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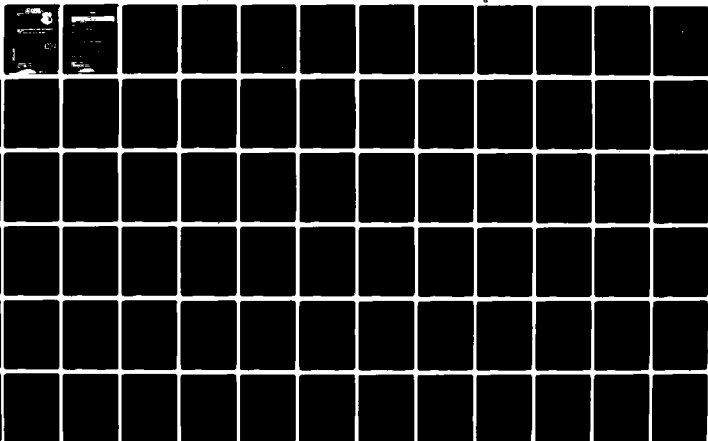
AEROSPACE MEDICAL RESEARCH LAB WRIGHT-PATTERSON AFB OH F/G 1/2
USAF BIODENVIRONMENTAL NOISE DATA HANDBOOK, VOLUME 75. C-141A AI--ETC(U)
JUL 79 R G POWELL
AMRL-TR-75-50-VOL-75

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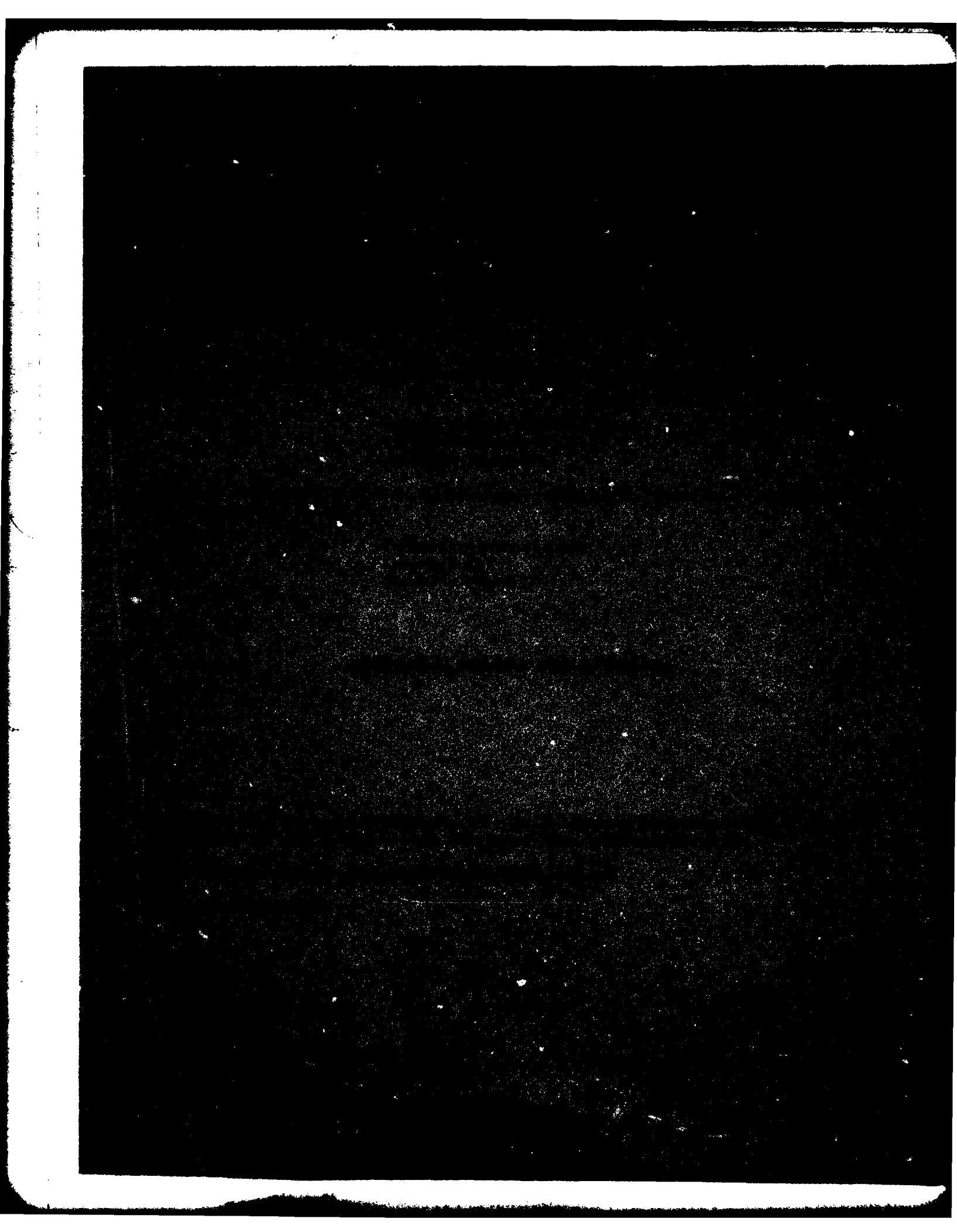
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REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER AMRL-TR-75-51 - VOL-75	2. GOVT ACCESSION NO.	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle) USAF BIOENVIRONMENTAL NOISE DATA HANDBOOK C-141A Aircraft, Near and Far-Field Noise		5. PERIOD COVERED Volume 75 Volume 75 of a series
6. AUTHOR(s) Robert G./Powell		7. PERFORMING ORG. REPORT NUMBER A048834
8. CONTRACT OR GRANT NUMBER(s)		
9. PERFORMING ORGANIZATION NAME AND ADDRESS Aerospace Medical Research Laboratory Aerospace Medical Division, Air Force Systems Command, Wright-Patterson AFB OH		10. PROGRAM ELEMENT, PROJECT, TASK AREA AND WORK UNIT NUMBERS 622021 7231-07-03 7231-08-07
11. CONTROLLING OFFICE NAME AND ADDRESS Same as above		12. REPORT DATE 11 July 79
13. NUMBER OF PAGES 85		
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)		15. SECURITY CLASS. (of this report) Unclassified
15a. DECLASSIFICATION/DOWNGRADING SCHEDULE		
16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited.		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from report) 17 07, 08		
18. SUPPLEMENTARY NOTES 9 Technical rept.		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Noise Noise Environments Bioenvironmental Noise C-141A Aircraft		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) → The USAF C-141A is a transport aircraft powered by four TF33-P-7 turbofan engines. This report provides measured and extrapolated data defining the bioacoustic environments produced by this aircraft operating on a concrete runup pad for three power conditions. Near-field data are reported for ten locations in a wide variety of physical and psychoacoustic measures: overall and band sound pressure levels, C-weighted and		

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A-weighted sound levels, preferred speech interference level, perceived noise level, and limiting times for total daily exposure of personnel with and without standard Air Force ear protectors. Far-field data measured at 19 locations are normalized to standard meteorological conditions and extrapolated from 75-8000 meters to derive sets of equal-value contours for these same seven acoustic measures as functions of angle and distance from the source. Refer to Volume 1 of this handbook, "USAF Bioenvironmental Noise Data Handbook, Vol 1: Organization, Content and Application", AMRL-TR-75-50(1) 1975, for discussion of the objective and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc.

PREFACE

This report was prepared by the Biodynamic Environment Branch, Aerospace Medical Research Laboratory, under Project/Task 723107, Technology to Define and Assess Environmental Quality of Noise from Air Force Operations and 723108, Crew Safety in Operational Noise Environments.

The author gratefully acknowledges Mr. John Cole for his assistance in preparing this report, Mr. Robert Lee, Mr. Jerry Speakman and Lt Thomas Rau for their assistance in acquiring the raw data, Mr. Henry Mohlman, Mr. Keith Kettler and Mr. Fred Lampley of the University of Dayton for assistance in the mechanics of data processing and Mrs. Peggy Massie for assistance in typing and preparation of the graphics.

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INTRODUCTION

The USAF C-141A is a transport aircraft powered by four TF33-P-7 turbofan engines. The aircraft was manufactured by the Lockheed Aircraft Corporation and the engines by United Aircraft, Pratt and Whitney Division.

This volume provides measured and extrapolated data defining bioacoustic environments produced by this aircraft during ground runup operations. Such data are essential to evaluate ear protection requirements, limiting personnel exposure times, voice communication capabilities, and annoyance problems associated with ground runups of the C-141A aircraft.

This volume is one of a series published by the Aerospace Medical Research Laboratory (AMRL) under the same report number (AMRL-TR-75-50) as a multi-volume handbook that quantifies the noise environments produced at flight/ground crew locations and in surrounding communities by operations of Air Force aircraft and ground support equipment. The far-field, community-type noise data in the handbook describe the noise produced during ground operations of aircraft, ground support equipment, and other ground-based equipment or facilities.

Volume 1 of this handbook discusses the objectives and design of the handbook, types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc. Volume 2 provides a method and data for adjusting the handbook's far-field noise data, which are for standard meteorological conditions (15 C temperature, 70% rel humidity, 0.760 meters Hg barometric pressure), to derive comparable data for other meteorological conditions. Refer to Volumes 1 and 2 (references 1 and 2) for such information because it is not repeated in other handbook volumes.

A cumulative index lists those aerospace systems contained in the handbook, and identifies the specific volumes containing each type of environmental noise data available (i.e., inflight/flight crew and passenger noise, near-field/ground crew noise, far-field/community noise). Volume numbers are assigned sequentially as individual volumes are published. This index is periodically updated as individual volumes are published and is available upon request from AMRL/BBE, Wright-Patterson AFB, OH 45433. Organizations on the distribution list for the handbook will automatically receive a copy of each updated index.

Direct any questions concerning the technical data in this report and other handbook volumes to: AMRL/BBE, Wright-Patterson AFB, OH 45433; AUTOVON 78-53675 or 78-53664; Commercial (513) 255-3675 or (513) 255-3664.

1. Cole, John N., *USAF Bioenvironmental Noise Data Handbook, Volume 1: Organization, Content and Application*, AMRL-TR-75-50 (1), Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975
2. Cole, John N., *USAF Bioenvironmental Noise Data Handbook Volume 2: Procedure to Evaluate Effects of Non-standard Meteorological Conditions on Far-Field Noise*, AMRL-TR-75-50 (2), AMRL, WPAFB, OH, 1975

NEAR-FIELD NOISE

MEASUREMENTS

AMRL acquired near-field noise data on the C-141A aircraft during ground runup operations of its turbofan engines. For these tests, the aircraft was located on a concrete parking apron at Wright-Patterson AFB along with other similar aircraft. Table 1 gives the engine power conditions. The ground-crew chief selected power conditions and near-field locations generally used during routine maintenance or engine runup for preflight checks.

At each near-field location a test engineer randomly moved a hand-held microphone in and around each location, probing all areas where a crew member's head would normally be located. He recorded all of the noise samples on magnetic tape. During analysis of each sample, he determined the root-mean square sound pressure using a 4- or 8-second integration time to derive a power-averaged level for each location.

Figure 1 shows the ten numbered near-field locations where ground crews are usually located for maintenance and/or preflight checkout operations. Estimates of noise levels at other locations in the near-field are difficult since the noise source is spatially distributed, i.e., not a point source. The noise levels at near-field locations can vary widely depending upon relative distances from each noise source (intake noise, exhaust noise, panel resonances, internal engine noise through the engine wall, etc.).

Table 1 lists the numeric/alphabetic designators used on the data pages in this report to identify the measurement locations and test conditions. For example, the designator 1/A means ground crew location 1 and test conditions A.

RESULTS

The measured data presented in Table 2 define the sound pressure levels (SPL) produced by the C-141A aircraft at the ten ground crew locations. This table includes the overall, 1/3 octave band and octave band levels. From these data one can calculate the variety of measures given in Table 3 which are widely used to assess the effects of noise on personnel and their performance.

All near-field data are for the meteorological conditions at the time of test but are valid for all typical airbase meteorology because of the short sound propagation distances involved.

TABLE 1
MEASUREMENT LOCATIONS AND TEST CONDITIONS
FOR NEAR-FIELD NOISE MEASUREMENT

C-141A Aircraft, Ground Runup, Wright-Patterson AFB, OH
8 and 14 May 1979
Tail #12777

Ground Crew Location

1	MD-3 Operator
2	Engine #4 Start
3	Engine #3 Start
4	Electric Disconnect
5	Engine #2 Start
6	Engine #1 Start
7	Right Wheel Well
8	Engine Trim
9	Telephone Talker
10	Left Wheel Well (Near APU)

Aircraft Engine and Ground Support Equipment Operation

A	MD-3
B	MD-3 and APU
C	MD-3 and Engine #4 Idle
D	MD-3 and Engine #3 and #4 Idle
E	Engines #2, #3, and #4 Idle
F	All Engines Idle
G	Engine #3 Maximum Power Other Engines Idle

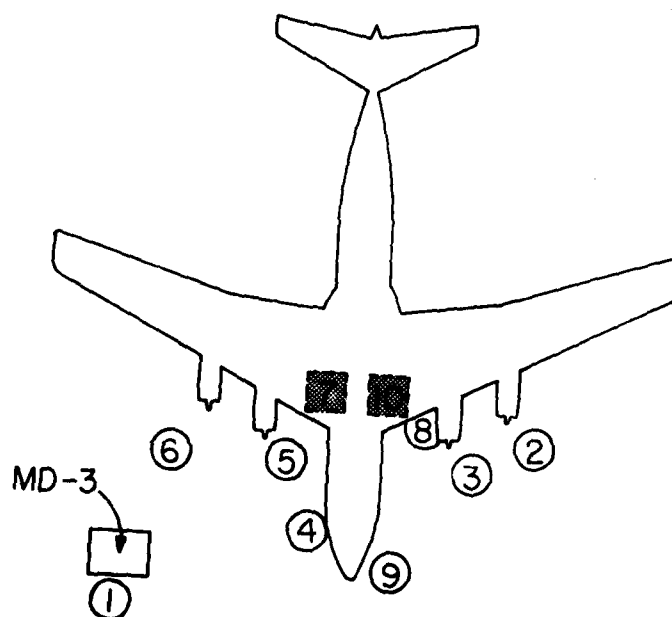
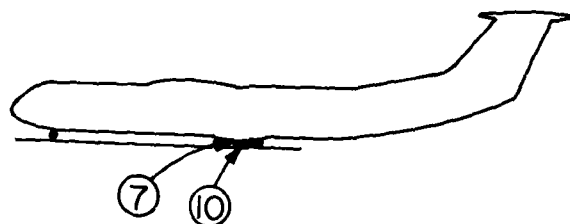


Figure 1. Near-Field Measurement Locations at Wright-Patterson AFB OH

FAR-FIELD NOISE

MEASUREMENTS

AMRL acquired far-field data during a one hour test period, thus keeping similar meteorological conditions throughout the test. Figure 2 shows the ground runup pad, ground cover, aircraft orientation and the 19 microphone measurement sites on a semicircle. The center of the 75 meter radius semicircle used in surveying the TF33-P-7 engines was on the ground directly below the intersection of the aircraft's centerline and the plane passing through the inboard engines' exhaust-nozzle exits. The ground runup area did not have a blast deflector; therefore, the engines' exhausts were in a "free-flow" condition.

Table 4 provides cockpit readouts of some engine characteristics (% RPM, fuel flow, etc.) for each power setting used in the far-field tests. Also listed in this table are the surface meteorological conditions during data acquisition.

All microphone measurement sites are in the acoustic far-field of the source where the sound wavefronts spherically diverge and the noise source may be regarded as a point source.

A portable microphone/tape-recorder system was used to sequentially record the noise at each far-field location. The microphone was attached to a hand-held pole, pointed at the source (0° angle of incidence) and vertically scanned from 0.5 to 3 meters for a period of 5-10 seconds during data acquisition at each microphone location. These samples were then time-integrated to derive a root-mean-square sound pressure level. Vertical scanning and time-integrating together reduce anomalies frequently present in data acquired by a fixed height microphone.

RESULTS

Table 5 lists the overall and 1/3 octave band SPL measured at the far-field locations under meteorological conditions at the time of the test. Data in all other figures and tables are based on these levels. These data were normalized to 100 meters distance and standard meteorological conditions (15C temperature, 70% relative humidity, 0.760 meter Hg barometric pressure) and used to derive the graphic data in Figure 3 which provides a compact summary of the far-field noise characteristics of the C-141A aircraft in a standard format.

Figure 4 and Table 6 present two basic acoustic measures, the acoustic power level and the directivity index, respectively. The acoustic power level describes the power radiated by the source as a function of frequency. The directivity index is a standard acoustical engineering measure which describes the geometric way in which the source radiates this power as a function of both frequency and angle from source. These basic source measures are primarily of interest for acoustical engineers and noise generation/control specialists.

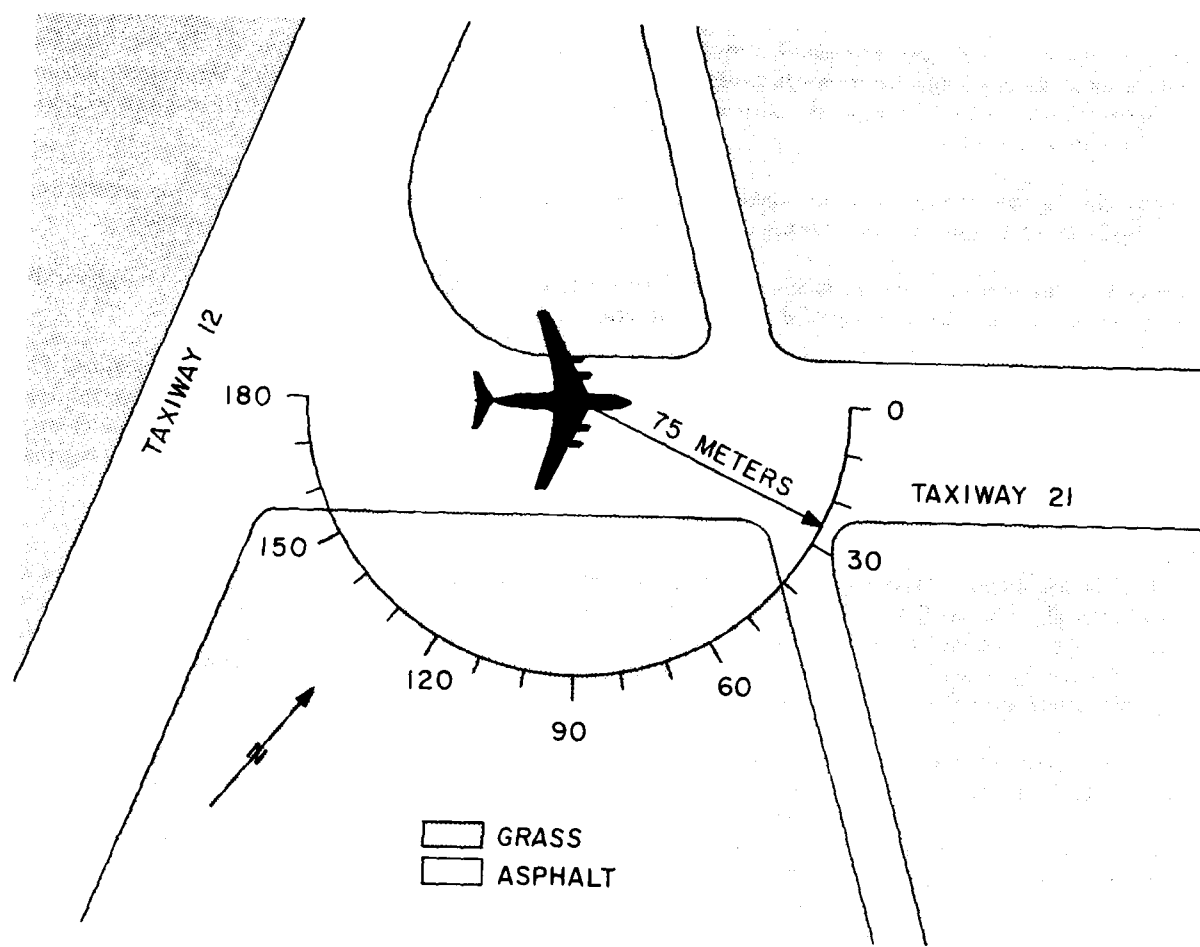


Figure 2. Far-Field Measurement Locations at Wright-Patterson AFB OH

Estimates of noise characteristics for intermediate power settings (e.g., 88% engine) and/or different number of engines operating (e.g., single engine) can be determined as explained in Volume 1 of this handbook.

Figures 5 through 11 are sets of equal noise contours describing seven different measures of noise as a function of angle and distance from the source for standard day meteorology. They are respectively, overall sound pressure level, C-weighted sound level, A-weighted sound level, perceived noise level, speech interference level, permissible exposure times for personnel and octave band sound pressure levels.

No data are presented at the 170 and 180 degree locations at idle power, 150 through 180 degree locations at 70% runup power nor 160 through 180 degree locations for military power because of turbulent air flow behind the aircraft. A-weighted levels at the 170 and 180 degree locations for idle power are 0 to 5 dBA lower than the 160 degree data. Typical A-weighted levels at the missing angles for the two higher power settings are 5 to 10 dBA below the level of the last measured angle.

Test personnel performed noise surveys during quiet periods when the background noise was minimal, e.g., early in the morning when no other aircraft or engine test stands were operating. Data eliminated because they were near the background electronic noise were generally not significant because the levels were so low (e.g., Table 5 at idle power).

Volume 2 of the handbook describes the influence of meteorology on far-field noise environments, and provides, if required, the factors necessary to adjust the handbook's standard meteorological day data.

TABLE: MEASURED SOUND PRESSURE LEVEL (DB)									
1/3 OCTAVE BAND									
2									
NOISE SOURCE/SUBJECT: (OPERATION:)									
C-141A AIRCRAFT									
GROUND CREW									
NEAR FIELD NOISE LEVELS									
IDENTIFICATION:)									
OMEGA 3.2									
TEST AP-027-001									
RUN 01									
23 MAY 79									
PAGE F1									
LOCATION/CONDITION									
FREQ (HZ)	1/A	2/C	3/D	4/D	5/E	6/F	7/B	8/F	9/F 10/B
25	78<	80	82	76<	77<	62	83	95	80
31.5	81	82	83	83	82	87	82	92	84
40	88	83	87	91	86	86	85	95	86
50	93	86	90	99	88	87	89	95	85
63	96	88	92	101	89	91	89	96	88
80	91	87	91	93	92	90	88	96	80<
100	101	91	92	101	88	88	89	97	79
125	106	95	95	105	88	87	91	99	86
160	96	90	94	95	90	65	93	97	85
200	97	92	95	94	93	87	94	97	83
250	105	96	99	100	95	94	99	101	86
315	90	96	99	97	95	97	94	102	90
400	89	99	103	94	101	100	95	105	83
500	89	99	103	91	101	99	92	104	87
630	90	99	103	93	101	100	93	106	83
800	91	101	105	91	103	101	93	106	87
1000	94	103	106	91	105	102	92	105	87
1250	88	100	104	89	101	99	98	105	76
1500	98	102	106	91	104	102	91	103	80
2000	91	102	107	93	104	103	93	104	82
2500	85	100	103	87	101	99	90	105	77
3150	84	99	105	88	102	101	96	106	77
4000	84	99	103	87	101	99	93	105	79
5000	81	98	102	85	99	98	91	105	78
6300	79	96	99	83	97	95	90	102	76
8000	74	95	99	81	96	94	89	102	74
10000	71	95	104	87	92	91	95	109	81
OVERALL	110	112	116	110	113	112	107	117	98

< LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

(	2	OCTAVE BAND											)	
(	-----													)
(	NOISE SOURCE/SUBJECT: (OPERATION:)													)
(	C-141A AIRCRAFT ()													)
(	GROUND CREW ()													)
(	23 MAY 79 ()													)
(	NEAR FIELD NOISE LEVELS ()													)
(	-----													)
(	LOCATION/CONDITION													)
(	1/A	2/C	3/O	4/D	5/E	6/F	7/B	8/F	8/G	9/F	10/B	)		
(	FREQ													)
(	(HZ)													)
(	31.5	89	86	89	91	87	90	88	99	117	88	89	)	
(	63	98	92	96	103	95	94	93	101	125	93	87	)	
(	125	107	97	98	107	94	91	96	102	131	95	90	)	
(	250	104	100	103	102	99	99	101	105	126	94	92	)	
(	500	94	104	108	98	106	104	98	110	125	102	88	)	
(	1000	96	106	110	95	108	106	96	110	126	104	90	)	
(	2000	94	106	110	95	108	106	96	109	121	105	85	)	
(	4000	88	103	108	92	105	104	99	110	126	103	82	)	
(	8000	80	100	106	89	100	98	97	110	125	99	82	)	
(	OVERALL	110	112	116	110	113	112	107	117	136	110	98	)	
(	-----													)

TABLE: MEASURES OF HUMAN NOISE EXPOSURE										IDENTIFICATION:	
3										OMEGA 3.2	
NOISE SOURCE/SUBJECT:										TEST AP-027-001	
C-141A AIRCRAFT										RUN 01	
GROUND CREW										23 MAY 79	
NEAR FIELD NOISE LEVELS										PAGE W1	
LOCATION/CONDITION											
1/A	2/C	3/D	4/O	5/E	6/F	7/B	8/F	8/G	9/F	10/B	
HAZARD/PROTECTION											
G-WEIGHTED OVERALL SOUND LEVEL (OASLC IN DB) AT EAR											
A-WEIGHTED OVERALL SOUND LEVEL (OASLA IN DB) AT EAR											
MAXIMUM PERMISSIBLE TIME (T IN MINUTES) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)											
NO PROTECTION											
OASLC	110	112	115	110	113	112	106	117	135	110	97
OASLA	101	112	116	102	113	112	104	116	132	110	94
T	25	3.8	P	21	3.2	3.8	15	P	P	5	85
MINIMUM QPL EAR MUFFS											
OASLA*	37	86	89	87	87	85	82	92	112	84	73
T	285	339	202	285	285	404	679	120	3.8	480	960
AMERICAN OPTICAL 1700 EAR MUFFS											
OASLA*	83	80	84	83	81	79	78	87	107	78	69
T	571	960	480	571	807	960	960	285	9	960	960
V-51R EAR PLUGS											
OASLA*	78	85	89	78	86	85	78	90	116	83	70
T	960	404	202	960	339	404	960	170	11	571	960
AMERICAN OPTICAL 1700 EAR MUFFS PLUS V-51R EAR PLUGS											
OASLA*	66	72	75	66	73	71	64	77	93	70	56
T	960	960	960	960	960	960	960	960	101	960	960
H-133 GROUND COMMUNICATION UNIT											
OASLA*	77	84	88	77	86	84	77	89	115	83	67
T	960	480	240	960	339	480	960	202	13	571	960
COMMUNICATION											
PREFERRED SPEECH INTERFERENCE LEVEL (PSIL IN DB)											
PSIL	95	105	109	96	107	105	97	109	124	104	88
ANNOYANCE											
PERCEIVED NOISE LEVEL, TONE CORRECTED (PNLT IN PND8)											
TONE CORRECTION (C IN DB)											
PNLT	117	125	129	118	126	125	121	131	148	125	108
C	2	1	0	1	0	0	2	1	1	1	1

* BASED ON CALCULATED SPL SPECTRUM UNDER PROTECTIVE DEVICE.
P ADDITIONAL EAR PROTECTION REQUIRED.

