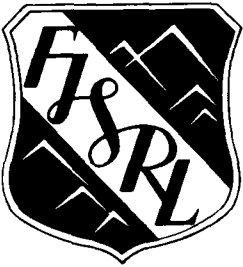


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FRANK J. SEILER RESEARCH LABORATORY

FJSRL-TR-86-0001 ✓

MARCH 1986

AD-A166 307

ADVANCED LASER BASED INERTIAL
INSTRUMENT DEVELOPMENT

FINAL REPORT

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MAJOR JAMES P. POTGE'

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PROJECT 2301-F1-71

AIR FORCE SYSTEMS COMMAND
UNITED STATES AIR FORCE

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FJSRL-TR-86-0001


This document was prepared by the Guidance and Control Division, Directorate of Lasers and Aerospace Mechanics, Frank J. Seiler Research Laboratory, United States Air Force Academy, Colorado Springs, CO. The research was conducted under Project Work Unit Number 2301-F1-71, Advanced Laser Based Inertial Instrument Development (ALBIID). Major Salvatore R. Balsamo and Major James R. Rotge were the Project Scientists in charge of the work.

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This report has been reviewed by the Commander and is released to the National Technical Information Service (NTIS). At NTIS it will be available to the general public, including foreign nations.

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INTRODUCTION

This work unit consisted of work done in three areas: (1) design and implementation of a passive resonant ring laser gyroscope based upon a high-finesse resonant cavity; (2) high accuracy clock based upon stabilizing a He-Ne laser and using the inter-mode beat as a clock signal; and (3) the "flip-flop" gyro - an attempt to suppress the lock-in phenomenon present in active ring laser gyros by alternating the traveling wave direction within the ring cavity. The bulk of the work performed on this effort was in the latter two areas. Papers describing each of these three areas are included in this final report. Discussion of (1) above will not be included here as the subject was addressed in greater detail under Work Unit 2301-F1-68, "Large Passive Resonant Ring Laser Gyro Project."

DISCUSSION

Laser Clock - The goal of this effort was to develop an inexpensive yet accurate and hardy clock to be used in avionics systems. This device should require very little or no warm up time and the stated accuracy goal was one part in 10^8 .

The principle behind this avionics clock is the beat signal obtained by heterodyning the output of a He-Ne laser lasing in two longitudinal modes on the He-Ne gain curve at $\lambda = 633$ nm. By controlling the cavity optical path length the positions of the two modes on the gain curve can be held relatively stable. The cavity (effective) length can be controlled by various means including discharge current control, active and passive

thermal control and physical length control of the laser gain tube. The experiment was never completed due to the late availability of various components, especially the high speed detector necessary to sense the intermode beat frequency (~ 1 GHz). Additional details of this effort are provided by the paper, "High Accuracy Clock Using a Stabilized HeNe Laser," by S. Balsamo (Appendix A).

Flip-Flop Gyro - Active ring laser gyros exhibit a lock-in phenomenon at near zero rotation rates. This is due to the fact that both running waves (CW and CCW) are being amplified by a common He-Ne gain medium (at near zero rotation rates). This lock-in phenomenon is common to all coupled oscillators with similar resonant frequencies. In laser gyros now in commercial operation, the technique used to get around this lock-in effect is mechanical dither. Other lock-in avoidance or compensation schemes have been proposed and tried but with limited success. This research effort was to demonstrate a different approach to lock-in avoidance; that of causing the ring laser to lase in only one direction (CW or CCW) at a time. By introducing a fixed optical delay equal to the reciprocal of the switching frequency one could then beat the clockwise and counter clockwise running waves together on a detector; the resulting AC signal from the detector being proportional to the rotation rate (re Sagnac Effect) of the device.

The method used to alternately suppress the counter-running waves is discussed in the paper attached as Appendix B, "An Active Ring Laser Gyroscope Without Lock-in: Concept, Design, and Feasibility," by S. Balsamo. Additional work is required to fully understand the characteristics of the system as described in this paper.

