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A COUPLED INTERIOR BALLISTICS-FINITE ELEMENT COMBUSTION INSTABI--ETC(U)

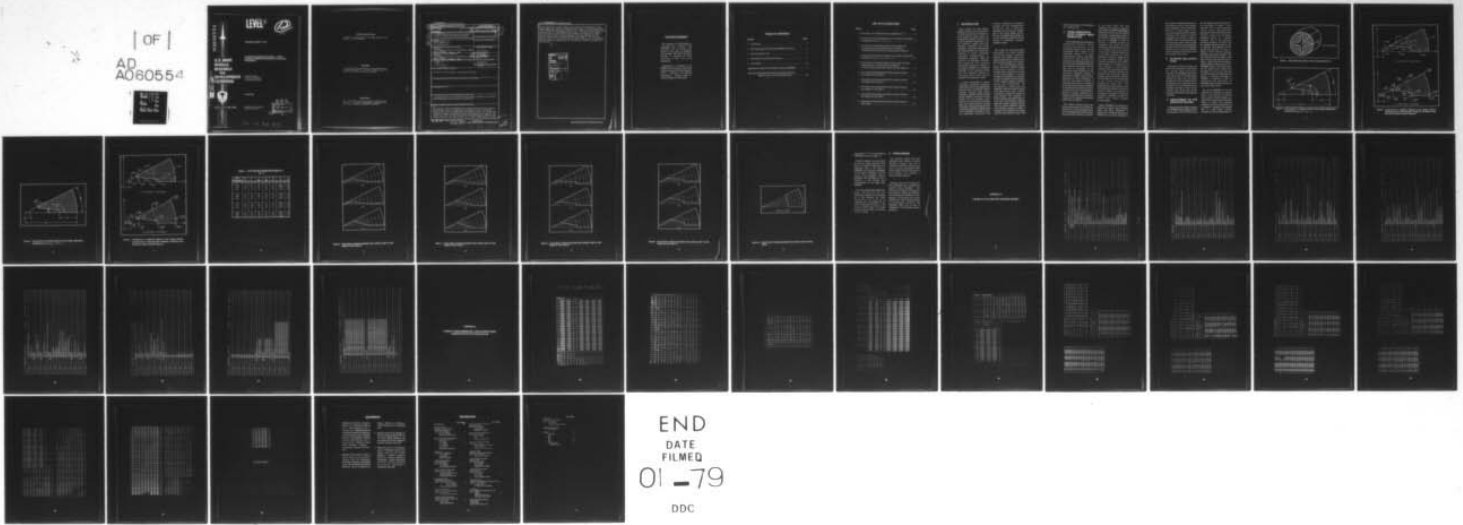
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Redstone Arsenal, Alabama 35809

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TECHNICAL REPORT T-78-72

**A COUPLED INTERIOR BALLISTICS — FINITE  
ELEMENT COMBUSTION INSTABILITY ANALYSIS  
PROCEDURE**

Robert M. Hackett  
Propulsion Directorate  
Technology Laboratory

14 July 1978

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surface regression, with the input to the existing three-dimensional finite element combustion instability prediction code, FLAP3. The three-dimensional finite element mesh and boundary conditions are generated from the grain surface regression data for the progressive burn times. The entire finite element mesh and boundary condition generation by FLESH3, the companion to FLAP3, is executed with the input of seven parameters, which are obtained from the ballistics code output or from the initial grain geometry. The use of the developed routine, in conjunction with the two existing codes, is demonstrated through a number of example cases of a star design, along with the case of a shell design.

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## **ACKNOWLEDGMENT**

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Appreciation is expressed to the MIRADCOM Propulsion Directorate Redstone Arsenal, the Army Research Office - Durham, and Battelle Laboratories - Durham Office for financial support of this project.

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## 1. INTRODUCTION

Until recently the term Interior Ballistics referred to the computation of head end pressure and thrust in solid propellant rocket motors by quasi-steady analysis. However, with recently developed technology<sup>1,2</sup>, it is now possible to predict, given the necessary propellant and nozzle admittance and response parameters, the linear stability of a rocket motor subjected to small, combustion amplified, disturbances, i.e., combustion instability. With the advent of low signature constraints, this capability has assumed considerable importance because of the high probability of combustion instability in low signature motors and the attendant jeopardy to successful motor (and system) operation that instability creates.

A major shortcoming of current combustion instability prediction technology is that it is computationally divorced from conventional interior ballistic analysis. This makes instability analysis difficult because interior ballistic data must be hand loaded into the stability codes. Moreover, it has fostered a functional split, with interior ballistic analysis the domain of the grain designer while combustion instability analysis is that of the "instability specialist". Due to the difficulties associated with

running combustion instability analyses and the aforementioned functional split, combustion instability has not been an integral part of the motor design process. Clearly, a successful motor design for any low signature application should include a combustion instability prediction aspect.

The remedy for this shortcoming would seem to be obvious: combine conventional interior ballistics analysis and stability codes into a single complete interior ballistics code that the motor designer can employ, i.e., a code that performs pressure, thrust, and linear stability margin versus time analyses virtually simultaneously. In this way, it is easy for the motor designer to consider combustion instability. It is with this concept in mind that the development of the computer program described herein was undertaken. Its development is based upon the utilization of the previously developed three-dimensional finite element combustion instability prediction code, FLAP3<sup>2</sup>. The utilization of FLAP3 requires automatic finite element mesh and boundary condition generation, and the coupling of FLAP3 with an interior ballistics analysis code, therefore, requires, in effect, a coupling of the ballistics code output with the mesh generator input. This

process is the essence of a development of a combined code.

## 2. THREE-DIMENSIONAL FINITE ELEMENT MESH GENERATION

FLAP3 (Fluid Analysis Program, 3 Dimensions) performs a linear acousto-modal analysis of the irrotational motions of an inviscid, compressible fluid coupled to the irrotational motions of a nearly incompressible, linearly viscoelastic solid, and a linear potential flow analysis of the irrotational motions of an inviscid, incompressible fluid, and then determines the effect of the flow field and of combustion on the calculated acoustic oscillations<sup>2,3</sup>. This combustion instability analysis is performed at different points in time, beginning at initial combustion, or time zero. The output of FLAP3 is modal frequency and an evaluation of the pressure growth/decay coefficient  $\alpha$  for each mode of vibration analyzed. A positive net value of  $\alpha$  indicates a growth of pressure oscillations and, therefore, instability while a negative value of  $\alpha$  is an indication of decaying oscillations, or stability.

The FLAP3 analysis utilizes the finite element method and models the acoustic cavity and propellant grain as an assemblage of three-dimensional quasi-hexahedral elements connected

at the corner nodes. The mesh generation code used in conjunction with FLAP3 is FLESH3 (Fluid Mesh, 3 Dimensions) which was developed for the purpose of generating input data for FLAP3. A detailed description of the use of FLESH3 is found in Reference 3. FLESH3 generates the numbered nodal points and nodal coordinates and identifies each node as to whether it lies in the acoustic cavity, on the cavity-grain interface, or in the grain; generates the quasi-hexahedral elements, designated by the eight numbered nodal points defining each element, and identifies each element as to whether it is a cavity element, a cavity element adjacent to the cavity-grain interface, or a solid propellant element; and generates the cavity-grain interface element surfaces (burning surfaces) and identifies each as to the direction of the surface normal. This information, along with a few additional input data relative to gas and propellant properties and type of acoustic mode(s) to be analyzed, is necessary and sufficient to activate a combustion instability analysis by FLAP3, given adequate computational facilities.

Input to FLESH3 is in the form of a definition of global curves, defining cavity and grain boundaries; a designation of blocks of points, as to whether they are cavity, interface or

grain points; a designation of blocks to be generated, by their nodal indices; and a designation of the number and location of longitudinal cross-sections, defining the number of layers of elements. The objective is that of minimizing the amount of input to FLESH3. This can be done by the use of an executive routine which accepts as input a small number of geometrical parameters from the interior ballistics code and, in turn, activates FLESH3. The development of this routine is the major thrust of this report.

### 3. INTERIOR BALLISTICS CODE

The interior ballistics analysis code used in the formulation was developed by Aerojet Solid Propulsion Company<sup>4</sup>. The FLESH3 geometrical input parameters are, therefore, those presented in that code, but similar parameters would be obtained from a consideration of a comparable interior ballistics code. The description of the developed executive routine will be related to the Aerojet code, but the procedure followed in the development is general.

### 4. DEVELOPMENT OF THE EXECUTIVE ROUTINE

A listing of the formulated computer code GRNMSH (Grain Mesh) is found in Appendix A. The code is based upon

the star design found in *Reference 4* and reproduced in *Figure 1*. One computer card of input parameters (seven values) automatically generates the three-dimensional finite element mesh and boundary conditions needed for a FLAP3 analysis. Due to the dihedral symmetry provision in FLAP3, only the smallest repeating segment need be analyzed. This segment, for the general star geometry, is shown in *Figures 2 and 4*; with the necessary geometry parameters identified. The only difference between *Figure 2* and *Figure 4* is that the angle  $\Phi$  is zero in *Figure 4*. This condition necessitated a slightly different internal provision in GRNMSH. *Figures 3 and 5* indicate: (a) the key global curves (others were generated, as can be seen from the included example), and (b) the key part indices. A working knowledge of FLESH3<sup>3</sup> is necessary in order to understand the intricacies of GRNMSH, and to modify and add to it, but not in order to use it.

