

AD-A163 282

MONOLITHIC CATHODES(U) OREGON GRADUATE CENTER BEAVERTON  
P R DAVIS ET AL. NOV 85 RADC-TR-85-211 F19628-84-K-0026

1/1

UNCLASSIFIED

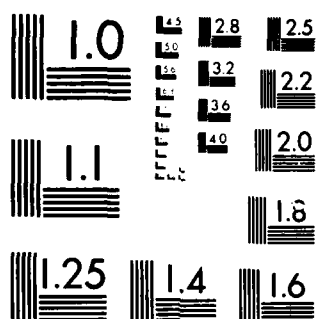
F/G 9/1

NL

|  |  |  |  |  |  |  |  |  |  |  |  |  |  |
|--|--|--|--|--|--|--|--|--|--|--|--|--|--|
|  |  |  |  |  |  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |  |  |  |  |

END

FILED  
-  
STC



MICROCOPY RESOLUTION TEST CHART  
NATIONAL BUREAU OF STANDARDS-1963-A

(12)



**RADC-TR-85-211**  
**Final Technical Report**  
**November 1985**

# ***MONOLITHIC CATHODES***

**AD-A163 282**

**Oregon Graduate Center**

**Paul R. Davis**  
**Lynwood W. Swanson**

**DTIC**  
**ELECTE**  
**JAN 22 1986**  
**B**

**APPROVED FOR PUBLIC RELEASE; DISTRIBUTION UNLIMITED**

**This effort was funded totally by the Laboratory Directors' Fund**

**DTIC COPY**

**ROME AIR DEVELOPMENT CENTER**  
**Air Force Systems Command**  
**Griffiss Air Force Base, NY 13441-5700**

This report has been reviewed by the RADC Public Affairs Office (PA) and is releasable to the National Technical Information Service (NTIS). At NTIS it will be releasable to the general public, including foreign nations.

RADC-TR-85-211 has been reviewed and is approved for publication.

APPROVED:



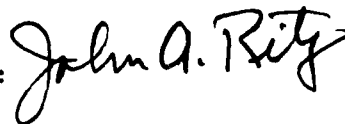
JOSEPH J. HUTTA  
Project Engineer

APPROVED:



HAROLD ROTH, Director  
Solid State Sciences Division

FOR THE COMMANDER:



JOHN A. RITZ  
Plans Office

If your address has changed or if you wish to be removed from the RADC mailing list, or if the addressee is no longer employed by your organization, please notify RADC (ESM) Hanscom AFB MA 01731. This will assist us in maintaining a current mailing list.

Do not return copies of this report unless contractual obligations or notices on a specific document requires that it be returned.

UNCLASSIFIED  
SECURITY CLASSIFICATION OF THIS PAGE

AD-A163282

| REPORT DOCUMENTATION PAGE                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         |                                             |                                                                                                  |                                                                                                                        |                                       |
|-------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|---------------------------------------------|--------------------------------------------------------------------------------------------------|------------------------------------------------------------------------------------------------------------------------|---------------------------------------|
| 1a. REPORT SECURITY CLASSIFICATION<br>UNCLASSIFIED                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                |                                             | 1b. RESTRICTIVE MARKINGS<br>N/A                                                                  |                                                                                                                        |                                       |
| 2a. SECURITY CLASSIFICATION AUTHORITY<br>N/A                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                      |                                             | 3. DISTRIBUTION/AVAILABILITY OF REPORT<br>Approved for public release;<br>distribution unlimited |                                                                                                                        |                                       |
| 2b. DECLASSIFICATION/DOWNGRADING SCHEDULE<br>N/A                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  |                                             |                                                                                                  |                                                                                                                        |                                       |
| 4. PERFORMING ORGANIZATION REPORT NUMBER(S)<br>N/A                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                |                                             | 5. MONITORING ORGANIZATION REPORT NUMBER(S)<br>RADC-TR-85-211                                    |                                                                                                                        |                                       |
| 6a. NAME OF PERFORMING ORGANIZATION<br>Oregon Graduate Center                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                     | 6b. OFFICE SYMBOL<br>(If applicable)        | 7a. NAME OF MONITORING ORGANIZATION<br>Rome Air Development Center (ESM)                         |                                                                                                                        |                                       |
| 6c. ADDRESS (City, State and ZIP Code)<br>19600 N.W., Von Neumann Drive<br>Beaverton, OR 97006-1999                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               |                                             | 7b. ADDRESS (City, State and ZIP Code)<br>Hanscom AFB MA 01731                                   |                                                                                                                        |                                       |
| 8a. NAME OF FUNDING/SPONSORING ORGANIZATION<br>Rome Air Development Center                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                        | 8b. OFFICE SYMBOL<br>(If applicable)<br>ESM | 9. PROCUREMENT INSTRUMENT IDENTIFICATION NUMBER<br>F19628-84-K-0026                              |                                                                                                                        |                                       |
| 8c. ADDRESS (City, State and ZIP Code)<br>Hanscom AFB MA 01731                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                    |                                             | 10. SOURCE OF FUNDING NOS.                                                                       |                                                                                                                        |                                       |
|                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   |                                             | PROGRAM ELEMENT NO.<br>61101F                                                                    | PROJECT NO.<br>LDFFP                                                                                                   | TASK NO.<br>01<br>WORK UNIT NO.<br>C5 |
| 11. TITLE (Include Security Classification)<br>MONOLITHIC CATHODES                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                |                                             |                                                                                                  |                                                                                                                        |                                       |
| 12. PERSONAL AUTHOR(S)<br>Paul R. Davis, Lynwood W. Swanson                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       |                                             |                                                                                                  |                                                                                                                        |                                       |
| 13a. TYPE OF REPORT<br>Final                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                      | 13b. TIME COVERED<br>FROM Dec 83 To Apr 85  | 14. DATE OF REPORT (Yr., Mo., Day)<br>November 1985                                              |                                                                                                                        | 15. PAGE COUNT<br>34                  |
| 16. SUPPLEMENTARY NOTATION<br>This effort was funded totally by the Laboratory Directors' Fund                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                    |                                             |                                                                                                  |                                                                                                                        |                                       |
| 17. COSATI CODES                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  |                                             |                                                                                                  | 18. SUBJECT TERMS (Continue on reverse if necessary and identify by block number)                                      |                                       |
| FIELD<br>21                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       | GROUP<br>02                                 | SUB. GR.<br>07                                                                                   | Lanthanum Hexaboride, Thermionic Cathode<br>Cerium Hexaboride, Thermionic Emission<br>Boride Cathodes, Electron Source |                                       |
| 19. ABSTRACT (Continue on reverse if necessary and identify by block number)<br><p>The purpose of this program was to evaluate LaB<sub>6</sub> (210) and (310) oriented single crystal material for flat cathode applications and to deliver mounted LaB<sub>6</sub> cathodes with (310) orientation for operation in RADC/Varian life test vehicles. To that end, the program was divided into four separate tasks, as shown in the report. This project was extended for 3 months past the initial program period, at no additional cost, so that unavoidable delays could be overcome and the project completed. The final report covers the entire period of the program, which terminated 29 March 1985.</p> |                                             |                                                                                                  |                                                                                                                        |                                       |
| 20. DISTRIBUTION/AVAILABILITY OF ABSTRACT<br>UNCLASSIFIED/UNLIMITED <input checked="" type="checkbox"/> SAME AS RPT <input type="checkbox"/> DTIC USERS <input type="checkbox"/>                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  |                                             |                                                                                                  | 21. ABSTRACT SECURITY CLASSIFICATION<br>UNCLASSIFIED                                                                   |                                       |
| 22a. NAME OF RESPONSIBLE INDIVIDUAL<br>Joseph J. Hutta                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                            |                                             |                                                                                                  | 22b. TELEPHONE NUMBER<br>(Include Area Code)<br>(617) 861-4247                                                         | 22c. OFFICE SYMBOL<br>RADC (ESM)      |

DD FORM 1473, 83 APR

EDITION OF 1 JAN 73 IS OBSOLETE

UNCLASSIFIED  
SECURITY CLASSIFICATION OF THIS PAGE

## FINAL TECHNICAL REPORT

### I. Introduction

The purpose of this program was to evaluate LaB<sub>6</sub>(210) and (310) oriented single crystal material for flat cathode applications and to deliver mounted LaB<sub>6</sub> cathodes with (310) orientation for operation in RADC/Varian life test vehicles. To that end, the program was divided into four separate tasks, as shown in Table I.

This project was extended for 3 months past the initial program period, at no additional cost, so that unavoidable delays could be overcome and the project completed. This final report covers the entire period of the program, which terminated 29 March 1985.

TABLE I  
PROGRAM TASK SUMMARY

- Task I: Investigate clean LaB<sub>6</sub>(210) and LaB<sub>6</sub>(310) surface work functions, thermal stabilities and other physical properties
- Task II: Investigate LaB<sub>6</sub>(210) and LaB<sub>6</sub>(310) interactions with tube gases at operating temperature
- Task III: Investigate clean CeB<sub>6</sub>(210) or CeB<sub>6</sub>(310) surface work function and thermal stability
- Task IV: Prepare, mount and deliver to RADC LaB<sub>6</sub>(210) or LaB<sub>6</sub>(310) cathodes suitable for operation in Varian life test vehicles

Pages 22 and 24 do not contain proprietary information. The information on these pages is in the public domain.

Per Ms. Joyce Watkins, RADC/TSR

|      |         |
|------|---------|
| Dist | Special |
| A-1  |         |

des or

## II. Task I: LaB<sub>6</sub> Clean Surface Studies

### A. Experimental Techniques

Both emission properties and vaporization rates (thermal stability) of the LaB<sub>6</sub>(210) and (310) surfaces have been investigated in ultra-high vacuum. An improved thermionic diode system was set up to allow current density stability to be studied over long periods of time in the pulsed mode. To study the cathode emission stability in the high power DC mode (~ 580 watts at 80 A/cm<sup>2</sup>, 1600 V, 1700 K) it is necessary to use water cooling to maintain a low collector temperature, thereby preventing the cathode from becoming contaminated with material evaporated from the collector surface. Figure 1 shows the water cooled collector which was constructed. The collector is electrically isolated from the rest of the system, by a glass tube between flanges, thus allowing the emission current to be monitored. The cathode is mounted on the two high voltage feedthroughs.

The vacuum system shown in Figure 2 was assembled to study emission characteristics in good vacuum and in partial pressures of various gases. In addition, this system was used for material evaporation (thermal stability) investigations, both in good vacuum and in partial pressures of O<sub>2</sub>. Emission measurements made in this system have been related to thermal stability results, as discussed later in this report.

### B. Experimental Results

Figure 3 shows J-F data obtained for a LaB<sub>6</sub>(310) cathode in the pulsed mode using this system. These results are presented here as log J vs F<sup>1/2</sup> plots, with the Child's law (space charge limited) regime at low fields and the Schottky regime at high fields. Note that the "knee," or transition region, occurs, for example, at about 4 A/cm<sup>2</sup> for a cathode temperature of 1500 K.

