

AD-A069 168

WOODS HOLE OCEANOGRAPHIC INSTITUTION MASS  
RAY CALCULATIONS OF OCEAN SOUND CHANNELS USING A POCKET PROGRAM--ETC(U)  
OCT 78 L BAXTER  
WHOT-78-65

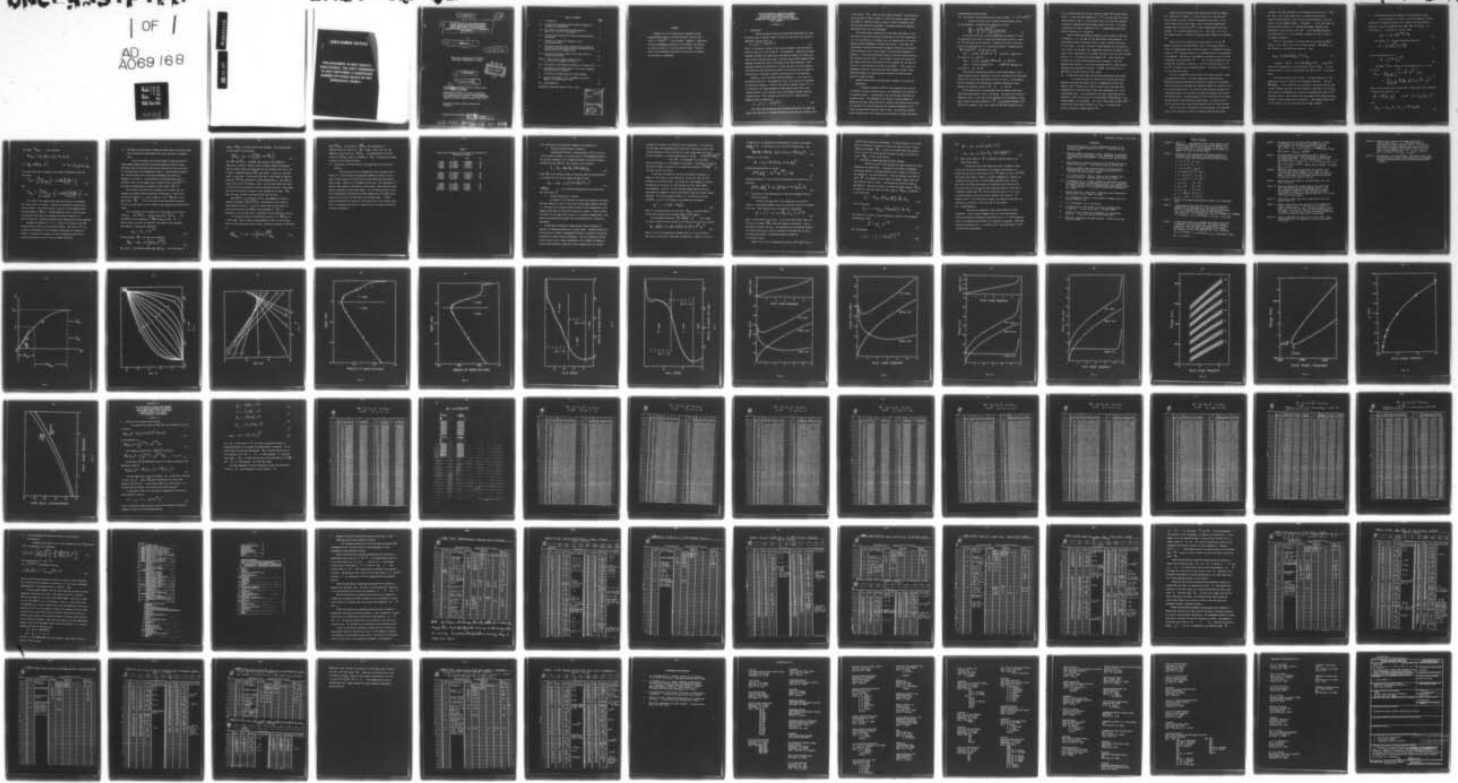
F/G 20/1

N00014-77-C-0196

NL

UNCLASSIFIED

OF 1  
AD A069 168



END  
DATE  
FILMED

7-79  
DDC

## **DISCLAIMER NOTICE**

**THIS DOCUMENT IS BEST QUALITY  
PRACTICABLE. THE COPY FURNISHED  
TO DDC CONTAINED A SIGNIFICANT  
NUMBER OF PAGES WHICH DO NOT  
REPRODUCE LEGIBLY.**

14 WHOI-78-65

12

6 RAY CALCULATIONS OF OCEAN SOUND CHANNELS  
USING A POCKET PROGRAMMABLE CALCULATOR  
AND EXTENDED FORMS OF THE HIRSCH-CARTER  
MATHEMATICAL MODEL WITH TABLES OF THE INCOMPLETE  
BETA FUNCTION.

by

10 Lincoln ~~and~~ Baxter, II

12 83 p.

WOODS HOLE OCEANOGRAPHIC INSTITUTION  
Woods Hole, Massachusetts 02543

DDC  
RECEIVED  
MAY 30 1979  
C

11 October 1978

9 TECHNICAL REPORT

15 Prepared for the Office of Naval Research under  
Contract N00014-77-C-0196

Reproduction in whole or in part is permitted  
for any purpose of the United States Government.  
This report should be cited as: Woods Hole Oceano-  
graphic Institution Technical Report WHOI-78-65.

Approved for public release; distribution  
unlimited.

Approved for Distribution Earl E. Hays  
Earl E. Hays, Chairman  
Department of Ocean Engineering

381 000 *Dist* 79 05 29 043

TABLE OF CONTENTS

	<u>Page</u>
I. Introduction	1
II. Incomplete Beta-function and Calculator Programs for Acoustic Ray Computations	2
III. The Geometry of Sound Speed Profile Layers in Which $c^2 = c_0^2 (1 - (\alpha z)^\beta)^{-1}$	3
IV. Fitting Ocean Sound Speed Profiles Using Hirsch-Carter Type Layers	4
V. Solutions for General Ray Segments in the Hirsch-Carter Model	7
VI. Calculation vs Axial Angle of Range and Travel Time at the End of Loops Above and Below the Sound Channel Axis and at the End of a Complete Cycle	9
VII. Calculation of Arrival Times for the Eigen Rays for a Source and Receiver	11
TABLE I Travel Time at 705 km of rays of order 14, 15 and 16 in Sargasso Sea Profile	12
VIII. Calculation of the Relative Intensity or Focusing Factor	13
IX. Calculation of New Axial Angles of a Ray that Propagates from one Profile to Another	
X. Calculation of Range Annotated Ray Angle Diagrams	
XI. Notes on the Values of $\beta$ in Asymmetric Profiles Based on the Hirsch-Carter Model	17
XII. Acknowledgements	17

Supplement follows main report in this volume

ACCESSION for	
NTIS	Write Section <input checked="" type="checkbox"/>
DDC	Buff Section <input type="checkbox"/>
ANNOUNCED	<input type="checkbox"/>
JUSTIFICATION	
BY	
DISTRIBUTION/AVAILABILITY CODES	
SPECIAL	
A 23 CP	

### ABSTRACT

Formulas for curve fitting and ray computation using compound models made up of several different layers of form  $c^2 = c_0^2 (1 - |\alpha z|^\beta)^{-1}$  are presented. Examples of computation by pocket programmable calculator on two Sargasso Sea profiles, one from the center of a cold ring eddy are given. Necessary tables of the incomplete beta-function and calculator programs are included in a supplement.

RAY CALCULATIONS OF OCEAN SOUND CHANNELS  
USING A POCKET PROGRAMMABLE CALCULATOR  
AND EXTENDED FORMS OF THE HIRSCH-CARTER  
MATHEMATICAL MODEL WITH TABLES OF THE INCOMPLETE  
BETA FUNCTION

L. Baxter, II

I. Introduction

Hirsch and Carter<sup>1</sup> have given closed form expressions for range and travel time of integral numbers of cycles of ray paths in the family of symmetrical profiles given by:

$$c^2 = c_0^2 (1 - |\alpha Z|^\beta)^{-1} \quad (1)$$

where  $c$  is the speed of sound at the vertical distance  $Z$  from the depth at which the speed is  $c_0$  and  $\alpha$  and  $\beta$  are parameters. Pedersen and Gordon<sup>2</sup>, Weinberg<sup>3</sup>, Stewart<sup>4</sup>, and others have developed the concept of fitting realistic acoustic profiles with layers of various curved profile segments while matching the speed and slope of the speed at the layer interfaces. This technique prevents the calculation of "false caustics" and other artifacts associated with less sophisticated profile fits and minimizes the number of layers needed to represent a natural sound speed profile realistically.

Equation 1 can be used with different parameters in each layer of a multilayer profile fit. The geometry of a ray in a layer may be understood by referring to Figure 1 in which a ray from the reference level ( $c = c_0$ ) is refracted as the sound propagates through higher speed levels, and  $Z$  (always positive) is the absolute value of the depth difference from the reference level. With  $Z$  defined in this way and  $\alpha$  always positive, Equation 1 may be rewritten as:

$$c^2 = c_0^2 (1 - (\alpha Z)^\beta)^{-1} \quad (1A)$$

The closed form expressions given by Hirsch and Carter<sup>1</sup> for range and travel time apply only to integral multiples of rays from the reference level

to the vertex,  $Z_v$ , where the ray becomes horizontal. The portions of this path which we need to compute for rays that traverse several layers can however be expressed almost as simply in terms of incomplete beta-functions which have been tabulated<sup>5</sup>. Convergent series for computing them directly have also been published<sup>6</sup>.

For profile models consisting of no more than four layers of this type with no more than two layers on each side of the sound channel axis, it is not too difficult or tiresome to do ray computations with a medium capacity pocket programmable calculator and tables. I have done such calculations, fitting various natural asymmetric profiles with approximations consisting of two or three layers, using different parameters of Equation 1 in each layer and matching speed of sound and its derivative at the interfaces between layers. In this paper I outline the methods and give sample results for two profiles from the Sargasso Sea, one from the center of a cold eddy and one outside of any eddies. I also outline an approximate method for calculating rays that propagate through a small horizontal gradient of sound speed and a method of calculating range annotated ray angle diagrams.

## II. Incomplete Beta-function and Calculator Programs for Acoustic Ray Computations

Although an extensive table<sup>6</sup> of the incomplete beta-function is available, the most important range of the variables for our purpose is too sparsely covered. A supplement\* to the present paper tabulates the necessary detail. The supplement also contains a Fortran program for generating any other values that may be required, and operating instructions and listings of the curve fitting and ray computation programs for the Texas Instruments SR56 calculator which I used. The calculator programs could be applied with little change to any equivalent or larger calculator

using algebraic operating system.

III. The Geometry of Sound Speed Profile Layers in Which  $c^2 = c_0^2 (1 - (\alpha z)^\beta)^{-1}$

We need the slope  $dc/dz$  in order to match different layers at the interfaces. Differentiating Equation 1A, we have:

$$\frac{dc}{dz} = \frac{\alpha^\beta c_0 (\alpha z)^{\beta-1}}{2 (1 - (\alpha z)^\beta)^{3/2}} \quad (2)$$

As we shall show later, the ray computations are simpler if we can fit the profile with layers in which the minimum speed of sound is equal to  $c_0$  and occurs at one interface. Therefore the limit of the slope,  $dc/dz$ , as  $Z$  approaches zero is an important parameter. Remembering that

$\alpha > 0$  and  $Z \geq 0$  we have three cases:

Case 1. If  $\beta > 1$ ,  $dc/dz \rightarrow 0$  as  $Z \rightarrow 0$ , regardless of the values of  $c_0$  and  $\alpha$ .

Case 2. If  $\beta = 1$ ,  $dc/dz \rightarrow \alpha c_0 / 2$  as  $Z \rightarrow 0$ .

Case 3. If  $\beta < 1$ ,  $dc/dz \rightarrow \infty$  as  $Z \rightarrow 0$  regardless of the values of  $c_0$  and  $\alpha$ .

For realistic sound speed profiles of ocean sound channels, Case 1 layers should be used to interface at the axis or minimum of the sound speed profile; the outer layers may belong to Case 2 or Case 3. This statement will be clarified by the following more detailed discussion of layer geometry for realistic values of  $\alpha$ ,  $\beta$ ,  $c_0$  and  $Z$ .

For refracted rays, the sound speed does not usually exceed about 102% of the axial speed of about 1.493 km/sec. The shape of the curves of Equations 1 and 2 in the range of the parameters for a 2% change in sound speed is most critically dependent on  $\beta$ , and reasonable changes in  $\beta$  may call for changes in  $\alpha$  over a range of  $10^{28}$  while changes in units

of  $Z$ , or depth variations of actual profiles, change  $\alpha$  by much smaller ratios. To show the shape changes due to  $\beta$  on the same axes for various values (Figures 2 and 3), I use arbitrary units for  $Z$  with  $\alpha$  adjusted to produce a maximum sound speed change of about 2% at  $Z=1$ . With these conventions, the order of magnitude of  $\alpha$  is approximately that which would be realistic for  $Z$  kilometers.

Figure 2 shows the geometry of Equation 1 while Figure 3 shows that of Equation 2, i.e. the slope for the same values of the parameters. In these figures curves 1-6 belong to Case 3, curve 7 is Case 2 and curves 8-11 are Case 1. For Case 1 and Case 2  $dc/dz$  increases with increasing  $Z$ , but in Case 2 the increase is not significant within the 2% change in sound speed. Case 2 approximates to a straight line and is the only case for which  $dc/dz$  at  $C_o$  is controlled by the parameter  $\alpha$ . Case 3 layers are the only ones in which  $dc/dz$  decreases with increasing  $Z$ . They are somewhat more difficult to use because matching  $dc/dz$  to a lower velocity adjoining layer requires an interface at  $Z > 0$ . The process will be explained in the next section of this report.

#### IV. Fitting Ocean Sound Speed Profiles Using Hirsch-Carter Type Layers

We can now see that the conditions of Pedersen and Gordon<sup>2</sup>, matching sound speed and slope, are met by asymmetrical profiles (see Figures 4 and 5) consisting of a Case 1 Hirsch-Carter type upper layer (designated by U) meeting a Case 1 lower layer (designated by L) at the sound channel axis. If the designations are used as subscripts and the subscript A refers to the axis of the sound channel  $(c_o)_U = (c_o)_L = c_A$ , but  $\alpha_U \neq \alpha_L$  and  $\beta_U \neq \beta_L$ . The U and L layers must belong to Case 1 because only a zero value of  $dc/dz$  at  $C_o = c_A$ , or minimum sound velocity, can give a common tangent at  $Z=0$ .

Figures 4 and 5 illustrate sound speed profiles from the Sargasso Sea. The profile in Figure 4 is from the center of a cold ring eddy; that in Figure 5 is from a location undisturbed by the eddy. To fit each of these with a U and L layer I used a calculator program which iterates from a trial value of  $\beta$  to place a Hirsch-Carter type curve through  $C_0$  at the axis and points 1 ( $C_1, Z_1$ ) and 2 ( $C_2, Z_2$ ) each marked with an X. The dots in Figures 4 and 5 indicate the resulting fit.

Where the fit is not perfect the exact values of  $\alpha$  and  $\beta$  depend of course on the chosen points 1 and 2, and an equally good or better fit might appear from a different choice. It simplifies the calculations if  $\beta$  corresponds exactly to a tabulated value in the supplement or in Pearson<sup>5</sup>. Therefore it is worthwhile to try such a value, as close as possible to the calculated  $\beta$ .  $\alpha$  is then recalculated to fit a point near the middle of the layer. The fit of the new values of  $\alpha$  and  $\beta$  is checked over the measured profile. The values are adopted if the fit seems good enough.

The fit is improved if one does not try to cover too great a range of depths with one layer. The depth range can be subdivided into additional layers but meeting the conditions of equality of sound speed and its derivative are somewhat more complicated when the interface is not at the sound channel axis. The sound ray calculations also become more complicated because the rays must be divided into segments that traverse the various layers. Inspecting Figures 4 and 5 we see that the fit above the axis would be much improved by another layer. In Figures 6 and 7 an "M" layer above the U layer has been added to each of these profiles.

Both in the curve fitting and in the calculations, to be discussed later it will be useful to think of a separate space for each layer.

In Figure 1 the layer interfaces with adjoining layers may occur at  $Z_Q$  and  $Z_S$ , but the layer space and its coordinate system extends beyond the portion  $Z_Q - Z_S$  that actually fits approximately to the real profile. The ray segments  $OZ_Q$  and  $Z_S Z_V$  in the layer space are only auxiliary constructs for computing the segment  $Z_Q Z_S$  which corresponds to the real ray in the layer. In the following discussion the subscripts U, M, or L are intended to indicate the space in which a coordinate is measured.

Figure 6 is an example of a profile that can be fitted with  $\beta_M = 1$ .  $(c_0)_M$  is set equal to  $C_U$  at the chosen interface.  $(dc/dz)_U$  at the interface is calculated from Equation 2. Then:

$$\alpha_M = 2 (dc/dz)_U / (c_0)_M \quad (3)$$

In Figure 7  $\beta_M < 1$ . Since  $(dc/dz)_M \rightarrow \infty$  as  $Z_M \rightarrow 0$ , slopes can be matched only if the reference velocity,  $(c_0)_M$ , is less than the velocity  $C_U$  at the interface and  $Z_M > 0$  at the same place.

The fit was carried out as follows: The layer interface in U was chosen near a point of inflection of the empirical profile.  $C_U$  and  $(dc/dz)_U$  at the interface were calculated. With some trial and error a layer thickness  $(Z_M)_{max}$ , and layer parameters  $\beta_M$ ,  $\alpha_M$  and  $(c_0)_M$  were selected to approximate the curvature and maximum sound speed in the empirical profile.  $(Z_M)$  interface was then computed from  $(dc/dz)_U = (dc/dz)_M$  interface using a program based on Equation 2. The program iterates from a trial value  $Z_t$  to a more accurate value of  $(Z_M)$  interface.

V. Solutions for General Ray Segments in the Hirsch-Carter Model

For our purposes it is useful to put the equations of Hirsch and Carter<sup>1</sup> in slightly different form. Within a layer described by Equation 1, a ray is designated by the angle  $\theta_0$  at velocity  $c_0$  (see Figure 1). It vertexes at  $Z = Z_v$  where

$$Z_v = (\sin \theta_0)^{2/\beta} / \alpha \quad (4)$$

The variable  $\xi$  defined by Hirsch and Carter<sup>1</sup> as

$$\xi = (\alpha Z)^\beta / \sin^2 \theta_0 \quad (5)$$

may also be expressed at Z by

$$\xi = \left( \frac{Z}{Z_v} \right)^\beta \quad (6)$$

The range,  $R_{0Z}$ , covered by the ray segment from 0 to Z, can be written:

$$\begin{aligned} R_{0Z} &= \frac{Z_v}{\beta \tan \theta_0} \int_0^\xi (1-x)^{\frac{1}{2}} x^{\frac{1}{\beta}-1} dx \\ &= \frac{Z_v}{\beta \tan \theta_0} \cdot B\left(\frac{1}{\beta}, \frac{1}{2}\right) \cdot I_\xi\left(\frac{1}{\beta}, \frac{1}{2}\right) \end{aligned} \quad (7)$$

where B is the complete beta function and I is the ratio of the incomplete to complete beta function. Let

$$B_1 = B\left(\frac{1}{\beta}, \frac{1}{2}\right) \quad \text{and} \quad I_1 = I_\xi\left(\frac{1}{\beta}, \frac{1}{2}\right)$$

then

$$R_{0Z} = Z_v \cdot B_1 \cdot I_1 / \beta \tan \theta_0 \quad (8)$$

The range,  $R_{ZZ_v}$ , can be written

$$R_{ZZ_v} = Z_v \cdot B(1 - I_1) / \beta \tan \theta_0 \quad (9)$$

Let  $B_2 = B(1 + \frac{1}{\beta}, \frac{1}{2})$  and  $I_2 = I_2(1 + \frac{1}{\beta}, \frac{1}{2})$ .

The travel times that correspond to the ranges of Equations 8 and 9 may be written

$$T_{CZ} = \frac{R_{CZ}}{c_0 \cos \theta_0} \left\{ 1 - \sin^2 \theta_0 \frac{I_2 B_2}{I_1 B_1} \right\} \quad (10)$$

and

$$T_{ZZ_v} = \frac{R_{ZZ_v}}{c_0 \cos \theta_0} \left\{ 1 - \sin^2 \theta_0 \frac{(1 - I_2) B_2}{(1 - I_1) B_1} \right\} \quad (11)$$

The values of the complete beta function,  $B_1$  and  $B_2$  are constants for a given layer of a profile. They may be calculated or taken at once from the tables. The relative values,  $I_1$  and  $I_2$ , of the incomplete beta function depend on  $\theta_0$  and  $Z$  through Equations 4 and 6, and the tables. The range and travel time of any segment such as Q-S (Figure 1) is easily obtained as a difference between values computed by the above equations.

Programs for Equations 4, 6, 8, 9, 10 and 11 fit easily in the SR56 calculator when  $I_1$  and  $I_2$  are entered from tables. Note that if  $Z = Z_v$ ,  $I_1$  and  $I_2 = 1$  and Equations 8 and 10 reduce to those given by Hirsch and Carter<sup>1</sup>. To obtain total ranges for  $N$  axis crossings the values can of course be multiplied by  $2N$  as is done by Hirsch and Carter.

VI. Calculation vs Axial Angle of Range and Travel Time at the End of Loops Above and Below the Sound Channel Axis and at the End of a Complete Cycle

Since the classical ray acoustics paper of Ewing and Worzell<sup>7</sup>, sound channel computations have often been presented by plots of range and travel time of loops above and below the axis and of a full ray cycle, all vs axial angle as the independent variable. These data are presented for the three-layer fits to the eddy and Sargasso Sea profiles in Figures 8, 9, 10 and 11. The procedure for calculating these plots is described: first for the simpler case of Figures 8 and 9 where  $\beta_M = 1$ , and then the modifications for Figures 10 and 11 where  $\beta_M < 1$ .

Axis to axis loops that do not penetrate into a second layer are computed by straight-forward application of Equations 4, 8 and 10 with the factor  $2N$  equal to 2,  $I_1$  and  $I_2$  equal to 1, and  $\theta_0$  equal to the axial angle,  $\theta_A$ . On the lower side of the axis where the profile fit has no second layer, the full range of axial angles may be covered this simply.

The Z coordinate of the interface in U layer space can be written  $(z_i)_U$ . When  $(z_v)_U$  becomes greater than  $(z_i)_U$ , the calculations can be simplified if  $\theta_A$  is chosen so that  $\gamma$  is exactly equal to a value of X printed in the tables of the incomplete beta-function. Omitting the subscript U:

$$z_v = z_i / X^{1/\beta} \tag{12}$$

The axial angle,  $\theta_A$ , for this ray is given by:

$$\theta_A = \text{arc sin} \left[ (\alpha z_v)^{\beta/2} \right] \tag{13}$$

When  $\beta = 1$ , the reference sound speed,  $(c_0)_M$ , for the M layer is

equal to  $(c_i)_U$ , the sound speed at the interface. The reference angle in the M layer is calculated by

$$(\theta_o)_M = \arccos \left[ \frac{(c_o)_M}{c_A} \cos \theta_A \right] \quad (14)$$

With  $\theta_A$  and  $(\theta_o)_M$  tabulated, one returns to the program for Equations 4, 8 and 10. Taking  $I_1$  and  $I_2$  directly without interpolation from the tables, range and travel time for the portion of the loop in the U layer is computed using the  $\theta_A$  just found.  $I_1$  and  $I_2$  in the layer equal 1 in this case. The portion of the ray in the M layer is computed using the  $(\theta_o)_M$  equivalent to  $\theta_A$  from Equation 14. The values of range, travel time, and distance,  $Z_V$ , in the U and M layers are added to obtain the values plotted for the ray loop above the axis. These range and travel time values are added to those computed for the same  $\theta_A$  below the axis to obtain the values for the full ray cycle.

When  $\beta_M < 1$ , as in Figures 10 and 11, the segments in M must be computed differently.  $(c_o)_M \neq (c_i)_U$  and Equations 9 and 11 must be used instead of 8 and 10.  $I_1$  and  $I_2$  in the M layer are not equal to 1. One could use directly the  $(\theta_o)_M$  that correspond to  $\theta_A$  by Equation 14, but one would have to interpolate in the tables for  $I_1$  and  $I_2$ .

