# Carderock Division, Naval Surface Warfare Center

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# A Contrarotating Propeller Design for a High Speed Patrol Boat with Pod Propulsion


by
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The main text of this report was presented at the FAST '95, Lubeck-Travemunde, Germany, 25-27 September 1995.





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## NOMENCLATURE

| | |
|---|---|
| $c/D$ | Chord to diameter ratio |
| $D_r$ | Rotor diameter |
| $EAR$ | Expanded area ratio |
| $f_m/c$ | Maximum camber to chord length ratio |
| $G$ | Nondimensional circulation |
| $i_T/D$ | Total rake to diameter ratio |
| $J_A$ | Advance coefficient |
| $K_Q$ | Torque coefficient |
| $K_T$ | Thrust coefficient |
| $N_r$ | Rotor RPM |
| $P/D$ | Pitch to diameter ratio |
| $P_D$ | Effective horsepower |
| $P_E$ | Delivered horsepower |
| $Q$ | Torque |
| $R_e$ | Reynolds Number |
| $t$ | Thrust deduction fraction |
| $t/c$ | Thickness to chord ratio |
| $T$ | Thrust |
| $u_{ij}$ | Velocity induced by Z blades of the forward propeller |
| $w_T$ | Taylor wake fraction |
| $X_R$ | Nondimensional radius measured from the shaft axis |
| $Z$ | Blade number |
| $\beta_i$ | Hydrodynamic pitch angle |
| $\delta_k$ | Angle between adjacent blades |
| $\eta_R$ | Relative rotative efficiency |
| $\eta_o$ | Open water efficiency |
| $\eta_D$ | Propulsive efficiency |
| $\theta_s$ | Skew angle |
| $\sigma_i$ | Cavitation index |

All other notation in this report is in accordance with the International Towing Tank Conference (ITTC) Standard Symbols.*

* "International Towing Tank Conference Standard Symbols 1976," The British Ship Research Association, BSRA Technical Memorandum No. 500 (May 1976).

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## ABSTRACT

*A contrarotating (CR) propeller design with a tractor pod for a high speed patrol boat is addressed. In the current arrangement, a CR propeller is placed at the forward end of a pod which is aligned with the local inflow. The powering and cavitation experiments show the performance prediction agree well with measurements. Compared to the existing controllable pitch propeller with shaft and strut configuration, the pod-mounted CR propeller shows a 28 % reduction in power consumption at design speed with a 7 knot improvement in cavitation inception speed. At full power, a larger pod is required, which will reduce the gain in power consumption.*


## ADMINISTRATIVE INFORMATION


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## INTRODUCTION

Over the past several years, there has been a renewed interest in finding more efficient and quieter propulsors for high speed patrol boats. Conventional propulsors mounted on inclined, strut supported shafts are the typical propulsion systems found on present patrol boats. The inclined flow results in the blade angle of attack variation and thus produce early blade cavitation.

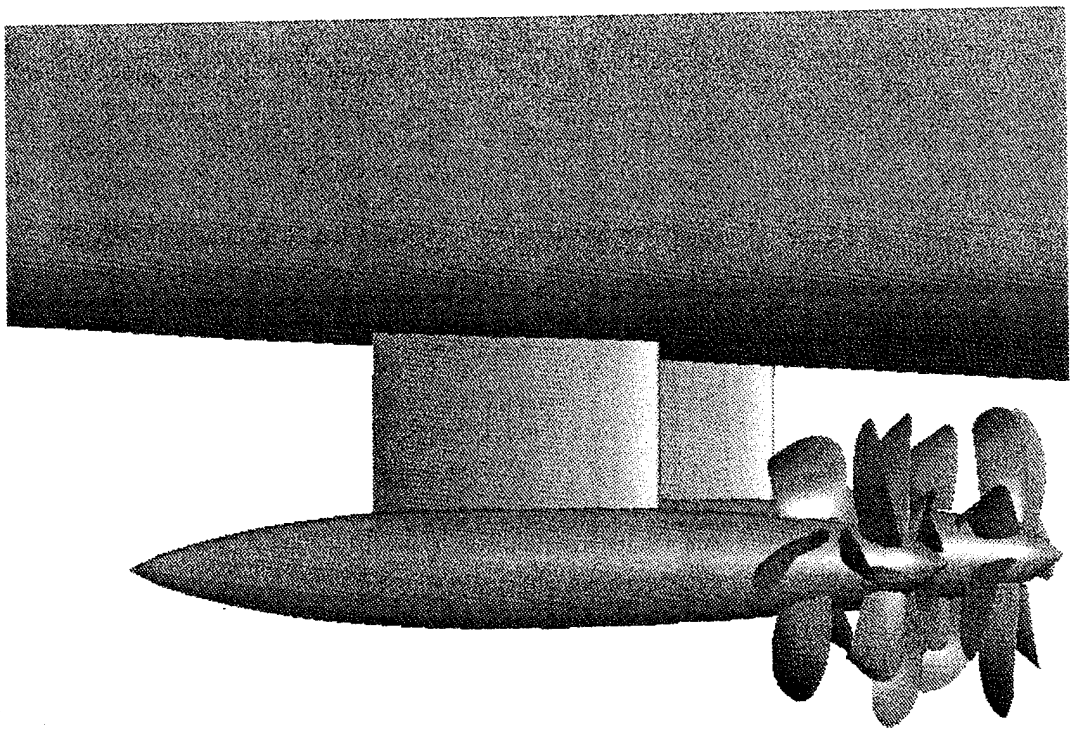

**Fig. 1.** An Arrangement of CR propeller with tractor pod.

In the current arrangement, contrarotating (CR) propellers are mounted at the forward end of a pod, as shown in Figure 1and are powered by an electric motor contained within the pod. The advantages of this arrangement are that the propulsor is placed outside the hull wake and no shaft and strut is forward of the propulsor to produce nonuniformities in the inflow. Elimination of nonuniformities in the inflow results in the propulsor blade sections having a nearly constant angle of attack, which greatly improves the cavitation performance. The CR propellers reduce rotational and axial kinetic energy losses in the propeller slip stream, as compared to single rotation (SR) propellers, and reduce power consumption. Additionally, in the current arrangement, the replacement of shaft and strut with the combination of pod and trim flaps also reduces power consumption at design speed. However, at full power, the benefit of power consumption gain will be reduced as a result of a larger pod to drive the propulsors.