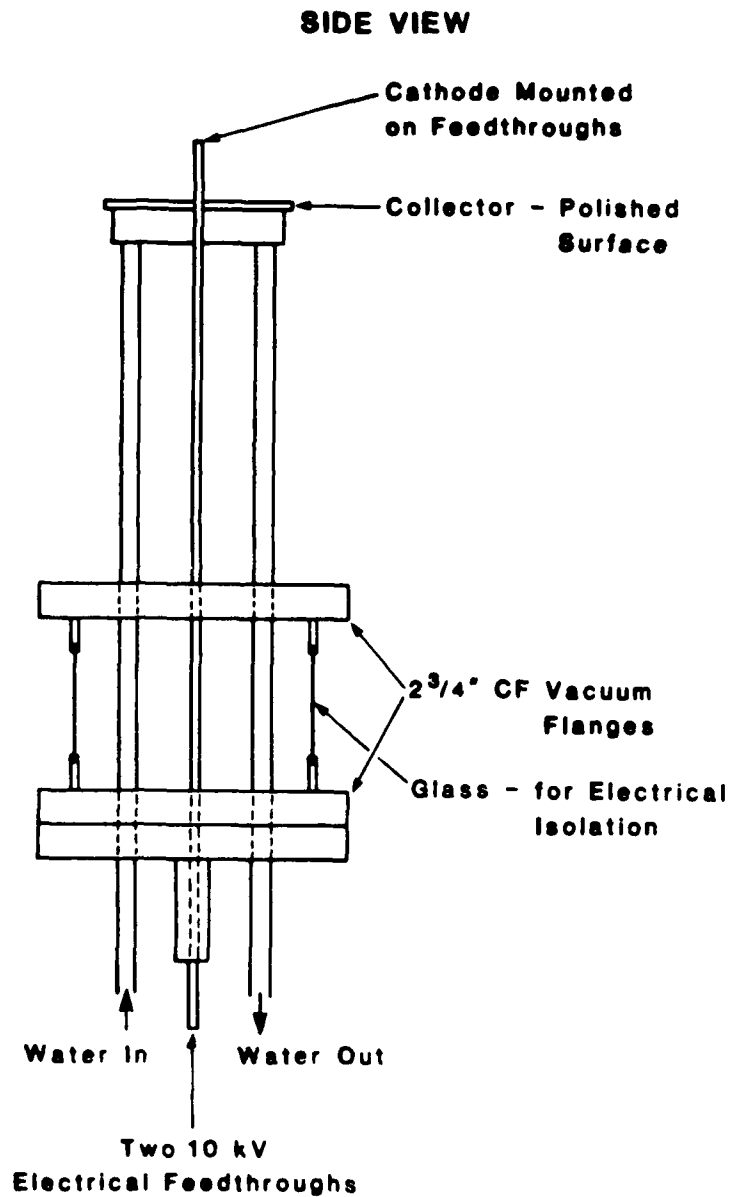


Figure 1. Water cooled collector used to study DC emission from  $\text{LaB}_6$  and  $\text{CeB}_6$  cathodes.

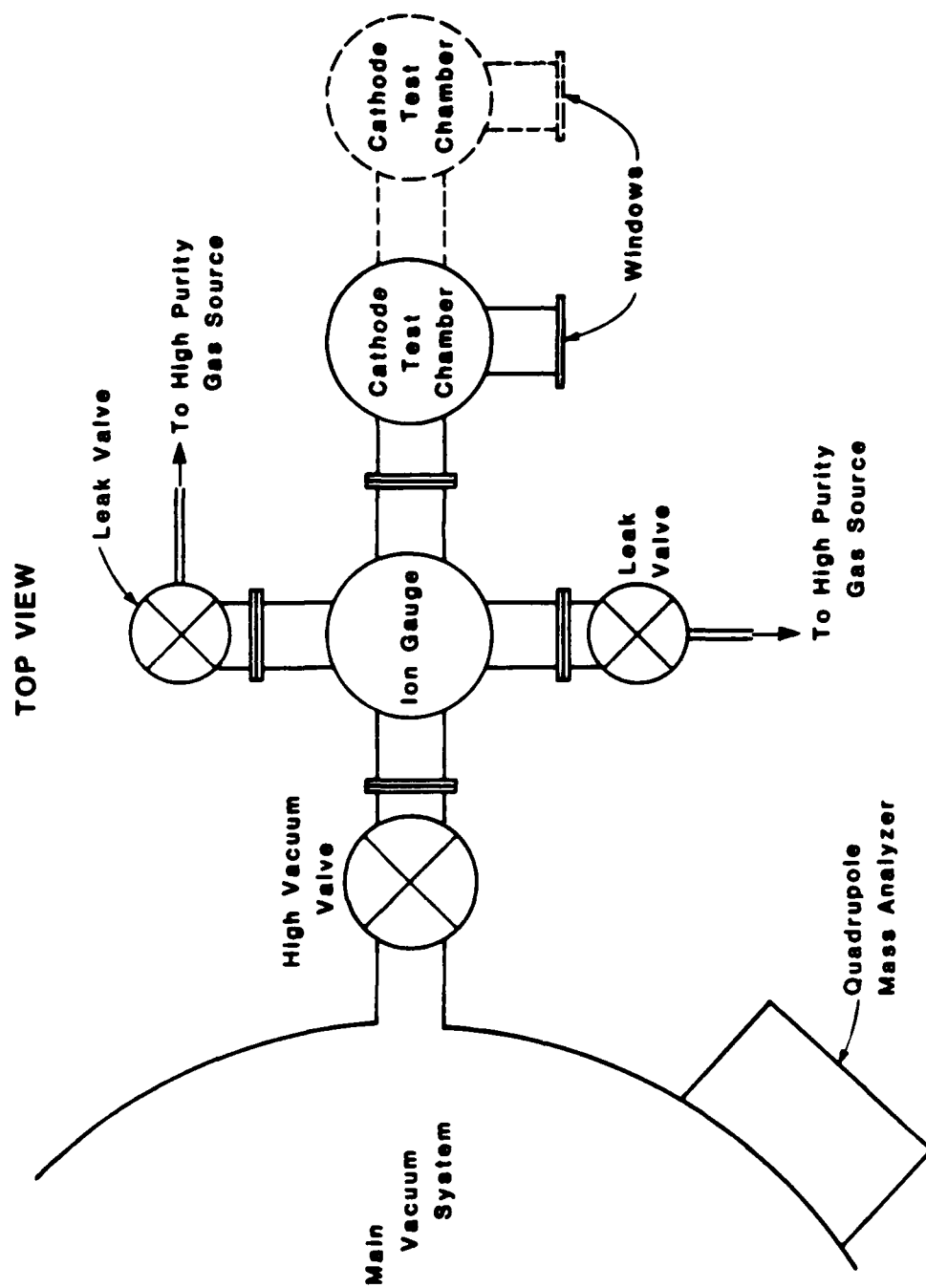


Figure 2. Vacuum system to study emission and thermal stability characteristics of  $\text{LaB}_6$  and  $\text{CeB}_6$  cathodes.

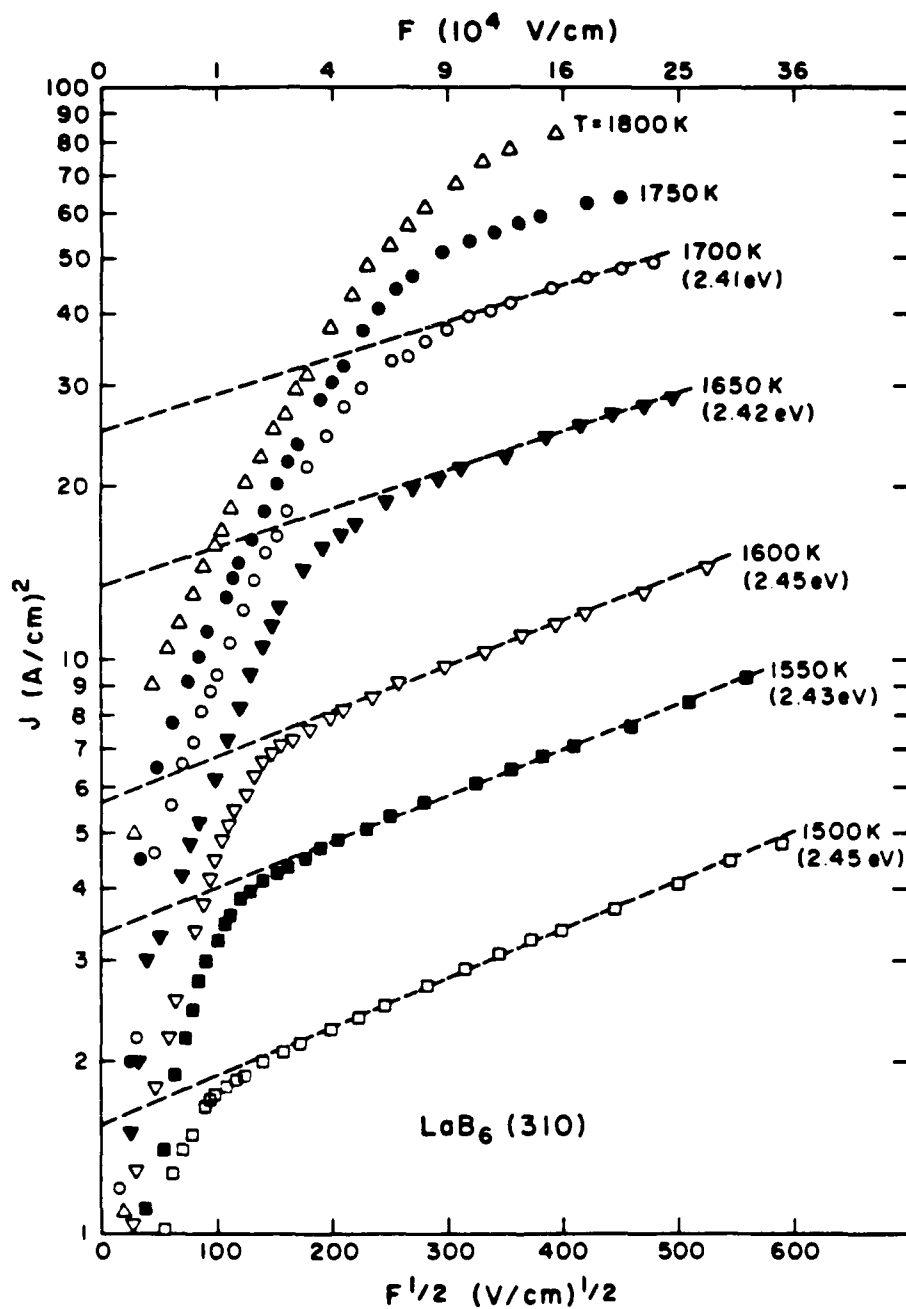


Figure 3. Emission characteristics of a  $\text{LaB}_{6.09}(310)$  surface, measured in a pulsed thermionic diode at various indicated temperatures. In the Schottky (high field) emission regime, the current density  $J$  is expected to be a linear function of the square root of the applied field  $F$ . At low values of  $F$ , the emission exhibits Child's Law (space-charge limited) behavior.

