

AL-TR-1992-0014

AD-A262 417



ARMSTRONG
LABORATORY

HYBRID OXYGEN SYSTEM

W. David Lee

Arthur D. Little, Incorporated
Acorn Park
Cambridge, MA 02140-2390

CREW SYSTEMS DIRECTORATE
Brooks Air Force Base, TX 78235-5000

DTIC
ELECTE
MAR 24 1993
S E D

October 1992

Final Technical Report for Period September 1988 - December 1991

20000920245

Approved for public release; distribution is unlimited.

Reproduced From
Best Available Copy

98-3 18 096

268550
93-05766



977

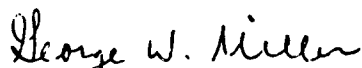
AIR FORCE MATERIEL COMMAND
BROOKS AIR FORCE BASE, TEXAS 78235-5000

NOTICES

When Government drawings, specifications, or other data are used for any purpose other than in connection with a definitely Government-related procurement, the United States Government incurs no responsibility or any obligation whatsoever. The fact that the Government may have formulated or in any way supplied the said drawings, specifications, or other data, is not to be regarded by implication, or otherwise in any manner construed, as licensing the holder or any other person or corporation; or as conveying any rights or permission to manufacture, use, or sell any patented invention that may in any way be related thereto.

The Office of Public Affairs has reviewed this report, and it is releasable to the National Technical Information Service, where it will be available to the general public, including foreign nationals.

This report has been reviewed and is approved for publication.



GEORGE W. MILLER, Major, USAF
Project Scientist



RICHARD L. MILLER, Ph.D.
Chief, Crew Technology Division



F. WESLEY BAUMGARDNER, Ph.D.
Chief, Crew Systems Branch

REPORT DOCUMENTATION PAGE

Form Approved
OMB No. 0704-0188

Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project (0704-0188), Washington, DC 20503.

1. AGENCY USE ONLY (Leave blank)		2. REPORT DATE October 1992		3. REPORT TYPE AND DATES COVERED Final September 1986 - December 1991	
4. TITLE AND SUBTITLE Hybrid Oxygen System				5. FUNDING NUMBERS C - F33615-86-C-4505 PE - 62202F PR - 7930 TA - 11 WU - 86	
6. AUTHOR(S) W. David Lee					
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) Arthur D. Little, Incorporated Acorn Park Cambridge, MA 02140-2390				8. PERFORMING ORGANIZATION REPORT NUMBER 56913-10	
9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) Armstrong Laboratory Crew Systems Directorate Brooks Air Force Base, TX 78235-5000				10. SPONSORING/MONITORING AGENCY REPORT NUMBER AL-TR-1992-0014	
11. SUPPLEMENTARY NOTES Armstrong Laboratory Technical Monitor: Maj. George W. Miller, (512)536-3361					
12a. DISTRIBUTION/AVAILABILITY STATEMENT Approved for public release; distribution is unlimited.				12b. DISTRIBUTION CODE	
13. ABSTRACT (Maximum 200 words) Investigation of concepts for generating oxygen on-board combat aircraft and development of a bleed air-driven refrigeration, liquefaction, and cryogenic storage system for oxygen were undertaken in this study. A number of alternative approaches were examined while considering size, weight and power consumption. An open-loop bleed air-driven system was selected for design, development, and testing. The bleed air-driven refrigeration unit achieved oxygen liquefaction temperatures of 90°K and liquefied and stored oxygen generated from a molecular sieve oxygen generating system (MSOGS). The oxygen was stored in cryogenic dewars, vaporized, and withdrawn from the system to simulate aircrew consumption. A heat exchanger flow reversing valving system was used to sublime and blow out condensates (water vapor and carbon dioxide) which normally collect in an open-loop refrigeration cycle operating from ambient air. The collection of condensate in the cryogenic system represented the largest technological area to overcome, and the reverse cycle system overcame the problem. The laboratory demonstrator utilized a helium cycle cold head refrigeration unit in conjunction with a J-T valve to simulate a cryogenic expander to be used in the flight system. Further work to incorporate a cryogenic expander in place of the cold head and J-T valve is recommended.					
14. SUBJECT TERMS on-board oxygen generation and liquefaction cryogenic storage of oxygen bleed air-driven liquefaction system				15. NUMBER OF PAGES 226	
				16. PRICE CODE	
17. SECURITY CLASSIFICATION OF REPORT Unclassified	18. SECURITY CLASSIFICATION OF THIS PAGE Unclassified	19. SECURITY CLASSIFICATION OF ABSTRACT Unclassified	20. LIMITATION OF ABSTRACT UL		

Table of Contents

1. Summary	1
2. Introduction	3
3. Phase I: Feasibility Study	5
3.1 Oxygen Requirements and Missions	5
3.1.1 Oxygen Backup for Combat Aircraft Crew	6
3.1.2 Demand Matching	8
3.2 Benefits of the Hybrid Oxygen System	9
3.2.1 Technical Risks	11
3.3 Hybrid System Refrigeration Trade Studies	12
3.4 Aircraft Integration	18
4. Phase II: Laboratory Demonstrator Design	21
4.1 Laboratory Demonstrator Objectives	21
4.2 System Schematic and Major Commercially Available Components	21
4.2.1 System Schematic	21
4.2.2 Heat Exchanger	21
4.2.3 System Deriving Function	24
4.2.4 Cold Head and Throttle Valve	25
4.2.5 Heat Balance	27
4.3 Piping Diagrams	28
4.3.1 Drawings	28
4.3.2 Pressure Losses	28
4.4 Mechanical System	31
4.4.1 Heat Exchangers and Liquid Storage Layout	31
4.4.2 Piping, Flow Controls and Instrumentation	33
4.4.3 Concentrator	33
4.4.4 Compressor	33
4.5 Electrical	33
4.6 Instrumentation	35
4.6.1 Temperature	35
4.6.2 Flow Rate	35
4.7 Safety Review	36
4.7.1 Safety Review Board	36
4.7.2 Materials	36
4.7.3 Pressure Certification	37
4.7.4 Modes of Failure	37
5. Phase III: Hybrid Oxygen Laboratory Test Procedure and Results	41
5.1 Test Program	41
5.2 Test Procedure	41
5.2.1 Instrument and System Checkout	41
5.2.2 Initial Cooldown (Room Temperature to 20°F)	41
5.2.3 Final Cooldown (270°K to 100°K)	41
5.2.4 Data Collection	42
5.3 Discussion of Test Results	42
5.3.1 Test No. 7 Results	42

6. Conclusions and Recommendations	53
6.1 Conclusions	53
6.2 Hybrid Oxygen - Turboexpander Demonstration	53
6.3 Flight System Design	53
6.3.1 Flight Heat Exchanger	54
6.3.2 Expander	54
7. Appendixes	57
7.1 Mission Scenario Specifications	58
7.2 Boeing Integration Report	69
7.3 Engineering Drawings	109
7.3.1 Mechanical and Piping Layout	111
7.3.2 Mechanical Drawings	115
7.3.3 Electrical Drawings	123
7.4 Essex Cryogenics Concentrator Performance	126
7.5 Spreadsheet Test Data	131
7.6 Test Data	139
7.6.1 Raw Test Data	139
7.6.2 Test Data Graphs	165
7.7 Nomenclature	213

Accession For	
NTIS CRA&I	☒
DTIC TAB	☒
Unannounced	☐
Justification	
By	
Distribution /	
Availability Codes	
Dist	Avail and/or Special
A-1	

DTIC QUALITY INSPECTED 1

List of Figures

Figure No.		
1	Fractional Oxygen Requirement vs. Cabin Altitude	7
2	Major Liquefaction Alternatives.....	13
3	Hybrid Oxygen System - Baseline System Layout.....	16
4	Bleed Air Refrigerator/Oxygen Liquefier System.....	17
5	System Schematic.....	22
6	Temperature Entropy Diagram of Process.....	26
7	Main Refrigeration Elements (Dewars not attached).....	30
8	Heat Exchanger and Liquid Storage Layout.....	32
9	Layout of Onboard Oxygen Enrichment System.....	34
10	Four-Way Reversing Valve System Schematic.....	43
11	Test Results Format Overview.....	44
12	Turboexpander Layout for Aircraft Oxygen Liquefier.....	55

List of Tables

Table No.		
1	Mission Profile in Minutes.....	5
2	Oxygen Breathing Requirements.....	6
3	Breathing Specifications.....	6
4	Fighter (MSOGS) Backup Requirements.....	8
5	Crew Stored Oxygen Available Liters/Man (NTP).....	8
6	Summary Specs--Backup/Peak Shaving Mission Scenarios.....	10
7	Boil-Off Replenishment (12-hr missions).....	11
8	Hybrid Oxygen Component Sizing.....	23
9	Mass Fraction Concentration of Condensibles	25
10	Expander Comparison.....	27
11	Heat Leak into Cold End in Watts.....	28
12	Heat Exchanger Design.....	29
13	System Pressure Losses in PSI.....	31
14	HOS Instrumentation	33
15	F&P Flow Meter Calibrations.....	35
16	Beckman Oxygen Analyzer.....	36
17	Safety Inspection List.....	37
18	Pressure Certification.....	37
19	Hazard Signals (Manually Monitored)	38
20	Valve Sequencing Hazards	39
21	Cooldown Time.....	46
22	Comparison of Longitudinal Conduction 2"x4" Heat Exchanger	54
23	Flight Heat Exchanger Size.....	54

Preface

This report was prepared under contract F33165-86-C-4505 to USAF Armstrong Laboratory. It represents the findings of a three-phase program to demonstrate the feasibility of a hybrid oxygen generation and storage system for combat aircraft. The first and second phases of the program centered on a study of alternative approaches to achieving the objective and the design of a laboratory demonstrator, respectively. Phase III was the culmination of the program in which a laboratory demonstrator was fabricated and tested.

The technical manager for USAF Armstrong Laboratory and point of contact was:

Major George W. Miller
Armstrong Laboratory (AL/CFTS)
Human Systems Division
Brooks Air Force Base, Texas 78235-5000
Tel. 512/536-3361

for Arthur D. Little, Inc.:

W. David Lee, Vice President and Managing Director
Technology and Product Development
20 Acom Park
Cambridge, Massachusetts 02140-2390
Tel. 617/864-5770

HYBRID OXYGEN SYSTEM

1. Summary

This program was conducted to examine the feasibility of liquefying and storing oxygen generated by a molecular sieve oxygen generating system (MSOCS). The principles of the liquefaction system were proven in this laboratory-based development using commercially available cryogenic components.

Generation and storage of oxygen may be important in a variety of military operations. This study is focused on the generation and storage of oxygen on-board aircraft. Engine bleed air is the feed gas for the molecular sieve oxygen generation unit and the liquefaction unit. A compressed air based liquefaction system was selected for this application for the following reasons:

- utilization of aircraft engine bleed air (compressed air) directly to produce refrigeration is the most efficient method for developing the required liquefaction;
- compressed air is a safe and readily available working fluid for the desired refrigeration system; and
- compressed air refrigeration cycle is tolerant of system leaks unlike the standard hermetic closed cycle refrigeration systems and may be more reliable and require less maintenance.

A variety of refrigeration systems were considered (see section 3.3). It was concluded that the best aircraft system from the basis of size, weight, and power consumption would be a liquefier driven solely by bleed air from the engine. Several mission profiles were examined, and it was concluded that a 5-liter storage system for fighter aircraft and a 10-liter system for transports would be appropriate for the missions. A Laboratory Demonstrator, capable of storing up to 10 liters of oxygen, was designed and successfully tested.

The major design and development issue identified during the course of the work was the steady deriming or removal of condensible buildup from the bleed air in the cryogenic system with a technique that would not adversely impact the system operation. A reversing derime cycle is an elegant technique in which the heat exchanger process stream is reversed, thus venting the high pressure side where condensibles have collected to low pressure and gradually vaporizing these into the exhaust stream. This (deriming) approach does not interrupt the refrigeration cycle or the rate of production of the liquefier. A more radical approach considered was a hot gas defrost in which hot bleed air is brought directly to the cold end of the heat exchanger, and all condensibles are quickly vaporized. As will be discussed later, the deriming cycle worked exceedingly well, and the defrost approach was never attempted.

Once the fundamental cycle with periodic deriming of the bleed air refrigeration system was proven, the demonstration of liquefaction, storage, revaporization, and delivery of high purity oxygen was the next test objective. Two 5-liter capacity dewars were successfully filled to the desired level, and then gaseous oxygen was withdrawn simulating the crew demand.

Ninety percent oxygen was generated by the MSOGS unit; the oxygen was liquefied at about 90°K and stored in the dewars. During liquefaction, nitrogen was separated from the oxygen stream as demonstrated by the consistently lower concentration of oxygen in the gas vented from the filling dewars than the oxygen supplied. During the withdrawal of oxygen from the dewar the initial concentration was approximately 90% to 92%, and the concentration at the end of the withdrawal was 98% to 100%.

2. Introduction

This study is focused on integrating an MSOGS which will generate an inexhaustible supply of oxygen and an on-board liquefier into a self-contained system—the **Hybrid Oxygen System (HOS)**.

The logistic benefits of producing oxygen on-board aircraft and at remote locations with molecular sieve technology has generated considerable interest in the military. Currently, several aircraft are equipped with MSOGS.^{1,2} The ability of MSOGS to produce oxygen in a safe, reliable fashion has created considerable interest. Further use of this oxygen generation technology could occur if a reliable and safe technique for storing large quantities of oxygen could be developed. Compression of oxygen for high pressure storage is hazardous and difficult to accomplish because standard compressor lubrication techniques are unacceptable with oxygen. Lubricants compatible with oxygen or nonlubricated compressors would need to be developed. Alternatively, gaseous oxygen generated by an MSOGS system can be liquefied by refrigeration at near-atmospheric pressure and stored.

A three-phase study was sponsored by the Armstrong Laboratory, AL/CFT, Crew Technology Division, Brooks Air Force Base (AFB), Texas, to demonstrate the feasibility of an HOS which would liquefy and store oxygen generated by an MSOGS. The scope of the work is summarized below.

Phase 1: The mission requirements and design trades for different types of HOS were examined. A variety of refrigeration approaches were investigated. Alternative refrigeration approaches are discussed in Section 3.3 (Chapter 3).

Phase 2: A Laboratory Demonstrator was designed with commercially available components to demonstrate the feasibility of the HOS. Also, a preliminary flight system was designed (Chapter 4).

Phase 3: The Laboratory Demonstrator was constructed and successfully tested (Chapter 5).

Computer simulations of the system components and laboratory test raw data, analyzed data, and graphics are contained on the Hybrid O₂ Master 3.5-inch diskette (available from AL/CFT) and are accessible with Lotus 1-2-3, Version 2 or later, or Quattro Pro, Version 2 or later.

¹ Routzahn, Richard L., "An Oxygen Enriched Air System for the AV-8A Harrier," Report No. NADC-81198-60, Naval Air Development Center, Warminster, PA (1981).

² Tedor, John B. and James Clink, "Manufacturing the B1-B Molecular Sieve Oxygen Generation System," Report No. USAFSAM-TR-87-4, USAF School of Aerospace Medicine, Brooks AFB, TX (1987).

3. Phase I: Feasibility Study

Essential questions of the method of liquefaction and storage of oxygen generated by the on-board MSOGS had to be answered. Phase I of the program was initiated to identify alternative system designs and choose the most promising. A systematic approach was adopted which started with the identification of the oxygen requirement and then selection of the liquefaction system and its design.

3.1 Oxygen Requirements and Missions

The oxygen requirement for crew and passengers is dictated by the type of aircraft and its missions. Two main classes of aircraft were chosen:

Combatfighter - Cabin pressure decreases with altitude requiring the use of oxygen masks to sustain about 195 mmHg of oxygen partial pressure to each individual.

Transport and Bomber - Cabin pressure is maintained at 8,000 ft cabin altitude throughout this mission, oxygen masks are needed for therapeutic uses (aeromedical evacuation) and for emergency conditions.

The mission duration will determine the total oxygen requirement, and typical missions are shown in Table 1.

Table 1. Mission Profile in Minutes

Aircraft No. of Crew	Fighter 1 or 2	Transport 9	Bomber 6
Pre-engine time on O ₂	15-30	-	15
Ground hold	15	30	15
Flight time	100	600	450
Ground hold	N/A	120	N/A
Return time	N/A	600	N/A

All aircraft have stored oxygen as either liquid oxygen (LOX) or high pressure bottles for use as the primary or secondary oxygen source.

Oxygen flow volume requirements are dictated by human oxygen consumption and ventilation rates. While the actual metabolic consumption of oxygen at modest levels of activity may be 1 to 2 LPM normal temperature and pressure (NTP), the volume of respired air required is substantially greater, probably 5 to 10 LPM. In simulated air combat, the O₂ uptake may reach an average of 5.3 LPM with peaks of 10 LPM (Table 2).

Table 2. Oxygen Breathing Requirements

Activity	Metabolic O ₂ Uptake LPM (NTP)	Ventilatio., Rate LPM (NTP)	Breaths/ Minute ³
Rest	1	6	12
Modest Activity	2	12	24
Combat and G's Average	5	32	64
Peak Activity (NATO)	10	50	
Instantaneous Peak Flow	N/A	150-200 ⁴	

Published oxygen flow rates for military aircraft are summarized in Table 3. The variability in the MIL-D-8683B accommodates increased activity. These flow rates are consistent with the time averaged values derived from known physiological O₂ uptake requirements.

Table 3. Breathing Specifications

Publication	Published Breathing Volume Requirements LPM (ATP)
MIL-D-8683B (Baseline & high activity)	13.3 - 28
B-1B MSOGS Specification	13.3 - 26.7 Max
F-16 (Rated LOX usage)	12.0
B-1B Backup Usage	12.4
NATO (STANAG 3865)	25 ⁵ - 50

Oxygen concentration requirements are generally consistent with the concentrations of the CRU-73 oxygen regulator, though variability between publications can be found. Figure 1 shows the MIL-D-8683B oxygen requirements. Curve B of this figure is the design concentration at each cabin altitude.

3.1.1 Oxygen Backup for Combat Aircraft Crew The current MSOGS systems are generally supported by a 200 liter (NTP) per man high pressure backup. Ideally, a backup supply of at least 850⁶ liters (NTP) per crew (MIL-D-8683B) for a 100-minute mission would be desired as shown in Table 4. To meet North Atlantic Treaty Organization (NATO) recommended levels of breathing, the backup requirement would probably be closer to 1,600 liters (NTP) per man. For a two-man crew, this amounts to a backup requirement of 3 to 4 liters of LOX.

³ Assuming .5 liters/breath.

⁴ NATO specification of 198 LPM ATPD for peak inspiration.

⁵ Minimum required at low bleed air pressure.

⁶ This value is derived for the 100-minute mission in Table A, Appendix 7.1, and provides 100% mission backup in the event of decompression and loss of MSOGS.

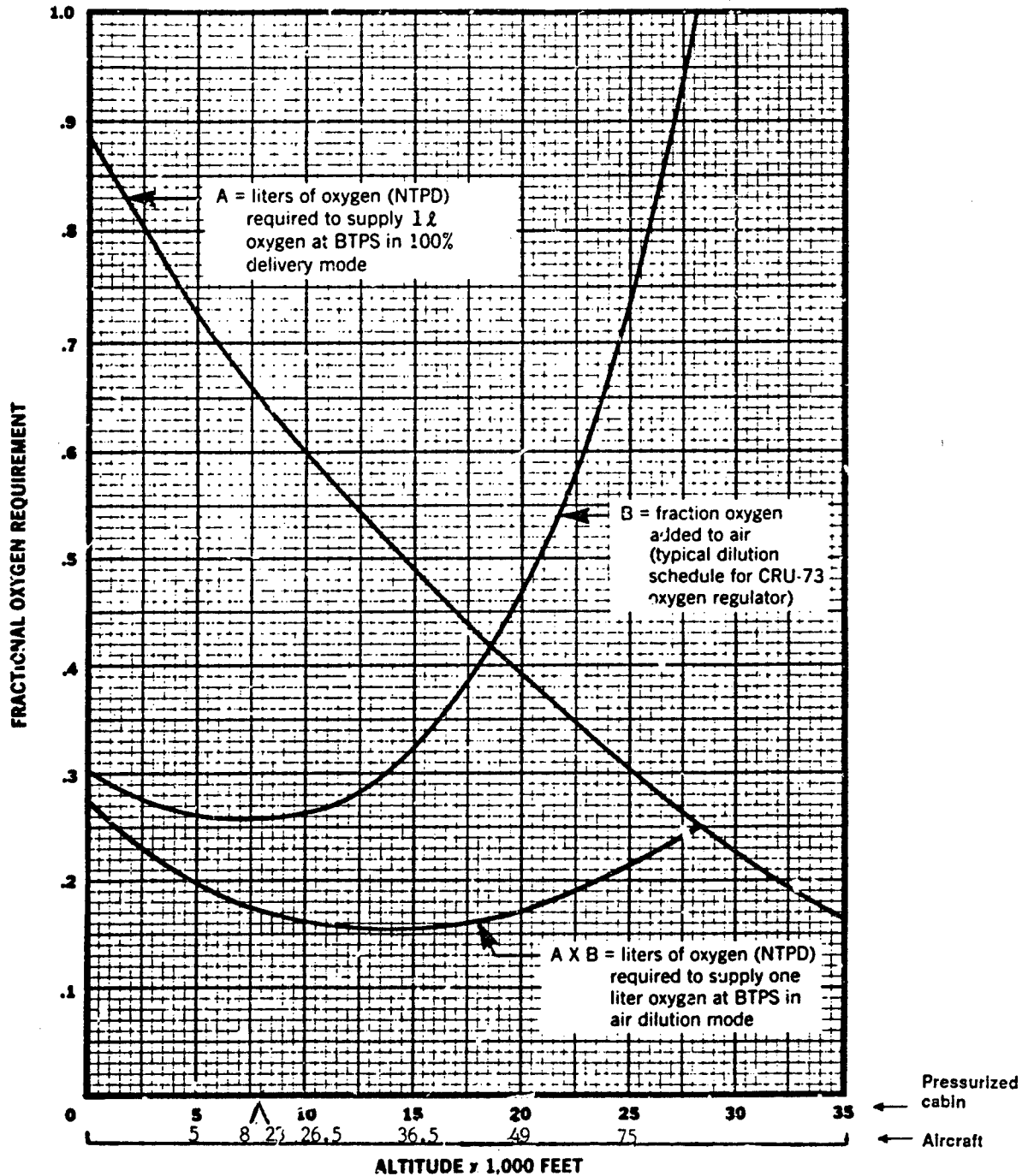


Figure 1. Fractional Oxygen Requirement vs. Cabin Altitude.

Table 4. Fighter (MSOGS) Backup Requirements

	O₂ Backup Liters @ NTP	Weight of High Pressure Bottle for Two Men
Practice	200 l/man	14 lbs
MIL-D-8683	850 l/man	26 lbs
NATO STANAG 3865	1,600 l/man	52 lbs

Backup for a transport crew amounts to 50% of the mission duration and is typically 25 liters of LOX (19,000 L NTP) which corresponds to more than 5 h of mission or more than one-half the total mission to comply with MIL-D-8683. The HOS will address this need.

Table 5 shows the typical stored supply of oxygen used with current MSOGS as compared to MIL-8683B requirements and to standard amounts carried as LOX. Higher back-up volumes are desirable to meet the total mission requirements for combat aircraft. An HOS could provide the backup oxygen through generation and storage of oxygen.

Table 5. Crew Stored Oxygen Available Liters/Man (NTP)

	Fighter	Bomber	Transport
Typical MSOGS Backup	200	550	Not established
MIL-8683B	850		
Standard LOX	2,200		2,100

Passenger and Therapeutic Oxygen or Aeromedical Evacuation Requirements

Stored oxygen requirements for transport aircraft are either:

- backup in case of cabin decompression or
- therapeutic (aeromedical evacuation).

The volume of passenger oxygen required is given as:

Passenger backup 2-75 liter LOX or 130,500 L-NTP

Therapeutic 15 liters LOX or 13,000 L-NTP

An HOS could be developed to provide part or all of the passenger or aeromedical evacuation requirements.

3.1.2 Demand Matching Another key function of the HOS would be to store oxygen to level the demand. A smaller MSOGS unit designed to meet the average (13 LPM/man) consumption rather than the peak (50 LPM/man) could be used.

The peak consumption rate creates a considerable problem at lower altitudes. An HOS storage system could draw on the stored liquid for peak demands and replenish the supply while at higher altitudes or lower O₂ consumption.

3.2 Benefits of the Hybrid Oxygen System

Logistic Autonomy

The HOS was conceived to achieve complete logistic autonomy for MSOGS while providing convenient LOX storage for reliability. Interviews with F-16 Special Project Office (SPO), AV8 SPO and others confirmed the perceived benefit of the stored LOX without the negative logistics. The HOS will eliminate routine servicing of the MSOGS backup oxygen system. Currently, the MSOGS backup oxygen system requires routine servicing due to usage and leakage. The HOS allows the crew to draw on the LOX at their own discretion without jeopardizing the backup volume necessary to meet the mission.

Engine OFF Supply of Breathing Gas

Pilots like to run breathing gas immediately upon entering the aircraft and prior to the operation of the engines. During this period, the MSOGS is not providing breathing gas and oxygen must be drawn from either a high pressure storage backup or LOX storage. The current oxygen demand of fighter aircraft is about 15 minutes of ground hold with the engine off. An HOS could meet this need with mission-to-mission storage of produced oxygen. With aircraft layovers of less than 3 to 5 days, the vaporization loss of the stored LOX will leave sufficient liquid oxygen to meet the preflight breathing needs.

Inflight Backup

The HOS will provide an inflight backup source of oxygen to allow the pilot to complete the mission should difficulties with the MSOGS occur.

Peak Shaving or Load Leveling

New NATO requirements indicate a need for 50 liters/min/man rating of MSOGS systems in order to meet peak combat crew oxygen breathing requirements while both MIL-D-19326F and MIL-D-8683B require an average of 13.3 liters/min. The HOS can be used to provide load leveling so that the MSOGS need only be sized for the mean load of 13.3 LPM ambient temperature and pressure (ATP). This leveling dramatically reduces its capacity requirement from 50 liters/min to approximately one-third that capacity.

To assess the possible benefit of the HOS, a fighter and bomber mission scenario was developed, and the system performance was simulated. Table 6 is a summary of the analysis, and additional tables (A-H) in Appendix 7.1 provide the analysis assumptions and the detailed data. The analysis includes the fighter and bomber on dilution, 95% oxygen at cabin altitude or 95% oxygen under decompression conditions. The simulation is contained on the Master disk under the file name HYBRIDOX.WK1.

Table 6. Summary Specifications Comparison for Backup and Peak Shaving Mission Scenarios

Table			% Oxy at Start	Liquefier Capacity g/s	System Size in Liters Storage, HX			System Weight in lbs	
					Std MSOGS	Hybrid System	Dewar	Std MSOGS	Hybrid
A	Fighter	Dilution air	75	.45	26	11	1.5	36	33
B	Fighter	95% oxygen	75	.9	54	20	3.2	84	47
C	Fighter	Decompression	75	.45	44	13	2.2	60	36
D	Fighter	Decompression	~0	.6	34	16	2.2	47	38
E	Bomber	Dilution air	75	.9	104	26	12	158	74
F	Bomber	95% oxygen	75	2.0	303	62	28	436	163
G	Bomber	Decompression	75	1.8	195	44	22	279	122
H	Bomber	Decompression	~0	1.8	144	49	22	211	86

Three cases for oxygen backup were examined: two missions in which dilution or 95% oxygen were consumed at cabin pressure; and another case in which cabin decompression and consumption at 13.3 LPM ATP are assumed. The decompression case represents the reasonable backup requirement and is the system design scenario.

The standard MSOGS includes the concentrator and the backup high pressure oxygen storage vessel. The HOS includes the concentrator, liquefier, and dewars for storage of the LOX.

Liquefier and Storage Dewar Size

The weight and volume benefit of the HOS designed to meet a cabin decompression is 40-50% weight reduction and 70-80% volume reduction over a standard MSOGS with high pressure bottle backup. For the fighter aircraft, two one-liter dewars integrated with a .5 g/sec liquefier provide sufficient backup and load leveling, while two ten-liter dewars are required for the bomber along with a 2 g/sec liquefier.

Aeromedical Evacuation

Considerable on-board oxygen is required for aeromedical evacuation transport aircraft for crew and passenger backup usage and as therapeutic oxygen. Crew and passenger backup usage is infrequent, and the current practice is to use one 25 and two 75 liter LOX dewars as backup. These dewars must be replenished on each flight due to boil-off (about 8 liters per day total). A small HOS could be used to maintain the stored LOX at desired levels, eliminating the regular logistic demand for LOX, depending on the interval between flights (the flights per week). Table 7 summarizes the size of the oxygen boil-off replenishment system as a function of the flights per week assuming the maximum boil-off of MIL-D-19326F. As the frequency of use increases, the dewars are replenished more often, and the size of the hybrid system is reduced. The system sizing assumptions for the aeromedical evacuation are the same as those used in Table 6 and are summarized in Appendix 7.1.

Table 7. Boil-Off Replenishment (12-hour missions)

Flights/Week	Liquefier LPM-NTP	Hybrid System Weight (lbs)
1	67	88
3	23	41
5	14	31
7	9	28

Retrofit

The HOS can supply oxygen pressures of 70 psi, hence, high pressure regulators can be used. Presently, MSOGS require special low pressure regulators.

High Purity

The HOS allows the MSOGS to run at high purity (about 95%) continuously, and peak oxygen demands are met by drawing on storage. The MSOGS concentration declines with flow rate above about 40 LPM normal temperature and pressure (NTP) at ground level. Oxygen can be liquefied for later higher demand, while the MSOGS is delivering high purity oxygen for current usage.

Electronics Cooling

Electronics cooling can be augmented by the use of this system. We estimate about 500 w of 40°F dry air will be available as excess cooling from the HOS.

Pilot Choices

The HOS allows the pilot to freely choose dilution or 95% oxygen during the mission because of the large stored volume of liquid oxygen. With a conventional MSOGS, capacity and concentration are linked to the performance of the MSOGS and the bleed air supply pressure. With the HOS, the stored oxygen allows the pilot to switch to 95% concentration while at high demand rates.

3.2.1 Technical Risks There are technical risks in the development phase as well as in the deployment phase of the HOS.

The major technological risk in the development phase is the performance and integration of the turboexpander envisioned for the bleed air liquefier. At the mass flow rates expected for the 40 LPM (NTP) liquefier, a very small high-speed turboexpander would be required. Expanders of this class are currently being developed by Creare, Inc., Hanover, NH, for cryogenic cooler applications in other military systems.

A miniature turboexpander concurrently was being developed by Creare during this project. Hence, we demonstrated HOS feasibility with standard components which simulate a turboexpander.

The second development risk is one of system failure due to contamination. Most cryogenic systems are closed cycle refrigerant systems in which the refrigeration fluid is a pure substance and continuously recycled inside of a hermetic unit. The bleed air-based liquefier is constantly processing ambient air, and along with it, considerable contaminants. Water vapor, carbon dioxide, and other condensibles will frost in the heat exchanger and ultimately deteriorate the

performance. This contamination process is inherent in the design, but can be used as a very important breathing gas purifier in a properly designed system. In this effort, management of the contaminants by the periodic deriming of the heat exchanger appeared successful.

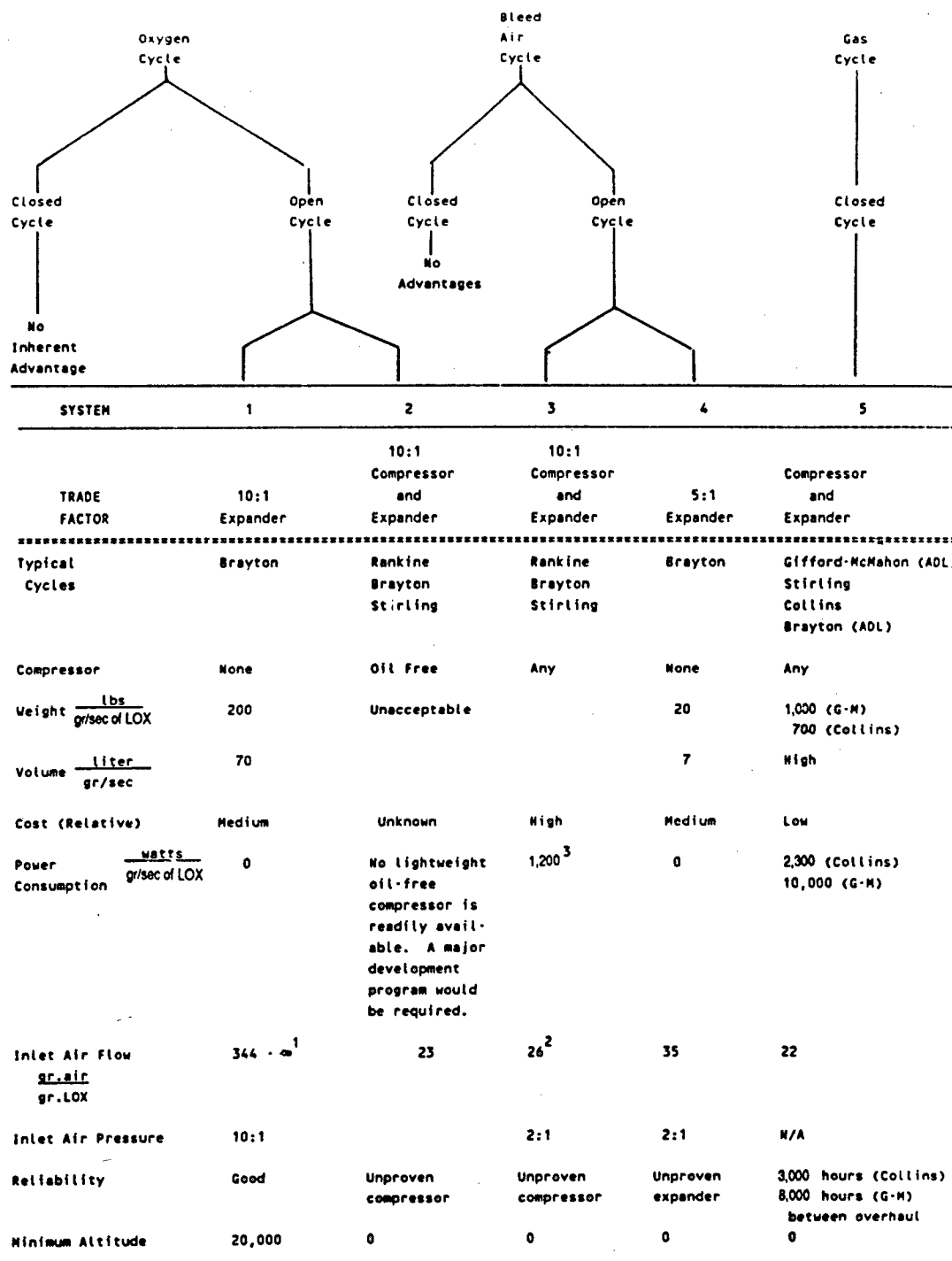
3.3 Hybrid System Refrigeration Trade Studies

The fighter aircraft application was chosen as more suitable for laboratory demonstration. This application requires a baseline module liquefier of 20-40 LPM-NTP (.45 to .9 g/sec). A number of refrigeration approaches were examined to provide the necessary liquefaction.

Figure 2 gives the summary comparison of alternative systems. There are six typical cryogenic refrigerator or liquefaction cycles that were considered.

- Reverse Brayton
- Reverse Rankine
- Stirling
- Gifford-McMahon
- Collins
- Closed Cycle Brayton

In general, the appropriateness of the cycle will be dictated by the availability or feasibility of components required. For instance, in column 2 the Rankine, Brayton, and Stirling cycle systems require a compressor which is probably infeasible for this project, since no oil-free, high pressure ratio oxygen rated compressor of a flight weight construction has been developed. Available oxygen compatible diaphragm compressors have limited life and are large mechanical systems probably weighing on the order of several hundred pounds for a relatively small flow volume. The Reverse Brayton cycle in an open cycle configuration is probably the approach most consistent with aircraft refrigeration. This cycle is used for environmental control on aircraft and offers the major advantage of small lightweight machinery. The Reverse Brayton cycle can take advantage of available bleed air for producing refrigeration, and, unlike the closed cycle units, will not degrade over time due to the gradual leakage loss of refrigerant which is generally encountered with closed cycle units, particularly helium-charged cryogenic refrigerators.



¹ 213 J/g liquefaction and 6.1 J/g sensible cooling from O₂ cycle at 82°K with a 97% expander and 20% O₂. Otherwise, a two-phase expander would be required.

² 213 J/g liquid O₂ liquefaction and 59.7 J/g cooling air cycle (84°K) = 3.5 $\frac{\text{grams bleed}}{\text{gram/LOX}}$ with 75% expander plus 22 $\frac{\text{g/air}}{\text{g/LOX}}$ to produce the electric power.

³ 125 J/g air compressor power x 6 g/gLOX.

Figure 2. Major Liquefaction Alternatives.

Compressor

Several of the systems (2 and 3) require a compressor. A cycle operating on the oxygen delivered from the concentrator and requiring additional compression will have substantial difficulties because it must be oil-free for safety reasons. Oil-free oxygen compressors capable of 10:1 compression ratios are highly specialized custom systems and not readily adaptable to lightweight operation. On the other hand, a compressor used in a bleed air system or a closed gas cycle can use any type of compressor including an oil-flooded unit because the hazard of explosion is eliminated. However, cleaning up of the entrained oil from the processed gas must be undertaken so that freeze out does not occur at the cold end. In many of the commercially available systems, the oil cleanup system is as large and as expensive as the compressor and expander combined. These deterrents require the addition of ancillary components, hence, making the compressor-free system alternatives (column 4) more attractive.

Newly available small high-speed turbocompressors can run oil free, but would require about five to eight stages of compression to raise the oxygen pressure from 5 ATM to 10 ATM, the pressure necessary to produce LOX directly. This type of system would certainly involve a major development.

Weight

Weight will depend on the cycle chosen and the need for a compressor and gas cleanup system. If the alternatives requiring a small turboexpander (columns 1 and 4) are used, the system weight is likely to be small relative to the other alternatives. An off-the-shelf closed cycle refrigerator using a standard oil-lubricated compressor and gas cleanup system may weigh as much as 1,000 lb/gram/second of LOX delivered. This result compares to the weight of an open cycle bleed air system of well under 30 lb/gram/second.

Volume

The relative volume of the alternative systems will be very similar to the relative weight of the systems. Turboexpanders will have extremely small rotors (1 inch diameter or less) for this application while the heat exchangers are likely to represent the largest component in the open cycle systems (1 and 4). A reciprocating expander has the advantage of much higher efficiency over a wider pressure ratio which means that less bleed air would be required for the open cycle systems than that shown in Figure 2. In addition, control start up and transient response of the expander will be more flexible than the very high speed turboexpanders which would be employed in such a small system. The reciprocating expander would, however, be heavier and bigger in volume than the turboexpander equivalent.

Cost

The ultimate cost of the system hardware will have two components: an initial development cost and a production cost. The values shown in Figure 2 are estimates of the relative cost of the different systems considering both development and production costs.

Power Consumption

The auxiliary electric power consumption for the different systems are dramatically different one from another. Systems 1 and 4 capitalize on the use of available bleed air pressure for the operation of the refrigerator or liquefier while Systems 2, 3, and 5 will require input power. The power requirements for a stand-alone refrigerator are quite high and, in conjunction with the weight penalty, make these Systems (2, 3, and 5) quite unattractive as compared to the open cycle Brayton systems.

Inlet Airflow Requirement

While Systems 1 and 4 do not require electric power consumption, they do require bleed air to operate the refrigerator or liquefier. While it is not intuitively obvious, there is a considerable difference between the two systems operating on the same bleed air pressure, System 1 employing oxygen after the concentrator and System 4 employing straight bleed air before the concentrator.

System 4 is essentially a nitrogen refrigerator and operates at refrigeration temperatures (79°K) below that of an oxygen-based working fluid for the same operating pressures. As a consequence, System 4 is about 17 times more efficient in the use of bleed air for producing LOX. In this system, the air would be used to produce a cold surface on which oxygen from the concentrator would be condensed.

System 1 operating on oxygen from the concentrator is likely to produce very little refrigeration for liquefying and storing oxygen at 1 ATM assuming realistic figures of 60%-70% expander efficiency (turboexpander at these pressure ratios). Systems 2, 3, and 5 use electric power, which can be derived from bleed air as well. We have assumed 54 w of electric power per 8 g/sec of 5 ATM bleed air.

Inlet Air Pressure

System 4 once again offers an advantage in that liquefaction can proceed at very modest inlet air pressures, though the amount of refrigeration would be proportional to the available pressure ratio, whereas System 1 must have an inlet pressure ratio of over 10:1 and then requires a nearly ideal expander to produce any refrigeration whatsoever.

The selection of the open cycle bleed air system (column 4), as the baseline system, was based on the forecasted weight and bleed air consumption. We believe that the system chosen is clearly more suitable for aircraft application because it will have the least impact on size.

Reverse Brayton Baseline System

Figure 3 shows the baseline system layout with the MSOGS conventional plumbing interfaced with the HOS.

Figure 4 shows the baseline system Temperature Entropy Diagram at the nominal design operating conditions. An inlet bleed air temperature of 300°K (70°F) or 533°K is assumed. The oxygen is cooled by heat exchange with the process stream and liquefied by the low pressure cold gas produced by the expander (point 3 of the cycle). The bleed air (process gas) is cooled by heat exchange to 120°K and then further cooled by the expander to about 84°K .

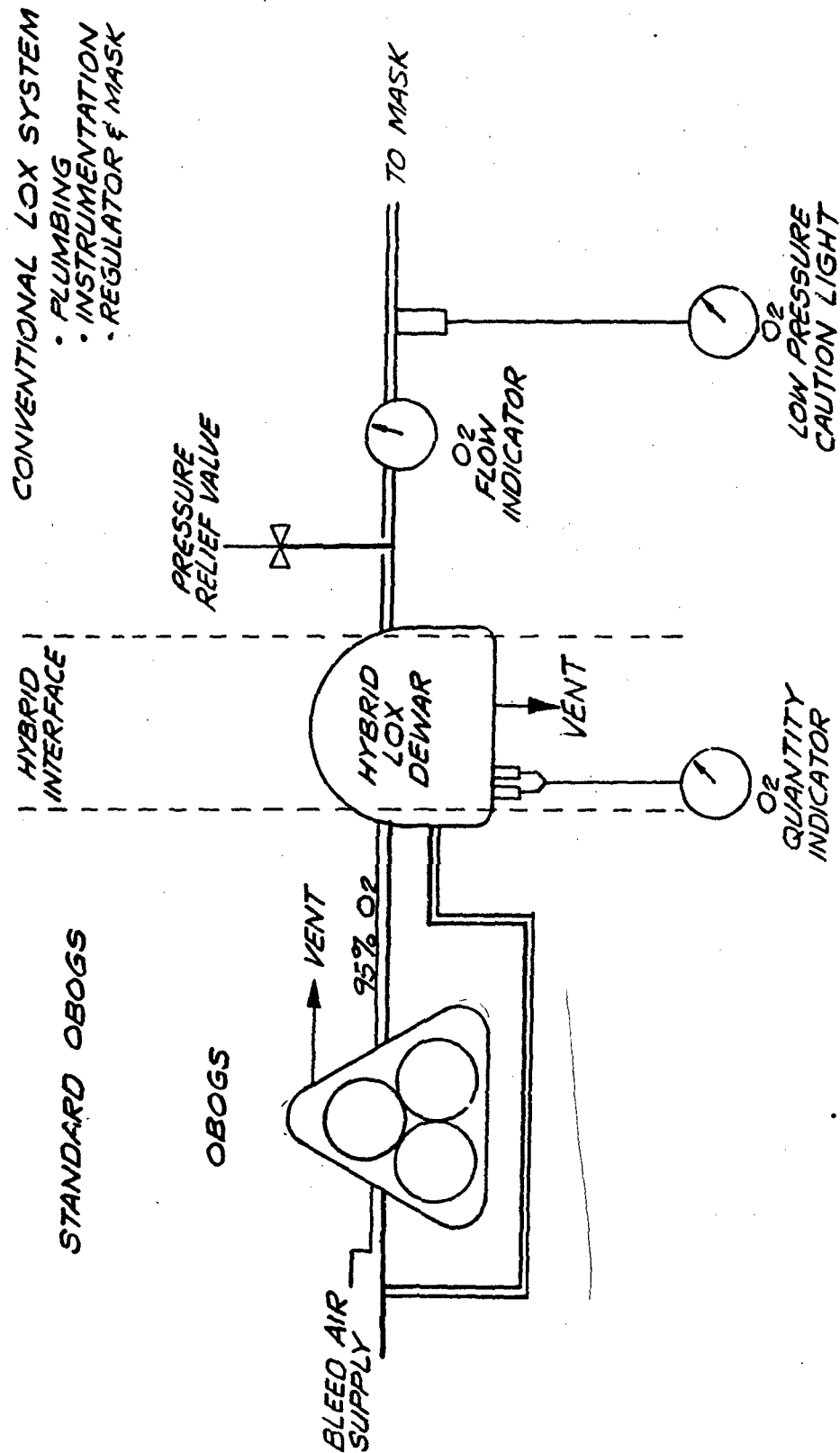


Figure 3. Hybrid Oxygen System - Baseline System Layout.

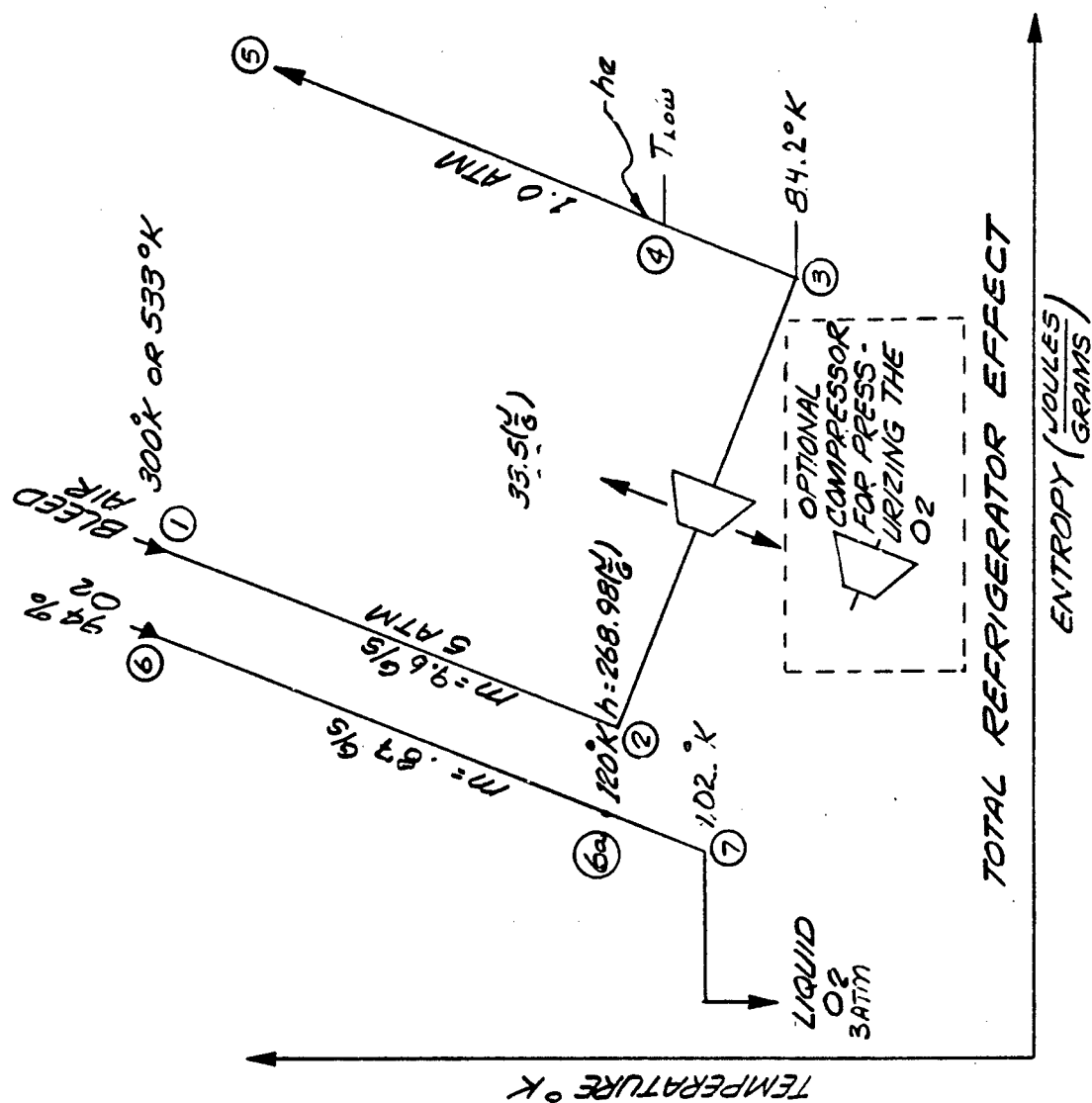


Figure 4. Bleed Air Refrigerator/Oxygen Liquefier System.

3.4 Aircraft Integration

As part of the study, Boeing Military Aircraft Company, Seattle, WA, was retained to evaluate the integration details for use of the HOS in a fighter aircraft. The summary of the report is reproduced here and the remainder of the report is given in Appendix 7.2.

1.0 Summary/Introduction

This report was prepared by the Boeing Advanced Systems Company to summarize the work conducted in support of Arthur D. Little, Inc. (ADL) on the HOS program. This report contains information on availability of bleed air for the HOS in typical fighter and bomber aircraft, discussions on system integration and feasibility, and identification of potential use and function of the additional cooling capability inherent in the HOS design.

The HOS discussed herein is based on the system shown in Figure 3. The HOS consists of an onboard oxygen liquefier which processes the concentrated oxygen from the onboard oxygen generation system (MSOGS) and stores it as liquid. Bleed air is required for oxygen generation and oxygen liquefaction.

Requirements for the MSOGS bleed air supply are based on oxygen breathing volume requirements. The design flow of oxygen from the MSOGS in the current HOS is set at 40 LPM (NTP). The required bleed air flow for liquefaction is a function of the MSOGS output, the temperature of the heat sink bleed air at the exit of the liquefier heat exchanger, and the heat removal required for oxygen liquefaction.

2.0 Integration Issues

Based on the liquefier design trade studies conducted by ADL, it was determined that the amount of bleed air required for liquefier operation ranged from 1.2 to 1.9 lb/min. The variation in flow rate is related to the selected design condition at the exit of the liquefier heat exchanger. An MSOGS unit was recently flight tested as part of the Tactical Life Support System demonstration on an F-15 aircraft. This unit required a flow rate of 2 lb/min and its installation in the F-15 and consequent flow extraction proved to have minimal effect on the aircraft environmental control system (ECS). This fact was verified by computer simulation and evaluation of flight test data. It is believed that retrofit installation of a 4 lb/min HOS in the F-15 and other aircraft will render no adverse effects either on the operation of the aircraft ECS or on aircraft performance. The HOS, with its low weight and volume characteristics, provides an excellent alternative to onboard stored oxygen. This system, because it produces oxygen at conventional pressure, will be compatible with either the CRU-73 or BRAG regulator. Appendix 1 examines the F-15 retrofit installation of the HOS in detail.

3.0 Bleed Air Availability

The quantity and thermodynamic state of the ECS bleed air supply is given in Tables 1 and 2 (pp. 80-81) for generic fighter and bomber missions, respectively. These conditions, as well as ECS bleed air location, were chosen based on design specification values for a typical (NGL) MSOGS unit (i.e., inlet pressure of 25 to 90 psig, required performance from 0°F to 100°F, operating to 160°F). The conditions from the fighter mission (Table 1, p. 80) most closely matching the required MSOGS conditions are at the exit of the secondary heat exchanger. A heat exchanger may be necessary prior to the MSOGS in the case of the bomber mission.

4.0 Excess Cooling Potential

Depending on the design and baseline operating state of the HOS, a range of options exist for generating excess cooling potential. For instance, the bleed air stream at the exit of the liquefier section, disregarding the low pressure, has excellent cooling potential. Also, the LOX, after extraction from the dewar, could be used as a heat sink prior to delivery to mask. Analysis of the HOS performance, as well as overall aircraft ECS performance, would have to be done to determine the worth or advantage of external use of excess cooling from the HOS. To determine where to tap off the HOS for additional cooling, the following questions must be answered or trades conducted:

1. What are the payoffs and/or weight penalties associated with increased cooling capability at the expense of increased sizing in heat exchangers or turboexpander?
2. What are the payoffs and/or weight penalties associated with increased cooling capability achieved through increased MSOGS output?

Additional cooling is necessary in present day and future electronics exhibiting high power density. Specific needs include cooling for avionics, advanced sensors, and applications for VHSIC (very high speed integrated circuit) and VLSIC (very large scale integrated circuits). The HOS could also be designed to use the vent bleed air as a bootstrap to precool the inlet bleed air. Use of the hybrid system as a bleed air conditioner and additional details on specific application of the excess cooling potential are found in Appendix 1 (p. 82).

5.0 Boeing Report Conclusions

The HOS being developed by A.D. Little, by processing the concentrated oxygen from the MSOGS and storing it as a liquid, preserves the conventional reliability and convenience of stored onboard oxygen while reducing the logistics burden of stored onboard LOX. The HOS, with its minimal bleed air requirements, can be used as a retrofit solution to LOX in current aircraft and as an alternative solution to LOX in future aircraft. The HOS also provides an excellent source of clean conditioned bleed air and inherently possesses potential for use in aircraft/avionics cooling.

4. Phase II: Laboratory Demonstrator Design

4.1 Laboratory Demonstrator Objectives

The purpose of this program is to develop and demonstrate a laboratory demonstrator HOS, as discussed in Chapter 3. The generation of oxygen from an MSOGS is to be coupled with a compressed air open loop refrigeration system. The compressed air will contain water vapor, trace gases, and possible chemical contaminants. These condensable gases are not too different than those experienced in stationary air liquefaction plants, but could be a considerable challenge for a lighter weight, compact mobile unit. Water vapor and other gases, which will condense at liquid nitrogen temperatures, could compromise the performance of the refrigeration cycle, if not adequately managed.

The objectives of the laboratory demonstration program are to:

- demonstrate cooldown of the system to the design temperature.
- demonstrate the production rate of LOX as a function of the amount of the supply bleed air flow. (Oxygen liquefaction (grams/second liquefied) with bleed air consumption required (grams/second))
- observe frost build up by monitoring the flow rate and pressure drop across the high pressure side of the heat exchanger.
- determine the feasibility of a deriming cycle to maintain free and clear refrigeration operation.
- determine the concentration of oxygen delivered from the storage dewars.

4.2 System Schematic and Major Commercially Available Components

Commercially available components were used throughout the system to reduce the overall cost. The demonstrator design was dictated by the performance of available heat exchangers and expanders.

4.2.1 System Schematic The system schematic is shown in Figure 5. The system consists of commercially available heat exchangers and components to simulate the reverse Brayton cycle selected in Phase I. Two 5-liter stainless steel dewars are shown, which are filled sequentially as outlined in Chapter 3.

4.2.2 Heat Exchanger The process and liquefaction stage heat exchangers are the central liquefier components so far as size and weight are concerned. Ideally, an actual aircraft HOS would require a lightweight stainless steel heat exchanger. A parametric analysis (Table 8) of the size and performance of the lightweight stainless steel heat exchanger can be performed through the selection of the temperature at station 4 (SW1) (SW or switch). This temperature dictates the sizing of the heat exchanger and the amount of bleed air required to liquefy the design flow of oxygen set at 40 LPM (NTP). An explanation of the calculation follows:

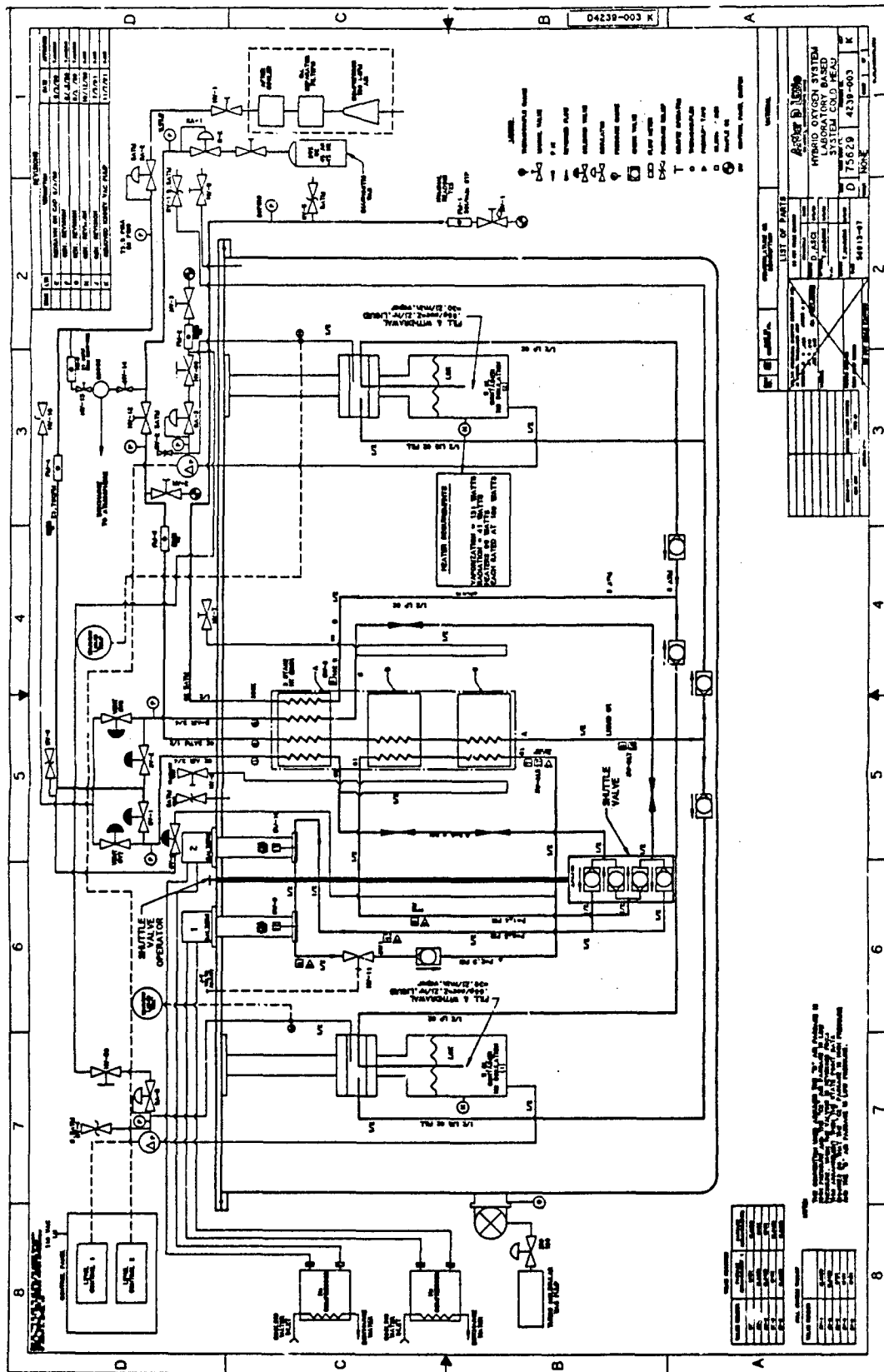


Figure 5. System Schematic.

Table 8. Hybrid Oxygen Component Sizing

08 Jan-88	File:Hybrid H wk1		Hybrid Oxygen Component Sizing					
02 09 PM			2.00 "X		4.00 "			
			2.00 "X		1.50 "			
STREAM	O2	HIP AIR	HIP AIR	LO P AIR	O2	LO P AIR		
T KELVIN	300	293.3	300	293.33	113	94.511		
Enthalpy(J/g)	272.133	420.917	426.705	420.917	78.104	217.88		
	\	\	\	\	\	\		
GRAMS/SEC	0.668	0.576	13.411	12.835	0.668	13.411		
	\	\	\	\	\	\		
T KELVIN	113	94.511	113	94.511	102	84.47		
Enthalpy(J/g)	78.104	217.881	232.337	217.88	-112.462	207.387		
.....								
REQUIRED CAPACITIES								
Duty Q (btu/hr)	441.97	398.80	8888.74	8886.39	434.09	479.8		
Effectiveness	0.9633		0.9637		0.3839			
M°C (btu/hr-ft)	1.3187	1.1193	26.5202	24.9371	22.0172	26.6634		
M°C minimum (btu/hr-ft)	1.1193		24.9371		22.0172			
Cmin/Cmax	0.8488		0.9403		0.8257			
NTU	10.60		15.90		0.59			
UA Required (btu/hr-F)	11.86		396.58		13.03			
.....								
PROPERTIES								
Viscosity (lbm/hr-ft)	0.037	0.033	0.033	0.033				
Cp (btu/lbm-F)	0.2479	0.2440	0.2483	0.2440	4.1386			
K (btu/hr-ft-F)	0.012	0.01	0.01	0.01				
Density (gr/cm3)	0.00586	0.00899	0.00899	0.00237				
Prandtl No. (1)	0.74	0.73	0.73	0.73				
.....								
CORE CHARACTERISTICS								
Width (in)	2	2	2	2	2			
Passage height	0.1	0.1	1.9	1.9	0.5			
Number of passages	2	2	38	38	10			
Free Flow Area (sq in)	0.0922	0.0922	1.7518	1.7518	0.4610	0.922		
Mass flux (lbm/hr in2)	57.7045	49.7572	60.9734	58.3546	11.5409	115.849		
Reynold's No	449.1593	434.2443	532.1319	509.2769	89.8319	1011.050		
h (btu/hr-in2-F)	0.6117	0.5335	0.5967	0.5744	4.8374	0.808		
fin ML	0.5682	0.5306	0.5612	0.5506	1.5978	0.653		
fin effectiveness	0.9047	0.9156	0.9068	0.9099	0.5766	0.878		
total effectiveness	0.9523	0.9578	0.9534	0.9549	0.7883	0.939		
UA/length(btu/hr-F-in)	8.9480	7.8489	166.0125	160.0770	292.8694	116.594		
.....								
Overall UA/inch	4.1812		81.4954		83.3943			
Length Required inches	2.8375		4.8663		0.1562			
.....								
LONGITUDINAL CONDUCTION								
Area header bars sq in.	0.010		0.190		0.075			
Area fins sq in.	0.016		0.298		0.118			
Area braze sq. in.	0.000		0.009		0.004			
Area side plates sq in.	0.160		0.160		0.160			
Area plates/braze sq.in.	0.033		0.623		0.164			
(ka)header	0.071		1.349		0.533			
(ka)fin	0.111		2.115		0.835			
(ka)braze	0.096		1.824		0.720			
(ka)side plate	1.136		1.136		1.136			
(ka)plates,splitters	0.307		5.833		1.535			
(ka) ALL	1.721		12.257		4.759			
Longitudinal loss ratio	0.0383		0.0079		0.1153			
Corrected Effectiveness	0.9277		0.9563		0.3669			
.....								
ITERATION OF LENGTH								
Length Multiplier	2.50		1.40		1.30			
Final NTU	26.5000		22.2645		0.7691			
Inc Effectiveness	0.9972		0.9790		0.4515			
Iteration Parameter	1.0194		1.0102		1.1288			
Final Core Length inches	7.0938		6.8128		0.2030			
.....								
Pressure Drop psi	0.1485	0.0741	0.0901	0.3247	0.0007	0.0211		
.....								
SIZE AND WEIGHT								
Add headers inches	3.000		3.000		3.000			
Total length inches	10.094		9.813		3.203			
Weight in lbs.	0.624		3.548		0.471			

1. Temperature at Station 4 is selected.
2. The ratio of bleed air mass flow rate to oxygen mass flow rate is calculated as the enthalpy ratios of the latent heat of vaporization of oxygen divided by the enthalpy difference of Station 4 minus Station 3. The sensible cooling of 6A to 7 is accomplished by the interchange with existing cold oxygen vapor.
3. The pounds/hour bleed air is calculated knowing the oxygen flow rate (40 LPM (NTP)).
4. An inlet temperature is selected, 300°K.
5. The required heat exchanger effectiveness (E) is calculated as $300 - 120/300 - T_4$.
6. The number of transfer units (NTU) is calculated from the formula $1/(1-E)$, resulting in a required UA equal to NTU times MC_p (mass flow rates times heat capacity); overall heat transfer coefficient U times area A .
7. A Reynolds number based on the hydraulic radius is calculated and used to estimate the heat transfer coefficient for the selected heat exchanger. For the purposes of this analysis, a plate fin compact heat exchanger on page 209 of Kays & London, Compact Heat Exchangers, McGraw Hill, 1961, was used.
8. The heat transfer coefficient in British thermal units per hour (Btu/h)/sq ft°F derived from the fluid properties, Reynolds number and heat transfer correlation is calculated. The required overall heat transfer coefficient is assumed to be one-half of each (the single-sided heat transfer coefficient).
9. The required heat transfer area as calculated by dividing the required UA by one-half H . This leads to the calculation of the heat exchanger size and weight directly from the Kays and London heat exchanger parametric data.

An identical series of calculations are made for the liquefier section of the heat exchanger. The entering temperature of the cold side to this portion of the heat exchanger is line #1, and the outlet condition is saturated liquid oxygen at atmospheric pressure and a temperature of 90°K.

These temperatures and flow rates set the heat exchanger performance, and assuming the heat transfer coefficient is governed by the same value as the gas heat transfer coefficient, the heat exchanger portion for liquefaction is calculated.

An additional heat exchange is required to precool the oxygen with the exiting nitrogen stream. Its size is calculated in the same fashion.

For the laboratory demonstration, the desired heat exchanger performance was achieved through the use of a commercially available aluminium cryogenic heat exchanger, as custom stainless steel heat exchangers were not available. The aluminium cryogenic heat exchanger was necessarily longer and heavier than a custom flight heat exchanger. This difference is a consequence of the extremely large longitudinal temperature difference that the heat exchanger must support. To counteract the adverse effect of high thermal conductivity of aluminium (nearly 16 times more conductive than stainless steel) on the end-to-end conduction, additional heat exchanger length is needed amounting to 4 times the length of a flight stainless steel unit. The added length increases the fluid pressure drop so that enlarged cross-sectional areas are needed, which again raises the size and weight of the unit.

4.2.3 System Deriming Function Bleed air from the engine is supplied to both the MSOGS and the liquefier refrigeration cycle. Condensibles and contaminants contained in the bleed air will be deposited in the process heat exchanger. Deriming techniques are needed to manage these condensibles in cryogenic liquefaction systems and can be very effective depending on the composition and amount of contaminant.

Deriming is accomplished by switching the heat exchanger flow path by changing the air flow solenoid switch positions and the manual 4-way reversing valve.⁷ Solenoids SV1 and SV8 (Figure 5) are operated together with the reversing valve in the "up" position, and SV2 and SV7 are set to the opposite condition. Reversing the flow is accomplished by changing the switch settings and the reversing valve. At each cross section, the heat exchanger suddenly experiences a drop in dew point on what was originally the high pressure side because the flow has switched from 5 ATM 100% saturated air to 1 ATM dry air.

Engine bleed air is simply compressed ambient air. Water vapor and some carbon dioxide (CO₂) represent the major condensibles which must be managed by the refrigeration system. The amount of water vapor and CO₂ deposited will depend on the concentration in the ambient air at altitude and the amount of water removal in the ECS. The low air temperatures at nominal flight altitude (25,000 and above) result in low humidity ratios relative to humidity at ground conditions but still contain substantial amounts of water vapor. The maximum condensable content (saturated) at altitude is given in Table 9.

Table 9. Mass Fraction Concentration of Condensibles

	Maximum Concentration (over 25,000 ft Altitude)	Typical Concentration in Lab Test
Water vapor	.0010	.023
CO ₂	.00031	.003

One of the major functions of the laboratory test program will be to evaluate the effect of high levels of water vapor and CO₂ on the air cycle (oxygen and nitrogen) refrigeration systems. The principal effect of condensibles and contaminants on the heat exchanger will be deposition of frost on the heat exchanger and subsequent reduction in mass flow rate due to the increased flow resistance and pressure drop.

4.2.4 Cold Head and Throttle Valve As described in section 3.3, a work-producing expansion device is necessary to achieve the refrigeration. Traditional reciprocating expanders operating at relatively low speeds would be enormous and impractical in the flight environment, hence, a miniature gas bearing turbine expander was identified as the appropriate component to achieve the desired expander performance, but the cost of such a development unit was beyond the budget of this program (see Table 10). For this program a simulation of the expander performance was devised using a cold head in series with an expansion valve. Figure 6 shows the design condition of a turboexpander and that of the cold head-throttle valve series. The simulation has the identical thermodynamic end points but is a different process. Two expansion processes are shown in the diagram. An ideal isentropic expansion and a realistic 70% of isentropic (ideal) expansion are shown. The isentropic expansion offers substantially greater refrigeration but can not be achieved. The nominal design point with a 70% expander will bring the bleed air to the saturation condition at about 85°K, which is sufficiently cold to liquefy pressurized oxygen.

⁷ The four-way reversing valve was added later in the program (see Figure 10).