## SUMMARY OF DESIGN METHOD

The design method used for the contrarotating propellers developed by Chen and Reed[1] will be briefly summarized in the following sections. A quasi-steady cavitation prediction method was developed in this report.

## DESIGN PRINCIPLE

The design method for contrarotating propellers is based on basic hydrodynamic principles of conservation of momentum, mass and circulation. To satisfy the conservation of momentum, the net force produced by the contrarotating propellers must overcome bare body drag and drag due to hull-propeller interactions. Conservation of mass is used to determine the circulation of the aft propeller once the circulation of the forward propeller is set. Conservation of circulation is used to determine the magnitude of the aft propeller circulation once the forward propeller circulation is set.

## DESIGN PROCEDURE

The design procedure includes three phases: specification of operating conditions, design, and analysis.

### *Specification of Operating Conditions*

In the first phase, the design requirements and wake survey data are provided. The effects of the hull on the flow and hull-propulsor interaction are traditionally represented by the nominal wake and two interaction coefficients: the thrust deduction factor and the wake fraction.

### *Design*

In the design phase, there are three design stages: preliminary, intermediate, and final.

*Preliminary Design.* Using lifting-line theory, the preliminary design stage employs a parametric study to determine optimum forward and aft propeller diameters, rotation speed, and number of blades. In this stage, the forward and aft propeller circulation distributions are also determined. Propulsive efficiency and cavitation are considered during this stage. The lifting-line theory developed by Kerwin et al.[2] was employed in the current study.

*Intermediate Design.* In the intermediate design stage, cavitation and strength are the major factors guiding the selection of chordlength, thickness, and blade loading distribution for the forward and aft propellers. Consideration is also given to strength requirements and propulsive efficiency which are affected by these parameters. Stress calculations for the forward and aft propellers were performed using a simple beam theory (Schott et al.[3]).

The cavitation prediction method for the forward propeller is the same as for conventional single rotation propellers, since there is no other component ahead of it. The cavitation inception prediction method for the aft propeller developed in this report is a quasi-steady prediction method. It is composed of two steps: inflow calculations and cavitation calculations. This method is considered quasi-steady because the induced velocities from the forward propeller are held steady for one calculation of the cavitation inception on the aft propeller. Then, the forward propeller is rotated $\Delta\theta$ and another calculation of the cavitation inception on the aft propeller is performed.

The calculation of the total inflow wake starts with the calculation of the induced velocities from the forward propeller on the aft propeller using a modified version of the lifting surface program developed by Chen and Reed[4]. These calculated induced velocities, though based only on theoretical prediction, show the same trends as induced velocities measured in experiments by Jessup[5]. The velocity field of a Z-bladed propeller is obtained by shifting the induced velocities from the key blade through one blade interval and superimposing the result Z times. Thus

$$u_{ij}(\theta) = \sum_{k=1}^{Z} u_{ij}^{*}(\theta + \delta_k) \quad , \tag{1}$$

where

$$\delta_k = \frac{2\pi(k-1)}{Z} \qquad k = 1,2,\ldots,Z \, . \tag{2}$$

$u_{ij}^{*}$ is the velocity induced by one blade and $u_{ij}$ is the velocity induced by Z blades of the forward propeller. $\delta_k$ is the angle between adjacent blades of the forward propeller. The addition of the induced velocities to the incoming wake is the total inflow wake to the aft propeller. For contrarotating propellers an additional step is required. Since the aft propeller rotates in the opposite direction of the forward propeller, the induced velocities must undergo a mirror image transformation before its addition to the incoming wake. The mirror image was done by the following procedure:

$$\theta = 360^{o} - \theta \tag{3}$$

$$u_{tij} = -u_{tij} \, . \tag{4}$$

The second step of the quasi-steady cavitation inception prediction method uses a two-dimensional airfoil theory developed by Brockett[6] to compute the blade surface cavitation using the total inflow wake described earlier. To simulate the rotation of the forward component, the key blade of the forward propeller is rotated in intervals of $\Delta\theta$ and cavitation analysis is performed after each rotation. The quasi-steady method is completed when the key blade is rotated so that it passes the location of the first blade.

The tip vortex cavitation index[7] was calculated as follows:

$$\sigma_i = K\left(\frac{G}{c/D}\right)^2_{0.9R} \left(\mathrm{Re}_{0.9R}\right)^{0.4} \quad , \qquad (5)$$

where K is an empirical coefficient determined from full scale experiments.

*Final Design.* The final design stage consists of using lifting-surface theory to incorporate three-dimensional effects in the design. The final pitch and camber distributions are determined using the CR lifting-surface program developed by Chen and Reed[4]. This program is a modified version of Wang's[8] SR lifting-surface program which includes hub effects.

*Analysis*

In the analysis phase, steady and unsteady forces and moments are computed. The inverse lifting-surface theory developed by Greeley and Kerwin[9] was employed.

## PODDED CONTRAROTATING PROPELLER DESIGN

### BOAT INFORMATION

The high speed patrol boat is a round bilge planing hull craft with a length of 154 ft (46.94 m) and displacement of 260 tons (264.2 tonnes). The existing hull has a diesel/gas turbine twin screw propulsion system. A controllable pitch propeller with shaft and strut system was mounted on each shaft.

