



Characterization of Raman Spectral Changes in Energetic Materials and Propellants During Heating

by Nicholas F. Fell, Jr., John A. Vanderhoff,
Rose A. Pesce-Rodriguez, and Kevin L. McNesby

ARL-TR-1743

August 1998

19980925 005

Approved for public release; distribution is unlimited.

DTIC QUALITY INSPECTED

The findings in this report are not to be construed as an official Department of the Army position unless so designated by other authorized documents.

Citation of manufacturer's or trade names does not constitute an official endorsement or approval of the use thereof.

Destroy this report when it is no longer needed. Do not return it to the originator.

Army Research Laboratory

Aberdeen Proving Ground, MD 21005-5066

ARL-TR-1743

August 1998

Characterization of Raman Spectral Changes in Energetic Materials and Propellants During Heating

Nicholas F. Fell, Jr., John A. Vanderhoff, Rose A. Pesce-Rodriguez,
Kevin L. McNesby
Weapons and Materials Research Directorate, ARL

Abstract

Raman spectroscopy has been shown to be a useful tool for characterizing neat crystalline explosive samples and for identifying principle components in many propellant and explosive formulations. Herein, we report recent measurements of Raman spectra of explosives and propellant formulations during bulk heating and recent measurements of laser heating of the samples during measurement of Raman spectra. The results of these measurements are important to investigators using Raman spectroscopy to measure vibrational spectra of burning propellant samples.

Acknowledgments

The authors would like to thank Dr. Jeffrey Morris (U.S. Army Research Laboratory [ARL]) for providing the RDX crystals used in this study, Dr. Richard Kranze for assistance in propellant sample preparation, and Drs. Betsy Rice and Cary Chabalowski for the normal-mode analysis of RDX. This work was performed while Dr. Nicholas F. Fell, Jr., held a National Research Council (NRC) ARL Research Associateship.

INTENTIONALLY LEFT BLANK.

Table of Contents

	<u>Page</u>
Acknowledgments	iii
List of Figures	vii
1. Introduction	1
2. Experimental	2
3. Results	4
4. Discussion	9
4.1 Laser Heating	10
4.2 Laser Damage	11
4.3 Increased Rayleigh Scatter	12
4.4 Density Changes	12
4.5 Population Change	12
4.6 Changes in Refractive Index	14
4.7 Premelt-Phase Transition	14
4.8 Fluorescence	14
4.9 Incandescence	14
5. Conclusions	16
6. References	19
Distribution List	21
Report Documentation Page	29

INTENTIONALLY LEFT BLANK.

List of Figures

<u>Figure</u>		<u>Page</u>
1.	Heatable Aluminum Sample Cell	3
2.	Raman Spectrum of RDX Crystal at 298 K	5
3.	Raman Spectra as a Function of Temperature	6
4.	Raman Spectra as a Function of Temperature: (a) XM39 Propellant and (b) C4 Explosive, NIR FTR System	7
5.	Raman Peak Shifts With Heating for RDX Crystal	8
6.	Raman Spectra of Naphthalene at Elevated Temperatures	13
7.	Raman Spectra of Naphthalene From 313 K to 373 K	15

INTENTIONALLY LEFT BLANK.

1. Introduction

Raman spectroscopy has been demonstrated to be useful for the identification of energetic materials and the characterization of propellants [1–3]. Since the Raman spectrum of most solid materials exhibits less spectral overlap than for the corresponding infrared (IR) absorption spectrum, Raman spectroscopy is the method of choice for the characterization of structural changes in these materials during slow heating to the melting point. In addition, the lower sensitivity of Raman spectroscopy, relative to IR absorption spectroscopy, minimizes interference from gas-phase species generated during thermal decomposition. For RDX (cyclo-1,3,5-trimethylene-2,4,6-trinitramine) and RDX-based propellants, thermal decomposition begins below the melting point [4].

