	10900.	90.
03	33411	33411	33411	33411	33411	33411	33411	33411	33411	15600.	10900.	90.
04	1	1	1	1	1	1	1	1	1	15600.	10900.	90.
13	1	1	1	1	1	1	1	1	1	15600.	10900.	90.
121111111	1	1	1	1	1	1	1	1	1	15600.	10900.	90.

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03 53211		12800.		7500.		20.		5 WEAPON		1		1.0		3		17.00	
04 VULNERABLE AREA TABLE		VS TYPE		VS TYPE		VS TYPE		VS TYPE		VS TYPE		VS TYPE		VS TYPE		VS TYPE	
07		55.370		55.370		55.370		55.370		55.370		55.370		55.370		55.370	
13	1	90.0	180.0	500.0	500.0	1.00E-06	1.00E-06	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
14	1	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
GEN	9	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
	7	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
	1	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
	1	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
	1	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0
	0.273	0.178	0.178	0.178	0.178	0.178	0.178	0.178	0.178	0.178	0.178	0.178	0.178	0.178	0.178	0.178	0.178
	0.235	0.047	0.110	0.110	0.110	0.110	0.110	0.110	0.110	0.110	0.110	0.110	0.110	0.110	0.110	0.110	0.110
	0.028	0.005	0.054	0.054	0.054	0.054	0.054	0.054	0.054	0.054	0.054	0.054	0.054	0.054	0.054	0.054	0.054
	0.112	0.032	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019
	0.033	0.102	0.509	0.509	0.509	0.509	0.509	0.509	0.509	0.509	0.509	0.509	0.509	0.509	0.509	0.509	0.509
	0.273	0.178	0.053	0.053	0.053	0.053	0.053	0.053	0.053	0.053	0.053	0.053	0.053	0.053	0.053	0.053	0.053
	0.235	0.047	0.020	0.020	0.020	0.020	0.020	0.020	0.020	0.020	0.020	0.020	0.020	0.020	0.020	0.020	0.020
	0.028	0.110	0.454	0.454	0.454	0.454	0.454	0.454	0.454	0.454	0.454	0.454	0.454	0.454	0.454	0.454	0.454
	1.0	0.173	0.073	0.093	0.093	0.093	0.093	0.093	0.093	0.093	0.093	0.093	0.093	0.093	0.093	0.093	0.093
	3	0.063	0.072	0.039	0.032	0.032	0.032	0.032	0.032	0.032	0.032	0.032	0.032	0.032	0.032	0.032	0.032
	17.00	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173
		0.019	0.019	0.068	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017
		0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063
		0.072	0.072	0.068	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072
		0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068
		0.017	0.017	0.173	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017
		0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173
		0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019
		0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063
		0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072
		0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068
		0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017
		0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173
		0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019	0.019
		0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063	0.063
		0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072	0.072
		0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068	0.068
		0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017	0.017
		0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173	0.173

100.000 100.000 100.000 100.000 100.000 100.000
 100.000 100.000 100.000 100.000 100.000 100.000
 100.000 100.000 100.000 100.000 100.000 100.000
 1000.000 1000.000 1000.000 1000.000 1000.000 1000.000
 1000.000 1000.000 1000.000 1000.000 1000.000 1000.000
 1000.000 1000.000 1000.000 1000.000 1000.000 1000.000
 1211111111

/*
 /*GD.FTC5F003 DD *
 /*

(NEW PAGE)

***** POOL FLIGHT PATH SCENARIO SUMMARY *****

THE FLIGHT PATH CONSISTS OF 20 MILESTONES WITH A TOTAL FLIGHT TIME OF
 114.5 SECONDS. BOMB RELEASE IS AT MILESTONE 15.

***** OPTION SUMMARY *****

CRUISE SPEED IS 232.0 METERS PER SECOND.
 A POOL INPUT LISTING IS PROVIDED AS OUTPUT.
 A PUNCHED DECK IS PROVIDED AS OUTPUT.
 A SCENARIO PLOT IS PROVIDED AS OUTPUT.
 EXTENDED PRINTOUT IS PROVIDED AS OUTPUT.
 PRESET GUN EMPLACEMENT IS BEING UTILIZED.
 AN AIRBORNE JAMMER IS BEING UTILIZED.
 JAMMER POWER IS 500.0 WATTS.
 JANTI-JAM FEATURE IS UTILIZED WHERE APPROPRIATE.
 MULTIPATH RADAR EFFECTS ARE CONSIDERED.

***** FLIGHT PATH ERRORS *****

***** NO FLIGHT PATH ERRORS IN THIS RUN *****

***** END OF POOL INPUT PROGRAM - SUMMARY COMPLETE *****

APPENDIX D

P001 INPUT GUIDE CHANGES

Revisions to P001:

1. Delete card 7A and 7B.
2. Insert cards 13 and 14 as given on the following pages.
3. The following list is given as an aid to facilitate the assignment of valid combinations of gun type, mode and operating characteristics to the ECM and multipath options:

Gun Type (IGT)	Mode (IEM)	Radar ID (IRECM)	IRMP	Option
1	1	-	-	-
2	1	-	-	-
3	1	-	-	-
3	2	-	-	-
3	3	-	-	-
3	3	1	-	Jam
3	3	1	1	Jam, Multipath
3	4	-	-	-
3	4	1	-	Jam
4	1	-	-	-
5	1	-	-	-
5	2	-	-	-
5	3	-	-	-
5	3	2	-	Jam
5	3	2	2	Jam, Multipath
5	3	3	-	Jam, Anti-jam
5	3	3	2	Jam, Anti-jam, Multipath
5	3	4	-	Jam
5	3	4	3	Jam, Multipath
5	4	-	-	-
5	4	2	-	Jam
5	4	3	-	Jam, Anti-jam
5	4	4	-	Jam

Radar Multipath Input Parameters					CARD: 13																
ID	PARA	UNITS	FORMAT	COLUMNS	DESCRIPTION																
A	I	ND	I2	1-2	Data group identification code. I=13 indicates that the remainder of the card contains radar multipath parameters.																
B	ICARD	ND	7A10,A8	3-80	Seventy-eight columns of alphanumeric data to be decoded and assigned as follows:																
B1	IMUL	ND	I3	3-5	IMUL=0, no multipath. Turn off multipath if previously used. IMUL=1, multipath desired.																
B2	IRMP	ND	I5	6-10	Radar type ID. Specifies the tracking radar.																
B3	REFC	ND	F10.0	11-20	Reflection coefficient. 0.35 is a typical value for terrain with vegetation.																
<p>NOTE: Multipath effects can only be applied to a system with a Mode ID (IEM) of 3.</p> <p>The value of IRMP selects the appropriate radar parameters:</p> <table border="1"> <thead> <tr> <th>IRMP</th> <th>Beamwidth (deg)</th> <th>Squint Angle (deg)</th> <th>Calibration Constant</th> </tr> </thead> <tbody> <tr> <td>1</td> <td>1.4</td> <td>0.5</td> <td>0.759</td> </tr> <tr> <td>2</td> <td>1.8</td> <td>0.6</td> <td>1.06</td> </tr> <tr> <td>3</td> <td>4.5</td> <td>1.4</td> <td>2.74</td> </tr> </tbody> </table>						IRMP	Beamwidth (deg)	Squint Angle (deg)	Calibration Constant	1	1.4	0.5	0.759	2	1.8	0.6	1.06	3	4.5	1.4	2.74
IRMP	Beamwidth (deg)	Squint Angle (deg)	Calibration Constant																		
1	1.4	0.5	0.759																		
2	1.8	0.6	1.06																		
3	4.5	1.4	2.74																		

CARD: 13

Radar Multipath Input Parameters					CARD: 13															
ID	PARA	UNITS	FORMAT	COLUMNS	DESCRIPTION															
					<p>The following table gives the relationship between the Radar ID (IRECM) and the corresponding radar parameters (IRMP):</p> <table border="0"> <tr> <td></td> <td><u>Radar ID</u></td> <td><u>IRMP</u></td> </tr> <tr> <td></td> <td><u>(IRECM)</u></td> <td></td> </tr> <tr> <td>1</td> <td>1</td> <td>1</td> </tr> <tr> <td>2,3</td> <td>2</td> <td>2</td> </tr> <tr> <td>4</td> <td>3</td> <td>3</td> </tr> </table>		<u>Radar ID</u>	<u>IRMP</u>		<u>(IRECM)</u>		1	1	1	2,3	2	2	4	3	3
	<u>Radar ID</u>	<u>IRMP</u>																		
	<u>(IRECM)</u>																			
1	1	1																		
2,3	2	2																		
4	3	3																		

CARD: 13

ECM (Jamming) Input Parameters		CARD: 14			
ID	PARA	UNITS	FORMAT	COLUMNS	DESCRIPTION
A	I	ND	I2	1-2	Data group identification code. I=14 indicates that the remainder of the card contains ECM parameters.
B	ICARD	ND	7A10,A8	3-80	Seventy-eight columns of alphanumeric data to be decoded and assigned as follows:
B1	IJAM	ND	I3	3-5	Jamming switch. 0=no jamming; 1=jamming; card must be reread with IJAM=0 to turn jamming off.
B2	IP	ND	I3	6-8	IP=0, no J/S printout. IP=n, print every nth value.
B3	IJ	ND	I2	9-10	IJ=0; then GAINJ is antenna gain of jammer.
B4	GAINJ	ND	F10.0	11-20	IJ=1; a 37x37 5° jammer table follows. Gain is in dB.
B5	PJW	watts	F10.0	21-30	Jammer power in watts.
B6	PLEN	sec	F10.0	31-40	Length of jammer cover pulse. Needed when IRECM=3. 1 microsecond is a standard value.
B7	IX	ND	I5	41-45	IX=0; no cross section table is needed.
B8	XSEC	m ²	F10.0	46-55	A constant cross section of XSEC m ² is used.
B9	CALX	ND	F10.0	56-65	IX=1, a cross section table will be read following the 14 card and a possible jammer table. The cross section table values will be multiplied by CALX. This allows the user to scale the cross section. If not used, it must be set to 1.

CARD: 14

ECM (Jamming) Input Parameters					CARD: 14																				
ID	PARA	UNITS	FORMAT	COLUMNS	DESCRIPTION																				
B10	IRECM	ND	I5	66-70	Radar type. This index calls up the following constants from a data statement: <table border="1"> <thead> <tr> <th>IRECM</th> <th>Gain (RGDB)</th> <th>Power (PRW)</th> <th>Frequency (FREQ)</th> <th>SJTMAX*</th> </tr> </thead> <tbody> <tr> <td>1</td> <td>40.0 dB</td> <td>150,000</td> <td>15.1E9</td> <td>3 dB</td> </tr> <tr> <td>2,3</td> <td>38.5 dB</td> <td>175,000</td> <td>9.38E9</td> <td>1.5, 17 dB</td> </tr> <tr> <td>4</td> <td>28.0 dB</td> <td>250,000</td> <td>2.638E9</td> <td>0 dB</td> </tr> </tbody> </table>	IRECM	Gain (RGDB)	Power (PRW)	Frequency (FREQ)	SJTMAX*	1	40.0 dB	150,000	15.1E9	3 dB	2,3	38.5 dB	175,000	9.38E9	1.5, 17 dB	4	28.0 dB	250,000	2.638E9	0 dB
IRECM	Gain (RGDB)	Power (PRW)	Frequency (FREQ)	SJTMAX*																					
1	40.0 dB	150,000	15.1E9	3 dB																					
2,3	38.5 dB	175,000	9.38E9	1.5, 17 dB																					
4	28.0 dB	250,000	2.638E9	0 dB																					
B11	SJTMAX	dB	F10.0	71-80	*SJTMAX must be entered as in B11. Threshold where tracking errors become significant. (dB) NOTE: The following table summarizes the valid combinations of radars, gun types and tracking modes: <table border="1"> <thead> <tr> <th>Radar ID (IRECM)</th> <th>Gun ID (IGT)</th> <th>Mode ID (IEM)</th> </tr> </thead> <tbody> <tr> <td>1</td> <td>3</td> <td>3,4</td> </tr> <tr> <td>2,3</td> <td>5</td> <td>3,4</td> </tr> <tr> <td>4</td> <td>5</td> <td>3,4</td> </tr> </tbody> </table> The two ID's on the second radar indicate the anti-jam capability. IRECM = 2, anti-jam off IRECM = 3, anti-jam on	Radar ID (IRECM)	Gun ID (IGT)	Mode ID (IEM)	1	3	3,4	2,3	5	3,4	4	5	3,4								
Radar ID (IRECM)	Gun ID (IGT)	Mode ID (IEM)																							
1	3	3,4																							
2,3	5	3,4																							
4	5	3,4																							

APPENDIX E

JCL CARD SETUPS FOR REFERENCES

THE FOLLOWING CARDS ARE THE SETUP TO RUN THE POOL INPUT PROGRAM (PIP)
FROM BATCH, UTILIZING THE CALCOMP PLOTTER:

```
( GREEN JOB CARD )  
// EXEC FORTCLGP  
// FORTCO1 DD SYSCUT=B  
// FORT.SYSIN DD *  
( PROGRAM SOURCE CARDS GO HERE )  
/* GO.SYSIN DD *  
( DATA DECK GOES HERE )  
/*
```

THIS PAGE IS BEST QUALITY PRACTICABLE
FROM COPY FURNISHED TO DDC

THE FOLLOWING CARDS ARE THE SETUP TO RUN THE POOL INPUT PRCGRAM (PIP)
FROM BATCH, UTILIZING THE VERSATEC PLOTTER:

```
// ( GREEN JOB CARD )  
// EXEC FOR TCLGV  
//FCRT.SYSIN DD *  
// ( PROGRAM SOURCE CARDS GO HERE )  
/*  
//GC.FTCTF001 DD SYSOUT=B  
//GO.SYSIN DD *  
// ( DATA DECK GOES HERE )  
/*
```

THE FOLLOWING CARDS ARE THE SETUP TO CREATE A LOAD MODULE FOR THE PC01
INPUT PROGRAM (PIP):

```
// ( GREEN JOB CARD )
// EXEC NVTECLNK
// FCRT.SYSIN DD *
// ( PROGRAM SOURCE CARDS GO HERE )
/*
//LINK.SYSLMOD DD DSN=F0559.PIP,SPACE=(CYL,(2,1,1)),
// UNIT=3330,VOL=SER=DISK02,DISP=(NEW,KEEP),
// LABEL=EXPDT=99360
//LINK.SYSIN DD *
ENTRY MAIN
NAME PIP(R)
/*
```

THE FOLLOWING CARDS ARE THE SETUP TO RUN THE POOL INPUT PROGRAM (PIP)
FROM A LOAD MODULE, UTILIZING THE VERSATEC PLOTTER.

```
// ( GREEN JOB CARD )  
// EXEC NVTECGO,NAME=PIP,REGION=200K  
// STEPLIB DD DSN=F0559.PIP,UNIT=3330,VOL=SER=DISK02,DISP=SHR  
//FT06F001 DD SYSCUT=A,DCB=(RECFM=FBA,LRECL=133,BLKSIZE=3325)  
//GO.FT07F001 DD SYSCUT=B  
//GO.SYSIN DD *  
// ( DATA DECK GOES HERE )  
/*
```

THIS PAGE IS BEST QUALITY PRACTICABLE
FROM COPY FURNISHED TO DDC

THE FOLLOWING CARDS ARE THE SETUP TO REMOVE THE PC01 INPUT PROGRAM
(PIP) FROM DISK 02:

```
/// JOB CARD )  
//SCRATCH EXEC PGM=IEHPROGM  
//SYSPRINT DD SYSCUT=A  
//DD1 DD UNIT=3330,VOL=SER=DISK02,DISP=CLD  
//SYSIN DD *  
SCRATCH DSN=SER=F0559.PIP,VOL=3330=DISK02,PURGE  
/*
```

THE FOLLOWING CARDS ARE THE SETUP TO RUN THE IMPRCVD P001 AAA SIMULATION PROGRAM (PIEW) FRM BATCH:

```

// ( GREEN JOB CARD )
// EXEC FORTCLG,REGICN.GO=250K
// FORT.SYSIN DD *
// ( PROGRAM SOURCE CARDS GO HERE )
/*
// GO.FTC4F001 DD UNIT=SYSDA,SPACE=(CYL,(1,1)),
// DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
// GO.FTC7F001 DD UNIT=SYSDA,SPACE=(CYL,(1,1)),
// DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
// GO.FTC8F001 DD UNIT=SYSDA,SPACE=(CYL,(1,1)),
// DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
// GO.FTC9F001 DD UNIT=SYSDA,SPACE=(CYL,(1,1)),
// DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
// GO.FT11F001 DD UNIT=SYSDA,SPACE=(CYL,(1,1)),
// DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
// GO.FTC5F001 DD *
// ( CATA DECK GOES HERE )
/*

```

THE FOLLOWING CARDS ARE THE SETUP TO CREATE A LOAD MODULE FOR THE IMPROVED POOL AAA SIMULATION PROGRAM (PIEW):

```
( GREEN JOB CARD )  
// EXEC FORTCL  
//FCRT.SYSIN DD *  
( PROGRAM SOURCE CARDS GO HERE )  
/* LINK.SYSLMOD DD DSN=F0559.PIEW,SPACE=(CYL,(2,1,1)),  
// UNIT=3330,VOL=SER=DISK02,DISP=(NEW,KEEP),  
// LABEL=EXPD1=99360  
// LINK.SYSIN DD *  
ENTRY MAIN  
NAME PIEW(R)  
/*
```

THE FOLLOWING CARDS ARE THE SETUP TO RUN THE IMPROVED POOL AAA SIMULATION PROGRAM (PIEW) FROM A LOAD MODULE:

```

( GREEN JOB CARD )
// EXEC PGM=PIEW, REGION=200K
// STEPLIB DD DSN=FO559.PIEW, UNIT=3330, VCL=SER=DISK02, DISP=SHR
// FT06FC01 DD SYSCUT=A, DCB=(RECFM=FBA, LRECL=133, BLKSIZE=3325)
// GO.FT04F001 DD UNIT=SYSDA, SPACE=(CYL,(1,1)),
// DCB=(RECFM=VBS, LRECL=404, BLKSIZE=3236)
// GO.FTC7F001 DD UNIT=SYSDA, SPACE=(CYL,(1,1)),
// DCB=(RECFM=VBS, LRECL=404, BLKSIZE=3236)
// GO.FTC8F001 DD UNIT=SYSDA, SPACE=(CYL,(1,1)),
// DCB=(RECFM=VBS, LRECL=404, BLKSIZE=3236)
// GO.FTC9F001 DD UNIT=SYSDA, SPACE=(CYL,(1,1)),
// DCB=(RECFM=VBS, LRECL=404, BLKSIZE=3236)
// GO.FT11F001 DD UNIT=SYSDA, SPACE=(CYL,(1,1)),
// DCB=(RECFM=VBS, LRECL=404, BLKSIZE=3236)
// GC.FTC5F001 DD *
( DATA DECK GOES HERE )
/*

```

THE FOLLOWING CARDS ARE THE SETUP TO REMOVE P1EW FROM DISK 02:

```
/// JOB CARD )  
/// SCRATCH EXEC PGM=IEHPRGM  
/// SYSPRINT DD SYSCUT=A  
/// DD1 DD UNIT=3330,VOL=SER=DISK02,DISP=CLD  
/// SYSPRINT DD *  
/// SCRATCH DSN=IEHPRGM.P1EW,VOL=3330=DISK02,PURGE  
/*
```

APPENDIX F
POQ1 ANALYSIS OUTPUT

MASK ANGLE FOR THIS RUN = 0.0 DEG.

TIME	AFATL P-001 AAASIM #	AIRCRAFT	CCPBT	SURVIVABILITY	SCENARIO	SPEED	XDOT	YDOT	ZDOT	HEADING	CLMR	PAGE
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0001	0001	0001	0001	0001	0001	0001	0001	0001	0001	0001	0001	0001
0002	0002	0002	0002	0002	0002	0002	0002	0002	0002	0002	0002	0002
0003	0003	0003	0003	0003	0003	0003	0003	0003	0003	0003	0003	0003
0004	0004	0004	0004	0004	0004	0004	0004	0004	0004	0004	0004	0004
0005	0005	0005	0005	0005	0005	0005	0005	0005	0005	0005	0005	0005
0006	0006	0006	0006	0006	0006	0006	0006	0006	0006	0006	0006	0006
0007	0007	0007	0007	0007	0007	0007	0007	0007	0007	0007	0007	0007
0008	0008	0008	0008	0008	0008	0008	0008	0008	0008	0008	0008	0008
0009	0009	0009	0009	0009	0009	0009	0009	0009	0009	0009	0009	0009
0010	0010	0010	0010	0010	0010	0010	0010	0010	0010	0010	0010	0010
0011	0011	0011	0011	0011	0011	0011	0011	0011	0011	0011	0011	0011
0012	0012	0012	0012	0012	0012	0012	0012	0012	0012	0012	0012	0012
0013	0013	0013	0013	0013	0013	0013	0013	0013	0013	0013	0013	0013
0014	0014	0014	0014	0014	0014	0014	0014	0014	0014	0014	0014	0014
0015	0015	0015	0015	0015	0015	0015	0015	0015	0015	0015	0015	0015
0016	0016	0016	0016	0016	0016	0016	0016	0016	0016	0016	0016	0016
0017	0017	0017	0017	0017	0017	0017	0017	0017	0017	0017	0017	0017
0018	0018	0018	0018	0018	0018	0018	0018	0018	0018	0018	0018	0018
0019	0019	0019	0019	0019	0019	0019	0019	0019	0019	0019	0019	0019
0020	0020	0020	0020	0020	0020	0020	0020	0020	0020	0020	0020	0020
0021	0021	0021	0021	0021	0021	0021	0021	0021	0021	0021	0021	0021
0022	0022	0022	0022	0022	0022	0022	0022	0022	0022	0022	0022	0022
0023	0023	0023	0023	0023	0023	0023	0023	0023	0023	0023	0023	0023
0024	0024	0024	0024	0024	0024	0024	0024	0024	0024	0024	0024	0024
0025	0025	0025	0025	0025	0025	0025	0025	0025	0025	0025	0025	0025
0026	0026	0026	0026	0026	0026	0026	0026	0026	0026	0026	0026	0026
0027	0027	0027	0027	0027	0027	0027	0027	0027	0027	0027	0027	0027
0028	0028	0028	0028	0028	0028	0028	0028	0028	0028	0028	0028	0028
0029	0029	0029	0029	0029	0029	0029	0029	0029	0029	0029	0029	0029
0030	0030	0030	0030	0030	0030	0030	0030	0030	0030	0030	0030	0030
0031	0031	0031	0031	0031	0031	0031	0031	0031	0031	0031	0031	0031
0032	0032	0032	0032	0032	0032	0032	0032	0032	0032	0032	0032	0032
0033	0033	0033	0033	0033	0033	0033	0033	0033	0033	0033	0033	0033
0034	0034	0034	0034	0034	0034	0034	0034	0034	0034	0034	0034	0034
0035	0035	0035	0035	0035	0035	0035	0035	0035	0035	0035	0035	0035
0036	0036	0036	0036	0036	0036	0036	0036	0036	0036	0036	0036	0036
0037	0037	0037	0037	0037	0037	0037	0037	0037	0037	0037	0037	0037
0038	0038	0038	0038	0038	0038	0038	0038	0038	0038	0038	0038	0038
0039	0039	0039	0039	0039	0039	0039	0039	0039	0039	0039	0039	0039
0040	0040	0040	0040	0040	0040	0040	0040	0040	0040	0040	0040	0040

TIME	XX	YY	ZZ	SPEED	XDOT	YDOT	ZDOT	HEADING	CLYPR	RCL
14.00	178	178	178	178	178	178	178	178	178	178
14.01	178	178	178	178	178	178	178	178	178	178
14.02	178	178	178	178	178	178	178	178	178	178
14.03	178	178	178	178	178	178	178	178	178	178
14.04	178	178	178	178	178	178	178	178	178	178
14.05	178	178	178	178	178	178	178	178	178	178
14.06	178	178	178	178	178	178	178	178	178	178
14.07	178	178	178	178	178	178	178	178	178	178
14.08	178	178	178	178	178	178	178	178	178	178
14.09	178	178	178	178	178	178	178	178	178	178
14.10	178	178	178	178	178	178	178	178	178	178
14.11	178	178	178	178	178	178	178	178	178	178
14.12	178	178	178	178	178	178	178	178	178	178
14.13	178	178	178	178	178	178	178	178	178	178
14.14	178	178	178	178	178	178	178	178	178	178
14.15	178	178	178	178	178	178	178	178	178	178
14.16	178	178	178	178	178	178	178	178	178	178
14.17	178	178	178	178	178	178	178	178	178	178
14.18	178	178	178	178	178	178	178	178	178	178
14.19	178	178	178	178	178	178	178	178	178	178
14.20	178	178	178	178	178	178	178	178	178	178
14.21	178	178	178	178	178	178	178	178	178	178
14.22	178	178	178	178	178	178	178	178	178	178
14.23	178	178	178	178	178	178	178	178	178	178
14.24	178	178	178	178	178	178	178	178	178	178
14.25	178	178	178	178	178	178	178	178	178	178
14.26	178	178	178	178	178	178	178	178	178	178
14.27	178	178	178	178	178	178	178	178	178	178
14.28	178	178	178	178	178	178	178	178	178	178
14.29	178	178	178	178	178	178	178	178	178	178
14.30	178	178	178	178	178	178	178	178	178	178

XR= 0.0 YR= 0.0 XR= 0.0 YR= 0.0 P1= 0.0 P2= 0.0

AFATL P-001 MAASIM " AIRCRAFT CCMBAT SURVIVABILITY SCENARIO

10 TIME INTERVALS FOR PK ACCUMULATION

0.00 11.47 22.95 34.42 45.85 57.37 68.84 80.31 91.78 103.26 999.99

VULNERABLE AREA (SQ METERS) AS A FUNCTION OF IMPACT SPEED (METERS/SEC) AND ASPECT VIEW

VIEW	0	152	305	457	610	762	914	1067	1219
1	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
2	0.00	0.70	0.70	0.70	0.70	0.70	0.70	0.70	0.70
3	0.00	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.90
4	0.00	0.66	0.66	0.66	0.66	0.66	0.66	0.66	0.66
5	0.00	0.70	0.70	0.70	0.70	0.70	0.70	0.70	0.70
6	0.00	0.66	0.66	0.66	0.66	0.66	0.66	0.66	0.66
7	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
8	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
9	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
10	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
11	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
12	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
13	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
14	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
15	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
16	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
17	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
18	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
19	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
20	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
21	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
22	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
23	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
24	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
25	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46
26	0.00	0.46	0.46	0.46	0.46	0.46	0.46	0.46	0.46

TIME/ROUND	MIN ELEV	MAX ELEV	AZIM RATE	ELEV RATE	TOFI	MAX TOF2	MUZ2	BALLISTIC CONSTANT 1		BALLISTIC CONSTANT 2		VFL MIN	VFL MAX	RANGE MIN	RANGE MAX	SCOTH COAST	MAXAZ FPRCF	MAXEI FPRCF
								CONSTANT 1	CONSTANT 2	MIN	MAX							
1	0.70	10.00	30.00	25.00	1.90	56.59	840.	0.180290	0.0123920	0.0	300.0	0.	300.0	0.	300.0	555.95	5.730	5.730
2	0.70	10.00	30.00	25.00	1.90	56.59	840.	0.180290	0.0123920	0.0	300.0	0.	300.0	0.	300.0	555.95	5.730	5.730
3	0.70	10.00	30.00	25.00	1.90	56.59	840.	0.180290	0.0123920	0.0	300.0	0.	300.0	0.	300.0	555.95	5.730	5.730
4	0.70	10.00	30.00	25.00	1.90	56.59	840.	0.180290	0.0123920	0.0	300.0	0.	300.0	0.	300.0	555.95	5.730	5.730
5	0.70	10.00	30.00	25.00	1.90	56.59	840.	0.180290	0.0123920	0.0	300.0	0.	300.0	0.	300.0	555.95	5.730	5.730
6	0.70	10.00	30.00	25.00	1.90	56.59	840.	0.180290	0.0123920	0.0	300.0	0.	300.0	0.	300.0	555.95	5.730	5.730

AFATL P-001 AAASIM = AIRCRAFT COMBAT SURVIVABILITY SCENARIO

ECM INPUTS (INITIAL OR CHANGED)

IP = 5
IJ = 0
GAIN(OBJ) = 1.00
PJMNI = 500.00
PLEN(S) = 0.0
IX = 1
ZSEC(SOM) = 0.0
ALC = 1.000
TREC = 1.000
SJTMAX(DB) = 3.00

JAMMER ANTENNA GAIN 1.000 DB

AIRCRAFT CROSS SECTION TABLE SPECIFIED. PRINTED VALUES WILL BE MULTIPLIED BY CALX. CALX = 1.00

TABLE DATA

19 ELEMENTS FROM
7 ELEMENTS FROM
ELSEWHERE TABLE IS

0.0 IC 1000.00
80.00 EV 30.00 AZ

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RCV MATRIX	0	10	20	30	40	50	60	70	80	90	100	110	120
-120.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
-60.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
-30.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
30.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
60.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
120.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00

BSS MATRIX	130	140	150	160	170	180
130	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
140	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
150	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
160	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
170	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
180	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
190	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
200	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
210	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
220	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
230	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
240	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
250	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
260	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
270	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
280	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
290	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
300	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
310	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
320	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
330	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
340	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
350	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
360	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
370	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
380	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
390	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
400	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
410	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
420	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
430	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
440	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
450	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
460	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
470	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
480	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
490	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
500	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00

VULNERABLE AREA (SQ METERS) AS A FUNCTION OF IMPACT SPEED (METERS/SEC) AND ASPECT VIEW

VIEW	0	152	305	457	510	762	914	1067	1219
N	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
E	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
S	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
W	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
C	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
L	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
T	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
B	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
A	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
M	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
I	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
N	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
E	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
S	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
W	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
C	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
L	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
T	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
B	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
A	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
M	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05
I	0.00	12.54	15.85	17.35	18.55	19.45	20.15	20.65	21.05

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AFATL P-001 AAASIM " AIRCRAFT-CHEAT SURVIVABILITY-SCENARIO

PAGE 7

MULTIPATH INPUTS (INITIAL CR-CHANGED

ISMP =
REFC = 0.350

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AFATL P-001 AAASIM - AIRCRAFT COMBAT SURVIVABILITY SCENARIO

MULTIPATH INPUTS (INITIAL OR CHANGE)

IRMP = 2
REFC = 0.350

ECM INPUTS (INITIAL OR CHANGE)

IP = 5
IJ = 0
GAIN(JDB) = 1.00
PJM(W) = 500.00
PLEA(S) = 0.0
IX = 1
ASCC(SCN) = 0.0
GAIN = 1.000
SINMAX(08) = 17.03

JAMMER ANTENNA GAIN 1,000 DB

AIRCRAFT CROSS SECTION TABLE SPECIFIED. PRINTED VALUES WILL BE MULTIPLIED BY CALC. CALX = 1.00

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TABLE DATA
19 ELEMENTS FROM
17 ELEMENTS FROM
ELSEWHERE TABLE IS

0-00	50.00	100.00	150.00	200.00	250.00	300.00	350.00	400.00	450.00	500.00	550.00	600.00	650.00	700.00	750.00	800.00	850.00	900.00	950.00	1000.00	

AD-A057 907

NAVAL POSTGRADUATE SCHOOL MONTEREY CALIF
ADAPTATION OF THE IMPROVED ANTI-AIRCRAFT ARTILLERY SIMULATION CO--ETC(U)
MAR 78 C F SWENSON

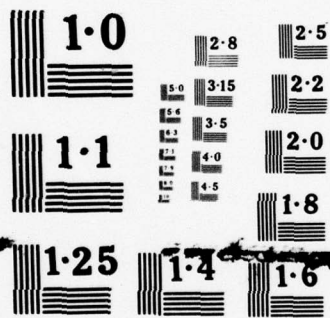
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NATIONAL BUREAU OF STANDARDS
MICROCOPY RESOLUTION TEST CHART

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BCS MATRIX	0	10	20	30	40	50	60	70	80	90	100	110	120
120.0	1000.00	1063.00	1066.00	1000.00	1000.00	1000.00	1205.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
120.0	1000.00	1063.00	1066.00	1000.00	1000.00	1000.00	1205.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
30.0	1000.00	1063.00	1066.00	1000.00	1000.00	1000.00	1205.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
40.0	1000.00	1063.00	1066.00	1000.00	1000.00	1000.00	1205.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
70.0	1000.00	1063.00	1066.00	1000.00	1000.00	1000.00	1205.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
90.0	1000.00	1063.00	1066.00	1000.00	1000.00	1000.00	1205.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
120.0	1000.00	1063.00	1066.00	1000.00	1000.00	1000.00	1205.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00

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KEY MATRIX 130 140 150 160 170 180
-120.0 100.00 100.00 100.00 100.00 100.00 100.00
-300.0 100.00 100.00 100.00 100.00 100.00 100.00
300.0 0.00 0.00 0.00 0.00 0.00 0.00
120.0 100.00 100.00 100.00 100.00 100.00 100.00

AFATL P-001 AAASIM - AIRCRAFT COMBAT SURVIVABILITY SCENARIO

VULNERABLE AREA (SQ METERS) AS A FUNCTION OF IMPACT SPEED (METERS/SEC) AND ASPECT VIEW

VIEW	0	152	305	457	610	762	914	1067	1219
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
0000	0000	0000	0000	0000	0000	0000	0000	0000	0000
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0000	0000	0000	0000	0000	0000	0000	0000	0000	0000

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AFATL P-001 AASIM - AIRCRAFT-COMBAT SURVIVABILITY SCENARIO

LOC	PKILL I	RCUNDS	FIRE TIME	XGUN	YGUN	ZGUN	RADIUS G1	Y REACT	Y TRACK	CH	CH	CH	CH	CH
1	0.0	0.0	0.0	14800.00	9000.00	40.00	0.0	0.0	0.0	1	1	1	1	1
2	0.0010946	17	5.25	12200.00	8700.00	40.00	0.0	0.0	0.0	1	1	1	1	1
3	0.0003691	0	6.80	12400.00	7200.00	20.00	0.0	0.0	0.0	1	1	1	1	1
4	0.0	0	0.0	11300.00	8000.00	20.00	0.0	0.0	0.0	1	1	1	1	1
5	0.0	0	0.0	11300.00	7000.00	20.00	0.0	0.0	0.0	1	1	1	1	1
6	0.012375	42	21.42	11300.00	1700.00	20.00	0.0	0.0	0.0	1	1	1	1	1
7	0.0884375	134	51.42	12800.00	17500.00	20.00	0.0	0.0	0.0	1	1	1	1	1

ATTRITION ACCRUED AS A FUNCTION OF TIME OF FIRE

TIME SEG	CUM FOR CLASS 1	DENSITY CLASS 1	DENSITY CLASS 2	DENSITY CLASS 3	DENSITY CLASS 4	DENSITY CLASS 5	DENSITY CLASS 6	DENSITY CLASS 7	DENSITY CLASS 8	DENSITY CLASS 9
1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0024045	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5	0.0033373	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0042701	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
7	0.0052029	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
8	0.0061357	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
9	0.0070685	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10	0.0080013	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

ATTRITION ACCRUED AS A FUNCTION OF TIME AT INTERCEPT

TIME SEG	CUM FOR CLASS 1	DENSITY CLASS 1	DENSITY CLASS 2	DENSITY CLASS 3	DENSITY CLASS 4	DENSITY CLASS 5	DENSITY CLASS 6	DENSITY CLASS 7	DENSITY CLASS 8	DENSITY CLASS 9
1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5	0.0025023	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0034351	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
7	0.0043679	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
8	0.0053007	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
9	0.0062335	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10	0.0071663	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
TOTALS	0.0802164	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

TOTAL PK FOR DENSITY CLASS 1 AS A FUNCTION OF ASPECT SECTOR AND IMPACT SPEED.

SECTOR	AZIMUTH	ELEV	0-152	152-305	305-457	457-610	610-762	762-914	914-1067	1067-1219	TOTAL PK
1	000-045	000-043	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2	045-130	000-043	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
3	090-135	000-045	0.0	0.0	0.0	0.0001340	0.0	0.0	0.0	0.0	0.0001340

AFATL P-001 AAASIM " AIRCRAFT-CCPBAT SURVIVABILITY SCENARIO

SECTOR	AZIMUTH	ELEV	0-152	152-305	305-457	457-610	610-762	762-914	914-1067	1067-1219	TOTAL PK
150-180	000-045	000-045	0.0003255	0.0012023	0.0002515	0.0002515	0.0002515	0.0002515	0.0002515	0.0002515	0.002515
180-210	000-045	000-045	0.000411	0.001518	0.000411	0.000411	0.000411	0.000411	0.000411	0.000411	0.00411
210-240	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
240-270	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
270-300	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
300-330	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
330-360	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
360-390	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
390-420	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
420-450	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
450-480	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
480-510	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
510-540	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
540-570	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
570-600	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
600-630	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
630-660	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
660-690	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
690-720	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
720-750	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
750-780	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
780-810	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
810-840	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
840-870	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
870-900	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
900-930	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
930-960	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
960-990	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
990-1020	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
1020-1050	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
1050-1080	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
1080-1110	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
1110-1140	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
1140-1170	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
1170-1200	000-045	000-045	0.000374	0.001374	0.000374	0.000374	0.000374	0.000374	0.000374	0.000374	0.00374
TOTALS			0.0001108	0.0047761	0.0075947	0.0079440	0.0180647	0.0438568	0.0	0.0	0.0902164

***** END OF JCB - - PC01 SCENARIO RUN COMPLETE *****

APPENDIX G

P001 ANALYSIS EXTENDED OUTPUT

MASK ANGLE FOR THIS RUN = 0.0 DEG.