In the current study, the shaft and strut system has been replaced by a podded system which is powered by an electrical motor contained within the pod. The pod length is 20 ft (6.10 m) with a length to diameter ratio of 7. Because of the replacement of the shaft and strut system, trim flaps and a transom extension were designed. The trim flaps decreased the boat resistance significantly, but the transom extension increased the resistance slightly. Compared to the existing shaft and strut system, the total resistance reduced substantially with a podded system at design speed.

However, the pod used in this study is not large enough to accommodate machinery to drive the propulsors at full power. A larger pod is required to house the machinery to go full power. It is expected that the increased resistance due to the larger pod will significantly reduce the gain in power consumption.

### DESIGN REQUIREMENTS

The podded CR propeller was designed at the operating point for a high speed patrol boat. The boat speed was chosen at 20 knots (10.3 m/s). The thrust loading coefficient, $C_{Th}$, is 0.280. The forward propeller diameter is 7.56 ft (2.30 m) and rotational speed is 117 rpm. The blade numbers of the forward and aft propellers are 7 and 5, respectively. The boat's full power condition is at a shaft horsepower of 2,970 Hp (2,216 KW) per pod (twin pods) and a rotational speed of 174 rpm.

### WAKE SURVEY

The resistance and stock powering tests and wake survey were performed at the David Taylor Model Basin (DTMB) towing tank. At the design boat speed of 20 knots, the effective horsepower is 1,550 hp. The thrust deduction and wake fraction are 0.885 and 1.0, respectively from the model propulsion test.

## PARAMETRIC STUDY

The design parameters for the present study were chosen based on a parametric study. The aft propeller diameter was determined through mass conservation and showed an optimum diameter of 95 % of the forward propeller diameter. In order to make certain that the tip vortex of the forward propeller does not impinge upon the aft propeller, the final aft propeller diameter, which is 85 % of the forward propeller diameter, was chosen to be slightly smaller than the preliminary diameter computed using mass conservation. The axial spacing between the proximate propellers was chosen to be one quarter of the forward propeller diameter. A summary of the design parameters for the podded CR propeller is given in Table 1.

**Table 1.** Podded CR propeller design—Summary.

| | Forward Propeller | Aft Propeller |
|---|---|---|
| Boat Speed (knots) | 20 | 20 |
| Rotational Speed (rpm) | 117 | 117 |
| Thrust Loading Coefficient | 0.2800 | |
| Geometry | | |
| Diameter (ft) | 7.56 | 6.43 |
| Number of blades | 7 | 5 |
| Expanded Area Ratio | 0.501 | 0.563 |
| Skew (deg) | 25 | 25 |
| Total Rake | 0 | 0 |
| Blade Sections | * | * |
| Axial Spacing (ft) | 1.89 | |

* NACA 66 (TMB Modified) Thickness, NACA a = 0.8 Meanline

PRELIMINARY DESIGN

The preliminary design consists of lifting-line calculations. The optimum circulation distributions and the unloaded circulation distributions for the forward and aft propellers are shown in Figure 2. For both the forward and aft propellers, the hub and the tip were unloaded; the loading was shifted inboard. The advantages of unloading the blade root and tip are to help delay blade hub and tip vortex cavitation inception and to reduce the tendency toward cavitation erosion near the blade root and tip.

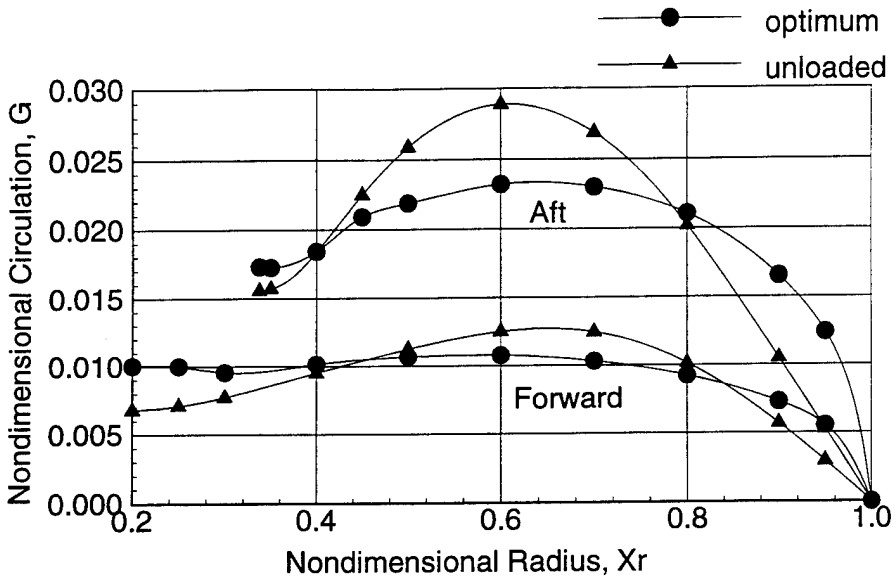


**Fig. 2.** Optimum and unloaded circulation distributions for forward and aft propellers.

The following guidelines for unloading the hub were employed: (1) The net circulation at the root is zero to minimize the hub vortex strength; (2) The slope of the circulation at the root is almost zero to minimize the trailing edge vortex sheet. The circulation distribution was constrained to keep the lift coefficient below 0.5. This constraint limited the amount of unloading at the tip since the increased loading inboard brought the lift coefficient to the 0.5 limit. The same guidelines were used in determining the circulation distributions of both propellers. Due to these constraints, there was a 5% loss in efficiency between the optimum and unloaded cases.

The thickness and chordlength distributions, shown in the Appendix, were determined from the strength analysis and cavitation performance predictions. When the final thickness and chordlength distributions were determined during the intermediate design phase, the lifting-line calculations were redone with the new geometry.