T-S Diagram Air

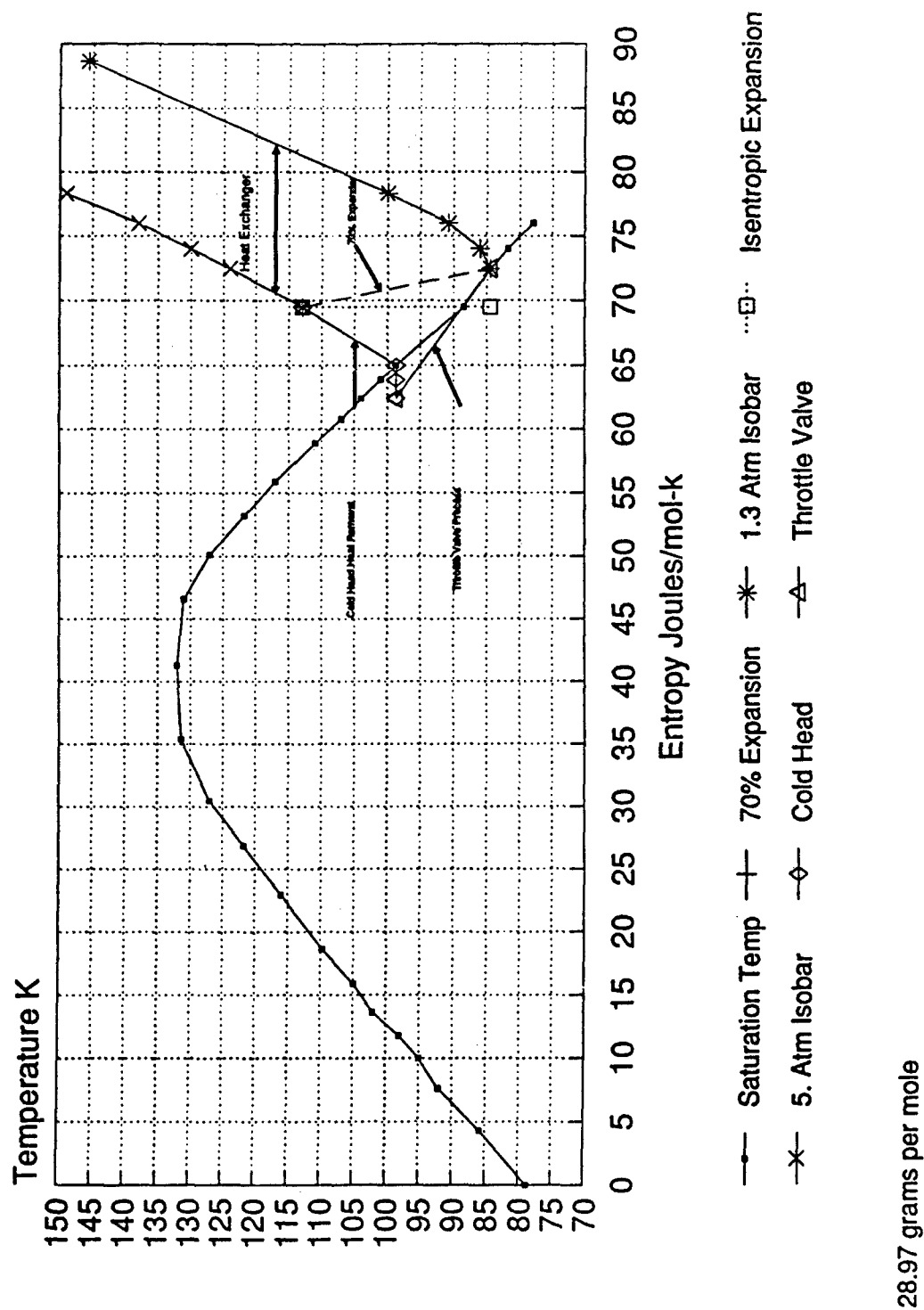


Figure 6. Temperature Entropy Diagram of Process.

Table 10. Expander Comparison

Manufacturer	Cost	Delivery Time	Weight	SIZE Dia. Length	η Max	Watts
Koch Process Systems Reciprocating Expander Model 1600	over \$50,000	16 wks	800 lb	34-1/4" 70"	50%	367 max.
Creare Turboexpander						
A. 18 watt	\$100,000	12 wks	10 lb	4" 11"	75%	18
B. 100 watt	\$350,000	20 wks	10 lb			100
C. 370 watt	\$500,000	24 wks	10 lb			370
Sulzer Bros. Turboexpander TGL-22-11/B2	\$80,000	8 mos	90 lb	10" 32"	~60%	~300
Balzer - 2 Cold Heads and JT Valve	\$60,000	1 mo	40 lb	10" 12"	NA	330

Two Balzers Helium Cryogenic Refrigerators and a Koch Process J-T valve were selected for the application. The two cold head units have a total refrigeration capacity of 330 w at about 80°K at the cold head. With the proper design of the cold head exchanger the cold heads will produce the required refrigeration.

4.2.5 Heat Balance Table 11 shows the heat leaks on the heat exchangers, and Table 12 summarizes the total heat balance. At the design (Table 12) flow rate of oxygen (.66 g/sec), 256 w of refrigeration are needed which requires 357 w of cooling at the cold head. The cold head exchanger operating at 85°K must have 6.2 NTU's. Table 12 shows the effects of running at reduced oxygen flow rate on the heat exchanger performance. For instance, at .55 g/sec, a cold head heat exchanger of 1.1 NTU is required.

Table 11. Heat Leak Into Cold End in Watts

Component	Conduction	Radiation	Total
Stem Valve	2.5	2.5	5
Piping- Air HP	.1	3.4	3.5
Piping-Air LP	.1	3.2	3.3
Piping-O ₂ -HP	.1	2.8	2.9
Piping-O ₂ -LP	.1	1.6	1.7
Pressure Relief Line	.1	2.2	2.3
Defrost Line	1.4	3.4	4.8
Pressure Taps	.2	3.6	3.8
Heat Exchanger		2	2
Heat Exchanger Supports	7.2	.8	8
Total w/o Dewars			37.30

The expected heat leak into the cold end of the system (exclusive of the dewars) is given in Table 11, which shows a predicted value of about 37 w. About 26 w of this heat leak flow into the heat exchanger and cold end and are included in the O2HXLQ.WK1 model (Table 12) which was used to size the cold head heat exchangers.

4.3 Piping Diagrams⁸

4.3.1 Drawings The piping diagrams for the system are given in Appendix 7.3.1. A photograph of the main elements is given in Figure 7. This figure shows the cold heads, cold end plumbing, and process heat exchanger all attached to the cover plate of the vacuum vessel. The dewars are not shown, but they attach to the two long stem flanges shown on either side of the heat exchanger.

4.3.2 Pressure Losses The piping diagrams specify the fluid components and piping lengths which, in turn, allow for the calculation of system pressure drop. Table 13 shows the calculated pressure drop through the system. The total pressure drop is estimated to be about 6 psi which is well within the 10 psi budget selected for this design.

⁸ The discharge of MSOGS and liquefier are attached to the same vacuum manifold.

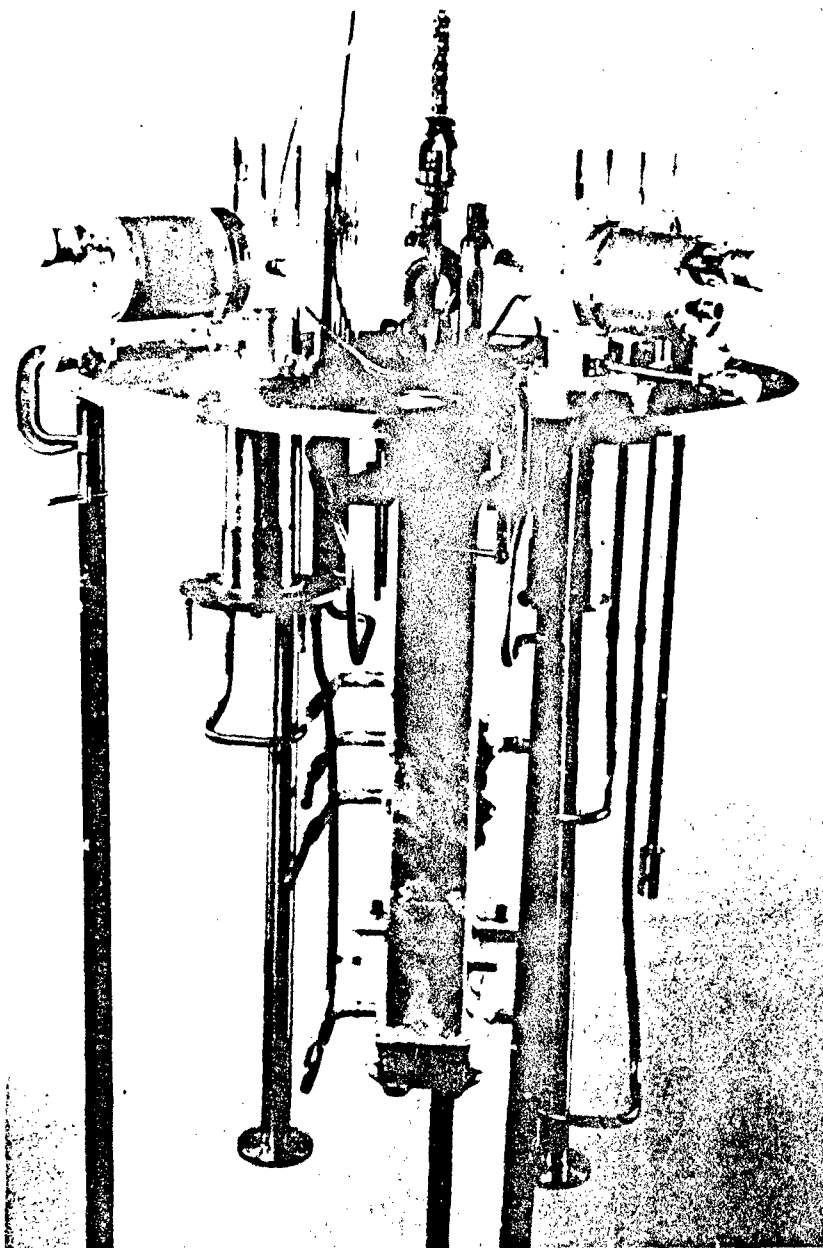


Figure 7. Main Refrigeration Elements (Dewars Not Attached).

Table 13. System Pressure Losses In PSI

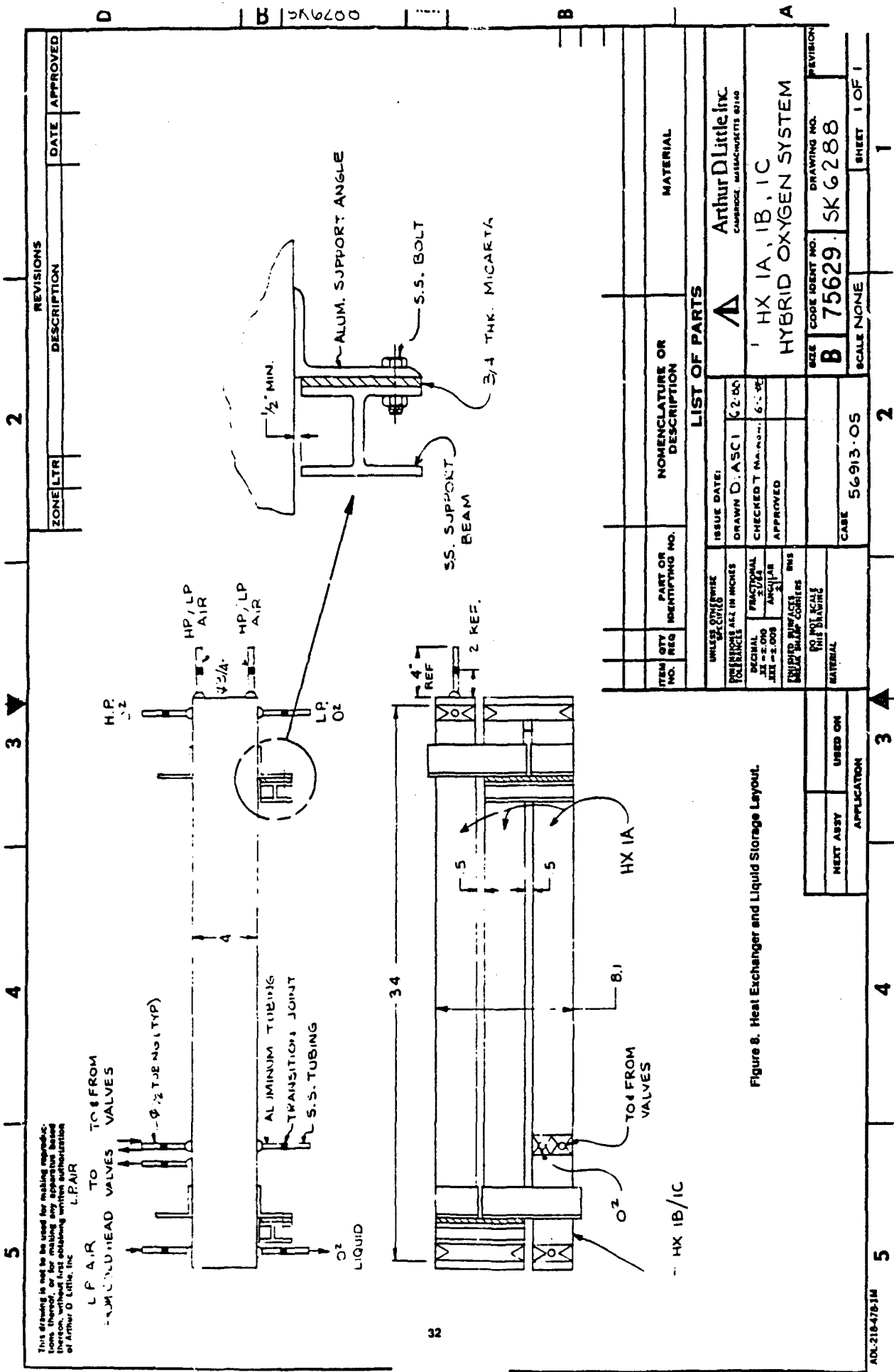
Component or Piping Section	Dia."	Len."	ΔP (psi)
Air-Low Pressure			
From Cold Head to HX	.5	48	2.3
from HX to check valve	.5	40	1.4
Check valve	.5		.14
from Check valve to Hx	.5	24	.8
from Hx to Vac Cover	.75	24	.1
Total			4.74
Air-High Pressure			
from Vac Shell to HX	.5	24	.01
from HX to Check valve	.5	12	.1
Check valve	.5		.2
from Check valve to cold head	.75	121	.5
Total			0.81
O ₂ -High pressure	.5	94	.05
O ₂ -Low pressure	.5	54	.11
Total			.16
O ₂ Needle Valve			0.14

4.4 Mechanical System

The mechanical system layout is shown in Appendix 7.3.1.

4.4.1 Heat Exchangers and Liquid Storage Layout The heat exchanger is four 36-in.-length sections as seen in Figure 8. Insulated stand-off mounting of the heat exchanger, dewars and plumbing follow standard cryogenic design practice to minimize heat leakage into the cold components.

Two 5-liter storage vessels are included to accommodate a range of liquid storage scenarios.



4.4.2 Piping, Flow Controls and Instrumentation The piping layout is given in Appendix 7.3.1. The piping approach incorporates the refrigeration cycle as well as the deriming cycle. A discussion of the controls and instrumentation are given in section 4.5. The piping analysis was outlined in section 4.3.

4.4.3 Concentrator An Essex concentrator will be used as provided by Essex Cryogenics, St. Louis, MO. The layout of the unit is shown in Figure 9 and the performance characteristics of the unit are contained in Appendix 7.4.

4.4.4 Compressor A compressor sized at 60 standard cubic feet per minute (SCFM) and 100 psig pressure will be used to provide the supply of bleed air to both the liquefier and the oxygen generating system.

4.5 Electrical

The diagrams in Appendix 7.3.3 (electrical schematic) shows the electrical system layout, including instrumentation sensors and controls for the operation of the compressor, vacuum and expander subsystems.

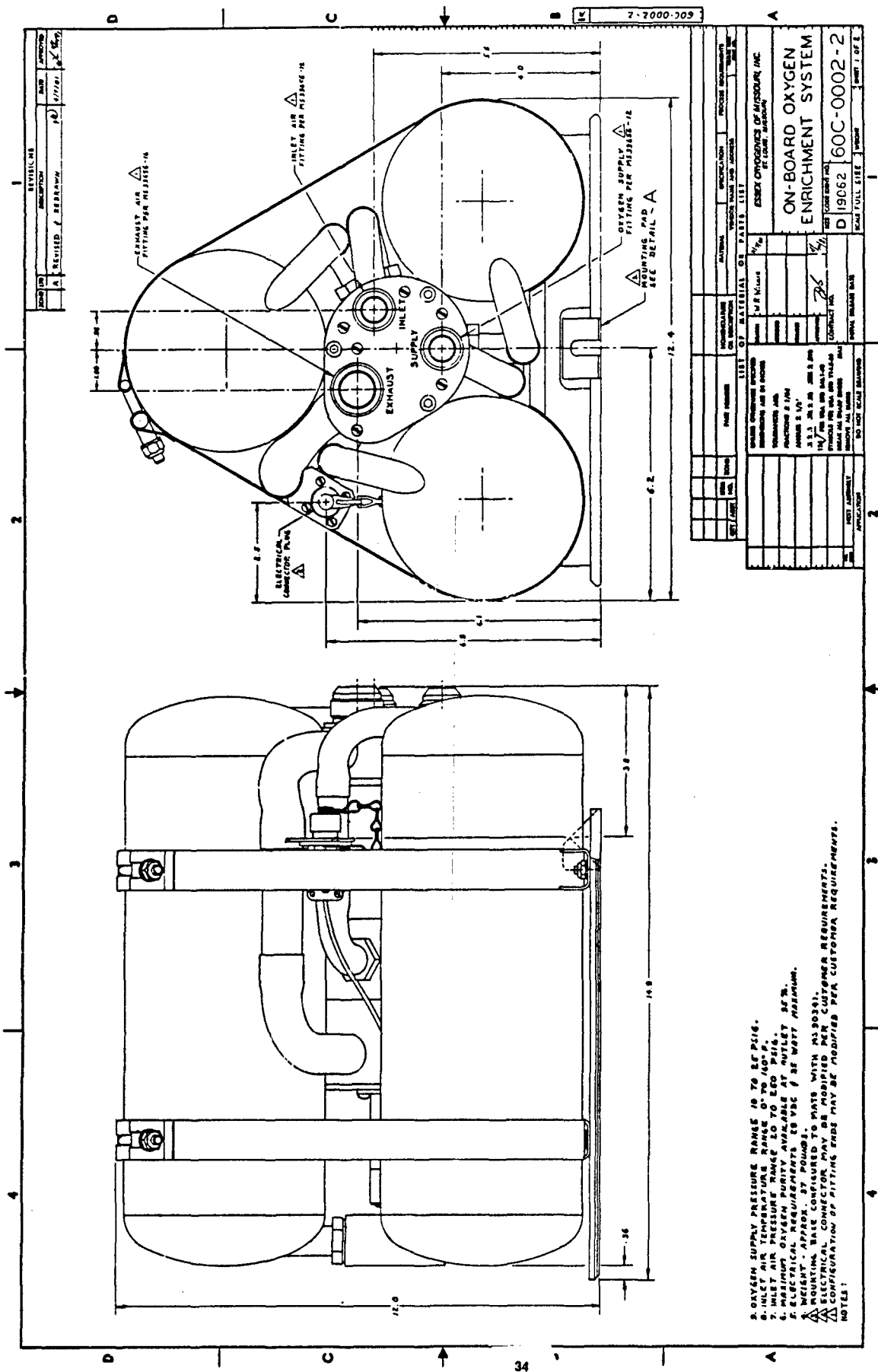
The instrumentation consisted of silicon diodes and thermocouples for the temperature measurements and capillary tubes for the pressure measurements. Table 14 summarizes the instrumentation selection.

Table 14. HOS Instrumentation

Flow Meters	Fisher-Porter, Model 10A 2700 (various diameters)
Regulator	Tescom Pressure Controls
Thermocouples	Arthur D. Little, Inc.
Diodes	Lake Shore Cryotronics
Pressure Taps	ADL Capillary
Manual Valves	Essex Industries
Pressure Gauge	1/2" NPT McDaniel Controls
Hermetic Connectors	Allied Amphenol Products

Silicon Diodes

Silicon diode temperature sensors were used because precise temperature measurements were required. We chose Lake Shore Cryotronics DT-470-DI-13-4L sensors which were individually calibrated in the 75 to 325 K range at an accuracy of 50-75 mK. These diodes were mounted on specially designed copper disks which have protruding sections into the gas flow. Diodes in pairs at each measurement location were used as a precaution in case one of the diodes lost its signal during testing.



A Lake Shore Cryotronics readout was used to determine the voltages of each station as a function of time. Twenty silicon diodes were used, consisting of a diode pair at each of nine locations (the inlet, intermediate, and exit of the heat exchanger streams) and two spares.

Thermocouples

Standard type "T" thermocouples are used at the warm end of the heat exchanger and on the dewar vents. The thermocouples were mounted to copper sleeves which were bent over the pipes at the desired locations. The thermocouple temperatures were recorded automatically with a Leeds and Northrop Strip-Chart Recorder.

Pressure Taps

Custom capillary tubes are used at locations where pressure measurements are desired. The taps were connected to a valving system with pressure gages to measure the pressure drops. For the HOS, we used six pressure taps (located at the inlet, intermediate, and exit of the heat exchanger streams).

4.6 Instrumentation

4.6.1 Temperature Five thermocouples and ten temperature sensing diodes were used in the system. Diodes were used principally at the cold end of the piping and thermocouples at the warm end of the heat exchanger. The selected diodes are Lake Shore Cryogenics DT-470-DI-13-4L sensors which were individually calibrated from 75°K to 325°K. The diodes are more precise at cryogenic temperatures with an accuracy of $0.1 \pm ^\circ\text{K}$, while the thermocouples are good to $\pm 1^\circ\text{K}$ or $\pm 3^\circ\text{K}$ @ 300 °K. The diodes were mounted on specially designed copper disks which have a section protruding into the gas flow. We estimate an uncertainty of absolute temperature of .2°K at 75°K with this method of measurement. The thermocouples are less expensive and more rugged; and were used at the warm end where there was not enough space to mount the diode disks as the thermocouples can be strapped directly on the gas tubing without being damaged.

The diode and thermocouple wires are routed through connectors in the cover plate. The diodes are connected to a single Lake Shore model 201 Thermometer with direct Kelvin readout, and selection of the sensor is done by a rotary manual switch. The thermocouples were monitored on a thermocouple strip-chart recorder reading directly in degrees centigrade.

4.6.2 Flow Rate Flow rates are determined by Fisher & Porter rotometer type flow meters. Three Model #10A4555x flow meters are used for the oxygen lines (FM-1, FM-2 and FM-6) and two for the air lines (FM-3 and FM-4). The typical meter accuracy is $\pm 2\%$. The design point reading of the meters is given below in Table 15.

Table 15. F&P Flow Meter Calibrations

Flow Meter	Pressure ATM	Design Flow Rate SCFM	% Scale
FM-1 O ₂ Mask	1	1.09	72
FM-2 O ₂ Vent	1	1.09	72
FM-3 MSOGS air	5	23.7	72
FM-4 Ref air	5	23.7	72
FM-6 O ₂ In	3	1.09	41

An on-line oxygen concentration monitor was used at all three NV locations during all tests. The monitor is a Beckman Oxygen Analyzer, digital display with characteristics shown in Table 16.

Table 16. Beckman Oxygen Analyzer

Reproducibility	$\pm .01 \%$
Response Time	7 seconds
Sample Flow Rate	250 cc/min
Output	Selectable DC & Alarm
Baromet mp	$\pm 1 \%$ F.S.

The Beckman oxygen monitor will track the concentration of oxygen during specific operations and for specific functions:

- **Dewar Filling:** oxygen percent is measured @ NV-2; in the event that the concentration falls below 90% the O₂ flow will be manually terminated (see section 4.7.4). Oxygen percent also is measured at NV-3 dewar vent.
- **Dewar Oxygen Withdrawal:** the concentration was monitored at NV-1.

The data taken during the course of the tests suggests that the oxygen monitor read 2% to 5% high on occasion when compared to the expected oxygen concentrations from the MSOGS unit.

4.7 Safety Review

Operation and handling of high purity oxygen can be hazardous, and careful attention to safety at the design stage is necessary to guarantee safe operation.

4.7.1 Safety Review Board A preliminary safety review was conducted as part of the Phase II design effort. A final safety analysis was conducted prior to assembly of the oxygen system plumbing and after parts were prepared. Members of the safety review board were:

- W. David Lee - Program Manager
- Thomas Maimoni - Manufacturing Specialist
- Arthur Post - Senior Cryogenics Engineer
- R. Warren Breckenridge - Senior Cryogenics Engineer
- Paul Croce - Senior Safety Engineer
- Tom McKelvey - Senior Safety Engineer

Safety action items and their resolutions are given in the next several subsections.⁹

4.7.2 Materials Inspection of all oxygen handling components and confirmation of oxygen rating were conducted and summarized in Table 17.

⁹ Smoking was not permitted in the pilot plant, and all oxygen was vented out of the building.

Table 17. Safety Inspection List

Component	Symbol or No.	Confirmation
O ₂ Pressure regulator valve	R-2	Supplied with O ₂ Bottle
O ₂ Pressure regulator	RA-1	Oxygen Regulator
Oxygen Pressure gages	5	316 SST Clean for O ₂
30 LPM std flow meters	4	316 SST Clean for O ₂
2.2 iph 3 ATM liquid check valves	4	O ₂ Service Essex Cryogenics
Liquid O ₂ Dewars	2	316 SST Clean for O ₂
Asco vent solenoid valves	2	O ₂ Service Asco
Magnehelic Liquid level gage	2	Dwyer Cleaned for O ₂
Atec Heat Exchanger Core	1	Inspection Certificate (IC)
Stainless .75 OD tubing		IC
Stainless .5 OD tubing		IC
Relief valves 5.5 ATM	5	316 SST cleaned by ADL
Needle & Hand valves	NV-3,NV-2 NV-1 HV-12	316 SST cleaned by ADL

All valves and pressure gages were calibrated on bench-top tests with an oil-free nitrogen source.

4.7.3 Pressure Certification The following components were certified or checked for the pressure rating and are summarized in Table 18.

Table 18. Pressure Certification

2 Oxygen Dewars	100 psi Pressure Vessel
Pressure relief valves on vacuum vessel 1-2 psig	The cover plate weight of 550 lbs and 34 in dia represents a .6 psig lift off pressure relief which is desirable.
Set all 4 oxygen line relief valves to 5.5 ATM	Calibrate relief valves #1,#2,#4,#5 to 5.5 ATM with oil free N ₂ .
Check RA-3, RA-5 to confirm can withstand 5 ATM.	100 psig 316 SST cleaned for O ₂ service

Note: RV3 and RV1 may not be necessary but will not harm the apparatus.

The vacuum cover plate is not fastened to the vessel and will release vessel pressure above about .6 psig. This safety feature will accommodate unexpected oxygen or pressurized air release in the system within the vacuum chamber.

4.7.4 Modes of Failure Liquid Level Failure Failure of the magnehelic gage to properly register the liquid level plus miscalculation of the accumulated liquid (using the differential flow rate from FM-6 minus FM-2) may result in overfilling the dewars. An overflow protection temperature sensor is located on the oxygen vent line. During filling the sensor will be monitored by the operator to be sure that it is well above the saturated LOX temperature, to

prevent venting of LOX which would be a hazard. When a temperature close to 103 K is detected, the liquid fill will be discontinued by either switching to the other dewar or shutting HV-12.

The safety review committee recommended the addition of a high liquid level interlock on the supply solenoid. The project team has decided to monitor the liquid level manually and not incorporate the automatic control because of budget limitations.

Similarly automatic shutdown of the dewar heaters (see section 5.1) was recommended, but because of budget limitations it was decided to carefully monitor the dewar vent line pressure manually and control the heaters manually.

Hazard Signals The test system has been designed to provide sensor data for evaluation of the performance. A number of sensors provide warning of a potentially hazardous condition, which will require human response as summarized in Table 19.

Table 19. Hazard Signals (Manually Monitored)

Monitor	Action
Loss of Vacuum @ HV-5	If sudden pressure rise occurs, immediately terminate compressor operation and close HV-12 oxygen valve.
Dewar Fill level High; either Magnehelic or flow rate * time.	Terminate oxygen flow @ HV-12
Dewar Fill level High; T_{11} or $T_{18} \approx 103$ K	Terminate oxygen flow @ HV-12
Dewar vent pressure rise above 5 ATM	Terminate heater input
O ₂ concentration drops below 90% @ NV-2	Terminate oxygen flow @ HV-12. A drop in concentration could indicate cross contamination in the heat exchanger.

Valve Sequencing-Oxygen Errors in the operation of the oxygen valve sequencing have been examined. No hazardous conditions could be identified.

Overfilling of a dewar is a potential hazard and the procedures to avoid that condition and the emergency response are given in the previous section.

Another potential valve operation error is to open the 3 ATM fill switch to a dewar already pressurized to 5 ATM. In that event the dewar will blow-down through the vent line to the pressure set of RA-3 or RA-5. The only hazard in this situation is handling an instantaneous release of up to 5 liters of 5 ATM oxygen or 25 liters at 1 ATM. The interior space of the building was sufficiently large as to make this release inconsequential.

Valve Sequencing-Air Improper valve sequencing of the pressurized air system can occur in the following manner shown in Table 20.

Table 20. Valve Sequencing Hazards

Valve		
SV-1 open SV-2 open	system pressurizes to RA-2 level if SV 7&8 are closed; otherwise system releases through vent	no hazard
All sv closed	compressor dead headed; no flow	no hazard if compressor shut off in reasonable time (minutes)
RA-2 fails	air side could pressurize to compressor max of 132 psig	RV-6 should relieve the line at 5 ATM set point

MSOGS/Compressor Oil Flow Through Probably the most significant hazard is a release of oil vapor by the compressor into the MSOGS unit, creating an explosion hazard. Under normal conditions the maximum release of oil from the compressor is well below the levels which can be tolerated in the MSOGS.

In the event of a sudden failure of the compressor oil clean-up system, considerable oil will flow into the MSOGS unit creating an explosion hazard. We monitored the glass air flow meter FM-3 for signs of oil contamination in the event that the flow meter shows significant oil, the MSOGS should be disconnected from the air flow immediately.

Unexpected Oxygen Spill The following conclusions were made concerning a LOX spill:

- An instantaneous release of 10 liters of LOX in the pilot plant would raise the oxygen concentration by $\approx 1.6\%$, which is not a hazard. While the local concentrations may be higher, the ventilation rates in the pilot plant should manage the vapor.
- A liquid spill on the floor could be a problem because of contaminants on the floor. A clean galvanized pan was fabricated and set on the floor to handle a spill.

5. Phase III: Hybrid Oxygen Laboratory Test Procedure and Results

5.1 Test Program

To meet the test objectives, a test program was implemented consisting of:

- Diagnostic tests (without oxygen) using research grade argon (Ar) to establish the operational procedures and any problems in the refrigeration system.
- Complete tests with MSOGS to measure oxygen concentration levels. During these tests, the MSOGS was vented to atmospheric pressure.

MSOGS system tests were conducted with an Essex MSOGS¹⁰. The unit was run with the bleed air refrigeration unit to produce about 2 liters of liquid in dewar No. 1. The flow was switched and the other dewar was filled with about 2 liters. Dewar No. 1 was allowed to self pressurize to 5 ATM and then liquid was withdrawn and the concentration measured. Self-pressurization required approximately 130 w of heat input. We calculated about 40 w of residual radiation heat input and have provided about 90 w additional heat with strip heaters wrapped on the outside of the dewar. Continuous gas concentration was monitored at NV-1 with a Beckman oxygen monitor. This instrument provides a continuous measure of the concentration percentage of the stream. The on-line oxygen instrument will provide a continuous check to detect major deviations from the condition when the chemical analysis samples were taken.

5.2 Test Procedure

The key processes for the safe and proper operation of the breadboard unit are given in this section.

5.2.1 Instrument and System Checkout Prior to operation, instruments and components were checked out by the test engineer and technician. All sensors were checked and the vacuum integrity of the pressure vessel was recorded. A room temperature pre-run vacuum of better than 1×10^{-4} torr should be achieved. With a cold heat exchanger, the vacuum should be 10^{-6} torr. The startup sequence is given in the next section.

5.2.2 Initial Cooldown (Room Temperature to 20°F) The initial cooldown is accomplished by alternate cold heat ON/OFF cycles to remove frost formation on the cold heads. The cold "ON" operation is initiated using SV1 and SV8. The JT valve is opened until a flow reading of 60 achieved at a supply pressure of 32 psig. Flow is continued until the air flow (FM4) drops by about 10% from a reading of about 60 to 50 with the same JT valve setting. The cold heads (SW 9) typically achieve a temperature of about 10°F. At this point, the cold heads are switched off until the cold head (SW9) is warmed up to equal the temperature at SW1. This operation is continued until temperatures SW 1, 2, 4, 8, 9 read below 20°F.

Once temperatures throughout the system are below 20°F, continue flow operation and valve down the JT until a 20 psi pressure differential is across the JT valve.

5.2.3 Final Cooldown (270°K to 100°K) During final cooldown, monitor the flow rate on FM 4, and when it drops by about 10% from 55 to 50, reverse the cycle. Observe the pressure taps, immediately after a cycle reversal, the high pressure side of the heat exchanger should rise approximately 10 psi to about 20 psi, representing a 10 psi frost formulation. During the defrost, pressure on the low pressure side should fall from 20 to 10 psig, at which point, another reverse cycle may be appropriate depending on FM 4 flow rate. Regular adjustment of the JT

¹⁰ The performance characteristics of the Essex unit are contained in Appendix 7.4.

valve must be made during the cooldown, as the density of the air is being reduced and the flow resistance drops. The target value should be about 35 psig on P3 and 15 psig on P7. (These values correspond to pressure tap positions 4 and 5, respectively, on Figure 5.)

Run Mode

Once SW4 diode has reached approximately 105°K, the oxygen can be started as it will begin to liquefy. Generally, open HV50 or Dewar No. 1 first and begin to fill it. Monitoring the dewar thermocouples is essential as it will indicate cooling of the vent in the dewar body itself. Considerable flow of oxygen is required before the dewar is chilled enough to allow for liquid build up. Generally the backflow regulators on the oxygen are set at approximately 15 psi gauge.

During the run mode, reverse the heat exchanger every 5 minutes when the flow rate on FM4 falls more than 10% of nominal or from 50 to 40.

5.2.4 Data Collection The test data are recorded on a personal computer (PC) spreadsheet according to the layout in Appendix 7.5.

5.3 Discussion of Test Results

The following tests were performed:

Four Ar diagnostics tests (not reported)

Seven oxygen tests

The Ar (not reported) diagnostic tests were performed in October and November 1989, at which point, the original check valves were replaced with hermetic check valves because of an unacceptable leakage to the vacuum. Subsequently, in the summer of 1990, the hermetic check valves failed as a result of sand from the heat exchanger fabrication lodging in the mechanisms. We suspect that the sand was either from the cleaning of the cold head heat exchanger or the main process heat exchanger by sand blasting. Finally, in the fall of 1990, a four-way sliding valve was designed and installed to overcome the check valve problems. The new flow schematic with the four-way reversing valve is shown in Figure 10. The modifications were completed in October 1990, and the oxygen tests were run.

A representation of system performance data recorded is displayed in Figure 11. The results of oxygen test No. 7 will be described for all six graphs as well as references made to the results of earlier tests. Test data for tests 1 through 6 are contained in Appendix 7.6.

5.3.1 Test No. 7 Results

Graph 1 - Cooling History

The cooldown curve for test no. 7 is given in Graph 1. The rise in the calculated effectiveness to the steady state value is a result of the heat capacity effect of the heat exchanger on the net heat exchange. The heat exchanger effectiveness is the ratio of the actual heat transferred between the ideal maximum as calculated:

$$\text{Effectiveness} = (T_{IN} - T_{SW1}) / (T_{IN} - T_{SW2})$$

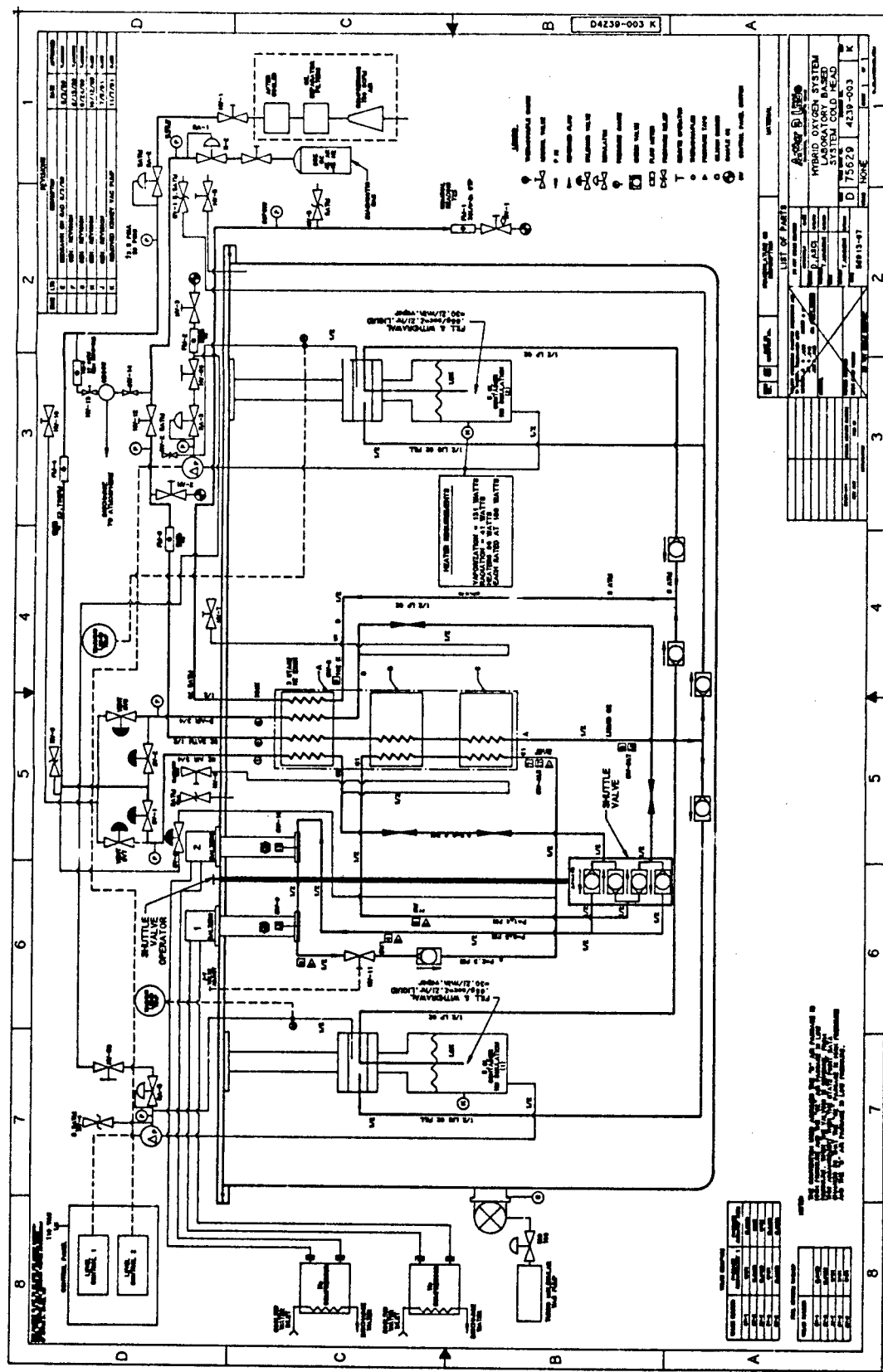


Figure 10. Four-Way Reversing Valve System Schematic.

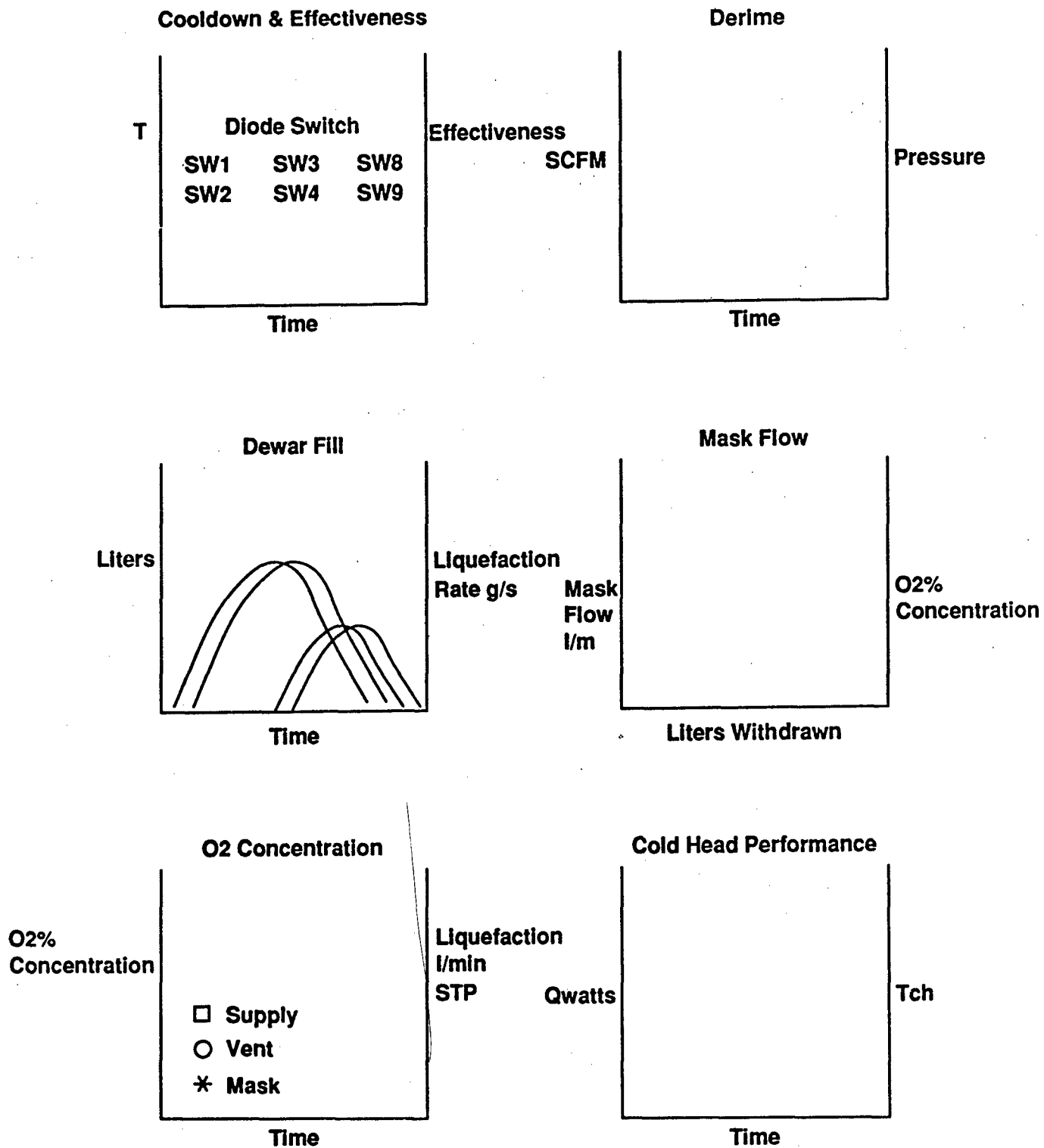
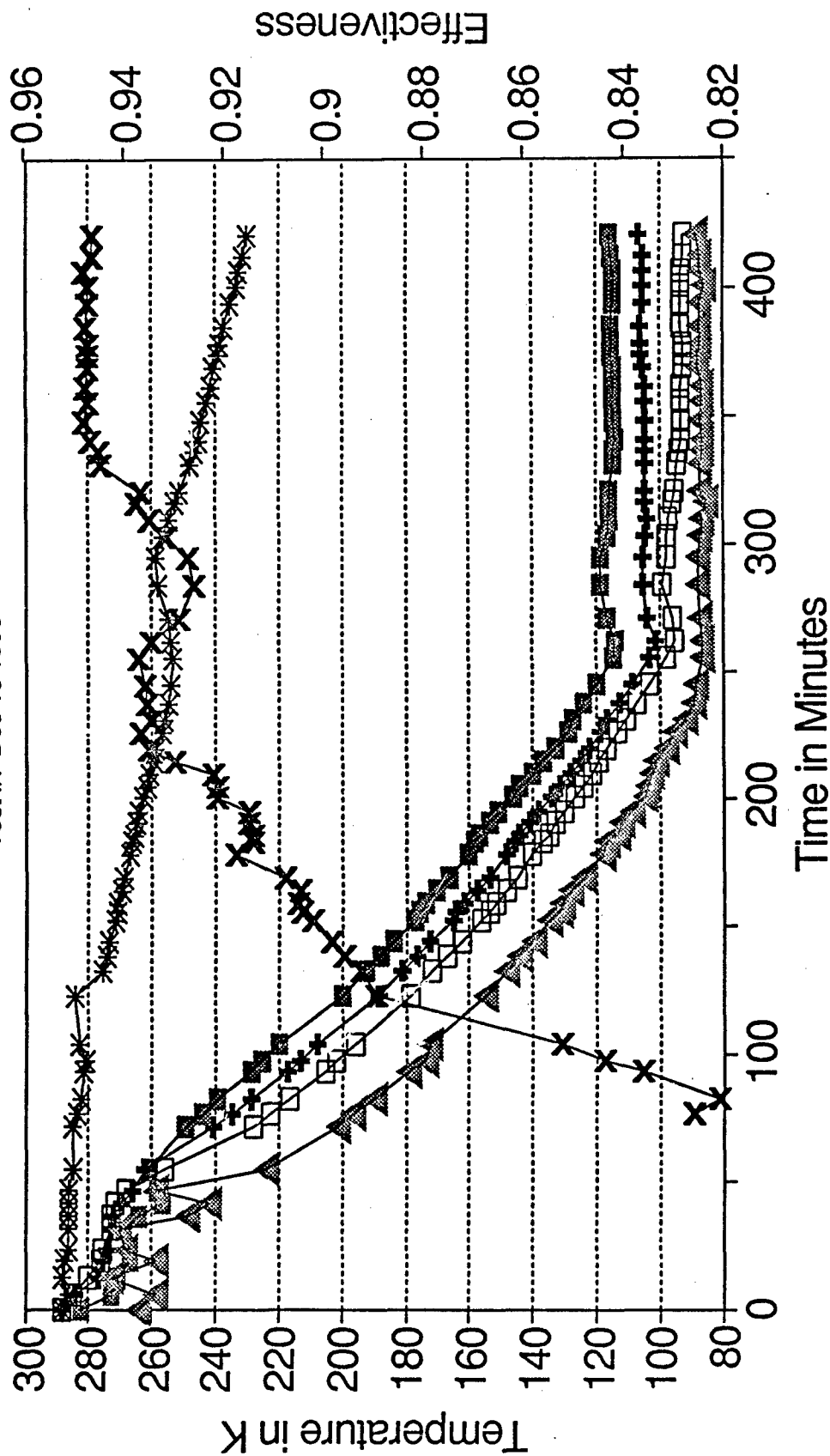


Figure 11. Test Results Format Overview.

Cooling History Test #7 Dec 16 1990

Graph 1



The design of effectiveness was .968 (excluding conduction heat gain). The measured value was about .95, which is acceptable.

During cooldown all diode temperatures were recorded. The expected cooldown period for the 120 lb heat exchanger, cold head, and other system hardware, was 75 min. The actual cooldown time is given below in Table 21. The increase in cooldown time is due to a loss in cold head refrigeration capacity due to the gradual loss of helium refrigerant as explained later in this section under the *Cold Head Performance* graph.

Table 21. Cooldown Time

	Nov 1989	Dec 13, 1989	Oct 1990	Dec 1990
Cooldown Time Mins.	100	225	225	250

The expected cooldown time for the 4 lb stainless steel flight model heat exchanger (Table 23) would be about 3 minutes.

Graph 2 - Dewar Fill

The amount of LOX produced and the corresponding bleed air flow requirement are essential test parameters.

Rate of Liquefaction = Mass Flow Rate @ FM 6 - Mass Flow Rate @ FM 2

In the oxygen test spreadsheet, the liquefaction rate is time integrated to provide the liquefied volume based upon the flow measurements. The magnehelic pressure gauge also gives cumulative liquid in the dewar, and both of these values are given in the dewar fill graph. The liquefaction rate in grams/second is also shown. There was generally good agreement between the flow measurement and magnehelic measurement of liquid level throughout the test program.

The fluctuation in oxygen flow rate and liquefaction rate seen throughout the test program is intentional operation over the range of capacity to detect system problems.

The unit has demonstrated a liquefaction rate of .1 to 1.0 g/sec with a nominal design value of about .5 g/sec in steady state.

We attribute the sudden jump in the magnehelic pressure gage reading seen in this test and others to the dynamic nature of the filling process and the tendency of the magnehelic pressure line to experience vapor-driven surges during the initial stages of dewar filling.

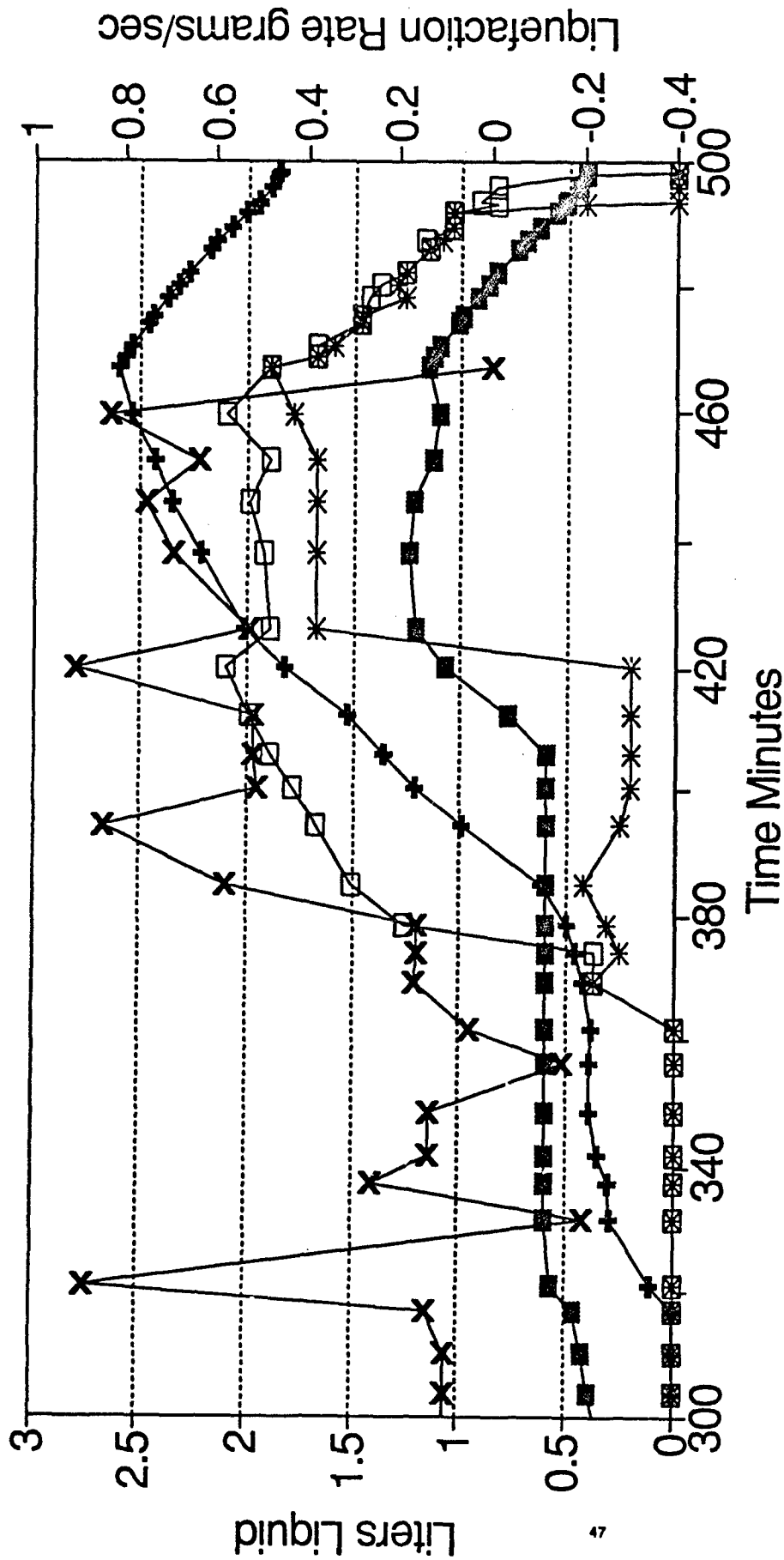
Graph 3 - Heat Exchanger Deriming

Periodic passage switching to derime the heat exchanger was effective as Graph 3 shows. After liquefaction temperature was achieved (about 200 min), regular deriming was necessary to maintain the flow. As seen in Graph 3, the derime cycle was able to rapidly restore the flow (+) to the design value of about 21 SCFM after falling to 18 or 19 SCFM due to frosting. We are confident of the success of the deriming, particularly, as the supply air was nearly 100% water saturated for all of the tests.

Dewar Fill

Test #7 Dec 16 1990

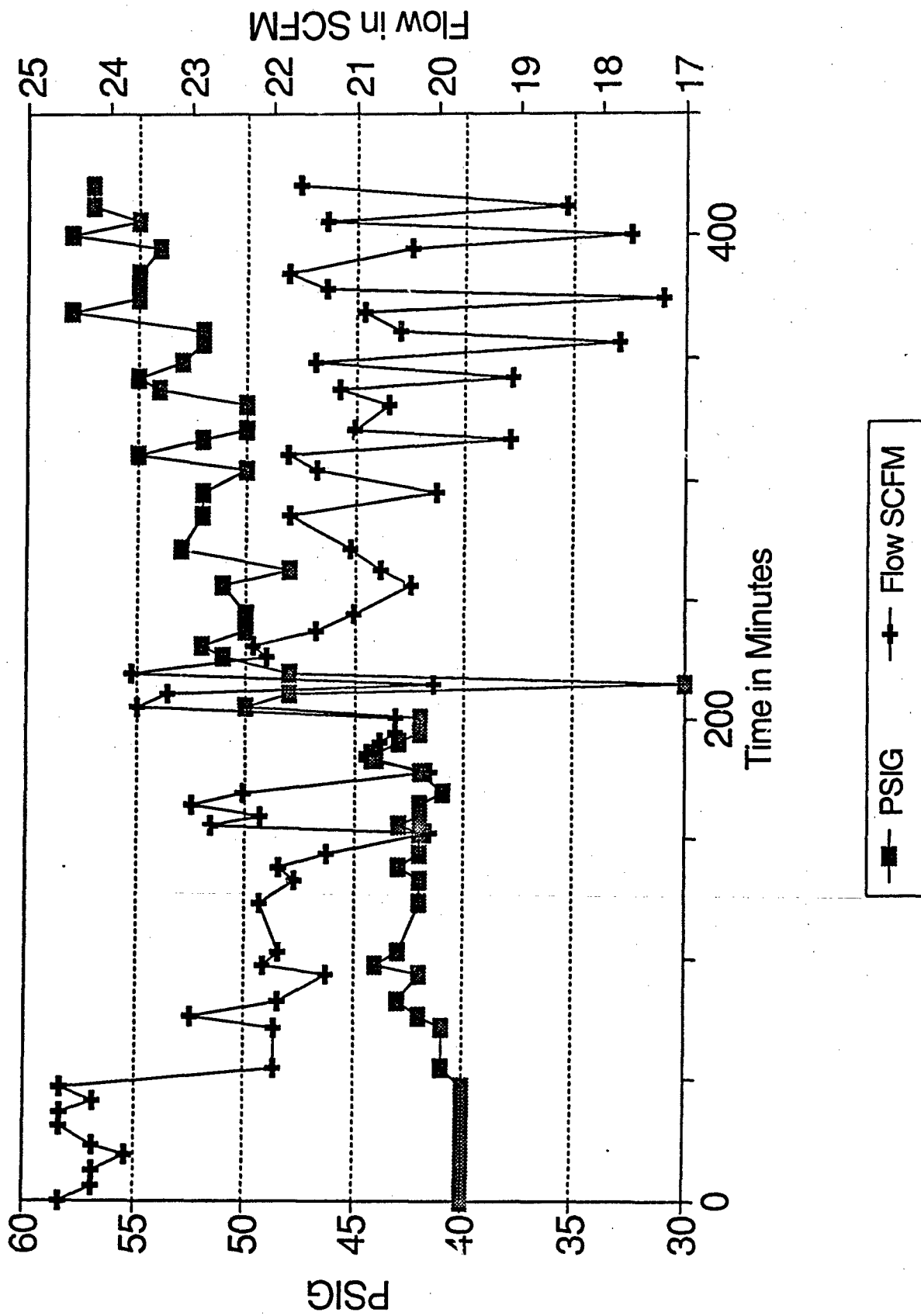
Graph 2



- Dewar 1-flow
- +— Dewar #2-flow
- *— Magnehelic #1
- Magnehelic #2
- x— Liq. Rate g/sec

Heat Exchanger Deriming Test #7 Dec 16 1990

Graph 3



The unit has demonstrated excellent deriming capability. A 5 min deriming cycle is necessary and adequate during the liquefaction period.

Graph 4 - Mask Oxygen Concentrations

The flow rate of oxygen to the mask from the dewars could be varied from 10 to 55 LPM by adjusting the back pressure regulator. The concentration of oxygen steadily increased as the last batch of oxygen was drawn from the dewar.

The level of concentration is shown to be well in excess of 95% based on measurements from the Beckman oxygen analyzer. This level, we believe, is a consequence of the oxygen separation during the liquefaction phase and batch distillation during the withdrawal phase. The average concentrations of the oxygen in the vent flow were consistently below that of the supply, indicating that separation was taking place. We expect that this phenomena would be observed from any stored LOX supply consisting of an oxygen, nitrogen, and argon mixture.

We suspect that the continuous reading oxygen concentrations are 1% to 3% higher than actual, since the MSOGS supply concentration shown in Graph 5 is 97%, while reliable manufacturers' data and theoretical limits to performance of the MSOGS of this type limit oxygen concentration supply levels to 95.1%.

Graph 5 - Oxygen Concentrations

The supply mask and vent concentrations with time are also given in Graph 5. In nearly all of the tests, the vent gas was about 1-2% lower in oxygen concentration than the supply, as we expect that some enrichment of oxygen takes place during liquefaction and filling. A vapor pressure separation occurred during mask liquid withdrawal which raised the oxygen concentration during the later stages of withdrawal, well above the MSOGS supply concentration. We do not know the composition of argon or nitrogen in the vent gas.

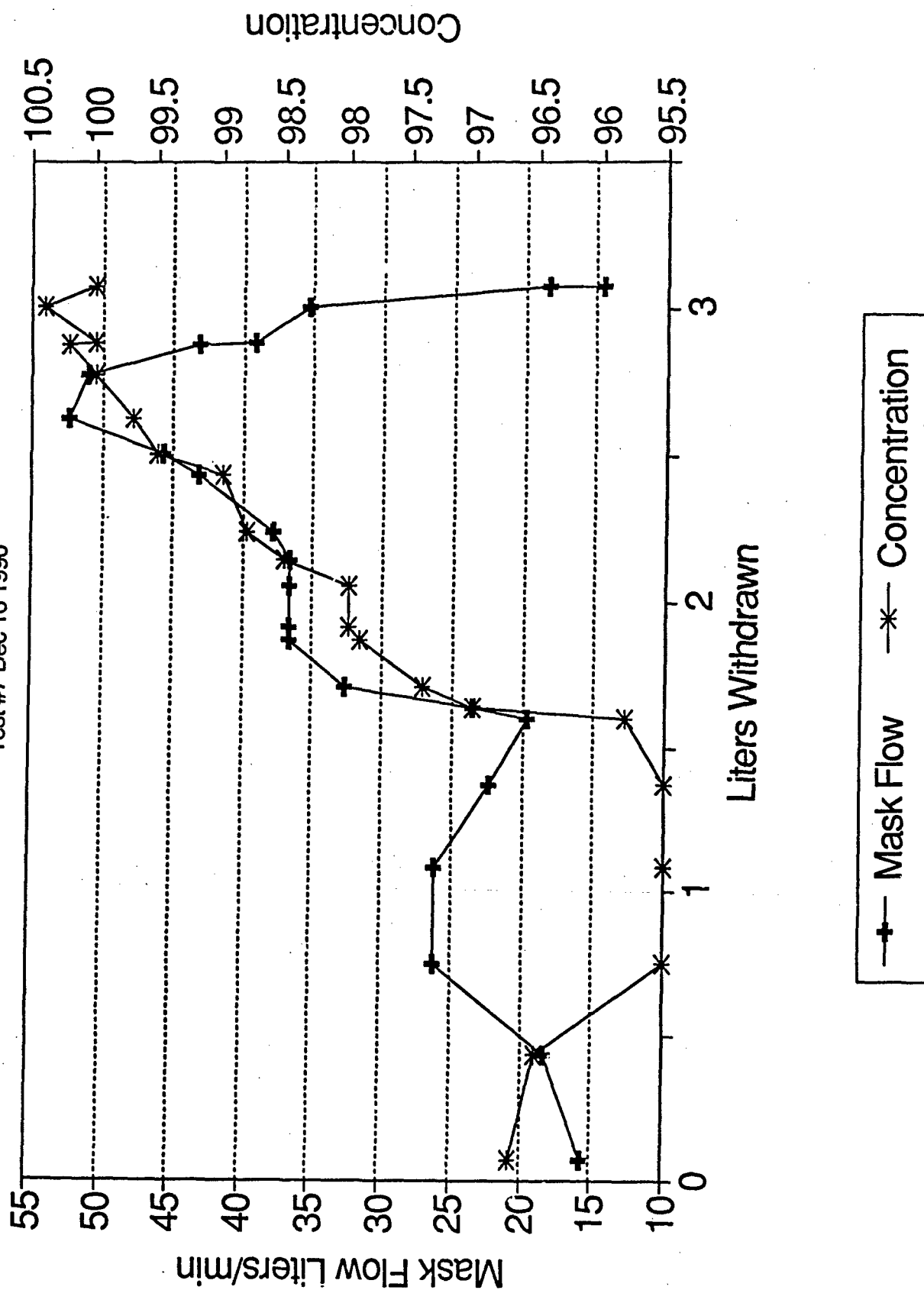
Graph 6 - Cold Head Performance

The predicted and actual cold head cooling performance and corresponding cold head temperature are shown in Graph 6. By oxygen Test No. 7, the cold head cooling capacity is about 50% of the original (and expected value). Loss of the helium charge as a result of connection and reconnection as well as leakage probably account for the loss. Scheduled recharging should be performed.

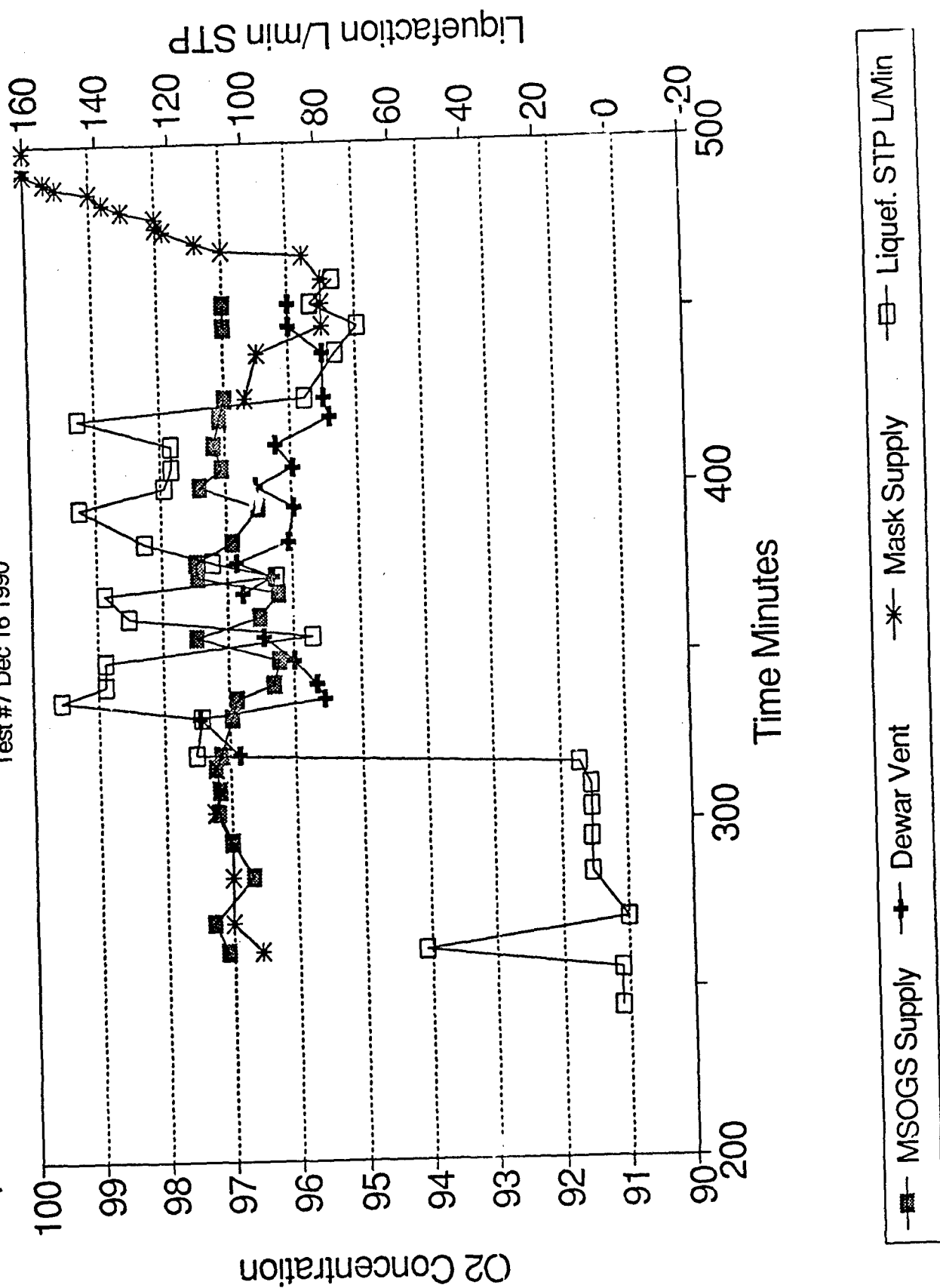
Graph 4

Mask O2 Concentrations

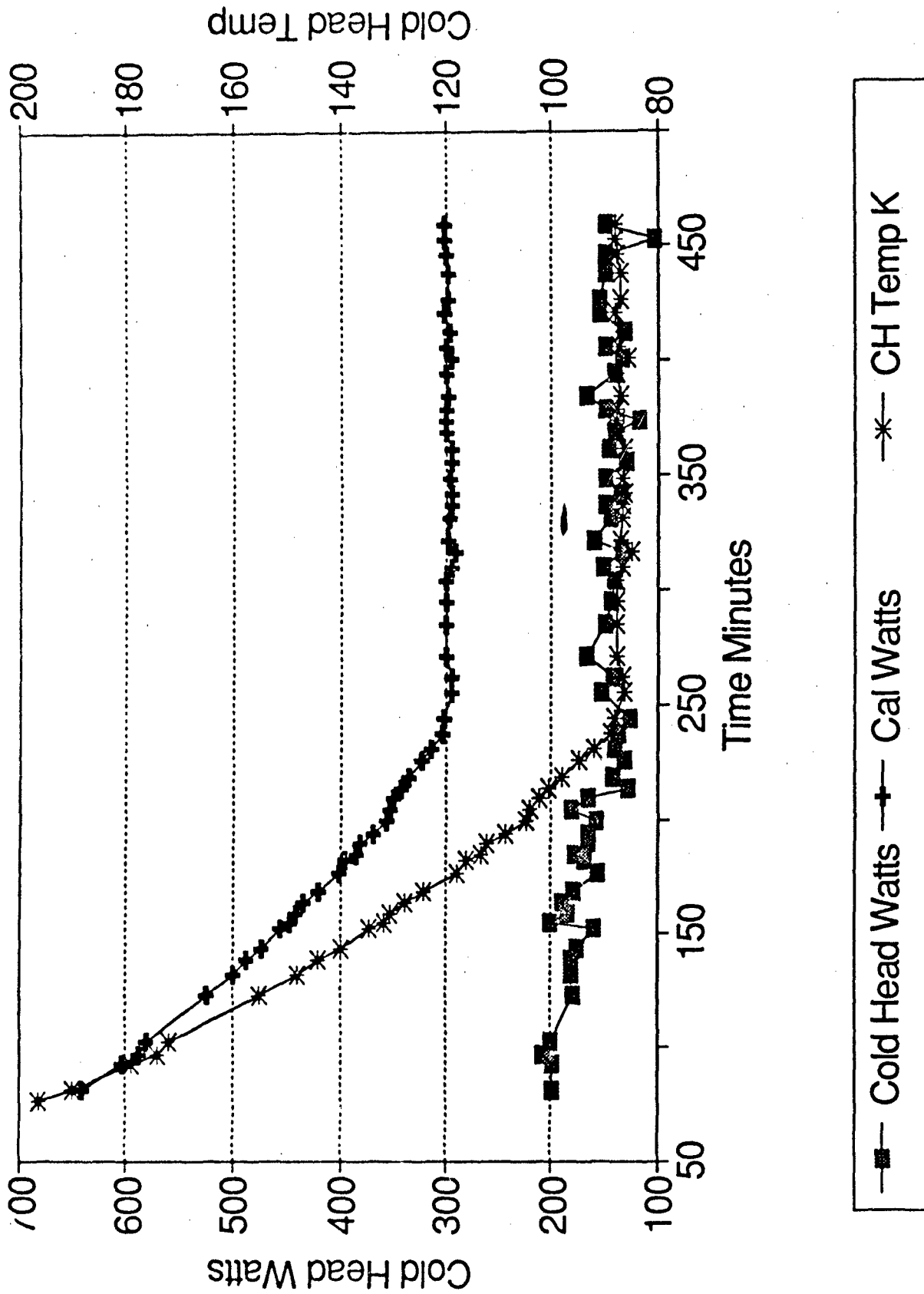
Test #7 Dec 16 1990



Graph 5
O2 Concentrations
Test #7 Dec 16 1990



Graph 6
Cold Head Performance
Test #7 Dec 16 1990



6. Conclusions and Recommendations

6.1 Conclusions

We have concluded that the open cycle HOS can liquefy and store oxygen from a MSOGS. We are pleased with the excellent performance of the reversing cycle through the initial cooldown (32°F), final cooldown (100°K), and liquefaction run operation (88°K). The HOS laboratory demonstrator has proven that the system can provide refrigeration with dirty (oil and particulates), humid (100% relative humidity) compressed air. We believe that the Air Force should continue the development of the open cycle hybrid system.

