

AD-A145 030

GEMSS (GROUND EMPLACED MINE SCATTERING SYSTEM) EXTENDED 171

RANGE TRIPLINE SE. (U) HONEYWELL INC MINNEAPOLIS MN

DEFENSE SYSTEMS DIV M B WEIDENBACH ET AL. JUL 84 47127

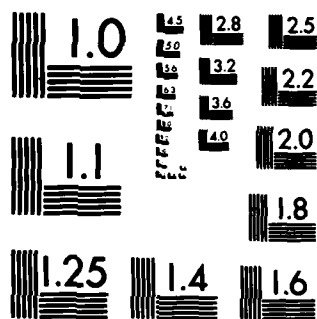
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CONTRACTOR REPORT ARLCD-CR-84020

AD-A145 030

**GEMSS EXTENDED RANGE TRIPLINE SENSOR (ERTS)
PRODUCT IMPROVEMENT PROGRAM (PIP)**

M.B. WEIDENBACH
B.J. AMUNDSON
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HONEYWELL INC.
DEFENSE SYSTEMS DIVISION
5640 SMETANA DRIVE
MINNETONKA, MN 55343

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U.S. ARMY ARMAMENT RESEARCH AND DEVELOPMENT CENTER

LARGE CALIBER WEAPON SYSTEMS LABORATORY

DOVER, NEW JERSEY

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20. ABSTRACT (CONTINUE ON REVERSE SIDE IF NECESSARY AND IDENTIFY BY BLOCK NUMBER) Three changes were incorporated into the GEMSS extended range tripline sensor (ERTS): (1) a stronger tripline, (2) a smaller breakwire, and (3) B. F. Goodrich Estane 58880 as the diaphragm material. A final successful first article acceptance test was run, as well as a successful lot acceptance test for the single delivery lot of 4,500 ERTS units.			

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INTRODUCTION

The extended range tripline sensor is used in the antipersonnel mines of the GEMSS, GATOR and MOPMS systems. The tripline is deployed when the bobbin on which it is wound is spring ejected from the sensor body. The bobbin is mechanically locked to the sensor body by a release mechanism which is designed to be activated by gas pressure above a threshold level. Activation of the release mechanism unlocks the bobbin. A pressure cartridge is used to supply the required threshold level of gas pressure. The deployed tripline is the mine's detection element; however, the sensitivity of this detection is dependent on how well the force of target/tripline interaction is transmitted through the tripline to a 0.0040-inch breakwire and the force needed to break this wire. The mine's electronic assembly monitors the continuity of this wire and initiates the self-destruct mode when wire continuity is lost.

In the subzero climate environments of the GEMSS DTII and OTII tests, the deployment of tripline from sensor was adequate but not 100 percent. Analysis of the hardware and data identified the reasons for this non-deployment as: the rigidity of the sensor's diaphragm and/or the lower output of the pressure cartridge at cold temperatures. The current material used for the diaphragm is Roylar E-82 and its flexibility is highly temperature dependent at cold temperatures; for example, its flexibility at the system's required low temperature operating limit of -35°F is approximately 52 percent of its flexibility at 0°F . Also, pressure cartridge closed bomb tests indicate that the average peak pressure at -35°F is approximately 65 percent of the average peak pressure at the system's high temperature operating limit of 125°F . The need for replacement is urgent because Roylar E-82 is no longer manufactured. It will be replaced by Estane 58880.

The detection sensitivity of the tripline/breakwire interface meets the requirements of the sensor specification; however, tactical field test data indicated that the strength of the tripline should be increased and that of the breakwire decreased. In the ideal detection situation, the minimum (1.8 lb) break strength of the tripline and the maximum (0.9 lb) break strength of the breakwire guarantee a detection if the tension force in the tripline exceeds 0.9 lb. In the tactical situation, it is often the case that between the source of the tension in the tripline and the breakwire, the tripline is in contact with other objects (vegetation, etc.). This means that the tension in the tripline at the source may exceed the minimum 1.8 lb needed to break the tripline before the tension force at the breakwire is sufficient to break the breakwire. An ideal tripline would be one that could not be broken. Another tactical situation that sometimes occurs is that the target detects the tension in

the tripline before it induces a tension of 0.9 lb, the maximum breakwire strength. Detection sensitivity for this last case would be improved if the breakwire had a maximum break strength less than the current 0.9 lb.

One of the principal objectives of this product improvement program was to incorporate the following changes into the extended range tripline (ERTS) technical data package and to demonstrate/characterize each for adequacy:

a. Alternate diaphragm material which is more flexible at cold temperature. (Original material in TDP is also no longer available.)

b. Weaker breakwire for increased sensitivity.

c. Stronger tripline thread to increase probability of breakwire opening before tripline breaks when pulled.

A second objective was to manufacture and deliver 4,500 ERTS with above modifications to Aerojet Corp., who had a parallel PIP contract to include improved sensors plus other modifications in the M74 mine.

The third objective was to provide a quantity of lithium cells to both Aerojet Corp. and Burroughs Corp. as government furnished material (GFM) to be used on PIP contracts for the M74 and M75 mines, respectively. The lithium cell is the power source for the GEMSS M74 and M75 mines and will be required in many of the mines manufactured under the GEMSS PIP program. The technical data package (TDP) for the lithium cell is a proven TDP; therefore, with respect to the lithium cell, Contract DAAK10-83-C-0049 was simply the vehicle to procure lithium cells and deliver them as GFM to the appropriate PIP contractors.

TECHNICAL REPORT

The principal objectives of this contract were to incorporate three changes into the Extended Range Tripline Sensor (ERTS), to test the effectiveness of these changes, and to ship 4500 ERTS to a specified contractor. An additional objective was to ship 2250 standard lithium batteries to the same contractor.

Changes Incorporated

The three changes incorporated into the ERTS in this contract were:

1. A stronger tripline (2.5 lb minimum break strength versus the original 1.8 lb minimum break strength).
2. A smaller breakwire (0.0035-inch diameter versus the original 0.0040 inch diameter).
3. A different diaphragm material (B.F. Goodrich Estane 58880 versus the original Uniroyal E-82).

Data Review

The initial effort on this contract consisted of a review of the technical data (drawings and specification) to determine changes necessary to make them compatible with the three changes specified above. Results of this review were submitted in a letter from J.H. Lundquist, dated 13 April 1983, Subject: Contract DAAK10-83-C-0049, GEMSS ERTS PIP, CRDL Requirements.

Drawing Changes

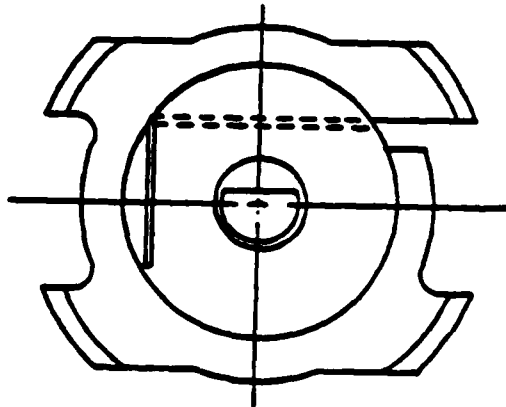
The drawing changes necessary to reflect the three changes and to correct minor errors are presented in Table 1.

All NEXT ASSEMBLY and USED ON boxes should be changed as necessary to assure that the drawings are identified with the proper mine.

Table 1. Drawing change summary

Drawing	Change required	Comments
9298576 (Wire, Magnet, Electrical)	In Note 1B, "#38 AWG" should be "#39 AWG". In Note 1C, "0.004 ± 0.0001 ROUND" should be "0.0035 ± 0.0001 ROUND".	This is an ADAM drawing. If this change is made the drawing cannot be used for ADAM. The drawing can either be changed to a tabulated drawing or a new drawing can be made.
9292972 (Tripline Sensor Extended Range)	In Note 3, "LOCATED 32 FEET" should be "LOCATED 29 FEET" and "0.3 TO 0.9 LB" should be "0.22 TO 0.82 LB".	See paragraphs entitled "Deployment Barrier Distance" and "New Minimum and Maximum Limits on Breakwire Strength" on pages 8 and 9, respectively.
9292991 (Release Mechanism Assembly)	In Zone C5 change "RING, LOCK - 9298586" to "RING, BALL LOCK - 9298586".	
9292998 (Diaphragm Assembly)	In Zone CD-3 change "DIAPHRAGM - 9298598-2" to "DIAPHRAGM - 9298598-1".	
9298592 (Thread, Polyester)	In Note 1A, "234 ± 8 DECITEX" should be "320 ± 8 DECITEX". In Note 1B, "1.8 LB MIN" should be "2.5 LB MIN".	This is an ADAM drawing. If this change is made it cannot be used for ADAM. The drawing can either be changed to a tabulated drawing or a new drawing can be made.
9298598 (Diaphragm)	In the table at the bottom of the drawing add "9292998" in the NEXT ASSEMBLY box for 9298598-1.	
9292982 (Bobbin Assembly)	In Note 2, "46 ± 3 FEET" should be "34 ± 2 FEET". In Note 7, "750 REVOLUTIONS" should be "580 REVOLUTIONS". Delete Note 8. Change left view as shown in Figure 1. Add Note 10: "ADD A DROP OF ADHESIVE AT EXIT POINT OF THREAD JUST PRIOR TO ASSEMBLY OF THE BOBBIN WEIGHT INTO THE BOBBIN".	Permission to incorporate this change was granted by ARRADCOM approval of Deviation 0297-008 (Ref PAN A3N7705).
XXXXXXX (Adhesive)	Make a new drawing like Figure 2 for the adhesive.	

