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ABSTRACT

Aeronautical Systems Division, ASD/SDML, requested assistance from the Air Force Packaging Evaluation Activity (AFPEA) to conduct certification and qualification testing on two new aluminum Maverick missile containers, the CNU-445/E and the CNU-447/E (see AFPEA Test Report 88-R-01). Part of the certification testing was the Performance Oriented Packaging (POP) testing on the container design.

The CNU-445/E and CNU-447/E containers were designed and fabricated by AD/YNP, Eglin AFB, FL, 32542-5000. The containers are environmentally sealed with a humidity indicator, desiccant port, and a pressure relief valve. Both containers are designed to protect one AGM-65A/B/C/D/E/F/G all-up-round Maverick missile during world-wide shipment, storage, and handling. The containers will also be used for one missile without the guidance unit and for one missile without the guidance unit and the hydraulic actuation system. The CNU-447/E is the Navy version and differs from the CNU-445/E only in some external Navy-specific handling features.

The test plan used for the container was derived from United Nations (UN) Standard (Ref. ICAD 4.3), UN "Transport of Dangerous Goods", and DOD Hazardous Materials Packaging Test Plan.

Results of the tests conducted on one CNU-447/E container were acceptable. The containers did successfully pass the POP tests, as prescribed by the UN test criteria.

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TABLE OF CONTENTS

	<u>Page</u>
ABSTRACT.....	i
TABLE OF CONTENTS.....	ii
INTRODUCTION	
BACKGROUND.....	1
PURPOSE.....	1
TEST SPECIMEN.....	1
TEST OUTLINE AND TEST EQUIPMENT.....	2
TEST PROCEDURES AND RESULTS.....	2
TEST NO. 1, UN DROP TEST.....	2
TEST NO. 2, UN STACKING TEST.....	3
CONCLUSIONS.....	3
TABLE I, CONTAINER TEST PLAN.....	4
FIGURE 1, CONTAINER CONFIGURATION.....	5
FIGURE 2, MISSILE ATTACHMENT AND TEST LOAD.....	6
FIGURE 3, CNU-447/E PROTOTYPE CONTAINER.....	6
FIGURE 4, FLAT DROP ON TOP SURFACE.....	7
FIGURE 5, CORNER DROP ON RIGHT REAR CORNER OF COVER.....	7
FIGURE 6, WELD CRACK FROM DROP ON FORWARD SURFACE.....	8
FIGURE 7, COVER DEFORMATION FROM CORNER DROP.....	8
FIGURE 8, INTERIOR DAMAGE AFTER TESTING.....	9
FIGURE 9, FOAM FAILURE IN CONTAINER BASE.....	9
DISTRIBUTION LIST.....	8

TEST OUTLINE AND TEST EQUIPMENT

TEST PLAN: Tests were conducted in accordance with table I. Test methods and procedures used were as outlined in UN Standard (Ref. ICAD 4.3), UN "Transport of Dangerous Goods", and DOD Hazardous Materials Packaging Test Plan.

TEST CONTAINERS: The tests in this report were performed on a prototype CNU-447/E container (see figure 3). Only one container was used for testing since the tests were severe and it would be too costly to provide a new container for each drop.

TEST LOADS: All tests were conducted using the heaviest missile the container was designed to hold. The test load was an inert missile weighing 675 pounds (see figure 2). A container base loaded to 1121 pounds was also used for test number 2 to simulate the stacking interface and weight of a stacked container. Steel plates and lead weights were used in test number 2 to simulate the stacked load.

TEST SITES: Testing was conducted at AFPEA, HQ AFLC/DSTZ, Building 70, Area C, Wright-Patterson AFB OH. The equipment required for testing was a forklift truck.

TEST PROCEDURES AND RESULTS

UN DROP TEST

Test No. 1: The CNU-447/E prototype container with the 675 pound test load was dropped from a height of 1.2 m as follows:

- a. Flat drop on bottom.
- b. Flat drop on top (see figure 4).
- c. Flat drop on left side (long side).
- d. Flat drop on forward end (short side).
- e. Corner drop on right rear corner of cover (see figure 5).

The container shall not spill its contents.

Results: The container was visually inspected following each drop:

- a. Bottom - several welds securing the skid to the container base broke.

b. Top - middle three latches on the right side of the container became unlatched. These were secured before the testing continued.

c. Left side - no additional visible damage.

d. Forward end - cover deformed and weld cracked at forward right corner of the cover on the right edge. Crack was two inches long (see figure 6).

e. Right rear corner of cover - desiccant port fell off and cover deformed at corner impacted. Cover also deformed on aft end along the flange (see figure 7). Both latches on the aft end became unlatched.