TABLE 4
TEST CONDITIONS
FOR FAR-FIELD NOISE MEASUREMENTS

C-141A Aircraft, Ground Runups, Wright-Patterson AFB, OH
14 August 1974
Tail #612775

Aircraft Engine Operation

Idle

All Engines

55 % RPM NC (Core Speed)
28 % RPM NF (Fan Speed)
310 C EGT (Exhaust Gas Temperature)
1.04 EPR (Engine Pressure Ratio)
1100 LBS/HR FF (Fuel Flow)

70% RPM

All Engines

87 % RPM NC
70 % RPM NF
345 C EGT
1.27 EPR
4100 LBS/HR FF

Military Power

All Engines

98 % RPM NC
95 % RPM NF
500 C EGT
1.85 EPR
10,000 LBS/HR FF

Meteorology

Temperature

25.6 C

Bar Pressure

0.743 M Hg

Rel Humidity

60 %

Wind — Speed

2.1 M/Sec (4 Kts)

— Direction

100 Deg

TABLE: MEASURED SOUND PRESSURE LEVEL (DB)																
1/3 OCTAVE BAND																
DISTANCE = 75 METERS																
NOISE SOURCE/SUBJECT:																
C-141A AIRCRAFT																
TF33-P-7 ENGINE																
FAR FIELD NOISE																
FREQ (HZ)																
ANGLE (DEGREES)																
0 10 20 30 40 50 60 70 80 90 100 110 120 130 140 150 160 170 180																
25	66<	65<	64<	63<	62<	61<	60<	59<	58<	57<	56<	55<	54<	53<	52<	51<
31.5	65<	64<	63<	62<	61<	60<	59<	58<	57<	56<	55<	54<	53<	52<	51<	50<
40	64<	63<	62<	61<	60<	59<	58<	57<	56<	55<	54<	53<	52<	51<	50<	49<
50	63<	62<	61<	60<	59<	58<	57<	56<	55<	54<	53<	52<	51<	50<	49<	48<
63	62<	61<	60<	59<	58<	57<	56<	55<	54<	53<	52<	51<	50<	49<	48<	47<
80	61<	60<	59<	58<	57<	56<	55<	54<	53<	52<	51<	50<	49<	48<	47<	46<
100	60<	59<	58<	57<	56<	55<	54<	53<	52<	51<	50<	49<	48<	47<	46<	45<
125	59<	58<	57<	56<	55<	54<	53<	52<	51<	50<	49<	48<	47<	46<	45<	44<
160	58<	57<	56<	55<	54<	53<	52<	51<	50<	49<	48<	47<	46<	45<	44<	43<
200	57<	56<	55<	54<	53<	52<	51<	50<	49<	48<	47<	46<	45<	44<	43<	42<
250	56<	55<	54<	53<	52<	51<	50<	49<	48<	47<	46<	45<	44<	43<	42<	41<
315	55<	54<	53<	52<	51<	50<	49<	48<	47<	46<	45<	44<	43<	42<	41<	40<
400	54<	53<	52<	51<	50<	49<	48<	47<	46<	45<	44<	43<	42<	41<	40<	39<
500	53<	52<	51<	50<	49<	48<	47<	46<	45<	44<	43<	42<	41<	40<	39<	38<
630	52<	51<	50<	49<	48<	47<	46<	45<	44<	43<	42<	41<	40<	39<	38<	37<
800	51<	50<	49<	48<	47<	46<	45<	44<	43<	42<	41<	40<	39<	38<	37<	36<
1000	50<	49<	48<	47<	46<	45<	44<	43<	42<	41<	40<	39<	38<	37<	36<	35<
1250	49<	48<	47<	46<	45<	44<	43<	42<	41<	40<	39<	38<	37<	36<	35<	34<
1600	48<	47<	46<	45<	44<	43<	42<	41<	40<	39<	38<	37<	36<	35<	34<	33<
2000	47<	46<	45<	44<	43<	42<	41<	40<	39<	38<	37<	36<	35<	34<	33<	32<
2500	46<	45<	44<	43<	42<	41<	40<	39<	38<	37<	36<	35<	34<	33<	32<	31<
3150	45<	44<	43<	42<	41<	40<	39<	38<	37<	36<	35<	34<	33<	32<	31<	30<
4000	44<	43<	42<	41<	40<	39<	38<	37<	36<	35<	34<	33<	32<	31<	30<	29<
5000	43<	42<	41<	40<	39<	38<	37<	36<	35<	34<	33<	32<	31<	30<	29<	28<
6300	42<	41<	40<	39<	38<	37<	36<	35<	34<	33<	32<	31<	30<	29<	28<	27<
8000	41<	40<	39<	38<	37<	36<	35<	34<	33<	32<	31<	30<	29<	28<	27<	26<
10000	40<	39<	38<	37<	36<	35<	34<	33<	32<	31<	30<	29<	28<	27<	26<	25<
OVERALL	100	100	99	96	93	93	93	91	89	89	87	89	90	90	89	86

< LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

TABLE: MEASURED SOUND PRESSURE LEVEL (DB)											IDENTIFICATION:									
1/3 OCTAVE BAND											OMEGA 1.4									
DISTANCE = 75 METERS											TEST 75-002-025									
NOISE SOURCE/SUBJECT:											RUN 02									
(OPERATION:											METEOROLOGY:									
(C-141A AIRCRAFT											TEMP = 26 C									
(TF33-P-7 ENGINE											BAR PRESS = .743 M HG									
(FAR FIELD NOISE											REL HUMID = 60 %									
(FREE FLOW											PAGE 2									
FREQ (HZ)											ANGLE (DEGREES)									
	0	10	20	30	40	50	60	70	80	90	100	110	120	130	140	150	160	170	180	
25	60	81	78	80	79	82	85	84	83	83										
31.5	78	80	79	81	81	82	84	84	84	85										
40	82	83	83	82	83	85	84	86	86	88										
50	82	82	83	82	82	83	85	86	87	88	91	92	94	98	102					
63	85	85	84	83	86	86	87	87	88	90	90	90	93	95	99	103				
80	86	86	85	86	87	88	88	88	90	91	91	91	93	96	99	102				
100	87	86	86	86	87	87	88	89	90	91	92	94	95	98	100					
125	87	87	86	87	86	86	87	87	89	90	92	93	94	96	97					
160	88	88	87	86	85	86	87	87	88	90	92	93	94	96	97					
200	87	87	86	86	85	85	85	85	88	89	90	91	92	94	92					
250	86	87	89	86	85	85	85	85	87	88	88	89	91	94	92					
315	87	86	86	85	82	81	82	81	82	83	85	86	88	91	88					
400	86	86	87	85	82	82	82	82	81	82	81	83	87	87	85					
500	86	87	87	85	82	81	82	82	80	82	80	82	84	82	81					
630	89	89	88	85	82	82	82	83	82	84	82	84	84	81	80					
800	90	90	89	87	84	83	83	83	82	84	82	85	83	79	79					
1000	93	94	95	95	93	92	92	89	88	86	83	84	84	79	79					
1250	104	104	103	100	97	96	95	93	91	89	85	86	86	80	80					
1600	99	98	99	99	97	96	95	94	90	89	87	87	87	81	81					
2000	99	98	97	97	93	94	93	93	91	89	90	89	89	83	81					
2500	103	102	108	114	104	100	100	99	99	97	99	100	102	92	90					
3150	100	100	102	106	100	98	97	97	95	93	96	97	98	92	91					
4000	101	101	100	100	98	97	97	97	95	93	97	96	95	89	87					
5000	102	102	102	106	100	100	101	99	99	96	98	97	98	90	89					
6300	99	99	99	101	98	97	97	97	95	93	95	94	94	87	86					
8000	96	97	97	101	96	96	96	96	94	93	97	96	96	89	87					
10000	93	93	94	97	93	91	91	92	89	88	93	92	92	85	83					
OVERALL	111	111	112	116	109	107	108	107	106	105	107	107	108	107	110					

LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

TABLE: MEASURED SOUND PRESSURE LEVEL (DB)																	IDENTIFICATIONS		
1/3 OCTAVE BAND																			
DISTANCE = 75 METERS																			
NOISE SOURCE/SUBJECT:																			
(OPERATION:																			
(MILITARY POWER																			
(982 RPM, 1.85 EPR																			
(ALL ENGINES																			
(FREE FLOW																			
METEOROLOGY:																			
TEMP = 26 C																			
BAR PRESS = .743 M HG																			
REL HUMID = 60 %																			
																	OMEGA 1.4		
																	TEST 75-002-025		
																	RUN 03		
																	06 MAY 75		
																	PAGE 2		

LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

FIGURE: NORMALIZED FARFIELD NOISE LEVELS

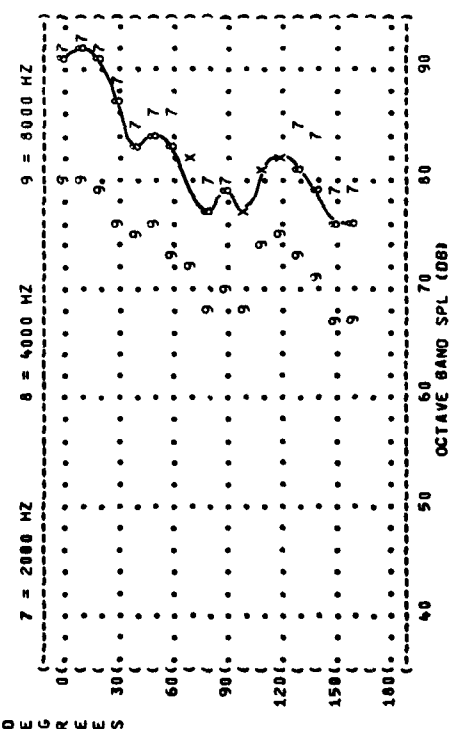
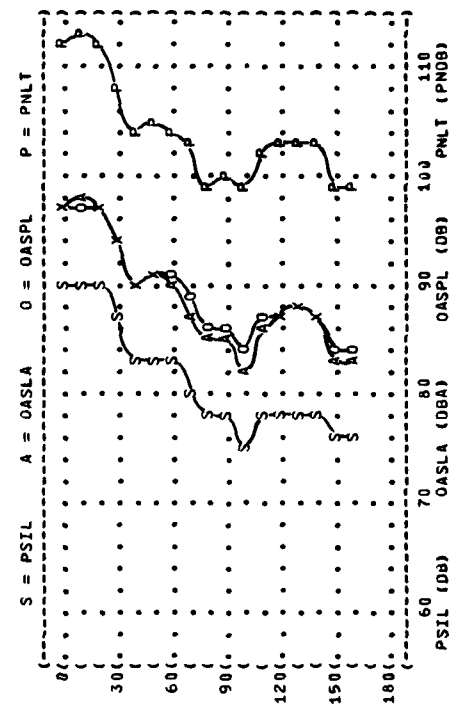
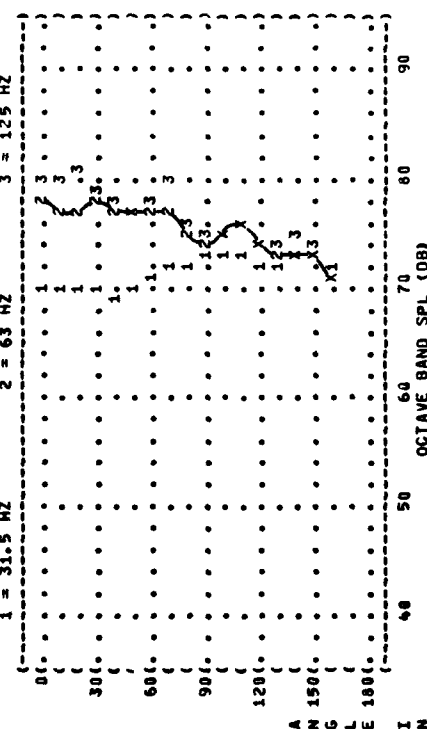
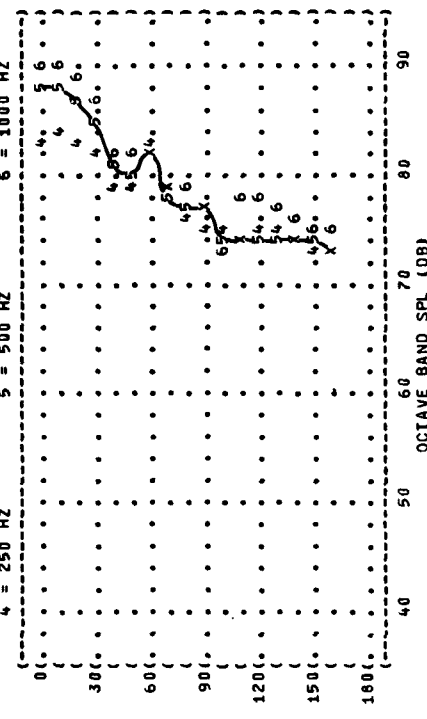
3 DISTANCE = 100 METERS

NOISE SOURCE/SUBJECT: C-1A1A AIRCRAFT
 1F33-P-7 ENGINE
 FAR FIELD NOISE

OPERATION: IDLE
 55% RPM, 1.04 EPR
 ALL ENGINES
 FREE FLOW

METEOROLOGY: TEMP = 15 C
 BAR PRESS = .760 M HG
 REL HUMID = 78 %

IDENTIFICATION: OMEGA 1.4
 TEST 75-002-025
 RUN 01
 06 MAY 75
 PAGE 6



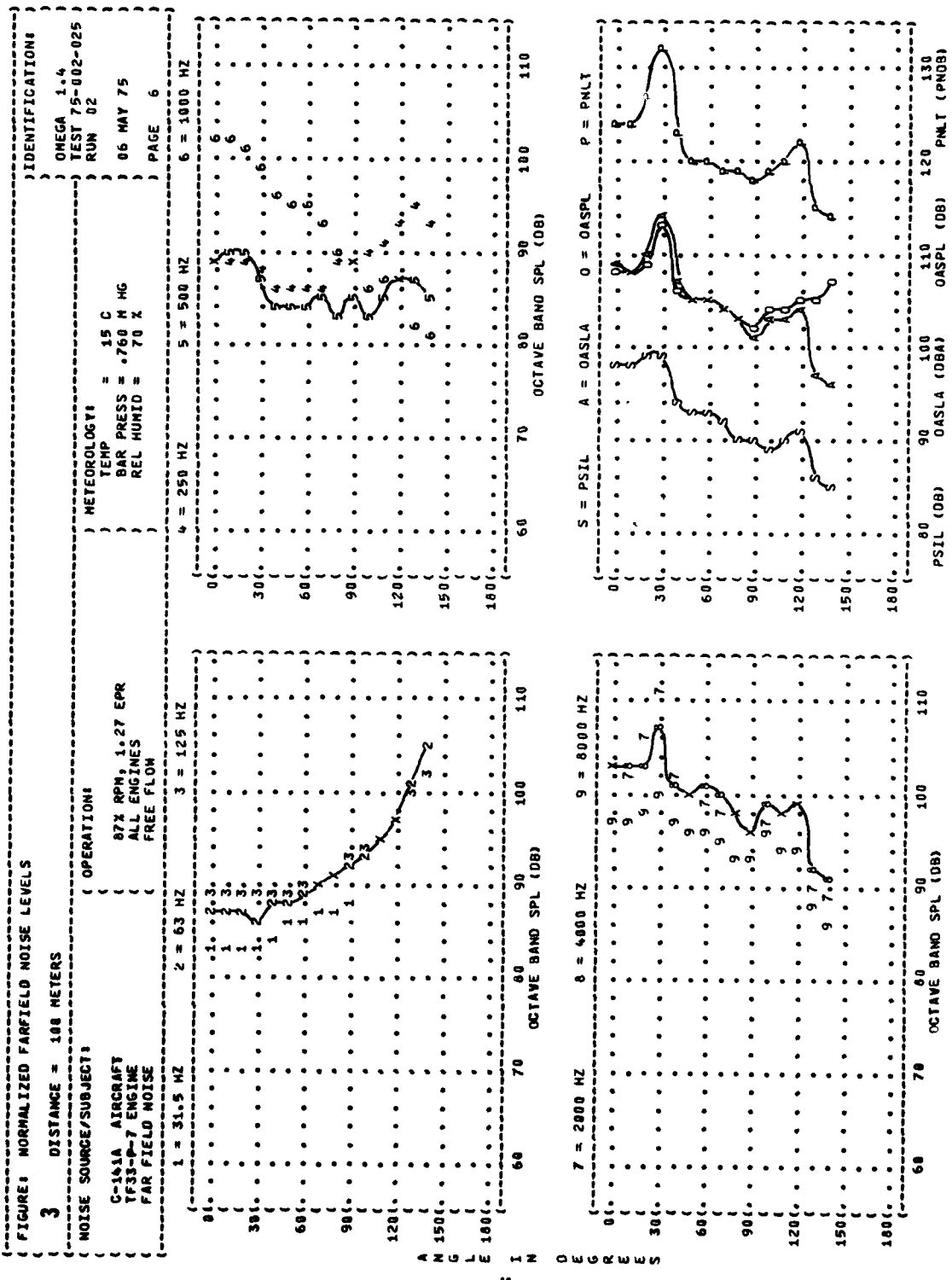


FIGURE 3 NORMALIZED FARFIELD NOISE LEVELS

3 DISTANCE = 100 METERS

NOISE SOURCE/SUBJECT:

C-130A AIRCRAFT

TF33-P-7 ENGINE

FAR FIELD NOISE

OPERATIONS:

MILITARY POWER

98X RPM, 1.85 EPR

ALL ENGINES

FREE FLOW

METEOROLOGICAL:

TEMP = 15 C

BAR PRESS = .760 Hg

REL HUMID = 70 %

IDENTIFICATIONS:

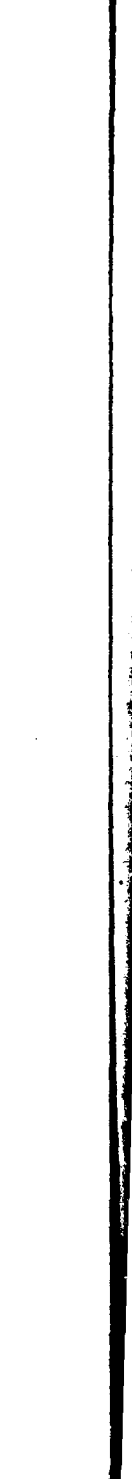
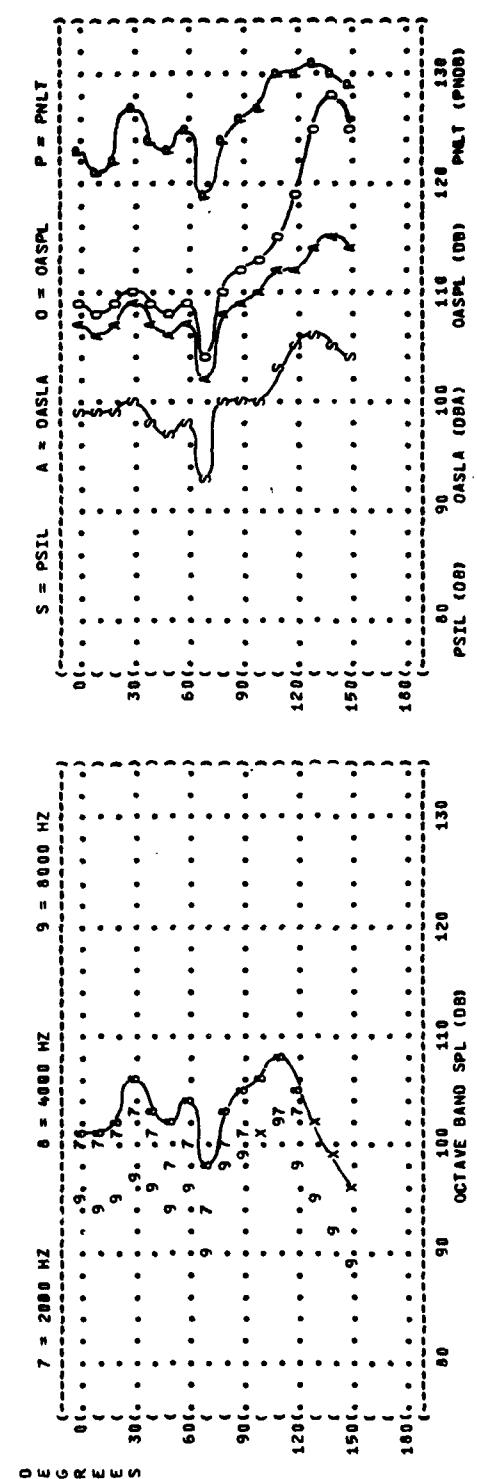
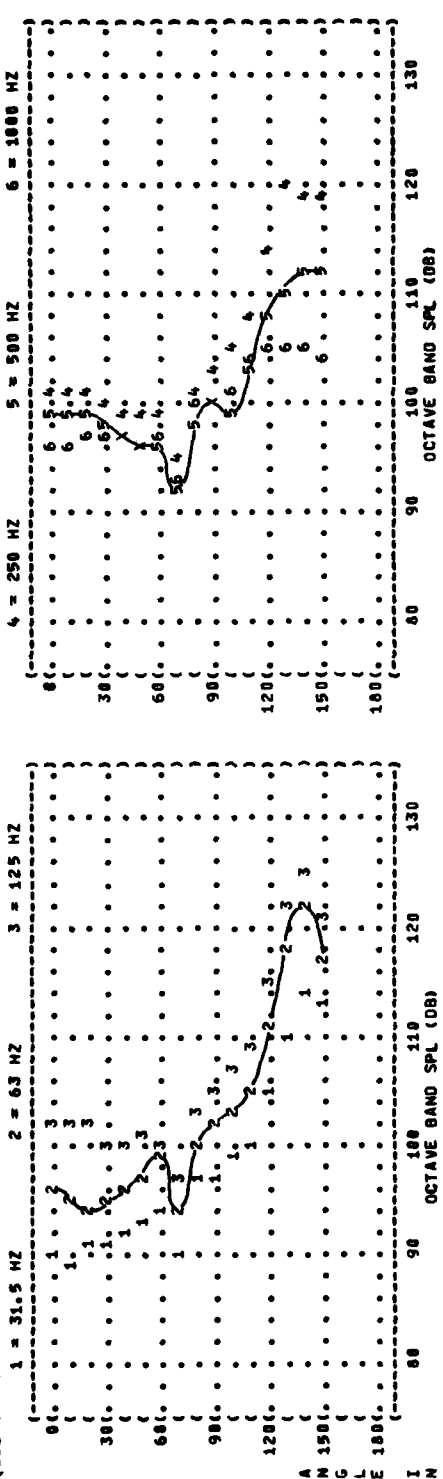
OMEGA 1.4

TEST 75-082-825

RUN 83

86 MAY 75

PAGE 6



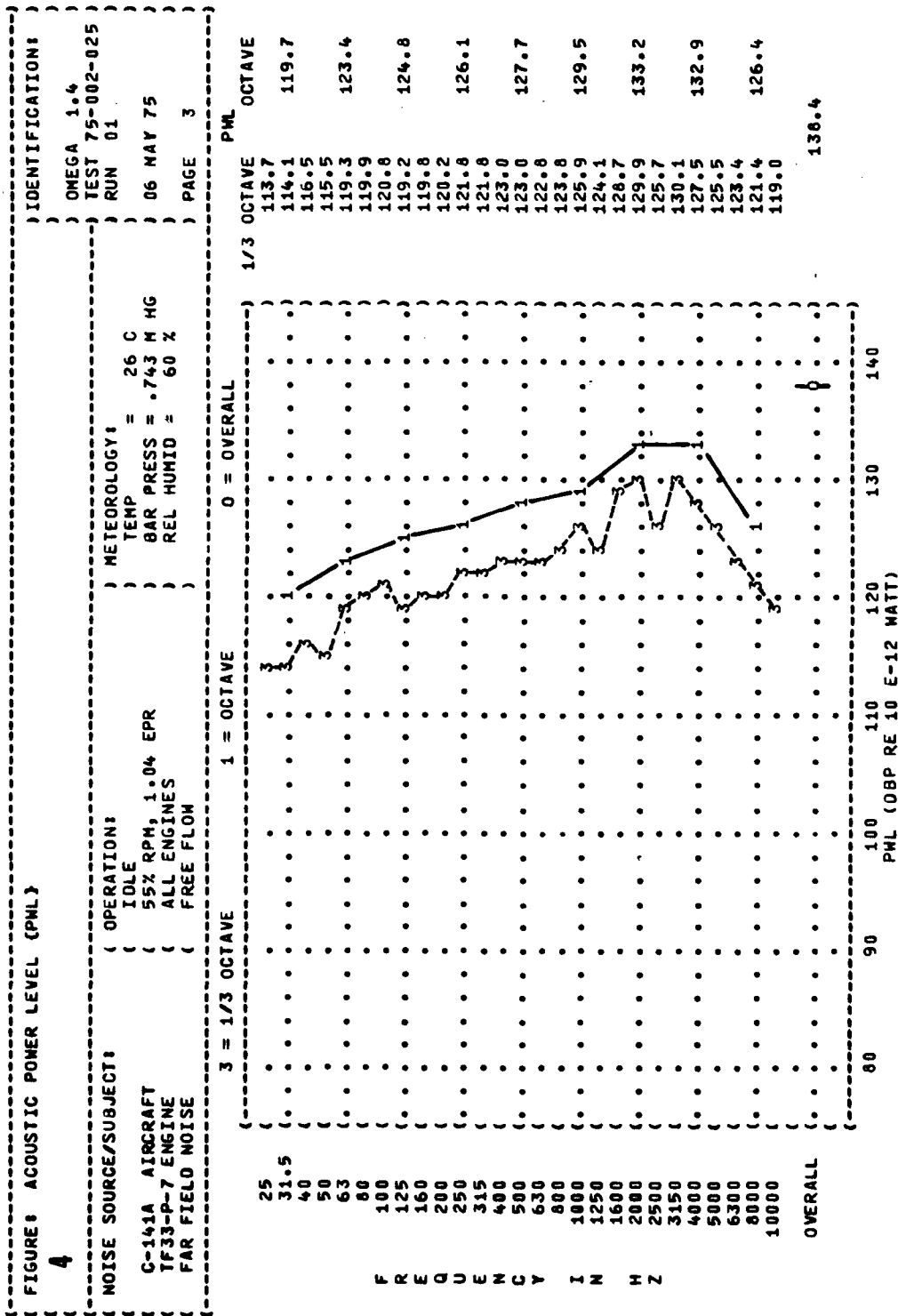


FIGURE: ACOUSTIC POWER LEVEL (PWL)

