

WL-TR-94-2025

DEVELOPMENT OF A BIPOLAR LEAD/ACID
BATTERY FOR THE MORE ELECTRIC AIRCRAFT

AD-A284 050



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MARCH 1994

INTERIM REPORT FOR 09/01/91-03/01/93

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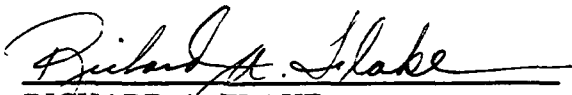
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13. ABSTRACT (Maximum 200 words) This report summarizes the development work completed under contract F33615-91-C-2142 for the time period of September 1991 through March 1993. The focus of the work was on the development of a filled polymeric composite substrate for use in a true bipolar lead acid battery. The contract goals for the development of the substrate material are as follows: Resistivity: ≤ 2 ohm-cm Thickness: ≤ 0.064 cm Weight: ≤ 150 mg/cm ² Area: ≥ 400 cm ² This report presents information towards achieving those goals.					
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Table of Contents

1. Introduction.....	1
2. Performance Projections.....	2
3. Conductive Filler Development.....	3
4. Fabrication Processing Techniques.....	9
5. Stability Testing.....	12
6. Battery Test Results.....	35
7. Next Steps.....	38

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List of Figures

Figure 1- Original Program Work Schedule.....	4
Figure 2- March 93 Work Schedule.....	5
Figure 3- Battery System Projections 5 Year.....	6
Figure 4- Battery System Projections 10 year.....	7
Figure 5- Stability Testing Fixture.....	13
Figure 6- Stability Testing Results.....	14
Figure 7- Typical Stability Test Graph.....	34
Figure 8- Battery Test Data.....	36

**WPAFB INTERIM TECHNOLOGY REPORT
BIPOLAR BATTERY DEVELOPMENT
SEPTEMBER 1991- MARCH 1993**

1. INTRODUCTION:

This report summarizes the development work completed by Johnson Controls Battery Group Inc., JCBGI, under U.S. Air Force Contract F33615-91-C-2142, Data Item #A006, for the time period of September 1991 through March 1993.

The focus of the work was on the development of a filled polymeric composite substrate for use in a true bipolar lead/acid battery. The role of the electrode in a bipolar design is paramount. The essential function of the substrate is to provide electronic conduction between the positive and negative active materials without permitting ionic conduction which would short out the cell. The substrate requirements are summarized below:

1. Electrical conductivity
2. Chemical insolubility in sulfuric acid in the potential windows of both electrodes
3. High oxygen and hydrogen overpotentials in sulfuric acid
4. Inertness to the electrode reactions
5. Ionically non-porous
6. Manufacturable

The contract goals for the development of the substrate material are found below.

Resistivity	$\leq 2 \Omega\text{-cm}$
Thickness	0.064 cm
Weight	$\leq 150 \text{ mg/cm}^2$
Area	$\geq 400 \text{ cm}^2$

JCBGI has focussed its development efforts on attaining or exceeding each of these goals. The original program schedule can be found in Figure 1, and the present program schedule can be found in Figure 2.

JCBGI has extensive experience in testing and developing potential conductive fillers for bipolar lead acid batteries. To date over 100 candidates have been screened and only a few have been identified as a possible filler materials. JCBGI concentrated on developing the most promising of these fillers for use in this program. A systematic approach was used to first, prove the stability of the conductive filler, and second, develop the processing and fabrication techniques needed to produce highly conductive, thin, non-porous substrates.

The stability of the conductive filler is tested through a series of tests developed at JCBGI. The most critical of which is exposing a compounded substrate to a constant potential of 1.5 volts for a period of time varying from 4 to 34 days. This method tests both the stability of the conductive filler at positive overpotentials and the porosity of the substrate.

After improving the compounding techniques to consistently produce non-porous parts, proof of concept testing was conducted on low voltage batteries. Four volt batteries were built to prove the viability of the substrate in an actual battery and to address the next major development issue- positive active material adhesion.

The above development issues will be discussed in further detail in the upcoming sections.

2. PERFORMANCE PROJECTIONS:

A set of preliminary performance requirements for the More Electric Aircraft (MEA) energy source were given to JCBGI, by Richard Flake of WPAFB, during the program kickoff meeting on December 12, 1991. The energy source requirements are summarized below:

Main Engine Starting:	150 kW, 30 sec
Ground Power:	25-75 kW, 30-45 min
Emergency Power:	75 kW, 10 min
APU Starting:	5-10 kW, 15 sec
Hybrid Emergency:	50-75 kW, 60 sec
Temperature Range:	-65F to 120F
Voltage Window:	270 Minimum; 330 Maximum

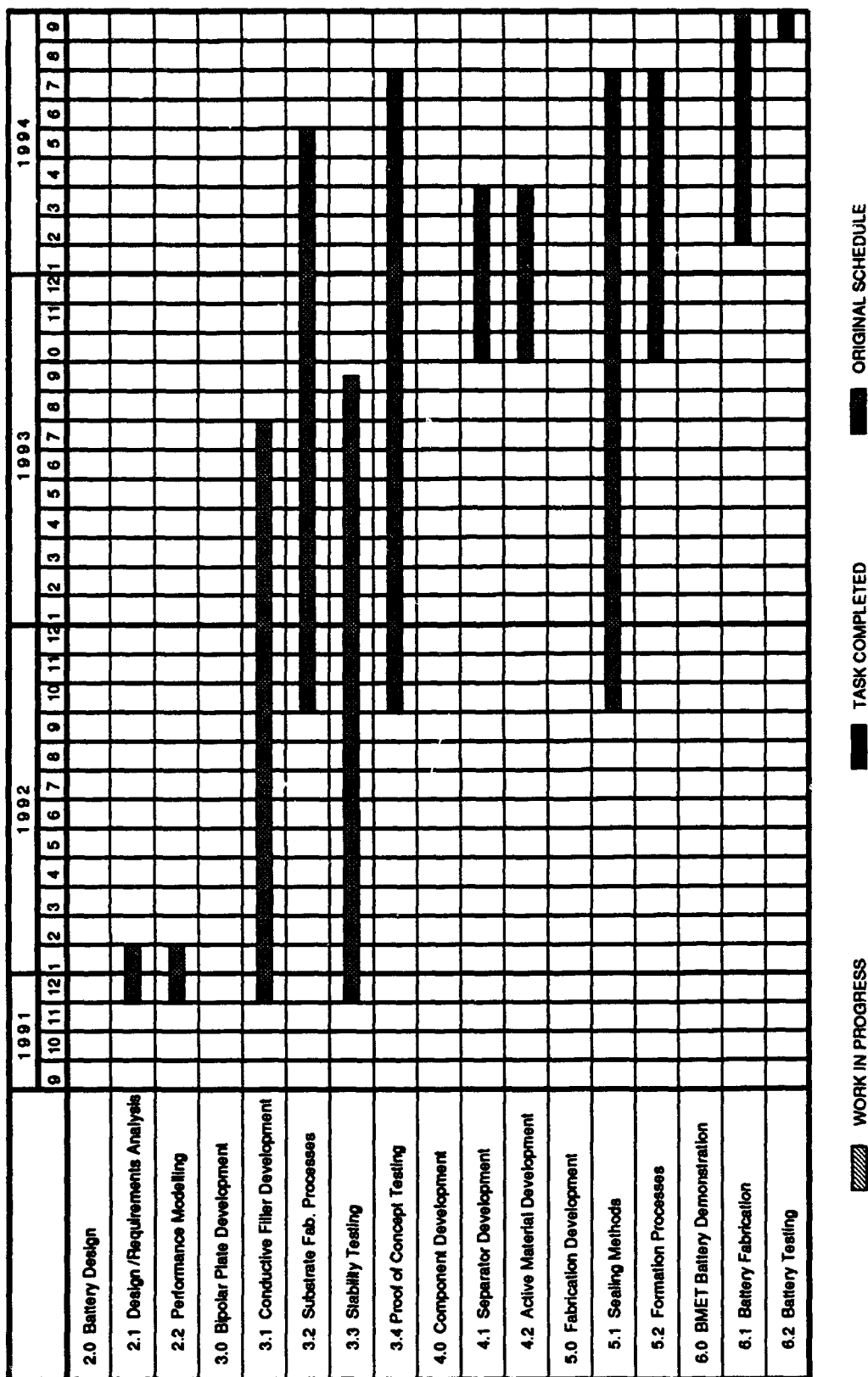
Given the above performance requirements, JCBGI using their proprietary lead/acid battery mathematical model (LABMM), designed four different bipolar batteries for both 5- and 10- year time frames. In designing the battery system for the 5-year time frame, JCBGI assumed that the program goals for the substrate development were reached and used conventional active materials. The 10-year battery systems were modelled assuming a thinner more conductive substrate and improved active materials. The results of the modelling can be found in Figures 3 and 4. The size of the battery can change dramatically depending upon the application ranging from as small as 0.18 ft³, 33 pounds to as large as 8.13 ft³, 1349 pounds.

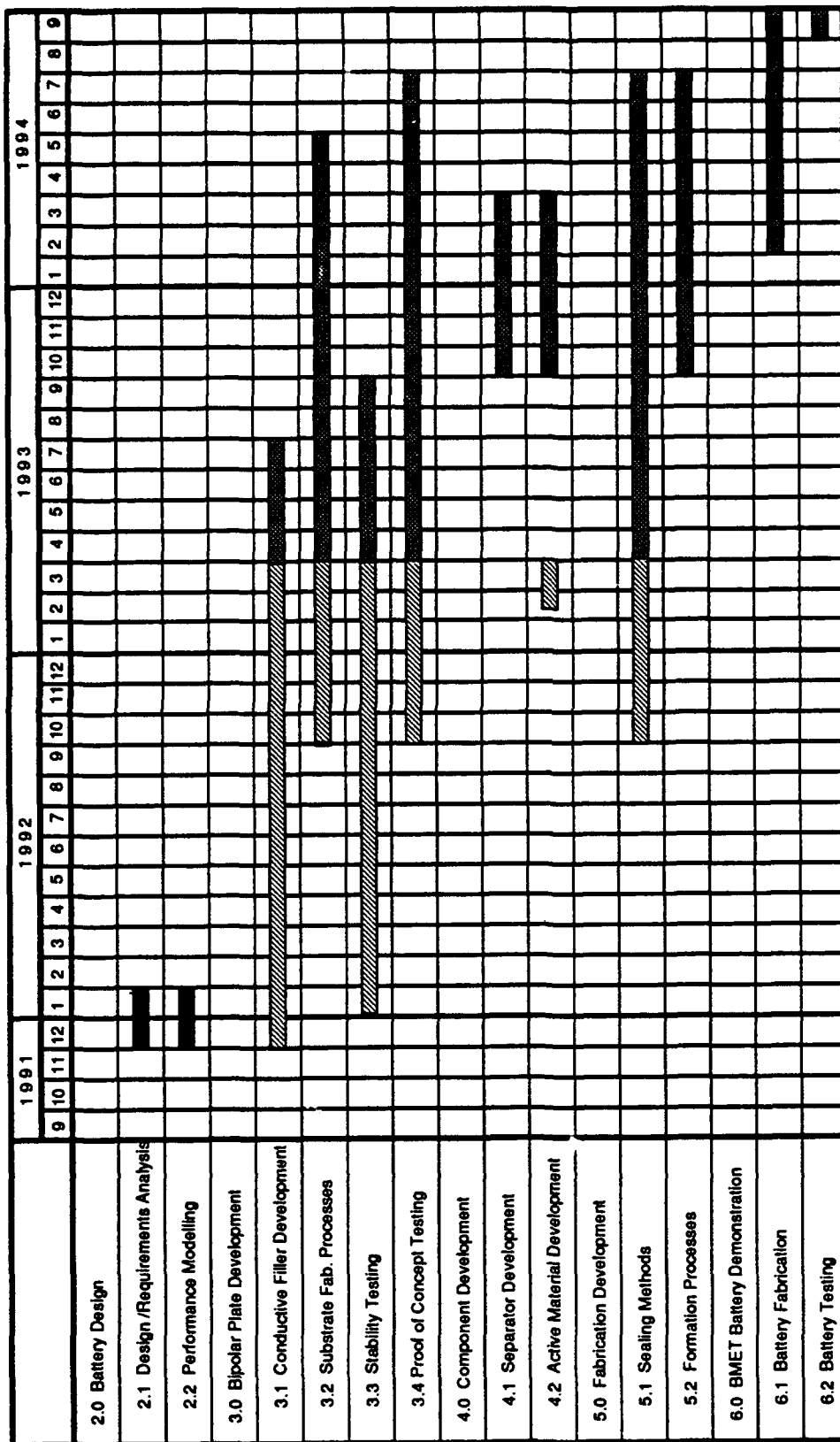
3. CONDUCTIVE FILLER DEVELOPMENT:

Building on the work performed in JCBGI's first bipolar lead/acid program with WPAFB, conductive filler development was continued in the same promising areas. Towards the end of the initial program JCBGI was having some success using one type of conductive filler that showed stability in the battery environment. Also, other conductive fillers were investigated including SiC, and Ceramic plaques and coated fibers using the same filler chemistry. JCBGI renewed working relationships with two suppliers of Conduflow, and found different companies that were interested in developing coated glass fibers with Conduflow and ceramic plaques with Conduflow.