The use of GRNMSH can best be understood by following an example. *Table 1*, from *Reference 4*, lists the twelve cases of the star design upon which the formulation of the executive routine is based. The generated meshes for the twelve cases are shown in *Figures 6 through 9*. The special case of a shell design (circular cylinder), for which the parameter  $\Phi$  is

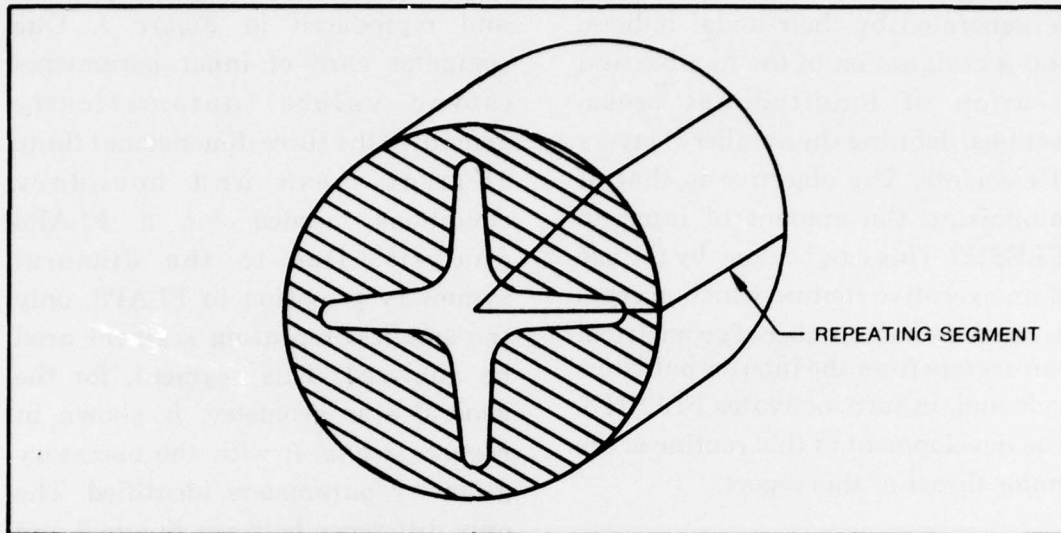


Figure 1. Star design with outside round on propellant tip;  $N=4$

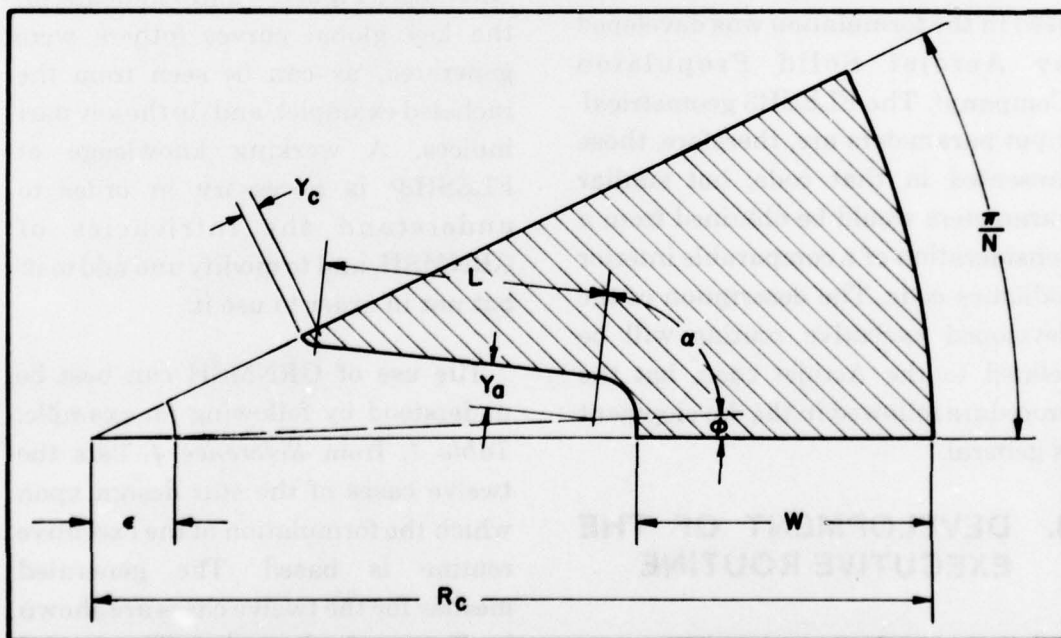


Figure 2. Cross-section of repeating segment of star design; independent parameters:  $N, R_c, w, \phi, Y_a, \alpha, L$

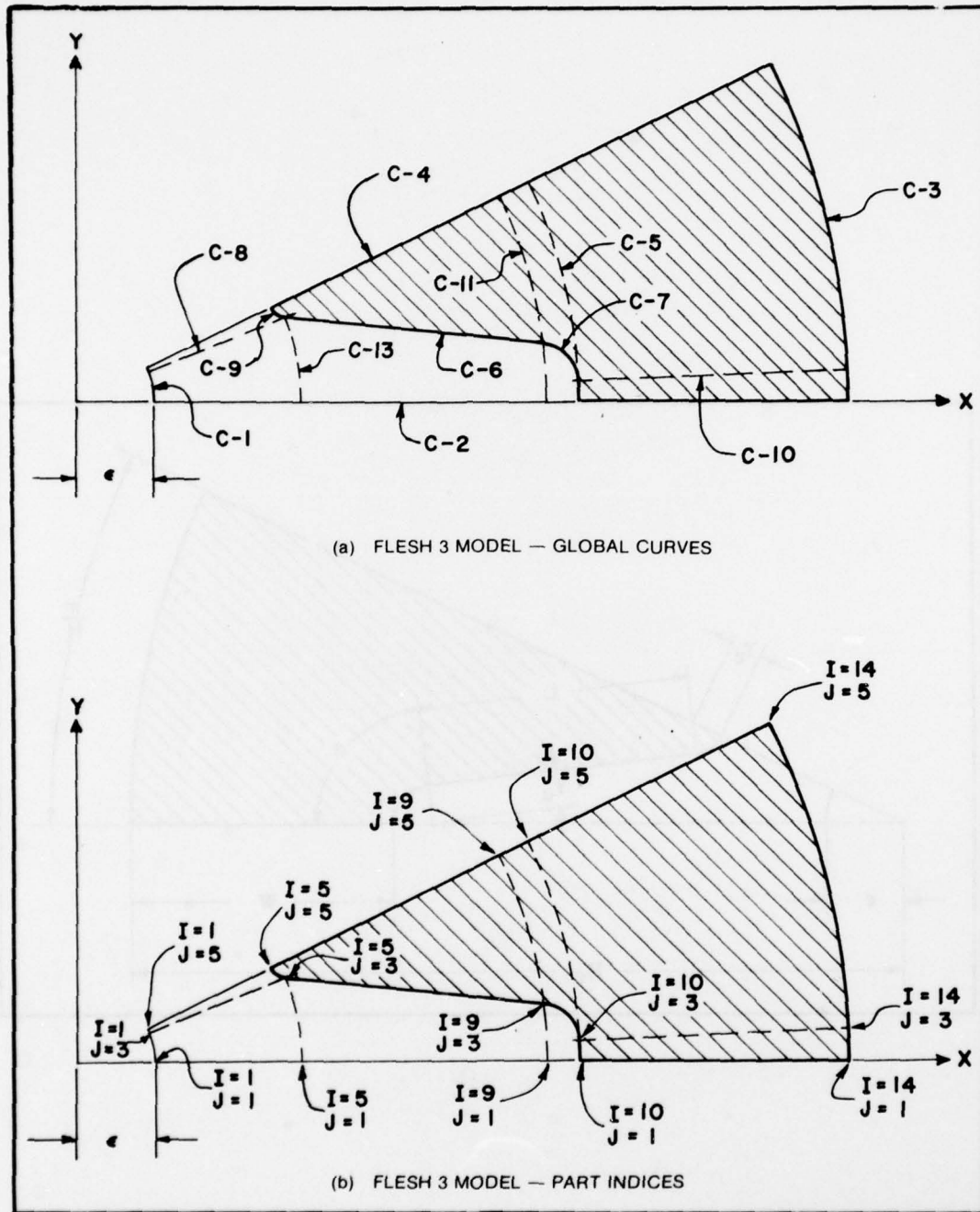


Figure 3. Cross-section of repeating segment of star design showing important input to mesh generator (FLESH 3): (a) Global curves and (b) Part indices (matches Figure 2).

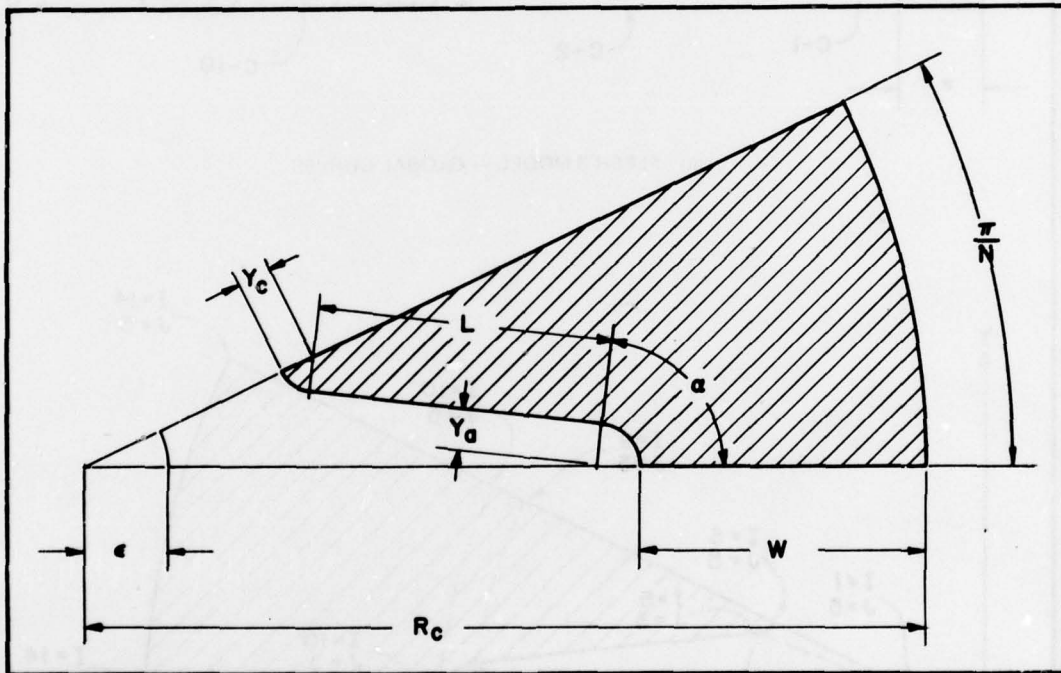


Figure 4. Cross-section of repeating segment of star design; independent parameters:  $N, R_c, w, Y_a, \alpha, L; \Phi = 0$