Figure 4 is a comparison of 1700 K data like that of Fig. 3 with the Child's law-Schottky line curve. The comparison of Fig. 4 is based on a planar diode configuration fit with a cathode work function of 2.36 eV. The observed deviation from the calculated curve is to be expected, even for an ideal cathode, in a planar diode, as has recently been pointed out by Scott,<sup>1</sup> since an abrupt transition from the retarding to accelerating regimes is an oversimplification of the true diode behavior. In fact, space charge effects are present within the accelerating field range, and these effects cause knee rounding and gradual approach to the Schottky line in J-F plots.

Figure 5 summarizes effective thermionic work function variation with temperature of the (210) and (310) surfaces under clean vacuum conditions with  $P < 1 \times 10^{-9}$  torr. The (100) surface behavior is included for comparison. Note that both the (210) and (310) surfaces exhibit considerably lower work functions than the (100) surface. At 1600 K,  $\phi(310)$  is approximately 0.25 eV lower than  $\phi(100)$ , corresponding to an increase in zero-field emitted current density by a factor of about six, from 1.4 to 8.3 A/cm<sup>2</sup>. This dramatic result suggests that cathodes with  $\langle 310 \rangle$  orientation should be able to emit current densities of 1 A/cm<sup>2</sup> at approximately 1450 K and 2 A/cm<sup>2</sup> at less than 1500 K. These are emission current densities used in standard life test studies.<sup>2</sup>

Emission capabilities comprise only half of the story for cathode characterization. The other parameter of interest is lifetime, which can be considered to be long-term stability at the operating temperature. Thermal stability for LaB<sub>6</sub> cathodes is equivalent to vaporization rate or material loss rate. All the LaB<sub>6</sub> results discussed in this report refer to single crystal samples of B/La = 6.09 stoichiometry, the best value we have found for maximizing electron emission and minimizing material loss.

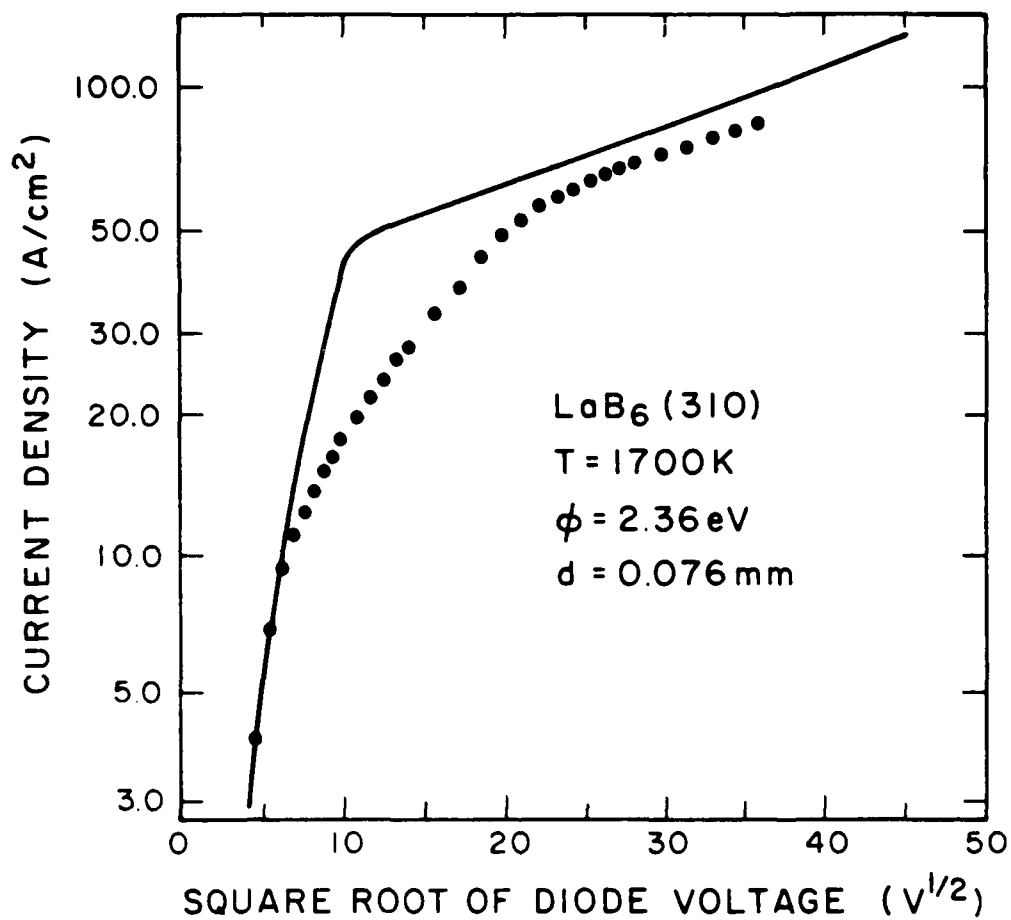


Figure 4. Comparison of LaB<sub>6</sub>(310) J-V data (points) at 1700 K with the predicted Child's law-Schottky line (solid), in the region of the knee. Data from Figure 3. A work function of 2.36 eV was used to calculate current density in the Schottky regime.

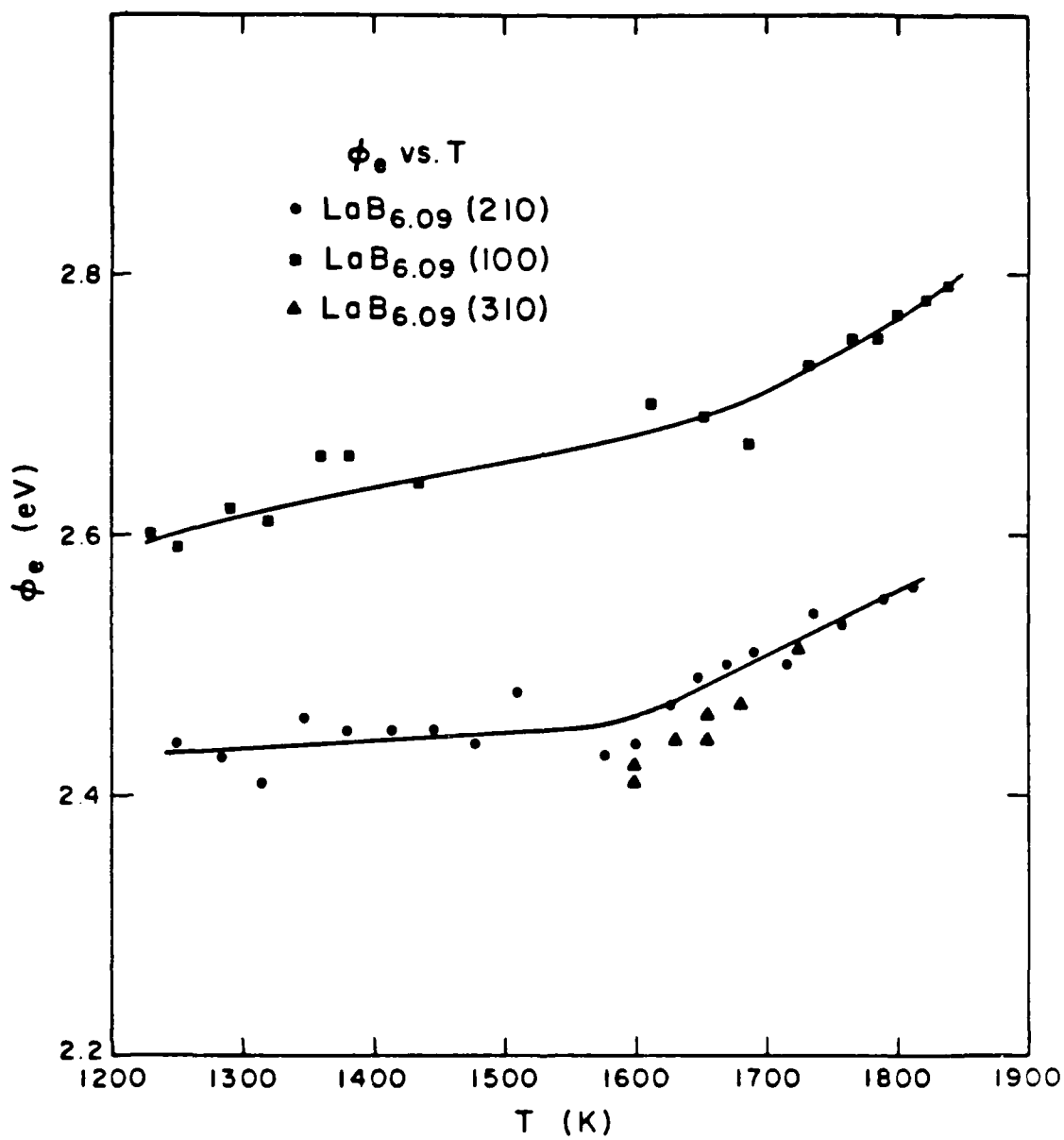


Figure 5. Temperature dependences of thermionic work functions,  $\phi_e$ , of the (100), (210) and (310) planes of  $\text{LaB}_{6.09}$ .

Figure 6 shows the results of evaporation rate experiments we have performed on a variety of  $\text{LaB}_6$  crystals, each point corresponding to one mounted cathode. Measurements of cathode dimensional changes were made following heating under various vacuum conditions at the indicated temperatures. Particular attention should be focused on curve (D), corresponding to ultra-high vacuum conditions. In this situation, material evaporation is isotropic, meaning the rate is the same from all crystal faces. Thus, faceting of the cathode surface does not occur under these conditions.

Data from Figures 5 and 6 can be combined into a figure of merit plot showing the ratio of emitted current density to material evaporation rate. On this basis, different types of cathodes may be compared, as in Figure 7, where  $\text{LaB}_6(210)$ ,  $\text{LaB}_6(310)$  and an impregnated dispenser cathode (IDC)<sup>3</sup> are compared. Note in Figure 7 that both (210) and (310)  $\text{LaB}_6$  have figures of merit superior to an M-type IDC over the entire range of operating temperatures. At 1400 K the M-type IDC undergoes rapid alloying of the Os layer with the W matrix, an effect which increases the cathode work function and reduces the emitted current density.