Raman spectra yield information about the lattice and internal vibrational modes of the sample. Changes in the Raman shifts of peaks measured during heating of the sample can be used to determine the effects of heating on the lattice and internal molecular vibrational parameters. For example, in a molecular crystal such as RDX, a Raman shift of a peak moving to lower frequency with heating may indicate that the vibrational motion giving rise to this spectral feature is less hindered sterically because of increases in the crystal lattice spacings.

In this paper, we describe the effects of slow, step-wise heating on Raman spectra of a crystal of RDX, a slice of XM39 gun propellant (76% RDX, 24% plasticizers and binders), and a small amount of the explosive C4 (94% RDX, 6% hydrocarbons) over the temperature range from 300 K to 453 K. The energetic materials literature contains many studies of heating to high temperatures, designed to simulate the conditions experienced during combustion events ranging from atmospheric pressure deflagration to weapons firing and explosion [5–12].

Typically, measurements of vibrational spectra of energetic materials during heating are undertaken to obtain experimental confirmation of the initial reactions and reaction products hypothesized to occur during thermal decomposition. We have chosen to investigate the changes in vibrational spectra, in both the peak positions and intensities, at temperatures below the

decomposition point to obtain insight into physical changes occurring prior to thermal decomposition.

2. Experimental

The Raman spectra for this study were measured using three different systems. System one measured spectra using a Bomem DA-8.02 Fourier transform spectrometer with a backscattering Raman accessory. The incident, unpolarized (1,064 nm, 9,394 cm^{-1}) radiation was provided by a Quantronix Series 100 continuous wave Nd: YAG laser. Two holographic, Rayleigh rejection filters (Kaiser Optical) were placed in the optical train to reduce the amount of Rayleigh scattered radiation reaching the liquid-N₂-cooled InGaAs detector. The detector sensitivity range was from 6,500 cm^{-1} to 10,000 cm^{-1} . Because of using this system, we were unable to measure spectral features with a Raman shift greater than approximately 2,900 cm^{-1} . All spectra were collected using 400 mW of incident laser power for 256 scans at 4- cm^{-1} resolution.

System two used a visible laser with an output wavelength of 785 nm as the scattering source (SDL, Inc.). Laser radiation was delivered to and collected from the sample using a lens-coupled fiber optic probe, filtered after scattering using a holographic rejection filter (Kaiser Optical), spectrally resolved using a monochromator (ISA, Inc.), and detected using a charge coupled device (CCD) camera (Princeton Instruments). Effective resolution was approximately 8 cm^{-1} . System three used a visible laser (Lexel Model 95 Argon ion laser) with an output wavelength of 514.5 nm. A linear diode array detector (EG&G Model 1455 intensified linear diode array) was used to detect the Raman-shifted radiation. The diode array detector consisted of 1,024 elements, of which the center 700 were intensified. A holographic Rayleigh rejection filter (Kaiser Optical) was also used with this system. The Raman scattered light was collected in a backscattering geometry. All spectra collected with this system were the result of averaging 100 one-second integrations on the array using 500 mW of incident, unpolarized 514.5-nm radiation as the scattering source. The use of a 150- μ entrance slit on the monochromator resulted in a spectral resolution of 12–18 cm^{-1} . The diode

array was calibrated for Raman shift from previously reported values for RDX at room temperature [13, 14]. All spectra displayed here were used without further correction.

For the experiments reported here using systems one and three, samples were placed in a heated aluminum cell, which was designed and built in-house. The cell (Figure 1) was conductively heated using a resistive cartridge heater embedded in the cell body. A Pt/Pt-10% Rh thermocouple was used to monitor the temperature and provided feedback to one of the two controllers used in these experiments (Omega Model CN2002 K-A-AT or Valco Instrument Co., Inc. Model ITC).