4

IDENTIFICATION:

OMEGA 1.4

TEST 75-002-025

RUN 03

06 MAY 75

PAGE 3

NOISE SOURCE/SUBJECT:

OPERATION: MILITARY POWER

TEMP = 26 C

98% RPM, 1.85 EPR

BAR PRESS = .743 M HG

ALL ENGINES

REL HUMID = 60 %

FREE FLOW

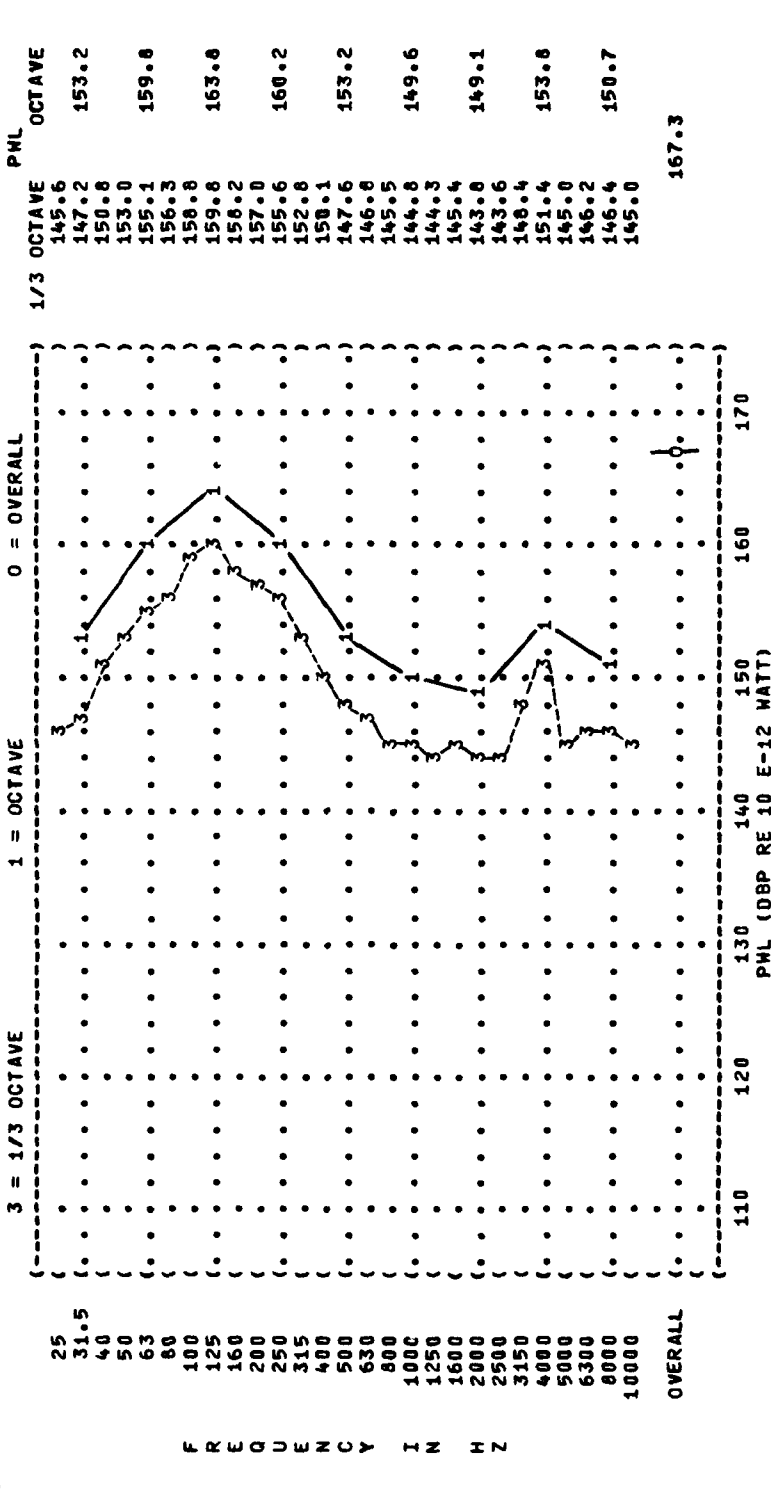


TABLE: DIRECTIVITY INDEX (DB)																
IDENTIFICATION:																
6																
NOISE SOURCE/SUBJECT:																
C-141A AIRCRAFT																
TF33-P-7 ENGINE																
FAR FIELD NOISE																
OPERATION:																
IDLE																
55% RPM, 1.04 EPR																
ALL ENGINES																
FREE FLOW																
METEOROLOGY:																
TEMP = 26 C																
BAR PRESS = .743 M HG																
REL HUMID = 60 %																
PAGE 4																
FREQ (HZ)																
0 10 20 30 40 50 60 70 80 90 100 110 120 130 140 150 160 170 180																
1/3 OCTAVE																
25	-2	-3	-3	-4	-2	-3	-2	1	-1	0	1	1	-1	-2	1	3
31.5	-3	-2	-1	-2	-3	-1	-1	-0	1	2	1	1	-1	-2	1	1
40	-1	-2	-2	-1	-3	-2	-1	-1	-0	1	2	0	0	-1	0	-2
50	1	0	1	0	1	1	-1	0	-1	-1	0	0	0	-1	-4	-6
63	2	1	1	2	3	1	2	1	-1	-1	0	-1	-1	-3	-3	-5
80	4	2	3	3	0	1	2	3	0	-1	-1	-2	-2	-2	-3	-6
100	3	2	4	2	0	0	0	3	-0	-2	-2	-1	-3	-1	-2	-4
125	3	3	3	2	1	1	2	3	-1	-2	-2	-2	-3	-2	-3	-6
160	3	4	4	2	0	0	5	3	-2	-3	-3	-4	-5	-4	-5	-6
200	4	4	4	3	2	0	1	6	-1	-3	-4	-5	-6	-5	-6	-5
250	4	4	4	3	2	0	1	3	-1	-3	-4	-5	-6	-5	-6	-5
315	6	7	6	5	2	1	2	2	-2	-3	-3	-4	-5	-4	-5	-6
400	8	8	8	6	1	1	2	2	-0	-3	-3	-4	-5	-4	-5	-6
500	8	9	8	6	1	1	3	2	-2	-3	-3	-4	-5	-4	-5	-6
630	9	9	8	5	0	-1	2	2	-2	-3	-6	-7	-7	-6	-7	-7
800	10	10	8	5	0	-1	1	3	-2	-4	-6	-7	-7	-6	-7	-7
1000	9	9	8	6	1	1	1	-3	-2	-4	-7	-7	-7	-6	-7	-7
1250	7	7	8	6	2	1	1	-2	-4	-9	-4	-4	-5	-4	-5	-6
1600	9	9	8	5	2	2	2	-3	-5	-5	-5	-4	-4	-3	-4	-5
2000	7	7	6	3	0	2	1	-3	-5	-4	-6	-2	-2	1	-4	-4
2500	9	9	8	5	1	2	0	-2	-5	-4	-8	-3	-2	-1	-5	-7
3150	9	10	9	4	-0	1	-1	-1	-6	-4	-6	-2	-1	-3	-5	-7
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5000	7	7	7	4	1	3	0	-1	-5	-4	-6	-1	-0	-4	-7	-7
6300	6	7	6	3	1	3	0	-1	-5	-3	-5	-0	1	-3	-6	-7
8000	6	6	6	2	1	1	-1	-2	-6	-4	-5	1	3	-1	-5	-6
10000	6	6	5	2	1	1	-2	-2	-5	-4	-4	2	3	1	-2	-5
OCTAVE																
31.5	-2	-2	-2	-2	-3	-2	-1	0	0	1	1	1	0	0	1	0
63	3	3	2	2	2	2	1	2	-0	-1	0	0	-2	-2	-3	-4
125	3	3	4	2	2	0	1	3	-0	-2	-2	-1	-2	-2	-3	-6
250	5	6	5	4	1	1	5	1	-2	-3	-3	-4	-3	-3	-5	-6
500	8	9	8	6	1	1	2	-1	-3	-3	-6	-5	-5	-6	-5	-6
1000	9	9	8	6	1	1	1	-3	-5	-4	-8	-4	-4	-5	-6	-6
2000	8	8	7	4	1	2	1	-3	-5	-4	-7	-3	-2	-1	-6	-6
4000	8	9	8	4	0	2	-0	-1	-6	-4	-6	-2	-1	-2	-7	-7
8000	6	6	6	3	1	2	-1	-2	-5	-4	-5	1	2	0	-6	-6
OVERALL																
6	6	7	4	1	1	1	1	-1	-3	-3	-5	-3	-2	-2	-3	-6

TABLE: DIRECTIVITY INDEX (DB)																
6																
IDENTIFICATION:																
OMEGA 1.4																
TEST 75-002-025																
RUN 02																
NOISE SOURCE/SUBJECT:																
(OPERATION:																
C-141A AIRCRAFT																
(87% RPM, 1.27 EPR																
TF33-P-7 ENGINE																
(ALL ENGINES																
FAR FIELD NOISE																
(FREE FLOW																
METEOROLOGY:																
TEMP = 26 C																
BAR PRESS = .743 M HG																
REL HUMID = 60 %																
PAGE 4																
FREQ																
(HZ)																
ANGLE (DEGREES)																
0 10 20 30 40 50 60 70 80 90 100 110 120 130 140 150 160 170 180																
1/3 OCTAVE																
25	0	1	-2	-0	-1	2	5	4	3	3						
31.5	-2	-1	-1	1	0	2	3	3	4	4						
40	-1	-0	0	-1	0	2	1	3	3	5						
50	-11	-11	-10	-11	-8	-8	-7	-7	-6	-5						
63	-9	-9	-10	-11	-8	-6	-6	-6	-4	-3						
80	-8	-8	-9	-8	-7	-6	-6	-5	-4	-3						
100	-7	-7	-7	-8	-7	-6	-6	-5	-4	-3						
125	-6	-5	-6	-6	-6	-7	-6	-5	-4	-3						
160	-3	-3	-4	-5	-6	-5	-4	-4	-3	-1						
200	-2	-2	-3	-3	-4	-4	-4	-4	-3	-1						
250	-2	-1	0	-2	-3	-4	-4	-4	-3	-1						
315	1	1	1	-0	-3	-4	-4	-4	-3	-1						
400	2	2	4	2	-2	-1	-1	-2	-3	-1						
500	4	5	5	3	-0	-1	-1	-0	-2	0						
630	6	6	5	2	1	-1	-1	-1	-1	1						
800	7	7	6	3	1	-1	-1	-0	-1	1						
1000	4	5	6	6	4	3	3	0	-1	-3						
1250	10	10	9	6	6	3	1	-1	-4	-5						
1600	6	5	6	6	4	3	2	1	-3	-6						
2000	7	6	6	5	2	3	2	1	-2	-2						
2500	0	-1	5	11	1	-3	-3	-4	-4	-6						
3150	2	2	4	8	2	0	-1	-1	-3	-5						
4000	5	5	4	4	2	1	1	1	-1	-3						
5000	3	3	3	7	1	1	2	0	-0	-3						
6300	3	4	3	5	3	2	2	1	-1	-2						
8000	1	1	2	6	1	0	0	0	-2	-2						
10000	1	2	3	6	2	-0	0	0	-2	-3						
OCTAVE																
31.5	-1	-0	-1	-0	0	2	3	3	3	5						
63	-9	-9	-9	-6	-8	-6	-7	-5	-4	-2						
125	-6	-5	-6	-6	-6	-6	-5	-5	-4	-2						
250	-1	-1	-1	-2	-4	-4	-4	-4	-1	-1						
500	4	5	5	2	-1	-1	-1	-0	-2	0						
1000	9	9	8	6	3	2	1	-1	-3	-4						
2000	2	1	5	10	1	-1	-2	-3	-3	-5						
4000	3	3	4	7	2	1	1	0	-1	-4						
8000	2	3	3	6	2	1	1	1	-1	-2						
OVERALL	3	2	4	8	1	-1	-1	-1	-2	-4						

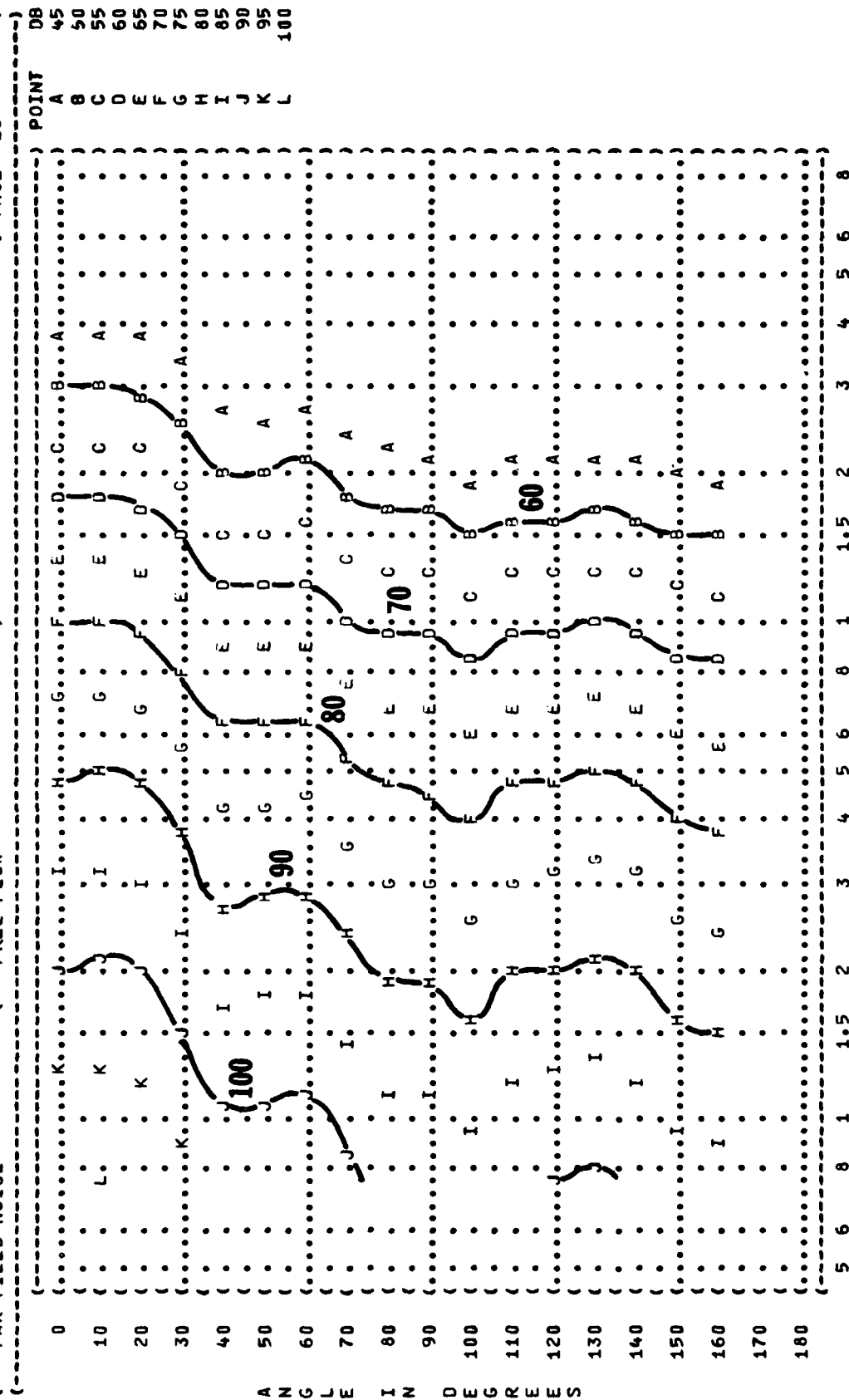
**FIGURE: OVERALL SOUND PRESSURE LEVEL (OASPL)
EQUAL LEVEL CONTOURS (DB)**

5

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FIGURE: OVERALL SOUND PRESSURE LEVEL {OASPL}
          EQUAL LEVEL CONTOURS (DB)
          5
-----
NOISE SOURCE/SUBJECT:      ) OPERATION:
                             ) IDLE
                             ) 55% RPM, 1.04 EPR
G-141A AIRCRAFT            )
TF33-P-7 ENGINE            ) ALL ENGINES
FAR FIELD NOISE            ) FREE FLOW
                             )
                             ) METEOROLOGY:
                             ) TEMP = 15 C
                             ) BAR PRESS = .760 M HG
                             ) REL HUMID = 70 %
                             )
                             ) 06 MAY 75
                             )
                             ) RUN 01
                             ) TEST 75-002-02
IDENTIFICATION:
) OMEGA 1.4
)
)
) PAGE 13

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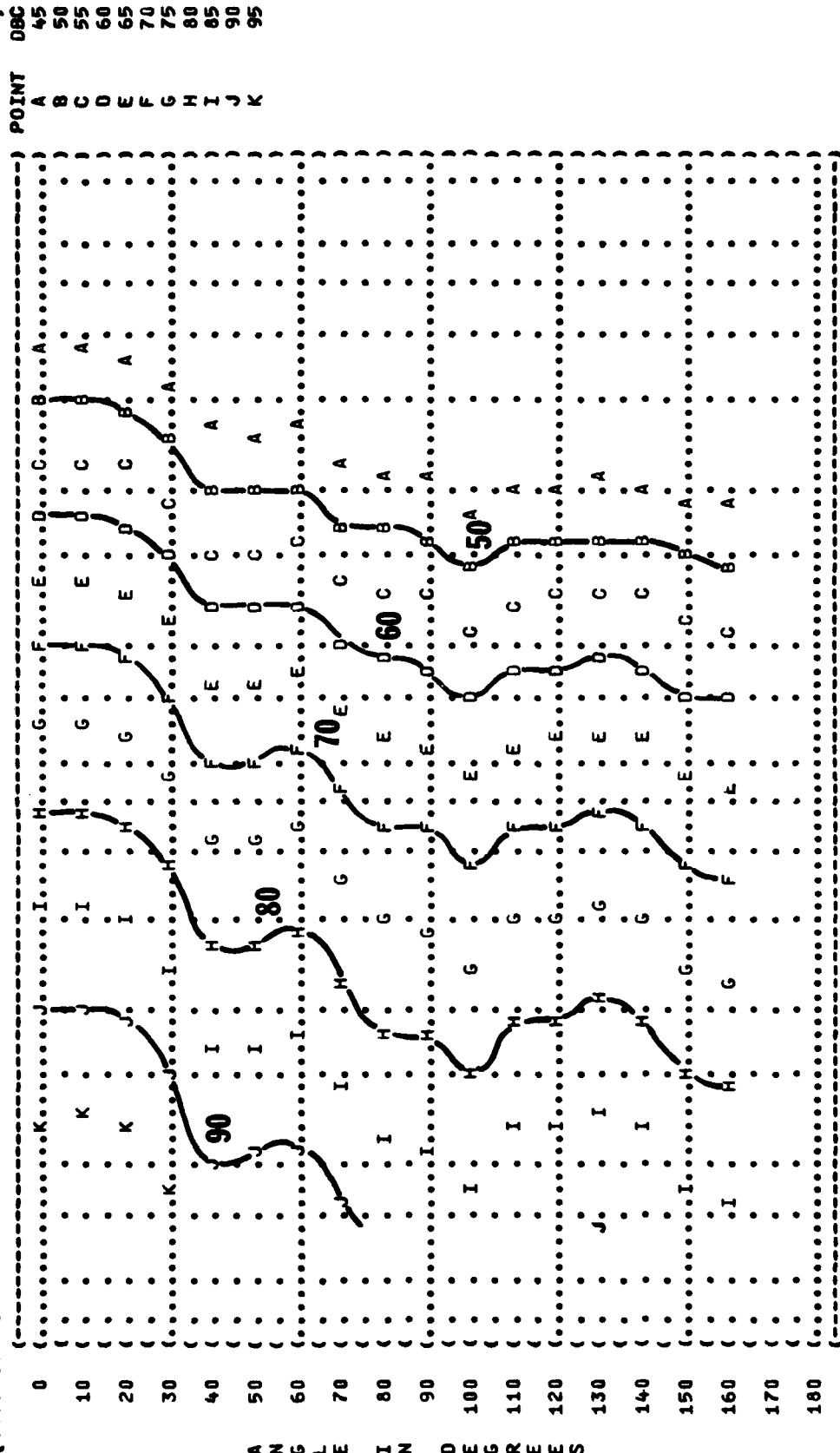


**1000
DISTANCE FROM SOURCE (METERS)**

ANGIE IN DEWEES



FIGURE: C-WEIGHTED OVERALL SOUND LEVEL (OASLC)
 6
 IDENTIFICATION:
 NOISE SOURCE/SUBJECT: (OPERATION:) METEOROLOGY: (TEMP = 15 C)
 ((IDLE))
 (C-141A AIRCRAFT (55% RPM, 1.04 EPR))
 (TF33-P-7 ENGINE (ALL ENGINES))
 (FAR FIELD NOISE (FREE FLOW))
 TEST 75-002-025
 RUN 01
 06 MAY 75
 PAGE 14



DISTANCE FROM SOURCE (METERS)

FIGURE: C-WEIGHTED OVERALL SOUND LEVEL (DASLC) EQUAL LEVEL CONTOURS (OBC)	IDENTIFICATION:
6	OMEGA 1.4
	TEST 75-002-025
	RUN 02
NOISE SOURCE/SUBJECT:	METEOROLOGY:
C-141A AIRCRAFT	TEMP = 15 C
TF33-P-7 ENGINE	BAR PRESS = .760 M HG
FAR FIELD NOISE	REL HUMID = 70 %
	PAGE 14

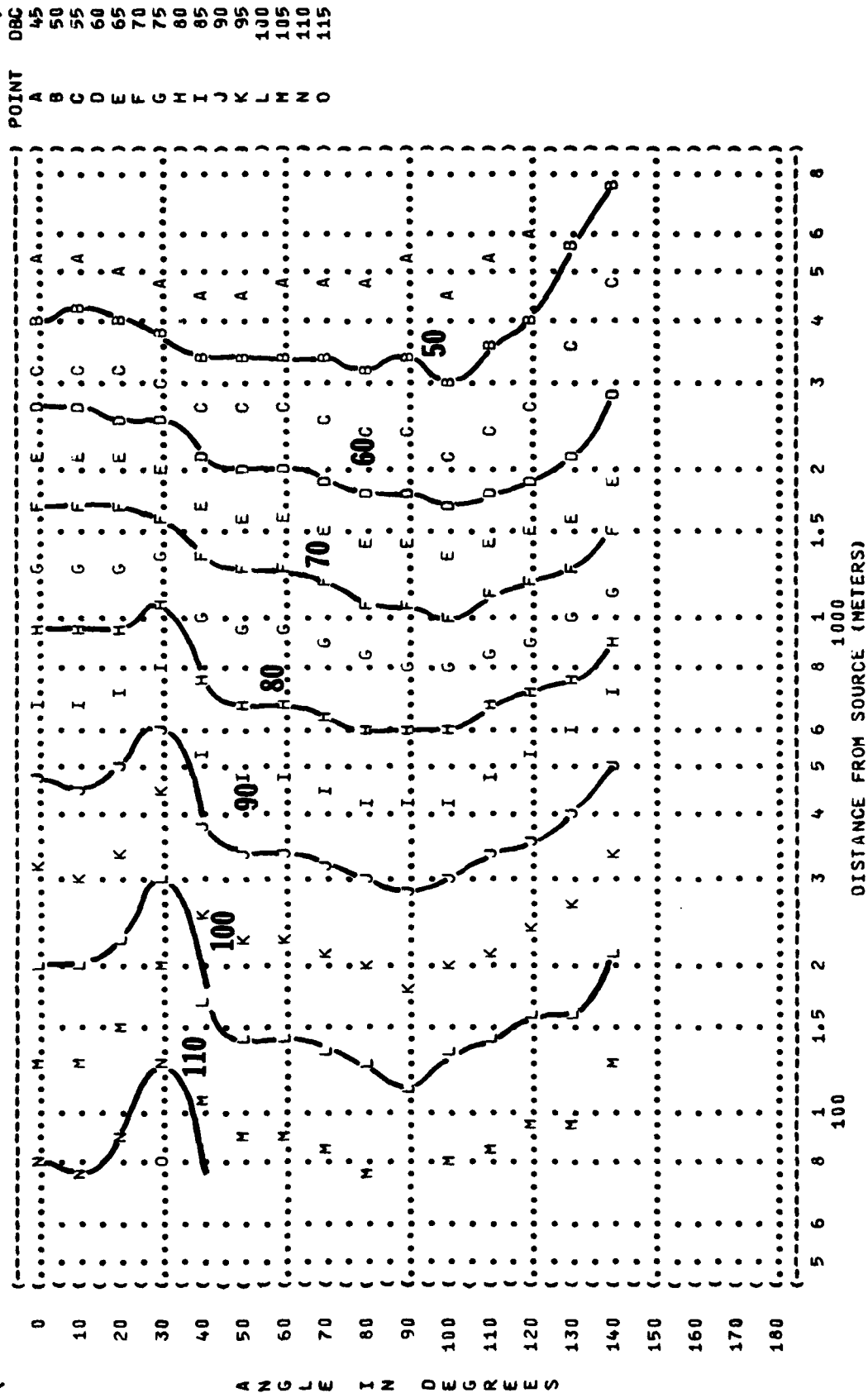
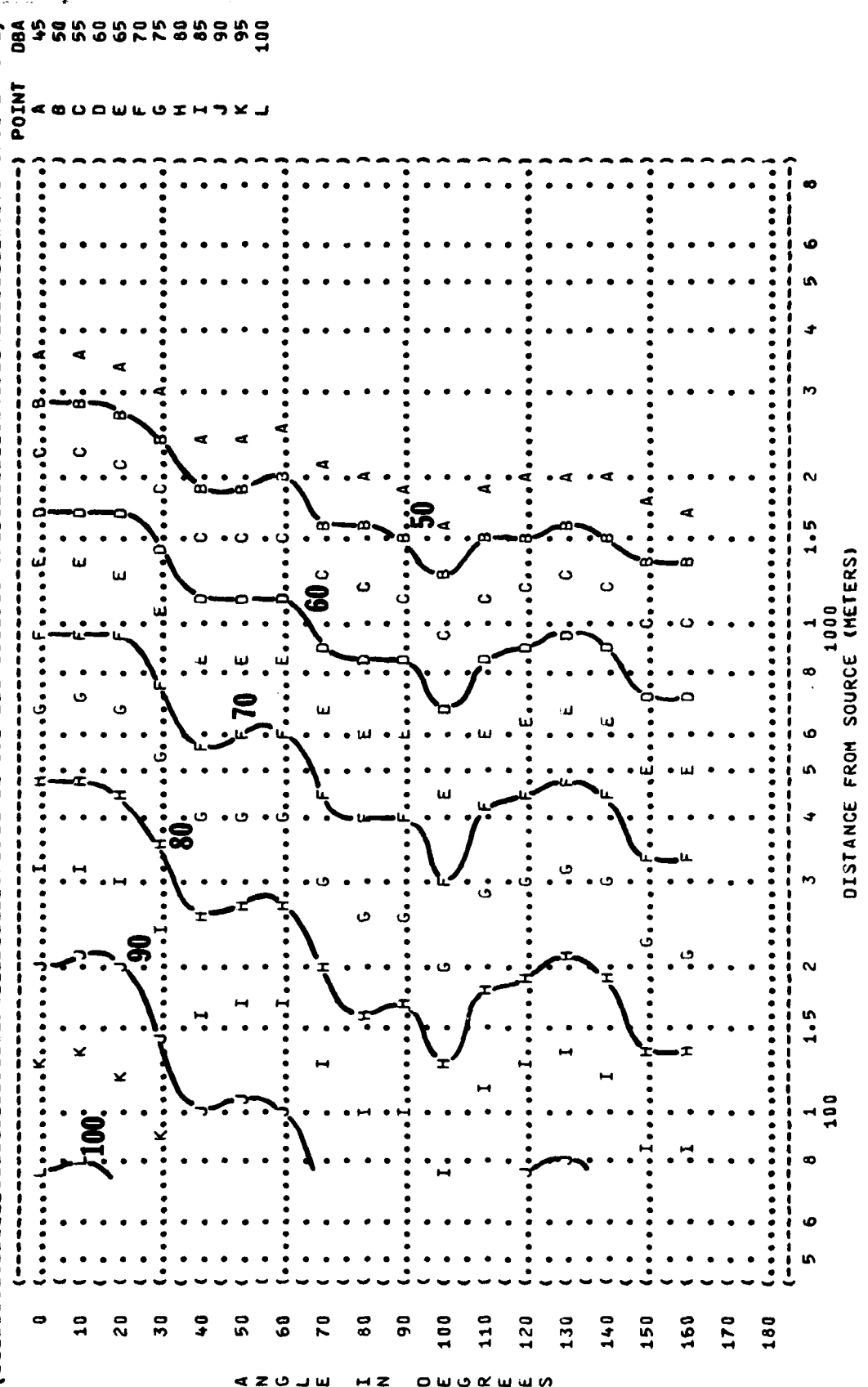