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AFATL P-001 AAASIM - AIRCRAFT COMBAT SURVIVABILITY SCENARIO

TIME	XXX	YYY	ZZZ	SPEED	XOCT	YDOT	ZDOT	HEADING	CLIMB	ROLL
30	500	500	700	300	90	50	500	45	200	75
31	500	500	700	300	90	50	500	45	200	75
32	500	500	700	300	90	50	500	45	200	75
33	500	500	700	300	90	50	500	45	200	75
34	500	500	700	300	90	50	500	45	200	75
35	500	500	700	300	90	50	500	45	200	75
36	500	500	700	300	90	50	500	45	200	75
37	500	500	700	300	90	50	500	45	200	75
38	500	500	700	300	90	50	500	45	200	75
39	500	500	700	300	90	50	500	45	200	75
40	500	500	700	300	90	50	500	45	200	75
41	500	500	700	300	90	50	500	45	200	75
42	500	500	700	300	90	50	500	45	200	75
43	500	500	700	300	90	50	500	45	200	75
44	500	500	700	300	90	50	500	45	200	75
45	500	500	700	300	90	50	500	45	200	75
46	500	500	700	300	90	50	500	45	200	75
47	500	500	700	300	90	50	500	45	200	75
48	500	500	700	300	90	50	500	45	200	75
49	500	500	700	300	90	50	500	45	200	75
50	500	500	700	300	90	50	500	45	200	75
51	500	500	700	300	90	50	500	45	200	75
52	500	500	700	300	90	50	500	45	200	75
53	500	500	700	300	90	50	500	45	200	75
54	500	500	700	300	90	50	500	45	200	75
55	500	500	700	300	90	50	500	45	200	75
56	500	500	700	300	90	50	500	45	200	75
57	500	500	700	300	90	50	500	45	200	75
58	500	500	700	300	90	50	500	45	200	75
59	500	500	700	300	90	50	500	45	200	75
60	500	500	700	300	90	50	500	45	200	75

XR= 0.0 YR= 0.0 ZR= 0.0
 XT= 0.0 YT= 0.0 ZT= 0.0
 XZ= 0.0 YZ= 0.0 ZY= 0.0
 XZ= 0.0 YZ= 0.0 ZY= 0.0

AFATL P-001 AAASIM - AIRCRAFT COMBAT-SURVIVABILITY SCENARIO

10 TIME INTERVALS FOR PK ACCUMULATION
0.00 11.47 22.95 34.42 45.89 57.37 68.84 80.31 91.78 103.26 599.99

VULNERABLE AREA (SQ METERS) AS A FUNCTION OF IMPACT SPEED (METERS/SEC) AND ASPECT VIEW
VULNERABLE AREA TABLE VS TYPE 1 AND 2 WEAPONS

VIEW	0	152	305	457	610	762	914	1067	1219	VEL MIN	VEL MAX	RANGE MIN	RANGE MAX	SPCCTH CONST	MAX-AZ ERROR	MAX-EL ERROR
1	0.00	0.46	0.92	1.38	1.84	2.30	2.76	3.22	3.68	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
2	0.00	0.70	1.40	2.10	2.80	3.50	4.20	4.90	5.60	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
3	0.00	0.94	1.88	2.82	3.76	4.70	5.64	6.58	7.52	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
4	0.00	1.18	2.36	3.54	4.72	5.90	7.08	8.26	9.44	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
5	0.00	1.42	2.84	4.26	5.72	7.18	8.64	10.10	11.56	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
6	0.00	1.66	3.32	5.04	6.72	8.40	10.08	11.76	13.44	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
7	0.00	1.90	3.80	5.70	7.60	9.50	11.40	13.30	15.20	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
8	0.00	2.14	4.28	6.44	8.64	10.64	12.68	14.72	16.88	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
9	0.00	2.38	4.76	7.18	9.76	11.92	14.24	16.48	18.24	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
10	0.00	2.62	5.24	7.92	10.92	13.24	15.92	18.24	20.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
11	0.00	2.86	5.72	8.66	12.16	14.56	17.68	20.32	22.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
12	0.00	3.10	6.20	9.40	13.40	15.90	19.32	22.72	24.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
13	0.00	3.34	6.68	10.14	14.64	17.24	21.16	25.44	26.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
14	0.00	3.58	7.16	10.88	15.88	18.58	22.80	27.88	28.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
15	0.00	3.82	7.64	11.62	17.12	20.00	24.64	30.32	30.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
16	0.00	4.06	8.12	12.36	18.36	21.42	26.48	32.84	32.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
17	0.00	4.30	8.60	13.10	19.60	22.80	28.32	35.28	34.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
18	0.00	4.54	9.08	13.84	20.84	24.18	30.16	37.72	36.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
19	0.00	4.78	9.56	14.58	22.08	25.56	32.00	40.16	38.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
20	0.00	5.02	10.04	15.32	23.32	26.94	33.84	42.60	40.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
21	0.00	5.26	10.52	16.06	24.56	28.32	35.68	45.04	42.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
22	0.00	5.50	11.00	16.80	25.80	29.70	37.52	47.48	44.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
23	0.00	5.74	11.48	17.54	27.04	31.08	39.36	50.00	46.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
24	0.00	5.98	11.96	18.28	28.28	32.46	41.20	52.44	48.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
25	0.00	6.22	12.44	19.02	29.52	33.84	43.04	54.88	50.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730
26	0.00	6.46	12.92	19.76	30.76	35.22	44.88	57.32	52.00	0.0	300.0	0.0	3000.0	999.99	5.730	5.730

AFATL P-001 AAASIM * AIRCRAFT-COMBAT-SURVIVABILITY-SCENARIO
 LOCATION 1 GUN TYPE 1 ERROR MODE 1 POSITION=(14800.0, 9000.0, 40.0) RADIUS= 0.0 M

FIRE FLY: INTCP FIRE INTCP CLOSE AZIM: ELEV: MEAN PEAN VULN
 TIME RANGE RANGE SIG1 SIG2 BIAS1 BIAS2 VEL. RATE RATE AZ-ERR EL-ERR AREA SHOT PK CUM.PK

PK AS A FUNCTION OF ASPECT SECTOR AND IMPACT SPEED.

SECTOR	MINIMUM RANGE=00	MAXIMUM RANGE=00	ELEV DUM=00	0-132	132-303	303-437	437-610	GT 1	EM 1	X 14800.	Y 5000.	Z 1067-1219	RADIUS	0.0 M	TOTAL PK
1	00-045	00-045	00-045	00	00	00	00	00	00	00	00	00	00	00	00
2	00-050	00-045	00-045	00	00	00	00	00	00	00	00	00	00	00	00
3	00-035	00-045	00-045	00	00	00	00	00	00	00	00	00	00	00	00
4	00-180	00-045	00-045	00	00	00	00	00	00	00	00	00	00	00	00
5	00-270	00-045	00-045	00	00	00	00	00	00	00	00	00	00	00	00
6	00-315	00-045	00-045	00	00	00	00	00	00	00	00	00	00	00	00
7	00-300	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
8	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
9	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
10	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
11	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
12	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
13	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
14	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
15	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
16	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
17	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
18	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
19	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
20	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
21	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
22	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
23	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
24	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
25	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
26	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
27	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
28	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
29	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
30	00-135	00-090	00-090	00	00	00	00	00	00	00	00	00	00	00	00
TOTALS	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00

AFATL P-001 AAASIM - AIRCRAFT COMBAT SURVIVABILITY SCENARIO

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PK AS A FUNCTION OF ASPECT SECTOR AND IMPACT SPEED.		LOC 3	GT 2	EM 1	X 13600.	Y 7200.	Z 20.	RADIUS 0.0		
SECTOR	ALTIMUTH	0-152	152-305	305-457	457-610	610-762	762-914	914-1067	1067-1219	TOTAL PK
28	180-225	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
29	225-270	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
30	270-315	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
31	315-360	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
TOTALS		0.0	0.0	0.000267	0.0003423	0.0	0.0	0.0	0.0	0.0003690

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AFATL P-001 AAASIM - AIRCRAFT COMBAT SURVIVABILITY SCENARIO
 LOCATION 4 GUN TYPE 2 ERROR MODE 1 POSITION=(1 13400.0, 8000.0, 20.0) RADIUS= 0.0 M

L DEM TIME FLT INTCP FIRE INTCP CLOSE AZIM ELEV MEAN VULN
 TIME RANGE RANGE RANGE VELOCITY RATE RATE EL.ERR AREA SHOT PK CUM.PK

PK AS A FUNCTION OF ASPECT SECTOR AND IMPACT SPEED. LOC 4 GT 2 EM 1 X 13400. Y 8000. Z 20. RADIUS 0.0
 SECTOR 0-152 152-305 305-457 457-610 610-762 762-914 914-1007 1007-1219 TOTAL PK

SECTOR	LOC	GT	EM	X	Y	Z	RADIUS	TOTAL PK
0-152	4	2	1	13400.0	8000.0	20.0	0.0	0.0
152-305	4	2	1	13400.0	8000.0	20.0	0.0	0.0
305-457	4	2	1	13400.0	8000.0	20.0	0.0	0.0
457-610	4	2	1	13400.0	8000.0	20.0	0.0	0.0
610-762	4	2	1	13400.0	8000.0	20.0	0.0	0.0
762-914	4	2	1	13400.0	8000.0	20.0	0.0	0.0
914-1007	4	2	1	13400.0	8000.0	20.0	0.0	0.0
1007-1219	4	2	1	13400.0	8000.0	20.0	0.0	0.0
TOTALS								0.0

ECM INPUTS (INITIAL OR CHANGED)

IP = 5
 IJ = 0
 GAIN(JDB) = 1.00
 PJW(L) = 500.00
 PLEN(S) = 0.0
 IX = 1
 SPEC(SOM) = 0.0
 CALX = 1.000
 JTRAX(DB) = 1.00

JAMMER ANTENNA GAIN 1.000 DB

AIRCRAFT CROSS SECTION TABLE SPECIFIED. PRINTED VALUES WILL BE MULTIPLIED BY CALX. CALX= 1.00

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TABLE DATA

19 ELEMENTS FROM 0.0 IO 100.00 BY 10.00 AZ
20 ELEMENTS FROM 100.00 IO 200.00 BY 20.00 AZ
ELSEWHERE TABLE IS 1000.00

REV	0	10	20	30	40	50	60	70	80	90	100	110	120
BZY MATRIX	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
-1	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
-2	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
30	0.27	0.18	0.05	0.02	0.14	0.07	0.04	0.27	0.23	0.03	0.02	0.01	0.01
60	0.27	0.18	0.05	0.02	0.14	0.07	0.04	0.27	0.23	0.03	0.02	0.01	0.01
90	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
120	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00

VA 1118 01111111

AFATL P-001 AAASIM - AIRCRAFT COMBAT SURVIVABILITY SCENARIO

VULNERABLE AREA (SQ METERS) AS A FUNCTION OF IMPACT SPEED (METERS/SEC) AND ASPECT VIEW
VULNERABLE AREA TABLE VS TYPE WEAPONS

VIEW	0	152	305	457	610	762	914	1067	1219
N	00	12	42	72	102	132	162	192	222
S	00	12	42	72	102	132	162	192	222
E	00	12	42	72	102	132	162	192	222
A	00	12	42	72	102	132	162	192	222
R	00	12	42	72	102	132	162	192	222
L	00	12	42	72	102	132	162	192	222
O	00	12	42	72	102	132	162	192	222
D	00	12	42	72	102	132	162	192	222
U	00	12	42	72	102	132	162	192	222
M	00	12	42	72	102	132	162	192	222
B	00	12	42	72	102	132	162	192	222
C	00	12	42	72	102	132	162	192	222
V	00	12	42	72	102	132	162	192	222
E	00	12	42	72	102	132	162	192	222
N	00	12	42	72	102	132	162	192	222
A	00	12	42	72	102	132	162	192	222
S	00	12	42	72	102	132	162	192	222
E	00	12	42	72	102	132	162	192	222
R	00	12	42	72	102	132	162	192	222
I	00	12	42	72	102	132	162	192	222
O	00	12	42	72	102	132	162	192	222
N	00	12	42	72	102	132	162	192	222
S	00	12	42	72	102	132	162	192	222
A	00	12	42	72	102	132	162	192	222
N	00	12	42	72	102	132	162	192	222

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AFATL P-001 AAASIM - AIRCRAFT COMBAT SURVIVABILITY SCENARIO

MULTIPATH INPUTS INITIAL OR CHANGED

IRMP =
REPC = 0.350

AFATL P-001 AAASIM - AIRCRAFT COMBAT SURVIVABILITY SCENARIO

LOCATION	6	GUN TYPE 3	ERROR MODE 3	SIGL	SI02	BLAS1	BLAS2	CLOSE VEL.	AZIM RATE	ELEV RATE	MEAN AZ. ERR	MEAN ELEV. ERR	O-C M	VOLUM AREA	SHOT PK	CUP.PK
	1	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	2	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	3	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	4	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	5	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	6	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	7	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	8	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	9	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	10	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	11	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	12	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	13	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	14	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	15	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	16	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	17	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	18	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	19	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	20	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	21	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	22	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	23	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	24	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	25	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	26	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	27	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	28	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	29	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00
	30	00	00	00	00	00	00	00	00	00	00	00	00	00	00	00

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AFATL P-001 AASIM - AIRCRAFT CP8AT-SURVIVABILITY SCENARIO

LOCATION 6 GUN TYPE 3 ERCR MODE 3 POSITION=(15600.0, 10900.0, 90.0) RADIUS= C.O.M

LOC	FIRE TIME	INTCP TIME	FIRE RANGE	INTCP RANGE	SIG1	SIG2	BIAS1	BIAS2	CLSE VCL	AZIM RATE	ELEV RATE	MEAN AZ. ERR	MEAN EL. ERR	WULN AREA	SHOT PK	CUM. PK
1	87	011	543	200	20	40	0	0	200	0	0	0	0	70	0003	00070
2	88	011	543	200	20	40	0	0	200	0	0	0	0	70	0004	00077
3	89	011	543	200	20	40	0	0	200	0	0	0	0	70	0005	00084
4	90	011	543	200	20	40	0	0	200	0	0	0	0	70	0006	00091
5	91	011	543	200	20	40	0	0	200	0	0	0	0	70	0007	00098
6	92	011	543	200	20	40	0	0	200	0	0	0	0	70	0008	00105
7	93	011	543	200	20	40	0	0	200	0	0	0	0	70	0009	00112
8	94	011	543	200	20	40	0	0	200	0	0	0	0	70	0010	00119
9	95	011	543	200	20	40	0	0	200	0	0	0	0	70	0011	00126
10	96	011	543	200	20	40	0	0	200	0	0	0	0	70	0012	00133
11	97	011	543	200	20	40	0	0	200	0	0	0	0	70	0013	00140
12	98	011	543	200	20	40	0	0	200	0	0	0	0	70	0014	00147
13	99	011	543	200	20	40	0	0	200	0	0	0	0	70	0015	00154
14	100	011	543	200	20	40	0	0	200	0	0	0	0	70	0016	00161
15	101	011	543	200	20	40	0	0	200	0	0	0	0	70	0017	00168
16	102	011	543	200	20	40	0	0	200	0	0	0	0	70	0018	00175
17	103	011	543	200	20	40	0	0	200	0	0	0	0	70	0019	00182
18	104	011	543	200	20	40	0	0	200	0	0	0	0	70	0020	00189
19	105	011	543	200	20	40	0	0	200	0	0	0	0	70	0021	00196
20	106	011	543	200	20	40	0	0	200	0	0	0	0	70	0022	00203
21	107	011	543	200	20	40	0	0	200	0	0	0	0	70	0023	00210
22	108	011	543	200	20	40	0	0	200	0	0	0	0	70	0024	00217
23	109	011	543	200	20	40	0	0	200	0	0	0	0	70	0025	00224
24	110	011	543	200	20	40	0	0	200	0	0	0	0	70	0026	00231
25	111	011	543	200	20	40	0	0	200	0	0	0	0	70	0027	00238
26	112	011	543	200	20	40	0	0	200	0	0	0	0	70	0028	00245
27	113	011	543	200	20	40	0	0	200	0	0	0	0	70	0029	00252
28	114	011	543	200	20	40	0	0	200	0	0	0	0	70	0030	00259
29	115	011	543	200	20	40	0	0	200	0	0	0	0	70	0031	00266
30	116	011	543	200	20	40	0	0	200	0	0	0	0	70	0032	00273
31	117	011	543	200	20	40	0	0	200	0	0	0	0	70	0033	00280
32	118	011	543	200	20	40	0	0	200	0	0	0	0	70	0034	00287
33	119	011	543	200	20	40	0	0	200	0	0	0	0	70	0035	00294
34	120	011	543	200	20	40	0	0	200	0	0	0	0	70	0036	00301
35	121	011	543	200	20	40	0	0	200	0	0	0	0	70	0037	00308
36	122	011	543	200	20	40	0	0	200	0	0	0	0	70	0038	00315
37	123	011	543	200	20	40	0	0	200	0	0	0	0	70	0039	00322
38	124	011	543	200	20	40	0	0	200	0	0	0	0	70	0040	00329
39	125	011	543	200	20	40	0	0	200	0	0	0	0	70	0041	00336
40	126	011	543	200	20	40	0	0	200	0	0	0	0	70	0042	00343
41	127	011	543	200	20	40	0	0	200	0	0	0	0	70	0043	00350
42	128	011	543	200	20	40	0	0	200	0	0	0	0	70	0044	00357
43	129	011	543	200	20	40	0	0	200	0	0	0	0	70	0045	00364
44	130	011	543	200	20	40	0	0	200	0	0	0	0	70	0046	00371
45	131	011	543	200	20	40	0	0	200	0	0	0	0	70	0047	00378
46	132	011	543	200	20	40	0	0	200	0	0	0	0	70	0048	00385
47	133	011	543	200	20	40	0	0	200	0	0	0	0	70	0049	00392
48	134	011	543	200	20	40	0	0	200	0	0	0	0	70	0050	00399
49	135	011	543	200	20	40	0	0	200	0	0	0	0	70	0051	00406
50	136	011	543	200	20	40	0	0	200	0	0	0	0	70	0052	00413
51	137	011	543	200	20	40	0	0	200	0	0	0	0	70	0053	00420
52	138	011	543	200	20	40	0	0	200	0	0	0	0	70	0054	00427
53	139	011	543	200	20	40	0	0	200	0	0	0	0	70	0055	00434
54	140	011	543	200	20	40	0	0	200	0	0	0	0	70	0056	00441
55	141	011	543	200	20	40	0	0	200	0	0	0	0	70	0057	00448
56	142	011	543	200	20	40	0	0	200	0	0	0	0	70	0058	00455
57	143	011	543	200	20	40	0	0	200	0	0	0	0	70	0059	00462
58	144	011	543	200	20	40	0	0	200	0	0	0	0	70	0060	00469
59	145	011	543	200	20	40	0	0	200	0	0	0	0	70	0061	00476
60	146	011	543	200	20	40	0	0	200	0	0	0	0	70	0062	00483
61	147	011	543	200	20	40	0	0	200	0	0	0	0	70	0063	00490
62	148	011	543	200	20	40	0	0	200	0	0	0	0	70	0064	00497
63	149	011	543	200	20	40	0	0	200	0	0	0	0	70	0065	00504
64	150	011	543	200	20	40	0	0	200	0	0	0	0	70	0066	00511
65	151	011	543	200	20	40	0	0	200	0	0	0	0	70	0067	00518
66	152	011	543	200	20	40	0	0	200	0	0	0	0	70	0068	00525
67	153	011	543	200	20	40	0	0	200	0	0	0	0	70	0069	00532
68	154	011	543	200	20	40	0	0	200	0	0	0	0	70	0070	00539
69	155	011	543	200	20	40	0	0	200	0	0	0	0	70	0071	00546
70	156	011	543	200	20	40	0	0	200	0	0	0	0	70	0072	00553
71	157	011	543	200	20	40	0	0	200	0	0	0	0	70	0073	00560
72	158	011	543	200	20	40	0	0	200	0	0	0	0	70	0074	00567
73	159	011	543	200	20	40	0	0	200	0	0	0	0	70	0075	00574
74	160	011	543	200	20	40	0	0	200	0	0	0	0	70	0076	00581
75	161	011	543	200	20	40	0	0	200	0	0	0	0	70	0077	00588
76	162	011	543	200	20	40	0	0	200	0	0	0	0	70	0078	00595
77	163	011	543	200	20	40	0	0	200	0	0	0	0	70	0079	00602
78	164	011	543	200	20	40	0	0	200	0	0	0	0	70	0080	00609
79	165	011	543	200	20	40	0	0	200	0	0	0	0	70	0081	00616
80	166	011	543	200	20	40	0	0	200	0	0	0	0	70	0082	00623
81	167	011	543	200	20	40	0	0	200	0	0	0	0	70	0083	00630
82	168	011	543	200	20	40	0	0	200	0	0	0	0	70	0084	00637
83	169	011	543	200	20	40	0	0	200	0	0	0	0	70	0085	00644
84	170	011	543	200	20	40	0	0	200	0	0	0	0	70	0086	00651
85	171	011	543	200	20	40	0	0	200	0	0	0	0	70	0087	00658
86	172	011	543	200	20	40	0	0	200	0	0	0	0	70	0088	00665
87	173	011	543	200	20	40	0	0	200	0	0	0	0	70	0089	00672
88	174	011	543	200	20	40	0	0	200	0	0	0	0	70	0090	00679
89	175	011	543	200	20	40	0	0	200	0	0	0	0	70	0091	00686
90	176	011	543	200	20	40	0	0	200	0	0	0	0	70	0092	00693
91	177	011	543	200	20	40	0	0	200	0	0	0	0	70	0093	00700
92	178	011	543	200	20	40	0	0	200	0	0	0	0	70	0094	00707
93	179	011	543	200	20	40	0	0	200	0	0	0	0	70	0095	00714
94	180	011	543	200	20	40	0	0	200	0	0	0	0</			

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AFATL P-001 AAASIM - AIRCRAFT-COMBAT-SURVIVABILITY-SCENARIO

M N 90.01 RADIUS=

90.01 POSITION=(15600.0, 10900.0,

15600.0, 10900.0,

G.C.H

LOCATION 6 GUN TYPE 5 ERROR CODE 3

N DEH	FIRE TIME	FLT TIME	INTCP TIME	FIRE RANGE	INTCP RANGE	SIGL	SIG2	BIAS1	BIAS2	CLOSE VEL.	AZTH RATE	ELEV RATE	MEAN AZERR	MEAN ELERR	VULN AREA	SHOT PK	CUM.PK
1	9999999999999999	9999999999999999	9999999999999999	9999999999999999	9999999999999999	9999999999999999	9999999999999999	9999999999999999	9999999999999999	9999999999999999	9999999999999999	9999999999999999	9999999999999999	9999999999999999	9999999999999999	9999999999999999	9999999999999999

LOCATION	6	GUN	TYPE	3	ERRC	MODE	3	POSITION	1	15600.0	10900.0	90.01	RAD	US	0.0	M	MEAN	ELERR	VUM	AREA	SHOT	PK	CUP	PK
AFATL	P-001	AAASIM	ARCRAFT	COMBAT	SURVIVABILITY	SCENARIO																		

AFATL P-001 AAASIM - AIRCRAFT COMBAT SURVIVABILITY SCENARIO

PK AS A FUNCTION OF ASPECT SECTOR AND IMPACT SPEED.		LOC 6	GT 3	EM 3	X 15600.	Y 16500.	Z 90.	RADIUS 0.4		
SECTOR	AZIMUTH ELEV	0-152	152-305	305-457	457-610	610-762	762-914	914-1067	1067-1219	TOTAL PK
32	315-360 135-180	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
TOTALS		0.0	0.0039258	0.0073024	0.0	0.0	0.0	0.0	0.0	0.0111996

MULTIPATH INPUTS INITIAL OR CHANGED

IBMP = 2
REFC = 0.350

ECM INPUTS (INITIAL OR CHANGED)

IP = 5
IJ = 0
GAIN(OBI) = 1.00
PJ(U) = 300.00
PL(NTS) = 0.0
XSEC(SOH) = 1
CALX = 1.00
TREC = 1.000
SUTRAX(OB) = 17.00

JAMMER ANTENNA GAIN 1.000 DB

AIRCRAFT CROSS SECTION TABLE SPECIFIED. PRINTED VALUES WILL BE MULTIPLIED BY CALX. CALX = 1.00

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TABLE DATA
19 ELEMENTS FROM 0-0 TO 30:00 AZ
7 ELEMENTS FROM 90:00 BY 30:00 AZ
ELSEWHERE TABLE IS 1000.00

	130	140	150	160	170	180
ECV MATRIX	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
-120.0	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
-50.0	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
-30.0	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
30.0	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
60.0	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00
120.0	1000.00	1000.00	1000.00	1000.00	1000.00	1000.00

AFATL P-001 AAASIM - AIRCRAFT COMBAT-SURVIVABILITY SCENARIO
LOCATION 7 GUN TYPE 5 ERROR MODE 3 POSITION=(12800.0, 7500.0, 20.01) RADIUS= 0.0 M

N	DER	FIRE TIME	FLY TIME	INCP TIME	INCP RANGE	INTCP	STGL	SIG2	BIAS1	BIAS2	CLOSE VEL	AZIM RATE	ELEV RATE	MEAN ALT. ERR	MEAN FEAR	VULN AREA	SHOT PK	GUN PK
0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1
2	2	2	2	2	2	2	2	2	2	2	2	2	2	2	2	2	2	2
3	3	3	3	3	3	3	3	3	3	3	3	3	3	3	3	3	3	3
4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4
5	5	5	5	5	5	5	5	5	5	5	5	5	5	5	5	5	5	5
6	6	6	6	6	6	6	6	6	6	6	6	6	6	6	6	6	6	6
7	7	7	7	7	7	7	7	7	7	7	7	7	7	7	7	7	7	7
8	8	8	8	8	8	8	8	8	8	8	8	8	8	8	8	8	8	8
9	9	9	9	9	9	9	9	9	9	9	9	9	9	9	9	9	9	9
0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1
2	2	2	2	2	2	2	2	2	2	2	2	2	2	2	2	2	2	2
3	3	3	3	3	3	3	3	3	3	3	3	3	3	3	3	3	3	3
4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4
5	5	5	5	5	5	5	5	5	5	5	5	5	5	5	5	5	5	5
6	6	6	6	6	6	6	6	6	6	6	6	6	6	6	6	6	6	6
7	7	7	7	7	7	7	7	7	7	7	7	7	7	7	7	7	7	7
8	8	8	8	8	8	8	8	8	8	8	8	8	8	8	8	8	8	8
9	9	9	9	9	9	9	9	9	9	9	9	9	9	9	9	9	9	9
0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1
2	2	2	2	2	2	2	2	2	2	2	2	2	2	2	2	2	2	2
3	3	3	3	3	3	3	3	3	3	3	3	3	3	3	3	3	3	3
4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4
5	5	5	5	5	5	5	5	5	5	5	5	5	5	5	5	5	5	5
6	6	6	6	6	6	6	6	6	6	6	6	6	6	6	6	6	6	6
7	7	7	7	7	7	7	7	7	7	7	7	7	7	7	7	7	7	7
8	8	8	8	8	8	8	8	8	8	8	8	8	8	8	8	8	8	8
9	9	9	9	9	9	9	9	9	9	9	9	9	9	9	9	9	9	9
0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1
2	2	2	2	2	2	2	2	2	2	2	2	2	2	2	2	2	2	2
3	3	3	3	3	3	3	3	3	3	3	3	3	3	3	3	3	3	3
4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4
5	5	5	5	5	5	5	5	5	5	5	5	5	5	5	5	5	5	5
6	6	6	6	6	6	6	6	6	6	6	6	6	6	6	6	6	6	6
7	7	7	7	7	7	7	7	7	7	7	7	7	7	7	7	7	7	7
8	8	8	8	8	8	8	8	8	8	8	8	8	8	8	8	8	8	8
9	9	9	9	9	9	9	9	9	9	9	9	9	9	9	9	9	9	9

AFATL P-001 AAASIM - AIRCRAFT COMBAT SURVIVABILITY SCENARIO

LOC	PKILL	RCUNDS	FIRE TIME	XCUM	YCUM	ZCUM	RAD: US GL	T REACT	T TRACK	GT	EM	CR	SR	LOC
1	0.0010546	17	5.75	1420.00	500.00	40.00	0.00	0.00	2.500	1	1	1	1	1
2	0.0003651	0	6.80	1360.00	720.00	20.00	0.00	0.00	2.500	1	1	1	1	1
3	0.0	0	0.0	1340.00	800.00	20.00	0.00	0.00	2.500	1	1	1	1	1
4	0.0	0	0.0	1130.00	970.00	20.00	0.00	0.00	2.500	1	1	1	1	1
5	0.0112059	422	21.10	1560.00	1090.00	20.00	0.00	0.00	6.000	3	3	3	3	3
6	0.0684375	134	21.42	1280.00	750.00	20.00	0.00	0.00	6.000	3	3	3	3	3

LOC	TIME 1	TIME 2	TIME 3	TIME 4	TIME 5	TIME 6	TIME 7	TIME 8	TIME 9	TIME10
1	11.47	22.95	34.42	45.89	57.37	68.84	80.31	91.78	103.26	599.99
2	11.47	22.95	34.42	45.89	57.37	68.84	80.31	91.78	103.26	599.99
3	11.47	22.95	34.42	45.89	57.37	68.84	80.31	91.78	103.26	599.99
4	11.47	22.95	34.42	45.89	57.37	68.84	80.31	91.78	103.26	599.99
5	11.47	22.95	34.42	45.89	57.37	68.84	80.31	91.78	103.26	599.99

LOC	PK(1F 1)	PK(1F 2)	PK(1F 3)	PK(1F 4)	PK(1F 5)	PK(1F 6)	PK(1F 7)	PK(1F 8)	PK(1F 9)	PK(1F10)
1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
7	0.0	0.0	0.0	0.0	0.024045	0.001261	0.2195878	0.0463743	0.008801	0.0
8	0.0	0.0	0.0	0.0	0.0024045	0.0025303	0.0220685	0.0676150	0.0084355	0.0

LOC	PK(1I 1)	PK(1I 2)	PK(1I 3)	PK(1I 4)	PK(1I 5)	PK(1I 6)	PK(1I 7)	PK(1I 8)	PK(1I 9)	PK(1I10)
1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.003690	0.0	0.0
5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

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LOC	PK(TI 1)	PK(TI 2)	PK(TI 3)	PK(TI 4)	PK(TI 5)	PK(TI 6)	PK(TI 7)	PK(TI 8)	PK(TI 9)	PK(TI 10)
7	0.0	0.0	0.0	0.0	0.0	0.0025023	0.016464	0.0491013	0.005465	0.0003027
7	CUM	0.0	0.0	0.0	0.0	0.0025023	0.0191087	0.0672899	0.0681927	0.0684956
LOC	RMD 1	RMD 2	RMD 3	RMD 4	RMD 5	RMD 6	RMD 7	RMD 8	RMD 9	RMD 9
1	1.00000									
2	1.00000									
3	1.00000									
4	1.00000									
5	1.00000									
6	1.00000									

ATTRITION ACCRUED AS A FUNCTION OF TIME OF FIRE

TIME SEG.	CUM FOR CLASS 1	DENSITY CLASS 1	DENSITY CLASS 2	DENSITY CLASS 3	DENSITY CLASS 4	DENSITY CLASS 5	DENSITY CLASS 6	DENSITY CLASS 7	DENSITY CLASS 8	DENSITY CLASS 9
1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0024045	0.0024045	0.001261	0.001261	0.001261	0.001261	0.001261	0.001261	0.001261	0.001261
5	0.023303	0.023303	0.013874	0.013874	0.013874	0.013874	0.013874	0.013874	0.013874	0.013874
6	0.0220885	0.0220885	0.0143268	0.0143268	0.0143268	0.0143268	0.0143268	0.0143268	0.0143268	0.0143268
7	0.0653287	0.0653287	0.0078437	0.0078437	0.0078437	0.0078437	0.0078437	0.0078437	0.0078437	0.0078437
8	0.0766286	0.0766286	0.0078437	0.0078437	0.0078437	0.0078437	0.0078437	0.0078437	0.0078437	0.0078437
9	0.0802163	0.0802163	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000
10	0.0802163	0.0802163	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000

ATTRITION ACCRUED AS A FUNCTION OF TIME AT INTERCEPT

TIME SEG.	CUM FOR CLASS 1	DENSITY CLASS 1	DENSITY CLASS 2	DENSITY CLASS 3	DENSITY CLASS 4	DENSITY CLASS 5	DENSITY CLASS 6	DENSITY CLASS 7	DENSITY CLASS 8	DENSITY CLASS 9
1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0025023	0.0025023	0.016464	0.016464	0.016464	0.016464	0.016464	0.016464	0.016464	0.016464
5	0.0191067	0.0191067	0.044522	0.044522	0.044522	0.044522	0.044522	0.044522	0.044522	0.044522
6	0.0676140	0.0676140	0.0090451	0.0090451	0.0090451	0.0090451	0.0090451	0.0090451	0.0090451	0.0090451
7	0.0760475	0.0760475	0.0045121	0.0045121	0.0045121	0.0045121	0.0045121	0.0045121	0.0045121	0.0045121
8	0.0802164	0.0802164	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000
9	0.0802164	0.0802164	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000
10	0.0802164	0.0802164	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000
TOTALS	0.0802243	0.0802243								

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TOTAL PK FOR DENSITY CLASS 1 AS A FUNCTION OF ASPECT SECTOR AND IMPACT SPEED.

SECTOR	ALGUTH	ELEV	U	0-152	152-305	305-457	457-610	610-762	762-914	914-1667	1067-1219	TOTAL PK
N	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
S	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.001340	0.00	0.00	0.00	0.001340
E	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.002519	0.00	0.00	0.00	0.002519
W	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
030	0.00	0.00	0.00	0.00	0.0003255	0.0012023	0.00	0.00	0.00	0.00	0.00	0.0015274
060	0.00	0.00	0.00	0.00	0.00070411	0.00	0.00	0.00	0.00	0.00	0.00	0.00070411
090	0.00	0.00	0.00	0.00	0.0001518	0.00	0.00	0.00	0.00	0.00	0.00	0.0001518
120	0.00	0.00	0.00	0.0001108	0.000003374	0.00	0.00	0.00	0.00	0.00	0.00	0.0001108
150	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
180	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.007089	0.00	0.00	0.00	0.007089
210	0.00	0.00	0.00	0.00	0.0058741	0.00	0.0025407	0.001866	0.0030211	0.00	0.00	0.007089
240	0.00	0.00	0.00	0.00	0.0016015	0.00	0.00	0.00	0.00	0.00	0.00	0.0016015
270	0.00	0.00	0.00	0.00	0.00	0.000250	0.00	0.00	0.00	0.00	0.00	0.000250
300	0.00	0.00	0.00	0.00	0.0006235	0.00	0.00	0.00	0.00	0.00	0.00	0.0006235
330	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
360	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
030	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
060	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
090	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
120	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
150	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
180	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.0001868	0.00	0.00	0.00	0.0001868
210	0.00	0.00	0.00	0.00	0.00	0.00	0.00054171	0.00	0.00	0.00	0.00	0.00054171
240	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
270	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
300	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
330	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
360	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
TOTALS	0.00	0.00	0.00	0.0001108	0.0047761	0.0075947	0.0079440	0.0180647	0.0436568	0.00	0.00	0.0802164

**** END OF JOB - P001 SCENARIO RUN COMPLETE ****

APPENDIX H

POOL INPUT PROGRAM (PIP) LISTING

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*****
THIS PROGRAM WILL PUNCH ALL REQUIRED CARDS FOR THE EXECUTION OF THE
POOL ANALYSIS WITH THE EXCEPTION OF THE FIRST CARD (THE GREEN JOB
CARD) AND THE LAST CARD (THE ORANGE END OF FILE CARD). THE MINIMAL
INPUT TO THIS PROGRAM IS THE X, Y AND Z COORDINATES FOR EACH OF THE
FLIGHT PATH MILESTONES. SIX GUN EMPLACEMENT LOCATIONS MAY BE
SPECIFIED IF THE SIX PRESET GUN LOCATIONS ARE NOT DESIRED. IF THE
PRESET GUN EMPLACEMENT LOCATIONS ARE USED, THE FINAL DATA CARD IS A
CONTROL CARD THAT SPECIFIES THE DESIRED INPUT, OUTPUT AND SCENARIO
OPTIONS. IF THE PRESET GUN LOCATIONS ARE NOT USED, THE FINAL DATA
CARDS WILL BE THE 6 INPUT GUN LOCATION CARDS.
MILESTONE CARDS: THE X, Y, Z COORDINATES FOR UP TO 199 FLIGHT PATH
MILESTONES MUST BE ENTERED IN 3F10.0 FORMAT, ONE MILESTONE PER INPUT
CARD. THE VALUES (IN METERS) MUST BE ENTERED IN DECIMAL FORM WITH THE
X COORDINATE IN COLUMNS 1-10, THE Y COORDINATE IN COLUMNS 11-20 AND
Z COORDINATE IN COLUMNS 21-30. THE DECIMAL POINT MUST APPEAR IN EACH
COORDINATE VALUE.
MILESTONE DELIMITER CARD: A CARD CONTAINING 99999. LEFT JUSTIFIED
(INPUT FORMAT F6.0) MUST BE PLACED AFTER THE FINAL MILESTONE CARD TO
MARK THE END OF THE MILESTONE DATA INPUT.
CONTROL CARD: A CONTROL CARD SPECIFYING THE CRUISE SPEED, THE NUMBER
OF THE BOMB RELEASE MILESTONE (COUNT THE INITIAL POSITION AS MILE-
STONE 1) THE "99999." CARD. THE DATA ON THE OPTION CARD MUST BE
FOLLOWING:
SPECIFIED IN F10.0, I2, 8I1, F10.0 FORMAT.
F10.0: THE AIRCRAFT CRUISE SPEED IN METERS PER SECOND.
I2: THE NUMBER OF THE BOMB RELEASE MILESTONE.
I1: EW OPTION - 0 FOR NO EW; 1 FOR EW (WITH EW, AN AIRBORNE
JAMMER IS USED TO JAM TRACKING RADARS.
I1: ANTI-JAM CAPABILITY - 0 FOR NO AJ; 1 FOR AJ (WITH AJ, RADAR
SYSTEMS WITH THE PATH OPTION USE AJ AGAINST THE JAMMER.)
I1: MULTIPATH OPTION - 0 FOR NO MULTIPATH; 1 FOR MULTIPATH (IF
SPECIFIED, THAT ARE AFFECTED BY RADAR MULTIPATH ARE INCLUDED IN THE RADAR
SYSTEMS THAT ARE AFFECTED BY MULTIPATH.)
I1: GUN EMPLACEMENT LOCATION INPUT OPTION - 0 FOR PRESET GUN
LOCATIONS; 1 FOR GUN EMPLACEMENT LOCATIONS INPUT AS DATA.
I1: LIST OF POOL INPUT DECK NO LISTED AS PART OF THE OUTPUT
1 FOR LIST OF POOL INPUT DECK NO PROVIDED AS PART OF THE OUTPUT;
I1: PUNCH OPTION - 0 FOR NO PUNCH INPUT PUNCHED CARDS DESIRED;
I1: PLOT OPTION - 0 FOR NO PLOT OF FLIGHT PATH AND GUN LOCATIONS
DESIRED; 1 FOR PLOT DESIRED.
*****

```

245
250
255
260
265
270
275
280
285
290
295
300
305
310
315
320
325
330
335
340
345
350
355
360
365
370
375
380
385
390
395
400
405
410
415
420
425
430
435
440
445
450
455
460
465
470
475
480

11: EXTENDED OUTPUT OPTION - 0 FCR NO EXTENDED OUTPUT DESIRED;
1 FOR EXTENDED PRINTOUT DESIRED AS OUTPUT.
F10.0: JAMMER PCWER - MUST BE BETWEEN 0 TC 1000 WATTS.

GUN EMPLOYMENT LOCATION CARDS: IF DESIRED AS INPUT DATA, THE 6 GUN
THE SAME FORMAT AS THE MILESTONE CARDS (3F10.0). THE GUN EMPLOYMENT
CARDS MUST BE PLACED IN THE FOLLOWING ORDER: 2 TYPE 1 MODE 1, 2 TYPE
2 MODE 1, 1 TYPE 3 MODE 4, 1 TYPE 3 MODE 3. WHETHER OR NOT THE
PRESET GUN LOCATION CPT ICH IS SPECIFIED, 1 TYPE 5 MODE 3 WEAPCN IS
ADDED TO THE SCENARIO. THE LOCATION OF THIS GUN IS FIXED AND IS
SPECIFIED BY THE PROGRAM.

INPUT DECK EXAMPLE: (FOR 10 MILESTONES)

```

1000. -200.
1200. -300.
1525. -350.
1800. -400.
2100. -525.
2350. -655.
2500. -803.
2819. -1055.
3000. -1100.
3200. -1215.
39999. 0711111110 500.0
1000. 2000.
1200. 1600.
2575. -550.
(ETC., UNTIL ALL 6 GUN EMPLOYMENTS ARE LOCATED.)

```

*****MAIN PROGRAM*****

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```

DIMENSION X10(200), Y10(200)
DIMENSION M(200), VEL(200), XDOT(200), YDOT(200), ZDGT(200), X(200)
1) Y(200), Z(200), HDG(200), CA(200), T(200), HDGDEG(200), CADEG(2
200), RA(200), TIME(10), XGUN(15), YGUN(15), ZGUN(15), XCIRC(25), Y
3CIRC(25)
DIMENSION TNRT(200), ITNMAX(200), TNRTDG(200), TNCCR(200), ISTALL(
1200), G(200), ABSRT(200), DT(200)
DIMENSION RCSTAB(1000)
DIMENSION VATIN2(208), VAT3(208), VAT5(208)

