

2

NAVAL POSTGRADUATE SCHOOL Monterey, California

AD-A278 581



*Original contains color
plates: All DTIC reproductions
will be in black and
white*

THESIS

DTIC
ELECTE
APR 26 1994
S G D

VALIDATION AND IMPLEMENTATION OF
OPTICAL DIAGNOSTICS FOR PARTICLE
SIZING IN ROCKET MOTORS

by

Paul V. Gomes

December, 1993

Thesis Advisor :

Professor David W. Netzer

Approved for public release; distribution is unlimited.

94-12647



4588

DTIC QUALITY INSPECTED 3

94 4 25 059

REPORT DOCUMENTATION PAGE			Form Approved OMB No. 0704-0188	
Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instruction, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project (0704-0188) Washington DC 20503.				
1. AGENCY USE ONLY (Leave blank)		2. REPORT DATE December 1993		3. REPORT TYPE AND DATES COVERED Master's Thesis
4. TITLE AND SUBTITLE Validation and Implementation of Optical Diagnostics for Particle Sizing in Rocket Motors.			5. FUNDING NUMBERS F04611-93-X-000505 F04611-93-X-000509	
6. AUTHOR(S) Paul V. Gomes				
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) Naval Postgraduate School Monterey CA 93943-5000			8. PERFORMING ORGANIZATION REPORT NUMBER -	
9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) Air Force Phillips Laboratory Edwards AFB, CA 93523-5000			10. SPONSORING/MONITORING AGENCY REPORT NUMBER -	
11. SUPPLEMENTARY NOTES The views expressed in this thesis are those of the author and do not reflect the official policy or position of the Department of Defense or the U.S. Government.				
12a. DISTRIBUTION/AVAILABILITY STATEMENT Approved for public release; distribution is unlimited.			12b. DISTRIBUTION CODE A	
13. ABSTRACT Aluminum oxide (Al_2O_3) particles of known size distribution were cast into a solid propellant which burned at temperatures less than the melting point of Al_2O_3 . Thus, particles of known size distribution existed at the nozzle inlet and in the plume. Malvern particle sizing instruments were used to make measurements at these two location using a windowed subscale motor and the results were compared to the known distribution. In the motor, measurements were limited due to disruptive flow from the window purge gas. However, the unaffected larger modes were properly measured. In the plume, the measurements of the modes were quite accurate, but low signal strength resulted in some inaccuracies for the mass contained in each mode. A phase Doppler particle analyzer was adapted to an existing plume probe. Initial measurements at two radial locations were in good agreement with the expected size distribution.				
14. SUBJECT TERMS Particle Sizers, Solid Rockets			15. NUMBER OF PAGES 46	
			16. PRICE CODE	
17. SECURITY CLASSIFICATION OF REPORT Unclassified	18. SECURITY CLASSIFICATION OF THIS PAGE Unclassified	19. SECURITY CLASSIFICATION OF ABSTRACT Unclassified	20. LIMITATION OF ABSTRACT UL	

NSN 7540-01-280-5500

DTIC QUALITY INSPECTED 3

Standard Form 298 (Rev. 2-89)

Prescribed by ANSI Std. Z39-18

Approved for public release; distribution is unlimited.

Validation and Implementation of Optical Diagnostics
for Particle Sizing in Rocket Motors

by

Paul V. Gomes
Lieutenant , United States Navy
B.S., University of Washington, 1986

Submitted in partial fulfillment
of the requirements for the degree of

MASTER OF SCIENCE IN AERONAUTICAL ENGINEERING

from the

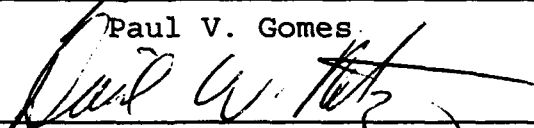
NAVAL POSTGRADUATE SCHOOL
December 1993

Author:

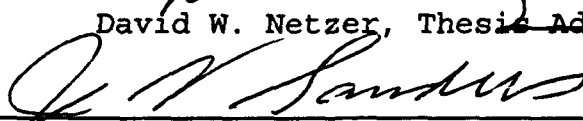


Paul V. Gomes

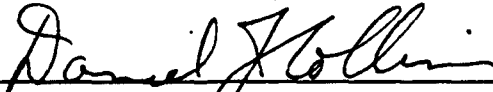
Approved by:



David W. Netzer, Thesis Advisor



James V. Sanders, Second Reader



Daniel J. Collins, Chairman
Department of Aeronautics and Astronautics

ABSTRACT

Aluminum oxide(Al_2O_3) particles of known size distribution were cast into a solid propellant which burned at temperatures less than the melting point of Al_2O_3 . Thus, particles of known size distribution existed at the nozzle inlet and in the plume. Malvern particle sizing instruments were used to make measurements at these two location using a windowed subscale motor and the results were compared to the known distribution. In the motor, measurements were limited due to disruptive flow from the window purge gas. However, the unaffected larger modes were properly measured. In the plume, the measurements of the modes were quite accurate, but low signal strength resulted in some inaccuracies for the mass contained in each mode. A phase Doppler particle analyzer was adapted to an existing plume probe. Initial measurements at two radial locations were in good agreement with the expected size distribution.

Accession For	
NTIS CRA&I	☒
DTIC TAB	☐
Unannounced	☐
Justification	
By	
Distribution /	
Availability Codes	
Dist	Avail and/or Special
A-1	

TABLE OF CONTENTS

I.	INTRODUCTION	1
II.	EQUIPMENT	4
	A. Subscale Motor	4
	B. Propellants	4
	C. Malvern2600HSD	6
	D. Malvern Mastersizer	7
	E. Aerometrics Phase Doppler Particle Analyzer	8
	F. Probe	9
III.	EXPERIMENTAL PROCEDURES	11
	A. Malvern Validation	11
	B. PDPA	12
IV.	RESULTS	14
	A. Malvern Measurements with Control Propellants	14
	B. PDPA Measurements	18
V.	CONCLUSIONS AND RECOMMENDATIONS	20
	APPENDIX A: FIGURES	22
	REFERENCES	37
	INITIAL DISTRIBUTION LIST	39

LIST OF TABLES

I.	PROPELLANT 1 COMPOSITION	5
II.	SIZE DISTRIBUTIONS OF AL_2O_3	6
III.	PROPELLANT 2 COMPOSITION	6
IV.	CALIBRATION DATA	15
V.	PDPA/PROBE DATA	18

I. INTRODUCTION

Metals are commonly added to solid rocket motor propellants to increase the specific impulse(I_{sp}) over the base propellant by adding energy to the combustion process. Metals are also added to increase the stability of the combustion by suppressing the transverse modes of oscillation (higher frequencies). The most commonly added metal is aluminum because of its high heat of combustion, low cost, and availability [ref. 1]. When aluminum is added to a solid propellant, there are associated problems such as primary smoke, two-phase flow losses and nozzle erosion [ref 2,3]. Two-phase flow losses are associated with lags in velocity and temperature between the condensed (Al_2O_3) particles and expanding gases and they are usually the greatest factor in determining nozzle efficiency.

There is an ongoing study of condensed aluminum oxide particle behavior in the chamber, across the exhaust nozzle and in the plume of solid rocket motors at the Naval Postgraduate School(NPS). Diagnostic techniques are directed at obtaining particle size distributions and the plume IR and visible signatures. To date, particle size measurements have been made using ensemble and single particle forward scattering of laser light, combined collection/optical probes and collection of particles on impact probes. Collected particles are analyzed using a scanning electron microscope(SEM). IR measurements are made using both a thermal imaging camera and a spectroradiometer.

Although good agreement has been obtained between the various

the probe to the PDPA instrument and to demonstrate its capabilities by making plume measurements.

II. EQUIPMENT

A. SUBSCALE MOTOR

An axisymmetric motor was used for the collection of data. The chamber(internal) diameter was 5.1 cm. The length was 25.4 cm with a residence time of 30 - 50 ms. The motor used viewing windows for collection of data in the motor chamber. It was modified with an additional window, offset by 50° to accommodate the PDPA measurements. To prevent excessive contamination of the viewing windows, a nitrogen purge system was included for each window. The flow rate of the nitrogen purge was approximately 10% of the propellant mass flow rate. The end-burning propellant grain was 5.1 cm in diameter and 2.54 cm thick. The exhaust nozzle was made of copper, had a converging half-angle of 45° and throat diameter of 0.5 cm.

B. PROPELLANTS

All propellants were provided by the Air Force Phillips Lab, Edwards AFB, California.

1. Calibration Propellants

These propellants were to be used for validation of the Malvern particle sizers for obtaining particle size distributions in the motor and plume.

TABLE I. PROPELLANT 1 COMPOSITION

PROPELLANT CONSTITUENTS	COMPOSITION	WEIGHT PERCENT
Aluminum Oxide	Al_2O_3	16%
Ammonium Perchlorate	NH_4ClO_4	32%
Ammonium Nitrate	NH_4NO_3	21.9%
GAP	$\text{C}_3\text{H}_5\text{N}_3\text{O}$	26.7%
IPDI	$\text{C}_{12}\text{H}_{18}\text{N}_2\text{O}_2$	3.2%
Others		0.2%

There were three specially formulated propellants (Table I) that differed only by the size distributions of the aluminum oxide (Table II). The various size distributions were selected to represent those that had been previously measured or predicted to occur at the nozzle inlet and exit. Spherical Al_2O_3 was not available. The specific sizes were provided by the manufacturer. The irregular shapes of Al_2O_3 can be seen in the SEM pictures in Figs 1-4. The propellant combustion temperature (2076K^1) was kept lower than the melting point of aluminum oxide (2318K) for a 450 psia combustion chamber pressure.

¹The temperature value was computed with the Micropep program [ref 8].

TABLE II. SIZE DISTRIBUTIONS OF Al_2O_3

1A	1B	1C
10% 2 micron	65% 2 micron	20% 2 micron
70% 5 micron	35% 122 micron	80% 122 micron
20% 20 micron		

2. Propellant 2

This propellant [Table III] was used to provide spherical Al_2O_3 particles at the entrance of the exhaust nozzle and in the plume. Firings were made in which the PDPA was used in the motor or with the probe in the plume.

TABLE III. PROPELLANT 2 COMPOSITION

PROPELLANT CONSTITUENTS	COMPOSITION	WEIGHT PERCENT
Aluminum (5-20 MICRONS)	Al	4.68%
Ammonium Perchlorate	NH_4ClO_4	70.31%
GAP	$C_3H_5N_3O$	14.67%
TEGDN	$C_6H_{12}N_2O_8$	8.49%
Others		1.84%

The density of the propellant was 1.76 g/cc.