FIGURE: A-WEIGHTED OVERALL SOUND LEVEL (OASLA)
 7
 EQUAL LEVEL CONTOURS (DBA)

IDENTIFICATION: OMEGA 1.4
 TEST 75-002-025
 RUN 01
 METEOROLOGY: TEMP = 15 C
 BAR PRESS = .760 M HG
 REL HUMID = 70 %
 OPERATION: IDLE
 55% RPM, 1.04 EPR
 ALL ENGINES
 FREE FLOW
 NOISE SOURCE/SUBJECT: C-141A AIRCRAFT
 TF33-P-7 ENGINE
 FAR FIELD NOISE
 PAGE 15



) FIGURE: A-WEIGHTED OVERALL SOUND LEVEL (OASLA)
) 7
) EQUAL LEVEL CONTOURS (DBA)
) IDENTIFICATION:
)
) OMEGA 1.4
) TEST 75-002-025
) RUN 02
)
) NOISE SOURCE/SUBJECT: (OPERATION:
)
) C-141A AIRCRAFT (87% RPM, 1.27 EPR
) TF33-P-7 ENGINE (ALL ENGINES
) FAR FIELD NOISE (FREE FLOW
)
) METEOROLOGY:
) TEMP = 15 C
) BAR PRESS = .760 M HG
) REL HUMID = 70 %
)
) PAGE 15

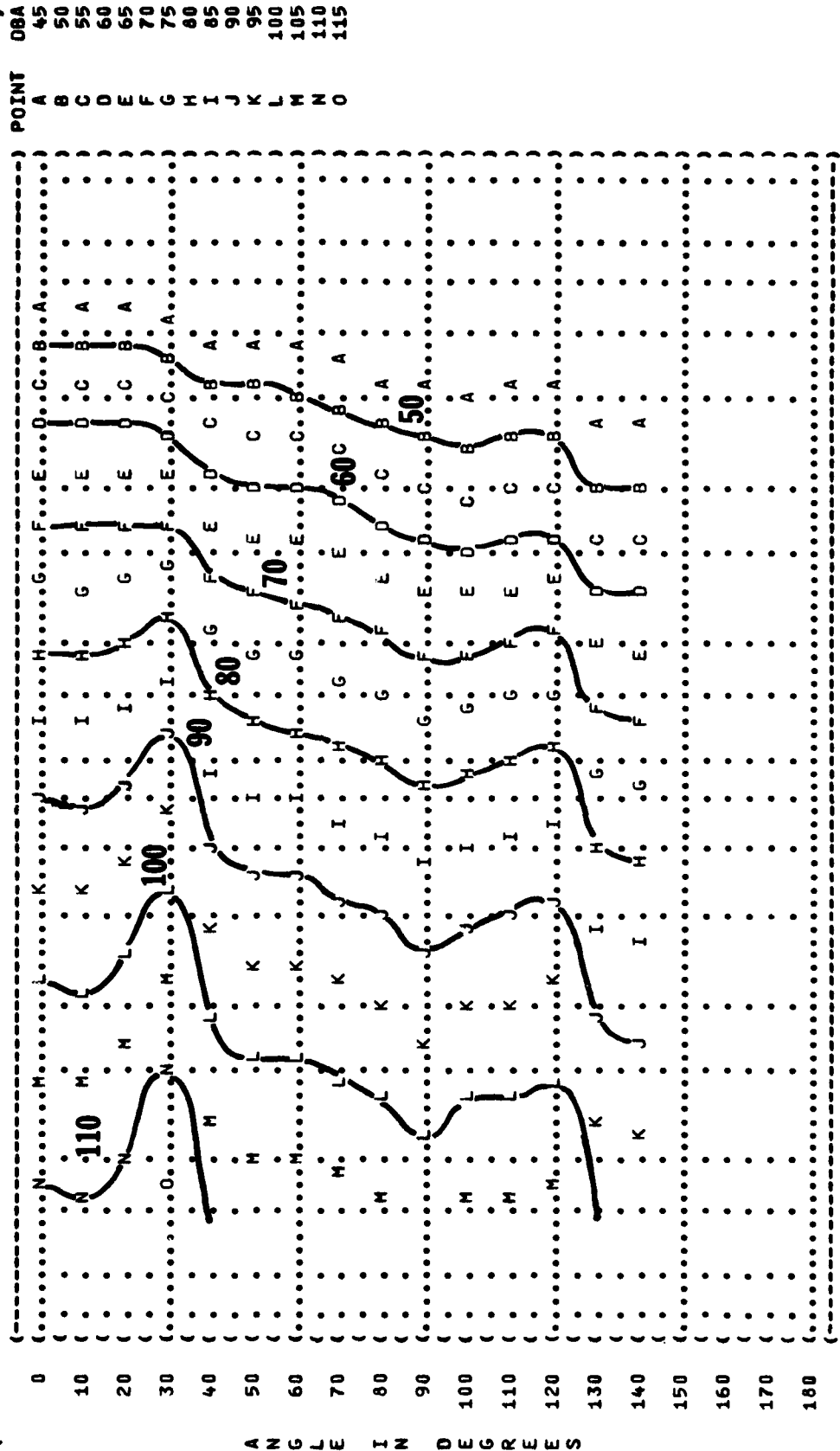
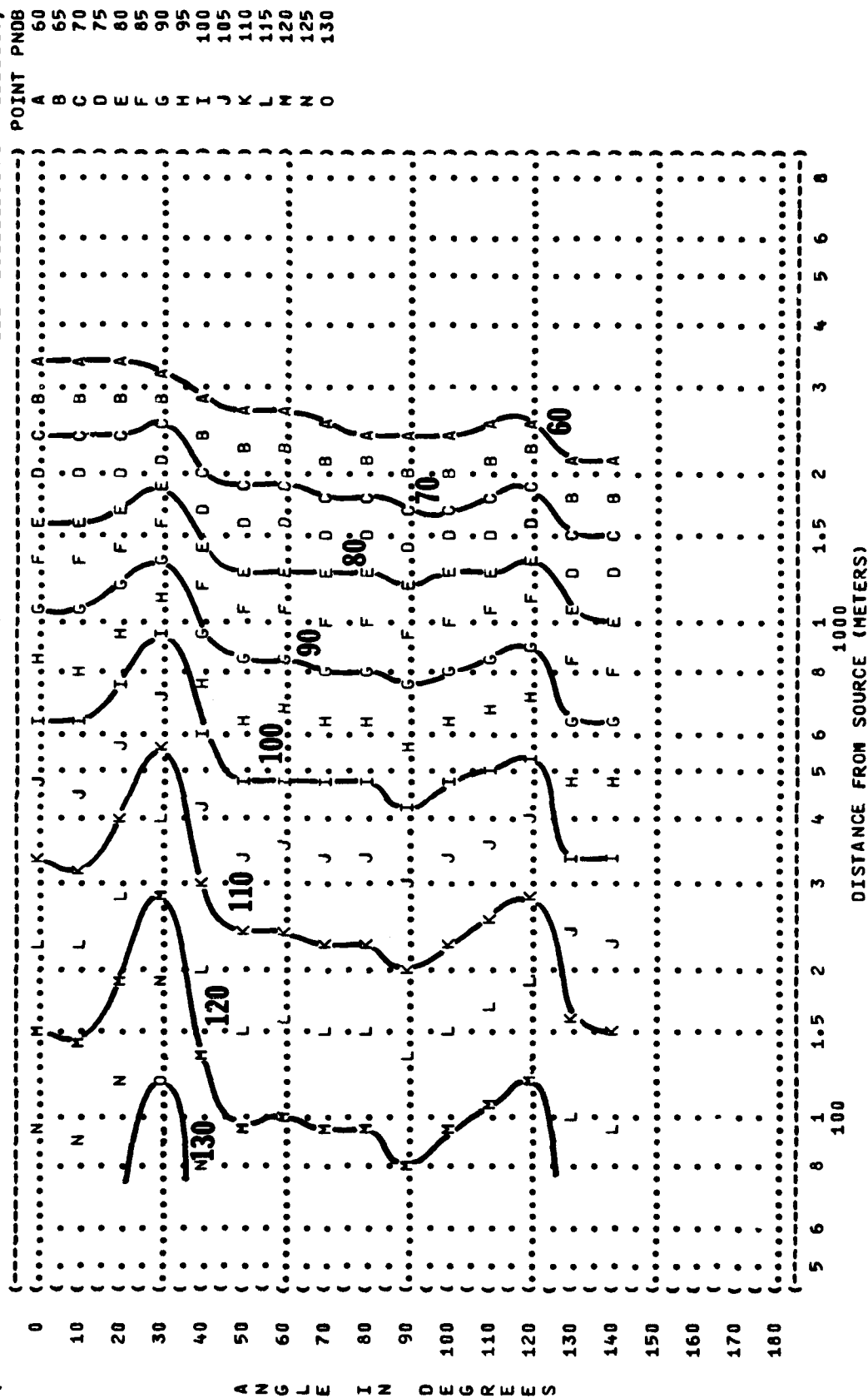


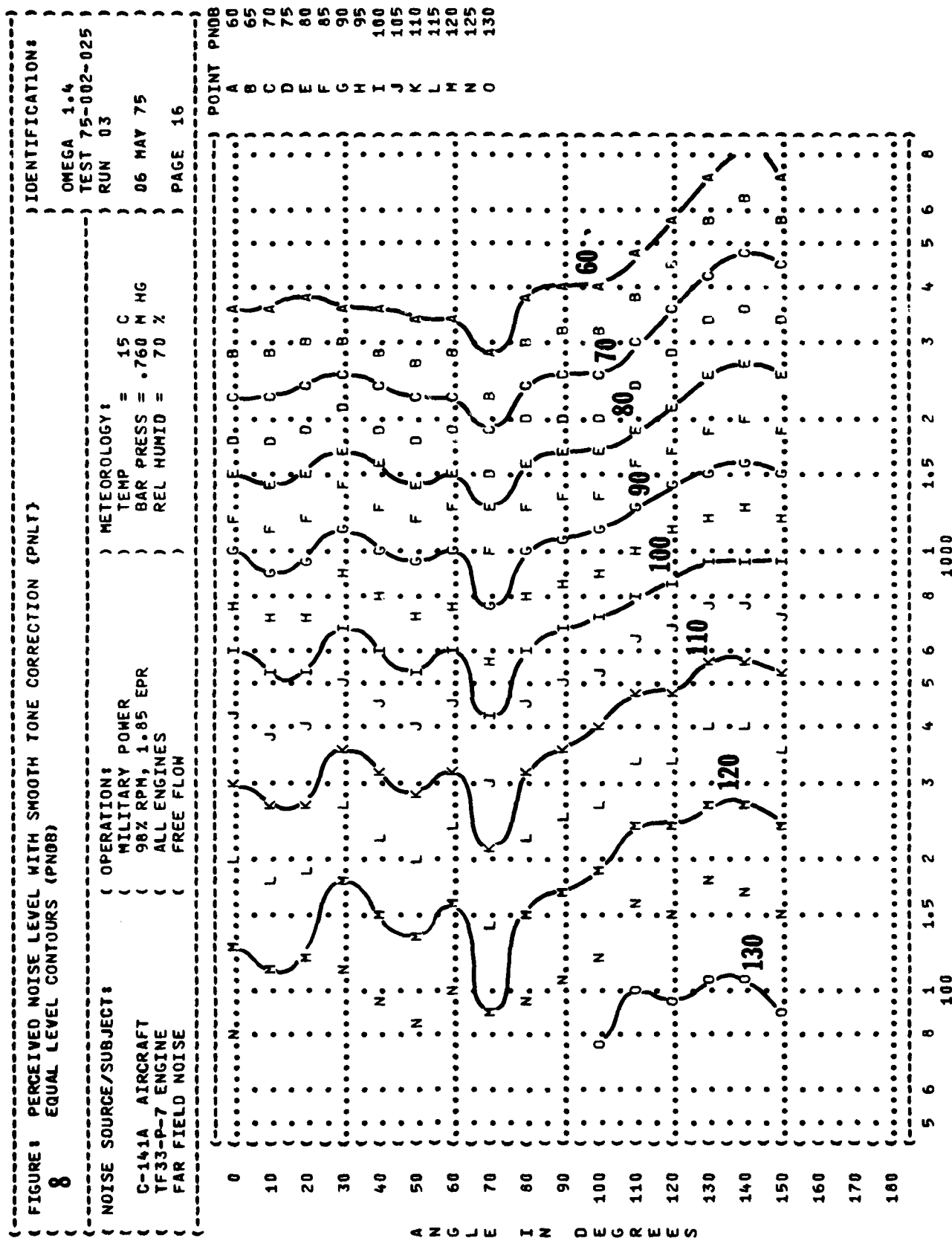
FIGURE 8 PERCEIVED NOISE LEVEL WITH SMOOTH TONE CORRECTION {PNLT}
O EQUAL LEVEL CONTOURS (PNDB)

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IDENTIFICATION:
OMEGA 1.4
TEST 75-002-029
RUN 02
06 MAY 75
PAGE 16
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NOISE SOURCE/SUBJECT:	OPERATION:	METEOROLOGY:
C-141A AIRCRAFT	(87% RPM, 1.27 EPR	TEMP = 15 C
TF33-P-7 ENGINE	(ALL ENGINES	BAR PRESS = .760 M HG
FAR FIELD NOISE	(FREE FLOW	REL HUMID = 70 %



AZGJE HZ DWDGEWWS



IDENTIFICATIONS

OMEGA 1.4

METEOROLOGY:

(OPERATION:

TEMP = 15 C
BAR PRESS = .760 M HG

REL HUMID = 70 %) PAGE 17

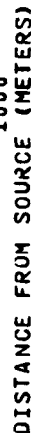
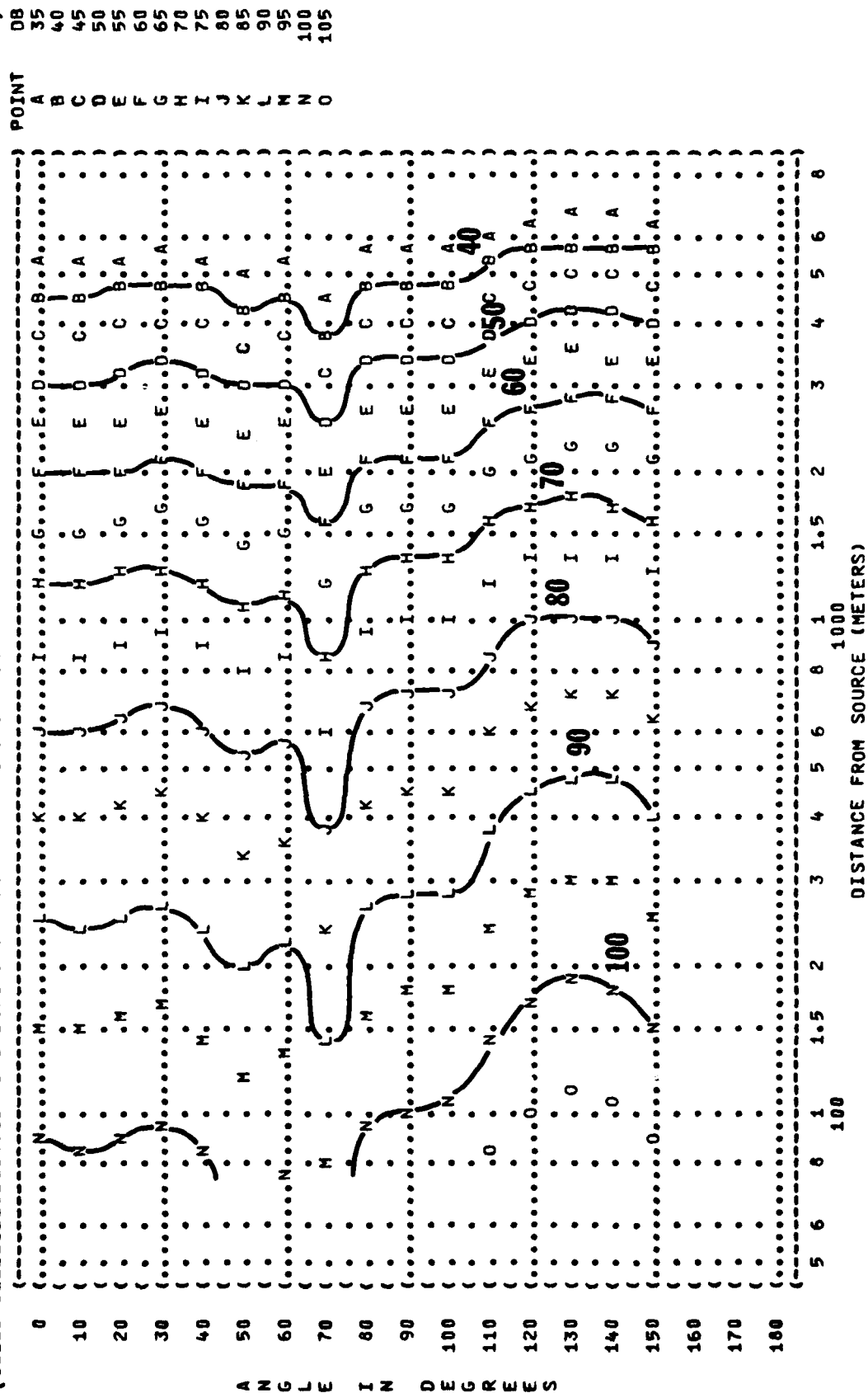


FIGURE: PREFERRED SPEECH INTERFERENCE LEVEL (PSIL)		IDENTIFICATION:
EQUAL LEVEL CONTOURS (DB)		
9		OMEGA 1.4
		TEST 75-002-025
NOISE SOURCE/SUBJECT:		RUN 03
(OPERATION:		METEOROLOGY:
(MILITARY POWER	TEMP = 15 C	
(98% RPM, 1.05 EPR	BAR PRESS = .760 M HG	06 MAY 75
(ALL ENGINES	REL HUMID = 70 %	
(FREE FLOW		PAGE 17
G-141A AIRCRAFT		
YF33-P-7 ENGINE		
FAR FIELD NOISE		



ANGIE IN DEGREES

FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)

IDENTIFICATION:

10 EQUAL TIME CONTOURS (MINUTES)

NO PROTECTION

NOISE SOURCE/SUBJECT: (OPERATION:) METEOROLOGY:)

(IDLE) TEMP = 15 C

(C-141A AIRCRAFT) 55% RPM, 1.04 EPR) BAR PRESS = .760 M HG

(TF33-P-7 ENGINE) ALL ENGINES) REL HUMID = 70 %

(FAR FIELD NOISE) FREE FLOW)

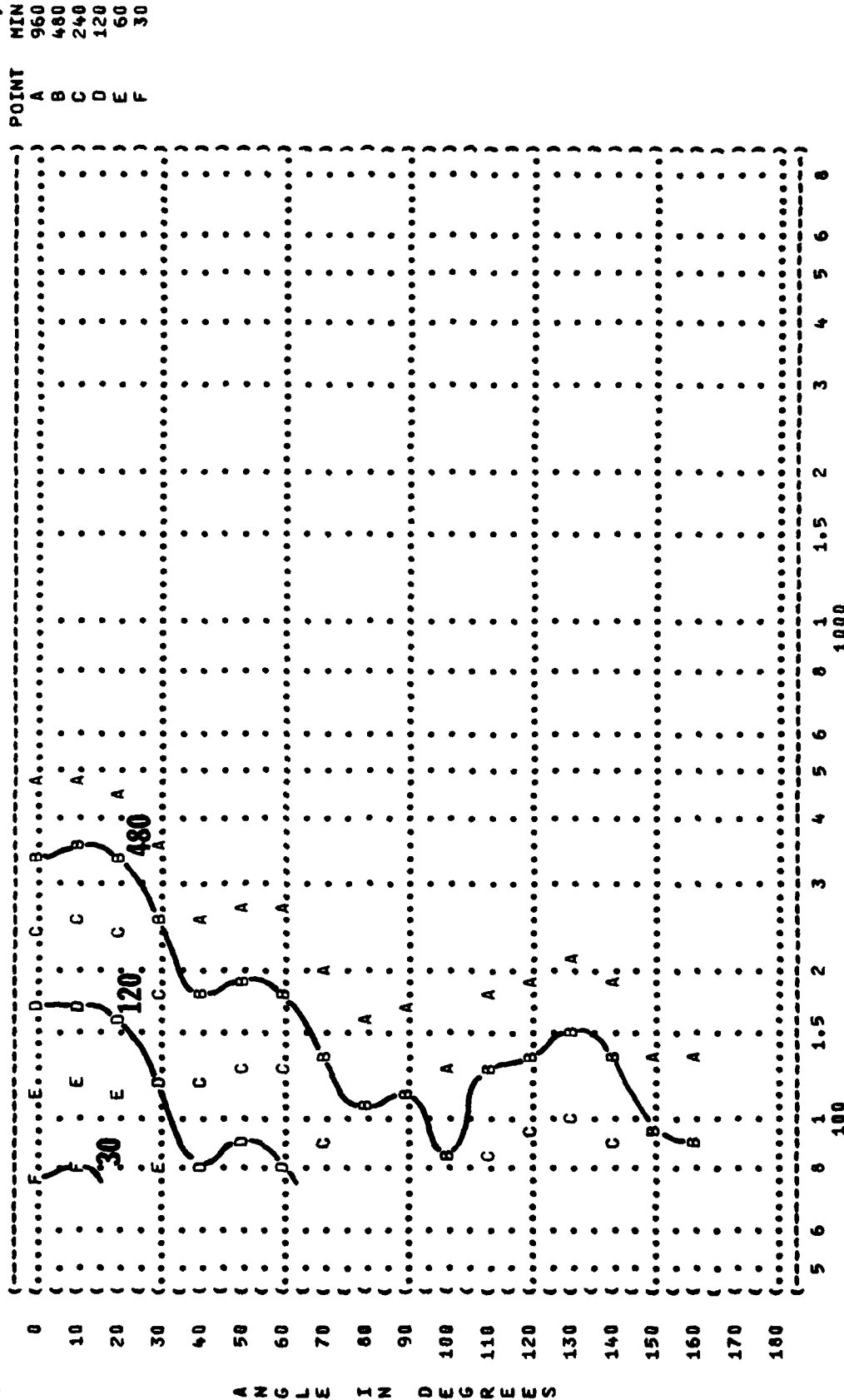
OMEGA 1.4

TEST 75-002-025

RUN 01

06 MAY 75

PAGE 7



ANGLING DEGREES

```
(-----)
( FIGURE: MAXIMUM PERMISSIBLE TIME {T} FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73) ) IDENTIFICATION:
(    10      EQUAL TIME CONTOURS (MINUTES) ) )
( ) OMEGA 1.4 )
(-----)
( NOISE SOURCE/SUBJECT: ) OPERATION: ) METEOROLOGY: ) RUN 01 ) TEST 75-002-025 )
( C-141A AIRCRAFT ) IDLE ) TEMP = 15 C ) )
( TF33-P-7 ENGINE ) 55% RPM, 1.04 EPR ) BAR PRESS = .760 M HG ) 06 MAY 75 )
( FAR FIELD NOISE ) ALL ENGINES ) REL HUMID = 70 % ) )
( FREE FLOW ) ) PAGE 8 )
(-----)
```

PERSONNEL MAY BE EXPOSED UP TO 960 MINUTES PER DAY
AT ALL DISTANCES FROM SOURCE EQUAL TO OR GREATER THAN 75 METERS
FOR ALL ANGLES EVALUATED (INDICATED BY $\angle$ AT LEFT)
UNDER THE FOLLOWING EAR PROTECTION CONDITIONS:

MINIMUM QPL EAR MUFFS
AMERICAN OPTICAL 1700 EAR MUFFS
V-51R EAR PLUGS
COMFIT TRIPLE FLANGE EAR PLUGS
H-133 GROUND COMMUNICATION UNIT

DISTANCE FROM SOURCE (METERS)

[illegible]

FIGURE 1 MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)