```

C PREDETERMINED GUN EMPLOYMENT LOCATIONS

```

485
490
495
500
505
510
515
520
525
530
535
540
545
550
555
560
565
570
575
580
585
590
595
600
605
610
615
620
625
630
635
640
645
650
655
660
665
670
675
680
685
690
695
700
705
710
715
720

C      DATA XGUN/14800.,16200.,13600.,13400.,11300.,15600.,12800./
C      DATA YGUN/9000.,8200.,7200.,8000.,9700.,10900.,7500./
C      DATA ZGUN/40.,40.,20.,20.,50.,90.,20./
C      C      VULNERABLE AREA TABLE VS TYPE 1 AND 2 WEAPONS
C      DATA VAT1N2/2*.4645,6*7.107,2*.6568,6*5.551,2*.6968,6*5.574,2*.656
18*7.357,2*.6968,6*5.574,2*.4645,6*7.432,2*.6568,6*2.858,2*.4645,6*
26*7.357,2*.6568,6*2.858,2*.4645,6*7.432,2*.6568,6*5.574,2*.6568,6*
33.298,2*.6568,6*2.858,2*.4645,6*5.551,2*.6968,6*5.574,2*.6568,6*
42.58,2*.6568,6*2.858,2*.6568,6*5.551,2*.6968,6*7.107/
C      C      VULNERABLE AREA TABLE VS TYPE 3 WEAPONS
C      DATA VAT3/2*12.54,6*13.47,2*9.853,6*10.51,2*9.639,6*11.15,2*12.64,
16*14.78,2*9.639,6*11.15,2*9.853,6*10.51,2*9.639,6*11.15,2*12.64,6*
214.78,2*9.639,6*11.15,15.8*1.394,2*4.762,6*6.240,2*5.342,2*4.762,6*
3762.6*6.240,8*1.394,2*4.762,6*6.240,2*5.342,6*7.432,2*4.762,6*
40,2*9.853,6*10.51,2*9.639,6*11.15,2*12.64,6*14.78,2*5.639,6*11.15,
52*9.853,6*10.51,2*5.639,6*11.15,2*12.64,6*14.78,2*5.639,6*11.15,2*
612.54,6*13.47/
C      C      VULNERABLE AREA TABLE VS TYPE 5 WEAPONS
C      DATA VAT5/8*55.37,8*43.22,8*47.10,8*62.53,8*47.10,8*43.22,8*47.10,
18*62.53,8*47.10,8*5.761,8*27.45,8*33.07,8*27.45,8*5.761,8*27.45,8*
233.07,8*27.45,8*5.761,8*47.10,8*62.53,8*47.10,8*43.22,8*47.10,8*62
3.53,8*47.10,8*55.37/
C      C      RADAR CROSS SECTION TABLE
C      DATA RCSTAB/19*1000.,19*100.,19*100.,178.,053.,023.,156.,166.,063.,1
173.,235.,047.,02.,015.,011.,073.,072.,019.,028.,11.,454.,005.,005.,
2.C07.,011.,093.,093.,039.,068.,112.,032.,015.,015.,009.,035.,032.,0
3017.,033.,102.,505.,273.,178.,053.,023.,156.,166.,063.,173.,235.,0
447.,02.,015.,011.,073.,072.,019.,028.,11.,454.,19*1000./
C      C      CALL ERRSET SUPPRESSES ANY POSSIBLE UNDERFLOW PROBLEMS THAT MAY
C      RESULT FROM MANIPULATION OF SCENARIO PARAMETERS.
C      CALL ERRSET (208,50,-1,1,1)
C      PI = 3.14159
C      MNUM = 0
C      IBR = 0

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C C C C C C
READ THE MILESTONE CARDS, COUNT THE NUMBER OF MILESTONES AND STOP
INPUT UPON REACHING THE '99999'. DELIMITER CARC.

DO 1 I=1,200 X(I),Y(I),Z(I)
READ (5,40) X(I),Y(I),Z(I)
IF (X(I).EQ.99999.) GO TO 2
MNUM = MNUM+1
X10(I) = X(I)/2000.0
Y10(I) = Y(I)/2000.0
1 CCNTINUE

C C C C C C
2 CCNTINUE

C C C C C C
READ THE CRUISE SPEED, BOMB RELEASE MILESTONE, EW OPTION, ANTI-JAM
OPTION, MULTIPATH OPTION, GUN LOCATION INPUT OPTION AND THE LIST
PUNCH, PLOT AND EXTENDED OUTPUT OPTICNS AND THE JAMMER POWER.

C C C C C C
READ (5,42) CVEL,MBR,IEW,IAJ,IMULT,IGUN,ILST,IPNCH,IPLT,IEXT,PJAM
JAMMER ON BUT JAMMER POWER = 0 ??? THEN TURN JAMMER OFF.

C C C C C C
IF (IEW.EQ.1.AND.PJAM.LE.0.0) IPJAM1=1
IF (IPJAM1.EQ.1) IEW=0
IF (IPJAM1.EQ.1) WRITE (6,43)

C C C C C C
JAMMER POWER GREATER THAN 1000 WATTS ??? THEN LIMIT IT TO 1000 WATTS

C C C C C C
IF (PJAM.GT.1000) IPJAM2=1
IF (IPJAM2.EQ.1) PJAM=1000
IF (IPJAM2.EQ.1.AND.IEW.EQ.1) WRITE (6,44)

C C C C C C
AJ ON BUT JAMMER OFF ??? THEN TURN AJ OFF.

C C C C C C
IF (IAJ.EQ.1.AND.IEW.EQ.0) WRITE (6,45)
IF (IEW.EQ.0) IAJ = 0

C C C C C C
IPTICN TO INPUT THE 6 ADDITIONAL GUN EMPACEMENT LOCATIONS

C C C C C C
IF (IGUN.NE.1) GO TO 3

C C C C C C
DC 3 I=1,6

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3 READ (5,41,END=38) XGUN(I),YGUN(I),ZGUN(I)
  CONTINUE
  CALCULATE THE X, Y AND Z VELOCITIES, THE CLIMB ANGLE, HEADING AND
  ROLL ANGLE AND THE TIME AT EACH MILESTONE ALONG WITH VARIOUS OTHER
  PARAMETERS FOR LATER USE.

  DO 9 I=1,MNUM
  IF (I.NE.MNUM) GO TO 4
  CALCULATIONS FOR THE FINAL MILESTONE
  HDG(I) = HDG(I-1)
  HCGDEG(I) = HDG(I)*57.29578
  IF (HCGDEG(I).LT.0) HDGDEG(I)=HDGDEG(I)+360
  RA(I) = 0
  CADEG(I) = 0
  ZCOT(I) = 0
  XDOT(I) = VEL(I)*COS(HDG(I))
  YDOT(I) = VEL(I)*SIN(HDG(I))
  TNRT(I) = 0
  GO TO 8

4 GENERAL COORDINATE CALCULATIONS FOR LATER USE
  CCNTINUE
  DX = X(I+1)-X(I)
  DX2 = DX**2
  CY = Y(I+1)-Y(I)
  DY2 = DY**2
  DZ = Z(I+1)-Z(I)
  DZ2 = DZ**2
  DIST = SQRT(DX2+DY2+DZ2)

  CLIMB ANGLE CALCULATIONS
  CA(I) = ATAN2(DZ, SQRT(DX2+DY2))
  IF (CA(I).GT.1.5533) CA(I)=1.5533
  CADEG(I) = CA(I)*57.29578

  HEADING CALCULATIONS
  IF (DX.NE.0.OR.DY.NE.0) GO TO 5
  HCG(I) = HDG(I-1)

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GC TO 6
5 HDG(I+1) = ATAN2(DY,DX)
  HDG(I) = HDG(2)
6 HCGDEG(I) = HDG(I)*57.29578
  IF (HCGDEG(I).LT.0) HDGDEG(I)=HCGCEG(I)+360

VELOCITY CALCULATIONS
  VEL(I) = CVEL
  VEL(I+1) = VEL(I)-TAN(CA(I))*DIST/100+(CVEL-VEL(I))*(DIST/VEL(I))/
130

BCMB RELEASE POINT CONSIDERATIONS
  IF (I.EQ.MBR) IBR = 1
  ***** RESTRICTION: MAX VEL PRIOR TO BOMB RELEASE POINT IS 260 MPS.
  IF (VEL(I+1).GT.260.AND.IBR.EQ.0) VEL(I+1)=260
  ***** RESTRICTION: MAX VEL AFTER BCMB RELEASE PCINT IS 310 MPS.
  IF (VEL(I+1).GT.310.AND.IBR.NE.0) VEL(I+1)=310
  VAVG = (VEL(I)+VEL(I+1))/2

VELOCITY COMPONENT CALCULATIONS
  ZCOT(I) = VEL(I)*SIN(CA(I))
  XYVEL = VEL(I)*COS(CA(I))
  XDOT(I) = XYVEL*COS(HDG(I))
  YDOT(I) = XYVEL*SIN(HDG(I))

MILESTONE TIME CALCULATIONS
  T(I) = 0
  T(I+1) = T(I)+DIST/VAVG
  DT(I) = 1
  IF (I.EQ.1) GO TO 7
  DT(I) = T(I)-T(I-1)

TURN RATE AND ROLL ANGLE CALCULATIONS
7 TAANG = HDG(I+1)-HDG(I)
  TNRT(I) = TNANG/DT(I)
  RA(I) = ATAN(TNRT(I)*VAVG/9.81)*57.29578
8 CONTINUE
9 CONTINUE

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CC CC CC
CARD 2 TIME INCREMENT CALCULATION
TINC = T(MNUM)/1000
CC CC
CARD 6 TIME INCREMENT CALCULATIONS
TINKI = 0
CC CC
DO 10 I=1,9
TINK(I) = TINKI+T(MNUM)/10
TINKI = TINK(I)
10 CONTINUE
CC CC CC CC CC CC
/////*****PRINT PROGRAM*****/////
OPTION TO LIST THE POOL CARD DECK
IF (ILST.EQ.0) GO TO 16
CC CC CC CC CC
COMMENT PRINTED OUTPUT OF THE POOL CARD DECK.
THE JCL CARDS.
WRITE (6,47)
WRITE (6,48)
WRITE (6,49)
WRITE (6,50)
WRITE (6,51)
WRITE (6,52)
WRITE (6,53)
WRITE (6,54)
WRITE (6,55)
WRITE (6,56)
CC CC
BLANK CARD FOR RADAR MASK ANGLE = 0.

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WRITE (6,57)
THE OUTPUT TITLE CARD.
WRITE (6,58)
CARD 2
WRITE (6,59) T(MNUM),TINC
THE 2A CARDS (MILESTONES).
DC 11 I=1,MNUM
WRITE (6,60) T(I),X(I),Y(I),Z(I),XDOT(I),YDOT(I),ZDOT(I),PCGDEG(I)
1,CADEG(I),RA(I)
11 CONTINUE
WRITE (6,61)
WRITE (6,62)
CARD 3 (GUN EMPLACEMENT CARD).
WRITE (6,63) XGUN(1),YGUN(1),ZGUN(1)
CARD 4 (GUN TYPE).
WRITE (6,64)
CARD 5
WRITE (6,65)
CARD 6
WRITE (6,66) (TINK(I),I=1,9)
CARD 7 (VULNERABLE AREA TABLE VS TYPE 1 AND 2 WEAPONS)
WRITE (6,67)
WRITE (6,68) (VATIN2(I),I=1,208)
CARD 12 (EXECUTE RUN).
EXTENDED OUTPUT OPTION

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C C IF (IEXT.NE.1) WRITE (6,70)
C C IF (IEXT.EQ.1) WRITE (6,69)
C C THE REMAINDER OF THE CARDS INTRODUCE NEW GUN LOCATIONS, GUN TYPES
C C AND VULNERABLE AREA TABLES TO BE EXECUTED BY THE PROGRAM.
C C WRITE (6,63) XGUN(2), YGUN(2), ZGUN(2)
C C EXTENDED OUTPUT OPTION
C C IF (IEXT.NE.1) WRITE (6,70)
C C IF (IEXT.EQ.1) WRITE (6,69)
C C WRITE (6,63) XGUN(3), YGUN(3), ZGUN(3)
C C WRITE (6,71)
C C EXTENDED OUTPUT OPTION
C C IF (IEXT.NE.1) WRITE (6,70)
C C IF (IEXT.EQ.1) WRITE (6,69)
C C WRITE (6,63) XGUN(4), YGUN(4), ZGUN(4)
C C EXTENDED OUTPUT OPTION
C C IF (IEXT.NE.1) WRITE (6,70)
C C IF (IEXT.EQ.1) WRITE (6,69)
C C WRITE (6,63) XGUN(5), YGUN(5), ZGUN(5)
C C WRITE (6,72)
C C CARD 7 (VULNERABLE AREA TABLE VS TYPE 3 WEAPONS)
C C WRITE (6,147)
C C WRITE (6,68) (VAT3(I), I=1,208)
C C EW (JAMMER) OPTION
C C IF (IEW.NE.1) GO TO 12
C C IRECM = 1
C C SJTMAX = 3
C C CARC 14 (SPECIFIES EW OPTION AND JAMMER INFO.)
C C WRITE (6,73) IEW, PJAM, IRECM, SJTMAX
C C WRITE THE RADAR CROSS SECTION TABLE.
C C WRITE (6,74) (RCSTAB(I), I=1,133)
C C 12 CONTINUE

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C C C EXTENDED OUTPUT OPTION  
C C C IF (IEXT.NE.1) WRITE (6,70)  
C C C WRITE (6,69)  
C C C WRITE (6,63) XGUN(6), YGUN(6), ZGUN(6)  
C C C WRITE (6,75)  
C C C MULTIPATH OPTICN  
C C C IF (IMULT.NE.1) GO TO 13  
C C C CARD 13 (MULTIPATH EFFECTS)  
C C C IRMP = 1  
C C C WRITE (6,76) IMULT,IRMP  
C C C 13 CONTINUE  
C C C EXTENDED OUTPUT OPTION  
C C C IF (IEXT.NE.1) WRITE (6,70)  
C C C IF (IEXT.EQ.1) WRITE (6,69)  
C C C WRITE (6,63) XGUN(7), YGUN(7), ZGUN(7)  
C C C WRITE (6,77)  
C C C CARD 7 (VULNERABLE AREA TABLE VS TYPE 5 WEAPONS)  
C C C WRITE (6,148)  
C C C WRITE (6,68) (VAT5(I),I=1,208)  
C C C MULTIPATH OPTICN  
C C C IF (IMULT.NE.1) GO TO 14  
C C C CARD 13 (MULTIPATH EFFECTS)  
C C C IRMP = 2  
C C C WRITE (6,76) IMULT,IRMP  
C C C 14 CCNTINUE  
C C C EW (JAMMER) OPTION  
C C C IF (IEW.NE.1) GO TO 15  
C C C IRECM = 2
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2405 ANTI-JAM OPTION
2415 IF (IAJ.EQ.1) IRECM=3
2425 IF (IRECM.EQ.2) SJTMAX=1.5
2435 IF (IRECM.EQ.3) SJTMAX=17
2445
2450 CARD 14 (SPECIFIES EW OPTION AND JAMMER INFO.)
2460 WRITE (6,73) IEW,PJAM,IRECM,SJTMAX
2470 WRITE THE RADAR CROSS SECTION TABLE.
2480 WRITE (6,74) (RCSTAB(I),I=1,133)
2490
2500 15 CCNTINUE
2510
2520 EXTENDED OUTPUT OPTICN
2530 IF (IEXT.NE.1) WRITE (6,70)
2540 IF (IEXT.EQ.1) WRITE (6,69)
2550 WRITE (6,61)
2560 WRITE (6,78)
2570 WRITE (6,61)
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WRITE (7,86)
WRITE (7,83)
WRITE (7,87)
WRITE (7,83)
WRITE (7,88)

LEADING BLANK DATA CARD SIGNIFIES RADAR MASKING ANGLE OF ZERO.
WRITE (7,89)

THE OUTPUT TITLE CARD.
WRITE (7,90)

CARD 2
WRITE (7,91) T(MNUM),TING
THE 2A CARDS (MILESTONES).

DC 17 I=1,MNUM
WRITE (7,92) T(I),X(I),Y(I),Z(I),XCOT(I),YDCT(I),ZDOT(I),HDGDEG(I)
1 CADEG(I),RA(I)
17 CCNTINUE

WRITE (7,93)
WRITE (7,94)

CARD 3 (GUN EMPLACEMENT CARD).
WRITE (7,95) XGUN(1),YGUN(1),ZGUN(1)

CARD 4 (GUN TYPE).
WRITE (7,96)

CARD 5
WRITE (7,97)

CARD 6
WRITE (7,98) (TINK(I),I=1,9)

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C C CARD 7 (VULNERABLE AREA TABLE VS TYPE 1 AND 2 WEAPONS)
      WRITE (7,99)
      WRITE (7,100) (VATIN2(I),I=1,208)
C C
C C CARD 12 (EXECUTE RUN).
C C EXTENDED OUTPUT OPTION
C C IF (IEXI.NE.1) WRITE (7,102)
C C IF (IEXI.EQ.1) WRITE (7,101)
C C THE REMAINDER OF THE CARDS INTRODUCE NEW GUN LOCATIONS, GUN TYPES
C C AND VULNERABLE AREA TABLES TO BE EXECUTED BY THE PROGRAM.
C C WRITE (7,95) XGUN(2),YGUN(2),ZGUN(2)
C C EXTENDED OUTPUT OPTION
C C IF (IEXI.NE.1) WRITE (7,102)
C C IF (IEXI.EQ.1) WRITE (7,101)
C C WRITE (7,95) XGUN(3),YGUN(3),ZGUN(3)
C C WRITE (7,103)
C C EXTENDED OUTPUT OPTION
C C IF (IEXI.NE.1) WRITE (7,102)
C C IF (IEXI.EQ.1) WRITE (7,101)
C C WRITE (7,95) XGUN(4),YGUN(4),ZGUN(4)
C C EXTENDED OUTPUT OPTION
C C IF (IEXI.NE.1) WRITE (7,102)
C C IF (IEXI.EQ.1) WRITE (7,101)
C C WRITE (7,95) XGUN(5),YGUN(5),ZGUN(5)
C C WRITE (7,104)
C C CARD 7 (VULNERABLE AREA TABLE VS TYPE 3 WEAPONS)
C C WRITE (7,145)
C C WRITE (7,100) (VAT3(I),I=1,208)
C C
C C EW (JAMMER) OPTION
C C IF (IEW.NE.1) GO TO 18
C C IRECM = 1
C C SJTMX = 3
C

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C C CARD 14 (SPECIFIES EW OPTION AND JAMMER INFO.)
C C WRITE (7,105) IEW,PJAM,IREFM,SJTMAX
C C WRITE THE RADAR CROSS SECTION TABLE.
C C WRITE (7,106) (RCSTAB(I),I=1,133)
C C 18 CCNTINUE
C C
C C EXTENDED OUTPUT OPTION
C C IF (IEXI.NE.1) WRITE (7,102)
C C IF (IEXT.EQ.1) WRITE (7,101)
C C WRITE (7,95) XGUN(6),YGUN(6),ZGUN(6)
C C WRITE (7,107)
C C
C C MULTIPATH OPTION
C C IF (IMULT.NE.1) GC TO 19
C C IRMP = 1
C C
C C CARD 13 (MULTIPATH EFFECTS)
C C WRITE (7,108) IMULT,IRMP
C C
C C 15 CCNTINUE
C C
C C EXTENDED OUTPUT OPTION
C C IF (IEXI.NE.1) WRITE (7,102)
C C IF (IEXT.EQ.1) WRITE (7,101)
C C WRITE (7,95) XGUN(7),YGUN(7),ZGUN(7)
C C WRITE (7,109)
C C
C C CARD 7 (VULNERABLE AREA TABLE VS TYPE 5 WEAPON)
C C WRITE (7,150)
C C WRITE (7,150) (VAT5(I),I=1,208)
C C
C C MULTIPATH OPTION
C C IF (IMULT.NE.1) GC TO 20
C C IRMP = 2
C C CARD 13 (MULTIPATH EFFECTS)

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C      WRITE (7,108) IMULT,IRMP
C      20 CCNTINUE
C      EW (JAMMER) OPTION
C      IF (IEW.NE.1) GO TC 21
C      IRECM = 2
C      ANTI-JAM OPTION
C      IF (IAJ.EQ.1) IRECM=3
C      IF (IRECM.EQ.2) SJTMAX=1.5
C      IF (IRECM.EQ.3) SJTMAX=17
C      CARC 14 (SPECIFIES EW OPTION AND JAMMER INFO.)
C      WRITE (7,105) IEW,PJAY,IRECM,SJTMAX
C      WRITE THE RADAR CROSS SECTION TABLE.
C      WRITE (7,106) (RCSTAB(I),I=1,133)
C      21 CONTINUE
C      EXTENDED OUTPUT OPTION
C      IF (IEXT.NE.1) WRITE (7,102)
C      IF (IEXT.EQ.1) WRITE (7,101)
C      WRITE (7,93)
C      WRITE (7,110)
C      WRITE (7,93)
C      ////*****PLOT SECTION*****////
C      OPTICN TC PLOT THE POOL SCENARIO
C      22 CCNTINUE
C      INITIALIZE PLOT
C      CALL PLOTS
C      IF (IPLOT.EQ.0) GO TO 25
C      ESTABLISH X AXIS
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C      REAL XTITLE/'
C      CALL AXIS (C.O,O.O,XTITLE,-1,9.O,C.O,O.O,2000.O)
C
C      ESTABLISH Y AXIS
C
C      REAL YTITLE/'
C      CALL AXIS (O.O,O.O,YTITLE,O.6.O,90.O,O.O,2000.O)
C
C      COMPLETE OUTLINE
C
C      REAL LX1(3)/O.O,9.O,9.O/
C      REAL LY1(3)/6.O,6.O,0.O/
C      CALL LINE (LX1,LY1,3,1,1)
C
C      PLOT THE ROAD
C
C      REAL RDX1(26)/O.O,O.O,0.5,1.0,1.5,2.0,2.5,3.0,3.5,4.0,4.5,5.0,5.5,5.75
C      1,6.O,6.5,6.86,7.0,7.2,7.3,7.4,7.5,7.6,7.7,7.8,C,8.5,9.C/
C      REAL RDX1(26)/2.75,2.82,2.88,2.92,2.90,2.85,2.80,2.78,2.79,2.
C      183,2.9,3.0,3.14,3.5,3.8,3.9,4.06,4.1,4.15,4.2,4.18,4.13,3.57,3.03,
C      22.48/
C      CALL LINE (RDX1,RDY1,26,1,1)
C      REAL RDX2(12)/O.O,0.5,1.0,1.5,2.0,2.5,3.0,3.5,4.0,4.5,5.0,5.3,5.75,
C      REAL RDX2(12)/2.55,2.65,2.72,2.76,2.73,2.7,2.6,2.5,2.25,2.0,1
C      1.75/
C      CALL LINE (RDX2,RCY2,12,1,1)
C      REAL RDX3(14)/4.O,3.75,3.45,3.25,3.5,4.O,4.5,5.O,5.3,5.6,5.45,5.3,
C      1,5.O/
C      REAL RDX3(14)/1.8,2.2,2.5,2.65,2.62,2.6,2.61,2.65,2.7,2.76,2.6,2.4
C      1,1.9,1.5/
C      CALL LINE (RDX3,RDY3,14,1,1)
C      REAL RDX4(19)/5.18,5.25,5.4,5.5,5.6,5.75,6.O,6.2,6.5,6.7,6.85,7.23
C      1,7.35,7.5,7.6,7.7,7.95,8.35,9.O/
C      REAL RDX4(19)/1.5,1.8,2.2,2.35,2.5,2.7,2.95,3.1,3.3,3.5,3.6,3.9,4.
C      10.4,04,4.0,3.9,3.5,3.0,2.25/
C      CALL LINE (RDX4,RDY4,19,1,1)
C
C      PLOT THE RIVER
C
C      REAL RVX1(23)/O.O,0.5,1.0,1.35,1.5,2.0,2.3,2.5,3.0,3.5,4.O,4.5,5.O
C      1,5.5,5.9,6.O,6.45,6.6,6.75,6.95,7.05,7.2,7.2/
C      REAL RVX1(23)/5.3,5.55,5.0,4.5,4.38,4.1,4.0,3.95,3.5,3.92,3.9,3.7
C      1,3.6,3.75,4.0,4.1,4.5,4.6,4.65,4.6,4.5,4.3,4.06/
C      CALL LINE (RVX1,RVY1,23,1,1)
C      REAL RVX2(27)/O.O,0.5,0.6,0.9,1.0,1.15,1.5,1.85,2.0,2.15,2.5,3.0,3
C      1,2,4.0,4.2,4.5,5.0,5.5,5.8,6.0,6.1,6.35,6.5,6.7,6.85,6.9,6.86/
C      REAL RVX2(27)/5.4,5.1,5.0,4.5,4.3,4.0,3.7,3.5,3.48,3.5,3.6,3.65,3.

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165,3.6,3.5,3.35,3.3,3.15,3.3,3.15,3.3,3.5,4.0,4.15,4.25,4.2,4.0,3.8/
CALL LINE (RVX2,RVY2,27,1,1)
REAL RVX3(11)/6.85,6.9,6.85,7.0,7.35,7.5,7.7,8.0,8.4,8.8,9.0/
CALL RVY3(11)/3.6,3.35,3.0,2.6,2.3,2.25,2.2,1.65,1.3,1.0,0.8/
CALL LINE (RVX3,RVY3,11,1,1)
REAL RVX4(7)/7.23,7.3,7.3,7.5,8.0,8.5,9.0/
CALL RVY4(7)/3.9,3.7,3.35,3.0,2.6,2.1,1.7/
CALL LINE (RVX4,RVY4,7,1,1)

CC PLOT THE BIG ISLAND
CC
REAL BIX1(12)/7.84,8.0,8.14,8.31,8.4,8.38,8.2,8.0,7.85,7.83,7.85,7
1.84/
1/ REAL BIY1(12)/2.3,2.22,2.18,1.8,1.7,1.63,1.68,1.82,2.0,2.1,2.2,2.3
1/ CALL LINE (BIX1,BIY1,12,1,1)

CC PLOT THE XY AXIS ARROWS
CC
REAL XXY1(9)/0.4,0.35,0.45,0.4,0.4,1.0,0.8,C,8,1.0/
REAL XYY1(9)/1.0,0.8,0.8,1.0,0.4,0.4,0.45,0.35,0.4/
CALL LINE (XXY1,XYY1,9,1,1)
REAL XCHAR(1)/'X'/
CALL SYMBOL (0.8,0.1,0.21,XCHAR,0.0,1)
REAL YCHAR(1)/'Y'/
CALL SYMBOL (0.1,0.8,0.21,YCHAR,0.0,1)

CC PLOT THE TRUE NORTH ARROW
CC
REAL TNX1(5)/2.0,2.0,1.95,2.05,2.0/
REAL TNY1(5)/4.8,5.4,5.2,5.2,5.4/
CALL LINE (TNX1,TNY1,5,1,1)
CALL TNCHAR(1)/'N'/
CALL SYMBOL (1.9,5.5,0.21,TNCHAR,0.0,1)

CC WRITE THE PLOT TITLE
CC
REAL PTITLE(6)/' AE ','3251',' POC','1 SC','ENAR','IO '/
CALL SYMBOL (2.3,6.5,0.21,PTITLE,0.0,24)

CC PLOT THE MILESTONES AND FLIGHT PATH
CC
CALL LINE (X10,Y10,MNUM,1,7)

CC PLOT THE BOMB RELEASE POINT
CC
RX = X10(MBR)
RY = Y10(MBR)

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CALL LINE (RX,RY,1,1,6)
PLOT THE GUN LCCATIONS AND RANGE RADIUS

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CC
CC

```

DC 24 I=1,7
TIMIX = XGUN(I)/2000.0
TIMIY = YGUN(I)/2000.0
IF (I.GE.1) MKR = 3
IF (I.GE.5) MKR = 4
IF (I.GE.7) MKR = 5
IF (I.GE.1) MKR = 2
IF (I.GE.3) RAD = 0.78
IF (I.GE.5) RAD = 0.91
IF (I.GE.7) RAD = 1.42
CALL LINE (TIMIX,TIMIY,1,1,MKR)

```

CC
CC

```

DO 23 J=1,25
XCIRC(J) = TIMIX+RAD*COS(J*PI/12)
YCIRC(J) = TIMIY+RAD*SIN(J*PI/12)
IF (XCIRC(J).LT.0) XCIRC(J)=0
IF (YCIRC(J).LT.0) YCIRC(J)=0
IF (XCIRC(J).GT.9) XCIRC(J)=9
IF (YCIRC(J).GT.6) YCIRC(J)=6
CCNTINUE

```

23

CC
CC

CALL LINE (XCIRC,YCIRC,25,1,MKR)
CCNTINUE

24

CC
CC

CALL PLOT (0.0,12.0,-3)
CALL PLOTE
CONTINUE

25

CC
CC
CC

/////***** ERROR CHECK SECTION *****/////

CPTICN SUMMARY

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WRITE (6,111) MNUM,T(MNUM),MBR
WRITE (6,112)
WRITE (6,113) CVEL

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IF (ILST.EQ.0) WRITE (6,114)
IF (IPNCH.EQ.0) WRITE (6,115)
IF (IPNCH.EQ.1) WRITE (6,116)
IF (IPLOT.EC.0) WRITE (6,117)
IF (IPLOT.EC.1) WRITE (6,118)
IF (ITEXT.NE.1) WRITE (6,119)
IF (ITEXT.EQ.1) WRITE (6,145)
IF (IGUN.EQ.0) WRITE (6,146)
IF (IGUN.EQ.1) WRITE (6,120)
IF (IEM.EQ.0) WRITE (6,121)
IF (IEM.EQ.1) WRITE (6,122)
IF (IAJ.EQ.0) WRITE (6,123) PJAM
IF (IAJ.EQ.1) WRITE (6,124)
IF (IMULT.EC.1) WRITE (6,125)
IF (IMULT.EC.0) WRITE (6,126)
IF (IMULT.EC.1) WRITE (6,127)
WRITE (6,129)

C IERR = 0
C POPALT = 0
C ***** RESTRICTION: NO TYPE 3 WEAPON MAY BE LCCATED WITHIN 3000
C METERS OF THE CENTER OF THE BRIDGE.
C
C XTGT = 1410C
C YGT = 7900
C GDIST = SQRT((XTGT-XGUN(5))**2+(YGT-YGUN(5))**2)
C IF (GDIST.LT.3000) WRITE (6,130) XGUN(5),YGUN(5),GDIST
C GDIST = SQRT((XTGT-XGUN(6))**2+(YGT-YGUN(6))**2)
C IF (GDIST.LT.3000) WRITE (6,130) XGUN(6),YGUN(6),GDIST
C
C ***** RESTRICTION: INITIAL CRUISE VELOCITY MUST BE BETWEEN 206
C AND 257 METERS PER SECOND.
C IF (CVEL.LT.206.OR.CVEL.GT.257) WRITE (6,131) CVEL
C
C DO 35 I=1,MNUM
C CALCULATE DISTANCE TC TARGET
C CISTGT = SQRT((XTGT-X(I))**2+(YGT-Y(I))**2)
C ***** RESTRICTION: ALT MAX PRIOR TC POP UP IS 457 METERS.
C

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```

IZMAX = 0
IF (DISTGT.GT.6000.AND.Z(I).GT.457) IZMAX=1
IF (IZMAX.NE.1) GO TO 26
IERR = 1
WRITE (6,132) I,Z(I)

```

C
C

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**** RESTRICTION: MIN ALT IS 61 METERS.

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26 IZMIN = 0
IF (Z(I).LT.61) IZMIN=1
IF (IZMIN.NE.1) GO TO 27
IERR = 1
WRITE (6,133) I,Z(I)

```

C
C

```

**** RESTRICTION: MAX ALT OVERALL IS 2134 METERS.

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27 IZZMAX = 0
IF (Z(I).GT.2134) IZZMAX=1
IF (IZZMAX.NE.1) GO TO 28
IERR = 1
WRITE (6,134) I,Z(I)

```

C
C

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**** RESTRICTION: MIN POP UP ALT IS 1219 METERS.

```

```

28 CCNTINUE
IF (DISTGT.GT.6000) GO TO 29
IF (Z(I).GT.POPALT) POPALT=Z(I)
IPOP = 0
IF (I.EQ.MBR.AND.POPALT.LT.1219) IPOP=1
IF (IPOP.NE.1) GO TO 29
IERR = 1
WRITE (6,135) POPALT

```

C
C
C

```

**** RESTRICTION: ACFT HEADING MUST BE WITHIN 5 DEGREES OF THE
CENTER OF THE BRIDGE DURING THE LEG PRIOR TO BCMB RELEASE POINT.

```

```

29 CCNTINUE
IF (I.NE.MBR) GO TO 33
DX = XTGT-X(I)
DY = YTGT-Y(I)
TGTHDG = ATAN2(DY,DX)*57.29578
IF (TGTHDG.LT.0) TGTHCG=TGTHDG+360
HDGLMT = ABS(HDGDEG(I)-TGTHDG)
IAIM = 0
IF (HDGLMT.GT.5) IAIM=1
IF (IAIM.NE.1) GO TO 30
IERR = 1
WRITE (6,136) HDGCEG(I),TGTHCG,HDGLMT