C. MALVERN 2600HSD

The 2600HSD[ref 9] measures forward scattered light with a maximum angle of 14° . The system has a transmitter and receiver.

The transmitter provided a 2 mW helium neon laser beam (633 nm wavelength) expanded to 9 mm. The receiver can detect a size range of .5 to 564 microns, broken up into three ranges that are determined by the choice of the receiver lens (63 mm, 100 mm, 300 mm). These Fourier transfer lenses focus the scattered light onto 31 concentric diodes and the data are processed using Fraunhofer diffraction theory. The read time for one sweep of the diode array is approximately eight milliseconds. This investigation used the 100 mm lens with a detection range of 1.9 to 188 microns and vignetting distance of 133 mm. The volume of particles in the range 0.5 to 1.9 microns is also estimated by the Malvern software.

D. MALVERN MASTERSIZER

The Mastersizer[ref 7] also measures forward scattered light, but to a maximum angle of approximately 50°. The transmitter provided a 2 mW helium-neon laser beam expanded to 18 cm. The receiver can detect particles with diameters from 0.1 to 600 microns depending on the focal length of the receiver lens(45 mm,100 mm,300 mm). The 45 mm lens uses reverse Fourier optics while the 100 mm and 300 mm lens use conventional Fourier optics. The processing of the scattered light data uses Mie corrections to Fraunhofer diffraction for smaller particles. The experiment used a 100 mm lens with range of 0.5 to 180 microns and a vignetting distance of 29 mm.

E. AEROMETRICS PHASE DOPPLER PARTICLE ANALYZER

The PDPA utilizes an argon ion laser with 2 watts of power at 514.5 nm[ref 10]. The transmitter splits the beam into two beams of equal intensity, 20 mm apart. One beam is an unshifted, zero order beam. The second beam is a first-order beam shifted by 40 MHz. The beams are crossed at a lens focal length of 250 mm to form the probe volume. As a particle moves through the probe volume, light is scattered. The doppler signal analyzer uses high speed analog-to-digital converters to record the signal. The analyzer uses a fast Fourier transform to determine the frequencies of the signal. It is capable of measuring particle sizes as small as 0.5 microns (with a dynamic range of 50:1). It can measure greater than 300,000 particles/sec with velocities up to 1900 m/s, depending upon the selected focal lengths of the transmitter and receiver lenses.

The PDPA measurement is based upon the phase-shift of scattered light from a particle. Geometric optics are assumed to apply, in which the scattered light consists of reflection, refraction, 2nd order refraction and diffraction. For a specific index of refraction, plots are made of scattered power vs. scattering angles (0-180°) for each of the individual types of scattering and the total (Mie) scattering. Angles are chosen where one type of scattering (reflection or refraction) dominates. Then the phase shift at multiple detectors produced by a particle as it passes through the crossed-beam probe volume is plotted against particle diameter for the chosen type of scattering and scattering angle. This plot turns out to be linear for non-absorbing particles, when

forward scattering measurements of refracted light are made (typically $\sim 50^\circ$). The same is true for highly absorbing particles when reflected light is measured in the backscattering mode.

Aluminum oxide is slightly absorbing and the index of refraction and absorption index vary considerably with particle temperature, particle size and the degree of contamination (soot, aluminum, etc). A "best" estimate for the plume particles index is $m = 1.74 - i(3 \times 10^{-5})$. Unfortunately, the low absorption index results in some non-linearity in the phase-shift vs. diameter correlation for backscattering, when particles are smaller than 40 microns. The software uses a linear correlation, so uncertainties in m can be translated to uncertainties in diameter. The more accurately m is known, the more accurately the diameter can be measured. In this experiment, a 50° backscattering angle was used.

F. PROBE

The dimensions of the probe are shown in Fig 5. The object of the probe is to isolate a small stream of the particles in the exhaust flow [ref 5,6]. The probe tip is designed to swallow the strong normal shock that could break up particles. Weaker oblique shocks form inside the probe tip. The window is large enough to allow for the incoming light source as well as the back-scattered light of the PDPA. Figure 6 shows the probe attachment which was designed and fabricated to protect the PDPA from exhaust products outside the probe. The probe initially had a nitrogen purge system

to keep the window relatively free of particles, but also to prevent recirculation of the particle as they exit the tip and pass into the probe chamber.

III. EXPERIMENTAL PROCEDURES

A. MALVERN VALIDATION

Spherical aluminum oxide particles were not available; instead, non-spherical aluminum oxide particles in four size distributions were used. The software used by the Malvern particle sizers assumes that particles are spherical. Therefore, calibration data were required to determine the "equivalent" spherical particle size distributions for the aluminum oxide cast into the propellants.

Individual particle size distributions were suspended in solution and data were collected using the Malvern particle sizers. Multi-modal size distributions were then prepared by first collecting data on individual size distributions and determining the volume concentrations of particles (V_p) with the following equation:

$$\%CONC = \frac{(V_p)}{(V_L)}$$

The percentage of particles in liquid (%CONC) was obtained from the Malvern particle sizer output and the volume of liquid (V_L) was measured with a disposable pipette. After determination of the first particle size distribution and volume (V_p), and knowing the desired percentage of the particle size distribution in the multi-modal size distribution, the required %CONC for the other size distributions could be calculated. The individual particle size distributions were mixed to less than 5% error of the required %CONC. The individual size distributions were then mixed together

and multi-modal data were collected by the Malvern particle sizers.

Multi-modal size distributions were measured at the nozzle entrance and exit [Fig. 7] during motor firings for comparison with the calibration data in order to assess the accuracy of the Malvern measurements in the rocket motor environment.

B. PDPA

The PDPA transmitter was arranged perpendicular to the probe/motor and the crossed-beam volume was parallel to the particle flow to acquire velocity and size distribution data. The receiver was placed above the transmitter at a 50° backscatter angle. The laser was directed through a beam waist adjuster and polarization rotator before being adjusted through three steering mirrors [Fig. 8] and passed into the transmitter. The system was aligned before each test.

1. Measurement in the Plume

The PDPA was used in conjunction with the combined collection/optical probe. The particle size distribution was measured on the centerline of the plume at 13.5 nozzle exit diameters (13.5 cm) downstream of the nozzle. Then, the particle size distribution was measured at 13.5 nozzle exit diameters downstream, but radially displaced by 1.3 cm (1.3 nozzle exit diameters). The data were compared to data obtained in another investigation which utilized the Malvern particle sizer.

Severe window contamination remained a problem. Therefore to permit longer data acquisition times, the windows were removed from the probe and protective windows were instead placed over the lenses of the receiver and transmitter. This also allowed the window purge system to be eliminated. Only a small ejector nozzle flow was required in the probe.

2. Measurements in the Motor

The modified window as well as the existing window restricted the width of the transmitted beams and reflected light. This allowed measured size distributions at $0.65R$ from the motor centerline, where R is the internal radius of the motor. Since the grain was an end-burner, the collected data should not be effected by the probe volume location because the size distribution is approximately constant throughout the motor.

IV. RESULTS

A. MALVERN MEASUREMENTS WITH CONTROL PROPELLANTS

1. Calibration

The 122 micron diameter particles were too heavy to suspend in distilled water. The use of a heavy liquid with a density of 2.45 was attempted. Problems were then encountered trying to suspend particles from the 2 microns size distribution. These particles could not be efficiently distributed in the heavy liquid. Other methods were attempted, but no reliable data were obtained. Accurate calibration data were only collected for the 2,5, and 20 micron tri-modal size distribution and, therefore, the only propellant that was fired in the motor contained this mixture of Al_2O_3 .

For the 2600HSD, individual size distributions (2,5,20) were multiplied by the percentage each had in the tri-modal distribution(8,65,27% respectively). These adjusted individual distributions are plotted[Fig. 9a] along with the sum of the distributions(e.g., the expected distribution for the tri-modal mix). The "expected" tri-modal distribution is plotted with the measured tri-modal distribution in Figure 10. There was good agreement. The small differences could be attributed to the process used when the 2 and 20 micron distributions were transferred from their optical vials to the optical vial that contained the 5 micron distribution.

The same procedure was repeated for the Mastersizer[Figs. 11b and 12]. The results were again in good agreement. Comparison between the results obtained with the two Malvern particle sizers for obscuration(OBS), Sauter Mean Diameter(D_{32}) and mode are shown in Table IV.

TABLE IV. CALIBRATION DATA

SIZE DISTRIBUTION	OBS	D_{32} (MICRONS)	Mode (MICRONS)	% V_p in the tri- modal
2 MICRON MS 2600	0.31 0.42	4.9 4.5	6.4 6.3	9 8
5 MICRON MS 2600	0.40 0.54	7.8 7.7	9.4 10.6	73 65
20 MICRON MS 2600	0.15 0.29	21.1 24.9	24.3 29.8	18 27
TRI-MODAL MS 2600	0.38 0.48	7.9 8.6	9.4 6.8, 11.4 , 32.0	

2. Motor firing data

Since the calibrations for the 2600HSD and Mastersizer were in good agreement and only limited propellant was available, the motor firing data were collected only with the Malvern 2600HSD.

a. Measurements in the Motor

As the particles pass through the beam volume, light is scattered in the forward direction at various angles. Smaller particles scatter more light at larger angles. The window cavity in the motor restricted the detector field of view. Therefore smaller particles must pass closer to the window cavity than larger

particles in order to prevent vignetting. Thus the motor geometry introduced some bias into the measurement process. For this experimental set-up, there was no bias for particles greater than 5 microns.

The expected tri-modal distribution for the propellant composition is shown in figure 13. Also shown is the "expected" tri-modal distribution with the window vignetting effect taken into consideration. It can be seen that vignetting had little effect on the "expected" distribution.

The measured distribution is also shown in figure 12. No particles were measured below 10 microns. This could have resulted from a motor plug that inadvertently vented during the run, just above the windows. Small particles may have followed the vented gas. Another possibility was that the window purge system pushed the smaller particles out of the beam volume, effectively preventing detection. The propellant had a very low burn rate, resulting in a chamber pressure(P_c) of approximately 200 psia(including the mass flow from the window purge system. The low propellant burning rate resulted in the window purge flow velocity being approximately equal to the gas velocity in the chamber. This problem of detecting no small particles was also evident in the motor run which used the PDPA. No particles passed through the crossed-beam volume located near the wall just outside of the window cavity. The mass of aluminum oxide contained in the propellant which had diameters less than 10 microns was approximately 38%. Assuming that these particles were not in the

measurement volume would result in a different "expected" distribution as shown in figure 13. It is seen that the two large modes were identified, though shifted toward each other.