CONCLUSION

Both the laser clock and flip-flop gyro efforts were discontinued prematurely. Major Balsamo retired from the Air Force and there was insufficient manpower to continue this work unit. There is interesting and potentially valuable experimental work to be performed to fully demonstrate both of these concepts. In the case of the laser clock, the basic question is how well can one lock the positions of the (two) longitudinal modes in the gain curve. A rough shot noise calculation for a typical device shows the minimum discernible frequency shift for a mode on the gain curve to be of order 10^{-3} Hz, and a more practical limit of 10 Hz or less. This calculation depends upon the assumption that the mode intensities are only functions of their positions on the gain curve. This implies a differential measurement to discriminate against intensity variations common to both modes. There is no fundamental reason why the laser clock would not work at the desired levels of precision (i.e., 1 part in 10^9). Whether this can be achieved in practice will depend upon frequency pulling effects and whether these can be controlled with sufficient precision.

Much remains to be done to fully demonstrate the flip-flop gyroscope. The transient behavior of the running wave laser at MHz switching speeds requires further investigation. High-Q cavities will have photon lifetimes approaching micro-seconds and may result in residual coupling of the running waves. This work might be advanced thru suitably managed AFIT graduate student efforts in terms of Master theses research. The transient behavior of a periodically switched running wave He-Ne ring oscillator would of itself be an interesting research topic.

APPENDIX A

HIGH ACCURACY CLOCK USING A STABILIZED HeNe LASER

by

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The objective of this effort is to demonstrate that a relatively low cost (less than \$1,000.00) HeNe laser and simple electronics can provide a high accuracy (1 part in 10^{11}) timing source.

The timing device is a HeNe laser stabilized with two adjacent orthogonally polarized modes. The beat frequency between the adjacent modes (Δf) is used as the timing reference.

Analysis indicates that the technique discussed here is indeed possible. In addition, a technique similar to the one proposed here is now commonly used as a basis for stabilizing HeNe lasers^{1,2}.

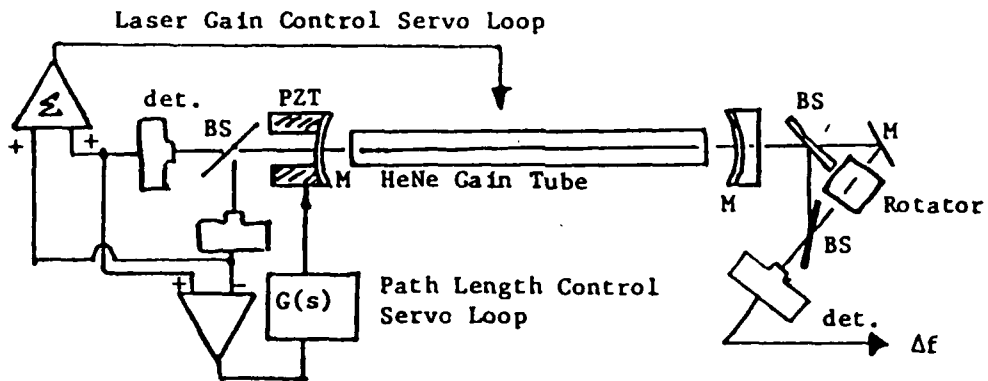


Figure 1. Clock Schematic Diagram

As shown in the figure above, Δf is the beat frequency between the two adjacent longitudinal modes. The laser length is approximately 15 cm allowing adjacent modes to be about 1 GHz. The beat frequency would be divided by 200 to provide the stable 5 MHz reference.

The problems can be seen by examining the adjacent modes as shown below in Figure 2.