6.2 Hybrid Oxygen - Turboexpander Demonstration

We recommend the next step in the open cycle hybrid oxygen program involve the replacement of the cold head and JT valve with a cryogenic expander under concurrent development by the USAF Armstrong Laboratory, Brooks AFB, Texas. The integration of the expander can be accomplished with the existing demonstrator configuration by removing one of the cold heads and replacing it with an appropriately flanged miniature turboexpander.

The procedure for cooldown and operation should not be significantly different than that of the current laboratory demonstrator, except that during the initial cooldown process, a bypass branch around the turbine would be operated so that defrost during initial cooldown can be achieved. After the system has passed through the 20°F point, the standard deriming cycle can be used for the remainder of the cooldown and run.

We recommend that a large flow area exhaust plenum be integrated with the turbine so that it has high frost tolerance even though we were able to achieve cooldown in the cold head demonstrator with a small diameter cold head heat exchanger.

Upon the completion of the test with an integral turbine, we recommend undertaking the design, development and testing with a lightweight cryogenic heat exchanger replacing the commercial aluminum unit currently in the laboratory demonstrator. We expect that a heat exchanger of stainless steel, weighing about 10 lb, would meet the design performance.

6.3 Flight System Design

We further recommend a detailed design and development program of a flight system based on the .5 g/sec liquefier and 2 liters of liquid storage. The major focus of this stage would be to develop a flight-qualified system, including:

- lightweight stainless steel heat exchanger
- integrated liquefier and dewar
- automated controls

Preliminary Flight System

As outlined earlier, the purpose of the laboratory system is to validate the technical feasibility of generating and storing LOX from a bleed air-powered system. The laboratory demonstrator consists of commercially available components and lacks the customization which would be necessary for a flight model. To achieve the size and weight which would be appropriate for a flight system, many of the key components would be different. The most important changes would be the use of the lightweight stainless steel process heat exchanger and the inclusion of a small turboexpander in place of the reciprocating unit.

6.3.1 Flight Heat Exchanger The major characteristics of the flight heat exchanger is that it would be composed of thin stainless steel plates and fins. The fin plates would reduce the axial conduction which represents an inefficiency in the heat exchanger while thin fin plates enhance the crosswise conductivity, improving the effectiveness of the heat exchanger. The use of thin plates and ultra fins (.001 inch thickness) reduces the size and weight of the heat exchanger for the given duty. Table 22 characterizes the main features of a stainless steel heat exchanger derived from a Garrett design outline in a report entitled, *3.6 K Closed Cycle Turbo-refrigerator*, as compared with the aluminum heat exchanger used in the laboratory test. As indicated earlier, the most important characteristic of this heat exchanger is the thin low axial conductivity metal components used in its fabrication. A more complete description of the stainless heat exchanger is contained in Section 3.2.2, Heat Exchanger, which outlines the characteristics of the interactive component sizing computer program and contains the specifications of the stainless heat exchanger. The key size characteristics of the flight heat exchanger are summarized in Table 23.

Table 22. Comparison of Longitudinal Conduction 2" x 4" Heat Exchanger

Fin	Brazed Aluminum	Stainless
Spacing (fins per inch)	20	40
Number of fins	80	160
Height (inches)	.281	.250
Thickness (inches)	.01	.001
K Btu/hr-ft-R	298	7.1
KA Btu ⁱⁿ /hr-R	5.6	.023
Header Bars		
Width (inches)	0.57	0.50
K Btu/hr-ft-R	298	7.1
KA Btu ⁱⁿ /hr-R	5.6	.118
Side Plates		
Thickness	NA	.02
K Btu/hr-ft-R	298	7.1
KA Btu ⁱⁿ /hr-R	NA	.047
Total KA Btu ⁱⁿ /hr-R	11.2	.188

Table 23. Flight Heat Exchanger Size

Core length	7.09 inches
Header length	3.0 inches
Total length	10.0 inches
Volume	2 in. x 5.5 in. or .063 cu ft
Weight	4.6 lbs

6.3.2 Expander Figure 12 shows a schematic of the turbocxpander appropriate for operation at the design point of the prototype flight system. The unit is dramatically smaller than the commercially available reciprocating unit, though there are attendant development risks associated with this subsystem which involves the use of an extremely small 500,000 rpm turbine wheel supported by an active gas-bearing system.

SEAL BALANCE PORTS

2 X $\frac{5}{16}$ -24 UNJF-3A
MS-16142-02

SPEED SENSOR
CONNECTION
MIL-C-26482

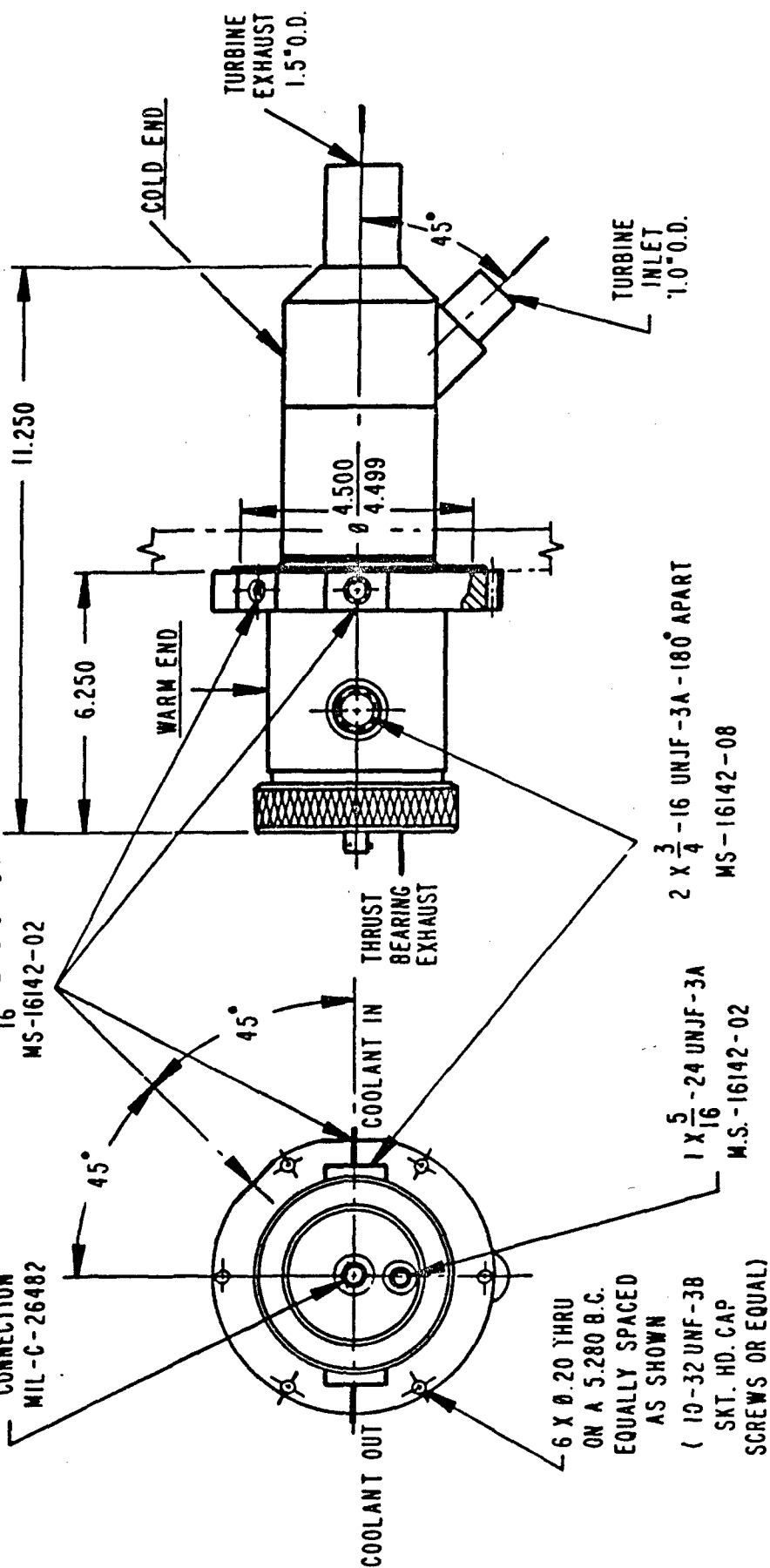


Figure 12. Turboexpander Layout for Aircraft Oxygen Liquefier.

7. Appendixes

7.1 Mission Scenario Specifications

The following tables are contained in the HYBRIDOX.WK1 document and summarize the expected size and weight of a HOS as compared to an MSOGS system without the liquefied storage.

Definition of the Columns on the Tables

Column

- 1 Altitude - aircraft altitude corresponding to the time in Column 4.
- 2 MSOGS Output - Essex MSOGS performance data for 94.5% purity MSOGS output in LPM-NTP. See Appendix 7.2 (Boeing Integration Report) for details of these characteristics.
- 3 O₂ Use Rate - LPM-NTP dilution air or 94.5% oxygen without dilution as indicated.
- 4 Time - in minutes.
- 5 Multiplier - activity and crew number multiplier taken from MIL-D-8683B.
- 6 O₂ - Liters-NTP consumed in each increment. The value at the top of the column is the total O₂ consumed in the entire mission.
- 7 MSOGS Output - LPM-NTP of LOX. The figures in the header refer to the liquefier rate and the MSOGS module fraction size 1 - standard size.
- 8 #1 dewar - the volume of LOX in the first dewar unit.
- 9 #2 dewar - the volume of LOX in the second dewar unit.
- 10 % backup - the fraction of mission remaining duration for which there is backup oxygen stored if the MSOGS is not operational. The dewars, MSOGS and liquefier were sized to meet the current demand, and 100% of the mission before 20,000 feet altitude is achieved in the initial climb.

Sizing Assumptions

The standard MSOGS system is assumed to consist of an Essex MSOGS (Figure 9) plus high pressure oxygen bottles of sufficient volume to allow completion of the mission in the event that the MSOGS fails to produce oxygen. Two standard high pressure bottles were chosen in the analysis. Bottles capable of supplying 2,200 liters NTP and 1,200 liters NTP were used throughout the analysis. These bottles weigh 26 and 24 lb, respectively.

The MSOGS unit used in the analysis is based on the 50 LPM-NTP Essex MSOGS unit which weighs 45 lb and is 35 liters in volume. It was assumed that smaller and larger MSOGS units could be provided, varying by size and weight in direct proportion to the rated oxygen capacity in LPM-NTP.

The hybrid unit sizing was developed in the tabular analysis of the mission. The dewar sizing was developed through the analysis of the mission profile assuming two dewars are used and each start with a small residual of LOX left over from the previous mission. The size of the

liquefier is based on the assumption that 2 liters and 10 lb of expander and related plumbing were necessary for a 40 LPM-NTP liquefier. (Later analysis of a stainless steel heat exchanger suggests that the weight would be closer to 4 lb.)

A minimum weight of 15 lb for the total dewar and liquefier for the minimum capacity unit was assumed.

The specific equations used in the spreadsheet analysis, file HYBRIDOX.WK1 follow:

Sizing Relations

Standard MSOGS

I. Volume and weight of backup high pressure oxygen bottle storage

Volume (liters) of backup = 17.99 or 9.48 liter volumes of high pressure gas for 2,200 liters NTP and 1,200 liters NTP high pressure backup bottles, respectively.

Weight (lbs) of backup = 26.04 or 23.9 lb for the 2,200 liters NTP and 1,200 liters NTP cylinders of high pressure gas, respectively.

II. Volume and weight of oxygen concentrator

Volume (liters) of MSOGS units = .7 liters per 1 LPM-NTP of oxygen

Weight (lbs) of MSOGS units = .9 lb per 1 LPM-NTP of oxygen

Based upon the specifications of the Essex Cryogenics 50 LPM concentrator data

III. Total volume and weight of MSOGS system with backup bottles

Total volume = volume backup gas cylinder + volume MSOGS unit

Total weight = weight backup gas cylinder + weight MSOGS units

Hybrid System

I. Volume and weight of liquefier

Volume (liters) of dewar and liquefier = liters dewar #1 + liters dewar #2 + 2.048 (liquefier capacity LPM-NTP/40) + total heat exchanger volume (based on compact heat exchanger design shown on Table 7.1, Preliminary Heat Exchanger Design, from spreadsheet HYBRIDOX.WK1)

Weight (lbs) of dewars and liquefier = 15 + (liquefier capacity LPM-NTP/40) 10 + total heat exchanger weight (based on compact heat exchanger also on HYBRIDOX.WK1)

II. Volume and weight of concentrator

Volume (liters) of MSOGS units = .7 liters per 1 LPM-NTP of oxygen

Weight (lbs) LPM MSOGS unit = .9 lb per 1 LPM-NTP of oxygen

III. Total volume of weight of HOS

Total hybrid volume = volume of dewars + liquefier + MSOGS units

Total hybrid weight = weight of dewars + liquefier + MSOGS units

Table A

MISSION PROFILE Fighter					% Dewar Remaining previous 75.00% liquefier 20 LPM-NTP 11.34				
Altitude	OBOGS Dilution Output O2 Consumed LPM-NTP LPM-NTP	Mins	Multi.	Dilution O2 L-NTP Used	OBOGS Mult #1 Output	0.11 dewar#2 0.75	0.11 dewar 0.75	%Back- up	
Mins.to fill				1142	NTP				
Ground	0 4.00	0	2.00	0.00	0.00	0.56	0.56	86%	
Eng On	40 4.00	15	2.00	120.06	4.46	0.43	0.64	91%	
Taxi	40 4.00	20	2.20	44.02	4.46	0.37	0.66	92%	
Take-off	40 4.00	30	2.80	112.06	4.46	0.25	0.72	97%	
5,000	50 2.50	31	2.60	10.40	5.58	0.25	0.70	97%	
10,000	60 2.51	33	2.20	11.00	6.69	0.27	0.69	99%	
15,000	70 2.51	35	2.00	10.06	7.81	0.29	0.68	101%	
20,000	75 2.51	36	2.00	5.03	8.37	0.30	0.67	102%	
25,000	80 2.51	37	2.00	5.03	8.92	0.31	0.67	104%	
25,000	80 2.51	38	2.00	5.03	8.92	0.32	0.66	104%	
25,000	80 2.51	39	2.00	5.03	8.92	0.33	0.66	105%	
25,000	80 2.51	40	2.00	5.03	8.92	0.34	0.65	106%	
30,000	90 2.38	41	2.00	5.03	10.04	0.35	0.65	107%	
35,000	95 2.33	43	2.00	9.53	10.60	0.37	0.63	110%	
40,000	100 2.40	45	2.00	9.34	11.16	0.40	0.62	113%	
45,000	110 2.58	47	2.00	9.60	12.27	0.43	0.61	116%	
50,000	120 2.65	49	2.00	10.32	13.39	0.46	0.60	120%	
60,000	120 3.23	60	2.00	58.34	13.39	0.62	0.53	143%	
60,000	120 3.23	80	3.25	210.21	13.39	0.93	0.29	214%	
60,000	120 3.23	105	3.25	262.76	13.39	0.63	0.68	485%	
40,000	120 3.23	115	3.25	105.11	13.39	0.51	0.83	901%	
20,000	120 3.23	125	2.00	64.68	13.39	0.66	0.76	1908%	
1,000	120 3.23	135	2.00	64.68	13.39	0.82	0.68	2015%	
Standard OBOGS					Hybrid				
Liters weight					Liters weight				
Backup + Liquefier					17.99 26.04	7.83 29.19			
50 LPM OBOGS Units					8.12 10.37	3.23 4.13			
					26.11 36.41	11.06 33.32			
						57.64%	8.48%		
Empty dewar fill time in mins.					216.83				

Table B

Altitude	MISSION PROFILE Fighter				% Dewar Remaining previous 75.00%				
	OBOGS Dilution		Mins	Multi.	95% O2 L-NTP Used	OBOGS #1 Output	liquefier		75.00% 12.69
	Output LPM-NTP	O2 Consumed LPM-NTP					OBOG Mult	40 LPM-NTP dewar#2	
Mins.to fill					2846	NTP		1.60	XBack- up
Ground	0	4.00	0	2.00	0.00	0.00		1.20	73%
Eng On	40	4.00	15	2.00	399.00	14.40		0.74	78%
Taxi	40	4.00	20	2.20	146.30	14.40		0.58	80%
Take-off	40	4.00	30	2.80	372.40	14.40		0.15	83%
5,000	50	2.50	31	2.60	28.93	18.00		0.17	84%
10,000	60	2.51	33	2.20	43.39	21.60		0.22	86%
15,000	70	2.51	35	2.00	39.45	25.20		0.28	88%
20,000	75	2.51	36	2.00	19.72	27.00		0.31	90%
25,000	80	2.51	37	2.00	19.00	28.80		0.34	93%
25,000	80	2.51	38	2.00	19.00	28.80		0.38	93%
25,000	80	2.51	39	2.00	19.00	28.80		0.41	94%
25,000	80	2.51	40	2.00	19.00	28.80		0.44	96%
30,000	90	2.38	41	2.00	17.01	32.40		0.48	98%
35,000	95	2.33	43	2.00	31.12	34.20		0.56	102%
40,000	100	2.40	45	2.00	28.59	36.00		0.64	106%
45,000	110	2.58	47	2.00	26.06	39.60		0.73	111%
50,000	120	2.65	49	2.00	24.25	40.00		0.82	116%
60,000	120	3.23	60	2.00	117.44	40.00		1.32	147%
60,000	120	3.23	80	3.25	346.98	40.00		0.93	233%
60,000	120	3.23	105	3.25	433.72	40.00		2.07	459%
40,000	120	3.23	115	3.25	232.30	40.00		1.60	601%
20,000	120	3.23	125	2.00	197.24	40.00		1.60	1047%
1,000	120	3.23	135	2.00	266.00	40.00		1.60	1047%
					Standard OBOGS		Hybrid		
					Liters	weight	Liters	weight	
Backup + Liquefier					27.47	49.94	9.53	33.76	
50 LPM OBOGS Units					27.00	34.45	10.44	13.32	
					54.47	84.39	19.97	47.08	
							63.35%	44.21%	
Empty dewar fill time in mins.					143.33				

Table C

MISSION PROFILE					% Dewar Remaining previous 75.00%				
Altitude	OBOGS Dilution		Fighter		Decompressliquefier		20 LPM-NTP		11.34
	Output	O2 Consumed	Mins	Multi.	95% OBOGS Mult	0.16	0.16	%Back-up	
	LPM-NTP	LPM-NTP			O2 L-NTP	OBOGS #1	dewar#2		dewar
Mins.to fill					Used	Output	1.10	1.10	
Ground	0	4.00	0	2.00	1692	NTP			
Eng On	40	4.00	15	2.00	0.00	0.00	0.83	0.83	85%
Taxi	40	4.00	20	2.20	399.00	6.60	0.37	0.94	88%
Take-off	40	4.00	30	2.80	146.30	6.60	0.20	0.98	89%
5,000	40	4.00	30	2.80	372.40	6.60	0.28	0.55	93%
5,000	50	2.50	31	2.60	28.93	8.24	0.29	0.52	94%
10,000	60	2.51	33	2.20	40.21	9.89	0.31	0.47	96%
15,000	70	2.51	35	2.00	30.40	11.54	0.34	0.44	99%
20,000	75	2.51	36	2.00	12.67	12.37	0.35	0.42	101%
25,000	80	2.51	37	2.00	10.13	13.19	0.36	0.41	107%
25,000	80	2.51	38	2.00	10.13	13.19	0.38	0.40	105%
25,000	80	2.51	39	2.00	10.13	13.19	0.37	0.41	108%
25,000	80	2.51	40	2.00	10.13	13.19	0.36	0.43	110%
30,000	90	2.38	41	2.00	7.78	14.84	0.37	0.42	112%
35,000	95	2.33	43	2.00	12.30	15.66	0.41	0.41	118%
40,000	100	2.40	45	2.00	9.41	16.49	0.45	0.40	124%
45,000	110	2.58	47	2.00	7.60	18.14	0.49	0.39	130%
50,000	120	2.65	49	2.00	6.51	19.79	0.53	0.38	137%
60,000	120	3.23	60	2.00	15.92	19.79	0.52	0.63	177%
60,000	120	3.23	80	3.25	47.05	19.79	0.46	1.08	260%
60,000	120	3.23	105	3.25	58.81	19.79	1.03	1.01	389%
40,000	120	3.23	115	3.25	76.45	19.79	0.94	1.24	497%
20,000	120	3.23	125	2.00	126.67	19.79	1.17	1.09	776%
1,000	120	3.23	135	2.00	253.33	19.79	0.88	1.32	754%

	Standard OBOGS		Hybrid	
	Liters	weight	Liters	weight
Backup + Liquefier	17.99	26.04	8.53	30.52
50 LPM OBOGS Units	27.00	34.45	4.78	6.10
	44.99	60.49	13.31	36.62
			70.42%	39.46%
Empty dewar fill time in mins.	215.14			

Table D

Altitude	MISSION PROFILE				% Dewar Remaining previous 1.00%				
	Fighter				Decompress	Liquefier	30 LPM-NTP	10.79	
	OBOGS Output	Dilution O2 Consumed	Mins	Multif.	95% OBOG Mult	OBOGS #1	0.35	0.20	
	LPM-NTP	LPM-NTP			O2 L-NTP Used	OBOGS #1 Output	dewar#2	dewar	%Back-up
Mins.to fill					775	NTP	1.10	1.10	
Ground	0	4.00	0	2.00	0.00	0.00	0.01	0.01	2%
Eng On	40	4.00	15	2.00	0.00	14.00	0.25	0.01	29%
Taxi	40	4.00	20	2.20	0.00	14.00	0.33	0.01	38%
Take-off	40	4.00	30	2.80	0.00	14.00	0.49	0.01	56%
5,000	50	2.50	31	2.60	28.93	17.50	0.51	-0.02	57%
10,000	60	2.51	33	2.20	40.21	21.00	0.47	0.03	61%
15,000	70	2.51	35	2.00	30.40	24.50	0.52	-0.01	66%
20,000	75	2.51	36	2.00	12.67	26.25	0.55	-0.02	69%
25,000	80	2.51	37	2.00	10.13	28.00	0.58	-0.03	75%
25,000	80	2.51	38	2.00	10.13	28.00	0.62	-0.05	77%
25,000	80	2.51	39	2.00	10.13	28.00	0.60	-0.01	81%
25,000	80	2.51	40	2.00	10.13	28.00	0.59	0.02	85%
30,000	90	2.38	41	2.00	7.78	30.00	0.63	0.01	90%
35,000	95	2.33	43	2.00	12.30	30.00	0.69	-0.01	100%
40,000	100	2.40	45	2.00	9.41	30.00	0.76	-0.02	110%
45,000	110	2.58	47	2.00	7.60	30.00	0.83	-0.02	120%
50,000	120	2.65	49	2.00	6.51	30.00	0.90	-0.03	131%
60,000	120	3.23	60	2.00	15.92	30.00	0.88	0.34	190%
60,000	120	3.23	80	3.25	47.05	30.00	0.83	1.03	314%
60,000	120	3.23	105	3.25	58.81	30.00	1.69	0.96	505%
40,000	120	3.23	115	3.25	76.45	30.00	1.60	1.31	665%
20,000	120	3.23	125	2.00	126.67	30.00	1.94	1.16	1065%
1,000	120	3.23	135	2.00	253.33	30.00	1.65	1.50	1084%
					Standard OBOGS	Hybrid			
					Liters weight	Liters weight			
Backup + Liquefier					17.99 26.04	5.93 25.85			
50 LPM OBOGS Units					16.78 21.41	10.15 12.95			
					34.77 47.45	16.08 38.80			
						53.76%	18.24%		

Table E

MISSION PROFILE B1-Bomber					% Dewar Remaining previous 75.00%				
Altitude	OBOGS Dilution		Mins	Multi.	Dilution O2 L-NTP Used	Liquefier		40 LPM-NTP	
	Output LPM-NTP	O2 Consumed LPM-NTP				OBOGS Mult Output	OBOGS #1 NTP	0.28 dewar#2	0.28 dewar
Mins.to fill								6.00	6.00
Ground	0	4.00	8	6	638.40	0.00		4.50	4.50
Eng On	40	4.00	15	6	168.08	11.10		4.31	4.59
Taxi	40	4.00	20	6	120.06	11.10		4.17	4.65
Take-off	40	4.00	30	8	324.16	11.10		3.80	4.78
5,000	50	2.50	31	7	28.00	13.88		3.82	4.75
10,000	60	2.51	33	6	30.00	16.65		3.85	4.71
15,000	70	2.51	35	6	30.17	19.43		3.90	4.68
20,000	75	2.51	37	6	30.17	20.82		3.95	4.64
25,000	80	2.51	100	6	950.29	22.20		5.54	3.56
25,000	80	2.51	150	6	754.20	22.20		4.68	4.83
25,000	80	2.51	200	6	754.20	22.20		3.82	6.10
25,000	80	2.51	250	6	754.20	22.20		2.96	7.36
25,000	80	2.51	300	6	754.20	22.20		4.23	6.50
30,000	90	2.38	340	6	603.36	24.98		5.37	5.81
35,000	95	2.33	350	6	142.92	26.37		5.67	5.65
40,000	100	2.40	360	7	163.38	27.75		5.99	5.46
45,000	110	2.58	380	8	384.00	30.53		6.69	5.02
50,000	120	2.65	440	11	1625.40	33.31		4.83	7.31
60,000	120	3.23	441	0	0.00	33.31		4.87	7.31
60,000	120	3.23	442	0	0.00	33.31		4.90	7.31
60,000	120	3.23	443	0	0.00	33.31		4.94	7.31
40,000	120	3.23	450	8	181.10	33.31		5.21	7.10
20,000	120	3.23	460	8	258.72	33.31		5.59	6.80
1,000	120	3.23	480	6	388.08	33.31		6.35	6.36
					Standard OBOGS		Hybrid		
					Liters	weight	Liters	weight	
Backup + Liquefier					81.46	128.06	18.33	63.83	
50 LPM OBOGS Units					23.50	29.98	8.05	10.27	
					104.96	158.04	26.38	74.10	
							74.87%	53.11%	
Empty dewar fill time in mins.					697.18				

Table F

MISSION PROFILE B1-Bomber					% Dewar Remaining previous 75.00% liquefier 100 LPM-NTP 36.72				
Altitude	OBOGS Output LPM-NTP	Dilution O2 Consumed LPM-NTP	Mins	Multi.	95% O2 L-NTP Used	OBOGS #1 Output	OBOGS Mult 0.83	#2 dewar 0.83	%Back- up
Mins.to fill					28752	NTP			
Ground	0	4.00	8	6	638.40	0.00	10.50	10.50	65%
Eng On	40	4.00	15	6	558.60	33.08	9.86	10.76	65%
Taxi	40	4.00	20	6	399.00	33.08	9.41	10.95	65%
Take-off	40	4.00	30	8	1077.30	33.08	8.17	11.33	65%
5,000	50	2.50	31	7	77.90	41.35	8.22	11.24	65%
10,000	60	2.51	33	6	118.34	49.62	8.34	11.11	65%
15,000	70	2.51	35	6	118.34	57.89	8.47	10.97	66%
20,000	75	2.51	37	6	118.34	62.03	8.61	10.84	66%
25,000	80	2.51	100	6	3591.00	66.16	13.37	6.73	107%
25,000	80	2.51	150	6	2850.00	66.16	10.12	10.51	93%
25,000	80	2.51	200	6	2850.00	66.16	6.86	14.29	110%
25,000	80	2.51	250	6	2850.00	66.16	3.60	18.07	140%
25,000	80	2.51	300	6	2850.00	66.16	7.38	14.82	181%
30,000	90	2.38	340	6	2041.14	74.43	10.78	12.48	235%
35,000	95	2.33	350	6	466.86	78.57	11.68	11.95	252%
40,000	100	2.40	360	7	500.33	82.70	12.63	11.38	273%
45,000	110	2.58	380	8	1042.29	90.97	14.71	10.19	328%
50,000	120	2.65	440	11	3819.00	99.24	10.34	16.99	854%
60,000	120	3.23	441	0	0.00	99.24	10.46	16.99	857%
60,000	120	3.23	442	0	0.00	99.24	10.57	16.99	861%
60,000	120	3.23	443	0	0.00	99.24	10.68	16.99	865%
40,000	120	3.23	450	8	400.27	99.24	11.48	16.54	1022%
20,000	120	3.23	460	8	788.95	99.24	12.61	15.63	1540%
1,000	120	3.23	480	6	1596.00	99.24	14.88	13.81	1564%
Standard OBOGS					Hybrid				
Liters					Liters				
weight					weight				
Backup + Liquefier					225.41 336.38 38.42 132.89				
50 LPM OBOGS Units					78.10 99.65 23.98 30.60				
					303.51 436.03 62.41 163.49				
					79.44% 62.50%				
Empty dewar fill time in mins.					545.94				

MISSION PROFILE					% Dewar Remaining previous 75.00%				
B1-Bomber					Decompressliquefier 80 LPM-NTP 25.37				
Altitude	OBOGS Dilution		Mins	Multi.	95% OBOG Mult 0.49 0.49		OBOGS #1 dewar#2 dewar	%Back-up	
	Output LPM-NTP	O2 Consumed LPM-NTP			O2 L-NTP Used	OBOGS Output			
Mins.to fill					15795	NTP	11.00	11.00	up
Ground	0	4.00	8	6	638.40	0.00	8.25	8.25	95%
Eng On	40	4.00	15	6	558.60	19.54	7.61	8.41	95%
Taxi	40	4.00	20	6	399.00	19.54	7.16	8.52	96%
Take-off	40	4.00	30	8	1077.30	19.54	5.92	8.74	97%
5,000	50	2.50	31	7	77.90	24.42	5.95	8.65	97%
10,000	60	2.51	33	6	109.66	29.31	6.02	8.53	98%
15,000	70	2.51	35	6	91.20	34.19	6.10	8.42	98%
20,000	75	2.51	37	6	76.00	36.64	6.18	8.34	99%
25,000	80	2.51	100	6	1915.20	39.08	8.99	6.15	169%
25,000	80	2.51	150	6	1520.00	39.08	11.23	4.41	146%
25,000	80	2.51	200	6	1520.00	39.08	9.49	6.64	180%
25,000	80	2.51	250	6	1520.00	39.08	7.75	8.88	230%
25,000	80	2.51	300	6	1520.00	39.08	6.02	11.11	312%
30,000	90	2.38	340	6	933.71	43.96	8.03	10.04	410%
35,000	95	2.33	350	6	184.57	46.41	8.56	9.83	438%
40,000	100	2.40	360	7	164.67	48.85	9.11	9.64	468%
45,000	110	2.58	380	8	304.00	53.73	10.34	9.30	537%
50,000	120	2.65	440	11	1026.00	58.62	14.36	8.12	906%
60,000	120	3.23	441	0	0.00	58.62	14.36	8.12	906%
60,000	120	3.23	442	0	0.00	58.62	14.36	8.12	906%
60,000	120	3.23	443	0	0.00	58.62	14.36	8.12	906%
40,000	120	3.23	450	8	131.73	58.62	14.21	8.59	979%
20,000	120	3.23	460	8	506.67	58.62	13.63	9.26	1310%
1,000	120	3.23	480	6	1520.00	58.62	11.90	10.60	1288%

	Standard OBOGS		Hybrid	
	Liters	weight	Liters	weight
Backup + Liquefier	117.44	180.14	30.37	104.29
50 LPM OBOGS Units	78.10	99.65	14.17	18.07
	195.55	279.79	44.54	122.37
			77.22%	56.27%
Empty dewar fill time in mins.	726.23			

Table H

Altitude	MISSION PROFILE				% Dewar Remaining previous 1.00%				
	B1-Bomber				Decompressliquefier	80 LPM-NTP	22.11		
	OBOGS Output	Dilution O2 Consumed	Mins	Multi.	95% OBOG Mult	0.76	0.76		
	LPM-NTP	LPM-NTP			O2 L-NTP Used	OBOGS #1 Output	dewar#2 Output	%Back-up	
Mins.to fill					13121	NTP	11.00	11.00	
Ground	0	4.00	8	6	0.00	0.00	0.11	0.11	1%
Eng On	40	4.00	15	6	0.00	30.45	0.11	0.11	1%
Taxi	40	4.00	20	6	0.00	30.45	0.28	0.11	3%
Take-off	40	4.00	30	8	0.00	30.45	0.63	0.11	5%
5,000	50	2.50	31	7	77.90	38.06	0.68	0.02	5%
10,000	60	2.51	33	6	109.66	45.67	0.55	0.13	5%
15,000	70	2.51	35	6	91.20	53.28	0.67	0.02	5%
20,000	75	2.51	37	6	76.00	57.09	0.80	-0.07	5%
25,000	80	2.51	100	6	1915.20	60.90	-1.39	4.32	33%
25,000	80	2.51	150	6	1520.00	60.90	2.09	2.58	44%
25,000	80	2.51	200	6	1520.00	60.90	0.36	6.06	71%
25,000	80	2.51	250	6	1520.00	60.90	3.44	4.32	113%
25,000	80	2.51	300	6	1520.00	60.90	2.10	7.80	181%
30,000	90	2.38	340	6	933.71	68.51	5.23	6.74	271%
35,000	95	2.33	350	6	184.57	72.32	6.06	6.53	300%
40,000	100	2.40	360	7	164.67	76.12	6.93	6.34	331%
45,000	110	2.58	380	8	304.00	80.00	8.76	5.99	403%
50,000	120	2.65	440	11	1026.00	80.00	14.24	4.82	768%
60,000	120	3.23	441	0	0.00	80.00	14.24	4.82	768%
60,000	120	3.23	442	0	0.00	80.00	14.24	4.82	768%
60,000	120	3.23	443	0	0.00	80.00	14.33	4.82	772%
40,000	120	3.23	450	8	131.73	80.00	14.18	5.46	843%
20,000	120	3.23	460	8	506.67	80.00	15.10	4.88	1143%
1,000	120	3.23	480	6	1520.00	80.00	13.36	6.71	1149%

	Standard OBOGS		Hybrid	
	Liters	weight	Liters	weight
Backup + Liquefier	99.45	154.10	27.77	57.94
50 LPM OBOGS Units	45.18	57.65	22.08	28.16
	144.63	211.75	49.85	86.11
			65.53%	59.34%
Empty dewar fill time in mins.	6.21			

7.2 Boeing Integration Report

HYBRID OXYGEN SYSTEM
PHASE I SUPPORT

TECHNICAL REPORT
REVISION A

Submitted to:

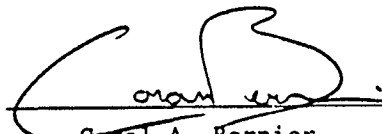
Arthur D. Little, Inc.
Acorn Park
Cambridge, Massachusetts 02140

November 25, 1987

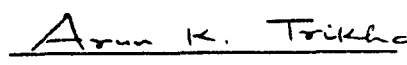
Submitted by:

The Boeing Advanced Systems Company
A Division of The Boeing Company
Seattle, Washington 98124

Prepared by:


Coral A. Bernier
Principal Investigator

Approved by:


Arun K. Trikha
Program Manager

FOREWORD

Revision A to the Technical Report is a major revision to the original report submitted in August 1987. It includes:

- (1) Updated bleed air analysis based on A. D. Little's configuration change reflected in the May 1987 version of the Phase I Report (Reference 1).
- (2) Appendix 1 containing detailed analysis of Hybrid Oxygen System retrofit in the F-15 aircraft.
- (3) Appendix 2 providing the methodology and resources to calculate partial pressures of atmospheric constituents as a function of geometric altitude.

TABLE OF CONTENTS

<u>Section</u>	<u>Page</u>
Foreword	ii
Table of Contents	iii
List of Tables	iv
List of Figures	v
1.0 Summary/Introduction	1
2.0 Integration Issues	1
3.0 Bleed Air Availability	2
4.0 Excess Cooling Potential	2
5.0 Conclusions	3
Appendix 1	6
Appendix 2	13
References	32

LIST OF TABLES

<u>Number</u>	<u>Title</u>	<u>Page</u>
1	Fighter Mission Bleed Air Availability	4
2	Bomber Mission Bleed Air Availability	5
1-1	F-15 ECS Item Description	10
2-1	Molecular Weights and Fractional Volume Concentration of Sea-Level Dry Air	14
2-2	Geometric Altitudes	15

LIST OF FIGURES

<u>Number</u>	<u>Title</u>	<u>Page</u>
1-1	F-15 Environmental Control System	11
1-2	F-15 Installation of the Hybrid Oxygen System	12

1.0 SUMMARY/INTRODUCTION

This report was prepared by the Boeing Advanced Systems Company to summarize the work conducted in support of Arthur D. Little, Inc. (ADL) on the Hybrid Oxygen System (HOS) program. This report contains information on availability of bleed air for the HOS in typical fighter and bomber aircraft, discussions on system integration and feasibility, and identification of potential use and function of the additional cooling capability inherent in the HOS design.

The HOS discussed herein is based on the system as described in ADL's Phase I Report (Reference 1). The HOS consists of an onboard oxygen liquifier which processes the concentrated oxygen from the onboard oxygen generation system (OBOGS) and stores it as liquid (see Figures 5 and 6 of Reference 1). Bleed air is required for two functions in the hybrid system:

1. Liquid oxygen generation, and
2. Heat sink for oxygen liquefaction.

Requirements for the bleed air supply for oxygen concentrating through the OBOGS are established based on oxygen breathing volume requirements. The design flow of oxygen from the OBOGS in the HOS is set at 40 LPM (NTP). The required bleed air flow for liquifaction is a function of the OBOGS output, the temperature of the heat sink bleed air at the exit of the liquifier heat exchanger, and the heat removal required for oxygen liquifaction.

2.0 INTEGRATION ISSUES

Based on the liquifier design trade studies conducted by ADL, it was determined that the amount of bleed air required for system operation ranged from 1.2 to 1.9 lbm/min. The variation in flow rate is related to the selected design condition at the exit of the liquifier heat exchanger. An OBOGS unit was recently flight tested as part of the Tactical Life Support System demonstration (Reference 2) on an F-15 aircraft. This unit required a flow rate of 2 lbm/min and its installation in the F-15 and consequent flow extraction proved to have minimal effect on the aircraft environmental control system (ECS). This fact was verified by computer

simulation and evaluation of flight test data. It is therefore believed that retrofit installation of the Hybrid Oxygen System in the F-15 and other aircraft will render no adverse effects either on the operation of the aircraft ECS or on aircraft performance. The HOS, with its low weight and volume characteristics, provides an excellent alternative to onboard stored oxygen. This system, because it produces oxygen at conventional pressure, will be compatible with either the CRU-73 or BRAG regulator. Appendix 1 examines the F-15 retrofit installation of the HOS in detail.

3.0 BLEED AIR AVAILABILITY

The quantity and thermodynamic state of the ECS bleed air supply is given in Tables 1 and 2 for generic fighter and bomber missions, respectively. These conditions, as well as ECS bleed air location, were chosen based on design specification values for a typical (NGL) OBOGS unit (i.e., 25 to 90 psig, required performance from 0°F to 100°F, operating to 160°F). The conditions from the fighter mission (Table 1) most closely matching the required OBOGS conditions are at the exit of the secondary heat exchanger. A heat exchanger may be necessary prior to OBOGS in the case of the bomber mission.

4.0 EXCESS COOLING POTENTIAL

Depending on the design and baseline operating state of the HOS, a range of options exist for generating excess cooling potential. For instance, the bleed air stream at the exit of the liquefier section, disregarding the low pressure, has excellent cooling potential. Also, the liquid oxygen after extraction from the dewar could be used as a heat sink prior to delivery to mask. Analysis of the HOS performance as well as overall aircraft ECS performance would have to be done to determine the worth or advantage of external use of excess cooling from the HOS. In order to determine where to tap off of the HOS for additional cooling the following questions must be answered or trades conducted:

1. What are the payoffs and/or weight penalties associated with increased cooling capability at the expense of increased sizing in heat exchangers or turboexpander?

2. What are the payoffs and/or weight penalties associated with increased cooling capability achieved through increased OBOGS output?

Additional cooling is necessary in present day and future electronics exhibiting high power density. Specific needs include cooling for avionics, advanced sensors, and applications for VHSIC (very high speed integrated circuit) and VLSIC (very large scale integrated circuits). The HOS could also be designed to use the vent bleed air as a bootstrap to precool the inlet bleed air. Use of the hybrid system as a bleed air conditioner and additional details on specific application of the excess cooling potential is found in Appendix 1.

5.0 CONCLUSIONS

The HOS being developed by A. D. Little, by processing the concentrated oxygen from the OBOGS and storing it as a liquid, preserves the conventional reliability and convenience of stored onboard oxygen while reducing the logistics burden of stored onboard liquid oxygen (LOX). The HOS, with its minimal bleed air requirements, can be used as a retrofit solution to LOX in current aircraft and as an alternative solution to LOX in future aircraft. The hybrid system also provides an excellent source of clean conditioned bleed air and inherently possesses potential for use in aircraft/avionics cooling.

TABLE 1: FIGHTER MISSION BLEED AIR AVAILABILITY

CONDITION	ALTITUDE 1000 Ft. No.	Single Primary HX			Secondary HX		
		POUT psia	TOUT OF	V lb/min	POUT psia	TOUT OF	V lb/min
Taxi	0	43	245	55	67	143	73
T/O	0	76	276	55	75	120	83
Climb	35	38	492	39	56	130	63
Supersonic Climb & Cruise	60	62	505	45	85	320	84
Descent	30	29	212	42	37	22	44

NOTES: Hot day condition
Fighter Mission
Air Cycle ECS

L-7170-AKT-87-046

TABLE 2: BOMBER MISSION BLEED AIR AVAILABILITY

Design Condition	ALTITUDE 1000 Ft. No.	Precooler Outlet			Primary HX Outlet		
		Pphx psia	Tphx OF	Vphx lb/min	Phx psia	Thx OF	Vhx lb/min
Taxi	0	88	250	125	77	195	123.5
T/O	0	131	234	126	77	187	124.5
LO Cruise	27	73	210	121	39	164	120
HI Cruise	47	29	222	111	28	136	104
Descent	47	24	282	77	23	166	70
Landing	0	105	175	77	48	142	76

NOTES: Hot Day Condition.
Bomber Mission
Air Cycle ECS

L-7170-AKT-87-046

APPENDIX 1: F-15 RETROFIT ANALYSIS

This appendix presents design integration issues and identifies the advantages of retrofit of the Hybrid Oxygen System (HOS) into the F-15 aircraft. The F-15 is only used as an example to illustrate the methodology required for retrofit analysis and to specify the potential worth of the HOS both as an air purification system and as a valuable source for air for avionics cooling. In this day and age when the threat of chemical, biological, and radiological (CBR) warfare is very real, any system offering inherent protection should be seriously evaluated. Contained within this appendix are the following specific items:

- (1) Retrofit of the HOS in the F-15
- (2) Use of the Conditioned Bleed Air
- (3) Use of Excess Cooling Potential

F-15 Retrofit of the Hybrid Oxygen System

The F-15 is a single-place supersonic fighter aircraft powered by two engines. The F-15 environmental control system (ECS) provides conditioned air to the cockpit for heating, cooling, pressurization, and ventilation; to the aircraft avionic compartments requiring environmental temperature control; and to those equipment units requiring direct forced air cooling. In addition, conditioned air is provided for windshield anti-icing and anti-fogging and for pressurization of the cockpit canopy seal, fuel tanks, anti-G suits and that avionic equipment requiring pressurization.

A schematic of the F-15 bleed air and environmental control system is given in Figure 1-1. A detailed description of the system and of each of the subsystems/components of the F-15 ECS may be found in References (3) and (4). Bleed air in the F-15 is obtained from the final compressor stage bleed port of each engine. The bleed control system is duplicated for each engine and consists of a primary bleed air pressure regulator/shutoff valve (75 ± 15 psig), a secondary bleed air pressure regulator/shutoff valve (120 ± 20 psig), a primary heat exchanger, an ejector, a primary heat exchanger bypass modulating valve, a preconditioned bleed air temperature sensor, and a preconditioned bleed air overtemperature sensor. The regulated bleed air passes through the primary heat exchange and is reduced to near ram air temperatures.

The F-15 was evaluated to determine the most appropriate location for retrofit of the HOS. Guidelines used for this determination were design specifications for an onboard oxygen generating system (OBOGS): pressure of from 25 to 90 psig and temperature of from 40° to 90°F.* These requirements are based on an OBOGS unit supplied by Normalair Garrett Ltd. (NGL) for the Tactical Life Support System (TLSS) and are consistent with the specifications of the Onboard Oxygen Enrichment System (OBOES) of Reference (1).

The best suited location for installation of the HOS in the F-15 aircraft is illustrated in Figure 1-2. This proposed installation for the HOS entails use of the pneumatic supply from the existing anti-fog heat exchanger. In the current system bleed air extracted from upstream of the compressor airflow modulating/shutoff valve enters the hot side of the anti-fog heat exchanger where it is cooled by conditioned air provided by the cabin supply duct. The conditioned air, after absorbing heat, is ducted for anti-fog of transparent surfaces. The bleed air, after rejecting its heat, flows either to the anti-G system and/or to the avionics supply duct where it merges with conditioned air in a mixing muff. A temperature sensor, located in the anti-fog duct downstream of the heat exchanger, controls the anti-fog modulating valve to ensure that air at a minimum temperature of $88 \pm 3^\circ \text{F}$ is available for anti-fog.

Installation of the HOS as illustrated in Figure 1-2 results in extraction of bleed air upstream of the modulating valve. The following impacts on the F-15 ECS are envisioned:

- (1) for extracted bleed air flows less than or equal to the hot side air flow required for proper anti-fog flow temperature control, no impact on anti-fog. Small effect on total avionics air flow (nominally 60 lbm/min.) and will result in only slight reduction of temperature to avionics.

*Performance specification: Deliver specified (rated O_2 concentration) from 0° to 100°F and operate from -65°F to 160°F)

- (2) for extracted bleed air flows greater than that required for maintenance of anti-fog air temperature, the anti-fog modulating valve will go closed and the anti-fog supply temperature will increase.

In either of the above cases, the avionics controller will signal the avionics hot air modulating valve (temperature and flow compensation) to go open to maintain temperature schedule.

Fluctuation in the temperature and pressure of this chosen pneumatic supply does occur. Performance data from Reference (3) reveals pressures ranging from 21 to 75 psig and temperatures ranging from 43 to 97°F. This variation in pneumatic supply state should be considered when evaluating the performance of the HOS. It may be desirable to install an over temperature protection system upstream of the OBOGS. In the event of F-15 ECS failure a probability would exist for high temperature air to flow to the OBOGS. To prevent OBOGS damage a means should be provided to automatically shut off the OBOGS/bleed air supply when the air temperature exceeds some preset threshold value.

Use of Conditioned Bleed Air

One very promising use of the conditioned bleed air is its use for body cooling and visor demist. Recent studies have been conducted to evaluate the feasibility of an air cooled garment compatible with the Tactical Life Support System. One of the biggest problems faced in the development of the air cooled vest concepts was the requirement for clean conditioned air. This requirement becomes even more stringent if the pneumatic supply for the cooling vest is also to be used for visor demist. The most prohibitive factor in retrofit/installation of the air cooled vest designs in current aircraft is the need for a CBR filter to clean the air delivered. Concerns with the CBR filter include service life (filter must be removed and replaced after a believed threat), sensitivity of the adsorbing media to humidity, and pressure drop through the filter. Use of the conditioned bleed air from the HOS offers an excellent solution to the aforementioned problems associated with the CD filter. The clean conditioned air from the

HOS would dispense with the logistics burden connected with CD filter and would increase the ease with which an air cooled vest/demist system could be retrofit into current aircraft. A broader use of the clean conditioned air from the HOS would be for supply of cabin air. This option, if exercised, would eliminate the need for filtration of ECS air and, in the limit, reduce the amount of equipment the crew is required to wear.

Use of Excess Cooling Potential

The excess cooling potential inherent in the HOS design could be used for avionics, advanced sensor, VHSIC, and VLSIC cooling. In the F-15, the avionics supply air temperature is controlled to $82.5 \pm 5^\circ\text{F}$ below 34,500 ft. altitude and to $53 \pm 3^\circ\text{F}$ above 34,500 ft. These temperatures are controlled at the downstream side of the radar heat exchanger. Cabin air flow demand is by design given priority over avionics air flow demand. The actuators of the cabin modulating valve and the avionics cold air modulating valve are pneumatically linked, so that the latter valve throttles to reduce avionics cold air flow when the cabin valve reaches the wide open position and cabin air flow still renders insufficient to meet the required flow rate schedule. This pneumatic coupling could be eliminated and the cold bleed air from the HOS could be used to supplement the cabin air supply or could be used to maintain adequate avionics cooling.

TABLE 1-1: F-15 ECS ITEM DESCRIPTION

<u>Item Number</u>	<u>Item Description</u>
103	Refrigeration Package
105	Bleed Air Preconditioning Package
123	(LH) Primary Heat Exchanger Installation Kits
125	(RH) Primary Heat Exchanger Installation Kits
201	Primary Heat Exchanger
203	Primary HX Modulating Bypass Valve
205	Bleed Air Regulator/Shutoff Valve, Primary
207	Bleed Air Regulator/Shutoff Valve (2nd Stage)
209	LH Primary HX Ejector
210	RH Primary HX Ejector
211	Preconditioned Bleed Air Temperature Sensor
217	Compressor Airflow
219	Cooling Turbine
221	Secondary Heat Exchanger
223	Regenerative Heat Exchanger
227	Primary Water Separator
229	Regenerating HX Modulating Valve
231	Avionics Cold Air Modulating Valve
235	Cabin Inlet Air Modulating/Shutoff Valve
237	Cabin Airflow/Temperature Sensor
239	Cabin Hot Air Modulating Valve
241	Cabin Water Separator
249	Cabin Inlet Air Overtemperature Sensor
253	Avionics Overtemperature Sensor
255	Avionics Air Flow/Temperature Sensor
263	Water Separator Discharge temperature Sensor
271	Cabin Controller
273	Avionics Controller
275	Control Box
277	Anti-Fog Modulating Valve
281	Anti-Fog Heat Exchanger
283	Ram Air Temperature Sensor
285	Preconditioned Bleed Air Overtemp Sensor
289	Avionics Hot Air Modulating Valve
293	Inlet Duct
351	Spray Ejector
369	Avionics Emergency Ram Air Converter Valve
379	Solenoid Shutoff Valve
443	Primary HX Ram Air Check Valve
451	Secondary HX Ejector

*ECS items appear in the schematic of Figure 1-1.

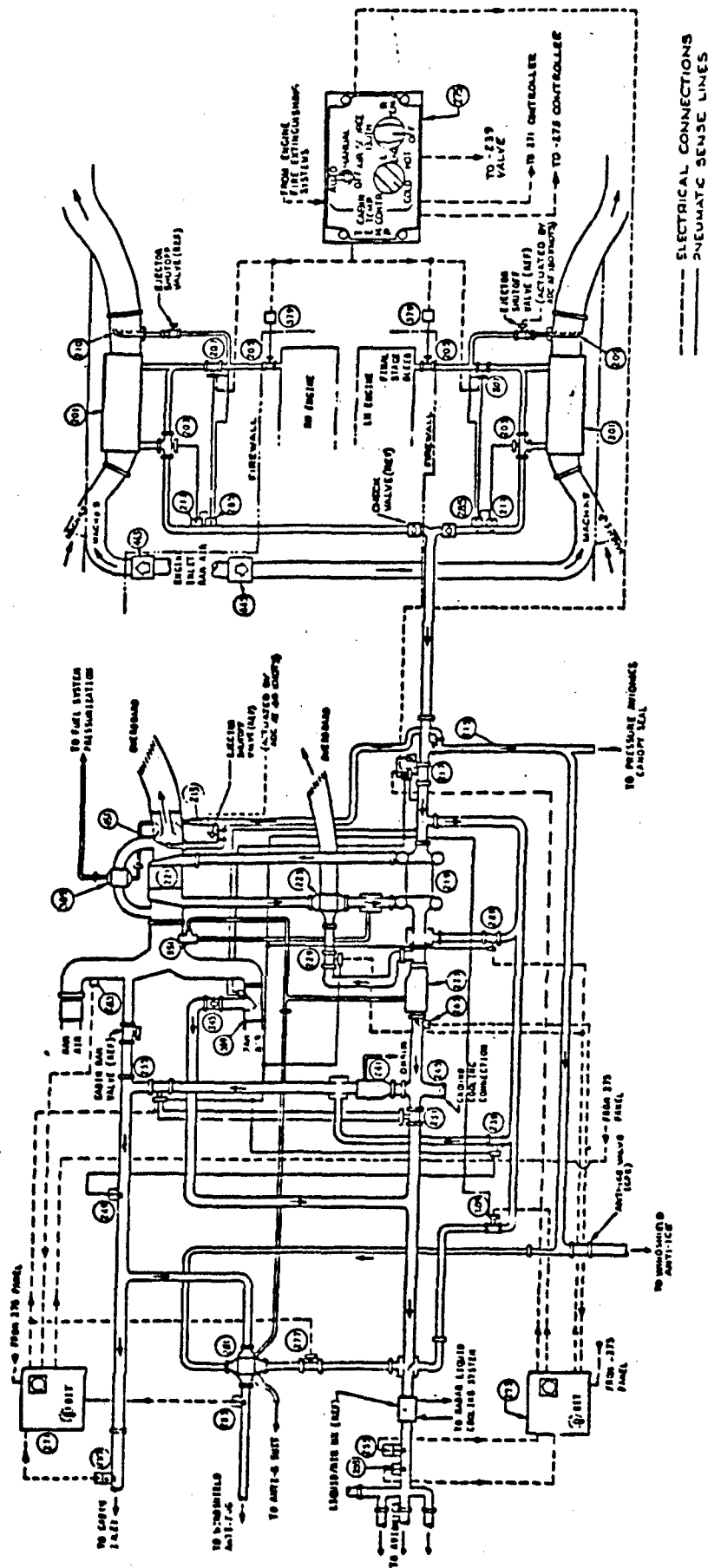
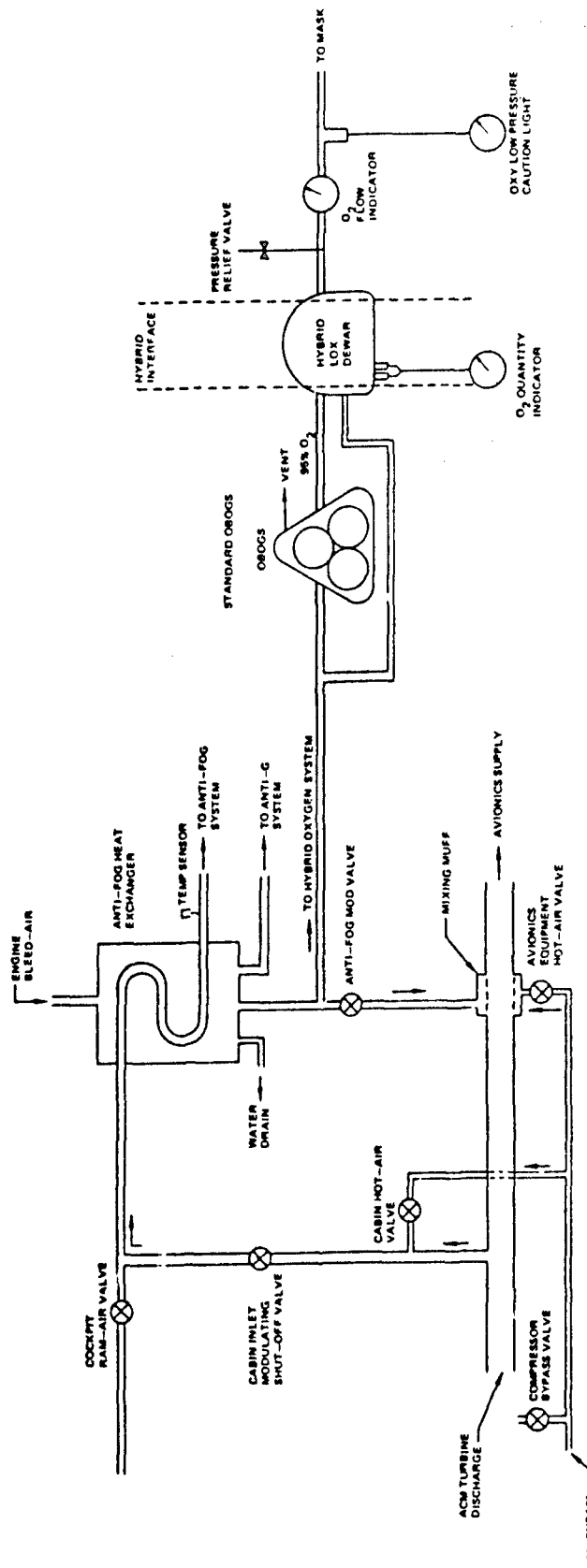


Figure 1-1 F-15 Environmental Control System



CONVENTIONAL LOX SYSTEM
 • PLUMBING
 • INSTRUMENTATION
 • REGULATOR AND MASK

Figure 1-2. F-15 Installation of the Hybrid Oxygen System

**APPENDIX 2: Use of tables to generate
partial pressures of atmospheric
constituents as a function of altitude.**

This appendix presents the methodology and resources for generation of partial pressures of atmospheric constituents as a function of geometric altitude. This investigation was conducted per Tom Maimoni's request. Tables 2-1 and 2-2 contained herein are reproduced from Tables 3 and IV, respectively, of Reference 5.

(1) Assumptions:

Per Neil Olien (Department of Commerce Library - research)
"Atmosphere contains same proportions of gas species up to 100,000 ft. After this altitude ionization and chemical reactions occur which alter atmospheric composition."

(2) Mole fractions of gas species per Table 2-1 extracted from Reference 5 (U. S. Standard Atmosphere, 1976, National Oceanic and Atmospheric Administration, National Aeronautics and Space Administration, United States Air Force, NOAA-S/T 76-1562, October 1976).

(3) Use geometric height, Z (ft.), for extraction of total pressure, P(mb), from Table 2-2

(4) Conversion of mb to psia: 14.504 psia/bar

(5) Sample:

$$P_{\text{total}} (\text{MF}_{\text{gas species}}) = P_{\text{total}} \text{ P}_{\text{gas species}}$$

where: P_{total} = total atmospheric pressure (Table IV)
 $\text{MF}_{\text{gas species}}$ = mole fraction of species (Table 3)
 $P_{\text{gas species}}$ = partial pressure gas species

Example:

@ 1,000 ft.

$$P = 9.7716 \times 10^2 \text{ mb}$$

$$\text{MF}_{\text{N}_2} = 0.78084$$

$$P_{\text{N}_2} = (9.7716 \times 10^2) (0.78084) \frac{(14.504)}{10^3}$$

$$P_{\text{N}_2} = 11.067 \text{ psia}$$

TABLE 2-1

MOLECULAR WEIGHTS AND FRACTIONAL VOLUME
CONCENTRATION OF SEA-LEVEL DRY AIR

<u>Gas Species</u>	<u>Molecular Weight M (kg/kmol)</u>	<u>Fractional Volume F (dimensionless)</u>
N ₂	28.0134	0.78084
O ₂	31.9988	.209476
Ar	39.948	.00934
CO ₂	44.00995	.000314
Ne	20.182	.00001818
He	4.0026	.00000524
Kr	83.80	.00000114
Xe	131.30	.00000087
CH ₄	16.04303	.000002
H ₂	2.01594	.0000005

L-7170-AKT-87-046

TABLE 2-2

GEOMETRIC ALTITUDES

Table 2-2:

Geometric Altitude, English Altitudes L-7170-AKT-87-046

Altitude		Temperature		Pressure		Density	
Z (ft)	H (ft)	T (K)	t (°C)	P (mb)	P/P ₀	ρ (kg/m ³)	ρ/ρ_0
-1000	-1000	290.131	16.981	1.0504 • 3	1.0366 • 0	1.2613 • 0	1.0290 • 0
-900	-900	289.933	16.783	1.0466	1.0329	1.2576	1.0266
-800	-800	289.735	16.585	1.0428	1.0292	1.2539	1.0236
-700	-700	289.537	16.387	1.0391	1.0255	1.2503	1.0206
-600	-600	289.339	16.189	1.0354	1.0218	1.2467	1.0177
-500	-500	289.140	15.990	1.0316	1.0182	1.2430	1.0147
-400	-400	288.942	15.792	1.0279	1.0145	1.2394	1.0118
-300	-300	288.744	15.594	1.0242	1.0108	1.2358	1.0088
-200	-200	288.546	15.396	1.0205	1.0072	1.2322	1.0059
-100	-100	288.348	15.198	1.0169	1.0036	1.2286	1.0029
0	0	288.150	15.000	1.01325 • 3	1.00000 • 0	1.2250 • 0	1.0000 • 0
100	100	287.952	14.802	1.0095	9.9639 - 1	1.2214	9.9708 - 1
200	200	287.754	14.604	1.0059	9.9279	1.2178	9.9410
300	300	287.556	14.406	1.0023	9.8920	1.2143	9.9125
400	400	287.357	14.207	9.9868 • 2	9.8562	1.2107	9.8845
500	500	287.159	14.009	9.9507	9.8206	1.2072	9.8565
600	600	286.961	13.811	9.9147	9.7850	1.2036	9.8286
700	700	286.763	13.613	9.8788	9.7496	1.2001	9.7968
800	800	286.565	13.415	9.8429	9.7142	1.1966	9.7680
900	900	286.367	13.217	9.8072	9.6790	1.1931	9.7393
1000	1000	286.169	13.019	9.7716 • 2	9.6438 - 1	1.1896 • 0	9.7107 - 1
1100	1100	285.971	12.821	9.7361	9.6088	1.1861	9.6821
1200	1200	285.773	12.623	9.7007	9.5739	1.1826	9.6536
1300	1300	285.574	12.424	9.6654	9.5390	1.1791	9.6251
1400	1400	285.376	12.226	9.6303	9.5043	1.1756	9.5967
1500	1500	285.178	12.028	9.5952	9.4697	1.1721	9.5684
1600	1600	284.980	11.830	9.5602	9.4352	1.1687	9.5402
1700	1700	284.782	11.632	9.5253	9.4008	1.1652	9.5120
1800	1800	284.584	11.434	9.4905	9.3664	1.1618	9.4838
1900	1900	284.386	11.236	9.4559	9.3322	1.1583	9.4558
2000	2000	284.188	11.038	9.4213 • 2	9.2981 - 1	1.1549 • 0	9.4278 - 1
2100	2100	283.990	10.840	9.3868	9.2641	1.1515	9.3999
2200	2200	283.792	10.642	9.3525	9.2302	1.1481	9.3720
2300	2300	283.594	10.444	9.3182	9.1964	1.1447	9.3442
2400	2400	283.396	10.246	9.2841	9.1627	1.1413	9.3164
2500	2500	283.197	10.047	9.2500	9.1291	1.1379	9.2887
2600	2600	282.999	9.849	9.2161	9.0955	1.1345	9.2611
2700	2700	282.801	9.651	9.1822	9.0621	1.1311	9.2336
2800	2800	282.603	9.453	9.1485	9.0288	1.1277	9.2061
2900	2900	282.405	9.255	9.1148	8.9956	1.1244	9.1787
3000	3000	282.207	9.057	9.0813 • 2	8.9625 - 1	1.1210 • 0	9.1513 - 1
3100	3100	282.009	8.859	9.0478	8.9295	1.1177	9.1240
3200	3200	281.811	8.661	9.0145	8.8966	1.1144	9.0967
3300	3300	281.613	8.463	8.9812	8.8638	1.1110	9.0696
3400	3400	281.415	8.265	8.9481	8.8311	1.1077	9.0425
3500	3500	281.217	8.067	8.9150	8.7984	1.1044	9.0154
3600	3600	281.019	7.869	8.8821	8.7659	1.1011	8.9884
3700	3700	280.821	7.671	8.8492	8.7335	1.0978	8.9615
3800	3800	280.623	7.473	8.8165	8.7012	1.0945	8.9346
3900	3900	280.425	7.275	8.7838	8.6689	1.0912	8.9078
4000	3999	280.227	7.077	8.7513 • 2	8.6368 - 1	1.0879 • 0	8.8811 - 1
4100	4099	280.029	6.879	8.7188	8.6048	1.0847	8.8544
4200	4199	279.831	6.681	8.6864	8.5728	1.0814	8.8278
4300	4299	279.632	6.482	8.6542	8.5410	1.0781	8.8012
4400	4399	279.434	6.284	8.6220	8.5093	1.0749	8.7747
4500	4499	279.236	6.086	8.5899	8.4776	1.0717	8.7483
4600	4599	279.038	5.888	8.5580	8.4461	1.0684	8.7219
4700	4699	278.840	5.690	8.5261	8.4146	1.0652	8.6956
4800	4799	278.642	5.492	8.4943	8.3832	1.0620	8.6693
4900	4899	278.444	5.294	8.4628	8.3520	1.0588	8.6431
5000	4999	278.246	5.096	8.4311 • 2	8.3208 - 1	1.0556 • 0	8.6170 - 1
5100	5099	278.048	4.898	8.3996	8.2897	1.0524	8.5909
5200	5199	277.850	4.700	8.3682	8.2587	1.0492	8.5649
5300	5299	277.652	4.502	8.3369	8.2279	1.0460	8.5390
5400	5399	277.454	4.304	8.3057	8.1971	1.0429	8.5131
5500	5499	277.256	4.106	8.2746	8.1664	1.0397	8.4873
5600	5599	277.058	3.908	8.2436	8.1358	1.0365	8.4615
5700	5699	276.860	3.710	8.2126	8.1052	1.0334	8.4358
5800	5799	276.662	3.512	8.1818	8.0746	1.0302	8.4102
5900	5899	276.464	3.314	8.1511	8.0445	1.0271	8.3846
6000	5998	276.266	3.116	8.1204 • 2	8.0142 - 1	1.0240 • 0	8.3590 - 1
6100	6098	276.068	2.918	8.0899	7.9841	1.0209	8.3336
6200	6198	275.870	2.720	8.0594	7.9541	1.0178	8.3082
6300	6298	275.672	2.522	8.0291	7.9241	1.0146	8.2828
6400	6398	275.474	2.324	7.9988	7.8942	1.0115	8.2575
6500	6498	275.276	2.126	7.9687	7.8644	1.0085	8.2323
6600	6598	275.078	1.928	7.9386	7.8348	1.0054	8.2071
6700	6698	274.880	1.730	7.9086	7.8052	1.0023	8.1820
6800	6798	274.682	1.532	7.8787	7.7757	9.9923 - 1	8.1570
6900	6898	274.484	1.334	7.8489	7.7463	9.9617	8.1320