It is apparent that a method needs to be found which greatly reduces or eliminates the window purge flow rate.

b. Measurements in the Plume.

Without window purge gases added in the chamber, this low burning rate propellant produced chamber pressures of only approximately 100 psia. This resulted in a very low mass flow rate, which translated into very weak light scattering signals from the particles in the plume. In the future, the mass flow rate should be increased, by reducing the diameter of the nozzle throat. However, this may cause problems with clogging of the throat due to the 16% aluminum oxide in the propellant.

Figure 13 compares the measured particle size distribution with the distribution in the propellant. All the modes were correctly measured as were the relative masses in the two smaller modes. However, the smallest particles were not detected. The measured mass in the largest mode was too high. Very small changes in the recorded intensity at small angles were observed to significantly change the measured distribution. Future tests should eliminate the laser line filter used in the front of the receiver to increase the signal strength. This coupled with higher propellant flow rates should eliminate the problem.

B. PDPA MEASUREMENTS

1. Measurements in the Plume

Three samples were taken with the PDPA/probe. Figures 14 and 15 are examples of the data output by the PDPA. The number of valid samples compared to the number of attempted measurements was very low. The Phase Doppler Analyzer has size and velocity limits for a given size of transmitter and receiver lens. Every particle that passed through the crossed-beam volume was recorded as an attempt. The Phase Doppler analyzer was designed with the ability to go back and review the recorded "buffer data". This was done in an attempt to determine the cause for the large number of rejections. It appeared that many(perhaps 50%) of the rejections resulted from particles smaller than 0.7 microns.

The data collected from the three runs showed a decrease in the number of larger particles as the probe was moved radially outward in the plume[Table V and Figs. 15 and 16]. This was

TABLE V. PDPA/PROBE DATA

PDPA				
RUN	LOCATION	D_{32} (microns)	V_{ave} (m/s)	Mode (microns)
1	centerline	18.8	227	32
2	centerline	17.9	209	28
3	radially spaced (1.3 ex. dia)	9.7	105	30
MALVERN				
1	across plume	1.3		28

expected since larger particles have been predicted and measured to

be concentrated along the plume centerline as a result of not being able to turn rapidly with the gases through the nozzle throat. The data from the Malvern[Ref. 11, Fig. 17] was obtained including all particles across the width of the plume and resulted in a $D_{3,2}$ of 1.3 microns. That measurement included the dominantly small(< 2.0 microns) particles in the outer plume region. The plume volume outside of the radius where the PDPA measurements were made was approximately 2.5 times as large as the region measured. The initial PDPA/probe results look quite realistic. Future measurements should be made at various radial locations for direct correlation with the Malvern data.

2. Measurements in the Motor

No valid data were obtained in the motor. As discussed above, it is believed that the high window purge velocity(relative to the low combustion flow velocity from the low burning rate propellant) removed all the particles from the near-wall probe volume location.

V. CONCLUSIONS AND RECOMMENDATIONS

Measurements made with the Malvern particle sizers were in good agreement with each other and the both were effective in correctly locating the modes of multi-modal distributions.

The Malvern and PDPA measurements made in the motor revealed a problem with the method of keeping the windows clean. Unless high propellant flow rates (pressure) are used, the window purge flow significantly disrupts the combustor flow, removing most particles near the wall and biasing the measurements toward the larger particles in the rest of the measurement volume. Within these hardware limitations, the Malvern measurements correctly located the modes of the distribution.

Malvern measurements in the plume accurately located the modes of the tri-modal distributions of the aluminum oxide in the control propellant. Low propellant flow rates resulted in low scattered light intensity in the plume. This apparently made the measured mass-in-mode very sensitive to small changes in the recorded intensity profile.

The PDPA was successfully adapted to the combined optical/collection probe and initial measurements were in agreement with the expected plume particle size distribution.

It is apparent that a better method is needed for motor measurements. The motor must eliminate the window purge system, but still reduce severe window contamination.

The PDPA/probe work should be continued in various locations

of the plume. In addition, a multiple-wavelength extinction measurement system should be incorporated in an attempt to measure the particles which are smaller than 0.5 microns.