### III. Task II: $\text{LaB}_6$ Interactions with Tube Gases

The thermal stability studies in  $\text{O}_2$  constitute an important part of the overall project. In order to determine the effect of  $\text{O}_2$  on the evaporation rate,  $\text{O}_2$  was leaked into the vacuum chamber during evaporation. The quantitative results of this investigation for  $\text{O}_2$  are given in Fig. 6 for  $1 \times 10^{-7}$ ,  $5 \times 10^{-7}$  and  $1 \times 10^{-6}$  torr  $\text{O}_2$ . As expected, the presence of  $\text{O}_2$  increases the evaporation rate due to the formation of volatile oxides. Furthermore, some interesting features of the  $\text{O}_2$  effect have been observed.

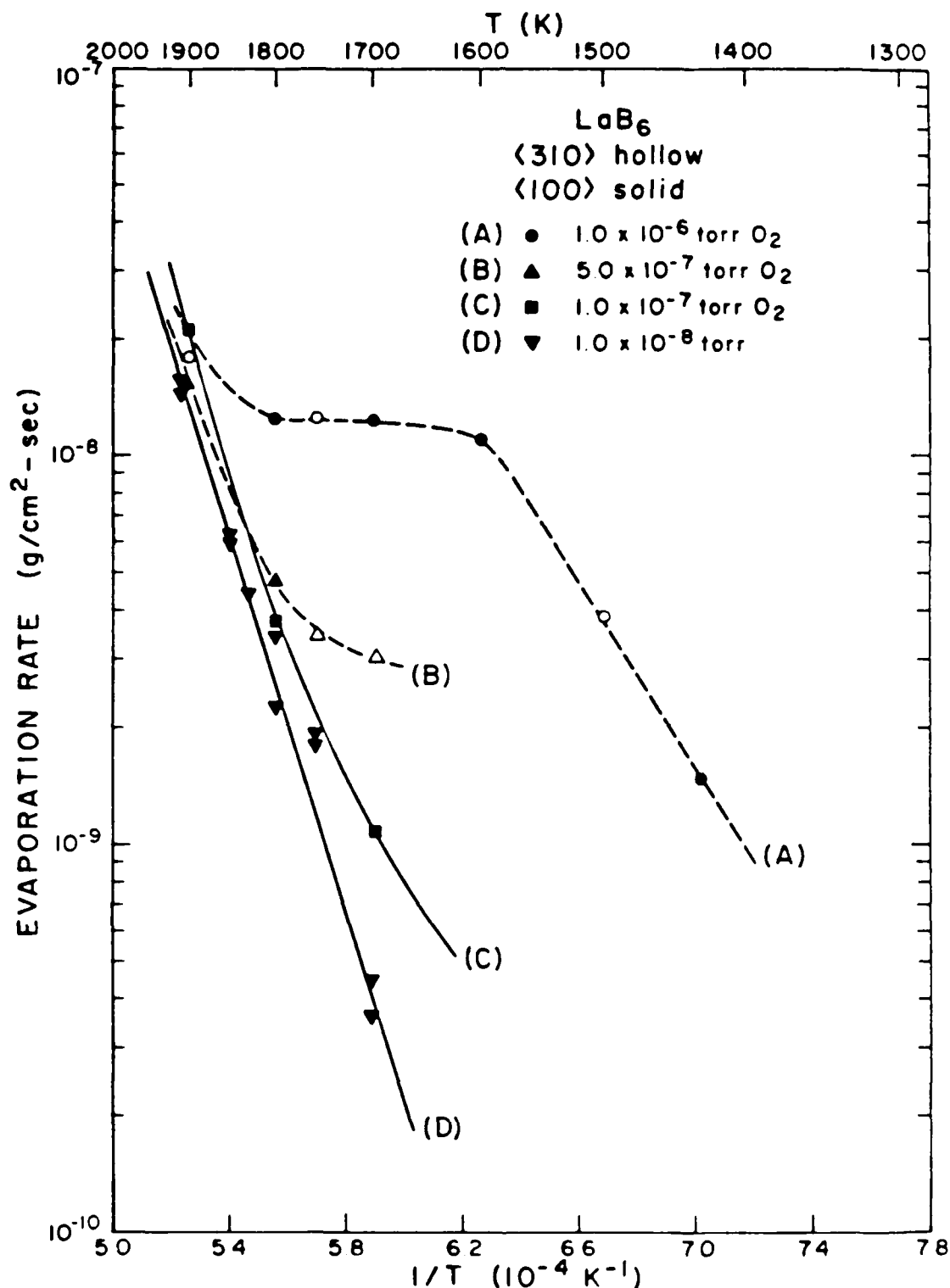


Figure 6. Results of long-term evaporation studies of  $\text{LaB}_{6.09}$  single crystals in good vacuum and in various  $\text{O}_2$  pressures, using conical conical specimens. The open points indicate  $\langle 310 \rangle$  and the solid points indicate  $\langle 100 \rangle$  axial specimen orientation.

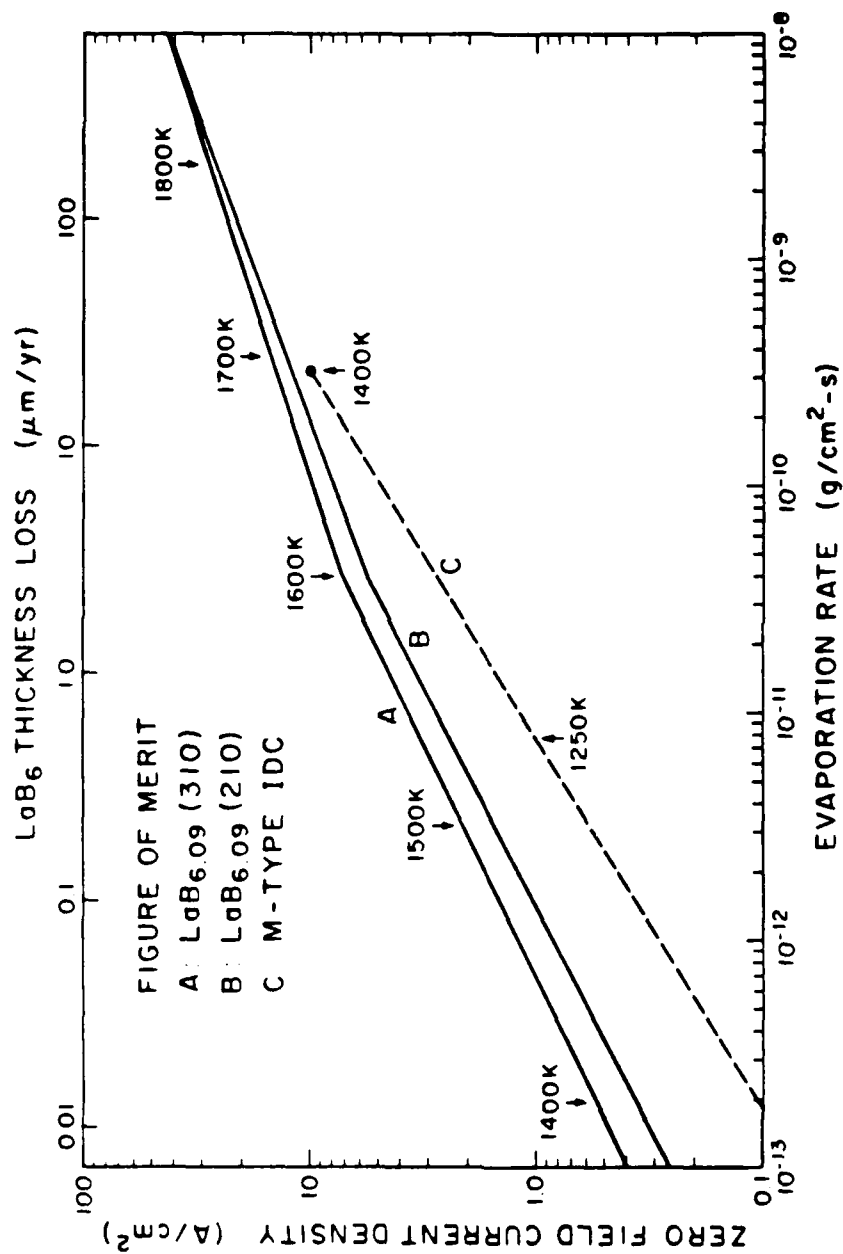


Figure 7. Figure of merit comparison of  $\text{LaB}_6$ (310) and (210) faces and a state-of-the-art M-type impregnated dispenser cathode (IDC).

As shown in Fig. 6, the effect of  $O_2$  on the evaporation enhancement is greatest for the lower temperature range. (In an earlier study<sup>4</sup> it was shown that no evaporation of La or B containing products occurred at  $P_{O_2} \leq 2 \times 10^{-6}$  and  $T < 1000$  K.) For  $T > 1850$  K there is a negligible effect of  $O_2$  on the evaporation rate. At the highest  $O_2$  pressure studied here ( $1 \times 10^{-6}$  torr), the rate of evaporation becomes virtually independent of temperature for  $1600 \leq T < 1850$  K. However, at 1600 K the evaporation rate increases more than 100 times when the partial pressure of  $O_2$  is increased from  $< 1 \times 10^{-8}$  to  $1 \times 10^{-6}$  torr. These results are indicated more clearly in Fig. 8, where the data of Fig. 6 are recast as evaporation rate vs  $O_2$  pressure for several temperatures of interest.

Besides the increase in evaporation rate, the presence of a partial pressure of  $O_2$  causes a significant alteration of the geometric shape of a single crystal cathode. With  $\langle 100 \rangle$  oriented emitters, evaporation of as little as  $15 \mu m$  of  $LaB_6$  in the presence of  $1 \times 10^{-7}$  torr of  $O_2$  is sufficient to initiate formation of an observable pyramidal structure. This faceting was found to be caused by a crystallographic anisotropy of enhancement of evaporation of  $LaB_6$  in  $O_2$ . The anisotropy in evaporation rates of the major crystal planes due to this effect has been determined to be  $\Delta r_{100} > \Delta r_{111} > \Delta r_{110}$ .