The investigation into coated glass fibers by Photon Energy Systems was not successful. They made four different attempts on making fiber, but each of the trials could not be used because of a number of problems. First, the glass used would not withstand the acid environment. Second, the material was not uniform and was highly resistive. Third, the adhesion of the coating to the glass was virtually nonexistent which made handling next to impossible and completely eliminated any chance of compounding the material into plastic. Finally, they could only make the material on long strands of glass, 2-6", which could not be used for our application. After the fourth shipment of subpar material, which they said would be the best that they could make, activity in this area was discontinued.

After the attempt at glass fibers was not successful, the focus of development efforts was placed on Conduflow powder. Two companies were contacted and samples obtained from each. The samples were extremely similar in particle size and appearance but the material from Magnesium Elektron Inc. (MEI) was much more conductive than the Crystal Research Inc. (CRI) material. Leach tests of both materials proved that they were stable in





WORK IN PROGRESS
 TASK COMPLETED
 ORIGINAL SCHEDULE

Figure 2
BMET Program Schedule

REQUIREMENTS MET	BATTERY DIMENSIONS	BATTERY VOLUME	BATTERY WEIGHT	W/kg	W/cm3	W-hr/kg	W- hr/cm3
Main Engine Starting APU Starting Hybrid Emergency	17.6"x15.5"x15.5"	2.45 ft3	450 lbs	747.9	2.2	12.25	0.036
Main Engine Starting Ground Power Emergency Power APU Starting Hybrid Emergency							
Scenario 1 30 minute ground power capacity	27.4"x19.7"x19.7"	6.15 ft3	1000 lbs	62.2	0.16	31.08	0.081
Scenario 2 45 minute ground power capacity	36.2"x19.7"x19.7"	8.13 ft3	1349 lbs	46.1	0.12	34.56	0.092
APU Starting	16.5"x4.33"x4.33"	0.18 ft3	33 lbs	705.0	2.1	11.75	0.036

Figure 3
 BMET PERFORMANCE REQUIREMENTS
 BIPOlar BATTERY SPECIFICATIONS
 Near Term Projections (within 5 years)
 330 Volt Battery Systems

REQUIREMENTS MET	BATTERY DIMENSIONS	BATTERY VOLUME	BATTERY WEIGHT	W/kg	W/cm ³	W-hr/kg	W- hr/cm ³
Main Engine Starting APU Starting Hybrid Emergency	14.4"x15.5"x15.5"	2.00 ft ³	389 lbs	895.3	2.8	14.17	0.044
Main Engine Starting Ground Power Emergency Power APU Starting Hybrid Emergency							
Scenario 1 30 minute ground power capacity	21.6"x19.7"x19.7"	4.85 ft ³	864 lbs	72.0	0.21	35.97	0.103
Scenario 2 45 minute ground power capacity	29.9"x19.7"x19.7"	6.72 ft ³	1235 lbs	50.6	0.15	37.77	0.111
APU Starting	15.2"x4.33"x4.33"	0.16 ft ³	31 lbs	772.0	2.3	12.87	0.041

Figure 4
 BMET PERFORMANCE REQUIREMENTS
 BIPOLAR BATTERY SPECIFICATIONS
 Far Term Projections (10 years)
 330 Volt Battery Systems

an acid environment, but determining their stability at the positive potential of a lead acid system was more difficult. The one problem with Conduflow was that it is not stable at the negative potentials of the battery. To be able to use Conduflow as a substrate material a multi layer system would have to be used. Drawing on JCBGI's experience from the initial WPAFB contract, C-black was determined to be the ideal second half of the substrate. C-black is highly conductive, lightweight, readily available, and stable at the negative potentials. However, C-black is not stable at positive potentials, making it an ideal complement to Conduflow. During the first contract much effort was made in developing the C-black material for use as the bipolar substrate before it was proved to be unstable at positive overpotentials. Because the development work was virtually completed for the C-black substrate, efforts were concentrated on proving the stability of Conduflow. The evolution of the substrate into a two piece laminate causes the interface between the halves to become a development issue. The manufacturing developments will be discussed later.

The attempt by Materials and Electrochemical Research Inc.(MER) to produce a dense plaque of Conduflow was also not successful. The plaque was found to be dissolve and become non conductive when placed in contact with sulfuric acid. After the initial trial no further attempts were made.

At the same time that efforts were shifted towards powder Conduflow, a concurrent effort was made to develop a SiC material that could be used as a conductive filler. Qualification screening at JCBGI showed that SiC was a possible material that would be stable at both the positive and negative potentials of a battery. Samples of SiC were received from MER and found to be too resistive to use as a conductive filler. Because of this, JCBGI's efforts became even more focussed on the promising Conduflow compound.

Initial compounding trials had yielded substrates that were tested to be non-porous and stable during 4- and 5-day stability testing. Through improvements in compounding procedures, which will be discussed later, consistently stable substrates were produced. After extensive testing it was concluded that Conduflow was indeed a viable conductive filler for a bipolar lead/acid battery. Development on different fillers were discontinued to concentrate on optimizing the Conduflow and the compounding parameters needed to produce working bipolar electrodes.

The Conduflow received from MEI proved to be consistently better than any other material tested. MEI had also supplied all of the material to JCBGI, worth over \$110,000, at no cost over the first 18 months of this contract. Because of the potential for this product, MEI and JCBGI entered into a joint effort for the development of Conduflow. MEI would

be responsible for developing the techniques of producing and improving the material, while JCBGI would test and qualify it. MEI would also lend their compounding expertise to the development effort.

The first area to be addressed in the development of Conduflow was the particle size. It was thought that smaller particle size would eliminate the chance for porosity by being more easily wetted by the plastic resin. Initial trials using very fine particle size (< 1 micron) material yielded interesting results. The material was not as conductive as thought and also displayed more porosity than previous trials. After discussion with MEI and their compounding consultants, it was determined that the ultra fine particle size caused more problems than it solved. The small particle size lead to many more interfaces between the plastic and Conduflow which lead to higher porosity. The small particle size also caused the increase in resistance of the part. Since the conductivity of the material is dependent upon particle to particle contact, having uniform, ultra-fine particles make it more difficult to have a chain of contact throughout the thickness of the material. In fact, after further investigation the optimized particle size may be in the 10-20 micron size. Presently the particle size used is 3-5 microns, which is a result of the type of manufacturing method used. An entirely different method of production would be needed to produce material with larger particle size. After discussion with MEI, it was decided because of the timeframe of this contract, investigation into larger particle size Conduflow would not be feasible. Efforts would be better spent on optimizing the present form of Conduflow and compounding parameters.

4. FABRICATION PROCESSING TECHNIQUES:

With the focus of this contract on producing a polymer based bipolar substrate, JCBGI again moved forward from where the initial contract with WPAFB left off. Resins investigated during the first 18 months of this contract were LDPE, LLDPE, HDPE, Kynar, and PTFE. In the past all of these materials were compounded by hand in small batches using a mortar and pestle. This process was very limited in batch size and caused a large degree of variability from trial to trial. To eliminate these problems, JCBGI had proposed to use contract funding to purchase a Brabender type of mechanical mixer. Prior to beginning the contract, another group in JCI had purchased a similar machine. Because of the high loading levels needed (75-85% by weight) to make the material conductive enough, special equipment is needed to compound the material. The Brabender mixer was proposed to do this compounding. However, initial trials with the mixer showed that it would not

work on loading levels higher than 70%. Because of this, the Brabender mixer was never purchased and JCBGI began looking for different methods to compound the material. It was recommended to JCBGI that the best equipment for JCBGI to use would be a twin screw extruder.

Prior to purchasing a twin screw extruder, JCBGI had material compounded by two different vendors using similar equipment to verify that it would work at the loading levels needed. Two trials were run by Dr. John Muzzy, of Georgia Tech University, and MEI to determine if a twin screw extruder would work for our application. The success of the trials convinced JCBGI to purchase an extruder instead of the Brabender mixer. The unit was ordered in December of 1991, and was ready to run at JCBGI in late March of 1993.

It was thought at the beginning of the contract that the cause of the porosity problems in the substrates was caused by trapped gases during compression molding of the material into sheets. To eliminate the trapped gas, JCBGI and a local vendor, Molded Rubber and Plastics, designed a vacuum compression mold. In theory, by compression molding under a vacuum, any trapped gas in the material would be removed and result in a porous free part. After many different trials, stability testing showed that the parts made with the vacuum mold were not any better than parts in the conventional manner. Because of the cost of the process, \$65-\$75 per 12" x 12" part, and the large amount of material needed per trial, 10+ pounds, work in this area was discontinued.

Because of the success late in the initial program using LDPE as the base resin, JCBGI began its development using the same resin. The plastic, called Microthene, is purchased from Quantum Chemical Corp. in powder form. Using a powder form of the plastic results in a much more uniform dispersion of the filler material and helps eliminate porosity. The conductive filler and the plastic are dry mixed prior to melt blending. The material is then compression molded, laminated to the C-black material, and cut to size for testing.

JCBGI's approach in developing processing parameters for a this contract were using one resin, LDPE, to first prove the stability of the conductive filler, second to optimize part conductivity and eliminate part porosity, and third to reduce part thickness from around 0.070" to the contract goal of 0.025". To prove the stability of the filler, test parts were hand compounded at lower loading levels 80-82.5% by weight, and compression molded to a thickness of 0.070" and stability tested. After a series of successful tests, no change in resistivity after 4-34 days while maintaining a positive potential of 1.5 volts, the emphasis changed towards making the parts thinner, and more conductive.

Because the properties of the filler dictate a loading level of 80-85% by weight is needed to produce part conductivity high enough for use in a battery, JCBGI began investigating additives that would improve the physical properties of the substrate. MEI has extensive experience with producing conductive materials and suggested the following additives which could improve part conductivity, reduce porosity, and/or improve manufacturing- coupling agents, paraffin oil, stearic acid, ethylene vinyl acetate, and fumed silica. After investigating all of these additives it was found that only coupling agents offered the only real advantage. The others while improving part conductivity caused additional problems with part porosity.

Since initial efforts with coupling agents were so promising, a series of trials were conducted to optimize the loading level and addition method of the material. Two different coupling agents were attempted per the advice of Kenrich Chemical Inc.. Of the two materials, one was clearly better at all loading levels. The coupling agents are used in small quantities, 0.01-1.00% by weight, of the filler. They are designed to bond between the filler and the base resin. The particular material used, LICA 38, is fumed silica which has been treated with the coupling agent. The coupling agent improved part conductivity and reduced part porosity at levels between 0.35%-1.00%. At levels higher than 1.00%, the additional material did not offer any improvement over the 1.00% level. It was also found that the order of addition was critical to the performance of the material. It was much more effective to blend the coupling agent with the conductive filler prior to adding the resin.

All of the development work was done using the powder form of the LDPE. This material had given JCBGI the best results to date. At the same time, other base resins were evaluated but none were found to have all the manufacturing advantages of LDPE. Recently, JCBGI has also been investigating HDPE which would give the substrate more high temperatures capabilities and a more rigid structure. Depending upon the application, the resin with the best properties can be used.

JCBGI also looked at using PTFE as a base resin. When loaded at 70-75%, the material was highly conductive, 1 Ω -cm, but was also highly porous. To combat the porosity, resin impregnation was attempted. The material was treated by Imprex Inc., with a polycarbonate based liquid resin under a vacuum. It was thought that the resin impregnation would remove the porosity while not hindering the conductivity. It did neither. The parts remained porous and became more resistive after the resin treatment. The activity as well as PTFE development was stopped.

Another resin investigated was Kynar. The material showed initial promise during hand compounding trials as producing conductive and non-porous material. However, the high temperatures needed to mold and compound material (375C) caused problems with the Conduflow. JCBGI attempted to use blends of LDPE and Kynar which resulted in conductive but highly porous material. Because of the problems with Kynar and the promise of LDPE, development in this area was discontinued.

The development efforts for the C-black part of the substrate were much easier because of the vast experience JCBGI has using C-black for the Zn/Br₂ battery development program. JCBGI had screened several different types of C-black before choosing a Ketjenblack material, EC600JD, from Azko Chemical. After several iterations, the loading level of C-black in the LDPE was found to be 19% by weight. This afforded parts with conductivity of 1-1.6 Ω -cm and enough flexibility to be used as a bipolar substrate.

The fact that Conduflow is not stable at negative potentials meant that a laminate system would have to be used to protect it. This led to development of the interface between the Conduflow substrate and the C-black substrate. Initial trials of compression molding the two halves together resulted in a part with higher resistivity than the sum of the pieces. This was caused by the "skins" which are formed on the surfaces of each of the flat sheets during compression molding. Two different methods were attempted to remove the effect of laminating the "skins" together. The first was to add C-black to the interface prior to compression molding the two pieces together. This proved to be effective but was rather difficult to apply a uniform coating of the C-black to the interface. The second method attempted was to remove the skins on the interface by using sand paper. This worked better than the C-black addition and was much easier to use. Sanding the faces of the sheet stock prior to lamination resulted in a 50-75% reduction in part resistivity and no effect on part stability.