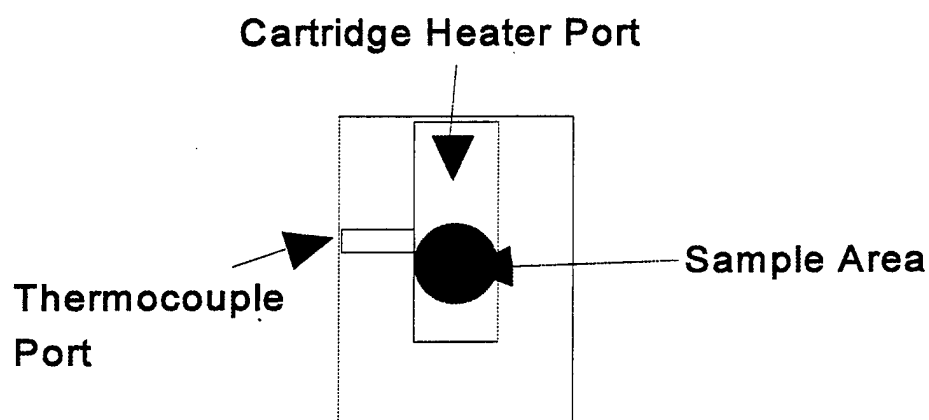


Figure 1. Heatable Aluminum Sample Cell.

The energetic materials examined included an RDX crystal (the RDX crystals were recrystallized from an acetone/ethanol solution), a thin slice of XM39 (a solid gun propellant), and a small (several milligram) sample of C4 explosive. XM39 contains 76% RDX, 12% cellulose acetate butyrate, 4% nitrocellulose, 7.6% acetyl triethyl citrate, and 0.4% ethyl centralite. C4 is a plastic explosive composed of 94% RDX and 6% oil/polymer binder. All energetic materials and compositions used in this study were obtained from in-house sources.

The experimental procedure was as follows. The sample was placed in the heatable cell. The heatable cell position was optimized by maximizing the Raman signal at room temperature. The sample cell temperature was then increased in steps of 10–20 K up to the onset of sample melting

(usually evidenced by large increases in detector output voltage, since RDX melts with decomposition and heat release swamps the detector). After the temperature equilibrated at each step value (~60-s soak), a Raman spectrum was collected. The process was continued until the sample melted or began to decompose prior to melting (usually evidenced by discoloration).

3. Results

A room temperature Raman spectrum (100 cm^{-1} to $1,600\text{ cm}^{-1}$) of a single crystal of RDX (typical size $2 \times 3 \times 1\text{ mm}$), with general band assignments, is shown in Figure 2. The inset shows the chemical structure of a single molecule of RDX. A normal-mode analysis [15] employing the density functional theory on an isolated RDX molecule yields frequencies in close agreement with the values observed in the room-temperature spectrum of the crystal, indicating that over the Raman-shift frequency range studied here, measured spectral features are attributable to internal molecular vibrations. Raman spectra of an RDX crystal, RDX powder, XM39, and C4, measured as a function of temperature, are shown in Figures 3 and 4. From these figures, two effects may be noticed. First, as temperature is increased, the intensity of Raman-shifted radiation decreases. Second, less apparent, is the dependence on temperature of the Raman shift of some spectral features.

Figure 5 shows the changes in Raman shifts of individual peaks as a function of temperature for a single crystal of RDX. As has been shown previously [1, 2], Raman spectra of nitramine-based energetic materials are dominated by features due to molecular vibrations of the crystalline components. Therefore, measured peak shifts are similar in all four samples; so, only changes for the RDX crystal are shown here. The Raman peak observed to change frequency with increasing temperature to an extent significantly greater than others is the peak near a shift of 150 cm^{-1} . This peak was not predicted by a normal-mode analysis of a single RDX molecule [15]. Because of the significant temperature dependence of the Raman shift and because this vibrational frequency was not predicted by the single RDX molecule normal-mode analysis, we believe this peak is due to a lattice vibration or combination of lower frequency internal and lattice vibrations. Significant temperature dependence of lattice-type modes in RDX have been measured by previous