FROM:



TO:

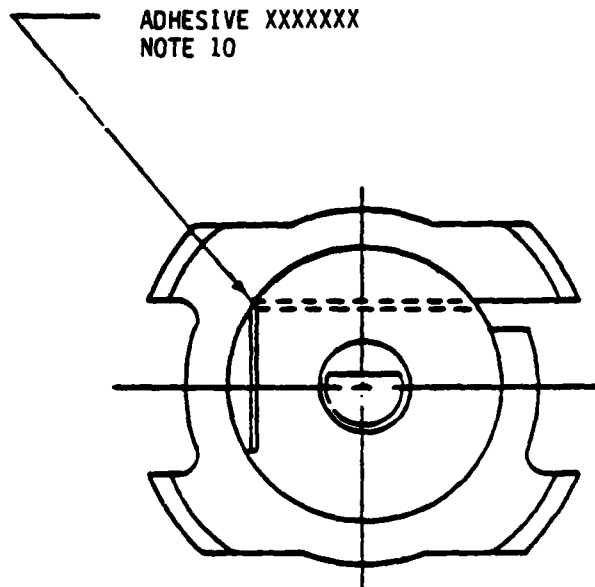


Figure 1. Changes to end view on drawing 9292982

PART NO	28201657-101
---------	--------------

APPROVED SOURCE OF SUPPLY

REV 1
 28201657

CONTROL		FSCM NO		PART NO		NAME AND ADDRESS	
28201657-101	61078	PERMABOND	910	PERMABOND INTERNATIONAL CORP	480 SOUTH DEAN STREET	ENGLEWOOD NEW JERSEY	07631

REV	DESCRIPTION	DATE	APPROVED

NOTES:

1. MATERIAL: PERMABOND 910 ADHESIVE.
2. MUST MEET REQUIREMENTS OF MIL-A-46050, TYPE 1 CLASS 2.
3. IDENTIFICATION OF THE SOURCE OF SUPPLY HEREON IS NOT TO BE CONSIDERED AS A GUARANTEE OF CONTINUED AVAILABILITY.

SOURCE CONTROL DRAWING

PART OR EQUIPMENT NO 	PART OR EQUIPMENT NO 28201657-101	DRAWING OR DETAIL NO 28201657-101	REV 1	DATE 10/1/77	BY J. C. C.	CHECKED J. C. C.	APPROVED J. C. C.
	PURCHASED FROM HONEYWELL						
	ADHESIVE						
	SCALE 1" = 1"						

CONTROL C

Figure 2. Adhesive drawing

During this contract it was apparent that vendors were having difficulty manufacturing the booster spring (9292990). The drawing had ambiguities in it as evidenced by various interpretations of what it meant. In addition, it was obvious that the spring load test fixture was not conducive to providing repeatable load readings. In order to clear up the ambiguities and improve the load test fixture, ECP 83-0007-045 was submitted. It was submitted on ERTS GATOR Contract DAAK10-83-C-0007 running concurrently with the ERTS GEMSS PIP contract which this report covers; however, the changes requested in it apply to the GEMSS PIP sensor because the same booster spring is used in both sensors. The ECP requested the following changes to drawing 9292990:

- a. In Zone C4 change "0.395 + 0.010 SEE NOTE 8" to "0.395 + 0.010 SEE NOTES 8 & 9."
- b. In Zone C4 add "SEE NOTE B" with an arrow pointing to the left end of the spring.
- c. In Zone C3 change "SEE NOTE 5" to "SEE NOTE B."
- d. In Zone C2 change "0.385 + 0.010" to "0.385 MIN."
- e. In Zone D2 change "0.455 MAX DIA, 8 COILS REF SEE NOTE 9" to "0.455 MAX DIA, 9 COILS REF SEE NOTE 9."
- f. In Zone D2 make the extension lines defining the 0.455 MAX DIA refer to the OD of the 9 smaller coils rather than to the OD of the 2 larger coils.
- g. In Note 7 change 0.475 to 0.484 and change 0.340 to 0.384.

The ECP was approved with modifications and the final changes are described in NOR A3N5166. The drawing should be changed per this NOR.

Specification Changes

The changes to Specification MIL-S-48755 (AR) necessary to reflect the three changes and to correct minor errors are:

- a. Delete paragraphs 3.3 and 4.5.1.1 and all further references to these paragraphs.
- b. In paragraph 3.5 change "thirty-two (32) feet" to "twenty-nine (29) feet."
- c. In paragraph 3.6 change "140 grams (0.3 pound) min and 410 grams (0.9 pound) max" to "100 grams (0.22 pound) min and 370 grams (0.82 pound) max."
- d. In paragraph 3.11.a change "Assemble" to "Assembly."
- e. In paragraph 3.11.f change "Ball Lock Ring" to "Ring, Ball Lock."
- f. In paragraph 4.4.2.22 Major 104 change "140 grams (0.3 pound) to 410 grams (0.9 pound)" to "100 grams (0.22 pound) to 370 grams (0.82 pound)." Delete Major 101 and Note 2.

- g. In paragraph 4.5.1.3.a change "32 feet" to "29 feet."
- h. In paragraphs 6.1.a, 6.1.b, and 6.1.c revise to reflect whatever mines use the smaller breakwire, stronger thread and different diaphragm material.

Bobbin Analysis

An analysis of the bobbin assembly was made to determine the amount of stronger thread that can be wound on the bobbin, and to determine the effect of this stronger thread on deployment barrier distance and bobbin assembly CG location. Results of this analysis were submitted to ARRADCOM in a letter from J.H. Lundquist dated 11 May 1983, Subject: Contract DAAK10-83-C-0049, GEMSS ERTS PIP CRDL Requirements.

Thread Length

The stronger thread used on this contract has a larger diameter than the standard thread. Thus, when the standard length (46 ± 3 feet) of the stronger thread is wound on the Bobbin (9292985), the diameter over the wound thread is significantly larger than when the standard thread is used. This results in interference between the thread and the inside of the Sleeve (9292988), causing assembly difficulties when assembling the Bobbin Assembly (9292982) into the Sleeve. Cut and nicked thread and friction between the thread and Sleeve ID results.

In order to eliminate these problems, less of the stronger thread must be wound on the bobbin. It has been experimentally determined that a maximum of 36 feet of the stronger thread should be used to provide proper assembly and release and to maintain thread integrity. Using a realistic overall length tolerance of ± 2 feet results in the recommended thread length of 34 ± 2 feet for the stronger thread.

Deployment Barrier Distance

Because the stronger thread is thicker, less thread can be wound on the bobbin, making the deployment distance less. Deployment test (OEXM 31388) shows that at the standard barrier deployment distance of 32 feet the barrier could not be cleared consistently. When the barrier distance was moved closer at 29 feet, the barrier was cleared consistently. In additional deployment tests on 36 sensors, all cleared the barrier at 29 feet. The three-foot reduction in barrier distance is strictly a function of thread length and has nothing to do with the weight and CG characteristics of the bobbin. The bobbin wound with the stronger thread clears the barrier located at 29 feet just as well as the bobbin wound with the standard thread clears the barrier located at 32 feet.

Center of Gravity

The bobbin analysis shows that the CG shift of the bobbin assembly wound with the stronger thread is 0.002 inch. This is insignificant with respect to sensor deployment characteristics. The weight of the bobbin assembly wound with the stronger thread is 0.0218 gram more than that of the bobbin assembly wound with the standard thread. This weight increase can be eliminated by removing material from the bobbin weight, either by making it shorter or by increasing the diameter of the hole in it. Since the heavier bobbin assembly clears the barrier located at 29 feet just as well as the standard bobbin assembly clears the barrier located at 32 feet, a change to lighten it is unnecessary and is not recommended.