Table 2-2: Geopotential Altitude, English Altitudes

Altitude		Temperature		Pressure		Density	
H (ft)	Z (ft)	T (K)	t (°C)	P (mb)	P/P ₀	ρ (kg/m ³)	ρ/ρ_0
7000	7002	274.282	1.132	7.8185 • 2	7.7162 - 1	9.9304 - 1	8.1065 - 1
7100	7102	274.084	.934	7.7889	7.6870	9.8999	8.0816
7200	7202	273.886	.736	7.7593	7.6578	9.8695	8.0567
7300	7303	273.687	.537	7.7299	7.6288	9.8391	8.0320
7400	7403	273.489	.339	7.7005	7.5998	9.8089	8.0072
7500	7503	273.291	.141	7.6712	7.5709	9.7787	7.9826
7600	7603	273.093	-.057	7.6420	7.5421	9.7485	7.9580
7700	7703	272.895	-.255	7.6129	7.5134	9.7185	7.9334
7800	7803	272.697	-.453	7.5839	7.4848	9.6885	7.9090
7900	7903	272.499	-.651	7.5550	7.4562	9.6586	7.8845
8000	8003	272.301	-.849	7.5262 • 2	7.4278 - 1	9.6287 - 1	7.8602 - 1
8100	8103	272.102	-1.048	7.4975	7.3994	9.5989	7.8359
8200	8203	271.904	-1.246	7.4688	7.3711	9.5692	7.8116
8300	8303	271.706	-1.444	7.4403	7.3430	9.5396	7.7874
8400	8403	271.508	-1.642	7.4118	7.3149	9.5100	7.7633
8500	8503	271.310	-1.840	7.3834	7.2868	9.4805	7.7392
8600	8604	271.112	-2.038	7.3551	7.2589	9.4511	7.7152
8700	8704	270.914	-2.236	7.3269	7.2311	9.4217	7.6912
8800	8804	270.716	-2.434	7.2988	7.2033	9.3924	7.6673
8900	8904	270.518	-2.632	7.2707	7.1757	9.3632	7.6434
9000	9004	270.319	-2.831	7.2428 • 2	7.1481 - 1	9.3341 - 1	7.6196 - 1
9100	9104	270.121	-3.029	7.2149	7.1206	9.3050	7.5959
9200	9204	269.923	-3.227	7.1872	7.0932	9.2760	7.5722
9300	9304	269.725	-3.425	7.1595	7.0659	9.2470	7.5486
9400	9404	269.527	-3.623	7.1319	7.0386	9.2182	7.5250
9500	9504	269.329	-3.821	7.1044	7.0115	9.1894	7.5015
9600	9604	269.131	-4.019	7.0770	6.9844	9.1606	7.4781
9700	9705	268.933	-4.217	7.0496	6.9574	9.1320	7.4547
9800	9805	268.734	-4.416	7.0224	6.9305	9.1034	7.4313
9900	9905	268.536	-4.614	6.9952	6.9037	9.0748	7.4080
10000	10005	268.338	-4.812	6.9681 • 2	6.8770 - 1	9.0464 - 1	7.3848 - 1
10100	10105	268.140	-5.010	6.9411	6.8504	9.0180	7.3616
10200	10205	267.942	-5.208	6.9142	6.8238	8.9897	7.3385
10300	10305	267.744	-5.406	6.8874	6.7973	8.9614	7.3154
10400	10405	267.546	-5.604	6.8606	6.7709	8.9332	7.2924
10500	10505	267.348	-5.802	6.8340	6.7446	8.9051	7.2695
10600	10605	267.149	-6.001	6.8074	6.7184	8.8770	7.2466
10700	10705	266.951	-6.199	6.7809	6.6922	8.8491	7.2237
10800	10806	266.753	-6.397	6.7545	6.6662	8.8211	7.2009
10900	10906	266.555	-6.595	6.7282	6.6402	8.7933	7.1782
11000	11006	266.357	-6.793	6.7019 • 2	6.6143 - 1	8.7655 - 1	7.1545 - 1
11100	11106	266.159	-6.991	6.6758	6.5885	8.7378	7.1329
11200	11206	265.961	-7.189	6.6497	6.5627	8.7102	7.1103
11300	11306	265.763	-7.387	6.6237	6.5371	8.6826	7.0878
11400	11406	265.565	-7.585	6.5978	6.5115	8.6551	7.0654
11500	11506	265.366	-7.784	6.5720	6.4860	8.6276	7.0429
11600	11606	265.168	-7.982	6.5462	6.4606	8.6002	7.0206
11700	11707	264.970	-8.180	6.5205	6.4353	8.5729	6.9983
11800	11807	264.772	-8.378	6.4950	6.4100	8.5457	6.9761
11900	11907	264.574	-8.576	6.4695	6.3849	8.5185	6.9539
12000	12007	264.376	-8.774	6.4440 • 2	6.3598 - 1	8.4914 - 1	6.9317 - 1
12100	12107	264.178	-8.972	6.4187	6.3348	8.4643	6.9097
12200	12207	263.980	-9.170	6.3934	6.3098	8.4373	6.8876
12300	12307	263.781	-9.369	6.3683	6.2850	8.4104	6.8657
12400	12407	263.583	-9.567	6.3432	6.2602	8.3836	6.8437
12500	12507	263.385	-9.765	6.3181	6.2355	8.3568	6.8219
12600	12608	263.187	-9.963	6.2932	6.2109	8.3301	6.8001
12700	12708	262.989	-10.161	6.2683	6.1864	8.3034	6.7783
12800	12808	262.791	-10.359	6.2435	6.1619	8.2768	6.7566
12900	12908	262.593	-10.557	6.2189	6.1375	8.2503	6.7349
13000	13008	262.395	-10.755	6.1942 • 2	6.1132 - 1	8.2238 - 1	6.7133 - 1
13100	13108	262.196	-10.954	6.1697	6.0890	8.1975	6.6918
13200	13208	261.998	-11.152	6.1452	6.0649	8.1711	6.6703
13300	13308	261.800	-11.350	6.1209	6.0408	8.1449	6.6489
13400	13409	261.602	-11.548	6.0965	6.0168	8.1187	6.6275
13500	13509	261.404	-11.746	6.0723	5.9929	8.0925	6.6061
13600	13609	261.206	-11.944	6.0482	5.9691	8.0665	6.5849
13700	13709	261.008	-12.142	6.0241	5.9453	8.0404	6.5636
13800	13809	260.810	-12.340	6.0001	5.9216	8.0145	6.5425
13900	13909	260.612	-12.538	5.9762	5.8980	7.9886	6.5213
14000	14009	260.413	-12.737	5.9523 • 2	5.8745 - 1	7.9628 - 1	6.5003 - 1
14100	14110	260.215	-12.935	5.9286	5.8511	7.9371	6.4792
14200	14210	260.017	-13.133	5.9049	5.8277	7.9114	6.4583
14300	14310	259.819	-13.331	5.8813	5.8044	7.8858	6.4373
14400	14410	259.621	-13.529	5.8578	5.7811	7.8602	6.4165
14500	14510	259.423	-13.727	5.8343	5.7580	7.8347	6.3957
14600	14610	259.225	-13.925	5.8109	5.7349	7.8093	6.3749
14700	14710	259.027	-14.123	5.7876	5.7119	7.7839	6.3542
14800	14811	258.828	-14.322	5.7644	5.6890	7.7586	6.3335
14900	14911	258.630	-14.520	5.7412	5.6661	7.7333	6.3129

Table 2-2: Geometric Altitude, English Altitudes

Altitude		Temperature		Pressure		Density					
Z (ft)	H (ft)	T (K)	t (°C)	P (mb)	P/P ₀	ρ (kg/m ³)	ρ/ρ ₀				
7000	6998	274.286	1.136	7.8192	• 2	7.7169	- 1	9.9311	- 1	8.1070	- 1
7100	7098	274.088	.938	7.7896		7.6877		9.9007		8.0822	
7200	7198	273.890	.740	7.7600		7.6586		9.8703		8.0573	
7300	7297	273.692	.542	7.7306		7.6295		9.8399		8.0326	
7400	7397	273.494	.344	7.7013		7.6005		9.8097		8.0079	
7500	7497	273.296	.146	7.6720		7.5717		9.7795		7.9832	
7600	7597	273.098	-.052	7.6428		7.5429		9.7494		7.9587	
7700	7697	272.900	-.250	7.6138		7.5142		9.7193		7.9341	
7800	7797	272.702	-.448	7.5848		7.4856		9.6894		7.9097	
7900	7897	272.504	-.646	7.5559		7.4571		9.6595		7.8853	
8000	7997	272.306	-.844	7.5271	• 2	7.4286	- 1	9.6296	- 1	7.8609	- 1
8100	8097	272.108	-1.042	7.4984		7.4003		9.5999		7.8366	
8200	8197	271.910	-1.240	7.4697		7.3720		9.5702		7.8124	
8300	8297	271.712	-1.438	7.4412		7.3439		9.5406		7.7882	
8400	8397	271.514	-1.636	7.4127		7.3156		9.5110		7.7641	
8500	8497	271.317	-1.833	7.3844		7.2878		9.4815		7.7400	
8600	8596	271.119	-2.031	7.3561		7.2599		9.4521		7.7160	
8700	8696	270.921	-2.229	7.3279		7.2321		9.4228		7.6921	
8800	8796	270.723	-2.427	7.2998		7.2044		9.3935		7.6682	
8900	8896	270.525	-2.625	7.2718		7.1767		9.3643		7.6443	
9000	8996	270.327	-2.823	7.2439	• 2	7.1492	- 1	9.3352	- 1	7.6206	- 1
9100	9096	270.129	-3.021	7.2160		7.1217		9.3061		7.5968	
9200	9196	269.931	-3.219	7.1883		7.0943		9.2772		7.5732	
9300	9296	269.733	-3.417	7.1606		7.0670		9.2482		7.5496	
9400	9396	269.535	-3.615	7.1331		7.0398		9.2194		7.5260	
9500	9496	269.337	-3.813	7.1056		7.0127		9.1906		7.5025	
9600	9596	269.139	-4.011	7.0782		6.9856		9.1619		7.4791	
9700	9695	268.941	-4.209	7.0509		6.9586		9.1333		7.4557	
9800	9795	268.743	-4.407	7.0236		6.9318		9.1047		7.4324	
9900	9895	268.545	-4.605	6.9965		6.9050		9.0762		7.4091	
10000	9995	268.347	-4.803	6.9694	• 2	6.8783	- 1	9.0477	- 1	7.3859	- 1
10100	10095	268.149	-5.001	6.9424		6.8516		9.0194		7.3627	
10200	10195	267.952	-5.198	6.9155		6.8251		8.9911		7.3396	
10300	10295	267.754	-5.396	6.8887		6.7987		8.9628		7.3166	
10400	10395	267.556	-5.594	6.8620		6.7723		8.9347		7.2936	
10500	10495	267.358	-5.792	6.8354		6.7460		8.9066		7.2707	
10600	10595	267.160	-5.990	6.8088		6.7198		8.8786		7.2478	
10700	10695	266.962	-6.188	6.7824		6.6937		8.8506		7.2250	
10800	10794	266.764	-6.386	6.7560		6.6676		8.8227		7.2022	
10900	10894	266.566	-6.584	6.7297		6.6417		8.7949		7.1795	
11000	10994	266.368	-6.782	6.7034	• 2	6.6158	- 1	8.7671	- 1	7.1568	- 1
11100	11094	266.170	-6.980	6.6773		6.5900		8.7394		7.1342	
11200	11194	265.972	-7.178	6.6513		6.5643		8.7118		7.1117	
11300	11294	265.774	-7.376	6.6253		6.5386		8.6843		7.0892	
11400	11394	265.577	-7.573	6.5994		6.5131		8.6568		7.0667	
11500	11494	265.379	-7.771	6.5736		6.4876		8.6294		7.0444	
11600	11594	265.181	-7.969	6.5479		6.4622		8.6020		7.0220	
11700	11693	264.983	-8.167	6.5222		6.4369		8.5747		6.9998	
11800	11793	264.785	-8.365	6.4967		6.4117		8.5475		6.9775	
11900	11893	264.587	-8.563	6.4712		6.3866		8.5203		6.9554	
12000	11993	264.389	-8.761	6.4458	• 2	6.3615	- 1	8.4933	- 1	6.9333	- 1
12100	12093	264.191	-8.959	6.4205		6.3365		8.4662		6.9112	
12200	12193	263.993	-9.157	6.3952		6.3116		8.4393		6.8892	
12300	12293	263.795	-9.355	6.3701		6.2868		8.4124		6.8672	
12400	12393	263.598	-9.552	6.3450		6.2620		8.3856		6.8453	
12500	12493	263.400	-9.750	6.3200		6.2374		8.3588		6.8235	
12600	12592	263.202	-9.948	6.2951		6.2128		8.3321		6.8017	
12700	12692	263.004	-10.146	6.2703		6.1883		8.3055		6.7800	
12800	12792	262.806	-10.344	6.2455		6.1638		8.2789		6.7583	
12900	12892	262.608	-10.542	6.2208		6.1395		8.2524		6.7367	
13000	12992	262.410	-10.740	6.1962	• 2	6.1152	- 1	8.2260	- 1	6.7151	- 1
13100	13092	262.212	-10.938	6.1717		6.0910		8.1996		6.6936	
13200	13192	262.015	-11.135	6.1473		6.0669		8.1733		6.6721	
13300	13292	261.817	-11.333	6.1229		6.0428		8.1471		6.6507	
13400	13391	261.619	-11.531	6.0986		6.0189		8.1209		6.6293	
13500	13491	261.421	-11.729	6.0744		5.9950		8.0948		6.6080	
13600	13591	261.223	-11.927	6.0503		5.9712		8.0688		6.5867	
13700	13691	261.025	-12.125	6.0263		5.9474		8.0428		6.5655	
13800	13791	260.827	-12.323	6.0023		5.9238		8.0169		6.5444	
13900	13891	260.630	-12.520	5.9784		5.9002		7.9910		6.5233	
14000	13991	260.432	-12.718	5.9546	• 2	5.8767	- 1	7.9652	- 1	6.5022	- 1
14100	14090	260.234	-12.916	5.9308		5.8533		7.9395		6.4812	
14200	14190	260.036	-13.114	5.9072		5.8299		7.9139		6.4603	
14300	14290	259.838	-13.312	5.8836		5.8066		7.8883		6.4394	
14400	14390	259.640	-13.510	5.8601		5.7834		7.8627		6.4186	
14500	14490	259.442	-13.708	5.8367		5.7603		7.8373		6.3978	
14600	14590	259.245	-13.905	5.8133		5.7373		7.8119		6.3770	
14700	14690	259.047	-14.103	5.7900		5.7143		7.7865		6.3563	
14800	14790	258.849	-14.301	5.7668		5.6914		7.7612		6.3357	
14900	14889	258.651	-14.499	5.7437		5.6686		7.7360		6.3151	

Table 2-2: Geopotential Altitude, English Altitudes

Altitude		Temperature		Pressure		Density					
H (ft)	Z (ft)	T (K)	t (°C)	P (mb)	P/P ₀	ρ/kg/m ³	ρ/ρ ₀				
15000	15011	258.432	-14.718	5.7182	• 2	5.6434	- 1	7.7082	- 1	6.2924	- 1
15100	15111	258.234	-14.916	5.6951		5.6207		7.6830		6.2719	
15200	15211	258.036	-15.114	5.6722		5.5980		7.6580		6.2514	
15300	15311	257.838	-15.312	5.6494		5.5755		7.6330		6.2310	
15400	15411	257.640	-15.510	5.6266		5.5530		7.6081		6.2107	
15500	15512	257.442	-15.708	5.6039		5.5306		7.5832		6.1904	
15600	15612	257.243	-15.907	5.5813		5.5083		7.5584		6.1701	
15700	15712	257.045	-16.105	5.5587		5.4860		7.5337		6.1499	
15800	15812	256.847	-16.303	5.5362		5.4638		7.5090		6.1298	
15900	15912	256.649	-16.501	5.5138		5.4417		7.4844		6.1097	
16000	16012	256.451	-16.699	5.4915	• 2	5.4197	- 1	7.4598	- 1	6.0896	- 1
16100	16112	256.253	-16.897	5.4692		5.3977		7.4353		6.0696	
16200	16213	256.055	-17.095	5.4470		5.3758		7.4109		6.0497	
16300	16313	255.857	-17.293	5.4249		5.3540		7.3865		6.0298	
16400	16413	255.659	-17.491	5.4029		5.3322		7.3622		6.0099	
16500	16513	255.461	-17.689	5.3809		5.3105		7.3379		5.9901	
16600	16613	255.262	-17.888	5.3590		5.2889		7.3137		5.9704	
16700	16713	255.064	-18.086	5.3372		5.2674		7.2896		5.9507	
16800	16814	254.866	-18.284	5.3155		5.2459		7.2655		5.9311	
16900	16914	254.668	-18.482	5.2937		5.2245		7.2415		5.9115	
17000	17014	254.470	-18.680	5.2721	• 2	5.2032	- 1	7.2176	- 1	5.8919	- 1
17100	17114	254.272	-18.878	5.2506		5.1819		7.1937		5.8724	
17200	17214	254.074	-19.076	5.2291		5.1607		7.1699		5.8530	
17300	17314	253.875	-19.275	5.2077		5.1396		7.1461		5.8336	
17400	17415	253.677	-19.473	5.1864		5.1186		7.1224		5.8142	
17500	17515	253.479	-19.671	5.1652		5.0976		7.0988		5.7949	
17600	17615	253.281	-19.869	5.1440		5.0767		7.0752		5.7757	
17700	17715	253.083	-20.067	5.1229		5.0559		7.0517		5.7565	
17800	17815	252.885	-20.265	5.1018		5.0351		7.0282		5.7373	
17900	17915	252.687	-20.463	5.0808		5.0144		7.0048		5.7182	
18000	18016	252.489	-20.661	5.0594	• 2	4.9938	- 1	6.9815	- 1	5.6941	- 1
18100	18116	252.290	-20.860	5.0391		4.9732		6.9582		5.6748	
18200	18216	252.092	-21.058	5.0183		4.9527		6.9349		5.6552	
18300	18316	251.894	-21.256	4.9976		4.9323		6.9118		5.6363	
18400	18416	251.696	-21.454	4.9770		4.9119		6.8887		5.6173	
18500	18516	251.498	-21.652	4.9565		4.8916		6.8656		5.5984	
18600	18617	251.300	-21.850	4.9360		4.8714		6.8426		5.5795	
18700	18717	251.102	-22.048	4.9156		4.8513		6.8197		5.5607	
18800	18817	250.904	-22.246	4.8952		4.8312		6.7968		5.5419	
18900	18917	250.706	-22.444	4.8749		4.8112		6.7740		5.5230	
19000	19017	250.507	-22.643	4.8547	• 2	4.7912	- 1	6.7513	- 1	5.5042	- 1
19100	19118	250.309	-22.841	4.8346		4.7713		6.7286		5.4857	
19200	19218	250.111	-23.039	4.8145		4.7515		6.7059		5.4672	
19300	19318	249.913	-23.237	4.7945		4.7318		6.6834		5.4488	
19400	19418	249.715	-23.435	4.7745		4.7121		6.6608		5.4304	
19500	19518	249.517	-23.633	4.7547		4.6925		6.6384		5.4121	
19600	19618	249.319	-23.831	4.7349		4.6729		6.6160		5.3938	
19700	19719	249.121	-24.029	4.7151		4.6534		6.5936		5.3756	
19800	19819	248.922	-24.228	4.6954		4.6340		6.5713		5.3574	
19900	19919	248.724	-24.426	4.6758		4.6147		6.5491		5.3392	
20000	20019	248.526	-24.624	4.6563	• 2	4.5954	- 1	6.5269	- 1	5.3211	- 1
20100	20119	248.328	-24.822	4.6368		4.5762		6.5048		5.3028	
20200	20220	248.130	-25.020	4.6174		4.5570		6.4828		5.2847	
20300	20320	247.932	-25.218	4.5980		4.5379		6.4608		5.2667	
20400	20420	247.734	-25.416	4.5788		4.5189		6.4388		5.2488	
20500	20520	247.536	-25.614	4.5596		4.4999		6.4169		5.2309	
20600	20620	247.337	-25.813	4.5404		4.4810		6.3951		5.2130	
20700	20721	247.139	-26.011	4.5213		4.4622		6.3733		5.1951	
20800	20821	246.941	-26.209	4.5023		4.4434		6.3516		5.1772	
20900	20921	246.743	-26.407	4.4834		4.4247		6.3300		5.1594	
21000	21021	246.545	-26.605	4.4645	• 2	4.4061	- 1	6.3084	- 1	5.1415	- 1
21100	21121	246.347	-26.803	4.4456		4.3875		6.2868		5.1237	
21200	21222	246.149	-27.001	4.4269		4.3690		6.2653		5.1060	
21300	21322	245.951	-27.199	4.4082		4.3505		6.2439		5.0883	
21400	21422	245.753	-27.397	4.3896		4.3322		6.2225		5.0706	
21500	21522	245.555	-27.596	4.3710		4.3138		6.2012		5.0529	
21600	21622	245.356	-27.794	4.3525		4.2956		6.1799		5.0352	
21700	21723	245.158	-27.992	4.3340		4.2774		6.1587		5.0175	
21800	21823	244.960	-28.190	4.3157		4.2592		6.1376		5.0000	
21900	21923	244.762	-28.388	4.2974		4.2412		6.1165		4.9825	
22000	22023	244.564	-28.586	4.2791	• 2	4.2231	- 1	6.0954	- 1	4.9650	- 1
22100	22123	244.366	-28.784	4.2609		4.2052		6.0744		4.9475	
22200	22224	244.168	-28.982	4.2428		4.1873		6.0535		4.9300	
22300	22324	243.969	-29.181	4.2247		4.1695		6.0326		4.9126	
22400	22424	243.771	-29.379	4.2067		4.1517		6.0118		4.8951	
22500	22524	243.573	-29.577	4.1888		4.1340		5.9910		4.8776	
22600	22625	243.375	-29.775	4.1709		4.1164		5.9703		4.8601	
22700	22725	243.177	-29.973	4.1531		4.0988		5.9497		4.8426	
22800	22825	242.979	-30.171	4.1353		4.0813		5.9291		4.8251	
22900	22925	242.781	-30.369	4.1177		4.0638		5.9085		4.8076	

Table 2-2: Geopotential Altitude, English Altitudes

Altitude		Temperature		Pressure		Density					
H (ft)	Z (ft)	T (K)	t (°C)	P (mb)	P/P ₀	ρ (kg/m ³)	ρ/ρ ₀				
23000	23025	242.583	-30.567	4.1000	• 2	4.0464	- 1	5.8880	- 1	4.8066	- 1
23100	23126	242.384	-30.766	4.0825		4.0291		5.8676		4.7899	
23200	23226	242.186	-30.964	4.0649		4.0118		5.8472		4.7732	
23300	23326	241.988	-31.162	4.0475		3.9946		5.8269		4.7566	
23400	23426	241.790	-31.360	4.0301		3.9774		5.8066		4.7401	
23500	23527	241.592	-31.558	4.0128		3.9603		5.7864		4.7236	
23600	23627	241.394	-31.756	3.9955		3.9433		5.7662		4.7071	
23700	23727	241.196	-31.954	3.9783		3.9263		5.7461		4.6907	
23800	23827	240.998	-32.152	3.9612		3.9094		5.7260		4.6743	
23900	23927	240.800	-32.350	3.9441		3.8925		5.7060		4.6580	
24000	24028	240.601	-32.549	3.9271	• 2	3.8757	- 1	5.6861	- 1	4.6417	- 1
24100	24128	240.403	-32.747	3.9101		3.8590		5.6662		4.6254	
24200	24228	240.205	-32.945	3.8932		3.8423		5.6463		4.6092	
24300	24328	240.007	-33.143	3.8763		3.8256		5.6265		4.5931	
24400	24429	239.809	-33.341	3.8595		3.8091		5.6068		4.5770	
24500	24529	239.611	-33.539	3.8428		3.7926		5.5871		4.5609	
24600	24629	239.413	-33.737	3.8261		3.7761		5.5675		4.5449	
24700	24729	239.215	-33.935	3.8095		3.7597		5.5479		4.5289	
24800	24830	239.016	-34.134	3.7930		3.7434		5.5284		4.5129	
24900	24930	238.818	-34.332	3.7765		3.7271		5.5089		4.4971	
25000	25030	238.620	-34.530	3.7600	• 2	3.7109	- 1	5.4895	- 1	4.4812	- 1
25100	25130	238.422	-34.728	3.7437		3.6947		5.4701		4.4654	
25200	25230	238.224	-34.926	3.7273		3.6786		5.4508		4.4496	
25300	25331	238.026	-35.124	3.7111		3.6625		5.4315		4.4339	
25400	25431	237.828	-35.322	3.6949		3.6466		5.4123		4.4182	
25500	25531	237.630	-35.520	3.6787		3.6306		5.3931		4.4026	
25600	25631	237.431	-35.719	3.6626		3.6147		5.3740		4.3870	
25700	25732	237.233	-35.917	3.6466		3.5989		5.3550		4.3714	
25800	25832	237.035	-36.115	3.6306		3.5831		5.3360		4.3559	
25900	25932	236.837	-36.313	3.6147		3.5674		5.3170		4.3404	
26000	26032	236.639	-36.511	3.5988	• 2	3.5518	- 1	5.2981	- 1	4.3250	- 1
26100	26133	236.441	-36.709	3.5830		3.5362		5.2792		4.3096	
26200	26233	236.243	-36.907	3.5673		3.5206		5.2604		4.2942	
26300	26333	236.045	-37.105	3.5516		3.5051		5.2417		4.2789	
26400	26433	235.847	-37.303	3.5359		3.4897		5.2230		4.2637	
26500	26534	235.648	-37.502	3.5204		3.4743		5.2043		4.2484	
26600	26634	235.450	-37.700	3.5048		3.4590		5.1858		4.2333	
26700	26734	235.252	-37.898	3.4894		3.4437		5.1672		4.2181	
26800	26834	235.054	-38.096	3.4739		3.4285		5.1487		4.2030	
26900	26935	234.856	-38.294	3.4586		3.4133		5.1303		4.1880	
27000	27035	234.658	-38.492	3.4433	• 2	3.3982	- 1	5.1119	- 1	4.1730	- 1
27100	27135	234.460	-38.690	3.4280		3.3832		5.0935		4.1580	
27200	27236	234.262	-38.888	3.4128		3.3682		5.0752		4.1431	
27300	27336	234.063	-39.087	3.3977		3.3532		5.0570		4.1282	
27400	27436	233.865	-39.285	3.3826		3.3384		5.0388		4.1133	
27500	27536	233.667	-39.483	3.3676		3.3235		5.0207		4.0985	
27600	27637	233.469	-39.681	3.3526		3.3087		5.0026		4.0837	
27700	27737	233.271	-39.879	3.3376		3.2940		4.9845		4.0690	
27800	27837	233.073	-40.077	3.3228		3.2793		4.9665		4.0543	
27900	27937	232.875	-40.275	3.3080		3.2647		4.9486		4.0397	
28000	28038	232.677	-40.473	3.2932	• 2	3.2501	- 1	4.9307	- 1	4.0251	- 1
28100	28138	232.478	-40.672	3.2785		3.2356		4.9129		4.0105	
28200	28238	232.280	-40.870	3.2638		3.2211		4.8951		3.9960	
28300	28338	232.082	-41.068	3.2492		3.2067		4.8773		3.9815	
28400	28439	231.884	-41.266	3.2347		3.1924		4.8596		3.9670	
28500	28539	231.686	-41.464	3.2202		3.1781		4.8420		3.9526	
28600	28639	231.488	-41.662	3.2057		3.1638		4.8244		3.9383	
28700	28740	231.290	-41.860	3.1913		3.1496		4.8068		3.9239	
28800	28840	231.092	-42.058	3.1770		3.1354		4.7893		3.9097	
28900	28940	230.894	-42.256	3.1627		3.1213		4.7719		3.8954	
29000	29040	230.695	-42.455	3.1485	• 2	3.1073	- 1	4.7545	- 1	3.8812	- 1
29100	29141	230.497	-42.653	3.1343		3.0933		4.7371		3.8670	
29200	29241	230.299	-42.851	3.1201		3.0793		4.7198		3.8529	
29300	29341	230.101	-43.049	3.1061		3.0654		4.7026		3.8388	
29400	29441	229.903	-43.247	3.0920		3.0516		4.6854		3.8248	
29500	29542	229.705	-43.445	3.0780		3.0378		4.6682		3.8108	
29600	29642	229.507	-43.643	3.0641		3.0240		4.6511		3.7968	
29700	29742	229.309	-43.841	3.0502		3.0103		4.6340		3.7829	
29800	29843	229.110	-44.040	3.0364		2.9967		4.6170		3.7690	
29900	29943	228.912	-44.238	3.0226		2.9831		4.6000		3.7551	
30000	30043	228.714	-44.436	3.0089	• 2	2.9696	- 1	4.5831	- 1	3.7413	- 1
30100	30144	228.516	-44.634	2.9952		2.9561		4.5663		3.7276	
30200	30244	228.318	-44.832	2.9816		2.9426		4.5494		3.7138	
30300	30344	228.120	-45.030	2.9680		2.9292		4.5327		3.7001	
30400	30444	227.922	-45.228	2.9545		2.9159		4.5159		3.6865	
30500	30545	227.724	-45.426	2.9410		2.9026		4.4992		3.6728	
30600	30645	227.525	-45.625	2.9274		2.8893		4.4826		3.6593	
30700	30745	227.327	-45.823	2.9142		2.8761		4.4660		3.6457	
30800	30846	227.129	-46.021	2.9009		2.8630		4.4495		3.6322	
30900	30946	226.931	-46.219	2.8876		2.8499		4.4330		3.6188	

Table 2-2 Geopotential Altitude, English Altitudes

Altitude		Temperature		Pressure		Density	
H (ft)	Z (ft)	T (K)	t (°C)	P (mb)	P/P ₀	ρ (kg/m ³)	ρ/ρ_0
31000	31046	226.733	-46.417	2.8744	• 2	2.8368	- 1
31100	31146	226.535	-46.615	2.8612		2.8238	- 1
31200	31247	226.337	-46.813	2.8481		2.8109	- 1
31300	31347	226.139	-47.011	2.8350		2.7980	- 1
31400	31447	225.941	-47.209	2.8220		2.7851	- 1
31500	31548	225.742	-47.408	2.8090		2.7723	- 1
31600	31648	225.544	-47.606	2.7961		2.7595	- 1
31700	31748	225.346	-47.804	2.7832		2.7468	- 1
31800	31849	225.148	-48.002	2.7704		2.7341	- 1
31900	31949	224.950	-48.200	2.7576		2.7215	- 1
32000	32049	224.752	-48.398	2.7448	• 2	2.7089	- 1
32100	32149	224.554	-48.596	2.7321		2.6964	- 1
32200	32250	224.356	-48.794	2.7195		2.6839	- 1
32300	32350	224.157	-48.993	2.7069		2.6715	- 1
32400	32450	223.959	-49.191	2.6944		2.6591	- 1
32500	32551	223.761	-49.389	2.6818		2.6468	- 1
32600	32651	223.563	-49.587	2.6694		2.6345	- 1
32700	32751	223.365	-49.785	2.6570		2.6222	- 1
32800	32852	223.167	-49.983	2.6446		2.6100	- 1
32900	32952	222.969	-50.181	2.6323		2.5979	- 1
33000	33052	222.771	-50.379	2.6200	• 2	2.5858	- 1
33100	33153	222.572	-50.578	2.6078		2.5737	- 1
33200	33253	222.374	-50.776	2.5956		2.5617	- 1
33300	33353	222.176	-50.974	2.5835		2.5497	- 1
33400	33454	221.978	-51.172	2.5714		2.5378	- 1
33500	33554	221.780	-51.370	2.5594		2.5259	- 1
33600	33654	221.582	-51.568	2.5474		2.5141	- 1
33700	33755	221.384	-51.766	2.5354		2.5023	- 1
33800	33855	221.186	-51.964	2.5235		2.4905	- 1
33900	33955	220.988	-52.162	2.5117		2.4788	- 1
34000	34056	220.789	-52.361	2.4999	• 2	2.4672	- 1
34100	34156	220.591	-52.559	2.4881		2.4555	- 1
34200	34256	220.393	-52.757	2.4764		2.4440	- 1
34300	34357	220.195	-52.955	2.4647		2.4325	- 1
34400	34457	219.997	-53.153	2.4531		2.4210	- 1
34500	34557	219.799	-53.351	2.4415		2.4095	- 1
34600	34658	219.601	-53.549	2.4299		2.3981	- 1
34700	34758	219.403	-53.747	2.4184		2.3868	- 1
34800	34858	219.204	-53.946	2.4070		2.3755	- 1
34900	34959	219.006	-54.144	2.3956		2.3642	- 1
35000	35059	218.808	-54.342	2.3842	• 2	2.3530	- 1
35200	35260	218.412	-54.738	2.3616		2.3307	- 1
35400	35460	218.016	-55.134	2.3391		2.3086	- 1
35600	35661	217.619	-55.531	2.3169		2.2866	- 1
35800	35862	217.223	-55.927	2.2948		2.2646	- 1
36000	36062	216.827	-56.323	2.2729		2.2432	- 1
36200	36263	216.430	-56.720	2.2511		2.2217	- 1
36400	36464	216.034	-57.116	2.2296		2.2004	- 1
36600	36664	215.638	-57.512	2.2083		2.1794	- 1
36800	36865	215.242	-57.908	2.1872		2.1585	- 1
37000	37066	214.846	-58.304	2.1662	• 2	2.1379	- 1
37200	37266	214.450	-58.700	2.1455		2.1174	- 1
37400	37467	214.054	-59.096	2.1250		2.0972	- 1
37600	37668	213.658	-59.492	2.1046		2.0771	- 1
37800	37869	213.262	-59.888	2.0845		2.0573	- 1
38000	38069	212.866	-60.284	2.0646		2.0376	- 1
38200	38270	212.470	-60.680	2.0448		2.0181	- 1
38400	38471	212.074	-61.076	2.0253		1.9988	- 1
38600	38672	211.678	-61.472	2.0059		1.9796	- 1
38800	38872	211.282	-61.868	1.9867		1.9607	- 1
39000	39073	210.886	-62.264	1.9677	• 2	1.9420	- 1
39200	39274	210.490	-62.660	1.9489		1.9234	- 1
39400	39475	210.094	-63.056	1.9302		1.9050	- 1
39600	39675	209.698	-63.452	1.9117		1.8867	- 1
39800	39876	209.302	-63.848	1.8935		1.8687	- 1
40000	40077	208.906	-64.244	1.8753		1.8508	- 1
40200	40278	208.510	-64.640	1.8574		1.8331	- 1
40400	40478	208.114	-65.036	1.8396		1.8156	- 1
40600	40679	207.718	-65.432	1.8220		1.7982	- 1
40800	40880	207.322	-65.828	1.8046		1.7810	- 1
41000	41081	206.926	-66.224	1.7873	• 2	1.7640	- 1
41200	41282	206.530	-66.620	1.7702		1.7471	- 1
41400	41482	206.134	-67.016	1.7533		1.7304	- 1
41600	41683	205.738	-67.414	1.7365		1.7138	- 1
41800	41884	205.342	-67.810	1.7199		1.6974	- 1
42000	42085	204.946	-68.206	1.7035		1.6812	- 1
42200	42286	204.550	-68.602	1.6872		1.6651	- 1
42400	42486	204.154	-68.998	1.6710		1.6492	- 1
42600	42687	203.758	-69.394	1.6550		1.6334	- 1
42800	42888	203.362	-69.790	1.6392		1.6178	- 1

Table 2-2:

Geometric Altitude, English Altitudes

L-7170-AKT-87-046

Altitude		Temperature		Pressure		Density					
Z (ft)	H (ft)	T (K)	t (°C)	P (mb)	P/P ₀	ρ (kg/m ³)	ρ/ρ ₀				
31000	30954	226.824	-46.326	2.8805	• 2	2.8426	- 1	4.4241	• 1	3.6115	- 1
31100	31054	226.626	-46.524	2.8673		2.8296		4.4077		3.5981	
31200	31153	226.429	-46.721	2.8542		2.8169		4.3914		3.5848	
31300	31253	226.231	-46.919	2.8412		2.8040		4.3751		3.5715	
31400	31353	226.034	-47.116	2.8281		2.7912		4.3589		3.5583	
31500	31452	225.836	-47.314	2.8152		2.7784		4.3427		3.5450	
31600	31552	225.639	-47.511	2.8023		2.7656		4.3265		3.5319	
31700	31652	225.441	-47.709	2.7894		2.7529		4.3105		3.5187	
31800	31752	225.244	-47.906	2.7766		2.7403		4.2944		3.5056	
31900	31851	225.046	-48.104	2.7638		2.7277		4.2794		3.4926	
32000	31951	224.849	-48.301	2.7511	• 2	2.7151	- 1	4.2624	- 1	3.4795	- 1
32100	32051	224.651	-48.499	2.7384		2.7026		4.2465		3.4666	
32200	32150	224.454	-48.696	2.7258		2.6901		4.2307		3.4536	
32300	32250	224.256	-48.894	2.7132		2.6777		4.2148		3.4407	
32400	32350	224.059	-49.091	2.7006		2.6653		4.1991		3.4278	
32500	32449	223.861	-49.289	2.6882		2.6530		4.1833		3.4150	
32600	32549	223.664	-49.486	2.6757		2.6407		4.1677		3.4022	
32700	32649	223.466	-49.684	2.6633		2.6285		4.1520		3.3894	
32800	32748	223.269	-49.881	2.6510		2.6163		4.1364		3.3767	
32900	32848	223.071	-50.079	2.6387		2.6042		4.1209		3.3640	
33000	32948	222.874	-50.276	2.6264	• 2	2.5921	- 1	4.1054	- 1	3.3513	- 1
33100	33048	222.676	-50.474	2.6142		2.5800		4.0899		3.3387	
33200	33147	222.479	-50.671	2.6020		2.5680		4.0745		3.3261	
33300	33247	222.281	-50.869	2.5899		2.5561		4.0591		3.3136	
33400	33347	222.084	-51.067	2.5779		2.5441		4.0438		3.3010	
33500	33446	221.886	-51.264	2.5658		2.5323		4.0285		3.2886	
33600	33546	221.689	-51.461	2.5538		2.5205		4.0133		3.2761	
33700	33646	221.491	-51.659	2.5419		2.5087		3.9981		3.2637	
33800	33745	221.294	-51.856	2.5300		2.4969		3.9829		3.2514	
33900	33845	221.096	-52.054	2.5182		2.4852		3.9678		3.2390	
34000	33945	220.899	-52.251	2.5064	• 2	2.4736	- 1	3.9528	- 1	3.2267	- 1
34100	34044	220.701	-52.449	2.4946		2.4620		3.9377		3.2145	
34200	34144	220.504	-52.646	2.4829		2.4504		3.9228		3.2023	
34300	34244	220.306	-52.844	2.4713		2.4389		3.9078		3.1901	
34400	34343	220.109	-53.041	2.4596		2.4275		3.8930		3.1779	
34500	34443	219.911	-53.239	2.4481		2.4160		3.8781		3.1658	
34600	34543	219.714	-53.436	2.4365		2.4047		3.8633		3.1537	
34700	34642	219.516	-53.634	2.4250		2.3933		3.8486		3.1417	
34800	34742	219.319	-53.831	2.4136		2.3820		3.8339		3.1297	
34900	34842	219.122	-54.028	2.4022		2.3708		3.8192		3.1177	
35000	34941	218.924	-54.226	2.3908	• 2	2.3596	- 1	3.8046	- 1	3.1058	- 1
35200	35141	218.529	-54.621	2.3683		2.3373		3.7754		3.0820	
35400	35340	218.134	-55.016	2.3459		2.3152		3.7465		3.0584	
35600	35539	217.739	-55.411	2.3236		2.2932		3.7177		3.0349	
35800	35739	217.344	-55.806	2.3016		2.2715		3.6891		3.0115	
36000	35938	216.950	-56.200	2.2797		2.2498		3.6607		2.9883	
36200	36137	216.650	-56.500	2.2579		2.2284		3.6308		2.9639	
36400	36337	216.650	-56.500	2.2364		2.2072		3.5962		2.9356	
36600	36536	216.650	-56.500	2.2151		2.1861		3.5619		2.9077	
36800	36735	216.650	-56.500	2.1940		2.1653		3.5279		2.8799	
37000	36934	216.650	-56.500	2.1731	• 2	2.1446	- 1	3.4943	- 1	2.8525	- 1
37200	37134	216.650	-56.500	2.1523		2.1242		3.4610		2.8253	
37400	37333	216.650	-56.500	2.1318		2.1039		3.4280		2.7984	
37600	37532	216.650	-56.500	2.1115		2.0839		3.3953		2.7717	
37800	37732	216.650	-56.500	2.0914		2.0640		3.3630		2.7453	
38000	37931	216.650	-56.500	2.0714		2.0443		3.3309		2.7191	
38200	38130	216.650	-56.500	2.0517		2.0249		3.2992		2.6932	
38400	38329	216.650	-56.500	2.0321		2.0056		3.2677		2.6675	
38600	38529	216.650	-56.500	2.0128		1.9864		3.2366		2.6421	
38800	38728	216.650	-56.500	1.9936		1.9675		3.2057		2.6169	
39000	38927	216.650	-56.500	1.9746	• 2	1.9488	- 1	3.1752	- 1	2.5920	- 1
39200	39126	216.650	-56.500	1.9558		1.9302		3.1449		2.5673	
39400	39326	216.650	-56.500	1.9371		1.9118		3.1149		2.5428	
39600	39525	216.650	-56.500	1.9187		1.8936		3.0852		2.5186	
39800	39724	216.650	-56.500	1.9004		1.8755		3.0558		2.4946	
40000	39923	216.650	-56.500	1.8823		1.8576		3.0267		2.4708	
40200	40123	216.650	-56.500	1.8643		1.8399		2.9979		2.4472	
40400	40322	216.650	-56.500	1.8466		1.8224		2.9693		2.4239	
40600	40521	216.650	-56.500	1.8290		1.8050		2.9410		2.4008	
40800	40720	216.650	-56.500	1.8115		1.7878		2.9130		2.3779	
41000	40920	216.650	-56.500	1.7943	• 2	1.7708	- 1	2.8852	- 1	2.3553	- 1
41200	41119	216.650	-56.500	1.7772		1.7539		2.8577		2.3328	
41400	41318	216.650	-56.500	1.7602		1.7372		2.8305		2.3106	
41600	41517	216.650	-56.500	1.7435		1.7207		2.8035		2.2886	
41800	41716	216.650	-56.500	1.7268		1.7043		2.7768		2.2668	
42000	41916	216.650	-56.500	1.7104		1.6880		2.7503		2.2452	
42200	42115	216.650	-56.500	1.6941		1.6719		2.7241		2.2238	
42400	42314	216.650	-56.500	1.6779		1.6560		2.6982		2.2026	
42600	42513	216.650	-56.500	1.6620		1.6402		2.6725		2.1816	
42800	42712	216.650	-56.500	1.6461		1.6246		2.6470		2.1608	

Table 2-2: Geopotential Altitude, English Altitudes

L-7170-AKT-87-046

Altitude		Temperature		Pressure		Density	
H (ft)	Z (ft)	T (K)	t (°C)	P (mb)	P/P ₀	ρ (kg/m ³)	ρ/ρ_0
43000	43089	216.650	-56.500	1.6235	• 2	1.6023	- 1
43200	43290	216.650	-56.500	1.6080		1.5870	- 1
43400	43491	216.650	-56.500	1.5926		1.5718	
43600	43691	216.650	-56.500	1.5774		1.5567	
43800	43892	216.650	-56.500	1.5623		1.5418	
44000	44093	216.650	-56.500	1.5473		1.5271	
44200	44294	216.650	-56.500	1.5325		1.5125	
44400	44495	216.650	-56.500	1.5179		1.4980	
44600	44696	216.650	-56.500	1.5033		1.4837	
44800	44896	216.650	-56.500	1.4890		1.4695	
45000	45097	216.650	-56.500	1.4747	• 2	1.4554	- 1
45200	45298	216.650	-56.500	1.4606		1.4415	- 1
45400	45499	216.650	-56.500	1.4466		1.4277	
45600	45700	216.650	-56.500	1.4328		1.4141	
45800	45901	216.650	-56.500	1.4191		1.4005	
46000	46102	216.650	-56.500	1.4055		1.3871	
46200	46303	216.650	-56.500	1.3921		1.3739	
46400	46503	216.650	-56.500	1.3788		1.3607	
46600	46704	216.650	-56.500	1.3656		1.3477	
46800	46905	216.650	-56.500	1.3525		1.3348	
47000	47106	216.650	-56.500	1.3396	• 2	1.3220	- 1
47200	47307	216.650	-56.500	1.3267		1.3094	- 1
47400	47508	216.650	-56.500	1.3140		1.2969	
47600	47709	216.650	-56.500	1.3015		1.2845	
47800	47910	216.650	-56.500	1.2890		1.2722	
48000	48111	216.650	-56.500	1.2767		1.2600	
48200	48312	216.650	-56.500	1.2645		1.2479	
48400	48513	216.650	-56.500	1.2524		1.2360	
48600	48714	216.650	-56.500	1.2404		1.2242	
48800	48914	216.650	-56.500	1.2285		1.2125	
49000	49115	216.650	-56.500	1.2168	• 2	1.2009	- 1
49200	49316	216.650	-56.500	1.2051		1.1894	- 1
49400	49517	216.650	-56.500	1.1936		1.1780	
49600	49718	216.650	-56.500	1.1822		1.1667	
49800	49919	216.650	-56.500	1.1709		1.1556	
50000	50120	216.650	-56.500	1.1597		1.1445	
50200	50321	216.650	-56.500	1.1486		1.1336	
50400	50522	216.650	-56.500	1.1376		1.1227	
50600	50723	216.650	-56.500	1.1267		1.1120	
50800	50924	216.650	-56.500	1.1159		1.1013	
51000	51125	216.650	-56.500	1.1053	• 2	1.0909	- 1
51200	51326	216.650	-56.500	1.0947		1.0804	- 1
51400	51527	216.650	-56.500	1.0842		1.0700	
51600	51728	216.650	-56.500	1.0738		1.0598	
51800	51929	216.650	-56.500	1.0636		1.0497	
52000	52130	216.650	-56.500	1.0534		1.0396	
52200	52331	216.650	-56.500	1.0433		1.0297	
52400	52532	216.650	-56.500	1.0333		1.0198	
52600	52733	216.650	-56.500	1.0234		1.0101	
52800	52934	216.650	-56.500	1.0137		1.0004	
53000	53135	216.650	-56.500	1.0040	• 2	9.9087	- 2
53200	53336	216.650	-56.500	9.9439	• 1	9.8139	- 1
53400	53537	216.650	-56.500	9.8888		9.7200	
53600	53738	216.650	-56.500	9.7546		9.6270	
53800	53939	216.650	-56.500	9.6613		9.5349	
54000	54140	216.650	-56.500	9.5688		9.4437	
54200	54341	216.650	-56.500	9.4773		9.3534	
54400	54542	216.650	-56.500	9.3866		9.2639	
54600	54743	216.650	-56.500	9.2968		9.1753	
54800	54944	216.650	-56.500	9.2079		9.0875	
55000	55145	216.650	-56.500	9.1198	• 1	9.0005	- 2
55200	55346	216.650	-56.500	9.0326		8.9144	- 1
55400	55548	216.650	-56.500	8.9461		8.8292	- 1
55600	55749	216.650	-56.500	8.8606		8.7447	
55800	55950	216.650	-56.500	8.7758		8.6610	
56000	56151	216.650	-56.500	8.6918		8.5782	
56200	56352	216.650	-56.500	8.6087		8.4961	
56400	56553	216.650	-56.500	8.5263		8.4148	
56600	56754	216.650	-56.500	8.4448		8.3343	
56800	56955	216.650	-56.500	8.3640		8.2546	
57000	57156	216.650	-56.500	8.2840	• 1	8.1756	- 2
57200	57357	216.650	-56.500	8.2047		8.0974	- 1
57400	57558	216.650	-56.500	8.1262		8.0199	
57600	57760	216.650	-56.500	8.0485		7.9432	
57800	57961	216.650	-56.500	7.9715		7.8672	
58000	58162	216.650	-56.500	7.8952		7.7920	
58200	58363	216.650	-56.500	7.8197		7.7174	
58400	58564	216.650	-56.500	7.7449		7.6436	
58600	58765	216.650	-56.500	7.6708		7.5705	
58800	58966	216.650	-56.500	7.5974		7.4980	
59000	59167	216.650	-56.500	7.5246		7.4257	
59200	59368	216.650	-56.500	7.4524		7.3544	
59400	59569	216.650	-56.500	7.3807		7.2837	
59600	59770	216.650	-56.500	7.3095		7.2135	
59800	59971	216.650	-56.500	7.2388		7.1438	
60000	60172	216.650	-56.500	7.1686		7.0746	
60200	60373	216.650	-56.500	7.0989		7.0060	
60400	60574	216.650	-56.500	7.0297		6.9377	
60600	60775	216.650	-56.500	6.9610		6.8694	
60800	60976	216.650	-56.500	6.8928		6.8017	
61000	61177	216.650	-56.500	6.8250		6.7345	
61200	61378	216.650	-56.500	6.7577		6.6681	
61400	61579	216.650	-56.500	6.6908		6.6024	
61600	61780	216.650	-56.500	6.6244		6.5374	
61800	61981	216.650	-56.500	6.5585		6.4728	
62000	62182	216.650	-56.500	6.4931		6.4087	
62200	62383	216.650	-56.500	6.4282		6.3448	
62400	62584	216.650	-56.500	6.3638		6.2818	
62600	62785	216.650	-56.500	6.2999		6.2193	
62800	62986	216.650	-56.500	6.2365		6.1572	
63000	63187	216.650	-56.500	6.1736		6.0956	
63200	63388	216.650	-56.500	6.1112		6.0345	
63400	63589	216.650	-56.500	6.0493		5.9740	
63600	63790	216.650	-56.500	5.9879		5.9142	
63800	63991	216.650	-56.500	5.9270		5.8549	
64000	64192	216.650	-56.500	5.8666		5.7961	
64200	64393	216.650	-56.500	5.8067		5.7377	
64400	64594	216.650	-56.500	5.7473		5.6797	
64600	64795	216.650	-56.500	5.6884		5.6227	
64800	64996	216.650	-56.500	5.6300		5.5664	
65000	65197	216.650	-56.500	5.5721		5.5133	
65200	65398	216.650	-56.500	5.5147		5.4564	
65400	65599	216.650	-56.500	5.4578		5.4000	
65600	65800	216.650	-56.500	5.4014		5.3446	
65800	66001	216.650	-56.500	5.3455		5.2893	
66000	66202	216.650	-56.500	5.2901		5.2346	
66200	66403	216.650	-56.500	5.2352		5.1805	
66400	66604	216.650	-56.500	5.1808		5.1270	
66600	66805	216.650	-56.500	5.1270		5.0738	
66800	67006	216.650	-56.500	5.0743		5.0211	
67000	67207	216.650	-56.500	5.0213		4.9694	
67200	67408	216.650	-56.500	4.9694		4.9180	
67400	67609	216.650	-56.500	4.9180		4.8671	
67600	67810	216.650	-56.500	4.8671		4.8167	
67800	68011	216.650	-56.500	4.8167		4.7668	
68000	68212	216.650	-56.500	4.7668		4.7174	
68200	68413	216.650	-56.500	4.7174		4.6685	
68400	68614	216.650	-56.500	4.6685		4.6201	
68600	68815	216.650	-56.500	4.6201		4.5722	
68800	69016	216.650	-56.500	4.5722		4.5248	
69000	69217	216.650	-56.500	4.5248		4.4779	
69200	69418	216.650	-56.500	4.4779		4.4315	
69400	69619	216.650	-56.500	4.4315		4.3856	
69600	69820	216.650	-56.500	4.3856		4.3402	
69800	70021	216.650	-56.500	4.3402		4.2953	
70000	70222	216.650	-56.500	4.2953		4.2509	
70200	70423	216.650	-56.500	4.2509		4.2070	
70400	70624	216.650	-56.500	4.2070		4.1636	
70600	70825	216.650	-56.500	4.1636		4.1207	
70800	71026	216.650	-56.500	4.1207		4.0783	
71000	71227	216.650	-56.500	4.0783		4.0364	
71200	71428	216.650	-56.500	4.0364		4.0000	
71400	71629	216.650	-56.500	4.0000		3.9641	
71600	71830	216.650	-56.500	3.9641		3.9287	
71800	72031	216.650	-56.500	3.9287		3.8938	
72000	72232	216.650	-56.500	3.8938		3.8594	
72200	72433	216.650	-56.500	3.8594		3.8255	
72400	72634	216.650	-56.500	3.8255		3.7921	
72600	72835	216.650	-56.500	3.7921		3.7592	
72800	73036	216.650	-56.500	3.7592		3.7268	
73000	73237	216.650	-56.500	3.7268		3.6949	
73200	73438	216.650	-56.500	3.6949		3.6635	
73400	73639	216.650	-56.500	3.6635		3.6326	
73600	73840	216.650	-56.500	3.6326			

Table 2-2: Geometric Altitude, English Altitudes

Altitude		Temperature		Pressure		Density	
Z (ft)	H (ft)	T (K)	t (°C)	P (mb)	P/P ₀	ρ (kg/m ³)	ρ/ρ_0
43000	42912	216.650	-56.500	1.6304 • 2	1.6091 - 1	2.6218 - 1	2.1402 - 1
43200	43111	216.650	-56.500	1.6149	1.5938	2.5968	2.1198
43400	43310	216.650	-56.500	1.5995	1.5786	2.5721	2.0996
43600	43509	216.650	-56.500	1.5843	1.5636	2.5476	2.0796
43800	43708	216.650	-56.500	1.5692	1.5487	2.5233	2.0598
44000	43907	216.650	-56.500	1.5542	1.5339	2.4993	2.0402
44200	44107	216.650	-56.500	1.5394	1.5193	2.4754	2.0208
44400	44306	216.650	-56.500	1.5248	1.5048	2.4519	2.0015
44600	44505	216.650	-56.500	1.5102	1.4905	2.4285	1.9825
44800	44704	216.650	-56.500	1.4959	1.4763	2.4054	1.9636
45000	44903	216.650	-56.500	1.4816 • 2	1.4622 - 1	2.3825 - 1	1.9449 - 1
45200	45102	216.650	-56.500	1.4675	1.4483	2.3598	1.9263
45400	45301	216.650	-56.500	1.4535	1.4345	2.3373	1.9080
45600	45501	216.650	-56.500	1.4397	1.4208	2.3150	1.8898
45800	45700	216.650	-56.500	1.4259	1.4073	2.2930	1.8718
46000	45899	216.650	-56.500	1.4124	1.3939	2.2711	1.8540
46200	46098	216.650	-56.500	1.3989	1.3806	2.2495	1.8363
46400	46297	216.650	-56.500	1.3856	1.3675	2.2281	1.8188
46600	46496	216.650	-56.500	1.3724	1.3544	2.2069	1.8015
46800	46695	216.650	-56.500	1.3593	1.3415	2.1858	1.7844
47000	46894	216.650	-56.500	1.3464 • 2	1.3288 - 1	2.1650 - 1	1.7674 - 1
47200	47093	216.650	-56.500	1.3336	1.3161	2.1444	1.7505
47400	47293	216.650	-56.500	1.3209	1.3036	2.1240	1.7339
47600	47492	216.650	-56.500	1.3083	1.2912	2.1038	1.7173
47800	47691	216.650	-56.500	1.2958	1.2789	2.0837	1.7010
48000	47890	216.650	-56.500	1.2835	1.2667	2.0639	1.6848
48200	48089	216.650	-56.500	1.2712	1.2546	2.0442	1.6687
48400	48288	216.650	-56.500	1.2591	1.2427	2.0248	1.6529
48600	48487	216.650	-56.500	1.2472	1.2308	2.0055	1.6371
48800	48686	216.650	-56.500	1.2353	1.2191	1.9864	1.6215
49000	48885	216.650	-56.500	1.2235 • 2	1.2075 - 1	1.9675 - 1	1.6041 - 1
49200	49084	216.650	-56.500	1.2119	1.1960	1.9487	1.5908
49400	49283	216.650	-56.500	1.2003	1.1846	1.9302	1.5756
49600	49482	216.650	-56.500	1.1889	1.1733	1.9118	1.5606
49800	49681	216.650	-56.500	1.1776	1.1622	1.8936	1.5458
50000	49880	216.650	-56.500	1.1664	1.1511	1.8756	1.5311
50200	50079	216.650	-56.500	1.1553	1.1401	1.8577	1.5165
50400	50278	216.650	-56.500	1.1443	1.1293	1.8400	1.5021
50600	50478	216.650	-56.500	1.1334	1.1185	1.8225	1.4878
50800	50677	216.650	-56.500	1.1226	1.1079	1.8051	1.4736
51000	50876	216.650	-56.500	1.1119 • 2	1.0973 - 1	1.7880 - 1	1.4596 - 1
51200	51075	216.650	-56.500	1.1013	1.0869	1.7709	1.4457
51400	51274	216.650	-56.500	1.0908	1.0765	1.7541	1.4319
51600	51473	216.650	-56.500	1.0804	1.0663	1.7374	1.4183
51800	51672	216.650	-56.500	1.0701	1.0561	1.7208	1.4048
52000	51871	216.650	-56.500	1.0600	1.0461	1.7045	1.3914
52200	52070	216.650	-56.500	1.0499	1.0361	1.6882	1.3782
52400	52269	216.650	-56.500	1.0399	1.0263	1.6722	1.3650
52600	52468	216.650	-56.500	1.0300	1.0165	1.6562	1.3520
52800	52667	216.650	-56.500	1.0202	1.0068	1.6405	1.3392
53000	52866	216.650	-56.500	1.0105 • 2	9.9729 - 2	1.6249 - 1	1.3264 - 1
53200	53065	216.650	-56.500	1.0008	9.9779	1.6094	1.3138
53400	53264	216.650	-56.500	9.9135 • 1	9.7839	1.5941	1.3013
53600	53463	216.650	-56.500	9.8192	9.6908	1.5789	1.2889
53800	53662	216.650	-56.500	9.7257	9.5985	1.5639	1.2766
54000	53861	216.650	-56.500	9.6332	9.5072	1.5490	1.2645
54200	54059	216.650	-56.500	9.5415	9.4167	1.5343	1.2525
54400	54258	216.650	-56.500	9.4507	9.3271	1.5197	1.2405
54600	54457	216.650	-56.500	9.3607	9.2383	1.5052	1.2287
54800	54656	216.650	-56.500	9.2716	9.1504	1.4909	1.2170
55000	54855	216.650	-56.500	9.1834 • 1	9.0633 - 2	1.4767 - 1	1.2055 - 1
55200	55054	216.650	-56.500	9.0960	8.9771	1.4626	1.1940
55400	55253	216.650	-56.500	9.0095	8.8916	1.4487	1.1826
55600	55452	216.650	-56.500	8.9237	8.8070	1.4349	1.1714
55800	55651	216.650	-56.500	8.8388	8.7232	1.4213	1.1602
56000	55850	216.650	-56.500	8.7547	8.6402	1.4077	1.1492
56200	56049	216.650	-56.500	8.6714	8.5580	1.3943	1.1382
56400	56248	216.650	-56.500	8.5889	8.4766	1.3811	1.1274
56600	56447	216.650	-56.500	8.5071	8.3959	1.3679	1.1167
56800	56646	216.650	-56.500	8.4262	8.3160	1.3549	1.1061
57000	56845	216.650	-56.500	8.3460 • 1	8.2369 - 2	1.3420 - 1	1.0955 - 1
57200	57044	216.650	-56.500	8.2666	8.1585	1.3293	1.0851
57400	57242	216.650	-56.500	8.1880	8.0809	1.3166	1.0748
57600	57441	216.650	-56.500	8.1101	8.0040	1.3041	1.0646
57800	57640	216.650	-56.500	8.0329	7.9278	1.2917	1.0544
58000	57839	216.650	-56.500	7.9565	7.8524	1.2794	1.0444
58200	58038	216.650	-56.500	7.8808	7.7777	1.2672	1.0345
58400	58237	216.650	-56.500	7.8058	7.7037	1.2552	1.0246
58600	58436	216.650	-56.500	7.7315	7.6304	1.2432	1.0149
58800	58635	216.650	-56.500	7.6580	7.5578	1.2314	1.0052

Table 2-2 Geopotential Altitude, English Altitudes

Altitude		Temperature		Pressure		Density	
H (ft)	Z (ft)	T (K)	t (°C)	P (mb)	P/P ₀	ρ (kg/m ³)	ρ/ρ_0
59000	59167	216.650	-56.500	7.5247 • 1	7.4263 - 2	1.2100 - 1	9.8773 - 2
59200	59369	216.650	-56.500	7.4527	7.3553	1.1984	9.7824
59400	59570	216.650	-56.500	7.3814	7.2849	1.1869	9.6892
59600	59771	216.650	-56.500	7.3108	7.2152	1.1756	9.5965
59800	59972	216.650	-56.500	7.2409	7.1462	1.1643	9.5047
60000	60173	216.650	-56.500	7.1716	7.0778	1.1532	9.4137
60200	60374	216.650	-56.500	7.1030	7.0101	1.1422	9.3237
60400	60575	216.650	-56.500	7.0350	6.9430	1.1312	9.2345
60600	60777	216.650	-56.500	6.9677	6.8766	1.1204	9.1462
60800	60978	216.650	-56.500	6.9011	6.8108	1.1097	9.0587
61000	61179	216.650	-56.500	6.8351 • 1	6.7457 - 2	1.0991 - 1	8.9720 - 2
61200	61380	216.650	-56.500	6.7697	6.6811	1.0886	8.8862
61400	61581	216.650	-56.500	6.7049	6.6172	1.0781	8.8011
61600	61782	216.650	-56.500	6.6408	6.5539	1.0678	8.7170
61800	61984	216.650	-56.500	6.5772	6.4912	1.0576	8.6336
62000	62185	216.650	-56.500	6.5143	6.4291	1.0475	8.5510
62200	62386	216.650	-56.500	6.4520	6.3676	1.0375	8.4692
62400	62587	216.650	-56.500	6.3903	6.3067	1.0275	8.3881
62600	62788	216.650	-56.500	6.3291	6.2464	1.0177	8.3079
62800	62990	216.650	-56.500	6.2686	6.1866	1.0080	8.2284
63000	63191	216.650	-56.500	6.2086 • 1	6.1274 - 2	9.9834 - 2	8.1497 - 2
63200	63392	216.650	-56.500	6.1492	6.0688	9.8879	8.0717
63400	63593	216.650	-56.500	6.0904	6.0107	9.7933	7.9965
63600	63795	216.650	-56.500	6.0321	5.9532	9.6996	7.9180
63800	63996	216.650	-56.500	5.9744	5.8963	9.6068	7.8423
64000	64197	216.650	-56.500	5.9173	5.8399	9.5149	7.7673
64200	64398	216.650	-56.500	5.8607	5.7840	9.4239	7.6930
64400	64599	216.650	-56.500	5.8046	5.7287	9.3337	7.6194
64600	64801	216.650	-56.500	5.7491	5.6739	9.2444	7.5465
64800	65002	216.650	-56.500	5.6941	5.6196	9.1560	7.4743
65000	65203	216.650	-56.500	5.6396 • 1	5.5658 - 2	9.0684 - 2	7.4028 - 2
65200	65404	216.650	-56.500	5.5856	5.5126	8.9816	7.3319
65400	65606	216.650	-56.500	5.5322	5.4599	8.8957	7.2618
65600	65807	216.650	-56.500	5.4793	5.4076	8.8106	7.1923
65800	66008	216.706	-56.444	5.4269	5.3559	8.7261	7.1217
66000	66210	216.767	-56.383	5.3750	5.3047	8.6382	7.0516
66200	66411	216.828	-56.322	5.3236	5.2540	8.5532	6.9822
66400	66612	216.889	-56.261	5.2727	5.2037	8.4691	6.9136
66600	66813	216.950	-56.200	5.2223	5.1540	8.3858	6.8466
66800	67015	217.011	-56.139	5.1724	5.1048	8.3034	6.7783
67000	67216	217.072	-56.078	5.1230 • 1	5.0560 - 2	8.2218 - 2	6.7117 - 2
67200	67417	217.133	-56.017	5.0741	5.0078	8.1410	6.6457
67400	67619	217.194	-55.956	5.0257	4.9600	8.0611	6.5805
67600	67820	217.255	-55.895	4.9777	4.9127	7.9819	6.5158
67800	68021	217.316	-55.834	4.9303	4.8658	7.9035	6.4519
68000	68222	217.377	-55.773	4.8833	4.8194	7.8260	6.3886
68200	68424	217.438	-55.712	4.8367	4.7734	7.7492	6.3259
68400	68625	217.499	-55.651	4.7906	4.7279	7.6732	6.2638
68600	68826	217.559	-55.591	4.7450	4.6829	7.5980	6.2024
68800	69028	217.620	-55.530	4.6998	4.6383	7.5235	6.1416
69000	69229	217.681	-55.469	4.6550 • 1	4.5941 - 2	7.4497 - 2	6.0814 - 2
69200	69430	217.742	-55.408	4.6107	4.5504	7.3767	6.0218
69400	69632	217.803	-55.347	4.5668	4.5071	7.3045	5.9629
69600	69833	217.864	-55.286	4.5233	4.4642	7.2330	5.9045
69800	70034	217.925	-55.225	4.4803	4.4217	7.1622	5.8467
70000	70236	217.986	-55.164	4.4377	4.3797	7.0921	5.7894
70200	70437	218.047	-55.103	4.3955	4.3380	7.0227	5.7328
70400	70638	218.108	-55.042	4.3537	4.2968	6.9540	5.6767
70600	70840	218.169	-54.981	4.3124	4.2560	6.8860	5.6212
70800	71041	218.230	-54.920	4.2714	4.2156	6.8187	5.5663
71000	71243	218.291	-54.859	4.2308 • 1	4.1755 - 2	6.7520 - 2	5.5119 - 2
71200	71444	218.352	-54.798	4.1917	4.1359	6.6861	5.4580
71400	71645	218.413	-54.737	4.1539	4.0966	6.6208	5.4047
71600	71847	218.474	-54.676	4.1115	4.0578	6.5561	5.3519
71800	72048	218.535	-54.615	4.0725	4.0193	6.4921	5.2997
72000	72249	218.596	-54.554	4.0339	3.9811	6.4288	5.2480
72200	72451	218.657	-54.493	3.9957	3.9434	6.3660	5.1968
72400	72652	218.718	-54.432	3.9578	3.9060	6.3039	5.1461
72600	72854	218.779	-54.371	3.9203	3.8690	6.2425	5.0959
72800	73055	218.840	-54.310	3.8831	3.8324	6.1816	5.0462
73000	73256	218.901	-54.249	3.8464 • 1	3.7961 - 2	6.1214 - 2	4.9970 - 2
73200	73458	218.962	-54.188	3.8100	3.7601	6.0617	4.9483
73400	73659	219.023	-54.127	3.7739	3.7245	6.0027	4.9001
73600	73861	219.083	-54.067	3.7382	3.6893	5.9442	4.8524
73800	74062	219.144	-54.006	3.7028	3.6544	5.8864	4.8052
74000	74264	219.205	-53.945	3.6678	3.6198	5.8291	4.7584
74200	74465	219.266	-53.884	3.6331	3.5856	5.7724	4.7121
74400	74666	219.327	-53.823	3.5988	3.5517	5.7162	4.6663
74600	74868	219.388	-53.762	3.5648	3.5182	5.6606	4.6209
74800	75069	219.449	-53.701	3.5311	3.4849	5.6056	4.5760

Table 2-2: Geometric Altitude, English Altitudes

Altitude		Temperature		Pressure		Density	
Z (ft)	H (ft)	T (K)	t (°C)	P (mb)	P/P ₀	ρ (kg/m ³)	ρ/ρ_0
59000	58834	216.650	-56.500	7.5851 • 1	7.4859 - 2	1.2147 - 1	9.9566 - 2
59200	59032	216.650	-56.500	7.5130	7.4147	1.2081	9.8618
59400	59231	216.650	-56.500	7.4415	7.3442	1.1966	9.7680
59600	59430	216.650	-56.500	7.3707	7.2743	1.1852	9.6751
59800	59629	216.650	-56.500	7.3006	7.2051	1.1739	9.5831
60000	59828	216.650	-56.500	7.2312	7.1366	1.1628	9.4919
60200	60027	216.650	-56.500	7.1624	7.0687	1.1517	9.4016
60400	60226	216.650	-56.500	7.0942	7.0015	1.1407	9.3122
60600	60424	216.650	-56.500	7.0268	6.9349	1.1299	9.2236
60800	60623	216.650	-56.500	6.9599	6.8689	1.1191	9.1359
61000	60822	216.650	-56.500	6.8937 • 1	6.8036 - 2	1.1085 - 1	9.0490 - 2
61200	61021	216.650	-56.500	6.8282	6.7389	1.0980	8.9629
61400	61220	216.650	-56.500	6.7632	6.6748	1.0875	8.8777
61600	61419	216.650	-56.500	6.6989	6.6113	1.0772	8.7933
61800	61617	216.650	-56.500	6.6352	6.5484	1.0669	8.7096
62000	61816	216.650	-56.500	6.5721	6.4861	1.0568	8.6268
62200	62015	216.650	-56.500	6.5096	6.4245	1.0467	8.5448
62400	62214	216.650	-56.500	6.4477	6.3634	1.0368	8.4635
62600	62413	216.650	-56.500	6.3864	6.3028	1.0269	8.3810
62800	62611	216.650	-56.500	6.3256	6.2429	1.0172	8.3033
63000	62810	216.650	-56.500	6.2655 • 1	6.1835 - 2	1.0075 - 1	8.2243 - 2
63200	63009	216.650	-56.500	6.2059	6.1247	9.9790 - 2	8.1461
63400	63208	216.650	-56.500	6.1469	6.0665	9.8841	8.0687
63600	63407	216.650	-56.500	6.0884	6.0088	9.7901	7.9919
63800	63605	216.650	-56.500	6.0305	5.9517	9.6970	7.9159
64000	63804	216.650	-56.500	5.9732	5.8951	9.6048	7.8407
64200	64003	216.650	-56.500	5.9164	5.8390	9.5135	7.7661
64400	64202	216.650	-56.500	5.8601	5.7835	9.4231	7.6923
64600	64401	216.650	-56.500	5.8044	5.7285	9.3335	7.6191
64800	64599	216.650	-56.500	5.7492	5.6741	9.2447	7.5467
65000	64798	216.650	-56.500	5.6946 • 1	5.6201 - 2	9.1568 - 2	7.4750 - 2
65200	64997	216.650	-56.500	5.6404	5.5667	9.0698	7.4039
65400	65196	216.650	-56.500	5.5868	5.5137	8.9835	7.3335
65600	65394	216.650	-56.500	5.5337	5.4613	8.8981	7.2638
65800	65593	216.650	-56.500	5.4811	5.4094	8.8135	7.1947
66000	65792	216.703	-56.447	5.4290	5.3580	8.7276	7.1266
66200	65991	216.764	-56.386	5.3774	5.3071	8.6423	7.0549
66400	66189	216.824	-56.326	5.3263	5.2567	8.5578	6.9859
66600	66388	216.885	-56.265	5.2757	5.2067	8.4741	6.9176
66800	66587	216.945	-56.205	5.2256	5.1573	8.3913	6.8500
67000	66785	217.006	-56.144	5.1760 • 1	5.1083 - 2	8.3094 - 2	6.7831 - 2
67200	66984	217.067	-56.083	5.1269	5.0599	8.2282	6.7169
67400	67183	217.127	-56.023	5.0783	5.0119	8.1479	6.6513
67600	67382	217.188	-55.962	5.0301	4.9643	8.0684	6.5864
67800	67580	217.248	-55.902	4.9824	4.9173	7.9896	6.5222
68000	67779	217.309	-55.841	4.9352	4.8707	7.9117	6.4585
68200	67978	217.369	-55.781	4.8885	4.8245	7.8346	6.3958
68400	68176	217.430	-55.720	4.8421	4.7788	7.7582	6.3332
68600	68375	217.491	-55.659	4.7963	4.7336	7.6826	6.2715
68800	68574	217.551	-55.599	4.7509	4.6888	7.6078	6.2104
69000	68772	217.612	-55.538	4.7059 • 1	4.6444 - 2	7.5337 - 2	6.1499 - 2
69200	68971	217.672	-55.478	4.6614	4.6004	7.4603	6.0900
69400	69170	217.733	-55.417	4.6173	4.5569	7.3877	6.0308
69600	69368	217.793	-55.357	4.5737	4.5139	7.3158	5.9721
69800	69567	217.854	-55.296	4.5304	4.4712	7.2446	5.9140
70000	69766	217.914	-55.236	4.4876	4.4289	7.1742	5.8565
70200	69964	217.975	-55.175	4.4452	4.3871	7.1044	5.7995
70400	70163	218.036	-55.114	4.4032	4.3457	7.0354	5.7432
70600	70362	218.096	-55.054	4.3617	4.3046	6.9670	5.6874
70800	70560	218.157	-54.993	4.3205	4.2640	6.8994	5.6321
71000	70759	218.217	-54.933	4.2797 • 1	4.2238 - 2	6.8324 - 2	5.5774 - 2
71200	70958	218.278	-54.872	4.2394	4.1839	6.7660	5.5213
71400	71156	218.338	-54.812	4.1994	4.1445	6.7004	5.4667
71600	71355	218.399	-54.751	4.1598	4.1054	6.6354	5.4126
71800	71554	218.459	-54.691	4.1206	4.0667	6.5710	5.3581
72000	71752	218.520	-54.630	4.0818	4.0284	6.5073	5.3021
72200	71951	218.581	-54.569	4.0433	3.9904	6.4442	5.2466
72400	72150	218.641	-54.509	4.0053	3.9529	6.3818	5.1906
72600	72348	218.702	-54.448	3.9675	3.9157	6.3200	5.1351
72800	72547	218.762	-54.388	3.9302	3.8788	6.2588	5.0802
73000	72745	218.823	-54.327	3.8932 • 1	3.8423 - 2	6.1982 - 2	5.0257 - 2
73200	72944	218.883	-54.267	3.8566	3.8062	6.1382	5.0107
73400	73143	218.944	-54.206	3.8204	3.7704	6.0788	4.9673
73600	73341	219.004	-54.146	3.7844	3.7350	6.0200	4.9143
73800	73540	219.065	-54.085	3.7489	3.6999	5.9618	4.8647
74000	73738	219.125	-54.025	3.7137	3.6651	5.9041	4.8147
74200	73937	219.186	-53.964	3.6788	3.6307	5.8471	4.7731
74400	74136	219.246	-53.904	3.6443	3.5966	5.7906	4.7270
74600	74334	219.307	-53.843	3.6100	3.5628	5.7346	4.6813
74800	74533	219.367	-53.783	3.5762	3.5294	5.6793	4.6361