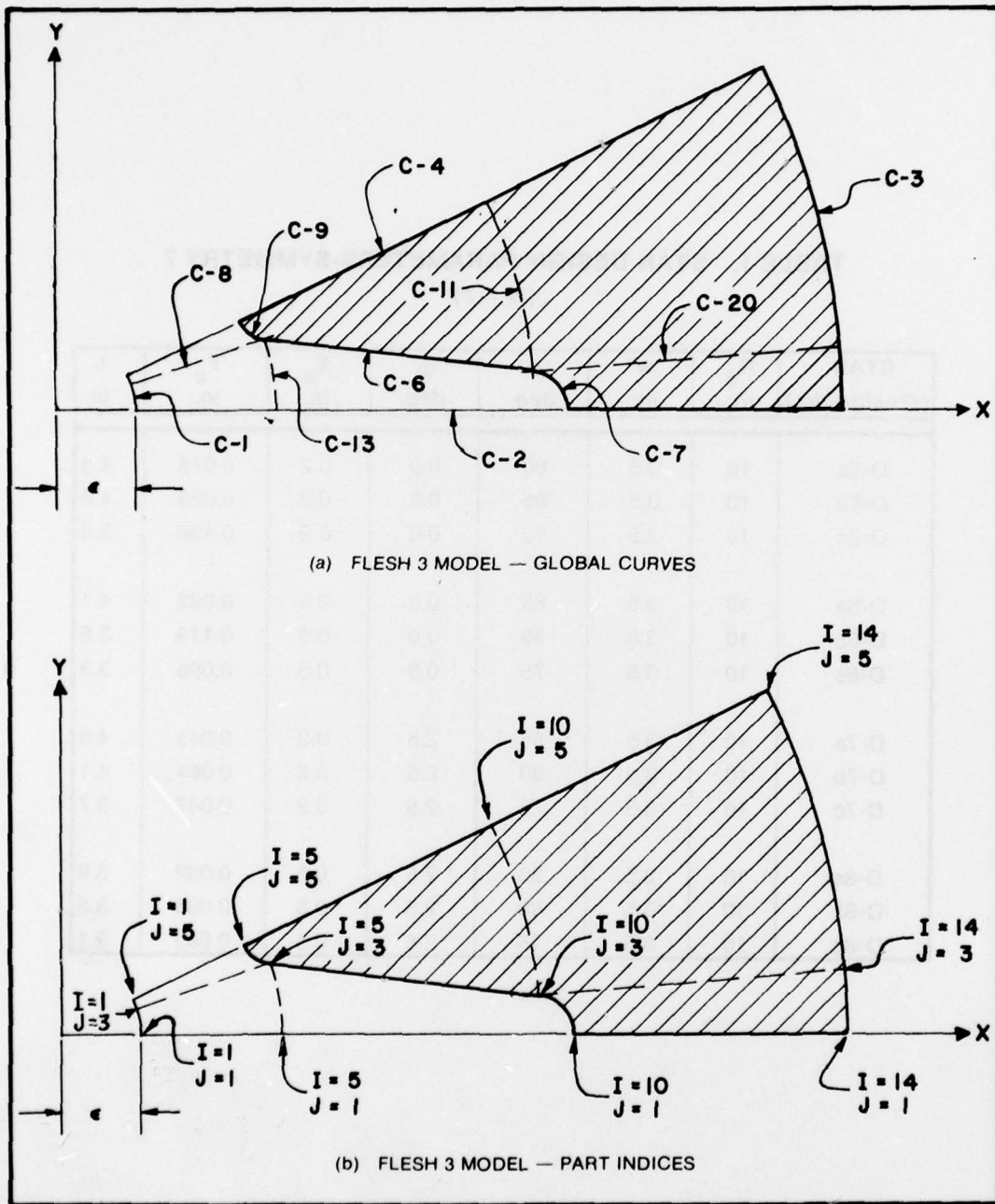
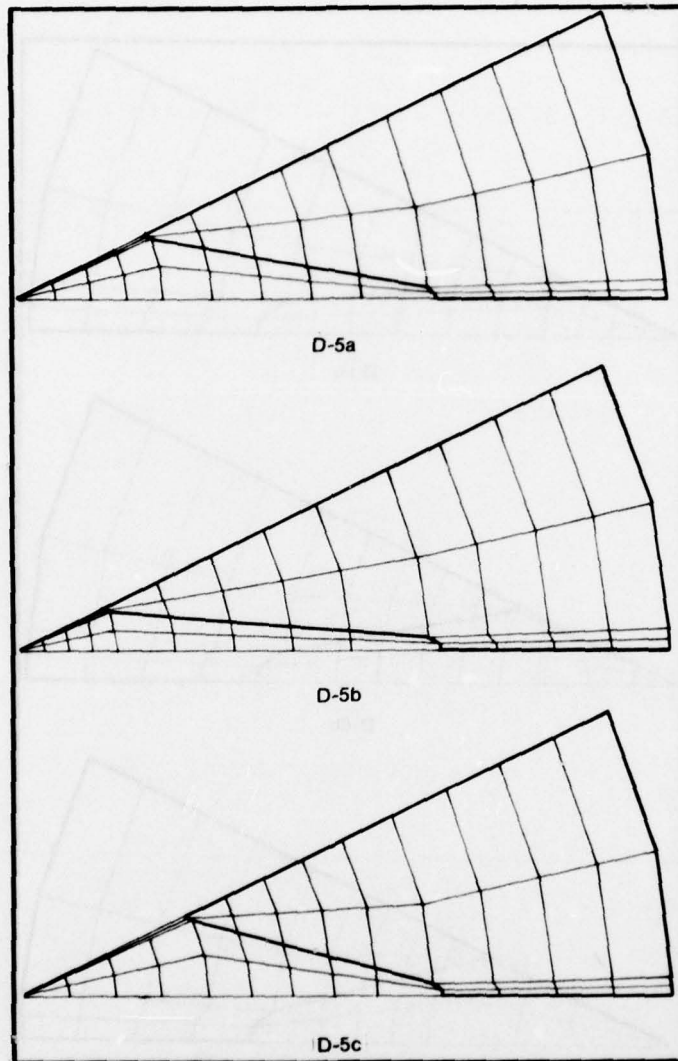


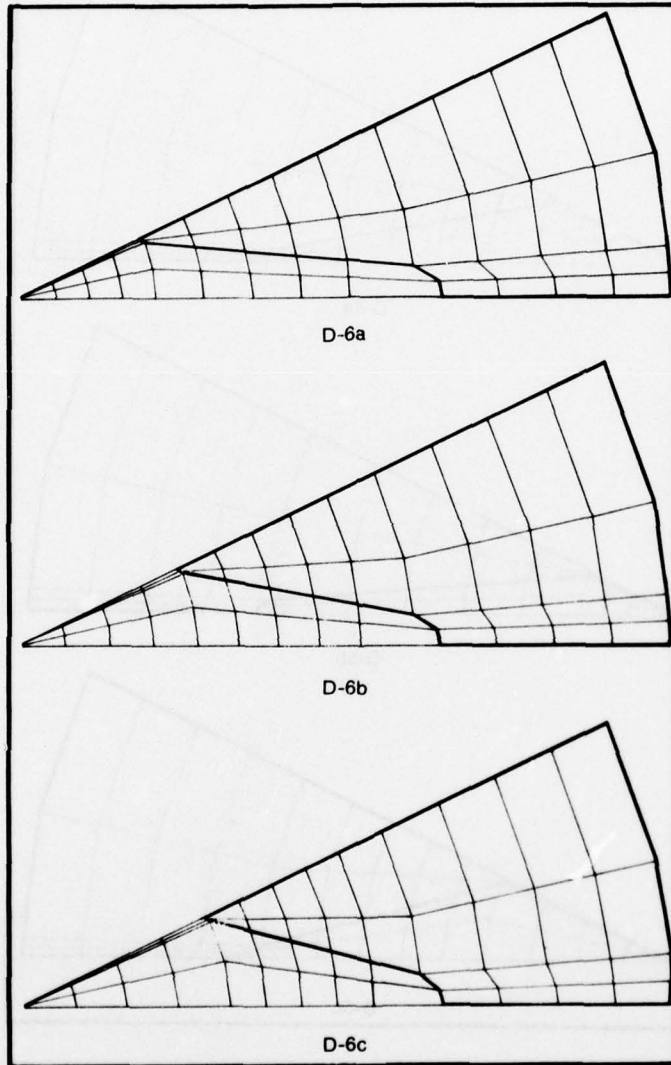
Figure 5. Cross-section of repeating segment of star design showing important input to mesh generator (FLESH3): (a) Global curves and (b) Part indices (matches Figure 4).

**TABLE 1. STAR DESIGN PARAMETERS-SYMMETRY 7**  
(N = 7)

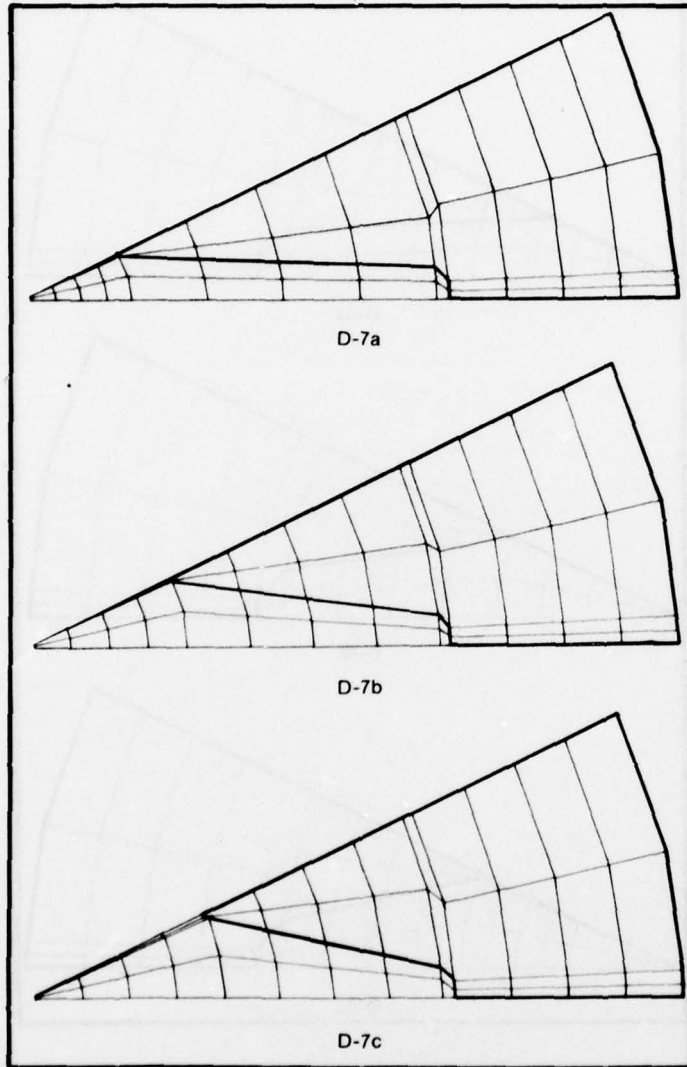
STAR (Design No.)	R <sub>c</sub> in.	w in.	α deg	Φ deg	Y <sub>a</sub> in.	Y <sub>c</sub> in.	L in.
D-5a	10	3.5	80	0.0	0.2	0.075	4.3
D-5b	10	3.5	85	0.0	0.2	0.068	4.9
D-5c	10	3.5	75	0.0	0.2	0.136	3.8
D-6a	10	3.5	85	0.0	0.5	0.092	4.1
D-6b	10	3.5	80	0.0	0.5	0.118	3.6
D-6c	10	3.5	75	0.0	0.5	0.095	3.3
D-7a	10	3.5	85	2.5	0.2	0.043	4.8
D-7b	10	3.5	80	2.5	0.2	0.084	4.1
D-7c	10	3.5	75	2.5	0.2	0.047	3.7
D-8a	10	3.5	85	2.5	0.5	0.092	3.9
D-8b	10	3.5	80	2.5	0.5	0.035	3.5
D-8c	10	3.5	75	2.5	0.5	0.069	3.1