It is easier to defer the interpolation, doing it in  $(\theta_o)_M$  at a later stage. This is done by using Equations 12 and 13 on the M layer after they have been used on the U layer. Equation 13 however is understood as:

$$(\theta_o)_{Mx} = \arcsin \left[ \left( \alpha Z_V \right)^{\beta/2} \right]_M \quad (15)$$

where  $(\theta_0)_{Mx}$  is the value of  $(\theta_0)_M$  that corresponds to a tabulated value of  $X$  and not to  $\theta_A$ . Range, travel time, and  $Z_v$  computed in the M layer for  $(\theta_0)_{Mx}$  are interpolated to find the values for  $(\theta_0)_M$  that do correspond to  $\theta_A$ . The results are added to those for the U layer as before.

#### VII. Calculation of Arrival Times for the Eigen Rays for a Source and Receiver

This problem is merely an extension of the techniques used in Section VI. First one adds the appropriate segments to obtain a plot of range vs axial angle for the source and receiver depths and the possible types of path. Figure 12 is an example of this step. One interpolates to find the axial angles of each path at a given range. Then range and travel time are computed for these axial angles. Due to limited precision in the first interpolation the ranges will differ slightly but the average sound speeds will be correct for each path at the desired range. A second linear interpolation will adjust all the travel times to the correct range. Table I illustrates the result at a range of 705 km in Figure 12 and rays of order 14 through 16.

TABLE I

Travel Time at 705 km of rays of order 14, 15 and 16 in  
Sargasso Sea Profile

	T	$\Delta T$	N
7.16964	471.65950	0.0000000	16
7.21127	471.65254	0.0069642	16
7.40994	471.604778	0.054726	16
7.46963	471.59383	0.1001213	16
8.29557	471.438324	0.2211799	15
8.40314	471.414127	0.2453772	15
8.57449	471.36485	0.2946512	15
8.69705	471.33525	0.324255	15
9.70808	471.11683	0.54267	14
9.87478	471.068211	0.59129	14
10.03846	471.015474	0.64403	14
10.21703	470.959929	0.69958	14

VIII. Calculation of the Relative Intensity or Focusing Factor

a. Relative Intensity Except at Caustics

Brekhovskikh<sup>8</sup> defines a "focusing factor"  $f=I/I_0$ , the ratio of the acoustic intensity  $I$  at a given point in the homogeneous medium to the acoustic intensity  $I_0$  in a homogeneous medium at the same distance. He shows that when  $R \gg Z$  and the point is not a caustic.

$$f = R / \sin \theta_p \left( \frac{dR}{d\theta_A} \right)_p \quad (16)$$

where  $\theta_p$  is the horizontal angle at the given point and the derivative is evaluated for the ray that passes through the point.

$$\theta_p = \arccos \left( c(\cos \theta_A) / c_0 \right) \quad (17)$$

$\left( \frac{dR}{d\theta_A} \right)_p$  may be obtained graphically as the slope from a plot like Figure 12.

b. Relative Intensity at a Caustic

In Figure 12 the four rays of a given order appear in two pairs. Each pair appears to join at a point for an axial angle slightly less than  $7^\circ$ . The scale is too coarse to show the detail in the neighborhood of the supposed point which is really the location of a caustic. Figure 13 shows the "point" of the lower pair of order 17 on a greatly expanded scale. The method of calculating this detail will be discussed after I outline its application.

We have been interested in comparing the relative intensity of caustics in differing profiles but at a given range. Although ordinary ray theory fails at a caustic, Brekhovskikh<sup>9</sup> discusses a method of calculating intensity at a caustic from ray parameters. The full expression involves an Airy function and is rather complicated, but to compare the maxima of caustics under different conditions without computing the true relative

intensity at any point the expressions can be shortened. In the notation of this paper, and discarding factors that don't vary much in actual sound velocity profiles, relative intensity at a given large range and a given acoustic frequency is inversely proportional to  $\tan \theta_A \sin \theta_P (d^2R/d\theta_A^2)_P^{2/3}$

where  $\theta_P$  (see Equation 17) is the angle with the horizontal of a ray tangent to the caustic. The method of computing the data for Figure 13 enables us to evaluate the derivative  $(d^2R/d\theta_A^2)_P$ ; the other factors are obvious.

To calculate the range of a ray near the caustic, we measure, in Figure 10, the slope,  $S_f$ , and intercept  $I_f$ , of the full cycle ray that vertexes at the receiver depth. The values are:  $S_f=3$  km/degree,  $I_f=22$  km. We measure also the slope,  $S_u$ , and intercept  $I_u$  of the upper branch.  $S_u=0$  km/degree.  $I_u=10$  km. We then calculate range from the vertex vs axial angle for segments to the receiver depth of rays that vertex slightly shallower. The result appears as Figure 14. We note that the range increment,  $r_x$ , due to this segment is approximately the parabola

$$r_x^2 = K (\theta_A - (\theta_A)_P)^2 \tag{18}$$

where, in the given example,  $K=2.9781$  and  $(\theta_A)_P$ , the axial angle of the ray that vertexes at the receiver depth, =6.892 degrees.

Let the angular difference,  $[\theta_A - (\theta_A)_P] = \varphi$ . Total range of a ray of order N in the vicinity of the vertex can be written as follows:

$$R = Q(I_u + S_u \cdot \varphi) + N(I_f + S_f \cdot \varphi) \pm K^{1/2} \varphi^{3/2} \tag{19}$$

where  $Q = 3/2$  or  $1/2$  depending on whether there is or is not an extra upper loop in the group of rays under consideration. Figure 13 is a plot

of Equation 19. As indicated by Brekhovskikh<sup>9</sup> the caustic occurs where  $dR/d\theta_A = 0$ , on the branch of the curve with the minus sign. Now:

$$dR/d\theta_A = dR/d\varphi = Q \cdot S_u + N \cdot S_f - K^{1/2} / 2 \varphi^{1/2} \quad (20)$$

Therefore, at the caustic

$$\varphi = K / 4 (Q \cdot S_u + N \cdot S_f)^2 \quad (21)$$

Differentiating Equation 20, we have

$$d^2R/d\theta_A^2 = K^{1/2} \varphi^{-3/2} / 4 \quad (22)$$

Evaluating Equation 22 at the caustic by substitution of Equation 21, we find that

$$(d^2R/d\theta_A^2) = 2 (Q \cdot S_u + N \cdot S_f)^3 / K \quad (23)$$

IX. Calculation of New Axial Angles of a Ray that Propagates from one Profile to Another

Milder<sup>10</sup> has shown that if the change from one profile to another is sufficiently gradual, there is an invariant called the characteristic time. This invariant can be calculated by the equation:

$$J = (T - X \cdot \cos \theta_A / c_A) / 2 \pi \quad (24)$$

where J is the characteristic time, T is the full cycle travel time, X is the full cycle range,  $\theta_A$  is the axial angle of the ray and  $c_A$  is the speed of sound at the axis. The conditions on the horizontal gradient for validity are given in detail by Milder for both wave and ray theory. In ocean sound channels a horizontal gradient as small as .03 m/sec/km is safe.

Figure 15 is a plot of characteristic time vs axial angle for the

profiles that we have been considering. To find the angle in one profile that is equivalent to an angle in another one finds the value  $J$  corresponding to the angle,  $\theta_{A_1}$ , in the first profile moves horizontally to the curve for the second profile and under the same  $J$  one finds the value,  $\theta_{A_2}$ , in the second profile.

X. Calculation of Range Annotated Ray Angle Diagrams

Flatte<sup>11</sup> and Cox<sup>12</sup> have discussed the range annotated ray angle diagram and its applications. A program (for the pocket programmable calculator) adapted to generating data for such a diagram is included in the supplement. The depth difference  $Z$  from the reference level and the angle  $\theta$  are computed from the range on a segment shorter than that from reference level to vertex. Longer paths are plotted by symmetry and addition. There are two cases: One where the given range is  $R_{OZ}$  in Equation 8, the other where the given range is  $R_{ZZ_V}$  in Equation 7. In the first case:

$$I_1 = R_{OZ} \beta(\tan \theta_o) / B_1 Z_V \quad (25)$$

In the second case:

$$I_1 = 1 - R_{ZZ_V} \beta(\tan \theta_o) / B_1 Z_V \quad (26)$$

The value of  $I_1$  is used to obtain  $X$  from the  $I_x$  tables of the incomplete beta-function. Then

$$Z = Z_V X^{1/\beta} \quad (27)$$

and from Equation 1

$$c/c_o = (1 - |\alpha Z|^\beta)^{-1/2} \quad (28)$$

but

$$\theta = \text{arc cos} (c \cos \theta_0 / c_0)$$

or

$$\theta = \text{arc cos} \left( \cos \theta_0 / \sqrt{1 - |\alpha z|^\beta} \right) \quad (29)$$

#### XI. Notes on the Values of $\beta$ in Asymmetric Profiles Based on the Hirsch-Carter Model

Hirsch and Carter have pointed out that, in symmetric models of the near axis sound transmission, the observed time dispersion of arrivals occurs only in that subset of the  $\beta$  family for which  $1 < \beta < 2$ . The actual sound channel, however, is grossly asymmetrical. Because the refraction below the axis is so much weaker than that above, rays at more than a very small axial angle will spend much more time below the axis than above it so that the overall dispersion pattern is like that of a symmetric channel with  $\beta$  near the below axis value of 1.25 or 1.26, the profiles above the axis are fitted, however, with values of  $\beta$  between 2 and 3. This would tend to reduce the dispersion below what one would get by reflection of the lower half of the channel.

#### XII. Acknowledgements

This work was supported by ONR Contract N00014-74-C-0262; NR 083-004. John C. Beckerle suggested the use of the Hirsch-Carter profile and did some preliminary calculations with it. P. Hirsch suggested use of the incomplete beta-function for computing ray segments. Additional helpful discussions with J. C. Beckerle, Earl E. Hays and George V. Frisk, are gratefully acknowledged.

REFERENCES

1. Peter Hirsch and Ashley H. Carter, "Mathematical Models for the Prediction of SOFAR Propagation Effects". J.Acoust.Soc.Am. 37, pp 90-94 (1965).
2. Melvin A. Pedersen and David F. Gordon, "Comparison of Curvilinear and Linear Profile Approximation in the Calculation of Underwater Sound Intensities by Ray Theory". J.Acoust.Soc.Am. 41, pp 419-438 (1967).
3. Henry Weinberg, "A Continuous Gradient Curve Fitting Technique for Acoustic Ray Analysis". J.Acoust.Soc.Am. 50, pp 975-984 (1971).
4. Kenneth R. Stewart, "Ray Acoustical Model of the Ocean Using a Depth/Sound-Speed Profile with a Continuous First Derivative". J.Acoust.Soc.Am. 38, pp 339-347 (1965).
5. E. S. Pearson and N.L. Johnson, "Tables of the Incomplete Beta-Function" 2nd Edition. Cambridge University Press (1968).
6. M. Abramowitz and I. A. Stegun, "Handbook of Mathematical Functions with Formulas, Graphs, and Mathematical Tables". U. S. Department of Commerce, National Bureau of Standards Applied Mathematical Series 55. Ninth printing, U. S. Government Printing Office, Washington, D.C. (1970).
7. Maurice Ewing and J. Lamar Worzel, "Long-Range Sound Transmission". Geological Soc. of Am., Memoir 27 (1948).
8. L. M. Brekhovskikh, "Waves in Layered Media". Academic Press, Inc., New York (1960), p.474.
9. L. M. Brekhovskikh. op.cit. pp 483-496.
10. D. Michael Milder, "Ray and Wave Invariants for SOFAR Channel Propagation". J.Acoust.Soc.Am. 46, -- 1259-1263 (1969).
11. Stanley M. Flatte, "Angle-depth diagram for use in underwater acoustics". J.Acoust.Soc.Am. 60, pp 1020-1023 (1976).
12. Henry Cox, "Approximate ray angle diagram". J.Acoust.Soc.Am. 61, pp 353-359 (1977).

CAPTIONS TO FIGURES

Figure 1 Path of a ray from reference level to vertex within a layer. Calculation of range and travel time of the segment from 0 to Z or that from Z to  $Z_v$  is discussed in the text. The general segment Q to S can be expressed as a difference of segments of either of the above types.

Figure 2 Geometry of sound speed profiles described by Equation 1A.  $C_0=1.49275$  in all the curves. The other parameters are listed after the indicated number of each curve as follows:

1.  $\alpha = 10^{-28}$ ,  $\beta = .05$  ;
2.  $\alpha = 10^{-14}$ ,  $\beta = .10$  ;
3.  $\alpha = 3 \times 10^{-6}$ ,  $\beta = .25$  ;
4.  $\alpha = 5.6 \times 10^{-5}$ ,  $\beta = .33$  ;
5.  $\alpha = 1.6 \times 10^{-3}$ ,  $\beta = .50$  ;
6.  $\alpha = 7.0 \times 10^{-3}$ ,  $\beta = .65$  ;
7.  $\alpha = .04$ ,  $\beta = 1.0$  ;
8.  $\alpha = .117$ ,  $\beta = 1.5$  ;
9.  $\alpha = .20$ ,  $\beta = 2.0$  ;
10.  $\alpha = .275$ ,  $\beta = 2.5$  ;
11.  $\alpha = .343$ ,  $\beta = 3.0$  .

Figure 3 Slopes of the sound speed profiles of Figure 2 on a logarithmic plot.

Figure 4 A sound speed profile from the center of a cold ring eddy is indicated by the continuous line. Two layers according to Equation 1 have been fitted at the axis and the points marked by X. The calculated sound speeds at other points are indicated by dots. The parameters are:  $\alpha_U = 0.428668$ ,  $\beta_U = 2.87968$ ,  $C_0 = 1.48867$ ,  $\alpha_L = .0325903$ , and  $\beta_L = 1.25083$

Figure 5 A sound speed profile in the Sargasso Sea outside of the eddy is indicated by the continuous line. Two layers according to Equation 1 have been fitted at the axis and at the points marked by X. The calculated sound speeds at other points are indicated by dots. The parameters are:  $\alpha_U = .335904$ ,  $\beta_U = 2.05506$ ,  $C_0 = 1.49275$ ,  $\alpha_L = .0321452$ , and  $\beta_L = 1.25931$

Figure 6 The upper portion of the eddy profile, Figure 4, is fitted by two layers. The previous fit is retained below the sound channel axis. The new parameters are as follows:  
 $\alpha_U = 0.44$ ,  $\beta_U = 2.857$ ,  $(c_o)_U = 1.48867$ ,  $\alpha_M = 0.061$   
 $\beta_M = 1.0000$ ,  $(c_o)_M = 1.49600$

Figure 7 The upper portion of the Sargasso Sea profile, Figure 5, is fitted by two layers. The previous fit is retained below the sound channel axis. The new parameters are as follows:  
 $\alpha_U = 0.315$ ,  $\beta_U = 2.000$ ,  $(c_o)_U = 1.49275$ ,  $\alpha_M = 4.01 \times 10^{-15}$ ,  
 $\beta_M = 0.100$ ,  $(c_o)_M = 1.4989$ ;  $(z_i)_M$  at the interface = 0.0216,  
 $(z_i)_U = 0.575$

Figure 8 Range and vertex depth vs axial angle for the three layer fit of the eddy profile (Figures 4, 6). Range of a loop above the axis, one below the axis, and a full ray cycle are shown. Vertex depth is for the upper loop and is therefore the shallowest point reached by the ray.

Figure 9 Travel time vs axial angle for the same profile, fit, and paths as Figure 8.

Figure 10 Range and vertex depth vs axial angle for the three layer fit of the Sargasso Sea profile (Figures 5 and 7). The data presented corresponds to that presented in Figure 8. The constancy of range of the upper loop over the axial angles 0-10.4 is a property of the fit with  $\beta = 2.0$  as noted in the Hirsch-Carter<sup>1</sup> paper.

Figure 11 Travel time vs axial angle for the same profile, fit, and paths as Figure 10.

Figure 12 Range vs axial angle of high order rays in the Sargasso Sea profile (Figures 5 and 7). The order is, of course, the number of loops below the axis. The receiver depth is .85 km. The source is on the axis. There are four rays belonging to each order. This figure may be used as described in the text to find axial angles of eigen rays at a given range.

Figure 13 Range vs axial angle for two rays of the 17th order. This figure demonstrates the formation of a caustic as discussed in the text.

Figure 14 Range vs axial angle of a ray segment from vertex to receiver depth (.85 km, i.e. .4 km above the axis) in the Sargasso Sea profile (Figures 5 and 7). The solid line shows values calculated using the Hirsch-Carter model with tables of the incomplete beta-function. The circles show values from the parabolic fit,  $r_x^2 = k (\theta_A - (\theta_A)_p)^2$ , with  $k = 2.9781$  and  $(\theta_A)_p = 6.892^\circ$ .

Figure 15 Characteristic time vs axial angle. This figure can be used for estimating the change in axial angle when a ray propagates through a transition region from one sound velocity profile to another.