5. STABILITY TESTING:

A diagram of the stability test fixture can be found in Figure 5. A summary of all stability tests run can be found in Figure 6. Figure 7 is a graph of a typical stability test.

- A. Bipolar Substrate
- B. Reference Electrode Socket
- C. Counter Electrode
- D. Current Collector
- E. Resistance Sensor

- F. Lexan Block
- G. Spacer with Sensor Socket
- H. Gasket
- I. Counter Electrode Bushing
- J. Nuts and Bolts for Clamping

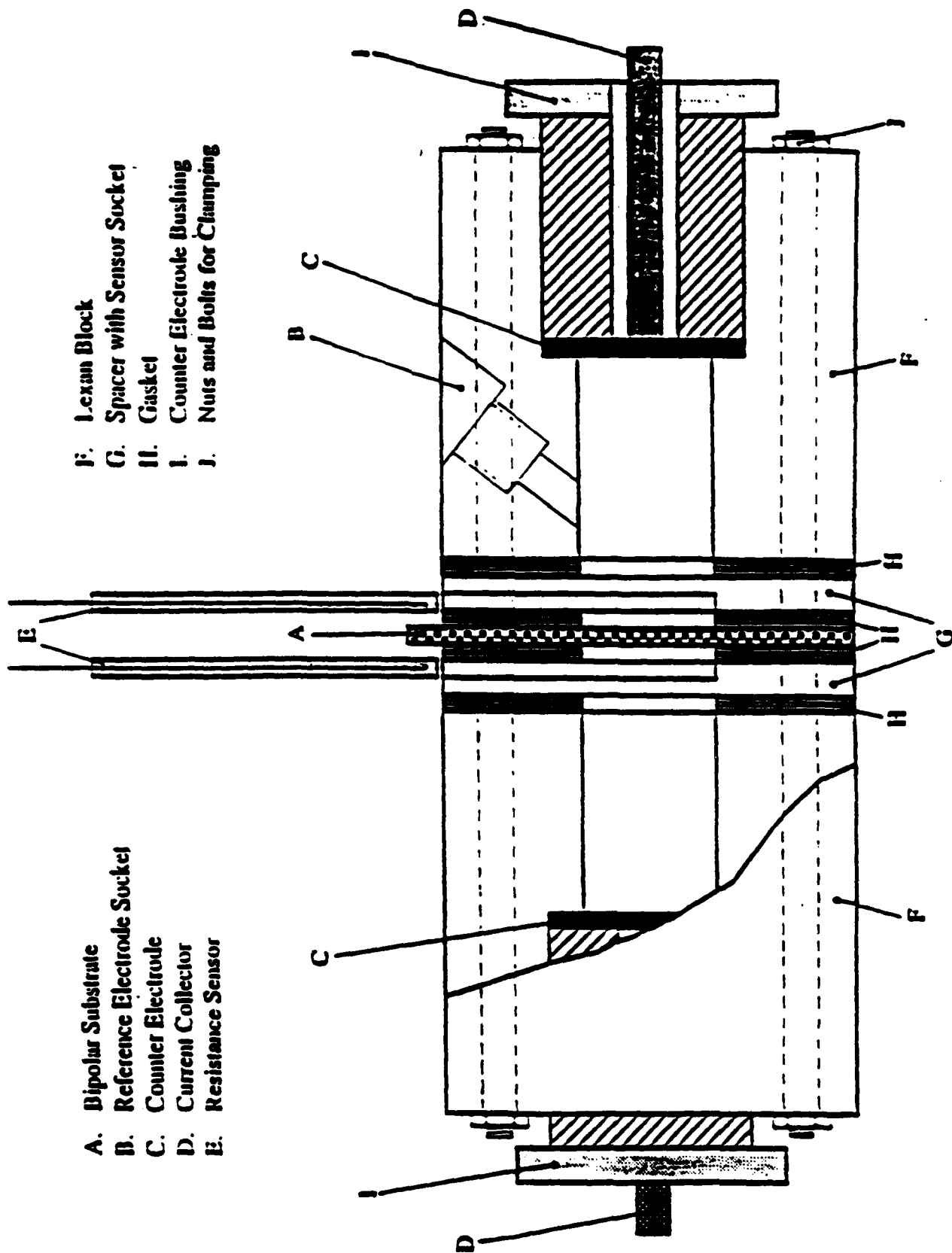


Figure 5 Stability Test Fixture

SAMPLE NO.	DATE	MATERIAL COMPOSITION	STABILITY TESTING				RESISTIVITY (OHM-IN) BEFORE	THICKNESS (INCH) BEFORE	RESISTIVITY (OHM-IN) AFTER	THICKNESS (INCH) AFTER	RESISTANCE (OHM) BEFORE	RESISTANCE (OHM) AFTER	PERCENT CHANGE (%)
			21-OCT-1992	RESISTIVITY (OHM-IN) BEFORE	THICKNESS (INCH) BEFORE	RESISTIVITY (OHM-IN) AFTER							
47A	4/2/92	LAMINATED 85% GC23N W/O CARON L/12N		6.248	0.023	0.960	0.960	0.042	0.999	0.042	0.960	0.999	
	RMI LAB BOOK PG.103	15% MICROTHENE 4.5 M.I. C-PLASTIC		7.302	0.024	0.640	0.640	0.042	9.999	0.042	0.640	9.999	
				4.622	0.023	1.060	1.060	0.042	9.936	0.042	1.060	9.936	
				6.004					0.311				36.616
48A	4/2/92	LAMINATED 85% GC23N WITH CARON L/12N		9.921	0.025	1.000	1.000	0.040	9.043	0.040	1.000	9.043	
	RMI LAB BOOK PG.103	15% MICROTHENE 4.5 M.I. C-PLASTIC		9.350	0.024	1.500	1.500	0.040	14.764	0.040	1.500	14.764	
				10.417	0.024	0.740	0.740	0.040	7.203	0.040	0.740	7.203	
				9.096					10.630				7.415
52	4/9/92	LAMINATED 84% GC23N & 16%PTFE DUPONT TO C-PLASTIC & pb FOIL SINGLE APPLICATION OF RESIN		0.765	0.053	19.500	19.500	0.062	123.825	0.062	19.500	123.825	
	RMI LAB BOOK PG.107					17.500	17.500	0.062	111.125	0.062	17.500	111.125	
						38.000	38.000	0.063	237.470	0.063	38.000	237.470	
									157.474				1696.534
53	4/9/92	LAMINATED 84% GC23N & 16%PTFE DUPONT TO C-PLASTIC & pb FOIL DOUBLE APPLICATION OF RESIN		23.420	0.061	169.000	169.000	0.054	1232.130	0.054	169.000	1232.130	
	RMI LAB BOOK PG.107					161.000	161.000	0.054	1173.012	0.054	161.000	1173.012	
						100.000	100.000	0.054	1370.662	0.054	100.000	1370.662	
									1250.870				5273.261
54B	4/14/92	LAMINATED 85% GC23N & 15% MICROTHENE 1.5 M.I. WITH pb FOIL		1.919	0.040	0.400	0.400	0.040	4.331	0.040	0.400	4.331	
	RMI LAB BOOK PG.113					0.403	0.403	0.040	4.754	0.040	0.403	4.754	
						0.470	0.470	0.040	4.626	0.040	0.470	4.626	
									4.570				130.120
55B	4/14/92	LAMINATED 85% GC23N-15% MICROTHENE 1.5 M.I. W/O pb FOIL		3.201	0.030	1.730	1.730	0.030	22.703	0.030	1.730	22.703	
	RMI LAB BOOK PG.113					2.350	2.350	0.030	30.840	0.030	2.350	30.840	
						3.600	3.600	0.030	47.244	0.030	3.600	47.244	
									33.596				924.000
71A	4/24/92	LAMINATED THICK/THICK GC23N-1 /C-PLASTIC		0.564	0.206	0.300	0.300	0.209	0.735	0.209	0.300	0.735	
	RMI LAB BOOK PG. 120			0.568	0.208	0.430	0.430	0.211	0.802	0.211	0.430	0.802	
				0.530	0.208	0.400	0.400	0.208	0.757	0.208	0.400	0.757	
				0.554					0.765				30.065
72A	4/24/92	LAMINATED THIN/THIN GC23N-2 /C-PLASTIC		0.521	0.208	0.495	0.495	0.210	0.920	0.210	0.495	0.920	
	RMI LAB BOOK PG. 120			0.502	0.208	0.410	0.410	0.210	0.769	0.210	0.410	0.769	
				0.523	0.207	0.500	0.500	0.209	0.942	0.209	0.500	0.942	
				0.515					0.880				70.763
73A	4/24/92	LAMINATED THICK/THIN GC23N-3 /C-PLASTIC		5.334	0.031	3.800	3.800	0.031	48.260	0.031	3.800	48.260	
	RMI LAB BOOK PG. 120			3.076	0.032	8.800	8.800	0.031	111.760	0.031	8.800	111.760	
				2.625	0.033	6.400	6.400	0.032	70.740	0.032	6.400	70.740	
				3.670					79.507				2063.769
74A	4/24/92	LAMINATED THIN/THIN GC23N-4 /C-PLASTIC		4.675	0.032	4.300	4.300	0.032	52.904	0.032	4.300	52.904	
	RMI LAB BOOK PG. 120			5.249	0.033	5.100	5.100	0.033	60.845	0.033	5.100	60.845	
				5.462	0.031	5.500	5.500	0.032	67.667	0.032	5.500	67.667	
									60.472				1079.129

REVISED- SAMPLE NO.	21-OCT-1992 DATE	MATERIAL COMPOSITION	STABILITY TESTING						PERCENT CHANGE (%)
			RESISTANCE (OHM) BEFORE	THICKNESS (INCH) BEFORE	RESISTIVITY (OHM-CM) BEFORE	RESISTANCE (OHM) AFTER	THICKNESS (INCH) AFTER	RESISTIVITY (OHM-CM) AFTER	
75A	4/24/92 RMI LAB BOOK Pg. 120	LAMINATED THICK/THIN GC23N-5 /C-PLASTIC	0.350 0.500 0.280	0.121 0.122 0.123	1.139 1.072 0.896 1.302	0.570 0.520 0.320	0.122 0.123 0.123	1.839 1.668 1.028 1.509	15.906
76A	4/24/92 RMI LAB BOOK Pg. 120	LAMINATED THIN/THICK GC23N-6 /C-PLASTIC	0.285 0.275 0.270	0.124 0.126 0.126	0.905 0.859 0.844 0.869	2.350 2.500 2.300	0.125 0.124 0.123	7.402 8.192 7.362 7.652	700.247
77A	5/12/92 RMI LAB BOOK Pg. 130	LAMINATED GC23N-A-3/92 PB-FOIL C-PLASTIC	0.360	0.026	5.451				
		LAMINATED GC23N-B-3/92 PB-FOIL C-PLASTIC	0.220	0.026	3.331				
		LAMINATED GC23N-B-3/92 PB-FOIL C-PLASTIC	0.660	0.027	9.624				
78A	6/5/92 RMI LAB BOOK Pg. 131	LAMINATED GC23N, MICROTHENE 6 C-PLASTIC	0.228 0.185 0.210	0.030 0.030 0.027	2.992 2.428 3.062	0.300 5.800 0.660	0.030 0.030 0.028	4.907 76.115 9.080	66.667 3035.135 203.061
		POROSITY TEST AFTER							
		RESIN TREATMENT 3/92							
		TESTING ON 2 & 3 INTERRUPTED AFTER 3 POINT							
79A	5/20/92 RMI LAB BOOK Pg. 117	LAMINATE GC23N-1-85% MICROTHENE/CAPON GC23N-2-85% MICROTHENE GC23N-3-80.3% KYNAR GC23N-4-80.3% KYNAR/CAPON	0.520 0.335 0.490 0.435	0.046 0.044 0.039 0.037	4.451 2.997 4.946 4.629	3.700 2.200 21.000 13.500	0.046 0.044 0.041 0.038	31.667 19.685 201.652 139.867	611.538 556.716 3976.655 2921.779
80A	5/27/92 RMI LAB BOOK Pg. 139/141	LAMINATE GC23N-1-85% MICROTHENE/CAPON GC23N-2-85% MICROTHENE GC23N-3-80.3% KYNAR GC23N-4-80.3%	0.360 0.495 0.223 0.305	0.040 0.039 0.044 0.042	3.543 4.997 1.995 2.859	0.445 0.525 0.253 0.350	0.040 0.038 0.041 0.041	4.380 5.439 2.264 3.361	23.611 8.852 13.453 17.553