```



```

55 FCRMAT (1X,10(FT05F001 DD UNIT=SYSDA,SPACE=(CYL,(1,1)),))
56 FCRMAT (1X,10(FT05F001 DD *))
57 FCRMAT (1X,01 AIRCRAFT COMBAT SURVIVABILITY SCENARIO)
58 FCRMAT (1X,02,4X,12,8X,F7.2,1X,F7.4)
59 FCRMAT (1X,10(F7.1,1X))
60 FCRMAT (1X,10(FT05F002 DD *))
61 FCRMAT (1X,03,6X,13(1X,F7.3))
62 FCRMAT (1X,04,11(11))
63 FCRMAT (1X,05,5X,11,0)
64 FCRMAT (1X,06,3X,9(F7.3,1X))
65 FCRMAT (1X,07,8F8.3)
66 FCRMAT (1X,12,11(1111))
67 FCRMAT (1X,12,21(1111))
68 FCRMAT (1X,04,34(411))
69 FCRMAT (1X,04,13,2X,5 0 1.0,7X,F6.1,4X,1.0E-06,6X,1,1X,
70 FCRMAT (1X,15,F10.2,1X,GEND,1X,3X,19,7 90.0 180.0,
71 FCRMAT (1X,15,F10.2,1X,GEND,1X,3X,19,7 90.0 180.0,
72 FCRMAT (1X,15,F10.2,1X,GEND,1X,3X,19,7 90.0 180.0,
73 FCRMAT (1X,15,F10.2,1X,GEND,1X,3X,19,7 90.0 180.0,
74 FCRMAT (1X,8F10.3,1X,8F10.3,1X,3F10.3)
75 FCRMAT (1X,04,33(411))
76 FCRMAT (1X,13,15,0.35)
77 FCRMAT (1X,04,53(211))
78 FCRMAT (1X,04,53(211))
79 FCRMAT (1X,04,53(211))
80 FCRMAT (1X,04,53(211))
81 FCRMAT (1X,04,53(211))
82 FCRMAT (1X,04,53(211))
83 FCRMAT (1X,04,53(211))
84 FCRMAT (1X,04,53(211))
85 FCRMAT (1X,04,53(211))
86 FCRMAT (1X,04,53(211))
87 FCRMAT (1X,04,53(211))
88 FCRMAT (1X,04,53(211))
89 FCRMAT (1X,04,53(211))
90 FCRMAT (1X,04,53(211))
91 FCRMAT (1X,04,53(211))
92 FCRMAT (1X,04,53(211))
93 FCRMAT (1X,04,53(211))
94 FCRMAT (1X,04,53(211))
95 FCRMAT (1X,04,53(211))
96 FCRMAT (1X,04,53(211))
97 FCRMAT (1X,04,53(211))
98 FCRMAT (1X,04,53(211))

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99 FCRMAT ('07 VULNERABLE AREA TABLE VS TYPE 1 AND 2 WEAPONS')
100 FCRMAT ('8F8.3)
101 FCRMAT ('1211111111')
102 FCRMAT ('04 211111')
103 FCRMAT ('04 34411')
104 FCRMAT ('14.13.5 0 1.0.7X,F6.1 ,4X,1.0E-06',6X,1.1,11X,
105 FCRMAT ('1.0.1300.0,15,F10.2 ,/GEND,/,3X,19 ,X,7 90.0 180.0',
106 FCRMAT ('8F10.3,/,8F10.3,/,3F10.3)
107 FCRMAT ('04 33411')
108 FCRMAT ('13.13,15, G.35')
109 FCRMAT ('04 53211')
110 FCRMAT ('/GO.FT05F003 DD *')
111 FCRMAT ('1. THE FLIGHT TIME IS AT MILESTONE',F6.1, SECONDS,/,1X,
112 FCRMAT ('1. BOMB RELEASE IS OPTIMUM SUMMARY',F6.1, METERS PER SECOND,))
113 FCRMAT ('1. CRUISE SPEED IS',F6.1, METERS PER SECOND,))
114 FCRMAT ('1. A POOL INPUT LISTING IS PROVIDED AS OUTPUT,))
115 FCRMAT ('1. A PUNCHED DECK IS PROVIDED AS OUTPUT,))
116 FCRMAT ('1. A PUNCHED DECK IS PROVIDED AS OUTPUT,))
117 FCRMAT ('1. A SCENARIO PLOT IS PROVIDED AS OUTPUT,))
118 FCRMAT ('1. A SCENARIO PLOT IS PROVIDED AS OUTPUT,))
119 FCRMAT ('1. PRESET GUN EMPLOYMENT LOCATIONS ARE UTILIZED,))
120 FCRMAT ('1. GUN EMPLOYMENT LOCATIONS ARE BEING UTILIZED,))
121 FCRMAT ('1. AN AIRBORNE JAMMER IS BEING UTILIZED,))
122 FCRMAT ('1. JAMMER POWER IS',F6.1, WATTS,))
123 FCRMAT ('1. ANTI-JAM FEATURE IS NOT SELECTED,))
124 FCRMAT ('1. MULTI-PATH RADAR EFFECTS ARE NOT CONSIDERED,))
125 FCRMAT ('1. MULTI-PATH FLIGHT PATTERN EFFECTORS',F7.1, IS,
126 FCRMAT ('1. GUN TYPE THE LOCATED AT',F7.1, IS,
127 FCRMAT ('1. METERS FROM OF 3000 METERS,/,1X, WHICH IS LESS THAN THE',
128 FCRMAT ('1. MINIMUM DISTANCE GIVEN',F6.1, METERS PER SECOND WHICH IS,
129 FCRMAT ('1. METERS WITHIN THE SECOND, AT MILESTONE',13, IS, METERS WHICH
130 FCRMAT ('1. METERS ABOVE THE MAX ALT PRIOR TO POP UP OF 457 METERS,))
131 FCRMAT ('1. METERS BELOW THE MIN ALT OF 61 METERS,/,1X, IS, METERS WHICH
132 FCRMAT ('1. METERS ABOVE THE MAX ALT OF 2134 METERS,/,1X, IS, METERS WHICH
133 FCRMAT ('1. METERS BELOW THE MIN ALT OF 2134 METERS,/,1X, IS, METERS WHICH
134 FCRMAT ('1. METERS ABOVE THE MAX ALT OF 2134 METERS,/,1X, IS, METERS WHICH

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135 FORMAT (1X, 'MAX ALTITUDE DURING FCP UP WAS ', F6.1, ' METERS WHICH',
1 ' OF 1219 METERS. ', //)
136 FORMAT (1X, 'THE AIRCRAFT HEADING INTO THE BCMB RELEASE PCINT IS ',
1 ' F5.1, ' WHICH IS GREATER THAN THE ',
2 ' THE DIFFERENCE IS ', F5.1, //)
137 FORMAT (1X, 'THE HEADING MAXIMUM DIFFERENCE LIMIT, ', //)
1 ' RELEASE PCINT IS ', F4.2, ' IMMEDIATELY PRIOR TO THE BCMB ',
2 ' WHICH IS LESS THAN THE MINIMUM OF 2 SECONDS. ', //)
138 FORMAT (1X, 'THE BCMB RELEASE ALTITUDE IS ', F7.1, ' METERS WHICH IS:
1 ' NOT IN METERS. ', //)
139 FORMAT (1X, 'THE BCMB WAS RELEASED AT A DISTANCE OF ', F7.1,
1 ' METERS FROM THE TARGET WHICH IS IN EXCESS ', //)
140 FORMAT (1X, 'MILES STALL OCCURS AT 90 METERS PER SECOND. DECREASE ', //)
1 ' SECOND, MB ANGLE PRIOR TO MILESTONE ', I3, ' METERS PER ', //)
141 ' THE CLIX, MILES PER SECOND WHICH RESULTS IN A TURN RATE OF ', I1X, F
2 ' F5.1, ' WHICH IS IN EXCESS OF ', F7.1, ' G MAX LOADING ', //)
3 ' DECREASES ', I3, ' TC BELOW ', F5.
4 ' DEGREES. ', //)
142 FORMAT (1X, '***** NO FLIGHT PATH ERRORS IN THIS RUN *****', //)
143 FORMAT (1X, '***** NO FURTHER FLIGHT PATH ERRORS *****', //)
144 FORMAT (1X, '***** END OF POOL INPUT PROGRAM - SUMMARY COMPLETE',
1 ' *****', //)
145 FORMAT (1X, 'THE EXTENDED OUTPUT OPTION IS NOT SPECIFIED. ')
146 FORMAT (1X, 'EXTENDED PRINTOUT IS PROVIDED AS OUTPUT. ')
147 FORMAT (1X, '07 VULNERABLE AREA TABLE VS TYPE 3 WEAPONS')
148 FORMAT (1X, '07 VULNERABLE AREA TABLE VS TYPE 5 WEAPONS')
149 FORMAT (1X, '07 VULNERABLE AREA TABLE VS TYPE 3 WEAPONS')
150 FORMAT (1X, '07 VULNERABLE AREA TABLE VS TYPE 5 WEAPONS')
END

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APPENDIX I

P001 PROGRAM LISTING (IBM)

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CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
AAASIM --- MULTIPLE GUN ANTI-AIRCRAFT ARTILLERY SIMULATION
WRITTEN BY THOMAS D. MCMURCHIE AND JAMES G. SEVERSON
AIRFORCE ARMAMENT LABORATORY (AFATL-DLYS) EGLIN AFB, FLORIDA
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
PROGRAM RECEIVED 25 OCT 72
CONVERTED AND MAINTAINED BY ASD/XROA
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
MODIFIED FOR USE ON IBM 360/370 AT NPS MONTEREY, CA., 7 MAR 78.
FOR INFORMATION CALL PROF R.E. BALL 408-646-2885, (AUTOVON 878).
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
COMMON/BLOCK1/ITITLE(20)
COMMON/BLOCK2/NEPA,TMIN,TMAX,DTFPA
COMMON/BLOCK3/XGUN,YGUN,ZGUN
COMMON/BLOCK4/IGT,IFEM,ICB,ISB,IGL,CIRCLE
COMMON/BLOCK5/NRHOS,RHO(9)
COMMON/BLOCK6/INTINTS,INTINTER(10),VA(9,5,9)
COMMON/BLOCK7/IVACOM(20),TRACK1,TRACK2
COMMON/BLOCK8/TROUND(6),THDMAX(6),PHDMAX(6),PHIMIN(6),PHIMAX(6),
1 VELMIN(6),VELMAX(6),RANMAX(6),RANMIN(6),RANMAX(6),
2 ATLAG(6),ETHMAX(6),EPHMAX(6),RMODES(6),
COMMON/BLOCK9/TFMAX1(6),TFMAX2(6),RVACON(6),RVBCON(6),VMUZEL(6)
COMMON/CONSTS/DEGREE,RADIAN,PI,PI2,QTRPI,SQRT2
COMMON/HEADFO/LINE,NUMBER
COMMON/NFPARM/XA,YA,ZA,RA2,RA,TIME
COMMON/IGXGYG/IG,XG(8),YG(8)
COMMON/MAGIC/FRACT,INDEX1,INDEX2
COMMON/VASBS/VMUZZ,ASHCON,BSHCON,CQUAD
COMMON/BUDGET/BDACON(6)
COMMON/CECMI/IRECM,IJ,GAINJ,IX,XSEC,CALX,PJW,
* * *
COMMON/FIGT,FJAM,GJ,SJT,SN
COMMON/NEWA/Y1,VY1,VY,VX,BETA,IRMP,PLEN,O1,O8,D1,JJJ,XR,ZT,PZI,IERR
COMMON/NEWB/Y1,P2,VX,B2,ALFA,J,IF6,IMUL,IP,N,O7,D,TRACK,KMODE,PSI,K
COMMON/NEWC/CP,X2,PI,B2,FILE,JFILE,IJAM,ISL,O4,CPS,TIPERS,JMODE,YI,T2
COMMON/NEWD/T33,X1,Z2,B1,VX2,C3,CPK,NROUND,IOFF,I,X,T,IM,SP,T1
COMMON/NEWE/Z1,A2,VX1,VZ2,PHI,IF5,IEOF,YR,IF2,T13,IFILE,Y2,A1
COMMON/NEWF/V2,VZ1,V,FUZZ,IF9,SD2RJ,SD2RJM,V1,VY2,F,J,P,IF7,REFC
COMMON/NEWG/SJTMAX,O2,PK,D2,ISW
COMMON/XFPA(1201),YFPA(1201),ZFPA(1201),BFPA(1201),AFPA(1201),

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1  VFPA(1201), VXFPA(1201), VYFPA(1201), VZFPA(1201)
   COMMON ICARD(20), PTOTTF(10), PTOTI(10)
   COMMON SPKTC(32,8), IPRINT(6), IFLAGS(4)
   COMMON PKTTDC(9), PKTFDC(10,9)
   COMMON INUNIT
   DIMENSION TEMP(16,6), SPKT(8,4,8), SPKT2(32,8)
   EQUIVALENCE (SPKT(1,1), SPKT(1,1,1), TEMP(1,1))

```

SOME EQUIVALENCES REMOVED BY B.E.E. BY CARRYING TWO VARIABLES THROUGH PROGRAM WITH SAME VALUES FOR CLARITY.

```

DATA ESPVCT/0.10/
DATA SD2J/0./
DATA IL LOOP/-1/
INUNIT = 5
IFLAGS(1) = 0
IFLAGS(2) = 0
IFLAGS(3) = 0
IFLAGS(4) = 0
CALL REREAD (208, 256, -1, 1, 1, 207)
ASSIGN 9996 TO IERR

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CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C  IIIII  N  N  IIIII  TTTT  IIIII  A  A  L  IIIII  ZZZZ  EEEEE  C
C  I  I  NN  N  N  I  I  T  I  I  A  A  L  I  I  ZZZZ  EEE  C
C  I  I  N  N  N  I  I  T  T  I  A  A  L  I  I  ZZZZ  EEE  C
C  IIIII  N  N  IIIII  T  IIIII  A  A  L  IIIII  ZZZZ  EEEEE  C
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
READ(INUNIT,1050) ISW,AMASK
AMASK1=AMASK*57.3
PRINT 38,AMASK1
FORMAT(5X,25HMASK ANGLE FOR THIS RUN =,F6.3,5H DEG.)
1050 95  FORMAT(A2,8X,F10.5)
      ISL=1
      IMUL = 0
      IJAM = 0
      SD2RJ = 0.
      DO 96 J=1,9
      PKTTDC(J)=0.0
      DO 96 I=1,10
      PKTIDC(I,J)=0.0
      PKTFDC(I,J)=0.0
      DO 94 I=1,32
      DO 94 J=1,8

```



```

RMIN =RANMIN(IGT)
RMAX =RANMAX(IGT)
VMIN =VELMIN(IGT)
VMAX =VELMAX(IGT)
IG=1
LINE=66
IF (IPRINT(6).LE.0) GO TO 60
CALL PAGES(5,0,JP)
WRITE (6,1013) ISL,IGT,IEM,XGUN,YGUN,ZGUN,CIRCLE
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
TTTTT RRRR R A A A C C C K K K
T R RRRR A A C C C K K K
T R R R A A A A C C C K K K
T R R R A A A A C C C K K K
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
COMPUTE ACTUAL AIRCRAFT PARAMETERS AT FIRE TIME FOR USE IN ERROR EQS
60 CALL INTERP(TIME/DTFPA)
X=GETVAL(XFFPA)-XGUN
Y=GETVAL(YFFPA)-YGUN
Z=GETVAL(ZFFPA)-ZGUN
VX=GETVAL(VXFFPA)
VY=GETVAL(VYFFPA)
VZ=GETVAL(VZFFPA)
ILOOP = ILOOP+1
RCL = GETVAL(PFFPA)
PIT = GETVAL(AFFPA)
HDG = GETVAL(BFFPA)
G2=X*X+Y*Y
G=SQRT(G2)
TFET=ATAN2(Y,X)
PHIT=ATAN2(Z,G)
IF MULTIPATH HAS BEEN SPECIFIED (IMUL=1) AND THE ELEVATION
ANGLE IS MEASURED BY RADAR (IDEM=3), COMPUTE THE ELEVATION
TRACKING BIAS (PBMP), VARIANCE (SP2MP), AND APPARENT
ALTITUDE (Z).
SP2MP = 0.
PBMP = J.
IF (IMUL.NE.1 .OR. IDEM.NE.3) GO TO 30
CALL MULTPTH(IRMP,REFC,PHIT,PBMP,SP2MP)
PHIT = PHIT+PBMP
Z = G*TAN(PHIT)

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```

30 CONTINUE (G2+Z*Z)
   RD = (X*VX+Y*VY+Z*VZ)/R
   TD = (X*VY-Y*VX)/G2
   PD = (VZ-Z*RD/R)/G
   IF (TIME.GT.TMIN) GO TO 58
   TLD = 0.0
   PCD = 0.0
   RANS = K
   THES = THET
   PHIS = PHIT
   ERAN2 = 0.0
   ERAN3 = 0.0
   ERAN4 = 0.0
   ETHE2 = 0.0
   ETHE3 = 0.0
   ETHE4 = 0.0
   EPHI2 = 0.0
   EPHI3 = 0.0
   EPHI4 = 0.0
   GO TO 59
58 RCD = (RD-RDS)/.064
   TDD = (TD-TDS)/.064
   PDD = (PD-PDS)/.064
   (STORE PREVIOUSLY OBSERVED MEAN TRACKING ERRORS FOR USE IN MEAN
   TRACKING ERROR EQUATIONS)
59 ERAN1 = ERAN2
   ERAN2 = ERAN3
   ERAN3 = ERAN4
   ERAN4 = R-RANS
   ETHE1 = ETHE2
   ETHE2 = ETHE3
   ETHE3 = ETHE4
   ETHE4 = (G/R)*ANGLIM(THET-THES)
   ETHE1 = ETHE1+.71875*(ETHE2-ETHE1)
   EPHI1 = EPHI2
   EPHI2 = EPHI3
   EPHI3 = EPHI4
   EPHI4 = EPHI1+.71875*(EPHI2-EPHI1)
   CHECK MASK ANGLE
   IF (PHIT.LE.AMASK) TFIRE = TIME+ TREAT+TTRACK
   (SKIP FIRE ATTEMPT IF INSUFFICIENT TRACKING TC FIRE)

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C
C
C

C
C
C

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C      IF(TIME-LE.TFIRE)GG TO 62
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C      FFFF  IIIII  RRRR  R  EEEEE
C      F      I      R      E
C      FFFF  I      RRRR  R  EEEEE
C      F      I      R      E
C      F      IIIII  R  EEEEE
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C      VG=SQR(VX*VX+VY*VY)
C      V =SQR(VG*VG+VZ*VZ)
C      PSI=ANGLIM(PFPA(INDEX1))+FRACT*ANGLIM(PFPA(INDEX2)-PFPA(INDEX1))
C      (SKIP FIRE ATTEMPT IF MAX ALLOWED TRACKING ERROR IS EXCEEDED)
C
C      54  IF(ABS(ETHE4).GT.ETMAX)GO TO 64
C          IF(ABS(EPHI4).GT.EPMAX)GO TO 64
C          IF(IDEEM.GT.1)GO TO 56
C      (LIMIT INPUT RANGE ESTIMATE)
C      RC=AMAX1(RMIN,AMINI(RMAX,RANS-0.575*RD))
C      COMPUTE MEAN ASSUMED TIME OF FIRE AIRCRAFT POSITION (MECHANICAL
C      COMPUTATION)
C      XF=RC*CTBCPB-XG(IG)
C      YF=RC*STBCPB-YG(IG)
C      ZF=RC*SPB
C      GF=SQR(XF*XF+YF*YF)
C      RF=SQR(GF*GF+ZF*ZF)
C      (SET UP MATRIX T, THE TRANSFORMATION BETWEEN THE LINE OF SIGHT
C      SYSTEM AND THE FALSE HORIZON SYSTEM)
C      T22 = X/G
C      CT  = X/G
C      ST  = Y/G
C      T33 = G/R
C      CP  = G/R
C      T13 = Z/R
C      SP  = Z/R
C      CS  = COS(PSI)
C      SS  = SIN(PSI)
C      CA  = VG/V
C      SA  = VZ/V

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CG=VX/VG
SG=VY/VG
T11=CT*CP
T12=ST*CP
T21=-ST
T31=-CT*SP
T32=-ST*SP
  (SET UP FALSE HORIZON SYSTEM VELOCITY COMPONENTS)
VXP=T11*VX+T12*VY+T13*VZ
VYP=T21*VX+T22*VY
VZP=T31*VX+T32*VY+T33*VZ
VGP=SQR T((VXP*VXP+VYP*VYP)
CAP=VGP/V
SAP=VZP/V
SBP=VYP/VGP
  (SET UP UNIT VECTOR OUT LEFT WING OF AIRCRAFT)
UX=-SA*CG*SS-SG*CS
UY= CG*CS-SA*SG*SS
UZ= CA*SS
UZP=T31*UX+T32*UY+T33*UZ
IF(VXP)31,32,31
CSP=-T11*UX-T12*UY-T13*UZ
GO TO 33
32 CSP=(VGP*(T21*UX+T22*UY)+UZP*VZP*SBP)/VXP
33 SSP=UZP/CAP
  (COMPUTE FALSE HORIZON SYSTEM MEAN AND STANDARD DEVIATION OF
  ERROR IN DIVE AND COURSE ANGLE ESTIMATES)
EMAP=SAP*(.3196*ABS(CBP)-.1859*ABS(SBP))
ESAP=.04712+.08063*ABS(SAP)*(1.0+1.16*ABS(CBP))
EMBP=.4060*CAP*SBP*CBP
ESBP=(.1670-.08098*ABS((CBP*CBP-SBP*SBP)*CSP)+
1 SEMAP=SIN(EMAP)
CEMAP=COS(EMAP)
SEMBP=SIN(EMBP)
CEMBP=COS(EMBP)
THE NEXT CARD ELIMINATES AN EQUIVALENCE BY USING A DOUBLE
REPLACEMENT.
A33 = SAP*CEMAP + CAP*SEMAP
SABP= SAP*CEMAP + CAP*SEMAP
CABP=CAP*CEMAP-SAP*SEMAP

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C C

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C C C

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CC      SBPP=SBP*CEMBP+CBP*SEMBP
        CBBP=CBP*CEMBP-SBP*SEMB
        (LIMIT VELOCITY ASSESSMENT (MECHANICAL COMPUTERS))
CC      VBP=AMINI(VMAX,AMAXI(VMIN,V))
        ESVP=ESVPC*V
        SET UP THE ELEMENTS OF THE MATRIX A.
        A CONTAINS THE PARTIALS OF VXE,VYE,VZE W.R.T. ALPHA,BETA,SPEED
CC      A31=CABP*CBBP
        A32=CABP*SBBP
        A21=-VBP*A32
        A22= VBP*A31
        A11=-VBP*SABP*CBBP
        A12=-VBP*SABP*SBBP
        A13= VBP*CABP
CC      COMPUTE MEAN ESTIMATED VELOCITY COMPONENTS (MECHANICAL
        COMPUTATION)
        VXE=(A31*T11+A32*T21+A33*T31)*VBP
        VYE=(A31*T12+A32*T22+A33*T32)*VBP
        VZE=(A31*T13
        GO TO 63
CC      COMPUTE MEAN ASSUMED TIME OF FIRE AIRCRAFT POSITION (ELECTRONIC
        COMPUTATION)
        56 XF=RANS*CTBCPB
        YF=RANS*STBCPB
        ZF=RANS*SPB
        GF=RANS*CPB
        RF=RANS
CC      ITERATION TO DETERMINE MEAN THEORETICAL INTERCEPT POINT
        63 RS=0.0
        VS=VMUZZ
        T=0.0
        21 XE=XF+VXE*T
        ZE=ZF+VZE*T
        YE2=XE*YE
        ZE2=ZE*ZE
        GE2=XE2+YE2

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```

RE2=GE2+ZE2
RE=SQRT(RE2)
RC=RE-RS
IF(RC.LT.1.0)GO TO 22
VC=VS-(XE*VXE+YE*VYE+ZE*VZE)/RE
IF(VD.LE.1.0)GO TO 64
T=T+RC/VD
IF(T.GT.TFMAX)GO TO 64
RS=RSHELL(T)
VS = VSHELL(T)
GC TO 21
22 T2=T*T
C
C
C
CHANGE 22 JAN 76
IF INTERCEPT POINT BELOW MASK*****SKIP FIRE
GE= SQRT(GE2)
IF(ATAN2(ZE,GE).LE. AMASK) GO TO 64
C
C
ITERATION TO DETERMINE ACTUAL INTERCEPT POSITION, RANGE, AND TIME
TU=AMINI(TFMAX, TMAX-TIME)
23 CALL RPLANE(TU)
IF(RSHELL(TU).GT.RA)GO TO 24
C
C
(SKIP FIRE ATTEMPT IF SHELL CANNOT CATCH AIRCRAFT)
IF((XA*GETVAL(VXFPA)+YA*GETVAL(VYFPA)+ZA*GETVAL(VZFPA))/RA .LT.
1 VSHELL(TU))GO TO 64
TU=TU-1.0
IF(TU)64,64,23
24 TL=0.0
25 T=0.5*(TL+TU)
CALL RPLANE(T)
RC=RA-RSHELL(T)
IF(RC.GT.1.0)GO TO 26
IF(RC.GT.-1.0)GO TO 37
TU=T
GC TO 25
26 TL=T
37 GC TO 25
GC=VS*RE-XE*VXE-YE*VYE-ZE*VZE
Q1=(VZE-VS*ZE/RE)/Q0
Q2=(XE*VYE-YE*VXE)/Q0
C
C
COMPUTE THE PARTIAL DERIVATIVES OF BIG THETA
DTDX=Q2*XE-YE

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DITDY=Q2*YE+XE
DITDZ=Q2*ZE
DITDR=(ZF*DITDZ+ YF*DITDY+XF*DIIDX)/RF
DITDI= XF*DITDY-YF*DIIDX
DITDP= GF*DITDZ-(YF*DITDY+XF*DIIDX)*ZF/GF
CC COMPUTE THE PARTIAL DERIVATIVES OF BIG PHI
DPPDX=Q1*XE
CPPDY=Q1*YE
DPPDZ=Q1*ZE+1.0
DPPDR=(ZF*DPPDZ+ YF*DPPDY+XF*DPPDX)/RF
DPPDT= XF*DPPDY-YF*DPPDX
DPPDP= GF*DPPDZ-(YF*DPPDY+XF*DPPDX)*ZF/GF
GE4=GE2*GE2
GC TO (210,220,230,240),IOEM
CC COMPUTATIONS FOR MODE 1 OPERATION
210 GO TO (211,212,213,214,215,999),IGT
CC TRACKING ERROR DISTRIBUTION SIZES (SPHERICAL COORDINATES)
(C FOR GT 1, 2, OR 3)
211 CONTINUE
212 SR2=(123.0+0.0225*R)**2
213 ST2=(.0643*TD)**2
SP2=(.1320*PD)**2
GO TO 219
(C FOR GT 4 AND 5)
214 CONTINUE
215 SR2=(123.0+0.0225*R)**2
ST2=(0.0167-.000710/(.0517+ABS(TD)))**2
SP2=(0.0116-.000216/(.0235+ABS(PD)-4.0*PDD))**2
CC SET UP THE ELEMENTS OF THE MATRIX B=AT
219 B11=A11*T11+A12*T21+A13*T31
B12=A11*T12+A12*T22+A13*T32
B13=A11*T13 +A13*T33
B21=A21*T11+A22*T21
B22=A21*T12+A22*T22
B23=A21*T13
B31=A31*T11+A32*T21+A33*T31
B32=A31*T12+A32*T22+A33*T32

```

```

C      B33=A31*T13      +A33*T33
C      COMPUTE THE PARTIAL DERIVATIVES OF BIG THETA AND BIG PHI W.R.T.
C      ON CARRIAGE INPUTS (ALPHA, BETA, SPEED)
      DTTDAP=B11*CTTDX+B12*DTTDX+B13*DTTDX
      DTTDBP=B21*DTTDX+B22*DTTDX+B23*DTTDX
      DTTDVP=B31*CTTDX+B32*DTTDX+B33*DTTDX
      DPPDAP=B11*CPPDX+B12*DPPDX+B13*DPPDX
      DPPDBP=B21*DPPDX+B22*DPPDX+B23*DPPDX
      LPPDVP=B31*CPPDX+B32*DPPDX+B33*DPPDX
C      COMPUTE THE VARIANCES OF BIG THETA AND BIG PHI
      STT2=((DTTDR**2)*SR2 + (DTTDT**2)*ST2 + (DTTDP**2)*SP2
      + ((DTTDAP**2)*SR2 + (DTTDBP**2)*ST2 + (DTTDP**2)*SP2
1     SPP2=((DPPDR**2)*SR2 + (DPPDT**2)*ST2 + (DPPDP**2)*SP2
1     + ((DPPDAP**2)*SR2 + (DPPDBP**2)*ST2 + (DPPDP**2)*SP2
      GO TO 29
C      COMPUTATIONS FOR MODE 2, 3, OR 4 OPERATION
220 GO TO (999,999,223,999,225,226),IGT
C      TRACKING ERROR DISTRIBUTIONS SIZES (SPHERICAL COORDINATES)
      (FOR MODE 2)
223 CONTINUE
225 CONTINUE
226 SR2=(41.0+0.0075*R)**2
      ST2=(.000982+.1681*TD*TD)**2
      SP2=(.000491+.033*ABS(ABS(PD)-4.0*PDD))**2
      GO TO 65
230 GO TO (999,999,233,999,235,236),IGT
      (FCR MODE 3)
C      CGNTINUE
233 CONTINUE
235 CONTINUE
236 SR2=(17.0+0.24*ABS(RDD)+0.018*RDD*RDD)**2+SC2RJ
      ST2=(0.00196+0.050*TD)**2
      SP2=(0.000982+0.11*ABS(ABS(POI)-2.0*PDD))**2+SP2MP
      GO TO 65
240 GO TO (999,999,243,999,245,246),IGT
      (FCR MODE 4)
C      CGNTINUE
243 CONTINUE

```

```

245 CONTINUE
246 SR2=(17.0+0.24*ABS(RDD)+0.018*RDD*RDD)**2+SC2RJ
ST2=(0.000982+0.1681*TD*TD)**2
SP2=(0.000491+J.033*ABS(PD)-4.0*PDD)**2+SP2MP

C
C
65 R8TD=RANS*THESD
R8PD=RANS*PHISD
VELOCITY COMPONENT ERROR DISTRIBUTION SIZES
SVX2=(SR2*(PHISD*CTBSPB+THESD*STBCPB)**2
+ST2*(R8PD*STBSPB-R8TD*CTBCPB-RANSD*STBCPB)**2
+SP2*(R8PD*CTBCPB-R8TD*STBSPB+RANSD*CTBSPB)**2)*ATLCON
1
2
SVY2=(SR2*(PHISD*STBSPB-THESD*CTBCPB)**2
+ST2*(R8PD*CTBSPB+R8TD*STBCPB-RANSD*CTBCPB)**2
+SP2*(R8PD*STBSPB-R8TD*CTBSPB+RANSD*STBSPB)**2)*ATLCON
1
2
SVZ2=(SR2*(PHISD*CPB)**2+SP2*(R8PD*SPB-RANS*CPB)**2)*ATLCON
C
C
C COMPUTE THE VARIANCES OF BIG THETA AND BIG PHI
STT2=((DTDR**2)*SR2+{(DTDT**2)*ST2+(DTDDP**2)*SP2
+{(DTDX**2)*SVX2+{(DTDY**2)*SVY2+{(DTDZ**2)*SVZ2}*T2)/GE4
1+SD2J
1
SPP2=((DPPDR**2)*SR2+{(DPPD**2)*ST2+{(DPPDP**2)*SP2
+{(DPPDX**2)*SVX2+{(DPPDY**2)*SVY2+{(DPPDZ**2)*SVZ2}*T2)/GE2
1+SD2J
1

```

COMPUTATION OF VULNERABLE AREA OF AIRCRAFT AT INTERCEPT

```

29 VP=VSHELL(T)
XU=XE/RE
YU=YE/RE
ZU=ZE/RE
XE=XU*RA
YE=YU*RA
ZE=ZU*RA
GE=SQRT(XE*XE+YE*YE)
VXA=GETVAL(VXFPA)
VYA=GETVAL(VYFPA)
VZA=GETVAL(VZFPA)
VGA=SQRT(VXA*VXA+VYA*VYA)
VA=SQRT(VGA*VGA+VZA*VZA)
VXI=VP*XU-VXA
VYI=VP*YU-VYA
VZI=VP*ZU-VZA
VI=SQRT(VXI*VXI+VYI*VYI+VZI*VZI)
ALFA=GETVAL(AFPA)
CA=COS(ALFA)

```

```

SA=SIN(ALFA)
BETA=BFPA(INDEX1)+FRACT*ANGLIM(BFPA(INDEX2)-BFPA(INDEX1))
CG=COS(BETA)
SG=SIN(BETA)
PZI=PFPA(INDEX1)+FRACT*ANGLIM(PFPA(INDEX2)-PFPA(INDEX1))
THE NEXT CARDS ELIMINATE AN EQUIVALENCE BY USING A DOUBLE
REPLACEMENT.
I33 = COS(PZI)
CP = COS(PZI)
TI3 = SIN(PZI)
SP = SIN(PZI)
Q1=VXI*CG+VYI*SG
Q2=VZI*CA-QI*SA
Q3=VYI*CG-VXI*SG
VXF=Q1*CA+VZI*SA
VYF=Q2*SP+Q3*CP
VZF=Q2*CP-Q3*SP

```

C

```

(SET UP INDICES FOR VULNERABLE AREA INTERPOLATION)

```

C C

```

F1=ATAN2(VYF,VXF)/QTRPI
IF(F1.LT.0.0)F1=F1+8.0
I1=F1-FLOAT(I1)
FI=F1+1
F2=ARCOS(VZF/VI)/QTRPI
I2=F2-FLOAT(I2)
I2=I2+1
F3=AMIN1(7.599999999,VI/152.4)
I3=F3-FLOAT(I3)
I3=I3+1
C1=1.0-F1
D2=1.0-F2
D3=1.0-F3
J1=I1+1
J2=I2+1
J3=I3+1

```

C C

```

(PERFORM LINEAR THREE DIMENSIONAL INTERPOLATION)

```

```

AVT=D3*(D2*(D1*VAT(I1,I2,I3)+F1*VAT(J1,I2,I3))+
F2*(D1*VAT(I1,J2,I3)+F1*VAT(J1,J2,I3))+
F3*(D2*(D1*VAT(I1,I2,J3)+F1*VAT(J1,I2,J3))+
F2*(D1*VAT(I1,J2,J3)+F1*VAT(J1,J2,J3)))

```

1
2
3

C SET UP DISTRIBUTION SIZES OF OTHER SOURCES OF RANDOM ERROR

```

C
SVA=XU*VXA+YU*VYA+ZU*VZA
CVA2=VA*VA-SVA*SVA
VMQ=.99*VMUZZ/RA-ASHCON
DTI=(VMQ-SQRT(VMQ*VMQ-4.0*B SHCON))/(2.0*B SHCON)--T
SLXMV2=CVA2*(DTI*VP/(VP-SVA/.99))**2
SLYFR2=(0.010*VA*T)**2
SLXFR2=SLYFR2*(1.0+CVA2/(VP-SVA)**2)
SAOGJ2=(0.003*RA)**2
SAOGJ2=(0.005*RA)**2
SAOBD2=(BDACON(IGT)*RA)**2
IF(10EM.EQ.1)GO TO 68
SAOPE2=(0.002*RA)**2
GO TO 67
68 SAOPE2=(V*V-((X*VX+Y*VY+Z*VZ)/R)**2)*
RA2*(0.0001463-7.478E-11*(R-1386.0)**2)**2
67 CCIIST=SAOAP2+SAOGJ2+SAOPE2+SAOBD2
C
C COMBINE ALL ERRORS INTO CNE DISTRIBUTION, COMPUTE BIAS
C
SXA2=STT2*RA2
SYA2=SPP2*RA2
SXL2=CDIST+SLXMV2+SLXFR2
CTT=XE/GE
CPTT=YE/GE
SPP=ZE/RE
T-YA*CTT
BXA=XA*STT-(YA*STT+XA*CTT)*SPP
BYA=ZA*CPTT-(YA*CPTT+XA*BYA)
BYA2=BYA*BYA
VAM=VZA*CPTT-(VYA*STT+VXA*CTT)*SPP
VAM2=VAM*VAM
VAI2=VAP*VAP
CD2=VAM2/VAI2
SD2=VAP2/VAI2
SXAT2=SXA2+CD2*SXL2+SD2*SYL2
SYAT2=SYA2+CD2*SYL2+SD2*SXL2
TWOCOV=2.0*VAM*VAP*(SXL2-SYL2)/VAI2
DIF=SXA2-SYAT2
DEN=2.0*SQRT(TWOCOV*TWOCOV+DIF*DIF)
HC2Z=DIF/DEN
CZ2=0.5+HC2Z
SZCZ=0.5-HC2Z
SZCZ=TWOCOV/DEN

```

```

STUFF=2.0*SZCZ*BXA*BYA
BXF2=CZ2*BXA2+SZ2*BYA2+STUFF
BYF2=CZ2*BYA2+SZ2*BXA2-STUFF
SYF2=CZ2*SYAT2+SZ2*SYAT2+SZCZ*TWOCOV
SVF2=CZ2*SYAT2+SZ2*SYAT2-SZCZ*TWOCOV
AVTPI=AVT/PI2

CC COMPUTE PROBABILITY OF KILL
STUFF=BXF2/(SXF2+AVTPI)+BYF2/(SYF2+AVTPI)
IF(STUFF.LT.50.0)GO TO 75
PK=0.0
GO TO 78
75 PK=AMIN1(1.0,EXP(-.5*STUFF)*AVTPI/SQRT((SXF2+AVTPI)*(SYF2+AVTPI)))
78 PS=(1.0-PK)**ISB
TI=TIME+T

CC ACCUMULATE PK AS A FUNCTION OF INPUT TIME INTERVALS
I=0
I=I+1
IF(TIME-GE.TINTER(I))GO TO 50
J=I
J=J+1
IF(TI.LT.TINTER(J))GO TO 52
GC TO 51
PTOTTF(I)=PK+PS*PTOTTF(I)
PTOTTI(J)=PK+PS*PTOTTI(J)
CPS=CPS#PS

CC ACCUMULATE PK FOR EACH SPHERICAL SECTOR
SPKT(I1,I2,I3)=PK+PS*SPKT(I1,I2,I3)

CC COMPUTE QUANTITIES FOR EXTENDED OUPUT, WHEN DESIRED
IF(IPRINT(6).LE.0)GO TO 20
O1=THESD#DEGREE
O2=PHISD#DEGREE
O3=ETHE4#DEGREE
C4=EPHI4#DEGREE
***** ABS(BXF2,BYF2,SXF2,SYF2) WERE ADDED BELCW AS A TEMPORARY
***** MEASURE OF NEGATIVE ARGUMENTS FOR A SQRT FUNCTION.
***** CAUSE OF NEGATIVE ARGUMENT: POSSIBLE ACCURACY DIFFERENCES
***** BETWEEN THE CDC AND IBM TYPE COMPUTERS.
***** CHANGE MADE BY LCDR C. SWENSON ON 3 MAR 1978.
BXF2 = ABS(BXF2)

```

```

05=SQRT(BXF2+FUZZ)
BYF2 = ABS(BYF2)
06=SQRT(BYF2+FUZZ)
SXF2 = ABS(SXF2)
07=SQRT(SXF2)
SYF2 = ABS(SYF2)
08=SQRT(SYF2)
09=1.0-CPS

CC
CC
WRITE EXTENDED OUTPUT

CALL PAGES(1,5,JP)
IF (JP.EQ.0) WRITE (6,1013) ISL,IGT,IEM,XGUN,YGUN,ZGUN,CIRCLE
WRITE(6,1014)IG,I0EM,TIME,T,II,R,RA,07,08,05,06,VI,01,02,03,04,
1
20 NROUND=NROUND+ISB

CC
CC
FIRE ADDITIONAL GUNS IN COMPLEX, IF ANY

64 IG=IG+1
IF(IG.GT.IGLJIG=1
TFIRE=TFIRE+TIPERS
IF(TIME.GT.TFIRE)GO TO 54
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
TTTT RRRR R A A CCC C K K
T R R A A C C K K
T RRRR A A A C C K K
T R R A A A A C C K K
T R R A A C C C K K
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
62 TIME=TIME+.064
IF(TIME.GT.TMAX)GO TO 69

CC
CC
SWITCH TO MODE 1 TRACKING IF JAMMING IS ABOVE THRESHOLD OR IF
RANGE IS TOO CLOSE

CC
CC
I0EM = IEM
SD2J = 0
SD2RJ = 0
IF(R.LT.RSMODE) GO TO 501
IF(IJAM.EQ.0) GO TO 502

C
C
CALL ECM2

IF(SJT.GT.SJTMAX) GO TO 501
IF(I0EM.EQ.3)CALL JAMER2(I0ECM,SJT,SD2J)

```



```

C 340 GO TO (999,999,343,999,345,346),IGT
343 CONTINUE
345 CONTINUE
346 THESD=0.910*TD+0.45*TDD+6.0*ETHE
THESD=0.75*PD-0.25*PDD+6.0*EPHI
RANS = 0.804*RD + 3.0*ERAN
C (LIMIT SLEW RATES AND ELEVATION ANGLE TO WEAPON MAXIMUMS)
C
73 THESD=SIGN(AMINI(TDMAX,ABS(THESD)),THESD)
PHISD=SIGN(AMINI(PDMAX,ABS(PHISD)),PHISD)
PHES=ANGLIM(THES+0.064*THESD)
PHIS=AMAX1(PHMIN,AMINI(PHMAX,PHIS+0.064*PHISD))
RANS=AMAX1(0.0,RANS+0.064*RANSD)
CTB=COS(THES)
STB=SIN(THES)
CPB=COS(PHIS)
SPB=SIN(PHIS)
CTBCPB=CTB*CPB
CTBSPB=CTB*SPB
STBSPB=STB*SPB
RCLS=RD
TCLS=TD
PCS=PD
IF(ITEM.LT.2)GO TO 60
C COMPUTE MEAN (SMOOTHED) VELOCITY COMPONENTS (ELECTRONIC
C COMPUTATION)
VXE=RANS*CTBCPB-RANS*(STBCPB*THESC+CTBSPB*PHISD)
VYE=RANS*STBCPB+RANS*(CTBCPB*THESC-CTBSPB*PHISD)
VZE=RANS*SPB+RANS*CPB*PHISD
VXE=VXE+EMDTA*(VYES-VXE)
VZE=VZE+EMDTA*(VZES-VZE)
VXES=VXE
VYES=VYE
VZE=VZE
GO TO 60
1013 FORMAT(LOCATION',15,6X,'GUN TYPE',12,6X,'ERROR MODE',12,6X,'OL
1 POSITION=(,F8.1,,F8.1,,)RADIUS=',F6.1,,M./,OL
2 FIRE INTCP',F8.1,,F8.1,,)RADIUS',F6.1,,M./,OL
3 MEAN FIRE INTCP',31X,'CLOSE AZIM. ELEV.
4 SIG1 MEAN BIAS1',N OEM VEL. TIME RATE AZ. ERR EL. ERR
5 AREA SIG2 CUM. PK./)
1014 FORMAT(I2,I3,2F7.2,F8.2,2F7.0,1X,2F6.1,1X,2F7.1,F8.1,1X,2F7.2,1X,

```

```

3F7.2,2F9.5)
1044 2 FORMAT(IX,F7.2,4X,F8.0,2X,F9.2,5X,F7.2,5X,F7.2,
15X,I2,3X,E10.4,5X,E10.4,5X,E10.4,5X,F7.2)
15 CALL $15
69 CALL $69
999 CALL $999
8888 CALL $8888
9994 CALL $9994
9995 CALL $9995
9996 CALL $9996
9997 CALL $9997
7777 CONTINUE
END

```

```

C
C
C
C
SUBROUTINE TOOBIG WAS CONSTRUCTED FROM A PORTION OF THE ORIGINAL POOL
MAIN PROGRAM IN ORDER TO ENABLE PROPER COMPILATION OF THE MAIN.