10 EQUAL TIME CONTOURS (MINUTES)

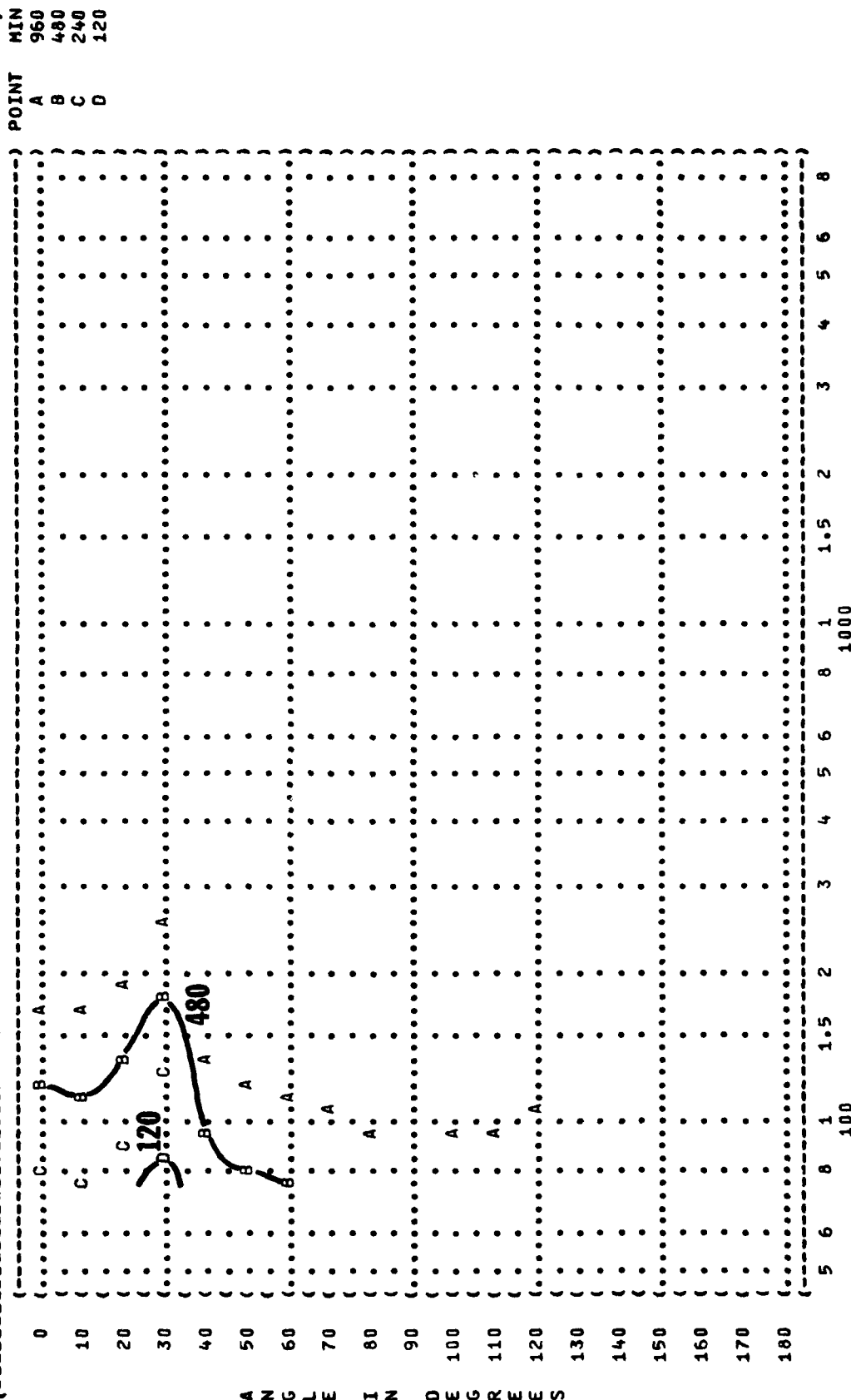
CONFIT TRIPLE FLANGE EAR PLUGS

NOISE SOURCE/SUBJECT: (OPERATION:) METEOROLOGY: (TEMP = 15 C)

C-141A AIRCRAFT (87% RPM, 1.27 EPR) BAR PRESS = .760 M HG

TF33-P-7 ENGINE (ALL ENGINES) REL HUMID = 70 %

FAR FIELD NOISE (FREE FLOW) PAGE 11



	(-----)	MIN	POINT
0	(.....A.....)	960	A
	(.....B.....)		
	(.....C.....)		
10	(.....A.....)	480	B
	(.....B.....)		
	(.....C.....)		
	(.....A.....)	240	C
	(.....B.....)		
	(.....C.....)		

[illegible][illegible][illegible][illegible][illegible][illegible][illegible]

200	(.)
170	(.)
180	(.)

(-----)

5	6	8	1	1.5	2	3	4	5	6	8
100										
1000										
DISTANCE FROM SOURCE (METERS)										

1

1

FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)

IDENTIFICATION:

10 EQUAL TIME CONTOURS (MINUTES)

NO PROTECTION

NOISE SOURCE/SUBJECT:

OPERATION:

MILITARY POWER

98% RPM, 1.85 EPR

ALL ENGINES

FREE FLOW

METEOROLOGY:

TEMP = 15 C

BAR PRESS = .760 M HG

REL HUMID = 70 %

OMEGA 1.4

TEST 75-002-025

RUN 03

06 MAY 75

PAGE 7

POINT MIN

A 960

B 480

C 240

D 120

E 60

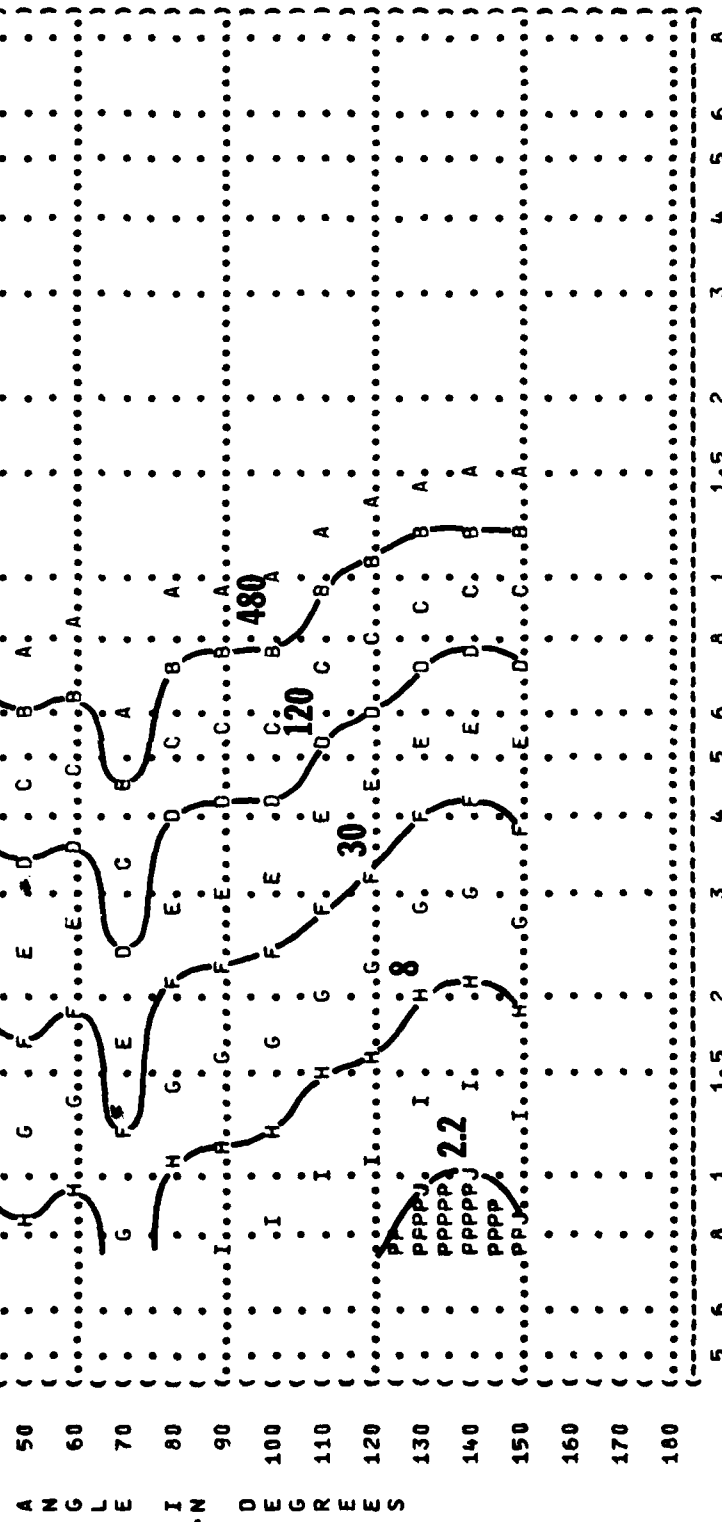
F 30

G 15

H 8

I 4

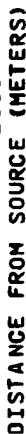
J 2.2



DISTANCE FROM SOURCE (METERS)

P ADDITIONAL EAR PROTECTION REQUIRED.

	(	-	-	-	-	-	-	)	MIN	POINT
0	(	.	.	.	.	.	.	)	A 960	A
	(	.	.	.	.	.	.	)	B 480	B
10	(	.	.	.	.	.	.	)	C 240	C
	(	.	.	.	.	.	.	)	D 120	D
20	(	.	.	.	.	.	.	)	E 60	E
	(	.	.	.	.	.	.	)	F 30	F
30	(	.	.	.	.	.	.	)	G 15	G
	(	.	.	.	.	.	.	)	H 8	H



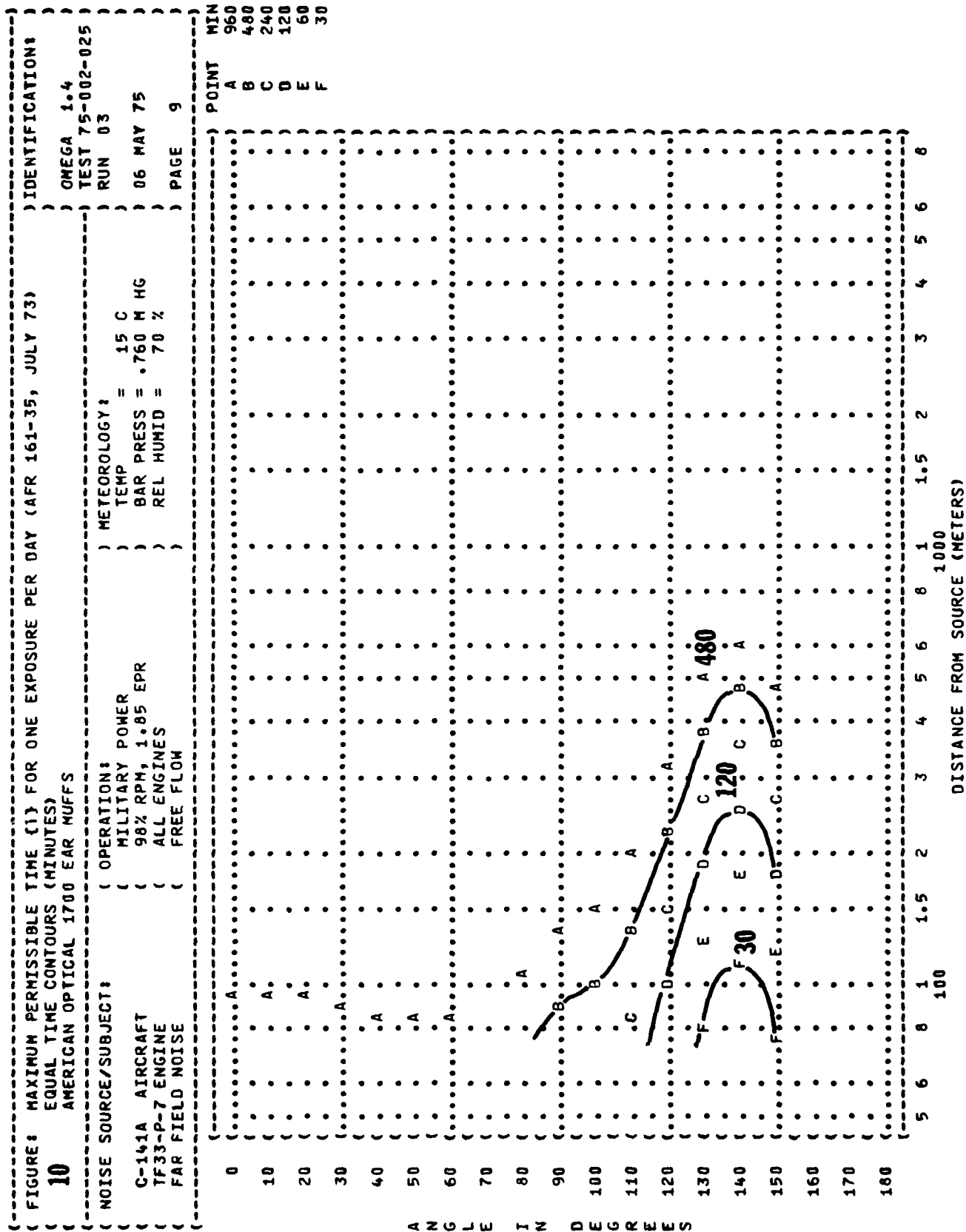


FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)
 EQUAL TIME CONTOURS (MINUTES)
 H-133 GROUND COMMUNICATION UNIT

10

IDENTIFICATION:)
)
 OMEGA 1.4)
 TEST 75-002-025)

NOISE SOURCE/SUBJECT:	OPERATION:	METEOROLOGY:	TIME
C-141A AIRCRAFT	MILITARY POWER	TEMP = 15 C	RUN 03
TF33-P-7 ENGINE	98% RPM, 1.05 EPR	BAR PRESS = .760 M HG	06 MAY 75
FAR FIELD NOISE	ALL ENGINES	REL HUMID = 70 %	PAGE 12
	FREE FLOW		

	MIN	POINT
0	A	A
10	A	B
20	A	C
		D
		E

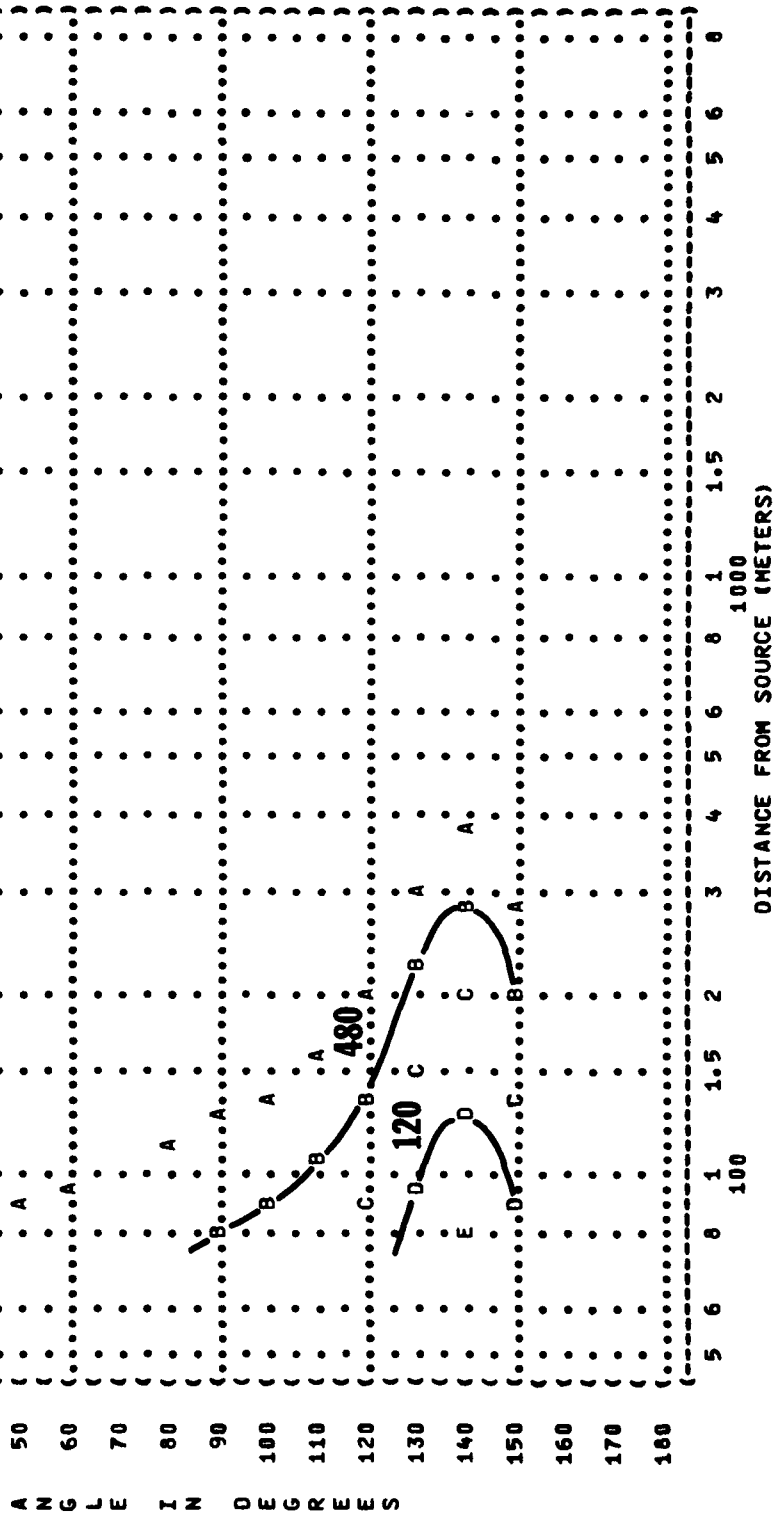
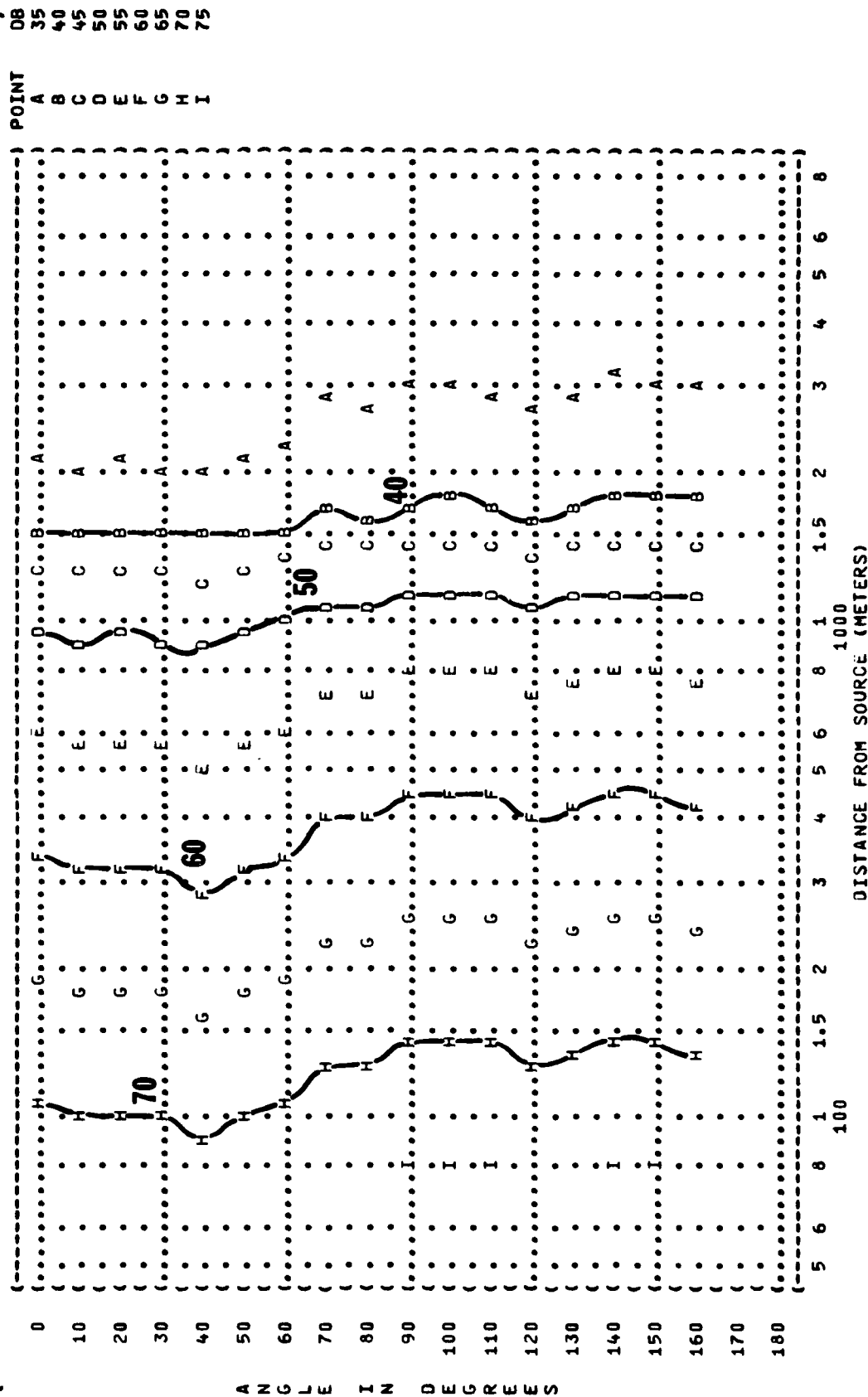


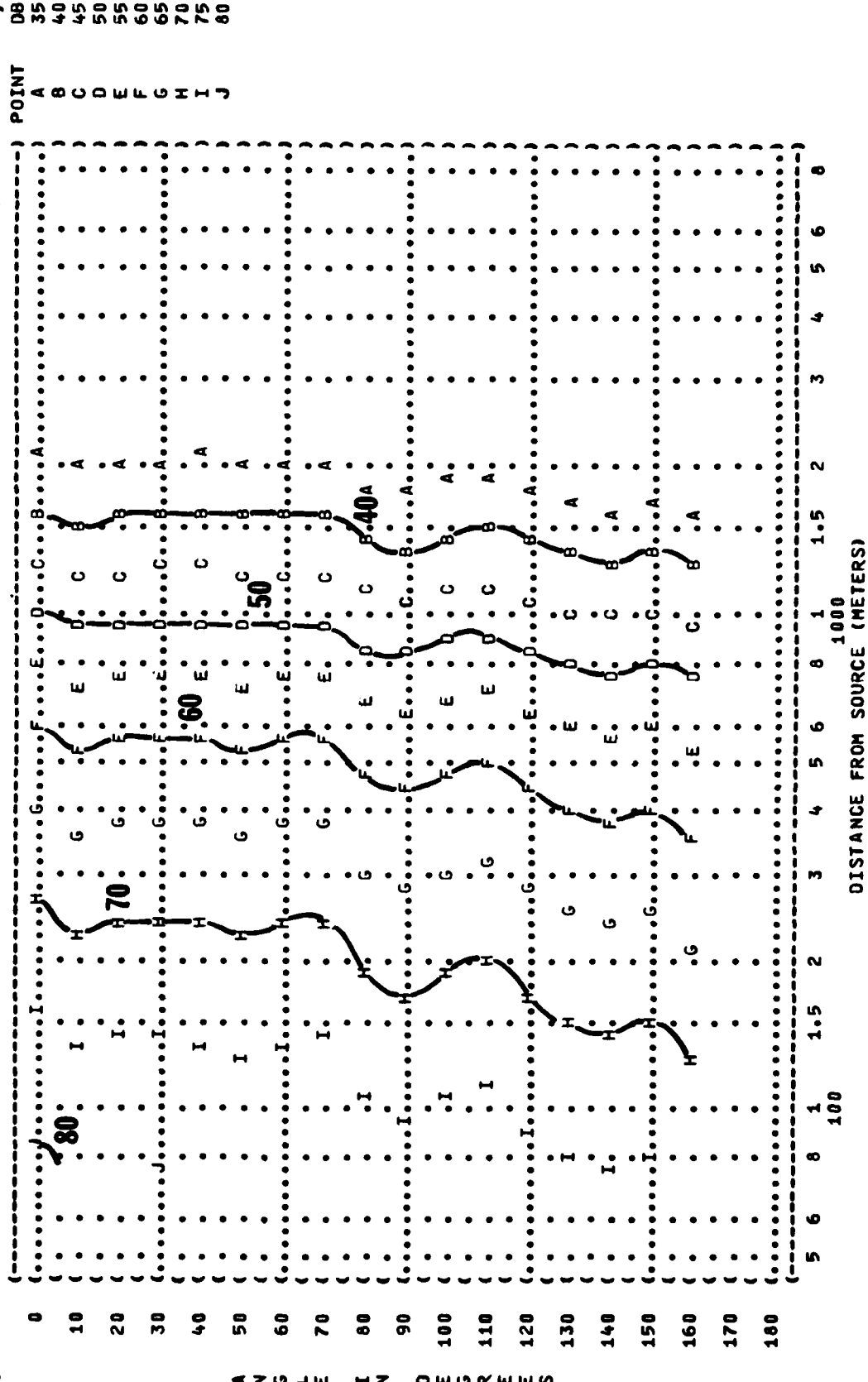
FIGURE: SOUND PRESSURE LEVEL (SPL)
 11 EQUAL LEVEL CONTOURS (DB)
 31.5 HZ OCTAVE BAND

NOISE SOURCE/SUBJECT: (OPERATION:) METEOROLOGY:)
 (C-141A AIRCRAFT (IDLE) TEMP = 15 C)
 (TF33-P-7 ENGINE (55% RPM, 1.04 EPR) BAR PRESS = .760 M HG)
 (FAR FIELD NOISE (ALL ENGINES) REL HUMID = 70 %)
 (FREE FLOW))

IDENTIFICATION:)
 OMEGA 1.4
 TEST 75-002-025
 RUN 01
 06 MAY 75
 PAGE 18



(FIGURE: SOUND PRESSURE LEVEL (SPL)
 (11 EQUAL LEVEL CONTOURS (DB)
 (63 HZ OCTAVE BAND
 (NOISE SOURCE/SUBJECT: (OPERATIONS:
 ((IDLE
 (C-141A AIRCRAFT (55% RPM, 1.04 EPR
 (TF33-P-7 ENGINE (ALL ENGINES
 (FAR FIELD NOISE (FREE FLOW
 (METEOROLOGY: (TEMP = 15 C
 (BAR PRESS = .760 M HG
 (REL HUMID = 70 %
 (IDENTIFICATION: (OMEGA 1.4
 (TEST 75-002-025
 (RUN 01
 (06 MAY 75
 (PAGE 19