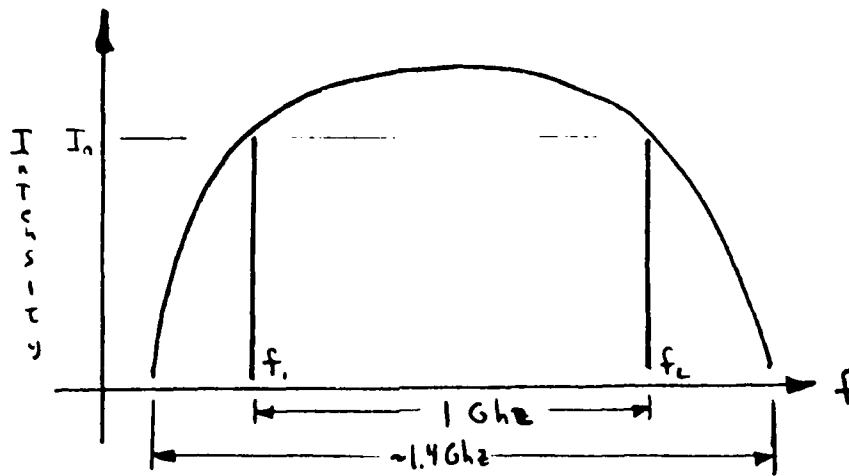


Figure 2. Placement of Modes on the HeNe Gain Curve

The two adjacent modes must be fixed on the gain curve. Any shift right or left will effect the 1 GHz beat frequency between the two adjacent modes because of variations in the index of refraction of the HeNe. Recall that the index of refraction is the first derivative of gain (to the first order), so that any drifting of the modes will effect the beat frequency. Locking the modes eliminates errors due to pulling and pushing. Similarly, any changes in gain will effect n and therefore the beat frequency (see Figure 3). A path length control loop such as the one shown in Figure 1 will fix the longitudinal modes on the gain curve.

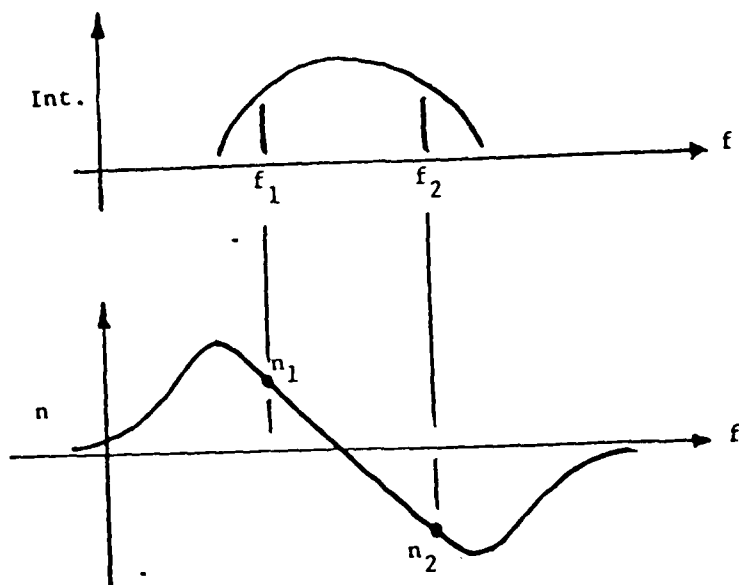


Figure 3. Laser Intensity and Index of Refraction versus Frequency

The technique is based upon the premise that the two adjacent longitudinal modes will be at orthogonal polarizations and thus be separable. This will be true if care is taken to insure that relatively equal losses for each mode are incurred in the gain tube, i.e., no Brewster windows, magnets, etc.

The approach is limited by the ability of the servo to fix the adjacent longitudinal modes. The servo is, in turn, limited in bandwidth by the mechanical noise in the laser. A very low noise, and very low thermal drift material, CERVIT, is used for the laser cavity.

The beat-frequency is derived by rotating the polarization of one of the adjacent modes and then sensing the interference of the two modes with a high frequency, low noise detector. A second method using a single polarizer at 45° to the two adjacent modes also yields a beat frequency but at a cost of a low signal to noise figure.

REFERENCES

1. R. Balhorn, et.al., "Frequency Stabilization of Internal Mirror Helium-Neon Lasers," Applied Optics, Vol II, No. 4, Apr 74.
2. Private conversation with Dr. Jan Hall, University of Colorado at Boulder and NBS, 24 Jan 83.
3. Lt Roger Facklam, "Ultra-Stable Laser Clock," presented at the 36th Annual Frequency Control Symposium.