New Minimum and Maximum Limits on Breakwire Strength

The new minimum and maximum breakwire break strength limits were established from the data on 182 ERTS sensors tested on the GATOR PIP program (Contract DAAK10-82-M-0504) and reported in test report OEXM 31388. The former limits of 140 grams minimum and 410 grams maximum resulted in a spread of 270 grams. Based on statistical analysis of the data and past experience on the ADAM program, it is felt that this spread should be maintained. This is also the basis for selecting the new limits of 100 grams (0.22 lb) minimum and 370 grams (0.82 lb) maximum.

QUALITY REPORT

Quality tests conducted for the GEMSS PIP contract were the standard tests: (1) First Article Acceptance Test (FAAT) and, (2) Lot Acceptance Test (LAT). A total of four tests were actually run, consisting of two unsuccessful FAATs, a final successful FAAT, and a single successful LAT for the single delivery lot containing a quantity of 4,500 units. See Appendix A for attachments dealing with quality tests. Attachment 1 is the conditional approval of the FAAT, Attachment 2 is the raw data sheet for this FAAT, Attachments 3 and 4 are the raw data sheets for the first two FAATs and Attachment 5 is the raw data sheet for the LAT.

First Article Acceptance Tests

The first FAAT occurred on 27 July 1983. Failures to release the bobbin (three units) were traced to a machine assembly problem that was damaging the diaphragm. Appropriate corrective action was taken, which eliminated this problem. All existing hardware was scrapped and a new set of units was built.

The second FAAT on 30 August 1983 was unsuccessful due mainly to test errors as well as sample preparation problems. Improper calibration of the pressure system, used in conjunction with the test equipment during cold temperature testing, caused an over-pressurization of the diaphragm assembly. When combined with inadequate sealing, it caused separation from the sensor case, resulting in release failures. Appropriate corrective action was taken and a third FAAT was scheduled.

The successful FAAT of 20 September 1983 had the following results:

- One of 82 units experienced post pullout below the 540-gram requirement (recorded for information only)
- One of 82 units failed breakwire force
- One of 32 units failed cold release (see Attachment 2 for test data).

The breakwire and release failures were caused by potting material intrusion under the post cover during sample preparation. The sample's post covers had not been sealed with epoxy prior to potting. This type of defect would not normally occur on shipped units since all sensors would be properly sealed with epoxy prior to shipment. These units were classed as "no tests" and spares were functioned in their place. All subsequent testing has been performed with epoxy sealed sensors.

Following this FAAT, a conditional approval was given by ARDC (see Appendix A). Honeywell's action for achieving full Governmental approval for the first article test was to ensure that all future LATs would be conducted using sample units sealed with epoxy before potting. The production layout was changed to formalize this action. The corrective action was completed and locally approved by DCAS on 14 October 1983.

Lot Acceptance Test

The LAT was conducted on 16 November 1983. The results of this test were as follows:

- One of 80 units failed breakwire force
- Three of 80 units failed post retention and thread strength (see Attachment 5 for test data).

Analysis of the breakwire failure showed that the breakwire retaining ring was not assembled properly, allowing the post to be pulled out of the sensor without breaking the breakwire. Thus, this unit was also counted in the second category as well, for a total of three defective units (not four). No cause was discerned for the other two units that experienced thread breakage below the 540-gram minimum. No corrective action for the above defects has been taken since additional quantities are not being built. The lot was accepted as tested since the defects identified were below allowable defects per the sample plan and AQL's specified.

SPECIAL TESTS

Leakage Test

The leakage test was conducted to determine: (1) the leak rate through undeployed sensors at a sensor pressurization of 75 ± 5 psi, and (2) the leak rate through deployed sensors at their deployment pressure or, if the sensors did not deploy before being pressurized to 300 ± 10 psi, their leak rate at 300 ± 10 psi. All tests were conducted at ambient temperature ($75^\circ \pm 10^\circ\text{F}$) and nitrogen gas was used as the pressure medium. Results are contained in test report OEXM 32211 (Appendix B). This test report was also submitted with a letter from J.H. Lundquist dated 6 January 1984, Subject: Contract DAAK10-83-C-0049 GEMSS ERTS PIP Test Report, Langlie Test.

Langlie Test

The Langlie test was conducted to determine sensor tripline deployment at three different square pressure pulse time durations at various pulse pressure levels. Results are also contained in test report OEXM 32211 (Appendix B).

APPENDIX A
FIRST ARTICLE TEST SUMMARY REPORT AND
LOT ACCEPTANCE TEST RESULTS

ATTACHMENT 1

1. First Article Test Summary Report		2. Date		3. Report No. 463-33	
US Army Armament Research and Development Center Dover, NJ 07801		4. In reply refer to DRSNC-OAM-VF(D)/92		5. Preliminary ☐ Supplemental ☐ Final ☒	
6. To Commander DCASPRO Honeywell 2701 Fourth Ave. So. Honeywell Plaza Minneapolis, MN 55408		7. Date initial production sample received at <u>New Brighton, NY</u> (insert location) Date <u>20 Sept. 1983</u>			
		8. Contract No. DAAG10-83-C-0049			
		9. Contractor Honeywell, Inc.			
10. Complete sample ☒ Partial sample ☐ 3rd Submission		11. Item nomenclature <u>Sensor Tripline, Extended Range.</u>			
12. Quantity <u>As per Contract</u>		13. Inspection in accordance with <u>MIL-S-48755(AR) w/amend. 5 dtd 5 May 82.</u>			
14. Name(s)		15. Representing			
16. Inspected by <u>R. Flanagan</u>		17. Submitted by <u>A. Siegfried/rfh/201-724-2458</u>			
18. Recommendation of Contracting Officer ☐ The contractor may proceed with production. ☒ The contractor may proceed with production provided the deviations cited under Remark of block 28 are corrected. ☐ Corrective action cited on block 28 is required ☒ Conditionally Approved.					
19. Distribution Cy Turn: Honeywell, Inc. R. Currie MN92-3553 B. Amundson MN29-3681 P. Rahvaldt MN29-3681 M. Weidenbach MN29-3680 J. Lundquist MN04-1200 C. Files MN04-1280 E. Ludke MN29-3561		20. RICHARD W. PORTER C. Mines & Sel Arm Br TISD, PAD <u>✓</u> CONCURS Signed: Gael C. Baker W. P. BELET GAEL C. BAKER Contracting Officer ActC, Product Verification Branch			

ARRADCOM FORM 51 JUN 78 replaces SHU 1031-2 AUG 66
which is obsolete

SHEET 1 OF 3

21. Item/Port Nomenclature	22. Part No.	23. Serial No.	24. Specification No.	25. Specification Paragraph No.	26. Identification	27. Test	
						Pass	Fail
Tripline, Sensor Extended Range 82 each	9292972 8-19-82	N/A	MIL-S-48755(AR) w/Amend. 5 dtd 5 May 82	4.5.1.2.1	Release of Sensor Assy. at Ambient Temperature	X	
				4.5.1.2.2	Release of Sensor Assy. at Cold Temperature		X
				4.5.1.3	Tripline Deploy- ment	X	
				4.5.1.4	Breakwire Function Post		X
				4.5.1.5	Retention & Thread Breaking Strength		X
				4.4.2.22	Examination	X	

ARRADCOM FORM 51a JUN 78 replaces SBU 1031-R AUG 66 which is obsolete

SHEET 2 OF 1

28. Summary of failures encountered and required corrective action.

Tripline Sensor, Extended Range, Dwg. 9292972 -

1. One unit failed to release at cold temperature. Major Defect
2. One unit failed breakwire functioning at cold temperature.
Breakwire broke at 800 grams in lieu of 100 to 378 grams. Major Defect
3. One unit failed post retention at cold temperature. Post released at 280 grams, min. requirement is 540 grams. Test performed for informational purposes only.

ACTION TO BE TAKEN:

The contractor may proceed with production provided the defects noted above in paragraphs 1 and 2 are corrected to the satisfaction of the QAR.