Visual inspection of the inside of the container after testing revealed the missile was still in the cradle. The cradle had become detached from the base of the container (see figure 8). Detachment was caused by foam shear and not adhesive failure (see figure 9). The forward end of the cover was deformed where the missile nose had impacted and the dome shattered. A weld crack was visible on the forward edge of the cover on the inside. The desiccant basket was deformed and weld cracks were found at the aft end of the cover.

The container did not spill its contents. Results of this test are acceptable.

UN STACKING TEST

Test No. 2: At ambient temperature, the container was subjected to a superimposed load of 2230 pounds through the stacking provisions for 24 hours. The container shall not permanently deform.

Results: The container dimensions were checked and no permanent deformation occurred during the stacking test. The results of this test are acceptable.

CONCLUSION

1. The container successfully passed the POP tests, as prescribed by the UN test criteria.

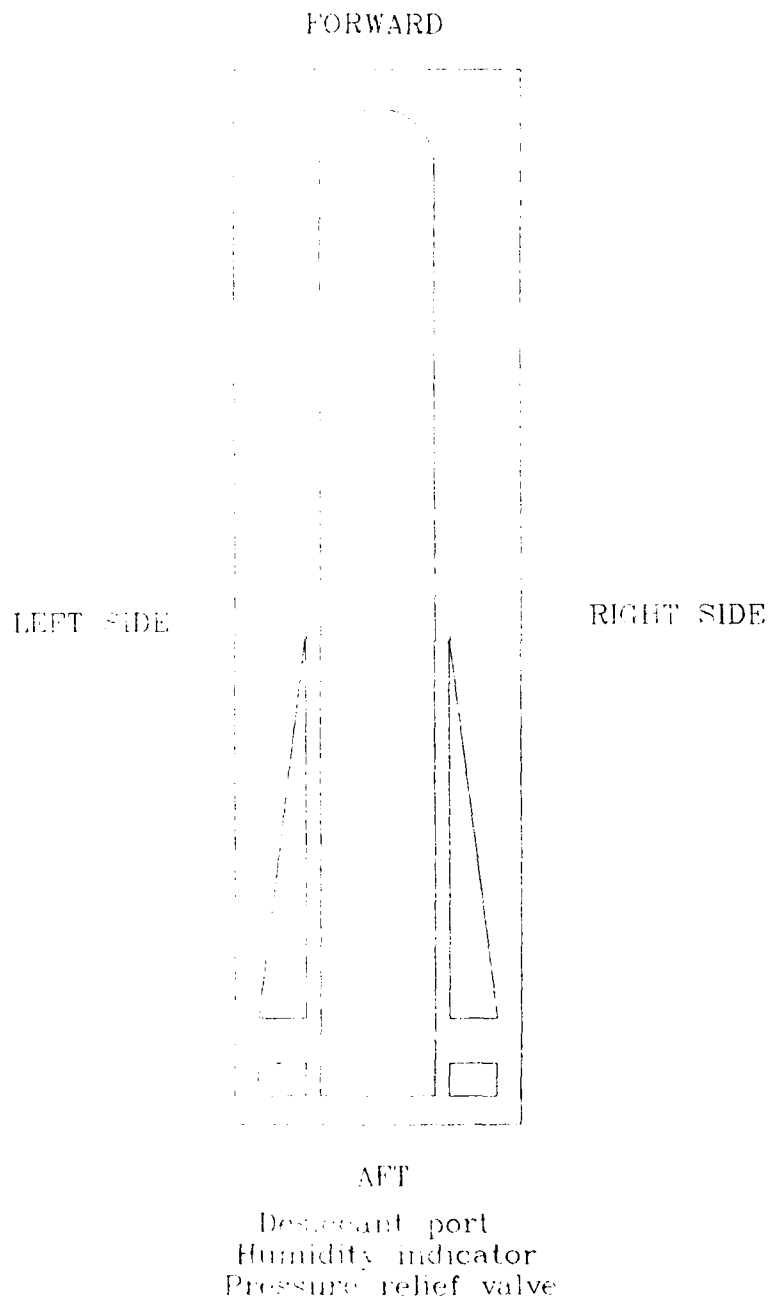


FIGURE 1.