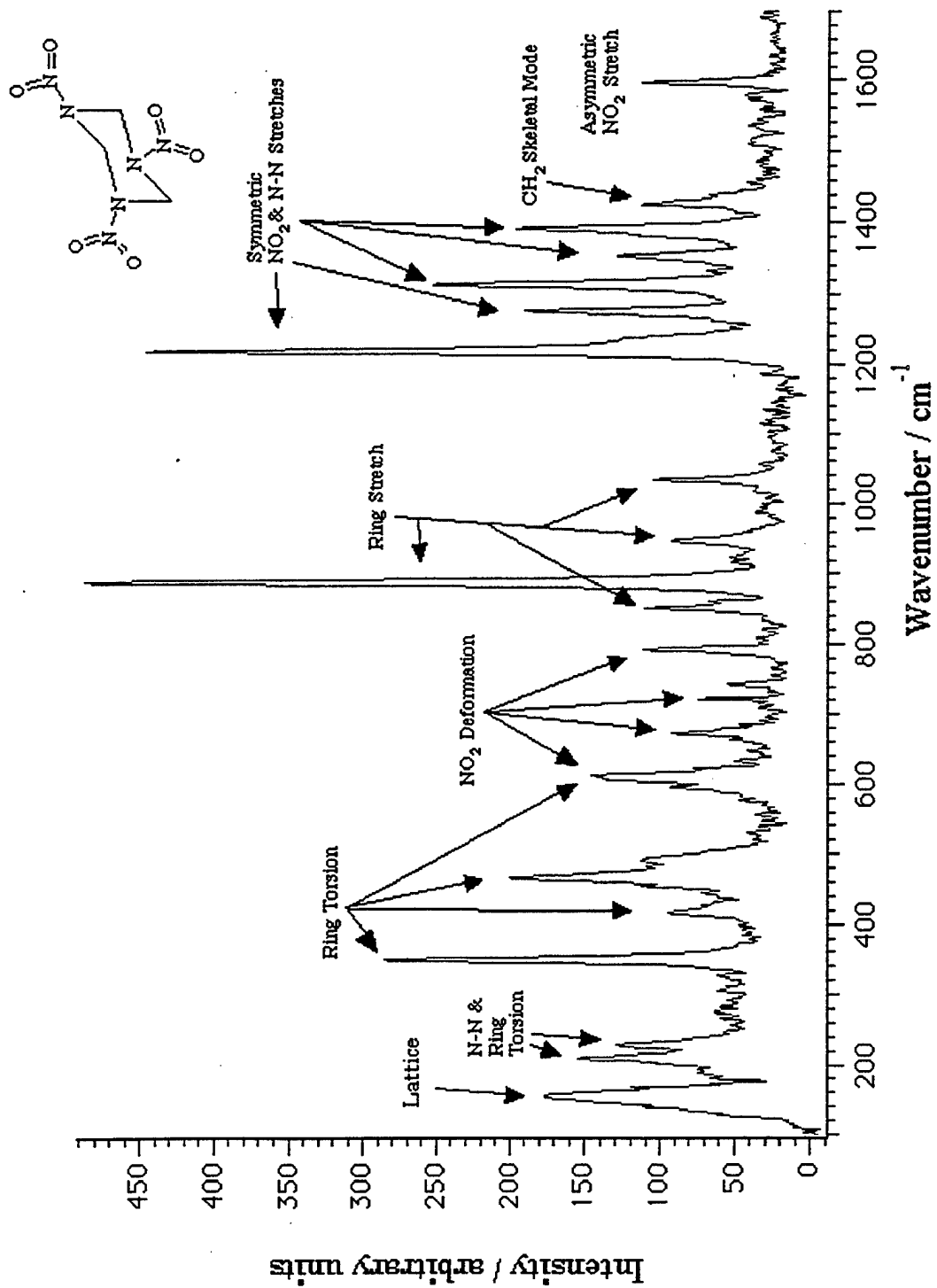


Figure 2. Raman Spectrum of RDX Crystal at 298 K. Inset: Structure of RDX Molecule.