A N G L E I N D E G R E E S

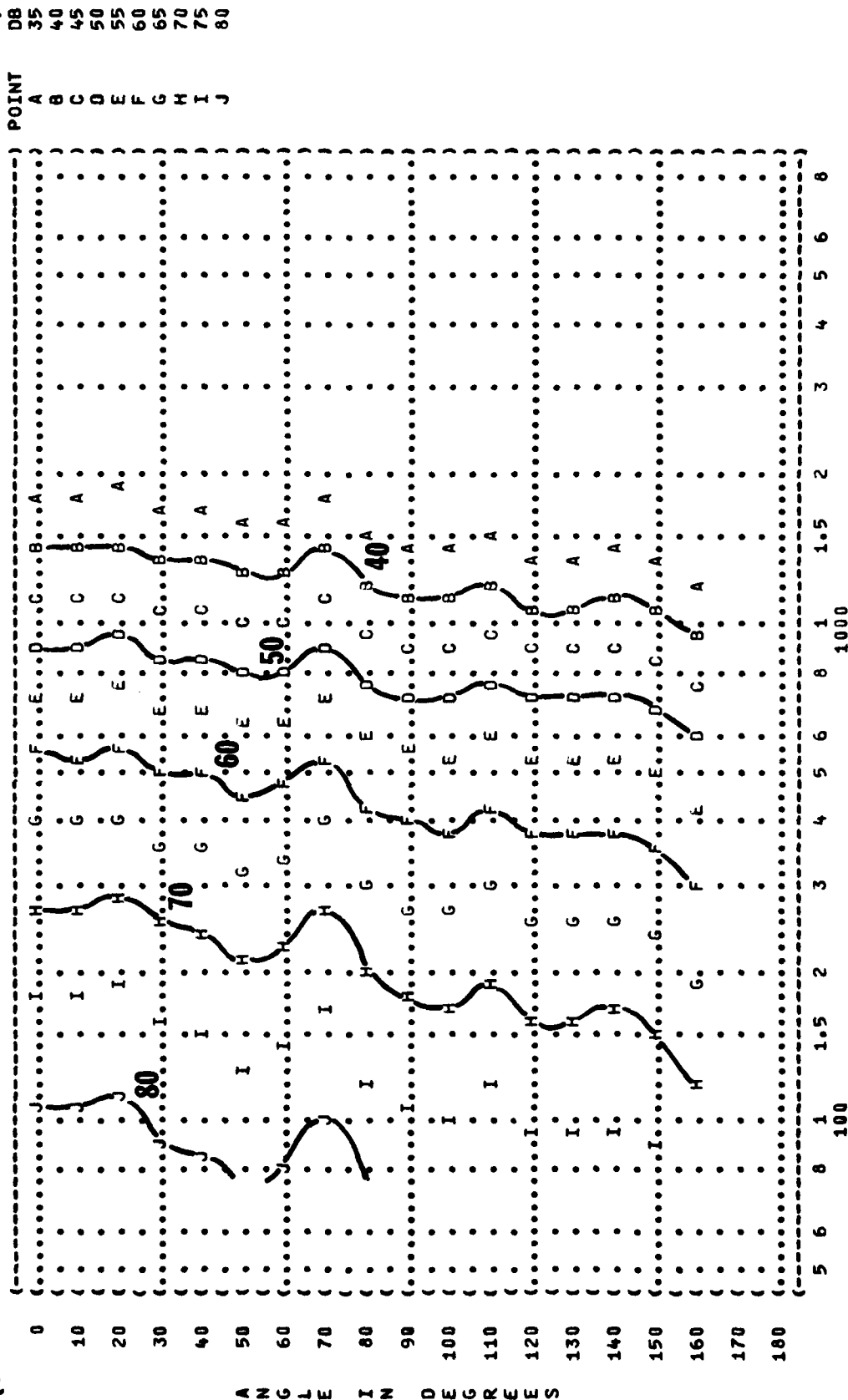
FIGURE 11 SOUND PRESSURE LEVEL (SPL)
EQUAL LEVEL CONTOURS
125 HZ OCTAVE BAND

NOISE SOURCE/SUBJECT:	OPERATION:
C-141A AIRCRAFT	IDLE
TF33-P-7 ENGINE	552 RPM
FAR FIELD NOISE	ALL ENGINE
	FREE FLOW

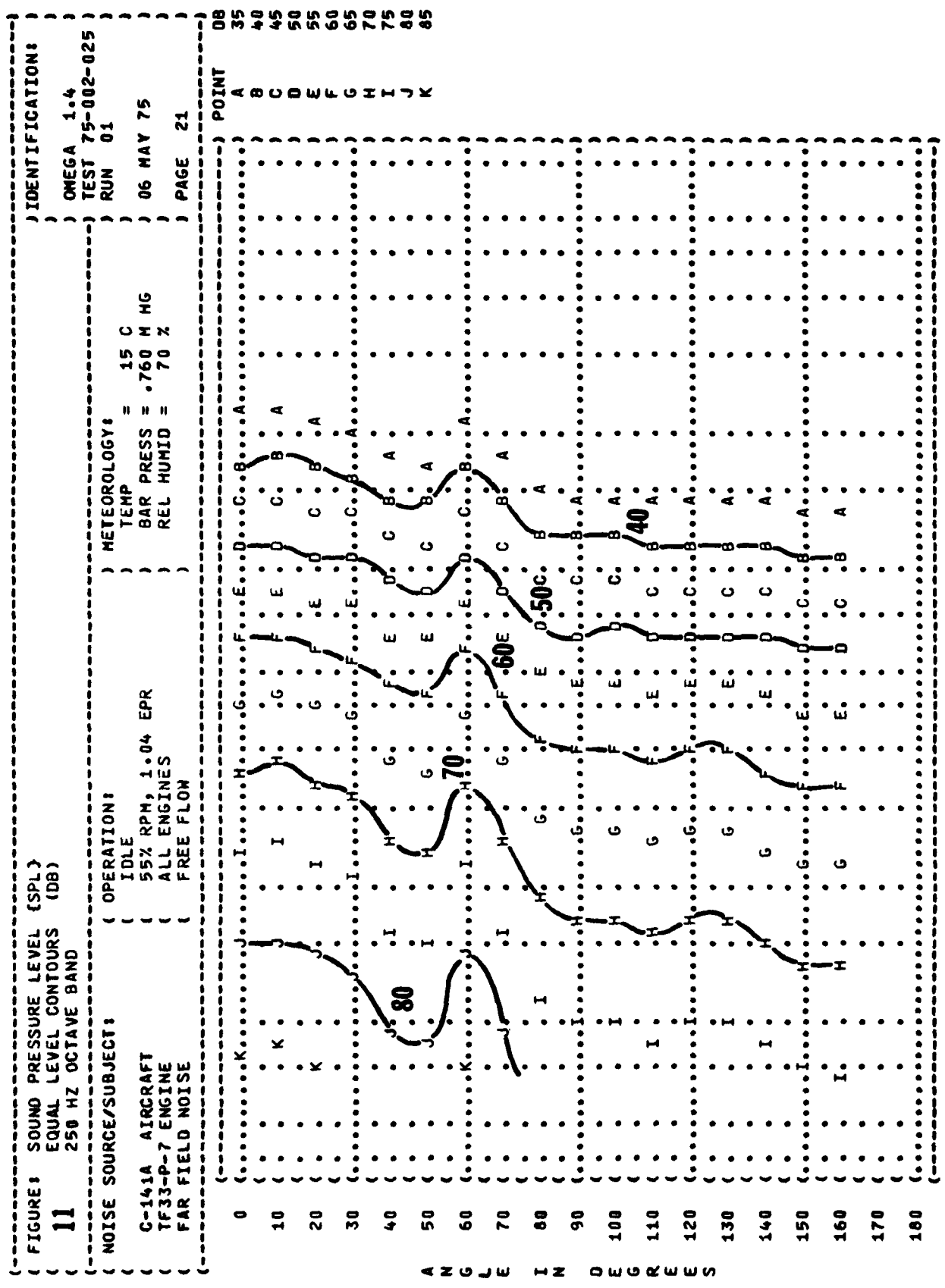
) METEOROLOGY: =
) TEMP =
) BAR PRESS =
) REL HUMID =
)

TEOROLOGY: = 15 C
TEMP = .760 M HG
BAR PRESS = 70 %
REL HUMID

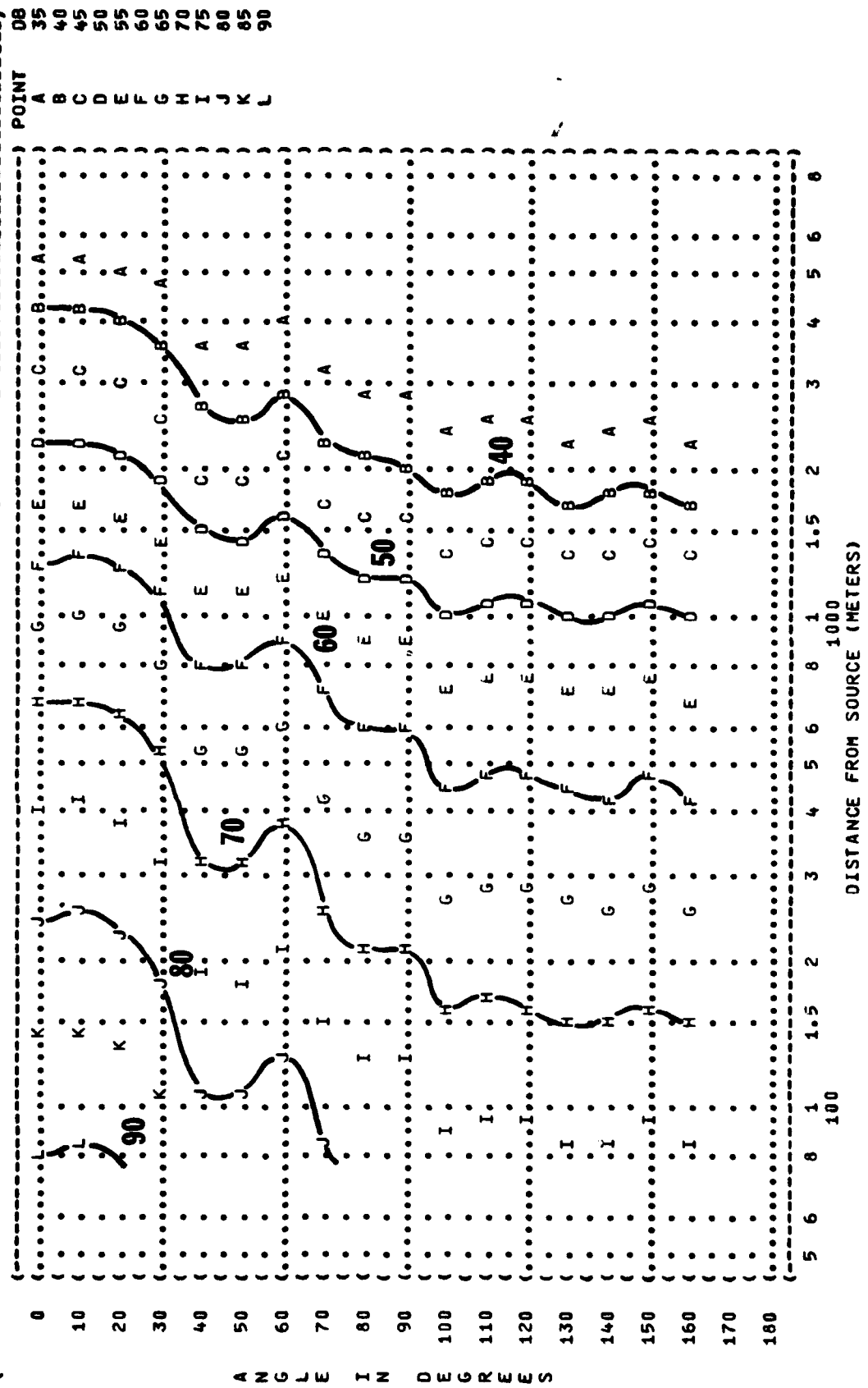
RUN 01
 06 MAY 75
 PAGE 20



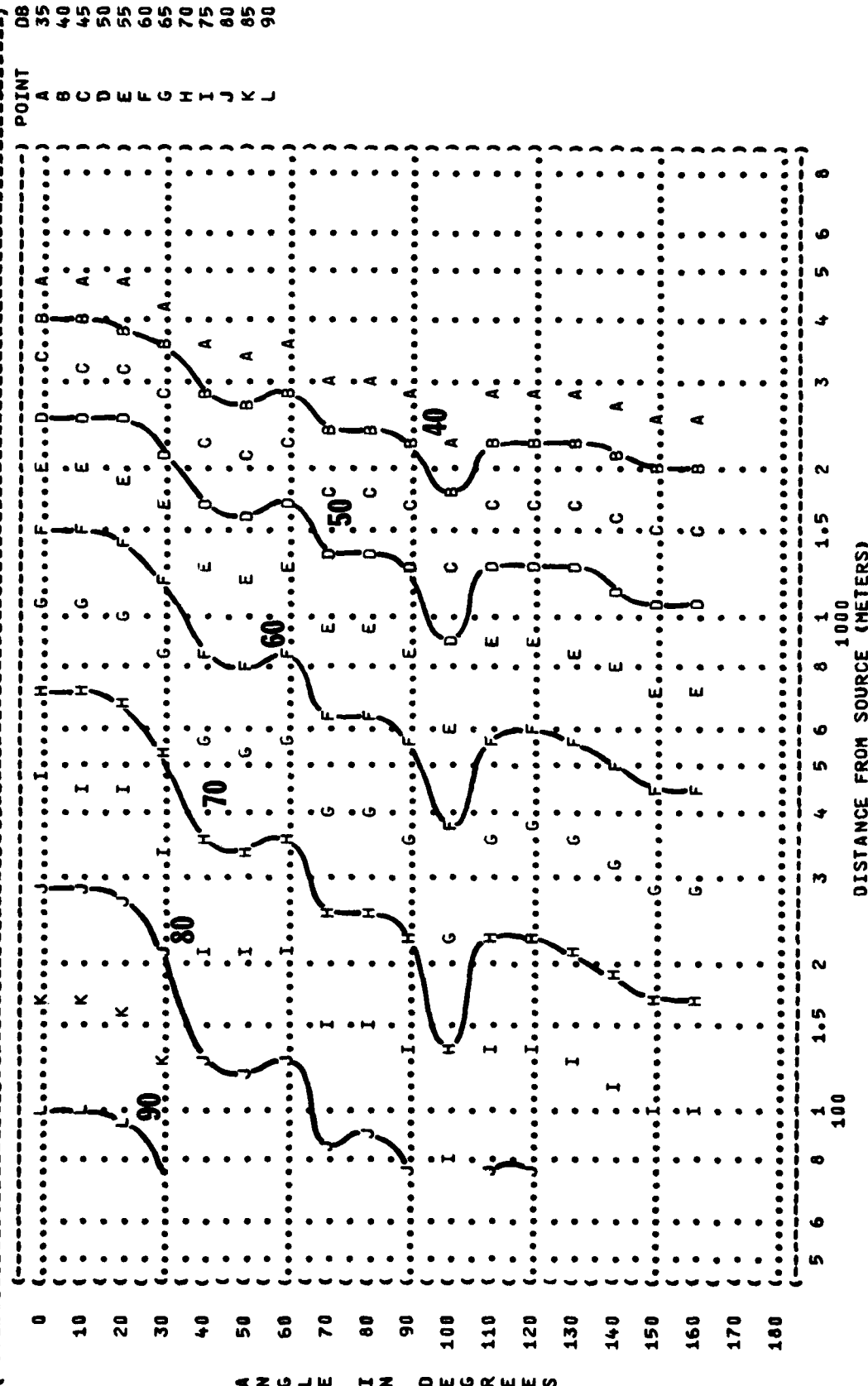
ANGLE IN DEGREES



(FIGURE: SOUND PRESSURE LEVEL (SPL)
 (11 EQUAL LEVEL CONTOURS (DB)
 (500 HZ OCTAVE BAND
 (NOISE SOURCE/SUBJECT: (OPERATION:
 ((IDLE
 ((55% RPM, 1.04 EPR
 ((TF33-P-7 ENGINE
 ((FAR FIELD NOISE
 ((FREE FLOW
 (METEOROLOGY:
 (TEMP = 15 C
 (BAR PRESS = .760 M HG
 (REL HUMID = 70 %
 (PAGE 22
 (IDENTIFICATION:
 (OMEGA 1.4
 (TEST 75-002-025
 (RUN 01
 (06 MAY 75
 (



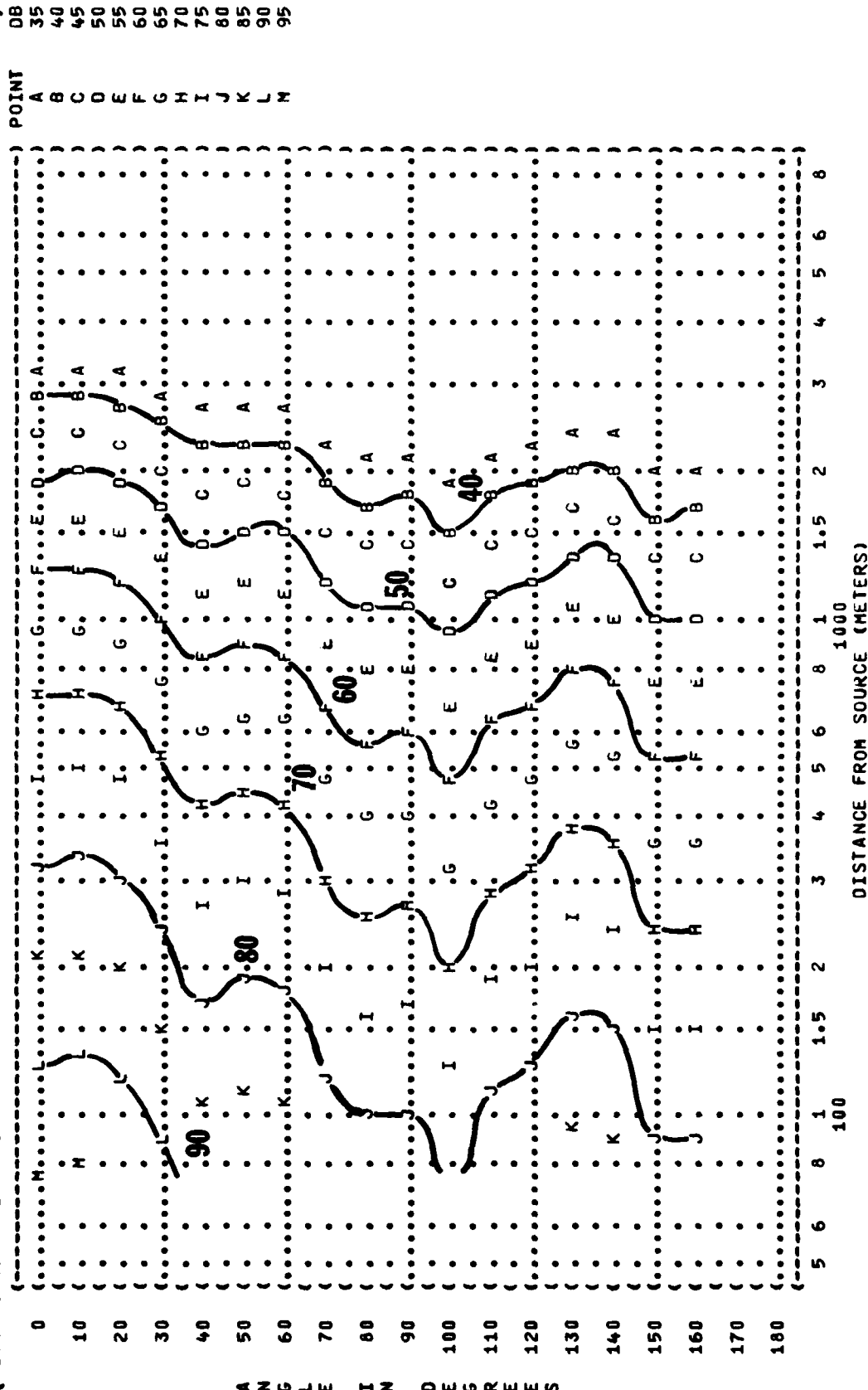
(FIGURE: SOUND PRESSURE LEVEL (SPL)
 (EQUAL LEVEL CONTOURS (DB)
 (11 1000 HZ OCTAVE BAND
 (NOISE SOURCE/SUBJECT: (OPERATION:
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 (C-141A AIRCRAFT (55% RPM, 1.04 EPR
 (TF33-P-7 ENGINE (ALL ENGINES
 (FAR FIELD NOISE (FREE FLOW
 (METEOROLOGY:
 (TEMP = 15 C
 (BAR PRESS = .760 M HG
 (REL HUMID = 70 %
 (IDENTIFICATION:
 (OMEGA 1.4
 (TEST 75-002-025
 (RUN 01
 (06 MAY 75
 (PAGE 23



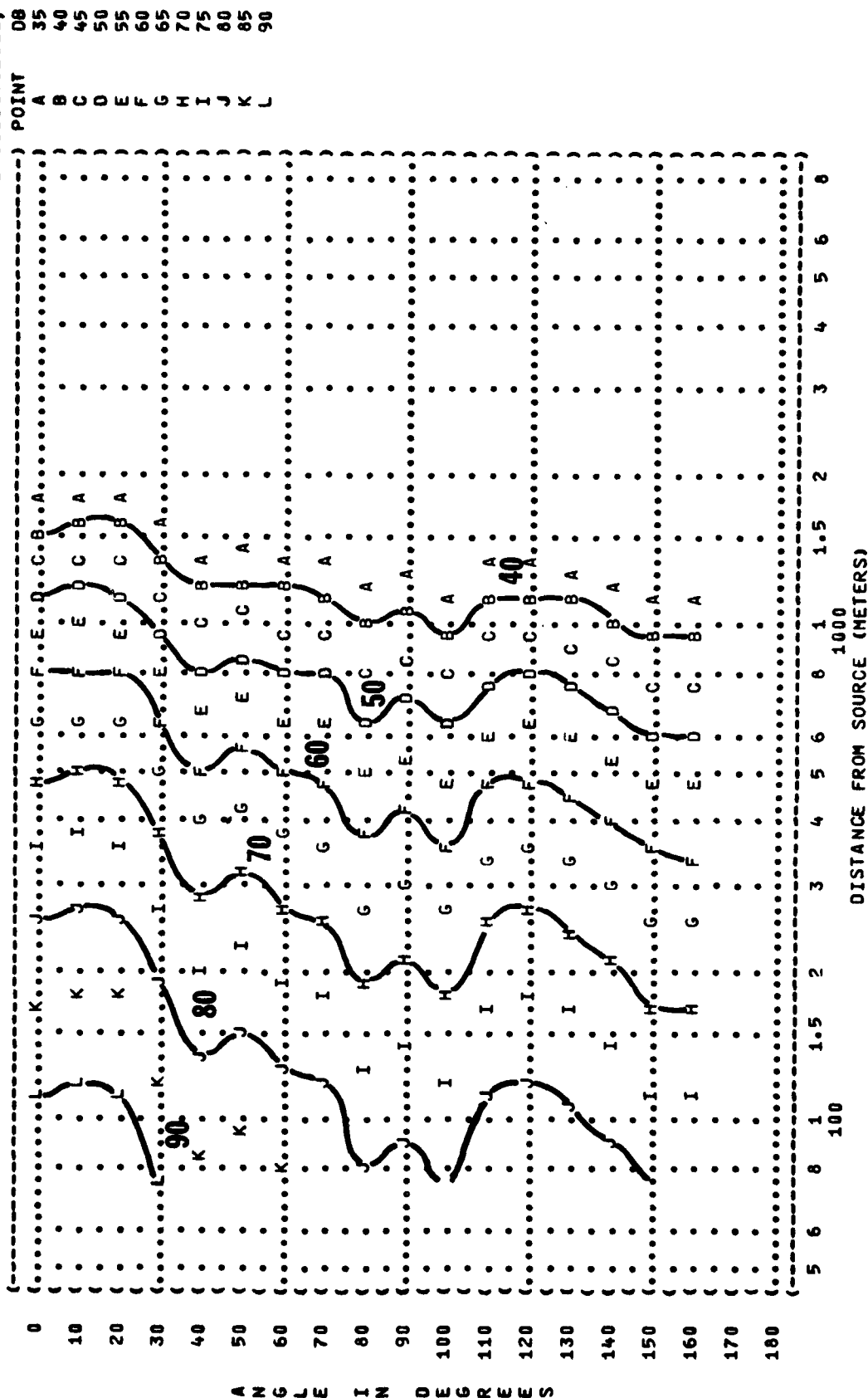
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(-----)
( FIGURE: SOUND PRESSURE LEVEL {SPL} ) IDENTIFICATION: )
( EQUAL LEVEL CONTOURS (DB) ) )
( 11 2000 HZ OCTAVE BAND ) OMEGA 1.4 )
( ) TEST 75-002-025 )
( NOISE SOURCE/SUBJECT: ) METEOROLOGY: ) RUN 01 )
( ) ) )
( C-141A AIRCRAFT ) TEMP = 15 C )
( TF33-P-7 ENGINE ) BAR PRESS = .760 M HG ) 06 MAY 75 )
( FAR FIELD NOISE ) REL HUMID = 70 % ) )
( ) ) ) PAGE 24 )
(-----)