APPENDIX B

AN ACTIVE RING LASER GYROSCOPE WITHOUT LOCK-IN:
CONCEPT, AND DESIGN

by

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At present, the limiting factor in active ring laser performance is related to the various methods used to get around the lock-in phenomenon. This phenomenon is directly attributable to the fact that two frequencies (10^{14} Hz) very close to each other (generally less than 10^3 Hz) are being amplified by the same amplifier at the same time. What occurs is simply that the frequencies lock together. This presentation discusses an approach that eliminates this problem. Simply put, this research demonstrates that inertial rotation can be measured by an active ring laser gyroscope without the lasing having to take place in both directions at the same time.

This active ring laser gyroscope concept is described with the aid of the diagram below.

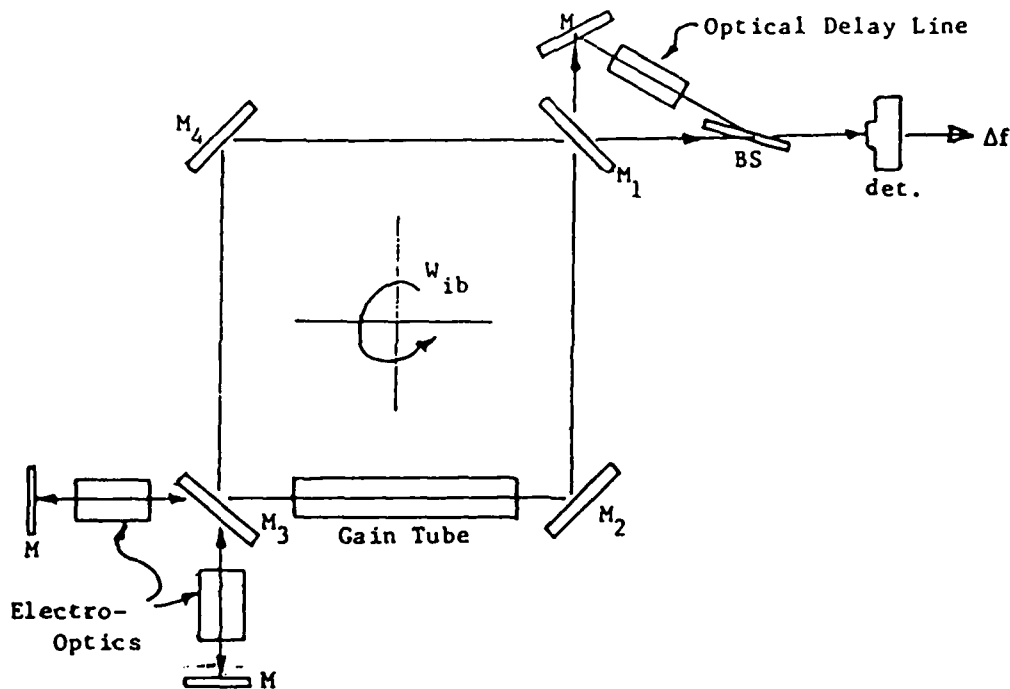


Figure 1. Gyroscope Schematic Diagram

In the diagram, a standard active ring laser is shown using the gain tube and the four mirrors M_1 , M_2 , M_3 , and M_4 .

First, the lasing action must be allowed to take place in only one direction at a time. This can be accomplished in several ways. In this diagram the technique used is that of reintroducing light into the cavity at the same frequency and polarization to cause destructive interference inside the cavity. Alternately, it can be viewed as introducing a directional loss in the cavity to stop lasing in either one direction or the other.

The electro-optic is used to rotate the polarization of the laser beam by 90° on reintroduction into the laser. This is accomplished by applying a $1/4$ wave voltage to the electro-optic. The polarization then changes from S (linear) to right hand circular. The reflection off the mirror then changes it to left hand circular which becomes p (linear) on passing through the electro-optic for the second time. Thus, with the electro-optic on the polarization is orthogonal to the polarization of the laser and therefore has no effect on the circulating beam (see Figure 2).