VI. APPENDIX A: FIGURES

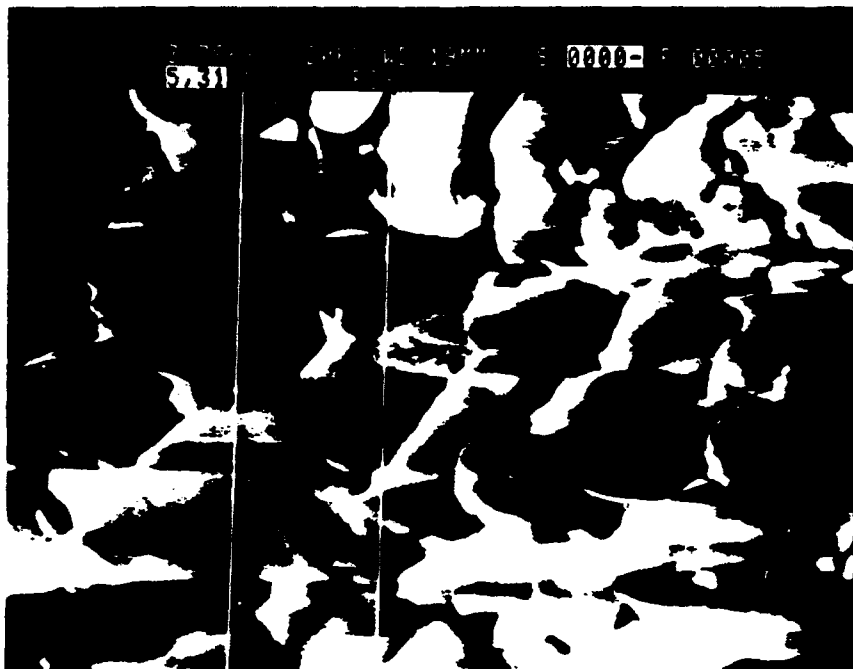


FIGURE 1: 2 MICRON Al_2O_3



FIGURE 2: 5 Micron Al_2O_3



FIGURE 3: 20 Micron Al_2O_3



FIGURE 4: 122 Micron Al_2O_3

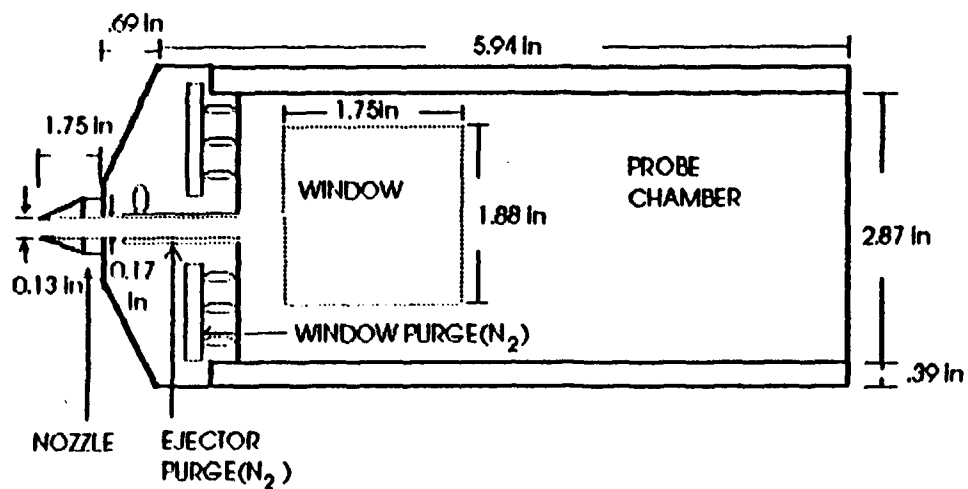


FIGURE 5: The Probe

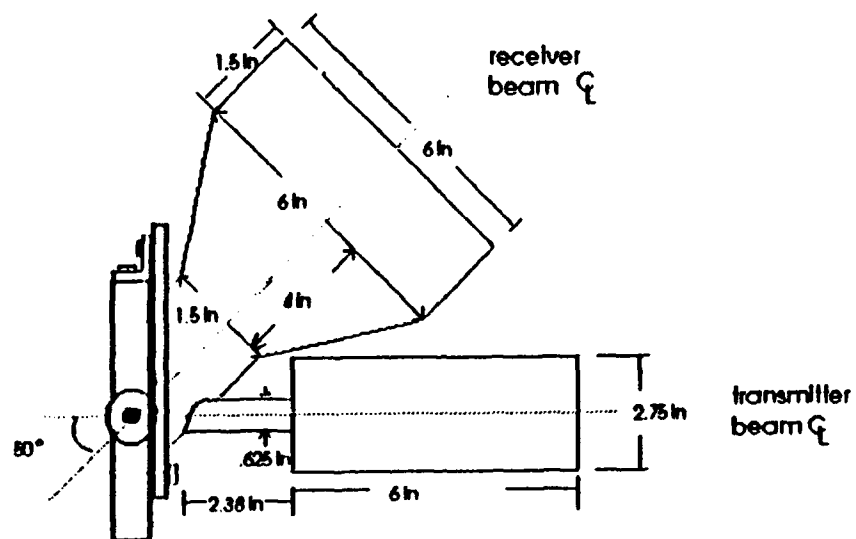


FIGURE 6: Probe Attachment

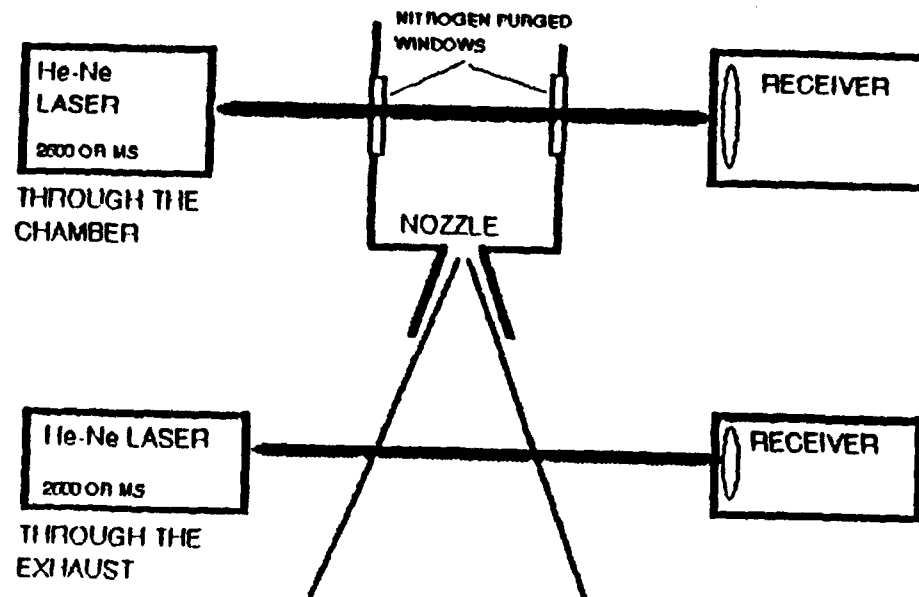


FIGURE 7: Malvern Set-up

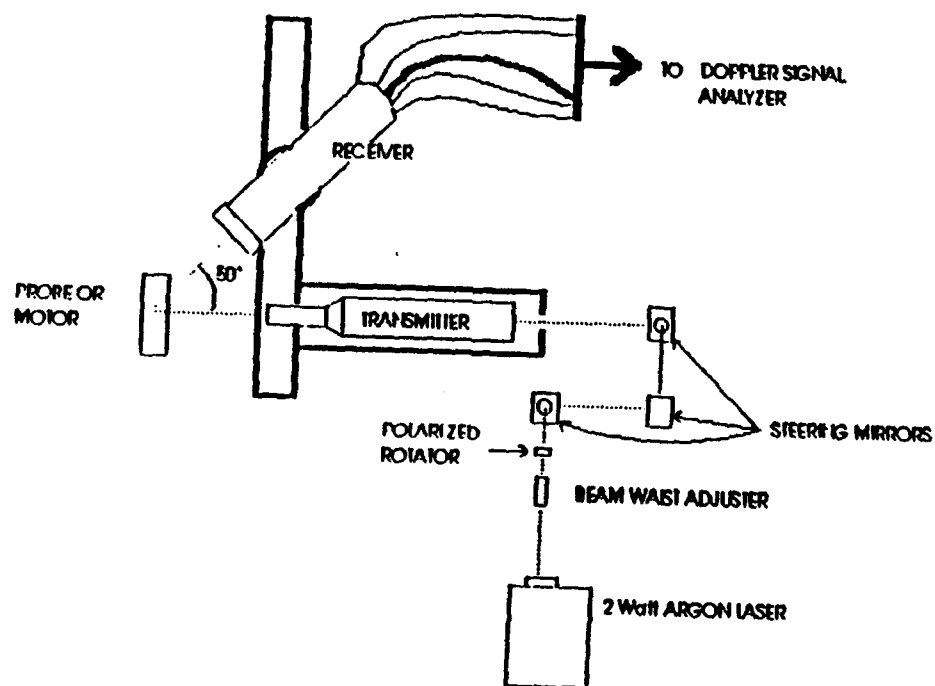


FIGURE 8: PDPA Set-up

Upper in Lower Under				Upper in Lower Under				Upper in Lower Under				Open	
100	100	100	100	100	100	100	100	100	100	100	100	1.25	
100	100	100	100	100	100	100	100	100	100	100	100	0.5	
100	100	100	100	100	100	100	100	100	100	100	100	6.12	
100	100	100	100	100	100	100	100	100	100	100	100	4.51	mu
100	100	100	100	100	100	100	100	100	100	100	100	0.93	
100	100	100	100	100	100	100	100	100	100	100	100	10.23	
100	100	100	100	100	100	100	100	100	100	100	100	0.11	
100	100	100	100	100	100	100	100	100	100	100	100	2.00	mu
Source =	Sample	Mean length =	5.6 g	Model indep								0.93	
Focal length =	100 m	Log. diff. =	0.4133	Values Conc. =	0.01615							5.69	
Presentation =	gill	Discardation =	0.4133	Sp. S.N.	1.3317 s/cc.							Shape Off	
		Values distribution											

Upper in Lower Under				Upper in Lower Under				Upper in Lower Under				Open	
												0.96	
188	0.0	0.0	100	37.7	0.0	0.0	100	1.82	12.0	0.47	31.6	B(4.3)	
140	0.0	0.0	100	43.8	0.0	0.0	100	0.97	9.9	0.30	22.6	3.44pp	
121	0.0	0.0	100	37.7	0.0	0.0	100	0.30	2.7	0.13	14.3	B(4.2)	
104	0.0	0.0	100	22.3	0.0	0.0	100	0.46	4.6	0.20	7.0	7.03pp	
77	0.0	0.0	100	11.4	0.0	0.0	100	0.71	7.1	0.33	1.0	B(7.5)	
57.7	0.0	0.0	100	1.4	0.0	0.0	100	0.04	0.4	0.01	0.1	14.08pp	
0.0	0.0	0.0	100	0.0	0.0	0.0	100	0.00	0.0	0.00	0.0	B(1.0)	
0.0	0.0	0.0	100	0.0	0.0	0.0	100	0.00	0.0	0.00	0.0	4.03pp	
Source = iSample				Bran length = 10.3 ug				Volume indp				B(1.0)	
				Cap length = 0.5185								9.41pp	
Focal length = 100 no				Obscuration = 0.518				Volume Conc. = 0.0185					
Presentation = pil				Volume distribution				Sp. B.R. = 0.7788 g/cc.				Shape Off	

[illegible]

Upper	in	Lower	Under	Upper	in	Lower	Under	Upper	in	Lower	Under	Spss
				57.7	0.7	45.0	98.0	9.82	0.3	8.47	36.0	2.24
				49.7	0.7	43.0	95.0	8.47	0.3	7.30	27.6	0(4.3)
				43.0	0.7	36.0	92.0	7.30	0.3	6.30	19.5	13.70p
				36.0	0.7	30.0	89.0	6.30	0.3	5.43	13.4	
188	0.0	162	100	29.7	0.7	26.0	86.0	5.43	0.3	4.65	9.1	0(1.2)
140	0.0	121	100	23.0	0.7	20.0	83.0	4.65	0.3	4.02	6.0	8.50p
121	0.0	99	99.9	20.0	0.7	17.7	80.0	4.02	0.3	3.51	3.6	28.04
99	0.0	77	99.9	17.7	0.7	15.3	77.0	3.51	0.3	3.12	1.2	0(0.0)
77	0.0	57.7	99.9	15.3	0.7	13.4	74.0	3.12	0.3	2.82	0.7	0(0.0)
57.7	0.0	45.0	98.0	13.4	0.7	11.4	70.0	2.82	0.3	2.50	0.4	0(0.0)
45.0	0.0	36.0	95.0	11.4	0.7	9.82	66.0	2.50	0.3	2.24	0.0	
Source = :Sample				Shape indep. = 10.34				Model indep				10(6.5)
Facial length = 100 mm				Log. Diff. = 1.267				Volume Conc. = 0.0183				10(6.5)
Presentation = pil				Observed = 0.4830				Sp. &A = 0.1995 s/cc				Shape Diff
				Volume distribution								