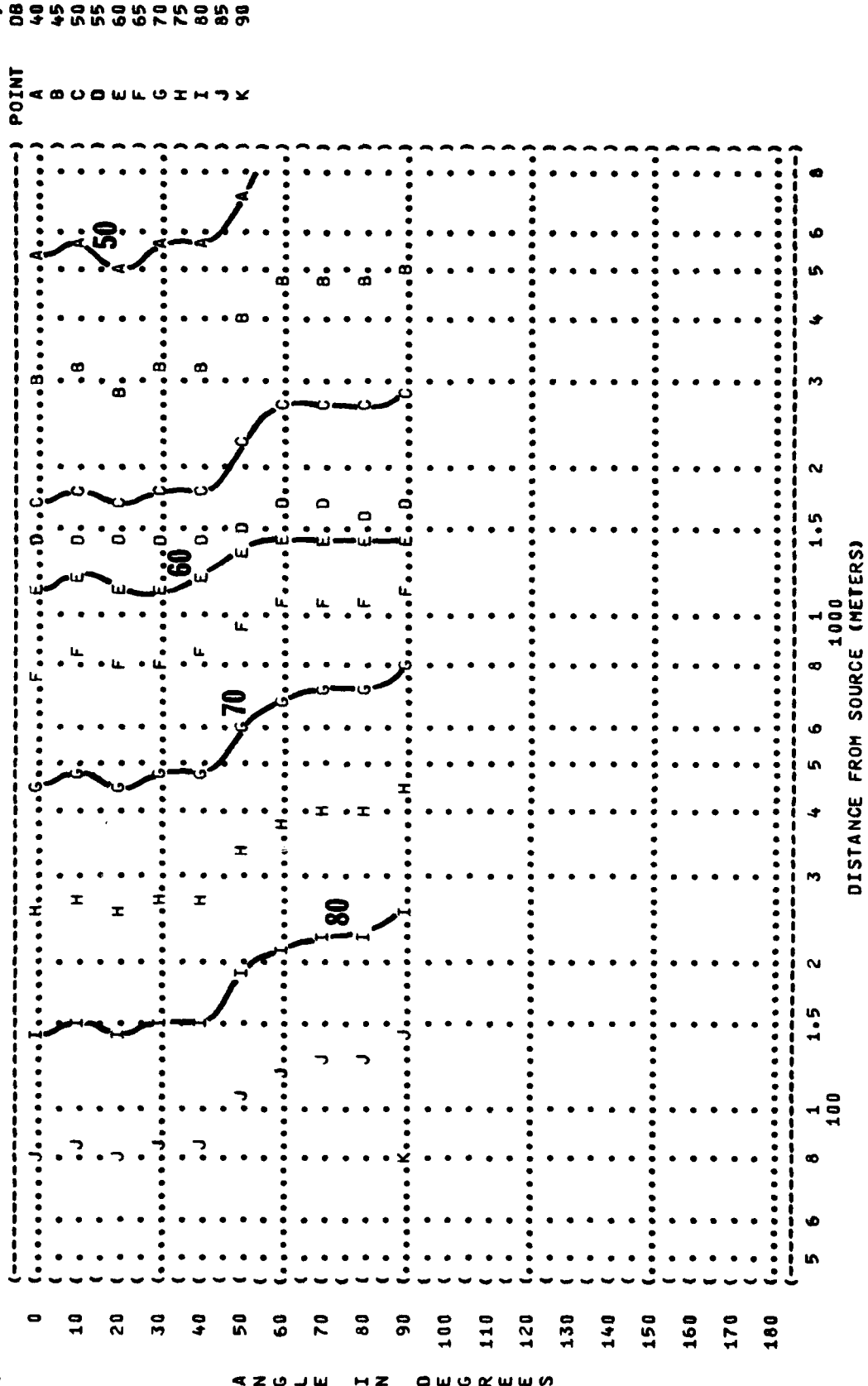
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(FIGURE: SOUND PRESSURE LEVEL (SPL))
 (11 EQUAL LEVEL CONTOURS (DB))
 (4000 HZ OCTAVE BAND)
 (NOISE SOURCE/SUBJECT:)
 (C-141A AIRCRAFT)
 (TF33-P-7 ENGINE)
 (FAR FIELD NOISE)
 (OPERATION:)
 (IDLE)
 (55% RPM, 1.04 EPR)
 (ALL ENGINES)
 (FREE FLOW)
 (METEOROLOGY:)
 (TEMP = 15 C)
 (BAR PRESS = .760 M HG)
 (REL HUMID = 70 %)
 (IDENTIFICATION:)
 (OMEGA 1.4)
 (TEST 75-002-025)
 (RUN 01)
 (06 MAY 75)
 (PAGE 25)

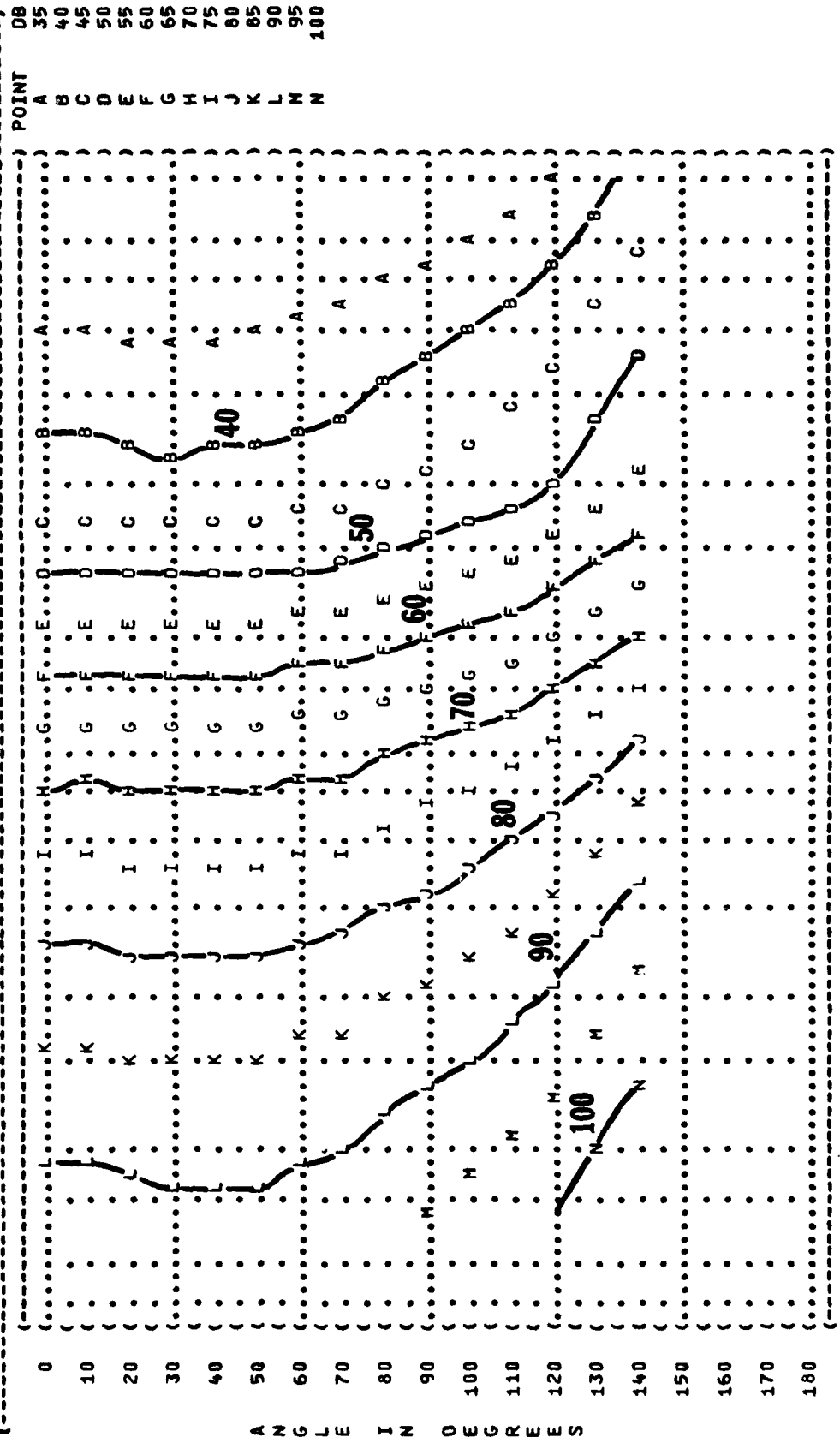


(FIGURE: SOUND PRESSURE LEVEL (SPL)
 (EQUAL LEVEL CONTOURS (DB)
 (11 31.5 HZ OCTAVE BAND
 (NOISE SOURCE/SUBJECT: (OPERATION:
 (C-141A AIRCRAFT (87% RPM, 1.27 EPR
 (TF33-P-7 ENGINE (ALL ENGINES
 (FAR FIELD NOISE (FREE FLOW
 (METEOROLOGY:
 (TEMP = 15 C
 (BAR PRESS = .760 M HG
 (REL HUMID = 70 %
 (IDENTIFICATION:
 (OMEGA 1.4
 (TEST 75-002-025
 (RUN 02
 (06 MAY 75
 (PAGE 18



A N G L E I N D E G R E E S

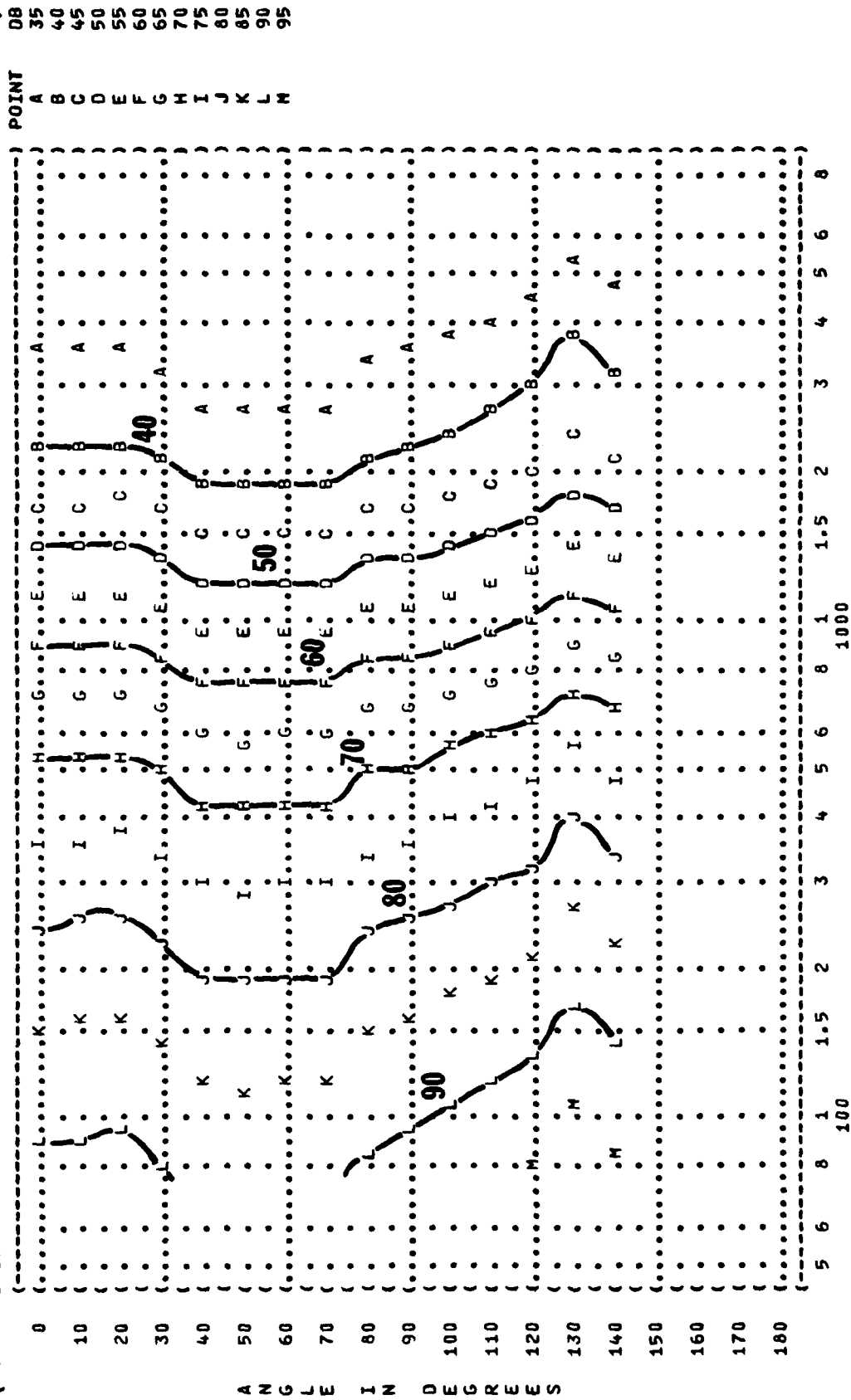
((FIGURE: SOUND PRESSURE LEVEL (SPL)
 ((EQUAL LEVEL CONTOURS (DB)
 ((11 125 HZ OCTAVE BAND
 ((NOISE SOURCE/SUBJECT: (OPERATION: (METEOROLOGY: (POINT DB
 ((C-141A AIRCRAFT (87% RPM, 1.27 EPR (TEMP = 15 C (A 35
 ((TF33-P-7 ENGINE (ALL ENGINES (BAR PRESS = .760 M HG (B 40
 ((FAR FIELD NOISE (FREE FLOW (REL HUMID = 70 % (C 45
 (((((((((D 50
 (((((((((E 55
 (((((((((F 60
 (((((((((G 65
 (((((((((H 70
 (((((((((I 75
 (((((((((J 80
 (((((((((K 85
 (((((((((L 90
 (((((((((M 95
 (((((((((N 100



((IDENTIFICATION: ((OMEGA 1.4
 ((TEST 75-002-025
 ((RUN 02
 ((06 MAY 75
 ((PAGE 20

DISTANCE FROM SOURCE (METERS)

(FIGURE: SOUND PRESSURE LEVEL (SPL))
 (11 EQUAL LEVEL CONTOURS (DB))
 (250 HZ OCTAVE BAND)
 (NOISE SOURCE/SUBJECT:)
 (C-141A AIRCRAFT)
 (TF33-P-7 ENGINE)
 (FAR FIELD NOISE)
 (OPERATION:)
 (87% RPM, 1.27 EPR)
 (ALL ENGINES)
 (FREE FLOW)
 (METEOROLOGY:)
 (TEMP = 15 C)
 (BAR PRESS = .760 M HG)
 (REL HUMID = 70 %)
 (IDENTIFICATION:)
 (OMEGA 1.4)
 (TEST 75-002-025)
 (RUN 02)
 (06 MAY 75)
 (PAGE 21)



(FIGURE: SOUND PRESSURE LEVEL (SPL)
 (EQUAL LEVEL CONTOURS (DB)
 (11 1000 HZ OCTAVE BAND
 (NOISE SOURCE/SUBJECT: (OPERATION:
 (C-141A AIRCRAFT (87% RPM, 1.27 EPR
 (TF33-P-7 ENGINE (ALL ENGINES
 (FAR FIELD NOISE (FREE FLOW
 (METEOROLOGY: (TEMP = 15 C
 (BAR PRESS = .760 M HG
 (REL HUMID = 70 %
 (IDENTIFICATION: (OMEGA 1.4
 (TEST 75-002-025
 (RUN 02
 (06 MAY 75
 (PAGE 23

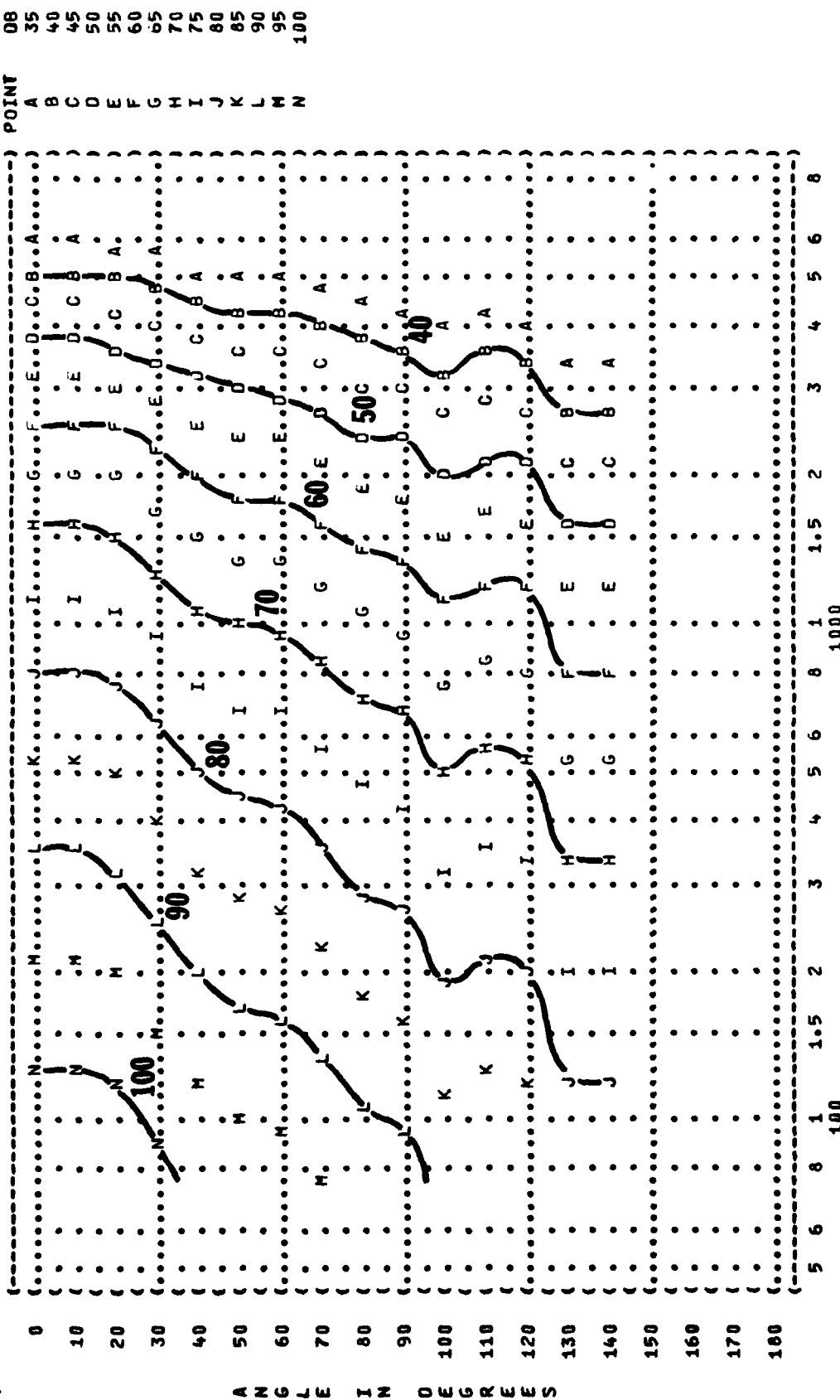
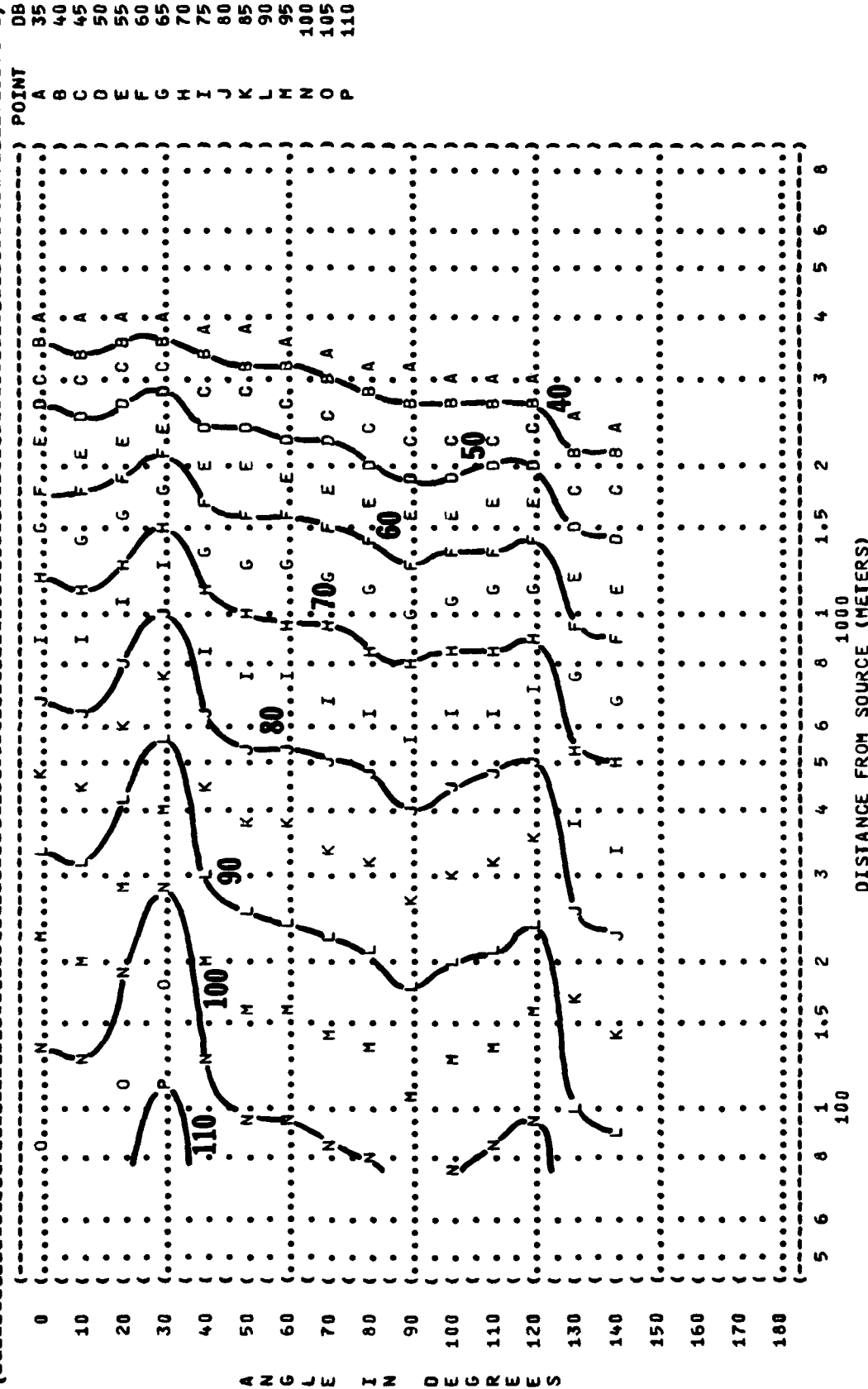
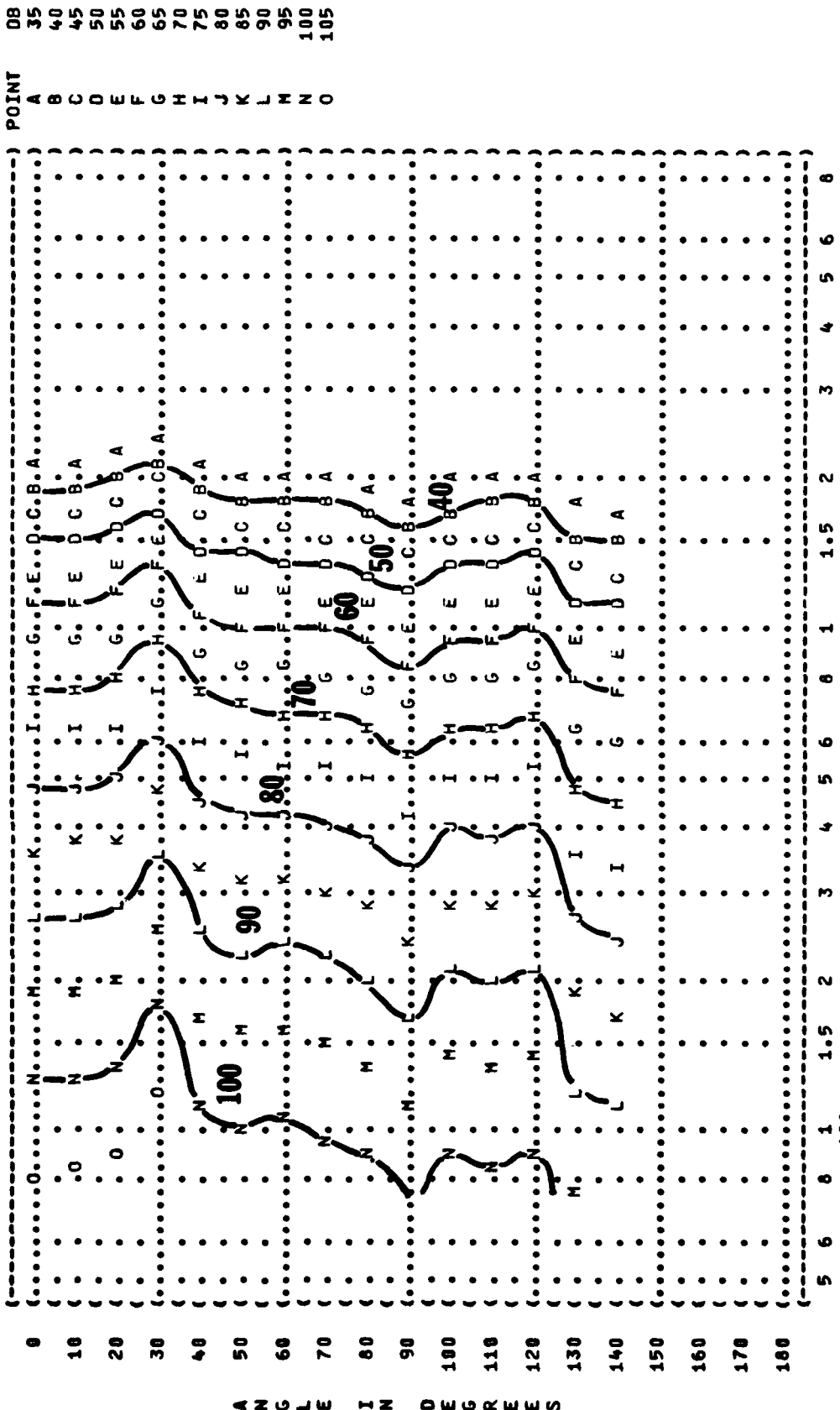


FIGURE 1 SOUND PRESSURE LEVEL (SPL)
EQUAL LEVEL CONTOURS (DB)
2000 HZ OCTAVE BAND

IDENTIFICATION:
OMEGA 1.4
TEST 75-002-025
RUN 02
METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 M HG
REL HUMID = 70 %
OPERATION:
C-141A AIRCRAFT
TF33-P-7 ENGINE
FAR FIELD NOISE
87% RPM, 1.27 EPR
ALL ENGINES
FREE FLOW



(FIGURE: SOUND PRESSURE LEVEL (SPL))
 (11 EQUAL LEVEL CONTOURS (DB))
 (4000 HZ OCTAVE BAND)
 (NOISE SOURCE/SUBJECT:)
 (C-141A AIRCRAFT)
 (TF33-P-7 ENGINE)
 (FAR FIELD NOISE)
 (OPERATION:)
 (87% RPM, 1.27 EPR)
 (ALL ENGINES)
 (FREE FLOW)
 (METEOROLOGY:)
 (TEMP = 15 C)
 (BAR PRESS = .760 M HG)
 (REL HUMID = 70 %)
 (IDENTIFICATION:)
 (OMEGA 1.4)
 (TEST 75-002-025)
 (RUN 02)
 (06 MAY 75)
 (PAGE 25)