#### IV. Task III: $CeB_6$ Clean Surface Studies

The thermal stability of  $CeB_6$  cathodes was studied in a preliminary fashion. Figure 9 shows the results of vaporization rate measurements on two different  $CeB_6$  cathodes in good vacuum, compared with results for  $LaB_6$ . These results suggest somewhat better thermal stability for  $CeB_6$  than for  $LaB_6$ , especially since the stoichiometry of the  $LaB_6$  cathodes has been optimized at

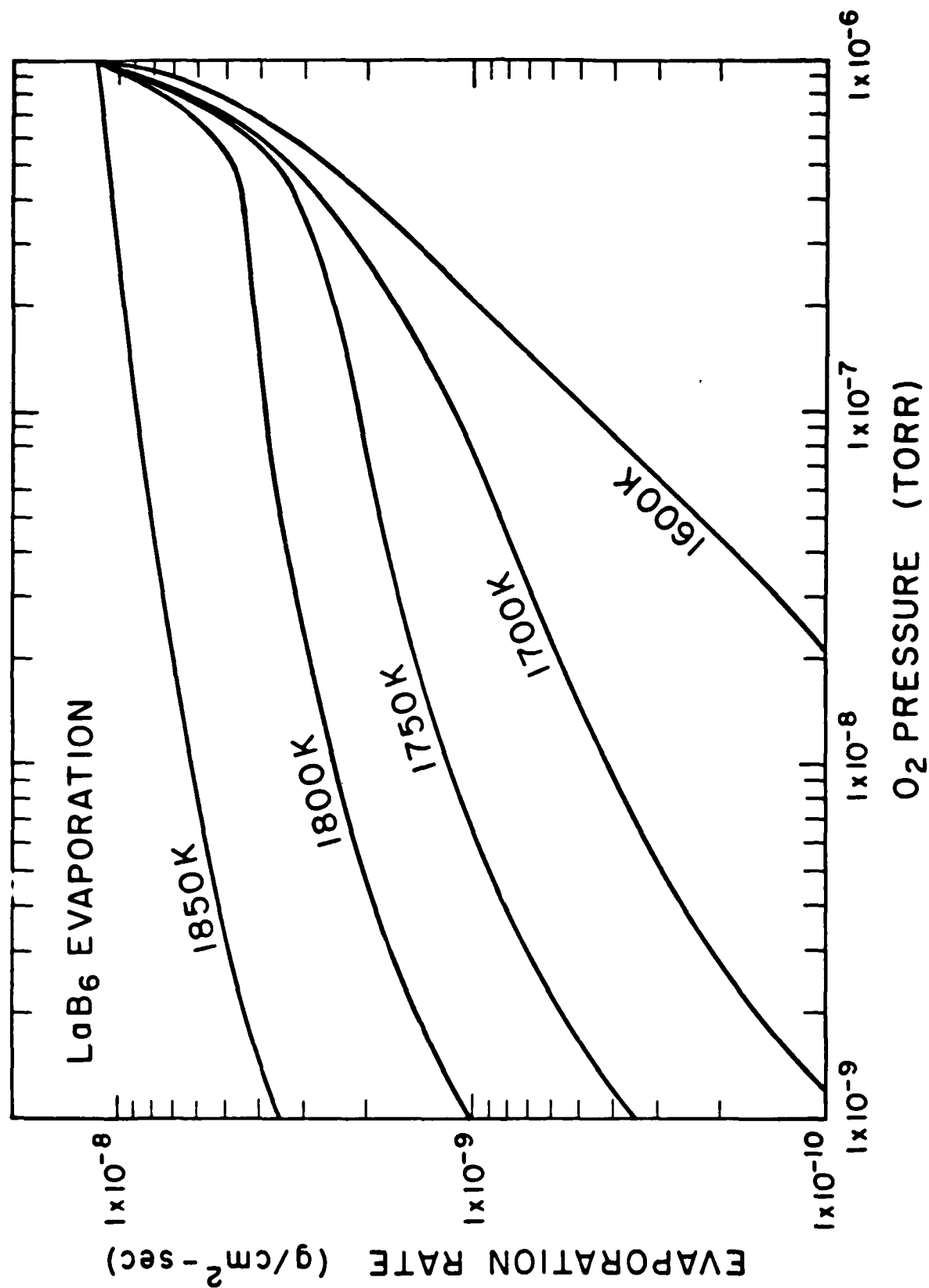


Figure 8. Variation of LaB<sub>6</sub> evaporation rate, at various temperatures, as a function of O<sub>2</sub> pressure.

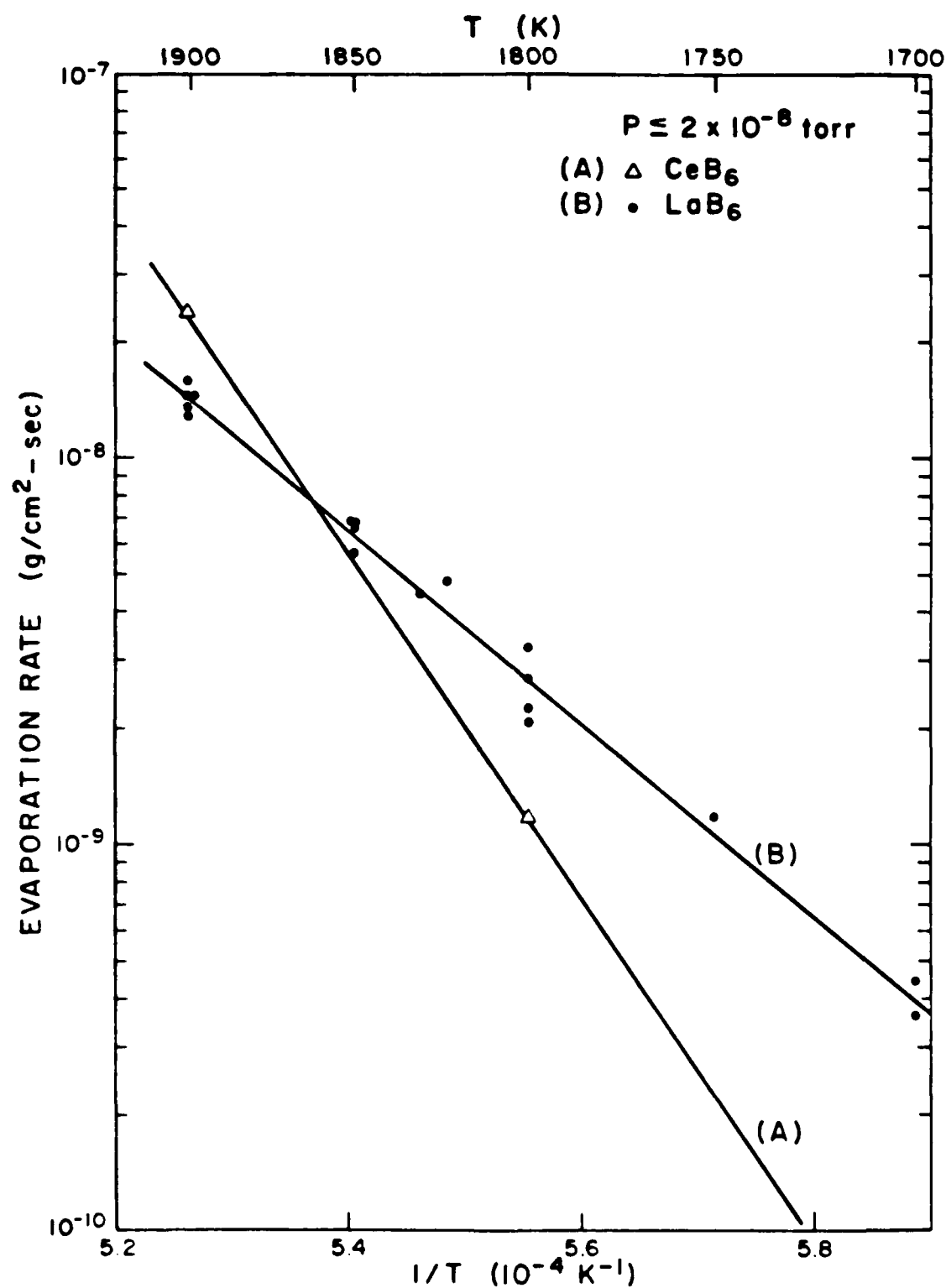


Figure 9. Vaporization rate comparison, in good vacuum, of  $\text{CeB}_6$  (A) and  $\text{LaB}_6$  (B). The stoichiometry of the  $\text{LaB}_6$  cathode is optimum at  $\text{B/La} = 6.09$ , while that of the  $\text{CeB}_6$  has not yet been optimized.

B/La = 6.09, while the least volatile  $\text{CeB}_6$  stoichiometry has not yet been determined.

After  $\text{LaB}_6$  .09 (310) was chosen as the optimum cathode surface for life test studies, investigations of  $\text{CeB}_6$  were discontinued, although this material still holds promise and warrants a long-term research project to determine its ultimate utility as a cathode material.

V. Task IV: Prepare, Mount and Deliver  $\text{LaB}_6$  Cathodes

A. Cathode Preparation and Mounting

The  $\text{LaB}_6$  cathode configuration to be used in the RADC life test facility was the subject of considerable deliberation. Figure 10 shows the important details of a preliminary test configuration selected. The cathode orientation was chosen to be (310), in order to obtain maximum current density at a given temperature. Dimensions of the test cathode are indicated in Figure 10.

The test cathode was fabricated and mounted by FEI Co., Hillsboro, OR to specifications provided by the contract principal and co-investigators. Figure 11 shows the design which was chosen. Tests on the expansion of this structure at operating temperature were conducted.

A miniature prototype cathode was prepared for and delivered to NASA, Lewis Research Center, Cleveland, OH under Grant NAG3-434. This cathode was fabricated and mounted by FEI Co., Hillsboro, OR. Results of studies on this cathode are reported here because the experience gained in its operation is germane to the discussion of  $\text{LaB}_6$  cathode development.

The miniature prototype sleeve mounted cathode structure is shown in Figure 12(a). It consisted of a .040" diameter Mo sleeve into which a .033" diameter Re cup had been spot welded. The cup held a (310)  $\text{LaB}_6$  single crystal as can be seen in Figure 12(b). The overall length of the cathode

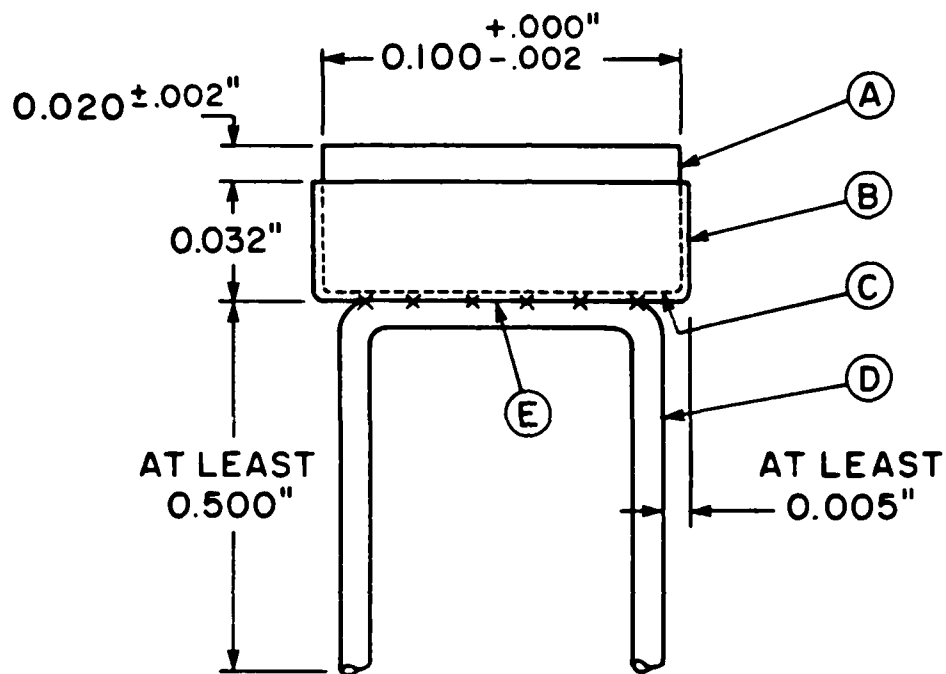


Figure 10. LaB<sub>6</sub>(310) test cathode with integral heater. A: LaB<sub>6</sub>(310) button; B: Re cup, 0.100 diam.  $\times$  .030 deep, approx. .002 wall thickness; C: Braze; D: Re heater wire, diam. chosen so cathode can attain 1800 K minimum; E: spot welds. Dimensions in inches.