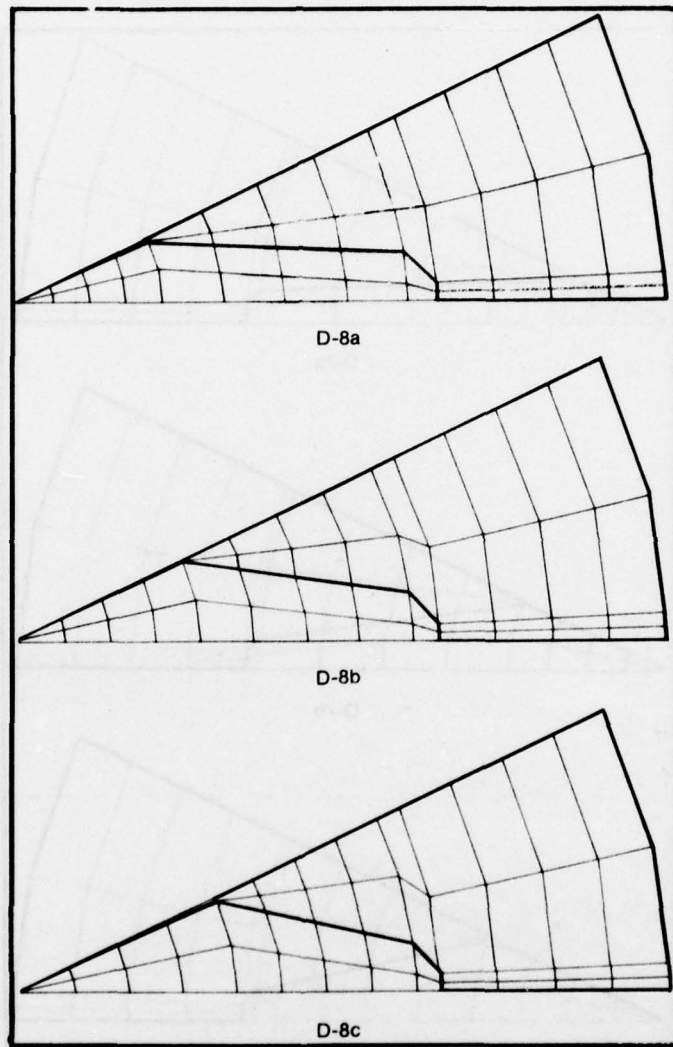
**Figure 6. Cross-section showing generated finite element mesh for star design D-5 (see Table 1).**



**Figure 7. Cross-section showing generated finite element mesh for star design D-6 (see Table 1).**



**Figure 8. Cross-section showing generated finite element mesh for star design D-7 (see Table 1).**



**Figure 9. Cross-section showing generated finite element mesh for star design D-8 (see Table 1).**

CONCLUSIONS

An extensive program has been developed which couples an interior ballistics analysis code with a computer capability analysis code. This is an important step toward more realistic and thorough exterior design analysis.

Basically, the point of interest for the analysis is the point of interest for the analysis. The program is written in FORTRAN and is being used to analyze the design of the gun barrel. The program is being used to analyze the design of the gun barrel. The program is being used to analyze the design of the gun barrel.

The program is being used to analyze the design of the gun barrel. The program is being used to analyze the design of the gun barrel. The program is being used to analyze the design of the gun barrel. The program is being used to analyze the design of the gun barrel. The program is being used to analyze the design of the gun barrel.

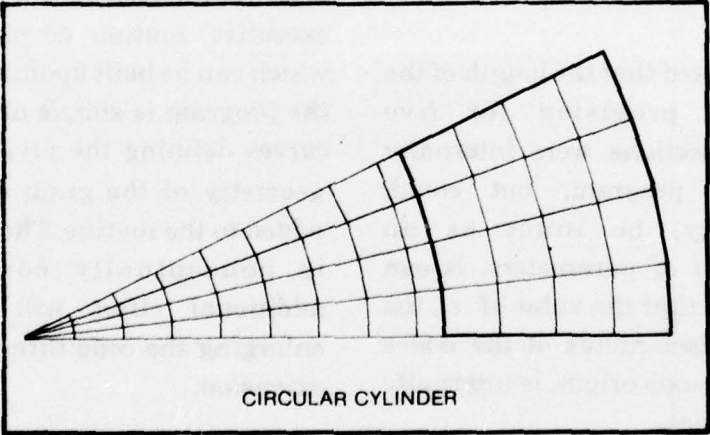


Figure 10. Cross-section showing generated finite element mesh for shell design.

input equal to  $\pi/N$ , as generated by GRNMSH is shown in *Figure 10*.

Sample computer input and output for the star design geometry D-5a (*Table 1*) is found in Appendix B. In sequential order is: the input to GRNMSH, the output from GRNMSH (or the input to FLESH 3), and the output from FLESH3. Again, familiarity with FLESH3, by way of *Reference 3*, will enhance one's understanding of the output from FLESH3.

It can be noted that the length of the grain, and provision for five equidistant sections were internally set in the program, but could, alternatively, be input as an additional set of parameters. It can also be noted that the value of  $\epsilon$ , the necessary offset radius of the nodes closest to the axis origin, is internally set at 0.1 inches.

## 5. CONCLUSIONS

An executive routine has been developed which couples an interior ballistics analysis code with a combustion instability analysis code; thus, in concept, providing the rocket motor designer with the potential for a more inclusive and, therefore, a superior design approach.

Basically, the proof of concept has been established and a fundamental executive routine developed — one which can be built upon. The format of the program is simple and additional curves defining the progressive burn geometry of the grain can be easily added to the routine. The development is conceptually complete; the additional effort will be that of enlarging the code through repetitive extension.

**APPENDIX A**

**LISTING OF THE COMPUTER PROGRAM GRNMSH**







```

172      I=22
      XN(I)=215
      XPALE=PI*AC
      175      AA(I)=SLOV*(AA(19)-AA(I))*0.75
      XN(I)=AA(I)
      TME(I)=TME*AD
      215      CONTINUE
      I=23
      180      GB=AA(I)*S*H-I*AC(I)*S*H*(PHI+0.0)
      XN(I)=GB/COS(PHI+0.5)-AA(I)*CPHI
      SLOPE=GB/H
      X(I)=AA(I)*CPHI-0.1
      X(I)=AA(I)*S*H-I*0.1*SLOPE4
      I=24
      185      X(I)=AA(I)*COS(PHI+0.5)*0.1
      X(I)=AA(I)*S*H*(PHI+0.5)-I*SLOPE4
      XN(I)=X
      217      CONTINUE
      I=25
      XN(I)=17
      XPALE=PI*AC
      AA(I)=AA(19)*(AA(19)-AA(I))*0.75
      XN(I)=AA(I)
      TME(I)=TME*AD
      218      CONTINUE
      I=27
      X(I)=SC
      X(I)=RC*TAN(PHI+0.5)
      XN(I)=X
      219      CONTINUE
      220      CONTINUE
      I=28
      XN(I)=X*(VA*SIN(ALPHA)/(RC+VA*VA-COS(ALPHA)))-C*VA*S*PHI+VA*S*FAP
      SLOPE=VV/VA
      X(I)=RC*VA*CPHI+VA*CFAP-0.1
      X(I)=RC*VA*S*PHI+VA*S*FAP-0.1*SLOPE5
      I=29
      X(I)=SC
      X(I)=RC*(VA+SIN(ALPHA)/(RC+VA*VA-COS(ALPHA)))
      XN(I)=X
      221      CONTINUE
      222      CONTINUE
      I=31
      X(I)=SC
      X(I)=RC*(VA+SIN(ALPHA)/(RC+VA*VA-COS(ALPHA)))
      XN(I)=X
      223      CONTINUE
      224      CONTINUE
      I=35
      XN(I)=X*(10)*(VA+2)*(1.0+0.0000)-(X(I)*2)*0.0000**0.5/(1.0+0.0000)
      ZZZ=X*(10)*X*(10)*X
      VV=AA(I)*S*H*(TME+0.5)*222
      XN(I)=VV/XXX
      SLOPE=VV/XXX
      X(I)=AA(19)*COS(THETA+0.5)-0.1