Based on the unsteady force calculation, a skew distribution, shown in the Appendix, was chosen for the forward and aft propellers to minimize unsteady forces. A nonlinear skew distribution with 25 degrees tip skew was selected. Zero total rake was used in this design.

## INTERMEDIATE DESIGN

The intermediate design phase includes strength analysis and cavitation performance predictions. The strength requirement for the propellers was 12,500 psi maximum stress for nickel aluminum bronze material at the full power condition. The stress distribution is shown in the Appendix.

The blade surface cavitation analysis of the forward propeller was straightforward since the inflow wake was nearly uniform and there were no components in front of the forward propeller. The cavitation analysis for the aft propeller was significantly more complicated because of the effect of the forward propeller wake on the aft propeller. The quasi-steady analysis method described previously was used for the aft propeller.

The calculation for tip vortex cavitation was specified in Eq. 5. Hub vortex cavitation prevention was also addressed in the preliminary design phase. The circulations at the roots of the forward and after propeller blades were chosen to be equal in magnitude and opposite in direction so that the net circulation from both propellers was zero. This procedure attempts to minimize the hub vortex strength. Also, the spanwise gradient of circulation at the root was chosen to be essentially zero for each propeller to inhibit trailing vortex sheet formation.

## FINAL DESIGN

The final design phase includes the lifting-surface design calculations. Both forward and aft propellers had a NACA a = 0.80 meanline chordwise loading distribution from cavitation and viscous flow points of view. A modified NACA 66 thickness distribution was selected for the present application. Figures 3 and 4 show the respective final pitch and camber distributions for the forward and aft propellers. A solid model rendering of the podded CR propellers is generated on a CAD system and is shown in Figure 5. The final geometry specifications are shown in the Appendix.