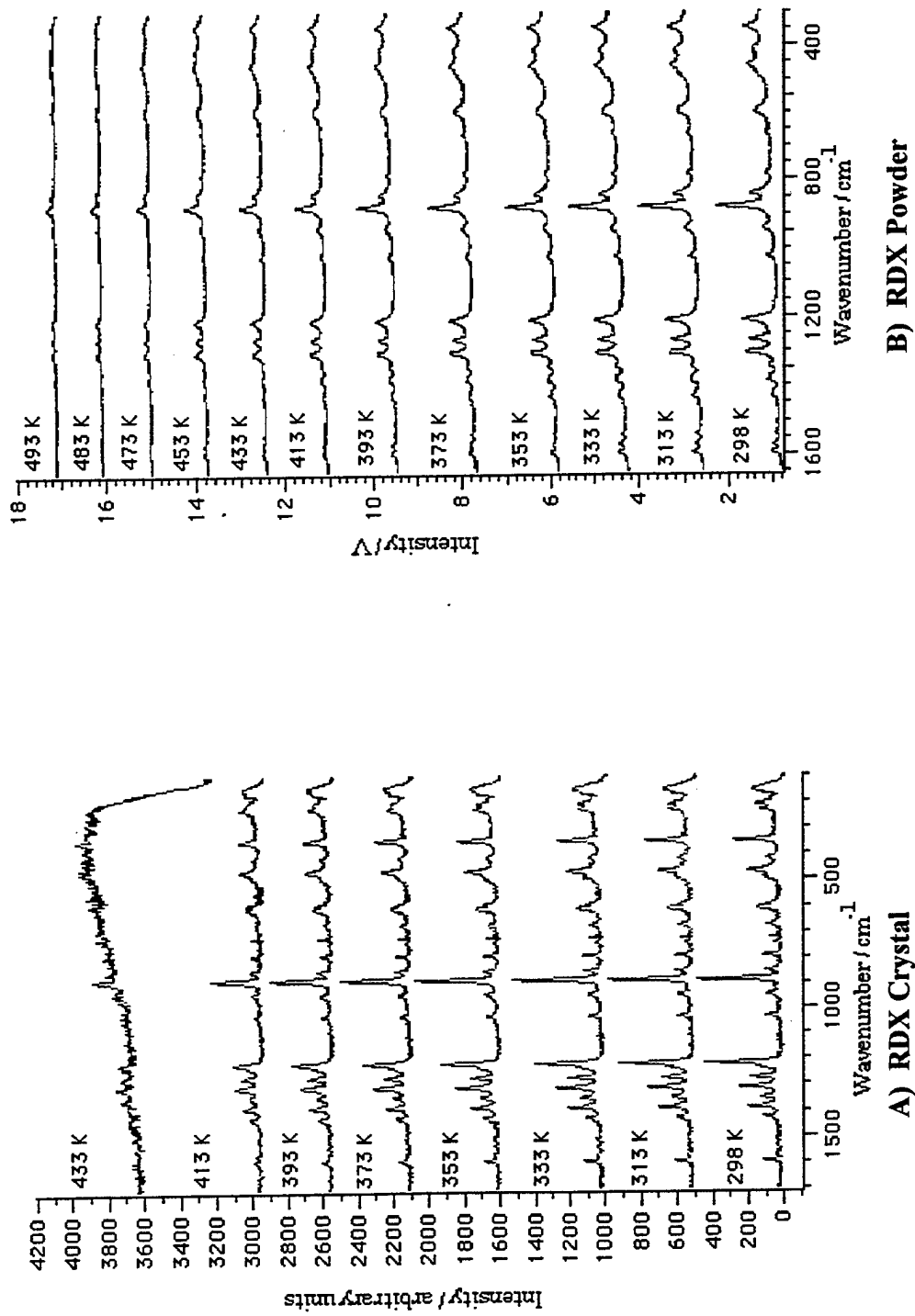


Figure 3. Raman Spectra as a Function of Temperature.