ATTACHMENT 2

ERTS
 SENSOR ASSEMBLY
 8292972
 LAT RESULTS SUMMARY
 DATE 9-20-83
 LOT # PIP (GENIS)
 LOT SIZE
 W. AMUNDSON MN29-3681
 G. HANSON MN29-3680
 R. JOHNSON MN29-3680
 W. McCABE MN29-3300
 J. SHEEHAN MN29-3606
 D. STACHOWSKI MN29-3680

MIL-S-48755 PARA. 4.4.2.22

- 102 AMBIENT RELEASE
- 102 COLD RELEASE
- 103 AMBIENT DEPLOYMENT
- 103 COLD DEPLOYMENT
- 104 BREAKURE
- 105 POST RETENTION AND THREAD STRENGTH

RESULTS	REQUIREMENT	REMARKS
50	50	BLU-02/B ONLY
32/0 *	32	
50	50	
32/0	32	
82/0 **	82	
82/1	82	INFORMATION ONLY

NOTES :

** H, 8 COLD B/W FAILURE DUE TO POTTING MAT'L UNDER FISHING
 * H, 9 COLD FTR DUE TO POTTING MAT'L IN UNIT BECAUSE FISHING MISSING

RELEASE	DEPLOYMENT	GRAND	100 TO 370	WEIGHING	540 GRAMS KCH.	RETENTION	170. BREAK	POST PULLOUT
1	✓	✓	✓	200	✓	1470	✓	✓
2	✓	✓	✓	140	✓	1480	✓	✓
3	✓	✓	✓	270	✓	1390	✓	✓
4	✓	✓	✓	170	✓	1270	✓	✓
5	✓	✓	✓	160	✓	1300	✓	✓
6	✓	✓	✓	290	✓	1520	✓	✓
7	✓	✓	✓	280	✓	1190	✓	✓
8	✓	✓	✓	200	✓	1450	✓	✓
9	✓	✓	✓	150	✓	1480	✓	✓
10	✓	✓	✓	220	✓	1440	✓	✓
11	✓	✓	✓	140	✓	1350	✓	✓
12	✓	✓	✓	140	✓	1500	✓	✓
13	✓	✓	✓	230	✓	1430	✓	✓
14	✓	✓	✓	160	✓	1380	✓	✓
15	✓	✓	✓	240	✓	1410	✓	✓
16	✓	✓	✓	230	✓	1370	✓	✓
17	✓	✓	✓	130	✓	990	✓	✓
18	✓	✓	✓	220	✓	1380	✓	✓
19	✓	✓	✓	210	✓	1050	✓	✓
20	✓	✓	✓	230	✓	1470	✓	✓
21	✓	✓	✓	190	✓	1410	✓	✓
22	✓	✓	✓	230	✓	1470	✓	✓
23	✓	✓	✓	350	✓	1520	✓	✓
24	✓	✓	✓	210	✓	1270	✓	✓
25	✓	✓	✓	210	✓	1480	✓	✓
26	✓	✓	✓	240	✓	1370	✓	✓
27	✓	✓	✓	230	✓	1450	✓	✓
28	✓	✓	✓	370	✓	1290	✓	✓
29	✓	✓	✓	240	✓	1190	✓	✓
30	✓	✓	✓	190	✓	1400	✓	✓
31	✓	✓	✓	160	✓	1370	✓	✓
32	✓	✓	✓	220	✓	1450	✓	✓
33	✓	✓	✓	180	✓	1250	✓	✓
34	✓	✓	✓	160	✓	1110	✓	✓
35	✓	✓	✓	250	✓	1310	✓	✓
36	✓	✓	✓	200	✓	990	✓	✓
37	✓	✓	✓	180	✓	1550	✓	✓
38	✓	✓	✓	200	✓	1510	✓	✓
39	✓	✓	✓	200	✓	1310	✓	✓
40	✓	✓	✓	190	✓	1500	✓	✓
41	✓	✓	✓	190	✓	1480	✓	✓
42	✓	✓	✓	240	✓	1470	✓	✓
43	✓	✓	✓	210	✓	1210	✓	✓
44	✓	✓	✓	180	✓	1450	✓	✓
45	✓	✓	✓	160	✓	1350	✓	✓
46	✓	✓	✓	220	✓	1110	✓	✓
47	✓	✓	✓	170	✓	1330	✓	✓
48	✓	✓	✓	250	✓	1370	✓	✓
49	✓	✓	✓	200	✓	1440	✓	✓
50	✓	✓	✓	180	✓	1560	✓	✓

* NOTE: NO MORE THAN ONE OF THE FIVE ALLOWABLE
DEFECTIVES SHALL EXCEED 544 GRAMS.

ATTACHMENT 3

ERTS
 SENSOR ASSEMBLY
 LAT RESULTS SUMMARY
 DATE 7/27/83
 LOT # 5000 3-1 FAAT
 LOT SIZE

B. AMUNDSON HN20-3000
 G. HANSON HN20-3000
 R. JOHNSON HN20-3000
 W. MCABE HN20-3000
 J. SHEEHAN HN20-3000
 D. STACHOWSKI HN20-3000

Genss ERTS PIP DAAH10-83-C-0049

MIL-S-48755 PARA. 4.4.2.22

- 102 AMBIENT RELEASE
- 102 COLD RELEASE
- 103 AMBIENT DEPLOYMENT
- 103 COLD DEPLOYMENT
- 104 BREAKTIRE
- 105 POST RETENTION AND THREAD STRENGTH

RESULTS	REQUIREMENT	REMARKS
1/50	50-1-2	
2/32	32-1-2	
1/50	50-3-4	
0/32	32-2-3	
0/82	82-5-6	
0/82	82	INFORMATION ONLY

NOTES :

RELEASE	DEPLOYMENT	GRAMS	100 TO 270	DEFECTIVES	540 GRAMS K21	INTENTION	110. BREAK	POST FALLOUT
1	✓	✓	260	✓	1250	✓		
2	✓	✓	240	✓	1380	✓		
3	✓	✓	290	✓	1300	✓		
4	✓	✓	220	✓	1340	✓		
5	✓	✓	330	✓	1330	✓		
6	✓	✓	260	✓	1350	✓		
7	✓	✓	280	✓	1350	✓		
8	✓	✓	240	✓	1450	✓		
9	✓	✓	250	✓	960	✓		
10	✓	✓	260	✓	1350	✓		
11	✓	✓	250	✓	1230	✓		
12	✓	✓	240	✓	1410	✓		
13	✓	✓	220	✓	1280	✓		
14	✓	✓	210	✓	1440	✓		
15	✓	✓	230	✓	1440	✓		
16	✓	✓	220	✓	1340	✓		
17	✓	✓	260	✓	1320	✓		
18	✓	✓	230	✓	1300	✓		
19	✓	✓	240	✓	1380	✓		
20	✓	✓	190	✓	1440	✓		
21	✓	✓	270	✓	1280	✓		
22	✓	✓	250	✓	1240	✓		
23	✓	✓	250	✓	1410	✓		
24	✓	✓	260	✓	1370	✓		
25	✓	✓	250	✓	1370	✓		
26	✓	✓	280	✓	1450	✓		
27	✓	✓	290	✓	1340	✓		
28	✓	✓	270	✓	1450	✓		
29	✓	✓	270	✓	1290	✓		
30	✓	✓	230	✓	1050	✓		
31	✓	✓	210	✓	1300	✓		
32	✓	✓	270	✓	1390	✓		
33	✓	✓	250	✓	1400	✓		
34	✓	✓	220	✓	1260	✓		
35	✓	✓	210	✓	1430	✓		
36	✓	✓	230	✓	1290	✓		
37	✓	✓	280	✓	1390	✓		
38	✓	✓	230	✓	1310	✓		
39	✓	✓	230	✓	1370	✓		
40	✓	✓	260	✓	1190	✓		
41	✓	✓	300	✓	1230	✓		
42	✓	✓	290	✓	980	✓		
43	✓	✓	270	✓	1340	✓		
44	✓	✓	330	✓	1310	✓		
45	✓	✓	260	✓	1320	✓		
46	✓	✓	230	✓	1080	✓		
47	✓	✓	270	✓	1450	✓		
48	✓	✓	220	✓	1490	✓		
49	✓	✓	240	✓	1250	✓		
50	✓	✓	280	✓	1300	✓		
7	✓	✓	260	✓	1450	✓		
26	✓	✓	280	✓	1450	✓		

* NOTE: NO MORE THAN ONE OF THE FIVE ALLOWABLE DEFECTIVES SHALL EXCEED 544 GRAMS.

ATTACHMENT 4

ERTS
 SENSOR ASSEMBLY
 829-3680
 LAT RESULTS SUMMARY
 DATE 8-30-83
 LOT # FART 42
 LOT SIZE 82

B. AMUNDSON MN29-3681
 G. HANSON MN29-3680
 R. JOHNSON MN29-3680
 W. McCABE MN29-3300
 J. SHEEHAN MN29-3690
 D. STACHOWSKI MN29-3680

MIL-S-48755 PARA. 4.4.2.22

- 102 AMBIENT RELEASE
- 102 COLD RELEASE
- 103 AMBIENT DEPLOYMENT
- 103 COLD DEPLOYMENT
- 104 BREAKWIRE
- 105 POST RETENTION AND THREAD STRENGTH

RESULTS	REQUIREMENT	REMARKS
0/50	50	BLU-92/B ONLY
0/32	32	
1/50	50	
0/32	32	
9/82	82	
2/82	82	INFORMATION ONLY

NOTES :

Nine "Blow Aparts" occurred during testing. These are not to be counted as
 any functional failures and splices were tested to finish the test.