```

```

SUBROUTINE TOOBIG (TEMP, SPKT, SPKT2)
COMMON/BLOCK1/ITITLE(20)
COMMON/BLOCK2/NFPA, TMIN, TMAX, DTFPA
COMMON/BLOCK3/YGUN, YGUN, ZGUN
COMMON/BLOCK4/IGT, IEM, ICB, ISB, IGL, CIRCLE
COMMON/BLOCK5/NRHOS, RHO(9)
COMMON/BLOCK6/INTINTS, TINTER(10)
COMMON/BLOCK7/IVACCM(20), VA(9,5,5)
COMMON/BLOCK8/TREACT, TRACK1, TRACK2
COMMON/BLOCK9/TROUND(6), THDMAX(6), PHDMAX(6), PHIMIN(6), PHIMAX(6),
VELMIN(6), VELMAX(6), RANMIN(6), RANMAX(6),
ATLAG(6), ETHMAX(6), EPHMAX(6), RVACCON(6), RVBCON(6), VMUZEL(6)
COMMON/BLOCKA/TFMAX1(6), TFMAX2(6), QTRPI, SQRT2
COMMON/CONSTS/DEGREE, RADIANS, PI, P12, QTRPI, SQRT2
COMMON/HEADFO/LINE, NUMBER
COMMON/NFPARM/XA, YA, ZA, RA2, RA, TIME
COMMON/IGXGYG/IG, XG(8), YG(8)
COMMON/MAGICC/FRACT, INDEX1, INDEX2
COMMON/VASBS/VMUZZ, ASHCON, BSHCON, CQUAD
COMMON/BUDGET/BDACON(6)
COMMON/X, Y, Z, ROL, PIT, HDG,
* * *
COMMON/NEWA/ FJAM, GJ, SJI, SN
COMMON/NEWB/ Y1, P2, VX, BEFA, IRMP, PLEN, O1, O8, D1, JJJ, XR, ZT, PZI, IERR
COMMON/NEWC/ CP, X2, P1, B2, JFILE, IJAM, ISL, O4, CPS, TPERS, JMCDE, YI, T2
COMMON/NEWD/ T33, X1, VZ2, B1, VX2, C3, CPK, NROUND, IOFF, I, Xf, TM, SP, T1
COMMON/NEWF/ Z1, A2, VZ1, VZ2, PHI, IF5, IEOF, YR, IF2, T13, IFILE, Y2, A1
COMMON/NEWG/ V2, VZ1, V, FUZZ, IF9, SD2RJ, SD2RJM, V1, VY2, F, JP, IF7, REFC

```

```

COMMON XFPA(1201), YFPA(1201), ZFPA(1201), BFPA(1201), AFPA(1201),
PFPA(1201), VFPA(1201), VYFPA(1201), VZFPA(1201)
1 CCOMMON ICARD(20), PTOTTF(10), PTOTTI(10)
CCCOMMON SPKTCI(32,8), IPRINT(6), IFLAGS(4)
CCCOMMON PKTIDC(9), PKTIDC(10,9), PKTFDC(10,9)
CCCOMMON INUNIT
DIMENSION TEMP(16,6), SPKT(8,4,8), SPKT2(32,8)
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
I I I I I N N N P P P P U U T T T T T
I I I I I N N N P P P P U U U U T T T
I I I I I N N N P P P P U U U U T T T
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
ENTRY S8888 (TEMP, SPKT, SPKT2, *, *, *)
ASSIGN 9997 TO IE0F
GO TO 15
12 ASSIGN 9999 TO IE0F
ENTRY S15 (TEMP, SPKT, SPKT2, *, *, *)
15 READ(INUNIT, 1007, ERR=9992, END=9993) I, ICARD
GO TO 14 OR I.GT.14 GO TO 9998
GO TO (101, 102, 103, 104, 105, 106, 107, 108, 109, 110, 111, 112,
1, 113, 114), I
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
CCODE DATA BLOCK 01 -- HEADER INFORMATION
XRQA VERSION ALLOWS ONLY 70 SPACES FOR TITLE.
101 DC 17 I=1,20
17 ITITLE(I)=ICARD(I)
GC TO 15
CCCCCCCC
CCODE DATA BLOCK 02 -- FLIGHT PATH INPUT, TRANSLATION, ROTATION,
AND INTERPLATION
SUGGEST SECTION ON TAPES BE RE-WRITTEN FOR COMPATIBILITY WITH
ASD FLIGHT PATH PROGRAMS. SEE P-1127, FOR EXAMPLE.
102 READ (99, 1028) JMODE, KMODE, TMIN, TMAX, CTFPA, XR, YR, XT, YT, PSI,
ZI
1 THIS CARD WAS ADDED
IF2=1
TMAX=TMAX-TMIN
TM=0.0
T2=-1.0
NFPA=0
K=0

```

```

PZI=PSI*RADIAN
T13 = SIN(PZI)
T33 = SIN(PZI)
CP = COS(PZI)
LINE=66
IF(JMODE)5,5,1
1 ASSIGN 4 TO IERR
  ASSIGN 3 TO IERR
  IF(JMODE.GT.IFILE)GO TO 2
  REWIND 9
  IFILE=1
  GO TO 4
  IFILE=IFILE+1
2 IF READ(9,ERR=9992,END=9993)
3 GO TO 3
4 ASSIGN 9999 TO IECF
  ASSIGN 28 TO IERR
  REAC(9,ERR=9992,END=9993)
28 ASSIGN 9996 TO IERR
  JMODE=6
5 T1=T2
  X1=X2
  Y1=Y2
  Z1=Z2
  P1=P2
  A1=A2
  B1=B2
  V1=V2
  VX1=VX2
  VY1=VY2
  VZ1=VZ2
  IF(JMODE)6,6,7
6 READ(INUNIT,1000,ERR=9992,END=9993)T2,X,Y,Z,VX,VY,VZ2,B2,A2,P2
  B2=B2*RADIAN
  A2=A2*RADIAN
  P2=P2*RADIAN
  V2=SQRT(VX*VX+VY*VY+VZ2*VZ2)
  GO TO 9
7 JMODE=JMODE+1
  JMODE=LE.6)GO TO 8
  JMODE=1
  ERR=9992,END=9993)TEMP
8 READ(TEMP( 1,JMODE)
  =TEMP( 2,JMODE)
  X =TEMP( 3,JMODE)
  Y =TEMP( 4,JMODE)
  Z =TEMP( 4,JMODE)

```

```

VX =TEMP( 5,JMODE )
VY =TEMP( 6,JMODE )
VZ2=TEMP( 7,JMODE )
V2 =TEMP( 11,JMODE )
B2 =TEMP( 13,JMODE )
A2 =TEMP( 14,JMODE )
9 X2=XT+(X-XR)*CP+(Y-YR)*SP
  Y2=YT+(Y-YR)*CP-(X-XR)*SP
  Z2=ZT+Z
  VX2=VX*CP+VY*SP
  VY2=VY*CP-VX*SP
  B2=ANGLIM(B2-PZ1)
  T2=T2-TMIN
10 IF (TM.GT. T2)GO TO 5
  F=((TM-T1)/(T2-T1)
  NFPA=NFPA+1
  XFPA(NFPA)=X1+F*(X2-X1)
  YFPA(NFPA)=Y1+F*(Y2-Y1)
  ZFPA(NFPA)=Z1+F*(Z2-Z1)
  VXFPA(NFPA)=VX1+F*(VX2-VX1)
  VYFPA(NFPA)=VY1+F*(VY2-VY1)
  VZFPA(NFPA)=VZ1+F*(VZ2-VZ1)
  VFPA(NFPA)=V1+F*(V2-V1)
  BFPA(NFPA)=ANGLIM(B1+F*ANGLIM(B2-B1))
  AFPA(NFPA)=A1+F*(A2-A1)
  PFPA(NFPA)=ANGLIM(P1+F*ANGLIM(P2-P1))
  K=K-1
  IF(K)34,34,35
34 K=KMODE
  PHI=DEGREE*PFPA(NFPA)
  BETA=DEGREE*BFPA(NFPA)
  ALFA=DEGREE*AFPA(NFPA)
  CALL PAGES(1,2,JP)
  IF (JP.EQ.0) WRITE (6,1003)
  WRITE(6,1004)TM,VXFFPA(NFPA),YFPA(NFPA),ZFPA(NFPA),BETA,ALFA,PHI
1  TM=OTFPA*FLOAT(NFPA)
  IF(TM.LE.TMAX)GO TO 10
  TMAX=TM-OTFPA-FUZZ
  CALL PAGES(4,0,JP)
  WRITE(6,1005)XR,YR,XT,YT,PSI,ZT
  CALL TPL0T(NFPA)
  IF(JMODE.GT.0)OR. INUNIT.EQ.8)GO TO 12
  ASSIGN 12 IC IE0F
13 READ(INUNIT,1000,ERR=9992,END=9993)
  GC TO 13

```

C

```

C DECODE DATA BLOCK 03 -- WEAPON LOCATION
C 103 READ (99,1008) XGUN,YGUN,ZGUN
C GO TO 12
C DECODE DATA BLOCK 04 -- WEAPON TYPE, MODE, NUMBER OF BARRELS (CYCLIC
C AND SIMULTANEOUS), NUMBER OF WEAPONS PER
C LOCATION, RADIUS OF CIRCLE OF WEAPON COMPLEX
C 104 READ (99,1009) IGT,IEM,ICB,ISB,IGL,CIRCLE
C 61 IF(IGL-1)9998,61,66
C XG(I)=0.0
C YG(I)=0.0
C GO TO 12
C 66 DC 16 I=1,IGL
C F=PI2*FLOAT(I)/FLOAT(IGL)
C XG(I)=CIRCLE*COS(F)
C YG(I)=CIRCLE*SIN(F)
C GO TO 12
C DECODE DATA BLOCK 05 -- WEAPON DENSITY FACTORS
C 105 READ (99,1015) IF5,NRHOS,(RFO(I),I=1,NRHOS)
C GO TO 12
C DECODE DATA BLOCK 06 -- PK ACCRUAL TIME INTERVALS
C 106 READ (99,1015) IF6,NTINTS,(TINTER(I),I=1,NTINTS)
C NTINTS=NTINTS+1
C TINTER(NTINTS)=999.99
C GO TO 12
C DECODE DATA BLOCK 07 --- AIRCRAFT VULNERABLE AREAS
C 107 DC 11 I=1,20
C 11 IVACOM(I)=ICARD(I)
C READ(INUNIT,1000,ERR=9992,END=9993)(VAT(1,1,K),K=2,9)
C DC 88 J=2,4
C DO 89 I=1,8
C READ(INUNIT,1000,ERR=9992,END=9993)(VAT(I,J,K),K=2,9)
C CONTINUE
C 89 DC 88 K=2,9
C 88 VAT(9,J,K)=VAT(1,J,K)
C READ(INUNIT,1000,ERR=9992,END=9993)(VAT(1,5,K),K=2,9)
C DC 18 K=2,9
C DO 18 I=2,9
C VAT(I,5,K)=VAT(1,5,K)
C 18 VAT(I,1,K)=VAT(I,1,K)

```

```

IF7=1
GO TO 12
C DECODE DATA BLOCK 08 -- WEAPON REACTICN AND TRACK TIMES
C 108 READ (99,1008) TTRACT,TRACK1,TRACK2
C DECODE DATA BLOCK 09 -- WEAPON PARAMETERS
C 109 READ (99,1008) TROUND(IGT),TFDMAX(IGT),PHDMAX(IGT),
PHIMIN(IGT),PHIMAX(IGT),VELMIN(IGT),VELMAX(IGT),
RANMIN(IGT),RANMAX(IGT)
1 READ(INUNIT,1000,ERR=9992,END=9993)ATLAG(IGT),ETHMAX(IGT),
2 EPHMAX(IGT),RMODES(IGT)
1 IF9=1
GC TO 12
C DECODE DATA BLOCK 10 -- SHELL PARAMETERS
C 110 READ (99,1008) TFMAX1(IGT),TFMAX2(IGT),RVACON(IGT),
RVBCON(IGT),VMUZEL(IGT)
1 IF9=1
GO TO 12
C DECODE DATA BLOCK 11 --- INPUT OPTION (CARD/TAPE)
C 111 READ (99,1028) I
81 INUNIT=5
82 GO TO 15
82 INUNIT=8
ASSIGN 83 TO IEOF
IF(I.GT.JFILE)GO TO 84
REWIND 8
JFILE=1
83 IF(I-JFILE)12,12,84
84 JFILE=JFILE+1
85 READ(8,1000,ERR=9992,END=83)
GC TO 85
C DECODE DATA BLOCK 13 --- LOW ALTITUDE RADAR MULTIPATH EFFECT
C 113 READ (99,1038) IMUL,IRMP,REFC
IF(IMUL.EQ.0) GO TO 12
CALL PAGES(12,0,JP)
WRITE(6,1098) IRMP,REFC
1098 FORMAT(//,' MULTIPATH INPUTS (INITIAL OR CFANGED ',//,

```

```

0: IRMP =      :15./,
1: REFC =      :F6.3,////
  GC TO 12

C
C
C  DECODE DATA BLOCK 14 --- ECM
114 SD2RJ = 0.
   SD2RJ = 0.
   READ (99,1041)
      * IF(IJAM.EQ.0) GO TO 12
        CALL PAGES(28,0,JP)
        WRITE(6,1096) IP,IJ,GAINJ,PJW,PLEN,IX,XSEC,CALX,IRECM,SJTMAX
1096 FORMAT(///,ECM INPUTS (INITIAL CR CHANGED) ,//,
0: IP =      :15./,
1: IJ =      :15./,
2: GAINJ(DB) = :F7.2,/,
3: PJW(W) =   :F9.2,/,
4: PLEN(S) =  :E12.6,/,
5: TX =      :15./,
6: XSEC(SQM) = :F10.3,/,
7: CALX =    :F10.3,/,
8: IRECM =   :15./,
9: SJTMAX(DB) = :F7.2,////
   CALL ECMI
   IF(IRECM.EQ.3) CALL JAMER1(PLEN,SD2RJM)
   IF(IEM.EQ.4) CALL JAMER1(PLEN,SD2RJM)
   GO TO 12

C
C
C  DECODE DATA BLOCK 12 -- PRINT OPTIONS FOR OUTPUT FORMAT - ALSO
   SIGNALS FOR RUN TO BEGIN
112 READ (99,1016) IPRINT

C
C
C  PRINT DATA BLOCKS 6, 7, 9, AND 10 (IF THEY CHANGE)
   "IF2" IS USED TO SET LINE COUNT TO PROPER VALUE. INPUT AND
   OUTPUT PRINT OUT START A NEW PAGE FOR EACH "12" CARD.
   IF (IF2.EQ.0) LINE = 66
   IF2=0
   IF(IJAM.EQ.0) GO TO 48
   IF(IP.EQ.0) GO TO 48
   WRITE(11,1043) ITITLE
   WRITE(11,1042)
   WRITE(11,1001) ISL,IGT,IEM,XGUN,YGUN,ZGUN,CIRCLE
   WRITE(11,1045)
48 CCNTINUE
   IF(IF5.EQ.0)GO TO 97

```

```

CALL PAGES(5,0,JP)
WRITE(6,1029)NRHOS,(RHO(I),I=1,NRHCS)
IF5=0
IF(IF6.EQ.0)GO TO 98
97 CALL PAGES(5,0,JP)
WRITE(6,1011)NTINTS,(TINTER(I),I=1,NTINTS)
IF6=0
IF(IF7.EQ.0)GO TO 99
58 CALL PAGES(7,0,JP)
WRITE(6,1010)IVACOM
DO 19 N=1,26
J=(N+14)/8
I=15+N-J*8
CALL PAGES(1,7,JP)
IF(JP.EQ.0)WRITE(6,1010)IVACOM
19 WRITE(6,1026)N,(VAT(I,J,K),K=2,9)
IF7=0
IF(IF9.EQ.0)GO TO 87
99 CALL PAGES(10,0,JP)
WRITE(6,1025)
DO 86 I=1,16
O1=PHIMIN(I)*DEGREE
O2=PHIMAX(I)*DEGREE
O3=THDMAX(I)*DEGREE
O4=PHDMAX(I)*DEGREE
O7=ETHMAX(I)*DEGREE
O8=EPHMAX(I)*DEGREE
86 WRITE(6,1012)I,TRCOUN(I),VELMIN(I),VELMAX(I),RANMIN(I),RANMAX(I)
1, RVACON(I),RVBCON(I),
2, ATLAG(I),O7,O8
IF9=0
RETURN
10E8.0)
1000 FORMAT(1, LOCAT ION', I5, 6X, 'GUN TYPE', I2, 6X, 'ERROR MCDE', I2, 9X,
1001 FORMAT(1, LOCAT ION', I5, 6X, 'F7.0', 'RADIUS=', F5.0, 'M./',
1, 'POSITION', I3X, TIME', I4X, 'XX', 7X, 'YY', 7X, 'ZZ', 12X, 'SPEED', 6X, 'XDOT',
1003 FORMAT(1, 'VDOT', 6X, 'ZDOT', 1X, 'HEADING', 5X, 'CLIMB', 6X, 'ROLL',
1004 FORMAT(1, 'FX', F6.2, 7X, 'FY', F6.2, 7X, 'FZ', F6.2, 7X, 'FYR', F9.2, 7X, 'FZR', F9.2, 7X, 'F9.2',
1005 FORMAT(1, 'OXR=', F9.2, 7X, 'DZ=', F9.2)
1, PSI=', I2, 15A4, A2)
1007 FORMAT(8X, 9E8.0)
1008 FORMAT(3X, 5I1, E8.0)
1009 FORMAT(1, 'VULNERABLE AREA (SQ. METERS) AS A FUNCTION OF IMPACT',
1010 FORMAT(1, 'SPEED (METERS/SEC) AND ASPECT VIEW', /5X, 20A4, /, 'VIEW', 7X, 0',
1, '305',
2, 152
1011 FORMAT(1H-, I2, ' TIME INTERVALS FOR PK ACCUMULATION', /4X, 0.00',
1, 10F8.2)

```

```

1012 FORMAT (I2,F7.3,3F7.2,3F6.2,F6.0,2F12.7,2F6.1,2F7.0,F8.2,2F8.3)
1015 FORMAT (5X,I1,IX,I1,9E8.0)
1016 FORMAT (2X,6I1)
1025 FORMAT (10G,BALLISTIC,MIN,MAX,AZIM,ELEV,MAX,SMOOTH,MAX,MUZZ,B
3X,EL,/,T,ROUND,ELEV,MIN,MAX,VEL,MAX,TOF1,TOF2,MAX,AZ,MA
4INSTANT,1,CONSTANT,2,MIN,MAX,ELEV,MIN,MAX,CONST,ERROR,CO
5RNROR,/)
1026 FORMAT (I5,4X,0.00,8F8.2)
1028 FCRMAT (2X,I4,I2,9E8.0)
1029 FCRMAT (1H-,I1,1,DENSITY CLASSES FOR PK ACCUMULATION'//9F12.5)
1038 FCRMAT (2X,I3,I5,F10.0)
1041 FCRMAT (2X,I3,I3,I2,3F10.0,I5,2F10.0,I5,F10.0)
1042 FCRMAT (//,ECM VARIABLES;)
1043 FCRMAT (//,AAASIM---,10A10)
1045 FCRMAT (//,TIME RANGE(M) X-SEC(SQM) JAM.GAIN(DB) J/S(DB)',
1, FIRE ADD.TRK.VAR ADD.RNG.VAR S/N(CB);)
2, MODE CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
COO U U TTTT P P P P U U U TTTT
0 0 U U U T T P P P U U U T
0 0 U U U T T P P P U U U T
0 00 U U U T T P P U U U T
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
ENTRY $69 (TEMP,SPKT,SPKT2,*,*,*)
92 IFLAGS(I)=IFLAGS(I)+IPRINT(I)
C C PRINT PK AS A FUNCTION OF AIRCRAFT ASPECT AND IMPACT SPEED
C C IF(IPRINT(5).EQ.0)GO TO 80
C C CALL PRSEGS(SPKT2,ISL)
C C COMPUTE PK AS A FUNCTION OF ASPECT AND IMPACT SPEED FOR ALL GUNS
80 DO 36 I=1,32
DC 36 J=1,8
PK=RHO(I)*SPKT2(I,J)
36 SPKTOT(I,J)=PK+(1.0-PK)*SPKTOT(I,J)
C C STORAGE OF PK VS DENSITY FACTOR AND TIME INTERVALS (AT FIRE AND
C C INTERCEPT) PER WEAPON OR WEAPON COMPLEX
CPK=1.0-CPS
DO 55 I=1,NRHOS

```

```

D=RHO(I)
PK=D*CPK
PKTTDC(I)=PK+(1.0-PK)*PKTTDC(I)
DC 55 J=1,NTINTS
D1=D*PTOTTF(J)
PKTFDC(J,I)=D1+(1.0-D1)*PKTFDC(J,I)
D2=D*PTOTTF(J)
55 PKTIDC(J,I)=D2+(1.0-D2)*PKTIDC(J,I)
C
C COMPUTE, STORE, AND WRITE TOTAL PKs FOR ENTIRE ARRAY OF WEAPONS
C
F=FLOAT(NROUND/ISB)*TPERS
WRITE(7)ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PTOTTF,PTOTTI,I,PRINT
ISL=ISL+1
ASSIGN 70 TO IEOF
GO TO 15 7
ENDFILE 7
LINE=66 7
REWIND 7
PUNCH2 CARDS WRITE ONLY ONE CARD IMAGE ON TAPE4 WHEN THERE IS
ONLY ONE DEFENSE PER EMPLOYMENT (I.E., ONLY ONE "12" CARD
BEFORE 7/8/9 END-OF-RECORD).
K = ISL PAGES(4,0,JP)
CALL PAGES(4,0,JP)
WRITE(6,1033)
ASSIGN 40 TO IEOF
79 READ(7,ERR=9992,END=9993)ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,
1 TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PTOTTF,PTOTTI,I,PRINT
CALL PAGES(1,4,JP)
IF (JP.EQ.0) WRITE (6,1033)
WRITE(6,1034) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,CIRCLE,IGL,
1 TREACT,TTRACK,PTOTTI,I,ITITLE(9),ITITLE(10),ISL,IGT,IEM,ICB,ISB,IGL,CPK,
1 WRITE(4,1060) ITITLE(9),ITITLE(10),ISL,IGT,IEM,ICB,ISB,IGL,CPK,
1 NROUND,XGUN,YGUN,ZGUN,F,NUMBER
GO TO 79
40 IF (K.NE.2) WRITE (4,1064) ITITLE(9),ITITLE(10),PKTTDC(1),NUMBER
IF (IFLAGS(2).LE.0) GO TO 76
REWIND 7
CALL PAGES(4,0,JP)
WRITE(6,1031) (I,I=1,10)
77 READ(7) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PTOTTF,PTOTTI,I,PRINT
IF (IPRINT(2).LE.0) GO TO 77
CALL PAGES(1,4,JP)
IF (JP.EQ.0) WRITE (6,1031) (I,I=1,10)
WRITE(6,1032)ISL,(TINTER(I),I=1,NTINTS)

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AD-A057 907

NAVAL POSTGRADUATE SCHOOL MONTEREY CALIF
ADAPTATION OF THE IMPROVED ANTI-AIRCRAFT ARTILLERY SIMULATION CO--ETC(U)
MAR 78 C F SWENSON

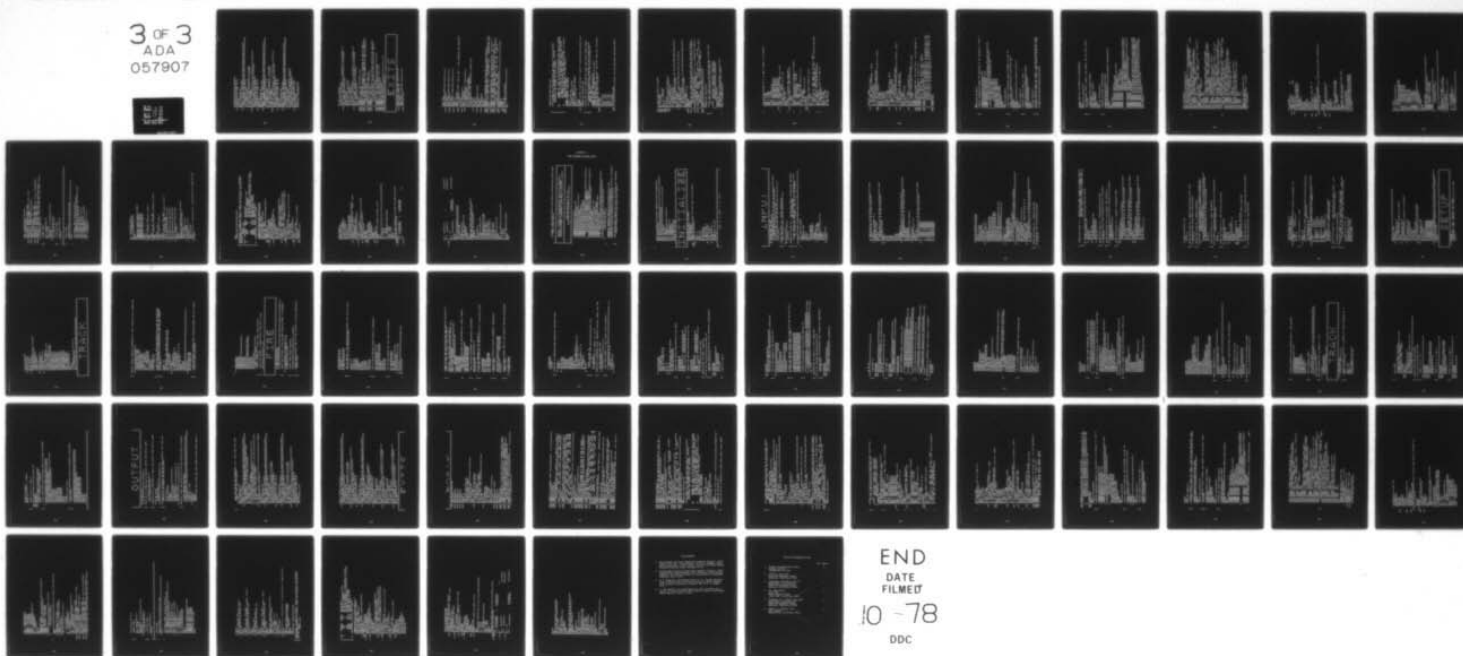
F/G 19/5

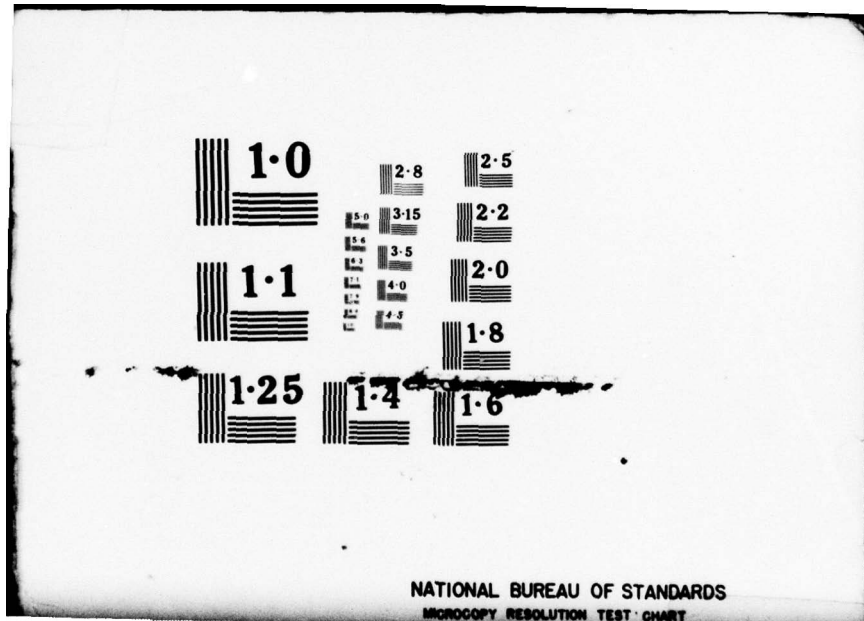
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NL

3 OF 3
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NL





1.0

2.8
3.15
3.5
4.0
4.5
5.0
5.6

2.5

2.2

1.1

2.0

1.8

1.25

1.4

1.6

NATIONAL BUREAU OF STANDARDS
MICROCOPY RESOLUTION TEST CHART

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IFLAGS(2)=IFLAGS(2)-IPRINT(2)
IF(IFLAGS(2).GT.0)GO TO 77
IF(IFLAGS(3).LE.0)GO TO 42
76 REWIND 7
CALL PAGES(4,0,JP)
WRITE(6,1035)(I,I=1,10)
41 READ(7) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 R CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PLOTF,PLOTTI,IPRINT
IF(IPRINT(3).LE.0)GO TO 41
CALL PAGES(2,4,JP)
IF(JP.EQ.0)WRITE(6,1035)(I,I=1,10)
WRITE(6,1018)ISL,(PLOTTF(I),I=1,NTINTS)
DO 410 I=2,NTINTS
410 PLOTF(I)=PTOTTF(I-1) + (1.0-PTOTTF(I-1))*PTOTTF(I)
IFLAGS(3)=IFLAGS(3)-IPRINT(3)
IF(IFLAGS(3).GT.0)GO TO 41
42 REWIND 7
CALL PAGES(4,0,JP)
WRITE(6,1036)(I,I=1,10)
43 READ(7) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 R CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PLOTF,PLOTTI,IPRINT
IF(IPRINT(4).LE.0)GO TO 43
CALL PAGES(2,4,JP)
IF(JP.EQ.0)WRITE(6,1036)(I,I=1,10)
WRITE(6,1018)ISL,(PLOTTF(I),I=1,NTINTS)
DO 430 I=2,NTINTS
430 PLOTTF(I)=PTOTTF(I-1) + (1.0-PTOTTF(I-1))*PTOTTF(I)
IFLAGS(4)=IFLAGS(4)-IPRINT(4)
IF(IFLAGS(4).GT.0)GO TO 43
71 REWIND 7
CALL PAGES(4,0,JP)
WRITE(6,1017)(I,I=1,9)
74 READ(7) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 R CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PLOTF,PLOTTI,IPRINT
CALL PAGES(1,4,JP)
IF(JP.EQ.0)WRITE(6,1017)(I,I=1,9)
WRITE(6,1059)ISL(RHO(I),I=1,NRHS)
IFLAGS(1)=IFLAGS(1)-IPRINT(1)
IF(IFLAGS(1).GT.0)GO TO 74
44 CALL PAGES(7+NTINTS,0,JP)
WRITE(6,1015)
WRITE(6,1021)(I,I=1,9)
PLOTTF(1)=PKTFDC(1,1)

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```

DC 440 I=1, NTINTS
IF (I.EQ.1) GO TO 440
PTOTTF(I) = PTOTTF(I-1) + (1.0-PTCTTF(I-1))*PKTFDC(I,1)
WRITE (6,1018) I, PTOTTF(I), (PKTFDC(I,J), J=1, NRHOS)
CALL PAGES (9+NTINTS,0,JP)
WRITE (6,1020)
WRITE (6,1021) (I,I=1,9)
PTOTTI(I) = PKTIDC(I,1)
DO 441 I=1, NTINTS
IF (I.EQ.1) GO TO 441
PTOTTI(I) = PTOTTI(I-1) + (1.0-PTOTTI(I-1))*PKTIDC(I,1)
441 WRITE (6,1018) I, PTOTTI(I), (PKTIDC(I,J), J=1, NRHOS)
CALL PRSEGS(SPKTOT,0)
GO TO 95
1017 FORMAT(15, LOC, 12X, 9(7X,RHO, I2)/)
1018 FORMAT(15, 10F12.7)
1019 FORMAT(15, -ATTRITION ACCRUED AS A FUNCTION OF TIME OF FIRE, )
1020 FORMAT(15, -ATTRITION ACCRUED AS A FUNCTION OF TIME AT INTERCEPT, )
1021 FORMAT(15, 9(5X,CLASS, I2)/)
1023 FORMAT(15, 10X, 10(6X, TIME, I2)/)
1031 FGRMAT(15, LCC, 10(6X, TIME, I2)/)
1032 FORMAT(15, LOC, 12X, 9(7X,RHO, I2)/)
1033 1 8X, YGUN, 8X, ZGUN, 8X, RADIUS, 3X, FIRE TIME, 8X, XGUN,
2 8X, GT, EM, CB, SB, LOC, )
1034 FORMAT(15, F12.7, I12.4, F12.2, F9.2, I3, 2F12.2, 4I3, I5)
1035 FORMAT(15, LOC, 10(4X, PK(TF, I2, I1)/))
1036 FORMAT(15, LOC, 10(4X, PK(TI, I2, I1)/))
1058 FGRMAT(15, LCC, 10(6X, TIME, I2)/)
1059 FGRMAT(15, 12X, 9F12.7)
1060 FGRMAT(2A9, I3, 5I2, F10.7, I5, 2F7.0, F6.0, F6.1, I6)
1064 FGRMAT(2A9, I3X, F10.7, 3I3, I6)
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
EEEE X X X IIII TTTT SSSS
EEE X X I I SSS
EEEE X X X IIII T SSSS
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
ENTRY S999 (TEMP, SPKT, SPKT2, *, *, *)
CALL EXIT
999 WRITE(6,1027) IGT, IEM
ENTRY S9994 (TEMP, SPKT, SPKT2, *, *, *)
9994 WRITE(6,1040)

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9955 CALL EXIT
      ENTRY S9995 (TEMP, SPKT, SPKT2, *, *, *, *)
      WRITE (6, 1039)
      CALL EXIT
9996 ENTRY S9996 (TEMP, SPKT, SPKT2, *, *, *, *)
      WRITE (6, 1037)
      CALL EXIT
9997 ENTRY S9997 (TEMP, SPKT, SPKT2, *, *, *, *)
      REWIND 4
      LINE = 66
6060 WRITE (6, 6060)
      FORMAT (///, 1X, '***** END OF JOB - - POOL SCENARIO RUN COMPLETE',
1)
      RETURN 3
6061 CALL PAGES(3, 0, JP)
      WRITE (6, 1061)
990 READ(4, 1062, END=991) ICARD, IPRINT, I
      CALL PAGES(1, 3, JP)
      IF (JP.EQ.0) WRITE (6, 1061)
      WRITE (6, 1063) ICARD, IPRINT, I
      IF (ISM.NE.10FF) CALL AVG(ICARD(3), IPRINT, ISH)
      GO TO 990
991 REWIND 4
      CALL EXIT
9998 RETURN 3
      WRITE (6, 1006) I, ICARD
9999 CALL EXIT
      WRITE (6, 1002)
      CALL EXIT
      RETURN 3
1002 FORMAT (///, 'UNEXPECTED END-OF-RECORD/FILE ENCOUNTERED. ')
1006 FORMAT (///, 'IMPROPER INPUT CARD ENCOUNTERED. ', I2, 19A4, A2, ' ')
1027 FORMAT (///, 'GUN TYPE ', I2, ', ERRCR MODE ', I2, ', COMBINATION INVAL
1) IC. ')
1037 FC RMAT (///, 'UNRECOVERABLE PARITY ERROR DETECTED. CALL EXIT. ')
1039 FC RMAT ( ' ECM SHOULD NOT BE SPECIFIED WHEN IEM IS 1 OR 2 ')
1040 FC RMAT ( ' MULTIPATH TRACKING ERROR SPECIFIED WITH IEM=3 ONLY ')
1061 FC RMAT (1H0, 4X, DATE, 7X, TIME, 4X, LCC, 5X, P(K',
      ' ILL) RDS XGUN YGUN ZGUN F TIME
1062 FC RMAT (2(A9, A3, 5A2, A10, A5, 2A7, 3A6)
1063 FC RMAT (2(2X, A9), 2X, A3, 5(2X, A2), 2X, A10, 1X, A5, 2(1X, A7), 3(2X, A6)
      87 RETURN 1
      55 RETURN 2
9992 GC TO IERR, (3, 28, 9996)
9993 GO TO IEOF, (4, 12, 40, 70, 83, 9997, 9999)
      END

```

```

SUBROUTINE PAGES(N,NT,JP)
  KEEPS NUMBER OF LINES PER PAGE LESS THAN 59, PRINTS HEADER,
  AND GETS TIME INFORMATION FROM SYSTEM. REPLACES
  NPAGE(MAX) AND HEADER IN AFATL PROGRAM.

  N      NUMBER OF LINES TO BE PRINTED BEFORE NEXT CALL TO PAGES.
  NT     NUMBER OF LINES IN TITLE OR HEADER CF DATA BEING PRINTED.
  JP     IF CALL TO PAGES IS TO PRINT HEADER ONLY, "N" SHCULD
        BE NUMBER OF LINES AND "NT" SHOULD BE ZERO.
        FLAG FROM PAGES, SET TO ZERO WHEN A NEW PAGE IS STARTED
        INDICATING NECESSITY TO PRINT HEADER.
  COMMON /BLOCK1/ TITLE(20)
  COMMON /BLOCK2/ LINE,NUMBER
  JP = 2
  LINE = LINE + N
  IF (LINE .LT. 59) RETURN
  NUMBER = NUMBER + 1
  WRITE (6,1000) TITLE,NUMBER
  LINE = 2 + N + NT
  JP = 0
  RETURN
1000 FORMAT('LAFATL P-001 AAASIM ',17A4,' ',A10,2(1XA9),2X'PAGE',14/)
SUBROUTINE PRSEGS(P,ISL)
  PRINTS THE PK AS A FUNCTION OF ASPECT AND IMPACT SPEED TABLES

  THIS SUBROUTINE EXTENSIVELY MODIFIED TO PRINT ASPECT SECTOR
  ANGLES AND PROPERLY LABEL THE TWO CASES FCR WHICH IT
  PRINTS TABLES.
  COMMON /BLOCK3/ XGUN,YGUN,ZGUN
  COMMON /BLOCK4/ IGT,IEM,ICB,ISB,IGL,CIRCLE
  DIMENSION P(32,8),PT(8)
  REAL*8 ANG(8),315-360,000-045,045-090,090-135,135-180,
1,180-225,225-270,270-315,
8888 CONTINUE
  PK = 0.0
  PT(1) = 0.0
  PT(2) = 0.0
  PT(3) = 0.0
  PT(4) = 0.0
  PT(5) = 0.0
  PT(6) = 0.0
  PT(7) = 0.0
  PT(8) = 0.0
111 CALL PAGES(60,JP)
222 IF (ISL.GT.0) WRITE (6,1001) ISL,IGT,IEM,XGUN,YGUN,ZGUN,CIRCLE

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333 IF (ISL.EQ.0) WRITE (6,1030)
444 WRITE(6,1040)
555 DO 2 I=1,32
666 IAZ = 1 + MCD(I,8)
777 IEL = 2 + (I-1)/8
888 PP=0.0
999 DO 3 J=1,8
112 PT(J)=PT(J)+(1.0-PT(J))*P(I,J)
113 PP=PP+(1.0-PP)*P(I,J)
CALL PAGES(1,6,JP)
IF (JP.NE.0) GO TO 2
IF (ISL.GT.0) WRITE (6,1001) ISL, IGT, IEM, XGUN, YGUN, ZGUN, CIRCLE
IF (ISL.EQ.0) WRITE (6,1030)
WRITE (6,1040)
WRITE (6,1041) I, ANG(IAZ), ANG(IEL), (P(I,J), J=1,8), PP
2 CALL PAGES(2,6,JP)
IF (JP.NE.0) GO TO 4
IF (ISL.GT.0) WRITE (6,1001) ISL, IGT, IEM, XGUN, YGUN, ZGUN, CIRCLE
IF (ISL.EQ.0) WRITE (6,1030)
WRITE (6,1040)
4 CONTINUE
889 RETURN
1001 FORMAT(' -PK AS A FUNCTION OF ASPECT SECTOR AND IMPACT SPEED: ',
5X, LOC, 14, 5X, IGT, 12, 5X, IEM, 12, 5X, X, 7, 0, 3X, Y, 7, 0, 3X, Z,
F7.0, C, 4X, RADII, F5.0, 'M. ')
1030 FORMAT(' -TOTAL PK FOR DENSITY CLASS 1 AS A FUNCTION OF ASPECT',
, SECTOR AND IMPACT SPEED',
, 305-457
1040 FORMAT('0 SECTOR 457-610
4X, 1067-1219 TOTAL PK, 610-762 152-305
, 4X, 1067-1219 TOTAL PK, 610-762 762-914 914-1067,
1041 FORMAT(14, 2X, A7, 8F12.7, F14.7)
1042 FORMAT(14, 15X, TOTALS, 8F12.7, F14.7)
END
SUBROUTINE TPLOT(NFPA)
C
C PLOTS X VS. Y AND X VS. Z ON PRINTER FOR EACH FLIGHT PATH.
C NO PLOT WHEN XMAX-XMIN IS LESS THAN 50.
C
COMMON XFPA(1201), YFPA(1201), ZFPA(1201),
BEFPA(1201), AEFPA(1201), PEFPA(1201),
VXFPA(1201), VYFPA(1201), VZFFPA(1201)
2 DIMENSION PLOT(111)
LOGICAL XA
DATA BLANK, YAXIS, XAXIS, POINT/1F, 1F, 1H-, 1H+/
PLOT ONLY ABOUT 50 POINTS WITH "IFPA".
IFPA = MAX0(NFPA/50, 1)