```

```

230      Y(1)=AA(19)*SIN(THETA*0.5)*0.1*SLOPE6
        I=33
        X(1)=MM*0.1
        Y(1)=ZZ*0.1*SLOPE5
        NMLI=24
235      CONTINUE
246      CONTINUE
        I=35
        SLOPE7=(Y(11)-YC*SAPT)/(X(11)-YC*CAPT)
        X(1)=X(11)-YC*CAPT*0.1
        Y(1)=Y(11)-YC*SAPT*0.1
        NMLI=24
240      CONTINUE
247      CONTINUE
248      CONTINUE
        I=36
        X(1)=X(11)-YC*CAPT*0.1
        Y(1)=Y(11)-YC*SAPT*0.1*TAN(THETA*0.25)
        I=37
        Y(1)=Y(11)-YC*SAPT*(1A(16)-X(11)+YC*CAPT)*TAN(THETA*0.25)
        X(1)=AA(16)
245      CONTINUE
        NMLI=24
250      CONTINUE
253      CONTINUE
        I=34
        AD=Y(13)-YC*CAPT
        AE=Y(13)-YC*SAPT
        AF=AA(16)
255      AG=TAN(THETA*0.25)
        AX=AG**2*0.1*0
        AY=2*0*0*0*0*0*0
        AZ=AD**2+AE**2+AF**2+0
        AV=(AY*(AZ**2+0+0+0+0+0+0+0+0+0.5)/(2.0*0.1))
        CW=AA(7)*SIN(THETA*0.5)-(Y(11)-YC*SAPT*0.1)
        DW=AA(7)*COS(THETA*0.5)-(X(11)-YC*CAPT*0.1)
260      SLOPE8=CW/DW
        X(1)=X(11)-YC*CAPT*0.1
        Y(1)=Y(11)-YC*SAPT*0.1*SLOPE8
        I=39
        X(1)=AA(7)*COS(THETA*0.5)*0.1
        Y(1)=AA(7)*SIN(THETA*0.5)*0.1*SLOPE8
        NMLI=30
270      CONTINUE
281      CONTINUE
282      CONTINUE
        I=40
        X(1)=RC*0.25
        Y(1)=RC*0.25*TAN(THETA*0.5)
275      CONTINUE
        I=41
        X(1)=RC
        Y(1)=RC*TAN(THETA*0.5)
        NMLI=32
280      CONTINUE
284      CONTINUE
        I=43
        X(1)=RC
        Y(1)=RC*TAN(THETA*0.25)
285      CONTINUE
    
```

```

235 CONTINUE
236 CONTINUE
      I=55
      X(I)=RC
      Y(I)=C*TA(X*THETA)+C*O
      N(I)=36
237 CONTINUE
238 CONTINUE
      I=57
      X(I)=RC
      Y(I)=C*TA(X*THETA)+C*O
      N(I)=36
239 CONTINUE
240 CONTINUE
      I=58
      AREA=1/2*(SIN(THETA)+SIN(THETA+O))-((Y(I)-Y(CASPT))
      SLOPE=C*A=ZAC
      X(I)=X(I)+Y(CASPT)-O*A
      I=59
      X(I)=A(16)*COS(THETA)+O*(1+O)*O
      Y(I)=A(16)*SIN(THETA)+O*(1+O)*O
      N(I)=40
      599 CONTINUE
      DO 331 I=1,54
      X(I)=C*O*(SIN(I*THETA)+SIN(I*THETA+O))-((Y(I)-Y(CASPT))
      SLOPE=C*A=ZAC
      X(I)=X(I)+Y(CASPT)-O*A
      I=59
      X(I)=A(16)*COS(THETA)+O*(1+O)*O
      Y(I)=A(16)*SIN(THETA)+O*(1+O)*O
      N(I)=40
      331 CONTINUE
      5331 FOR=ATGCFIQ*.215*.9F10.61
      I=61
      I=ZERO
      I=61
      I=62
      I=63
      I=64
      I=65
      I=66
      I=67
      I=68
      I=69
      I=70
      I=71
      I=72
      I=73
      I=74
      I=75
      I=76
      I=77
      I=78
      I=79
      I=80
      I=81
      I=82
      I=83
      I=84
      I=85
      I=86
      I=87
      I=88
      I=89
      I=90
      I=91
      I=92
      I=93
      I=94
      I=95
      I=96
      I=97
      I=98
      I=99
      I=100

```





**APPENDIX B**  
**EXAMPLE THREE-DIMENSIONAL FINITE ELEMENT MESH**  
**GENERATION FOR STAR DESIGN (D-5A)**

7.0	10.0	3.5	0.0	0.2	80.0	4.3
-----	------	-----	-----	-----	------	-----

GRAIN MESH													
14	5	29	7	1.00	5								
0.0000	0.0000	1	25	.1000	.1000	0.0000	25.7143						
0.0000	0.0000	0	0	0.0000	0.0000	0.0000	0.0000						
10.0000	0.0000	2	0	0.0000	0.0000	0.0000	0.0000						
0.0000	0.0000	1	25	10.0000	10.0000	0.0000	25.7143						
0.0000	0.0000	0	0	0.0000	0.0000	0.0000	0.0000						
10.0000	0.0000	4	0	0.0000	0.0000	0.0000	0.0000						
0.0000	0.0000	5	25	6.5000	6.5000	0.0000	25.7143						
2.0000	0.0000	0	0	0.0000	0.0000	0.0000	0.0000						
6.4347	.1793	6	0	0.0000	0.0000	0.0000	0.0000						
6.3000	0.0000	7	50	.2000	.2000	0.0000	25.7143						
0.0000	0.0000	0	0	0.0000	0.0000	0.0000	0.0000						
2.2001	.3000	8	0	0.0000	0.0000	0.0000	0.0000						
2.1131	1.0176	9	54	.0751	.0751	204.7143	251.0000						
0.0000	0.0000	0	0	0.0000	0.0000	0.0000	0.0000						
10.0000	0.0000	10	0	0.0000	0.0000	0.0000	0.0000						
0.0000	0.0000	11	25	6.3378	6.3378	0.0000	25.7143						
0.0000	0.0000	0	0	0.0000	0.0000	0.0000	0.0000						
10.0000	2.2824	12	0	0.0000	0.0000	0.0000	0.0000						
0.0000	0.0000	13	25	2.3023	2.3023	0.0000	25.7143						
2.1000	.5200	0	0	0.0000	0.0000	0.0000	0.0000						
6.4378	-.0125	14	0	0.0000	0.0000	0.0000	0.0000						
0.0000	0.0000	15	25	1.7517	1.7517	0.0000	25.7143						
6.2378	0.0000	0	0	0.0000	0.0000	0.0000	0.0000						
6.6000	0.0000	16	0	0.0000	0.0000	0.0000	0.0000						
0.0000	0.0000	17	25	5.3289	5.3289	0.0000	25.7143						
0.0000	0.0000	0	0	0.0000	0.0000	0.0000	0.0000						
10.0000	0.0000	18	0	0.0000	0.0000	0.0000	0.0000						
6.2347	.1834	0	0	0.0000	0.0000	0.0000	0.0000						
10.0000	.3109	20	0	0.0000	0.0000	0.0000	0.0000						
0.0000	0.0000	0	0	0.0000	0.0000	0.0000	0.0000						
10.0000	.1555	22	0	0.0000	0.0000	0.0000	0.0000						
2.1000	.5221	0	0	0.0000	0.0000	0.0000	0.0000						
6.5728	.0909	24	0	0.0000	0.0000	0.0000	0.0000						
0.0000	0.0000	0	0	0.0000	0.0000	0.0000	0.0000						
2.1022	.9947	26	0	0.0000	0.0000	0.0000	0.0000						
1.9022	.9379	0	0	0.0000	0.0000	0.0000	0.0000						
6.3378	1.4286	28	0	0.0000	0.0000	0.0000	0.0000						
6.0700	1.3000	0	0	0.0000	0.0000	0.0000	0.0000						
6.4370	1.4689	30	0	0.0000	0.0000	0.0000	0.0000						
2.5000	.5700	0	0	0.0000	0.0000	0.0000	0.0000						
10.0000	2.2824	32	0	0.0000	0.0000	0.0000	0.0000						
0.0000	0.0000	0	0	0.0000	0.0000	0.0000	0.0000						
10.0000	1.1267	34	0	0.0000	0.0000	0.0000	0.0000						
0.0000	0.0000	0	0	0.0000	0.0000	0.0000	0.0000						
10.0000	2.2824	36	0	0.0000	0.0000	0.0000	0.0000						
0.0000	0.0000	0	0	0.0000	0.0000	0.0000	0.0000						
10.0000	5.4992	38	0	0.0000	0.0000	0.0000	0.0000						
1.9022	.9379	0	0	0.0000	0.0000	0.0000	0.0000						
6.2709	1.4215	40	0	0.0000	0.0000	0.0000	0.0000						
1	1	1	4	5									
2	9	1	9	2									
2	5	3	5	5									
2	6	3	9	5									
3	10	1	10	5									
3	6	4	14	5									
3	11	1	14	5									
3	1	1	5	2	2	13	12	1	1	0	0	0	0
3	1	2	5	3	12	13	8	1	1	0	0	0	0
3	1	3	4	4	8	15	26	1	1	0	0	0	0
3	1	4	4	5	26	15	8	1	1	0	0	0	0
3	5	1	9	2	2	17	24	13	1	0	0	0	0
3	4	3	5	4	4	9	26	15	2	0	0	0	0

7	4	4	5	5	26	9	4	15	2	0	0	0	0
8	5	2	9	3	24	17	6	13	2	0	0	0	0
9	9	1	10	2	2	7	24	17	2	0	0	0	0
10	3	2	10	3	24	7	6	17	2	0	0	0	0
11	5	3	9	4	6	17	40	9	3	0	0	0	0
12	5	4	9	5	40	17	4	9	3	0	0	0	0
13	9	3	10	4	6	11	40	17	3	0	0	0	0
14	9	4	10	5	40	11	4	17	3	0	0	0	0
15	10	1	14	2	2	3	22	7	3	0	0	0	0
16	10	2	14	3	22	3	20	7	3	0	0	0	0
17	10	3	14	4	20	3	32	11	3	0	0	0	0
18	10	4	14	5	32	3	4	11	3	0	0	0	0
-1	0	0	0	0	0	0	0	0	0	0	0	0	0
A.0000													
1	1	1	5	2	2	13	12	1	1	0	0	0	0
2	1	2	5	3	12	13	8	1	1	0	0	0	0
3	1	3	4	4	8	15	26	1	1	0	0	0	0
4	1	4	4	5	26	15	4	1	1	0	0	0	0
5	5	1	9	2	2	17	24	13	1	0	0	0	0
6	4	3	5	4	8	9	26	15	2	0	0	0	0
7	4	4	5	5	26	9	4	15	2	0	0	0	0
8	5	2	9	3	24	17	6	13	2	0	0	0	0
9	9	1	10	2	2	7	24	17	2	0	0	0	0
10	9	2	10	3	24	7	6	17	2	0	0	0	0
11	5	3	9	4	6	17	40	9	3	0	0	0	0
12	5	4	9	5	40	17	4	9	3	0	0	0	0
13	9	3	10	4	6	11	40	17	3	0	0	0	0
14	9	4	10	5	40	11	4	17	3	0	0	0	0
15	10	1	14	2	2	3	22	7	3	0	0	0	0
16	10	2	14	3	22	3	20	7	3	0	0	0	0
17	10	3	14	4	20	3	32	11	3	0	0	0	0
18	10	4	14	5	32	3	4	11	3	0	0	0	0
-1	0	0	0	0	0	0	0	0	0	0	0	0	0
A.0000													
1	1	1	5	2	2	13	12	1	1	0	0	0	0
2	1	2	5	3	12	13	8	1	1	0	0	0	0
3	1	3	4	4	8	15	26	1	1	0	0	0	0
4	1	4	4	5	26	15	4	1	1	0	0	0	0
5	5	1	9	2	2	17	24	13	1	0	0	0	0
6	4	3	5	4	8	9	26	15	2	0	0	0	0
7	4	4	5	5	26	9	4	15	2	0	0	0	0
8	5	2	9	3	24	17	6	13	2	0	0	0	0
9	9	1	10	2	2	7	24	17	2	0	0	0	0
10	9	2	10	3	24	7	6	17	2	0	0	0	0
11	5	3	9	4	6	17	40	9	3	0	0	0	0
12	5	4	9	5	40	17	4	9	3	0	0	0	0
13	9	3	10	4	6	11	40	17	3	0	0	0	0
14	9	4	10	5	40	11	4	17	3	0	0	0	0
15	10	1	14	2	2	3	22	7	3	0	0	0	0
16	10	2	14	3	22	3	20	7	3	0	0	0	0
17	10	3	14	4	20	3	32	11	3	0	0	0	0
18	10	4	14	5	32	3	4	11	3	0	0	0	0
-1	0	0	0	0	0	0	0	0	0	0	0	0	0
A.0000													
1	1	1	5	2	2	13	12	1	1	0	0	0	0
2	1	2	5	3	12	13	8	1	1	0	0	0	0
3	1	3	4	4	8	15	26	1	1	0	0	0	0
4	1	4	4	5	26	15	4	1	1	0	0	0	0
5	5	1	9	2	2	17	24	13	1	0	0	0	0
6	4	3	5	4	8	9	26	15	2	0	0	0	0
7	4	4	5	5	26	9	4	15	2	0	0	0	0
8	5	2	9	3	24	17	6	13	2	0	0	0	0
9	9	1	10	2	2	7	24	17	2	0	0	0	0
10	9	2	10	3	24	7	6	17	2	0	0	0	0
11	5	3	9	4	6	17	40	9	3	0	0	0	0
12	5	4	9	5	40	17	4	9	3	0	0	0	0
13	9	3	10	4	6	11	40	17	3	0	0	0	0
14	9	4	10	5	40	11	4	17	3	0	0	0	0
15	10	1	14	2	2	3	22	7	3	0	0	0	0
16	10	2	14	3	22	3	20	7	3	0	0	0	0
17	10	3	14	4	20	3	32	11	3	0	0	0	0
18	10	4	14	5	32	3	4	11	3	0	0	0	0
-1	0	0	0	0	0	0	0	0	0	0	0	0	0
A.0000													
1	1	1	5	2	2	13	12	1	1	0	0	0	0
2	1	2	5	3	12	13	8	1	1	0	0	0	0
3	1	3	4	4	8	15	26	1	1	0	0	0	0
4	1	4	4	5	26	15	4	1	1	0	0	0	0
5	5	1	9	2	2	17	24	13	1	0	0	0	0
6	4	3	5	4	8	9	26	15	2	0	0	0	0
7	4	4	5	5	26	9	4	15	2	0	0	0	0
8	5	2	9	3	24	17	6	13	2	0	0	0	0
9	9	1	10	2	2	7	24	17	2	0	0	0	0
10	9	2	10	3	24	7	6	17	2	0	0	0	0
11	5	3	9	4	6	17	40	9	3	0	0	0	0
12	5	4	9	5	40	17	4	9	3	0	0	0	0