ALLCARE	DEPLOYMENT	GRAMS	100 TO 370	MEASUREMENT*	S40 GRAMS MIN.	RETENTION	170. BREAK	POST PULLOUT
1	✓	✓	✓	240	1460	✓	✓	✓
2	✓	✓	✓	190	1540	✓	✓	✓
3	✓	✓	✓	170	1460	✓	✓	✓
4	✓	✓	✓	160	1450	✓	✓	✓
5	✓	✓	✓	210	1280	✓	✓	✓
6	✓	✓	✓	170	1380	✓	✓	✓
7	✓	✓	✓	190	1240	✓	✓	✓
8	✓	✓	✓	210	1470	✓	✓	✓
9	✓	✓	✓	240	1440	✓	✓	✓
10	✓	✓	✓	180	1450	✓	✓	✓
11	✓	✓	✓	220	1510	✓	✓	✓
12	✓	✓	✓	270	1400	✓	✓	✓
13	✓	✓	✓	210	1470	✓	✓	✓
14	✓	✓	✓	180	1420	✓	✓	✓
15	✓	✓	✓	260	1280	✓	✓	✓
16	✓	✓	✓	180	1460	✓	✓	✓
17	✓	✓	✓	210	1440	✓	✓	✓
18	✓	✓	✓	170	1580	✓	✓	✓
19	✓	✓	✓	170	1400	✓	✓	✓
20	✓	✓	✓	NONE	380	✓	✓	✓
21	✓	✓	✓	160	1390	✓	✓	✓
22	✓	✓	✓	190	1360	✓	✓	✓
23	✓	✓	✓	170	1360	✓	✓	✓
24	✓	✓	✓	190	1470	✓	✓	✓
25	✓	✓	✓	160	1420	✓	✓	✓
26	✓	✓	✓	190	1440	✓	✓	✓
27	✓	✓	✓	120	1500	✓	✓	✓
28	✓	✓	✓	210	1450	✓	✓	✓
29	✓	✓	✓	220	1410	✓	✓	✓
30	✓	✓	✓	160	1400	✓	✓	✓
31	✓	✓	✓	380	1570	✓	✓	✓
32	✓	✓	✓	180	1400	✓	✓	✓
33	✓	✓	✓	170	1230	✓	✓	✓
34	✓	✓	✓	150	1400	✓	✓	✓
35	✓	✓	✓	120	1510	✓	✓	✓
36	✓	✓	✓	210	1460	✓	✓	✓
37	✓	✓	✓	360	1310	✓	✓	✓
38	✓	✓	✓	130	1340	✓	✓	✓
39	✓	✓	✓	240	1490	✓	✓	✓
40	✓	✓	✓	220	1210	✓	✓	✓
41	✓	✓	✓	640	1180	✓	✓	✓
42	✓	✓	✓	250	1230	✓	✓	✓
43	✓	✓	✓	140	1440	✓	✓	✓
44	✓	✓	✓	170	1170	✓	✓	✓
45	✓	✓	✓	160	1350	✓	✓	✓
46	✓	✓	✓	210	1110	✓	✓	✓
47	✓	✓	✓	180	1580	✓	✓	✓
48	✓	✓	✓	550	1450	✓	✓	✓
49	✓	✓	✓	180	1440	✓	✓	✓
50	✓	✓	✓	210	1410	✓	✓	✓
51	✓	✓	✓	210	1180	✓	✓	✓

* NOTE: NO MORE THAN ONE OF THE FIVE ALLOWABLE DEFECTIVES SHALL EXCEED S44 GRAMS.

RELEASE	DEPLOYMENT	GRAMS	100 TO 370	BREAKAGES	540 GRAMS MIN.	RETENTION	170 - BREAK	POST PULLOUT
BA 1	✓	✓	✓	NONE OR	1450	✓	✓	✓
BA 2	✓	✓	✓	NONE OR	1100	✓	✓	✓
3	✓	✓	✓	220	1340	✓	✓	✓
4	✓	✓	✓	220	1440	✓	✓	✓
5	✓	✓	✓	180	1310	✓	✓	✓
BA 6	✓	✓	✓	NONE OR	NONE	✓	✓	✓
7	✓	✓	✓	430	430	✓	✓	✓
8	✓	✓	✓	NONE	1370	✓	✓	✓
9	✓	✓	✓	220	1380	✓	✓	✓
10	✓	✓	✓	180	1390	✓	✓	✓
11	✓	✓	✓	220	1460	✓	✓	✓
BA 12	✓	✓	✓	NONE OR	510	✓	✓	✓
13	✓	✓	✓	880	1390	✓	✓	✓
14	✓	✓	✓	220	1380	✓	✓	✓
15	✓	✓	✓	220	1220	✓	✓	✓
SPARKS								
1	✓	✓	✓	280	1370	✓	✓	✓
2	✓	✓	✓	220	1450	✓	✓	✓
3	✓	✓	✓	240	1340	✓	✓	✓
4	✓	✓	✓	240	1350	✓	✓	✓
5	✓	✓	✓	150	1360	✓	✓	✓
6	✓	✓	✓	170	1300	✓	✓	✓
7	✓	✓	✓	220	1240	✓	✓	✓
8	✓	✓	✓	290	1400	✓	✓	✓
9	✓	✓	✓	180	1370	✓	✓	✓
16	✓	✓	✓	240	1420	✓	✓	✓
BA 17	✓	✓	✓	NONE OR	880	✓	✓	✓
18	✓	✓	✓	220	1280	✓	✓	✓
19	✓	✓	✓	190	1230	✓	✓	✓
20	✓	✓	✓	1430	1430	✓	✓	✓
21	✓	✓	✓	230	1400	✓	✓	✓
22	✓	✓	✓	220	1300	✓	✓	✓
23	✓	✓	✓	120	1310	✓	✓	✓
BA 24	✓	✓	✓	NONE OR	NONE	✓	✓	✓
25	✓	✓	✓	190	1250	✓	✓	✓
26	✓	✓	✓	210	1360	✓	✓	✓
27	✓	✓	✓	170	1370	✓	✓	✓
28	✓	✓	✓	220	1200	✓	✓	✓
29	✓	✓	✓	220	950	✓	✓	✓
BA 30	✓	✓	✓	NONE OR	NONE	✓	✓	✓
BA 31	✓	✓	✓	NONE OR	NONE	✓	✓	✓
BA 32	✓	✓	✓	NONE OR	NONE	✓	✓	✓

* NOTE: NO MORE THAN ONE OF THE FIVE ALLOWABLE
DEFECTIVES SHALL EXCEED 544 GRAMS.
BA = BLOW AWAY
OR = ON RELEASE

ATTACHMENT 5

ERIS
 SENSOR ASSEMBLY
 809297
 LAT RESULTS SUMMARY
 DATE 11-16-83
 LOT # 3-1
 LOT SIZE 4621

B. AMUNDSON MN29-3681
 G. HANSON MN29-3680
 R. JOHNSON MN29-3680
 W. McCABE MN29-3300
 J. SHEEHAN MN29-3600
 D. STACHOWSKI MN29-3680

NIL-S-48755 PARA. 4.4.2.22

- 101 NON-RELEASE
- 102 AMBIENT RELEASE
- 102 COLD RELEASE
- 103 AMBIENT DEPLOYMENT
- 103 COLD DEPLOYMENT
- 104 BREAKURE
- 105 POST RETENTION AND THREAD STRENGTH

RESULTS	REQUIREMENT	REMARKS
N/A	80-2-3	BLU-82/B ONLY
0/50	50-1-2	
0/50	30-1-2	
0/50	50-3-4	
0/50	30-2-3	
1/80	80-5-8	
3/80	80	INFORMATION ONLY

NOTES :