FIGURE 9a: Calibration:2600HSD:Data Analysis Results

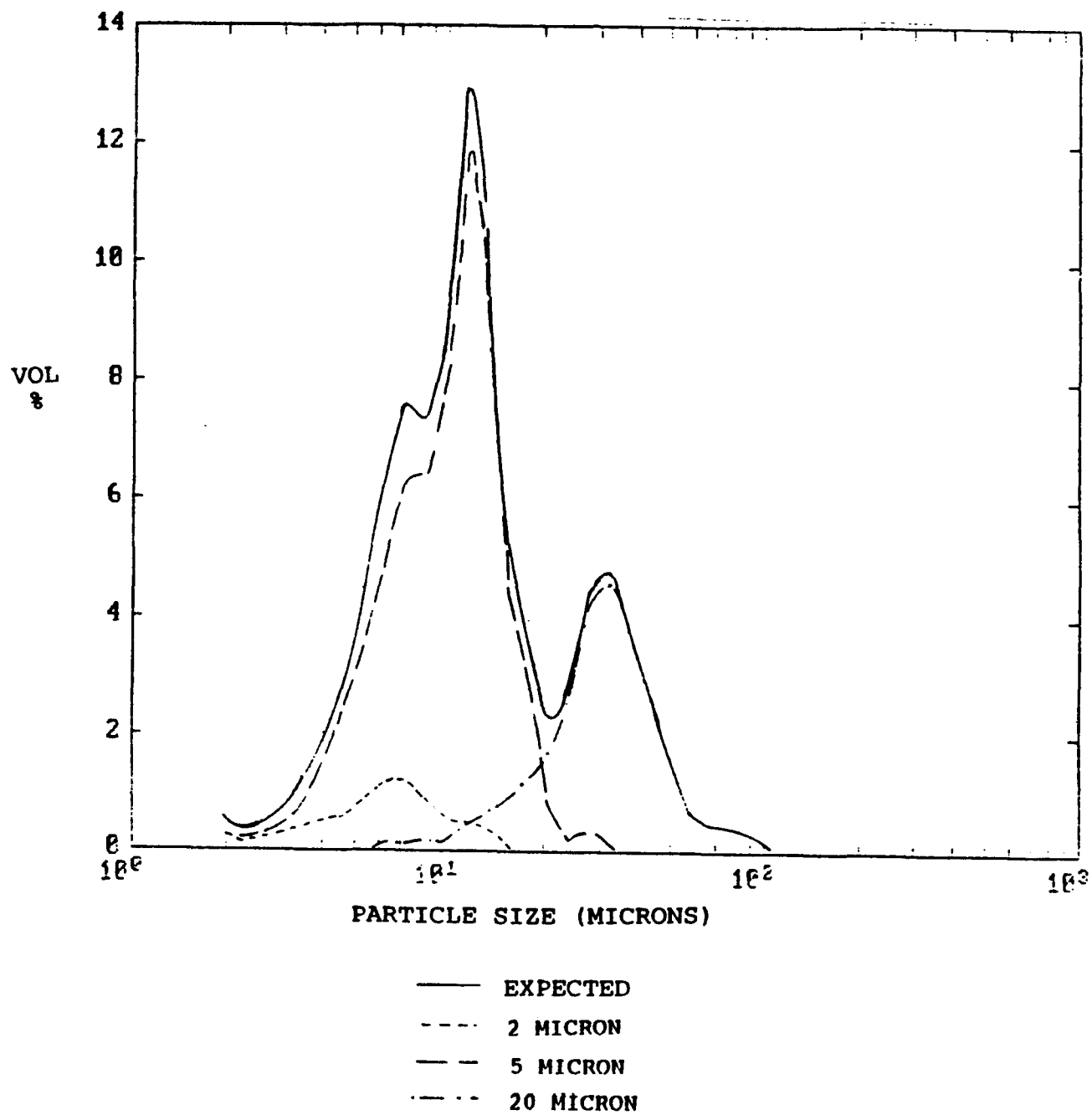


FIGURE 9b: Calibration:2600HSD:"Expected" Distribution

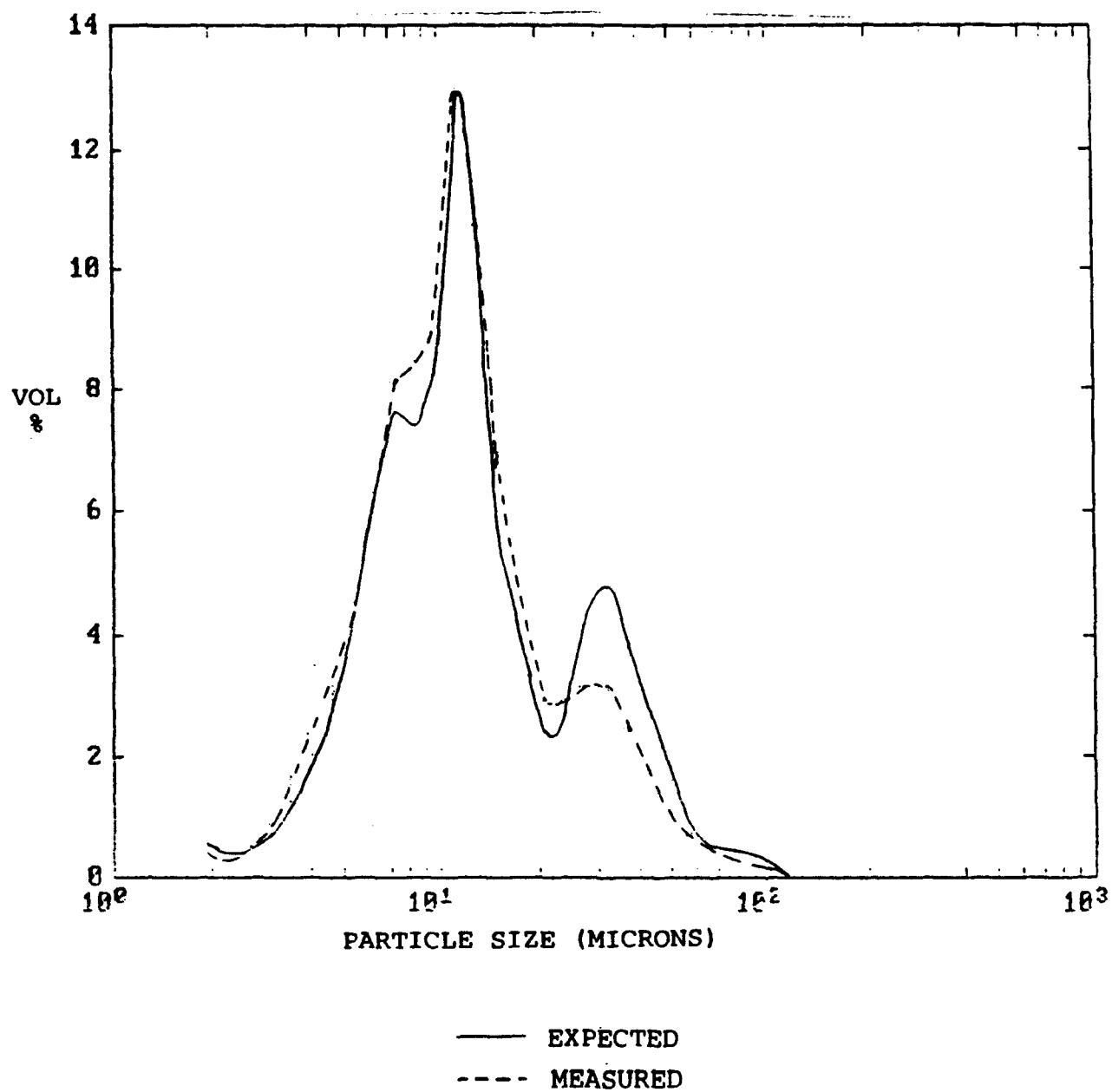


FIGURE 10: Calibration:2600HSD:"Expected" vs. Measured

Upper	in	Lower	Under	Upper	in	Lower	Under	Upper	in	Lower	Under	Span
				39.1	0.4	32.3	99.3	1.95	7.9	1.27	14.1	1.70
				32.3	0.6	28.7	98.7	1.27	3.55	2.70	8.6	D(4,3)
				28.7	1.1	25.0	97.7	2.70	3.85	2.23	7.4	7.50µm
180	0.0	149	100	25.0	1.8	21.2	96.8	1.84	1.33	1.52	1.1	D(3,2)
149	0.0	123	100	21.2	3.0	18.2	95.8	1.52	0.9	1.26	1.1	4.90µm
123	0.0	102	100	18.2	4.8	15.4	94.0	1.26	0.55	1.04	0.5	D(2,0.9)
102	0.0	83.9	100	15.4	7.0	12.4	91.0	1.04	0.3	0.86	0.2	13.34µm
83.9	0.0	69.3	100	12.4	9.4	9.4	88.0	0.86	0.2	0.71	0.1	D(0.9,1)
69.3	0.0	57.3	100	9.4	15.4	7.01	84.8	0.71	0.1	0.59	0.1	D(0.1,1)
57.3	0.0	47.3	99.9	7.01	18.2	4.79	82.4	0.59	0.1	0.48	0.1	D(0.1,1)
47.3	0.2	39.1	97.7	4.79	10.3	3.95	22.0	0.48	0.1	0.20	0.0	2.44µm
Source = Sample				Beas length = 5.0 mm				Model indp				D(0.5,1)
Focal length = 100 mm				Residual = 0.931 %				Value Conc. = 0.0107%				6.16µm
Presentation = 1403				Obscuration = 0.3112				Sp.S.A 1.2253 m ² /cc.				Shape Off
				Value distribution								

2 MICRON

Upper	in	Lower	Under	Upper	in	Lower	Under	Upper	in	Lower	Under	Span
				39.1	0.4	32.3	99.3	1.95	7.9	1.27	14.1	1.38
				32.3	0.6	28.7	98.7	1.27	3.55	2.70	8.6	D(4,3)
				28.7	1.1	25.0	97.7	2.70	3.85	2.23	7.4	11.04µm
180	0.0	149	100	25.0	1.8	21.2	96.8	1.84	1.33	1.52	1.6	D(3,2)
149	0.0	123	100	21.2	3.0	18.2	95.8	1.52	0.9	1.26	1.6	7.84µm
123	0.0	102	100	18.2	4.8	15.4	94.0	1.26	0.55	1.04	0.3	D(2,0.9)
102	0.0	83.9	100	15.4	7.0	12.4	91.0	1.04	0.3	0.86	0.2	18.04µm
83.9	0.0	69.3	100	12.4	9.4	9.4	88.0	0.86	0.2	0.71	0.1	D(0.9,1)
69.3	0.0	57.3	100	9.4	15.4	7.01	84.8	0.71	0.1	0.59	0.1	D(0.1,1)
57.3	0.0	47.3	99.9	7.01	18.2	4.79	82.4	0.59	0.1	0.48	0.0	D(0.1,1)
47.3	0.5	39.1	98.9	4.79	10.3	3.95	22.0	0.48	0.0	0.20	0.0	4.78µm
Source = Sample				Beas length = 10.6 mm				Model indp				D(0.5,1)
Focal length = 100 mm				Residual = 0.232 %				Value Conc. = 0.0116%				9.59µm
Presentation = 1403				Obscuration = 0.4026				Sp.S.A 0.7657 m ² /cc.				Shape Off
				Value distribution								

5 MICRON

Upper	in	Lower	Under	Upper	in	Lower	Under	Upper	in	Lower	Under	Span
				39.1	12.3	32.3	74.8	1.95	0.3	1.27	0.5	1.09
				32.3	18.1	25.0	56.7	1.27	0.2	2.70	0.3	D(4,3)
				25.0	19.9	22.0	36.7	2.70	0.2	3.85	0.3	27.11µm
180	0.1	149	99.9	22.0	15.9	18.2	20.9	2.23	0.1	1.84	0.1	D(3,2)
149	0.0	123	99.9	18.2	3.7	15.1	11.2	1.84	0.1	1.52	0.0	21.05µm
123	0.0	102	99.9	15.1	4.9	12.4	8.3	1.52	0.0	1.26	0.0	D(2,0.9)
102	0.0	83.9	99.9	12.4	2.4	10.3	3.9	1.26	0.0	1.04	0.0	41.84µm
83.9	0.0	69.3	99.0	10.3	1.5	8.4	2.0	1.04	0.0	0.86	0.0	D(0.9,1)
69.3	0.0	57.3	97.6	8.4	0.7	7.01	1.5	0.86	0.0	0.71	0.0	D(0.1,1)
57.3	0.0	47.3	94.3	7.01	0.5	5.79	1.1	0.71	0.0	0.59	0.0	D(0.1,1)
47.3	0.2	39.1	87.1	5.79	0.4	4.79	1.1	0.59	0.0	0.48	0.0	14.46µm
				4.79	0.3	3.95	0.8	0.48	0.0	0.20	0.0	
Source = Sample				Beas length = 10.0 mm				Model indp				D(0.5,1)
Focal length = 100 mm				Residual = 0.601 %				Value Conc. = 0.0106%				25.05µm
Presentation = 1403				Obscuration = 0.1469				Sp.S.A 0.2850 m ² /cc.				Shape Off
				Value distribution								