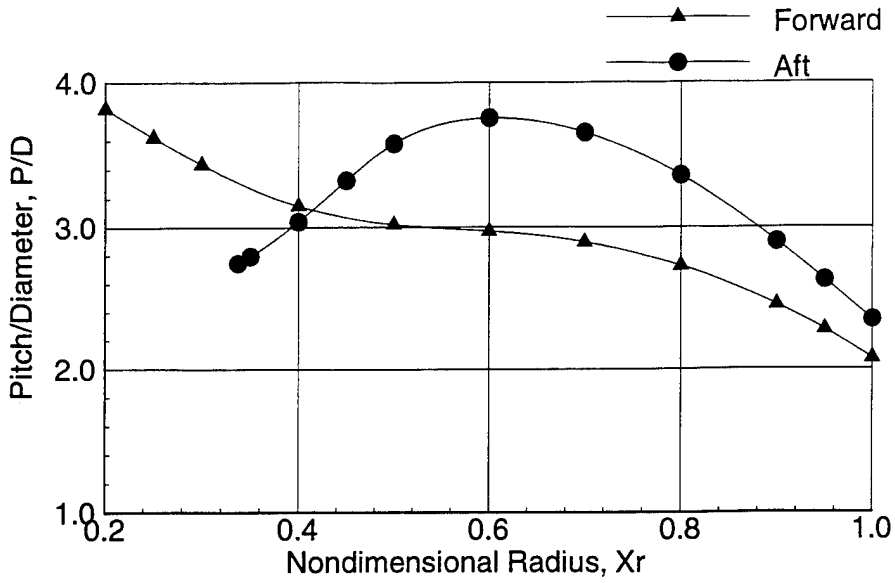


**Fig. 3.** Pitch distribution for forward and aft propellers.

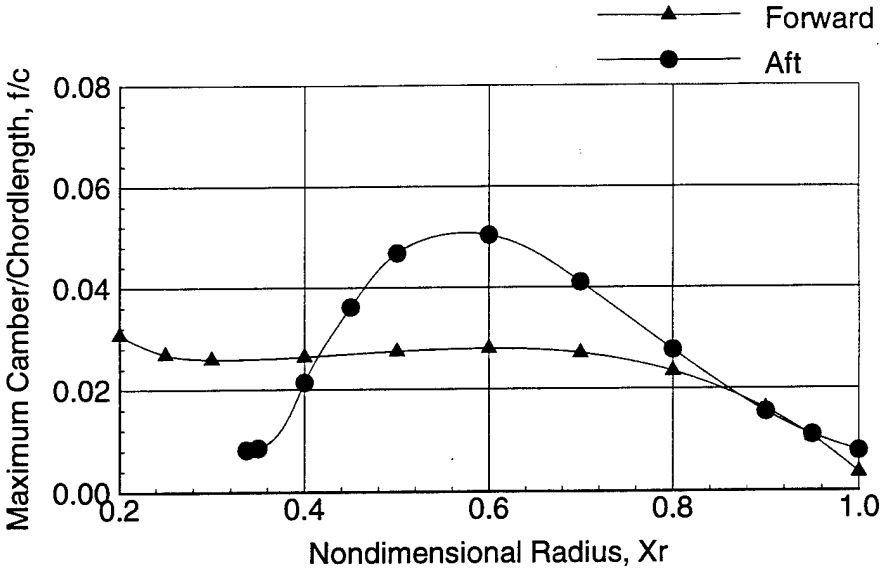


**Fig. 4.** Camber distribution for forward and aft propellers.

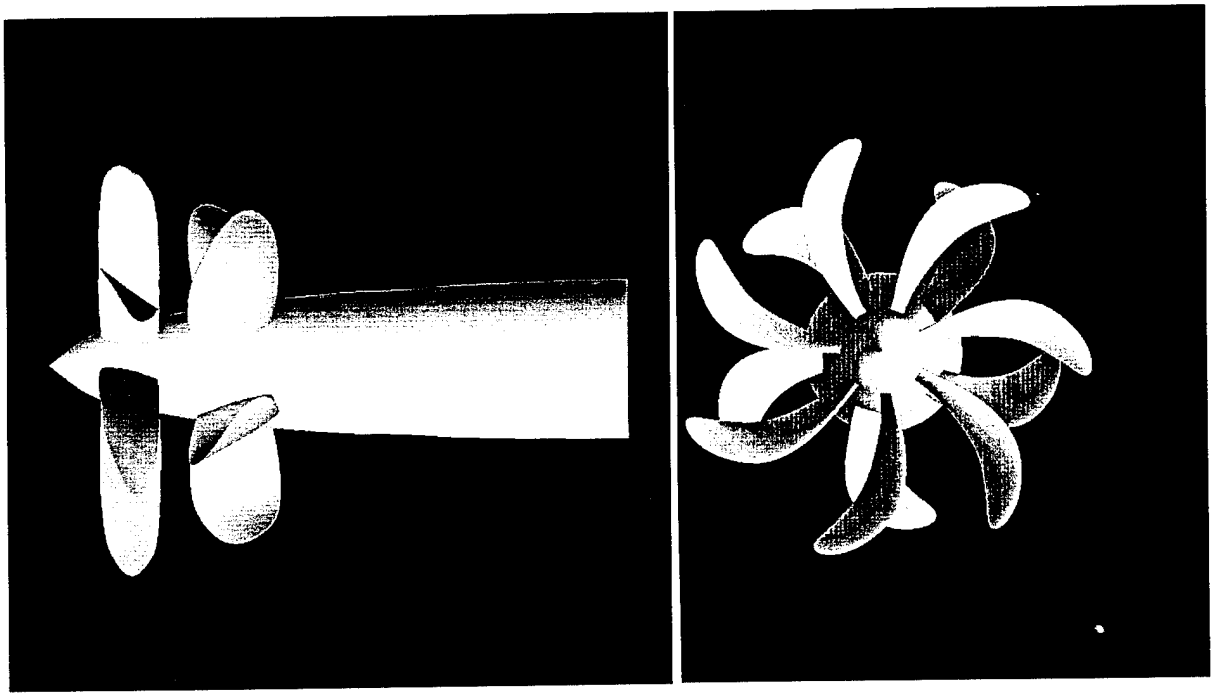

**Fig. 5.** Side and forward views of podded CR propellers.

## PERFORMANCE PREDICTIONS AND EXPERIMENTAL RESULTS

Aluminum models of the forward and aft propellers were manufactured based on the final design geometry. The model propellers 5112 and 5114 represent the forward propellers and 5113 and 5115 represent aft propellers.

### SELF-PROPULSION TESTS

Self-propulsion tests were conducted in the David Taylor Model Basin (DTMB) towing tank. The boat model 5365-A used for the resistance and self-propulsion tests of the podded CR propellers had previously been used for the resistance and self-propulsion tests of the SR propeller. Figure 6(a) shows the measured delivered power as a function of boat speed and compares the predicted value with the measured. Figure 6(b) shows the measured rotational speed as a function of boat speed and compares the predicted value with the measured. The predicted and measured self-propulsion performance at the design boat speed of the podded CR propeller is shown in Table 2.