11-21-83
 11-21-83
 11-21-83
 11-21-83
 11-22-83

RELEASE	DEPLOYMENT	GRAMS	100 TO 370	MEASUREMENTS	540 GRAMS NOM.	RETENTION	110. BREAK	POST FALLOUT
1	✓	✓	✓	190	✓	1190	✓	✓
2	✓	✓	✓	210	✓	1200	✓	✓
3	✓	✓	✓	210	✓	1330	✓	✓
4	✓	✓	✓	180	✓	1380	✓	✓
5	✓	✓	✓	160	✓	1330	✓	✓
6	✓	✓	✓	130	✓	1310	✓	✓
7	✓	✓	✓	230	✓	1210	✓	✓
8	✓	✓	✓	170	✓	1460	✓	✓
9	✓	✓	✓	220	✓	910	✓	✓
10	✓	✓	✓	200	✓	1260	✓	✓
11	✓	✓	✓	220	✓	1560	✓	✓
12	✓	✓	✓	170	✓	1510	✓	✓
13	✓	✓	✓	180	✓	1450	✓	✓
14	✓	✓	✓	170	✓	1480	✓	✓
15	✓	✓	✓	210	✓	1230	✓	✓
16	✓	✓	✓	150	✓	1390	✓	✓
17	✓	✓	✓	240	✓	1440	✓	✓
18	✓	✓	✓	200	✓	1000	✓	✓
19	✓	✓	✓	200	✓	880	✓	✓
20	✓	✓	✓	210	✓	1370	✓	✓
21	✓	✓	✓	250	✓	1250	✓	✓
22	✓	✓	✓	160	✓	1180	✓	✓
23	✓	✓	✓	260	✓	1330	✓	✓
24	✓	✓	✓	170	✓	1250	✓	✓
25	✓	✓	✓	230	✓	1420	✓	✓
26	✓	✓	✓	290	✓	1180	✓	✓
27	✓	✓	✓	160	✓	1370	✓	✓
28	✓	✓	✓	220	✓	1410	✓	✓
29	✓	✓	✓	210	✓	1280	✓	✓
30	✓	✓	✓	150	✓	1450	✓	✓
31	✓	✓	✓	230	✓	1250	✓	✓
32	✓	✓	✓	240	✓	1330	✓	✓
33	✓	✓	✓	280	✓	370	✓	✓
34	✓	✓	✓	280	✓	1390	✓	✓
35	✓	✓	✓	190	✓	1500	✓	✓
36	✓	✓	✓	120	✓	1580	✓	✓
37	✓	✓	✓	270	✓	1170	✓	✓
38	✓	✓	✓	140	✓	1480	✓	✓
39	✓	✓	✓	230	✓	1380	✓	✓
40	✓	✓	✓	210	✓	370	✓	✓
41	✓	✓	✓	260	✓	1470	✓	✓
42	✓	✓	✓	110	✓	1370	✓	✓
43	✓	✓	✓	210	✓	1380	✓	✓
44	✓	✓	✓	230	✓	1470	✓	✓
45	✓	✓	✓	120	✓	1430	✓	✓
46	✓	✓	✓	220	✓	670	✓	✓
47	✓	✓	✓	190	✓	920	✓	✓
48	✓	✓	✓	200	✓	1420	✓	✓
49	✓	✓	✓	240	✓	1310	✓	✓
50	✓	✓	✓	220	✓	1180	✓	✓

* NOTE: NO MORE THAN ONE OF THE FIVE ALLOWABLE DEFECTIVES SHALL EXCEED 544 GRAMS.

RELEASE	DEPLOYMENT	GRADE	100 TO 375	MEASUREMENTS	5-6 GRADE NDA	RETENTION	100 - BREAK	POST FALLOUT
1	✓	✓	✓	260	1510	✓		
2	✓	✓	✓	300	1310	✓		
3	✓	✓	✓	240	1170	✓		
4	✓	✓	✓	240	1440	✓		
5	✓	✓	✓	230	1410	✓		
6	✓	✓	✓	210	1460	✓		
7	✓	✓	✓	260	1270	✓		
8	✓	✓	✓	270	1480	✓		
9	✓	✓	✓	210	1400	✓		
10	✓	✓	✓	210	1530	✓		
11	✓	✓	✓	320	1400	✓		
12	✓	✓	✓	220	1130	✓		
13	✓	✓	✓	290	1380	✓		
14	✓	✓	✓	230	1450	✓		
15	✓	✓	✓	260	1440	✓		

RELEASE	DEPLOYMENT	GRADE	100 TO 375	MEASUREMENTS	5-6 GRADE NDA	RETENTION	100 - BREAK	POST FALLOUT
16	✓	✓	✓	260	1210	✓		
17	✓	✓	✓	190	1390	✓		
18	✓	✓	✓	190	1360	✓		
19	✓	✓	✓	180	1450	✓		
20	✓	✓	✓	230	1440	✓		
21	✓	✓	✓	POST PULLED ON RELEASE		✓		
22	✓	✓	✓	170	1440	✓		
23	✓	✓	✓	180	1220	✓		
24	✓	✓	✓	190	1330	✓		
25	✓	✓	✓	260	1410	✓		
26	✓	✓	✓	160	1340	✓		
27	✓	✓	✓	260	1250	✓		
28	✓	✓	✓	300	1070	✓		
29	✓	✓	✓	270	1212	✓		
30	✓	✓	✓	260	1400	✓		
31	✓	✓	✓	170	1490	✓		

* NOTE: NO MORE THAN ONE OF THE FIVE ALLOWABLE DEFECTIVES SHALL EXCEED 544 GRAMS.

APPENDIX B
ENGINEERING TEST REPORT OEXM 32211

Honeywell**ENGINEERING TEST REPORT**

COPYLIST:

B. Amundson MN29-3681
R. Currie MN29-3553
J. Funk MN29-3682
J. Haley MN29-3553
J. Lundquist(3) MN04-1200
T. Martorano MN29-3682
D. Stachowski MN29-3680
D. Swanson MN11-1430
M. Weidenbach MN29-3680
L. Wilder MN11-1430
D & E File MN11-1430
Uniterm File MN11-1430

KEYWORDS:

GEMSS, XM74
Sensor, Extended
Leakage Rate

ATTACHMENTS:

I Leakage Test Plan (2)
II Langlie Test Plan (2)

☐ AVIONICS	DATE 12-29-83	REPORT NUMBER OEXM 32211
☒ DEFENSE SYSTEMS	DEVELOPMENT NUMBER S4594-AA-6000-2759	PAGE 1 OF 7
ISSUED BY NB103 - D & E Lab		CONTRACT NUMBER DAAK10-83-C-0049

UNITS TESTED:

One hundred thirty-five Extended Range Tripline Sensors (9292972 Modified). Modifications consist of the 9298598-1 diaphragm, 0.0035 $\pm$.0001 diameter breakwire and 320 $\pm$ 8 Decitrex Thread.

OBJECT OF TEST:

Conduct Leakage Rate and Langlie One-Shot Release Tests as outlined in Attachments I and II.

DOCUMENTATION:

See attached data sheets.

PROCEDURE AND RESULTS:Leakage Rate:

Each unit was tested as shown on the attached Test Plan except the Nicolet Oscilloscope was set on a slow trace and started before applying pressure to the sensor. The starting pressure (supply pressure cut-off) and the fixture pressure at 30 second intervals as shown in the data sheets were read from the scope trace.

Langlie Test:

Each test was conducted as shown on the attached Test Plan. Initial "no release" and "all release" limits were set at 50 psi and 350 psi respectively.

DATA BOOK NUMBER 0-2431	PAGE 112-117	TEST STARTED 12-5-83	TEST COMPLETED 12-16-83
REQUESTED BY M. Weidenbach	DATE 12-1-83	WRITTEN BY J. Funk/T. Martorano	
DEPARTMENT Production Engineering		APPROVED BY L. D. Wilder	

HE-44B REV 12/78

Change Pressure (Psi)

	Initial	300m	600m	900m	1200m	1500m	1800m	2100m	2400m	2700m	3000m	3300m	3600m	3900m
4.1	74.6	71.3	62.7	55.4	47.6	35.6	42.1	37.6	32.8	36.5	35.7	35.0	34.5	34.4
1	74.6	71.3	64.8	64.8	61.3	58.2	56.1	52.7	51.2	49.7	48.3	47.1	46.0	45.5
2	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
3	75.0	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
4	75.0	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
5	75.0	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
6	75.0	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
7	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
8	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
9	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
10	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
11	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
12	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
13	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
14	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
15	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
16	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
17	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
18	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
19	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
20	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
21	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
22	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
23	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
24	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
25	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
26	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
27	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
28	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
29	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5
30	75.1	71.3	63.3	63.0	58.1	53.7	50.0	47.7	45.2	43.1	42.0	40.6	39.3	38.5

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Release Pressure Temperature Test