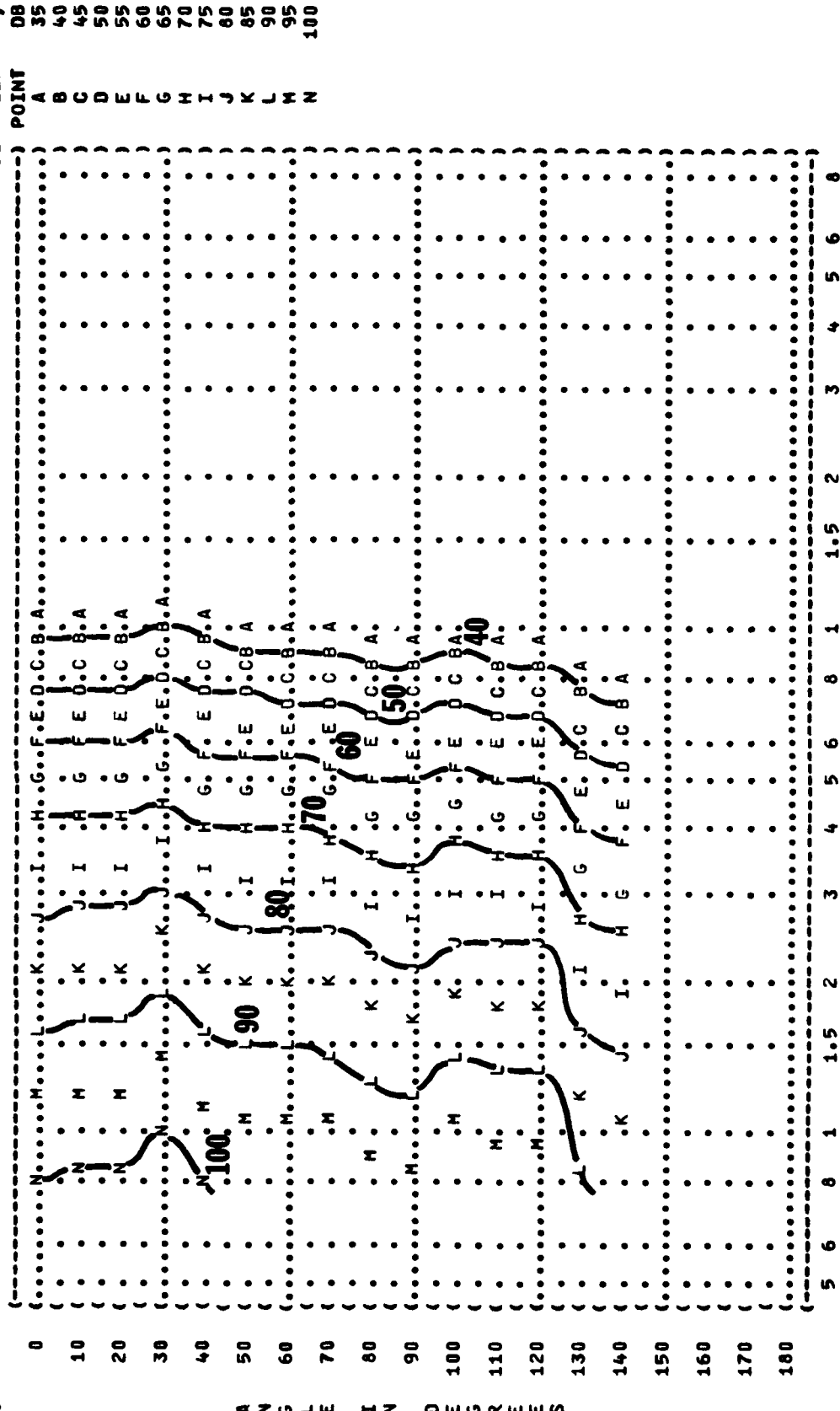


DISTANCE FROM SOURCE (METERS)

FIGURE: SOUND PRESSURE LEVEL (SPL)
 11 EQUAL LEVEL CONTOURS (DB)
 8000 HZ OCTAVE BAND

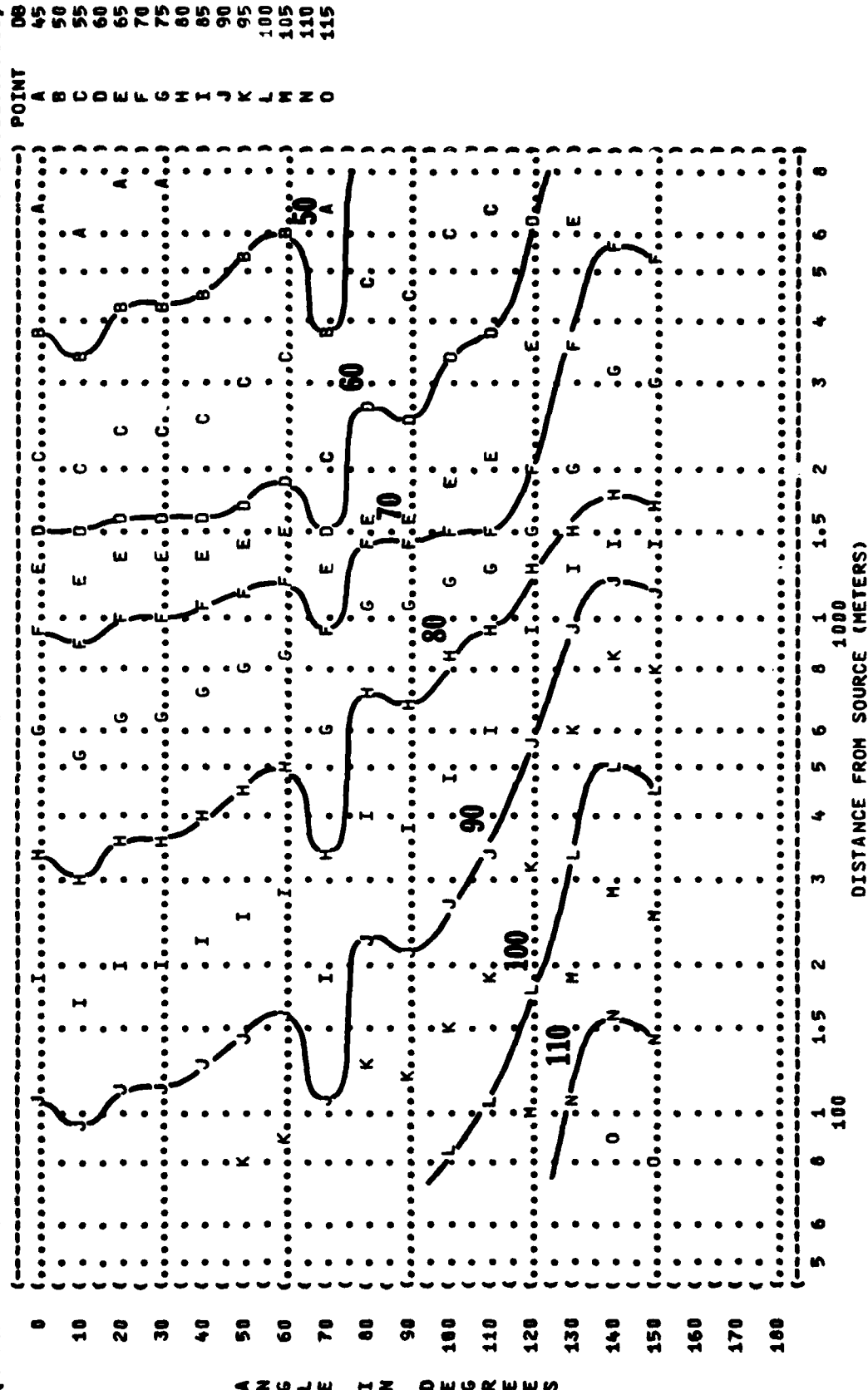
NOISE SOURCE/SUBJECT: (OPERATION:) METEOROLOGY:)
 (C-141A AIRCRAFT (87% RPM, 1.27 EPR) TEMP = 15 C)
 (TF33-P-7 ENGINE (ALL ENGINES) BAR PRESS = .760 M HG)
 (FAR FIELD NOISE (FREE FLOW) REL HUMID = 70 %)

IDENTIFICATION:)
) OMEGA 1.4
) TEST 75-002-025
) RUN 02
) 06 MAY 75
) PAGE 26

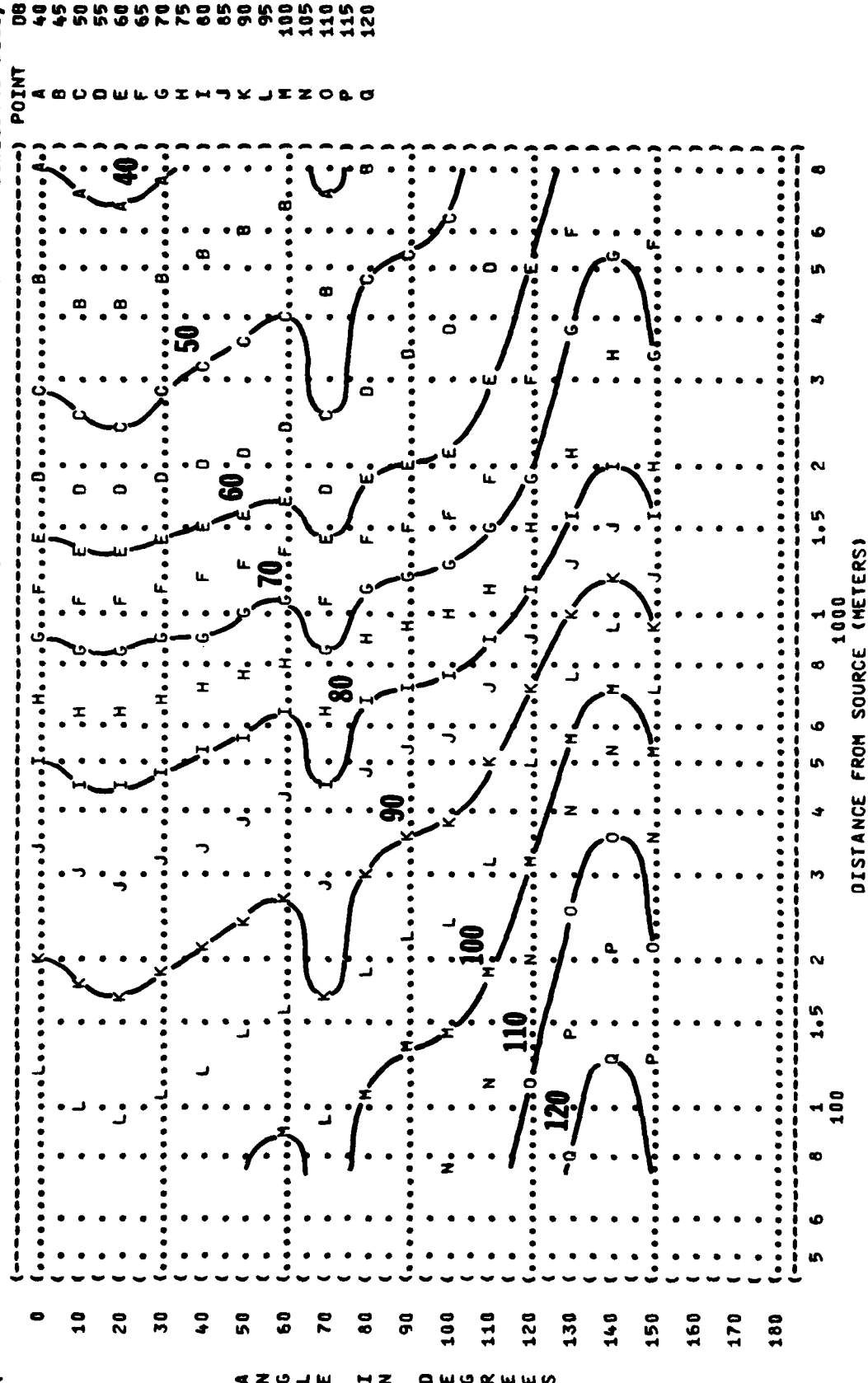


A N G L E I N D E G R E E S

(FIGURE: SOUND PRESSURE LEVEL (SPL)
 (11 EQUAL LEVEL CONTOURS (DB)
 (31.5 HZ OCTAVE BAND
 (NOISE SOURCE/SUBJECT: (OPERATION: (METEOROLOGY: (POINT DB
 (C-141A AIRCRAFT (MILITARY POWER (TEMP = 15 C
 (TF33-P-7 ENGINE (98% RPM, 1.85 EPR (BAR PRESS = .760 M HG
 (FAR FIELD NOISE (ALL ENGINES (REL HUMID = 70 %
 (FREE FLOW (PAGE 18



(FIGURE: SOUND PRESSURE LEVEL (SPL)
 (11 EQUAL LEVEL CONTOURS (DB)
 (63 HZ OCTAVE BAND
 (NOISE SOURCE/SUBJECT: (OPERATION: (METEOROLOGY:
 ((MILITARY POWER (TEMP = 15 C
 (C-141A AIRCRAFT (98% RPM, 1.85 EPR (BAR PRESS = .760 M HG
 (TF33-P-7 ENGINE (ALL ENGINES (REL HUMID = 70 %
 (FAR FIELD NOISE (FREE FLOW (PAGE 19)



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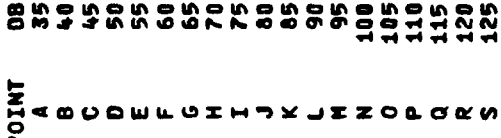
IDENTIFICATION#
)
)
) OMEGA 1.4
)
) TEST 75-002-025
)
) RUN 03
)
)
) 06 MAY 75
)
)
)
) PAGE 20
)

```

METEOROLOGY: 15 C
TEMP =
BAR PRESS = .760 M
REL HUMID = 70 %

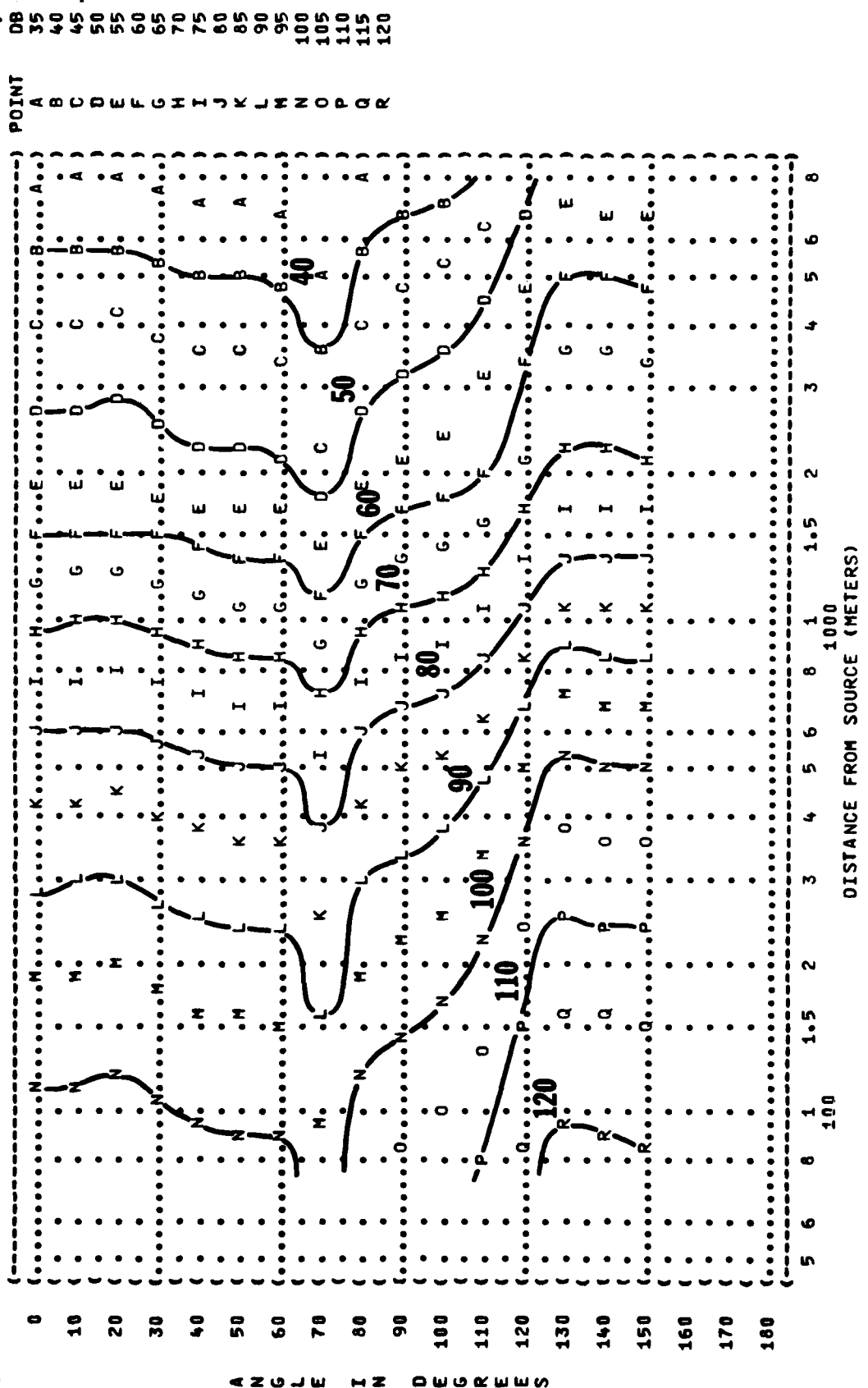
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) PAGE 20

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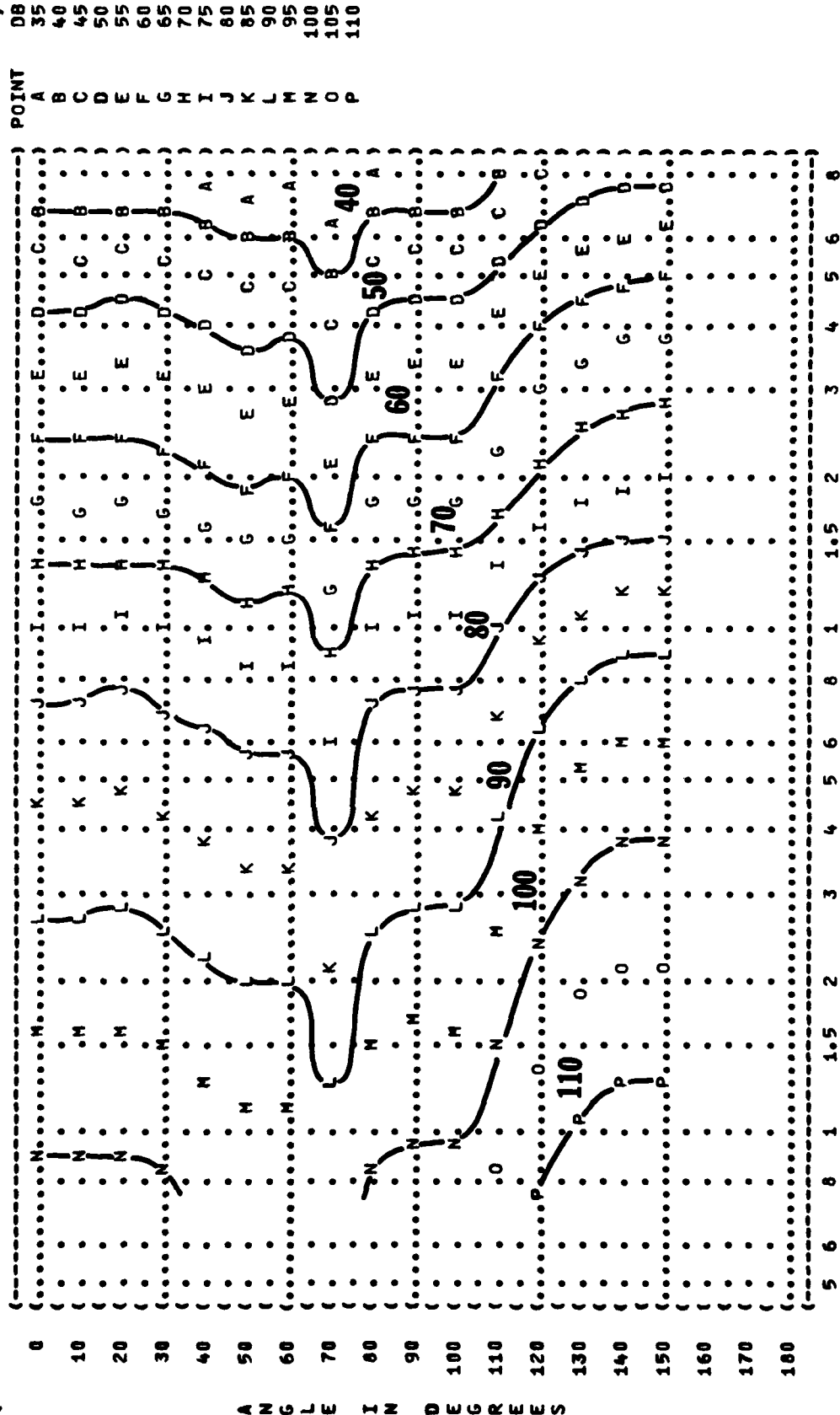
DISTANCE FROM SOURCE (METERS)

(FIGURE: SOUND PRESSURE LEVEL (SPL))
 (11 EQUAL LEVEL CONTOURS (DB))
 (250 HZ OCTAVE BAND)
 (NOISE SOURCE/SUBJECT:)
 ((OPERATION:)
 ((MILITARY POWER)
 ((98% RPM, 1.85 EPR)
 ((ALL ENGINES)
 ((FREE FLOW)
 (C-141A AIRCRAFT)
 (TF33-P-7 ENGINE)
 (FAR FIELD NOISE)
 (METEOROLOGY:)
 (TEMP = 15 C)
 (BAR PRESS = .760 M HG)
 (REL HUMID = 70 %)
 (RUN 03)
 (OMEGA 1.4)
 (TEST 75-002-025)
 (PAGE 21)



A N G L E I N D E G R E E S

(FIGURE: SOUND PRESSURE LEVEL (SPL)
 (11 EQUAL LEVEL CONTOURS (DB)
 (500 HZ OCTAVE BAND
 (NOISE SOURCE/SUBJECT: (OPERATION:
 (C-141A AIRCRAFT (MILITARY POWER
 (TF33-P-7 ENGINE (98% RPM, 1.85 EPR
 (FAR FIELD NOISE (ALL ENGINES
 (FREE FLOW
 (METEOROLOGY:
 (TEMP = 15 C
 (BAR PRESS = .760 M HG
 (REL HUMID = 70 %
 (IDENTIFICATION:
 (OMEGA 1.4
 (TEST 75-002-025
 (RUN 03
 (06 MAY 75
 (PAGE 22



A N G L E I N D E G R E E S

1. PRESSURE LEVEL (SPL)
2. LEVEL CONTOURS (DB)
3. 1/2 OCTAVE BAND

IDENTIFICATIONS
OMEGA 1.4
TEST 75-002-025
RUN 03

OPERATION:

MILITARY POWER

98% RPM, 1.85 EPR

ALL ENGINES

FREE FLOW

METEOROLOGY:

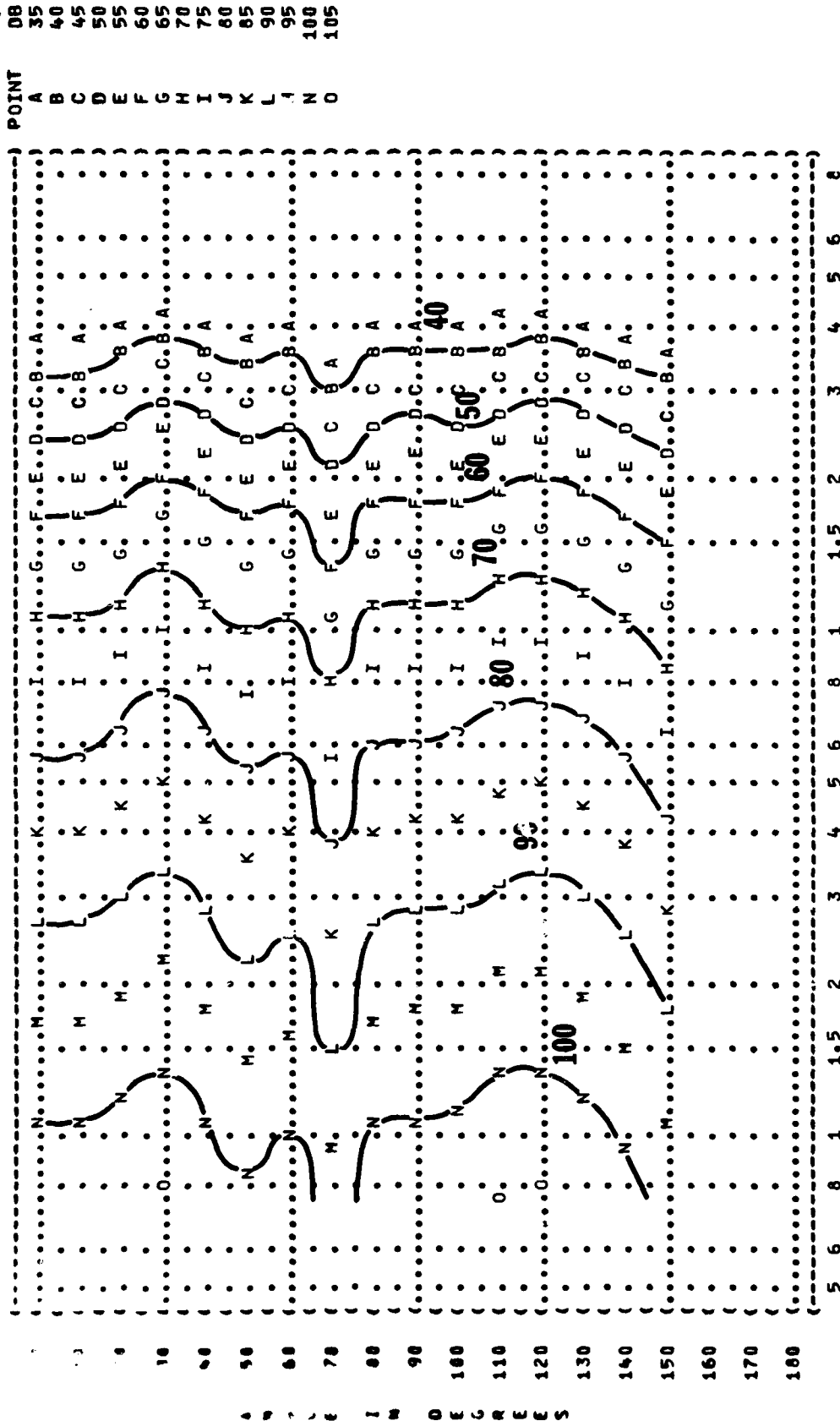
TEMP = 15 C

BAR PRESS = .760 M HG

REL HUMID = 70 %

PAGE 24

AIRCRAFT
ENGINE
FIELD NOISE



DISTANCE FROM SOURCE (METERS)