20 MICRON

Upper	in	Lower	Under	Upper	in	Lower	Under	Upper	in	Lower	Under	Span
				39.1	5.4	32.3	95.0	1.95	3.0	1.27	5.0	2.00
				32.3	5.4	25.0	91.4	1.27	1.9	2.70	3.0	D(4,3)
				25.0	5.1	22.0	86.2	2.70	1.1	2.23	1.9	13.11µm
180	0.0	149	100	22.0	6.9	18.2	79.3	2.23	0.6	1.84	1.3	D(3,2)
149	0.0	123	100	18.2	8.7	15.1	70.4	1.84	0.5	1.52	0.8	7.84µm
123	0.0	102	100	15.1	10.2	12.4	60.4	1.52	0.3	1.26	0.5	D(2,0.9)
102	0.0	83.9	100	12.4	11.1	10.3	49.2	1.26	0.2	1.04	0.3	25.21µm
83.9	0.1	69.3	97.9	10.3	11.2	8.4	38.0	1.04	0.1	0.86	0.2	D(0.9,1)
69.3	0.1	57.3	95.6	8.4	10.3	7.01	27.8	0.86	0.1	0.71	0.1	D(0.1,1)
57.3	0.1	47.3	96.8	7.01	8.4	5.79	13.0	0.71	0.1	0.59	0.1	D(0.1,1)
47.3	1.5	39.1	97.4	5.79	6.4	4.79	16.5	0.59	0.1	0.48	0.0	4.35µm
				4.79	4.5	3.95	8.0	0.48	0.0	0.20	0.0	
Source = Sample				Beas length = 10.6 mm				Model indp				D(0.5,1)
Focal length = 100 mm				Residual = 0.948 %				Value Conc. = 0.0106%				10.41µm
Presentation = 1403				Obscuration = 0.3753				Sp.S.A 0.7611 m ² /cc.				Shape Off
				Value distribution								