TABLE I. Container Test Plan.

AIR FORCE PACKAGING EVALUATION ACTIVITY (Container Test Plan)				AFPEA PROJECT NUMBER	
CONTAINER SIZE (L x W x D)(INCHES)				87-P-130	
INTERIOR:		EXTERIOR:		QUANTITY	DATE
		108x28.5x30			7 Dec 88
ITEM NAME			MANUFACTURER		
AGM-65 Maverick Missile					
CONTAINER NAME				CONTAINER COST	
CNU-445/E and CNU-447/E					
PACK DESCRIPTION					
Aluminum Container					
CONDITIONING					
Ambient					
TEST NO.	REF STD/SPEC AND TEST METHOD OR PROCEDURE NO'S	TEST TITLE AND PARAMETERS	CONTAINER ORIENTATION	INSTRUMENTATION	
1.	<u>UN DROP TEST</u> UN "Orange Book", Requirement 9.7.3	Drop container from a height of 1.2 meters (3.94 feet), as required for Packing Group II. The container shall not spill its contents. A different container may be used for each drop.	One each flat drop on the bottom, top, long side, and short side; and a corner drop. Total of 5 drops. Test with heaviest All-up-round (AUR) missile.	Visual inspection	
2.	<u>UN STACKING TEST</u> UN "Orange Book", Requirement 9.7.6	Simulate stacking to a minimum height of 3 m (9.84 ft) for 24 hours. There shall be no permanent deformation.	Test with heaviest AUR.	Visual inspection	
COMMENTS:					
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Figure 2. Missile attachment and test load.

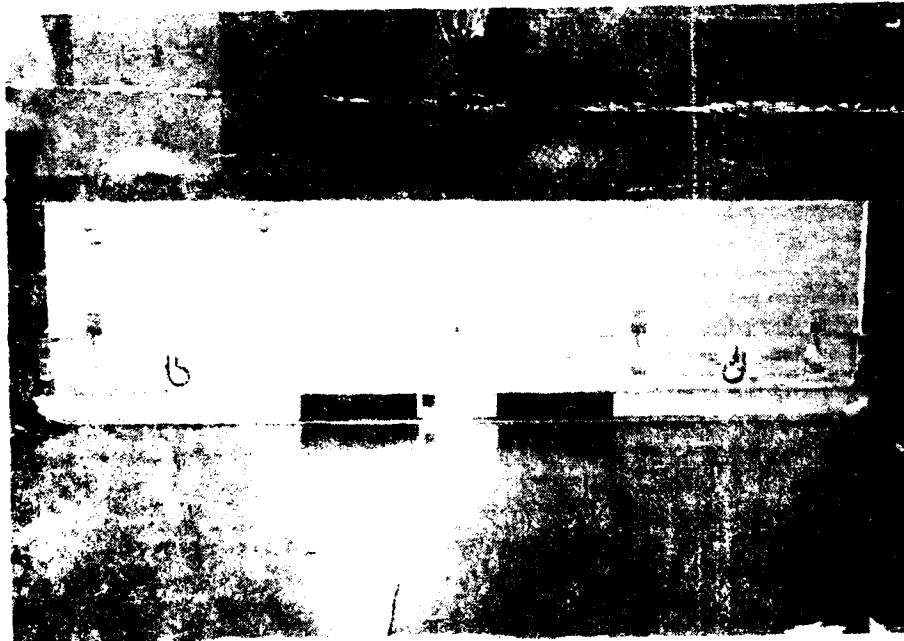


Figure 3. NU-44 /E prototype container.