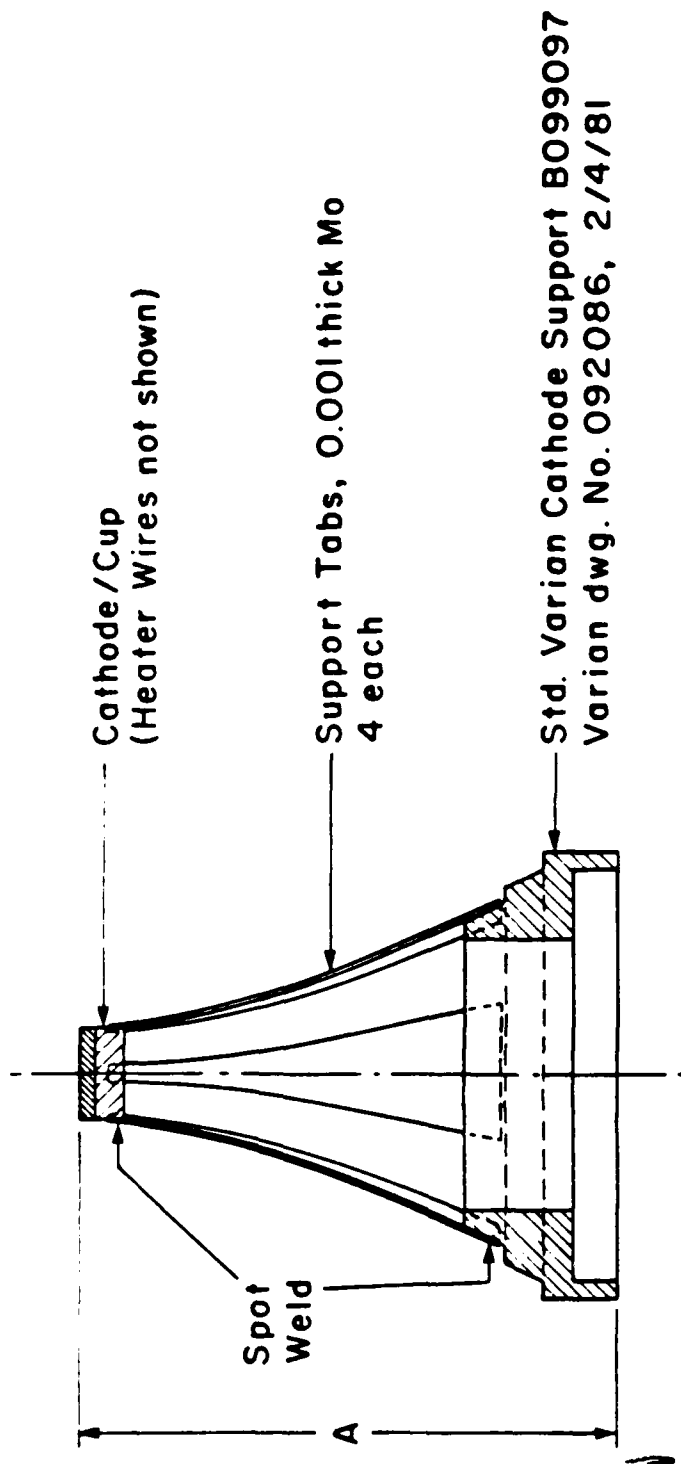
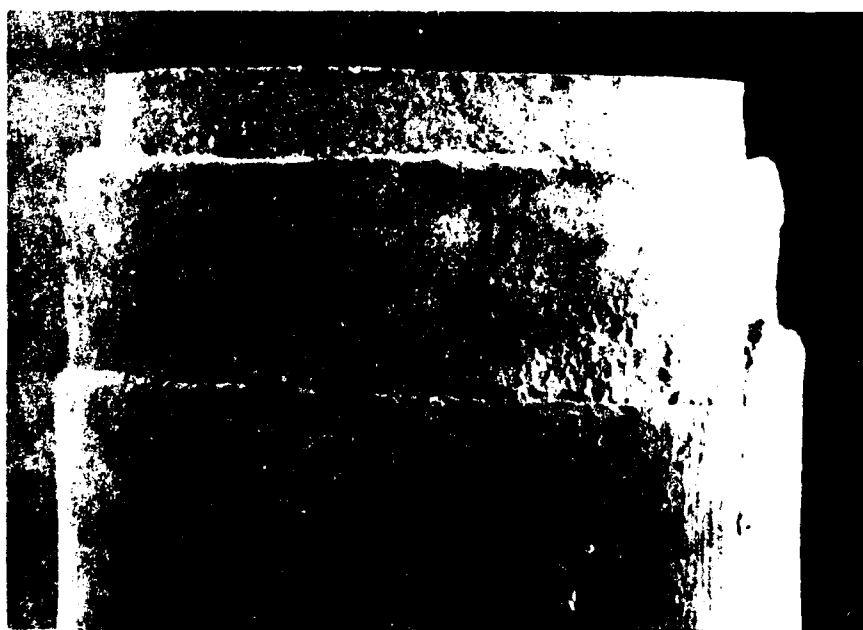


Figure 11. Detail of test cathode mounting structure.



(a)

1 mm



(b)

0.1 mm

structure was .47". Heating of the cathode was accomplished by resistively heating a .008" diameter Re wire which was spot welded to the bottom of the Re cup and held in place by a ceramic tube. The ceramic tube was press fit into a larger diameter stainless steel collar that can also be seen at the bottom of the cathode structure in Figure 12(a).

This miniature cathode structure, although very different in size from the one finally chosen for the life test vehicle allowed us to conveniently test the sleeve cathode and Re cup concepts. The heating characteristics of this structure and its structural integrity over short heating intervals at the anticipated operating temperatures were examined. Figure 13 shows the heating power and current as a function of temperature. From results discussed under Task I, we estimated that in order to achieve a current density of  $5 \text{ A/cm}^2$  at the knee of the J(F) characteristics we would need to operate at a cathode temperature of 1500 to 1550 K. At this temperature the heating power requirement is ~ 3 W and the current required is ~ 1.7 A for the miniature prototype. For these tests the prototype cathode structure was supported by the heating leads and the temperature at the base of the sleeve was 1340 K for an emitter temperature of 1500 K.

These tests confirmed that the general sleeve and Re cup concepts used here for mounting the  $\text{LaB}_6$  are sound. For the cathode structure used in the life test facility a .100" diameter  $\text{LaB}_6$  crystal has been employed. Thus, the heating power requirements are correspondingly larger than for the miniature prototype. Nevertheless, on the basis of the tests on the smaller prototype structure and larger test cathode we believe that the basic sleeve concept utilizing the Re cup mount will be successful.

We have determined, as a result of our cathode mounting experiments and through discussions with personnel of RADC and Varian, Inc., that the most

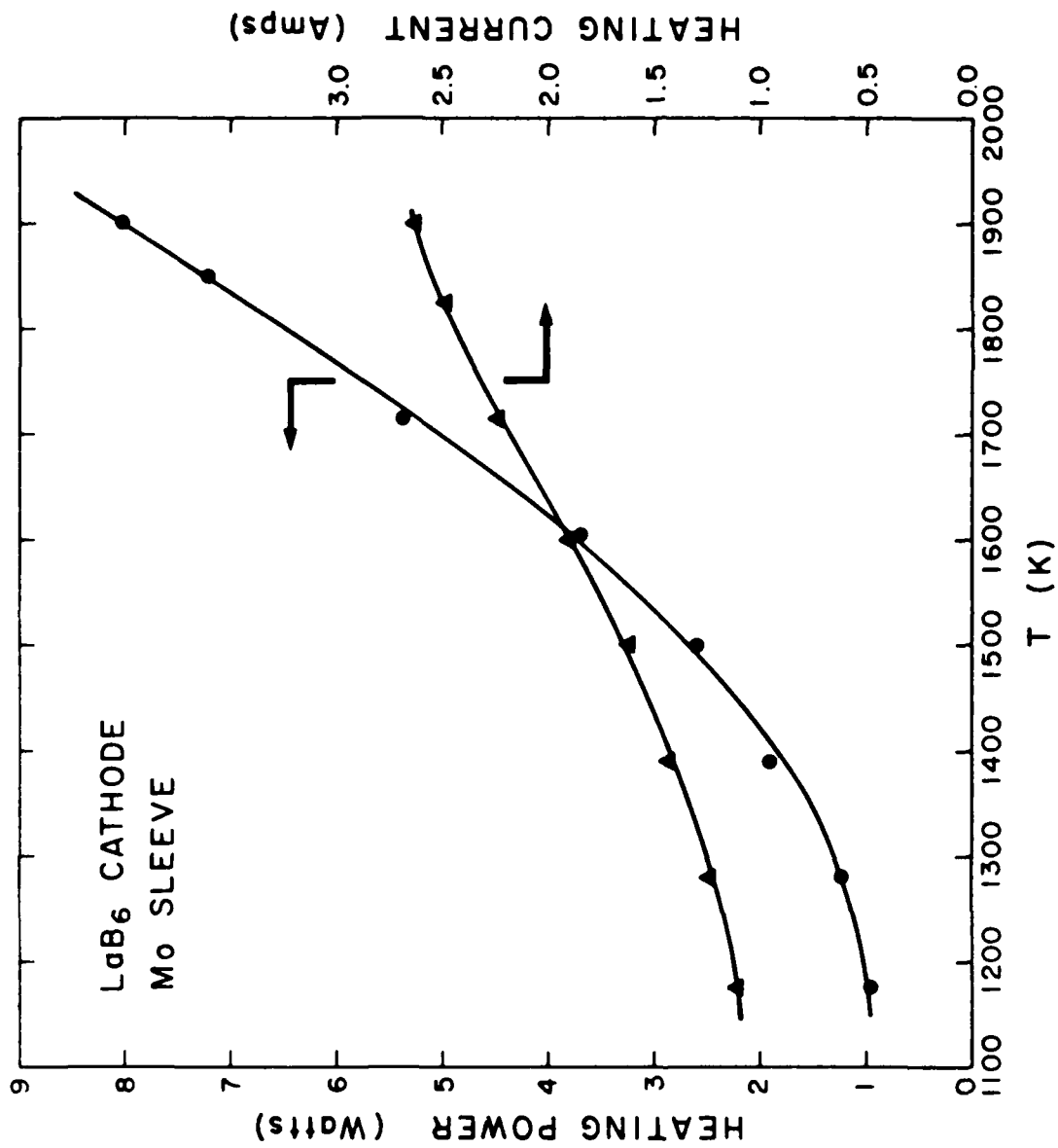


Figure 13. Heating characteristics of prototype sleeve mounted LaB<sub>6</sub> cathode structure of Figure 12.

stable, most practical and least expensive mounting structure for  $\text{LaB}_6$  cathodes in RADC/Varian life test vehicles is a modified version of the standard Varian cathode assembly.