SHEET 5

No.	Time Release	Change Pressure (Psi)												300m	350m	400m
		300m	350m	400m	450m	500m	550m	600m	650m	700m	750m	800m	850m			
1	1840	73.5	57.0	31.0	23.0	12.5	0	—	—	—	—	—	—	—	—	—
2	1840	102.0	128.0	153.0	172.5	136.5	131.5	125.5	121.5	117.0	112.0	107.0	102.0	97.0	92.0	87.0
3	1845	115.5	142.5	169.0	188.0	154.0	149.0	144.0	139.0	134.0	129.0	124.0	119.0	114.0	109.0	104.0
4	1850	108.0	135.0	162.0	181.0	147.0	142.0	137.0	132.0	127.0	122.0	117.0	112.0	107.0	102.0	97.0
5	1910	120.5	147.5	174.5	193.5	159.5	154.5	149.5	144.5	139.5	134.5	129.5	124.5	119.5	114.5	109.5
6	1910	51.0	41.0	31.0	21.0	11.0	0	—	—	—	—	—	—	—	—	—
7	1925	116.0	143.0	170.0	189.0	155.0	150.0	145.0	140.0	135.0	130.0	125.0	120.0	115.0	110.0	105.0
8	1910	96.5	71.0	46.0	31.0	16.0	0	—	—	—	—	—	—	—	—	—
9	1955	74.0	54.5	35.0	25.5	16.0	11.0	6.0	1.0	—	—	—	—	—	—	—
10	1920	99.5	70.0	40.5	21.0	11.5	0	—	—	—	—	—	—	—	—	—
11	1915	102.0	82.5	63.0	43.5	24.0	14.5	9.5	4.5	—	—	—	—	—	—	—
12	1900	64.5	31.0	13.5	0	—	—	—	—	—	—	—	—	—	—	—
13	1925	108.5	81.0	53.5	26.0	13.5	0	—	—	—	—	—	—	—	—	—
14	1945	102.5	84.0	65.5	47.0	28.5	19.0	10.5	1.0	—	—	—	—	—	—	—
15	1915	87.5	54.5	22.0	11.0	0	—	—	—	—	—	—	—	—	—	—
16	1920	102.5	81.0	53.5	26.0	13.5	0	—	—	—	—	—	—	—	—	—
17	1910	88.5	64.0	39.5	25.0	10.5	0	—	—	—	—	—	—	—	—	—
18	1915	104.5	85.5	66.5	47.5	28.5	19.0	10.5	1.0	—	—	—	—	—	—	—
19	1920	105.5	86.5	67.5	48.5	29.5	20.0	11.0	1.5	—	—	—	—	—	—	—
20	1920	104.5	85.5	66.5	47.5	28.5	19.0	10.5	1.0	—	—	—	—	—	—	—
21	1910	76.5	57.5	38.5	19.5	10.5	0	—	—	—	—	—	—	—	—	—
22	1910	92.0	73.5	55.0	36.5	18.0	9.0	0	—	—	—	—	—	—	—	—
23	1945	104.5	85.5	66.5	47.5	28.5	19.0	10.5	1.0	—	—	—	—	—	—	—
24	1925	116.5	94.0	71.5	49.0	26.5	13.5	0	—	—	—	—	—	—	—	—
25	1910	104.0	87.0	69.5	52.0	34.5	17.0	8.0	0	—	—	—	—	—	—	—
26	1925	100.5	83.5	66.5	49.5	32.5	15.5	6.5	—	—	—	—	—	—	—	—
27	1930	75.5	58.0	40.5	23.0	15.5	7.5	—	—	—	—	—	—	—	—	—
28	1915	104.0	87.0	69.5	52.0	34.5	17.0	8.0	0	—	—	—	—	—	—	—
29	1925	72.5	55.5	38.5	21.5	14.5	7.5	—	—	—	—	—	—	—	—	—
30	1940	79.0	60.0	41.0	22.0	13.5	0	—	—	—	—	—	—	—	—	—

SHEET 22

Release Pressure - Average Test

[illegible]

LEAKAGE TEST - BREAK WIRE & TRIP LINE FORCE

Unit	Break Wire (Grams)	Trip Line (Grams)
1	230	920
2	250	740
3	260	1590
4	180	1020
5	240	1040
6	250	1200
7	280	1020
8	200	1400
9	190	1310
10	220	1340
11	240	1130
12	270	1200
13	240	1150
14	210	1330
15	180	1040
16	240	1120
17	250	1260
18	240	1330
19	220	1140
20	230	1440
21	240	1200
22	230	1310
23	240	1230
24	220	1140
25	260	1320
26	240	1120
27	200	1040
28	260	1220
29	230	1250
30	250	1160
31	230	1330
32	260	1410
33	230	1240
34	280	1180
35	220	650

LANGUAGE RELIABILITY TEST

50 ms Pulse				75 ms Pulse				100 ms Pulse			
Unit	Pass (Psi)	Release	Non Release	Unit	Pass (Psi)	Release	Non Release	Unit	Pass (Psi)	Release	Non Release
1	200	X		31	200	X		61	200	X	
2	125	X		32	125	X		62	125	X	
3	87.5		0	33	87.5		0	63	87.5		0
4	106.25	X		34	106.25		0	64	106.25		0
5	96.25		0	35	228	X		65	228	X	
6	101.5		0	36	139	X		66	139	X	
7	225	X		37	94.5		0	67	94.5		0
8	160	X		38	116.5		0	68	116.5	X	
9	130		0	39	123	X		69	103		0
10	145	X		40	132.5		0	70	110		0
11	127.5	X		41	146.5	X		71	149	X	
12	93.75		0	42	143	X		72	144.5	X	
13	116		0	43	140	X		73	127	X	
14	138	X		44	95		0	74	115.5		0
15	127	X		45	117		0	75	122.5	X	
16	121		0	46	128	X		76	120	X	
17	128	X		47	122.5	X		77	117		0
18	124.5	X		48	120	X		78	240	X	
19	87		0	49	118.5		0	79	115	X	
20	105		0	50	119		0	80	225		0
21	116		0	51	130	X		81	121	X	
22	127	X		52	125		0	82	102.5	X	
23	121	X		53	121.5	X		83	115.5		0
24	118		0	54	126.5	X		84	118		0
25	123	X		55	122.5		0	85	121.5	X	
26	121	X		56	124.5		0	86	125		0
27	119.5		0	57	127	X		87	128	X	
28	123		0	58	126	X		88	126.5	X	
29	125.5		0	59	125	X		89	121		0
30	126.2	X		60	52.5		0	90	124	X	

TEST PLAN
GEMMS/ERTS LEAKAGE TEST

1.0 Purpose

The purpose of the test is to determine the rate of pressure leakdown through the sensor.

2.0 Test Procedure**2.1 Initial leakage.**

- 2.1.1 Install sensor in release socket of pressure chamber.
- 2.1.2 Pressurize chamber with nitrogen gas to 75 ± 5 psi.
- 2.1.3 Close shut off valve to chamber and start Nicolet Oscilloscope trace.
- 2.1.4 Record pressure trace on Floppy Disc.

2.2 Function and leakage

- 2.2.1 Attach continuity meter to breakwire leads for continuous monitoring.
- 2.2.2 Increase pressure in the chamber at a steady rate until tripline deploys or to 300 ± 10 psi.
- 2.2.3 Close shutoff valve to chamber and start Nicolet Oscilloscope trace.
- 2.2.4 Record pressure trace on Floppy Disc.
- 2.2.5 Pull trip line along longitudinal axis of sensor and record the force to break the break wire and the force to break the trip line.

2.3 Repeat (2.1) and (2.2) on 35 sensors at ambient temperature ($75 \pm 10^\circ\text{F}$).**3.0 Fixture Design**

The fixture design is shown on the attached drawing.