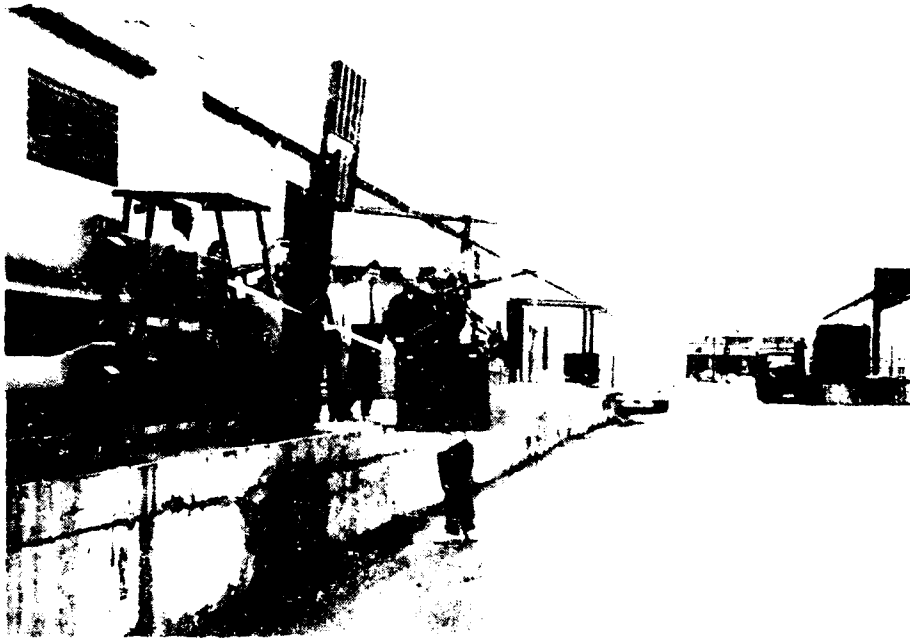


Figure 4. Flat drop on top surface.

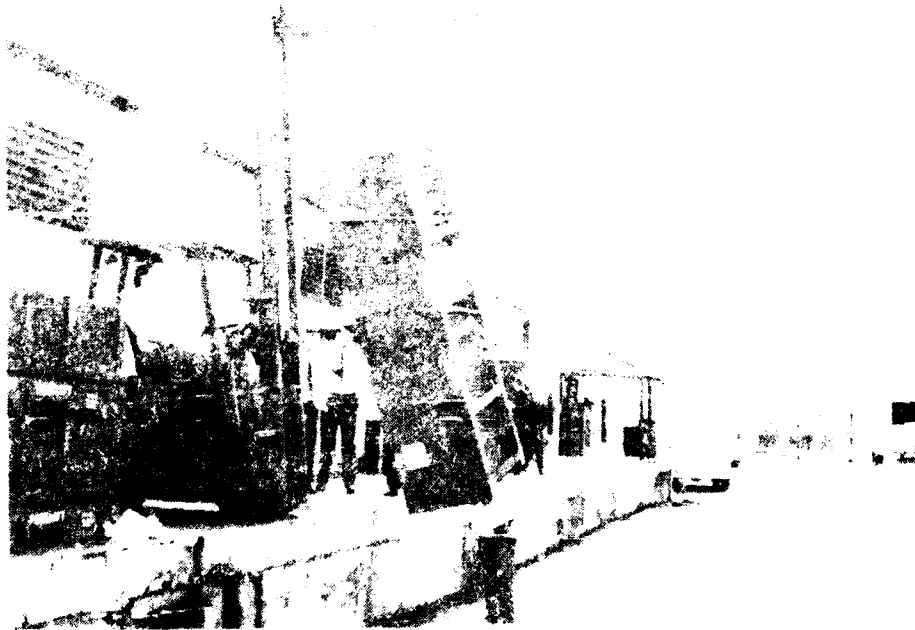
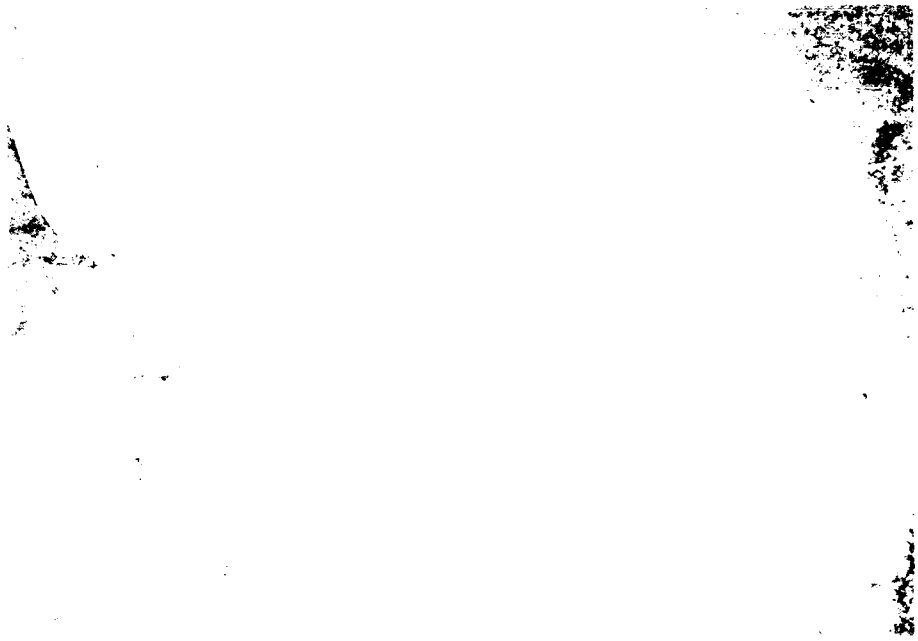


Figure 5. Corner drop on right rear corner of cover.



at ice.