Table 2-2: Geopotential Altitude, English Altitudes

Altitude		Temperature		Pressure		Density	
H (ft)	Z (ft)	T (K)	t (°C)	P (mb)	P/P ₀	ρ (kg/m ³)	ρ/ρ_0
75000	75271	219.510	-53.640	3.4978	• 1	5.5511	- 2
75200	75472	219.571	-53.579	3.4647		5.4972	4.4475
75400	75674	219.632	-53.518	3.4320		5.4438	4.4439
75600	75875	219.693	-53.457	3.3996		5.3909	4.4007
75800	76077	219.754	-53.396	3.3676		5.3386	4.3580
76000	76278	219.815	-53.335	3.3358		5.2868	4.3157
76200	76479	219.876	-53.274	3.3044		5.2355	4.2738
76400	76681	219.937	-53.213	3.2732		5.1847	4.2324
76600	76882	219.998	-53.152	3.2424		5.1344	4.1913
76800	77084	220.059	-53.091	3.2118		5.0846	4.1507
77000	77285	220.120	-53.030	3.1816	• 1	5.0353	- 2
77200	77487	220.181	-52.969	3.1516		4.9866	4.0707
77400	77688	220.242	-52.908	3.1220		4.9382	4.0312
77600	77890	220.303	-52.847	3.0926		4.8904	3.9922
77800	78091	220.364	-52.786	3.0635		4.8431	3.9535
78000	78293	220.425	-52.725	3.0347		4.7962	3.9153
78200	78494	220.486	-52.664	3.0061		4.7498	3.8774
78400	78696	220.547	-52.603	2.9779		4.7038	3.8399
78600	78897	220.607	-52.543	2.9499		4.6584	3.8027
78800	79099	220.668	-52.482	2.9222		4.6133	3.7660
79000	79300	220.729	-52.421	2.8947	• 1	4.5687	- 2
79200	79502	220.790	-52.360	2.8676		4.5246	3.6935
79400	79703	220.851	-52.299	2.8406		4.4809	3.6579
79600	79905	220.912	-52.238	2.8140		4.4376	3.6225
79800	80107	220.973	-52.177	2.7876		4.3948	3.5876
80000	80308	221.034	-52.116	2.7614		4.3523	3.5529
80200	80510	221.095	-52.055	2.7355		4.3103	3.5186
80400	80711	221.156	-51.994	2.7099		4.2688	3.4847
80600	80913	221.217	-51.933	2.6845		4.2276	3.4511
80800	81114	221.278	-51.872	2.6594		4.1868	3.4178
81000	81316	221.339	-51.811	2.6344	• 1	4.1465	- 2
81200	81517	221.400	-51.750	2.6098		4.1065	3.3523
81400	81719	221.461	-51.689	2.5853		4.0669	3.3200
81600	81921	221.522	-51.628	2.5612		4.0278	3.2880
81800	82122	221.583	-51.567	2.5372		3.9890	3.2563
82000	82324	221.644	-51.506	2.5135		3.9506	3.2250
82200	82525	221.705	-51.445	2.4900		3.9126	3.1939
82400	82727	221.766	-51.384	2.4667		3.8749	3.1632
82600	82928	221.827	-51.323	2.4436		3.8377	3.1328
82800	83130	221.888	-51.262	2.4208		3.8008	3.1027
83000	83332	221.949	-51.201	2.3982	• 1	3.7642	- 2
83200	83533	222.010	-51.140	2.3758		3.7281	3.0633
83400	83735	222.071	-51.079	2.3536		3.6922	3.0341
83600	83936	222.131	-51.019	2.3316		3.6568	2.9851
83800	84138	222.192	-50.958	2.3099		3.6217	2.9565
84000	84340	222.253	-50.897	2.2883		3.5869	2.9281
84200	84541	222.314	-50.836	2.2670		3.5525	2.9000
84400	84743	222.375	-50.775	2.2459		3.5184	2.8722
84600	84945	222.436	-50.714	2.2249		3.4846	2.8446
84800	85146	222.497	-50.653	2.2042		3.4512	2.8173
85000	85348	222.558	-50.592	2.1837	• 1	3.4181	- 2
85200	85550	222.619	-50.531	2.1633		3.3854	2.7903
85400	85751	222.680	-50.470	2.1432		3.3530	2.7737
85600	85953	222.741	-50.409	2.1232		3.3208	2.7571
85800	86154	222.802	-50.348	2.1035		3.2890	2.7409
86000	86356	222.863	-50.287	2.0839		3.2575	2.7252
86200	86558	222.924	-50.226	2.0645		3.2264	2.7099
86400	86759	222.985	-50.165	2.0453		3.1955	2.6946
86600	86961	223.046	-50.104	2.0263		3.1649	2.6800
86800	87163	223.107	-50.043	2.0075		3.1347	2.6559
87000	87364	223.168	-49.982	1.9888	• 1	3.1047	- 2
87200	87566	223.229	-49.921	1.9704		3.0750	2.5344
87400	87768	223.290	-49.860	1.9521		3.0456	2.5102
87600	87970	223.351	-49.799	1.9340		3.0165	2.4862
87800	88171	223.412	-49.738	1.9160		2.9877	2.4625
88000	88373	223.473	-49.677	1.8982		2.9592	2.4390
88200	88575	223.534	-49.616	1.8806		2.9310	2.4157
88400	88776	223.595	-49.555	1.8632		2.9030	2.3926
88600	88978	223.655	-49.495	1.8459		2.8753	2.3698
88800	89180	223.716	-49.434	1.8288		2.8479	2.3472
89000	89381	223.777	-49.373	1.8119	• 1	2.8207	- 2
89200	89583	223.838	-49.312	1.7951		2.7934	2.3248
89400	89785	223.899	-49.251	1.7785		2.7672	2.3026
89600	89987	223.960	-49.190	1.7620		2.7409	2.2807
89800	90188	224.021	-49.129	1.7457		2.7148	2.2590
90000	90390	224.082	-49.068	1.7295		2.6889	2.2374
90200	90592	224.143	-49.007	1.7135		2.6633	2.2161
90400	90794	224.204	-48.946	1.6977		2.6380	2.1950
90600	90995	224.265	-48.885	1.6820		2.6129	2.1741
90800	91197	224.326	-48.824	1.6665		2.5880	2.1534
91000							2.1330
91200							2.1127

Table 2-2:

Geopotential Altitude, English Altitudes

Altitude		Temperature		Pressure		Density	
H (ft)	Z (ft)	T (K)	t (°C)	P (mb)	P/P ₀	ρ (kg/m ³)	ρ/ρ_0
91000	91399	224.387	-48.763	1.6511	• 1	1.6295	- 2
91200	91601	224.448	-48.702	1.6358		1.6144	- 2
91400	91802	224.509	-48.641	1.6207		1.5995	- 2
91600	92004	224.570	-48.580	1.6057		1.5847	- 2
91800	92206	224.631	-48.519	1.5909		1.5701	- 2
92000	92408	224.692	-48.458	1.5762		1.5556	- 2
92200	92609	224.753	-48.397	1.5617		1.5413	- 2
92400	92811	224.814	-48.336	1.5473		1.5271	- 2
92600	93013	224.875	-48.275	1.5330		1.5130	- 2
92800	93215	224.936	-48.214	1.5189		1.4990	- 2
93000	93417	224.997	-48.153	1.5049	• 1	1.4852	- 2
93200	93618	225.058	-48.092	1.4910		1.4715	- 2
93400	93820	225.119	-48.031	1.4773		1.4580	- 2
93600	94022	225.179	-47.971	1.4637		1.4444	- 2
93800	94224	225.240	-47.910	1.4502		1.4313	- 2
94000	94426	225.301	-47.849	1.4369		1.4181	- 2
94200	94627	225.362	-47.788	1.4237		1.4051	- 2
94400	94829	225.423	-47.727	1.4106		1.3921	- 2
94600	95031	225.484	-47.666	1.3976		1.3793	- 2
94800	95233	225.545	-47.605	1.3848		1.3666	- 2
95000	95435	225.606	-47.544	1.3720	• 1	1.3541	- 2
95200	95637	225.667	-47.483	1.3594		1.3416	- 2
95400	95838	225.728	-47.422	1.3469		1.3293	- 2
95600	96040	225.789	-47.361	1.3346		1.3171	- 2
95800	96242	225.850	-47.300	1.3223		1.3050	- 2
96000	96444	225.911	-47.239	1.3102		1.2930	- 2
96200	96646	225.972	-47.178	1.2982		1.2812	- 2
96400	96848	226.033	-47.117	1.2863		1.2694	- 2
96600	97050	226.094	-47.056	1.2745		1.2578	- 2
96800	97251	226.155	-46.995	1.2628		1.2463	- 2
97000	97453	226.216	-46.934	1.2512	• 1	1.2348	- 2
97200	97655	226.277	-46.873	1.2397		1.2235	- 2
97400	97857	226.338	-46.812	1.2284		1.2123	- 2
97600	98059	226.399	-46.751	1.2171		1.2012	- 2
97800	98261	226.460	-46.690	1.2060		1.1902	- 2
98000	98463	226.521	-46.629	1.1949		1.1793	- 2
98200	98665	226.582	-46.568	1.1840		1.1685	- 2
98400	98866	226.643	-46.507	1.1732		1.1578	- 2
98600	99068	226.703	-46.447	1.1624		1.1472	- 2
98800	99270	226.764	-46.386	1.1518		1.1368	- 2
99000	99472	226.825	-46.325	1.1413	• 1	1.1264	- 2
99200	99674	226.886	-46.264	1.1309		1.1161	- 2
99400	99876	226.947	-46.203	1.1205		1.1059	- 2
99600	100078	227.008	-46.142	1.1103		1.0958	- 2
99800	100280	227.069	-46.081	1.1002		1.0858	- 2
100000	100482	227.130	-46.020	1.0901		1.0759	- 2
100200	100684	227.191	-45.959	1.0802		1.0660	- 2
100400	100886	227.252	-45.898	1.0703		1.0563	- 2
100600	101088	227.313	-45.837	1.0605		1.0467	- 2
100800	101290	227.374	-45.776	1.0509		1.0371	- 2
101000	101492	227.435	-45.715	1.0413	• 1	1.0277	- 2
101200	101693	227.496	-45.654	1.0318		1.0183	- 2
101400	101895	227.557	-45.593	1.0224		1.0090	- 2
101600	102097	227.618	-45.532	1.0131		9.9986	- 3
101800	102299	227.679	-45.471	1.0039		9.9978	- 3
102000	102501	227.740	-45.410	9.9877	• 0	9.9876	- 3
102200	102703	227.801	-45.349	9.9771		9.9782	- 3
102400	102905	227.862	-45.288	9.9674		9.9697	- 3
102600	103107	227.923	-45.227	9.9578		9.9520	- 3
102800	103309	227.984	-45.166	9.9483		9.9451	- 3
103000	103511	228.045	-45.105	9.9388	• 0	9.9391	- 3
103200	103713	228.106	-45.044	9.9293		9.9298	- 3
103400	103915	228.167	-44.983	9.9198		9.9204	- 3
103600	104117	228.227	-44.923	9.9103		9.9125	- 3
103800	104319	228.288	-44.862	9.9008		9.9028	- 3
104000	104521	228.349	-44.801	9.8913		9.8947	- 3
104200	104723	228.410	-44.740	9.8818		9.8874	- 3
104400	104925	228.471	-44.679	9.8723		9.8788	- 3
104600	105127	228.532	-44.618	9.8628		9.8710	- 3
104800	105329	228.593	-44.557	9.8533		9.8639	- 3
105000	105531	228.654	-44.496	9.8438	• 0	9.8565	- 3
105200	105733	228.715	-44.435	9.8343		9.8490	- 3
105400	105935	228.776	-44.374	9.8248		9.8417	- 3
105600	106137	228.837	-44.313	9.8153		9.8344	- 3
105800	106339	228.898	-44.252	9.8058		9.8271	- 3
106000	106541	228.959	-44.191	9.7963		9.8198	- 3
106200	106743	229.020	-44.130	9.7868		9.8125	- 3
106400	106945	229.081	-44.069	9.7773		9.8052	- 3
106600	107147	229.142	-44.008	9.7678		9.7979	- 3
106800	107349	229.203	-43.947	9.7583		9.7906	- 3
107000	107551	229.264	-43.886	9.7488		9.7833	- 3
107200	107753	229.325	-43.825	9.7393		9.7760	- 3
107400	107955	229.386	-43.764	9.7298		9.7687	- 3
107600	108157	229.447	-43.703	9.7203		9.7614	- 3
107800	108359	229.508	-43.642	9.7108		9.7541	- 3
108000	108561	229.569	-43.581	9.7013		9.7468	- 3
108200	108763	229.630	-43.520	9.6918		9.7395	- 3
108400	108965	229.691	-43.459	9.6823		9.7322	- 3
108600	109167	229.752	-43.398	9.6728		9.7249	- 3
108800	109369	229.813	-43.337	9.6633		9.7176	- 3
109000	109571	229.874	-43.276	9.6538		9.7103	- 3
109200	109773	229.935	-43.215	9.6443		9.7030	- 3
109400	109975	229.996	-43.154	9.6348		9.6957	- 3
109600	110177	230.057	-43.093	9.6253		9.6884	- 3
109800	110379	230.118	-43.032	9.6158		9.6811	- 3
110000	110581	230.179	-42.971	9.6063		9.6738	- 3

Table 2-2: Geometric Altitude, English Altitudes

Altitude		Temperature		Pressure		Density	
Z (ft)	H (ft)	T (K)	t (°C)	P (mb)	P/P ₀	ρ (kg/m ³)	ρ/ρ_0
75000	74731	219.428	-53.722	3.5426	1	5.6244	1
75200	74930	219.488	-53.662	3.5094	0.999	5.5701	0.999
75400	75128	219.549	-53.601	3.4765	0.998	5.5164	0.998
75600	75327	219.610	-53.540	3.4439	0.997	5.4632	0.997
75800	75525	219.670	-53.480	3.4117	0.996	5.4105	0.996
76000	75724	219.731	-53.419	3.3797	0.995	5.3584	0.995
76200	75923	219.791	-53.359	3.3481	0.994	5.3067	0.994
76400	76121	219.852	-53.298	3.3167	0.993	5.2556	0.993
76600	76320	219.912	-53.238	3.2857	0.992	5.2050	0.992
76800	76518	219.973	-53.177	3.2549	0.991	5.1549	0.991
77000	76717	220.033	-53.117	3.2245	0.990	5.1053	0.990
77200	76915	220.094	-53.056	3.1943	0.989	5.0561	0.989
77400	77114	220.154	-52.996	3.1645	0.988	5.0075	0.988
77600	77312	220.215	-52.935	3.1349	0.987	4.9593	0.987
77800	77511	220.275	-52.875	3.1056	0.986	4.9117	0.986
78000	77709	220.336	-52.814	3.0766	0.985	4.8645	0.985
78200	77908	220.396	-52.754	3.0479	0.984	4.8177	0.984
78400	78106	220.457	-52.693	3.0195	0.983	4.7714	0.983
78600	78305	220.517	-52.633	2.9913	0.982	4.7256	0.982
78800	78503	220.578	-52.572	2.9634	0.981	4.6803	0.981
79000	78702	220.638	-52.512	2.9357	0.980	4.6353	0.980
79200	78900	220.699	-52.451	2.9084	0.979	4.5909	0.979
79400	79099	220.759	-52.391	2.8813	0.978	4.5468	0.978
79600	79297	220.820	-52.330	2.8544	0.977	4.5032	0.977
79800	79496	220.880	-52.270	2.8278	0.976	4.4601	0.976
80000	79694	220.941	-52.209	2.8015	0.975	4.4173	0.975
80200	79893	221.001	-52.149	2.7754	0.974	4.3750	0.974
80400	80091	221.062	-52.088	2.7496	0.973	4.3331	0.973
80600	80290	221.122	-52.028	2.7240	0.972	4.2916	0.972
80800	80488	221.183	-51.967	2.6987	0.971	4.2505	0.971
81000	80687	221.243	-51.907	2.6736	0.970	4.2099	0.970
81200	80885	221.304	-51.846	2.6487	0.969	4.1696	0.969
81400	81084	221.364	-51.786	2.6241	0.968	4.1297	0.968
81600	81282	221.425	-51.725	2.5997	0.967	4.0902	0.967
81800	81480	221.485	-51.665	2.5756	0.966	4.0511	0.966
82000	81679	221.546	-51.604	2.5517	0.965	4.0124	0.965
82200	81877	221.606	-51.544	2.5280	0.964	3.9741	0.964
82400	82076	221.667	-51.483	2.5045	0.963	3.9362	0.963
82600	82274	221.727	-51.423	2.4813	0.962	3.8986	0.962
82800	82473	221.788	-51.362	2.4583	0.961	3.8614	0.961
83000	82671	221.848	-51.302	2.4355	0.960	3.8245	0.960
83200	82869	221.908	-51.242	2.4129	0.959	3.7880	0.959
83400	83068	221.969	-51.181	2.3905	0.958	3.7519	0.958
83600	83266	222.029	-51.121	2.3684	0.957	3.7162	0.957
83800	83465	222.090	-51.060	2.3465	0.956	3.6807	0.956
84000	83663	222.150	-51.000	2.3248	0.955	3.6457	0.955
84200	83861	222.211	-50.939	2.3032	0.954	3.6109	0.954
84400	84060	222.271	-50.879	2.2819	0.953	3.5766	0.953
84600	84258	222.332	-50.818	2.2608	0.952	3.5425	0.952
84800	84457	222.392	-50.758	2.2399	0.951	3.5088	0.951
85000	84655	222.453	-50.697	2.2192	0.950	3.4754	0.950
85200	84853	222.513	-50.637	2.1987	0.949	3.4424	0.949
85400	85052	222.574	-50.576	2.1784	0.948	3.4096	0.948
85600	85250	222.634	-50.516	2.1583	0.947	3.3772	0.947
85800	85448	222.695	-50.455	2.1383	0.946	3.3451	0.946
86000	85647	222.755	-50.395	2.1186	0.945	3.3134	0.945
86200	85845	222.815	-50.335	2.0990	0.944	3.2819	0.944
86400	86044	222.876	-50.274	2.0797	0.943	3.2507	0.943
86600	86242	222.936	-50.214	2.0605	0.942	3.2199	0.942
86800	86440	222.997	-50.153	2.0415	0.941	3.1893	0.941
87000	86639	223.057	-50.093	2.0227	0.940	3.1591	0.940
87200	86837	223.118	-50.032	2.0040	0.939	3.1291	0.939
87400	87035	223.178	-49.972	1.9856	0.938	3.0994	0.938
87600	87234	223.239	-49.911	1.9673	0.937	3.0701	0.937
87800	87432	223.299	-49.851	1.9492	0.936	3.0410	0.936
88000	87630	223.360	-49.790	1.9312	0.935	3.0122	0.935
88200	87829	223.420	-49.730	1.9135	0.934	2.9836	0.934
88400	88027	223.480	-49.670	1.8959	0.933	2.9554	0.933
88600	88225	223.541	-49.609	1.8784	0.932	2.9274	0.932
88800	88423	223.601	-49.549	1.8611	0.931	2.8997	0.931
89000	88622	223.662	-49.488	1.8440	0.930	2.8723	0.930
89200	88820	223.722	-49.428	1.8271	0.929	2.8451	0.929
89400	89018	223.783	-49.367	1.8103	0.928	2.8182	0.928
89600	89217	223.843	-49.307	1.7937	0.927	2.7916	0.927
89800	89415	223.904	-49.246	1.7772	0.926	2.7652	0.926
90000	89613	223.964	-49.186	1.7609	0.925	2.7391	0.925
90200	89812	224.024	-49.126	1.7448	0.924	2.7132	0.924
90400	90010	224.085	-49.065	1.7287	0.923	2.6876	0.923
90600	90208	224.145	-49.005	1.7129	0.922	2.6623	0.922
90800	90406	224.206	-48.944	1.6972	0.921	2.6372	0.921

Table 2-2: Geometric Altitude, English Altitudes

L-7170-AKT-87-046

Altitude		Temperature		Pressure		Density	
Z (ft)	H (ft)	T (K)	t (°C)	P (mb)	P/P ₀	ρ (kg/m ³)	ρ/ρ_0
91000	90605	224.266	-48.884	1.6816	• 1	1.6596	- 2
91200	90803	224.327	-48.823	1.6662		1.6444	
91400	91001	224.387	-48.763	1.6510		1.6294	
91600	91199	224.447	-48.703	1.6358		1.6145	
91800	91398	224.508	-48.642	1.6209		1.5997	
92000	91596	224.568	-48.582	1.6060		1.5850	
92200	91794	224.629	-48.521	1.5913		1.5705	
92400	91992	224.689	-48.461	1.5768		1.5562	
92600	92191	224.750	-48.400	1.5624		1.5419	
92800	92389	224.810	-48.340	1.5481		1.5278	
93000	92587	224.870	-48.280	1.5339	• 1	1.5139	- 2
93200	92785	224.931	-48.219	1.5199		1.5000	
93400	92984	224.991	-48.159	1.5060		1.4863	
93600	93182	225.052	-48.098	1.4923		1.4728	
93800	93380	225.112	-48.038	1.4787		1.4593	
94000	93578	225.173	-47.977	1.4652		1.4460	
94200	93776	225.233	-47.917	1.4518		1.4328	
94400	93975	225.293	-47.857	1.4386		1.4198	
94600	94173	225.354	-47.796	1.4254		1.4060	
94800	94371	225.414	-47.736	1.4125		1.3940	
95000	94569	225.475	-47.675	1.3996	• 1	1.3813	- 2
95200	94767	225.535	-47.615	1.3868		1.3687	
95400	94966	225.595	-47.555	1.3742		1.3562	
95600	95164	225.656	-47.494	1.3617		1.3439	
95800	95362	225.716	-47.434	1.3491		1.3317	
96000	95560	225.777	-47.373	1.3370		1.3195	
96200	95758	225.837	-47.313	1.3249		1.3075	
96400	95956	225.897	-47.253	1.3128		1.2956	
96600	96155	225.958	-47.192	1.3009		1.2839	
96800	96353	226.018	-47.132	1.2890		1.2722	
97000	96551	226.079	-47.071	1.2773	• 1	1.2606	- 2
97200	96749	226.139	-47.011	1.2657		1.2492	
97400	96947	226.199	-46.951	1.2542		1.2378	
97600	97145	226.260	-46.890	1.2428		1.2266	
97800	97343	226.320	-46.830	1.2316		1.2155	
98000	97542	226.381	-46.769	1.2204		1.2044	
98200	97740	226.441	-46.709	1.2093		1.1935	
98400	97938	226.501	-46.649	1.1984		1.1827	
98600	98136	226.562	-46.588	1.1875		1.1720	
98800	98334	226.622	-46.528	1.1767		1.1613	
99000	98532	226.682	-46.468	1.1661	• 1	1.1508	- 2
99200	98730	226.743	-46.407	1.1555		1.1404	
99400	98928	226.803	-46.347	1.1450		1.1301	
99600	99127	226.864	-46.286	1.1347		1.1198	
99800	99325	226.924	-46.226	1.1244		1.1097	
100000	99523	226.984	-46.166	1.1142		1.0997	
100200	99721	227.045	-46.105	1.1041		1.0897	
100400	99919	227.105	-46.045	1.0942		1.0799	
100600	100117	227.166	-45.984	1.0843		1.0701	
100800	100315	227.226	-45.924	1.0745		1.0604	
101000	100513	227.286	-45.864	1.0648	• 1	1.0508	- 2
101200	100711	227.347	-45.803	1.0551		1.0413	
101400	100909	227.407	-45.743	1.0456		1.0319	
101600	101107	227.467	-45.683	1.0362		1.0226	
101800	101305	227.528	-45.622	1.0268		1.0134	
102000	101504	227.588	-45.562	1.0176		1.0043	
102200	101702	227.649	-45.501	1.0084		0.9954	- 3
102400	101900	227.709	-45.441	9.9934	• 0	9.9827	
102600	102098	227.769	-45.381	9.9833		9.9738	
102800	102296	227.830	-45.320	9.9740		9.9657	
103000	102494	227.890	-45.260	9.9650	• 0	9.9584	- 3
103200	102692	227.950	-45.200	9.9560		9.9512	
103400	102890	228.011	-45.139	9.9472		9.9463	
103600	103088	228.071	-45.079	9.9384		9.9375	
103800	103286	228.131	-45.019	9.9297		9.9288	
104000	103484	228.192	-44.958	9.9211		9.9202	
104200	103682	228.252	-44.898	9.9125		9.9116	
104400	103880	228.312	-44.838	9.9040		9.9031	
104600	104078	228.373	-44.777	9.8954		9.8945	
104800	104276	228.433	-44.717	9.8869		9.8860	
105000	104474	228.494	-44.656	9.8784	• 0	9.8791	- 3
105200	104672	228.554	-44.596	9.8699		9.8706	
105400	104870	228.614	-44.536	9.8614		9.8621	
105600	105068	228.675	-44.475	9.8529		9.8536	
105800	105266	228.735	-44.415	9.8444		9.8451	
106000	105464	228.795	-44.355	9.8359		9.8366	
106200	105662	228.855	-44.295	9.8274		9.8281	
106400	105860	228.915	-44.235	9.8189		9.8196	
106600	106058	228.975	-44.175	9.8104		9.8111	
106800	106256	229.035	-44.115	9.8019		9.8026	
107000	106454	229.095	-44.055	9.7934		9.7941	
107200	106652	229.155	-43.995	9.7849		9.7856	
107400	106850	229.215	-43.935	9.7764		9.7771	
107600	107048	229.275	-43.875	9.7679		9.7686	
107800	107246	229.335	-43.815	9.7594		9.7601	
108000	107444	229.395	-43.755	9.7509		9.7516	
108200	107642	229.455	-43.695	9.7424		9.7431	
108400	107840	229.515	-43.635	9.7339		9.7346	
108600	108038	229.575	-43.575	9.7254		9.7261	
108800	108236	229.635	-43.515	9.7169		9.7176	
109000	108434	229.695	-43.455	9.7084		9.7091	
109200	108632	229.755	-43.395	9.6999		9.7006	
109400	108830	229.815	-43.335	9.6914		9.6921	
109600	109028	229.875	-43.275	9.6829		9.6836	
109800	109226	229.935	-43.215	9.6744		9.6751	
110000	109424	229.995	-43.155	9.6659		9.6666	

Table 2-2: Geopotential Altitude, English Altitudes

L-7170-AKT-87-046

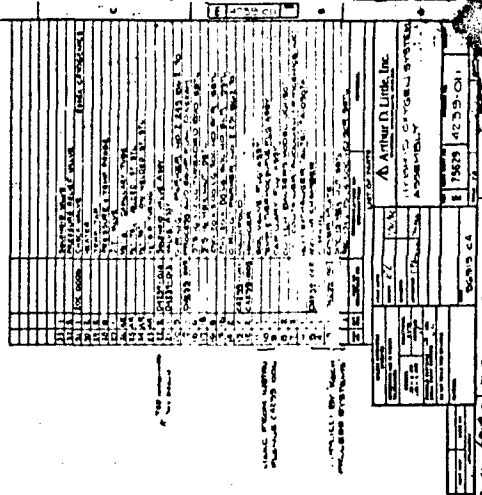
Altitude		Temperature		Pressure		Density	
H (ft)	Z (ft)	T (K)	t (°C)	P (mb)	P/P ₀	ρ (kg/m ³)	ρ/ρ_0
110000	110583	232.929	-40.221	6.9230	• 0	6.8325	- 3
110500	111089	233.355	-39.795	6.7701		6.6816	1.0107
111000	111594	233.782	-39.368	6.6209		6.5343	9.8661
111500	112099	234.209	-38.941	6.4752		6.3905	9.6314
112000	112605	234.635	-38.515	6.3329		6.2501	9.4027
112500	113110	235.062	-38.088	6.1941		6.1131	9.1799
113000	113616	235.489	-37.661	6.0585		5.9793	8.9627
113500	114121	235.916	-37.234	5.9262		5.8487	8.7510
114000	114627	236.342	-36.808	5.7969		5.7211	8.5447
114500	115132	236.769	-36.381	5.6707		5.5966	8.3436
115000	115638	237.196	-35.954	5.5475	• 0	5.4749	- 3
115500	116143	237.623	-35.527	5.4272		5.3562	7.9566
116000	116649	238.049	-35.101	5.3096		5.2402	7.7704
116500	117154	238.476	-34.674	5.1949		5.1269	7.5888
117000	117660	238.903	-34.247	5.0828		5.0163	7.4118
117500	118166	239.329	-33.821	4.9733		4.9083	7.2392
118000	118671	239.756	-33.394	4.8664		4.8027	7.0710
118500	119177	240.183	-32.967	4.7619		4.6997	6.9069
119000	119683	240.610	-32.540	4.6599		4.5990	6.7469
119500	120189	241.036	-32.114	4.5602		4.5006	6.5910
120000	120695	241.463	-31.687	4.4629	• 0	4.4045	- 3
120500	121200	241.890	-31.260	4.3678		4.3106	6.2905
121000	121706	242.316	-30.834	4.2748		4.2189	6.1458
121500	122212	242.743	-30.407	4.1840		4.1293	6.0047
122000	122718	243.170	-29.980	4.0953		4.0418	5.8671
122500	123224	243.597	-29.553	4.0087		3.9562	5.7329
123000	123730	244.023	-29.127	3.9240		3.8726	5.6019
123500	124236	244.450	-28.700	3.8412		3.7910	5.4742
124000	124742	244.877	-28.273	3.7603		3.7111	5.3496
124500	125248	245.303	-27.847	3.6813		3.6331	5.2280
125000	125754	245.730	-27.420	3.6040	• 0	3.5569	- 3
125500	126260	246.157	-26.993	3.5285		3.4824	4.9937
126000	126766	246.584	-26.566	3.4547		3.4096	4.8809
126500	127272	247.010	-26.140	3.3826		3.3384	4.7707
127000	127778	247.437	-25.713	3.3121		3.2688	4.6632
127500	128284	247.864	-25.286	3.2432		3.2008	4.5584
128000	128790	248.291	-24.859	3.1759		3.1343	4.4560
128500	129297	248.717	-24.433	3.1100		3.0694	4.3562
129000	129803	249.144	-24.006	3.0456		3.0058	4.2587
129500	130309	249.571	-23.579	2.9827		2.9437	4.1636
130000	130815	249.997	-23.153	2.9212	• 0	2.8830	- 3
130500	131322	250.424	-22.726	2.8610		2.8236	3.9801
131000	131828	250.851	-22.299	2.8022		2.7656	3.8916
131500	132334	251.278	-21.872	2.7447		2.7088	3.8053
132000	132841	251.704	-21.446	2.6885		2.6533	3.7210
132500	133347	252.131	-21.019	2.6335		2.5991	3.6387
133000	133854	252.558	-20.592	2.5797		2.5459	3.5584
133500	134360	252.984	-20.166	2.5271		2.4940	3.4800
134000	134867	253.411	-19.739	2.4757		2.4433	3.4034
134500	135373	253.839	-19.312	2.4254		2.3936	3.3286
135000	135880	254.265	-18.885	2.3762	• 0	2.3451	- 3
135500	136386	254.691	-18.459	2.3280		2.2976	3.1844
136000	136893	255.118	-18.032	2.2810		2.2511	3.1148
136500	137399	255.545	-17.605	2.2349		2.2057	3.0468
137000	137906	255.971	-17.179	2.1899		2.1612	2.9804
137500	138413	256.398	-16.752	2.1458		2.1178	2.9156
138000	138919	256.825	-16.325	2.1027		2.0752	2.8523
138500	139426	257.252	-15.898	2.0606		2.0336	2.7905
139000	139933	257.678	-15.471	2.0193		1.9929	2.7301
139500	140439	258.105	-15.045	1.9790		1.9531	2.6711
140000	140946	258.532	-14.618	1.9395	• 0	1.9141	- 3
140500	141453	258.959	-14.191	1.9008		1.8760	2.5572
141000	141960	259.385	-13.765	1.8630		1.8387	2.502
141500	142467	259.812	-13.338	1.8200		1.8021	2.4485
142000	142974	260.239	-12.911	1.7798		1.7664	2.3960
142500	143480	260.665	-12.485	1.7444		1.7315	2.3448
143000	143987	261.092	-12.058	1.7197		1.6972	2.2947
143500	144494	261.519	-11.631	1.6858		1.6638	2.2457
144000	145001	261.946	-11.204	1.6526		1.6310	2.1979
144500	145508	262.372	-10.778	1.6201		1.5989	2.1512
145000	146015	262.799	-10.351	1.5883	• 0	1.5675	- 3
145500	146522	263.226	-9.924	1.5572		1.5368	2.0609
146000	147029	263.652	-9.498	1.5267		1.5067	2.0173
146500	147536	264.079	-9.071	1.4969		1.4773	1.9747
147000	148044	264.506	-8.644	1.4677		1.4485	1.9331
147500	148551	264.933	-8.217	1.4391		1.4203	1.8924
148000	149058	265.359	-7.791	1.4111		1.3926	1.8526
148500	149565	265.786	-7.364	1.3837		1.3656	1.8137
149000	150072	266.213	-6.937	1.3569		1.3391	1.7757
149500	150579	266.639	-6.511	1.3306		1.3132	1.7385
150000	151086	267.066	-6.084	1.3044		1.2876	1.7022
150500	151593	267.493	-5.657	1.2792		1.2632	1.6669
151000	152100	267.920	-5.230	1.2540		1.2386	1.6326
151500	152607	268.347	-4.803	1.2292		1.2143	1.5992
152000	153114	268.774	-4.376	1.2044		1.1899	1.5667
152500	153621	269.201	-3.949	1.1800		1.1664	1.5350
153000	154128	269.628	-3.522	1.1560		1.1433	1.5041
153500	154635	270.055	-3.095	1.1320		1.1203	1.4740
154000	155142	270.482	-2.668	1.1080		1.0969	1.4447
154500	155649	270.909	-2.241	1.0840		1.0736	1.4161
155000	156156	271.336	-1.814	1.0600		1.0502	1.3882
155500	156663	271.763	-1.387	1.0360		1.0267	1.3610
156000	157170	272.190	-0.960	1.0120		1.0033	1.3346
156500	157677	272.617	-0.533	0.9880		0.9797	1.3089
157000	158184	273.044	-0.106	0.9640		0.9563	1.2840
157500	158691	273.471	0.321	0.9400		0.9328	1.2597
158000	159198	273.898	0.748	0.9160		0.9093	1.2360
158500	159705	274.325	1.175	0.8920		0.8858	1.2128
159000	160212	274.752	1.602	0.8680		0.8623	1.1901
159500	160719	275.179	2.029	0.8440		0.8388	1.1679
160000	161226	275.606	2.456	0.8200		0.8153	1.1462

REFERENCES

1. Little, Arthur D., Inc., "Hybrid Oxygen Systems Analysis", Phase I Report, Contract Number F33615-86-C-4504, May 1987.
2. Eng, E. G., A. Gupta, A. J. P. Lloyd, J. K. Robinson, "Tactical Life Support System", Final Report for August 1983 - April 1987, D449-10021, Boeing Military Airplane Company, May 1987.
3. Garrett AiResearch, "Performance data, Environmental Control System Performance Characteristics for McAir F-15 Aircraft", Volume I of II, 12 April 1973.
4. Letton, George C., "Survey of Fighter Aircraft Environmental Control Systems", Aeronautical Systems Division, Wright-Patterson AFB, presented at SAE AC-9 Committee Meeting, San Francisco, CA., 6 May 1976.
5. U. S. Standard Atmosphere, 1976, National Oceanic and Atmospheric Administration, National Aeronautics and Space Administration, United States Air Force, NOAA-S/T 76-1562, Washington, D. C., October 1976.

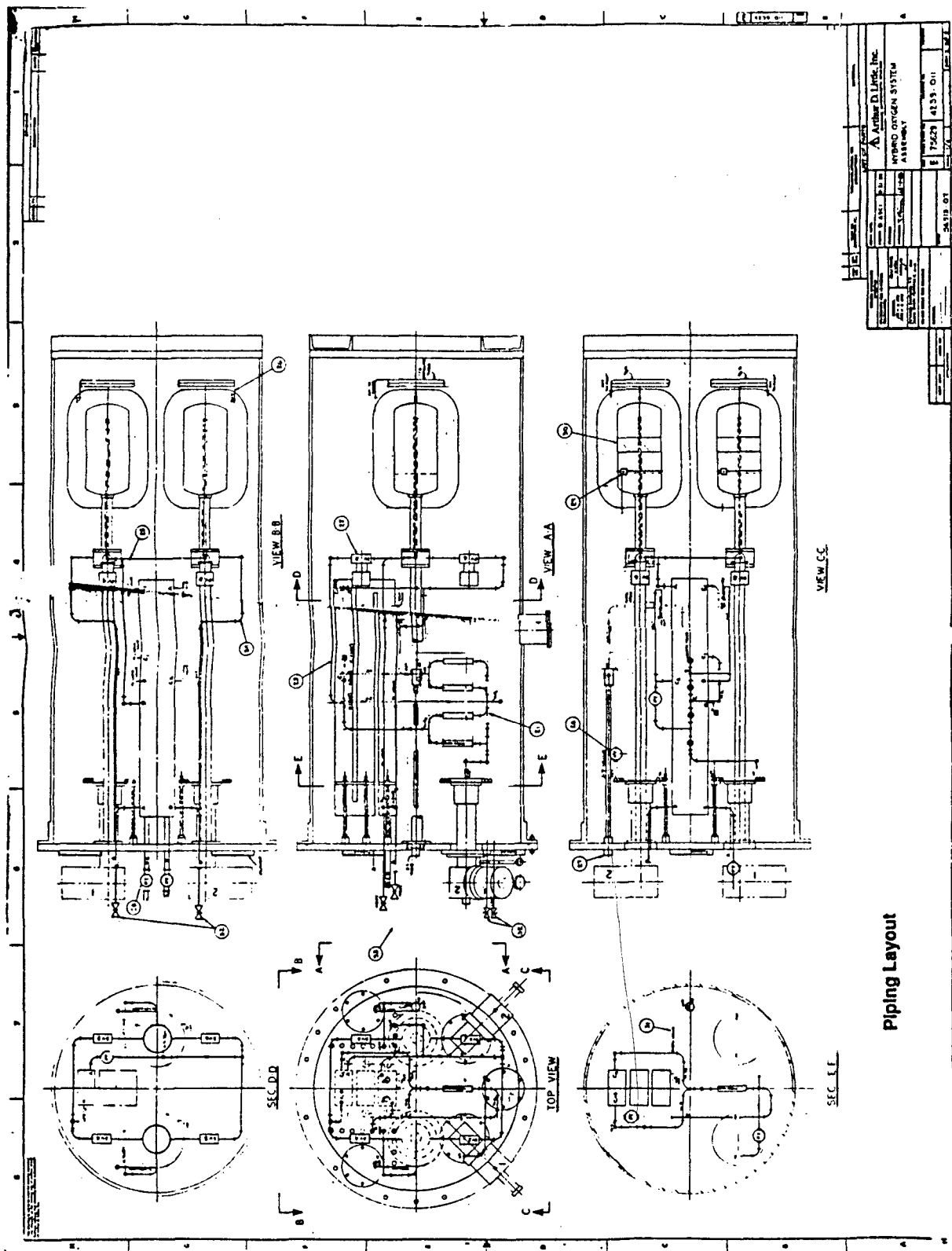
7.3 Engineering Drawings

7.3.1 Mechanical and Piping Layout

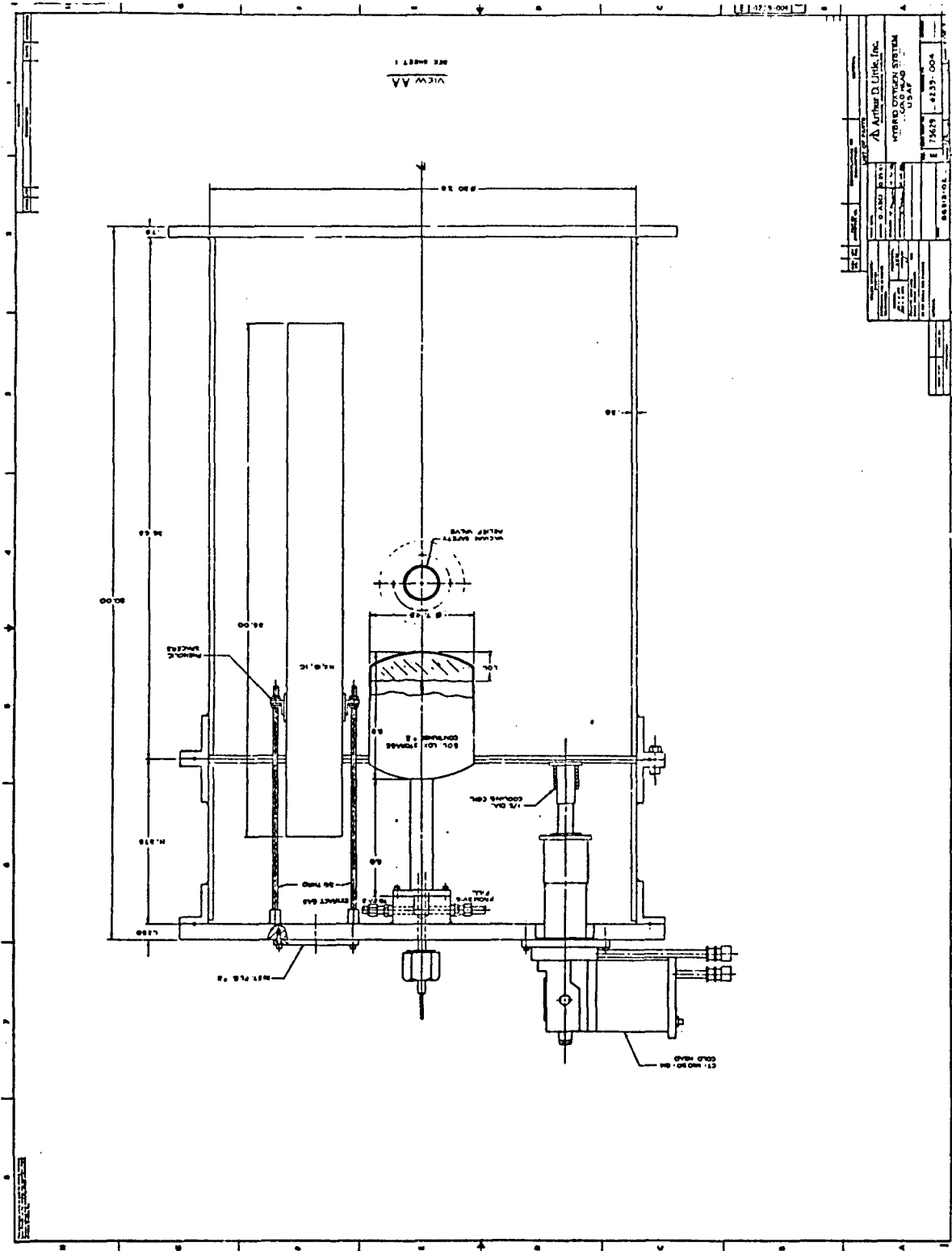


Mechanical Layout

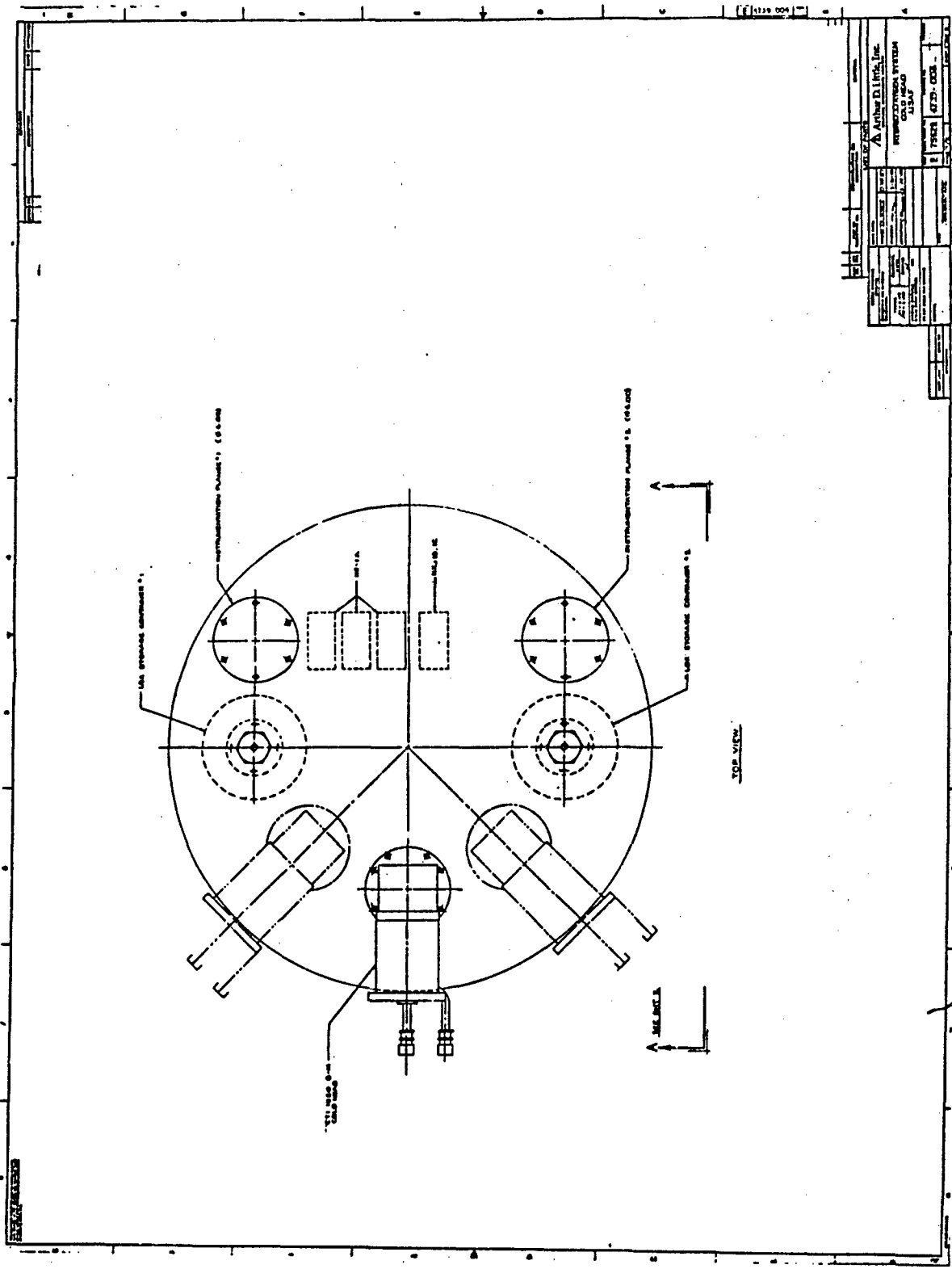
SEE SHIT. 2 FOR PRINTING ASSY



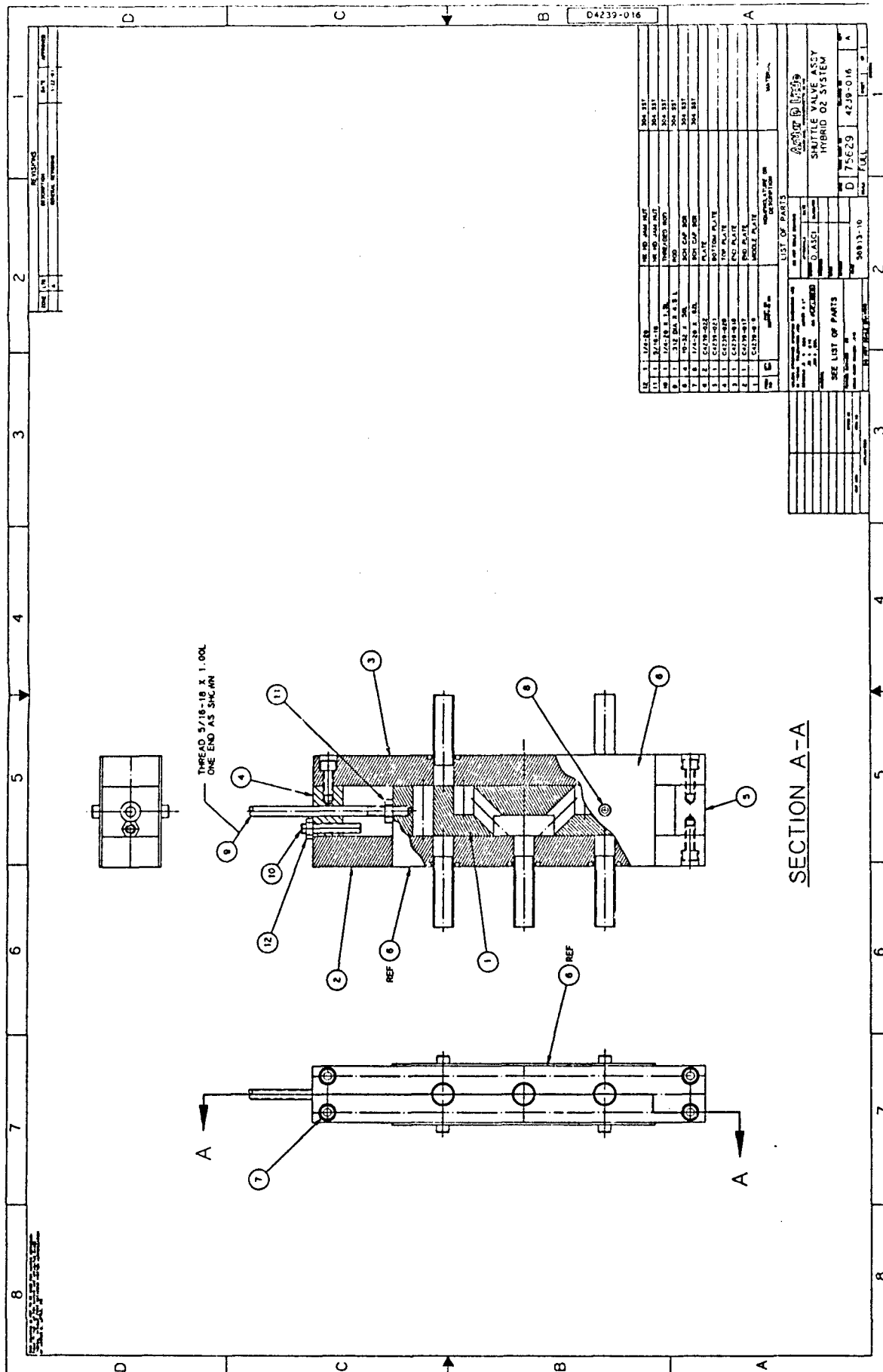
7.3.2 Mechanical Drawings



Arthur D. Little, Inc. HYBRID SYSTEM 1000 WASHINGTON ST. CAMBRIDGE, MASS. 02142	
PROJECT NO. 75629-4239-004	DRAWING NO. 75629-4239-004
DATE 10/1/68	BY J. D. LITTLE
CHECKED BY J. D. LITTLE	APPROVED BY J. D. LITTLE



ARTHUR D. LITTLE, Inc. 100 STATE STREET CAMBRIDGE, MASSACHUSETTS 02142	
PROJECT NO. 4220-001 DRAWING NO. 4220-001-001	SHEET NO. 1 OF 1

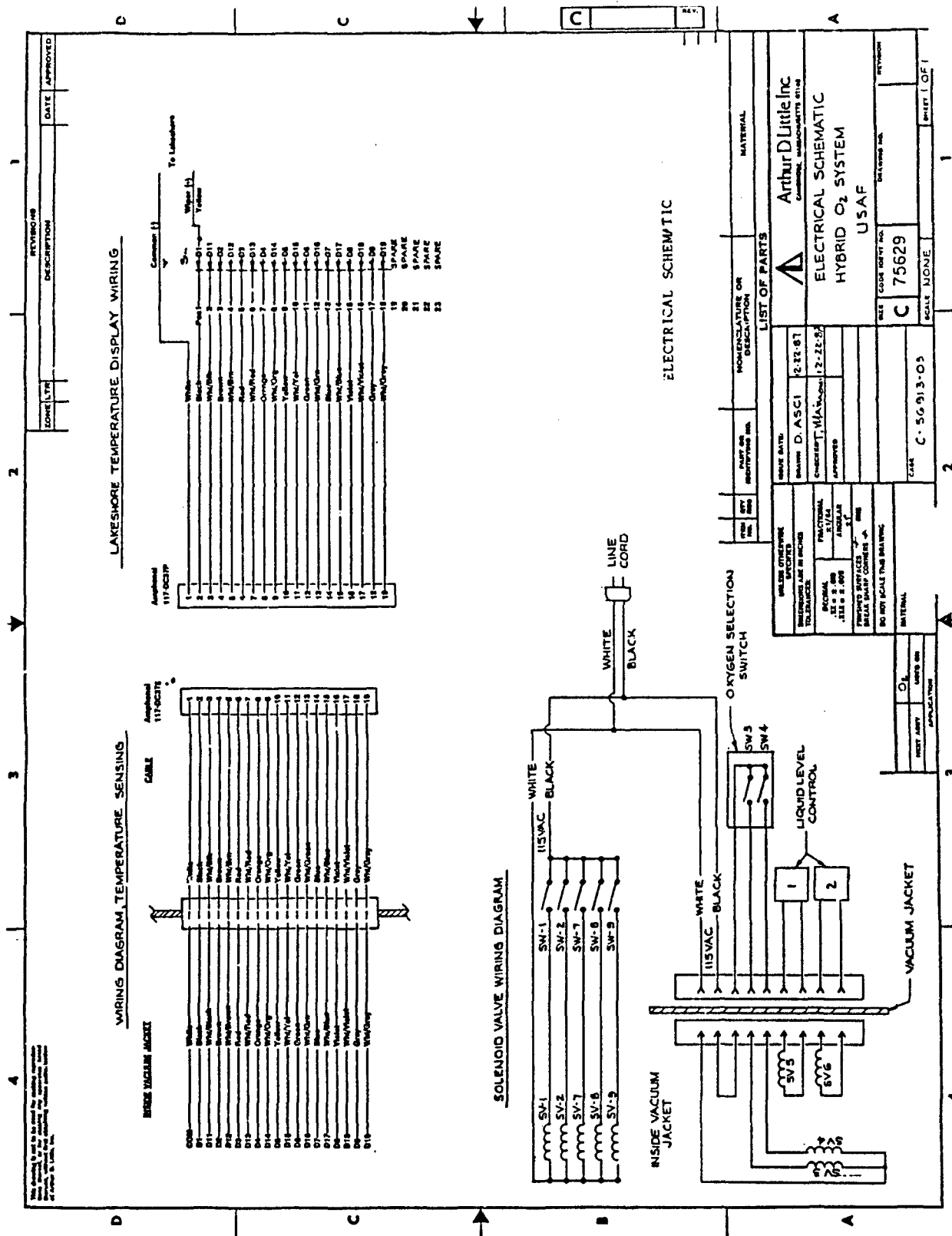


D4239-016

QTY	DESCRIPTION	QTY	DESCRIPTION
12	1. 1/4-20 X 1.00L	10	10. 1/4-20 X 1.00L
11	1. 1/4-20 X 1.00L	10	10. 1/4-20 X 1.00L
10	1. 1/4-20 X 1.00L	10	10. 1/4-20 X 1.00L
9	1. 1/4-20 X 1.00L	10	10. 1/4-20 X 1.00L
8	1. 1/4-20 X 1.00L	10	10. 1/4-20 X 1.00L
7	1. 1/4-20 X 1.00L	10	10. 1/4-20 X 1.00L
6	1. 1/4-20 X 1.00L	10	10. 1/4-20 X 1.00L
5	1. 1/4-20 X 1.00L	10	10. 1/4-20 X 1.00L
4	1. 1/4-20 X 1.00L	10	10. 1/4-20 X 1.00L
3	1. 1/4-20 X 1.00L	10	10. 1/4-20 X 1.00L
2	1. 1/4-20 X 1.00L	10	10. 1/4-20 X 1.00L
1	1. 1/4-20 X 1.00L	10	10. 1/4-20 X 1.00L

LIST OF PARTS	
QTY	DESCRIPTION
1	1. 1/4-20 X 1.00L
2	2. 1/4-20 X 1.00L
3	3. 1/4-20 X 1.00L
4	4. 1/4-20 X 1.00L
5	5. 1/4-20 X 1.00L
6	6. 1/4-20 X 1.00L
7	7. 1/4-20 X 1.00L
8	8. 1/4-20 X 1.00L
9	9. 1/4-20 X 1.00L
10	10. 1/4-20 X 1.00L
11	11. 1/4-20 X 1.00L
12	12. 1/4-20 X 1.00L

7.3.3 Electrical Drawings



7.4 Essex Cryogenics Concentrator Performance

60004 = 5.08 ATM

Essex Cryogenics Molecular Sieve
Unit Performance

60C 0002-2

OBOES

	ALT		AIR IN		O ₂ ENRICHED AIR OUT		%	%	%	AIRS
	A/C	CABIN	PRES. (PSIG)	FLOW (LPM)	PRES. (PSIG)	FLOW (LPM)				
10	0	0	10	192	3.0	10	70.7	3.27	26.0	0.43
20	0	0	10	204	2.7	25	37.2	1.72	60.7	0.43
30	0	0	10	204	2.3	40	29.8	1.36	68.8	0.43
40	0	0	10	204	2.0	50	27.6	1.25	71.0	0.43
50	0	0	10	204	2.0	52	27.1	1.23	71.6	0.43
60	0	0	20	396	6.9	10	94.5	5.11	0.4	0.50
70	0	0	20	396	6.6	25	84.3	3.92	11.7	0.50
80	0	0	20	396	6.4	40	58.9	2.73	38.3	0.50
90	0	0	20	396	6.1	50	50.3	2.33	47.0	0.50
100	0	0	20	403	5.9	60	44.5	2.06	53.3	0.50
110	0	0	20	408	5.5	75	39.2	1.81	58.7	0.50
120	0	0	20	406	5.0	91	35.3	1.63	63.0	0.50
130	0	0	30	589	10.9	10	93.5	5.94	2.5	0.57
140	0	0	30	589	10.7	25	95.0	4.64	0.3	0.57
150	0	0	30	589	10.6	40	87.5	4.08	8.2	0.57
160	0	0	30	589	10.3	50	75.4	3.50	20.9	0.57
170	0	0	30	589	10.0	60	65.8	3.05	31.2	0.57
180	0	0	30	589	9.6	75	56.2	2.63	40.8	0.57
190	0	0	30	600	8.9	100	46.3	2.20	51.3	0.57
200	0	0	40	770	15.6	10	93.5	6.10	0.3	0.67
210	0	0	40	775	15.2	25	94.7	4.99	0.3	0.67
220	0	0	40	781	14.9	40	95.0	4.59	0.3	0.67
230	0	0	40	781	14.7	50	91.1	4.34	4.3	0.67
240	0	0	40	781	14.6	60	86.4	4.03	9.4	0.67
250	0	0	40	781	14.1	75	73.4	3.40	23.1	0.67
260	0	0	40	792	13.4	100	59.1	2.75	38.0	0.67
270	0	0	60	883	18.7	10	93.8	5.83	0.3	0.70
280	0	0	60	889	18.4	25	94.7	5.00	0.3	0.70
290	0	0	60	900	18.2	40	95.0	4.62	0.4	0.70
300	0	0	60	883	18.0	50	94.1	4.47	1.3	0.70
310	0	0	60	894	17.8	60	90.2	4.24	5.5	0.70
320	0	0	60	894	17.4	75	80.0	3.70	16.3	0.70
330	0	0	60	883	16.8	100	65.0	3.00	32.0	0.70
340	10.0	8.0	10	197	3.2	10	85.1	4.02	11.1	0.43
350	10.0	8.0	10	192	2.8	25	51.3	2.35	46.5	0.43
360	10.0	8.0	10	177	2.5	40	38.3	1.74	59.8	0.43
370	10.0	8.0	10	199	2.4	50	34.4	1.55	63.9	0.43
380	10.0	8.0	10	199	2.2	54	32.7	1.47	65.6	0.43
390	10.0	8.0	20	389	7.3	10	94.2	5.40	0.3	0.47
400	10.0	8.0	20	389	7.1	25	93.9	5.51	1.5	0.47
410	10.0	8.0	20	389	7.0	40	81.8	5.79	14.4	0.47
420	10.0	8.0	20	389	6.7	50	67.8	5.12	28.9	0.47
430	10.0	8.0	20	389	6.4	60	59.6	4.73	37.5	0.47
440	10.0	8.0	20	392	6.2	75	50.5	4.32	46.5	0.47
450	10.0	8.0	20	396	5.5	100	41.5	3.92	56.2	0.47
460	10.0	8.0	30	566	11.6	10	93.7	5.08	0.2	0.53
470	10.0	8.0	30	573	11.4	25	94.8	4.96	0.2	0.53
480	10.0	8.0	30	577	11.2	40	95.1	4.57	0.3	0.53
490	10.0	8.0	30	575	11.0	50	93.6	4.38	1.9	0.53
500	10.0	8.0	30	573	10.8	60	86.8	4.02	9.2	0.53
510	10.0	8.0	30	568	10.6	75	74.0	3.42	22.3	0.53
520	10.0	8.0	30	571	10.1	100	59.5	2.75	37.2	0.53
530	10.0	8.0	40	724	16.2	10	93.4	5.33	0.2	0.60
540	10.0	8.0	40	736	15.9	25	94.6	5.17	0.2	0.60
550	10.0	8.0	40	736	15.6	40	95.0	4.76	0.2	0.60
560	10.0	8.0	40	733	15.4	50	93.1	4.59	0.1	0.60
570	10.0	8.0	40	733	15.2	60	92.1	4.47	0.5	0.60
580	10.0	8.0	40	736	15.0	75	81.6	4.16	5.8	0.60
590	10.0	8.0	40	731	14.6	100	74.1	3.92	22.5	0.60
600	10.0	8.0	60	838	19.4	10	93.6	5.19	0.2	0.65
610	10.0	8.0	60	833	19.2	25	94.6	4.76	0.2	0.65
620	10.0	8.0	60	842	19.1	40	95.0	4.78	0.2	0.65

Essex Cryogenics Molecular Sieve
Unit Performance (con. inued)

	Mo	Calor	Area ft.	Dist. in ft.	Product press.	Flow rate lpm	% O ₂	% A	% H ₂	Temp
630	10.0	8.0	60	838	19.0	50	94.9	4.63	0.42	0.65
640	10.0	8.0	60	849	18.6	60	94.5	4.51	0.9	0.65
650	10.0	8.0	60	842	18.4	75	91.7	4.28	3.9	0.65
660	10.0	8.0	60	826	18.0	100	79.4	3.68	16.9	0.65
670	20.0	8.0	20	363	7.3	10	94.3	5.50	0.2	0.45
680	20.0	8.0	20	365	7.0	25	94.8	4.55	0.6	0.45
690	20.0	8.0	20	365	6.7	40	83.5	3.90	12.2	0.45
700	20.0	8.0	20	367	6.5	50	71.5	3.29	25.3	0.45
710	20.0	8.0	20	369	6.2	60	60.6	2.83	36.5	0.45
720	20.0	8.0	20	369	6.2	63	59.7	2.76	37.5	0.45
730	20.0	8.0	30	532	12.0	10	93.6	6.17	0.2	0.51
740	20.0	8.0	30	543	11.7	25	94.7	5.03	0.2	0.51
750	20.0	8.0	30	548	11.4	40	95.1	4.61	0.2	0.51
760	20.0	8.0	30	548	11.1	50	95.0	4.46	0.5	0.51
770	20.0	8.0	30	548	10.9	60	92.3	4.27	3.8	0.51
780	20.0	8.0	30	550	10.6	75	78.9	3.65	17.3	0.51
790	20.0	8.0	30	552	10.1	100	63.4	2.94	33.4	0.51
800	20.0	8.0	40	668	17.1	10	93.7	6.03	0.2	0.61
810	20.0	8.0	40	670	16.9	25	94.7	5.08	0.2	0.61
820	20.0	8.0	40	668	16.6	40	95.0	4.68	0.3	0.61
830	20.0	8.0	40	677	16.5	50	94.9	4.52	0.5	0.61
840	20.0	8.0	40	679	16.3	60	93.9	4.38	1.7	0.61
850	20.0	8.0	40	684	15.8	75	88.3	4.10	7.6	0.61
860	20.0	8.0	40	691	15.2	100	74.6	3.42	21.7	0.61
870	20.0	8.0	60	781	20.4	10	93.7	6.02	0.2	0.64
880	20.0	8.0	60	792	20.0	25	94.6	5.16	0.2	0.64
890	20.0	8.0	60	804	19.7	40	94.9	4.78	0.2	0.64
900	20.0	8.0	60	804	19.6	50	95.0	4.64	0.3	0.64
910	20.0	8.0	60	804	19.5	60	94.6	4.52	0.8	0.64
920	20.0	8.0	60	797	19.0	75	91.9	4.20	3.9	0.64
930	20.0	8.0	60	797	18.6	100	79.4	3.68	16.0	0.64
940	20.0	20.0	10	181	3.3	10	94.9	4.80	0.3	0.41
950	20.0	20.0	10	181	3.2	25	79.6	3.75	16.4	0.41
960	20.0	20.0	10	181	2.9	40	55.5	2.55	41.5	0.41
970	20.0	20.0	10	181	2.8	50	48.9	2.20	49.0	0.41
980	20.0	20.0	10	183	2.6	60	42.5	2.00	55.0	0.41
990	20.0	20.0	10	183	2.4	75	38.0	1.76	59.8	0.41
1000	20.0	20.0	10	183	2.1	86	35.3	1.70	62.9	0.41
1010	20.0	20.0	20	357	7.7	10	93.2	6.04	0.7	0.46
1020	20.0	20.0	20	357	7.5	25	94.6	4.92	0.5	0.46
1030	20.0	20.0	20	357	7.3	40	93.9	4.51	1.4	0.46
1040	20.0	20.0	20	357	7.2	50	92.4	4.30	2.9	0.46
1050	20.0	20.0	20	357	7.1	60	88.3	4.10	7.2	0.46
1060	20.0	20.0	20	360	6.9	75	75.1	3.50	19.9	0.46
1070	20.0	20.0	20	362	6.4	100	60.5	2.83	36.5	0.46
1080	20.0	20.0	30	537	12.4	10	93.0	6.82	0.2	0.52
1090	20.0	20.0	30	532	12.3	25	94.3	5.49	0.2	0.52
1100	20.0	20.0	30	534	12.1	40	94.8	4.98	0.2	0.52
1110	20.0	20.0	30	537	12.0	50	95.0	4.77	0.2	0.52
1120	20.0	20.0	30	534	11.9	60	95.0	4.63	0.2	0.52
1130	20.0	20.0	30	532	11.8	75	94.3	4.45	1.1	0.52
1140	20.0	20.0	30	532	11.5	100	87.3	4.04	8.5	0.52
1150	20.0	20.0	40	688	17.2	10	92.9	6.95	0.2	0.57
1160	20.0	20.0	40	679	17.2	25	94.1	5.70	0.2	0.57
1170	20.0	20.0	40	684	17.0	40	94.6	5.20	0.1	0.57
1180	20.0	20.0	40	688	16.8	50	94.8	4.96	0.2	0.57
1190	20.0	20.0	40	688	16.7	60	95.0	4.81	0.2	0.57
1200	20.0	20.0	40	691	16.6	75	95.1	4.64	0.2	0.57
1210	20.0	20.0	40	691	16.4	100	94.9	4.45	0.5	0.57
1220	20.0	20.0	50	781	20.8	10	93.6	6.15	0.12	0.62
1230	20.0	20.0	50	770	20.4	25	94.2	5.60	0.2	0.62
1240	20.0	20.0	50	777	20.3	40	94.7	5.16	0.1	0.62
1250	20.0	20.0	50	779	20.2	50	94.8	5.00	0.1	0.62