15	0	5	10	4	6	11	40	17	5	0	0	0	0	0
14	9	4	10	5	40	11	4	17	5	0	0	0	0	0
15	10	1	14	2	2	5	22	7	5	0	0	0	0	0
16	10	2	14	3	22	5	20	7	5	0	0	0	0	0
17	10	5	14	4	20	5	32	11	5	0	0	0	0	0
18	10	4	14	5	32	5	4	11	5	0	0	0	0	0
-1	0	0	0	0	0	0	0	0	0	0	0	0	0	0
6*0000														
1	1	1	5	2	2	15	12	1	1	0	0	0	0	0
2	1	2	5	3	12	15	8	1	1	0	0	0	0	0
3	1	3	4	4	8	15	26	1	1	0	0	0	0	0
4	1	4	4	5	26	15	4	1	1	0	0	0	0	0
5	1	9	2	2	17	24	15	1	0	0	0	0	0	0
6	4	3	5	4	5	9	26	15	2	0	0	0	0	0
7	4	4	5	4	26	9	4	15	2	0	0	0	0	0
8	5	2	9	3	24	17	6	15	2	0	0	0	0	0
9	9	1	10	2	2	7	24	17	2	0	0	0	0	0
10	2	2	10	3	24	7	6	17	2	0	0	0	0	0
11	5	3	9	4	6	17	40	9	5	0	0	0	0	0
12	5	4	9	5	40	17	4	9	5	0	0	0	0	0
13	9	3	10	4	6	17	40	17	5	0	0	0	0	0
14	5	4	10	5	40	17	4	17	5	0	0	0	0	0
15	10	1	14	2	2	5	22	7	5	0	0	0	0	0
16	10	2	14	3	22	5	20	7	5	0	0	0	0	0
17	10	5	14	4	20	5	32	11	5	0	0	0	0	0
18	10	4	14	5	32	5	4	11	5	0	0	0	0	0
-1	0	0	0	0	0	0	0	0	0	0	0	0	0	0
6*0000														
END														

CARD I. TITLE.  
GREEN WASH

CARD II. CONTROL.  
INCH UNITS MNC IBC SCALE NLAZ  
14 5 34 7 1.

CARD III. PART BOUNDARY CURVE.

X - COORD	Y - COORD	MN	NP	A	X-AXIS	B	Y-AXIS	THETA1	THETA2
0.000	0.000	1	25		1.0000		1.0000	0.00000	25.71430
0.000	0.000	0	0		0.00000		0.00000	0.00000	0.00000
10.000	0.000	2	0		0.00000		0.00000	0.00000	0.00000
0.000	0.000	4	25		10.00000		10.00000	0.00000	25.71430
0.000	0.000	0	0		0.00000		0.00000	0.00000	0.00000
10.000	4.416	4	0		0.00000		0.00000	0.00000	0.00000
0.000	0.000	5	25		6.50000		6.50000	0.00000	25.71430
0.000	4.416	0	0		0.00000		0.00000	0.00000	0.00000
6.435	4.179	6	0		0.00000		0.00000	0.00000	0.00000
6.300	0.000	7	0		2.0000		2.0000	0.00000	80.00000
0.000	0.000	0	0		0.00000		0.00000	0.00000	0.00000
2.200	4.416	8	0		0.00000		0.00000	0.00000	0.00000
2.115	1.010	9	54		0.7510		0.7510	204.71430	261.00000
0.000	0.000	0	0		0.00000		0.00000	0.00000	0.00000
10.000	0.000	10	0		0.00000		0.00000	0.00000	0.00000
0.000	0.000	11	25		6.33780		6.33780	0.00000	25.71430
0.000	0.000	7	0		0.00000		0.00000	0.00000	0.00000
10.000	2.242	12	0		0.00000		0.00000	0.00000	0.00000
0.000	0.000	13	25		2.30230		2.30230	0.00000	25.71430
2.145	4.525	0	0		0.00000		0.00000	0.00000	0.00000
5.434	-0.113	14	0		0.00000		0.00000	0.00000	0.00000
0.000	0.000	15	25		1.75170		1.75170	0.00000	25.71430
5.232	0.000	0	0		0.00000		0.00000	0.00000	0.00000
5.600	0.000	16	0		0.00000		0.00000	0.00000	0.00000
0.000	0.000	17	25		5.32890		5.32890	0.00000	25.71430
0.000	0.000	0	0		0.00000		0.00000	0.00000	0.00000
10.000	0.000	18	0		0.00000		0.00000	0.00000	0.00000
6.230	4.183	0	0		0.00000		0.00000	0.00000	0.00000
10.000	4.111	20	0		0.00000		0.00000	0.00000	0.00000
0.000	0.000	0	0		0.00000		0.00000	0.00000	0.00000
10.000	4.156	22	0		0.00000		0.00000	0.00000	0.00000
2.145	4.222	0	0		0.00000		0.00000	0.00000	0.00000
5.573	4.031	24	0		0.00000		0.00000	0.00000	0.00000
0.000	0.000	0	0		0.00000		0.00000	0.00000	0.00000
2.182	4.225	26	0		0.00000		0.00000	0.00000	0.00000
1.782	4.238	0	0		0.00000		0.00000	0.00000	0.00000
6.338	1.429	28	0		0.00000		0.00000	0.00000	0.00000
6.079	1.188	0	0		0.00000		0.00000	0.00000	0.00000
6.437	1.469	30	0		0.00000		0.00000	0.00000	0.00000
2.500	4.571	0	0		0.00000		0.00000	0.00000	0.00000
10.000	2.282	32	0		0.00000		0.00000	0.00000	0.00000
0.000	0.000	0	0		0.00000		0.00000	0.00000	0.00000
10.000	1.127	34	0		0.00000		0.00000	0.00000	0.00000
0.000	0.000	0	0		0.00000		0.00000	0.00000	0.00000
10.000	2.282	36	0		0.00000		0.00000	0.00000	0.00000
0.000	0.000	0	0		0.00000		0.00000	0.00000	0.00000
10.000	3.499	38	0		0.00000		0.00000	0.00000	0.00000
1.982	4.238	0	0		0.00000		0.00000	0.00000	0.00000
4.279	1.422	40	0		0.00000		0.00000	0.00000	0.00000

CARD IV. NODE CODE SEQUENCE.

IC	I1	J1	I2	J2
1	1	1	4	5
1	5	1	9	2
2	5	3	5	5
2	5	3	9	4
3	10	1	10	1
3	1	4	14	5
3	11	1	14	3

CARD V. PART DEFINITION.

NP	T1	J1	T2	J2	L1	L2	L3	L4	MT	NH	I3	J3	IK
1	1	1	5	2	2	13	12	1	1	0	0	0	0
2	1	2	5	3	12	13	8	1	1	0	0	0	0
3	1	3	4	4	8	15	26	1	1	0	0	0	0
4	1	4	4	5	26	15	9	1	1	0	0	0	0
5	5	1	9	2	2	17	24	13	1	0	0	0	0
6	4	3	5	4	8	9	26	15	2	0	0	0	0
7	4	4	5	5	26	9	4	15	2	0	0	0	0
8	5	2	9	3	24	17	6	13	2	0	0	0	0
9	9	1	10	2	2	7	24	17	2	0	0	0	0
10	9	2	10	3	24	7	6	17	2	0	0	0	0
11	5	3	9	4	5	17	40	9	3	0	0	0	0
12	5	4	9	5	40	17	4	9	3	0	0	0	0
13	9	3	10	4	6	11	40	17	3	0	0	0	0
14	9	4	10	5	40	11	4	17	3	0	0	0	0
15	10	1	14	2	2	3	22	7	3	0	0	0	0
16	10	2	14	3	22	3	20	7	3	0	0	0	0
17	10	3	14	4	20	3	32	11	3	0	0	0	0
18	10	4	14	5	32	3	4	11	3	0	0	0	0
-1	0	0	0	0	0	0	0	0	0	0	0	0	0

CARD VI. SECTION LOCATION.