REVISED- 21-OCT-1992

STABILITY TESTING

SAMPLE NO.	DATE	MATERIAL COMPOSITION	RESISTANCE (OHM) BEFORE	THICKNESS (INCH) BEFORE	RESISTIVITY (OHM-CM) BEFORE	RESISTANCE (OHM) AFTER	THICKNESS (INCH) AFTER	RESISTIVITY (OHM-CM) AFTER	PERCENT CHANGE (%)
KYNAR/CAPOW									
LAMINATED									
81A	6/9/92	RMI Lab Book GC23N, MICROTHENE & Pg.145 Remake of test 78 Using 5/92							
C-PLASTIC									
5/92-1R									
5/92-2R									
5/92-3R									
5/92-4R									
82A	6/10/92	RMI Lab Book Pg.147 Tested sample from [redacted] Corp. 6/26/92	0.380	0.098	1.527	23.300	0.098	93.604	6031.579
laminated									
and C-Plastic									
Kynar 7201 & 711									
C-Plastic									
Capow									
701-7201									
751-7201									
851-711									
701-7201 & Capow									
751-7201 & Capow									
851-711 & Capow									
LAMINATES									
83A	6/30/92	R.M.H. lab book pg.154	1.700	0.031	21.590	26.300	0.031	334.011	1447.050
C-Plastic, Pb Foil									
711 Kynar & Pb Dust									
701-W/CA-FOIL									
701-W/CA-DUST									
701-W/O CA-FOIL									
701-W/O CA-DUST									
751-W/CA-FOIL									
751-W/CA-DUST									
751-W/O CA-FOIL									
751-W/O CA-DUST									
84A	7/13/92	R.M.H. lab book pg.159/160	3.700	0.022	66.213	71.500	0.025	1125.986	22243.750
Kynar-711									
C-Plastic									
.013-C-Plastic									
.020-									
.030-									
.040-									
.050-									
The sample .050									
was									
repressed at 540 F									
for better									
resistivity									
85A	7/22/92	R.M.H. lab book pg.167	0.110	0.013	3.331	0.115	0.012	3.773	13.250
LEAD DUST &									
RADEL(POLYSULFONE)									
PREMIEX W/1.1.1									
DATED PREPARED AT									

REVISED- 21-OCT-1992

STABILITY TESTING

SAMPLE NO.	DATE	MATERIAL COMPOSITION	RESISTANCE (OHM) BEFORE	THICKNESS (INCH) BEFORE	RESISTIVITY (OHM-CM) BEFORE	RESISTANCE (OHM) AFTER	THICKNESS (INCH) AFTER	RESISTIVITY (OHM-CM) AFTER	PERCENT CHANGE (%)
94A	7/28/92	599E 30 TONS 55% BY WT. LAMINATE R.M.H. lab book Pg. 170/171							
		KYNAR-711							
		KETJENBLACK WITH KYNAR-711							
		14% KET/KYN-050	0.305	0.068	1.766	0.480	0.068	2.779	57.377
		14% KET/KYN-040	0.380	0.061	2.453	0.400	0.061	2.582	5.263
		14% KET/KYN-030	0.500	0.051	3.060	0.740	0.051	5.713	46.000
		Pb/RADEL	0.066	0.025	1.039	0.570	0.025	0.976	763.636
95A	7/30/92	LAMINATE R.M.H. lab book Pg. 173/174							
		KYNAR-7201							
		KETJENBLACK WITH KYNAR-7201							
		14% KET/KYN-050	0.305	0.066	1.819	0.345	0.066	2.858	13.115
		14% KET/KYN-040	0.295	0.061	1.904	0.400	0.061	2.502	35.993
		14% KET/KYN-030	0.270	0.052	2.044	1.150	0.052	8.707	325.926
		14% KET/KYN-020	0.255	0.043	2.335	4.200	0.043	38.454	1547.059
		14% KET/KYN-026	0.243	0.047	2.036	SAMPLE FOR SHOW			
96A	8/10/92	LAMINATES W/Microthene Ketjenblack & Microthene R.M.H. lab book Pg. 175/178							
		325E/3000E used to make laminates	0.620	0.071	3.438	0.640	0.071	3.549	3.226
		80% KYNAR-96A-1	0.460	0.063	2.875	0.440	0.063	2.750	-4.348
97A	8/18/92	LAMINATES Kynar (8/92) RMI lab book Pg. 181							
		400E/3000E used to make laminates	0.210	0.052	1.590	1.650	0.052	12.492	685.714
		75% KYNAR-97A-1	0.230	0.063	1.437	0.430	0.063	2.687	86.957
		75% KYNAR-97A-2	0.218	0.067	1.281	0.290	0.067	1.794	33.828
98A	8/21/92	LAMINATES Kynar RMI lab book Pg. 185/186							
		400E/3000E used to make laminates	0.250	0.074	1.330	0.310	0.074	1.649	26.000
		75% KYNAR-98A-1	0.220	0.076	1.140	0.320	0.076	1.656	45.459
		75% KYNAR-98A-2	0.225	0.060	1.476	0.510	0.060	3.346	126.667
		75% KYNAR-98A-3	0.185	0.060	1.214	0.330	0.060	2.165	78.378
102A	9/16/92	LAMINATES MICROTHENE RMI lab book Pg. 189							
		350E/3000E used to make laminates	1.550	0.061	10.004	2.400	0.062	15.240	52.341
		80% KYNAR-102A-1	1.150	0.073	6.202	1.630	0.077	8.334	34.376
		80% KYNAR-102A-2							68.396

REVISED- SAMPLE NO.	21-DEC-1992 DATE	STABILITY TESTING							
		MATERIAL COMPOSITION	RESISTANCE (OHM) BEFORE	THICKNESS (INCH) BEFORE	RESISTIVITY (OHM-CM) BEFORE	RESISTANCE (OHM) AFTER	THICKNESS (INCH) AFTER	RESISTIVITY (OHM-CM) AFTER	PERCENT CHANGE (%)
103A	9/23/92	material from 801-102A-4 ALCAN CHEM. LTD. laminates	1.130	0.046	9.671	1.830	0.048	15.010	53.199
103A	9/23/92	WASHED PRECOMPOUNDED C-PLASTIC							
103A-1			0.580	0.080	2.854	1.500	0.080	7.382	150.621
103A-2			0.595	0.083	3.718	6.800	0.083	37.495	900.403
103A-3			0.375	0.050	2.953	2.800	0.050	22.047	646.647
103A-4			0.355	0.040	3.494	12.500	0.040	123.031	3421.327
104A	9/29/92	laminates							
104A-1			0.330	0.047	2.764	8.500	0.047	71.201	2475.750
104A-2			0.440	0.056	2.987	3.200	0.056	21.721	627.273
104A-3			0.310	0.064	1.907	5.200	0.064	31.988	1577.819
104A-4			0.355	0.073	1.915	2.900	0.073	15.640	716.901
104A-5			0.720	0.048	5.906	10.300	0.048	84.482	1330.556
104A-6			0.700	0.062	4.445	5.500	0.062	34.925	605.714
104A-7			0.455	0.066	2.714	5.500	0.066	32.888	1100.791
104A-8			0.540	0.066	3.221	4.300	0.066	25.650	696.296
105A	10/9/92	ETAR (7/92) & MICROTHERM (5/92)							
105A	10/9/92	80% LOADING (5/92)							
105A-1			0.170	0.056	1.195	0.450	0.056	3.164	164.706
105A-2			0.185	0.053	1.374	0.780	0.053	5.794	321.622
105A-3			0.173	0.053	1.285	1.850	0.053	13.742	969.364
105A-4			0.165	0.050	1.299	2.800	0.050	22.047	1506.970
109A	10/26-92	ETAR (7/92) & MICROTHERM (5/92)							
109A	10/26-92	80% LOADING (5/92)							
109A-1			0.290	0.041	2.785	0.870	0.040	8.563	141.667
109A-2			0.360	0.040	3.543	4.000	0.042	412.448	12915.873
109A-3			0.330	0.041	3.169	0.850	0.041	8.162	83.758
109A-4			0.440	0.039	4.442				
110A	10/26-92	LAMINATES							
110A	10/26-92	80% LOADING (5/92) MICRO. (5/92) & ETAR (7/92)							
110A-1			0.225	0.038	2.331	4.900	0.038	50.767	2077.778
110A-2			0.350	0.039	3.933	4.800	0.039	40.455	1271.429
110A-3			0.220	0.042	2.662	1.750	0.042	16.484	695.455
110A-4			0.330	0.041	3.169	0.570	0.041	5.473	72.727
111A	10/29/92	LAMINATES							
111A	10/29/92	400F/3 TONS 400F/3 TONS 400F/3 TONS 400F/3 TONS							

REVISED-	SAMPLE NO.	DATE	MATERIAL COMPOSITION	STABILITY TESTING						PERCENT CHANGE (%)
				RESISTANCE (OHM) BEFORE	THICKNESS (INCH) BEFORE	RESISTIVITY (OHM-CM) BEFORE	RESISTANCE (OHM) AFTER	THICKNESS (INCH) AFTER	RESISTIVITY (OHM-CM) AFTER	
115A	NO HEAT FIRST WASH THEN FILTERED	11/24/92	LAMINATES 80% LANDING (7/92) H2O WASHED 20% MICROTHENE DIFFERENT H2O WASHING TECH. PRECOMPOUNDED C-PLASTIC .25/.07GMS CAPOW L38/H							
	325F/3 TONS		325F/3 TONS							
	115A-1 COARSE-X		115A-1 COARSE-X	0.380	0.073	2.049	0.520	0.072	2.843	30.743
	115A-2 COARSE-X		115A-2 COARSE-X	0.350	0.072	1.914	0.520	0.072	2.843	48.571
	115A-3 MEDIUM-X		115A-3 MEDIUM-X	0.460	0.062	2.921	0.960	0.062	6.096	100.696
	115A-4 MEDIUM-X		115A-4 MEDIUM-X	0.580	0.067	3.400	1.100	0.067	6.934	103.400
116A	HEAT - ALL THREE WASHES THEN FILTERED	11/25/92	LAMINATES 80% LANDING (7/92) H2O WASHED 20% MICROTHENE DIFFERENT H2O WASHING TECH. PRECOMPOUNDED C-PLASTIC .25/.07GMS CAPOW L38/H							
	325F/3 TONS		325F/3 TONS							
	116A-1 COARSE		116A-1 COARSE	0.370	0.077	1.892				
	116A-2 COARSE		116A-2 COARSE	0.300	0.071	1.664				
	116A-3 MEDIUM		116A-3 MEDIUM	0.800	0.067	5.171				
	116A-4 MEDIUM		116A-4 MEDIUM	0.610	0.066	3.639				
117A	CAPOW/MICROTHENE MIXED FIRST THEN INTO	12/03/92	LAMINATES 80% LANDING (7/92) H2O WASHED 20% MICROTHENE PRECOMPOUNDED C-PLASTIC .15% TO .45% CAPOW L38/H							
	325F/3 TONS		325F/3 TONS							
	117-1A (.15%)		117-1A (.15%)	0.380	0.071	2.107	0.980	0.071	5.434	157.895
	117-2A (.20%)		117-2A (.20%)	0.510	0.071	2.820	1.150	0.071	6.377	125.490
	117-3A (.25%)		117-3A (.25%)	0.420	0.068	2.432	0.700	0.066	4.176	71.717
	117-4A (.30%)		117-4A (.30%)	0.560	0.068	3.242	0.900	0.068	5.674	75.000
	117-5A (.35%)		117-5A (.35%)	0.420	0.071	2.320				
	117-6A (.40%)		117-6A (.40%)	0.460	0.065	2.706				
	117-7A (.45%)		117-7A (.45%)	0.640	0.064	3.937				
118A	CAPOW/SB02 MIXED FIRST THEN MICROTHENE	12/07/92	LAMINATES 80% LANDING (7/92) H2O WASHED 20% MICROTHENE							
	325F/3 TONS		325F/3 TONS							
	118A-1 COARSE		118A-1 COARSE	0.380	0.071	2.107	0.980	0.071	5.434	157.895
	118A-2 COARSE		118A-2 COARSE	0.510	0.071	2.820	1.150	0.071	6.377	125.490
	118A-3 MEDIUM		118A-3 MEDIUM	0.420	0.068	2.432	0.700	0.066	4.176	71.717
	118A-4 MEDIUM		118A-4 MEDIUM	0.560	0.068	3.242	0.900	0.068	5.674	75.000

PROCEEDINGS
CHARGE
(3)