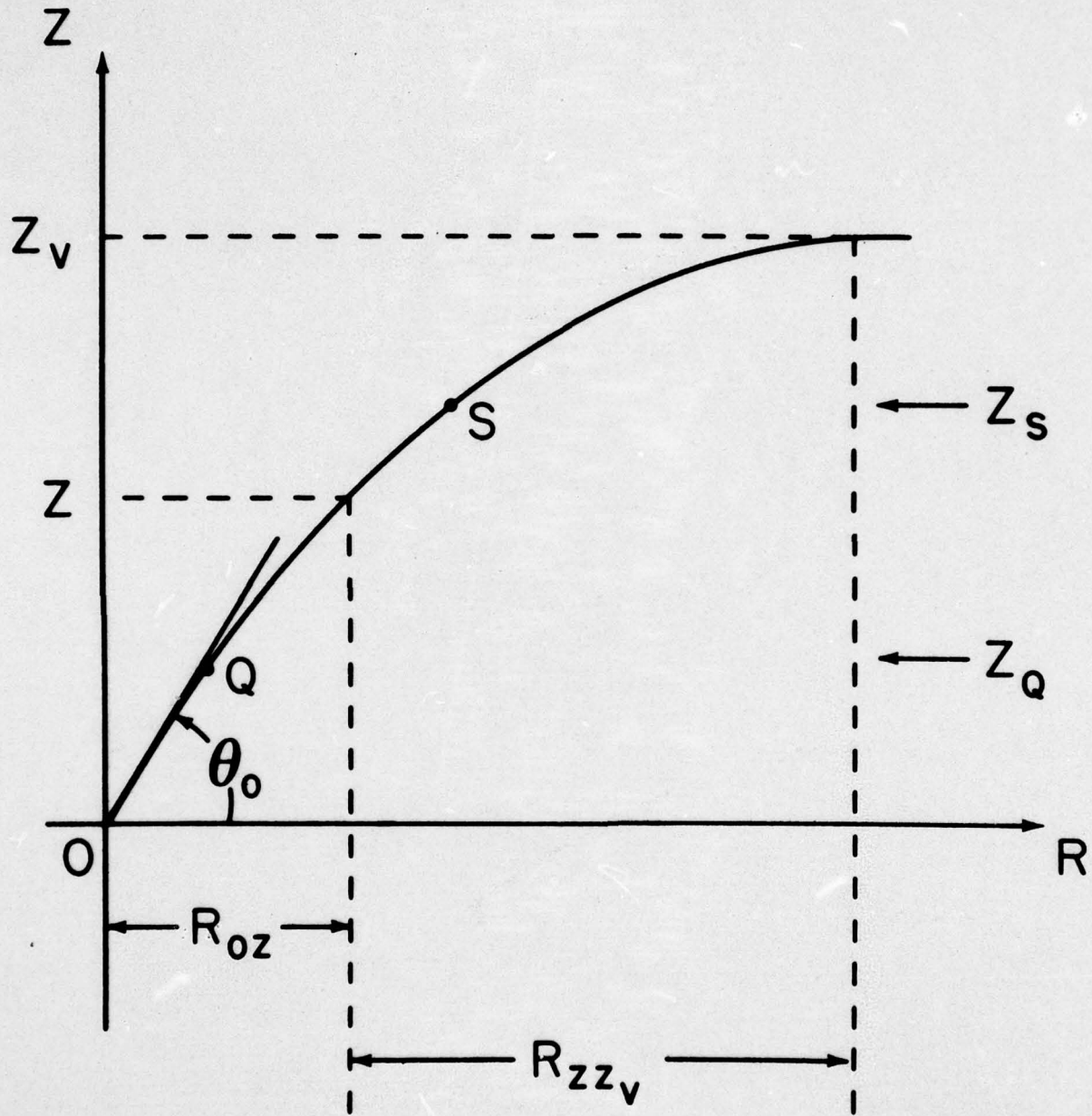


Fig. 1

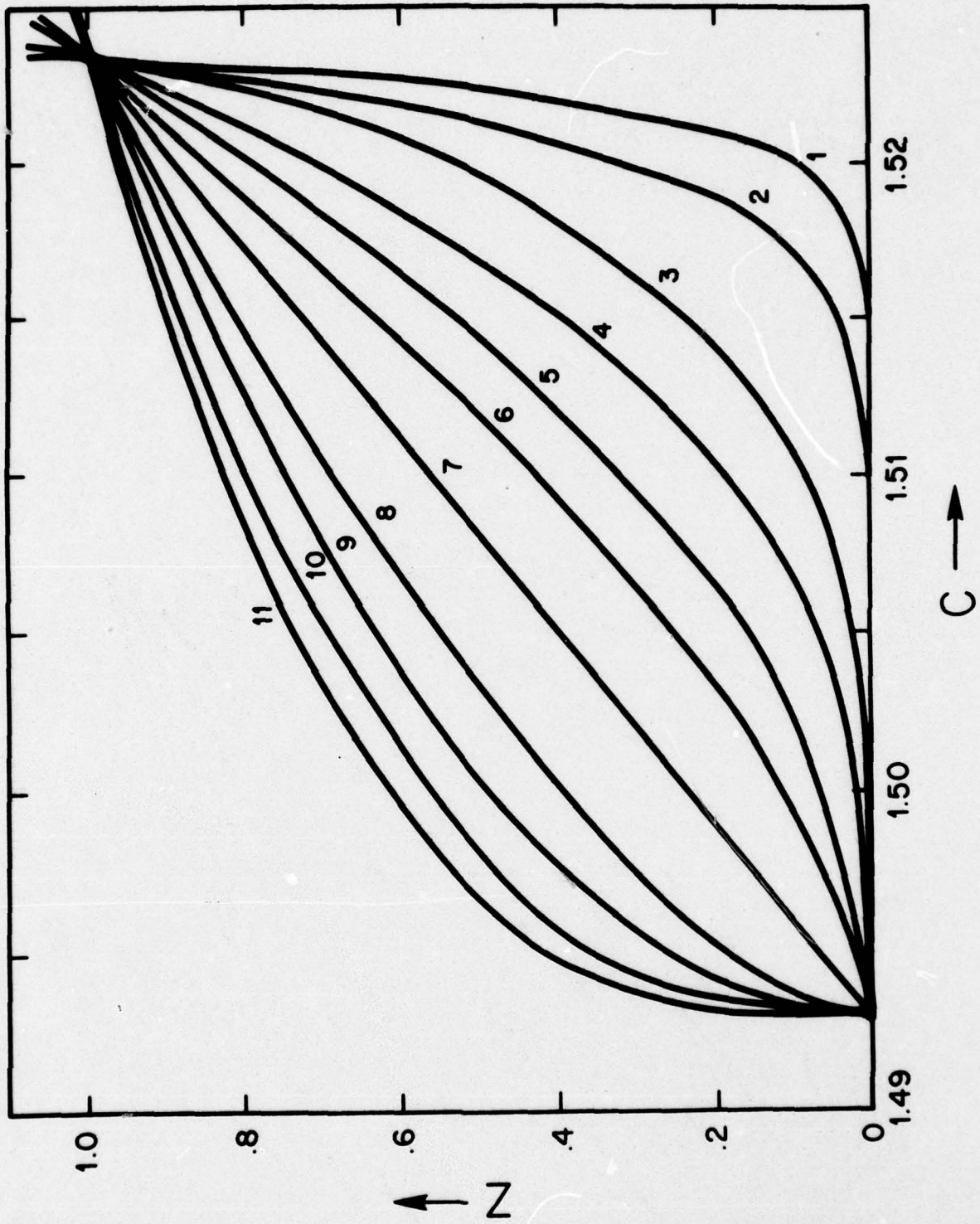


Fig. 2

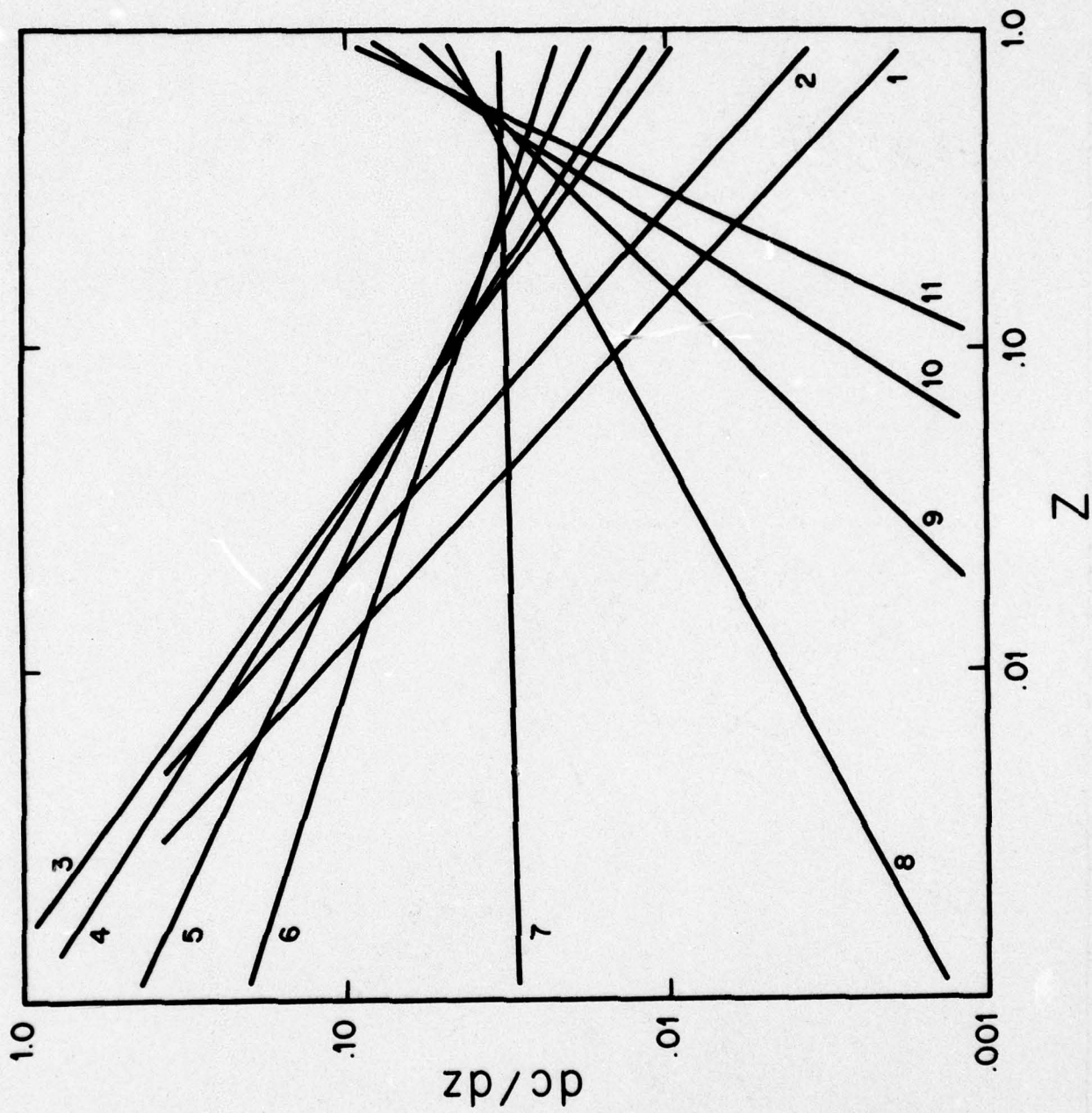


Fig. 3

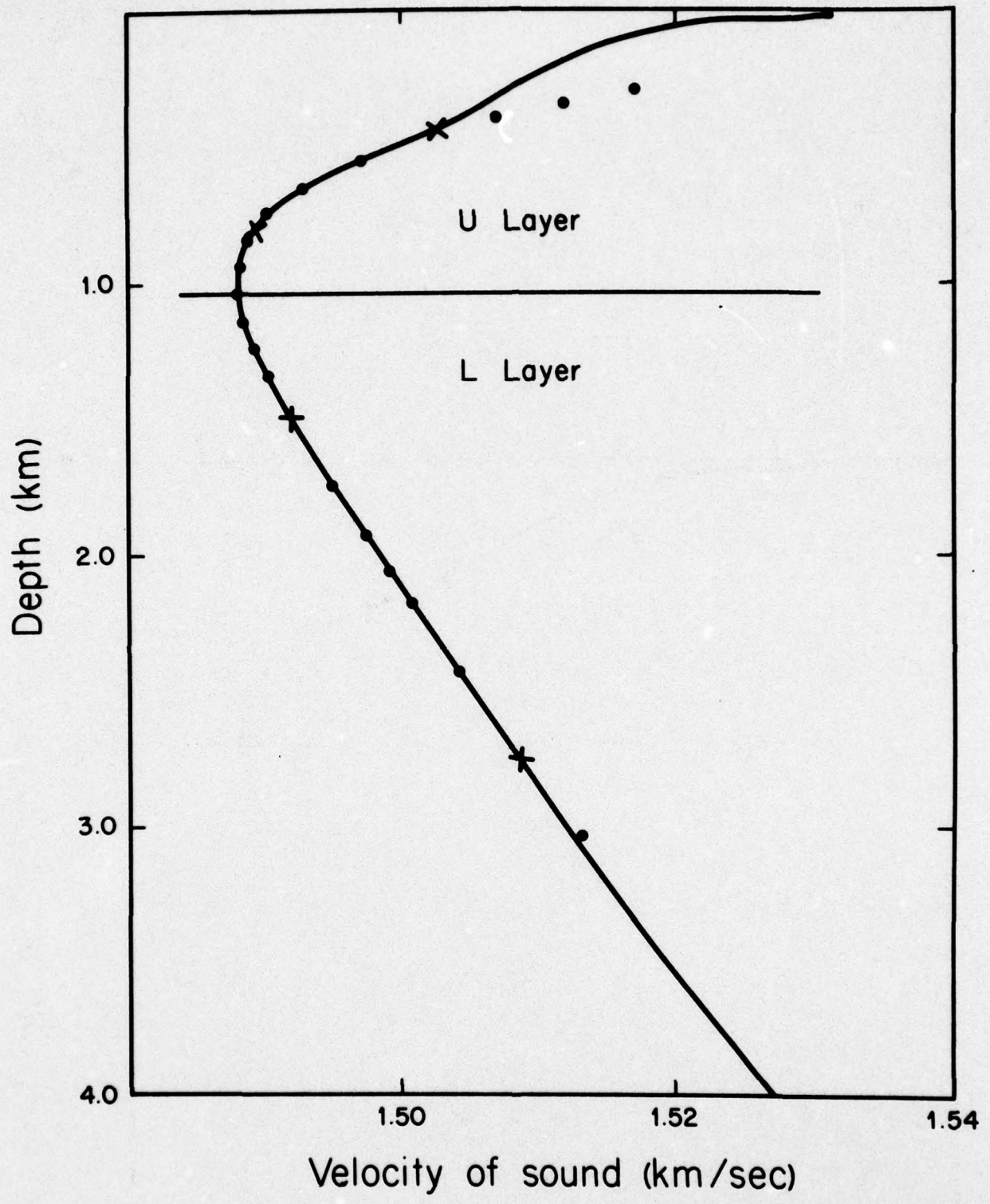


Fig. 4

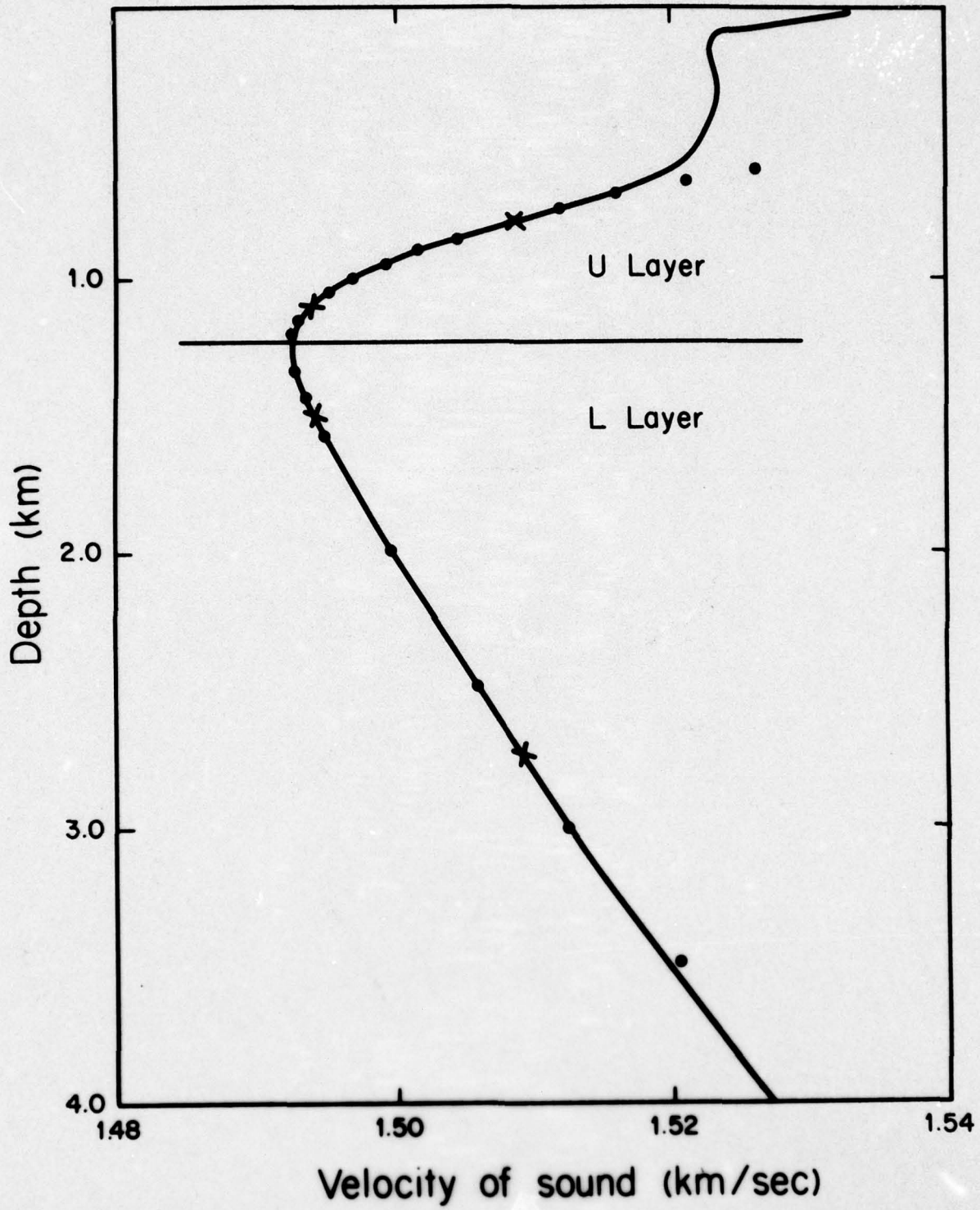


Fig. 5

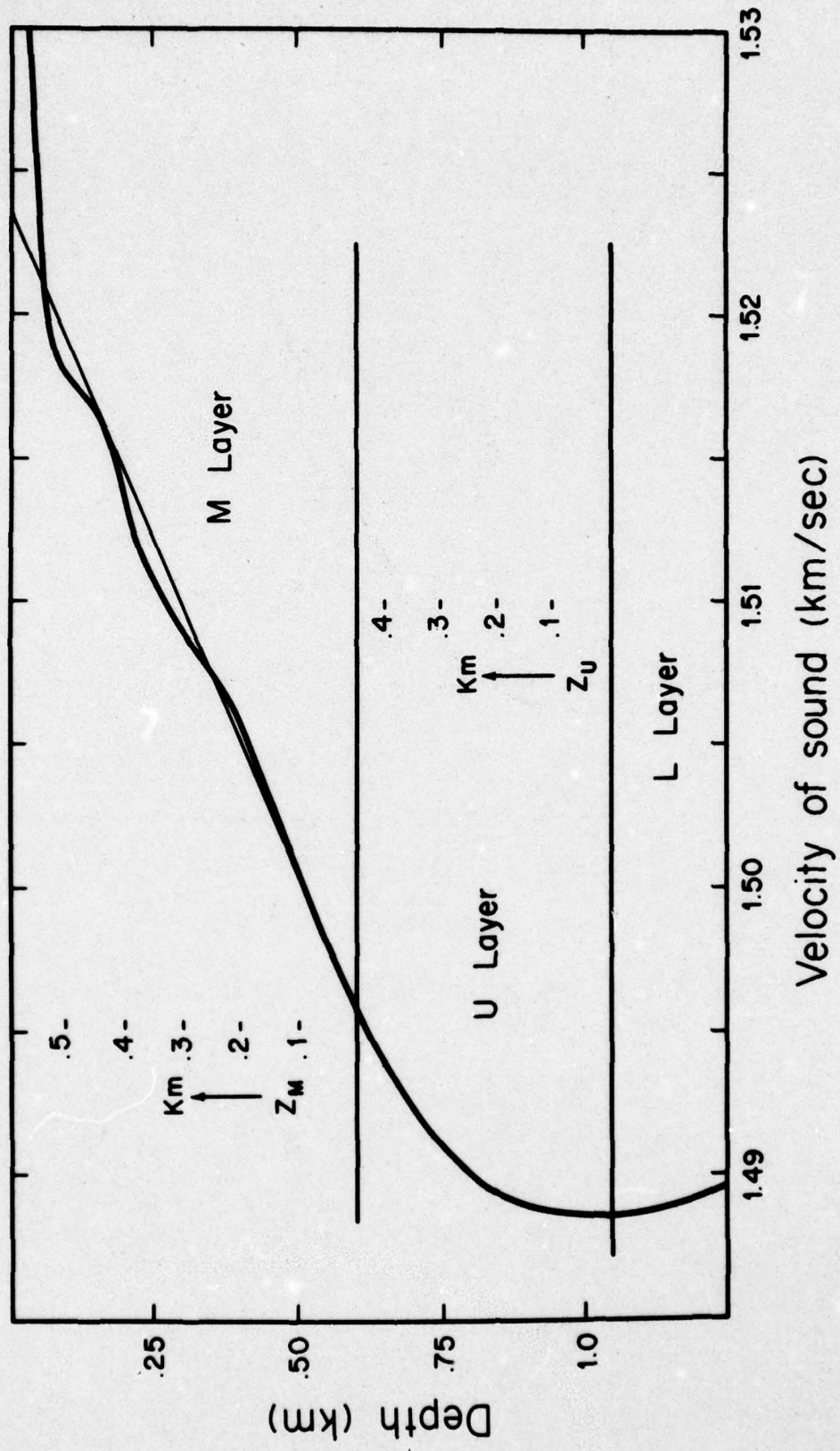


Fig. 6

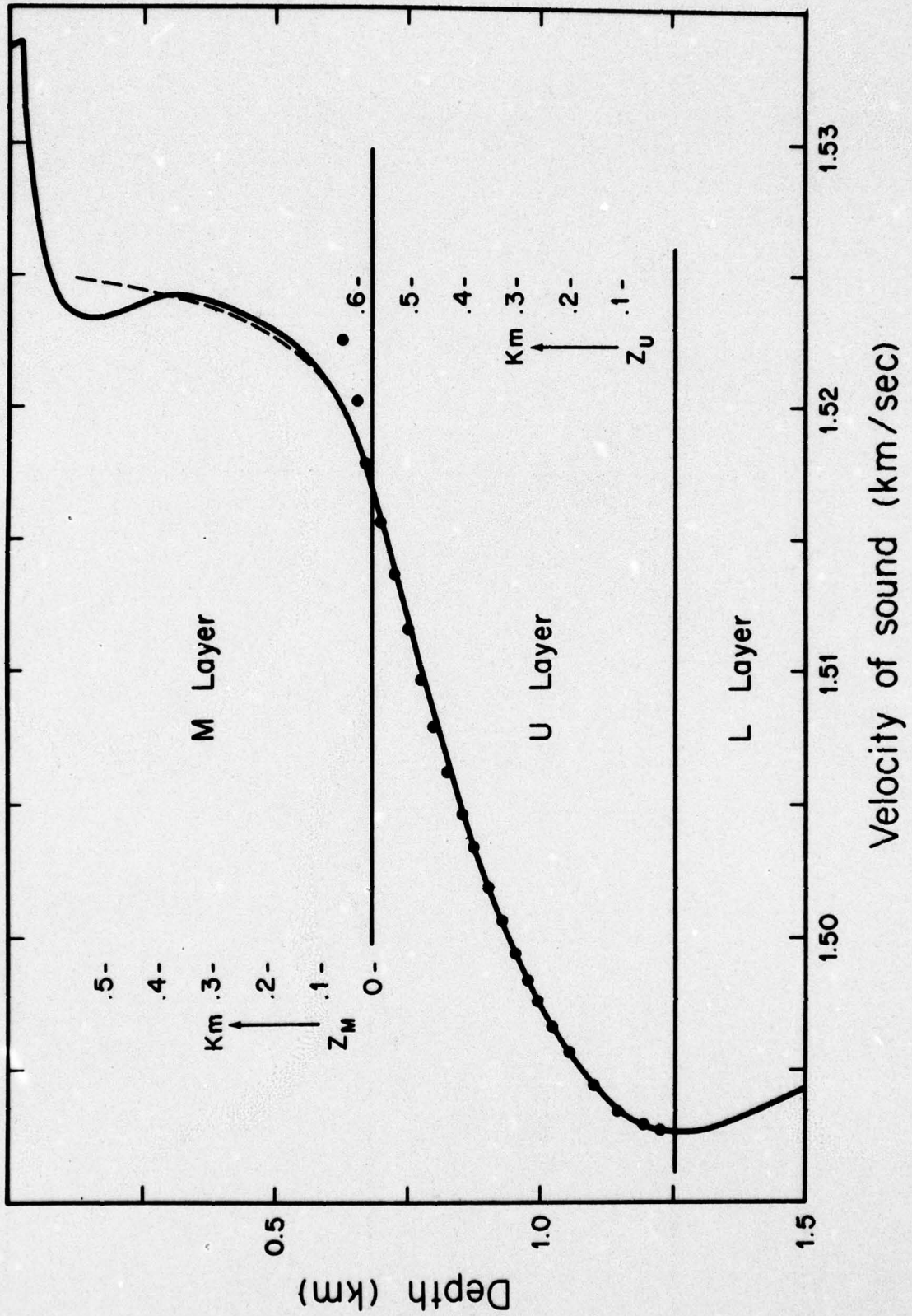


Fig. 7

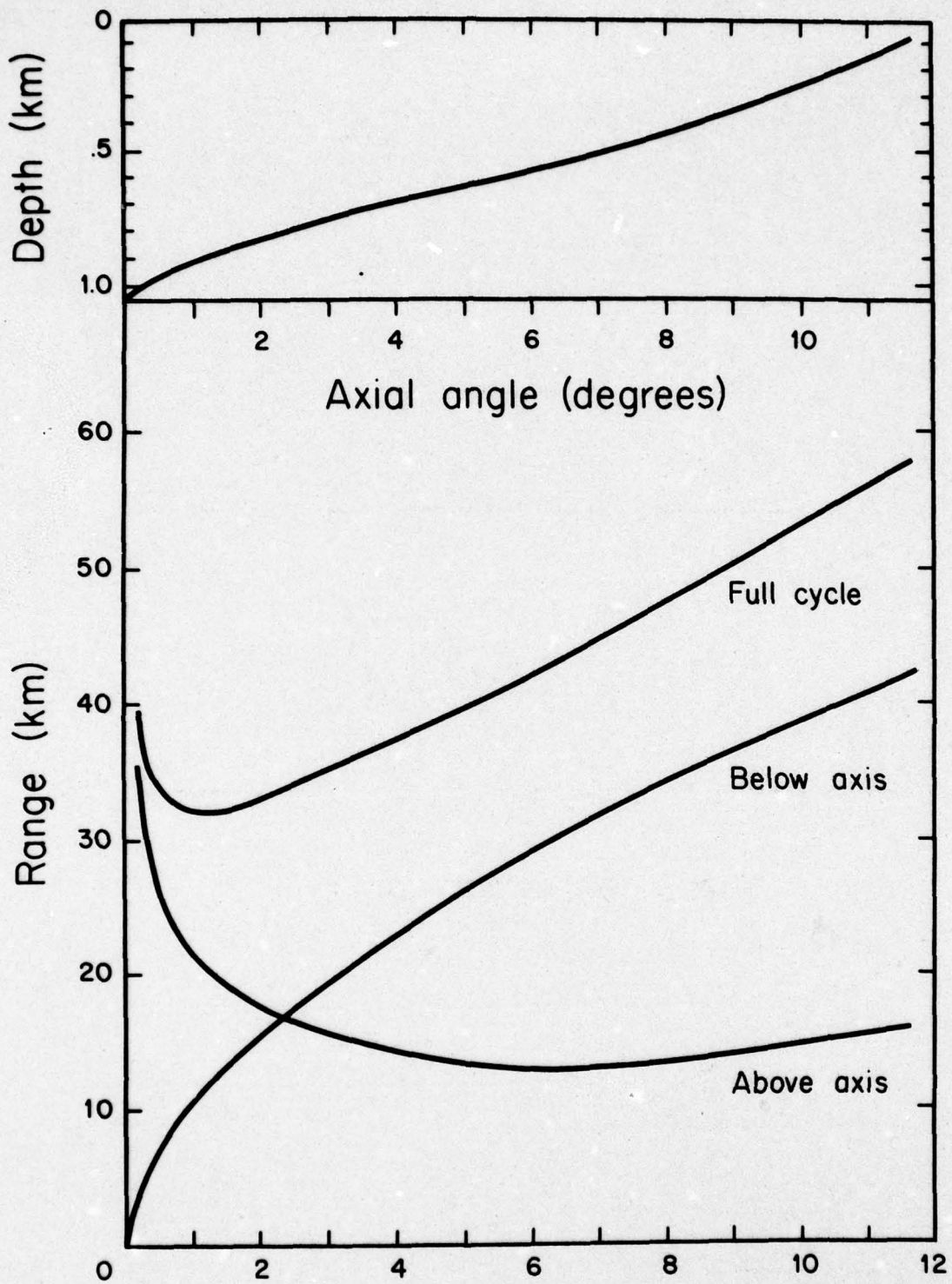


Fig. 8

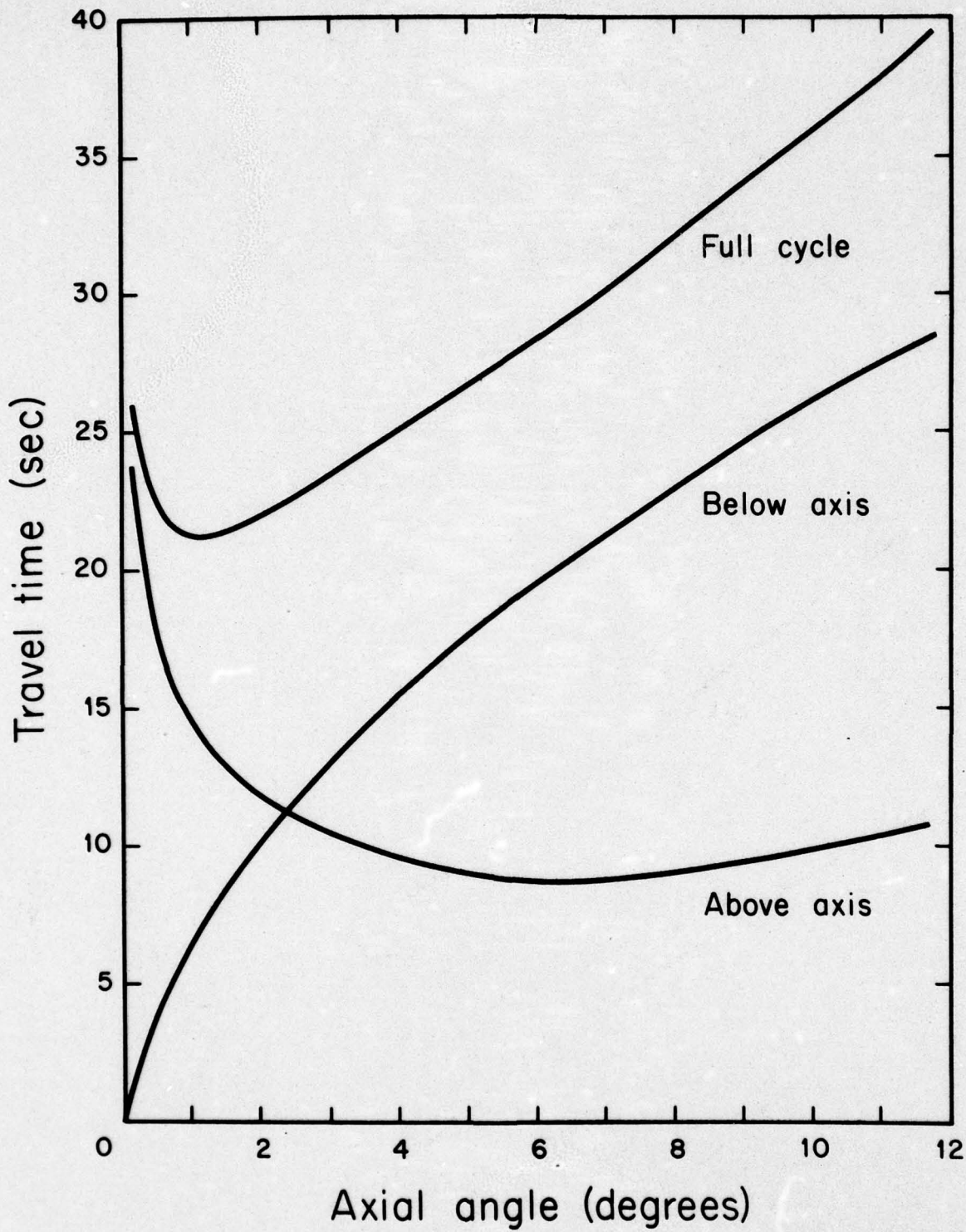


Fig. 9

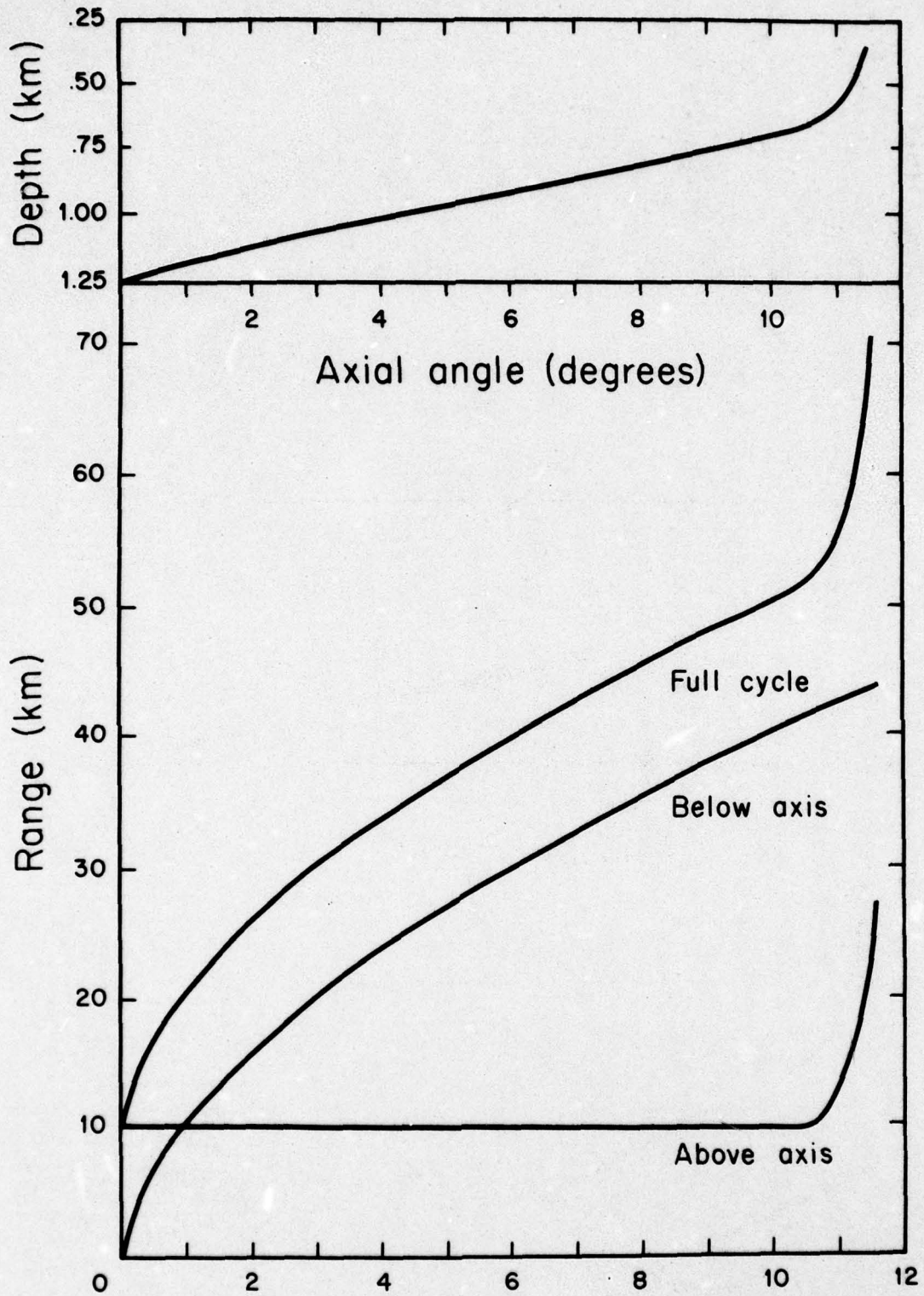


Fig. 10

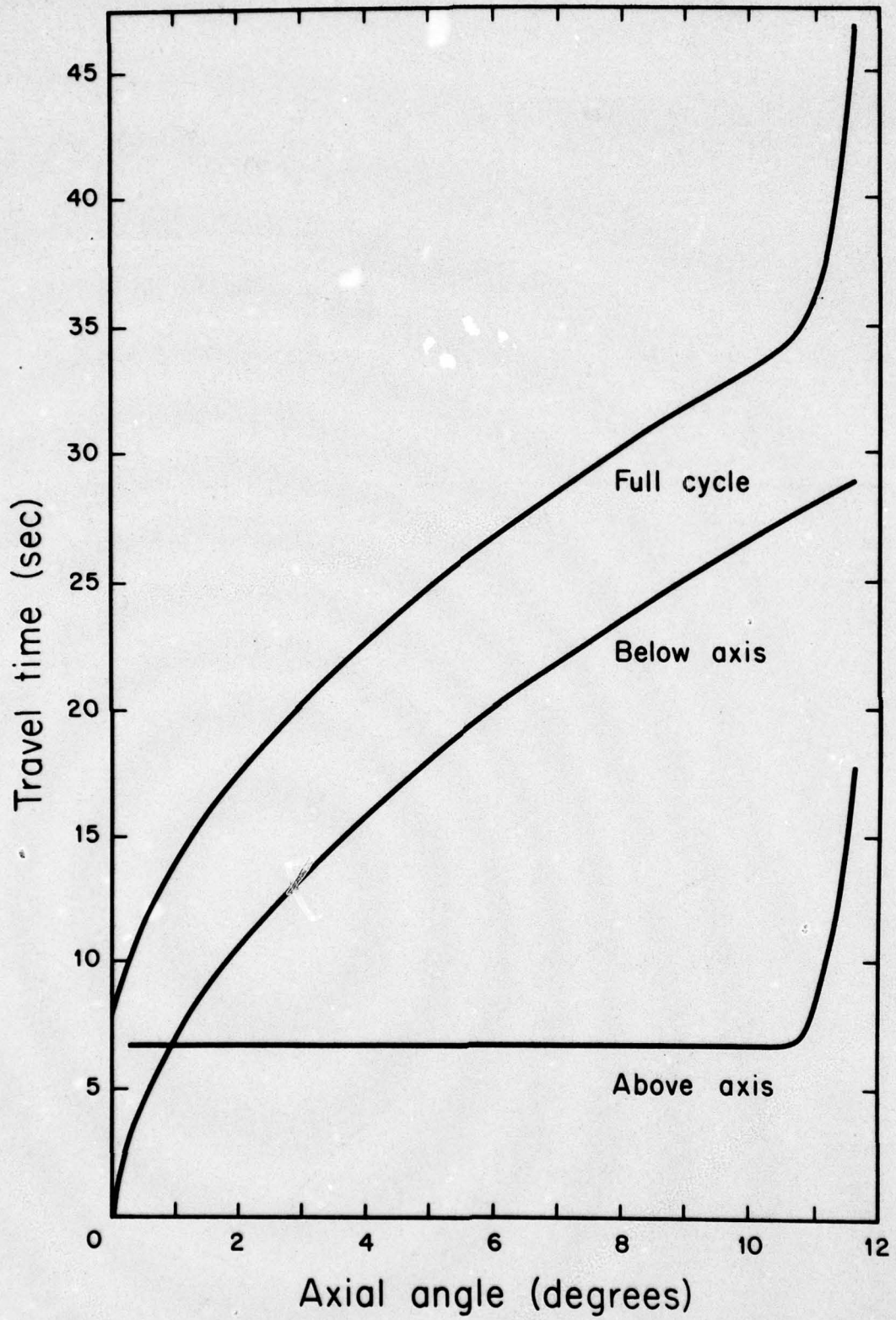


Fig. 11

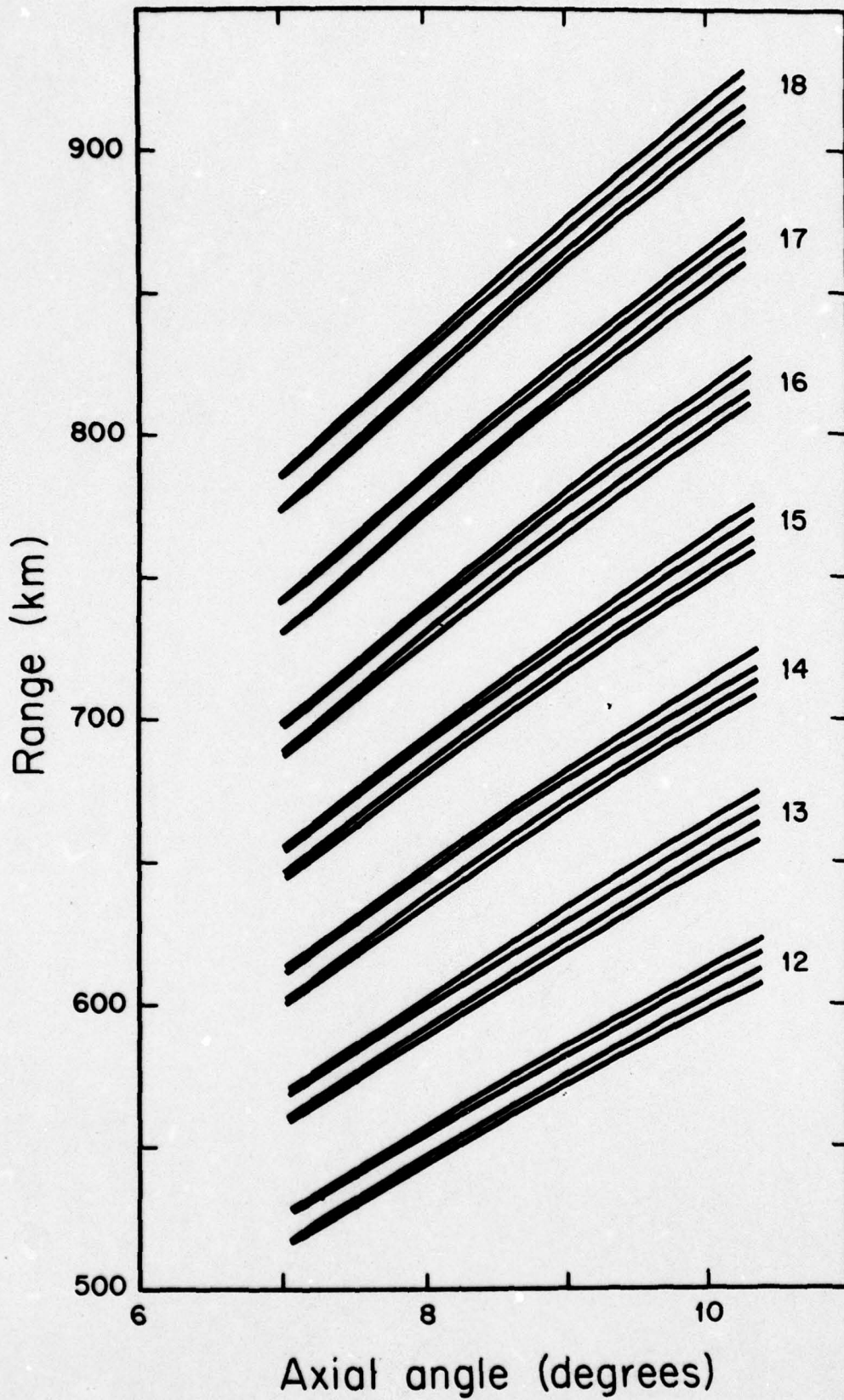


Fig. 12

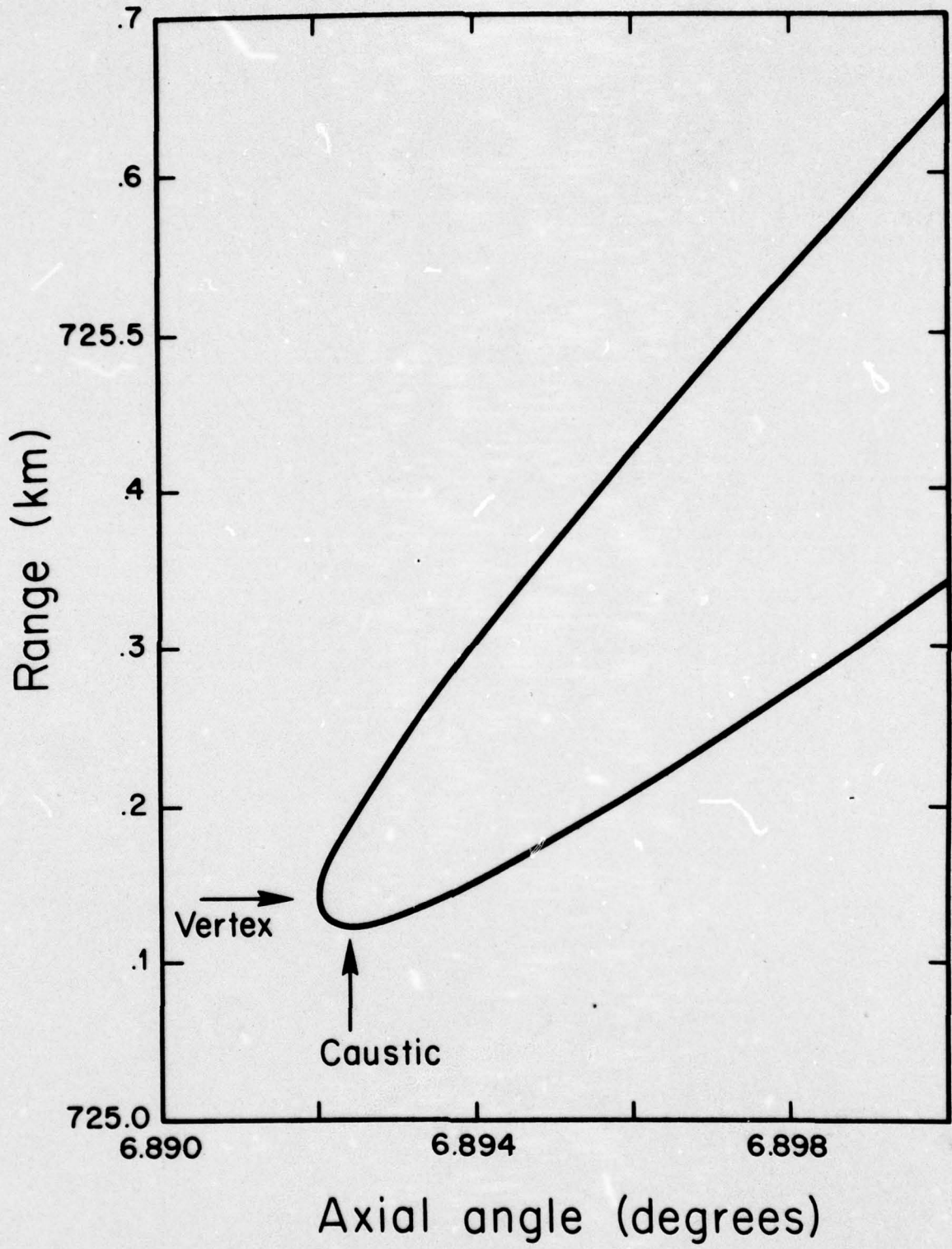


Fig. 13

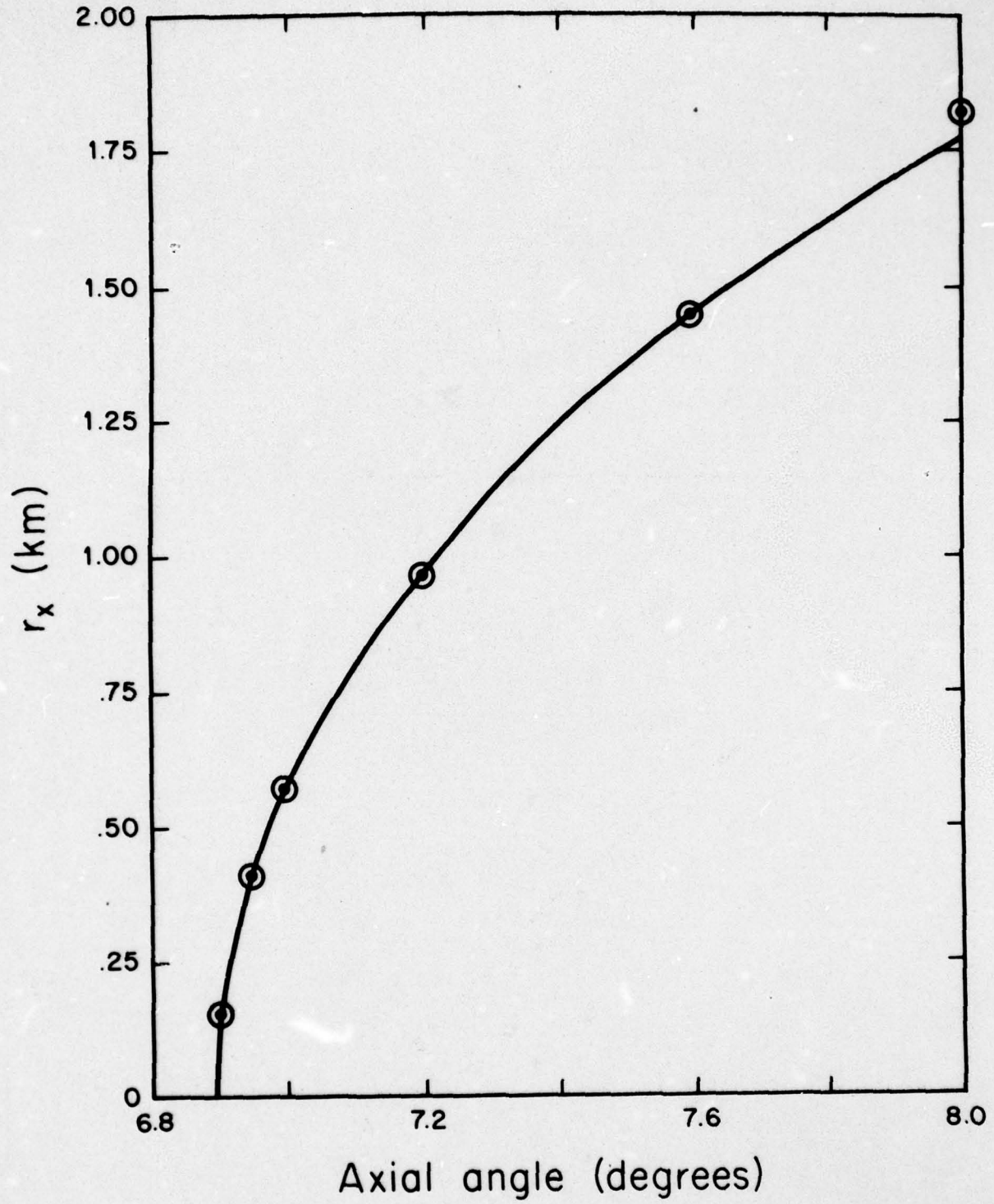


Fig. 14

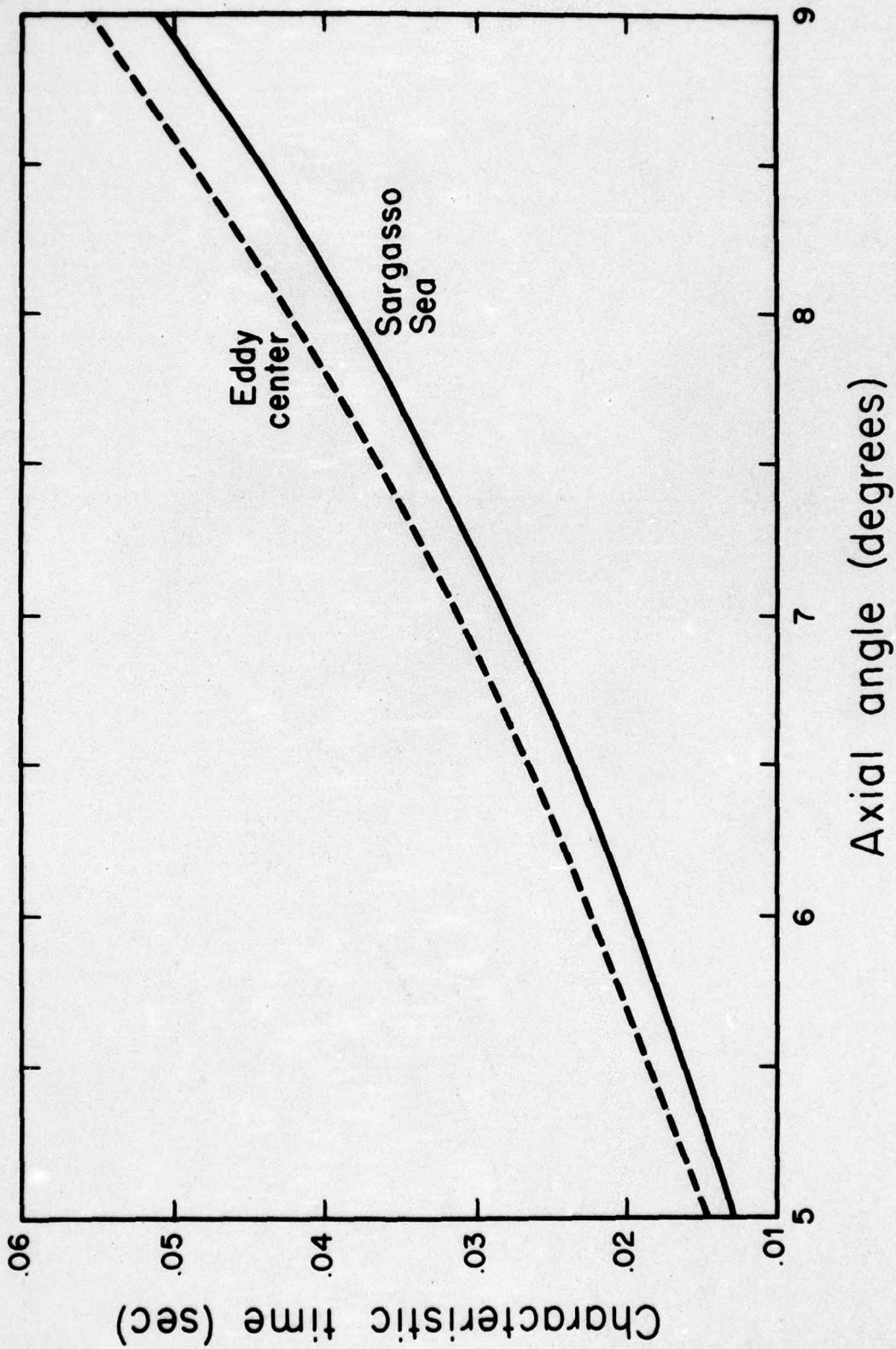


Fig. 15

Supplement to

RAY CALCULATIONS OF OCEAN SOUND CHANNELS  
USING A POCKET PROGRAMMABLE CALCULATOR  
AND EXTENDED FORMS OF THE HIRSCH-CARTER  
MATHEMATICAL MODEL WITH TABLES  
OF THE INCOMPLETE BETA-FUNCTION

L. Baxter, II

A. Tables of the Incomplete Beta-function

The complete beta-function,  $B(p, q)$  of the variables  $p$  and  $q$  is given by

$$B(p, q) = \Gamma(p) \cdot \Gamma(q) / \Gamma(p+q) \quad (1)$$

or alternatively by:

$$B(p, q) = \int_0^1 x^{(p-1)} (1-x)^{(q-1)} dx \quad (2)$$

The incomplete beta-function,  $B_x(p, q)$  is given by

$$B_x(p, q) = \int_0^x y^{(p-1)} (1-y)^{(q-1)} dy, \quad 0 < x < 1 \quad (3)$$

The function  $I_x(p, q)$  sometimes called the relative incomplete beta-function is given by

$$I_x(p, q) = B_x(p, q) / B(p, q) \quad (4)$$

In these tables and in those of Pearson<sup>1</sup>  $I_x$  is given as a function of  $p, q$  and  $x$ , while  $B(p, q)$  is tabulated at the top of each column of  $I_x(p, q)$ . In the present tables  $q$  is taken equal to .5, the only value it assumes in the Hirsch-Carter model equations.