```

```

C      MIN-MAX VALUES OF X, Y, AND Z (ZMIN = 0. BY DEFN.)
      FIND = 1.E99
      XMIN = 1.E99
      YMIN = -1.E99
      XMAX = -1.E99
      YMAX = -1.E99
      ZMAX = 1.E99
      DO 100 I=1,NFPA,IFPA,IFPA
      XMIN = AMINI(XMIN,XFPA(I))
      YMIN = AMINI(YMIN,YFPA(I))
      XMAX = AMAXI(XMAX,XFPA(I))
      YMAX = AMAXI(YMAX,YFPA(I))
      ZMAX = AMAXI(ZMAX,ZFPA(I))
      CHECK X RANGE TO AVOID FUNNY X-Z PLOT.
      XRNG = XMAX - XMIN
      IF (XRNG.GT.49.999) GO TO 110
      CALL PAGES(3,0,I)
      WRITE (6,210)
      RETURN
      X,Y TC METERS PER CHARACTER.
      SCALE = AMAXI(XRNG/11.,(YMAX-YMIN)/7.25)
      DX = D/10.
      DY = D/8.
      CENTER = (XMAX+XMIN)/2. - 55.*DX
      XMIN = (YMAX+YMIN)/2. - 29.*DY
      XMAX = XMIN + 110.*DX
      YMAX = YMIN + 58.*DY
      SCALE Z (NOT TO SAME SCALE AS X-Y).
      DO 120 I=1,100
      D = FLOAT(I*500)
      IF (D.GT. ZMAX) GO TO 130
      CCNTINUE
      DZ = D/10.
      SEE IF AXES SHOULD APPEAR ON PLOT.
      XA = .FALSE.
      NYA (XMIN .LE. 0. .AND. XMAX .GE. 0.) XA = .TRUE.
      IF (YMIN .LE. 0. .AND. YMAX .GE. 0.) NYA = .TRUE.
      IF (XA) NXA = 1 + INT(0.5-XMIN/DX)
      SET 8 LINES PER INCH SPACING ON PRINTER.
      WRITE (6,280)
      SET PAGE CONTROL; PLOT X-Z, THEN X-Y.
      CALL PAGES(56,0,I)
      WRITE (6,220)
      IFPA = 2 * IFPA
      DC 160 K=1,11
      DO 140 I=1,111

```

```

140 PLOT(I) = BLANK
    IF (K.EQ.11) PLOT(I) = XAXIS
    IF (XA.AND. K.NE.11) PLOT(NXA) = YAXIS
    Z = FLOAT(11-K)*DZ
    DO I=1,NFPA,IFPA
    IX = 1 + INT(0.5+(XFPA(I)-XMIN)/DX)
    IF (K.EQ.230) 11 - INT(0.5+ZFPA(I)/DZ) PLOT(IX) = POINT
150 WRITE (6,220)
160 WRITE (6,220)
    WRIFPA = IFPA/2
    DO I=1,59
    DO I=1,111
    PLOT(I) = BLANK
170 IF (K.EQ. NYA) PLOT(I) = XAXIS
    IF (XA) PLOT(NXA) = YAXIS
    IF (YMIN + FLOAT(59-K)*DY
180 DO I=1,NFPA,IFPA
    IX = 1 + INT(0.5+(XFPA(I)-XMIN)/DX)
    IF (K.EQ.2) PLOT(IX) = POINT
    I = MOD(K,2)
190 IF (I.NE.1) WRITE (6,240) PLOT
    IF (I.EQ.1) WRITE (6,260) PLOT, Y
    C PRINT X, AXIS VALUES.
    DC=10.*DX
    DC 200 I=1,12
    PLOT(I) = XMIN + D*FLOAT(I-1)
    WRITE (6,270) (PLOT(I),I=1,12),DX,DY
    C RE-SET PRINTER SPACING BACK TO 6 LINES PER INCH.
    WRITE (6,290)
    RETURN
    C ONEARLY CONSTANT X IN FLIGHT PATH. NO PLOT PRINTED.
210 FORMAT(1X,113(IH#))
220 FORMAT(2H*,111A1,1H*)
230 FORMAT(2H*,111A1,1H*)
240 FORMAT(2H*,111A1,1H*)
250 FORMAT(2H*,111A1,1H*)
260 FORMAT(2H*,111A1,1H*)
270 FORMAT(4X,X(A)=,F8.1,
1 4X,X(D)=,F8.1,
2 4X,X(G)=,F8.1,
3 4X,X(J)=,F8.1,
4 4X,DX =,F8.1,
280 FORMAT(1HT)
290 FGMAT(1HS)

```

```

END
SUBROUTINE RPLANE(T)
  COMPUTES INFORMATICN ABOUT THE POSITION OF THE AIRCRAFT AT TIME =T
  COMMON/BLOCK2/NFPA,TMIN,TMAX,DTFPA
  COMMON/BLOCK3/XGUN,YGUN,ZGUN
  COMMON/NFARM/XA,YA,ZA,RA2,RA,TIME
  COMMON/IGXGYG/IG,XG(8),YG(8)
  COMMON/XFPA(1201),YFPA(1201),VFPA(1201),VZFPA(1201),
  1  COMMON/PFPA(1201),VXFPA(1201),VYFPA(1201),VZFPA(1201),
  *  COMMON/CECM1/IRECM,I,J,GAINJ,IX,XSEC,CALX,PJK,
  *  X,Y,Z,ROL,PIT,FDG,
  *  FGT,FJAM,GJ,SJT,SN
  CALL INTERP((T+TIME)/DTFPA)
  XA=GETVAL(XFPA)-XGUN-XG(IG)
  YA=GETVAL(YFPA)-YGUN-YG(IG)
  ZA=GETVAL(ZFPA)-ZGUN
  RA=XA*XA+YA*YA+ZA*ZA
  RETURN
END
FUNCTION ANGLIM(X)
  LIMITS ANGLES TO PRINCIPAL ANGLES BETWEEN -PI AND +PI
  COMMON /CONSTS/ DEGREE,RADIAN,PI,PI2,QTRPI,SQRT2
  IF(ABS(X)-PI)1,1,2
  1 ANGLIM=X
  2 ANGLIM=X-PI2*FLOAT(IFIX((X+SIGN(PI,X))/PI2))
  RETURN
END
FUNCTION RSHELL(T)
  COMPUTES RANGE TO SHELL AT TIME = T
  COMMON/VASBS/VMUZZ,ASHCON,BSHCON,DQUAD
  DQUAD=1.0+T*(ASHCON+T*BSHCON)
  RSHELL=T*VMUZZ/DQUAD
  RETURN
END
FUNCTION VSHELL(T)
  COMPUTES SPEED OF SHELL AT TIME=T. CAN ONLY BE USED AFTER A CALL
  TO RSHELL AT THE SAME TIME, SINCE RSHELL COMPUTES DQUAD FOR VSHELL
  COMMON/VASBS/VMUZZ,ASHCON,BSHCON,DQUAD

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```

VSHLL=VMUZZ*(1.0-BSHCON*T*T)/(DQUAD*DQUAD)
RETURN
END
SUBROUTINE INTERP(FINT)
SETS CONSTANTS (FRACT, INDEX1, AND INDEX2) FOR TWO PCINT INTER-
POLATION
COMMON/MAGIC/FRACT,INDEX1,INDEX2
INDEX1=FINT
FRACT=FINT-FLOAT(INDEX1)
INDEX1=INDEX1+1
INDEX2=INDEX1+1
RETURN
END
FUNCTION GETVAL(ARRAY)
PERFORMS TWO POINT INTERPOLATION
COMMON/MAGIC/FRACT,INDEX1,INDEX2
DIMENSION ARRAY(1201)
GETVAL=ARRAY(INDEX1)+FRACT*(ARRAY(INDEX2)-ARRAY(INDEX1))
RETURN
END
BLOCK DATA
COMMON /BLOCK1/ ITITLE, XGUN, YGUN, ZGUN, IRMP, ILEN, O1, O8, D1, JJJ, XR, ZT, PZI, IERR
COMMON /BLOCK3/ IGT, IEM, IC8, ISB, IGL, CIRCLE, IF6, IMUL, IP, N, O7, CC, TTRACK, KMODE, PSI, T2
COMMON /BLOCK4/ NRHOS, RHO(9), TROUND(6), THD, MAX(6), PHD, MAX(6), PHMIN(6), PHMAX(6),
COMMON /BLOCK5/ NTINTS, TINTER(10), IVACOM, TTRACK, TRACK1, TRACK2, VELMIN(6), VELMAX(6), RANMIN(6), RANMAX(6), ATLAG(6),
COMMON /BLOCK6/ TREAT, TRAC, VAT(9,5,9), VELMIN(6), VELMAX(6), EPHMAX(6), RMODES(6)
COMMON /BLOCK7/ TFM, MAX1(6), TFM, MAX2(6), RVACON(6), RVBCCN(6), VMUZEL(6)
COMMON /BLOCK8/ TFM, MAX(6), TFM, MAX(6), RVACON(6), RVBCCN(6), RVBCCN(6), VMUZEL(6)
COMMON /BLOCK9/ TFM, MAX(6), TFM, MAX(6), RVACON(6), RVBCCN(6), RVBCCN(6), VMUZEL(6)
COMMON /BLOCKA/ TFM, MAX(6), TFM, MAX(6), RVACON(6), RVBCCN(6), RVBCCN(6), VMUZEL(6)
COMMON /CONSTS/ DEGREE, RADIAN, PI, PI2, QTRPI, SQRT2
COMMON /HEADFO/ LINE, NUMBER
COMMON /BUDGET/ BCACON(6)
COMMON /NEWA/ VY1, VY, ALFA, J, IRMP, ILEN, O1, O8, D1, JJJ, XR, ZT, PZI, IERR
COMMON /NEWB/ Y1, P2, VX, BETA, J, IF6, IMUL, IP, N, O7, CC, TTRACK, KMODE, PSI, T2
COMMON /NEWC/ CP, X2, P1, B2, J, FILE, C4, CFS, TPER, S, JMODE, YTI, T2
COMMON /NEWD/ T33, X1, Z2, B1, VX2, C3, CPK, NR, OUND, IOFF, I, XI, TM, SP, TI
COMMON /NEWF/ Z1, A2, VX1, VZ2, PH1, IF5, IE, OF, YR, IF2, T13, IFILE, Y2, AL
COMMON /NEWG/ V2, VZ1, V, FUZZ, IF9, SC2RJ, SD2RJ, V1, VY2, F, F, J, P, IF7, REFC
COMMON /JAM/O/, SD2RJ/O./
DATA IMUL/O/

```

CC C

CC C


```

CALL REREAD
90C READ (99,900) PK
   FORMAT (F10.7)
910 READ (99,910) YGUN
   FORMAT (F7.0)
   IF (ISW.EQ.IALL) GOTO 600
   IF (ITST.EQ.IBLNK) GOTO 200
   IF (YGUN.NE.0.0) GOTO 150
   CNT=CNT+0.5
   SUM=SUM+PK/2.
   POLD=0.0
   GOTO 500
150 CGNT INUE
   CNT=CNT+1.0
   SUM=SUM+PK
   POLD=PK
   GOTO 500
200 IF (CNT.LE.0.5) GOTO 500
   CNT=CNT-0.5
   SUM=SUM-POLD/2.0
201 AVERG=SUM/CNT
   WRITE (6,800) CNT,AVERG
800 X FORMAT (75X,18H AVERAGE P(KILL) ON,F6.1,20H OFFSET LOCATIONS IS,
   F10.7/)
   SUM=0.0
   CNT=0.0
500 RETURN
600 IF (ITST.NE.IBLNK) GOTO 610
   IF (CNT.NE.0.0) GOTO 201
   GOTO 500
610 SUM=SUM+PK
   CNT=CNT+1.0
   GOTO 500
END
SUBROUTINE JAMER1(PLEN,SDSQ)
SDR = PLEN*0.6826/2.*2.998E8
SDSQ = SDR*SDR
RETURN
END
SUBROUTINE MULPTH(I,REFC,EL,BIAS,SD2)
DIMENSION C(3),S(3),B(3)
DATA
* AK/-0.6931471806/
* ,SQR2/1.414213562/
* ,B/0.0244346,0.0314159,0.0785398/
* ,C/0.0132557,0.0185345,0.0478558/
* ,S/0.00872665,0.0104720,0.0244346/
* BM = B(I)

```

```

CAL = C(I)
SQ = S(I)
DIR = EXP(AK*(SQ/BW)**2)
RL2 = 2*EL*((EL2+SQ)/BW)**2)
RL = EXP(AK*((EL2-SQ)/BW)**2)
DRUIS = (DIR+REFC*RU)**2
DRLIS = (DIR-REFC*RU)**2
DRL2S = (DIR-REFC*RL)**2
DIF1 = (DRUIS+DRLIS)
SUM1 = (DRUIS-DRLIS)
DIF2 = (DRU2S+DRL2S)
SUM2 = (DRUIS-DRLIS)
SIGER1 = DIF1/SUM1
SIGER2 = DIF2/SUM2
ANGERR1 = CAL*SIGER1
ANGERR2 = CAL*SIGER2
PPBY2 = ABS(ANGERR2-ANGERR1)/2.
BIAS = ANGER1+PPBY2
SD2 = PPBY2/SQRT2
SD2 = SD*SD
RETURN
END
SUBROUTINE ECM1
DIMENSION RGDB(3), PRW(3), FREQ(3), IRTYP(4), RNOISE(3)
DIMENSION TABJ(37,37), TABX(37,37)
COMMON /BLOCK1/ ITITLE(10)
COMMON /HEADFO/ LINE, NUMBER
COMMON
/CECML/ IREC, IJ, GAINJ, IX, XSEC, CALX, PJW,
X, Y, Z, ROL, PIT, HDG,
FTGT, FJAM, GJ, SJT, SN
**
NAMELIST /NAML/ RGDB, PRW, FREQ, IRTYP, I, RG, WL, FTGT, FJAM, PJW
DATA
RGDB/40.0, 38.5, 28./
PRW/10500., 175000., 250000./
FREQ/15.1E9, 9.3805E9, 2.838E9/
IRTYP/1, 2, 2, 3/
PI4/12.56637061/
RNOISE/-123.0, -130.6, -132.2/
I = IRTYP(IREC)
RG = 10.**(RGDB(I)/10.)
WL = 2.998E8/FREQ(I)
RN = RNOISE(I)
FTGT = PRW(I)*RG*WL*WL/PI4/PI4

```

C

C

```

FJAM = PJW*RG*WL*WL/PI4/PI4
CALL PAGES(3,0,J,JP)
IF(IJ.EQ.0) WRITE(6,9003) GAINJ
9003 FORMAT(/,/, JAMMER ANTENNA GAIN',F7.3,' DB')
IF(IJ.NE.0) WRITE(6,9004)
9004 FORMAT(/,/, JAMMER TABLE SPECIFIED')
CALL PAGES(3,0,J,JP)
IF(IX.EQ.0) WRITE(6,9005) XSEC
9005 FORMAT(/,/, AIRCRAFT CROSS SECTION ',F9.2,' SQ.METERS')
IF(IX.NE.0) WRITE(6,9006) CALX
9006 * FORMAT(/,/, AIRCRAFT CROSS SECTION TABLE SPECIFIED.',F9.2)
* PRINTED VALUES WILL BE MULTIPLIED BY CALX. CALX=.,F9.2)
IF(IJ.EQ.0) GO TO 1

C
C JAMMER TABLE
CALL TABLR(TABJ,37)
DC 3 I=1,37
DO 3 J=1,37
TABJ(I,J) = 10.**(TABJ(I,J)/10.)
3 CONTINUE
1 GJ = 10.**(GAINJ/10.)
IF(IX.EQ.0) GO TO 2

C
C X-SECTION TABLE
CALL TABLR(TABX,37)
2 RETURN

C
C ENTRY ECM2
NAMELIST/NAM2/ X,Y,Z,ROL,PIT,HDG,
*CX1,CY1,CZ1,CX2,CY2,CZ2,AZ,EL,GAINJ,XSEC,D2,SJ,ST,SJT
*,GJ

C
IF(IX.EQ.0 .AND. IJ.EQ.0) GO TO 5
CALL DIRCOS(X,Y,Z,0.,0.,0.,CX1,CY1,CZ1)
CALL CARROT(CX1,CY1,CZ1,ROL,PIT,HDG,CX2,CY2,CZ2)
CALL RECSPH(CX2,CY2,CZ2,AZ,EL)
IF(IJ.EQ.0) GO TO 6
CALL INTRP(TABJ,AZ,EL,37,GJ)
6 IF(IX.EQ.0) GO TO 5
CALL INTRP(TABX,AZ,EL,37,XSEC)
XSEC = XSEC*CALX
5 D2 = DIST2(X,Y,Z,0.,0.,0.)
SJ = FJAM*GJ/D2

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```

ST = FTGT*XSEC/D2/D2
SN = 10.*ALOG10(ST)-RN
SJ = 10.*ALOG10(SJ/ST)
RETURN
END
SUBROUTINE JAMER2(IRAD,AJS,SDSQ)
DIMENSION AJS(3),SD1(3)
DIMENSION AJS2(4),SD2(4)
DIMENSION AJS3(2),SD3(2)
DIMENSION AJS4(4),SD4(4)
DATA
* N1/3/ -5. 6. 9./
* ,AJS1/ -5. 6. 9./
* ,SD1/ 0.002963, 0.01185, 0.1374 /
DATA
* N2/4/ -2. 6. 10. 16./
* ,AJS2/ -2. 6. 10. 16./
* ,SD2/ 0.002963, 0.009877, 0.08278, 0.1374 /
DATA
* N3/2/ 30./
* ,AJS3/ 15. 30./
* ,SD3/ 0.0018, 0.01441 /
DATA
* N4/4/ -2. 6. 10. 16./
* ,AJS4/ -2. 6. 10. 16./
* ,SD4/ 0.006914, 0.02173, 0.1024, 0.1374 /
GO TO (1,2,3,4),IRAD
1 CALL INT2(N1,AJS1,SD1,AJS,SD)
2 CALL INT2(N2,AJS2,SD2,AJS,SD)
3 CALL INT2(N3,AJS3,SD3,AJS,SD)
4 CALL INT2(N4,AJS4,SD4,AJS,SD)
5 SDSQ = SD*SD
RETURN
END
SUBROUTINE INT2(NVAL,X,Y,XVAL,YVAL)
DIMENSION X(NVAL),Y(NVAL)
YVAL = Y(1)
IF(X(1)-XVAL) 4,4,3
DC 1 I=1,NVAL
IF(X(I)-XVAL) 1,1,2
1 CONTINUE
YVAL = Y(NVAL)
GO TO 3
2 YVAL = Y(I-1) + (Y(I)-Y(I-1)) / (X(I)-X(I-1)) * (XVAL-X(I-1))
3 RETURN

```

```

C SUBROUTINE TABLR(TABX, IDIM)
C SUBROUTINE TO READ AND PRINT A TABLE CF UP TO 37 X 37 ELEMENTS
C THE PROGRAM PROVIDES A DEFAULT VALUE FOR ELEMENTS OUTSIDE THE
C DEFINED TABLE.
C INPUTS ARE:
C CARD VARIABLE FORMAT DEFINITION IDENTIFICATION
C 1 INAME 8A10 NO. OF AZ ELEMENTS
C 2 NAZ I5 (ASSUMING ELEMENTS 1 CORRESPONDS TO AZ=0)
C 3+ TABX(IDIM,1) 8F10.0 DATA TABLE NO. OF EL ELEMENTS, EL GCES-ELEND TO +ELEND
C 4 NEL I5 MAXIMUM ENTRY EL(DEG)
C 5 ELEND F10.4 MAXIMUM ENTRY AZ(DEG)
C 6 AZEND F10.4 DEFAULT VALUE
C 7 DEFALT F10.4 DEFAULT TABLE
C 8 COMMON/TABLES/ELO,DELAZ,DELEL,JFL
C 9 DIMENSION TABX(IDIM, IDIM), INAME(8)
C 10 DATA LE, LZ/2HEL, 2HAZ/
C 11 READ(5, 99) INAME
C 12 READ(5, 99) NAZ, NEL, ELEND, AZEND, DEFALT
C 13 FCRMAT(215, 3F10.4)
C 14 NOTE IMPLIED INCREMENT
C 15 DELAZ=AZEND/(INAZ-1)
C 16 DELEL=(2.*ELEND)/(NEL-1)
C 17 LOCATE FIRST ELEVATION ENTRY ETC
C 18 JEL=(IDIM-NEL)/2 +1
C 19 MEL=JEL-NEL-1
C 20 ELO=-ELEND
C 21 WRITE(6, 101) INAME, NAZ, AZO, AZEND, DELAZ,
C 22 LE, LZ, NEL, ELO, ELEND, DELEL, LE, DEFALT
C 23 101 FCRMAT(1, 1) TABLE DATA, /, 1X, 8A10, /, 2(1X, 15,
C 24 1, ELEMENTS FROM, F10.2, /, TO, F10.2, /, BY,
C 25 2F10.2, 2X, A2, /), ELSEWHERE TABLE IS, F10.2}
C 26 INSERT DEFAULT
C 27 MAZ=(180./DELAZ)+1
C 28 IF ((MAZ.GT.37).OR.(MEL.GT.37)) GO TO 999
C 29 DO 8 I=1, MAZ
C 30 DO 8 J=1, IDIM
C 31 TABX(I, J)=DEFALT
C 32 READ DC 14 I=JEL, MEL
C 33 READ(5, 102) (TABX(I, J), J=1, NAZ)
C 34 CONTINUE
C 35 102 FORMAT(8F10.0)

```

```

ICELAZ=DELAZ
NPAGE=NAZ/13+1
DO 20 LP=1, NPAGE
JH1=(LP-1)*13+IDELAZ
JF2=JH1+12*IDELAZ
KH1=(LP-1)*13+1
KF2=KH1+12
IF(LP.EQ.NPAGE)KH2=MAZ
IF(LP.EC.NPAGE)JH2=180
WRITE(6,106)(JH,JH=JF1,JH2,IDE LAZ),
ELEV,1319)
106 FORMAT(11 RCS MATRIX,%,
JELO=MAX(XO(JEL-1,1),
MEL=MINO(MEL+1,1DIM)
ELPT=ELO-DELEL#{JEL-JELO)
C PRINT TABLE
DO 15 J=JELO,MELO
WRITE(6,104)ELPT,(TABX(J,K),K=KF1,KH2)
104 FORMAT(1X,F7.1,2X,13FS.2)
15 CONTINUE
20 CONTINUE
C CONVERT TO RADIANS
DELEL=DELEL*CDTR
DELAZ=DELAZ*CDTR
ELC=ELO*CDTR
RETURN
CONTINUE
999 WRITE(6,105)
105 FORMAT(1 >>> ERRCR IN INPUT <<<<')
END
SUBROUTINE DIRCOS(X1,Y1,Z1,X2,Y2,Z2,COSA,COSB,COSG)
XC = X2-X1
YD = Y2-Y1
ZD = Z2-Z1
D = SQRT(XD*XD+YD*YD+ZD*ZD)
COSA = XD/D
COSB = YD/D
COSG = ZD/D
RETURN
END
SUBROUTINE CARROT(X1,Y1,Z1,ROL,PIT,HDG,X2,Y2,Z2)
C CARROT
C HEAD X = X1*COS(HDG) + Y1*SIN(HDG)
C HEAD Y = -X1*SIN(HDG) + Y1*COS(HDG)
C HEAD Z =
C PITCH

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C ROLL
XX = X*COS(PIT)
YY = X*SIN(PIT)
ZZ = X*SIN(PIT)
X2 = XX
Y2 = -YY*SIN(ROL)
Z2 = YY*COS(ROL)
      + ZZ*COS(ROL)
      + ZZ*SIN(ROL)
      -Z*SIN(PIT)
      Z*COS(PIT)
SUBROUTINE RECSPH(X,Y,Z,PHI,THE)
THE = ARCCOS(Z/SQRT(X*X+Y*Y+Z*Z))
THE=THE-1.5708
PHI=0.0
SB=SQRT(X*X+Y*Y)
IF(SB.NE.C.0)PHI=X/SB
PHI=ARCCOS(PHI)
PHI=ABS(PHI)
RETURN
END
SUBROUTINE INTRP(TAB,AZ,EL,NVAL,VALUE)
COMMON/TABLES/ELO,DELAZ,DELEL,JEL
DIMENSION TAB (NVAL,NVAL)
A=EL
E=EL
AAZ=A/DELAZ+1.
IAZ=AAZ
IEL=(E-ELO)/DELEL+JEL
IEL=EEL
IAZ=MIN0(MAX0(IAZ,1),36)
IEL=MIN0(MAX0(IEL,1),36)
V1=TAB( IEL, IAZ+1)
V2=TAB( IEL+1, IAZ)
V3=TAB( IEL+1, IAZ+1)
V4=TAB( IEL, IAZ)
S=AAZ-I*IAZ
V12=V1+(V2-V1)*S
V34=V3+(V4-V3)*S
S=EEL-IEL
VALUE=V12+(V34-V12)*S
RETURN
END
FUNCTION DIST2(X1,Y1,Z1,X2,Y2,Z2)
XD = X2-X1
YD = Y2-Y1
ZD = Z2-Z1
DIST2 = XD*XD+YD*YD+ZD*ZD
RETURN
END

```

APPENDIX J

P001 PROGRAM LISTING (CDC)

```

PROGRAM P7022(INPUT,OUTPUT,TAPE5=INPUT,TAPE6=OUTPUT,TAPE7,TAPE8,
1 TAPE9,TAPE4,TAPE11)
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
AAASIM -- MULTIPLE GUN ANTI-AIRCRAFT ARTILLERY SIMULATION
WRITTEN BY THOMAS D. MCMURCHIE AND JAMES O. SEVERSON
AIRFORCE ARMAMENT LABORATORY (AFATL-DLYS) EGLIN AFB, FLORIDA
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
PROGRAM RECEIVED 25 OCT 72
CONVERTED AND MAINTAINED BY ASD/XROA
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
COMMON/BLOCK1/TITLE(10)
COMMON/BLOCK2/NFPA,TMIN,TMAX,DTFPA
COMMON/BLOCK3/XGUN,YGUN,ZGUN
COMMON/BLOCK4/IGT,IEM,ICB,ISB,IGL,CIRCLE
COMMON/BLOCK5/NRHCS,RHO(9)
COMMON/BLOCK6/NTINTS,TINTER(10)
COMMON/BLOCK7/IVACOM(8),VAT(9,5,5)
COMMON/BLOCK8/TREACT,TRACK1,TRACK2,THDMAX(6),PHIMIN(6),PHIMAX(6),
1 ATLAG(6),VELMAX(6),RANMIN(6),RANMAX(6),
2 VELMIN(6),ETHMAX(6),RPHMAX(6),RMODES(6)
COMMON/BLOCK9/TROUND(6),THDMAX(6),RANMIN(6),RANMAX(6),
COMMON/BLOCK10/TFMAX2(6),RVBACON(6),VMUZEL(6)
COMMON/CONSTS/DEGREE,RADIAN,PI,PI2,QTRPI,SQRT2
COMMON/HEADFO/LINE,NUMBER
COMMON/NFPARM/XA,YA,ZA,RA2,RA,TIME
COMMON/IGXGYG/IG,XG(8),YG(8)
COMMON/MAGIC /FRACT,INDEX1,INDEX2
COMMON/VASASBS/VMUZZ,ASHCON,BSHCON,DQUAD
COMMON/BUDGET/BDACCN(6)
COMMON XFFPA(1201),VXFFPA(1201),VFPPA(1201),BFPPA(1201),AFPPA(1201),
1 CCOMMON/CECM1/IREC,M,I,J,GAINJ,IX,XSEC,CALX,PJH,
* X,Y,Z,ROL,PIT,HOG,
* FIGT,FJAN,GJ,SJT,SN
DIMENSION ICARD(8),TEMP(16,6),PTOTIF(10),PTCTTI(10)
DIMENSION PKTDC(9),PKTIDC(10,9),PKTFDC(10,9)
DIMENSION SPKT(8,4,8),SPKT2(32,8),SPKTOT(32,8),IPRINT(6),IFLAGS(4)
EQUIVALENCE (SPKT2(1,1),SPKT(1,1,1),TEMP(1,1))
SOME EQUIVALENCES REMOVED BY B.E.E. BY CARRYING TWO VARIABLES
THROUGH PROGRAM WITH SAME VALUES FOR CLARITY.

```

C

DATA IFILE/9999/,JFILE/9999/,INUNIT/5/,FUZZ/0.000000001/
DATA ESVPCT/0.10/
DATA X2/0.0/,Y2/0.0/,Z2/0.0/,V2/0.0/,P2/0.0/,A2/0.0/,B2/0.0/
DATA VX2/0.0/,VY2/0.0/,VZ2/0.0/
DATA IF2,IF5,IF6,IF7,IF9,IFLAGS/1,1,1,1,1,0,0,0,0/
DATA IOFF/2HDF/
DATA IMUL/0/
DATA IJAM/0/,SD2RJ/0./,SD2J/0./

DATA ILOOP/-1/
CALL FTNBNIN(1,0,CPU)
CALL SECOND(CPU)
CALL TIMREM(CIO)
CALL DATE(I,TITLE(9))
ASSIGN 9996 TO IERR

CC
C I I I I I N N I I I I I T T T T I I I I I A A L L I I I I I Z Z Z Z Z E E E E E C
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READ(INUNIT,1050) ISW,AMASK
AMASK1=AMASK*57.3
PRINT 38,AMASK1
FORMAT(5X,25HMASK ANGLE FOR THIS RUN =,F6.3,5H DEG.)
FORMAT(A2.8X,F10.5)

38 95 ISL=1
96 IPUL = 0
94 IJAM = 0
SD2RJ = 0
DC 96 J=1,9
DC 96 J=1,10
PKTIDC(I,J)=0.0
PKTFDC(I,J)=0.0
DC 94 I=1,32
DC 94 J=1,8
SPKTOT(I,J)=0.0
REWIND 7
ASSIGN 9997 TO IEOP
GO TO 15

CC
C I I I I N N P P P P U U T T T T T C
C


```

IF (EOF(9).NE.0.) GO TO IEOF,(4,12,40,70,83,5557,9599)
IF (IOCHEC(9).NE.0.) GO TO IERR,(3,28,9996)
GO TO 3
4 IF(JMODE.GT.IFILE)GO TO 2
  ASSIGN 28 TO IERR
  READ (9)
28 IF (EOF(9).NE.0.) GO TO IEOF,(4,12,40,70,83,9997,9999)
  IF (IOCHEC(9).NE.0.) GO TO IERR,(3,28,9996)
  ASSIGN 9996 TO IERR
5 JMODE=6
  T1=T2
  X1=X2
  Y1=Y2
  Z1=Z2
  P1=P2
  A1=A2
  B1=B2
  V1=V2
  VX1=VX2
  VY1=VY2
  VZ1=VZ2
6 IF(JMODE)6,6,7
  READ(INUNIT,1000)T2,X,Y,Z,VX,VY,VZ2,B2,A2,P2
  IF (EOF(INUNIT).NE.0.) GO TO IEOF,(4,12,40,70,83,9997,9999)
  IF (IOCHEC(INUNIT).NE.0.) GO TO IERR,(3,28,9996)
  B2=B2*RADIAN
  A2=A2*RADIAN
  P2=P2*RADIAN
  V2=SQRT(VX*VX+VY*VY+VZ2*VZ2)
  GO TO 9
7 JMODE=JMODE+1
  IF(JMODE.LE.6)GO TO 8
  JMODE=1
  READ(9) TEMP
  IF (EOF(9).NE.0.) GO TO IEOF,(4,12,40,70,83,9597,9999)
  IF (IOCHEC(9).NE.0.) GO TO IERR,(3,28,9996)
8 T2 =TEMP( 1,JMODE)
  X =TEMP( 2,JMODE)
  Y =TEMP( 3,JMODE)
  Z =TEMP( 4,JMODE)
  VX =TEMP( 5,JMODE)
  VY =TEMP( 6,JMODE)
  VZ2 =TEMP( 7,JMODE)
  B2 =TEMP(11,JMODE)
  A2 =TEMP(13,JMODE)
  P2 =TEMP(15,JMODE)

```

```

9 X2=XT+(X-XR)*CP+(Y-YR)*SP
  Y2=YT+(Y-YR)*CP-(X-XR)*SP
  Z2=ZT+Z
  VX2=VX*CP+VY*SP
  VY2=VY*CP-VX*SP
  B2=ANGLIM(B2-PZ1)
  T2=T2-TMIN
10 IF(TM.GT.T2)GO TO 5
   F=(TM-T1)/(T2-T1)
   NFPA=NFPA+1
   XFPA(NFPA)=X1+F*(X2-X1)
   YFPA(NFPA)=Y1+F*(Y2-Y1)
   ZFPA(NFPA)=Z1+F*(Z2-Z1)
   VXFPA(NFPA)=VX1+F*(VX2-VX1)
   VYFPA(NFPA)=VY1+F*(VY2-VY1)
   VZFPANFPA)=VZ1+F*(VZ2-VZ1)
   V=V1+F*(V2-V1)
   BFPA(NFPA)=ANGLIM(B1+F*ANGLIM(B2-B1))
   AFPA(NFPA)=A1+F*(A2-A1)
   PFPA(NFPA)=ANGLIM(P1+F*ANGLIM(P2-P1))
   K=K-1
34 IF(K)34,34,35
   K=KMODE
   PHI=DEGREE*PFPA(NFPA)
   BETA=DEGREE*BFPA(NFPA)
   ALFA=DEGREE*AFPA(NFPA)
   CALL PAGES(1,2,JP)
   IF(JP.EQ.0)WRITE(6,1003)
   WRITE(6,1004)TM,V,VXFPA(NFPA),YFPA(NFPA),ZFPA(NFPA),
     BETA,ALFA,PHI
1  TM=DTFPA#FLCAT(NFPA)
   IF(TM.LE.TMAX)GO TO 10
   TMAX=TM-DTFPA-FUZZ
   CALL PAGES(4,0,JP)
   WRITE(6,1005)XR,YR,XT,YT,PSI,ZT
   CALL TPLTOT(NFPA)
   IF(JMODE.GT.0)OR.INUNIT.EQ.8)GO TO 12
   ASSIGN 12 TO IEOP
13 READ(INUNIT,1000)
   IF(EOF(INUNIT).NE.0.)GO TO IEOP,(4,12,40,70,83,9997,9999)
   IF(IOCHEC(INUNIT).NE.0.)GO TO IEAR,(3,28,55,56)
   GO TO 13
C DECODE DATA BLOCK 03 -- WEAPON LOCATION
C 103 DECODE(30,1008,ICARD)XGUN,YGUN,ZGUN
C GO TO 12
C

```

```

C LECODE DATA BLOCK 04 -- WEAPON TYPE, MODE, NUMBER OF BARRELS (CYCLIC
C AND SIMULTANEOUS), NUMBER OF WEAPONS PER
C LOCATION, RADIUS OF CIRCLE OF WEAPON COMPLEX
104 DECODE(14,1009,ICARD)IGT,IEM,ICB,ISB,IGL,CIRCLE
61 XG(1)=0.0
   YG(1)=0.0
   GO TO 12
66 CO 16 I=1,IGL
   F=PI2*FLOAT(I)/FLOAT(IGL)
   XG(I)=CIRCLE#COS(F)
   YG(I)=CIRCLE#SIN(F)
   GO TO 12

C DECODE DATA BLOCK 05 -- WEAPON DENSITY FACTORS
105 DECODE(78,1C15,ICARD)IF5,NRHOS,(RPO(I),I=1,NRHOS)
   GO TO 12

C DECODE DATA BLOCK 06 -- PK ACCRUAL TIME INTERVALS
106 DECODE(78,1C15,ICARD)IF6,NTINTS,(TINTER(I),I=1,NTINTS)
   NTINTS=NTINTS+1
   TINTER(NTINTS)=999.99
   GO TO 12

C DECODE DATA BLOCK 07 -- AIRCRAFT VULNERABLE AREAS
107 DC 11 I=1,8
   11 IVACOM(I)=ICARD(I)
   READ(INUNIT,1000)(VAT(1,1,K),K=2,9)
   IF (EOF(INUNIT).NE.0.) GO TO IEOF,(4,12,40,70,83,9997,9999)
   IF (IOCHEC(INUNIT).NE.0.) GO TO IERR,(3,28,5996)
   DC 88 J=2,4
   DO 89 I=1,8
   READ (INUNIT,1000) (VAT(I,J,K),K=2,9)
   IF (EOF(INUNIT).NE.0.) GO TO IEOF,(4,12,40,70,83,9997,9999)
   IF (IOCHEC(INUNIT).NE.0.) GO TO IERR,(3,28,5996)
   89 CONTINUE
   DC 88 K=2,9
   88 VAT(9,J,K)=VAT(1,J,K)
   READ(INUNIT,1000)(VAT(1,5,K),K=2,9)
   IF (EOF(INUNIT).NE.0.) GO TO IEOF,(4,12,40,70,83,9997,9999)
   IF (IOCHEC(INUNIT).NE.0.) GO TO IERR,(3,28,5996)
   DC 18 K=2,9
   DO 18 I=2,9
   VAT(I,5,K)=VAT(1,5,K)

```

```

18 VAT(I,1,K)=VAT(1,1,K)
   IF7=1
   GO TO 12
C
C   DECODE DATA BLOCK 08 -- WEAPON REACTION AND TRACK TIMES
C
108 DECODE(30,1008,ICARD)TREAT,TRACK1,TRACK2
   GO TO 12
C
C   DECODE DATA BLOCK 09 -- WEAPON PARAMETERS
C
109 DECODE(78,1008,ICARD)TROUND(IGT),THDMAX(IGT),PHDMAX(IGT),
1   PHIMIN(IGT),PHIMAX(IGT),VELMIN(IGT),VELMAX(IGT),
2   RANMIN(IGT),RANMAX(IGT)
   READ(INUNIT,1000)ATLAG(IGT),ETHMAX(IGT),EPHMAX(IGT),RMODES(IGT)
   IF (EOF(INUNIT).NE.0.) GO TO IEOF,(4,12,40,70,83,9997,9999)
   IF (IOCHECK(INUNIT).NE.0.) GO TO IERR,(3,28,5956)
   IF9=1
   GO TO 12
C
C   DECODE DATA BLOCK 10 -- SHELL PARAMETERS
C
110 DECODE(46,1008,ICARD)TFMAX1(IGT),TFMAX2(IGT),RVACON(IGT),
1   RVBCON(IGT),VMUZEL(IGT)
   IF9=1
   GO TO 12
C
C   DECODE DATA BLOCK 11 -- INPUT OPTION (CARD/TAPE)
C
111 DECODE(4,1028,ICARD)I
81  INUNIT=5
82  INUNIT=8
   ASSIGN 83 TC IEOF
   IF(I.GT.JFILE)GO TO 84
   REWIND 8
   JFILE=1
83  IF(I-JFILE)12,12,84
84  JFILE=JFILE+1
85  READ(8,1000)
   IF (EOF(8).NE.0.) GO TO 83
   IF (IOCHECK(8).NE.0.) GO TO IERR,(3,28,9996)
   GO TO 85
C
C   DECODE DATA BLOCK 13 --- LOW ALTITUDE RADAR MULTIPATH EFFECT
C
113 DECODE(18,1038,ICARD) IMUL,IRMP,REFC