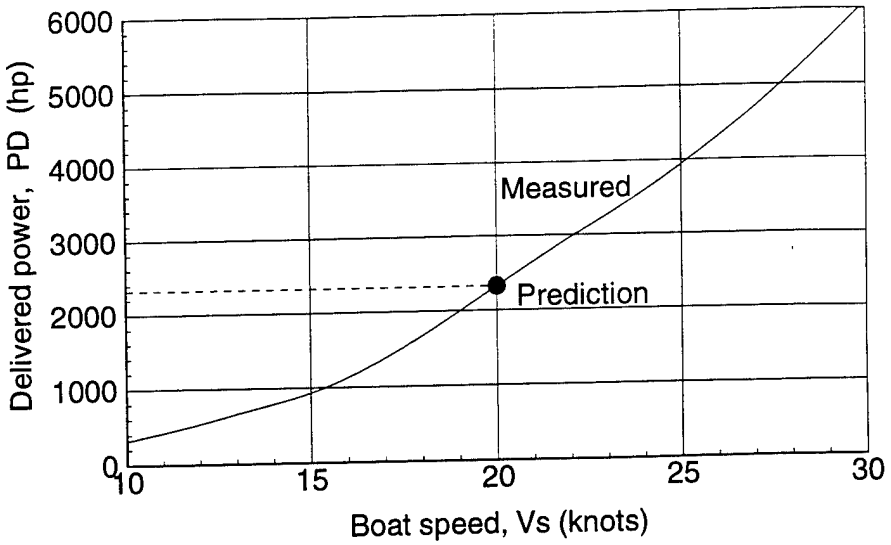


**Fig. 6a.** Delivered Power.

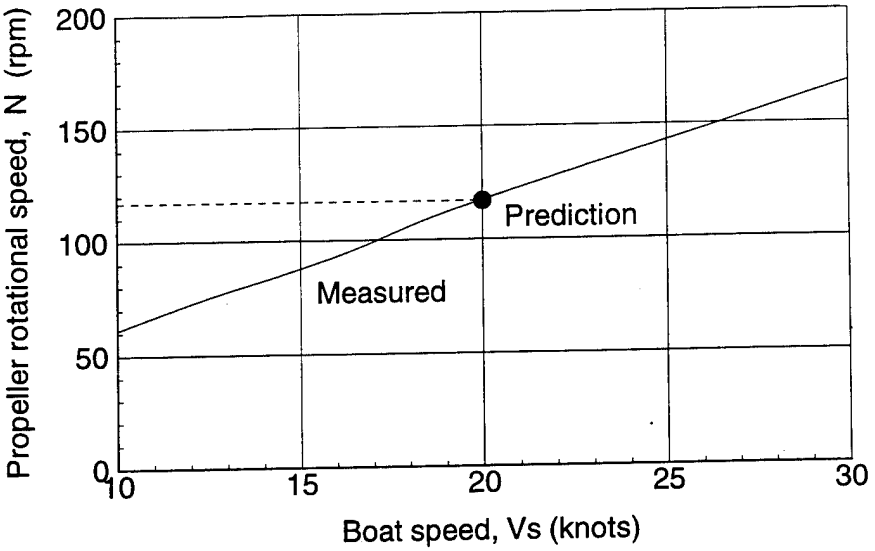


**Fig. 6b**. Rotation speed.

**Fig. 6.** Predicted and measured self propulsion test results.

**Table 2.** Predicted and measured powering performance of podded CR propeller design represented as a unit.

| | Design | Self-propulsion Experiment |
|---|---|---|
| $V_S$ (knots) | 20 | 20 |
| $P_E$ (hp) | 1,550 (-4.3%) | 1,620 |
| $P_D$ (hp) | 2,323 ( 0.1%) | 2,320 |
| 1-t | 0.885 (-2.7%) | 0.910 |
| 1-$w_T$ | 1.000 (-2.0%) | 1.020 |
| $J_A$ | 2.290 (-1.5%) | 2.325 |
| N (rpm) | 117( -0.3%) | 117.4 |
| $K_T$ | 0.580 ( -0.3%) | 0.582 |
| $K_Q$ | 0.279( 1.0%) | 0.276 |
| $\eta R$ | 1.010 (2.0%) | 0.990 |
| $\eta O$ | 0.756(-4.9%) | 0.795 |
| $\eta D$ | 0.670 (-4.3%) | 0.700 |

The effective horsepower (EHP) used in the design is 4.3% lower than the measurement. Modifications made to the test boat most likely account for the higher experimental EHP. The predicted delivered power is almost identical to the measured value. The estimated thrust deduction and wake fraction for the design are 2.7% and 2.0% lower than the measurement. Therefore, the actual drag due to the presence of the propeller was lower than the drag used in the design. This miscalculation probably resulted in lessening the difference between the measured and estimated EHP.

The design advance coefficient is 1.5 % lower than the measurement,and the predicted rotation speed is almost equivalent to the measurement. The predicted thrust is almost identical, and the predicted torque is 1.0 % higher than the measurement. The predicted relative rotative efficiency is 2.0 % higher than the measurement. The predicted open water efficiency and propulsive efficiency are 4.9% and 4.3 % lower than the measurement. These discrepancies primarily result from the difference between the design and the measured effective horsepower. In general, the accuracy of the experimental measurements with the CR propeller is $\pm 2$ % on thrust and torque. Overall the predicted values agree well with the experimental measurements, and in general are within the accepted accuracy of the experimental measurements.

## CAVITATION TESTS

Cavitation tests were performed at the Applied Research Laboratory/Pennsylvania State University (ARL/PSU) 48 inch water tunnel. Table 3 shows the cavitation inception speed difference between the design and the measured values (designed value - measured value). There were no predicted leading edge suction side (LESS), leading edge pressure side (LEPS), pressure side back bubble (PSBB), and pressure side tip vortex (PSTV) cavitation for the forward and aft propellers as the measurements showed. The predicted suction side back bubble (SSBB) cavitation for the forward and aft propellers were 3 knots and 1 knot higher than the measurement. Calculated suction side tip vortex (SSTV) for the aft propeller came in 2 knots higher than the measurement, and there was no predicted and measured SSTV for the forward propeller. The good agreement between the experimental data and the predicted cavitation inception speeds gives us confidence in the quasi-steady cavitation prediction method developed for this design.

**Table 3.** Comparisons of predicted and measured cavitation inception speeds of podded CR propeller design.

| | Forward propeller | Aft Propeller |
|---|---|---|
| SSBB (knots) | 3 | 1 |
| SSTV (knots) | -- | 2 |

Note: Inception speed comparisons are
designed value-measured value
SSBB - suction side back bubble cavitation
SSTV - suction side tip vortex cavitation

## CONCLUSIONS AND RECOMMENDATIONS

The following conclusions can be drawn from the present study.

- Self-propulsion and cavitation experiments show that the performance predictions agree well with the experimental measurements.

- The pod-mounted CR propeller design achieves the design goals of reducing power consumption and increasing cavitation inception speed with no degradation in overall performance. Compared to the existing controllable pitch propeller with shaft and strut configuration, the pod-mounted CR propeller shows a 28 % reduction in power consumption with a 7 knot improvement in cavitation inception speed. Rotational energy recovery of the CR propeller results in energy saving of 10 %. The larger CR propeller diameter contributes an additional 4 % energy saving over the SR propeller. At design speed, the combined effect of pod and trim flaps results in 14 % energy saving due to removing shaft and strut. However, a larger pod is needed to obtain full power and will reduce the improvement in energy saving.

A steering pod which can be adjusted with the inflow direction was not considered in the current study but has been recommended for future work. This arrangement improves cavitation performance during maneuvering.

## ACKNOWLEDGMENT

The authors would like to thank Mr. Ken Remmers and Mr. Bill Cave of DTMB for carrying out the open-water and self-propulsion experiments. The authors would also like to thank Mr. J. Oefelein of ARL/PSU for conducting the cavitation measurements.

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## APPENDIX A

## DETAILED DESIGN INFORMATION

This section provides additional information for the original report. Based on cavitation, flow separation, and efficiency considerations, the chord-length distributions of the forward and aft propellers were chosen. The chord distributions for the forward and aft propellers are shown in Figure A.1.

The thickness distribution was selected based on strength and cavitation considerations. Figure A.2 shows the thickness distributions for the forward and aft propellers.