In the paper to which this supplement is appended the Hirsch-Carter model equation is written

$$c^2 = c_0^2 (1 - |\alpha z|^\beta)^{-1} \quad (5)$$

and it is shown that range and travel time can be written for sound ray segments in terms of the following quantities

$$B_1 = B\left(\frac{1}{p}, .5\right) \tag{6}$$

$$I_1 = I_x\left(\frac{1}{p}, .5\right) \tag{7}$$

$$B_2 = B\left(1 + \frac{1}{p}, .5\right) \tag{8}$$

$$I_2 = I_x\left(1 + \frac{1}{p}, .5\right) \tag{9}$$

where  $\chi = \left(\frac{z}{z_v}\right)^p$  (10)

and  $z_v$  is the value of  $z$  for which a sound ray vertexes (i.e. becomes horizontal at its maximum or minimum depth of excursion). To use these tables for sound ray calculations,  $B_1$  is taken from the top of the column with  $p = \frac{1}{p}$  ;  $I_1$  is taken opposite  $\chi$  from the same column;  $B_2$  is taken from the top of the column with  $p = 1 + \frac{1}{p}$  ; and  $I_2$  is taken opposite  $\chi$  from that column.

For some computations it may be necessary to enter the tables with a value of  $I_1$  and interpolate to find a value of  $\chi$  .

The  $I_x(p, q)$  Function  
 $q = 0.5$        $p = .20$  to  $.30$



	X	P = .2	.25	.30		X	P = .2	.25	.30
	B(p, q)	6.268655		4.554444			6.268655		4.554444
	.01	.3178035		.184054		.51	.7357336		.6443241
	.02	.3653681		.2268620		.52	.7396286		.6473406
	.03	.3965678		.2565072		.53	.7435044		.6543426
	.04	.4204138		.2799599		.54	.7473630		.6593315
	.05	.4399803		.2997001		.55	.7512063		.6643100
	.06	.4567164		.3169302		.56	.7550366		.6692804
	.07	.4714288		.3323333		.57	.7588556		.6742451
	.08	.4846165		.3463399		.58	.7626655		.6792067
	.09	.4966091		.3592383		.59	.7664682		.6841673
	.10	.5076395		.3712348		.60	.7702656		.6891297
	.11	.5178767		.3824811		.61	.7740599		.6940961
	.12	.5274482		.3930920		.62	.7778531		.6990691
	.13	.5364528		.4031581		.63	.7816472		.7040516
	.14	.5449675		.4127504		.64	.7854444		.7090461
	.15	.5530558		.4219277		.65	.7892470		.7140552
	.16	.5607692		.4307382		.66	.7930571		.7190821
	.17	.5681497		.4392212		.67	.7968770		.7241296
	.18	.5752329		.4474108		.68	.8007092		.7292010
	.19	.5820503		.4553367		.69	.8045565		.7342995
	.20	.5886254		.4630221		.70	.8084209		.7394283
	.21	.5949836		.4704910		.71	.8123058		.7445919
	.22	.6011419		.4777536		.72	.8162140		.7497936
	.23	.6071190		.4848451		.73	.8201486		.7550378
	.24	.6129291		.4917640		.74	.8241130		.7603270
	.25	.6185856		.4985279		.75	.8281112		.7656721
	.26	.6241008		.5051492		.76	.8321468		.7710727
	.27	.6294845		.5116379		.77	.8362240		.7765365
	.28	.6347481		.5180048		.78	.8403481		.7820698
	.29	.6398979		.5242568		.79	.8445238		.7876795
	.30	.6449436		.5304035		.80	.8487573		.7933743
	.31	.6498911		.5364509		.81	.8530544		.7991621
	.32	.6547480		.5424062		.82	.8574232		.8050534
	.33	.6595198		.5482757		.83	.8618715		.8110591
	.34	.6642126		.5540657		.84	.8664085		.8171923
	.35	.6688304		.5597800		.85	.8710454		.8234678
	.36	.6733773		.5654249		.86	.8757945		.8299029
	.37	.6778630		.5710042		.87	.8806710		.8365184
	.38	.6822851		.5765225		.88	.8856927		.8433383
	.39	.6866506		.5819842		.89	.8908808		.8503926
	.40	.6909627		.5873926		.90	.8962620		.8577176
	.41	.6952244		.5927516		.91	.9018698		.8653592
	.42	.6994404		.5980653		.92	.9077468		.8733366
	.43	.7036116		.6033359		.93	.9139497		.8818481
	.44	.7077423		.6085674		.94	.9205569		.8908814
	.45	.7118349		.6137625		.95	.9276810		.9006315
	.46	.7158926		.6189235		.96	.9354948		.9113368
	.47	.7199167		.6240546		.97	.9442899		.9233995
	.48	.7239106		.6291571		.98	.9546366		.9376044
	.49	.7278763		.6342343		.99	.9620097		.9559839
	.50	.7318162		.6392893		1.00	1.0000000		1.0000000

$$B = 4.0566228$$

A	B
.350000	.5000000
N	K
25	9
K	I
.000100	.0280396
.000200	.0357387
.000300	.0411886
.000400	.0455524
.000500	.0492533
.000600	.0524994
.000700	.0554103
.000800	.0580623
.000900	.0605066

A	B
.350000	.5000000
N	K
25	9
X	I
.001000	.0627804
.002000	.0800277
.003000	.0922420
.004000	.1020267
.005000	.1103288
.006000	.1176139
.007000	.1241499
.008000	.1301070
.009000	.1356003

The  $I_x(p, q)$  Function  
 $q = 0.5$   $p = .35$  to  $.45$



x	P =			x	P =		
	.35	.40	.45		.35	.40	.45
B(p, q)	4.0566228	3.6790923	3.3820539		4.0566228	3.6790923	3.3820539
.01	.1407124	.1078504	.0828482	.51	.6050637	.5692794	.5365068
.02	.1795816	.1425152	.1133517	.52	.6105126	.5750914	.5426226
.03	.2072370	.1678532	.1362559	.53	.6159506	.5808963	.5487374
.04	.2294953	.1886002	.1553352	.54	.6213784	.5866976	.5548532
.05	.2484703	.2065132	.1720190	.55	.6268009	.5924978	.5609750
.06	.2652006	.2224676	.1870295	.56	.6322194	.5982989	.5671022
.07	.2802834	.2369720	.2007909	.57	.6376363	.6041037	.5732393
.08	.2940971	.2503520	.2135768	.58	.6430547	.6099150	.5793825
.09	.3068985	.2628293	.2255753	.59	.6484768	.6157355	.5855528
.10	.3188713	.2745643	.2369230	.60	.6539054	.6215677	.5917344
.11	.3301514	.2856761	.2477220	.61	.6593431	.6274146	.5979367
.12	.3408430	.2962561	.2580513	.62	.6647923	.6332787	.6041629
.13	.3510282	.3063771	.2679737	.63	.6702569	.6391632	.6104158
.14	.3607711	.3160962	.2775390	.64	.6757377	.6450714	.6166986
.15	.3701264	.3254619	.2867891	.65	.6812396	.6510065	.6230150
.16	.3791373	.3345131	.2957588	.66	.6867654	.6569715	.6293678
.17	.3878407	.3432824	.3044766	.67	.6923178	.6629704	.6357617
.18	.3962678	.3517990	.3129677	.68	.6979008	.6690063	.6422001
.19	.4044461	.3600866	.3212536	.69	.7035174	.6750832	.6486871
.20	.4123976	.3681658	.3293527	.70	.7091722	.6812059	.6552272
.21	.4201434	.3760556	.3372816	.71	.7148689	.6873782	.6618254
.22	.4277001	.3837714	.3450539	.72	.7206117	.6935952	.6684865
.23	.4350841	.3913282	.3526835	.73	.7264055	.6998915	.6752161
.24	.4423097	.3987384	.3601815	.74	.7322553	.7062432	.6820199
.25	.4493878	.4060128	.3675573	.75	.7381666	.7126657	.6889045
.26	.4563307	.4131626	.3748212	.76	.7441452	.7191660	.6958767
.27	.4631477	.4201962	.3819810	.77	.7501978	.7257510	.7029445
.28	.4698485	.4271229	.3890450	.78	.7563317	.7324284	.7101169
.29	.4764413	.4339495	.3960196	.79	.7625540	.7392069	.7174021
.30	.4829331	.4406840	.4029116	.80	.7688748	.7460967	.7248112
.31	.4893316	.4473321	.4097267	.81	.7753026	.7531077	.7323560
.32	.4956425	.4539000	.4164705	.82	.7818495	.7602529	.7400497
.33	.5018722	.4603935	.4231485	.83	.7885278	.7675460	.7479073
.34	.5080266	.4668183	.4297656	.84	.7953320	.7750028	.7559460
.35	.5141098	.4731785	.4363254	.85	.8023385	.7826417	.7641864
.36	.5201274	.4794785	.4428332	.86	.8095509	.7904840	.7726508
.37	.5260835	.4857231	.4492924	.87	.8168805	.7985555	.7813678
.38	.5319828	.4919162	.4557069	.88	.8244484	.8068863	.7903700
.39	.5378288	.4980616	.4620807	.89	.8323582	.8155129	.7996970
.40	.5436252	.5041628	.4684167	.90	.8405365	.8244806	.8093985
.41	.5493761	.5102239	.4747186	.91	.8490732	.8338465	.8195361
.42	.5550854	.5162477	.4809893	.92	.8580347	.8436838	.8301899
.43	.5607549	.5222378	.4872322	.93	.8675088	.8540894	.8414656
.44	.5663887	.5281965	.4934502	.94	.8776165	.8651971	.8533508
.45	.5719893	.5341268	.4996456	.95	.8885322	.8771970	.8665273
.46	.5775608	.5400325	.5058219	.96	.9005239	.8903904	.8808478
.47	.5831097	.5459157	.5119815	.97	.9140428	.9052704	.8970028
.48	.5886244	.5517796	.5181271	.98	.9299712	.9229111	.9160601
.49	.5941222	.5576259	.5242614	.99	.9505906	.9455295	.9407555
.50	.5996014	.5634587	.5303868	1.00	1.0000000	1.0000000	1.0000000

The  $I_x(p, q)$  Function  
 $q = 0.5$   $p = .50$  to  $.70$



X	P =			X	P =		
	.50	.60	.70		.50	.60	.70
B(p, q)	3.141593	2.7745031	2.5057947		3.141593	2.7745031	2.5057947
.01	.0637686	.0379735	.0227433	.51	.5063666	.4527878	.4065981
.02	.0903345	.0576663	.0370236	.52	.5127358	.4595364	.4135910
.03	.1108247	.0736896	.0492781	.53	.5191101	.4663035	.4206160
.04	.1281884	.0877416	.0603989	.54	.5254920	.4730921	.4276764
.05	.1435663	.1005064	.0707605	.55	.5318843	.4799036	.4347748
.06	.1575424	.1123443	.0805656	.56	.5382845	.4867417	.4419128
.07	.1704634	.1234739	.0899398	.57	.5447103	.4936084	.4490941
.08	.1825549	.1340395	.0989678	.58	.5511494	.5005074	.4563214
.09	.1939734	.1441428	.1077098	.59	.5576098	.5074403	.4635974
.10	.2048328	.1538592	.1162111	.60	.5640942	.5144110	.4709253
.11	.2152190	.1632460	.1245067	.61	.5706057	.5214230	.4783086
.12	.2251989	.1723479	.1326241	.62	.5771474	.5284784	.4857498
.13	.2348255	.1812013	.1405862	.63	.5837226	.5355818	.4932539
.14	.2441418	.1898354	.1484110	.64	.5903345	.5427362	.5008236
.15	.2531833	.1982753	.1561146	.65	.5969867	.5499452	.5084635
.16	.2619798	.2065412	.1637101	.66	.6036829	.5572135	.5161774
.17	.2705563	.2146510	.1712089	.67	.6104271	.5645447	.5239704
.18	.2789343	.2226199	.1786208	.68	.6172233	.5719434	.5318468
.19	.2871326	.2304615	.1859550	.69	.6240760	.5794148	.5398124
.20	.2951672	.2381868	.1932187	.70	.6309899	.5869637	.5478718
.21	.3030525	.2458069	.2004193	.71	.6379699	.5945959	.5560322
.22	.3108011	.2533303	.2075625	.72	.6450216	.6023171	.5642993
.23	.3184242	.2607658	.2146543	.73	.6521506	.6101336	.5726806
.24	.3259319	.2681208	.2217001	.74	.6593633	.6180531	.5811834
.25	.3333333	.2754016	.2287039	.75	.6666667	.6260829	.5898160
.26	.3406367	.2826152	.2356706	.76	.6740681	.6342311	.5985883
.27	.3478494	.2897665	.2426040	.77	.6815758	.6425074	.6075099
.28	.3549784	.2968611	.2495079	.78	.6891989	.6509222	.6165923
.29	.3620301	.3039036	.2563858	.79	.6969475	.6594862	.6258479
.30	.3690101	.3108989	.2632412	.80	.7048328	.6682124	.6352907
.31	.3759240	.3178504	.2700768	.81	.7128674	.6771146	.6449362
.32	.3827767	.3247637	.2768956	.82	.7210657	.6862097	.6548027
.33	.3895729	.3316408	.2837010	.83	.7294437	.6955155	.6649100
.34	.3963171	.3384864	.2904952	.84	.7380202	.7050531	.6752819
.35	.4030133	.3453030	.2972809	.85	.7468167	.7148473	.6859450
.36	.4096655	.3520944	.3040609	.86	.7558582	.7249259	.6969310
.37	.4162774	.3588630	.3108370	.87	.7651745	.7353230	.7082773
.38	.4228526	.3656125	.3176118	.88	.7748011	.7460788	.7200286
.39	.4293943	.3723454	.3243879	.89	.7847810	.7572420	.7322327
.40	.4359058	.3790641	.3311675	.90	.7951672	.7688726	.7449745
.41	.4423902	.3857726	.3379527	.91	.8060266	.7810467	.7583203
.42	.4488506	.3924716	.3447459	.92	.8174451	.7938616	.7723840
.43	.4552897	.3991648	.3515493	.93	.8295361	.8074467	.7873092
.44	.4617105	.4058544	.3583647	.94	.8424576	.8219792	.8032923
.45	.4681157	.4125430	.3651944	.95	.8561337	.8377155	.8206178
.46	.4745080	.4192328	.3720406	.96	.8707116	.8550479	.8397213
.47	.4808899	.4259264	.3789058	.97	.8861753	.8746394	.8613367
.48	.4872642	.4326264	.3857918	.98	.9026655	.8977824	.8868971
.49	.4936334	.4393345	.3927010	.99	.9362314	.9278185	.9201049
.50	.5000000	.4460540	.3996360	1.00	1.0000000	1.0000000	1.0000000

The  $I_x(p, q)$  Function  
 $q = 0.5$        $p = .80$  to  $1.0$



X	P =			X	P =		
	.80	.90	1.0		.80	.90	1.0
B(p, q)	2.2992875	2.1347606	2.000000		2.2992875	2.1347606	2.000000
.01	.0136863	.0082688	.0050126	.51	.3663957	.3311312	.3000000
.02	.0238828	.0154671	.0100505	.52	.3735271	.3383188	.3071797
.03	.0331087	.0223325	.0151142	.53	.3807054	.3455681	.3143345
.04	.0417718	.0290028	.0202041	.54	.3879334	.3528807	.3217670
.05	.0500507	.0355405	.0253206	.55	.3952132	.3602604	.3291796
.06	.0580445	.0419816	.0304640	.56	.4025477	.3677084	.3366750
.07	.0658163	.0483495	.0356349	.57	.4099399	.3752282	.3442561
.08	.0734091	.0546607	.0408337	.58	.4173921	.3828230	.3519259
.09	.0808541	.0609272	.0460608	.59	.4249077	.3904952	.3596876
.10	.0881753	.0671584	.0513167	.60	.4324888	.3982482	.3675445
.11	.0953912	.0733620	.0566019	.61	.4401416	.4060860	.3755002
.12	.1025169	.0795439	.0619168	.62	.4478667	.4140115	.3835586
.13	.1095647	.0857094	.0672621	.63	.4556689	.4220294	.3917237
.14	.1165446	.0918629	.0726382	.64	.4635524	.4301435	.4000000
.15	.1234656	.0980081	.0780456	.65	.4715214	.4383522	.4083920
.16	.1303352	.1041486	.0834849	.66	.4795797	.4466783	.4169048
.17	.1371599	.1102871	.0889566	.67	.4877332	.4551088	.4255437
.18	.1439455	.1164266	.0944615	.68	.4959861	.4636554	.4343146
.19	.1506971	.1225693	.1000000	.69	.5043447	.4723230	.4432236
.20	.1574192	.1287174	.1055728	.70	.5128148	.4811201	.4522774
.21	.1641163	.1348737	.1111806	.71	.5214022	.4900523	.4614835
.22	.1707919	.1410391	.1168239	.72	.5301144	.4991264	.4708497
.23	.1774496	.1472163	.1225036	.73	.5389596	.5083517	.4803248
.24	.1840928	.1534067	.1282202	.74	.5479448	.5177364	.4900000
.25	.1907242	.1596119	.1339746	.75	.5570799	.5272899	.5000000
.26	.1973468	.1658340	.1397675	.76	.5663755	.5370244	.5101021
.27	.2039632	.1720740	.1455996	.77	.5758410	.5469501	.5204168
.28	.2105759	.1783337	.1514719	.78	.5854923	.5570816	.5309524
.29	.2171872	.1846145	.1573850	.79	.5953357	.5674319	.5417424
.30	.2237997	.1909180	.1633400	.80	.6053935	.5780193	.5527864
.31	.2304150	.1972454	.1693376	.81	.6156796	.5888605	.5641101
.32	.2370357	.2035984	.1753789	.82	.6262141	.5999773	.5757359
.33	.2436638	.2099783	.1814647	.83	.6370200	.6113940	.5876894
.34	.2503012	.2163866	.1875962	.84	.6481212	.6231374	.6000000
.35	.2569497	.2228247	.1937742	.85	.6595480	.6352395	.6127017
.36	.2636115	.2292939	.2000000	.86	.6713347	.6477369	.6258343
.37	.2702887	.2357957	.2062746	.87	.6835220	.6606746	.6394449
.38	.2769818	.2423317	.2125992	.88	.6961580	.6741049	.6535898
.39	.2836944	.2489033	.2189750	.89	.7093040	.6880913	.6683375
.40	.2904274	.2555121	.2254033	.90	.7230307	.7027128	.6837722
.41	.2971830	.2621593	.2318854	.91	.7374305	.7180686	.7000000
.42	.3039628	.2688471	.2384227	.92	.7526220	.7342862	.7171573
.43	.3107689	.2755765	.2450166	.93	.7687612	.7515343	.7354249
.44	.3176032	.2823495	.2516685	.94	.7860636	.7700454	.7550510
.45	.3244675	.2891679	.2583802	.95	.8048385	.7901537	.7763932
.46	.3313637	.2960333	.2651531	.96	.8255625	.8123723	.8000000
.47	.3382940	.3029472	.2719890	.97	.8490350	.8375651	.8267949
.48	.3452601	.3099120	.2788897	.98	.8768218	.8674173	.8585786
.49	.3522643	.3169293	.2858572	.99	.9129584	.9062814	.9000000
.50	.3593085	.3240014	.2928932	1.00	1.0000000	1.0000000	1.0000000

The  $I_x(p, q)$  Function  
 $q = 0.5$   $p = 1.20$  to  $1.30$



X	P =			X	P =		
	1.20	1.25	1.30		1.20	1.25	1.30
B(e, q)	1.791044		1.7079163		1.791044		1.7079163
.01	.0018574		.0011545	.51	.2477474		.2257117
.02	.0042789		.0028016	.52	.2547680		.2326015
.03	.0069799		.0047596	.53	.2618896		.2396036
.04	.0098854		.0069383	.54	.2691146		.2467207
.05	.0129573		.0093005	.55	.2764461		.2539558
.06	.0161721		.0118229	.56	.2838863		.2613124
.07	.0195142		.0144893	.57	.2914383		.2687922
.08	.0229719		.0172878	.58	.2991059		.2764006
.09	.0265367		.0202093	.59	.3068916		.2841390
.10	.0302018		.0232466	.60	.3147995		.2920125
.11	.0339619		.0263940	.61	.3228343		.3000247
.12	.0378126		.0296468	.62	.3309989		.3081808
.13	.0417505		.0330009	.63	.3392990		.3164848
.14	.0457724		.0364532	.64	.3477383		.3249422
.15	.0498761		.0400009	.65	.3563224		.3335584
.16	.0540595		.0436418	.66	.3650570		.3423386
.17	.0583208		.0473736	.67	.3739483		.3512894
.18	.0626586		.0511949	.68	.3830017		.3604180
.19	.0670718		.0551043	.69	.3922251		.3697312
.20	.0715591		.0591004	.70	.4016259		.3792372
.21	.0761202		.0631824	.71	.4112118		.3889443
.22	.0807540		.0673494	.72	.4209920		.3988621
.23	.0854602		.0716008	.73	.4309763		.4090011
.24	.0902387		.0759362	.74	.4411750		.4193720
.25	.0950887		.0803550	.75	.4515998		.4299871
.26	.1000105		.0848571	.76	.4622641		.4408605
.27	.1050039		.0894423	.77	.4731808		.4520065
.28	.1100691		.0941107	.78	.4843680		.4634424
.29	.1152061		.0988623	.79	.4958407		.4751861
.30	.1204153		.1036973	.80	.5076206		.4872589
.31	.1256968		.1086157	.81	.5197282		.4996834
.32	.1310512		.1136181	.82	.5321901		.5124872
.33	.1364789		.1187049	.83	.5450344		.5257003
.34	.1419806		.1238767	.84	.5582944		.5393571
.35	.1475578		.1291340	.85	.5720084		.5534986
.36	.1532084		.1344775	.86	.5862206		.5681710
.37	.1589358		.1399079	.87	.6009847		.5834306
.38	.1647403		.1454263	.88	.6163640		.5993444
.39	.1706225		.1510334	.89	.6324342		.6159931
.40	.1765838		.1567302	.90	.6492921		.6334761
.41	.1826252		.1625180	.91	.6670551		.6519184
.42	.1887478		.1683981	.92	.6858768		.6714818
.43	.1949528		.1743714	.93	.7059602		.6923794
.44	.2012416		.1804397	.94	.7275835		.7149035
.45	.2076157		.1866042	.95	.7511480		.7394756
.46	.2140769		.1928666	.96	.7772681		.7667414
.47	.2206264		.1992287	.97	.8069774		.7977864
.48	.2272663		.2056922	.98	.8422912		.8347259
.49	.2339981		.2122587	.99	.8884081		.8830156
.50	.2408242		.2189312	1.00	1.0000000		1.0000000