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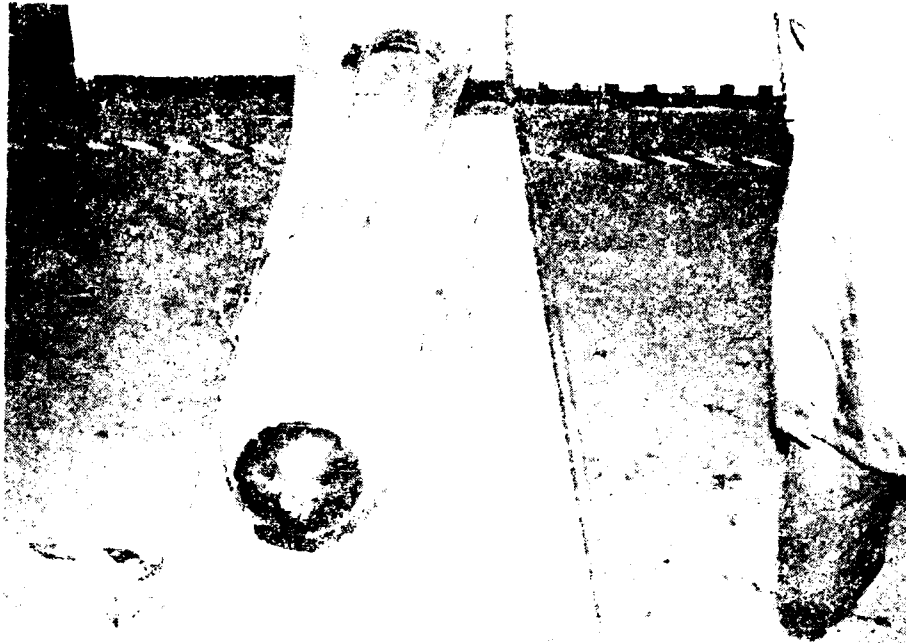


Figure 8. Interior damage after testing.

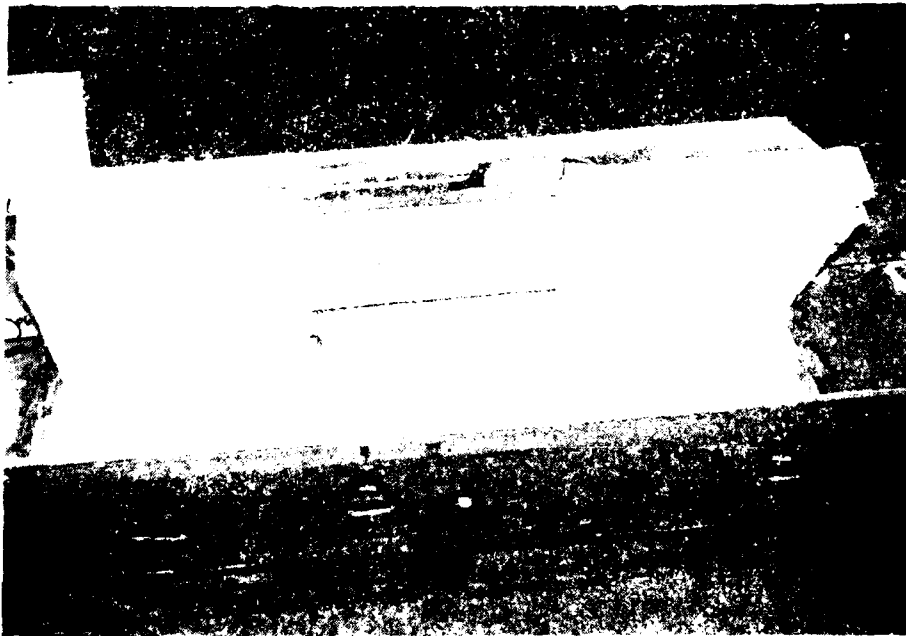


Figure 9. Foam failure in container base.

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