4.0 Pass/Fail Criteria.

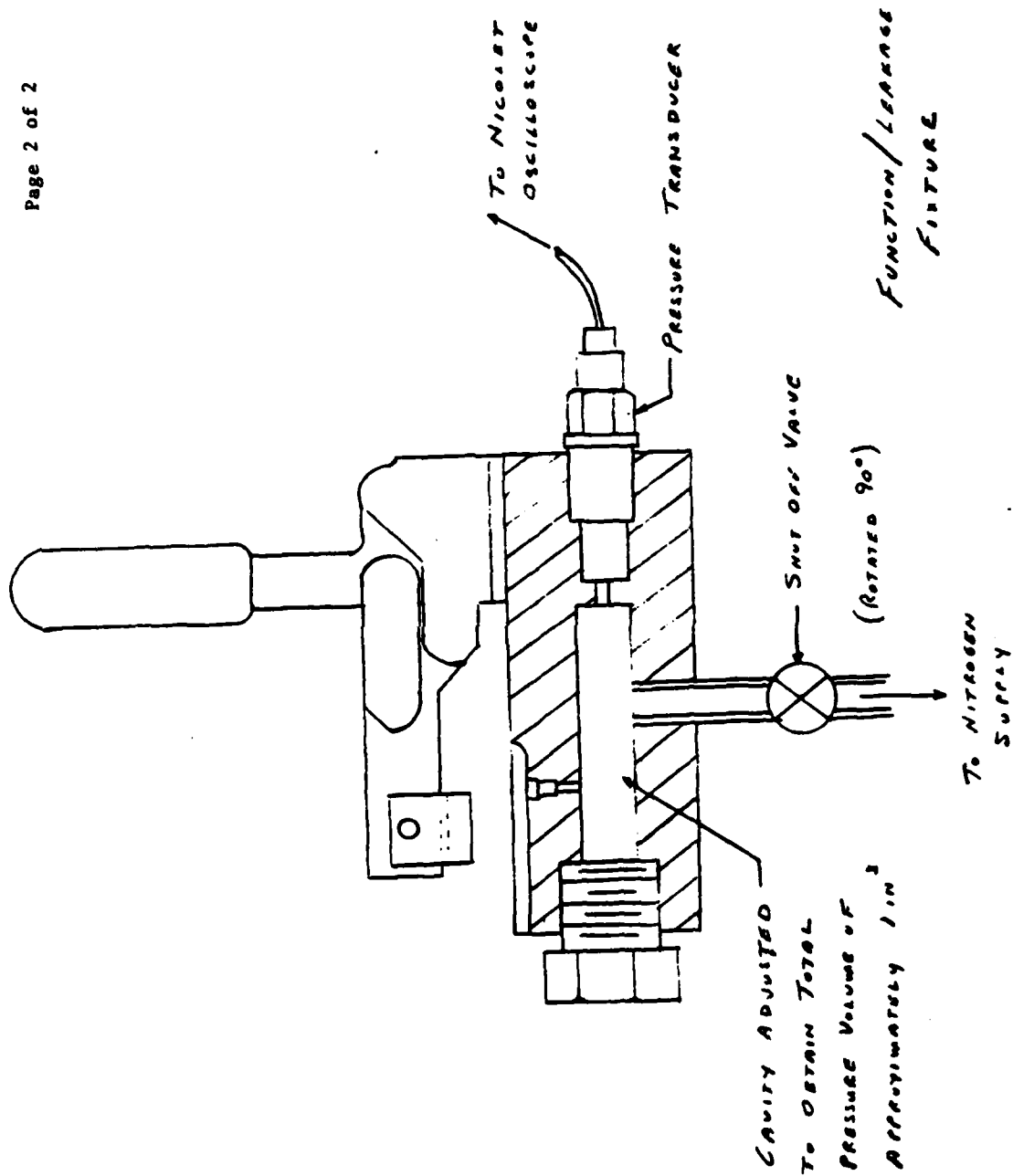
There is no pass/fail criteria.

Submitted by: J. E. Funk
Principal Engineer

ATTACHMENT I

OEXM 32211

Page 2 of 2



TEST PLAN

GEMMS/ERTS LANGLEIE TEST

1.0 Purpose

The purpose of the test is to determine the pressure level/pulse time characteristics to produce trip line deployment.

2.0 Test Procedure

2.1 Test A

- 2.1.1 Adjust pressure input solenoid to produce a square wave pulse of 50 ± 5 ms duration. Check that pressure rise and fall rates are 50 psi/ms minimum.
- 2.1.2 Adjust pressure input level desired. (Estimate expected release level from results of leakage test for first trial).
- 2.1.3 Install sensor in release fixture, pressurize and record result.
- 2.1.4 Repeat 2.1.2 and 2.1.3 (for a total of 30 units) using the Langlie test procedure per MIL-STD-331A and a new sensor for each trial.

2.2 Test B

- 2.2.1 Repeat 2.1 except the pulse duration will be adjusted to 75 ± 5 ms duration.

2.3 Test C

- 2.3.1 Repeat 2.1 except the pulse duration will be adjusted to 100 ± 5 ms duration.

3.0 Fixture Design

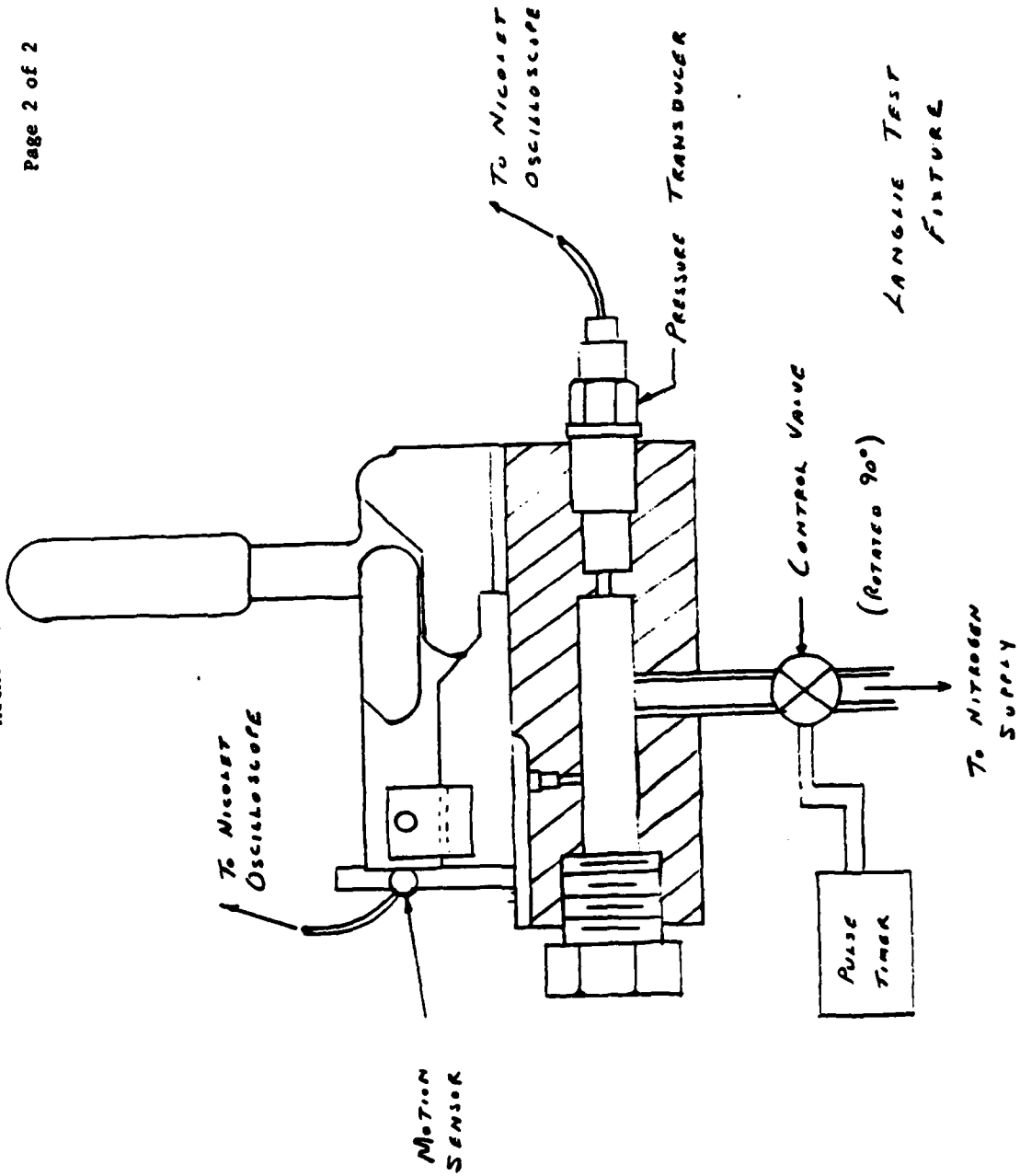
The fixture design is shown on the attached drawing.

4.0 Pass/Fail Criteria

There is no pass/fail criteria.

Submitted by: J. E. Funk
Principal Engineer

ATTACHMENT II



Distribution List

Commander
Armament Research and Development Center
U.S. Army Armament, Munitions and Chemical Command
ATTN: DRSMC-TSS(D) (5)
DRSMC-LCU-CM(D) (2)
Dover, NJ 07801

Administrator
Defense Technical Information Center
ATTN: Accessions Division (12)
Cameron Station
Alexandria, VA 22314

Director
U.S. Army Materiel Systems Analysis Activity
ATTN: DRXSY-MP
Aberdeen Proving Ground, MD 21005

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Chemical Research and Development Center
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ATTN: DRSMC-CLJ-L(A)
DRSMC-CLB-PA(A)
APG, Edgewood Area, MD 21010

Director
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Armament Research and Development Center
U.S. Army Armament, Munitions and Chemical Command
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