The  $I_x(p, q)$  Function  
 $q = 0.5$   
 $p = 1.35$  to  $1.45$



X	P =			X	P =		
	1.35	1.40	1.45		1.35	1.40	1.45
B(p, q)	1.6703740	1.6351522	1.6020257		1.6703740	1.6351522	1.6020257
.01	.0008874	.0006944	.0005436	.51	.2155571	.2059278	.1967939
.02	.0022686	.0018378	.0014895	.52	.2223721	.2126637	.2034436
.03	.0039333	.0032518	.0026896	.53	.2293076	.2195208	.2102207
.04	.0058170	.0048791	.0040942	.54	.2363578	.2265037	.2171289
.05	.0078854	.0066885	.0056757	.55	.2435353	.2336167	.2241715
.06	.0101162	.0086598	.0074161	.56	.2508384	.2408605	.2313510
.07	.0124942	.0107787	.0093026	.57	.2582709	.2482399	.2386709
.08	.0150080	.0130346	.0113255	.58	.2658378	.2557528	.2461355
.09	.0176488	.0154194	.0134773	.59	.2735407	.2634196	.2537482
.10	.0204095	.0179265	.0157521	.60	.2813853	.2712272	.2615131
.11	.0232847	.0205506	.0181450	.61	.2893745	.2791267	.2694353
.12	.0262696	.0232873	.0206520	.62	.2975137	.2873013	.2775186
.13	.0293605	.0261329	.0232696	.63	.3058068	.2955766	.2857687
.14	.0325541	.0290844	.0259952	.64	.3142600	.3040184	.2941917
.15	.0358477	.0321393	.0288262	.65	.3228786	.3126311	.3027925
.16	.0392390	.0352953	.0317608	.66	.3316685	.3214223	.3115772
.17	.0427260	.0385506	.0347971	.67	.3406358	.3303978	.3205535
.18	.0463070	.0419035	.0379337	.68	.3497876	.3395653	.3297280
.19	.0499808	.0453527	.0411695	.69	.3591318	.3489316	.3391093
.20	.0537461	.0488972	.0445033	.70	.3686761	.3585057	.3487052
.21	.0576019	.0525360	.0479344	.71	.3784298	.3682969	.3585258
.22	.0615473	.0562683	.0514621	.72	.3884017	.3783144	.3685805
.23	.0655819	.0600936	.0550858	.73	.3986034	.3885694	.3788804
.24	.0697051	.0640113	.0588053	.74	.4090455	.3990735	.3894376
.25	.0739163	.0680212	.0626202	.75	.4197404	.4098393	.4002657
.26	.0782157	.0721231	.0665303	.76	.4307030	.4208814	.4113792
.27	.0826022	.0763167	.0705357	.77	.4419478	.4322158	.4227939
.28	.0870777	.0806023	.0746365	.78	.4534926	.4438601	.4345284
.29	.0916405	.0849798	.0788327	.79	.4653359	.4558328	.4466020
.30	.0962913	.0894495	.0831247	.80	.4775593	.4681573	.4590377
.31	.1010304	.0940117	.0875128	.81	.4901259	.4808563	.4718592
.32	.1058583	.0986667	.0919974	.82	.5030084	.4939588	.4850919
.33	.1107752	.1034150	.0965791	.83	.5161462	.5074972	.4987826
.34	.1157818	.1082574	.1012586	.84	.5303036	.5215068	.5129538
.35	.1208788	.1131943	.1060364	.85	.5446417	.5360308	.5276538
.36	.1260666	.1182264	.1109135	.86	.5595267	.5511180	.5429326
.37	.1313462	.1233548	.1158905	.87	.5750166	.5668273	.5587512
.38	.1367186	.1285801	.1209688	.88	.5911802	.5832292	.5754806
.39	.1421845	.1339035	.1261490	.89	.6080991	.6004074	.5929075
.40	.1477450	.1393261	.1314325	.90	.6258759	.6184667	.6112385
.41	.1534014	.1448489	.1368206	.91	.6446393	.6375388	.6306080
.42	.1591549	.1504736	.1423145	.92	.6645540	.6577923	.6511885
.43	.1650068	.1562012	.1479156	.93	.6858379	.6794502	.6732085
.44	.1709585	.1620332	.1536256	.94	.7087910	.7028193	.6967807
.45	.1770114	.1679712	.1594459	.95	.7333846	.7283403	.7229561
.46	.1831674	.1740170	.1653785	.96	.7616591	.7566887	.7518244
.47	.1894282	.1801724	.1714252	.97	.7933456	.7890002	.7817455
.48	.1957954	.1864392	.1775879	.98	.8310679	.8274868	.8239784
.49	.2022713	.1928194	.1838686	.99	.8804061	.8778507	.8753457
.50	.2088517	.1993156	.1902697	1.00	1.0000000	1.0000000	1.0000000

The  $I_x(p, q)$  Function  
 $q = 0.5$   $P = 1.50$  to  $1.70$



$x$	$P =$			$x$	$P =$		
	1.50	1.60	1.70		1.50	1.60	1.70
$B(p, q)$	1.5707965	1.51336534	1.46171357		1.5707965	1.51336534	1.46171357
.01	.0002257	.0002264	.0001607	.51	.1881204	.1720482	.1575098
.02	.0012077	.0007948	.0005238	.52	.1946807	.1784197	.1636832
.03	.0022255	.0015254	.0010470	.53	.2013737	.1849339	.1700056
.04	.0034369	.0024247	.0017130	.54	.2082024	.1915915	.1764812
.05	.0048182	.0034762	.0025114	.55	.2151699	.1983976	.1831132
.06	.0063536	.0046686	.0034352	.56	.2222797	.2053550	.1899049
.07	.0080318	.0059939	.0044792	.57	.2295352	.2124681	.1968598
.08	.0098443	.0074459	.0056395	.58	.2369403	.2197405	.2039849
.09	.0117844	.0090197	.0069129	.59	.2444990	.2271770	.2112819
.10	.0138468	.0107115	.0082971	.60	.2522155	.2347810	.2187560
.11	.0160272	.0125179	.0097900	.61	.2600945	.2425588	.2264131
.12	.0183220	.0144365	.0113901	.62	.2681408	.2505137	.2342590
.13	.0207281	.0164651	.0130960	.63	.2763598	.2586527	.2422987
.14	.0232430	.0186017	.0149068	.64	.2847570	.2669817	.2505393
.15	.0258646	.0208450	.0168215	.65	.2933384	.2755065	.2589867
.16	.0285911	.0231936	.0188396	.66	.3021105	.2842342	.2676487
.17	.0314210	.0256466	.0209605	.67	.3110804	.2931721	.2765328
.18	.0343530	.0282032	.0231840	.68	.3202554	.3023285	.2856469
.19	.0373861	.0308625	.0255098	.69	.3296437	.3117115	.2950010
.20	.0405193	.0336242	.0279379	.70	.3392541	.3213303	.3046035
.21	.0437521	.0364879	.0304683	.71	.3490960	.3311948	.3144661
.22	.0470837	.0394533	.0331010	.72	.3591800	.3413163	.3245945
.23	.0505139	.0425203	.0358364	.73	.3695172	.3517061	.3350160
.24	.0540424	.0456889	.0386747	.74	.3801201	.3623778	.3457297
.25	.0576689	.0489591	.0416163	.75	.3910022	.3733449	.3567551
.26	.0613934	.0523313	.0446617	.76	.4021785	.3846241	.3681090
.27	.0652160	.0558055	.0478114	.77	.4136655	.3962313	.3798093
.28	.0691369	.0593823	.0510660	.78	.4254815	.4081877	.3918763
.29	.0731562	.0630620	.0544264	.79	.4376470	.4205124	.4043317
.30	.0772743	.0668452	.0578931	.80	.4501849	.4332315	.4172013
.31	.0814916	.0707325	.0614671	.81	.4631209	.4463699	.4305125
.32	.0858087	.0747245	.0651494	.82	.4764843	.4599597	.4442974
.33	.0902262	.0788222	.0689408	.83	.4903085	.4740353	.4585929
.34	.0947447	.0830263	.0728427	.84	.5046316	.4886364	.4734396
.35	.0993650	.0873379	.0768559	.85	.5194980	.5038095	.4888867
.36	.1040880	.0917580	.0809819	.86	.5349594	.5196079	.5049890
.37	.1089147	.0962873	.0852219	.87	.5510771	.5360966	.5218147
.38	.1138459	.1009276	.0895772	.88	.5679242	.5533510	.5394423
.39	.1188830	.1056799	.0940495	.89	.5855812	.5714641	.5579674
.40	.1240271	.1105455	.0986403	.90	.6041813	.5905489	.5775089
.41	.1292794	.1155259	.1033511	.91	.6238377	.6107485	.5982147
.42	.1346415	.1206231	.1081840	.92	.6447345	.6322467	.6202759
.43	.1401147	.1258382	.1131406	.93	.6671049	.6552858	.6439439
.44	.1457008	.1311731	.1182229	.94	.6912688	.6801987	.6695614
.45	.1514014	.1366298	.1234331	.95	.7176856	.7074636	.6976339
.46	.1572183	.1422103	.1287733	.96	.7470601	.7378134	.7289120
.47	.1631535	.1479168	.1342458	.97	.7805761	.7724779	.7646744
.48	.1692091	.1537511	.1398531	.98	.8205388	.8138530	.8074039
.49	.1753872	.1597160	.1455976	.99	.8728886	.8681089	.8634939
.50	.1816901	.1658139	.1514823	1.00	1.0000000	1.0000000	1.0000000

The  $I_x(p, q)$  Function  
 $q = 0.5$   $P = 1.80$  to  $2.00$



$x$	$P = 1.80$	$1.90$	$2.00$	$x$	$P = 1.80$	$1.90$	$2.00$
$B(p, q)$	1.41494618	1.3723461	1.3333333		1.41494618	1.3723461	1.3333333
.01	.0000989	.0000610	.0000376	.51	.1443350	.1323756	.1215000
.02	.0003457	.0002284	.0001510	.52	.1503035	.1381338	.1270464
.03	.0007195	.0004950	.0003409	.53	.1564277	.1440543	.1327597
.04	.0012116	.0008580	.0006092	.54	.1627111	.1501398	.1386441
.05	.0018166	.0013155	.0009536	.55	.1692588	.1563963	.1447040
.06	.0025307	.0018664	.0013779	.56	.1757737	.1628265	.1509441
.07	.0033513	.0025102	.0018821	.57	.1825609	.1694359	.1573691
.08	.0042765	.0032465	.0024670	.58	.1895243	.1762289	.1639844
.09	.0053046	.0040750	.0031335	.59	.1966690	.1832110	.1707954
.10	.0064347	.0049458	.0038825	.60	.2039998	.1903864	.1778078
.11	.0076658	.0060090	.0047150	.61	.2115223	.1977630	.1850278
.12	.0089973	.0071148	.0056319	.62	.2192430	.2053451	.1924618
.13	.0104287	.0083137	.0066341	.63	.2271671	.2131401	.2001167
.14	.0119599	.0096059	.0077228	.64	.2353022	.2211552	.2080000
.15	.0135906	.0109920	.0088990	.65	.2436540	.2293973	.2161194
.16	.0153208	.0124725	.0101636	.66	.2522317	.2378753	.2244834
.17	.0171506	.0140481	.0115180	.67	.2610427	.2465960	.2331009
.18	.0190802	.0157194	.0129630	.68	.2700951	.2555725	.2419815
.19	.0211098	.0174871	.0145000	.69	.2793995	.2648070	.2511357
.20	.0232399	.0193521	.0161301	.70	.2889651	.2743174	.2605745
.21	.0254708	.0213152	.0178545	.71	.2988036	.2841128	.2703102
.22	.0278031	.0233772	.0196745	.72	.3089266	.2942055	.2803556
.23	.0302373	.0255393	.0215915	.73	.3193473	.3046099	.2907252
.24	.0327742	.0278022	.0236066	.74	.3300791	.3153396	.3014343
.25	.0354142	.0301671	.0257214	.75	.3411386	.3264115	.3125000
.26	.0381585	.0326352	.0279372	.76	.3525430	.3378441	.3239408
.27	.0410077	.0352075	.0302556	.77	.3643304	.3496556	.3357773
.28	.0439627	.0378853	.0326779	.78	.3764628	.3618702	.3480322
.29	.0470246	.0406698	.0352059	.79	.3890021	.3745097	.3607307
.30	.0501944	.0435624	.0378410	.80	.4020156	.3876027	.3739010
.31	.0534732	.0465645	.0405847	.81	.4154717	.4011788	.3875747
.32	.0568622	.0496775	.0434395	.82	.4294238	.4152731	.4017277
.33	.0603626	.0529029	.0464084	.83	.4439105	.4299245	.4165806
.34	.0639758	.0562424	.0494875	.84	.4589740	.4451785	.4320000
.35	.0677031	.0596975	.0526847	.85	.4746650	.4610863	.4480999
.36	.0715462	.0632700	.0560000	.86	.4910412	.4777085	.4649430
.37	.0755063	.0669616	.0594354	.87	.5081726	.4951177	.4826134
.38	.0795853	.0707744	.0629931	.88	.5261414	.5133982	.5011694
.39	.0837942	.0747101	.0666752	.89	.5450468	.5326538	.5207477
.40	.0881067	.0787708	.0704840	.90	.5650113	.5530114	.5414697
.41	.0925527	.0829587	.0744219	.91	.5861892	.5744297	.5635000
.42	.0971250	.0872720	.0784915	.92	.6087779	.5970741	.5870496
.43	.1018255	.0917250	.0826951	.93	.6330386	.6225358	.6123974
.44	.1066565	.0963082	.0870356	.94	.6593288	.6494586	.6399250
.45	.1116202	.1010220	.0915157	.95	.6881628	.6790203	.6701200
.46	.1167193	.1058871	.0961383	.96	.7203268	.7120304	.7040000
.47	.1219559	.1108884	.1009064	.97	.7571400	.7498522	.7427905
.48	.1273328	.1161346	.1058233	.98	.8011712	.7951359	.7892822
.49	.1328531	.1216289	.1108922	.99	.8519072	.8547016	.8505000
.50	.1385196	.1273745	.1161165	1.00	1.0000000	1.0000000	1.0000000

The  $I_x(p, q)$  Function

$q = 0.5$

Miscellaneous values of  $p$  corresponding to Profile Fits used in this paper



$x$	$p = .347261$			$x$	$p = .347261$		
	$B(p, q)$	.7994692	.7994086		$B(p, q)$	.7994692	.7994086
.01	.1427377	.0137231	.0141019	.51	.6071199	.3665951	.3686270
.02	.1818832	.0239382	.0245072	.52	.6125470	.3737262	.3757522
.03	.2096584	.0331782	.0338926	.53	.6179621	.3809040	.3829242
.04	.2319922	.0418532	.0426879	.54	.6233685	.3881313	.3901449
.05	.2510183	.0501422	.0510803	.55	.6287684	.3954104	.3974165
.06	.2677842	.0581450	.0591742	.56	.6341641	.4027442	.4047421
.07	.2828922	.0659247	.0670354	.57	.6395582	.4101353	.4121239
.08	.2967235	.0735247	.0747091	.58	.6449534	.4175867	.4195657
.09	.3095368	.0809764	.0822279	.59	.6503519	.4251009	.4270695
.10	.3215170	.0883037	.0896167	.60	.6557562	.4326820	.4346387
.11	.3328013	.0955252	.0968951	.61	.6611701	.4403325	.4422768
.12	.3434938	.1026560	.1040784	.62	.6665949	.4480568	.4499882
.13	.3536778	.1097087	.1111798	.63	.6720340	.4558575	.4577752
.14	.3634175	.1166930	.1182095	.64	.6774905	.4637395	.4656427
.15	.3727677	.1236182	.1251771	.65	.6829672	.4717069	.4735942
.16	.3817722	.1304916	.1320901	.66	.6884671	.4797636	.4816351
.17	.3904679	.1373200	.1389552	.67	.6939934	.4879155	.4897695
.18	.3988863	.1441089	.1457789	.68	.6995499	.4961667	.4980025
.19	.4070547	.1508637	.1525660	.69	.7051400	.5045234	.5063485
.20	.4149955	.1575889	.1593215	.70	.7107675	.5129913	.5147885
.21	.4227298	.1642887	.1660497	.71	.7164364	.5215768	.5233532
.22	.4302744	.1709670	.1727545	.72	.7221513	.5302869	.5320419
.23	.4376459	.1776272	.1794396	.73	.7279164	.5391296	.5408618
.24	.4448579	.1842726	.1861083	.74	.7337371	.5481128	.5498211
.25	.4519219	.1909061	.1927635	.75	.7396188	.5572455	.5589292
.26	.4588504	.1975307	.1994084	.76	.7455676	.5665383	.5681958
.27	.4656526	.2041490	.2060454	.77	.7515891	.5760014	.5776317
.28	.4723383	.2107636	.2126776	.78	.7576916	.5856477	.5872502
.29	.4789149	.2173766	.2193067	.79	.7638822	.5954907	.5970632
.30	.4853911	.2239904	.2259357	.80	.7701701	.6055452	.6070870
.31	.4917730	.2306071	.2325661	.81	.7765645	.6158283	.6173379
.32	.4980673	.2372290	.2392008	.82	.7830769	.6263596	.6278355
.33	.5042799	.2438582	.2458416	.83	.7897199	.6371616	.6386024
.34	.5104166	.2504966	.2524906	.84	.7965077	.6482593	.6496632
.35	.5164825	.2571461	.2591496	.85	.8034570	.6596825	.6610475
.36	.5224823	.2638087	.2658204	.86	.8105869	.6714650	.6727895
.37	.5284202	.2704863	.2725055	.87	.8179204	.6836480	.6849296
.38	.5343004	.2771806	.2792064	.88	.8254851	.6962802	.6975169
.39	.5401278	.2838936	.2859251	.89	.8333138	.7094208	.7106096
.40	.5459054	.2906271	.2926633	.90	.8414471	.7231429	.7242806
.41	.5516372	.2973831	.2994230	.91	.8499366	.7375373	.7386209
.42	.5573268	.3041633	.3062061	.92	.8588481	.7527231	.7537487
.43	.5629767	.3109696	.3130145	.93	.8682693	.7688563	.7698193
.44	.5685908	.3178040	.3198502	.94	.8783201	.7861518	.7870466
.45	.5741713	.3246684	.3267149	.95	.8891742	.8049197	.8057395
.46	.5797223	.3315645	.3336106	.96	.9010978	.8256348	.8263712
.47	.5852456	.3384948	.3405395	.97	.9145396	.8490986	.8497386
.48	.5907446	.3454606	.3475033	.98	.9303765	.8768733	.8773978
.49	.5962209	.3524644	.3545042	.99	.9508770	.9129950	.9133672
.50	.6016788	.3595085	.3615447	1.00	.0000000	.0000000	.0000000

The  $I_x(p, q)$  Function

$q = 0.5$

Miscellaneous values of  $p$  corresponding to Profile fits used in this paper

$x$	$P =$			$x$	$P =$		
	1.347261	1.799469	1.794086		1.347261	1.799469	1.794086
$B(p, q)$	1.6723683	1.4151829	1.4175897	1.6723683	1.4151829	1.4175897	
.01	.0008994	.0001015	.0001018	.51	.2160995	.1444021	.1458788
.02	.0022949	.0003534	.0003543	.52	.2229184	.1503714	.1516597
.03	.0039745	.0007340	.0007356	.53	.2298547	.1564965	.1571959
.04	.0058734	.0012341	.0012367	.54	.2369121	.1627814	.1634912
.05	.0079569	.0018480	.0018516	.55	.2440923	.1692297	.1699497
.06	.0102029	.0025720	.0025768	.56	.2513986	.1758463	.1765757
.07	.0125959	.0034032	.0034092	.57	.2588345	.1826336	.1833724
.08	.0151246	.0043396	.0043469	.58	.2664031	.1895980	.1903456
.09	.0177800	.0053796	.0053882	.59	.2741089	.1967434	.1974995
.10	.0205553	.0065219	.0065319	.60	.2819546	.2040750	.2048326
.11	.0234448	.0077658	.0077772	.61	.2899455	.2115980	.2123704
.12	.0264439	.0091104	.0091233	.62	.2980856	.2193196	.2200985
.13	.0295488	.0105553	.0105698	.63	.3063802	.2272443	.2280303
.14	.0327561	.0121002	.0121163	.64	.3148338	.2353803	.2361718
.15	.0360632	.0137450	.0137627	.65	.3234524	.2437328	.2445301
.16	.0394678	.0154895	.0155088	.66	.3322414	.2523106	.2531132
.17	.0429678	.0173338	.0173547	.67	.3412082	.2611220	.2619285
.18	.0465618	.0192781	.0193007	.68	.3503591	.2701749	.2709857
.19	.0502482	.0213226	.0213469	.69	.3597018	.2794799	.2802933
.20	.0540258	.0234676	.0234936	.70	.3692442	.2890458	.2898528
.21	.0578938	.0257136	.0257413	.71	.3789955	.2988842	.2997020
.22	.0618511	.0280610	.0280965	.72	.3889650	.3090073	.3098260
.23	.0658974	.0305105	.0305417	.73	.3991631	.3194278	.3202471
.24	.0700320	.0330626	.0330956	.74	.4096017	.3301599	.3309790
.25	.0742545	.0357181	.0357529	.75	.4202924	.3412192	.3420370
.26	.0785648	.0384777	.0385142	.76	.4312505	.3526232	.3534389
.27	.0829626	.0413422	.0413804	.77	.4424899	.3643901	.3652029
.28	.0874481	.0443126	.0443526	.78	.4540293	.3765423	.3773589
.29	.0920212	.0473898	.0474315	.79	.4658860	.3891014	.3899945
.30	.0966820	.0505749	.0506183	.80	.4780828	.4020841	.4029916
.31	.1014309	.0538689	.0539140	.81	.4906418	.4155508	.4163398
.32	.1062682	.0572730	.0573199	.82	.5035923	.4295012	.4302226
.33	.1111944	.0607885	.0608371	.83	.5169641	.4439365	.4447539
.34	.1162101	.0644166	.0644670	.84	.5307927	.4590490	.4599183
.35	.1213158	.0681589	.0682109	.85	.5451205	.4747386	.4756077
.36	.1265121	.0720166	.0720703	.86	.5599939	.4911138	.4919835
.37	.1318001	.0759913	.0760467	.87	.5754718	.5082437	.5091228
.38	.1371804	.0800848	.0801418	.88	.5916219	.5262103	.5270919
.39	.1426541	.0842985	.0843572	.89	.6085261	.5451139	.5459956
.40	.1482221	.0886344	.0886947	.90	.6262873	.5650762	.5659362
.41	.1538858	.0930944	.0931562	.91	.6450332	.5862517	.5870971
.42	.1596463	.0976804	.0977437	.92	.6649289	.6088380	.6096455
.43	.1655050	.1023944	.1024593	.93	.6861922	.6330959	.6338721
.44	.1714631	.1072387	.1073051	.94	.7091221	.6593826	.6599239
.45	.1775222	.1122154	.1122833	.95	.7341498	.6882126	.6887136
.46	.1836842	.1173272	.1173966	.96	.7619348	.7203716	.7208263
.47	.1899506	.1225765	.1226472	.97	.7935863	.7571795	.7575785
.48	.1963232	.1279657	.1280378	.98	.8312662	.8012036	.8015341
.49	.2028041	.1334980	.1335714	.99	.8805479	.8590524	.8592893
.50	.2093955	.1391760	.1392509	1.00	1.0000000	1.0000000	1.0000000

## B. Fortran Programs for Calculating Tables of the Incomplete

## Beta-Function

Abramowitz and Stegun<sup>2</sup> give a series expansion for the  $I_x$  function which is equivalent to the following:

$$I_x(p, q) = \frac{x^p (1-x)^q}{p B(p, q)} \left[ 1 + \sum_{n=1}^{\infty} \frac{B(p+1, n)}{B(p+q, n)} x^n \right] \quad (11)$$

This converges well if  $x < .5$ .

For  $.5 < x < 1$  the symmetry relation:

$$I_x(a, b) = 1 - I_{(1-x)}(b, a) \quad (12)$$

may be used to evaluate Equation 11 within its region of good convergence. The following Fortran programs use equations 1, 11, and 12 and a polynomial approximation<sup>2</sup> to the gamma function to tabulate  $I_x$ .

The main program "TABLE" calls the other functions, accepts the input parameters, and prints the output. When TABLE is called it requests a logical unit number for the output with "ENTER IPRNT". When this is typed in, the line "ENTER A, B, N, X0, K" appears. Type in, in free field form, the quantities p, q, N, X0, K where p and q are the quantities we have been using (p and .5), N is the number of terms in the expansion (25 for 6 place accuracy), X0 is just smaller than the smallest desired value (X0=0 for a complete table column), and K equals the number of tabular entries desired in the column to be printed. When these are typed in, the line "ENTER DELX" appears. Type in the increment between successive values of  $x$  and the program will type out the table with

$$x = x_0 + I * DELX$$

with  $I = 1$  to  $K$

Finally, the line "NEW VALUES, YE or NO" appears. Type in NO to exit the program or YE to repeat.