Z = 0.000

0.000

	X	Y	Z	MT
1	.10000	0.00000	0.00000	1
2	.65057	0.00000	0.00000	1
3	1.20113	0.00000	0.00000	1
4	1.75173	0.00000	0.00000	1
5	2.30230	0.00000	0.00000	1
6	2.85285	0.00000	0.00000	1
7	3.40340	0.00000	0.00000	1
8	3.95395	0.00000	0.00000	1
9	4.50450	0.00000	0.00000	1
10	5.05505	0.00000	0.00000	2
11	5.60560	0.00000	0.00000	3
12	6.15615	0.00000	0.00000	3
13	6.70670	0.00000	0.00000	3
14	7.25725	0.00000	0.00000	3
15	7.80780	.02225	0.00000	1
16	8.35835	.14476	0.00000	1
17	8.90890	.26727	0.00000	1
18	9.45945	.38977	0.00000	1
19	10.00999	.51228	0.00000	1
20	10.56054	.63479	0.00000	1
21	11.11109	.75730	0.00000	1
22	11.66164	.87980	0.00000	1
23	12.21219	.12245	0.00000	1
24	12.76274	.10065	0.00000	2
25	13.31329	.11436	0.00000	3
26	13.86384	.12807	0.00000	3
27	14.41439	.14177	0.00000	3
28	14.96494	.15548	0.00000	3
29	15.51549	.16919	0.00000	1
30	16.06604	.18290	0.00000	1
31	16.61659	.19660	0.00000	1

CARD V. PART DEFINITION.

	NP	11	J1	12	J2	L1	L2	L3	L4	WT	MN	I3	J5	IR
1	1	1	1	1	1	1	1	1	1	1	1	1	1	1
2	1	1	2	5	2	2	13	12	1	1	0	0	0	0
3	1	1	3	4	4	8	15	26	1	1	0	0	0	0
4	1	1	4	4	4	5	26	15	4	1	1	0	0	0
5	5	1	9	2	2	17	24	13	1	1	0	0	0	0
6	4	1	5	2	2	17	24	15	2	0	0	0	0	0
7	4	2	5	5	5	26	9	4	15	2	0	0	0	0
8	5	2	10	2	2	23	17	6	13	2	0	0	0	0
9	2	2	10	3	2	23	7	6	17	2	0	0	0	0
10	2	2	10	3	2	23	7	6	17	2	0	0	0	0
11	5	3	9	4	6	17	40	9	3	0	0	0	0	0
12	5	4	9	5	4	17	4	9	3	0	0	0	0	0
13	9	3	10	4	6	11	40	17	3	0	0	0	0	0
14	9	4	10	5	4	11	4	17	3	0	0	0	0	0
15	10	1	14	2	2	2	22	7	3	0	0	0	0	0
16	10	2	14	3	2	2	22	7	3	0	0	0	0	0
17	10	3	14	4	2	0	32	11	3	0	0	0	0	0
18	10	4	14	5	3	2	3	4	11	3	0	0	0	0
19	0	0	0	0	0	0	0	0	0	0	0	0	0	0
20	0	0	0	0	0	0	0	0	0	0	0	0	0	0

CARD VI. SECTION LOCATION.

Z = 0.050

Z = 0.000

	X	Y	Z	WT
71	1.0000	0.0000	2.0000	1
72	0.5000	0.0000	2.0000	1
73	1.0000	0.0000	2.0000	1
74	1.75175	0.0000	2.0000	1
75	2.50350	0.0000	2.0000	1
76	3.25525	0.0000	2.0000	1
77	4.00700	0.0000	2.0000	1
78	4.75875	0.0000	2.0000	1
79	5.51050	0.0000	2.0000	1
80	6.26225	0.0000	2.0000	1
81	7.01400	0.0000	2.0000	1
82	7.76575	0.0000	2.0000	1
83	8.51750	0.0000	2.0000	1
84	9.26925	0.0000	2.0000	1
85	10.02100	0.0000	2.0000	1
86	10.77275	0.0000	2.0000	1
87	11.52450	0.0000	2.0000	1
88	12.27625	0.0000	2.0000	1
89	13.02800	0.0000	2.0000	1
90	13.77975	0.0000	2.0000	1
91	14.53150	0.0000	2.0000	1
92	15.28325	0.0000	2.0000	1
93	16.03500	0.0000	2.0000	1
94	16.78675	0.0000	2.0000	1
95	17.53850	0.0000	2.0000	1
96	18.29025	0.0000	2.0000	1
97	19.04200	0.0000	2.0000	1
98	19.79375	0.0000	2.0000	1
99	20.54550	0.0000	2.0000	1
100	21.29725	0.0000	2.0000	1
101	22.04900	0.0000	2.0000	1

32	1.50774	0.71793	0.00000	1
33	2.01548	0.43481	0.00000	2
34	2.52322	0.09152	0.00000	2
35	3.03096	0.60317	0.00000	2
36	3.53870	0.51041	0.00000	2
37	4.04644	0.37666	0.00000	2
38	4.55418	0.18679	0.00000	2
39	5.06192	0.21788	0.00000	3
40	5.56966	0.28883	0.00000	3
41	6.07740	0.29378	0.00000	3
42	6.58514	0.31073	0.00000	3
43	7.09288	0.31848	0.00000	1
44	7.60062	0.26983	0.00000	1
45	8.10836	0.39318	0.00000	1
46	8.61610	0.72653	0.00000	1
47	9.12384	0.93916	0.00000	2
48	9.63158	1.07600	0.00000	3
49	10.13932	1.12285	0.00000	3
50	10.64706	1.20969	0.00000	3
51	11.15480	1.23653	0.00000	3
52	11.66254	1.41022	0.00000	3
53	12.17028	1.61193	0.00000	3
54	12.67802	1.81364	0.00000	3
55	13.18576	2.01535	0.00000	3
56	13.69350	2.21706	0.00000	3
57	14.20124	2.41877	0.00000	3
58	14.70898	2.62048	0.00000	3
59	15.21672	2.82219	0.00000	3
60	15.72446	3.02390	0.00000	3
61	16.23220	3.22561	0.00000	3
62	16.73994	3.42732	0.00000	3
63	17.24768	3.62903	0.00000	3
64	17.75542	3.83074	0.00000	3
65	18.26316	4.03245	0.00000	3
66	18.77090	4.23416	0.00000	3
67	19.27864	4.43587	0.00000	3
68	19.78638	4.63758	0.00000	3
69	20.29412	4.83929	0.00000	3
70	20.80186	5.04100	0.00000	3

CARD V. PART DEFINITION.

	NP	IT	J1	J2	L1	L2	L3	L4	L5	L6	L7	L8	L9	L10	L11	L12	L13	L14	L15	L16	L17	L18	L19	L20
102	1.98774	.71765																						
103	2.03202	.64409																						
104	2.50082	.60132																						
105	3.00172	.55017																						
106	4.51164	.51149																						
107	5.31556	.47565																						
108	6.58841	.44448																						
109	7.25294	.41742																						
110	8.16541	.39448																						
111	9.28046	.37378																						
112	10.50000	.35410																						
113	11.81932	.33639																						
114	13.23202	.32049																						
115	14.73156	.30617																						
116	16.30922	.29315																						
117	17.95600	.28115																						
118	19.66281	.27000																						
119	21.42970	.25964																						
120	23.24700	.25000																						
121	25.11482	.24112																						
122	27.02300	.23287																						
123	28.97121	.22519																						
124	30.95877	.21795																						
125	32.98544	.21112																						
126	34.95000	.20469																						
127	36.85210	.19863																						
128	38.69154	.19297																						
129	40.46700	.18769																						
130	42.18800	.18278																						
131	43.85444	.17822																						
132	45.46600	.17399																						
133	47.02257	.17007																						
134	48.52400	.16644																						
135	49.97000	.16308																						
136	51.36000	.15997																						
137	52.69400	.15709																						
138	53.97200	.15442																						
139	55.20400	.15195																						
140	56.39000	.14966																						

CARD VI. SECTION LOCATION.

	Z	X	Y	Z	X	Y
	4.000					
141	1.0000	0.0000	0.0000	1	0	0
142	6.5057	0.0000	0.0000	1	1	0
143	1.20115	0.0000	0.0000	1	2	0
144	1.75173	0.0000	0.0000	1	3	0
145	2.30230	0.0000	0.0000	1	4	0
146	3.05895	0.0000	0.0000	1	5	0
147	3.81361	0.0000	0.0000	1	6	0
148	4.56725	0.0000	0.0000	1	7	0
149	5.32089	0.0000	0.0000	1	8	0
150	6.07453	0.0000	0.0000	1	9	0
151	6.82817	0.0000	0.0000	1	10	0
152	7.58181	0.0000	0.0000	1	11	0
153	8.33545	0.0000	0.0000	1	12	0
154	9.08909	0.0000	0.0000	1	13	0
155	9.84273	0.0000	0.0000	1	14	0
156	10.59637	0.0000	0.0000	1	15	0
157	11.34901	0.0000	0.0000	1	16	0
158	12.10265	0.0000	0.0000	1	17	0
159	12.85629	0.0000	0.0000	1	18	0
160	13.60993	0.0000	0.0000	1	19	0
161	14.36357	0.0000	0.0000	1	20	0
162	15.11721	0.0000	0.0000	1	21	0
163	15.87085	0.0000	0.0000	1	22	0
164	16.62449	0.0000	0.0000	1	23	0
165	17.37813	0.0000	0.0000	1	24	0
166	18.13177	0.0000	0.0000	1	25	0
167	18.88541	0.0000	0.0000	1	26	0
168	19.63905	0.0000	0.0000	1	27	0
169	20.39269	0.0000	0.0000	1	28	0
170	21.14633	0.0000	0.0000	1	29	0
171	21.89997	0.0000	0.0000	1	30	0

CARD V. PART DEFINITION.