Figure 14 is a modified Varian drawing showing the Mo button which holds the  $\text{LaB}_6$  (310) cathode. This cathode was brazed into a Re cup, which was in turn brazed into the recess shown in the button in Fig. 14.

Figure 15 shows heating test data for the cathode assembly with  $\text{LaB}_6$  cathode installed. For these measurements the assembly was clamped by its base in a copper holder, in order to simulate mounting in an actual test vehicle. The power requirements for heating this cathode are somewhat different when installed in the actual test vehicle, since additional heat shielding is used in that case.

Figure 16 illustrates the final assembly, which is standard in all respects, except for the  $\text{LaB}_6$  (310) cathode (not shown). Standard Varian heaters will be used with these cathodes. The illustrated cathode assembly fits the standard RADC/Varian life test vehicle. Two of these life test vehicles were reserved at Varian for use with the  $\text{LaB}_6$  (310) cathodes.

#### B. Delivery of $\text{LaB}_6$ Cathodes

Two  $\text{LaB}_6$  (310) cathodes, mounted as illustrated in Fig. 16, were delivered to Varian, Palo Alto, CA on 13 May 1985. As of 26 July, one cathode had been installed in a test vehicle and had undergone preliminary testing. The second cathode was still being mounted in a tube. Varian personnel estimate that both vehicles will be delivered to RADC before the end of August 1985.

Figure 17 is a copy of the Varian Final Data Sheet for the first test vehicle with installed  $\text{LaB}_6$  cathode. Figure 18 is an emission rolloff plot for this tube at 2.0, 3.5 and 5.0 kV beam voltage. Note that it is possible

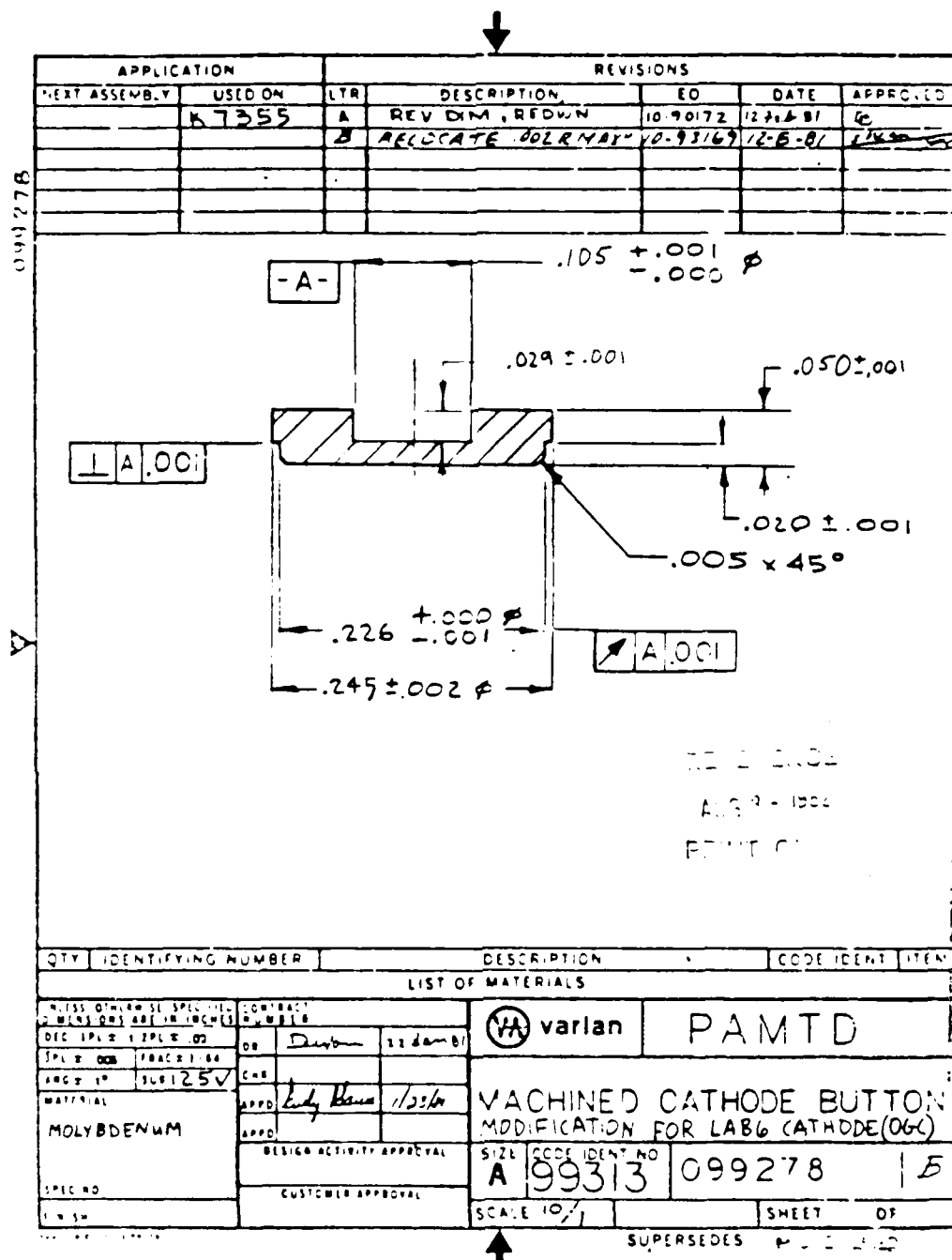
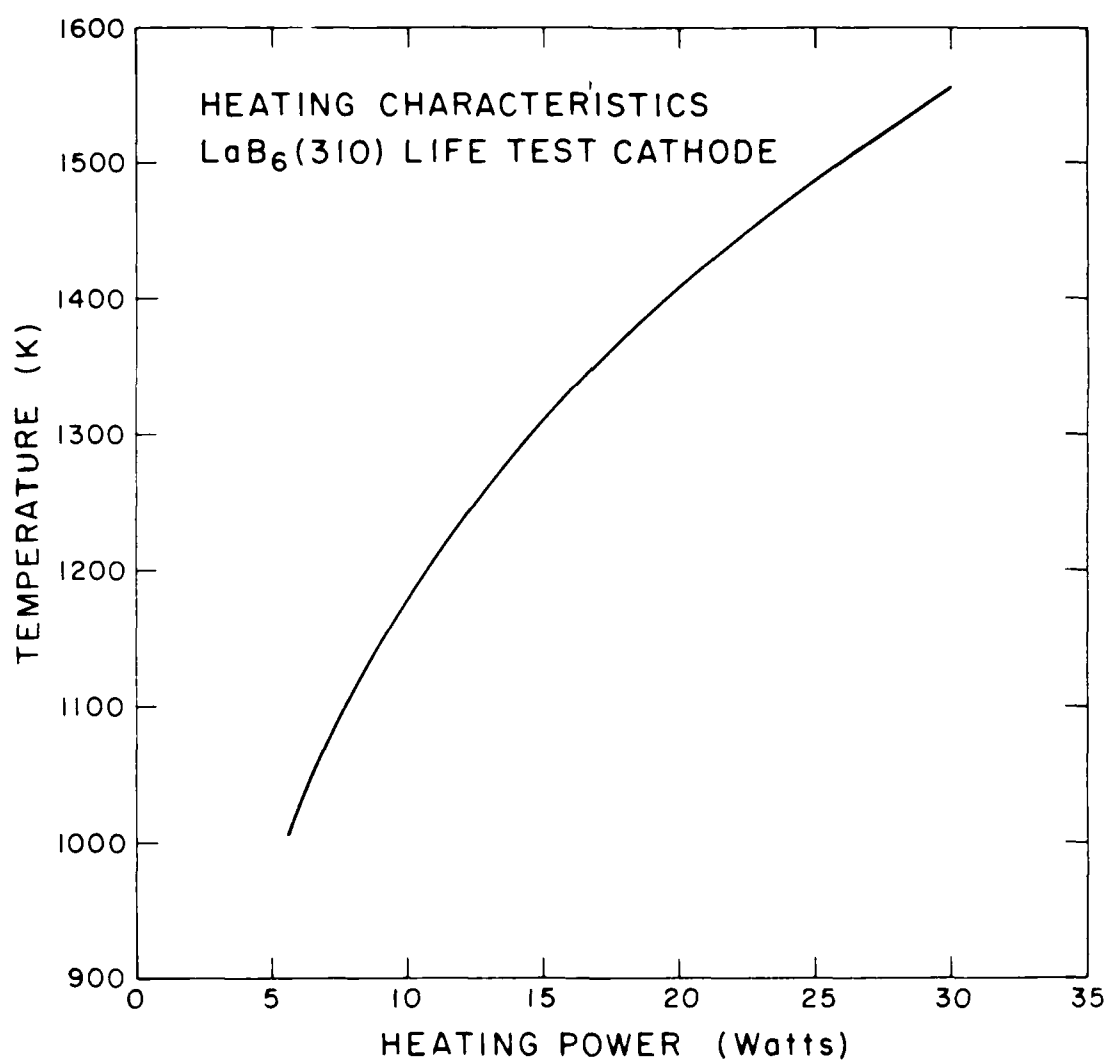


Figure 14. Modified Varian drawing of machined Mo button. A LaB<sub>6</sub> (310) cathode in a Re cup was brazed into the recess at the top of the button.