Essex Cryogenics Molecular Sieve
Unit Performance (continued)

9. The Pressure

1260	20.0	20.0	60	768	20.1	60	94.9	4.62	0.2	0.62
1270	20.0	20.0	60	792	20.0	75	94.7	4.66	0.1	0.62
1280	20.0	20.0	60	768	19.7	100	94.5	4.87	0.3	0.62
1290	30.0	11.9	20	328	7.9	10	93.5	5.48	0.2	0.43
1300	30.0	11.9	20	335	7.6	25	94.2	5.05	0.8	0.43
1310	30.0	11.9	20	335	7.3	40	90.9	4.45	4.5	0.43
1320	30.0	11.9	20	335	7.1	50	79.1	3.69	16.8	0.43
1330	30.0	11.9	20	335	6.8	61	67.0	3.05	29.8	0.43
1340	30.0	11.9	30	496	12.7	10	93.1	4.73	0.2	0.48
1350	30.0	11.9	30	503	12.3	25	94.2	5.54	0.2	0.48
1360	30.0	11.9	30	507	12.1	40	94.8	5.11	0.2	0.48
1370	30.0	11.9	30	509	12.0	50	94.6	4.97	0.4	0.48
1380	30.0	11.9	30	509	11.9	60	94.0	4.80	1.2	0.48
1390	30.0	11.9	30	507	11.5	75	84.2	3.89	11.8	0.48
1400	30.0	11.9	30	509	10.9	100	67.0	3.10	30.0	0.48
1410	30.0	11.9	40	661	17.7	10	93.0	6.85	0.2	0.56
1420	30.0	11.9	40	656	17.4	25	94.0	5.81	0.1	0.56
1430	30.0	11.9	40	666	17.2	40	94.5	5.31	0.2	0.56
1440	30.0	11.9	40	668	17.0	50	94.7	5.12	0.2	0.56
1450	30.0	11.9	40	668	16.8	60	94.7	5.00	0.3	0.56
1460	30.0	11.9	40	668	16.6	75	94.3	4.81	0.9	0.56
1470	30.0	11.9	40	668	16.1	100	83.3	3.80	12.9	0.56
1480	30.0	11.9	60	758	21.2	10	93.4	6.36	0.2	0.62
1490	30.0	11.9	60	756	20.8	25	94.0	5.76	0.1	0.62
1500	30.0	11.9	60	763	20.4	40	94.5	5.35	0.1	0.62
1510	30.0	11.9	60	765	20.2	50	94.6	5.20	0.2	0.62
1520	30.0	11.9	60	765	20.1	60	94.5	5.05	0.4	0.62
1530	30.0	11.9	60	770	20.0	75	93.1	4.75	2.1	0.62
1540	30.0	11.9	60	761	19.9	100	86.8	4.19	8.7	0.62
1550	30.0	30.0	10	149	3.4	10	94.2	5.25	0.3	0.40
1560	30.0	30.0	10	152	3.3	25	93.8	4.46	1.7	0.40
1570	30.0	30.0	10	152	3.3	40	78.4	3.60	18.3	0.40
1580	30.0	30.0	10	154	3.1	50	67.0	3.07	30.0	0.40
1590	30.0	30.0	10	154	2.9	60	58.5	2.75	38.7	0.40
1600	30.0	30.0	10	152	2.7	75	50.5	2.38	47.7	0.40
1610	30.0	30.0	10	161	2.4	100	42.5	1.98	55.2	0.40
1620	30.0	30.0	20	317	8.0	10	92.8	6.92	0.3	0.44
1630	30.0	30.0	20	319	7.9	25	94.3	5.41	0.2	0.44
1640	30.0	30.0	20	321	7.8	40	94.9	4.90	0.2	0.44
1650	30.0	30.0	20	326	7.7	50	95.0	4.72	0.2	0.44
1660	30.0	30.0	20	328	7.6	60	95.1	4.60	0.2	0.44
1670	30.0	30.0	20	324	7.4	75	94.9	4.46	0.5	0.44
1680	30.0	30.0	20	328	7.2	100	84.0	3.92	11.9	0.44
1690	30.0	30.0	30	475	13.0	10	92.2	7.62	0.2	0.52
1700	30.0	30.0	30	475	13.0	25	93.8	6.02	0.1	0.52
1710	30.0	30.0	30	475	12.9	40	94.4	5.40	0.1	0.52
1720	30.0	30.0	30	475	12.8	50	94.7	5.11	0.1	0.52
1730	30.0	30.0	30	475	12.7	60	94.9	4.93	0.1	0.52
1740	30.0	30.0	30	475	12.6	75	95.1	4.74	0.1	0.52
1750	30.0	30.0	30	488	12.4	100	95.3	4.57	0.1	0.52
1760	30.0	30.0	40	634	18.4	10	92.3	7.46	0.1	0.60
1770	30.0	30.0	40	634	18.3	25	93.5	6.39	0.1	0.60
1780	30.0	30.0	40	634	18.3	40	94.1	5.70	0.1	0.60
1790	30.0	30.0	40	634	18.3	50	94.4	5.40	0.1	0.60
1800	30.0	30.0	40	634	18.2	60	94.7	5.17	0.1	0.60
1810	30.0	30.0	40	634	18.1	75	94.9	4.96	0.1	0.60
1820	30.0	30.0	40	634	18.1	100	95.1	4.72	0.1	0.60
1830	30.0	30.0	60	720	21.6	10	93.1	6.78	0.1	0.64
1840	30.0	30.0	60	720	21.4	25	93.7	6.16	0.1	0.64
1850	30.0	30.0	60	713	21.2	40	94.2	5.66	0.1	0.64
1860	30.0	30.0	60	713	21.2	50	94.4	5.40	0.1	0.64
1870	30.0	30.0	60	715	21.2	60	94.6	5.20	0.1	0.64
1880	30.0	30.0	60	715	21.1	75	94.8	5.00	0.1	0.64

Press
(PSIG)

Flow
(LPM)

~ 14.85

Crack

7.5 Spreadsheet Test Data

Time Minutes	%O2 FM-6 supply	%O2 FM2 vent	%O2 FM1 withdraw	Q FM4	Phig R42	Q FM-6	Phig R45	Q FM2	Q FM1	Phig RA-FM1	1	2	Temperatures 3	4
01				61	40						282.6	286.0	288.6	288.9
6.7				60	40						273.0	284.0	286.0	285.0
13.1				60	40						271.1	277.3	280.3	281.1
19.3				59	40						267.0	275.5	287.8	276.8
23.8				60	40						266.8	274.1	286.0	275.6
31.5				61	40						269.0	273.4	286.0	273.4
36.5				61	40						264.0	272.0	285.0	273.0
41.4				60	40						257.5	269.7	286.7	271.0
47.2				61	40						258.0	266.0	286.0	286.0
54.8				54	41						261.0	262.0	283.0	286.0
71.6				56	41						249.7	241.0	285.0	282.0
76.3				56	42						244.0	234.7	283.5	282.0
82.0				53	43						239.4	228.8	282.9	216.5
93.3				52	42						228.8	217.2	281.5	204.9
97.3				53	44						225.4	213.8	281.0	201.3
103.3				53	43						228.0	208.2	283.0	196.0
123.5				54	42						199.5	188.5	284.0	178.6
132.7				53	43						192.7	181.2	275.5	171.3
138.5				53	43						188.2	176.6	274.5	167.0
144.3				52	42						183.8	172.0	273.5	162.5
152.6				49	42						177.1	165.1	271.9	156.4
155.7				55	43						175.6	163.7	271.4	153.6
159.5				54	42						173.3	161.3	270.8	151.5
164.3				56	42						170.0	157.6	269.9	148.6
169.3				55	41						165.7	153.3	268.6	144.8
177.9				49	42						160.0	148.5	267.0	139.8
183.0				50	44						157.7	145.5	266.2	136.9
183.5				50	44						156.8	144.5	265.6	134.9
190.3				50	43						153.2	140.7	264.7	132.3
194.4				50	42						151.0	138.3	264.0	130.1
200.4				50	42						146.2	134.1	262.7	126.7
210.1				54	50						143.8	131.5	261.9	124.1
214.3				55	50						140.1	127.7	260.8	121.4
219.9				55	50						136.5	125.1	259.9	119.4
226.6				50	51						132.6	121.8	258.7	116.6
231.5				50	52						129.1	118.4	257.3	113.1
237.7				50	52						127.5	116.3	256.4	110.5
244.7				49	50						123.9	112.6	255.2	107.2
246.0				48	50						120.0	108.5	254.2	103.1
256.0				46	51						114.9	103.3	253.6	96.0
262.9		NA	96.6	48	48	35.0	10.0	na	20.0	11.0	113.7	101.5	254.0	95.5
271.1		97.3	97.0	47	53	33.0	15.0		36.0	14.0	117.0	104.0	255.0	96.0
285.1	96.7	97.5	97.0	49	52	35.0	10.0		35.0	10.0	119.0	105.0	258.0	98.6
295.3	97.0	96.9	97.0	45	52	35.0	10.0		35.0	10.0	118.7	105.4	258.5	98.0
304.0	97.2	97.2	97.3	49	50	35.0	10.0		35.0	10.0	117.0	104.5	257.1	98.0
310.4	97.2		97.2	48	55	35	10				115.9	103.9	255.3	97.0
317.1	97.2		97.2	43	52	35	10				113.7	104.2	253.0	96.0
321.2	97.2	96.9		48	50	33	12	19			115.8	104.2	251.8	95.4
331.7	97.0	97.5		47	50	33	11	53			114.6	104.5	248.0	94.3
337.4	96.9	95.6		47	54	45	10	54			114.3	104.3	246.7	93.9
341.8	96.3	95.7		42	55	40	12	53			114.0	104.3	245.6	93.3
348.6	96.3	96.5		48	53	40	12	53			114.1	104.5	244.5	92.5
356.2	97.5	96.0		40	52	23	14	39			114.4	104.8	242.8	92.8

i-Dec-90

Spreadsheet Test Input and Results - O₂ Test No. 7 (continued)

Time Minutes		%O2 FM6 supply	%O2 FM2 vent	%O2 FM1 withdraw	Q FM4	P _{sig} R ₄₂	Q FM-6	P _{sig} R ₄₅	Q FM2	Q FM1	P _{sig} RA-FM1	1	2	Temperatures		3	4
361.6	rv1-8 down hv-60 open	96.5			46	52	38	12	53			114.3	104.8	241.6	92.6		
369.2	rv2-7 up	96.3	96.8		45	58	40	12	52			114.9	105.3	240.5	92.4		
374.1	rv1-8 down	97.5	96.3		38	55	25	16	33			115.2	105.6	239.6	92.3		
378.3	rv2-7 up	97.5	96.9		47	55	30	15	40			115.4	105.9	238.8	92.3		
384.7	rv1-8 down	96.9	96.1		48	55	35	15	34			115.2	105.7	237.4	92.7		
394.5	rv2-7 up	96.5	96.0		45	54	40	15	33			114.9	105.3	235.4	93.0		
400.8		97.4	96.6		38	58	33	16	34			114.9	105.3	233.7	92.6		
405.0	rv1-8 down	97.1	96.0		47	55	33	15	33			114.9	105.5	232.8	92.8		
412.5		97.2	96.3		40	57	33	15	33			115.1	105.4	231.4	92.5		
420.3	rv2-7 up	97.1	95.4		47	57	40	15	31			115.8	106.1	230.1	92.5		
426.1	hv-60 closed	97.0	95.5	96.7	45	57	35	13	34	12		116.7	106.2	227.0	94.0		
438.4			95.5	96.5	45	57	35	12	28	14		116.0	106.0	226.0	94.0		
446.0		97.0	96.0	95.5	43	57	40	14	35	20		117.0	106.2	225.2	93.0		
452.6		97.0	96.0	95.5	46.0	55.0	45.0	12.0	44.0	20.0		109.8	106.5	224.0	92.9		
459.4	rv1-8 down			95.5	42.0	58.0	35.0	15.0	26.0	17.0	9.0	118.0	107.0	223.3	93.2		
467.1				95.8						15.0							
468.6				97.0						18.0							
470.6				97.4						25.0							
474.2				97.9						28.0							
475.3																	
478.4				98.0						28.0							
480.3				98.0						28.0							
482.3				98.5						28.0							
482.3				98.8						29.0							
486.1				99.0						31.0							
487.4				99.5						33.0							
489.3				99.7						48.0							
491.7				100.0						39.0							
493.4				100.2						33.0							
493.6				100.0						30.0							
496.2				100.4						27.0							
498.1				100.0						14.0							
498.2				100.0						11.0							

Spreadsheet Test Input and Results - O₂ Test No. 7 (continued)

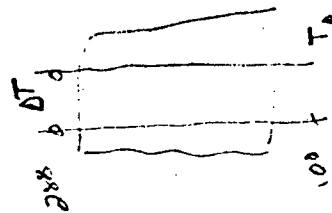
O₂ test 427

Time Minutes	6	8	9	P3	P7	L1	L2	Drygas No.	Vaporized Liquid to cool down			PM4 Gr/min	PM6 Gr/min	Liquors Dewar	Liquors L1	Liquors L2	Liquors Liq withdrawn	Liquors Liquors withdrawn
									PM4 SCFM	PM4 Gr/min	PM6 Gr/min							
0.1	258.0	258.0	258.0	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
6.7	259.0	259.0	259.0	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
13.1	271.5	271.5	271.5	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
19.3	276.7	276.7	276.7	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
23.8	277.5	277.5	277.5	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
31.5	277.3	277.3	277.3	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
36.5	277.7	277.7	277.7	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
41.4	271.7	271.7	271.7	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
47.2	268.0	268.0	268.0	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
54.8	252.0	252.0	252.0	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
71.6	225.0	225.0	225.0	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
76.3	219.6	219.6	219.6	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
82.0	213.2	213.2	213.2	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
93.3	201.8	201.8	201.8	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
97.3	198.2	198.2	198.2	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
103.3	191.3	191.3	191.3	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
123.5	176.1	176.1	176.1	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
132.7	168.7	168.7	168.7	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
138.5	164.3	164.3	164.3	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
144.3	159.9	159.9	159.9	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
152.6	154.1	154.1	154.1	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
155.7	150.1	150.1	150.1	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
159.5	149.0	149.0	149.0	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
164.3	146.1	146.1	146.1	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
169.3	142.5	142.5	142.5	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
177.9	137.4	137.4	137.4	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
183.0	134.4	134.4	134.4	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
185.5	132.1	132.1	132.1	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
190.3	129.9	129.9	129.9	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
194.4	127.5	127.5	127.5	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
200.4	124.0	124.0	124.0	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
204.9	121.8	121.8	121.8	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
210.1	119.5	119.5	119.5	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
214.3	117.8	117.8	117.8	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
219.9	115.3	115.3	115.3	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
226.6	111.8	111.8	111.8	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
231.5	109.2	109.2	109.2	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
237.7	105.4	105.4	105.4	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
244.7	102.6	102.6	102.6	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
256.0	93.1	93.1	93.1	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
262.9	91.9	91.9	91.9	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
271.1	88.8	88.8	88.8	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
285.1	99.0	99.0	99.0	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
295.3	98.0	98.0	98.0	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
304.0	98.1	98.1	98.1	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
310.4	95.9	95.9	95.9	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
317.1	95.6	95.6	95.6	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
321.2	93.9	93.9	93.9	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
331.7	94.3	94.3	94.3	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
337.4	93.8	93.8	93.8	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
341.8	93.6	93.6	93.6	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
346.6	94.0	94.0	94.0	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					
356.2	93.3	93.3	93.3	22.0	29.8			0.0	0.0	0.0	0.0	0.0	0.0					

5-Dec-90

Spreadsheet Test Input and Results - O₂ Test No. 7 (continued)

Time Minutes	6	8	9	P5	P7	L1	L2	Dryan No.	FM4 SCPM	FM4 -FM1 Cfm/min	FM2 Cfm/min	FM6 Cfm/min	Liters Dewar1	1.00 Liters Dewar2	Liters L1	Liters L2	Liters Liq withdrawn	L-STP withdraw
361.6	95.6	98.1	35.0	14.0	0.9	0.9	0.9	40.7	0.0	0.0	99.1	0.0	0.6	0.4	0.0	0.0	0.0	0.0
369.2	95.3	97.8	45.0	18.0	0.9	0.9	0.9	34.9	0.0	0.0	97.2	101.8	0.6	0.4	0.4	0.4	0.0	0.0
374.1	95.7	97.9	45.0	15.0	0.6	0.6	0.6	26.3	0.0	0.0	61.7	71.3	0.6	0.5	0.3	0.3	0.0	0.0
378.3	95.2	97.4	35.0	20.0	0.8	0.8	0.8	40.2	0.0	0.0	74.8	84.3	0.6	0.5	0.3	1.3	0.0	0.0
384.7	91.1	97.0	40.0	25.0	1.0	1.0	1.0	41.9	0.0	0.0	63.5	98.3	0.6	0.6	0.4	1.5	0.0	0.0
394.5	94.8	97.5	40.0	23.0	0.6	0.6	0.6	37.5	0.0	0.0	61.7	112.4	0.6	1.0	0.3	1.7	0.0	0.0
400.8	93.0	95.7	33.0	16.0	0.5	0.5	0.5	24.9	0.0	0.0	63.5	94.1	0.6	1.2	0.2	1.8	0.0	0.0
406.0	94.7	97.3	40.0	20.0	0.5	0.5	0.5	40.2	0.0	0.0	61.7	92.7	0.6	1.4	0.2	1.9	0.0	0.0
412.5	94.4	96.8	39.0	15.0	0.5	0.5	0.5	28.1	0.0	0.0	61.7	92.7	0.8	1.5	0.2	2.0	0.0	0.0
420.3	95.1	98.1	43.0	23.0	0.5	0.5	0.5	38.8	0.0	0.0	57.9	112.4	1.1	1.8	0.2	2.1	0.0	0.0
426.1	95.2	97.0	43.0	23.0	4.0	4.0	4.0	35.5	-30.8	0.0	63.5	95.3	1.2	2.0	1.7	1.9	0.1	15.7
438.4	95.0	97.0	44.0	17.0	4.0	4.0	4.0	35.5	-35.9	0.0	52.3	93.8	1.2	2.2	1.7	1.9	0.4	19.3
446.0	95.2	97.9	43.0	20.0	4.0	4.0	4.0	32.4	-57.4	0.0	63.4	110.6	1.2	2.4	1.7	2.0	0.7	26.2
452.6	95.4	98.1	43.0	15.0	4.0	4.0	4.0	38.5	-57.4	0.0	82.2	120.5	1.1	2.4	1.7	1.9	1.1	26.2
459.4	95.9	98.1	43.0	15.0	4.3	4.3	4.3	30.4	-40.3	0.0	48.6	98.3	1.1	2.5	1.8	2.1	1.4	22.2
467.1	95.0	97.0	44.0	17.0	4.5	4.5	4.5	0.0	-28.0	0.0	0.0	0.0	1.2	2.6	1.9	1.9	1.6	19.6
468.6	95.0	97.0	44.0	17.0	4.0	4.0	4.0	0.0	-33.6	0.0	0.0	0.0	1.1	2.6	1.7	1.7	1.6	23.5
470.6	95.0	97.0	44.0	17.0	4.0	4.0	4.0	0.0	-46.7	0.0	0.0	0.0	1.1	2.5	1.6	1.7	1.7	32.7
474.2	95.0	97.0	44.0	17.0	3.5	3.5	3.5	0.0	-52.3	0.0	0.0	0.0	1.0	2.5	1.5	1.5	1.9	36.6
478.4	95.0	97.0	44.0	17.0	3.5	3.5	3.5	0.0	-52.3	0.0	0.0	0.0	1.0	2.4	1.5	1.5	1.9	36.6
480.3	95.0	97.0	44.0	17.0	3.0	3.0	3.0	0.0	-52.3	0.0	0.0	0.0	0.9	2.4	1.3	1.4	2.1	36.6
482.3	95.0	97.0	44.0	17.0	3.0	3.0	3.0	0.0	-54.2	0.0	0.0	0.0	0.9	2.3	1.3	1.4	2.1	36.6
486.1	95.0	97.0	44.0	17.0	2.8	2.8	2.8	0.0	-61.7	0.0	0.0	0.0	0.8	2.3	1.3	1.3	2.2	37.9
487.4	95.0	97.0	44.0	17.0	2.6	2.6	2.6	0.0	-65.4	0.0	0.0	0.0	0.7	2.2	1.2	1.2	2.4	43.2
489.3	95.0	97.0	44.0	17.0	2.5	2.5	2.5	0.0	-74.8	0.0	0.0	0.0	0.6	2.1	1.1	1.1	2.5	45.8
491.7	95.0	97.0	44.0	17.0	2.5	2.5	2.5	0.0	-72.9	0.0	0.0	0.0	0.6	2.0	1.0	1.0	2.6	52.3
493.4	95.0	97.0	44.0	17.0	2.0	2.0	2.0	0.0	-61.7	0.0	0.0	0.0	0.5	2.0	1.0	1.0	2.8	51.0
493.6	95.0	97.0	44.0	17.0	2.2	2.2	2.2	0.0	-56.1	0.0	0.0	0.0	0.5	2.0	0.8	0.8	2.9	43.2
496.2	95.0	97.0	44.0	17.0	2.0	2.0	2.0	0.0	-50.5	0.0	0.0	0.0	0.4	1.9	0.0	0.0	2.9	39.2
498.1	95.0	97.0	44.0	17.0	1.0	1.0	1.0	0.0	-25.2	0.0	0.0	0.0	0.4	1.9	0.0	0.0	3.0	35.3
498.2	95.0	97.0	44.0	17.0	0.0	0.0	0.0	0.0	-20.6	0.0	0.0	0.0	0.4	1.9	0.0	0.0	3.1	18.3
																		14.4



139.5°C

5 1/3, 18 ~ 19°C

139.5, 10°C ~ 0.25

4 1/3, 10.25 ~ 32 watts

Hot Look 37 watts

69

Time Minutes	L-STEP Liq	Effectiveness	O ₂ Net Flow g/sec
0.1		0.226	
6.7		0.421	
13.1		0.836	
19.3		0.844	
23.8		0.852	
31.5		0.889	
36.5		0.892	
41.4		0.896	
47.2		0.898	
54.8		0.902	
71.6		0.904	
76.3		0.905	
82.0		0.905	
93.3		0.908	
97.3		0.914	
103.3		0.914	
123.5		0.915	
132.7		0.921	
138.5		0.921	
144.3		0.921	
152.6		0.923	
153.7		0.930	
159.5		0.935	
164.3		0.936	
169.3		0.936	
177.9		0.937	
183.0		0.937	
185.5		0.937	
190.3		0.937	
194.4		0.937	
200.4		0.937	
204.9		0.937	
210.1		0.937	
214.3		0.937	
219.9		0.937	
226.6		0.937	
231.5		0.937	
237.7		0.937	
244.7		0.937	
256.0		0.937	
262.9		0.937	
271.1		0.937	
283.1		0.937	
293.3		0.937	
304.0		0.937	
310.4		0.937	
317.1		0.937	
321.2		0.937	
331.7		0.937	
337.4		0.937	
341.8		0.937	
348.6		0.937	
356.2		0.937	

O2 Test 457

Spreadsheet Test Input and Results - O₂ Test No. 7 (continued)

Time Minutes	LSFP Liq	Efficiency	O2 Net Flow g/min
361.6	133.1	0.948	0.05
369.2	140.1	0.947	0.17
374.1	93.2	0.947	0.16
378.3	110.2	0.948	0.16
384.7	128.6	0.948	0.38
394.5	146.9	0.947	0.84
400.8	123.0	0.947	0.51
406.0	121.2	0.948	0.52
412.5	121.2	0.947	0.52
420.3	146.9	0.947	0.91
426.1	84.4	0.942	0.53
438.4	75.6	0.945	0.69
446.0	69.6	0.941	0.75
452.6	82.5	0.982	0.64
459.4	75.8	0.939	0.83
467.1			0.00
468.6			
470.6			
474.2			
475.3			
478.4			
480.3			
482.3			
486.1			
487.4			
489.3			
491.7			
493.4			
495.6			
496.2			
498.1			
499.2			

i-Dec-90

7.6 Test Data

7.6.1 Raw Test Data

O2 Test No. 1

sat Sept 17 1980

TO- Minutes	14:00 Time		%O2 FM6 supply	%O2 FM2 vent	%O2 FM1 withdraw	O FM4	Psig RM2	FM-6	O R45	Psig FM2	O FM1	Psig RA-FM1	2	3	4	6	8	9	P3
0	15:35	ev1-8 warm	95	84		35	50	50	12	75			282.0	274.0	268.0	260.0	254.0	182.0	
0	15:35	LN2 ev-8	82	95									279.0	282.0	245.8	283.0	225.0	145.0	
6	15:41	LN2 ev-8											270.0	277.0	225.0	283.0	201.0	94.0	
12	15:47	LN2 ev-8											260.0	280.0	208.0	283.0	175.0	87.0	
14	15:50	LN2 ev-8											256.0	281.0	194.0	282.0	124.0	78.9	
16	15:51	LN2 & CH ev-8											179.0	280.0	176.0	282.0	82.0	77.2	
20	15:55	LN2 only											85.0	279.0	167.0	281.0	81.0	78.0	
25	16:01	LN2 only											82.0	278.0	148.0	281.0	82.0	82.0	
28	16:03	LN2 only											81.0	277.0	130.0	281.0	81.0	81.0	
33	16:09	LN2 only											79.0	276.0	124.0	281.0	79.0	80.0	
36	16:11	dry N2 ev1-8				48	10						79.0	275.0	118.0	281.0	80.0	80.9	
40	16:15	dry N2 ev1-8				70	30	70	0	0			84.6	271.0	79.0	267.0	77.6	70.5	27.0
46	16:21	O2 on	95.0			58	40	30	10	0			81.0	268.0	79.0	265.0	77.8	70.4	25.0
49	16:24	O2 on	97.2			60	40	40	10	0			80.5	254.0	78.0	260.0	74.0	69.0	40.0
57	16:32	O2 on	95.0			35	55	50	15	0			79.0	233.0	78.0	252.0	73.0	67.0	
60	16:36	ev2-7	93.9	67.0		55	48	50	5	80			79.0	223.0	77.0	240.0	75.0	68.8	
66	16:41		94.0	67.5		58	48	25	15	62			79.0	202.0	77.0	214.0	73.0	68.8	35.0
68	16:43		92.1	67.0		61	45	50	15				79.0	190.0	78.0	194.0	75.0	71.0	
74	16:49	fm-6 oscillating	74.7	67.0		60	45	50	18	44			80.0	176.0	79.0	184.0	73.0	71.0	
77			90.0	67.0		55	45	100	10	7			82.0	172.0	81.0	117.0	80.0	70.0	25.0
80	16:55		95.0	67.0		50	50	50	10	0			90.0	165.0	83.0	96.8	82.0	78.0	40.0
86	17:01	ev1-8	95.1	no flow	70	50	50	45	18	0	15		89.0	158.0	83.0	96.8	82.0	78.0	45.0
89	17:04	hw 60 on	96.8	no flow		45	50	35	5	0	0		87.0	152.0	83.7	99.0	82.0	73.6	50.0
92	17:07		96.5	no flow		38	50	40	10	0	8		84.9	146.0	82.2	94.0	79.4	75.0	50.0
101	17:16	hw 60 on	95.0	no flow	81.00	38	53	50	10	5	44		82.9	143.0	82.8	91.1	81.0	73.0	50.0
109	17:24	ev2-7	95.0	no flow	95.00	37	53	40	5	0	36		83.0	147.0	83.0	89.0	82.0	73.5	50.0
113	17:28		94.5	no flow	94.90	38	53	35	5	0	36		83.7	147.0	84.0	88.5	82.4	74.0	50.0
116	17:31		95.2	no flow	94.91	33	53	40	5	0	33		84.4	152.0	84.4	88.0	82.2	74.0	50.0
124	17:39	fm-1 off to pressurize	96.1	no flow	no flow	32	53	30	10	0	0		84.4	158.0	84.6	89.5	82.0	73.0	50.0
131	17:47		96.0	87.3	no flow	33	50	23	20	40	0		83.0	160.0	83.0	90.0	81.2	73.8	50.0
137	17:52		96.0	87.3	no flow	32	52	25	12	35	0		83.0	165.0	83.2	90.9	80.0	72.0	50.0
140	17:55		95.3	88.0	no flow	28	58	50	15	48	0		83.6	168.1	82.9	89.9	78.4	71.1	48.0
145	18:01		90.2	91.0	no flow	34	52	38	15	35	0		84.5	166.0	84.6	88.6	82.6	74.4	50.0
150	18:06	heater #1 on 50%	91.5	89.2		35	50	35	15	28	0		87.0	172.0	84.9	88.8	82.5	74.6	50.0
155	18:10	heater #1 on 50%	78.8	85.1		33	50	50	20	40	0		89.9	173.0	84.8	88.2	82.7	76.0	50.0
158	18:13	heater #2 on also	no flow								12	0							
166	18:21	terminated vacuum			92.1						12	0							
172	18:27				92.1						60	40							
185	18:50				85.8														

$\frac{d}{dt} \left(\frac{1}{2} m v^2 \right) = \frac{1}{2} m \frac{d}{dt} (v^2)$
 $= \frac{1}{2} m \frac{d}{dt} (v_x^2 + v_y^2 + v_z^2)$
 $= \frac{1}{2} m \left(2 v_x \frac{d v_x}{dt} + 2 v_y \frac{d v_y}{dt} + 2 v_z \frac{d v_z}{dt} \right)$
 $= m \left(v_x \frac{d v_x}{dt} + v_y \frac{d v_y}{dt} + v_z \frac{d v_z}{dt} \right)$
 $= m \left(v_x a_x + v_y a_y + v_z a_z \right)$
 $= m \mathbf{v} \cdot \mathbf{a}$
 $= m \mathbf{v} \cdot \frac{d \mathbf{v}}{dt}$
 $= \frac{d}{dt} \left(\frac{1}{2} m v^2 \right)$

50	45
31	15
28	12
41	10
28	10
17	0
15	0
8	0

82.9
85.3
90.0
91.2
93.3
94.1
93.4
92.2

heaters on 100%

202	1857
209	1904
223	1919
232	1927
234	1930
238	1933
245	1941
250	1945

4.4	4.0	0.0	-93.5	0.0	1.5	2.2	1.8	1.7	3.4	65
3.5	3.0	0.0	-57.9	0.0	1.2	2.0	1.5	1.3	3.9	41
3.0	2.0	0.0	-52.3	0.0	0.9	1.6	1.3	0.8	4.6	37
3.3	1.0		-76.6	0.0	0.6	1.4	1.4	0.4	5.0	54
3.1	1.0		-52.3	0.0	0.6	1.3	1.3	0.4	5.2	37
3.0	0.9		-31.8	0.0	0.5	1.2	1.3	0.4	5.3	22
2.5	0.9		-28.0	0.0	0.4	1.1	1.0	0.4	5.5	20
0.5	0.7		-15.0	0.0	0.4	1.1	0.2	0.3	5.6	10

星一又三三三

144

TO- Minutes	1425 Time	%O2 FIM supply	%O2 FIM2 vent	%O2 FIM1 withdraw	O FIM4	Pdg FIM2	O FIM-6	Pdg FIM-6	O FIM2	O FIM1	Pdg FIM-Flu	Temperature 2	3	4	6	8	9	P3	P7	L1	L2	Vaporized No.	Delta P FLU	Liq. level	Gravim
0	1425				30	50						285.0	287.0	283.0	286.0	272.0	210.0					0.0			
9	1434	CH + H2 only			30	40						257.0	286.0	240.0	286.9	230.8	174.0					0.0			
12	1437	CH + H2 only			29	39						253.0	285.0	222.0	283.0	221.0	172.0					22.5			
16	1441	LM2 + CH										238.0	285.0	205.0	263.0	212.0	161.0					21.6			
25	1456	LM2 + CH										234.0	284.0	207.0	285.0	195.0	98.0					0.0			
33	1458	LM2 + CH										216.0	282.0	182.0	285.0	141.0	90.0					0.0			
38		n2 + ch										206.0	282.0	188.0	284.0	90.0	84.0					27.5			
40		n2 + ch										185.0	281.0	156.0	284.0	84.0	84.0					29.2			
43	1508	air on			30	55			38			97.0	280.0	132.0	283.0	97.0	90.4					16.4			
45		air - 8			25	60			18			97.4	275.0	123.0	279.0	94.0	90.4					10.4			
48	1513	CH F1 off	93.70	91.00																	0.0				
51	1516	air-1	94.20	94.00	20	60	30	14	22			97.0	278.0	118.0	289.0	95.0	90.0					0.0			
53	1518	air-1	95.00	94.40			30	11				97.0	277.0	116.0	263.0	97.0	90.0					0.0			
61	1528	bag #1 M/SOG supply	94.00	95.90	20	60	30	11	35			97.4	278.0	114.0	251.0	97.1	89.8					6.7			
64	1529	air-2-7	94.00	94.50	22	60	30	11	35			97.0	278.0	112.0	238.0	96.9	89.5					6.7			
69	1534	air-2-7	94.56	94.00	20	62	30	15	26			96.0	273.0	109.0	210.0	96.7	89.0					0.5			
72	1537	air-2-7	95.60	95.00	25	60	30	10	24			97.0	273.0	109.0	210.0	96.7	89.0					0.5			
78	1543	air-1-8	95.60	95.40	18	62	30	15	25			96.0	271.0	107.8	192.0	96.0	89.0					0.5			
82	1547	air-1-8	96.00	95.80	15	65	40	10	80			96.0	270.0	106.0	175.0	97.0	84.0					0.7			
85	1550	air																							

Code	186.7	AR	0.1034	4.4521	GR/min	GRSec1	1290
1142	C2	0.0028	4.9748	208.8	1303	1142	
3.80	Ar	0.0748	14.8000	486.2			
Liquid to cool down =			1.00				
FMS	FMS-FM2	Libers	Libers	Libers	Libers Lq	LSTP	
SCFM	Gr/min	Dewar2	L1	L2	withdraw	withdraw	
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0
6.3	0.0	0.0	0.0	0.0	0.0	0.0	0
6.3	0.0	0.0	0.0	0.0	0.0	0.0	0
6.1	0.0	0.0	0.0	0.0	0.0	0.0	0
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0
13.6	0.0	0.0	0.0	0.0	0.0	0.0	0
7.3	0.0	0.0	0.0	0.0	0.0	0.0	0
6.3	0.0	78.1	0.1	0.0	0.0	0.0	0
5.2	0.0	83.1	0.3	0.0	0.0	0.0	0
0.0	0.0	82.9	0.4	0.0	0.0	0.0	0
8.2	0.0	83.4	0.8	0.0	0.0	0.0	0
0.0	0.0	79.4	0.9	0.0	0.0	0.0	0
9.4	0.0	79.4	1.0	0.0	0.0	0.0	0
4.2	0.0	79.4	1.2	0.0	0.0	0.0	0
10.4	0.0	79.4	1.5	0.0	0.0	0.0	0
5.2	0.0	78.1	1.7	0.0	0.0	0.0	0
9.4	0.0	84.6	2.1	0.3	0.0	0.0	0
8.5	0.0	84.6	2.3	0.3	0.0	0.0	0
7.1	0.0	104.1	2.8	0.3	0.0	0.0	0
6.6	0.0	78.1	3.2	0.3	0.0	0.0	0
7.2	0.0	78.1	3.3	0.4	0.0	0.0	0
7.8	0.0	94.2	3.6	0.4	0.0	0.0	0
14.6	0.0	82.1	3.8	1.3	0.0	0.0	0
7.2	0.0	86.6	4.0	1.7	0.0	0.0	0
7.3	0.0	87.8	4.4	1.9	0.0	0.0	0
15.9	0.0	91.1	4.8	1.9	0.0	0.0	0
8.8	0.0	50.2	5.4	0.2	0.0	0.0	0
7.3	0.0	56.7	5.7	0.2	0.0	0.0	0
5.4	0.0	78.1	5.9	0.2	1.6	0.0	0
0.0	0.0	0.0	6.3	0.2	1.7	0.0	0
0.0	-37.4	0.0	6.2	0.2	0.0	0.1	28
0.0	-18.7	0.0	5.8	0.2	0.0	0.3	13
0.0	-18.7	0.0	5.9	0.2	0.0	0.4	13
0.0	-18.7	0.0	5.9	0.2	0.0	0.4	13
0.0	-18.7	0.0	5.8	0.2	0.0	0.4	13
0.0	-9.3	0.0	5.8	0.2	0.0	0.5	7
0.0	-9.3	0.0	7.1	0.2	0.0	-0.8	7
0.0	0.0	0.0	7.1	0.2	0.0	-0.8	0
0.0	0.0	0.0	7.1	0.2	0.0	-0.8	0
0.0	0.0	0.0	7.1	0.2	0.0	-0.8	0
0.0	0.0	0.0	7.1	0.2	0.0	-0.8	0
0.0	0.0	0.0	7.1	0.2	0.0	-0.8	0
0.0	0.0	0.0	7.1	0.2	0.0	-0.8	0

TD- miles	1425 Time	%O2 FMS supply	%O2 FMS vent	%O2 FMS submers	O FMS	Pdg P4.2	O FMS4	Pdg P4.5	O FMS2	O FMS1	Pdg P4-FH	1	2	3	4	5	6	7	8	9	P5	P4	L1
-47	1033				37	52						14.9	130.0	272.0	118.0	262.0	118.0						
-35					34	52						142.0	129.0	272.0	118.0	262.0	118.0						
-24					31	52						134.0	131.0	272.0	122.8	262.0	122.8						
-21					28	55						137.5	133.6	272.0	128.0	262.0	128.0						
-19					28	55						134.0	131.0	272.0	122.0	262.0	122.0						
-17					30	58						137.2	137.5	272.0	129.5	262.0	129.5						
-0					33	55						137.0	138.0	272.0	132.0	262.0	132.0						
6					35	55						138.0	132.0	272.0	118.0	262.0	117.0						
15					33	55						136.1	123.8	270.0	111.0	262.0	112.0						
16					33	55						133.0	121.0	268.0	105.0	262.0	105.0						
18					34	55						129.0	120.0	268.0	104.0	262.0	101.0						
23					32	54						124.0	111.0	267.0	96.0	278.0	97.0						
26					28	55						124.0	111.0	267.0	96.0	271.0	97.5						
28					25	54						124.0	111.0	267.0	95.9	265.0	97.5						
31					24	58						124.8	112.0	268.0	95.3	257.0	96.0						
33					23	60						125.0	112.0	265.0	96.0	240.0	105.0						
36					27	58						125.4	116.0	265.0	104.0	232.0	106.0						
41					23	58						125.0	119.0	264.0	115.0	222.0	115.0						
46					22	58						125.0	120.0	264.0	123.0	210.0	121.0						
52					23	59						127.0	129.0	262.0	123.0	203.0	121.0						
54					27	59						140.7	134.8	260.0	119.0	192.0	119.0						
62					28	56						141.0	138.0	260.0	114.0	185.0	114.0						
71					34	55						138.0	127.0	258.0	112.0	183.0	113.0						
81					30	50						133.0	119.0	256.0	108.0	179.0	108.0						
84					50	45						130.0	116.0	257.0	108.0	178.0	108.0						
87					48	45						128.0	114.0	256.0	107.0	176.0	106.0						
96					48	45						127.0	121.0	252.0	114.0	173.0	112.0						
105					48	48						125.0	121.0	252.0	113.0	173.0	111.0						
108					45	50						126.5	118.0	254.0	111.4	173.0	109.0						
119					55	45						123.0	116.7	257.0	108.9	172.8	108.5						
120					53	45						115.0	118.0	253.0	110.0	170.0	108.0						
127					54	45						118.0	120.0	254.0	117.0	167.0	115.0						
132					53	44						125.0	129.0	251.0	125.0	165.0	126.0						
144					58	45						138.0	140.0	250.0	136.0	163.0	134.0						
145					50	45						137.0	142.0	250.0	136.0	163.0	135.0						
145					58	45						142.0	139.0	250.0	132.0	164.0	130.0						
154					54	45						137.0	133.0	248.0	122.0	168.0	119.0						
164					49	55						130.6	121.5	248.0	116.0	168.0	115.0						
171					53	43						126.0	117.0	248.0	111.0	168.0	107.0						
178					58	45						121.8	111.0	246.0	99.6	175.0	89.0						
184					50	30						118.0	108.5	248.0	96.6	178.0	84.0						
188					48	30						117.0	106.3	246.0	94.2	177.0	87.0						
195					35	35						116.8	108.0	248.0	92.4	178.0	87.7						
204					35	50						116.5	107.5	246.0	91.0	168.0	87.4						
208					32	50						116.5	107.5	246.0	91.0	168.0	87.4						
212					28	53						116.0	107.0	244.0	91.0	159.0	86.0						
1405					97.7	97.0																	

215	216	217	218	219	220	221	222	223	224	225	226	227	228	229	230	231	232	233	234	235	236	237	238	239	240	241	242	243	244	245	246	247	248	249	250	251	252	253	254	255	256	257	258	259	260	261	262	263	264	265	266	267	268	269	270	271	272	273	274	275	276	277	278	279	280	281	282	283	284	285	286	287	288	289	290	291	292	293	294	295	296	297	298	299	300	301	302	303	304	305	306	307	308	309	310	311	312	313	314	315	316	317	318	319	320	321	322	323	324	325	326	327	328	329	330	331	332	333	334	335	336	337	338	339	340	341	342	343	344	345	346	347	348	349	350	351	352	353	354	355	356	357	358	359	360	361	362	363	364	365	366	367	368	369	370	371	372	373	374	375	376	377	378	379	380	381	382	383	384	385	386	387	388	389	390	391	392	393	394	395	396	397	398	399	400	401	402	403	404	405	406	407	408	409	410	411	412	413	414	415	416	417	418	419	420	421	422	423	424	425	426	427	428	429	430	431	432	433	434	435	436	437	438	439	440	441	442	443	444	445	446	447	448	449	450	451	452	453	454	455	456	457	458	459	460	461	462	463	464	465	466	467	468	469	470	471	472	473	474	475	476	477	478	479	480	481	482	483	484	485	486	487	488	489	490	491	492	493	494	495	496	497	498	499	500	501	502	503	504	505	506	507	508	509	510	511	512	513	514	515	516	517	518	519	520	521	522	523	524	525	526	527	528	529	530	531	532	533	534	535	536	537	538	539	540	541	542	543	544	545	546	547	548	549	550	551	552	553	554	555	556	557	558	559	560	561	562	563	564	565	566	567	568	569	570	571	572	573	574	575	576	577	578	579	580	581	582	583	584	585	586	587	588	589	590	591	592	593	594	595	596	597	598	599	600	601	602	603	604	605	606	607	608	609	610	611	612	613	614	615	616	617	618	619	620	621	622	623	624	625	626	627	628	629	630	631	632	633	634	635	636	637	638	639	640	641	642	643	644	645	646	647	648	649	650	651	652	653	654	655	656	657	658	659	660	661	662	663	664	665	666	667	668	669	670	671	672	673	674	675	676	677	678	679	680	681	682	683	684	685	686	687	688	689	690	691	692	693	694	695	696	697	698	699	700	701	702	703	704	705	706	707	708	709	710	711	712	713	714	715	716	717	718	719	720	721	722	723	724	725	726	727	728	729	730	731	732	733	734	735	736	737	738	739	740	741	742	743	744	745	746	747	748	749	750	751	752	753	754	755	756	757	758	759	760	761	762	763	764	765	766	767	768	769	770	771	772	773	774	775	776	777	778	779	780	781	782	783	784	785	786	787	788	789	790	791	792	793	794	795	796	797	798	799	800	801	802	803	804	805	806	807	808	809	810	811	812	813	814	815	816	817	818	819	820	821	822	823	824	825	826	827	828	829	830	831	832	833	834	835	836	837	838	839	840	841	842	843	844	845	846	847	848	849	850	851	852	853	854	855	856	857	858	859	860	861	862	863	864	865	866	867	868	869	870	871	872	873	874	875	876	877	878	879	880	881	882	883	884	885	886	887	888	889	890	891	892	893	894	895	896	897	898	899	900	901	902	903	904	905	906	907	908	909	910	911	912	913	914	915	916	917	918	919	920	921	922	923	924	925	926	927	928	929	930	931	932	933	934	935	936	937	938	939	940	941	942	943	944	945	946	947	948	949	950	951	952	953	954	955	956	957	958	959	960	961	962	963	964	965	966	967	968	969	970	971	972	973	974	975	976	977	978	979	980	981	982	983	984	985	986	987	988	989	990	991	992	993	994	995	996	997	998	999	1000	1001	1002	1003	1004	1005	1006	1007	1008	1009	1010	1011	1012	1013	1014	1015	1016	1017	1018	1019	1020	1021	1022	1023	1024	1025	1026	1027	1028	1029	1030	1031	1032	1033	1034	1035	1036	1037	1038	1039	1040	1041	1042	1043	1044	1045	1046	1047	1048	1049	1050	1051	1052	1053	1054	1055	1056	1057	1058	1059	1060	1061	1062	1063	1064	1065	1066	1067	1068	1069	1070	1071	1072	1073	1074	1075	1076	1077	1078	1079	1080	1081	1082	1083	1084	1085	1086	1087	1088	1089	1090	1091	1092	1093	1094	1095	1096	1097	1098	1099	1100	1101	1102	1103	1104	1105	1106	1107	1108	1109	1110	1111	1112	1113	1114	1115	1116	1117	1118	1119	1120	1121	1122	1123	1124	1125	1126	1127	1128	1129	1130	1131	1132	1133	1134	1135	1136	1137	1138	1139	1140	1141	1142	1143	1144	1145	1146	1147	1148	1149	1150	1151	1152	1153	1154	1155	1156	1157	1158	1159	1160	1161	1162	1163	1164	1165	1166	1167	1168	1169	1170	1171	1172	1173	1174	1175	1176	1177	1178	1179	1180	1181	1182	1183	1184	1185	1186	1187	1188	1189	1190	1191	1192	1193	1194	1195	1196	1197	1198	1199	1200	1201	1202	1203	1204	1205	1206	1207	1208	1209	1210	1211	1212	1213	1214	1215	1216	1217	1218	1219	1220	1221	1222	1223	1224	1225	1226	1227	1228	1229	1230	1231	1232	1233	1234	1235	1236	1237	1238	1239	1240	1241	1242	1243	1244	1245	1246	1247	1248	1249	1250	1251	1252	1253	1254	1255	1256	1257	1258	1259	1260	1261	1262	1263	1264	1265	1266	1267	1268	1269	1270	1271	1272	1273	1274	1275	1276	1277	1278	1279	1280	1281	1282	1283	1284	1285	1286	1287	1288	1289	1290	1291	1292	1293	1294	1295	1296	1297	1298	1299	1300	1301	1302	1303	1304	1305	1306	1307	1308	1309	1310	1311	1312	1313	1314	1315	1316	1317	1318	1319	1320	1321	1322	1323	1324	1325	1326	1327	1328	1329	1330	1331	1332	1333	1334	1335	1336	1337	1338	1339	1340	1341	1342	1343	1344	1345	1346	1347	1348	1349	1350	1351	1352	1353	1354	1355	1356	1357	1358	1359	1360	1361	1362	1363	1364	1365	1366	1367	1368	1369	1370	1371	1372	1373	1374	1375	1376	1377	1378	1379	1380	1381	1382	1383	1384	1385	1386	1387	1388	1389	1390	1391	1392	1393	1394	1395	1396	1397	1398	1399	1400	1401	1402	1403	1404	1405	1406	1407	1408	1409	1410	1411	1412	1413	1414	1415	1416	1417	1418	1419	1420	1421	1422	1423	1424	1425	1426	1427	1428	1429	1430	1431	1432	1433	1434	1435	1436	1437	1438	1439	1440	1441	1442	1443	1444	1445	1446	1447	1448	1449	1450	1451	1452	1453	1454	1455	1456	1457	1458	1459	1460	1461	1462	1463	1464	1465	1466	1467	1468	1469	1470	1471	1472	1473	1474	1475	1476	1477	1478	1479	1480	1481	1482	1483	1484	1485	1486	1487	1488	1489	1490	1491	1492	1493	1494	1495	1496	1497	1498	1499	1500	1501	1502	1503	1504	1505	1506	1507	1508	1509	1510	1511	1512	1513	1514	1515	1516	1517	1518	1519	1520	1521	1522	1523	1524	1525	1526	1527	1528	1529	1530	1531	1532	1533	1534	1535	1536	1537	1538	1539	1540	1541	1542	1543	1544	1545	1546	1547	1548	1549	1550	1551	1552	1553	1554	1555	1556	1557	1558	1559	1560	1561	1562	1563	1564	1565	1566	1567	1568	1569	1570	1571	1572	1573	1574	1575	1576	1577	1578	1579	1580	1581	1582	1583	1584	1585	1586	1587	1588	1589	1590	1591	1592	1593	1594	1595	1596	1597	1598	1599	1600	1601	1602	1603	1604	1605	1606	1607	1608	1609	1610	1611	1612	1613	1614	1615	1616	1617	1618	1619	1620	1621	1622	1623	1624	1625	1626	1627	1628	1629	1630	1631	1632	1633	1634	1635	1636	1637	1638	1639	1640	1641	1642	1643	1644	1645	1646	1647	1648	1649	1650	1651	1652	1653	1654	1655	1656	1657	1658	1659	1660	1661	1662	1663	1664	1665	1666	1667	1668	1669	1670	1671	1672	1673	1674
-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	-----	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------

87.2	21.1	0.0	30.2	0.0	0.0	0.0	0.0	0.0	0.0
87.2	24.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
87.2	24.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
87.2	24.8	0.0	53.8	0.0	0.0	0.0	0.0	0.0	0.0
82.4	22.3	0.0	53.6	0.2	0.1	0.0	0.0	0.0	0.0
59.5	22.3	0.0	70.5	0.4	0.1	0.0	0.0	0.0	0.0
52.1	21.0	0.0	60.5	0.7	0.2	0.0	0.0	0.0	0.0
50.0	20.6	0.0	54.9	0.8	0.2	0.0	0.0	0.0	0.0
24.5	14.7	0.0	54.4	1.1	0.4	0.0	0.0	0.0	0.0
8.3	9.5	0.0	56.4	1.3	1.7	0.0	0.0	0.0	0.0
0.0	0.0	-54.1	0.0	1.3	1.5	0.0	0.1	39.2	0.0
0.0	0.0	-52.3	0.0	0.9	1.5	0.0	0.4	36.6	0.0
0.0	0.0	-54.1	0.0	0.9	1.5	0.0	0.4	39.2	0.0
0.0	0.0	-54.1	0.0	0.9	1.5	0.0	0.5	39.2	0.0
0.0	0.0	-43.5	0.0	0.6	1.5	0.0	0.6	44.5	0.0
0.0	0.0	-18.7	0.0	0.0	0.0	0.0	1.6	13.1	0.0

O₂ TEST # 5.007
NOV 30

02 Test No. 5

1990

TD= Minutes	14:25 Time	%O ₂ FM6 supply	%O ₂ FM2 vent	%O ₂ FM1 withdraw	Q FM4	Psig R2	Q FM-6	Psig R2	Q FM2	Q FM1
2	9.77				55	33				
16	9.80				53	30				
19	10.03				53	30				
23	10.08				52	30				
26	10.15				53	30				
34	10.20				53	30				
36	10.33				52	30				
39	10.37				50	30				
42	10.42				53	30				
46	10.47				53	30				
52	10.53				53	30				
64	10.63				52	30				
69	10.83				53	30				
74	10.92				53	30				
79	11.00				51	30				
84	11.08				52	30				
89	11.17				52	30				
94	11.25				52	30				
99	11.33				52	30				
104	11.42				52	30				
109	11.50				52	30				
114	11.58				52	30				
119	11.67				52	30				
129	11.75				52	30				
134	11.92				44	30				
144	12.00				44	32				
154	12.17				44	32				
164	12.33				46	31				
174	12.50				46	31				
179	12.67				45	32				
184	12.75				47	32				
204	12.83				47	30				
214	1.17				47	30				
224	1.33				48	30				
242	1.50				48	32				
248	1.80				45	30				
254	1.90				46	30				
261	2.00				48	31				
270	2.12				48	31				
284	2.27				49	31				
289	2.50				53	45				
	2.58				44	45				

95.70

95.51

95.40

95.70 95 94.70 95.80 93.40 93.90 94.50

Bag Sample #1 @ FM-6

25 15 15 15 15 12 17 35 38 38 37 35 28

03-Dec-90

TD= Minutes	Prig RA-FM1	Temperatures			4	6	8	9	P3	P5	P4	L1	L2	Gr/min Liq liter= Delta P FM-6=	Drym No.	Vaporized Li FM4 SCFM
		2	3													
2		284.7	284.0	284.0	281.3	284.1	284.1	278.6							0.0	0.0
16		274.0	284.0	284.0	281.3	280.8	280.8	269.1							0.0	0.0
19		274.7	284.0	284.0	280.0	280.0	280.0	270.0							0.0	0.0
23		273.0	284.0	284.0	280.0	280.0	280.0	270.0							91.7	11.5
26		268.6	278.0	284.0	279.7	280.2	280.2	273.6							93.6	11.1
34		271.9	278.0	284.0	279.7	283.0	283.0	273.6							93.6	11.1
36		274.7	277.4	284.0	279.1	283.0	283.0	280.2							90.1	18.9
39		274.1	277.0	284.0	279.0	280.0	280.0	255.2							83.3	18.2
42		271.3	276.3	284.0	277.4	279.7	279.7	254.7							93.6	19.3
46		266.9	274.7	284.0	276.3	277.4	277.4	268.6							90.1	18.9
52		268.6	273.6	284.0	274.1	277.0	277.0	268.6							90.1	18.9
64		271.3	273.0	284.0	273.0	273.0	273.0	273.0							93.6	19.3
69		273.0	273.0	284.0	273.0	273.0	273.0	255.2							93.6	19.3
74		273.0	273.0	284.0	273.0	273.0	273.0	241.9							86.7	18.6
79		273.0	272.0	284.0	273.0	273.0	273.0	231.0							90.1	18.9
84		271.3	264.7	284.0	250.8	250.8	250.8	223.0							90.1	18.9
89		266.9	257.4	280.0	243.0	243.0	243.0	214.7							90.1	18.9
94		262.0	254.0	278.0	239.0	239.0	239.0	207.0							90.1	18.9
99		257.5	247.0	277.0	225.0	225.0	225.0	196.0							90.1	18.9
104		250.0	234.6	276.0	219.0	219.0	219.0	189.0							90.1	18.9
109		245.0	234.0	275.0	219.0	219.0	219.0	189.2							90.1	18.9
114		240.0	229.0	274.0	214.0	214.0	214.0	184.6							90.1	18.9
119		235.5	223.6	273.0	208.0	208.0	208.0	179.3							90.1	18.9
129		224.0	216.0	272.0	199.0	199.0	199.0	165.9							64.5	16.0
134		220.0	208.0	270.0	192.0	192.0	192.0	161.0							60.5	16.0
144		210.0	197.0	269.0	182.0	182.0	182.0	154.0							60.5	16.0
154		204.0	191.0	268.0	176.0	176.0	176.0	147.0							68.3	16.7
164		194.0	179.0	267.0	166.9	166.9	166.9	139.0							68.3	16.7
174		184.0	169.0	266.0	157.0	157.0	154.0	131.0							63.3	16.7
179		180.0	167.0	265.0	154.2	154.2	151.7	127.0							69.0	17.5
184		176.0	161.7	264.0	148.9	148.9	146.0	124.7							73.6	17.3
204		160.0	146.0	263.0	135.5	135.5	132.6	111.0							73.6	17.3
214		154.0	140.0	263.0	129.0	129.0	127.0	103.7							72.0	17.9
224		147.0	134.0	262.0	124.0	124.0	122.0	99.0							72.0	17.9
242		134.0	121.0	260.0	112.0	112.0	110.0	89.8							67.5	16.4
248		132.0	119.0		109.0	109.0	107.0	86.7							70.5	16.7
254		131.0	117.0	259.0	108.0	108.0	106.0	88.0							74.3	17.5
261		131.0	117.0	257.0	108.0	108.0	106.0	88.0							74.3	17.9
270		127.6	113.6	254.0	106.0	106.0	104.0	86.0							77.5	17.8
284		122.0	110.0	253.0	98.5	98.5	95.9	86.9							62.4	19.3
289		121.0	108.0	251.0	97.6	97.6	94.4	86.4							43.0	16.2

Oct 4 1990

TO= Minutes	Psig RA-FM1	Temperatures		9		P5	P4	L1	L2	Gr/min Liq liter= Delta P FM-6		Vaporized Li FM4 SCFM
		2	3	4	6	8				Drym No.		
310		121.0	108.0	96.5		94.0	22.0	37.0	0.3	43.0		16.2
318		122.0	109.0	92.0		95.0	8.0	40.0	0.8	23.1		12.5
325		123.0	111.0	89.5		95.4	18.0	45.0	0.8	18.0		12.6
334		124.0	112.0	88.0		95.3	7.0	44.0	0.8	16.8		12.2
348		126.0	113.0	88.5		95.0	12.0	48.0	4.0	16.8		12.2
362		126.0	113.0	88.4		95.4	9.0	50.0	4.3	13.5		11.4
374		126.7	113.0	87.3		95.7	5.0	45.0	4.0	13.5		11.4
389		127.0	113.0	86.8		94.8	4.0	40.0	5.0	12.0		11.0
394		126.7	113.7	89.2		95.5	19.0	46.0	4.6	15.7		12.3
404		124.7	112.0	90.7		94.0	10.0	35.0	5.8	20.0		13.6
413		123.8	111.6	91.5		94.0	25.0	40.0	0.8	29.1	0.8	16.2
421		123.5	110.9	91.3		95.4	15.0	43.0	0.8	20.5	0.8	14.2
428		123.7	111.0	89.7		94.0	14.0	40.0	0.8	19.2	0.8	13.6
439	30	124.5	112.0	88.6		95.2	10.0	40.0	5.5	12.5	1.5	10.8
450	25	124.8	112.0	91.0		95.6	9.0	30.0	5.0	26.1	2.0	15.0
458	20	123.1	111.0	93.0		93.9	40.0	24.0	5.0	35.6	3.5	17.5
464	20	122.0	110.0	94.0		93.1	13.0	33.0	5.0	35.6	3.8	17.5
469	20	122.0	110.5	93.5		93.8	26.0	40.0	5.0	31.4	4.0	16.6
474	20	122.5	110.0	93.7		93.3	12.0	35.0	4.5	33.8	4.0	16.7
479	20	122.7	111.0	93.1		94.1	23.0	40.0	4.5	30.7	4.0	16.0
484	20	123.7	111.0	91.8		94.1	9.0	40.0	1.3	22.8	4.5	13.5
489	20	123.8	111.9	91.2		95.4	20.0	44.0	4.3	25.5	4.5	14.8
494	20	124.5	112.5	90.4		95.2	10.0	40.0	4.3	18.4	4.5	12.6
499	20								4.0	0.0	4.5	0.0
506	20								3.5	0.0	4.0	0.0
511	20								3.5	0.0	4.0	0.0
514	18								2.5	0.0	4.0	0.0
518	15								1.5	0.0	4.0	0.0
520	11								0.0	0.0	3.0	0.0
524	15								0.0	0.0	3.5	0.0
534	15								0.0	0.0	2.8	0.0
539	15								0.0	0.0	2.0	0.0
550	15								0.0	0.0	0.0	0.0
550	15								0.0	0.0	0.0	0.0

03-Dec-90

Oct 4 1990	TD= Minutes	Coeff= 3.50 aid to cool dewar=	FM1 Gr/min	FM2 Gr/min	FM6 Gr/min	AR O2 Air	NTP #/l3 0.1034 0.0828 0.0749	Coeff 4.4521 4.9749 14.6000	GR/min 208.8 186.9 496.2	GR/liq l 1393 1142	1290
						Liters Dewar1	Liters Dewar2	Liters L1	Liters L2	Liters Liq withdrawn	L STP withdrawn
	2	0.0	0.0	0.0	0.0			0.0	0.0		
	16	0.0	0.0	0.0	0.0			0.0	0.0		
	19	0.0	0.0	0.0	0.0			0.0	0.0		
	23	0.0	0.0	0.0	0.0			0.0	0.0		
	26	0.0	0.0	0.0	0.0			0.0	0.0		
	34	0.0	0.0	0.0	0.0			0.0	0.0		
	36	0.0	0.0	0.0	0.0			0.0	0.0		
	39	0.0	0.0	0.0	0.0			0.0	0.0		
	42	0.0	0.0	0.0	0.0			0.0	0.0		
	46	0.0	0.0	0.0	0.0			0.0	0.0		
	52	0.0	0.0	0.0	0.0			0.0	0.0		
	64	0.0	0.0	0.0	0.0			0.0	0.0		
	69	0.0	0.0	0.0	0.0			0.0	0.0		
	74	0.0	0.0	0.0	0.0			0.0	0.0		
	79	0.0	0.0	0.0	0.0			0.0	0.0		
	84	0.0	0.0	0.0	0.0			0.0	0.0		
	89	0.0	0.0	0.0	0.0			0.0	0.0		
	94	0.0	0.0	0.0	0.0			0.0	0.0		
	99	0.0	0.0	0.0	0.0			0.0	0.0		
	104	0.0	0.0	0.0	0.0			0.0	0.0		
	109	0.0	0.0	0.0	0.0			0.0	0.0		
	114	0.0	0.0	0.0	0.0			0.0	0.0		
	119	0.0	0.0	0.0	0.0			0.0	0.0		
	129	0.0	0.0	0.0	0.0			0.0	0.0		
	134	0.0	0.0	0.0	0.0			0.0	0.0		
	144	0.0	0.0	0.0	0.0			0.0	0.0		
	154	0.0	0.0	0.0	0.0			0.0	0.0		
	164	0.0	0.0	0.0	0.0			0.0	0.0		
	174	0.0	0.0	0.0	0.0			0.0	0.0		
	179	0.0	0.0	0.0	0.0			0.0	0.0		
	184	0.0	0.0	0.0	0.0			0.0	0.0		
	204	0.0	0.0	0.0	0.0			0.0	0.0		
	214	0.0	0.0	0.0	0.0			0.0	0.0		
	224	0.0	0.0	0.0	0.0			0.0	0.0		
	242	0.0	0.0	0.0	0.0			0.0	0.0		
	248	0.0	0.0	0.0	0.0			0.0	0.0		
	254	0.0	0.0	0.0	0.0			0.0	0.0		
	261	0.0	0.0	0.0	0.0			0.0	0.0		
	270	0.0	0.0	0.0	0.0			0.0	0.0		
	284	0.0	0.0	0.0	0.0			0.0	0.0		
	289	0.0	0.0	0.0	0.0			0.0	0.0		

Oct 4 1990	Coef=	uid to cool dewar=	FM2	FM6	AR	NTP #/G	Coef	GR/min	GR/liq l	1290
Minutes	Gr/min	Gr/min	Gr/min	Gr/min	Dewar1	Liters Dewar2	Liters L1	Liters L2	Liters Liq withdrawn	L-STP withdrawn
310	0.0	78.5	71.1	0.5			0.3	0.0	0.0	0.0
318	0.0	78.5	75.0	0.4			0.3	0.0	0.0	0.0
325	0.0	72.9	84.3	0.5			0.3	0.0	0.0	0.0
334	0.0	71.0	90.6	0.6			1.7	0.0	0.0	0.0
348	0.0	69.2	95.8	0.9			1.7	0.0	0.0	0.0
362	0.0	71.0	94.1	1.2			1.8	0.0	0.0	0.0
374	0.0	65.4	97.3	1.5			1.7	0.0	0.0	0.0
389	0.0	63.5	99.8	1.9			2.1	0.0	0.0	0.0
394	0.0	20.6	73.8	2.1			1.9	0.0	0.0	0.0
404	0.0	56.1	102.7	2.1			2.4	0.3	0.0	0.0
413	0.0	97.2	102.7	2.1		0.1	2.1	0.3	0.0	0.0
421	0.0	97.2	93.8	2.1		0.1	2.1	0.3	0.0	0.0
428	0.0	93.5	95.3	2.1		0.1	2.2	0.3	0.0	0.0
439	-26.1	74.8	95.3	2.0		0.2	2.3	0.6	0.1	10.5
450	-24.6	56.1	84.3	1.7		0.4	2.1	0.8	0.4	10.5
458	-28.7	56.1	98.3	1.6		0.7	2.1	1.5	0.6	13.1
464	-31.6	56.1	98.3	1.4		0.9	2.1	1.6	0.7	14.4
469	-31.6	55.4	98.3	1.3		1.1	2.1	1.7	0.9	14.4
474	-34.5	65.4	98.3	1.1		1.2	1.9	1.7	1.0	15.7
479	-34.5	65.4	98.3	1.0		1.4	1.9	1.7	1.1	15.7
484	-34.5	65.4	98.3	0.8		1.5	1.8	1.9	1.3	15.7
489	-34.5	65.4	98.3	0.7		1.7	1.8	1.9	1.5	15.7
494	-34.5	65.4	84.3	0.5		1.8	1.8	1.9	1.6	15.7
499	-34.5	0.0	0.0	0.4		1.7	1.7	1.9	1.8	15.7
506	-34.5	0.0	0.0	0.1		1.4	1.6	0.0	2.0	15.7
511	-34.5	0.0	0.0	0.0		1.3	1.5	1.7	2.1	15.7
514	-30.7	0.0	0.0	0.0		1.2	1.5	1.7	2.2	14.4
518	-21.3	0.0	0.0	0.0		1.1	1.0	1.7	2.3	10.5
522	-49.4	0.0	0.0	0.0		1.1	0.6	1.7	2.4	26.2
527	-21.3	0.0	0.0	0.0		0.9	0.0	1.5	2.5	10.5
539	-26.6	0.0	0.0	0.0		0.7	0.0	1.3	2.7	13.1
550	-34.5	0.0	0.0	0.0		0.6	0.0	1.2	2.8	17.0
550	-34.5	0.0	0.0	0.0		0.3	0.0	0.8	3.2	17.0
550	-34.5	0.0	0.0	0.0		0.2	0.0	0.0	3.2	17.0

The Maine	Line number	L. STP number
317.0	12	
323.1	12	
330.4	12	
339.9	12	
348.2	12	
351.4	12	
362.3	12	
371.5	12	
381.8	12	
391.3	12	
398.4	12	
407.8	13	131
411.9	15	131
424.5	17	131
429.9	17	131
429.4	18	
435.4	18	105
442.5	19	131
444.5	19	196
445.0	20	432
446.4	21	523
448.4	22	458
449.3	23	654
451.9	26	628
452.7	27	432
454.7	28	379
455.0	28	340
456.2	29	258
457.1	29	183
460.5		

TO- L. STP
Minutes withdrawn

@Pro-

@Pro-
[/ Block, Values] -

3.4 7.1 11.6 13.2 26.6 29.0 32.2 36.7 39.7 42.1 44.8 48.6 50.2 53.4 56.5 64.1 70.6 83.5 100.6 115.0 125.0 135.0 136.7 141.7 145.6 154.0 160.4 166.7 171.3 176.5 182.1 186.5 190.9 196.1 198.8 204.3 222.6 227.1 232.9 236.1 241.6 246.2 252.6 257.4 262.5 270.3 275.4 279.6 289.1 298.7 306.2 312.0