TRI-MODAL

FIGURE 11a:Calibration:MS:Data Analysis Results

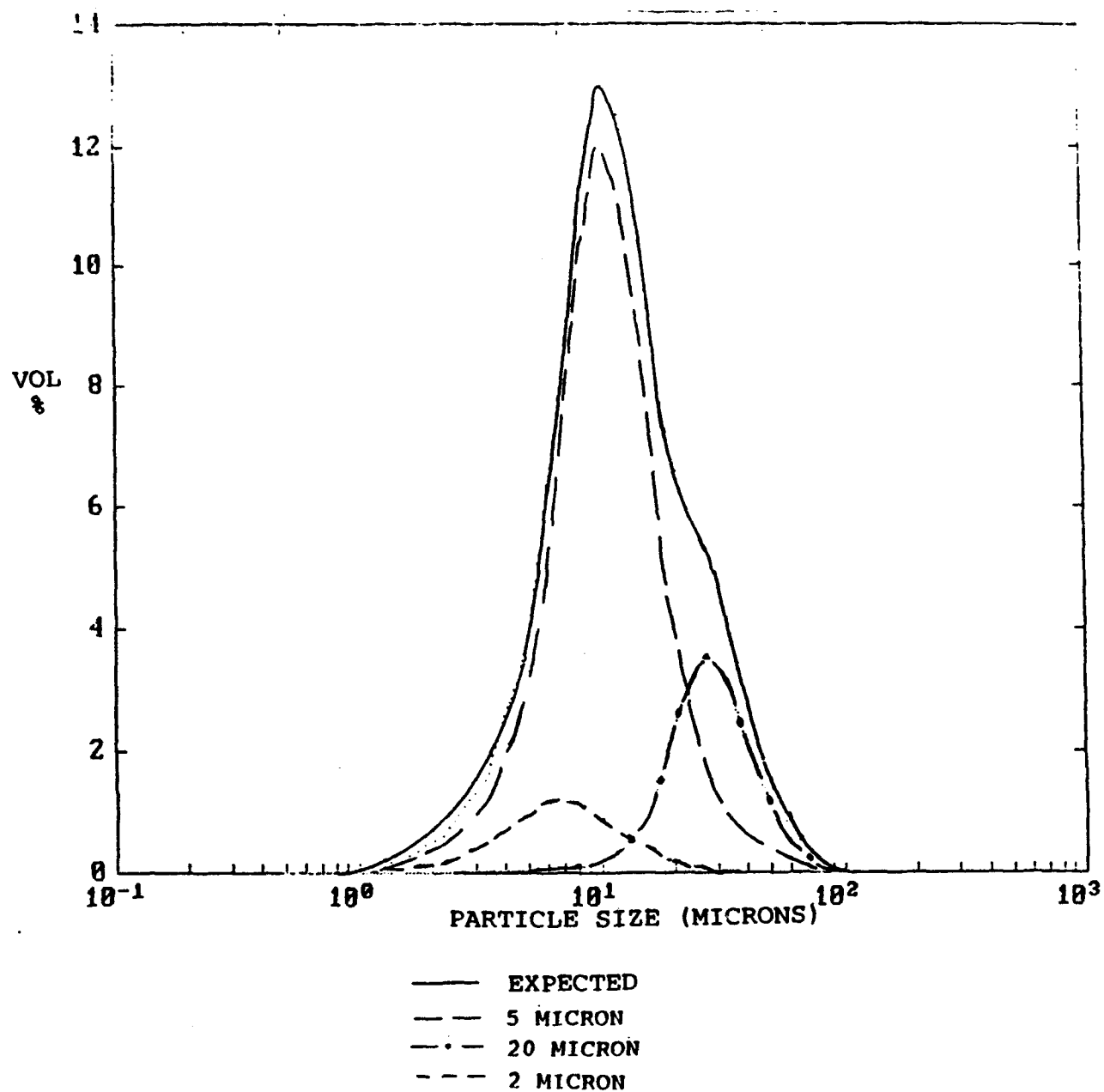


FIGURE 11b:Calibration:MS:"Expected" Distribution

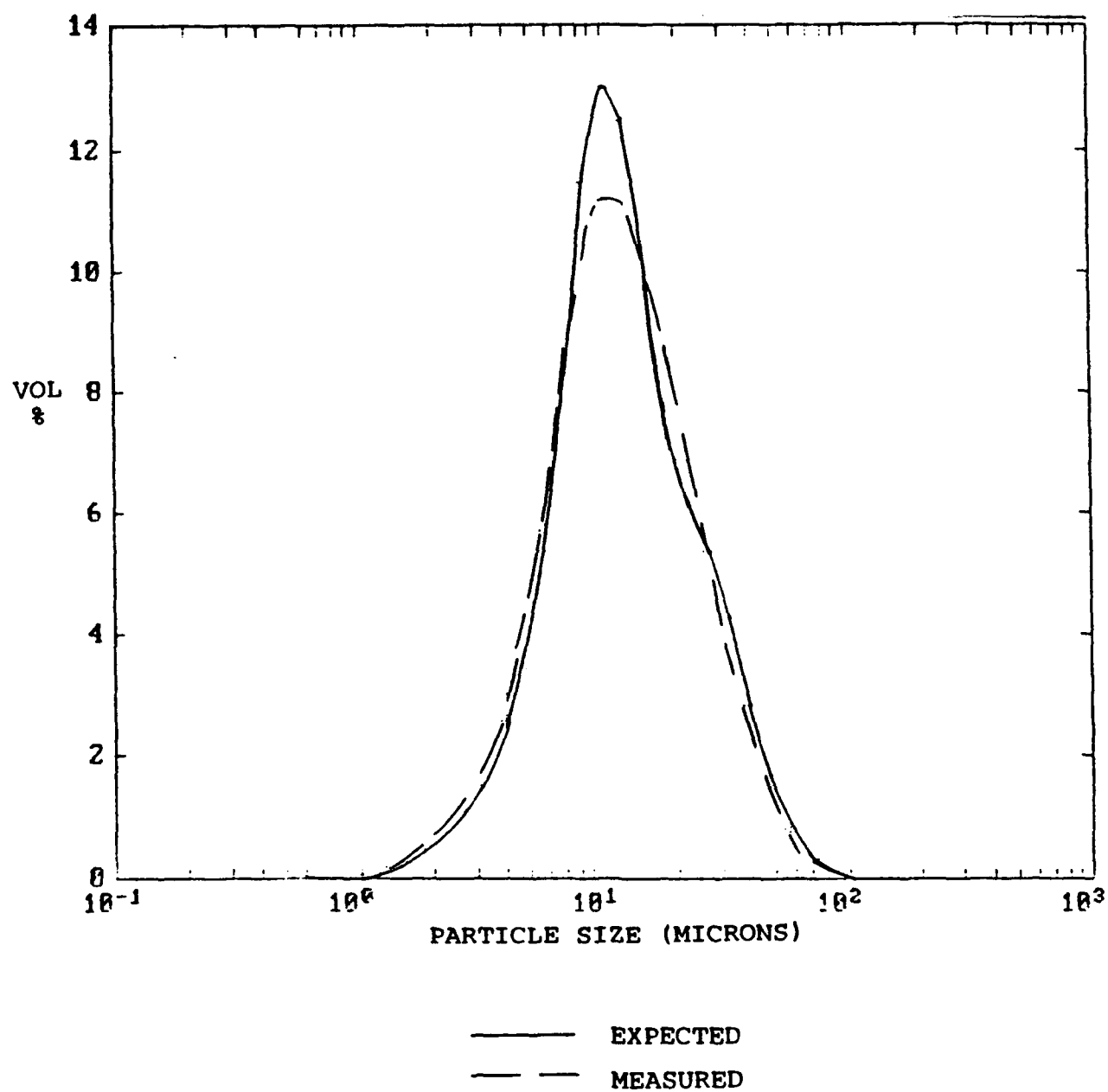


FIGURE 12: Calibration:MS:"Expected" vs. Measured

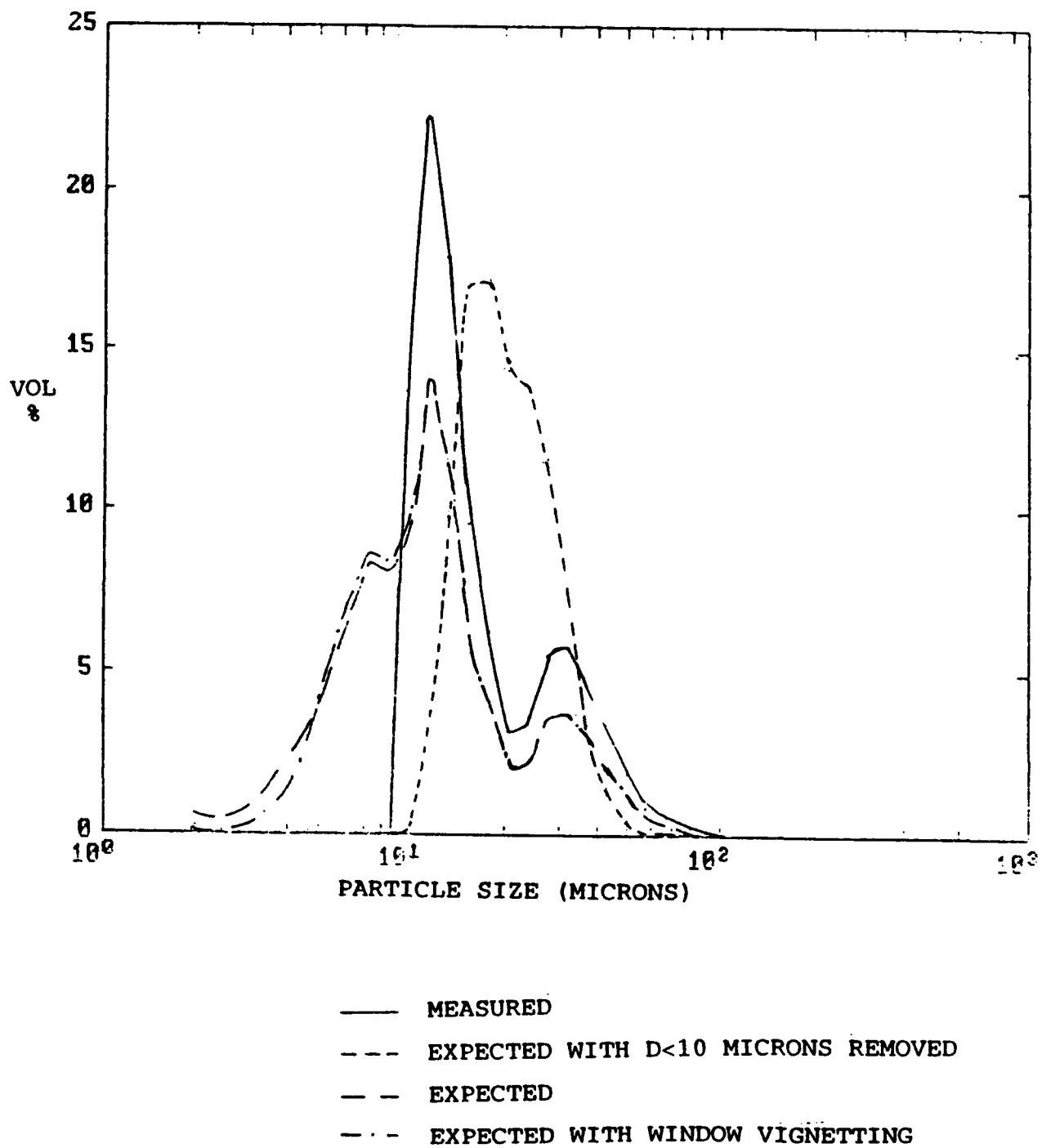


FIGURE 13: Malvern Measurements in the Motor

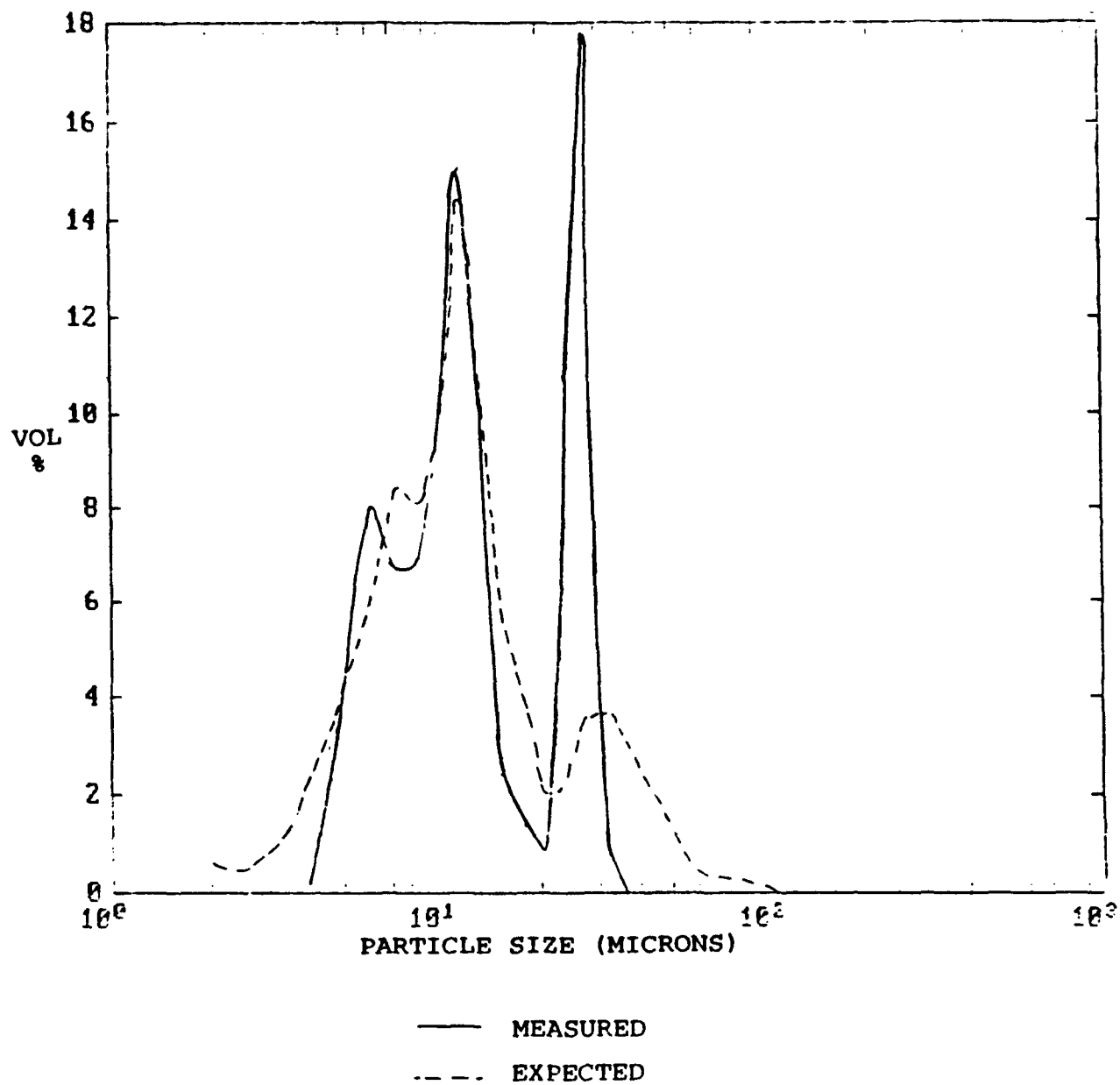


FIGURE 14: Malvern Measurements in the Plume

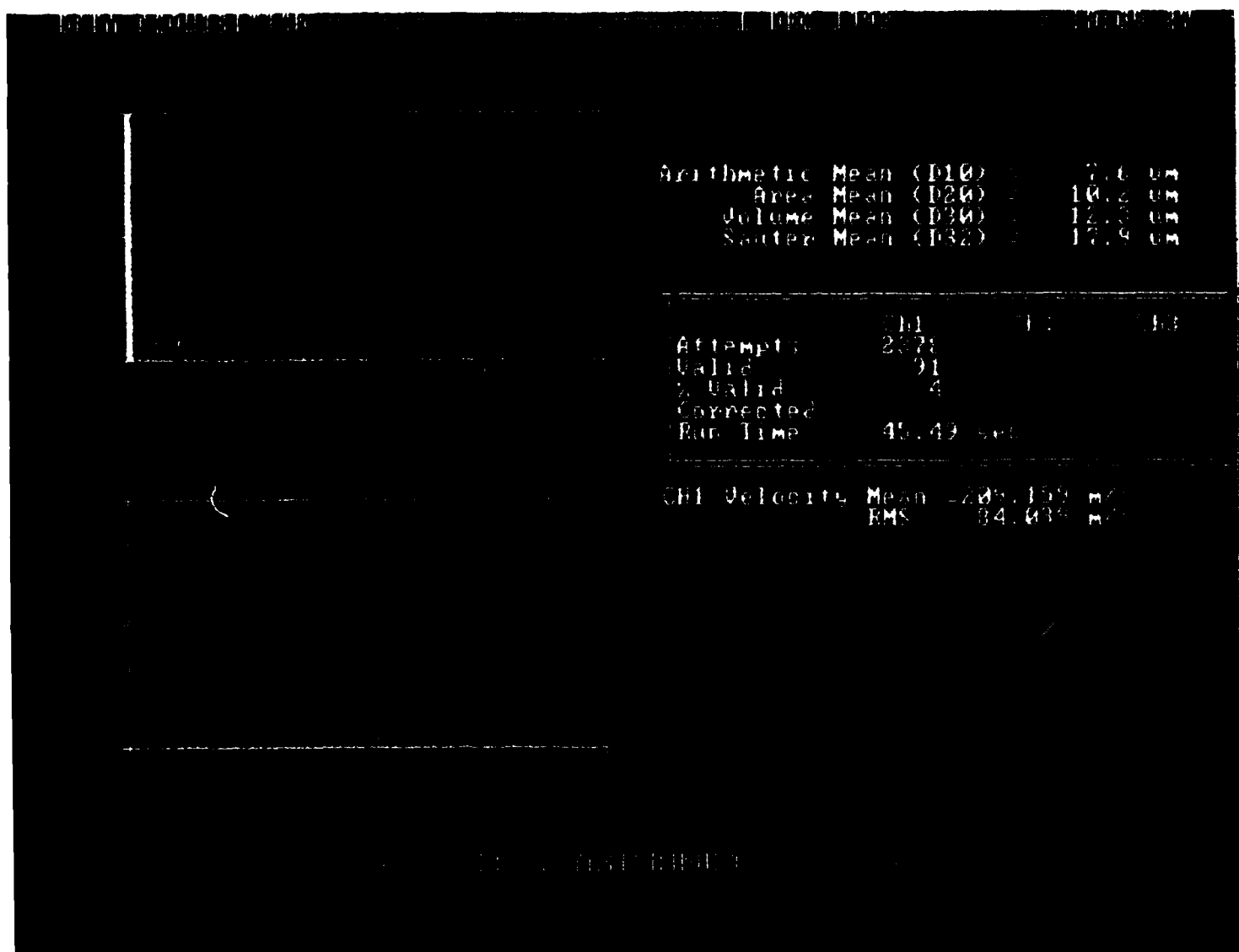


FIGURE 15: PDPA Results: Centerline

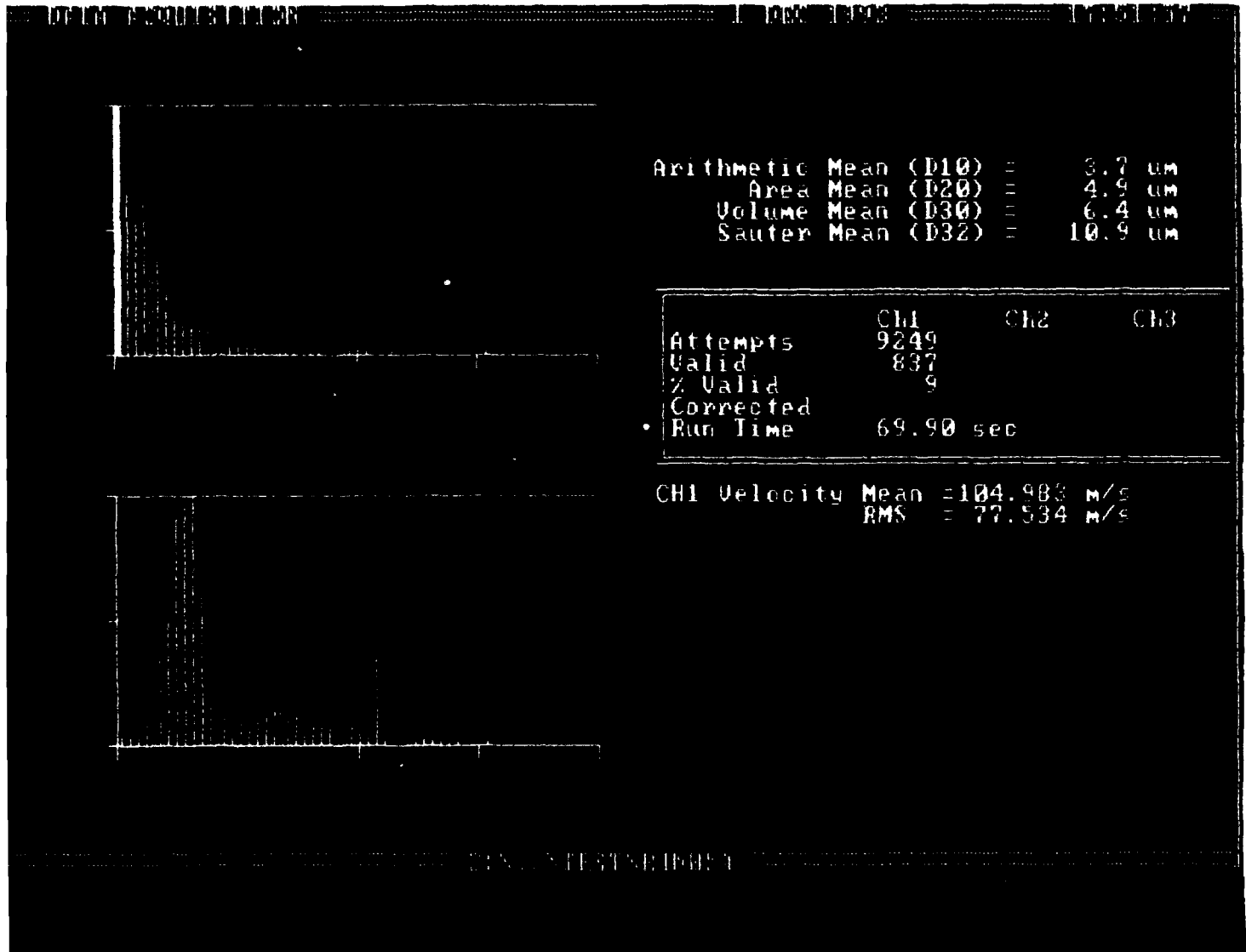
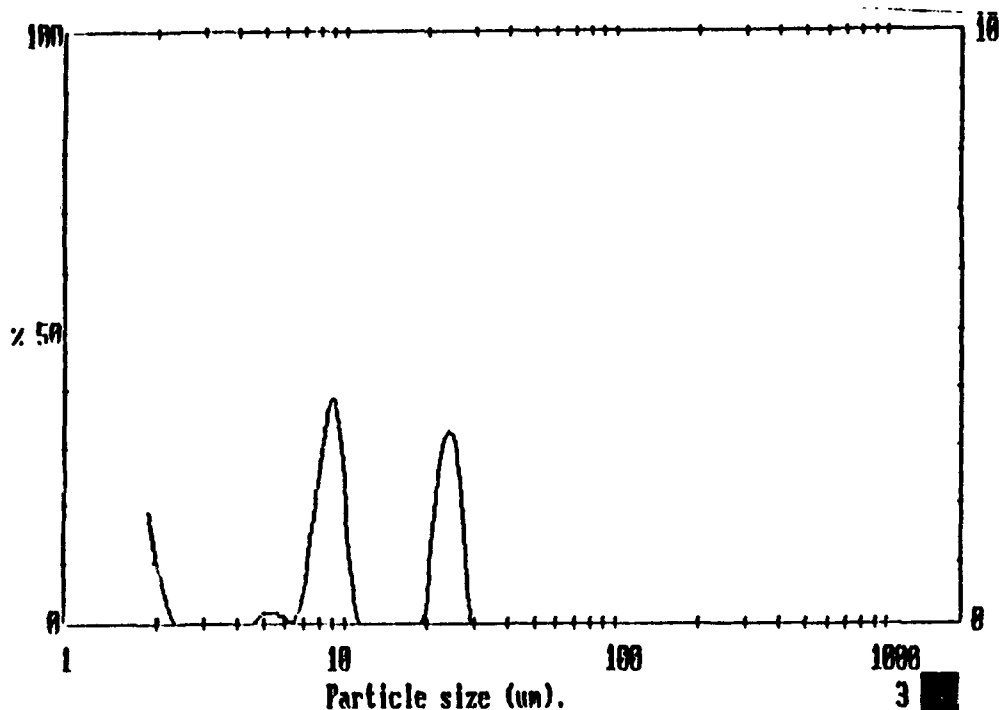


FIGURE 16: PDPA Results: Radially Spaced



Upper	in	Lower	Under	Upper	in	Lower	Under	Upper	in	Lower	Under	Span
				57.7	0.0	49.8	100	9.82	7.2	8.47	80.1	16.66
				49.8	0.0	43.0	100	8.47	4.9	7.30	75.2	D[4,3]
				43.0	0.0	37.0	100	7.30	0.7	6.30	74.5	4.70um
				37.0	0.0	32.0	100	6.30	0.2	5.43	74.2	
188	0.0	162	100	32.0	0.2	27.5	99.8	5.43	0.3	4.68	73.9	D[3,2]
162	0.0	140	100	27.5	5.3	23.8	94.5	4.68	0.0	4.05	73.9	1.31um
140	0.0	121	100	23.8	5.4	20.5	89.0	4.05	0.0	3.48	73.9	
121	0.0	104	100	20.5	0.3	17.7	88.8	3.48	0.0	3.02	73.9	D[y,0.9]
104	0.0	89.8	100	17.7	0.0	15.3	88.8	3.02	0.0	2.60	73.9	21.33um
89.8	0.0	77.5	100	15.3	0.0	13.2	88.8	2.60	0.1	2.23	73.8	
77.5	0.0	66.8	100	13.2	0.0	11.4	88.8	2.23	1.4	1.93	72.4	D[y,0.1]
66.8	0.0	57.7	100	11.4	1.5	9.82	87.3	1.93	72.4	0.50	0.0	0.70um
Source = Data:Input				Beam length = 50.0 mm				Model indep [4, 0]				D[y,0.5]
Focal length = 100 mm				Log. Diff. = 3.651				Volume Conc. = 0.0004%				1.24um
Presentation = pia				Obscuration = 0.3347				Sp.S.A 4.5951 m ² /cc.				Shape OFF
				Value distribution								

FIGURE 17: Malvern Data For PDPA [Ref. 11]

REFERENCES

1. Netzer, D. W., "*Classnotes, AE4452, Tactical Missile Propulsion*", Naval Postgraduate School, Monterey, California.
2. Price, E.W., "*Combustion of Metalized Propellants*", Fundamentals of Solid-Propellant Combustion, edited by Kuo, K.K. and Summerfield, M., AIAA Progress in Astronautics and Aeronautics v.90, AIAA, 1984, pp 479-513.