1. The cathode was heated in a vacuum furnace at a rate of 100°C/min. The temperature was measured by a thermocouple.

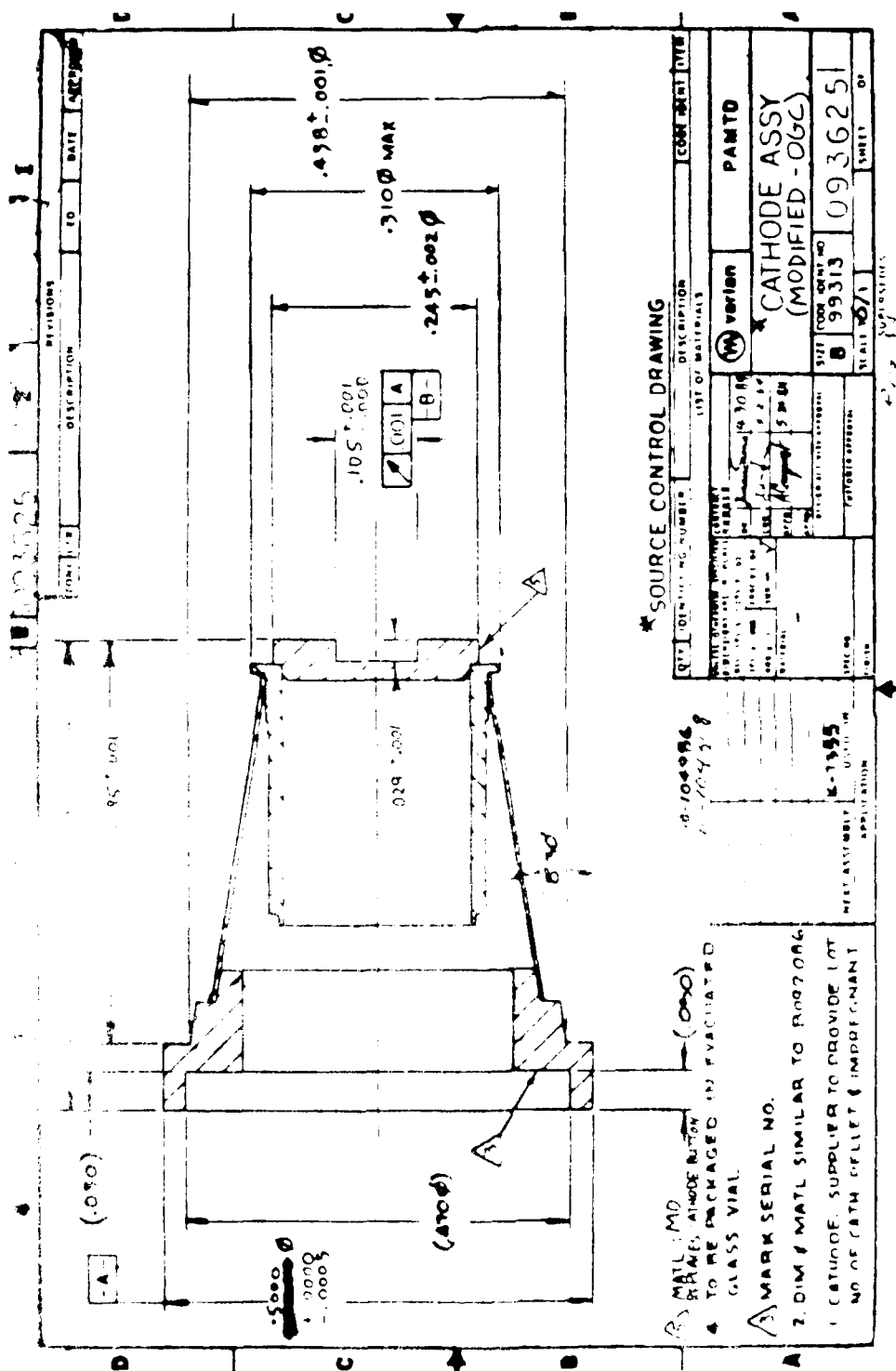


Figure 16. Modified Varian drawing of the standard cathode assembly for the RADC/Varian life test vehicle. The modification shows the button of Figure 14.

# V-7355 LIFE TEST VEHICLE FINAL DATA SHEET

|                   |                          |
|-------------------|--------------------------|
| Test Vehicle S/N: | 216                      |
| Gun S/N:          | 103AA                    |
| Cathode S/N:      | LaB <sub>6</sub><br>Blue |

## CONFIGURATION:

- Life Test Vehicle (V-7355) P/N B099265 Rev. \_\_\_\_\_
- Final Cathode Assembly (Gun) P/N C092095 Rev. \_\_\_\_\_
- Cathode Support Assembly (Cathode) P/N B092086 Rev. \_\_\_\_\_

## ACCEPTANCE DATA:

| Item | Test Parameter                            | Test Specification                                   | Accept             |
|------|-------------------------------------------|------------------------------------------------------|--------------------|
| A    | Cathode Temperature                       | <del>1500 ± 25</del> SPECIAL                         | ✓                  |
| B    | 2 liter/sec. Pump Pressure                | $1 \times 10^{-7}$ Torr max                          | ✓                  |
| C    | Perveance                                 | $0.5 \pm 10\%$ $\mu$ perv                            | 0.436 $\mu$ P      |
| D    | Leakage Current<br>( $E_b = E_c = 10kV$ ) | Collector 150 $\mu$ A max<br>Cathode 150 $\mu$ A max | 10 $\mu$ A<br>0 AA |
| E    | Total Age Time                            | 20 Hr. min                                           | 203.7 Hr           |

## ATTACHMENTS:

- ~~Mitron Plot~~ ROLL-OFF CURVE ✓
- Cathode Fabrication Data Sheet ✓

Tested By: RS / EA Conquest Date: 7-24-85  
 QA Approval: \_\_\_\_\_ Date: \_\_\_\_\_

87-800-746 5/84

Fig. 17. Final data sheet for LaB<sub>6</sub>(310) test vehicle S/N 216.

to achieve space-charge limited operation only at 2.0 kV within the heater limitation of about  $1270^{\circ}\text{C}_B$  with this cathode. The position of the knee in Fig. 17 ( $1250^{\circ}\text{C}_B$  or 1550 K) implies a cathode work function of 2.6 eV, considerably greater than the 2.4 eV observed in carefully controlled laboratory experiments (see Section II). This higher work function value may be attributed to contamination present on the  $\text{LaB}_6$  surface, which needs to be heated to  $> 1600$  K for cleaning.

We expect to maintain contact with RADC personnel during the startup and test periods of these vehicles, in order to minimize possible problems.

# EMISSION PLOT SPECIAL CATHODE (LaB<sub>6</sub>)

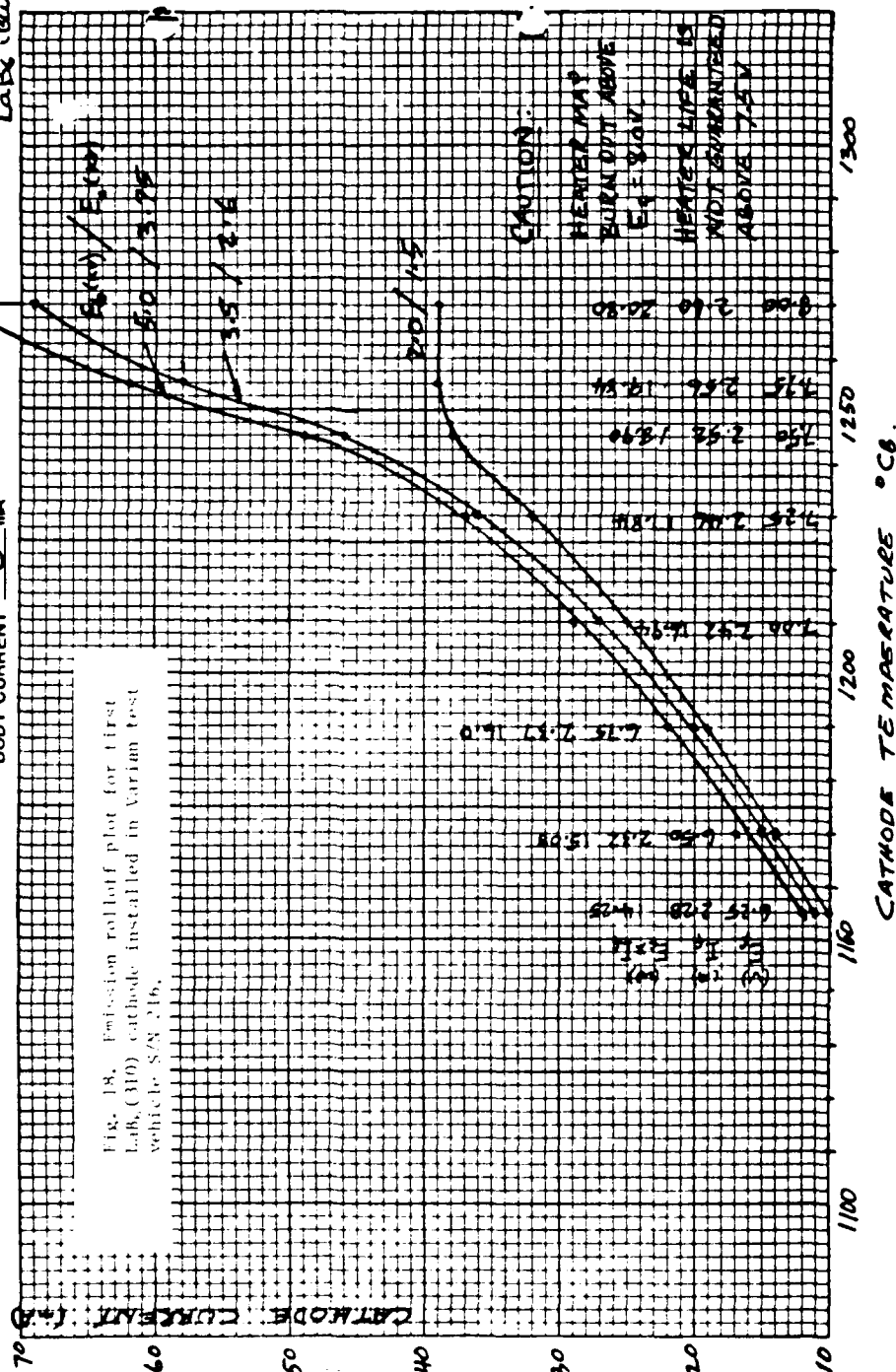
V-7355 SERIAL NO. 216

FILAMENT VOLTAGE 7.5 V  
FILAMENT CURRENT 2.51 A

BEAM VOLTAGE ASSUME KV  
BEAM CURRENT AS SPEC mA  
BODY CURRENT 0 mA

DATE: 7-24-85 BY: SAR

GUN S/N: 103AA  
CATHODES/N: 02021 'G.C.  
LaB<sub>6</sub> (BLUE)



#### REFERENCES

1. J. B. Scott, J. Appl. Phys. 52 (1981) 4406.
2. D. Bussey, RADC. (Private communication)
3. IDC vaporization data taken from R. A. Tuck, Appl. Surf. Sci. 2 (1979) 128.
4. P. R. Davis and S. A. Chambers, Appl. Surf. Sci. 8 (1981) 197.



*MISSION  
of  
Rome Air Development Center*

RADC plans and executes research, development, test and selected acquisition programs in support of Command, Control, Communications and Intelligence (C<sup>3</sup>I) activities. Technical and engineering support within areas of competence is provided to ESD Program Offices (POs) and other ESD elements to perform effective acquisition of C<sup>3</sup>I systems. The areas of technical competence include communications, command and control, battle management, information processing, surveillance sensors, intelligence data collection and handling, solid state sciences, electromagnetics, and propagation, and electronic, maintainability, and compatibility.

**END**

**FILMED**

**2-86**

**DTIC**