As shown in Figure A.3, a tip skew distribution of 25 degrees, varying nonlinearly from zero at the hub, was selected for the forward and aft propellers. The total rake for both forward and aft propellers was zero. Therefore, they have negative rake to offset the skew-induced rake. The stress distributions computed by beam theory corresponding to these choices of geometry are given in Figure A.4.

The final geometric specifications of the podded CR propulsor, including the details of the leading and trailing edges, were computed using the computer code, XYZ-PROP, developed by Brockett[10]. All the input data: chord length, thickness, skew, pitch, and camber distributions were faired by a cubic spline procedure before being input to XYZ-PROP. A list of the chord length, thickness, skew, pitch and camber distributions are given in Table A.1.

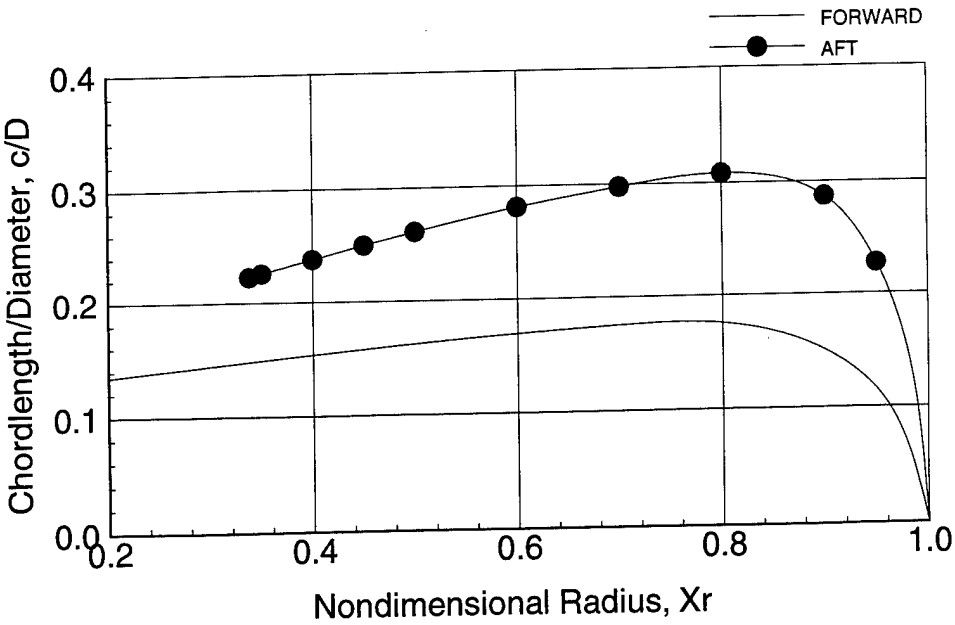


**Fig. A.1.** Chordlength distribution for forward and aft propellers.

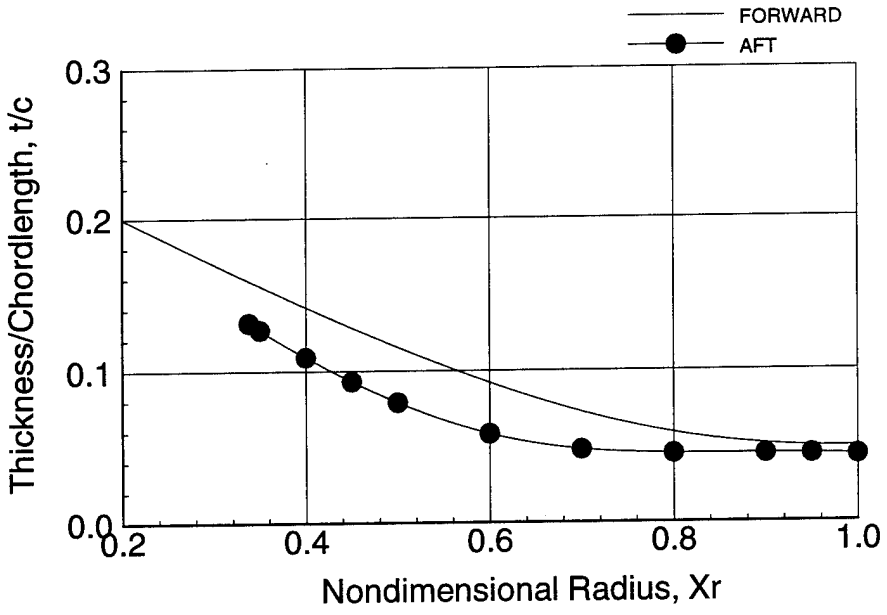


**Fig. A.2.** Thickness distribution for forward and aft propellers.

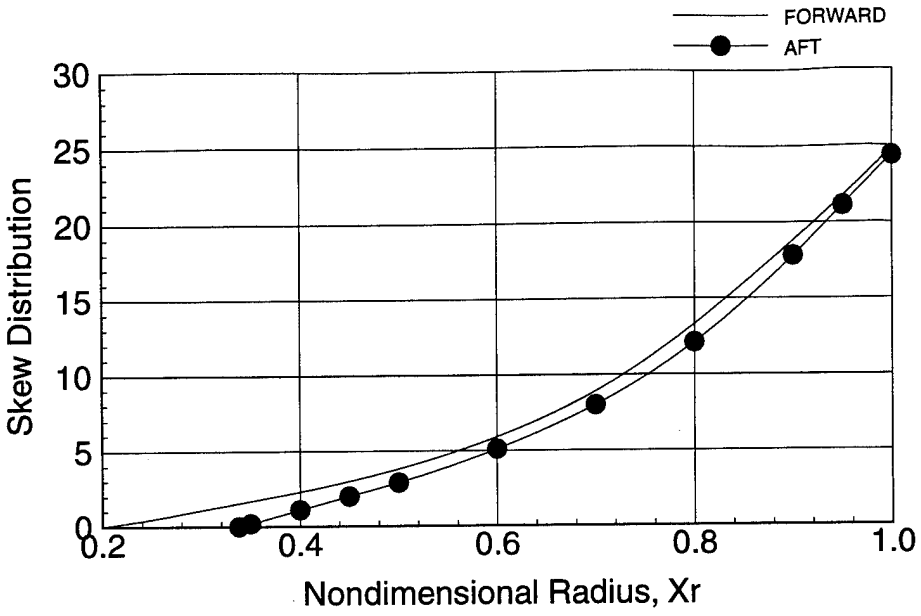


**Fig. A.3.** Skew distribution for forward and aft propellers.

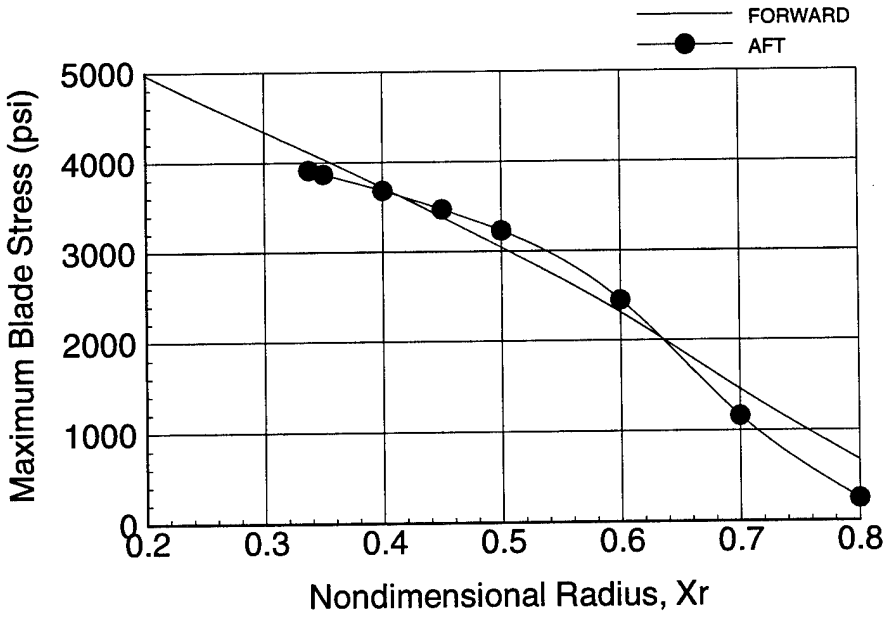


**Fig. A.4.** Maximum blade stress distribution for forward and aft propellers.

**Table A.1.** Final design geometry for CR propulsor.

**Table A.1a.** Forward propeller.

| r/R | c/D | P/D | $i_T/D$ | $\theta_s$ | t/c | $f_m/c$ |
|---|---|---|---|---|---|---|