	NP	11	J1	12	J2	L1	L2	L3	L4	MT	NN	13	J3	14	15	16
172	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1
173	2	1	2	1	2	1	2	1	2	1	2	1	2	1	2	1
174	3	1	3	1	3	1	3	1	3	1	3	1	3	1	3	1
175	4	1	4	1	4	1	4	1	4	1	4	1	4	1	4	1
176	5	1	5	1	5	1	5	1	5	1	5	1	5	1	5	1
177	6	1	6	1	6	1	6	1	6	1	6	1	6	1	6	1
178	7	1	7	1	7	1	7	1	7	1	7	1	7	1	7	1
179	8	1	8	1	8	1	8	1	8	1	8	1	8	1	8	1
180	9	1	9	1	9	1	9	1	9	1	9	1	9	1	9	1
181	10	1	10	1	10	1	10	1	10	1	10	1	10	1	10	1
182	11	1	11	1	11	1	11	1	11	1	11	1	11	1	11	1
183	12	1	12	1	12	1	12	1	12	1	12	1	12	1	12	1
184	13	1	13	1	13	1	13	1	13	1	13	1	13	1	13	1
185	14	1	14	1	14	1	14	1	14	1	14	1	14	1	14	1
186	15	1	15	1	15	1	15	1	15	1	15	1	15	1	15	1
187	16	1	16	1	16	1	16	1	16	1	16	1	16	1	16	1
188	17	1	17	1	17	1	17	1	17	1	17	1	17	1	17	1
189	18	1	18	1	18	1	18	1	18	1	18	1	18	1	18	1
190	19	1	19	1	19	1	19	1	19	1	19	1	19	1	19	1
191	20	1	20	1	20	1	20	1	20	1	20	1	20	1	20	1
192	21	1	21	1	21	1	21	1	21	1	21	1	21	1	21	1
193	22	1	22	1	22	1	22	1	22	1	22	1	22	1	22	1
194	23	1	23	1	23	1	23	1	23	1	23	1	23	1	23	1
195	24	1	24	1	24	1	24	1	24	1	24	1	24	1	24	1
196	25	1	25	1	25	1	25	1	25	1	25	1	25	1	25	1
197	26	1	26	1	26	1	26	1	26	1	26	1	26	1	26	1
198	27	1	27	1	27	1	27	1	27	1	27	1	27	1	27	1
199	28	1	28	1	28	1	28	1	28	1	28	1	28	1	28	1
200	29	1	29	1	29	1	29	1	29	1	29	1	29	1	29	1
201	30	1	30	1	30	1	30	1	30	1	30	1	30	1	30	1
202	31	1	31	1	31	1	31	1	31	1	31	1	31	1	31	1
203	32	1	32	1	32	1	32	1	32	1	32	1	32	1	32	1
204	33	1	33	1	33	1	33	1	33	1	33	1	33	1	33	1
205	34	1	34	1	34	1	34	1	34	1	34	1	34	1	34	1
206	35	1	35	1	35	1	35	1	35	1	35	1	35	1	35	1
207	36	1	36	1	36	1	36	1	36	1	36	1	36	1	36	1
208	37	1	37	1	37	1	37	1	37	1	37	1	37	1	37	1
209	38	1	38	1	38	1	38	1	38	1	38	1	38	1	38	1
210	39	1	39	1	39	1	39	1	39	1	39	1	39	1	39	1

CARD VI. SECTION LOCATION.

	Z	Y	X	Z	Y	X	Z	Y	X
211	1	1	1	1	1	1	1	1	1
212	2	1	2	2	1	2	2	1	2
213	3	1	3	3	1	3	3	1	3
214	4	1	4	4	1	4	4	1	4
215	5	1	5	5	1	5	5	1	5
216	6	1	6	6	1	6	6	1	6
217	7	1	7	7	1	7	7	1	7
218	8	1	8	8	1	8	8	1	8
219	9	1	9	9	1	9	9	1	9
220	10	1	10	10	1	10	10	1	10
221	11	1	11	11	1	11	11	1	11
222	12	1	12	12	1	12	12	1	12
223	13	1	13	13	1	13	13	1	13
224	14	1	14	14	1	14	14	1	14
225	15	1	15	15	1	15	15	1	15
226	16	1	16	16	1	16	16	1	16
227	17	1	17	17	1	17	17	1	17
228	18	1	18	18	1	18	18	1	18
229	19	1	19	19	1	19	19	1	19
230	20	1	20	20	1	20	20	1	20
231	21	1	21	21	1	21	21	1	21
232	22	1	22	22	1	22	22	1	22
233	23	1	23	23	1	23	23	1	23
234	24	1	24	24	1	24	24	1	24
235	25	1	25	25	1	25	25	1	25
236	26	1	26	26	1	26	26	1	26
237	27	1	27	27	1	27	27	1	27
238	28	1	28	28	1	28	28	1	28
239	29	1	29	29	1	29	29	1	29
240	30	1	30	30	1	30	30	1	30
241	31	1	31	31	1	31	31	1	31
242	32	1	32	32	1	32	32	1	32
243	33	1	33	33	1	33	33	1	33
244	34	1	34	34	1	34	34	1	34
245	35	1	35	35	1	35	35	1	35
246	36	1	36	36	1	36	36	1	36
247	37	1	37	37	1	37	37	1	37
248	38	1	38	38	1	38	38	1	38
249	39	1	39	39	1	39	39	1	39
250	40	1	40	40	1	40	40	1	40



312	1.59774	.71793	8.00000	1	28	72	77	87	96	142	143	157	156	1
313	2.09822	.84405	8.00000	2	29	73	74	88	87	143	144	158	157	1
314	2.59870	.97018	8.00000	2	30	74	75	89	88	144	145	159	158	1
315	2.70773	1.04017	8.00000	2	31	75	76	90	89	145	146	160	159	1
316	4.51164	1.51441	8.00000	2	32	76	77	91	90	146	147	161	160	1
317	5.31552	1.57444	8.00000	2	33	77	78	92	91	147	148	162	161	1
318	6.13441	1.63447	8.00000	2	34	78	79	93	92	148	149	163	162	1
319	7.25224	1.71448	8.00000	3	35	79	80	94	93	149	150	164	163	2
320	8.16691	1.74448	8.00000	3	36	80	81	95	94	150	151	165	164	1
321	2.38884	1.27378	8.00000	3	37	81	82	96	95	151	152	166	165	1
322	3.49445	1.31077	8.00000	3	38	82	83	97	96	152	153	167	166	1
323	4.19053	1.34143	8.00000	1	39	83	84	98	97	153	154	168	167	1
324	5.39175	1.36483	8.00000	1	40	84	85	99	98	154	155	169	168	2
325	1.00222	1.40218	8.00000	1	41	85	86	100	99	155	156	170	169	2
326	1.53388	1.41653	8.00000	1	42	86	87	101	100	156	157	171	170	2
327	2.12222	1.44214	8.00000	1	43	87	88	102	101	157	158	172	171	1
328	3.38342	1.47300	8.00000	1	44	88	89	103	102	158	159	173	172	2
329	4.52541	1.42225	8.00000	3	45	89	90	104	103	159	160	174	173	1
330	4.39730	1.20964	8.00000	3	46	90	91	105	104	160	161	175	174	1
331	5.16353	1.20251	8.00000	3	47	91	92	106	105	161	162	176	175	2
332	4.17953	1.41222	8.00000	3	48	92	93	107	106	162	163	177	176	2
333	7.07121	1.62127	8.00000	3	49	93	94	108	107	163	164	178	177	2
334	7.96477	1.51745	8.00000	3	50	94	95	109	108	164	165	179	178	2
335	8.38453	2.00137	8.00000	3	51	95	96	110	109	165	166	180	179	1
336	9.74444	2.22509	8.00000	3	52	96	97	111	110	166	167	181	180	2
337	1.09212	1.04139	8.00000	1	53	97	98	112	111	167	168	182	181	1
338	1.54514	1.22227	8.00000	1	54	98	99	113	112	168	169	183	182	1
339	1.08214	1.52115	8.00000	1	55	99	100	114	113	169	170	184	183	1
340	1.57522	1.75003	8.00000	1	56	100	101	115	114	170	171	185	184	1
341	2.05544	1.92102	8.00000	2	57	101	102	116	115	171	172	186	185	1
342	2.73437	1.31475	8.00000	3	58	102	103	117	116	172	173	187	186	2
343	3.42331	1.64556	8.00000	3	59	103	104	118	117	173	174	188	187	1
344	4.11224	1.97637	8.00000	3	60	104	105	119	118	174	175	189	188	1
345	4.80118	2.30718	8.00000	3	61	105	106	120	119	175	176	190	189	2
346	5.49011	2.63799	8.00000	3	62	106	107	121	120	176	177	191	190	1
347	6.17905	2.96880	8.00000	3	63	107	108	122	121	177	178	192	191	1
348	7.45394	3.34433	8.00000	3	64	108	109	123	122	178	179	193	192	1
349	8.14287	3.24156	8.00000	3	65	109	110	124	123	179	180	194	193	1
350	9.00470	4.53480	8.00000	3	66	110	111	125	124	180	181	195	194	2
1	1	16	15	71	72	86	85	1	67	160	161	175	174	2
2	2	17	16	72	73	87	86	1	68	161	162	176	175	2
3	3	18	17	73	74	88	87	1	69	162	163	177	176	2
4	4	19	18	74	75	89	88	1	70	163	164	178	177	2
5	5	20	19	75	76	90	89	1	71	164	165	179	178	1
6	6	21	20	76	77	91	90	1	72	165	166	180	179	1
7	7	22	21	77	78	92	91	1	73	166	167	181	180	1
8	8	23	22	78	79	93	92	1	74	167	168	182	181	2
9	9	24	23	79	80	94	93	2	75	168	169	183	182	1
10	10	25	24	80	81	95	94	1	76	169	170	184	183	1
11	11	26	25	81	82	96	95	1	77	170	171	185	184	1
12	12	27	26	82	83	97	96	1	78	171	172	186	185	1
13	13	28	27	83	84	98	97	1	79	172	173	187	186	2
14	14	29	28	84	85	99	98	1	80	173	174	188	187	1
15	15	30	29	85	86	100	99	2	81	174	175	189	188	1
16	16	31	30	86	87	101	100	2	82	175	176	190	189	1
17	17	32	31	87	88	102	101	2	83	176	177	191	190	1
18	18	33	32	88	89	103	102	2	84	177	178	192	191	1
19	19	34	33	89	90	104	103	2	85	178	179	193	192	1
20	20	35	34	90	91	105	104	2	86	179	180	194	193	1
21	21	36	35	91	92	106	105	2	87	180	181	195	194	1
22	22	37	36	92	93	107	106	2	88	181	182	196	195	1
23	23	38	37	93	94	108	107	2	89	182	183	197	196	1
24	24	39	38	94	95	109	108	2	90	183	184	198	197	1
25	25	40	39	95	96	110	109	1	91	184	185	199	198	1
26	26	41	40	96	97	111	110	1	92	185	186	200	199	1
27	27	42	41	97	98	112	111	1	93	186	187	201	200	1
28	28	43	42	98	99	113	112	1	94	187	188	202	201	1
29	29	44	43	99	100	114	113	1	95	188	189	203	202	1
30	30	45	44	100	101	115	114	1	96	189	190	204	203	1
31	31	46	45	101	102	116	115	1	97	190	191	205	204	1
32	32	47	46	102	103	117	116	2	98	191	192	206	205	1
33	33	48	47	103	104	118	117	1	99	192	193	207	206	1
34	34	49	48	104	105	119	118	1	100	193	194	208	207	1
35	35	50	49	105	106	120	119	1	101	194	195	209	208	1
36	36	51	50	106	107	121	120	1	102	195	196	210	209	1
37	37	52	51	107	108	122	121	1	103	196	197	211	210	1
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