```

```

IF (IMUL.EQ.0) GO TO 12
CALL PAGES(12,0,JP)
WRITE(6,1098) IRMP,REFC
FORMAT(///,*,I5,/,*,F6.3,///)
1098 Q* IRMP =
1* REFC =
GO TO 12

C
C DECODE DATA BLOCK 14 --- ECM
C
114 SD2RJ = 0.
SC2RJ = 0.
DECODE(78,1041,ICARD) IJAM,IP,IJ,GAINJ,PJW,PLEN,IX,XSEC,CALX,
IRECM,SJTMAX
* IF (IJAM.EQ.0) GO TO 12
CALL PAGES(28,0,JP)
WRITE(6,1096) IP,IJ,GAINJ,PJW,PLEN,IX,XSEC,CALX,IRECM,SJTMAX
FORMAT(///,*,I5,/,*,F7.2,/,*,F9.2,/,*,E12.6,/,*,I5,/,*,I5,/,*,F7.2,/,*,F9.2,/,*,E12.6,/,*,I5,/,*,I5,/,*,F10.3,/,*,F10.3,/,*,I5,/,*,I5,/,*,F7.2,///)
1096 Q* IP =
1* IJ =
2* GAINJ(DB) =
3* PJW(W) =
4* PLEN(S) =
5* IX =
6* XSEC (SQM) =
7* CALX =
8* IRECM =
9* SJTMAX(DB) =
CALL ECM1
IF (IRECM.EQ.3) CALL JAMER1(PLEN,SC2RJM)
IF (IRECM.EQ.4) CALL JAMER1(PLEN,SD2RJM)
GO TO 12

C
C DECODE DATA BLOCK 12 -- PRINT OPTIONS FOR OUTPUT FORMAT - ALSO
C SIGNALS FOR RUN TO BEGIN
C
112 DECODE(6,1016,ICARD)IPRINT
C
PRINT DATA BLOCKS 6, 7, 9, AND 10 (IF THEY CHANGE)
"IF2" IS USED TO SET LINE COUNT TO PROPER VALUE. INPUT AND
OUTPUT PRINT CUT START A NEW PAGE FOR EACH "12" CARD.
IF (IF2.EQ.0) LINE = 66
IF2=0
IF (IJAM.EQ.0) GO TO 48
IF (IP.EQ.0) GO TO 48
WRITE(11,1043) ITITLE

```



```

14 DO I=1,NTINTS
PTOTFF(I)=0.0
27 DO J=1,32
DC 27 J=1,8
SPK2(I,J)=C.0
TPERS=TROUND(IGT)/FLOAT(IGL*ICB)
IF(IEM.GT.1)GO TO 90
TFMAX=TFMAX1(IGT)
TTRACK=TRACK1
GC TO 91
TFMAX=TFMAX2(IGT)
TTRACK=TRACK2
VXES=VYES=VZES=0.0
90 TIME=TM IN+TTRACK
TFIRE=TFIRE.GT.TMAX)GO TO 69
IF(TFMAX=1.2*TFMAX)
ATLCON=0.125/ATLAG(IGT)
EMDTA=EXP(-0.512*ATLCON)
ETMAX=ETHMAX(IGT)
EPMUZZ=VMUZZEL(IGT)
TDCMAX=THDMAX(IGT)
PCMAX=PHDMAX(IGT)
PHMIN=PHMIN(IGT)
PFMAX=PHIMAX(IGT)
ASHCON=RVACCN(IGT)
BSHCON=RVBCCN(IGT)
RSMODE=RMODES(IGT)
RMIN =RANMIN(IGT)
RMAX =RELMAX(IGT)
VMIN=VELMIN(IGT)
VMAX=VELMAX(IGT)
IG=1
LINE=66
LIF (IPRINT(6).LE.0) GO TO 60
CALL PAGES(5,0,JP)
WRITE (6,1013) ISL,IGT,IEM,XGUN,YGUN,ZGUN,CIRCLE

```

```

CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
CC      TTTT  RRRR  A  A  C  C  C  C  K  K  K  K
CC      T      RRRR  A  A  C  C  C  C  K  K  K  K
CC      T      R  R  A  A  A  A  C  C  C  C  K  K  K  K
CC      T      R  R  A  A  A  A  C  C  C  C  K  K  K  K
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC

```

C C COMPUTE ACTUAL AIRCRAFT PARAMETERS AT FIRE TIME FOR USE IN ERROR EQS

```
60 CALL INTERP(TIME/DTFPA)
X=GETVAL(XFPA)-XGUN
Y=GETVAL(YFPA)-YGUN
Z=GETVAL(ZFPA)-ZGUN
VX=GETVAL(VXFPA)
VY=GETVAL(VYFPA)
VZ=GETVAL(VZFPA)
ILOOP = ILOOP+1
ROT = GETVAL(PFFPA)
PHIT = GETVAL(AFFPA)
HCG = GETVAL(BFPA)
G2=X*X+Y*Y
G=SQRT(G2)
THET=ATAN2(Y,X)
PHIT=ATAN2(Z,G)
```

C C C C C

IF MULTIPATH HAS BEEN SPECIFIED (IMUL=1) AND THE ELEVATION ANGLE IS MEASURED BY RADAR (IOEM=3) COMPUTE THE ELEVATION TRACKING BIAS (PBMP), VARIANCE (SP2MP), AND APPARENT TARGET ALTITUDE (Z).

```
SP2MP = 0.
PBMP = 0.
IF (IMUL.NE.1 .OR. IOEM.NE.3) GO TO 30
CALL MULTPTH(IRMP,REFC,PHIT,PBMP,SP2MP)
PHIT = PHIT+PBMP
Z = G*TAN(PHIT)
30 CONTINUE
R = SQRT(G2+Z*Z)
RC=(X*VX+Y*VY+Z*VZ)/R
TD=(X*VY-Y*VX)/G2
PD=(VZ-Z*RD/R)/G
IF(TIME.GT.TMIN) GO TO 58
RCD=TD*PDD=0.0
RANS=R
THES=THET
PHIS=PHIT
ERAN2=ERAN3=ERAN4=ETHE2=ETHE3=ETHE4=EPHI2=EPHI3=EPHI4=0.0
GO TO 59
```

58

```
RDD=(RD-RDS)/.064
TDD=(TD-TDS)/.064
PDD=(PD-PDS)/.064
```

C C

(STORE PREVIOUSLY OBSERVED MEAN TRACKING ERRORS FOR USE IN MEAN TRACKING ERROR EQUATIONS)

```

C      59  ERAN1=ERAN2
          ERAN2=ERAN3
          ERAN3=ERAN4
          ERAN4=R-RANS
          ERAN=ERAN1+.71875*(ERAN2-ERAN1)
          ETHE1=ETHE2
          ETHE2=ETHE3
          ETHE3=ETHE4
          ETHE4=(G/R)*ANGLIM(THET-THES)
          EPHI1=EPHI2
          EPHI2=EPHI3
          EPHI3=EPHI4
          EPHI4=EPHI1-PHIS
          EPHI=EPHI1+.71875*(EPHI2-EPHI1)
          CHECK MASK ANGLE
          IF(PHIT.LE.AMASK) TFIRE= TIME+ TREAT+TTRACK
          (SKIP FIRE ATTEMPT IF INSUFFICIENT TRACKING TO FIRE)
          IF(TIME.LE.TFIRE)GO TO 62
          CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
          C      FFFF IIIII RRRR EEEEE
          C      F    I    R    E
          C      FFFF I    RRRR EEEE
          C      F    I    R    E
          C      F    IIII R    EEEEE
          CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
          VG=SQRT(VX*VX+VY*VY)
          V =SQRT(VG*VG+VZ*VZ)
          PSI=ANGLIM(PFPA(INDEX1))+FRACT*ANGLIM(PFPA(INDEX2))-PFPA(INDEX1))
          (SKIP FIRE ATTEMPT IF MAX ALLOWED TRACKING ERROR IS EXCEEDED)
          C      54  IF(ABS(ETHE4).GT.ETMAX)GO TO 64
          IF(ABS(EPHI4).GT.EPMAX)GO TO 64
          IF(IOEM.GT.1)GO TO 56
          (LIMIT INPUT RANGE ESTIMATE)
          RC=AMAX1(RMIN,AMIN1(RMAX,RANS-0.575*RD))
          COMPUTE MEAN ASSUMED TIME OF FIRE AIRCRAFT POSITION (MECHANICAL
          COMPUTATION)
          CCCCC

```

XF=RC*CTBCPB-XG(IG)
YF=RC*STBCPB-YG(IG)
ZF=RC*SPB
GF=SQRT(XF*XF+YF*YF)
RF=SQRT(GF*GF+ZF*ZF)

C
C
C

(SET UP MATRIX T, THE TRANSFORMATION BETWEEN THE LINE OF SIGHT SYSTEM AND THE FALSE HORIZON SYSTEM)

T22=CT=X/G
T1=Y/G
T33=CP=G/R
T13=SP=Z/R
CS=COS(PSI)
SS=SIN(PSI)
CA=VG/V
SA=VZ/V
CG=VX/VG
SG=VY/VG
T11=CT*CP
T12=ST*CP
T21=-ST
T31=-CT*SP
T32=-ST*SP

C
C

(SET UP FALSE HORIZON SYSTEM VELOCITY COMPONENTS)

VXP=T11*VX+T12*VY+T13*VZ
VYP=T21*VX+T22*VY
VZP=T31*VX+T32*VY+T33*VZ
VGP=SQRT(VXP*VXP+VYP*VYP)
CAP=VGP/V
SAP=VZP/V
CBP=VXP/VGP
SBP=VYP/VGP

C
C

(SET UP UNIT VECTOR OUT LEFT WING OF AIRCRAFT)

UX=-SA*CG*SS-SG*CS
UY=CG*CS-SA*SG*SS
UZ=CA*SS
UZP=T31*UX+T32*UY+T33*UZ
IF(VXP) 31, 32 31
CSP=-T11*UX-T12*UY-T13*UZ
GC TO 33
31 CSP=(VGP*(T21*UX+T22*UY)+UZP*VZP*SBP)/VXP
33 SSP=UZP/CAP

C

CC (COMPUTE FALSE HORIZON SYSTEM MEAN AND STANCRD DEVIATION CF
C ERROR IN DIVE AND COURSE ANGLE ESTIMATES)

EMAP=SAP*(.3196*ABS(CBP)-.1859*ABS(SBP))
ESAP=.04712+.08063*ABS(SAP)*(1.0+1.16*ABS(CBP))
EMBP=.4060*CAP*SBP*CBP
ESBP=(.1670-.08098*ABS((CBP*CBP-SBP*SBP)*CSP)+
1.09006*ABS(SBP*SSP*CSP))/CAP

1 SEMAP=\$IN(EMAP)
CEMAP=\$COS(EMAP)
SEMBP=\$SIN(EMBP)
CEMBP=\$COS(EMBP)
THE NEXT CARD ELIMINATES AN EQUIVALENCE BY USING A DOUBLE
REPLACEMENT.

A33=SABP=SAP*CEMAP+CAP*SEMAP
CABP=CAP*CEMAP-SAP*SEMAP
SBBP=SBP*CEMBP+CBP*SEMBP
CBBP=CBP*CEMBP-SBP*SEMB

(LIMIT VELOCITY ASSESSMENT (MECHANICAL COMPUTERS))

VBP=AMIN,(VMAX,AMAX1(VMIN,V))
ESVP=ESVPCT*V

CC SET UP THE ELEMENTS CF THE MATRIX A.
C A CONTAINS THE PARTIALS OF VXE,VYE,VZE W.R.T. ALPHA,EETA,SPEED

A31=CABP*CBBP
A32=CABP*SBBP
A21=-VBP*A32
A22=VBP*A31
A11=-VBP*SABP*CBBP
A12=-VBP*SABP*SBBP
A13=VBP*CAP

CC COMPUTE MEAN ESTIMATED VELOCITY COMPONENTS (MECHANICAL
C COMPUTATION)

VXE=(A31*T11+A32*T21+A33*T31)*VBP
VYE=(A31*T12+A32*T22+A33*T32)*VBP
VZE=(A31*T13
+A33*T33)*VBP
GO TO 63

CC COMPUTE MEAN ASSUMED TIME OF FIRE AIRCRAFT POSITION (ELECTRONIC
C COMPUTATION)

56 XF=RANS*CTBCPB
YF=RANS*STBCPB

ZF=RANS*SPB
GF=RANS*CPB
RF=RANS

C C ITERATION TO DETERMINE MEAN THEORETICAL INTERCEPT POINT

63 RS=0.0
VS=VMUZZ
T=0.0
21 XE=XF+VXE*T
ZE=ZF+VZE*T
XE2=XE*XE
YE2=YE*YE
ZE2=ZE*ZE
GE2=XE2+YE2
RE2=GE2+ZE2
RC=RE-RS
IF(RC.LT.1.0)GO TO 22
IF(VS-(XE*VXE+YE*VYE+ZE*VZE))/RE
IF(VD.LE.1.0)GO TO 64
IF(T.GT.TFMAX)GO TO 64
RS=RSHELL(T)
VS = VSHELL(T)
GO TO 21
22 T2=T*T

C C CHANGE 22 JAN 76
C C IF INTERCEPT POINT BELOW MASK*****SKIP FIRE

GE=SQRT(GE2)
IF(ATAN2(ZE,GE).LE.AMASK)GO TC 64

C C ITERATION TO DETERMINE ACTUAL INTERCEPT POSITION, RANGE, AND TIME

TU=AMIN1(TFMAX,TMAX-TIME)
23 CALL RPLANE(TU)
IF(RSHELL(TU).GT.RA)GO TO 24

C C (SKIP FIRE ATTEMPT IF SHELL CANNOT CATCH AIRCRAFT)
C C IF((XA*GETVAL(VXFPA)+YA*GETVAL(VYFPA)+ZA*GETVAL(VZFFPA))/RA.LT.
1 TU=TU-1.0
IF(TU)64,64,23
24 TL=0.0

```

25 T=0.5*(TL+TU)
   CALL RPLANE(T)
   RC=RA-RSHELL(T)
   IF(RC.GT.1.0)GO TO 26
   TU=T
   IF(RC.GT.-1.0)GO TO 37
   GC TO 25
26 TL=T
   GC TO 25
37 Q0=VS*RE-XE*VXE-YE*VYE-ZE*VZE
   Q1=(VZE-VS*7E/RE)/Q0
   Q2=(XE*VYE-YE*VXE)/Q0
   C COMPUTE THE PARTIAL DERIVATIVES OF BIG THETA
   C
   DTIDX=Q2*XE-YE
   DTIDY=Q2*YE+XE
   DTIDZ=Q2*ZE
   DIIDR=(ZF*DTIDZ+ YF*DTIDY+XF*DTIDX)/RF
   DIIDT= XE*DTIDY-YF*DTIDZ
   DIIDP= GF*DTIDZ-(YF*DTIDY+XF*DTIDZ)*ZF/GF
   C COMPUTE THE PARTIAL DERIVATIVES OF BIG PHI
   C
   DPPDX=Q1*XE
   DPPDY=Q1*YE
   DPPDZ=Q1*ZE+1.0
   DPPDR=(ZF*DPPDZ+ YF*DPPDY+XF*DPPDX)/RF
   DPPDT= XE*DPPDY-YF*DPPDZ
   DPPDP= GF*DPPDZ-(YF*DPPDY+XF*DPPDX)*ZF/GF
   GE4=GE2*GE2
   GC TO (210,220,230,240),IOEM
   C COMPUTATIONS FCR MODE 1 OPERATION.
   C
210 GC TO (211,212,213,214,215,999),IGT
   C TRACKING ERROR DISTRIBUTION SIZES (SPHERICAL COORDINATES)
   C (FOR GT 1, 2, OR 3)
   C
211 CCNTINUE
212 CCNTINUE
213 SR2=(123.0+0.0225*R)**2
   ST2=(.0643*TD)**2
   SP2=(.1320*PD)**2
   GC TO 219
   C (FOR GT 4 AND 5)
   C

```

```

C 214 CONTINUE
215 SR2=(123.0+0.0225*R)**2
ST2=(0.0167-.000710/(.0517+ABS(TC1)))*2
SP2=(0.0116-.000216/(.0235+ABS(PD)-4.0*FDD))**2
C
C SET UP THE ELEMENTS OF THE MATRIX E=AT
C
219 B11=A11*T11+A12*T21+A13*T31
B12=A11*T12+A12*T22+A13*T32
B13=A11*T13+A12*T23+A13*T33
B21=A21*T11+A22*T21
B22=A21*T12+A22*T22
B23=A21*T13+A22*T23
B31=A31*T11+A32*T21+A33*T31
B32=A31*T12+A32*T22+A33*T32
B33=A31*T13+A32*T23+A33*T33
C
C COMPUTE THE PARTIAL DERIVATIVES OF BIG, THETA AND BIG PHI W.R.T.
C ON CARRIAGE INPUTS (ALPHA, BETA, SPEED)
C
DTTDAP=B11*DTTDX+B12*DTTDY+B13*DTTDZ
DTTDBP=B21*DTTDX+B22*DTTDY+B23*DTTDZ
DTTDVP=B31*DTTDX+B32*DTTDY+B33*DTTDZ
DPPDAP=B11*CPPDX+B12*CPPDY+B13*CPPDZ
DPPDBP=B21*CPPDX+B22*CPPDY+B23*CPPDZ
DPPDVP=B31*CPPDX+B32*CPPDY+B33*CPPDZ
C
C COMPUTE THE VARIANCES OF BIG, THETA AND BIG PHI
C
STT2=((DTTDR**2)*SR2+(DTTDT**2)*ST2+(DTTDP**2)*SP2
+((DTTDAP*ESAP)**2+(DTTDBP*ESBP)**2+(DTTDVP*ESVP)**2)*T2)/GE4
1 SPP2=((DPPDR**2)*SR2+(DPPDT**2)*ST2+(DPPDP**2)*SP2
+((DPPDAP*ESAP)**2+(DPPDBP*ESBP)**2+(DPPDVP*ESVP)**2)*T2)/GE2
1 GO TO 29
C
C COMPUTATIONS FOR MODE 2, 3, OR 4 OPERATION
C
220 GO TO (999,999,223,999,225,226),IGT
C
C TRACKING ERROR DISTRIBUTIONS SIZES (SPHERICAL COORDINATES)
C (FOR MODE 2)
C
223 CONTINUE
225 CCNTINUE
226 SR2=(41.0+0.0075*R)**2
SP2=(.000982+.1681*TD*TD)**2

```

```

GG TO 65
230 GG TO (999,999,233,999,235,236),IGT
CC
   (FOR MODE 3)
233 CONTINUE
235 SR2=(17.0+0.24*ABS(RDD)+0.018*RDD*RDD)**2+SD2RJ
236 ST2=(0.00196+0.050*TD)**2
   SP2=(0.000982+0.11*ABS(ABS(PD)-2.0*PDD))**2+SP2MP
GG TO 65
240 GG TO (999,999,243,999,245,246),IGT
CC
   (FOR MODE 4)
243 CONTINUE
245 SR2=(17.0+0.24*ABS(RDD)+0.018*RDD*RDD)**2+SD2RJ
246 ST2=(0.000982+0.1681*TD*TD)**2
   SP2=(0.000491+0.033*ABS(ABS(PD)-4.0*PDD))**2+SP2MP
C
65 R8TD=RANS*THESD
   R8PD=RANS*PHISD
CC
VELOCITY COMPONENT ERROR DISTRIBUTION SIZES
SVX2=(SR2*(PHISD*CTBSPB+THESD*STBCPB)**2
1 +ST2*(R8PD*STBSPB-R8TD*CTBCPB-RANSD*STBCPB)**2
2 +SP2*(R8PD*CTBCPB-R8TD*STBSPB+RANSD*CTBSPB)**2)*ATLCON
SVY2=(SR2*(PHISD*STBSPB-THESD*CTBCPB)**2
1 +ST2*(R8PD*CTBSPB+R8TD*STBCPB-RANSD*CTBCPB)**2
2 +SP2*(R8PD*STBCPB+R8TD*CTBSPB+RANSD*STBSPB)**2)*ATLCON
SVZ2=(SR2*(PHISD*CPB)**2+SP2*(R8PD*SPB-RANSC*CPB)**2)*ATLCON
CC
CGMPUTE THE VARIANCES OF BIG THETA AND BIG PHI
STT2={{(DTDR**2)*SR2 + (DTTD**2)*ST2 + (DTDP**2)*SP2
1 +{(DTDX**2)*SVX2+(DTDY**2)*SVY2+(DTDZ**2)*SVZ2}/GE4
1 +SD2J
SPP2={{(DPPDR**2)*SR2 +{(DPPD**2)*ST2 +{(DPPD**2)*SP2
1 +{(DPPDX**2)*SVX2+(DPPDY**2)*SVY2+(DPPDZ**2)*SVZ2}/GE2
1 +SD2J
CC
COMPUTATION OF VULNERABLE AREA OF AIRCRAFT AT INTERCEPT
29 VP=VSHELL(T)
   XU=XE/RE
   YU=YE/RE

```

```

ZU=ZE/RE
XE=XU*RA
YE=YU*RA
ZE=ZU*RA
GE=SQRT(XE*XE+YE*YE)
VXA=GETVAL(VXFP)
VYA=GETVAL(WYFPA)
VZA=GETVAL(VZFP)
VGA=SQRT(VXA*VXA+VYA*VYA)
VA=SQRT(VGA*VGA+VZA*VZA)
VXI=VP*XU-VYA
VYI=VP*YU-VZA
VI=SQRT(VXI*VXI+VYI*VYI+VZI*VZI)
ALFA=GETVAL(AFPA)
CA=COS(ALFA)
SA=SIN(ALFA)
BETA=BFPA(INDEX1)+FRACT*ANGLIM(BFPA(INDEX2))-BFPA(INDEX1)
CG=COS(BETA)
SG=SIN(BETA)
PZI=PFPA(INDEX1)+FRACT*ANGLIM(PFPA(INDEX2))-PFPA(INDEX1)
THE NEXT CARDS ELIMINATE AN EQUIVALENCE BY USING A DOUBLE
REPLACEMENT.
T33=CP=COS(PZI)
T13=SP=SIN(PZI)
Q1=VXI*CG+VYI*SG
Q2=VZI*CG-Q1*SA
Q3=VYI*CG-VXI*SG
VXF=Q1*CA+VZI*SA
VYF=Q2*CP+Q3*CP
VZF=Q2*CP-Q3*SP
(SET UP INDICES FOR VULNERABLE AREA INTERPOLATION)
F1=ATAN2(VYF,VXF)/QTRPI
IF(F1-LT.0.0)F1=F1+8.0
I1=F1-FLOAT(I1)
I1=I1+1
F2=ACOS(VZF/VI)/QTRPI
I2=F2-FLOAT(I2)
I2=I2+1
F3=AMIN1(7.599999999,VI/152.4)
I3=F3-FLOAT(I3)
DI=1.0-F1

```

C C

C C

```

D2=1.0-F2
D3=1.0-F3
J1=I1+1
J2=I2+1
J3=I3+1
C C C (PERFORM LINEAR THREE DIMENSIONAL INTERPOLATION)
AVT=D3*(D2*(D1*VAT(I1,I2,I3)+F1*VAT(J1,I2,I3))+
1 F2*(D1*VAT(I1,J2,I3)+F1*VAT(J1,J2,I3))+
2 F3*(D1*VAT(I1,I2,J3)+F1*VAT(J1,I2,J3))+
3 F2*(D1*VAT(I1,J2,J3)+F1*VAT(J1,J2,J3)))
C C C SET UP DISTRIBUTION SIZES OF OTHER SOURCES OF RANDOM ERROR
SVA=XU*VXA+YU*VYA+ZU*VZA
CVA2=VA*VA-SVA*SVA
VMQ=.99*VMUZZ/RA-ASHCON
DTI=(VMQ-SQRT(VMQ*VMQ-4.0*BSHCON))/(2.0*BSHCON)-T
SLXMV2=CVA2*(DTI*VP/(VP-SVA/.99))**2
SLYFR2=(0.010*VA*T)**2
SLXFR2=SLYFR2*(1.0+CVA2/(VP-SVA))**2
SACAP2=(0.003*RA)**2
SAOGJ2=(0.005*RA)**2
SACBD2=(BDACCN(IGT)*RA)**2
IF(I0EM.EQ.1)GO TO 68
SAOPEZ=(0.002*RA)**2
GC TO 67
68 SAOPEZ=(V*V-((X*VX+Y*VY+Z*VZ)/R)**2)*
1 RA2*(0.0001463-7.478E-11*(R-1386.0)**2)**2
67 CDIST=SAOAP2+SAOGJ2+SAOPE2+SAOBD2
C C C COMBINE ALL ERRORS INTO ONE DISTRIBUTION, COMPUTE BIAS
SXA2=STT2*RA2
SYA2=SPP2*RA2
SXL2=CDIST+SLXMV2+SLXFR2
SYL2=CDIST+SLYFR2
CTT=XE/GE
CPTT=YE/RE
CFPP=ZE/RE
SFP=ZE/RE
T-YA*CTT
BXA=XA*CPP-(YA*STT+XA*CTT)*SPP
BYA=ZA*CPP-(YA*STT+XA*CTT)*SPP
BXA2=BXA*BXA
BYA2=BYA*BYA
VAM=VXA*STT-VYA*CTT
VAP=VZA*CPP-(VYA*STT+VXA*CTT)*SPP

```

```

VAM2=VAM*VAM
VAP2=VAP*VAP
VAI2=VAM2+VAP2
CD2=VAM2/VAI2
SXA2=SYA2+CD2*SXL2+SD2*SYL2
SYAT2=SYA2+CD2*SYL2+SD2*SXL2
TWOCOV=2.0*VAM*VAP*(SXL2-SYL2)/VAI2
DIF=SXA2-SYAT2
DEN=2.0*SQR(TWOCOV*TWOCOV+DIF*DIF)
HC2Z=DIF/DEN
CZ2=0.5+HC2Z
SZ2=0.5-HC2Z
SZCZ=TWOCOV/DEN
STUFF=2.0*SZCZ*BXA*BYA
BXF2=CZ2*BXA2+SZ2*BYA2+STUFF
BYF2=CZ2*BYA2+SZ2*BXA2-STUFF
SYF2=CZ2*SYA2+SZ2*SYAT2+SZCZ*TWOCOV
AVTPI=AVT/PI2

CC
CC
      COMPUTE PROBABILITY OF KILL
      STUFF=BXF2/(SXF2+AVTPI)+BYF2/(SYF2+AVTPI)
      IF(STUFF.LT.50.0)GO TO 75
      PK=0.0
      GO TO 78
75  PK=AMINI(1.0,EXP(-.5*STUFF)*AVTPI/SQRT((SXF2+AVTPI)*(SYF2+AVTPI)))
78  PS=(1.0-PK)*ISB
      PK=1.0-PS
      TI=TIME+T

CC
CC
      ACCUMULATE PK AS A FUNCTION OF INPUT TIME INTERVALS
      I=0
      I=I+1
50  IF(TIME-GE.TINTER(I))GO TO 50
      J=1
51  IF(TI-LT.TINTER(J))GC TO 52
      J=J+1
      GO TO 51
52  PTOTTF(I)=PK+PS*PTOTTF(I)
      PTOTTI(J)=PK+PS*PTOTTI(J)
      CPS=CPS*PS

CC
CC
      ACCUMULATE PK FOR EACH SPHERICAL SECTOR
      SFKT(I1,I2,I3)=PK+PS*SPKT(I1,I2,I3)

```

```

CC
CC      COMPUTE QUANTITIES FOR EXTENDED OUPUT,  WHEN DESIRED
IF(IPRINT(6).LE.0)GO TO 20
C1=THESD#DEGREE
C2=PHISD#DEGREE
C3=ETHI4#DEGREE
C4=EPHI4#DEGREE
O5=SQRT(BXF2+FUZZ)
O6=SQRT(BYF2+FUZZ)
O7=SQRT(SXF2)
O8=SQRT(SYF2)
O9=1.0-CPS

CC      WRITE EXTENDED OUTPUT
CALL PAGES(1,5,JP)
IF(JP.EQ.0)WRITE(6,1013) ISL,IGT,IEM,XGUN,YGUN,ZGUN,CIRCLE
WRITE(6,1014)IG,IEM,TIME,T,II,R,RA,07,08,05,06,VI,01,02,03,04,
1
20 NROUND=NROUND+ISB

CC      FIRE ADDITIONAL GUNS IN COMPLEX, IF ANY
64 IG=IG+1
IF(IG.GT.IGL)IG=1
IF(FIRE.TFIRE+TPERS)
IF(TIME.GT.TFIRE)GO TO 54
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
CC      TTTT      RRRR      R      A      A      A      CCC      K      K
CC      T      F      RRRR      A      A      C      C      C      K      K
CC      T      R      RRRR      A      A      C      C      C      K      K
CC      T      R      R      A      A      A      C      C      K      K
CC      T      R      R      A      A      A      C      C      K      K
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
62 TIME=TIME+.064
62 IF(TIME.GT.TMAX)GO TO 69

CC      SWITCH TO MCDE 1 TRACKING IF JAMMING IS ABOVE THRESHOLD OR IF
CC      RANGE IS TOO CLOSE
IEM = IEM
SD2J = 0.
SC2RJ = 0.
IF(R.LT.RSMCDE) GO TO 501
IF(IJAM.EQ.0) GO TO 502

```

```

C
C
CALL ECM2
IF(SJT.GT.SJTMAX) GO TO 501
IF(IOEM.EQ.3)CALL JAMER2(IRECM,SJT,SD2J)
IF(IOEM.EQ.4) SD2RJ=SD2RJM
IF(IRECM.EQ.3) SD2RJ = SD2RJM
GC TO 502
501 IOEM = 1
502 CCNTINUE
IF(IJAM.EQ.0) GO TO 53
IF(IP.EQ.0) GO TO 53
IF( (ILOOP/IP)*IP.NE.ILOOP) GO TO 53
DUM = 10.*ALOG10(GJ)
WRITE(11,1044) TIME,R,XSEC,DUM,SJT,IOEM,SD2J,SD2RJ,SN
CC
CC
CC
CC
COMPUTE MEAN TRACKING ERRORS
53 GO TO(310,320,330,340),IOEM
(MODE 1, GT 1 - 5)
310 GC TO (311,312,313,314,315,999),IGT
311 CCNTINUE
312 CCNTINUE
313 CCNTINUE
314 CCNTINUE
315 THESD=1.11*TD+0.9*TDD+6.0*ETHE
PHISD=1.10*PD-0.7*PDD+6.0*EPHI
RANSD=RD+3.*ERAN
GC TO 73
CC
CC
(MODE 2, GT 3, 5, AND 6)
320 GC TO (999,999,323,999,325,326),IGT
321 CCNTINUE
322 CCNTINUE
323 CCNTINUE
325 THESD=0.91*TD+0.45*TDD+6.0*ETHE
PHISD=0.75*PD-0.25*PDD+6.0*EPHI
RANSD=RD+3.*ERAN
GC TO 73
CC
CC
(MODE 3, GT 3, 5, AND 6)
330 GC TO (999,999,333,999,335,336),IGT
331 CCNTINUE
332 CCNTINUE
333 CCNTINUE
334 CCNTINUE
335 CCNTINUE
336 THESD=TD+6.0*ETHE

```

```
PHISD=PD+6.*EPIH  
RANSQ = 0.804*RD + 3.0*ERAN  
GO TO 73
```

```
CC
```

```
(MODE 4, GT 3, 5, AND 6)
```

```
340 GO TO (999,999,343,959,345,346),IGT  
343 CONTINUE  
345 CONTINUE  
346 THESD=0.910*TD+0.45*TD+6.0*ETHE  
PHISD=0.75*PD-0.25*PDD+6.0*EPIH  
RANSQ = 0.804*RD + 3.0*ERAN
```

```
CC
```

```
(LIMIT SLEW RATES AND ELEVATION ANGLE TO WEAPCN MAXIMUMS)
```

```
73 THESD=SIGN(AMINI(TDMAX,ABS(THESD)),THESD)  
PHISD=SIGN(AMINI(PDMAX,ABS(PHISD)),PHISD)  
THES=ANGLIM(THES+0.064*THESD)  
PHIS=AMAXI(PHMIN,AMINI(PHMAX,PHIS*0.064*PHISD))  
RANS=AMAXI(0.0,RANS+0.064*RANSQ)  
CTB=COS(THES)  
STB=SIN(THES)  
CPB=COS(PHIS)  
SPB=SIN(PHIS)  
CTBCPB=CTB*CPB  
CTBSPB=CTB*SPB  
STBCPB=STB*SPB  
STBSPB=STB*SPB  
RCS=RD  
TCS=TD  
PLS=PD  
IF(IEM.LT.2)GO TO 60
```

```
CC
```

```
CCOMPUTE MEAN (SMOOTHED) VELOCITY COMPONENTS (ELECTRONIC  
COMPUTATION)
```

```
VXE=RANSQ*CTBCPB-RANS*(STBCPB*THESD+CTBSPB*PHISD)  
VYE=RANSQ*STBCPB+RANS*(CTBCPB*THESD-STBSPB*PHISD)  
VZE=VXE+EMDTA*(VYES-VXE)  
VYE=VYE+EMDTA*(VYES-VYE)  
VZE=VZE+EMDTA*(VZES-VZE)  
VXES=VXE  
VYES=VYE  
VZES=VZE  
GO TO 60
```

```
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC  
C
```

```

CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
65 DC 92 I=1,4
92 IFLAG(I)=IFLAGS(I)+IPRINT(I)
C
C PRINT PK AS A FUNCTION OF AIRCRAFT ASPECT AND IMPACT SPEED
C
C IF(IPRINT(5).EQ.0)GO TO 80
C CALL PRSEGS(SPKT2,ISL)
C
C COMPUTE PK AS A FUNCTION OF ASPECT AND IMPACT SPEED FOR ALL GUNS
C
80 DC 36 I=1,32
DC 36 J=1,8
PK=RHO(I)*SPKT2(I,J)
36 SPKTOT(I,J)=PK+(1.0-PK)*SPKTOT(I,J)
C
C STORAGE OF PK VS DENSITY FACTOR AND TIME INTERVALS (AT FIRE AND
C INTERCEPT) PER WEAPON OR WEAPON COMPLEX
C
CPK=1.0-CPS
DC 55 I=1,NRHOS
D=RHO(I)
PK=D*CPK
PKTIDC(I)=PK+(1.0-PK)*PKTIDC(I)
DC 55 J=1,NTINTS
D1=D*PTOTIF(J)
PKTFDC(J,I)=D1+(1.0-D1)*PKTFDC(J,I)
D2=D*PTOTI(J)
55 PKTIDC(J,I)=D2+(1.0-D2)*PKTIDC(J,I)
C
C COMPUTE, STORE, AND WRITE TOTAL PKs FOR ENTIRE ARRAY OF WEAPONS
C
F=FLOAT(NROUND/ISB)*IPERS
WRITE(7)ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,I,TRACK,
1 I,CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PTOTIF,PTOTI,IPRINT
ASSIGN 70 TC IE0F
GC TO 15
70 ENDFILE 7
LINE=66
REWIND 7
PUNCH2 CARDS WRITE ONLY ONE CARD IMAGE ON TAPE4 WHEN THERE IS
C