LAMINATES
2.600 KETELACK
.370 MICRO (5/92)
325F/15 TONS
0.060" SHIM
122-1A
122-2A

SECRET

REVISED- SAMPLE NO.	DATE	MATERIAL COMPOSITION	STABILITY TESTING						PERCENT CHANGE (%)
			RESISTANCE (OHM) BEFORE	THICKNESS (INCH) BEFORE	RESISTIVITY (OHM-CM) BEFORE	RESISTANCE (OHM) AFTER	THICKNESS (INCH) AFTER	RESISTIVITY (OHM-CM) AFTER	
851	16-FEB-1993	PELLETS PRESSED TO HANDCOMPOUNDED CARBON PLASTIC SHEET	122-3A	0.410	0.050	3.220	1.450	0.051	246.724
			122-4A	0.430	0.052	3.256	0.820	0.053	90.698
123A	01/04/93	LAMINATES							
		80% LOADING							
		(7/92) H2O WASHED							
		20% MICROTHENE							
		HANDCOMPOUNDED							
		G-PLASTIC							
		.30% TO 1.00% CAPOM							
		L38/H							
		325F/3 TONS							
		123-1A (.30%)	0.490	0.080	2.411	0.580	0.080	2.854	18.367
		123-2A (.35%)	0.360	0.081	1.750	0.370	0.081	1.798	2.778
		123-3A (.40%)	0.580	0.080	2.054	0.740	0.081	3.597	26.011
		123-4A (.45%)	0.430	0.081	2.090	0.510	0.080	2.510	20.007
		123-5A (.50%)	0.440	0.078	2.221				
		123-6A (.55%)	0.650	0.078	3.281				
		123-7A (.60%)	0.620	0.076	3.212				
		123-8A (.65%)	0.600	0.076	3.108				
		123-9A (.70%)	0.660	0.078	3.331				
		123-10A (.75%)	0.600	0.076	3.108				
		123-11A (.80%)	0.900	0.080	4.429				
		123-12A (.85%)	0.680	0.080	3.346				
		123-13A (.90%)	0.600	0.072	3.281				
		123-14A (.95%)	0.520	0.075	2.730				
		123-15A (1.00%)	0.540	0.075	2.635				
124A	07-JAN-93	LAMINATES							
		80% LOADING							
		(7/92) H2O WASHED							
		20% MICROTHENE							
		HANDCOMPOUNDED							
		G-PLASTIC							
		1.5% TO 3.0% CAPOM							
		L38/H							
		325F/3 TONS							
		124-1A (1.5%)	0.620	0.075	3.255				
		124-2A (2.0%)	0.980	0.077	5.011				
		124-3A (2.5%)	0.830	0.076	4.300				
		124-4A (3.0%)	0.660	0.077	3.375				
125A	01/12/93	LAMINATES							
		TEMP 230F TO 400F							
		85% PELLETS							
		HANDCOMPOUNDED							
		G-PLASTIC							
		125-1A (300F)	0.410	0.057	2.832				
		125-2A (300F)	0.730	0.057	5.042				
		FROM							

REVISED-	16-FEB-1993	STABILITY TESTING									
SAMPLE NO.	DATE	MATERIAL COMPOSITION	RESISTANCE (OHM) BEFORE	THICKNESS (INCH) BEFORE	RESISTIVITY (OHM-CM) BEFORE	RESISTANCE (OHM) AFTER	THICKNESS (INCH) AFTER	RESISTIVITY (OHM-CM) AFTER	PERCENT CHANGE (%)		
126A											
01/14/93											
80% TO 90% LOADING											
DIFFERENT & (7/92)											
OF .35% CAPOW L30/H											
SAMPLES 162											
.30% CAPOW L30/H											
SAMPLES 3-7											
NEW BAG MICROTHIENE											
(10/92)											
HANDCOMPOUNDED											
C-PLASTIC											
MICROTHIENE (5/92)											
325P/3 TONS											
126-1A (80%)			1.450	0.061	9.350						
126-2A (80%)			2.850	0.061	10.394						
126-3A (85%)			0.32	0.08	1.575	0.4	0.08	1.969	25.000		
126-4A (85%)			0.27	0.076	1.399	0.33	0.076	1.709	22.222		
PRESS UPSIDE DOWN											
275P/3 TONS											
126-6A (82.5%)			1.4	0.073	7.550	1.45	0.073	7.820	3.571		
126-7A (82.5%)			0.52	0.062	3.302	0.53	0.062	3.366	1.923		
PRESS UPSIDE DOWN											
SIDE DOWN											
129A											
01/15/93											
LAMINATES											
85% PELLETS											
14% TO 22%											
KETJENBLACK (9/92)											
325P/3 TONS											
129-1A (15%)			0.54	0.05	4.252						
129-2A (15%)			0.64	0.048	9.249						
129-3A (16%)			0.55	0.049	4.419						
129-4A (16%)			0.56	0.049	4.499						
129-5A (16%)			0.75	0.05	5.906						
129-6A (18%)			0.66	0.049	5.303						
129-7A (22%)			0.63	0.051	4.663						
129-8A (22%)			0.38	0.05	2.992						
130A											
01/19/93											
LAMINATE 325P/3 TONS											
130-1A (18%)			0.46	0.049	3.696	0.71	0.049	5.705	54.340		
130-2A (18%)			0.46	0.044	4.116	0.76	0.045	6.649	61.546		
130-3A (16%)			0.43	0.044	3.848	0.53	0.044	4.742	23.256		
130-4A (16%)			0.49	0.043	4.406	0.64	0.044	5.727	27.644		
131A											
01/27/93											
LAMINATE 325P/3 TONS											
131-1A(3 TONS)			0.58	0.061	3.743						
131-3A(15 TONS)			0.74	0.055	5.297						
131-4A(15 TONS)			0.41	0.052	3.861						
132A											
01/27/93											
85% PELLETS											
(7/92)											
KETJENBLACK(9/92)											
NEW MICRO(10/92)											
325P/3 TONS											
85% PELLETS											
FROM											

REVISED- SAMPLE NO.	16-FEB-1993	DATE	MATERIAL COMPOSITION	STABILITY TESTING						PERCENT CHANGE (%)
				RESISTANCE (OHM) BEFORE	THICKNESS (INCH) BEFORE	RESISTIVITY (OHM-CM) BEFORE	RESISTANCE (OHM) AFTER	THICKNESS (INCH) AFTER	RESISTIVITY (OHM-CM) AFTER	
132A		AND 15 TONS 16% C-PLASTIC	131-5A(3 TONS)	0.70	0.076	4.041				
132A		132A	LAMINATE 325P/3 TONS							
		01/28/93	132-1A(3 TONS)	1.3	0.057	8.979				
		82.5% (7/92)	132-2A(3 TONS)	1.5	0.048	12.303				
		PRESSED AT 3 TONS	132-3A(15 TONS)	0.96	0.05	7.559				
		AND 15 TONS	132-4A(15 TONS)	0.79	0.051	6.099				
133A		16% C-PLASTIC								
133A		133A	LAMINATE 325P/3 TONS							
		01/28/93	133-1A(3 TONS)	0.36	0.069	2.054				
		85% (7/92)	133-2A(3 TONS)	0.32	0.065	1.938				
		PRESSED AT 3 TONS	133-3A(15 TONS)	0.44	0.051	3.397				
		AND 15 TONS	133-4A(15 TONS)	0.5	0.052	3.786				
134A		18% C-PLASTIC								
134A		134A	LAMINATE 325P/3 TONS							
		01/28/93	134-1A(3 TONS)	0.76	0.058	5.159	0.83	0.058	5.634	9.211
		82.5% (7/92)	134-2A(3 TONS)	0.63	0.057	4.351	0.69	0.058	4.684	7.635
		PRESSED AT 3 TONS	134-3A(15 TONS)	0.85	0.049	6.830	0.72	0.049	5.785	-15.284
		AND 15 TONS	134-4A(15 TONS)	0.76	0.051	5.867	0.68	0.05	5.354	-8.737
135A		18% C-PLASTIC								
135A		01/28/93	LAMINATE 325P/3 TONS							
		THIN (85%) AND	135-1A(.010")	0.65	0.029	8.824	0.61	0.03	8.005	-9.282
		C-PLASTIC(18%)	135-2A(.010")	0.6	0.034	6.948	0.57	0.034	6.600	-5.000
		.010" AND .006"	135-3A(.006")	0.56	0.029	7.602	0.51	0.029	6.928	-8.929
		SHIMS	135-4A(.006")	0.62	0.029	8.417	0.72	0.029	9.775	16.129
136A		02/01/93	LAMINATE 325P/3 TONS							
		22% AND 24%	136-1A(22%)	0.39	0.034	4.516				
		C-PLASTIC	136-2A(22%)	0.57	0.033	6.800				
		THIN (85%)	136-3A(24%)	0.34	0.035	3.825				
			136-4A(24%)	0.39	0.034	4.516				
137A		02/03/93	LAMINATE 325P/3 TONS							
		87% (7/92)	137-1A	0.24	0.041	2.305				
		18% AND 22%	137-2A	0.235	0.043	2.152				
		C-PLASTIC	137-3A	0.305	0.042	2.859				
			137-4A	0.215	0.043	1.969				
138A		02/03/93	LAMINATE 325P/3 TONS							
		85% FROM PELLETS	MRP-1	0.48	0.056	3.375	0.59	0.056	4.148	22.917
		FIRST RUN	MRP-2	0.41	0.061	2.646	0.48	0.06	3.150	19.024
		FROM 0.030" THICK	MRP-3	0.49	0.054	3.972	0.53	0.054	3.864	0.163
			MRP-4	0.43	0.058	2.919	0.47	0.058	3.190	9.302

REVISED-	12-MAR-1993	STABILITY TESTING							
SAMPLE NO.	DATE	MATERIAL COMPOSITION	RESISTANCE (OHM) BEFORE	THICKNESS (INCH) BEFORE	RESISTIVITY (OHM-CM) BEFORE	RESISTANCE (OHM) AFTER	THICKNESS (INCH) AFTER	RESISTIVITY (OHM-CM) AFTER	PERCENT CHANGE (%)
139A 02/05/93 LAMINATE 325P/3 TONS									
DUPLICATE OF 135A									
85% (THIN)		139-1A	0.34	0.03	4.462				
101 C-PLASTIC (THIN)		139-2A	0.6	0.032	7.382				
		139-3A	0.43	0.036	4.703				
		139-4A	0.35	0.035	3.937				
139A 02/05/93 LAMINATE 325P/3 TONS									
85% PELLETS FROM NO SHIM		139-1A(18)	0.39	0.022	6.979	4.8	0.022	85.898	1130.769
18 & 228 C-PLASTIC		139-2A(18)	0.36	0.025	5.669	5.8	0.024	95.144	1578.241
PRESSED TWICE		139-3A(228)	0.33	0.026	4.997	1.95	0.026	29.528	490.909
		139-4A(228)	0.4	0.027	5.833	1.65	0.027	24.059	312.500
BR AND R3 02/05/93 LAMINATE 325P/3 TONS									
C-PLASTIC FROM ZMR		BR-1	0.76	0.039	7.672	1.15	0.039	11.609	51.316
		BR-2	0.05	0.039	8.581	1.15	0.039	11.609	35.294
1.3 AND 5% STEARIC ACID		R3-1	0.47	0.042	4.406	0.81	0.042	7.593	73.340
163 PLUS 24% C-PLASTIC		R3-2	0.51	0.049	4.098	1.1	0.049	8.838	115.606
3 PIECE LAMINATE									
142A 02/11/93 LAMINATE 325P/3 TONS									
85% STEARIC ACID		142-1A(18)	0.135	0.033	1.611	24.5	0.035	275.591	17011.111
		142-2A(18)	0.12	0.033	1.432	38	0.032	467.520	32556.250
		142-3A(38)	0.115	0.033	1.372				
		142-4A(38)	0.102	0.033	1.217				
PELLETS(2/9)		142-5A(58)	0.115	0.032	1.425				
		142-6A(58)	0.14	0.033	1.678				
143A 02/12/93 LAMINATE 325P/3 TONS									
3.6 AND 10% PARAFFIN OIL		143-1A(38)	0.102	0.036	1.115	68	0.032	836.614	74900.000
		143-2A(38)	0.102	0.037	1.005	27.5	0.036	300.744	27609.695
		143-3A(68)	0.13	0.036	1.422				
		143-4A(68)	0.115	0.037	1.224				
		143-5A(108)	0.098	0.035	1.102				
		143-6A(108)	0.1	0.035	1.125				
144A 02/12/93 LAMINATE 325P/3 TONS									
30% CAPOW 1% STEARIC ACID 3% PARAFFIN OIL		144-1A	0.170	0.030	2.231	0.270	0.030	3.543	58.824
		144-2A	0.210	0.029	2.851	0.255	0.029	3.462	21.429
		144-3A	0.160	0.028	2.250	0.210	0.028	2.953	31.250
		144-4A	0.190	0.029	2.579	0.215	0.029	2.919	13.158
146A 02/12/93 LAMINATE 325P/3 TONS									
101% PARAFFIN OIL		146-1A(108)	0.16	0.024	2.625				
		146-2A(108)	0.14	0.025	2.205				
		146-3A(38)	0.145	0.025	2.283				
		146-4A(38)	0.155	0.025	2.441				
3% STEARIC ACID		PELLETS(2/9)							
147A 02/16/93 LAMINATE 325P/3 TONS									