0.1 15.7
0.3 15.7
0.9 49.7
1.2 1.2

Time Minutes	Temp °C	4	6	8	9	P3	P7	L1	L2	Drym No.	Vaporized Liquid to cool down		PM2 Or/min	PM6 Or/min	Liters Dewar1	Liters Dewar2	Liters L1	Liters L2
											PM4 SCFM	PM1 Or/min						
317.0	255.8	97.1		97.7	96.5	33.0	14.0	0.9		44.1	20.1		65.4	99.8	0.4		0.4	
323.1	255.5	96.3		96.8	96.5	30.0	12.0	0.9		40.5	19.7		65.4	98.3	0.6		0.4	
338.4	254.5	94.8		92.8	95.7	30.0	11.0	1.0		32.0	17.1		65.4	98.3	1.1		0.4	
339.9	253.6	94.8		94.2	97.7	35.0	18.0	1.0		48.0	20.5		65.4	98.3	1.1		0.4	
348.2	252.1	95.7		95.8	95.7	35.0	18.0	1.0		38.7	19.0		65.4	98.3	1.4		0.4	
353.4	251.1	94.9		94.0	97.0	35.0	17.0	4.0		48.0	21.0		65.4	98.3	1.5		1.7	
362.3	248.3	93.9		94.3	97.3	40.0	15.0	4.0		48.0	21.0		65.4	96.8	1.8			
373.5	246.2	93.4		94.3	97.4	35.0	10.0	4.5		38.5	18.5		65.4	99.1	2.1		1.9	
383.8	244.0	92.3		94.6	97.6	40.0	18.0	4.8		40.5	19.7		65.4	107.2	2.1		2.0	
393.3	241.0	92.1		94.6	97.5	38.0	10.0	4.8		35.3	18.1		65.4	98.3	2.1		2.0	
398.4	240.4	92.3		95.3	97.9	43.0	13.0	4.8		35.3	18.1		93.5	112.1	2.1		2.0	
403.8	238.8	92.0		95.2	98.1	40.0	12.0	4.2		35.3	18.1	-28.7	87.8	117.3	2.0		1.8	
411.9	237.9	92.1		95.4	91.1	40.0	10.0	3.5		26.3	16.6	-31.5	65.4	118.5	1.9		1.5	
424.5	237.1	92.1		96.3	98.9	48.0	12.0	4.0		18.8	14.5		69.2	108.2	1.8		1.6	
425.9	236.0	91.7		96.3	98.8	50.0	10.0	3.5		26.3	16.6	-29.9	65.4	98.3	1.7		1.5	
429.4								4.5		18.2	14.5		65.4	101.1	1.8		1.9	
435.4	235.0	94.3		96.2	97.8	48.0	15.0	4.0		4.5	22.3	-15.0			1.8		1.7	
442.5								3.5		3.5		-18.7			1.7		1.5	
445.0								3.5		3.5		-28.0			1.7		1.5	
445.0								3.5		3.5		-116.8			1.7		1.5	
446.4								3.0		3.0		-74.8			1.7		1.4	
448.4								2.9		2.8		-65.4			1.6		1.3	
449.3								2.9		2.9		-143.6			1.5		1.2	
451.9								2.5		2.5		-127.5			1.4		1.0	
452.7								2.4		2.4		-71.4			1.4		1.0	
454.7								2.5		2.5		-62.7			1.3		0.8	
455.0								2.5		2.5		-51.8			1.3		0.8	
456.2								1.9				-41.1			1.3		0.7	
457.1												-26.2			1.2		0.7	
460.5																		

Time Minutes	Temperatures 3	4	6	8	9	P3	P7	L1	L2	Drymo No.	Vaporized Liquid to cool down				FM6 Gr/min	Liers Dewar1	1.00 Liers Dewar2	Liers L1	Liers L2
											FM4 SCFM	-FM1 Gr/min	FM2 Gr/min	FM2 Gr/min					
3.4	283.0	284.5		285.1	268.0	17.0	22.0			94.5	11.5								
7.1	284.3	283.0		281.5	263.0	18.0	23.0			88.4	11.3								
11.6	283.0	281.8		282.0	272.5	18.0	24.0			91.1	11.3								
15.2	282.0	280.0		280.0	265.0	18.0	23.0			91.1	11.3								
26.6	281.1	282.2		276.2	273.8	18.0	23.0			97.6	20.5								
29.0	284.4	277.0		275.9	272.5	18.0	23.0			97.6	20.7								
32.2	282.0	274.5		275.5	256.0	18.0	23.0			94.5	20.5								
36.7	284.0	275.3		275.2	269.8					94.5	20.5								
39.7	285.0	274.7		274.7	258.6					97.6	20.2								
42.1	285.0	274.0		274.2	255.7					94.1	19.9								
44.8	285.0	275.0		274.0	271.0					101.2	20.8								
48.6	285.0	274.0		278.2	279.5					97.6	20.5								
50.2	285.0	273.0		277.0	280.0					97.6	20.2								
53.4	285.0	273.4		276.3	257.0					94.5	20.2								
56.5	285.0	272.2		273.6	265.2					91.7	20.2								
64.1	283.4	270.0		267.8	245.5	31.0	13.0			94.1	19.9								
70.6	283.0	260.1		256.2	230.0	30.0	12.0			62.4	19.5								
83.5	282.3	241.1		237.7	212.8	28.0	12.0			62.4	19.7								
100.6	280.8	219.3		216.2	193.0					58.8	19.6								
113.0	280.0	208.5		205.8	186.7	27.0	12.0			54.2	18.8								
125.0	278.0	207.7		204.8	183.4	32.0	20.0			56.3	21.9								
135.0	277.7	200.6		197.8	174.8	35.0	15.0			46.1	20.2								
136.7	276.6	191.9		189.0	165.3	38.0	14.0			53.2	21.2								
141.7	275.9	187.1		184.1	160.4	37.0	13.0			40.5	19.4								
145.6	275.3	183.4		180.1	157.1	37.0	13.0			43.2	19.8								
154.0	273.9	174.3		171.0	147.7	37.0	14.0			46.1	21.0								
160.4	273.0	168.7		166.2	144.7	35.0	13.0			46.0	20.1								
166.7	271.9	163.5		161.2	140.5	35.0	13.0			50.0	21.9								
171.3	271.3	159.1		156.8	136.5	35.0	13.0			50.0	21.3								
176.5	270.7	156.3		154.1	134.3	35.0	13.0			40.0	21.0								
182.1	269.9	152.0		149.8	131.2	33.0	13.0			50.0	21.6								
186.5	269.4	149.2		147.0	127.6	33.0	13.0			48.0	21.0								
190.6	268.8	146.6		144.1	124.7	33.0	13.0			48.0	20.7								
196.1	267.4	139.7		137.2	119.0	36.0	10.0			44.1	19.8								
198.8	267.4	139.7		137.2	119.0	36.0	20.0			50.0	21.5								
204.3	266.6	134.7		132.9	115.0	34.0	15.0			53.2	21.6								
222.6	263.5	127.5		126.0	108.0	30.0	12.0			51.0	21.6								
227.1	262.0	125.5		124.8	102.9	35.0	20.0			57.3	22.9								
232.9	260.0	122.4		121.3	103.0	33.0	19.0			58.5	22.9								
236.1	260.5	120.5		119.7	101.5	33.0	18.0			60.8	23.1								
241.6	259.8	117.8		116.9	98.8	34.0	19.0			61.1	23.0								
246.2	259.1	115.5		114.7	96.9	32.0	18.0			67.7	24.8								
252.6	257.9	111.9		110.9	93.4	32.0	18.0			65.3	24.1								
257.4	257.2	109.7		108.7	92.0	32.0	12.3			64.4	23.7								
262.5	256.2	106.8		105.8	89.3	32.0	18.0			63.0	23.3								
270.3	254.7	102.7	101.4	101.4	89.2	30.0	18.0			63.0	23.7								
275.4	253.8	100.3		99.2	86.7	30.0	17.0			46.1	20.5								
279.6	253.5	98.2		96.6	87.6	40.0	14.0			46.1	20.5								
289.1	254.1	97.6		97.3	87.6	34.0	12.0			40.5	19.4								
298.7	255.2	97.7		98.2	87.6	31.0	11.0			58.3	23.3								
308.2	256.0	98.4		99.0	88.3	35.0	13.0			57.5	22.4								
312.0	256.0	98.1		98.8	87.2	35.0	15.0	0.6		49.0	21.0							0.3	

02 Test No. 6

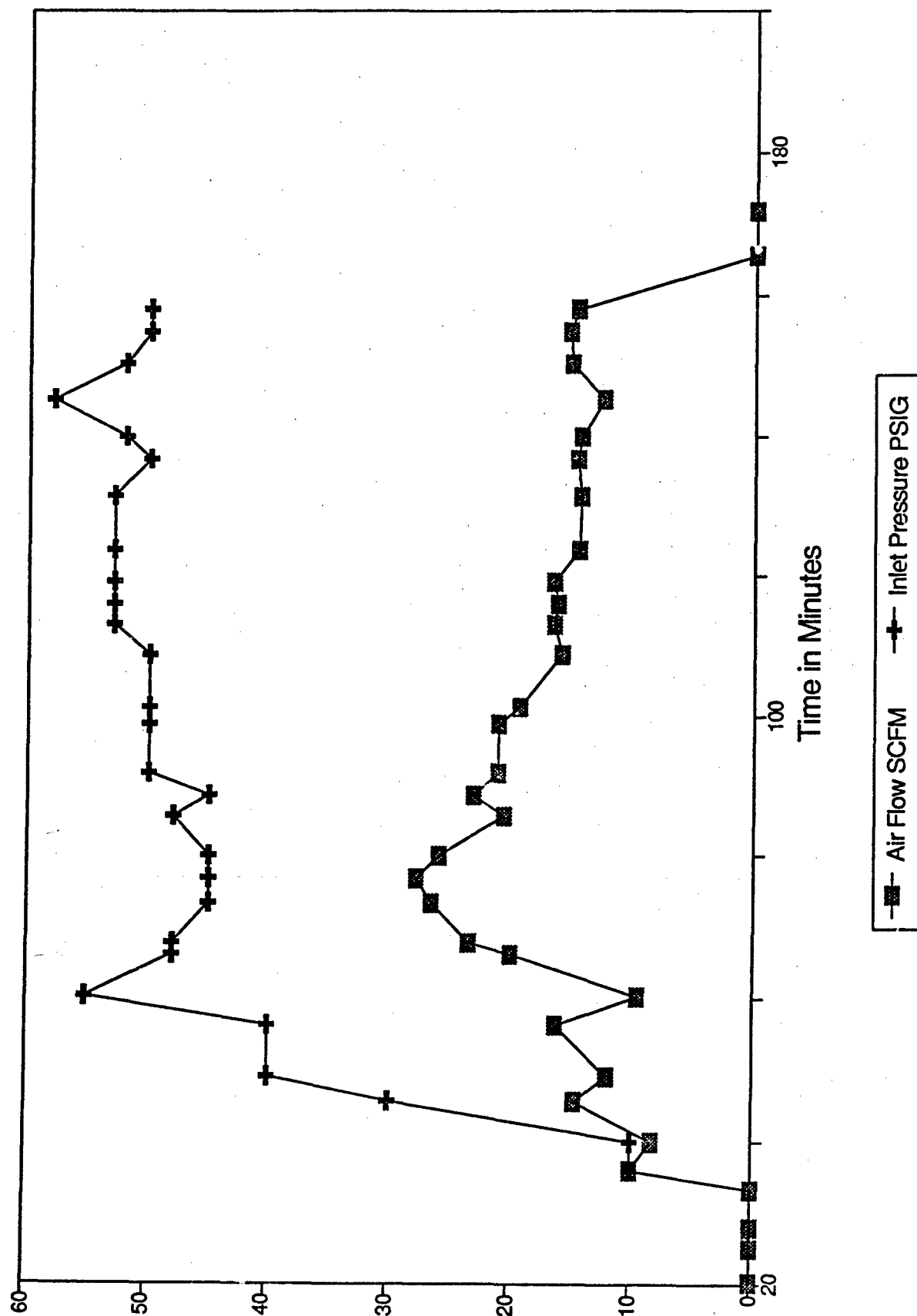
TD- Minutes	14:25 Time	%O2 FM6 supply	%O2 FM2 vent	%O2 FM1 withdraw	Q FM4	Psig Ra2	Q FM-5	Psig Ra5	Q FM2	Q FM1	Psig RA-FM1	1	2
3.4	12:14:49 PM												
7.1	12:18:16 PM											285.0	281.8
11.6	12:21:55 PM											279.0	283.0
15.2	12:26:22 PM											271.0	281.0
20.6	12:30:02 PM											271.9	277.0
29.0	12:41:23 PM											271.9	275.5
32.2	12:43:50 PM											277.0	275.9
36.7	12:46:58 PM											270.0	275.3
39.7	12:51:30 PM											268.9	274.5
42.1	12:54:31 PM											269.3	274.7
44.8	12:56:55 PM											267.6	273.8
48.6	12:59:38 PM											267.8	273.4
50.2	01:03:22 PM											270.9	273.6
53.4	01:04:59 PM											269.0	273.0
56.5	01:08:15 PM											268.8	273.4
64.1	01:11:17 PM											268.4	272.4
70.6	01:18:56 PM											270.0	271.1
83.5	01:25:25 PM											269.4	268.2
100.6	01:38:16 PM											268.2	268.2
115.0	01:55:21 PM											258.9	232.2
125.0												241.9	231.1
135.0												231.0	219.8
136.7	02:29:49 PM											229.0	218.6
141.7	02:31:31 PM											223.0	211.7
145.6	02:36:29 PM											214.9	203.5
154.0	02:40:22 PM											210.4	199.0
160.4	02:48:50 PM											206.8	195.2
166.7	02:55:12 PM											198.7	187.2
171.3	03:01:32 PM											193.0	180.5
176.5	03:06:04 PM											186.9	174.3
182.1	03:11:21 PM											183.1	170.4
186.5	03:16:55 PM											178.9	166.1
190.6	03:21:19 PM											174.2	161.3
196.1	03:25:23 PM											171.1	158.1
198.8	03:30:32 PM											168.1	155.4
204.3	03:33:35 PM											162.1	149.3
222.6	03:39:06 PM											162.1	149.3
227.1	03:57:26 PM											159.3	146.0
232.9	04:01:58 PM											148.9	135.8
236.1	04:07:42 PM											147.0	134.3
241.6	04:10:52 PM											143.0	130.3
246.2	04:16:24 PM											140.2	127.8
252.6	04:21:03 PM											137.7	125.3
257.4	04:27:27 PM											134.9	122.7
262.5	04:32:10 PM											130.3	118.6
262.5	04:37:20 PM											127.2	116.0
270.3	04:43:09 PM											124.0	112.9
273.4	04:50:00 PM											119.3	108.3
279.6	04:54:22 PM											116.8	106.0
289.1	05:03:53 PM											116.9	106.2
298.7	05:13:28 PM											118.7	106.7
306.2	05:20:58 PM											119.7	106.6
312.0	05:26:49 PM											119.8	106.6

@no--
@no--
(/ Block;Values)--

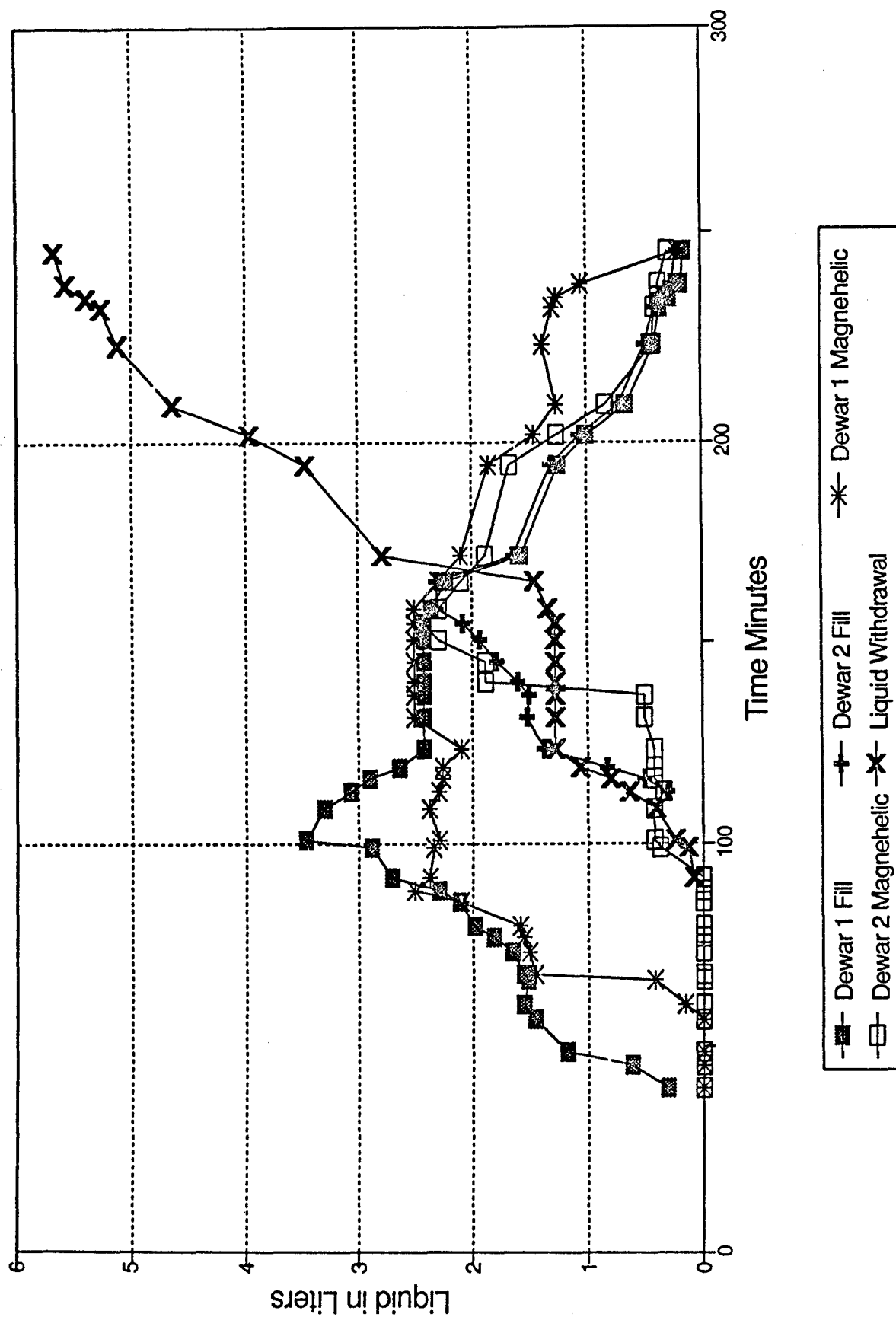
7.6.2 Test Data Graphs

02 Test No. 1 Graphs

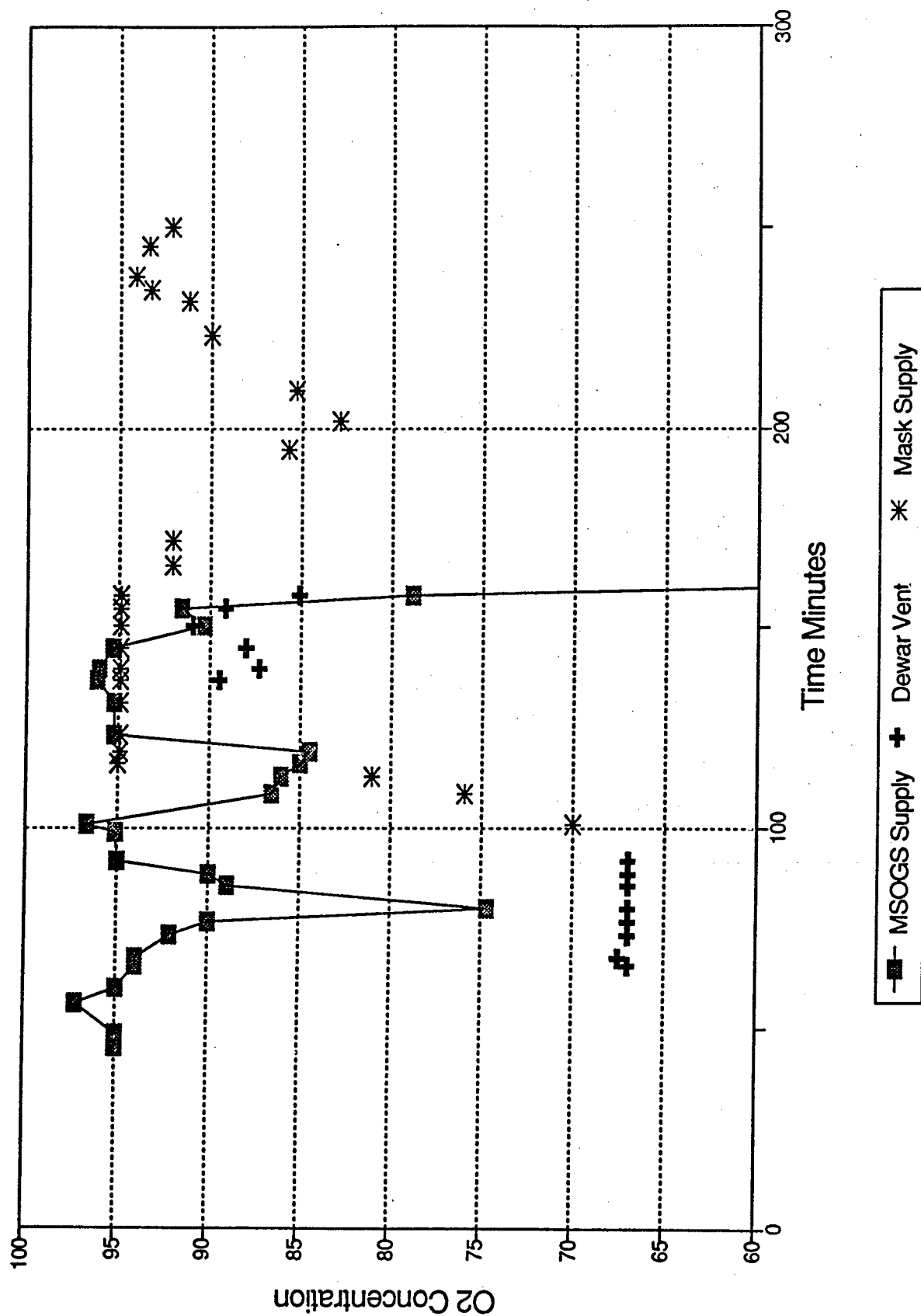
Heat Exchanger Deriming O2Test #1



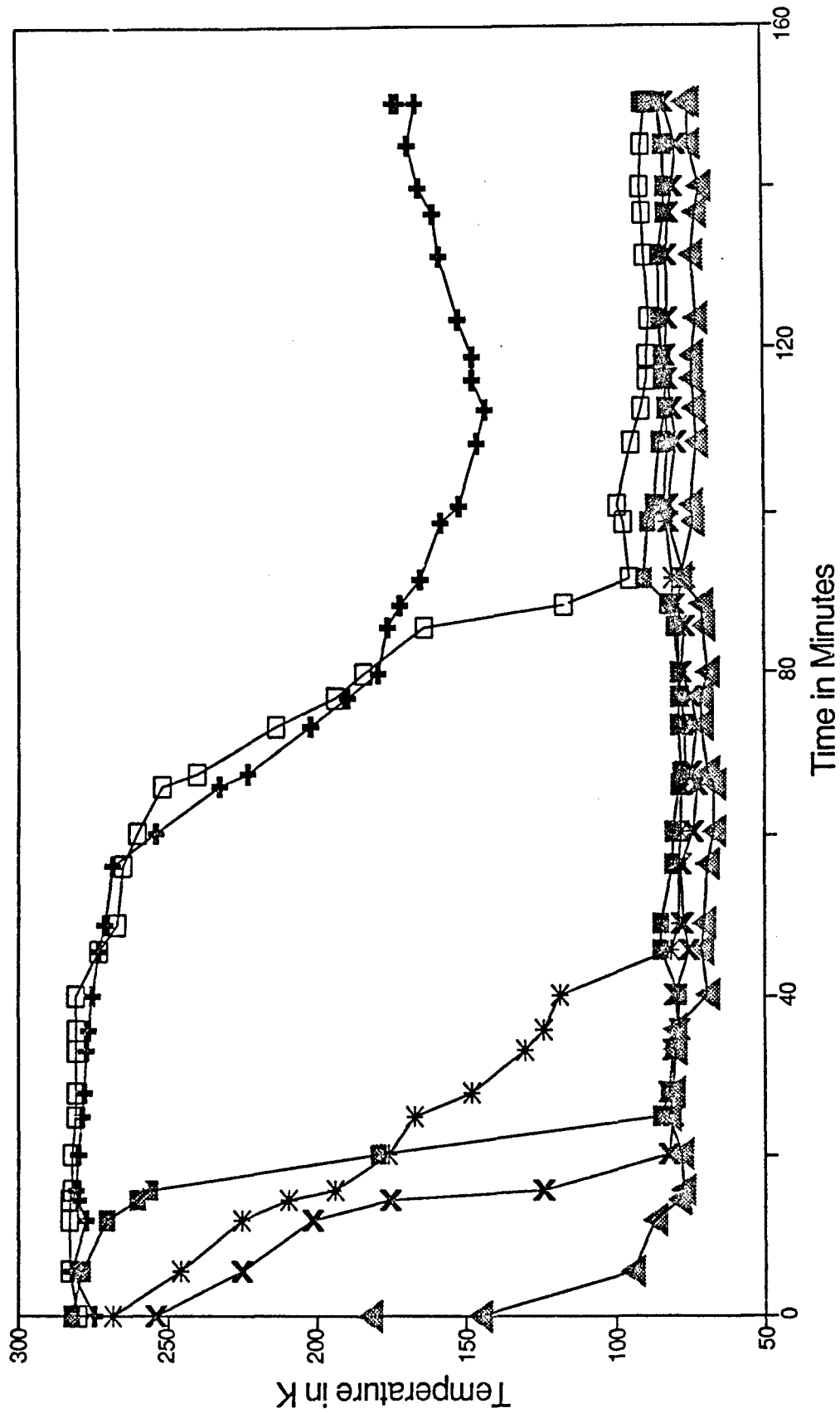
Dewar Fill O2 Test #1



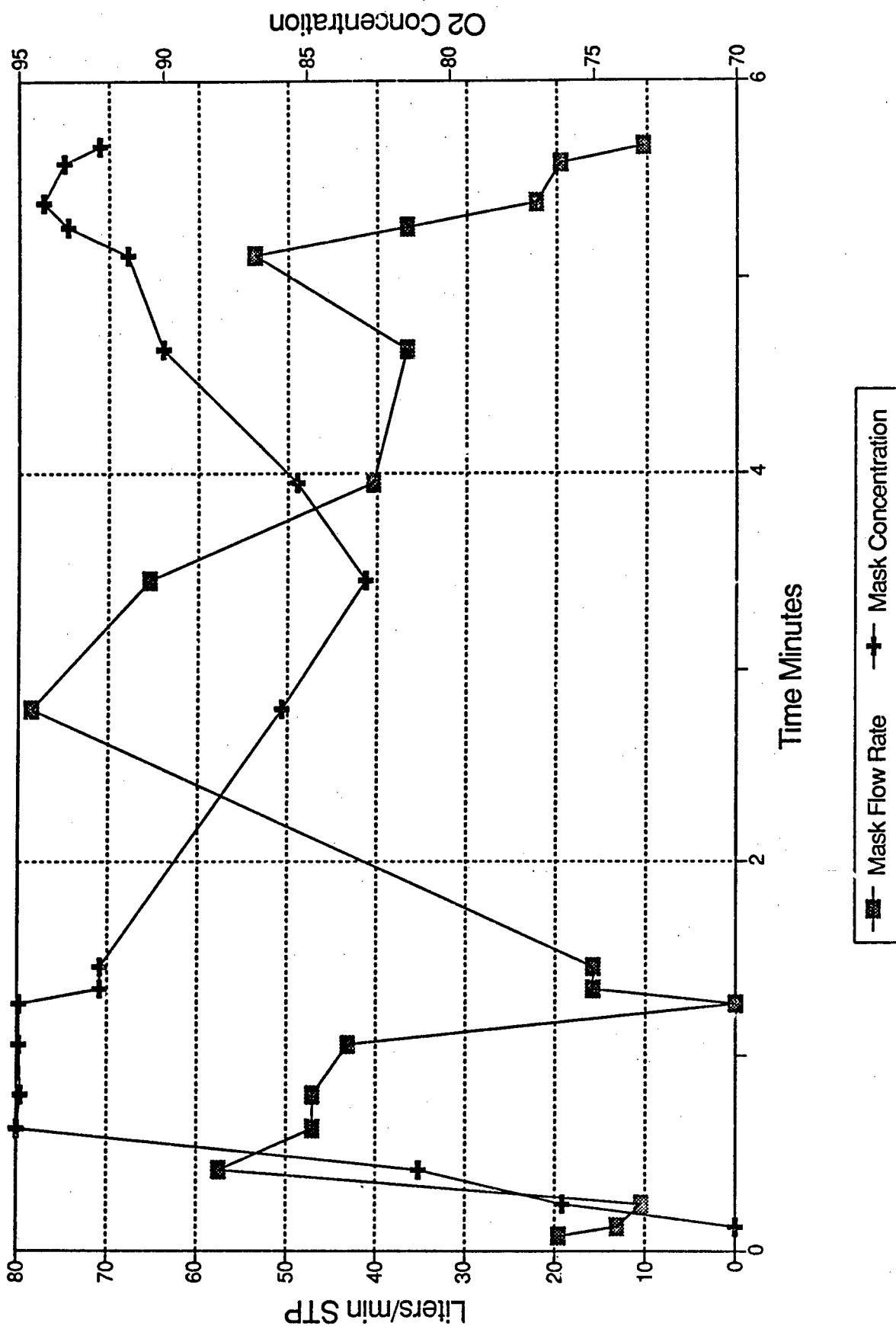
O2 Concentrations O2 Test #1



Cooling History O2 Test #1

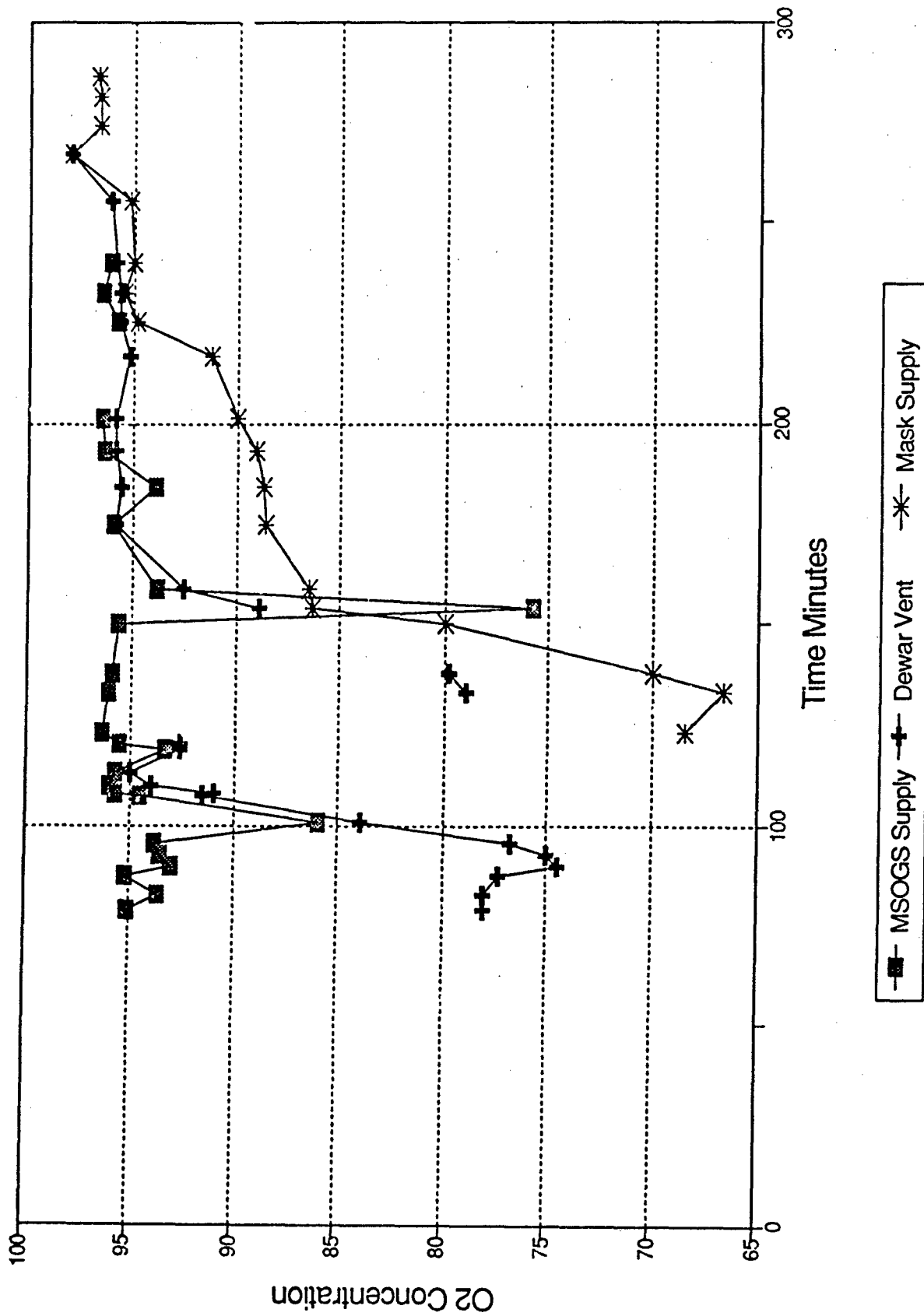


O2 Concentrations O2 Test #1

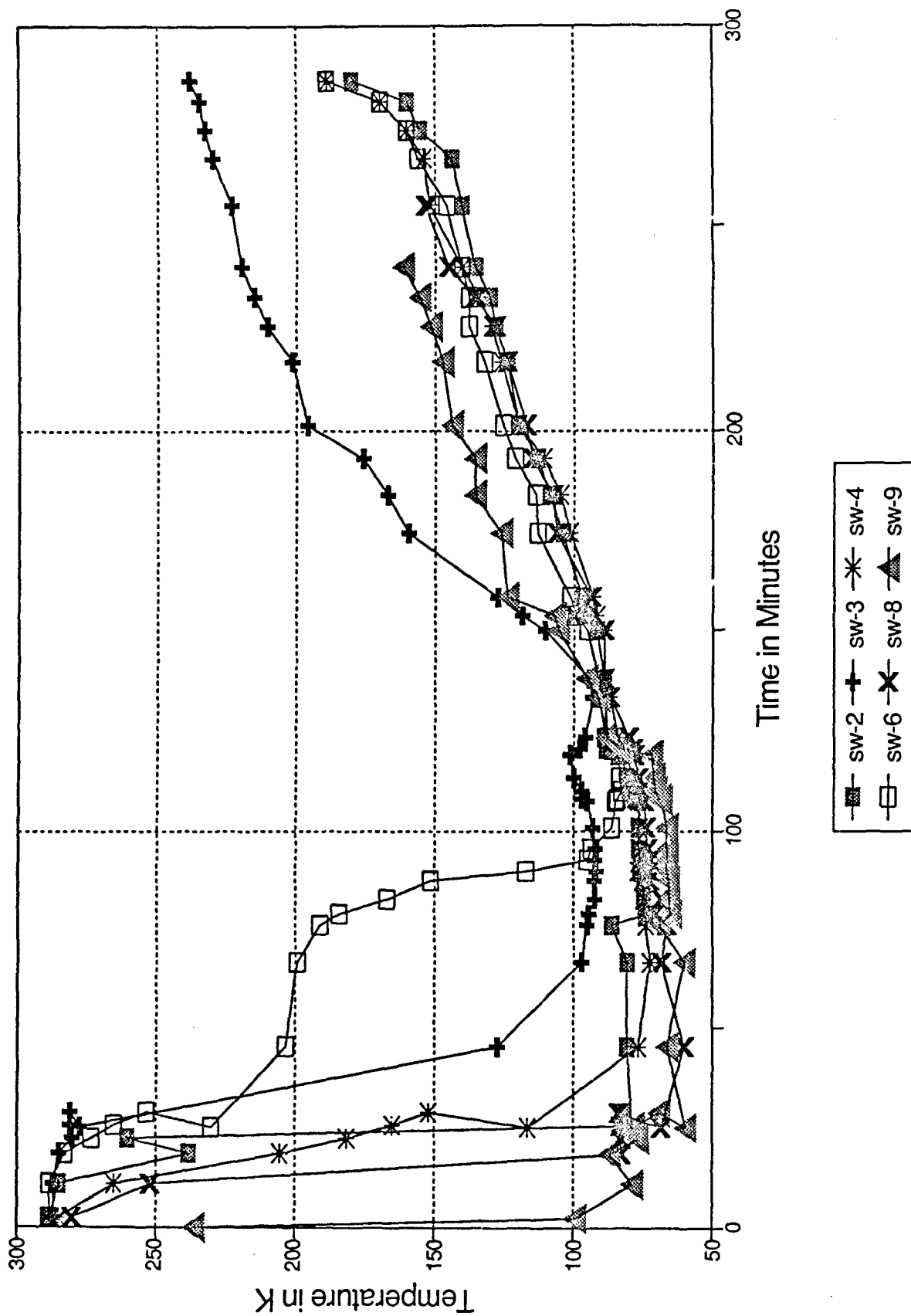


02 Test No. 2 Graphs

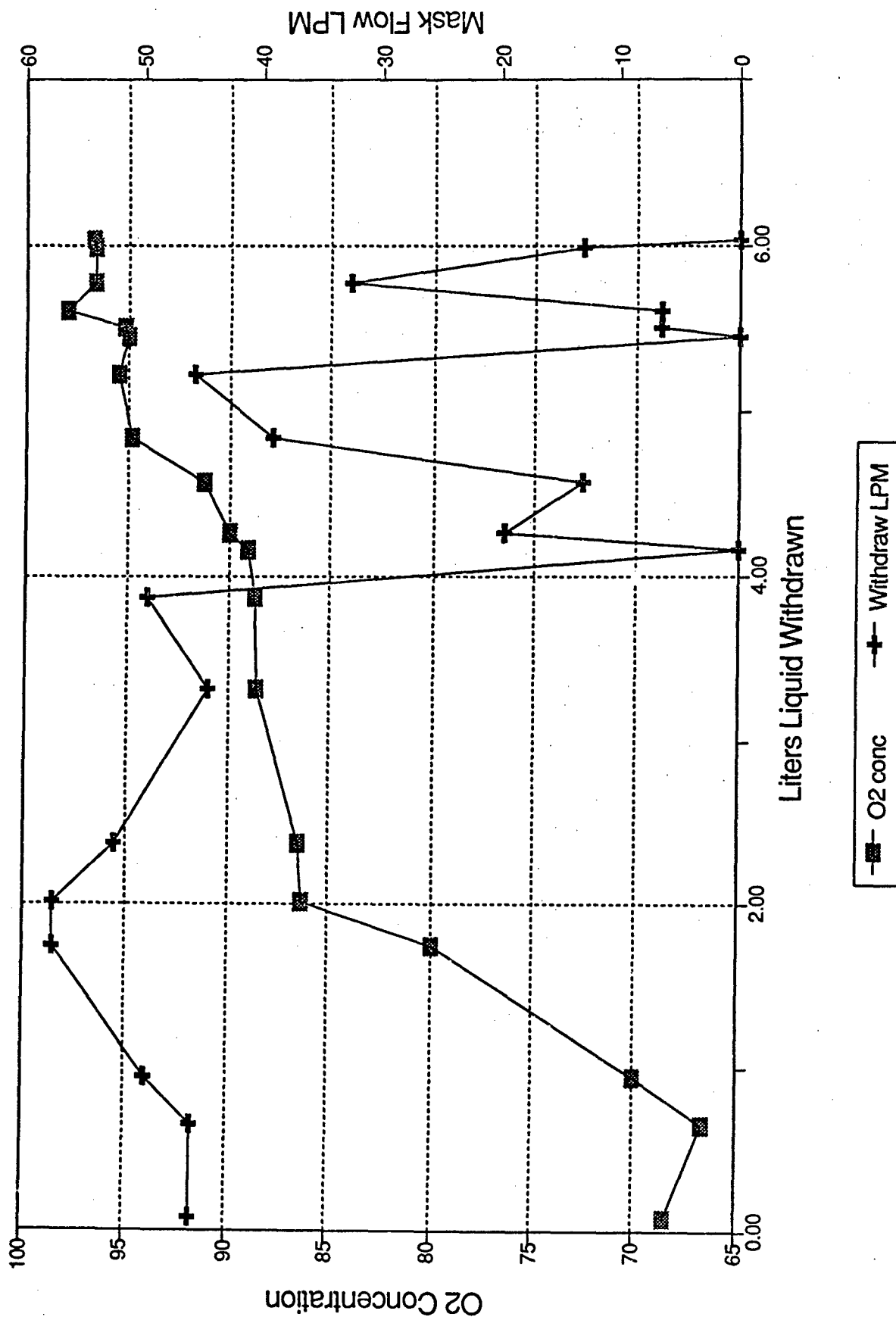
O2 Concentrations O2 Test #2



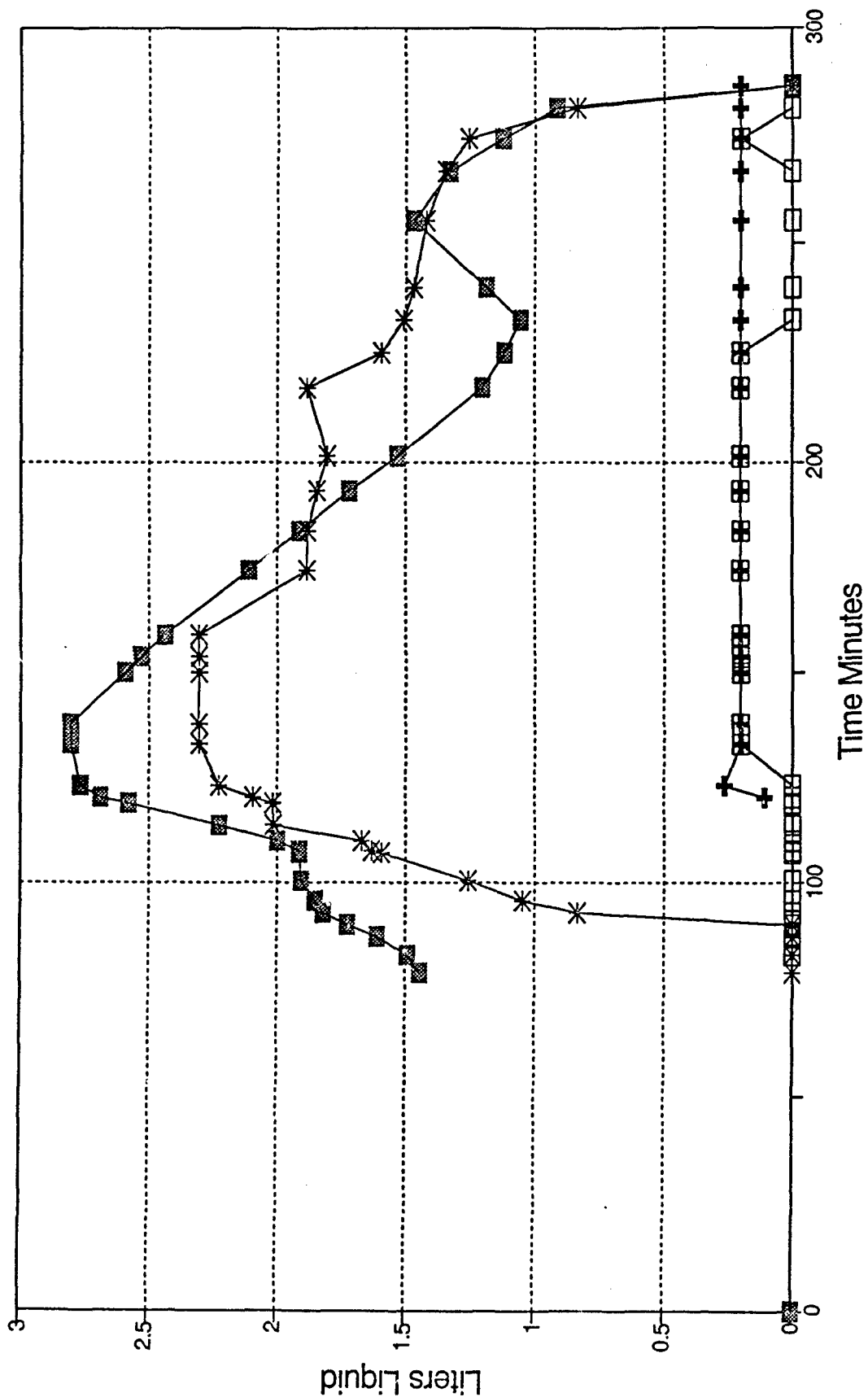
Cooling History O2 Test #2



Mask Flow O2 Test #2

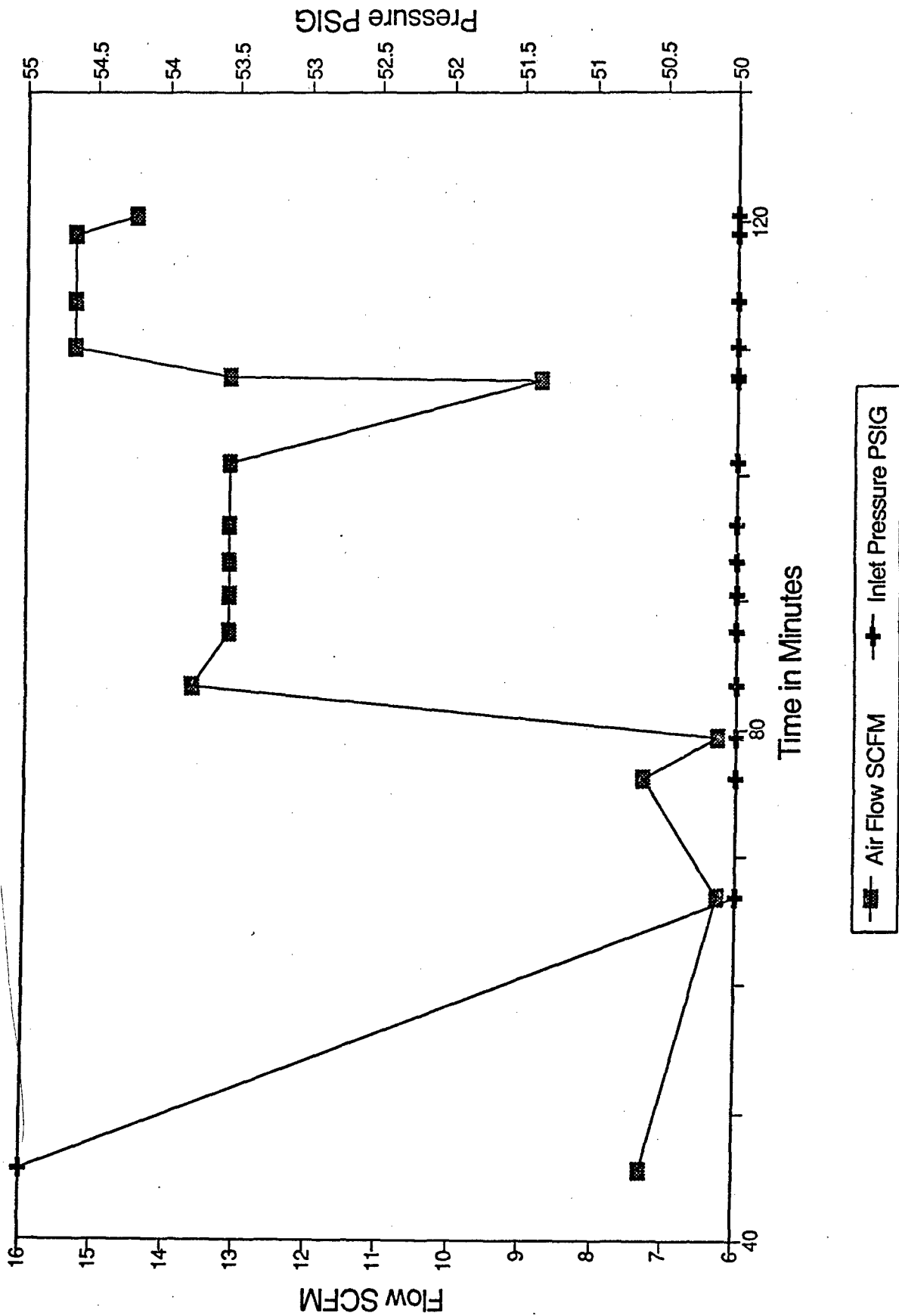


Dewar Filling-Oxygen O2 Test #2



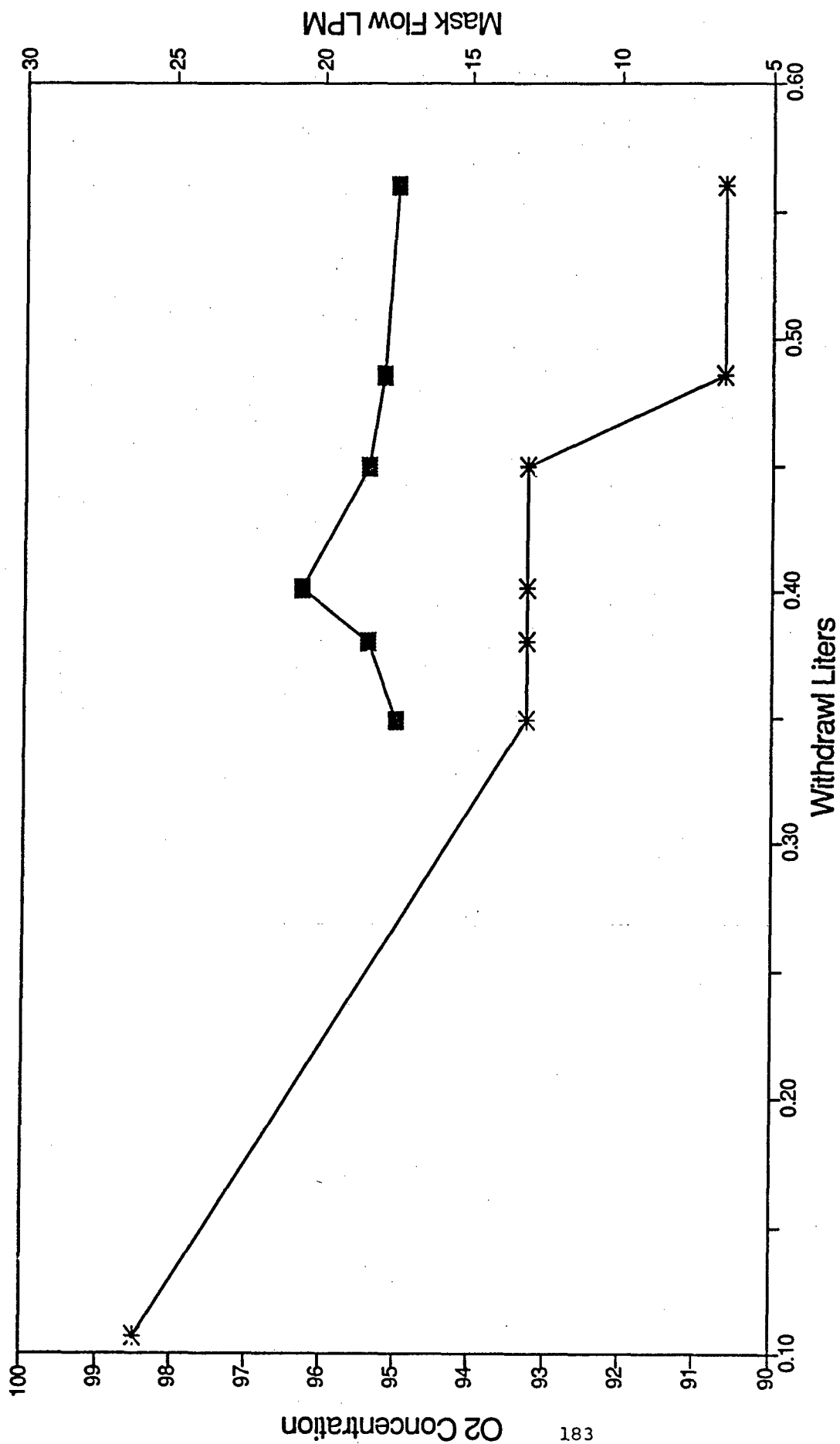
Dewar 1-flow
 Dewar 2-flow
 Magnehelic Dewar 1
 Magnehelic Dewar 2