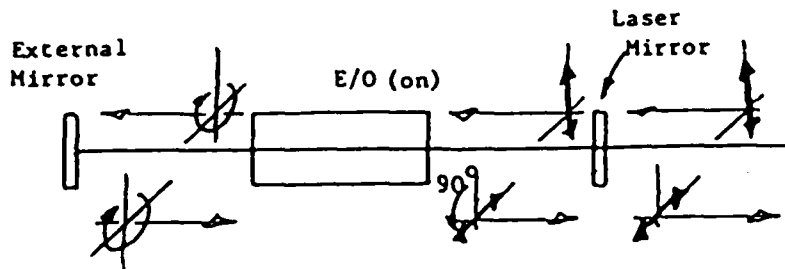


Figure 2. Polarization Changes for the System with the E/O "On"

With the E/O off there is no rotation and so the returning beam interferes and causes the lasing action to stop. In the ring laser there are two external E/O's. One for each direction which are alternately switched on and off.

If we look at the ring laser output at mirror M_1 in Figure 1, it would alternate as shown in Figure 3.

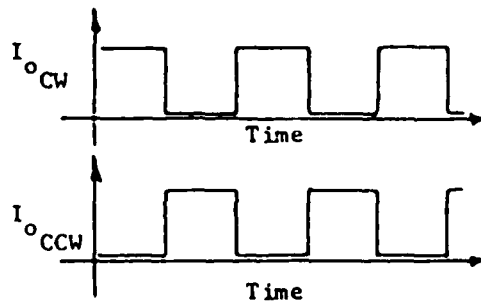


Figure 3. Output Beam Intensities at M_1

The lasing now takes place in only one direction at a time and so there will be no lock-in. There is no beat frequency at the detector because of the differing arrival times of the counter-rotating beams. To solve this problem a time delay must be introduced to time synchronize the two output beams. (See the optics at the mirror M_1 in Figure 1.) The delay time must be half the E/O switching time to produce the time synchronized output as shown below.

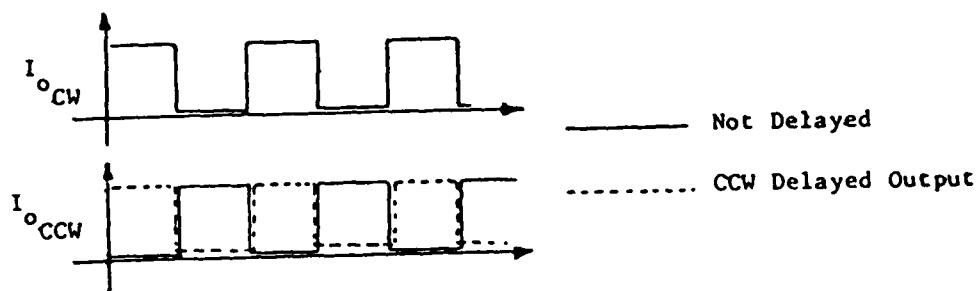


Figure 4. CCW Delayed Output in Synchronization with the CW Output

The delayed output in the CCW direction and the non-delayed output in the CW direction will not produce a beat frequency (f) at the detector.

The optical delay line is a length of glass (or plastic) fiber. For an E/O switching frequency of 1 MHz would require a half cycle delay of 0.5 usec. Using a glass fiber as the delay line implies a fiber length of 100 meters.

Recall that current active ring lasers sense beat frequencies on the order of 10 Hz or less in the face of mechanical and other noise sources of from 1 to 10 KHz on a carrier frequency of 5×10^{14} Hz! This measurement is possible because all noise sources are perfectly correlated in both the CW and CCW directions. This particular ring laser must retain the perfect noise correlation. This implies an E/O switching frequency of greater than 100 KHz, i.e., 10 times the highest noise frequency. The upper limit of the E/O switching frequency will be due to the cavity relaxation time. This will allow an E/O switching rate at least as high as 1 MHz. The output detector will not see the 100 KHz or greater switching frequency since it is only required to observe beat frequencies which are less than 20 KHz for most applications.

Experiments have already shown that the active ring laser can be controlled using the E/O's as described above and the use of fiber and/or a glass wedge as a time delay device for the frequencies involved here has already been demonstrated.

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