REVISED- SAMPLE NO.	12-MAR-1993	DATE	MATERIAL COMPOSITION	STABILITY TESTING						PERCENT CHANGE (%)
				RESISTANCE (OHM)	THICKNESS (INCH)	RESISTIVITY (OHM-IN)	RESISTANCE (OHM)	THICKNESS (INCH)	RESISTIVITY (OHM-IN)	
				BEFORE	BEFORE	BEFORE	AFTER	AFTER	AFTER	
140A		85% SHEET FROM MRP	147-1A	0.66	0.041	6.338	20.5	0.041	196.850	3006.061
		22% C-PLASTIC	147-2A	0.86	0.049	6.910	44.5	0.049	357.545	5074.419
		PELLETS(2/93)	147-3A	0.78	0.043	7.142	3.8	0.043	34.792	367.179
			147-4A	0.44	0.049	3.535	1.1	0.049	8.838	150.000
		02/17/93 LAMINATE 325P/3 TONS								
		86% LOADING	148-1A(3A)	0.105	0.037	1.117				
		3% STEARIC ACID	148-2A(3A)	0.089	0.039	0.898				
		6% PARAFFIN OIL	148-3A(6A)	0.1	0.043	0.916				
		22% C-PLASTIC	148-4A(6A)	0.12	0.049	0.964				
		PELLETS								
149A		02/17/93 LAMINATE 325P/3 TONS								
		86% LOADING	149-1A(3A)	0.275	0.032	3.383				
		3% STEARIC ACID	149-2A(3A)	0.175	0.033	2.088				
		6% PARAFFIN OIL	149-3A(6A)	0.210	0.032	2.504				
		22% C-PLASTIC	149-4A(6A)	0.185	0.033	2.207				
		PELLETS								
150A		02/18/93 LAMINATE 325P/3 TONS								
		85% PELLETS	150-1A	0.350	0.025	5.512				
		FROM 25G SAMPLE	150-2A							
		C-PLASTIC PELLETS								
151A		02/19/93 LAMINATE 325P/3 TONS								
		85% GRID PATTERN	151-1A	0.2	0.039	2.019	0.34	0.042	3.187	57.857
		PRESSED IN HANDCOMPOUNDED	151-2A	0.205	0.043	1.877	0.32	0.045	2.800	49.160
		LAMINATE	151-3A	0.245	0.042	2.297				
		C-PLASTIC PELLETS	151-4A	0.225	0.043	2.060				
152A		02/19/93 LAMINATE 325P/3 TONS								
		85% GRID PATTERN	152-1A	0.380	0.035	4.274	0.960	0.042	8.999	110.526
		PRESSED IN SHEET FROM MRP	152-2A	0.450	0.039	4.543	1.050	0.039	10.600	133.333
		LAMINATE								
153A		02/22/93 LAMINATE 325P/3 TONS								
		82.5% 3% PARAFFIN	153-1A(3A)	0.300	0.018	6.562				
		.30% CAPOW 133 II	153-2A(3A)	0.500	0.026	7.571				
		1% STEARIC ACID	153-3A(1A)	0.530	0.026	8.025				
		.30% CAPOW 133 II	153-4A(1A)	0.400	0.026	6.057				
		C-PLASTIC PELLETS								
154A		02/23/93 LAMINATE 325P/3 TONS								
		85% 1% STEARIC ACID	154-1A(1A)	0.580	0.028	8.155	0.740	0.028	10.405	27.506
		WASHED	154-2A(1A)	0.320	0.027	4.666	0.410	0.027	5.978	20.125

REVISED- SAMPLE NO.	12-MAR-1993 DATE	STABILITY TESTING						PERCENT CHANGE (%)
		MATERIAL COMPOSITION	RESISTANCE (OHM) BEFORE	THICKNESS (INCH) BEFORE	RESISTIVITY (OHM-CM) BEFORE	RESISTANCE (OHM) AFTER	THICKNESS (INCH) AFTER	RESISTIVITY (OHM-CM) AFTER
159A	W/ 1% CAPOW 139/H 3% PARAFFIN	154-3A(3%) 154-4A(3%)	0.400 0.275	0.027 0.027	5.833 4.010	0.570 0.430	0.029 0.029	32.672 45.500
159A	C-PLASTIC PELLETS 02/25/93 LAMINATE 325P/3TOMS							
159A	WASHED WITH HEXANE C-PLASTIC PELLETS NO CAPOW	155-1A 155-2A	0.160 0.155	0.034 0.034	1.853 1.795	9.200 9.400	0.035 0.034	5405.714 5964.516
159A	85% (HEXANE WASH) SAME AS 85% PELLETS FROM SLUGS FROM	156-1A 156-2A	0.550 0.480	0.025 0.026	8.661 7.268			
157A	C-PLASTIC PELLETS 02/26/93 LAMINATE 325P/3TOMS							
157A	85% STEARIC ACID 0.031" SHIMS	157-1A(1%) 157-2A(1%)	0.580 0.235	0.031 0.030	7.366 3.084	0.960 0.420	0.031 0.030	12.192 5.512
159A	3% PARAFFIN OIL C-PLASTIC PELLETS 02/26/93 LAMINATE 325P/3TOMS	157-31(3%) 157-4A(3%)	0.310 0.215	0.031 0.030	3.937 2.822	0.470 0.340	0.031 0.030	5.969 4.462
159A	85% STEARIC ACID 0.031" SHIMS	158-1A(1%) 158-2A(1%)	0.235 0.210	0.039 0.042	2.372 1.969			
159A	3% PARAFFIN OIL C-PLASTIC PELLETS *PARAFFIN LAMINATE STUCK TO GRID	158-3A(3%) 158-4A(3%)	0.230 0.245	0.039 0.039	2.322 2.473			
159A	82.5% LAMINATE FOR 4V BATTERY	159-1A(STABILITY) 159-2A(BATTERY) 159-3A(STABILITY) 159-4A(BATTERY)	0.380 0.400 0.420 0.370	0.073 0.086 0.086 0.087	2.049 1.831 1.923 1.674			
160A	C-PLASTIC PELLETS 03/08/93 LAMINATE 325P/3TOMS							
160A	80% LAMINATE FOR 4V BATTERY	160-1A(STABILITY) 160-2A(STABILITY) 160-3A(BATTERY) 160-4A(BATTERY)	0.740 0.840 0.670 0.570	0.085 0.085 0.084 0.086	3.428 3.891 3.140 2.609			

REVISED-	SAMPLE NO.	DATE	MATERIAL COMPOSITION	STABILITY TESTING						PERCENT CHANGE (%)
				RESISTANCE (OHM) BEFORE	THICKNESS (INCH) BEFORE	RESISTIVITY (OHM-CM) BEFORE	RESISTANCE (OHM) AFTER	THICKNESS (INCH) AFTER	RESISTIVITY (OHM-CM) AFTER	
	161A	03/15/93	LAMINATE 325F/3TORS							
	FIRST RUN OF 85% FROM THE EXTRUDER	85% LOADING .024" ± .005" EXTRUDED IN-HOUSE	161-1A(.050") 161-2A(.050")	0.780 0.620	0.055 0.054	5.583 4.530	1.400	0.056	9.843	76.282
	162A	C-PLASTIC PELLETS 03/17/93	161-3A(.025") 161-4A(.025")	0.580 0.620	0.035 0.037	6.524 6.597	5.400	0.038	55.947	708.048
	MRP SAMPLE RECEIVED 3/17/93	85% SHEET FROM MRP C-PLASTIC PELLETS	162-1A 162-2A 162-3A 162-4A	0.480 0.480 0.480 0.540	0.032 0.036 0.036 0.032	5.906 5.249 5.249 6.644	12.000 1.350	0.033 0.037	143.164 14.365	2324.242 173.649
	163A	03/22/93	LAMINATE 325F/3TORS							
	85% FROM IN-HOUSE EXTRUDER	SAMPLE EXTRUDED 3/18/93	163-1A 163-2A	0.970 0.680	0.037 0.037	10.321 7.236	1100 61.000	0.036 0.036	667.104	9119.771
	164A	C-PLASTIC PELLETS 03/22/93	LAMINATE 325F/3 TORS							
	85% FROM MRP	RECEIVED 3/19/93	164-1A 164-2A	0.550 0.760	0.032 0.033	6.767 9.667	18.500 15.500	0.031 0.033	234.950 184.920	3372.141 1939.874
	165A	C-PLASTIC PELLETS 03/22/93	LAMINATE 325F/3 TORS							
	EXTRUDED 3/22/93	85% LOADING .35% CAPON 100/110/120/125	165-1A 165-2A	0.800 0.700	0.032 0.031	9.843 9.906	0.840 0.700	0.031 0.030	10.668 10.236	8.307 3.333
	166A	C-PLASTIC PELLETS 03/23/93	LAMINATE 325F/3 TORS							
	EXTRUDED 3/23/93	85% LOADING .35% CAPON 100/110/120/125	166-1A 166-2A	0.590 0.700	0.039 0.038	5.956 7.252	0.700 0.740	0.039 0.038	7.066 7.667	18.644 5.714
	167A	C-PLASTIC PELLETS 03/24/93	LAMINATE 325F/3 TORS							
	EXTRUDED 3/24/93	85% LOADING .35% CAPON L38/H 100/105/105/105	167-1A 167-2A	0.710 0.820	0.036 0.039	7.983 8.278				
		C-PLASTIC PELLETS								

REVISED-		STABILITY TESTING									
SAMPLE NO.	DATE	MATERIAL COMPOSITION	RESISTANCE (OHM)		THICKNESS (INCH)		RESISTIVITY (OHM-IN)		THICKNESS (INCH)		PERCENT CHANGE (%)
			BEFORE	AFTER	BEFORE	AFTER	BEFORE	AFTER	BEFORE	AFTER	
EXTRUDED 3/24/93	85% LOADING	168-1A	1.200								
	.35% CAPOW L38/H	100/115/120/125									
	10% PARAFFIN OIL	168-2A	IR too high.								
VARYING TEMP.	C-PLASTIC PELLETS	100/110/120/125	Lamination								
		168-3A	stopped.								
		100/110/115/115									
169A	03/25/93	LAMINATE 325P/3 TONS									
EXTRUDED 3/25/93	85% LOADING	169-1A	0.490	0.546	0.041	0.041	0.530	0.660	0.041	0.338	-2.941
	NO CAPOW/ADDITIVES	169-2A	0.520	4.993	0.041	0.041	8.642	0.860	0.041	8.258	-4.446
HIGH TEMP C-PLASTIC PELLETS											
170A	03/26/93	LAMINATE 325P/3 TONS									
EXTRUDED 3/26/93	85% LOADING	170-1A	0.680	0.041	0.041	0.041	6.530	0.660	0.041	6.338	-2.941
	.35% CAPOW L38/H	170-2A	0.900	0.041	0.041	0.041	8.642	0.860	0.041	8.258	-4.446
HIGH TEMP C-PLASTIC PELLETS											
171A	03/30/93	LAMINATE 325P/3 TONS									
MRP RECEIVED 3/25/93	85% VACUUM PRESSED PELLETS FROM	171-1A	0.350	3.937	0.035	0.035	0.740	0.035	0.035	8.324	111.429
		171-2A	0.680	7.649	0.035	0.035	1.050	0.036	0.036	11.483	50.123
		171-3A	0.520	5.249	0.039	0.039					
		171-4A	0.550	5.698	0.038	0.038					
C-PLASTIC PELLETS											
172A	03/31/93	LAMINATE 325P/3 TONS									
EXTRUDED 3/31/93	85% LOADING	172-1A	0.460	4.024	0.045	0.045	0.530	0.045	0.045	4.637	15.217
	.35% CAPOW L38/H	172-2A	0.560	4.899	0.045	0.045	0.800	0.045	0.045	6.999	42.857
HIGH TEMP C-PLASTIC PELLETS											
173A	04/2/93	LAMINATE 325P/3 TONS									
EXTRUDED 4/2/93	85% LOADING	173-1A	0.340	3.432	0.039	0.039	0.360	0.040	0.040	3.543	3.235
	.35% CAPOW L38/H	173-2A	0.410	4.139	0.039	0.039	0.480	0.041	0.041	4.609	11.362
HIGH TEMP C-PLASTIC PELLETS											
174A	04/3/93	LAMINATE 325P/3 TONS									
EXTRUDED 4/2/93	85% LOADING	174-1A	0.410	4.139	0.039	0.039	0.400	0.040	0.040	3.937	-4.878
	.35% CAPOW L38/H	174-2A	0.425	4.522	0.037	0.037	0.480	0.040	0.040	4.724	4.478
HIGH TEMP C-PLASTIC PELLETS											
175A	04/05/93	LAMINATE 325P/3 TONS									
EXTRUDED	85% LOADING	175-1A(160)	0.550	5.281	0.041	0.041	0.790	0.041	0.041	7.986	43.636