Heat Exchanger Deriming O2 Test #2

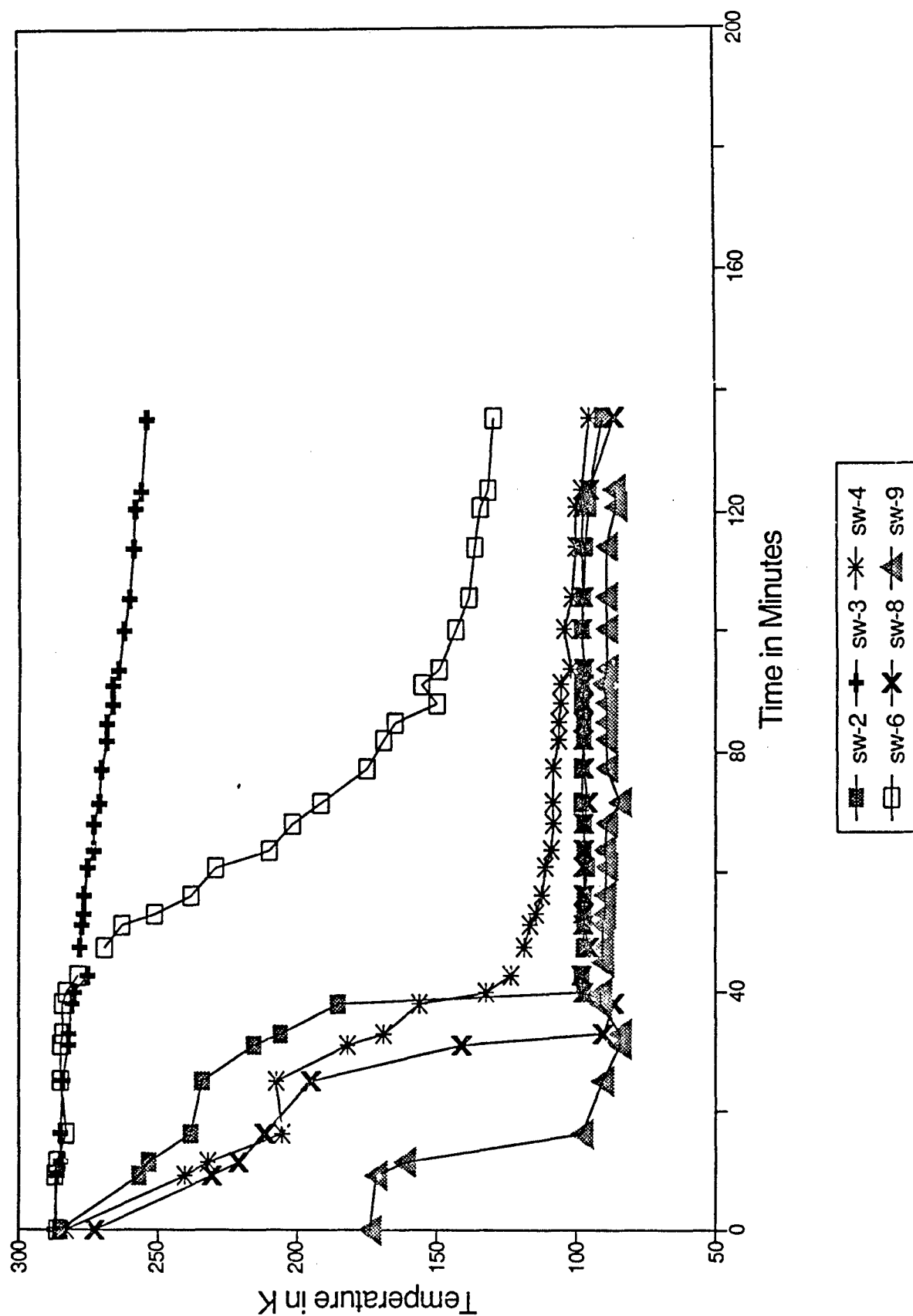


02 Test No. 3 Graphs

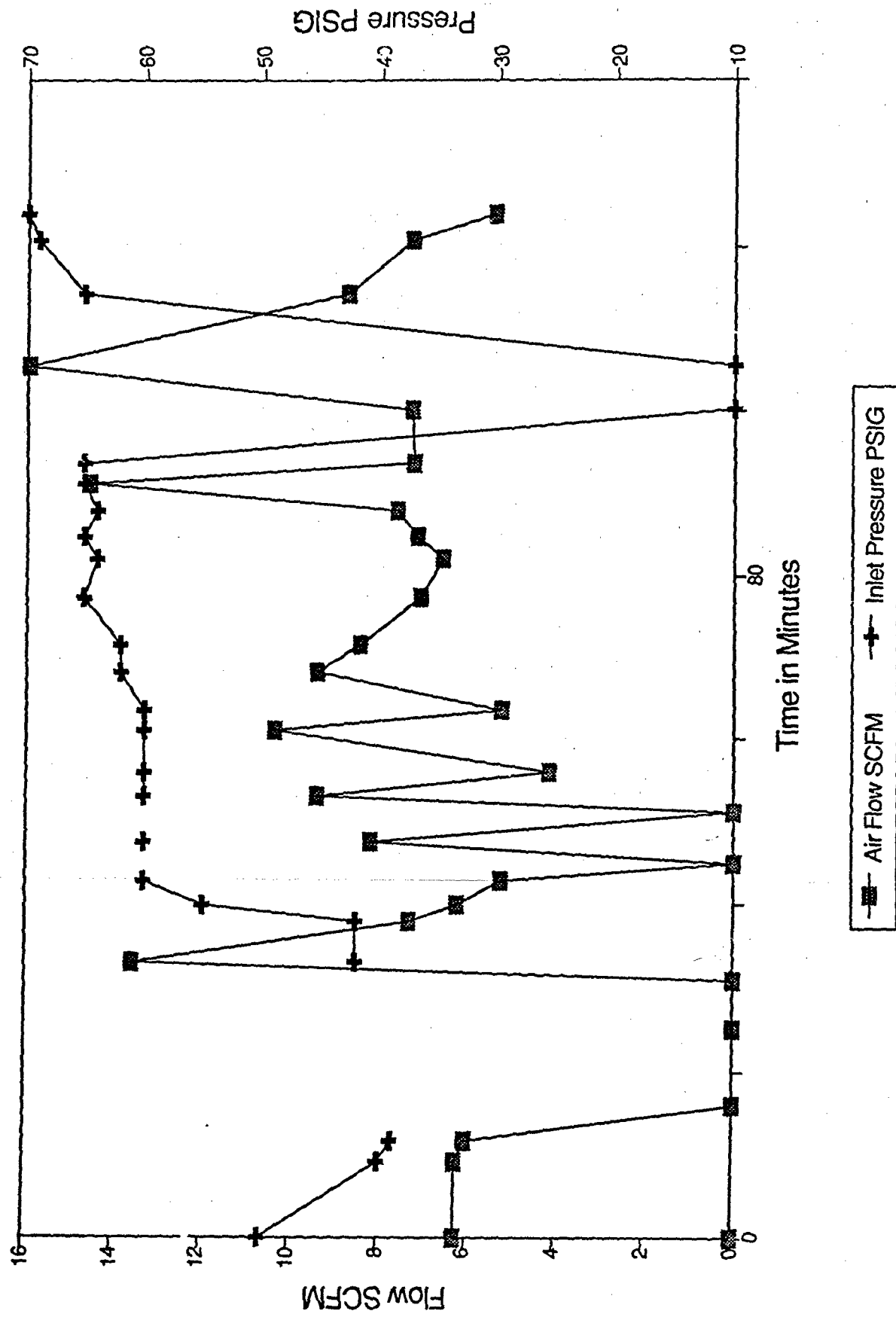
O2 Concentrations O2 Test #3



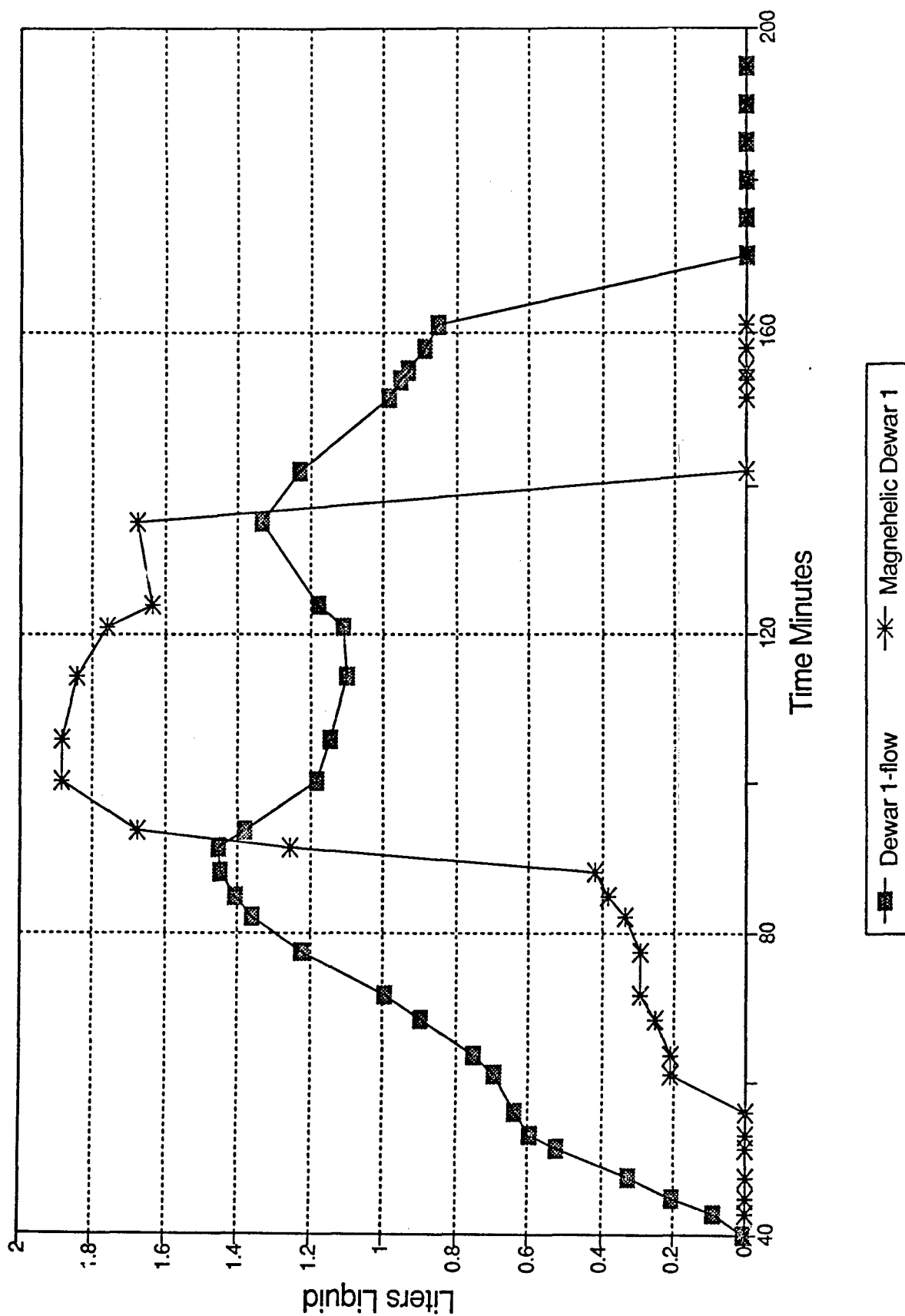
Cooling History O2 Test #3



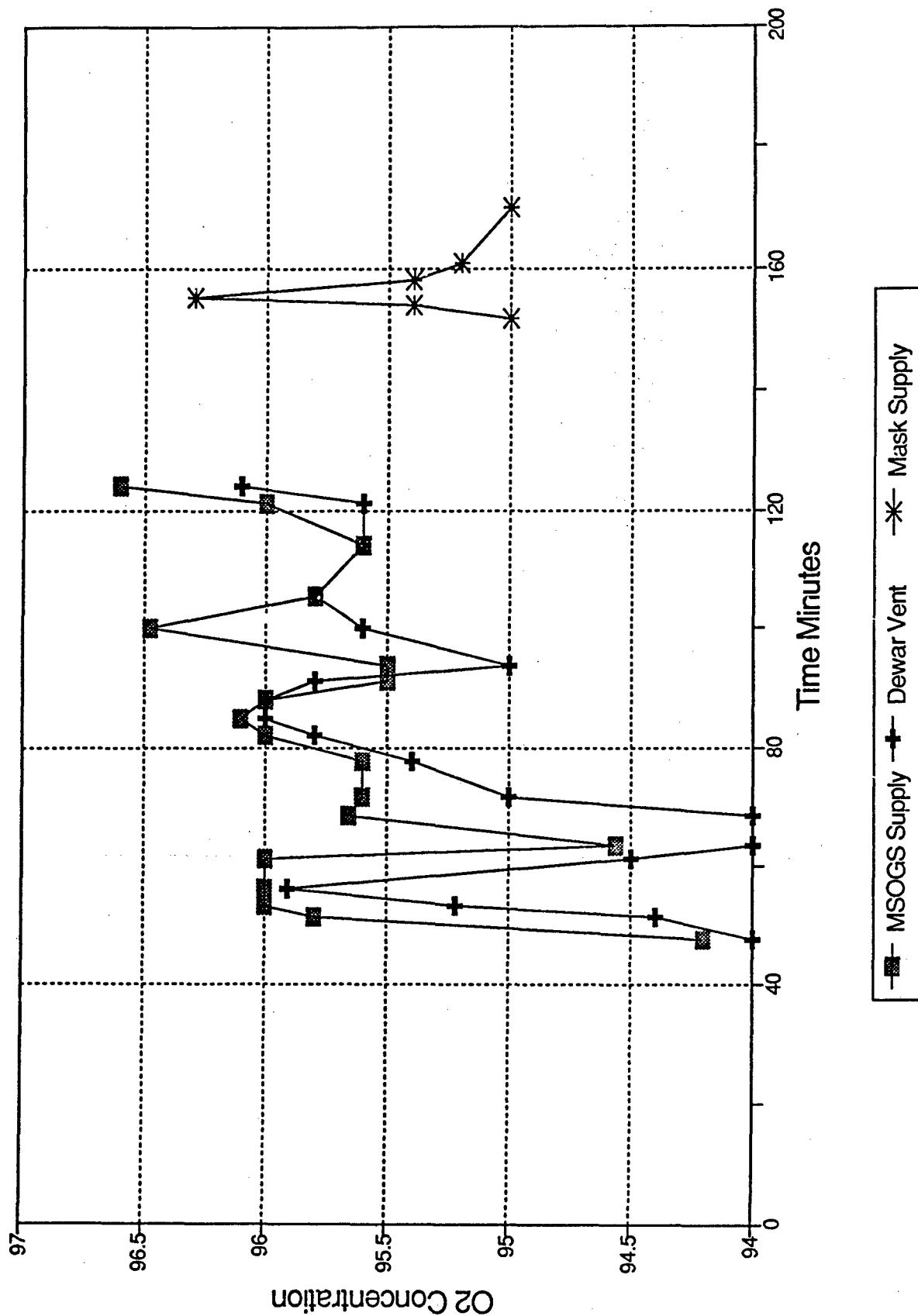
Heat Exchanger Deriming O2 Test #3



Dewar Filling-Oxygen O2 Test #3

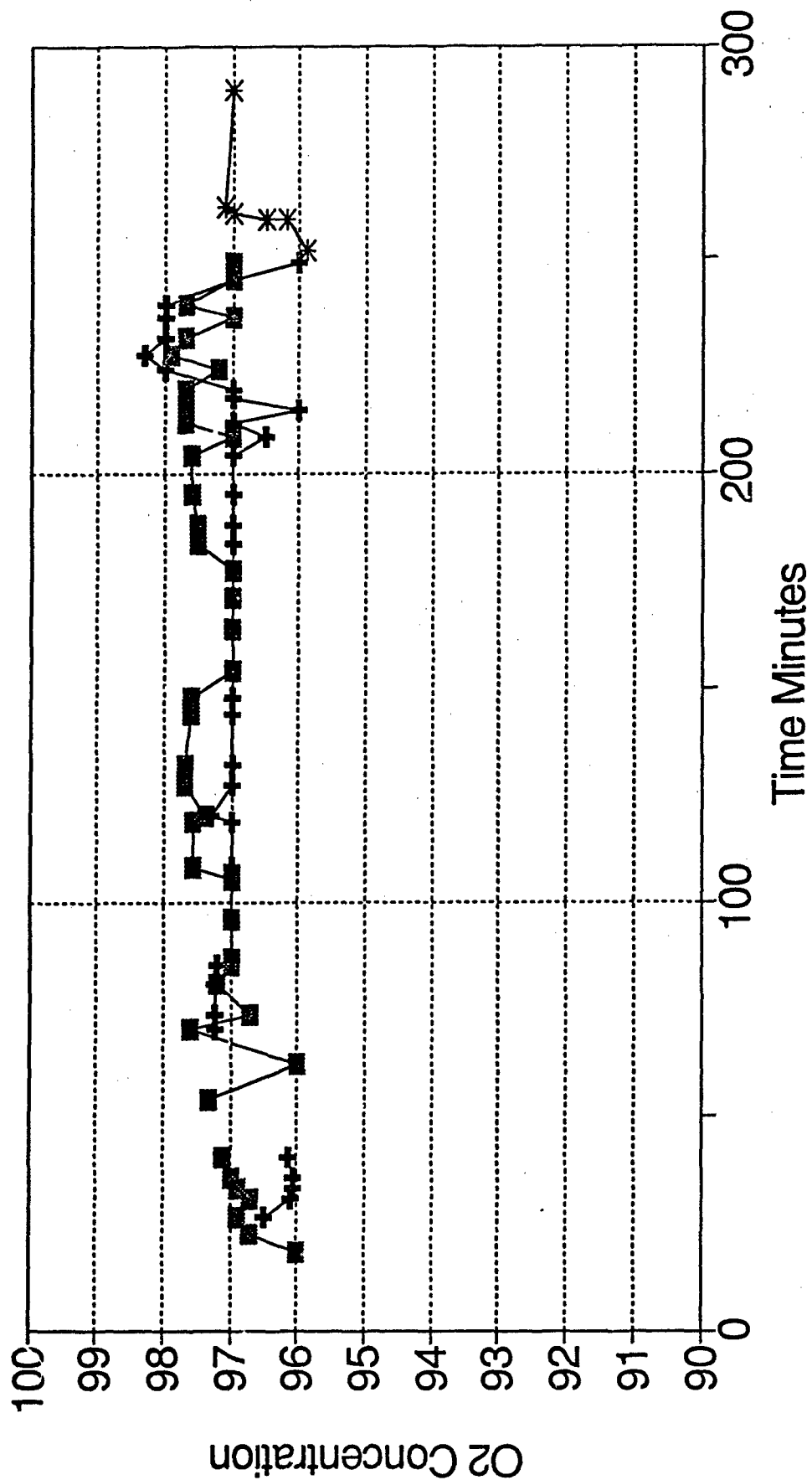


O2 Concentrations O2 Test #3



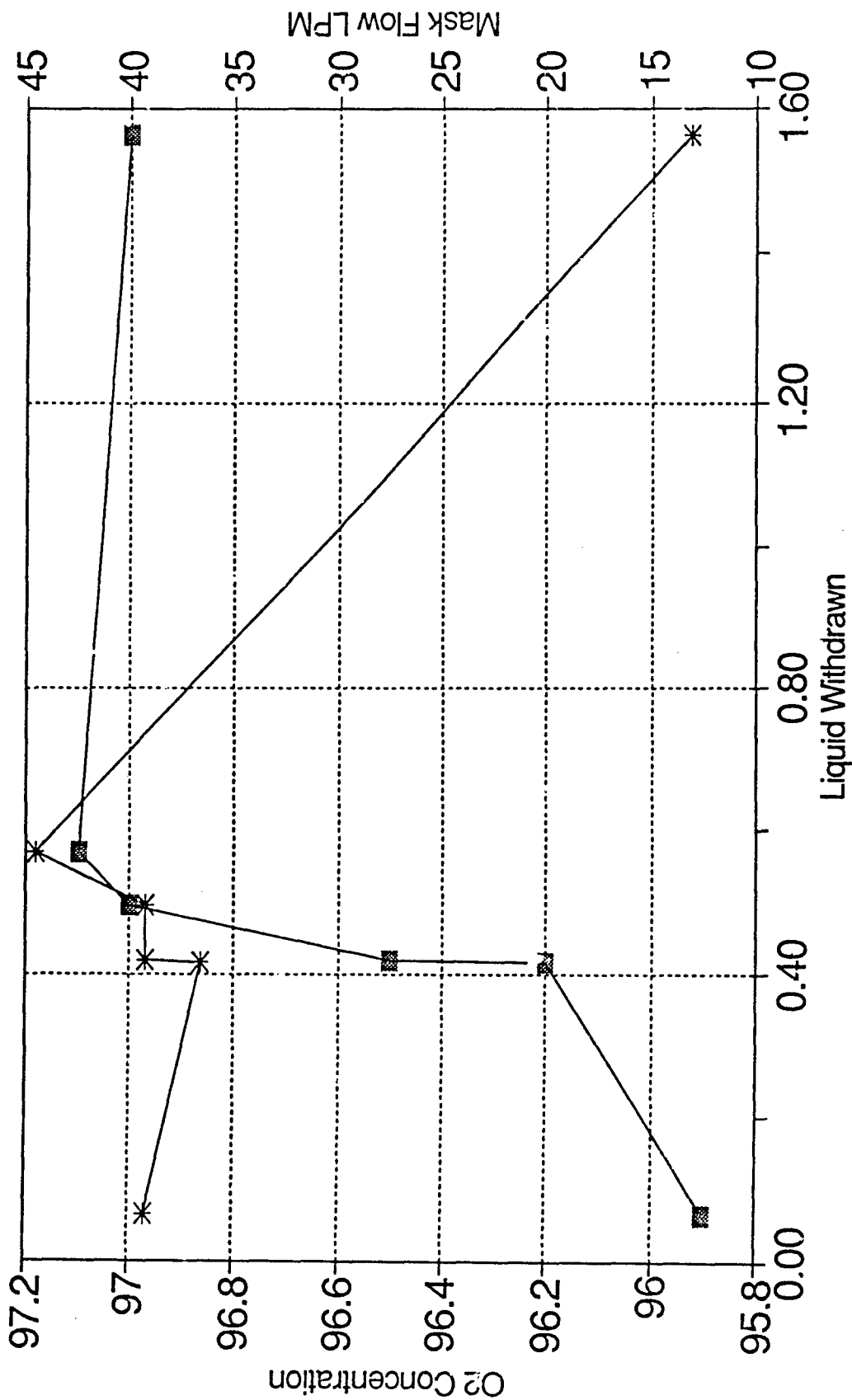
02 Test No. 4 Graphs

O2 Concentrations O2 Test # 4



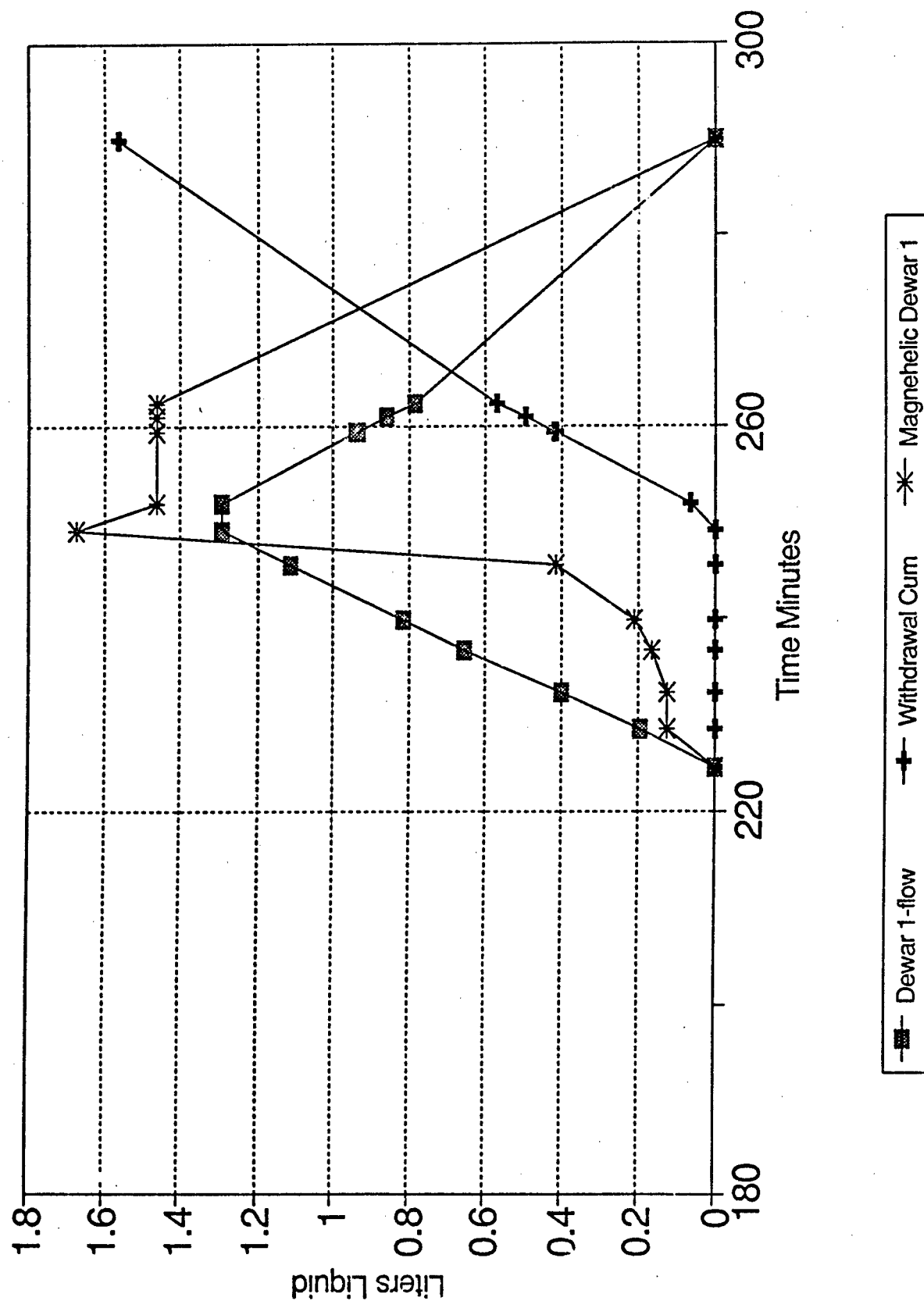
MSOGS Supply
 Dewar Vent
 Mask Supply

O2 Concentrations O2 Test #4

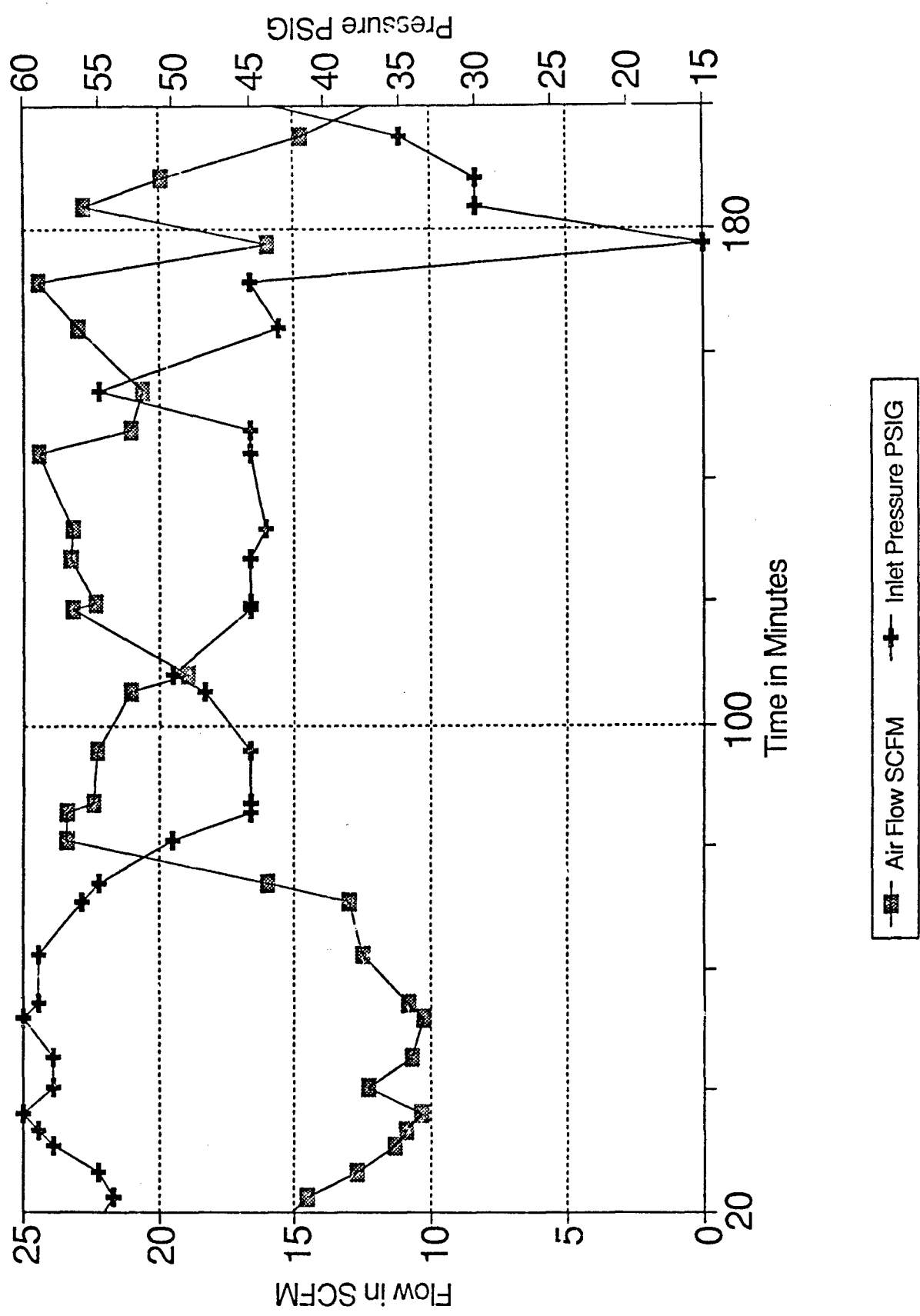


O2 conc
 Mask Supply

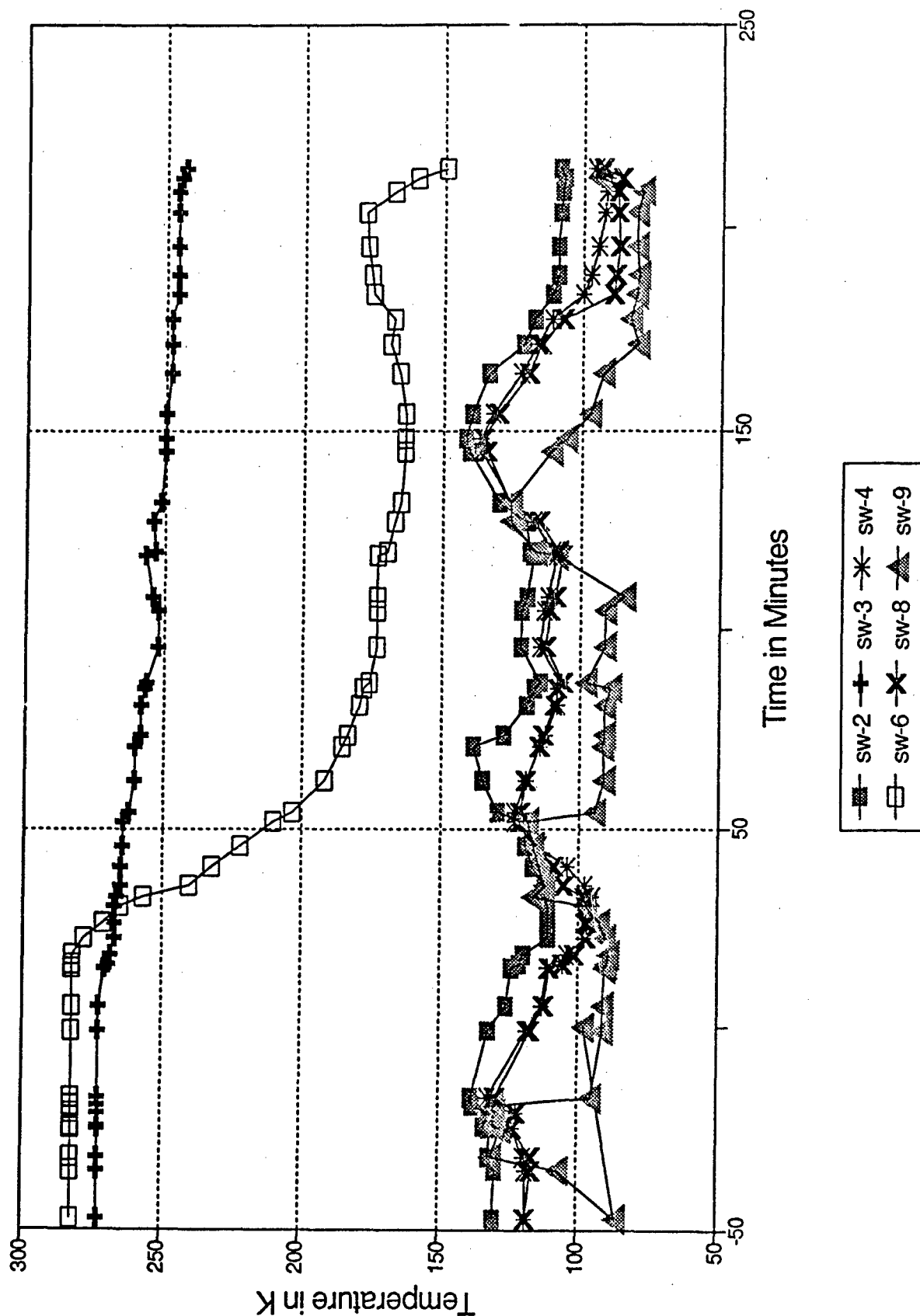
Dewar Filling-Oxygen O2 Test #4



Heat Exchanger Deriming
O2 Test # 4



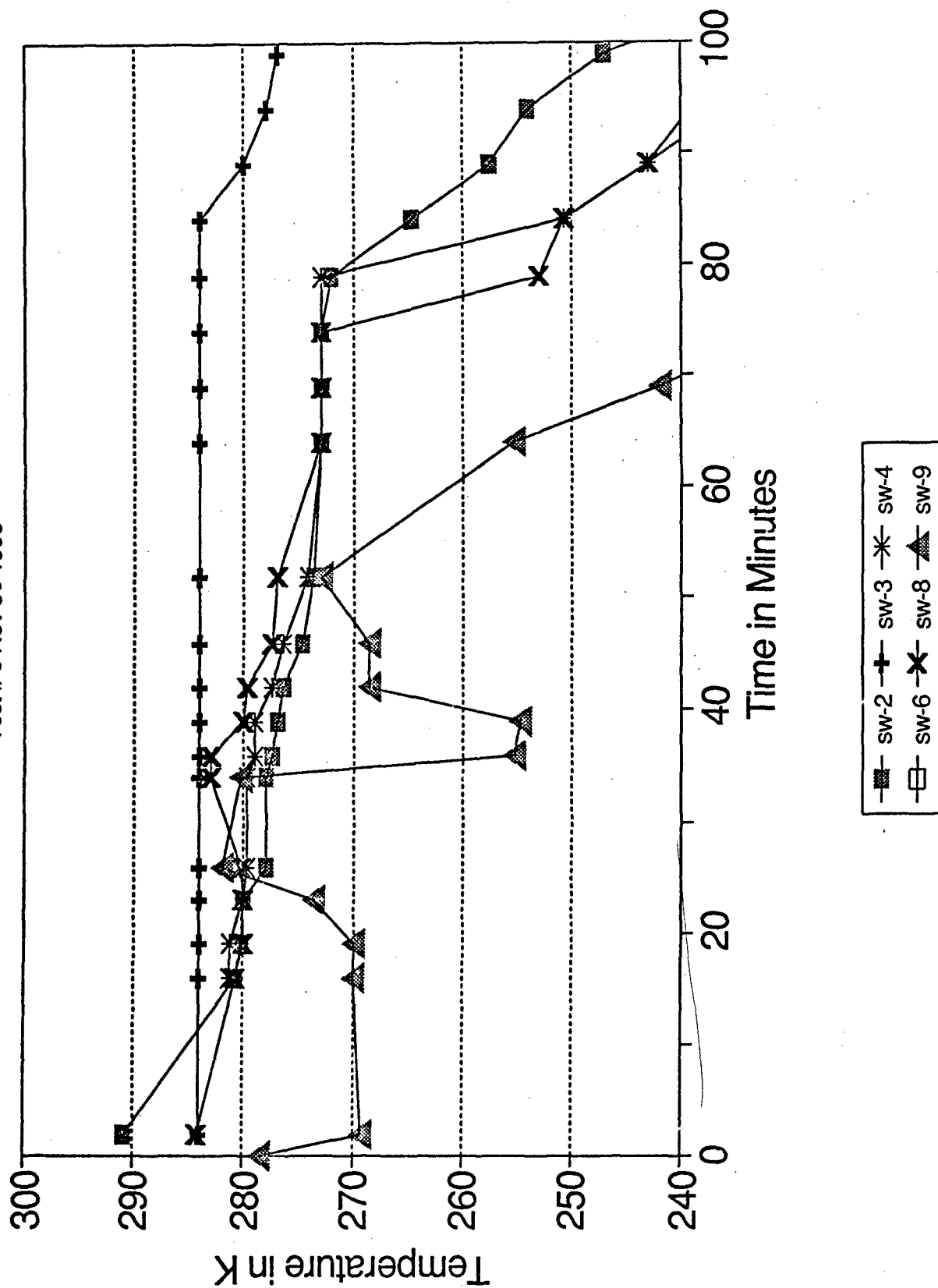
Cooling History O2 Test #4



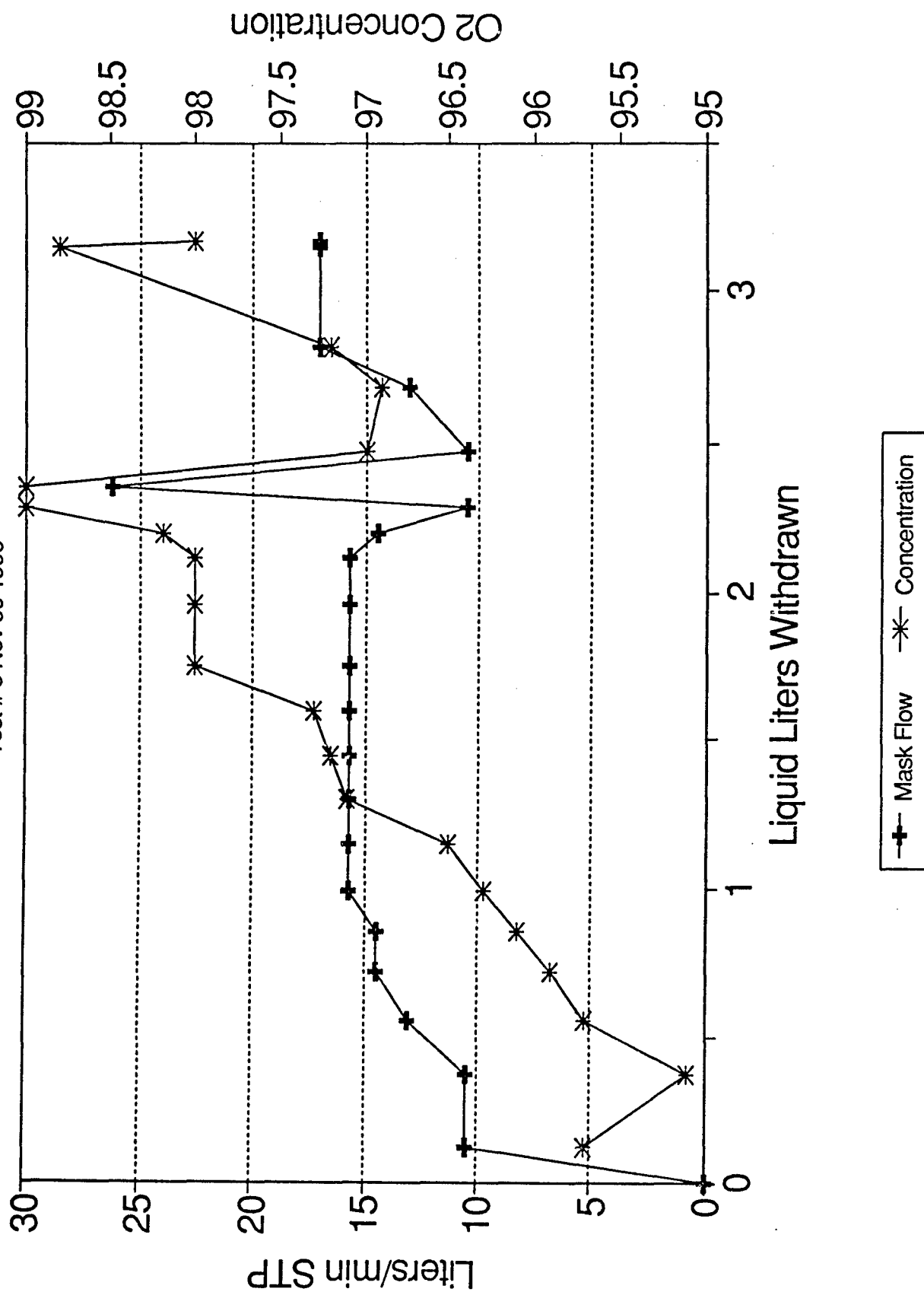
02 Test No. 5 Graphs

Cooling History-Initial Cooldown

Test # 5 Nov 30 1990

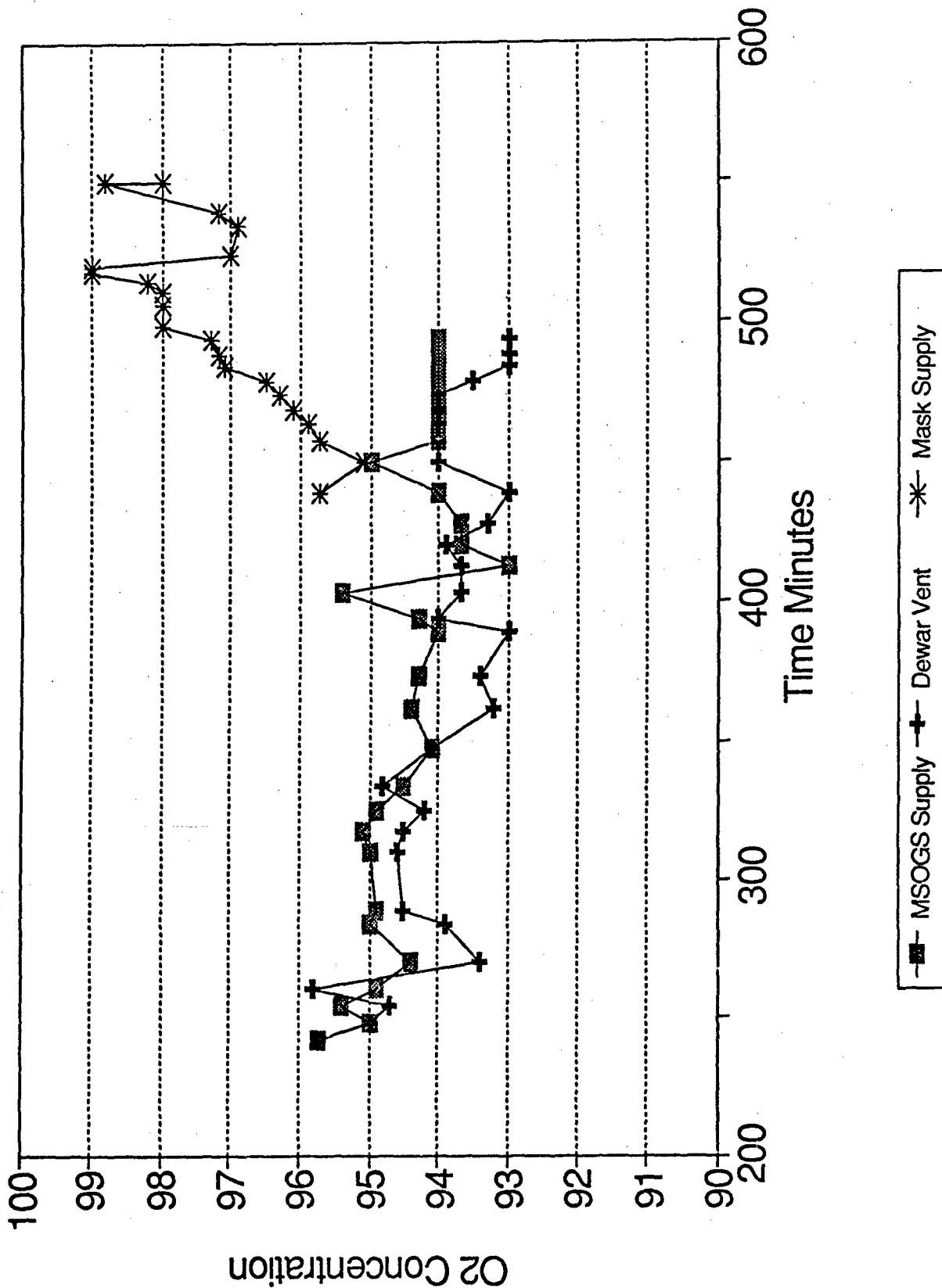


O2 Concentrations Test # 5 Nov 30 1990



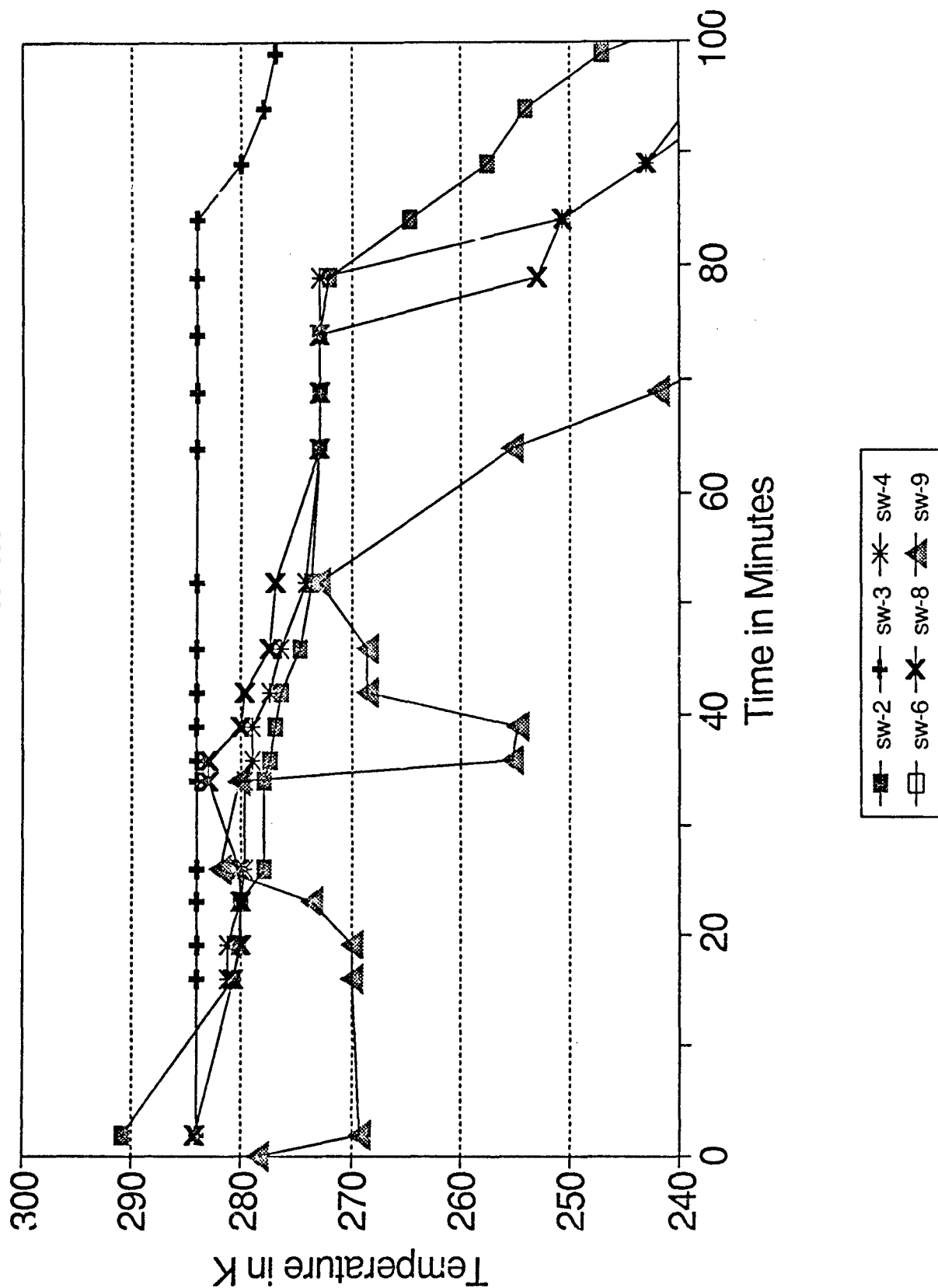
O2 Concentrations

Test # 5 Nov 30 1990



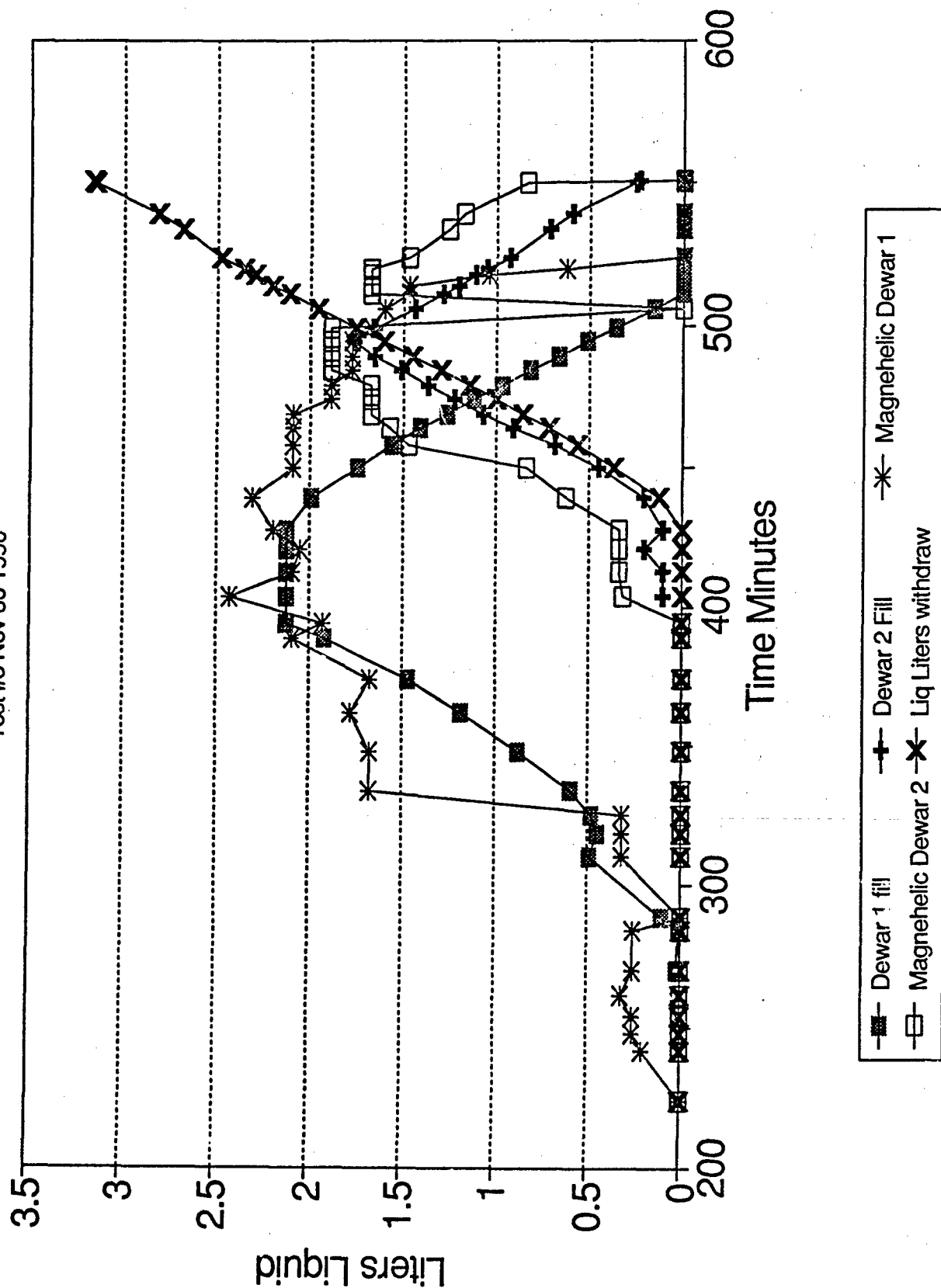
Cooling History-Initial Cooldown

Test # 5 Nov 30 1990

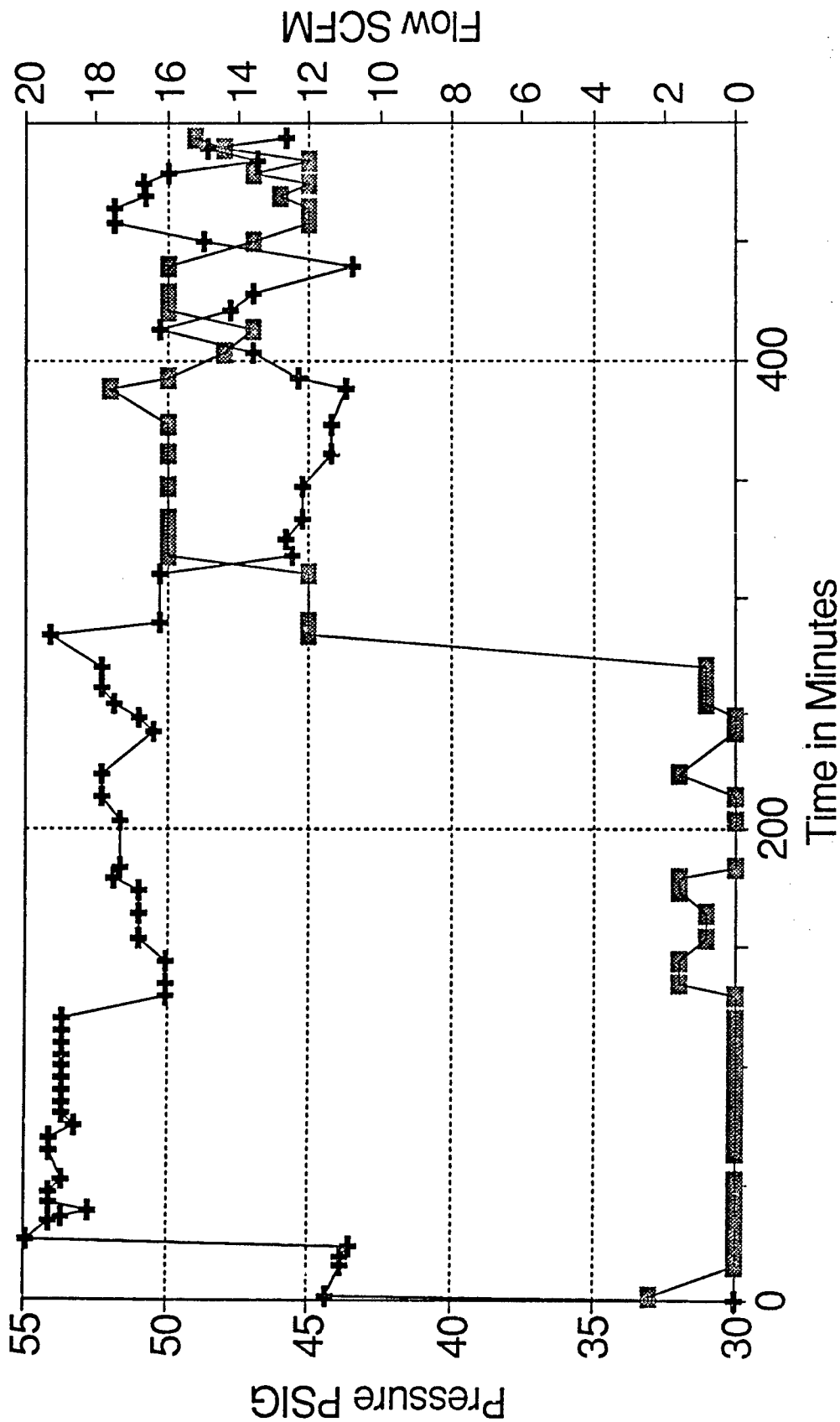


Dewar Filling-Oxygen

Test #5 Nov 30 1990



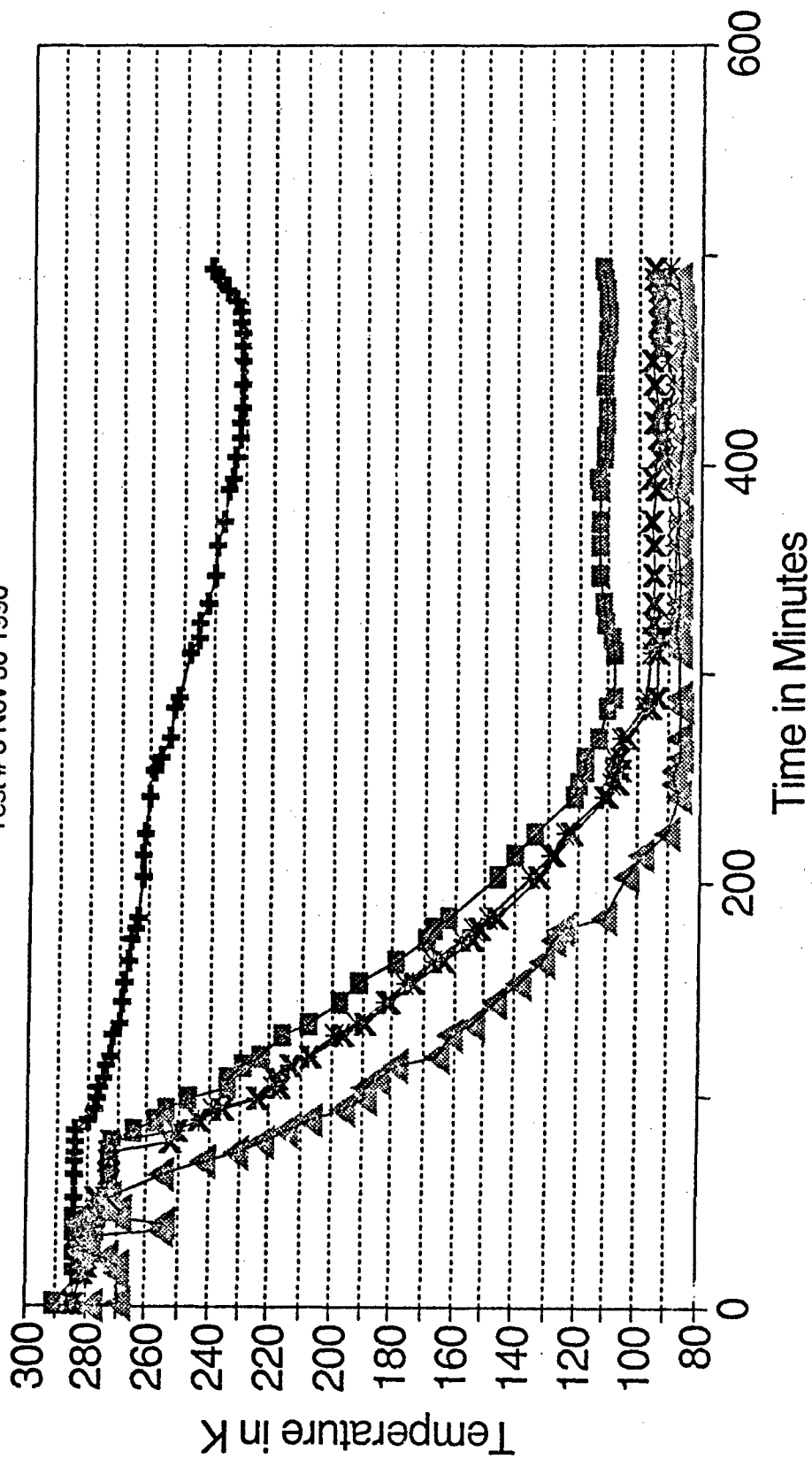
Heat Exchanger Deriming O2 Test #5



Air Flow SCFM
 Inlet Pressure PSIG

Cooling History

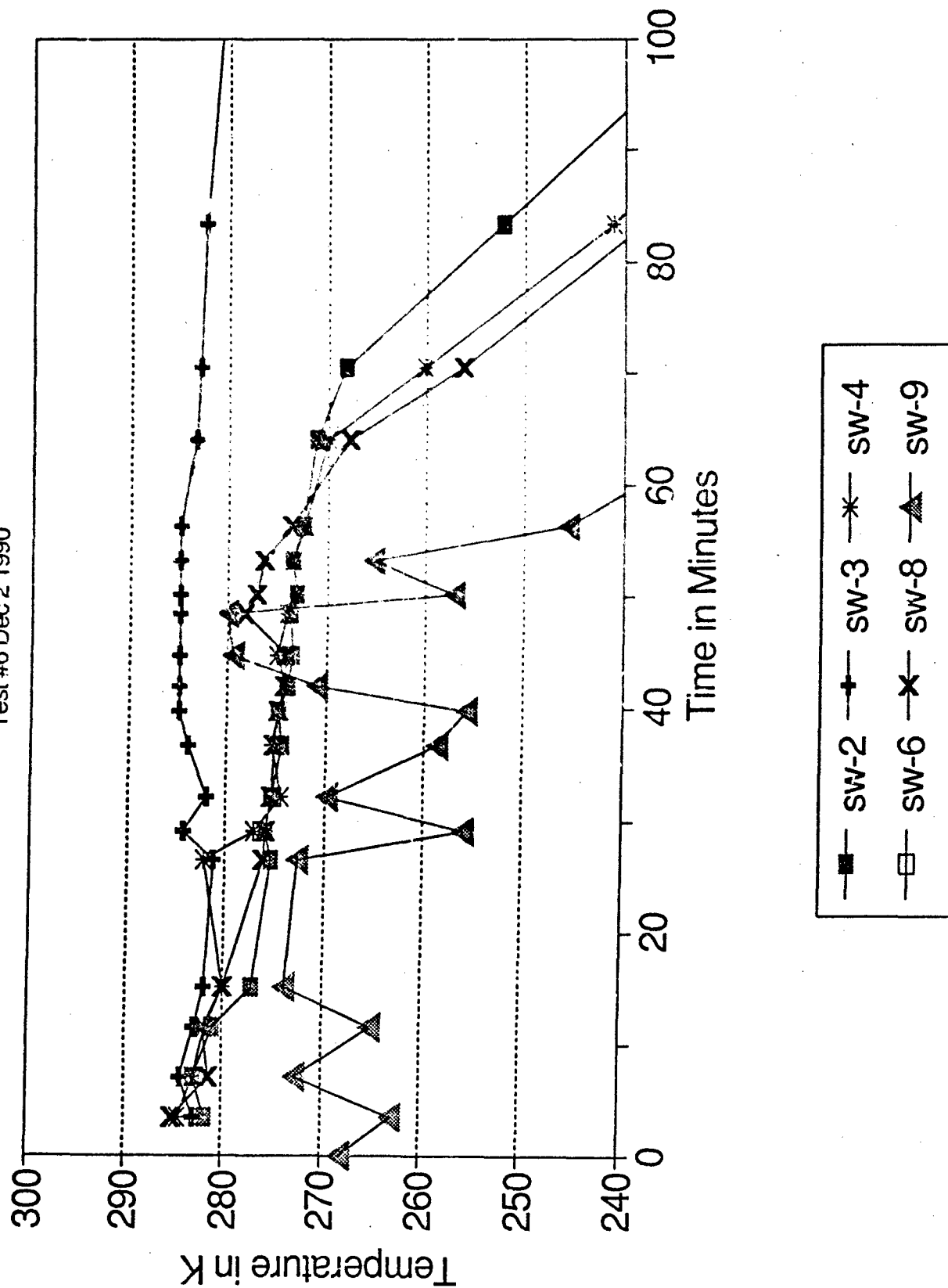
Test # 5 Nov 30 1990



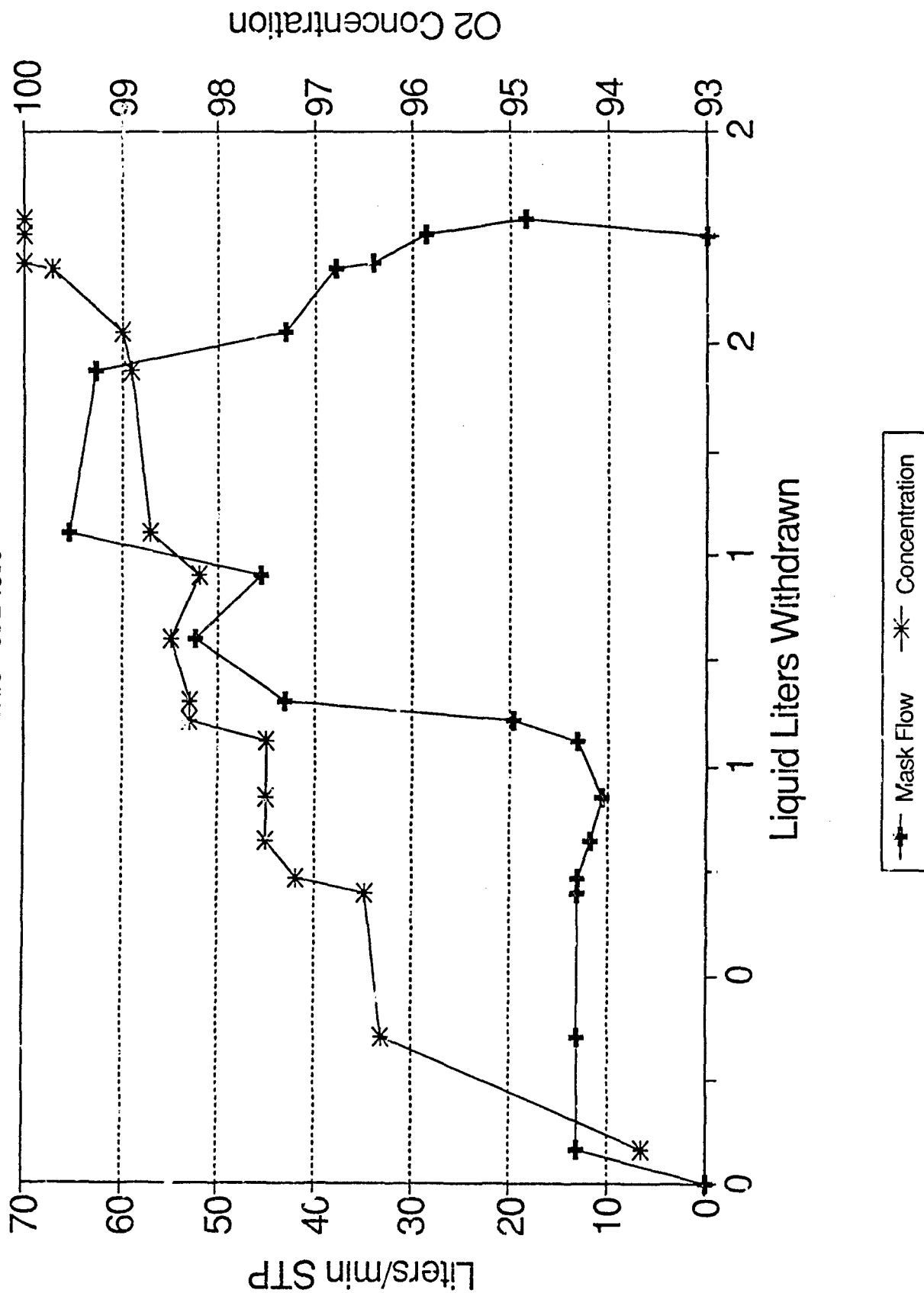
02 Test No. 6 Graphs

Cooling History

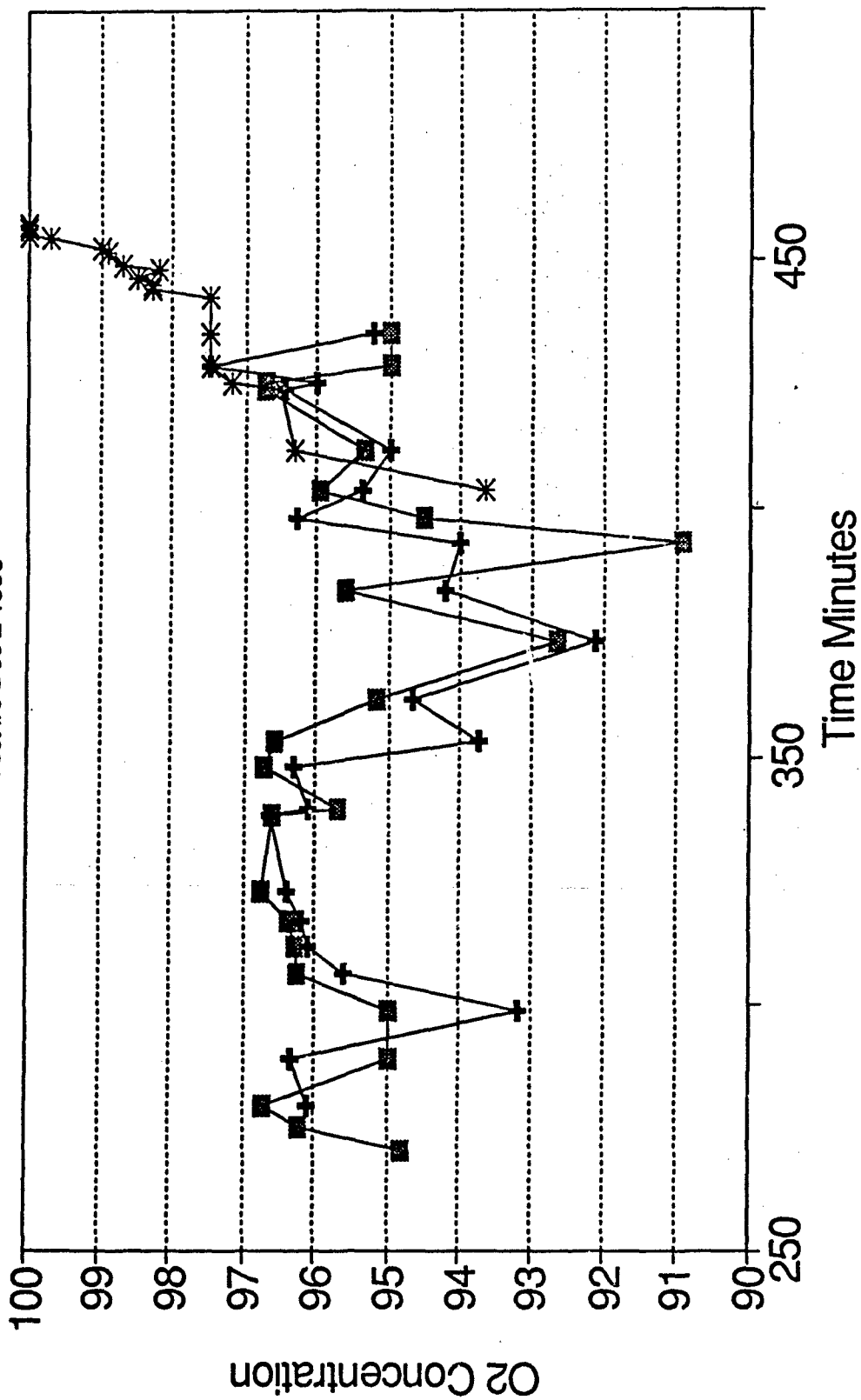
Test #6 Dec 2 1990



O2 Concentrations Test #6 Dec 2 1990

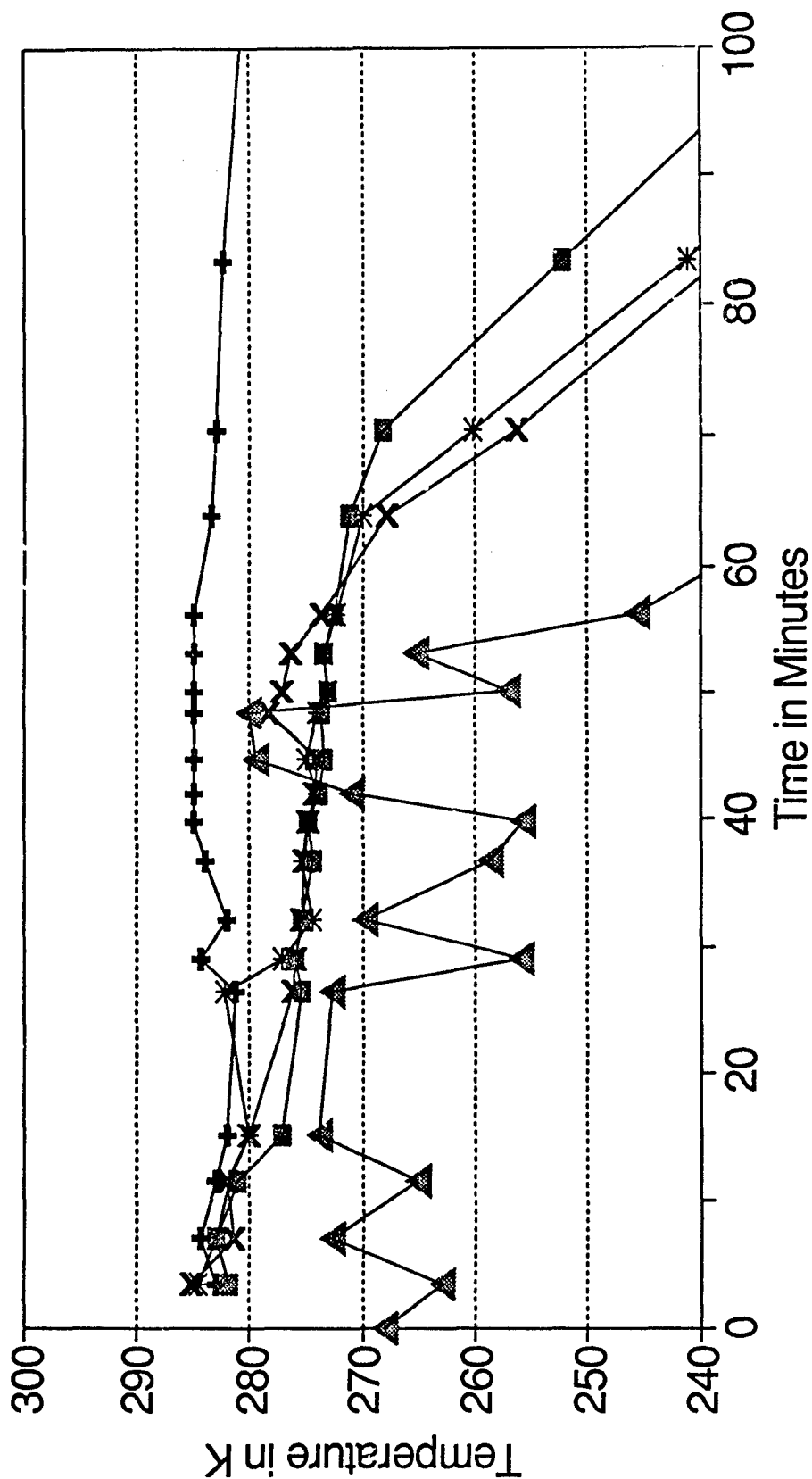


O2 Concentrations Test #6 Dec 2 1990

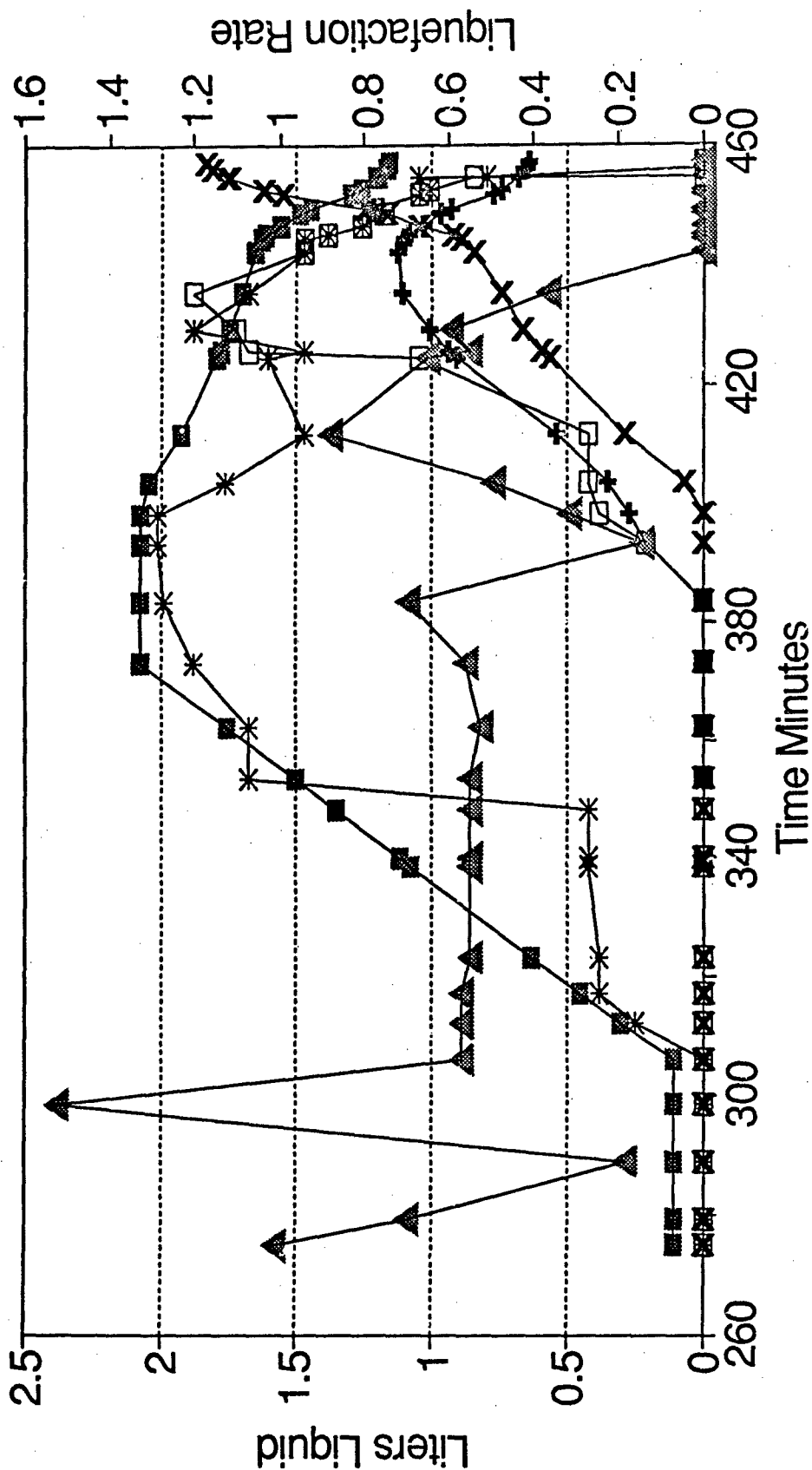


—■— MISOGS Supply —+— Dewar Vent —*— Mask Supply

Cooling History Test #6 Dec 2 1990

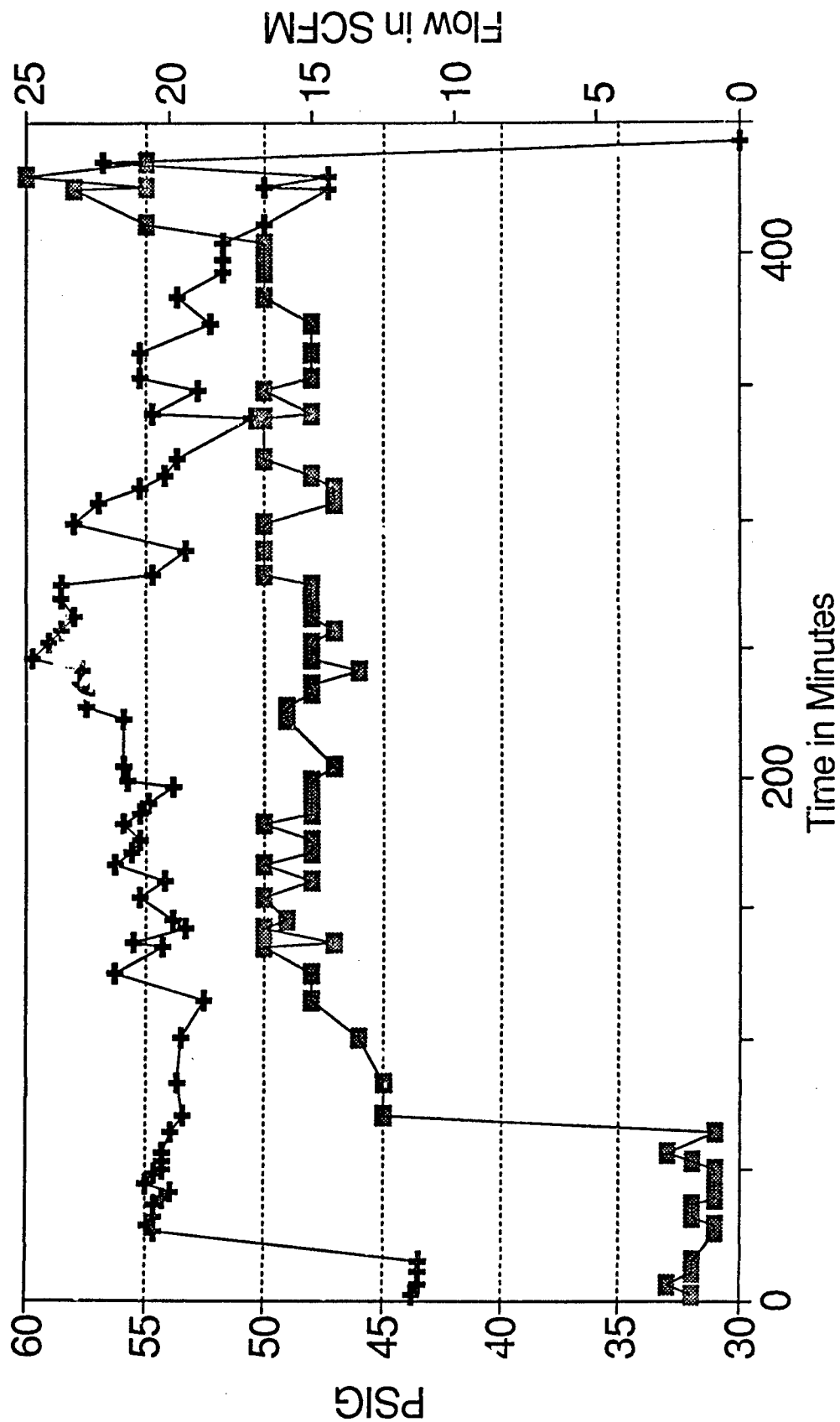


Dewar Filling-Oxygen O2 Test #6 Dec 21 1990



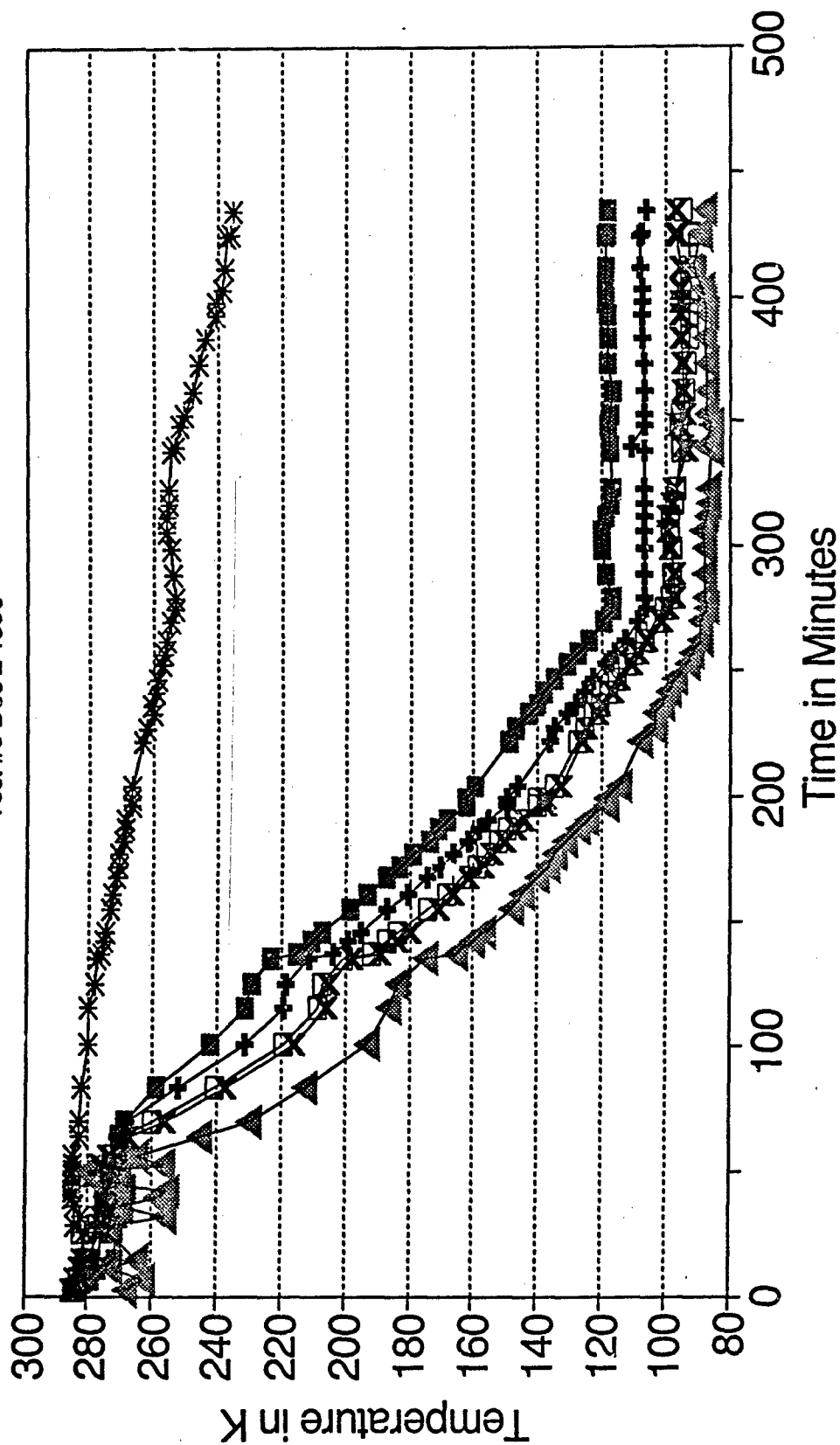
—■— Dewar 1 fill —+— Dewar 2 Fill —*— Mannehelic Dewar 1
 —□— Mannehelic Dewar 2 —x— Lia Liens withdraw —▲— Liq. Rate Gr/sec

Heat Exchanger Deriming Test #6 Dec 2 1990



Cooling History

Test #6 Dec 2 1990



7.7 Nomenclature

Note on Abbreviations

The following abbreviations are used throughout the report:

mm Hg is millimeter of mercury - pressure

LPM is liter per minute oxygen flow

NTP is normal temperature (70°F) and pressure (14.7 psia)

ATP is ambient (in cabin) temperature and pressure

**END
FILMED**

DATE:

4-93

DTIC