3. Victor, A.C., "*Solid Rocket Plumes*", To be published in forthcoming AIAA Progress in Astronautics and Aeronautics volume.
4. Traineau, J.C., Kuentzmann, P., Prevost M., Tarrin, P. and Delfour, A., "*Particle Size Distribution Measurements in a Subscale Motor for Ariane 5 Solid Rocket Booster*", AIAA 92-3049.
5. Eno, T.J., "*A Combined Optical and Collection Probe for Solid Propellant Exhaust Particle Analysis*", Master's Thesis, Naval Postgraduate School, Monterey, California, December 1989.
6. Kellman, L.J., "*Modification and Experimental Validation of a Combined Optical and Collection Probe for Solid Propellant Exhaust Analysis*", Master's Thesis, Naval Postgraduate School, Monterey, California, March 1991.

7. *"Mastersizer Instruction Manual"*, Malvern Instruments Ltd, Version IM100 Issue 3, October 1989.
8. McAtee, A.M., *"Adaptations to "Micopep" and "Rocket" to Allow Performance Evaluation of Multiple Grain and/or Airbreathing Motors"*, Master's Thesis, Naval Postgraduate School, Monterey, California, December 1991.
9. *"System 2600 Instruction Manual"*, Malvern Instrument Ltd, Version IM026 Issue 2.2, November 1991.
10. *"Laser Doppler Velocimeter/Doppler Signal Analyzer: Component User Manual"*, Aerometrics.
11. Yakin, B., *"Combuster And Nozzle Effects On Particle Behavior in Solid Rocket Motors"*, Master's Thesis, Naval Postgraduate School, Monterey, California, December 1993.

INITIAL DISTRIBUTION LIST

	No. Copies
1. Defense Technical Information Center Cameron Station Alexandria, VA 22304-6145	2
2. Superintendent ATTN: Library, Code 052 Naval Postgraduate School 833 Dyer Rd Monterey, CA 93943	2
3. Department Chairman, Code AA Department of Aeronautics and Astronautics Naval Postgraduate School 833 Dyer Rd Monterey, CA 93943	1
4. Weapon Systems Curriculum Office, Code 033 Naval Postgraduate School 833 Dyer Rd Monterey, CA 93943	1
5. Prof. D.W. Netzer, Code AA/Nt Department of Aeronautics and Astronautics Naval Postgraduate School 833 Dyer Rd Monterey, CA 93940	2
6. Lt. Paul Gomes 322 Davis Rd Portsmouth, RI 02871	2