FTN4. L

PROGRAM TABLE

```

5005 FORMAT("ENTER IPRNT")
5000 FORMAT("ENTER A, B, N, X0, K")
5012 FORMAT(5X, I4, 10X, F14.7)
5010 FORMAT(5X, F9.6, 4X, F14.7)
5020 FORMAT(7X, "X", 20X, "I")
5030 FORMAT("NEW VALUES, YE OR NO")
5040 FORMAT(A2)
5045 FORMAT(7X, "A", 20X, "B")
5047 FORMAT(7X, "N", 20X, "K")
5060 FORMAT("ENTER DELX")

```

IYES=2HYE

IITTY=1

IOTTY=1

WRITE(IOTTY, 5005)

READ(IITTY, \*)IPRNT

100 CONTINUE

WRITE(IOTTY, 5060)

READ(IITTY, \*)DELX

WRITE(IOTTY, 5000)

READ(IITTY, \*)A, B, N, X0, K

WRITE(IPRNT, 5045)

WRITE(IPRNT, 5010)A, B

WRITE(IPRNT, 5047)

WRITE(IPRNT, 5048)N, K

5048 FORMAT(5X, I4, 10X, I14)

WRITE(IPRNT, 5020)

DO 200 I=1, K

X=X0+DELX\*I

OUT=BI(X, A, B, N)

WRITE(IPRNT, 5010)X, OUT

200 CONTINUE

WRITE(IOTTY, 5030)

READ(IITTY, 5040)IQUER

IF(IQUER. EQ. IYES)GO TO 100

END

FUNCTION BI(X, A, B, N)

Xa=X\*\*A

Xb=(1.0-X)\*\*B

FCTR=Xa\*Xb/(A\*B\*BETA(A, B))

IF(X.GT.0.50)FCTR=Xa\*Xb/(B\*BETA(A, B))

BIN=1.0

AP=A+1.0

IF(X.GT.0.50)AP=B+1.0

AB=A+B

Xu=X

IF(X.GT.0.50)Xu=1.0-X

DO 10 I=1, N

XI=XQ\*\*I

YI=I

BE1=BETA(AP, YI)

BE2=BETA(AB, YI)

BIN=BIN+XI\*BE1/BE2

10 CONTINUE

BI=BIN+FCTR

IF(X.GT.0.50)BI=1.0-BI

RETURN

END

FUNCTION BETA(P,Q)

PG=GAMMA(P)  
QG=GAMMA(Q)  
PQG=GAMMA(P+Q)  
BETA=PG\*QG/PQG  
RETURN  
END

FUNCTION GAMMA(A)

DOUBLE PRECISION AK(8),RG,AD,ALD,DJ,AI,AG  
DATA AK/-.577191652D0,.988205891D0,  
1 -.897056937D0,.918206857D0,-.756704078D0,  
2 .482199394D0,-.193527818D0,.035868343D0/

AD=A

AG=1.0

KG=1.0

IF(AD.GE.1.0D0)GO TO 19

DO 10 I=1,8

RG=RG+AD\*\*I\*AK(I)

10 CONTINUE

RG=RG/AD

CONTINUE

15 GAMMA=RG

RETURN

19 J=AD

IF(AD.EQ.1.0D0)J=0

IF(AD.EQ.2.0D0)J=1

DJ=J

ALD=AD-DJ

DO 25 I=1,8

RG=RG+ALD\*\*I\*AK(I)

25 CONTINUE

IF(AD.LE.2.0D0)GO TO 15

J=J-1

26 CONTINUE

AI=J

AG=AG\*(ALD+AI)

J=J-1

IF(J.GT.0)GO TO 26

RG=RG\*AG

GO TO 15

END

ENDS

C. Programs for Profile Fitting and Acoustic Ray Plotting in Long Range Deep Ocean Sound Transmission Studies

The following programs written for the Texas Instruments SR56 programmable pocket calculator would be easily adaptable to other programmables using algebraic notation.

Tables C1A and C1B give operating instructions and a listing of a program that finds values of  $\alpha$  and  $\beta$  for a Hirsch-Carter model fit to the three points:  $c_0, 0$ ;  $c_1, z_1$ ; and  $c_2, z_2$ . The program iterates from a trial value of  $\beta$  to find  $\alpha$  and a more accurate value of  $\beta$ . The iteration can be continued to any desired precision. The operating steps 7 and 8 use the final values of  $\alpha$  and  $\beta$  to calculate  $c$  as a function of  $z$  for comparison with the empirical profile.

Tables C2A and C2B give operating instructions and a listing of a program that calculates slope ( $dc/dz$ ) and sound speed as a function of  $z$  in a Hirsch-Carter type profile with parameters  $\alpha$ ,  $\beta$ , and  $c_0$ .

Tables C3A and C3B give operating instructions and a listing of a program that calculates the depth increment  $z$  that corresponds to a given sound velocity in a Hirsch-Carter type profile with parameters  $\alpha$ ,  $\beta$ , and  $c_0$ .

Tables C4A and C4B give operating instructions and a listing of a program that calculates the depth increment,  $z$ , that corresponds to a given slope  $dc/dz$ , in a Hirsch-Carter type profile with parameters  $\alpha$ ,  $\beta$ , and  $c_0$ . The program iterates from a trial value of  $z$  to find a more accurate value. The iteration can be continued to any desired precision.

Tables C5A and C5B give operating instructions and a listing of a program that computes range and travel time of a ray segment or multiple ray segments in a Hirsch-Carter type profile. See Section V of the basic paper for more detail of the equations programmed. The parameters

TABLE C-1A HIRSCH-CARTER PROFILE CURVE FITTING



OPERATING INSTRUCTIONS

STEP	PROCEDURE	ENTER	PRESS				Display
1	KEY IN PROGRAM		2 <sup>nd</sup> CP	LRN			00 00
			ALL KEY	ENTRIES	IN TABLE	IB	98 00
			LRN	RST			0
2	PRELOAD REGISTERS ( $P_2 < P$ is a test value. Final value will replace it)	$C_1$	$X^2$	STO	1		$C_1$
		$C_2$	$X^2$	STO	2		$C_2$
		$Z_1$	STO	3			$Z_1$
		$C_3$	$X^2$	STO	7		$C_3$
		$Z_2$	STO	8			$Z_2$
		$A$	STO	9			A
		$R_1$	STO	4			
3	compute $P$		R/S				$P_1$
SEE NOTE	( $P$ will increment by A until value is just greater than ideal. Pauses show each test.)						$P_2$
							$P_3$
							$P_4$
							...
							$P$
4	CYCLE FOR NEXT SIGNIFICANT DIGIT		—	RCL	9	$\equiv$	$P - A$
			STO	4	RCL	9	
			$\div$	1	0	$\equiv$	$A_n = A/10$
			STO	9	RST	R/S	$P$
	The same as STEP 3						$P_1$
							$P_2$
							$P_3$
5	REPEAT 3,4 UNTIL $P$ HAS ENOUGH DIGITS						$P$
6	COMPUTE $\alpha$		RCL	6	2 <sup>nd</sup> $\sqrt{y}$		
			RCL	4	$\equiv$		$\alpha$
7	CALCULATE SVP FOR VARIOUS $Z_a$ , $Z_b$ , etc.	$Z_a$	GTO	7	7	R/S	$C_a$
		$Z_b$	R/S				$C_b$
		$Z_c$	R/S				$C_c$
		$Z_d$	R/S				$C_d$
8	REPEAT FITTING PROCESS FOR OTHER BRANCH OF SVP (LOOPS BACK TO STEP 2)		RST				

Note: if  $P$  does not change from  $P_2$ , either  $P_2$  is already larger than  $P$ , or trial points 1+2 are in the wrong order in memory. To reverse trial points in memory, key in GTO 57 R/S.

TABLE C-18 HIRSCH-CARTER PROFILE CURVE FITTING

		Register Contents									
		0	1	2	3	4	5	6	7	8	9
		$C_0^2$	$C_1^2$	$Z_1$	$\beta$	$\alpha^p$	$\alpha^{p_{prev}}$	$C_2^2$	$Z_2$	$\Delta$	
Preloaded		Program									
Loc.	Code	Key Entry	Comments	Loc.	Code	Key Entry	Comments				
			MAIN LOOP								
00	57	2nd subr		50	01	1	$C_0^2$				
01	03	3	to calculate	51	94	=	$C_2^2 - C_1^2$				
02	04	4		52	30	2nd PROD	$\alpha^p$				
03	33	STO		53	05	5					
04	06	6	$\alpha^p$ from $C_1, Z_1$	54	34	RCL					
05	57	2nd subr		55	05	5	$\alpha^p$				
06	05	5	to rotate registers	56	58	2nd rtn					
07	07	7	2-7, 3-8	57	34	RCL					
08	57	2nd subr		58	02	2	This subr rotates registers				
09	03	3	$\alpha^p$ from $C_2, Z_2$	59	39	2nd EXC					
10	04	4		60	07	7					
11	54	÷		61	33	STO	2-7,				
12	34	RCL		62	02	2	3-8				
13	06	6		63	34	RCL					
14	74	-		64	03	3					
15	01	1		65	39	2nd EXC					
16	94	=	$\alpha_2^p / \alpha_1^p - 1$	66	08	8					
17	56	2nd CP	clear test reg.	67	33	STO					
18	47	2nd X>T	if $\alpha_2^p > \alpha_1^p$	68	03	3					
19	07	7		69	58	2nd rtn					
20	00	0	Go TO 70	70	32	X>T					
21	34	RCL		71	57	2nd subr	RESTORE ORIG. Pos of register.				
22	04	4	$\beta$	72	05	5					
23	84	+		73	07	7					
24	34	RCL	$\Delta$	74	34	RCL	$\beta$ to x reg.				
25	09	9		75	04	4					
26	94	=	new $\beta = \beta + \Delta$	76	41	R/S					
27	59	2nd Pause	display $\beta$	77	45	$2^x$	2nd LOOP with Z in x reg. calculate C				
28	33	STO	STORE new $\beta$	78	34	RCL					
29	04	4		79	04	4					
30	57	2nd subr	to rotate registers to original position	80	64	X					
31	05	5		81	34	RCL					
32	07	7		82	06	6					
33	42	RST	TO NEXT MAIN LOOP	83	94	=					
34	34	RCL	THIS subr CALCULATES $\alpha^p$	84	93	+/-					
35	03	3		85	84	+					
36	45	$4^x$		86	01	1					
37	34	RCL		87	94	=					
38	04	4		88	54	÷					
39	94	=	$Z_1^p$	89	34	RCL					
40	20	2nd 1/x		90	01	1					
41	33	STO	$1/2,^p$ in 5	91	94	=					
42	05	5		92	48	2nd Vx					
43	34	RCL		93	20	2nd 1/x					
44	02	2	$C_1^2$	94	41	R/S					
45	12	INV		95	22	GTO	2nd LOOP TO calculate next C				
46	30	2nd PROD	$1/C_1^2 Z_1^p$ in 5	96	07	7					
47	05	5		97	07	7					
48	74	-		98							
49	34	RCL		99							

TABLE C-2 A  $dc/dz$  &  $c$  OF HIRSCH-CARTER PROFILE



OPERATING INSTRUCTIONS

STEP	PROCEDURE	ENTER	PRESS		Display
1	KEY IN PROGRAM		2nd CP	LRN	00 00
			ALL KEY	ENTRIES IN	
			TABLE	2 B	75 00
			LRN	RST	0
2	LOAD REGISTERS	$\alpha$	R/S		$\alpha$
		$\beta$	R/S		$\beta$
		$c_0$	R/S		$c_0$
3	CALCULATE $dc/dz$	Z	R/S		$dc/dz$
4	CALCULATE $c$		R/S		$c$
	REPEAT STEPS 3, 4 WITH NEW Z AS DESIRED				
5	TO CHANGE ALL PARAMETERS CONTINUE AT STEP 2		RST		

TABLE C-2 B  $dc/dz$  &  $c$  OF HIRSCH-CARTER PROFILE

		Register Contents									
		0	1	2	3	4	5	6	7	8	9
		$\alpha$	$\beta$	$c$	$Z$	$\alpha Z$					
Program											
Loc.	Code	Key Entry	Comments	Loc.	Code	Key Entry	Comments				
			LOAD REGISTERS								
00	33	STO	$\leftarrow \alpha$	39	34	RCL					
01	01	1		40	05	5					
02	41	R/S	$\leftarrow \beta$	41	45	$y^x$					
03	33	STO		42	34	RCL					
04	02	2		43	02	2					
05	41	R/S	$\leftarrow c$	44	53	)					
06	33	STO		45	45	$y^x$					
07	03	3		46	01	1					
08	41	R/S	$\leftarrow Z$	47	92	.					
09	33	STO	MAIN LOOP	48	05	5					
10	04	4		49	94	=					
11	64	X	THIS SECTION CALCULATES $dc/dz$	50	41	R/S		$dc/dz \rightarrow$			
12	34	RCL		51	34	RCL		THIS SECTION CALCULATE C			
13	01	1		52	04	4					
14	94	=		53	64	X					
15	33	STO		54	34	RCL					
16	05	5		55	01	1					
17	45	$y^x$		56	94	=					
18	52	(		57	45	$y^x$					
19	34	RCL		58	34	RCL					
20	02	2		59	02	2					
21	74	-		60	74	-					
22	01	1		61	01	1					
23	53	)		62	94	=					
24	64	X		63	93	+/-					
25	34	RCL		64	20	2nd $\sqrt{x}$					
26	03	3		65	64	X					
27	64	X		66	34	RCL					
28	34	RCL		67	03	3					
29	02	2		68	43	$x^2$					
30	64	X		69	94	=					
31	34	RCL		70	48	2nd $\sqrt{x}$					
32	01	1		71	41	R/S		$c \rightarrow$			
33	54	$\div$		72	22	GTO		$\leftarrow Z$			
34	02	2		73	00	0		RETURN TO MAIN LOOP			
35	54	$\div$		74	09	9					
36	52	(		75							
37	01	1		76							
38	74	-		77							

TABLE C-3A Depth for a given sound velocity. Hirsch-Carter Profile



OPERATING INSTRUCTIONS							
STEP	PROCEDURE	ENTER	PRESS			Display	
1	KEY IN PROGRAM		2nd CP	LRN		00 00	
			ALL ENTRIES	IN			
			TABLE	C-3B		32 00	
			LRN	RST		0	
2	LOAD REGISTERS	$\alpha$	R/S			$\alpha$	
		$\beta$	R/S			$\beta$	
		$c_0$	R/S			$c_0$	
3	CALCULATE	$c$	R/S			$Z$	
	2						
4	REPEAT STEP 3 AS DESIRED						
5	TO LOAD NEW CONSTANTS AND REPEAT STEPS 2, 3		RST				

TABLE C-3B Depth for a given sound velocity. Hirsch-Carter Profile



Register Contents									
0	1	2	3	4	5	6	7	8	9
	$\alpha$	$\beta$	$c_0$						

Program							
Loc.	Code	Key Entry	Comments	Loc.	Code	Key Entry	Comments
			LOAD REGISTERS	17	94	=	
			$\leftarrow \alpha$	18	93	+/-	$1 - c_0^2/c^2$
00	33	STO		19	45	$\chi^2$	
01	01	1		20	34	RCL	
02	41	R/S	$\leftarrow \beta$	21	02	2	
03	33	STO		22	20	2nd 1/X	$(1 - c_0^2/c^2)^{1/2}$
04	02	2		23	94	=	
05	41	R/S	$\leftarrow c_0$	24	54	$\div$	
06	33	STO		25	34	RCL	
07	03	3		26	01	1	
08	41	R/S	$\leftarrow c$	27	94	=	
09	20	2nd 1/X	MAIN LOOP	28	41	R/S	$Z \rightarrow$
10	64	$\chi^2$		29	22	GTO	$\leftarrow c$
11	34	RCL		30	00	0	RETURN TO
12	03	3		31	09	9	MAIN LOOP
13	94	=	$c^2/c^2$				
14	43	$\chi^2$					
15	74	-					
16	01	1					

TABLE C-4A Depth for a given slope. Hirsch-Carter Profile.



OPERATING INSTRUCTIONS

STEP	PROCEDURE	ENTER	PRESS				Display
1	KEY IN PROGRAM		2nd CP	LRN			00 00
			ALL KEY	ENTRIES	IN		
			Table	C-4B			70 00
			LRN	RST			0
2	LOAD REGISTERS	$\alpha$	R/S				$\alpha$
		$\beta$	R/S				$\beta$
		$C_0$	R/S				$C_0$
3	Put $dc/dz$ in t register	$dc/dz$	X $\Sigma$ T				0
4	Initialize $\Delta Z$ , $Z_t$ & START	$\Delta Z$	STO	6			$\Delta Z$
		$Z_t$	R/S				
5	PROGRAM PAUSES AT EACH NEW TRIAL Z HALTS AT FIRST Z WITH $dc/dz$ G.E. t						$Z_1$ $Z_2$ $Z_3$ $Z$
6	if desired restart for more accurate Z		-	RCL	6	=	$dc/dz$ *
			2nd EXC	4			
			RCL	6	$\div$	1	$\Delta Z / 10$
			0	=			
			STO	6	RCL	4	
			R/S				
	RECYCLES TO STEP 5						
	* $dc/dz$ here is trial value from last run						

TABLE C-4B Depth for a given slope. Mirsch-Carter Profile

		Register Contents									
		0	1	2	3	4	5	6	7	8	9
		$\alpha$	$\beta$	$C_0$	$Z$	$\alpha Z$	$\Delta Z$				
		Program									
Loc.	Code	Key Entry	Comments	Loc.	Code	Key Entry	Comments				
			LOAD REGISTERS	36	52	(					
00	33	STO	$\leftarrow \alpha$	37	01	1					
01	01	1		38	74	-					
02	41	R/S	$\leftarrow \beta$	39	34	RCL					
03	33	STO		40	05	5					
04	02	2		41	45	$\beta^x$					
05	41	R/S	$\leftarrow C_0$	42	34	RCL					
06	33	STO		43	02	2					
07	03	3		44	53	)	$(1 - (\alpha Z)^{\beta})$				
08	41	R/S	put $dc/dz$ in t register	45	45	$\beta^x$					
			$\Delta Z$ in 6	46	01	1					
			Initiate MAIN LOOP with trial $Z_t$	47	92	.					
				48	05	5	1.5				
				49	94	=	$dc/dz$				
09	33	STO	$\leftarrow Z_t$	50	12	INV					
10	04	4		51	47	2nd XZt	test whether to iterate $Z_t$				
11	64	X		52	06	6					
12	34	RCL		53	00	0					
13	01	1		54	39	2nd EXC	on final iteration				
14	94	=	$\alpha Z_t$	55	04	4					
15	33	STO		56	41	R/S	$Z \rightarrow$				
16	05	5					NEW $\Delta Z$ trial $Z$				
17	45	$\beta^x$									
18	52	)					REPEAT FOR MORE ACCURATE $Z$				
19	34	RCL		57	22	GTO					
20	02	2		58	00	0					
21	74	-		59	09	9					
22	01	1									
23	53	)		60	34	RCL					
24	64	X	$(\alpha Z)^{\beta-1}$	61	04	4	$Z_t$				
25	34	RCL		62	84	+					
26	03	3		63	34	RCL	$\Delta Z$				
27	64	X		64	06	6					
28	34	RCL		65	94	=					
29	02	2		66	59	2nd Pass	$Z + \Delta Z$				
30	64	X		67	22	GTO	To MAIN LOOP for next iteration				
31	34	RCL		68	00	0					
32	01	1		69	09	9					
33	54	$\div$									
34	02	2									
35	54	$\div$	$\alpha p C_0 (\alpha Z)^{\beta-1} / 2$								

$\alpha$  ,  $\beta$  ,  $c_0$  and values  $B_1$  and  $B_2$  of the beta-function (see Section 1 of this supplement are needed for initialization. If the segment is not an integral multiple of the path from reference level to vertex, values of  $I_1$  and  $I_2$  must be entered from a table of the incomplete beta-function (section 1 of this supplement). Otherwise  $I_1 = I_2 = 1$  . The program recycles for each new value of the reference angle  $\theta_0$  , which in a case 1 profile, fitted at the axis, is the axial angle,  $\theta_A$  .

Tables C6A and C6B give operating instructions and a listing of a program that calculates angles  $\theta_0$  and  $\theta_j$  at points  $c_0$  ,  $0$  and  $C_j$  ,  $Z_j$  of a Hirsch-Carter type profile with parameters  $\alpha$  ,  $\beta$  . The angles are calculated for given values of the  $X$  parameter of the incomplete beta-function to avoid interpolation in the tables when using the previous program (C5A,C5B) in this section.

Tables C7A and C7B give operating instructions and a listing of a program that calculates the characteristic time  $J$  (see Milder<sup>3</sup> and section IX of the basic paper) of a sound ray path when the axial sound speed  $c_0$  , the axial angle  $\theta_A$  , the full cycle range  $X$  and the full cycle travel time  $T$  are known. The characteristic time is needed to convert axial angles in one profile to those in the next when a ray propagates through a horizontal gradient.

Tables C8A and C8B give operating instructions and a listing of a program that calculates ray angle with the horizontal, and depth increment, of a ray in a Hirsch-Carter type profile. The independent variable is range, which may be specified from an axis crossing or a vertex. The program is initialized with values of  $\alpha$  ,  $\beta$  ,  $c_0$  , and the beta-function,  $B(\frac{1}{p}, \frac{1}{2})$  . The ray is designated by its reference angle  $\theta_0$  ,

TABLE C-5A Ray plot by Hirsch-Carter profile



NOTE: IF LOC 86 is 2nd rtn Computes Equations 8-10  
 IF LOC 86 is ( computes equations 9-11  
 of the main report

OPERATING INSTRUCTIONS

STEP	PROCEDURE	ENTER	PRESS	Display
1	KEY IN PROGRAM		2nd CP LRN	00 00
			ALL KEY ENTRIES IN	
			TABLE C-5B	93 00
			LRN RST	
2	PRELOAD 2N = 1 single segment 2N = 2 axis to axis or even number for higher order	2N	STO 6	2N
3	LOAD REGISTERS	$\alpha$	R/S	$\alpha$
		$\beta$	R/S	2/ $\beta$
		$B_1$	R/S	$B_1$
		$B_2$	R/S	$B_2$
		$C_0$	R/S	$C_0$
4	RUN MAIN LOOP	$\theta$	R/S	$Z_v$
		I	R/S	R
			R/S	
		$I_2$	R/S	T
5	REPEAT STEP 4 AS DESIRED WITH ANY $\theta$			
6	TO CHANGE CONSTANTS AND REPEAT STEPS 2, 5		RST	

TABLE C-5B Ray Plot by Hirsch-Carter Profile

Register contents									
0	1	2	3	4	5	6	7	8	9
	$\theta$	$B_1(1-I_1)$	$\alpha$	$\beta$	$c_0$	$2XN$	$2/\beta$	$B_1$	$B_2$
Program									
Loc.	Code	Key Entry	Comments	Loc.	Code	Key Entry	Comments		
			LOAD REGISTERS	49	54	$\div$			
00	33	STO	$\leftarrow \alpha$	50	34	RCL			
01	03	3		51	01	1			
02	41	R/S	$\leftarrow \beta$	52	25	tan			
03	33	STO		53	94	=			
04	04	4		54	41	R/S	$R \rightarrow$		
05	20	2nd 1/x		55	54	$\div$			
06	64	X		56	34	RCL			
07	02	2		57	05	5			
08	94	=	$2/\beta$	58	54	$\div$			
09	33	STO		59	34	RCL			
10	07	7		60	01	1			
11	41	R/S	$\leftarrow B_1$	61	24	cos			
12	33	STO		62	64	X			
13	08	8		63	52	(			
14	41	R/S	$\leftarrow B_2$	64	01	1			
15	33	STO		65	74	-			
16	09	9		66	34	RCL			
17	41	R/S	$\leftarrow c_0$	67	01	1			
18	33	STO		68	23	sin			
19	05	5		69	43	$x^2$			
20	41	R/S		70	54	$\div$			
			MAIN LOOP	71	34	RCL			
21	33	STO	$\leftarrow \theta$	72	02	2			
22	01	1		73	64	X			
23	23	sin		74	41	R/S	$\leftarrow I_2$		
24	45	$y^2$		75	57	2nd subtr	for choice of ray segmen		
25	34	RCL		76	08	8			
26	07	7		77	06	6			
27	54	$\div$		78	64	X			
28	34	RCL		79	34	RCL			
29	03	3		80	09	9			
30	94	=		81	94	=			
31	64	X		82	41	R/S	$T \rightarrow$		
32	52	(					$\leftarrow \theta$		
33	41	R/S	$Z_v \rightarrow$	83	22	GTO	RETURN TO BEGIN MAIN LOOP AGAIN		
			$\leftarrow I_1$	84	02	2			
34	57	2nd subtr	for choice of ray segment	85	01	1			
35	08	8					for segment ref. level to intermediate depth or		
36	06	6		86	58	2nd rtn			
37	64	X					for segment intermediate level to vertex		
38	34	RCL		87	52	(			
39	08	8		88	51	CE			
40	53	)	$B_1(1-I_1)$	89	74	-			
41	33	STO		90	01	1			
42	02	2		91	53	)			
43	64	X		9	93	+/-	$(1-I)$		
44	34	RCL			58	2nd rtn			
45	06	6							
46	54	$\div$							
47	34	RCL							
48	04	4							