REVISED-	06-MAY-1993	STABILITY TESTING								
SAMPLE NO.	DATE	MATERIAL COMPOSITION	RESISTANCE (OHM) BEFORE	THICKNESS (INCH) BEFORE	RESISTIVITY (OHM-CM) BEFORE	RESISTANCE (OHM) AFTER	THICKNESS (INCH) AFTER	RESISTIVITY (OHM-CM) AFTER	PERCENT CHANGE (%)	
4/5/93	35% CAPOM L38/H 3% STERIC ACID 100/130/150/160 100/140/160/180	175-2A(160)	0.390	0.042	3.556	0.530	0.042	4.968	35.897	
VARY EXTR.		175-3A(180)	0.470	0.043	4.303	0.600	0.043	6.226	44.601	
TEMP		175-4A(180)	0.440	0.042	6.124	0.500	0.043	5.310	28.753	
C-PLASTIC PELLETS										
176A	04/06/93	LAMINATE 325P/3 TONS								
EXTRUDED	85% CAPOM L38/H	176-1A(160)	0.420	0.040	4.134	0.430	0.040	4.232	2.301	
4/5/93	35% CAPOM L38/H 10% PARAFFIN OIL 100/130/150/160 100/140/160/180	176-2A(160)	0.490	0.040	4.823	0.490	0.041	4.705	-2.429	
VARY EXTR.		176-3A(180)	0.380	0.039	3.836	0.390	0.039	3.937	2.632	
TEMP		176-4A(180)	0.350	0.038	3.626	0.360	0.040	3.543	-2.286	
C-PLASTIC PELLETS										
177A	04/12/93	LAMINATE 325P/3 TONS								
EXTRUDED	85% CAPOM L38/H	177-1A(160)	0.610	0.041	5.857	0.530	0.041	5.009	-13.115	
4/5/93	35% CAPOM L38/H 10% PARAFFIN OIL 100/130/150/160 100/140/160/180	177-2A(160)	0.810	0.044	7.200	0.740	0.042	6.937	-4.292	
VARY EXTR.		177-3A(180)	1.050	0.043	9.614	0.770	0.042	7.218	-24.921	
TEMP		177-4A(180)	0.840	0.044	7.516	0.650	0.043	5.951	-20.819	
C-PLASTIC										
178A	04/14/93	LAMINATE 325P/3 TONS								
EXTRUDED	85% CAPOM L38/H	178-1A(160)	0.540	0.046	4.622	0.580	0.046	4.964	7.007	
3/31/93	35% CAPOM L38/H 100/130/150/160 AND	178-2A(160)	0.640	0.047	5.361	0.600	0.045	5.949	10.972	
4/2/93	100/140/160/180	178-3A(180)	0.530	0.045	4.637	0.480	0.045	4.199	-9.434	
IN-HOUSE		178-4A(180)	0.450	0.041	4.321	0.400	0.041	4.609	6.667	
C-PLASTIC										
179A	04/15/93	LAMINATE 325P/3 TONS								
EXTRUDED	85% CAPOM L38/H	179-1A(160)	0.390	0.045	3.412	0.460	0.045	4.024	17.949	
3/31/93	35% CAPOM L38/H 100/130/150/160 AND	179-2A(160)	0.310	0.043	2.838	0.390	0.043	3.571	25.906	
4/2/93	EXTRUDED 3/31/93 100/140/160/180 BEFORE	179-3A(180)	0.280	0.043	2.564	0.340	0.043	3.113	21.429	
LAMINATION	EXTRUDED 4/2/93	179-4A(180)	0.310	0.043	2.838	0.300	0.043	3.479	22.501	
C-PLASTIC										
180A	04/16/93	LAMINATE 325P/3 TONS								
EXTRUDED	85% CAPOM L38/H	180-1A(160)	0.300	0.041	2.881	0.390	0.042	3.656	26.905	
3/31/93	35% CAPOM L38/H 100/130/150/160 AND	180-2A(160)	0.310	0.041	2.881	0.310	0.041	3.212	16.901	
4/2/93	EXTRUDED 3/31/93 100/140/160/180 BEFORE	180-3A(180)	0.280	0.041	2.564	0.340	0.041	3.113	21.429	
LAMINATION	EXTRUDED 4/2/93	180-4A(180)	0.310	0.041	2.838	0.300	0.041	3.479	22.501	

REVISED- SAMPLE NO.	17-MAY-1993 DATE	STABILITY TESTING						PERCENT CHANGE (%)
		MATERIAL COMPOSITION	RESISTANCE (OHM) BEFORE	THICKNESS (INCH) BEFORE	RESISTIVITY (OHM-CM) BEFORE	RESISTANCE (OHM) AFTER	THICKNESS (INCH) AFTER	RESISTIVITY (OHM-CM) AFTER
AND C-PLASTIC BEFORE LAMINATION	100/130/150/160 100/140/160/180 EXTRUDED 4/5/93	180-3A(180) 180-4A(180)	0.310 0.280	0.039 0.037	3.129 2.979	0.390 0.320	0.039 0.039	3.937 3.236
IM-HOUSE EXTRUDED C-PLASTIC	100/140/160/180							
101A	04/28/93 85% CARBON LAMINATE 325P/3 TONS							
SISTER TO 4V BATTERY LAMINATE	10% PARAFFIN OIL .5% CARBON	181-1A(200) 181-2A(200) 181-3A(180) 181-4A(180)	0.470 0.400 0.540 0.550	0.063 0.059 0.064 0.064	2.937 2.669 3.322 3.383	0.580 0.560 0.610 0.580	0.062 0.057 0.064 0.064	3.683 3.868 3.752 3.568
IM-HOUSE EXTRUDED C-PLASTIC	100/140/160/180							
102A	04/28/93 85% CARBON LAMINATE 325P/3 TONS							
4V BATTERIES FOR PASTE ADHESION STUDIES	10% PARAFFIN OIL .5% CARBON	182-1A(200) SAND 182-2A(200) PB THEN SANDED 182-3A(180) SAND 182-4A(180) PB THEN SANDED	0.680 0.700 0.680 0.600	0.073 0.060 0.071 0.058	3.667 4.593 3.771 4.073			
103A	04/29/93 85% CARBON 10% PARAFFIN OIL .35% CARBON HEXANE WASH EXTRUDED 4/29/93	LAMINATE 325P/3 TONS 183-1A 183-2A 183-3A 183-4A	0.380 0.380 0.380 0.380	0.043 0.043 0.059 0.057	3.479 3.479 2.536 2.625	0.540 0.550 0.550 0.580	0.044 0.040 0.059 0.059	4.832 4.531 3.670 3.870
104A	05/04/93 80% CARBON .35% CARBON HANDCOMPOUNDED NOT SANDED	LAMINATE 325P/3 TONS 184-1A 184-2A 184-3A	0.580 0.500 0.580	0.046 0.046 0.050	4.964 4.279 4.567	0.780 0.800 0.760	0.046 0.047 0.051	6.676 6.761 5.867
105A	05/05/93 82.5% CARBON .35% CARBON HANDCOMPOUNDED NOT SANDED	LAMINATE 325P/3 TONS 185-1A 185-2A THICK SUBSTRATE	0.380 0.290	0.055 0.048	2.720 2.379	0.580 0.410	0.055 0.050	4.152 3.228
106A	05/05/93 82.5% CARBON .35% CARBON HANDCOMPOUNDED NOT SANDED	LAMINATE 325P/3 TONS 186-1A 186-2A THICK SUBSTRATE	0.380 0.290	0.055 0.048	2.720 2.379	0.580 0.410	0.055 0.050	4.152 3.228
107A	05/05/93 82.5% CARBON .35% CARBON HANDCOMPOUNDED NOT SANDED	LAMINATE 325P/3 TONS 187-1A 187-2A THICK SUBSTRATE	0.380 0.290	0.055 0.048	2.720 2.379	0.580 0.410	0.055 0.050	4.152 3.228
108A	05/05/93 82.5% CARBON .35% CARBON HANDCOMPOUNDED NOT SANDED	LAMINATE 325P/3 TONS 188-1A 188-2A THICK SUBSTRATE	0.380 0.290	0.055 0.048	2.720 2.379	0.580 0.410	0.055 0.050	4.152 3.228
109A	05/05/93 82.5% CARBON .35% CARBON HANDCOMPOUNDED NOT SANDED	LAMINATE 325P/3 TONS 189-1A 189-2A THICK SUBSTRATE	0.380 0.290	0.055 0.048	2.720 2.379	0.580 0.410	0.055 0.050	4.152 3.228
110A	05/05/93 82.5% CARBON .35% CARBON HANDCOMPOUNDED NOT SANDED	LAMINATE 325P/3 TONS 190-1A 190-2A THICK SUBSTRATE	0.380 0.290	0.055 0.048	2.720 2.379	0.580 0.410	0.055 0.050	4.152 3.228

REVISED-	15-JUL-1993	STABILITY TESTING										
SAMPLE NO.	DATE	MATERIAL COMPOSITION	RESISTANCE (OHM) BEFORE	THICKNESS (INCH) BEFORE	RESISTIVITY (OHM-CM) BEFORE	THICKNESS (INCH) AFTER	RESISTIVITY (OHM-CM) AFTER	PERCENT CHANGE (%)				
85% CAPOW		186-1A	1.000	0.041	9.602	0.042	14.529	51.310				
10% PARAFFIN OIL		186-2A	0.820	0.041	7.874	0.043	11.903	51.163				
NOT SANDED												
19% C-PLASTIC												
85% CAPOW		LAMINATE 330P/2 TONS										
187A		187-1A	0.155	0.030	2.034	0.030	137.795	6674.194				
RIBBON		187-2A	0.135	0.029	1.833	0.029	271.518	14714.815				
REC' 5/17/93		187-3A	0.135	0.029	1.833	0.029	84.171	4692.593				
HAND PRESS		187-4A	0.155	0.030	2.034	0.030	106.299	5125.006				
19% C-PLASTIC												
85% CAPOW		LAMINATE 330P/2 TONS										
188A		188-1A	0.140	0.028	1.969	0.030	78.740	3900.000				
RIBBON		188-2A	0.140	0.031	1.778	0.030	118.110	6542.857				
REC' 5/17/93		188-3A	0.135	0.031	1.715	0.031	102.870	5000.000				
HAND PRESS		188-4A	0.155	0.031	1.969	0.031	121.920	6093.548				
19% C-PLASTIC												
85% CAPOW		LAMINATE 295P/3 TONS										
189A		189-1A	0.970	0.045	8.486							
EXTRUDED												
06/14/93												
19% C-PLASTIC												
85% CAPOW		LAMINATE 295P/3 TONS										
189-1A(SANDED)			0.470	0.051	3.628	0.051	5.404	48.936				
189-4A(SANDED)			0.540	0.045	4.724	0.045	7.262	53.704				
06/16/93												
LAMINATE 295P/3 TONS												
190A		190-1A	0.740	0.086	3.388							
4V BATTERY		190-2A	0.780	0.092	3.338							
SUBSTRATES		190-3A	0.730	0.086	3.342							
		190-4A	0.660	0.086	3.021							
19% C-PLASTIC												
04/2/93												
191A		LAMINATE 295P/3 TONS										
85% CAPOW		191-1A(006)SANDED	0.250	0.044	2.237	0.044	14.316	540.000				
35% CAPOW L38/H		191-2A(006)	0.255	0.045	2.231	0.044	8.679	289.037				
19% C-PLASTIC												
85% CAPOW		LAMINATE 295P/3 TONS										
191-3A(007)SANDED			0.230	0.043	2.106	0.044	4.295	103.953				
191-4A(007)			0.290	0.044	2.595	0.044	5.190	100.000				
06/18/93												
LAMINATE 295P/3 TONS												
192A		192-1A	1.300	0.051	10.036	0.049	28.121	188.220				
80% CAPOW		192-2A	1.900	0.049	15.266	0.050	34.646	126.947				
POWDER												
35% CAPOW/L38H												
HEATED ON V-BLENDED 6/1/93												
NOT PLATE												
19% C-PLASTIC												
85% CAPOW		LAMINATE 295P/3 TONS										
193A												
POWDER		193-1A(SANDED)	0.190	0.062	1.207	0.062	5.207	331.579				
HEATED ON V-BLENDED 6/1/93		193-2A	0.260	0.058	1.765	0.059	14.680	731.812				
NOT PLATE												
19% C-PLASTIC												
85% CAPOW		LAMINATE 295P/3 TONS										
193-1A(SANDED)			0.190	0.062	1.207	0.062	5.207	331.579				
193-2A			0.260	0.058	1.765	0.059	14.680	731.812				