| 0.140 | 0.1296 | 4.064 | 0.0 | -0.6 | 0.2182 | 0.03764 |
| 0.200 | 0.1350 | 3.820 | 0.0 | 0.0 | 0.2000 | 0.03070 |
| 0.250 | 0.1396 | 3.623 | 0.0 | 0.5 | 0.1849 | 0.02690 |
| 0.300 | 0.1442 | 3.438 | 0.0 | 1.1 | 0.1700 | 0.02595 |
| 0.400 | 0.1533 | 3.150 | 0.0 | 2.3 | 0.1417 | 0.02640 |
| 0.500 | 0.1619 | 3.019 | 0.0 | 3.8 | 0.1155 | 0.02745 |
| 0.600 | 0.1692 | 2.971 | 0.0 | 5.9 | 0.0921 | 0.02795 |
| 0.700 | 0.1749 | 2.895 | 0.0 | 8.9 | 0.0727 | 0.02705 |
| 0.800 | 0.1754 | 2.727 | 0.0 | 13.3 | 0.0590 | 0.02340 |
| 0.900 | 0.1499 | 2.455 | 0.0 | 18.8 | 0.0514 | 0.01600 |
| 0.950 | 0.1152 | 2.279 | 0.0 | 21.7 | 0.0498 | 0.01050 |
| 1.000 | 0.0000 | 2.074 | 0.0 | 24.8 | 0.0497 | 0.00350 |

**Table A.1b.** Aft propeller.

| r/R | c/D | P/D | $i_T/D$ | $\theta_s$ | t/c | $f_m/c$ |
|---|---|---|---|---|---|---|
| 0.338 | 0.2230 | 2.750 | 0.0 | 0.0 | 0.1317 | 0.00821 |
| 0.350 | 0.2260 | 2.800 | 0.0 | 0.2 | 0.1270 | 0.00860 |
| 0.400 | 0.2380 | 3.040 | 0.0 | 1.1 | 0.1091 | 0.02138 |
| 0.450 | 0.2500 | 3.325 | 0.0 | 2.0 | 0.0932 | 0.03618 |
| 0.500 | 0.2613 | 3.581 | 0.0 | 2.9 | 0.0793 | 0.04682 |
| 0.600 | 0.2813 | 3.755 | 0.0 | 5.1 | 0.0587 | 0.05038 |
| 0.700 | 0.2976 | 3.655 | 0.0 | 8.0 | 0.0482 | 0.04103 |
| 0.800 | 0.3082 | 3.360 | 0.0 | 12.1 | 0.0453 | 0.02767 |
| 0.900 | 0.2872 | 2.899 | 0.0 | 17.8 | 0.0450 | 0.01545 |
| 0.950 | 0.2274 | 2.630 | 0.0 | 21.1 | 0.0448 | 0.01092 |
| 1.000 | 0.0000 | 2.350 | 0.0 | 24.4 | 0.0440 | 0.00776 |

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