TABLE C-6A Given  $x$  and  $c_j$ , find  $\theta_a$  and  $\theta_j$ . Hirsch-Carter profile



OPERATING INSTRUCTIONS

STEP	PROCEDURE	ENTER	PRESS			Display
			2nd CP	LRN	IN	
1	KEY IN PROGRAM			LRN		00 00
				ALL KEY	ENTRIES	
				TABLE	C-6B	
				LRN	RST	57 00
2	LOAD REGISTERS	$\alpha$	R/S			0
		$\beta$	R/S			$\alpha$
		$c_j$	R/S			$\beta$
		$z_j$	R/S			$c_j$
		$c_j$	R/S			$z_j$
3	RUN MAIN LOOP	$x$	R/S			$\theta_a$
			R/S			$\theta_j$
4	REPEAT STEP 3 WITH NEW $x$ AS DESIRED					
5	FOR NEW CONSTANTS AND REPEAT STEPS 2-4		RST			

TABLE C-6 B Given  $x$  and  $C_j$ , find  $\theta_a$  and  $\theta_j$ . Hirsch-Carter profile  
Register Contents



0	1	2	3	4	5	6	7	8	9
	$\alpha$	$\beta$	$C_0$	$Z_j$	$\theta_a$	$C_j$			

Program

Loc.	Code	Key Entry	Comments	Loc.	Code	Key Entry	Comments
			LOAD REGISTERS	29	52	(	
00	33	STO	$\leftarrow \alpha$	30	34	RCL	
01	01		1	31	02	2	
02	41	R/S	$\leftarrow \beta$	32	54	$\div$	
03	33	STO		33	02	2	
04	02		2	34	53	)	
05	41	R/S	$\leftarrow C_0$	35	94	=	$(\alpha Z_v)^{n/2}$
06	33	STO		36	12	INV	
07	03		3	37	23	SIN	
08	41	R/S	$\leftarrow Z_j$	38	33	STO	
09	33	STO		39	05	5	
10	04		4	40	41	R/S	$\theta_a \leftarrow$
11	41	R/S	$\leftarrow C_j$	41	34	RCL	
12	33	STO		42	05	5	
13	06		6	43	24	COS	
14	41	R/S		44	64	X	
			MAIN LOOP	45	34	RCL	
			$\leftarrow X$	46	06	6	
15	48	2nd $\sqrt{y}$		47	54	$\div$	
16	34	RCL		48	34	RCL	
17	02		2	49	03	3	
18	94	=		50	94	=	
19	20	2nd $1/x$		51	12	INV	
20	64	X		52	24	COS	
21	34	RCL		53	41	R/S	$\theta_j \leftarrow$
22	04		4				$\leftarrow X$
23	94	=	$Z_v$	54	22	GTO	LOOP AGAIN WITH NEW $X$
24	64	X		55	01	1	
25	34	RCL		56	05	5	
26	01		1				
27	94	=					
28	45	$y^x$					

TABLE C-7A Given full cycle range,  $X$ , and Travel Time,  $T$ , find characteristic time,  $J$ .  
see Milder, J. Acoust. Soc. Am., 46, 1259-1263 (1969)



OPERATING INSTRUCTIONS

STEP	PROCEDURE	ENTER	PRESS				Display
1	KEY IN PROGRAM		2nd CP	LRN			00 00
			ALL	KEY	ENTRIES	IN	
			TABLE	C-7B			18 00
			LRN	RST			0
2	LOAD $1/c_0$	$c_0$	2nd $1/x$	STO	1		$1/c_0$
3	AXIAL ANGLE	$\theta_A$	R/S				$n_m$
4	RANGE	$X$	R/S				$X \cdot n_m$
5	TRAVEL TIME	$T$	R/S				$J$
6	REPEAT STEPS 3-5 FOR NEW $\theta_A$ AS DESIRED						
7	IF NEW $c_0$ REPEAT STEPS 2-5						

TABLE C-7B Given full cycle range,  $X$ ,  $\theta$  Travel Time,  $T$ , Find Char. Time

Register Contents									
0	1	2	3	4	5	6	7	8	9
$1/c_0$									
Preloaded					Program				
Loc.	Code	Key Entry	Comments		Loc.	Code	Key Entry	Comments	
00	24	cos	← $\theta_A$		10	54	÷		
01	64	X			11	69	2nd $\pi$		
02	34	RCL			12	54	÷		
03	01	=			13	02	2		
04	94	=	$n_m$		14	94	=		
05	64	X			15	93	+/-		
06	41	R/S	← $X$		16	41	R/S	J →	
07	74	-			17	42	RST		
08	41	R/S	← $T$						
09	94	=							

which for a case 1 profile (see Section IV of the basic paper) fitted at the axis is the axial angle  $\theta_A$ . Range may be specified from an axis crossing or a vertex. Tables of the incomplete beta-function are used to convert  $I_1 = I_x\left(\frac{1}{p}, \frac{1}{2}\right)$  to  $x$ . This program can be used to generate data for a range annotated ray angle diagram as described by Flatte<sup>4</sup> and Cox<sup>5</sup>.

TABLE C-8A RANGE ANNOTATED RAY ANGLE DIAGRAM  
 COMPUTED BY HIRSCH-CARTER MODEL

\* IF  $I_1$  COMES OUT NEGATIVE OR G.T. 1 RANGE IS LARGER  
 THAN THAT OF REFERENCE LEVEL TO VERTEX PATH.

OPERATING INSTRUCTIONS

STEP	PROCEDURE	ENTER	PRESS				Display
			2nd CP	LRN	ENTRIES	IN	
1	KEY IN PROGRAM		ALL	LRN KEY			00 00
			TABLE	C-8B			86 00
			LRN	RST			
2	LOAD REGISTERS	$\alpha$	R/S				$\alpha$
		$\beta$	R/S				$\beta$
		$c_0$	R/S				$c_0$
		$B_1$	R/S				$B_1$
3	SELECT RAY BY REF. ANGLE	$\theta_0$	R/S				$Z_v$
4	CONTINUE		R/S				
5	SUPPLY RANGE READ $I_1$	R	R/S				$I_1$ *
6	LOOK UP X IN TABLE OF $I_x$ FUNCTION AND ENTER	X	R/S				Z
7	COMPUTE $\theta$ AT RANGE R AND DEPTH Z		R/S				$\theta$
8	REPEAT STEPS 4-7 AS NECESSARY TO GET DATA FOR PLOT						
9	FOR NEW RAY AND REPEAT STEPS 4-7	$\theta_0$	GTO	1	2	R/S	$Z_v$

TABLE C-8B RANGE ANNOTATED RAY ANGLE DIAGRAM

Register Contents.



0	1	2	3	4	5	6	7	8	9
	$\alpha$	$\beta$	$c_0$	$B_1$	$\theta_0$	$Z_v$	$Z$		

Program

Loc.	Code	Key Entry	Comments	Loc.	Code	Key Entry	Comments
			LOAD REGISTERS	44	94	=	$R \tan \theta_0 / B_1 Z$
00	33	STO	$\leftarrow \alpha$	45	46	NOP	for R
01	01	1		46	46	NOP	measured
02	41	R/S	$\leftarrow \beta$	47	46	NOP	from ref level
03	33	STO		48	46	NOP	or
04	02	2		45	93	+/-	For R
05	41	R/S	$\leftarrow c_0$	46	84	+	measured
06	33	STO		47	01	1	from vertex
07	03	3		48	94	=	
08	41	R/S	$\leftarrow B_1$				
09	33	STO		49	41	R/S	I, $\rightarrow$ FIND
10	04	4					$\leftarrow X$
11	41	R/S	SELECT RAY BY REF. ANGLE				IN TABLE
			$\leftarrow \theta_0$	50	40	2nd $\sqrt{y}$	
12	33	STO		51	34	RCL	
13	05	5		52	02	2	
14	23	SIN		53	64	X	$X^{1/p}$
15	45	$y^x$		54	34	RCL	
16	52	(		55	06	6	
17	02	2		56	94	=	
18	54	$\div$		57	33	STO	
19	34	RCL		58	07	7	
20	02	2		59	41	R/S	$Z \rightarrow$
21	53	)		60	34	RCL	
22	54	$\div$	$(\sin \theta_0)^{2/p}$	61	07	7	
23	34	RCL		62	64	X	
24	01	1		63	34	RCL	
25	94	=		64	01	1	
26	33	STO		65	94	=	$\alpha Z$
27	06	6		66	45	$y^x$	
28	41	R/S	$Z_v \rightarrow$	67	34	RCL	
			MAIN LOOP STARTS HERE	68	02	2	
29	34	RCL		69	74	-	
30	06	6		70	01	1	
31	20	2nd $1/x$		71	94	=	
32	54	$\div$		72	93	+/-	$(1 - (\alpha Z)^p)$
33	34	RCL		73	20	2nd $1/x$	
34	04	4		74	48	2nd $\sqrt{x}$	
35	64	X	$1/B_1 Z_v$	75	64	X	
			$\leftarrow R$	76	34	RCL	
36	41	R/S		77	05	5	
37	64	X		78	24	COS	
38	34	RCL		79	94	=	$\cos \theta$
39	02	2		80	12	INV	
40	64	X	$R \beta / B_1 Z_v$	81	24	COS	
41	34	RCL		82	41	R/S	$\theta \rightarrow$
42	05	5		83	22	GTO	RETURN TO START OF MAIN LOOP
43	25	TAN		84	02	2	
				85	09	9	

REFERENCES FOR SUPPLEMENT

1. E. S. Pearson and N. L. Johnson, "Tables of the Incomplete Beta-function 2nd Edition". Cambridge University Press (1968).
2. M. Abramowitz and I. A. Stegun, "Handbook of Mathematical Functions with Formulas, Graphs, and Mathematical Tables". U. S. Department of Commerce, National Bureau of Standards Applied Mathematics Series 55. Ninth printing. U. S. Government Printing Office, Washington, D.C. (1970).
3. D. Michael Milder, "Ray and Wave Invariants for SOFAR Channel Propagation". J.Acoust.Soc.Am. 46, pp 1259-1263. (1969).
4. Stanley M. Flatte, "Angle-depth diagram for use in underwater acoustics". J.Acoust.Soc.Am. 60, pp 1020-1023. (1976).
5. Henry Cox, "Approximate ray angle diagram". J.Acoust.Soc.Am. 61, pp 353-359. (1977).

DISTRIBUTION LIST

Director  
Strategic Systems Project Office (PM-1)  
Department of the Navy  
Washington, DC 20360

OASN (RE&S)  
Room 4D, 745 Pentagon  
Washington, DC 20301  
Attn: G. A. Cann

Assistant Director  
Ocean Control DDR&E  
Room 3D, 1048 Pentagon  
Washington, DC 20301

Chief of Naval Operations  
Department of the Navy  
Washington, DC 20350  
Attn: Code OP 02

OP 03  
OP 023  
OP 032  
OP 092U  
OP 095  
OP 095E  
OP 098  
OP 951  
OP 952  
OP 955  
OP 955F  
OP 966  
OP 967  
OP 981

Chief of Naval Material  
Department of the Navy  
Washington, DC 20360  
Attn: Code NMAT 08T  
NMAT 98T1  
NMAT 08T2  
NMAT 08T24  
NMAT 08T4

Commander  
Pacific Missile Test Center  
Point Mugu, CA 93047  
Attn: Library

Commanding Officer  
Fleet Numerical Weather Central  
Monterey, CA 93940

President  
Naval War College  
Newport, RI 02840  
Attn: Library

Commanding Officer  
Fleet ASW Training Center, Pacific  
San Diego, CA 92147

Commanding Officer  
Fleet ASW Training Center, Atlantic  
Naval Station  
Norfolk, VA 23511

Commanding Officer and Director  
Defense Documentation Center  
Cameron Station, Bldg. 5  
5010 Duke St.  
Alexandria, VA 22314

Commander  
Naval Oceanography Command  
NSTL Station, MS 39529

Anti-Submarine Warfare Systems  
Project Office  
Department of the Navy  
Washington, DC 20360  
Attn: Code PM-4 (A. Bernard)

Naval Ocean Systems Center  
San Diego, CA 92152

U. S. Coast Guard HQ  
400 Seventh St., SW  
Washington, DC 20591  
Attn: R & D Director

National Oceanic & Atmos. Admin.  
6001 Executive Blvd.  
Rockville, MD 20852

Applied Physics Laboratory  
University of Washington  
1013 NE Fortieth St.  
Seattle, WA 98195  
Attn: Library

Applied Research Laboratories  
University of Texas  
P. O. Box 8029  
Austin, TX 78712  
Attn: L. D. Hampton  
G. E. Ellis  
S. K. Mitchell  
K. C. Focke  
K. E. Hawker  
S. R. Rutherford  
T. G. Muir  
D. J. Shirley  
Library

Applied Research Laboratory  
Pennsylvania State Univ.  
P. O. Box 30  
State College, PA 16801  
Attn: Library

Applied Physics Laboratory  
Johns Hopkins University  
Laural, MD 20810  
Attn: G. L. Smith  
J. R. Austin  
Library

Marine Physical Laboratory  
University of California, San Diego  
San Diego, CA 92152  
Attn: F. N. Spiess  
V. C. Anderson  
F. H. Fisher  
Library

Lamont-Doherty Geol. Obs.  
Palisades, NY 10964  
Attn: G. M. Bryan  
W. J. Ludwig  
B. Tucholke  
R. E. Houtz  
J. E. Nafe  
H. R. Kutschale

Woods Hole Oceanographic Inst.  
Woods Hole, MA 02543  
Attn: J. Ewing  
E. E. Hayes

Library

TRACOR, Inc.  
6500 Tracor Lane  
Austin, TX 78721  
Attn: A. F. Wittenborn  
Library

Mechanics Research, Inc.  
7929 Westpark Dr.  
McLean, VA 22101  
Attn: N. Peck

Western Electric Co.  
Gelford Center  
P. O. Box 20046  
Greensboro, NC 27420  
Attn: R. Scudder

Applied Hydroacoustics, Inc.  
Montgomery Plaza Bldg.  
656 Quince Orchard Rd.  
Gaithersburg, MD 20760  
Attn: F. Ryder

TRW  
7600 Colshire Dr.  
McLean, VA 22101  
Attn: R. Murawski  
I. B. Gereben

TRACOR Inc.  
1601 Research Blvd.  
Rockville, MD 20850  
Attn: R. J. Urick

Planning Systems Inc.  
7900 Westpark Dr.  
McLean, VA 22101  
Attn: L. Solomon

Arthur D. Little, Inc.  
15 Acorn Park  
Cambridge, MA 02140  
Attn: W. G. Sykes  
G. Raisbeck

Commander  
Naval Electronic Systems Command  
Department of the Navy  
Washington, DC 20360  
Attn: Code 035  
320  
320 (J. Cybulski)  
320 (J. A. Sinsky)  
PME 124  
124-20  
124-30  
124-34 (J. Reeves)  
124-40  
124-60  
124TA

Commander  
Naval Sea Systems Command  
Department of the Navy  
Washington, DC 20362  
Attn: Code 06H1  
06H1-4

Commander  
Naval Air Systems Command  
Department of the Navy  
Washington, DC 20360  
Attn: Code 350  
360  
370  
604  
PMA 264

Office of Naval Research  
Department of the Navy  
Arlington, VA 22217  
Attn: Code 102-OS  
480  
483  
485  
486

U. S. Naval Oceanographic Office  
NSTL Station, MS 39522  
Attn: Code 3400  
3721 (R. Anderson)

Commander  
New London Laboratory  
Naval Underwater Systems Center  
New London, CT 06320  
Attn: P. Herstein  
F. R. DiNapoli  
S. R. Santaniello  
R. L. Deavenport  
R. W. Hasse  
J. J. Hanrahan  
R. K. Dullea  
B. F. Cole  
T. G. Bell

Commanding Officer  
Newport Laboratory  
Naval Underwater Systems Center  
Newport, RI 02840  
Attn: Library

Commander  
Naval Air Development Center  
Warminster, PA 18974  
Attn: J. Howard  
C. L. Bartberger  
P. S. Haas  
J. H. Rubisch  
Library

Director  
Naval Research Laboratory  
Department of the Navy  
Washington, DC 20375  
Attn: Code 2627  
8000  
8100  
8106 (R. K. Perry)  
8106 (H. S. Fleming)  
8108  
8120 (R. L. Dicus)  
8120 (R. H. Ferris)  
8160 (B. B. Adams)  
8403 (R. T. Swim)  
Library  
B. G. Hurdle  
O. I. Diachok

Chief Scientist  
Naval Underwater Sound Reference Division  
Naval Research Laboratory  
P. O. Box 8337  
Orlando, FL 32806  
Attn: Library

Commanding Officer  
Naval Coastal Systems Laboratory  
Panama City, FL 32401  
Attn: E. G. Mcleroy, Jr.  
W. Tolbert  
Library

Commander  
David W. Taylor Naval Ship  
Research and Development Center  
Bethesda, MD 20084  
Attn: Library

Officer in Charge  
Naval Ship R & D Center  
Annapolis, MD 21402  
Attn: Library

Superintendent  
Naval Postgraduate School  
Monterey, CA 93940  
Attn: H. Medwin  
O. B. Wilson, Jr.  
R. S. Andrews  
Library

Commander  
Naval Surface Weapons Center  
White Oak Laboratory  
Silver Spring, MD 20910

Commanding Officer  
Civil Engineering Laboratory  
Naval Construction Bn. Center  
Port Hueneme, CA 93043  
Attn: Library

Defense Advanced Research Projects Agency  
1400 Wilson Blvd.  
Arlington, VA 22209  
Attn: Dr. Gustafson

ARPA Research Center  
Unit 1, Bldg. 301A  
NAS Moffet Field, CA 94035  
Attn: E. L. Smith

Science Applications, Inc.  
8400 Westpark Dr.  
McLean, VA 22101  
Attn: J. S. Hanna  
C. W. Spofford

Bell Telephone Laboratories  
Whippany, NJ 07981  
Attn: R. Lauver  
Mr. Pennotti

Commander-in-Chief, Atlantic Fleet  
Naval Base  
Norfolk, VA 23511

Commander-in-Chief, U. S. Navy Europe  
FPO  
San Francisco, CA 96610

Commander-in-Chief, Pacific Fleet  
Box 11, FPO  
San Francisco, CA 96610  
Attn: Code 352

Commander  
Operational Test & Eval. Force  
Naval Base  
Norfolk, VA 23511

Commander  
Submarine Development Group 1  
FPO  
San Diego, CA 92132

Commander  
Submarine Development Group 2  
Naval Submarine Base-New London  
Groton, CT 06340

Office of Naval Research  
Department of the Navy  
NORDA Liaison Office  
Arlington, VA 22217  
Attn: K. W. Lackie

Office of Naval Research  
Resident Representative  
University of California  
La Jolla, CA 92093

Commander  
Naval Intelligence Support Center  
4301 Suitland Road  
Washington, DC 20390  
Attn: Code 222

Director of Naval Warfare Analysis  
Institute of Naval Studies  
1401 Wilson Blvd.  
Arlington, VA 22209

Director for Naval Matters  
Center for Naval Analysis  
Arlington, VA 22209  
Attn: C. E. Woods

Commander  
Naval Oceanographic Office  
Washington, DC 20374  
Attn: Code 3410 (J. L. Carrol)

Commanding Officer  
Naval Ocean Research and Development Activity  
NSTL Station, MS 39529  
Attn: Code 110

	Code 410
200 (K. W. Mackensie)	420
200 (CDR T. McClosky)	440
230 (H. E. Morris)	460
300 (H. Eppert)	500
320	500 (E. M. Stanley)
330	500 (M. G. Lewis)
340	600
341 (D. F. Fenner)	Library
350	
360	
360 (G. L. Maynard)	
300 (T. L. Holcomb)	
301 (A. Lowrie)	
362 (J. E. Matthews)	

ADDITIONAL DISTRIBUTION LIST-----

Dr. D. C. Stickler  
Courant Inst. Math. Sciences  
New York, NY 10012

Commander, Third Fleet  
FPO  
San Francisco, CA 96610  
Attn: N-32

Dr. G. H. Sutton  
Hawaii Inst. of Geophysics  
The University of Hawaii  
2525 Correa Rd.  
Honolulu, HI 96822

Commander, Seventh Fleet  
FPO  
San Francisco, CA 96610  
Attn: N-34

Dr. Morris Schulkin  
9325 Orchard Brook Dr.  
Potomac, MD 20854

Commander, Submarine Force  
U. S. Pacific Fleet  
FPO  
San Francisco, CA 96601

Dr. P. A. Rona  
National Oceanic and Atmos. Admin.  
15 Rickenbacker Causeway  
Miami, FL 33149

Marvin S. Weinstein  
Underwater Systems, Inc.  
3121 Georgia Ave.  
Silver Spring, MD 20910

Director  
Geophysics Laboratory  
University of Texas  
700 The Strand  
Galveston, TX 77550

Dr. C. S. Clay  
Dept. of Geology and Geophysics  
University of Wisconsin  
Madison, WI 53700

Dr. A. O. Williams, Jr.  
Dept. of Physics  
Brown University  
Providence, RI 02912

Dr. H. M. Uberall  
The Catholic University of America  
6220 Michigan Ave., NE  
Washington, DC 20017



<p>Woods Hole Oceanographic Institution WHOI-78-65</p> <p>RAY CALCULATIONS OF OCEAN SOUND CHANNELS USING A POCKET PROGRAMMABLE CALCULATOR AND EXTENDED FORMS OF THE HIRSCH-CARTER MATHEMATICAL MODEL WITH TABLES OF THE INCOMPLETE BETA FUNCTION by Lincoln Baxter II. Prepared for the Office of Naval Research under Contract N00014-77-C-0196. (70 pages)</p>	<p>1. Acoustic Ray Computations Sound Channels</p> <p>2.</p> <p>3. Programmable Pocket Calculators</p> <p>I. Baxter, Lincoln, II</p> <p>II. N00014-77-C-0196</p>	<p>Woods Hole Oceanographic Institution WHOI-78-65</p> <p>RAY CALCULATIONS OF OCEAN SOUND CHANNELS USING A POCKET PROGRAMMABLE CALCULATOR AND EXTENDED FORMS OF THE HIRSCH-CARTER MATHEMATICAL MODEL WITH TABLES OF THE INCOMPLETE BETA FUNCTION by Lincoln Baxter II. Prepared for the Office of Naval Research under Contract N00014-77-C-0196. (70 pages)</p>	<p>1. Acoustic Ray Computations Sound Channels</p> <p>2.</p> <p>3. Programmable Pocket Calculators</p> <p>I. Baxter, Lincoln, II</p> <p>II. N00014-77-C-0196</p>
<p>This card is UNCLASSIFIED</p>		<p>This card is UNCLASSIFIED</p>	
<p>Woods Hole Oceanographic Institution WHOI-78-65</p> <p>RAY CALCULATIONS OF OCEAN SOUND CHANNELS USING A POCKET PROGRAMMABLE CALCULATOR AND EXTENDED FORMS OF THE HIRSCH-CARTER MATHEMATICAL MODEL WITH TABLES OF THE INCOMPLETE BETA FUNCTION by Lincoln Baxter II. Prepared for the Office of Naval Research under Contract N00014-77-C-0196. (70 pages)</p>	<p>1. Acoustic Ray Computations Sound Channels</p> <p>2.</p> <p>3. Programmable Pocket Calculators</p> <p>I. Baxter, Lincoln, II</p> <p>II. N00014-77-C-0196</p>	<p>Woods Hole Oceanographic Institution WHOI-78-65</p> <p>RAY CALCULATIONS OF OCEAN SOUND CHANNELS USING A POCKET PROGRAMMABLE CALCULATOR AND EXTENDED FORMS OF THE HIRSCH-CARTER MATHEMATICAL MODEL WITH TABLES OF THE INCOMPLETE BETA FUNCTION by Lincoln Baxter II. Prepared for the Office of Naval Research under Contract N00014-77-C-0196. (70 pages)</p>	<p>1. Acoustic Ray Computations Sound Channels</p> <p>2.</p> <p>3. Programmable Pocket Calculators</p> <p>I. Baxter, Lincoln, II</p> <p>II. N00014-77-C-0196</p>
<p>This card is UNCLASSIFIED</p>		<p>This card is UNCLASSIFIED</p>	