STABILITY TESTING

15-JUL-1993

REVISED-

SAMPLE NO.	DATE	MATERIAL COMPOSITION	RESISTANCE (OHM)	THICKNESS (INCH)	RESISTIVITY (OHM-IN)	THICKNESS (INCH)	RESISTIVITY (OHM-IN)	PERCENT CHANGE (%)
194A	06/24/93	LAMINATE 295P/3 TONS						
	85% RIBBON FROM	194-1A(.008")	0.265	0.031	3.366	0.032	104.577	3007.311
	CRISS-CROSS PRESS OF	194-2A(.008")	0.320	0.042	3.000	0.043		
	.008" AND .010" THICK	194-3A(.010")	0.290	0.040	2.854	0.040		
	.008" AND .010" RIBBON	194-4A(.010")	0.310	0.034	3.590	0.034	509.495	14093.540
195A	06/28/93	LAMINATE 295P/3 TONS						
	19% C-PLASTIC	195-1A	0.460	0.046	3.937	0.046	5.563	41.304
	85% .5% CAPON EXTRUDED 6-16-93	195-2A	0.500	0.046	4.964	0.046	6.162	24.130
196A	06/28/93	LAMINATE 295P/3 TONS						
	19% C-PLASTIC	196-1A	1.150	0.044	10.290	0.045	10.061	-2.222
	80% .35% CAPON EXTRUDED 6-16-93	196-2A	1.050	0.045	9.106	0.045	11.374	23.810
197A	06/29/93	LAMINATE 295P/3 TONS						
	19% C-PLASTIC	197-1A(315P)	0.240	0.044	2.147	0.045	6.124	185.185
	85% .35% CAPON EXTRUDED 6-16-93	197-2A(315P)	0.275	0.045	2.406	0.045	5.687	136.364
	SP006 VARY TEMP COMPOUNDED	197-3A(335P)	0.225	0.045	1.969	0.045	6.387	234.444
		197-4A(335P)	0.360	0.051	2.779	0.051	6.253	125.000
		197-5A(355P)	0.240	0.044	2.147	0.045	3.237	50.741
		197-6A(355P)	0.220	0.045	1.925	0.045	2.406	25.000
		197-7A(375P)	0.235	0.045	2.056	0.045	2.537	23.404
		197-8A(375P)	0.215	0.045	1.881	0.045	2.625	39.535
		197-9A(400P)	0.205	0.045	1.794	0.045	2.406	34.146
		197-10A(400P)	0.200	0.046	1.712	0.045	2.406	40.956
198A	06/29/93	LAMINATE 295P/3 TONS						
	19% C-PLASTIC	198-1A(315P)	0.430	0.049	3.455	0.049	6.910	100.000
	85% .5% CAPON EXTRUDED 6-16-93	198-2A(315P)	0.410	0.050	3.228	0.050	9.843	204.670
	SP007 VARY TEMP COMPOUNDED	198-3A(335P)	0.295	0.047	2.471	0.047	6.069	177.966
		198-4A(335P)	0.320	0.052	2.423	0.052	4.000	68.750
		198-5A(355P)	0.280	0.046	2.396			
		198-6A(355P)	0.230	0.046	1.969			
		198-7A(375P)	0.210	0.047	1.759			
		198-8A(375P)	0.360	0.049	2.892			
		198-9A(400P)	0.245	0.048	2.010			
		198-10A(400P)	0.240	0.045	2.100			

STABILITY TESTING WAS SHORTERED BY ONE DAY ON SAMPLES 1-4

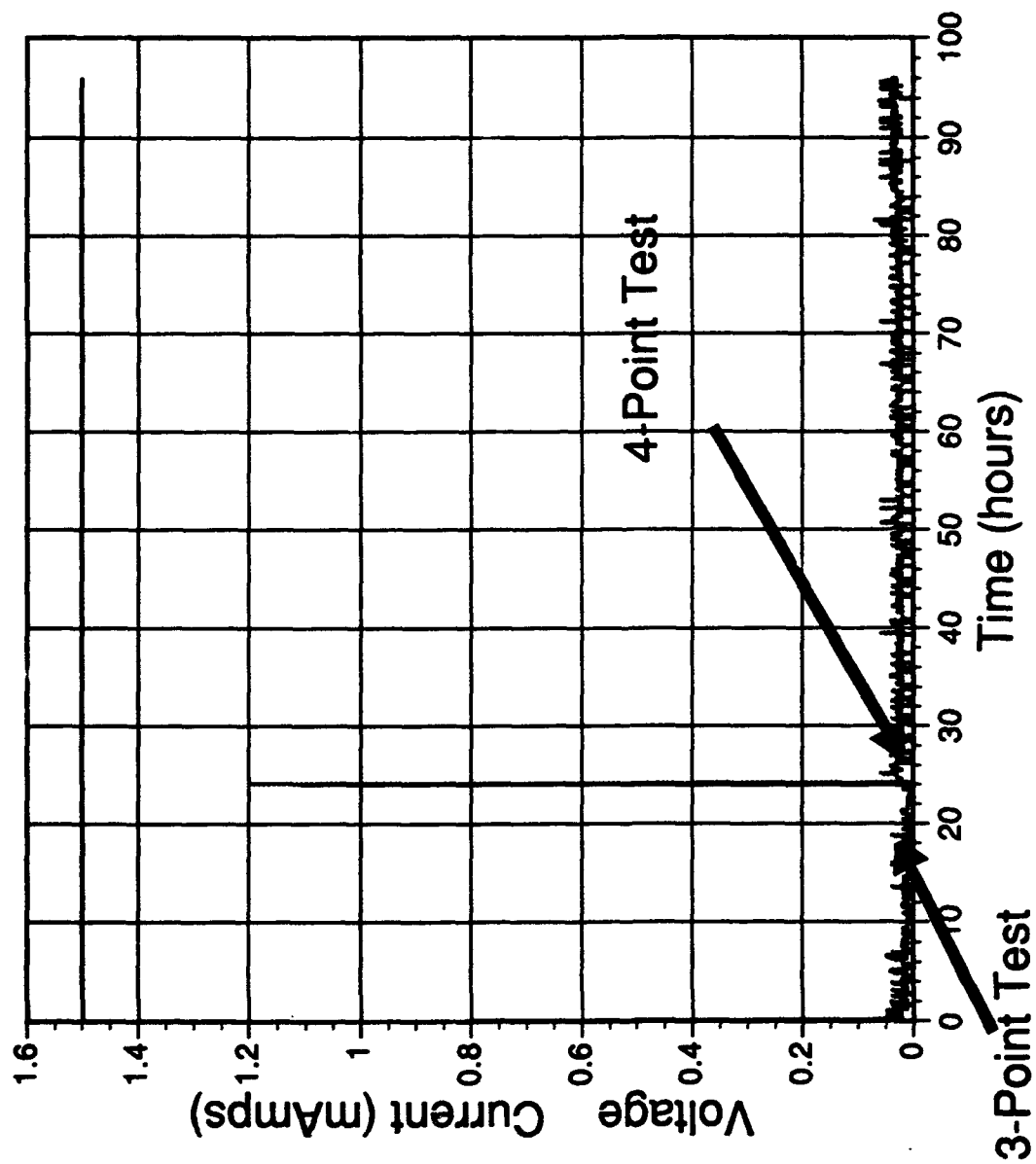


Figure 7
 Sample 96a-1
 0.050" Microthene 80% Loading
 3.2% Resistivity Increase

Stability testing a bipolar substrate and/or conductive filler has been developed at JCBGI over many years. The method used for this contract is the 3 and 4 point tests where the bipolar substrate is exposed to a constant potential of 1.5 volts. The substrate is sandwiched between the two cells and serves as the working electrode. A constant potential of 1.5 volts is applied to the substrate. In the three electrode system where the current is collected at the top of the substrate. After 24 hours the test is switched to a four point test with the current being collected after it passes through the substrate. With the potential remaining at 1.5 volts, the test continues for a minimum of 3 additional days. Ideally, if the substrate was not porous, the current would remain the same for both the 3 and 4 point tests. A rising current during testing would suggest a porous substrate or a non-stable conductive filler.

The second test the substrate must pass is the conductivity test. The conductivity of the part is measured before and after the stability test. If the conductivity increases more than 20%, the accuracy of the conductivity test $\pm 10-20\%$, the part has exhibited either porosity or the conductive filler is not stable. Since, Conduflow has been successfully tested, an increase in resistivity would indicate a porosity problem. The increase in resistance can be explained by the fact that the C-black side would be exposed to the positive potential by the porosity in the Conduflow part, which causes the C-black to oxidize and become non-conductive. In parts where the conductive filler is not stable the increase in resistivity is explained by the non-stability of the filler.

JCBGI's stability testing station consists of one four channel power supply which is controlled using a Macintosh computer and Labview software. The software was written by JCBGI and in conjunction with the computer makes the power supply act like a 4 channel potentiostat. The computer also the data collection device, recording voltage and current acceptance every 30 seconds.

6. BATTERY TEST RESULTS:

Four different four volt batteries have been tested. A summary of the test results and battery information can be found in Figure 8. The batteries are made using lead sheet terminal electrodes and one bipolar substrate. On the bipolar substrate 1g of Pb powder was sintered on to help promote paste adhesion. A number of pasting trials have been conducted and it was found that the PAM would not adhere to the bipolar substrate without some form of Pb adhered to the surface. Parts pasted without any Pb displayed no paste adhesion, the PAM would shear off at the substrate to PAM interface in sheet form.

<u>Battery ID</u>	<u>% Filler</u>	<u>% CA</u>	<u>Thick</u>	<u>Ω-cm</u>	<u>PAM Weight</u>	<u>NAM Weight</u>	<u>Active Area</u>
159	82.5	0.35	0.086"	1.83	34g	34g	12"
160	80.0	0.35	0.087"	3.14	34g	34g	12"

Battery terminals are lead sheets with 12g of active material on each. Separator thickness is 0.030", with the battery being assembled using gasket material and a glue to prohibit leaking. Less than 1g of lead dust was sintered on each bipolar plate to aid in paste adhesion. Acid was introduced via a vacuum fill technique. Formation took place over 48 hours with a total of 79 ah/# PAM (Positive Active Material). Battery 160 developed a short thorough the separator during formation. The separator was replaced and the formation was completed. Discharge data from both batteries follow.

Battery #159

<u>Cycle</u>	<u>Date</u>	<u>IR</u>	<u>Dis I</u>	<u>D Time</u>	<u>% Recharge</u>	<u>Comments</u>
1	3/25	40	10	41s	1120	
2	3/26	40	10	85s	1116	
3	3/29	43	20	31s	643	
4	3/30	51	10	22s	256	Cold Crank
5	3/31	45	10	35s	243	
6	4/1	57	10	21s	1134	
7	4/2	61	10	53s	1103	
8	4/5	76	10	15s	2971	
9	4/6	76	10	65s	596	
10	4/7	88	10	69s	494	
11	4/8	95	10	70s	140	
12	4/9	105	10	24s	1100	
13	4/12	105	10	76s	140	
14	4/13	115	10	70s	140	
15	4/14	110	5	286s	140	
16	4/15	120	5	284s	140	
17	4/16	130	10	89s	140	
18	4/19	130	10	40s	140	
19	4/20	150	10	43s	200	
20	4/21	145	10	60s	200	
21	4/22	150	10	65s	200	
22	4/26	170	10	19s	200	
23	4/27	170	10	89s	200	
24	4/28	170	10	89s	200	
25	4/29	175	10	91s	200	
26	4/30	180	10	99s	300	

Figure 8: Battery Test Data

Battery #160

<u>Cycle</u>	<u>Date</u>	<u>IR</u>	<u>Dis</u> <u>Current</u>	<u>Dis</u> <u>Time</u>	<u>%</u> <u>Recharge</u>	<u>Comments</u>
1	3/26	40	10	150s	553	
2	3/29	49	20	69s	221	
3	3/30	56	10	109s	122	Cold Crank
4	3/31	56	10	123s	127	
5	4/1	80	20	14s	530	
6	4/2	84	10	125s	193	
7	4/5	105	10	108s	167	
8	4/6	115	10	127s	133	
9	4/7	140	10	148s	133	
10	4/8	160	10	158s	140	
11	4/9	170	10	148s	200	
12	4/12	205	10	120s	140	
13	4/13	220	10	135s	140	
14	4/14	220	10	150s	125	
15	4/15	255	10	241s		

Teardown Performed on 4/16/93

Battery #159-B

<u>Cycle</u>	<u>Date</u>	<u>IR</u>	<u>Dis</u> <u>Current</u>	<u>Dis</u> <u>Time</u>	<u>%</u> <u>Recharge</u>	<u>Comments</u>
1	4/8	42	10	214s	271	
2	4/9	42/84	24	2s	388	Cold Crank
3	4/12	42	20	56s	140	
4	4/13	49	24	24s	140	
5	4/14	49	10	67s	200	
6	4/15	55	5	224s	200	
7	4/16	56	10	113s	140	
8	4/19	65	10	57s	140	
9	4/20	72	10	39s	200	
10	4/21	67	10	39s	200	
11	4/22	80	10	38s	200	
12	4/26	96	10	9s		
13	4/27	72	10	132s	200	
14	4/28	74	10	97s	200	
15	4/29	76	10	83s	200	

Teardown Performed on 4/30/93

Figure 8 Cont.

Battery teardowns have determined that failure has been caused by lack of active material adhesion. The PAM would not be adhered to the substrate. This causes excessive resistance between the active material/substrate interface and severely hinders battery performance.

7. NEXT STEPS:

The three areas to be addressed in the remaining months of the contract are- Produce thinner more conductive substrates; Decide on a method to produce cell to cell sealing; Improve positive active material adhesion to the substrate. Of the three issues, the most critical is the adhesion issue. The first issue should, without any material improvements or breakthroughs, evolve simply from process optimization. The sealing issue is not seen as a major problem because of the material used as a base resin, PE, can be easily thermally bonded too. JCBGI has extensive experience in using vibration welding and infrared welding techniques to provide cell to cell sealing for bipolar batteries. These techniques can be easily modified for use in this project. The main development issue will be PAM adhesion to the substrate. Initial battery testing has shown that this has been the performance limiting area on all of the batteries made. Future development efforts will be concentrated on this area.