```

```

C ONLY ONE DEFENSE PER EMPLOYMENT (I.E., ONLY ONE "12" CARD
BEFORE 7/8/9 END-OF-RECORD).
K = ISL
CALL PAGES(4,0,JP)
WRITE (6,1033)
ASSIGN 40 TC IE0F
75 READ(7) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PLOTF,PLOTTI,IPRINT
IF (IE0F(7).NE.0.) GO TO IE0F(4,12,40,70,83,9597,9959)
IF (IOCHECK(7).NE.0.) GO TO IERR(3,28,9996)
CALL PAGES(1,4,JP)
IF (JP.EQ.0) WRITE (6,1033)
WRITE (6,1034) ISL,CPK,NROUND,F,XGUN,YGUN,ZGUN,CIRCLE,IGL,
1 TREAT,TTRACK,IGT,IEM,ICB,ISB,ISL
WRITE (4,1060) ITITLE(9),ITITLE(10),ISL,IGT,IEM,ICB,ISB,IGL,CPK,
1 NROUND,XGUN,YGUN,ZGUN,F,NUMBER
GO TO 79
40 IF (K.NE.2) WRITE (4,1064) ITITLE(9),ITITLE(10),PKTDC(1),NUMBER
IF (IFLAGS(2).LE.0) GO TO 76
REWIN 7
CALL PAGES(4,0,JP)
WRITE (6,1031) (I,I=1,10)
77 READ(7) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PLOTF,PLOTTI,IPRINT
IF (IPRINT(2).LE.0) GO TO 77
CALL PAGES(1,4,JP)
IF (JP.EQ.0) WRITE (6,1031) (I,I=1,10)
WRITE(6,1032) ISL(TINTER(I),I=1,NTINTS)
IFLAGS(2)=IFLAGS(2)-IPRINT(2)
IF (IFLAGS(2).GT.0) GO TO 77
76 IF (IFLAG(3).LE.0) GO TO 42
REWIN 7
CALL PAGES(4,0,JP)
WRITE (6,1035) (I,I=1,10)
41 READ(7) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PLOTF,PLOTTI,IPRINT
IF (IPRINT(3).LE.0) GO TO 41
CALL PAGES(2,4,JP)
IF (JP.EQ.0) WRITE (6,1035) (I,I=1,10)
WRITE(6,1018) ISL(PLOTF(I),I=1,NTINTS)
DO 410 I=2,NTINTS
410 PLOTF(I)=PLOTF(I-1) + (1.0-PLOTF(I-1))*PLOTF(I)
WRITE (6,1058) ISL(PLOTF(I),I=2,NTINTS)
IFLAGS(3)=IFLAGS(3)-IPRINT(3)
IF (IFLAGS(3).GT.0) GO TO 41
42 IF (IFLAG(4).LE.0) GO TO 71
REWIN 7
CALL PAGES(4,0,JP)

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WRITE (6,1036) (I,I=1,10)
43 READ(7) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PTOTTF,PTOTI,I,PRINT
IF(I,PRINT(4).LE.0) GO TO 43
CALL JPAGES(2,4,JP)
IF(I,JP,EQ.0) WRITE (6,1036) (I,I=1,10)
WRITE(6,1018) ISL, (PTOTI(I),I=1,NTINTS)
DC 430 I=2,NTINTS
PTOTI(I) = PTOTI(I-1) + (1,0-PTOTI(I-1))*PTOTI(I)
430 WRITE (6,1058) ISL,(PTOTI(I),I=2,NTINTS)
IFLAGS(4)=IFLAGS(4)-IPRINT(4)
IF(IFLAGS(4).GT.0)GO TO 43
71 IF(IFLAGS(1).LE.0)GO TO 44
REWIND 7
CALL PAGES(4,0,JP)
WRITE (6,1017) (I,I=1,9)
44 READ(7) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PTOTTF,PTOTI,I,PRINT
IF(I,PRINT(1).LE.0)GO TO 74
CALL JPAGES(1,4,JP)
IF(I,JP,EQ.0) WRITE (6,1017) (I,I=1,9)
WRITE(6,1055) ISL,(RHO(I),I=1,NRHOS)
IFLAGS(1)=IFLAGS(1)-IPRINT(1)
IF(IFLAGS(1).GT.0)GO TO 74
44 CALL PAGES(7,NTINTS,0,JP)
WRITE (6,1019)
PTOTTF(1) = PKTFDC(1,1)
DC 440 I=1,NTINTS
IF (I,EQ.1) GO TO 440
PTOTTF(I) = PTOTTF(I-1) + (1,0-PTOTTF(I-1))*PKTFDC(I,1)
440 WRITE (6,1018) I,PTOTTF(I),(PKTFDC(I,J),J=1,NRHOS)
CALL PAGES (9+NTINTS,0,JP)
WRITE(6,1020)
WRITE (6,1021) (I,I=1,9)
PTOTTI(1) = PKTIDC(1,1)
DC 441 I=1,NTINTS
IF (I,EQ.1) GO TO 441
PTOTTI(I) = PTOTTI(I-1) + (1,0-PKTII(I-1))*PKTIDC(I,1)
441 WRITE (6,1018) I,PTOTTI(I),(PKTIDC(I,J),J=1,NRHOS)
CALL PRSEGS(5PKTOT,0)
GO TO 95
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C
C
C
EEEE X X X IIII TTTT SSSS
EEE X X X I I T SSS
C
C
C

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1006 FORMAT(///)*-IMPRCPR INPUT CARD ENCOUNTEREC. ** , I2, 7A10, A8, ***)
1007 FORMAT( I2, 7A10, A8)
1008 FORMAT( 6X, 9E8, 0)
1009 FORMAT( 1X, 5I1, E8, 0)
1010 FORMAT(*-VULNERABLE AREA (SQ. METERS) AS A FUNCTION OF IMPACT *,
1 152 305 613 762 1067 1219**/)
2 *SPEED (METERS/SEC) AND ASPECT VIEW*/5X, 8A10/** VIEW*, 7X*0
1011 FORMAT(1H-, I2, * TIME INTERVALS FOR PK ACCUMULATION*/4X*0.00*,
1 10F8, 2)
1012 FORMAT( I2, LOCATION*, I5, 6X, #GUN TYPE*, I2, 6X, #ERRCR MODE*, I2, 6X, 3)
1013 FORMAT(* LOCATION*, I5, #GUN TYPE*, I2, 6X, #ERRCR MODE*, I2, 6X, #M/*0L
1 #POSITION=(*, F8, 1, #, F8, 1, #) RADILCS=*, F6, 1, # M/*0L
2 FIRE INTCP* FIRE TIME RANGE AZIM ELEV
3 MEAN FIRE MEAN DEM VEL. VULN/* N DEM TIME RANGE EL. ERR
4 SIG1 SIG2 BIAS1 BIAS2
5 AREA SHOT PK CUM. PK**/)
1014 2 FORMAT( I2, I3, 2F7.2, F8.2, 2F7.0, 1X, 2F6.1, 1X, 2F7.1, F8.1, 1X, 2F7.2, 1X,
3F7.2, 2F10.6)
1015 FCRMAT( 3X, 1)
1016 FCRMAT( 6I1)
1017 FCRMAT(*- LCC*, I2X, 9(7X**RHO*, I2) //)
1018 FCRMAT( I5, LCC*, I2)
1019 FCRMAT(*- ACCRUED AS A FUNCTION OF TIME AT INTERCEPT*)
1020 FCRMAT(*- ACCRUED AS A FUNCTION OF TIME AT INTERCEPT*)
1021 FCRMAT(*0TIME*, 5X*CUM FOR*, 9(5X*DENSITY*)/* SEG. *, 5X*CLASS 1*,
9(5X*CLASS*, I2) //)
1023 FCRMAT(1H0, 10X, *TOTALS*, 9F12.7)
1024 FCRMAT(///)*-END OF JOB. * 9X*EXECUTION CPU TIME *, F7.2, 9X*EXEC*,
*UTION (CPU + 10/2) TIME *, F7.2, 9X*TOTAL CPU TIME *, F7.2)
1025 FCRMAT(*0G BALLISTIC MIN VEL MAX AZIM ELEV MAX MUZZ
2 ALLISTIC BALLISTIC MIN VEL MAX RANGE RATE SMOOTH MAX AZ
3X EL*/ * T CONSTANT 2 MIN MAX ELEV RATE TOF1 TOF2 VELOCITY
4N STANT 1 CONSTANT 2 MIN MAX CCNST ERROR
5RROR**/)
1026 FCRMAT( I5, 4X, #0.00*, 8F8.2)
1027 FCRMAT(///)*-GUN TYPE*, I2, *, ERRCR MODE*, I2, *. COMBINATION INVAL
1 IC. *)
1028 FCRMAT( I4, I2, 9E8, 0)
1029 FCRMAT(1H-, I1, * DENSITY CLASSES FOR PK ACCUMULATION*/9F12.5)
1031 FCRMAT(*- LCC*, I0(6X, *TIME*, I2) //)
1032 FCRMAT( I5, LCC*, I2)
1033 FCRMAT(*- LCC*, 5X*P(KILL) *, 6X*ROUNDS*, 3X*FIRE TIME*, 8X*XGUN*,
1 8X*YGUN*, 8X*ZGUN
2 * GT EM CB SB LOC*)
1034 FCRMAT( I5, F12.7, I12, 4F12.2, I3, 2F12.2, 4I3, I5)
1035 FCRMAT(*- LCC*, I0(4X, *PK(TI*, I2, *) //)
1036 FCRMAT(*- LCC*, I0(4X, *PK(TI*, I2, *) //)
1037 FCRMAT(///)*-UNRECOVERABLE PARITY ERROR DETECTED. CALL EXIT.*)

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1038 FCRMAT(13,15,F10.0)
1039 FCRMAT(* ECM SHOULD NOT BE SPECIFIED WHEN IEM IS 1 OR 2 *)
1040 FCRMAT(* MULTIPATH TRACKING ERROR SPECIFIED WITH IEM=3 ONLY *)
1041 FCRMAT(13,12,3F10.0,15,2F10.0,15,F10.0)
1042 FCRMAT(// * ECM VARIABLES *)
1043 FCRMAT(* AAASIM-----*,10A10)
1044 FCRMAT(1X,F7.2,4X,F8.0,2X,F9.2,5X,F7.2,5X,F7.2,
15X,12,3X,E10.4,5X,E10.4,5X,F7.2)
1045 FCRMAT(// TIME RANGE(M) X-SEC(SQM) JAM.GAIN(DB) J/S(DB)*,
FIRE ADD.TRK.VAR. ADD.RNG.VAR. S/N(DB)*,
MODE *CUM* 9X,9F12.7)
1058 FCRMAT(15,*CUM* 9X,9F12.7)
1059 FCRMAT(15,12X,9F12.5)
1060 FCRMAT(2A9,13,5I2,F10.7,15,2F7.0,F6.0,F6.1,16)
1061 FCRMAT(1H0,4X*DATE*,7X*TIME* 4X*LCC GT EM CB SB GL*,5X*P(K*,
*ILL) RDS XGUN YGUN ZGUN F TIME PAGE*/)
1062 FCRMAT(2A9,A3,5A2,A10,A5,2A7,3A6)
1063 FCRMAT(2(2X,A9),2X,A3,5(2X,A2),2X,A10,1X,A5,2(1X,A7),3(2X,A6))
1064 FCRMAT(2A9,13X,F10.7,31X,16)
END
SUBROUTINE PAGES(N,NT,JP)

```

CCCCCCCCCCCC
KEEPS NUMBER OF LINES PER PAGE LESS THAN 59, PRINTS HEADER,
AND GETS TIME INFORMATION FROM SYSTEM. REPLACES
NPAGE(MAX) AND HEADER IN AFATL PROGRAM.

NT NUMBER OF LINES TO BE PRINTED BEFORE NEXT CALL TO PAGES.
NUMBER OF LINES IN TITLE OR HEADER OF DATA BEING PRINTED.
IF CALL TO PAGES IS TO PRINT HEADER ONLY, "N" SHOULD
BE NUMBER OF LINES AND "NT" SHOULD BE ZERO.
JP FLAG FROM PAGES, SET TO ZERO WHEN A NEW PAGE IS STARTED
COMMON /BLOCK1/ TITLE(10)
COMMON /HEADFO/ LINE,NUMBER
JP = 2
LINE = LINE + N
IF (LINE .LT. 59) RETURN
NUMBER = NUMBER + 1
CALL TIME (ITITLE(10))
WRITE (6,1000) ITITLE,NUMBER
LINE = 2 + N + NT
JP = 0
RETURN
FCRMAT(*1AFATL P-COI AAASIM ** ,7A10,** ,A10,2(1XA9),4X*PAGE*,14/)
END
SUBROUTINE PRSEGS(P,ISL)

C PRINTS THE PK AS A FUNCTION OF ASPECT AND IMPACT SPEED TABLES

C
C
C

```
THIS SUBROUTINE EXTENSIVELY MODIFIED TO PRINT ASPECT SECTOR
ANGLES AND PROPERLY LABEL THE TWO CASES FOR WHICH IT
PRINTS TABLES
COMMON /BLOCK3/ XGUN, YGUN, ZGUN, IGL, CIRCLE
COMMON /BLOCK4/ IGT, IEM, ICB, ISB, I, ANG(18)
DIMENSION P(32, 8), PT(8), ANG(18)
DATA ANG/7H315-360, 7H000-045, 7H045-090, 7H090-135, 7H135-180,
7H180-225, 7H225-270, 7H270-315/
1 PK=PT(1)=PT(2)=PT(3)=PT(4)=PT(5)=PT(6)=PT(7)=PT(8)=0.0
CALL PAGES(6, 0, JP)
IF (ISL.GT.0) WRITE (6, 1001) ISL, IGT, IEM, XGUN, YGUN, ZGUN, CIRCLE
IF (ISL.EQ.0) WRITE (6, 1030)
WRITE(6, 1040)
DC 2 I=1, 32
IAZ = 1 + MCD(I, 8)
IEL = 2 + (I-1)/8
PP=0.0
DC 3 J=1, 8
PT(J)=PT(J)+(1.0-PT(J))*P(I, J)
PK = PK + (1.0-PP) * P(I, J)
3 PP=PP+(1.0-PP)*P(I, J)
CALL PAGES(1, 6, JP)
IF (JP.NE.0) GO TO 2
IF (ISL.GT.0) WRITE (6, 1001) ISL, IGT, IEM, XGUN, YGUN, ZGUN, CIRCLE
IF (ISL.EQ.0) WRITE (6, 1030)
WRITE (6, 1040)
2 WRITE (6, 1041) I, ANG(IAZ), ANG( IEL), (P(I, J), J=1, 8), PP
CALL PAGES(2, 6, JP)
IF (JP.NE.0) GO TO 4
IF (ISL.GT.0) WRITE (6, 1001) ISL, IGT, IEM, XGUN, YGUN, ZGUN, CIRCLE
IF (ISL.EQ.0) WRITE (6, 1030)
WRITE (6, 1040)
4 RETURN
1001 FCFORMAT(*-PK AS A FUNCTION OF ASPECT SECTOR AND IMPACT SPEED.*,
1 5X*LOC*, I4, 5X*GT*, I2, 5X*EM*, I2, 5X*X*, F7.0, 3X*Y*, F7.0, 3X*Z*,
2 F7.0, 5X*RADIUS*, F5.0, *M*)
1030 FCFORMAT(*-TOTAL PK FOR DENSITY CLASS 1 AS A FUNCTION OF ASPECT*,
1 *SECTOR AND IMPACT SPEED.*)
1040 FCFORMAT(*SECTOR AZIMUTH ELEV.*, 8X*0-152 152-305 *?,
1 *305-457 457-610 610-762 762-914 914-1067*,
2 3X*1067-1219 TOTAL PK*/ 9X*REAR=00 DOWN=00*)
1041 FCFORMAT(I7, 2X, A7, 8F12.7, F14.7)
1042 FCFORMAT(IH0, 18X*TOTALS*, 8F12.7, F14.7)
END
SUBROUTINE TPLOT(NFPA)
```

C

C PLOTS X VS. Y AND X VS. Z ON PRINTER FOR EACH FLIGHT PATH.
C NO PLOT WHEN XMAX-XMIN IS LESS THAN 50.
C

CCOMON XFPA(1201), YFPA(1201), ZFPA(1201),
BFPA(1201), AFPA(1201), PFPA(1201),
VFPA(1201), VVFPA(1201), VZFFPA(1201)
1 PLOT(111)
2

DIMENSION XA, YAXIS, XAXIS, POINT/1H, 1H-, 1H+, 1H+/
LGGICAL XA, YAXIS, XAXIS, POINT/1H, 1H-, 1H+, 1H+/
DATA BLANK, ABOUT 50 POINTS WITH "IFPA".
C PLOT = MAXO(NFPA/50, 1)
C IFPA = MIN-MAX VALUES OF X, Y, AND Z (ZMIN = 0. BY DEFN.)

1 FIND MIN-MAX VALUES OF X, Y, AND Z (ZMIN = 0. BY DEFN.)
2 XMIN = YMIN = 1.E99
XMAX = YMAX = ZMAX = -1.E99
DO 100 I=1, NFPA, IFPA
XMIN = AMIN1(XMIN, XFPA(I))
YMIN = AMIN1(YMIN, YFPA(I))
XMAX = AMAX1(XMAX, XFPA(I))
YMAX = AMAX1(YMAX, YFPA(I))
ZMAX = AMAX1(ZMAX, ZFPA(I))
C CHECK X RANGE TO AVOID FUNNY X-Z PLOT.
C XCRNG = XMAX - XMIN
C IF (XCRNG.GT. 49.999) GO TO 110
C CALL PAGES(3,0,1)
C WRITE(6,210)
C RETURN

C 110 SCALE X, Y TE METERS PER CHARACTER.
C D = AMAX1(XRNG/11., (YMAX-YMIN)/7.25)
C DX = D/10.
C DY = D/8.

C PLOT BY RE-CALCULATING X-Y MIN-MAX.
C CENTER = (XMAX+XMIN)/2. - 55.*DX
C XMIN = (YMAX+YMIN)/2. - 29.*DY
C XMAX = XMIN + 110.*DX
C YMAX = YMIN + 58.*DY
C SCALE Z (NOT TO SAME SCALE AS X-Y).
C DO 120 I=1, 100
C D = FLOAT(I*500)

120 IF (D.GT. ZMAX) GC TO 130
C CONTINUE

130 DZ = D/10.
C SEE IF AXES SHOULD APPEAR ON PLOT.
C XA = .FALSE.
C NYA = -10

IF (XMIN.LE. 0. .AND. XMAX.GE. 0.) XA = .TRUE.
IF (YMIN.LE. 0. .AND. YMAX.GE. 0.) NYA = .TRUE.
IF (XA) NXA = 1 + INT(0.5-XMIN/DX)
IF (NYA) NYA = 1 + INT(0.5-YMIN/DY)
C SET 8 LINES PER INCH SPACING ON PRINTER.

```

C      .
WRITE (6,280)
SET PAGE CONTROL, PLOT X-Z, THEN X-Y.
CALL PAGES(56,0,1)
WRITE (6,220)
IFPA = 2 * IFPA
DO 160 K=1,11
  DO 140 I=1,111
    PLOT(I) = BLANK
  IF (K.EQ.11) PLOT(I)=XAXIS
  IF (K.EQ.11) AND (K.NE.11) PLOT(NXA) = YAXIS
  Z = FLOAT(11,NFPA) IFPA
  DO 150 I=1,INT(0.5+Z*FPA(I)/DZ) PLOT(IX) = POINT
  IF (K.EQ.11) PLOT,Z
  WRITE (6,230)
  WRITE (6,220)
  WRITE (6,220)
  IFPA = IFPA/2
  DO 190 K=1,59
    DO 170 I=1,111
      PLOT(I) = BLANK
    IF (K.EQ.59) PLOT(NXA) = YAXIS
    Y = (XMIN + FPA(I)-XMIN)/DX
    DO 180 I=1,NFPA
      IF (K.EQ.59-INT(0.5+(YFPA(I)-YMIN)/DY)) PLOT(IX) = POINT
      I = MOD(K,2)
      WRITE (6,240) PLOT
      IF (I.EQ.1) WRITE (6,260) PLOT,Y
      PRINT *DX
    C      D=10.00
  DC 200 I=1,12
  PLOT(I) = XMIN + D*FLOAT(I-1)
  WRITE (6,270) (PLOT(I),I=1,12),DX,DY
  C      RE-SET PRINTER SPACING BACK TO 6 LINES PER INCH.
  RETURN
  C      FORMAT(*ONEARLY CONSTANT X IN FLIGHT PATH.  NC PLOT PRINTED.*/)
  210 FORMAT(1X,113(1H*))
  220 FORMAT(2H *,111(1H*))
  230 FORMAT(2H *,111(1H*))
  240 FORMAT(2X*A*, 9X*B*, 9X*H*, 9X*I*, 9X*J*, 9X*K*, 9X*L*,
  250 1
  9X*G*, 9X*H*, 9X*I*, 9X*J*, 9X*K*, 9X*L*)

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260 FORMAT(2H * 111A1,1H* 3X*Y ** ,F8.1,* M*)
270 FORMAT(#JX(A)=*,F8.1,* M*,4X*(B)=*,F8.1,* M*,4X*(C)=*,F8.1,* M*,
      4X*(D)=*,F8.1,* M*,4X*(E)=*,F8.1,* M*,4X*(F)=*,F8.1,* M*,
      4X*(G)=*,F8.1,* M*,4X*(H)=*,F8.1,* M*,4X*(I)=*,F8.1,* M*,
      4X*(J)=*,F8.1,* M*,4X*(K)=*,F8.1,* M*,4X*(L)=*,F8.1,* M*,
      4X*(M)=*,F8.1,* M*,4X*(N)=*,F8.1,* M*,4X*(O)=*,F8.1,* M*)
280 FORMAT(1HT)
290 FORMAT(1HS)
      END
      SUBROUTINE RPLANE(T)
      C
      C
      C COMPUTES INFORMATICN ABOUT THE POSITION OF THE AIRCRAFT AT TIME =T
      COMMON/BLOCK2/NFPA,TMIN,TMAX,DTFPA
      COMMON/BLOCK3/XGUN,YGUN,ZGUN
      COMMON/NFPARM/XA,YA,ZA,RA2,RA,TIME
      COMMON/IGXGYG/IG,XG(8),YG(8)
      COMMON XFPA(1201),YFPA(1201),ZFPA(1201),BFPA(1201),AFPA(1201),
      COMMON PFPA(1201),VXFPA(1201),VYFPA(1201),VZFPA(1201)
      COMMON/CECM1/TRECM,I,J,GAINJ,IX,XSEC,CALX,PJh,
      X,Y,Z,ROL,PIT,FDG,
      X,FTGT,FJAM,GJ,SJT,SN
      *
      CALL INTERP((T+TIMEI)/DTFPA)
      XA=GETVAL(XFPA)-XGUN-YG(IG)
      YA=GETVAL(YFPA)-YGUN-XG(IG)
      ZA=GETVAL(ZFPA)-ZGUN
      RA2=XAT*XA+YA*YA+ZA*ZA
      RA=SQRT(RA2)
      RETURN
      END
      FUNCTION ANGLIM(X)
      C
      C
      C LIMITS ANGLES TO PRINCIPAL ANGLES BETWEEN -PI AND +PI
      COMMON /CONSTS/ DEGREE,RADIAN,PI,PI2,QTRPI,SQRT2
      IF(ABS(X)-PI)1,1,2
      ANGLIM=X
      RETURN
      1 ANGLIM=X-PI2*FLOAT(IFIX((X+SIGN(PI,X))/PI2))
      2 ANGLIM=X
      RETURN
      END
      FUNCTION RSHELL(T)
      C
      C
      C COMPUTES RANGE TO SHELL AT TIME = T
      COMMON/VASBS/VMUZZ,ASHCON,BSHCON,CQUAD
      DQUAD=1.0+T*(ASHCON+T*BSHCON)
      RSHELL=T*VMUZZ/DQUAD

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```

RETURN
END
FUNCTION VSHELL(T)
  CCMPUTES SPEED OF SHELL AT TIME=T. CAN ONLY BE USED AFTER A CALL
  TO RSHELL AT THE SAME TIME, SINCE RSHELL COMPUTES DQUAD FOR VSHELL
  COMMON/VSASBS/VMUZZ,ASHCON,BSHCON,CQUAD
  VSHELL=VMUZZ*(1.0-BSHCON*T)/(DQUAD*DQUAD)
  RETURN
END
SUBROUTINE INTERP(FINT)
  SETS CONSTANTS (FRACT, INDEX1, ANC INDEX2) FOR TWO POINT INTER-
  POLATION
  COMMON/MAGIC/FRACT,INDEX1,INDEX2
  INDEX1=FINT
  FRACT=FINT-FLOAT(INDEX1)
  INDEX1=INDEX1+1
  INDEX2=INDEX1+1
  RETURN
END
FUNCTION GETVAL(ARRAY)
  PERFORMS TWC POINT INTERPOLATION
  COMMON/MAGIC/FRACT,INDEX1,INDEX2
  DIMENSION ARRAY(1,201)
  GETVAL=ARRAY(INDEX1)+FRACT*(ARRAY(INDEX2)-ARRAY(INDEX1))
  RETURN
END
BLOCK DATA
COMMON /BLOCK1/ ITITLE(10)
COMMON /BLOCK3/ XGUN,YGUN,ZGUN
COMMON /BLOCK4/ IGT,IEM,ICB,ISB,IGL,CIRCLE
COMMON /BLOCK5/ NRHOS,RHO(9)
COMMON /BLOCK6/ NTINTS,TTINTER(10)
COMMON /BLOCK7/ IVACOM(8),VAT(9,5,9)
COMMON /BLOCK8/ TROUND(6),TCK1,TRACK2
COMMON /BLOCK9/ VELMIN(6),VELMAX(6),PHDMAX(6),PHIMIN(6),PHIMAX(6),
1 TFMAX1(6),TFMAX2(6),RANMAX(6),ATLAG(6),
2 ETHMAX(6),EPHMAX(6),RMODES(6)
COMMON /BLOCKA/ RFBACON(6),VMUZEL(6)
COMMON /CONSTS/ DEGREE,RADIAN,PI,FI2,QTRPI,SQRT2
COMMON /HEADFO/ LINE,NUMBER
COMMON /BUDGET/ BDAICON(6)
VALUES ON "ASIDQC" CARDS FROM A.S.I. DOCUMENTATION DRAFT. (BEE)

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C

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DATA ATLAG /2*9999.99,1.33,9999.99,2*1.33/
DATA BDACON /00265,00501,0031,00697,00113,00113,00113/
DATA EPHMAX /57.295779513082/, RACIAN/0.01745325251994/
DATA IGT /6*.1/ IEM/4/, ICB/1/, ISB/4/, IGL/1/, CIRCLE/0./
DATA IVACOM /3/ (1H), 10H ASD/XRDA,1H,1H /
DATA LINE /66/, NUMBER/0/
DATA NRHOS /9/
DATA NTINTS /10/
DATA PHDMAX /43633, 43633, 78540, 34910, 31416, 3491/
DATA PHIMAX /1.48353, 1.57079, 1.48353, 1.48353, 1.51844, 1.43117/
DATA PHIMIN /-1.17453, -1.14835, -1.17453, -0.8727, -0.6981, -0.5236/
DATA PI /3.1415926535898/, PI2/6.2831853071796/
DATA QTRPI /0.78539816339745/
DATA RANMAX /3000, 3000, 3300, 5000, 5500, 59599./
DATA RANMIN /0.0, 400, 500, 3333, 25, 2, 1666666666, 1, 02, 01/
DATA RHO /1.5, -1, 400, 1, 1500, 0.0/
DATA RMODES /-1, 209, 251499, 22958, 089321, 07845, 050694/
DATA RVACON /0.12392, -0.06259, -0.06889, 004262, -0.00421, -0.000357/
DATA SORT2 /1.4142135623731/
DATA SQRD CASES BUT NOT EXACT VALUES FCR MAX RANGE. 13 FEB 73 (BEE)
DATA TFMX2 /1.6, 2.2, 3.8, 4.1, 6.2, 9.99/
DATA THDMAX /99.99, 99.99, 7.5, 99.99, 11.6, 19.2/
DATA TINTER /0.5236, 0.5236, 1.39626, 0.5236, 0.5236, 0.5236/
DATA TREAT /10, 20, 30, 40, 50, 60, 70, 80, 90, 99/
DATA TROUND /0.75, 4, 2, 75, 857, 4./
DATA VAI MAX /405*0./
DATA VELMIN /300, 300, 350, 250, 300, 999.9/
DATA VMUZEL /840, 1000, 930, 880, 560.0, 800./
DATA XGUN /0./
DATA YGUN /0./
DATA ZGUN /0./
END
SUBROUTINE AVG(ITST, IPRINT, ISM)
DIMENSION IPRINT(6)
DATA IBLNK, SUM, CNT/2H ,0,0,0,0,0,0/
CECODE(10,900, IPRINT(1)) PK
FORMAT(10,7)
FORMAT(7,910, IPRINT(4)) YGUN
FORMAT(7,0)
IF (ISM.EQ.IALL) GOTO 600
IF (ITST.EQ.IBLNK) GOTO 200

```

C

900
910

```

IF(YGUN-NE,0.0) GOTO 150
CNT=INT+0.5
SUM=SUM+PK/2.
POLD=0.0
GOTO 500
CINT INUE
CNT=INT+1.0
SUM=SUM+PK
PCLD=PK
GOTO 500
IF(CNT-LE,0.5) GOTO 500
CNT=INT-0.5
SUM=SUM-POLD/2.0
AVERG=SUM/CNT
WRITE(6,800) CNT,AVERG
FORMAT(75X,18H AVERAGE P(KILL) ON,F6.1,20H OFFSET LOCATIONS IS,
800 X
SUM=0.0
CNT=0.0
RETURN
IF(ITST-NE,IBLNK) GOTO 610
IF(CNT-NE,0.0) GOTO 201
GOTO 500
SUM=SUM+PK
CNT=INT+1.0
GOTO 500
END
SUBROUTINE MULTPH(I,REFC,EL,BIAS,SC2)
DIMENSION C(3),S(3),B(3)
DATA
*,AK/-0.6931471806/
*,SQR2/1.414213562/
*,B/0.0244346,0.0314159,0.0785398/
*,C/0.0132557,0.0185345,0.0478558/
*,S/0.00872665,0.0104720,0.0244346/
BW = B(I)
CAL = C(I)
SQ = S(I)
DIR = EXP(AK*(SQ/BW)**2)
EL2 = 2.*EL
RRL = EXP(AK*((EL2+SQ)/BW)**2)
DRU1S = EXP(AK*(DIR+REFC*RU)**2)
DRU2S = EXP(AK*(DIR+REFC*RU)**2)
DRL1S = EXP(AK*(DIR+REFC*RL)**2)
DRL2S = EXP(AK*(DIR+REFC*RL)**2)
DIF1 = (DRU1S-DRL1S)
SUM1 = (DRU1S+DRL1S)

```

```

CIF2 = (DRU2S-DRL2S)
SUM2 = (DRU2S+DRL2S)
SIGER1 = DIF1/SUM1
SIGER2 = DIF2/SUM2
ANGER1 = CAL*SIGER1
ANGER2 = CAL*SIGER2-ANGER1/2.
PPBY2 = ANGER1+PPBY2
BIAS = ANGER2/SQRT2
SD = SD*SD
RETURN
ENDROUTINE ECM1
SUBROUTINE RGB(3), PRW(3), FREQ(3), IRTYP(4), RAOISE(3)
DIMENSION TABJ(37,37), TABX(37,37)
DIMENSION /BLOCK/ IITILE(10)
COMMON /HEADFO/ LINE, NUMBER
COMMON
CCECM1/ IREC, IJ, GAINJ, IX, XSEC, CALX, PJW,
X, Y, Z, ROL, PIT, HDG,
FTGT, FJAM, GJ, SGT, SN
**
**
NAMELIST/NAME1/RGB, PRW, FREQ, IRTYP, I, RG, WL, FTGT, FJAM, PJW
DATA
* RGB/40., 38.5, 28./
* PRW/10500., 17500., 25000./
* FREQ/15.1E9, 9.3805E9, 2.838E9/
* IRTYP/1, 2, 2, 3/
* PI4/12.56637061/
* RNOISE/-123.0, -130.6, -132.2/

C
I = IRTYP(IREC)
RG = 10.*(RGB(1)/10.)
WL = 2.998E8/FREQ(I)
RN = RNOISE(I)
FTGT = PRW(I)*RG*WL*WL/PI4/PI4
CALL PAGES(3, 0, I, PJ)
IF(IJ.EQ.0) WRITE(6, 9003) GAINJ
FORMAT(/, *, JAMMER ANTENNA GAIN*, F7.3, * DB*)
IF(IJ.NE.0) WRITE(6, 9004)
FORMAT(/, *, JAMMER TABLE SPECIFIED*)
CALL PAGES(3, 0, I, PJ)
IF(IX.EQ.0) WRITE(6, 9005) XSEC SECTION *, F9.2, * SQ.METERS*
FORMAT(/, *, AIRCRAFT CROSS SECTION *, F9.2, * SQ.METERS*)
IF(IX.NE.0) WRITE(6, 9006) CALX SECTION TABLE SPECIFIED. *, F9.2)
FORMAT(/, *, AIRCRAFT CROSS SECTION TABLE SPECIFIED. *, F9.2)
** PRINTED VALUES WILL BE MULTIPLIED BY CALX. CALX= *, F9.2)

```

```

C
C
IF(IJ.EQ.0) GO TO 1
JAMMER TABLE
CALL TABLR(TABJ,37)
DO 3 I=1,37
DO 3 J=1,37
TABJ(I,J) = 10.**((TABJ(I,J)/10.)
CCONTINUE
1 GJ = 10.**((GAINJ/10.)
IF(IX.EQ.0) GO TO 2
C
C
X-SECTION TABLE
CALL TABLR(TABX,37)
2 RETURN
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C
ENTRY ECM2
C
NAMELIST/NAM2/ X,Y,Z,ROL,PIT,HDG,
*CXI,CYI,CZI,CX2,CY2,CZ2,AZ,EL,GAINJ,XSEC,D2,SJ,ST,SJT
*,GJ
C
IF(IX.EQ.0 .AND. IJ.EQ.0) GO TO 5
CALL DIRCOS(X,Y,Z,0,0,0,CX1,CY1,CZ1)
CALL CARROT(CX1,CY1,CZ1,ROL,PIT,HDG,CX2,CY2,CZ2)
CALL RECSPH(CX2,CY2,CZ2,AZ,EL)
IF(IJ.EQ.0) GO TO 6
CALL INTRP(TABJ,AZ,EL,37,GJ)
IF(IX.EQ.0) GO TO 5
CALL INTRP(TABX,AZ,EL,37,XSEC)
XSEC = XSEC*CALX
D2 = DIST2(X,Y,Z,0,0,0.)
5 SJ = FJAM*GJ/D2
ST = FTGT*XSEC/D2/D2
SN = 10.*ALOG10(ST)-RN
SJT = 10.*ALOG10(SJ/ST)
RETURN
END
SUBROUTINE JAMER2 (IRAD,AJS,SDSQ)
DIMENSION AJS1(3),SD1(3)
DIMENSION AJS2(4),SD2(4)
DIMENSION AJS3(2),SD3(2)
DIMENSION AJS4(4),SD4(4)
GC TO (1,2,3,4), IRAD
DATA

```

```

* N1/3/ -5. 6. 9. /
* ,AJS1/ 0.002963, 0.01185, 0.1374 /
* ,SD1/ INT2(N1,AJS1,SD1,AJS,SD)
1 GC TO 5
DATA
* N2/4/ -2. 6. 10. 16. /
* ,AJS2/ 0.002563, 0.009877, 0.08278, 0.1374 /
* ,SD2/ INT2(N2,AJS2,SD2,AJS,SD)
2 GC TO 5
DATA
* N3/2/ 15. 30. /
* ,AJS3/ 0.0018, 0.01441 /
* ,SD3/ INT2(N3,AJS3,SD3,AJS,SD)
3 GC TO 5
DATA
* N4/4/ -2. 6. 10. 16. /
* ,AJS4/ 0.006914, 0.02173, 0.1024, 0.1374 /
* ,SD4/ INT2(N4,AJS4,SD4,AJS,SD)
4 SCSQ = SD*SD
5 RETURN
END
SUBROUTINE INT2(NVAL,X,Y,XVAL,YVAL)
DIMENSION X(NVAL),Y(NVAL)
YVAL = Y(1)
IF (X(1)-XVAL) 4,4,3
DO 1 I=1,NVAL
IF (X(I)-XVAL) 1,1,2
1 CONTINUE
YVAL = Y(NVAL)
GO TO 3
2 YVAL = Y(I-1) + (Y(I)-Y(I-1)) / (X(I)-X(I-1)) * (XVAL-X(I-1))
3 RETURN
END
SUBROUTINE JAMER1(PLEN,SDSQ)
SDR = PLEN*0.6826/2.*2.998E8
SDSQ = SDR*SDR
RETURN
END
SUBROUTINE TABLR(TABX,IDIM)
C SUBROUTINE TO READ AND PRINT A TABLE CCF UP TO 37 X 37 ELEMENTS
C THE PROGRAM PROVIDES A DEFAULT VALUE FOR ELEMENTS OUTSIDE THE
C DEFINED TABLE
C INPUTS ARE:

```

```

C CARD VARIABLE FORMAT DEFINITION IDENTIFICATION
C 1 INAME 8A10 ARBITRARY ELEMENTS
C 2 NAZ 15 NO. OF AZ ELEMENTS (ASSUMING ELEMENTS EL GOES-ELEND TO +ELEND
C 3+ TABX(IDIM,1) 8F10.0 DATA TABLE NO. OF ELEMENTS, EL GOES-ELEND TO +ELEND
C 4 NEL 15 NO. OF ELEMENTS, EL GOES-ELEND TO +ELEND
C 5 ELEND F10.4 MAXIMUM ENTRY EL(DEG)
C 6 AZEND F10.4 MAXIMUM ENTRY AZ(DEG)
C 7 DEFALT F10.4 DEFAULT VALUE
C 8 TABX(IDIM,1) 8F10.0 DATA TABLE
C 9 COMMON/TABLES/ELO,DELAZ,DELEL,JEL
C 10 DIMENSION TABX(IDIM,1),INAME(8)
C 11 DATA LE,LZ/2HEL,2HAZ/
C 12 DATA AZO,CDTR/O,.,.0174533/
C 13 READ(5,99) INAME
C 14 FORMAT(8A10)
C 15 FCFORMAT(215,3F10.4)
C 16 IMPLIED INCREMENT
C 17 DELAZ=AZEND/(NAZ-1)
C 18 DELEL=(2.*ELEND)/(NEL-1)
C 19 LOCATE FIRST ELEVATION ENTRY ETC
C 20 JEL=(IDIM-NEL)/2 +1
C 21 MEL=JEL+NEL-1
C 22 ELO=-ELEND
C 23 WRITE(6,101)INAME,NAZ,AZO,AZEND,DELAZ,
C 24 LZ,NEL,ELO,ELEND,DELEL,LE,DEFALT
C 25 101 FORMAT(*1, TABLE DATA*,/,1X,8A10,/,2(1X,15,
C 26 1* ELEMENTS FROM *,F10.2,* TO *,F10.2,* BY *
C 27 2F10.2,2X,A2,/),* ELSEWHERE TABLE IS*,F10.2)
C 28 INSERT DEFAULT
C 29 MAZ=((180./DELAZ)+1
C 30 IF ((MAZ.GT.37).OR.(MEL.GT.37)) GG TO 999
C 31 DO 8 IAZ=1,MAZ
C 32 DC 8 IEL=1,IDIM
C 33 TABX(IEL,IAZ)=DEFALT
C 34 8 TABX(I,J)=JEL,MEL
C 35 DO 14 I=1,NEL
C 36 READ(5,102) (TABX(I,J),J=1,NAZ)
C 37 CONTINUE
C 38 14 FORMAT(8F10.0)
C 39 IDELAZ=DELAZ
C 40 NPAGE=NAZ/13+1
C 41 DO 20 LP=1,NPAGE
C 42 JH1=(LP-1)*13+1
C 43 JH2=JH1+12*IDELAZ
C 44 KH1=(LP-1)*13+1
C 45 KH2=KH1+12

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```

IF(LP.EQ.NPAGE)KH2=MAZ
IF(LP.EQ.NPAGE)JH2=180
WRITE(6,106)(JH,JH=JF1,JH2,IDE LAZ)
FORMAT(106) RCS MATRIX*,/*, *ELEV *,1319)
106 JELU=MAXO(JEL-1,1)
MELO=MINO(MEL+1,ICIM)
ELPT=ELO-DELEL*(JEL-JELO)
C PRINT TABLE
DO 15 J=JELC,MELO
WRITE(6,104)ELPT,(TABX(J,K),K=KH1,KH2)
FORMAT(104)F7.1,2X,13F9.2)
104 ELPT=ELPT+DELEL
C CONTINUE
20 CONTINUE
C CONVERT TO RADIAN
DELEL=DELEL*CDTR
DELAZ=DELAZ*CDTR
ELO=ELO*CDTR
RETURN
999 CONTINUE
105 WRITE(6,105)
FORMAT(105) >>> ERROR IN INPUT <<<<*)
STOP
SUBROUTINE CIRCOS(X1,Y1,Z1,X2,Y2,Z2,COSA,COSB,COSG)
XD = X2-X1
YD = Y2-Y1
ZD = Z2-Z1
DCSA = SQRT(XD*XD+YD*YD+ZD*ZD)
COSB = XD/D
COSG = YD/D
RETURN
END
SUBROUTINE CARROT(X1,Y1,Z1,ROL,PIT,HDG,X2,Y2,Z2)
C CARROT CARTESIAN ROTATION
C HEADING X = X1*COS(HDG) + Y1*SIN(HDG)
Y = -X1*SIN(HDG) + Y1*COS(HDG)
Z = Z1
C PITCH XX = X*COS(PIT)
YY = Y
ZZ = X*SIN(PIT) + Y
Z1 -Z*SIN(PIT)
Z*COS(PIT)
C ROLL X2 = XX
Y2 = YY
Z2 = -YY*SIN(ROL) + YY*COS(ROL)
+ ZZ*COS(ROL) + ZZ*SIN(ROL)

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```

RETURN
END
SUBROUTINE RECSPH(X,Y,Z,PHI,THE)
THE = ACOS(Z/SQRT(X*X+Y*Y+Z*Z))
THE = THE - 1.5708
PHI = 0.0
SB = SQRT(X*X+Y*Y)
IF(SB .NE. C.O) PHI = X/SB
PHI = ACOS(PHI)
PHI = ABS(PHI)
RETURN
END
SUBROUTINE INTRP(TAB,AZ,EL,NVAL,VALUE)
COMMON/TABLES/ELO,DELAZ,DELEL,JEL
DIMENSION TAB(NVAL,1)
A = ABS(AZ)
E = EL
AAZ = A/DELAZ+1.
IAZ = AAZ
IEL = (E-ELO)/DELEL+JEL
IEL = IEL
IAZ = MINO(MAXO(IAZ,1),36)
IEL = MINO(MAXO(IEL,1),36)
V1 = TAB(IEL,IAZ)
V2 = TAB(IEL,IAZ+1)
V3 = TAB(IEL+1,IAZ)
V4 = TAB(IEL+1,IAZ+1)
S = AAZ-IAZ
V12 = V1+(V2-V1)*S
V34 = V3+(V4-V3)*S
S = EEL-IEL
VALUE = V12+(V34-V12)*S
RETURN
END
FUNCTION DIST2(X1,Y1,Z1,X2,Y2,Z2)
XC = X2-X1
YC = Y2-Y1
ZC = Z2-Z1
DIST2 = XC*XC+YC*YC+ZC*ZC
RETURN
END

```

BIBLIOGRAPHY

1. Antiaircraft Artillery Simulation Computer Program - AFTAL Program P001 Vol. I User Manual, Air Force Armament Laboratory, Eglin AFB, Florida (Joint Aircraft Attrition Program Advanced Planning Group, September 1973).
2. Antiaircraft Artillery Simulation Computer Program - AFATL Program P001 Program Update, Joint Technical Coordinating Group for Aircraft Survivability (Survivability Assessment Subgroup, April 1976).
3. M. E. Ramaccia, ATS Working Paper No. 9, Calspan Modification to Antiaircraft Artillery Simulation, AFATL Program P001 (Calspan Corporation, Buffalo, New York, 11 August 1977).
4. G. Gary Maxwell, The Development of Class Problems for a Course in Aircraft Combat Survivability (Naval Postgraduate School Master's Thesis, 1978).

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