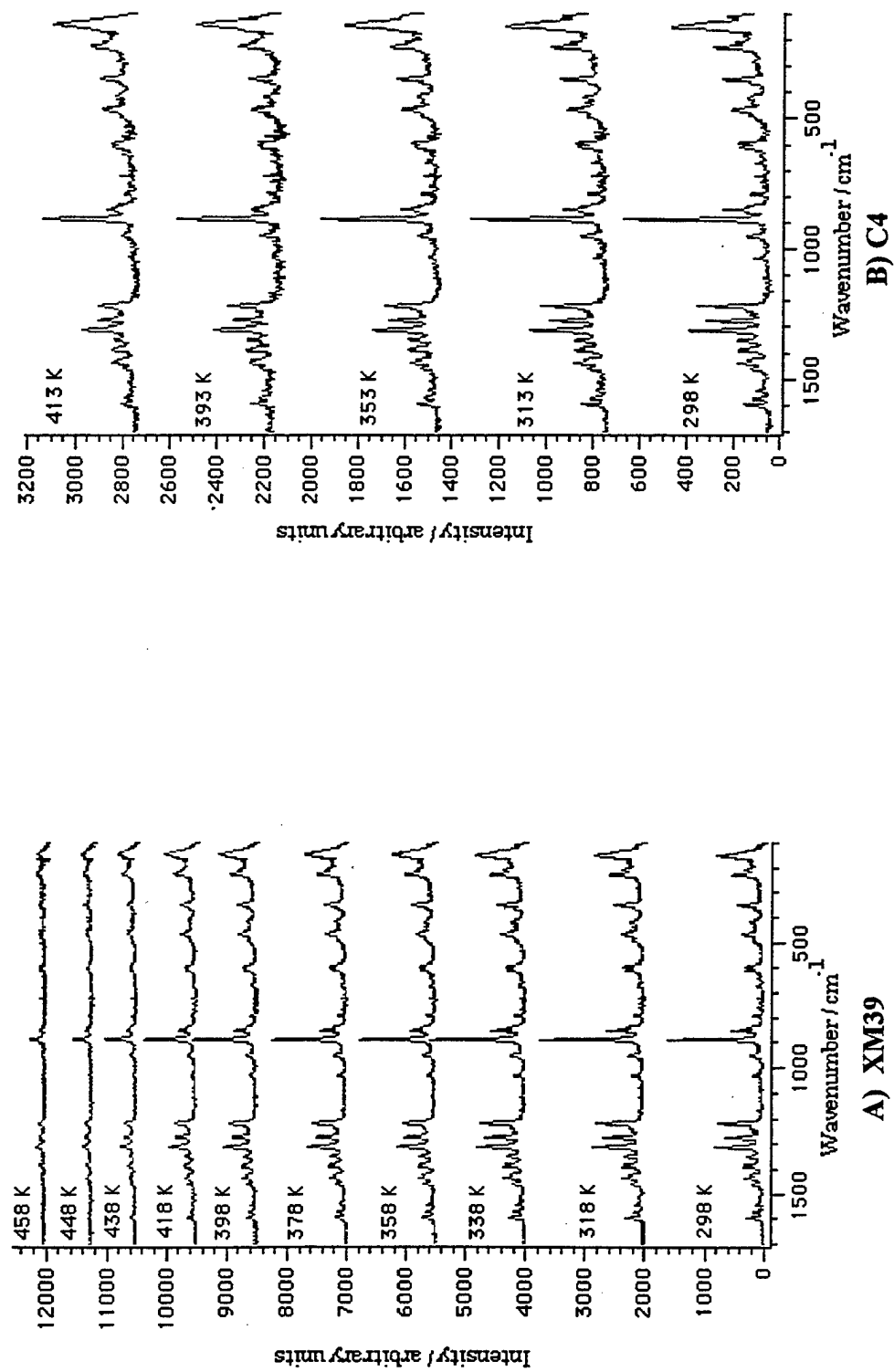


Figure 4. Raman Spectra as a Function of Temperature: (a) XM39 Propellant and (b) C4 Explosive, NIR FTIR System.

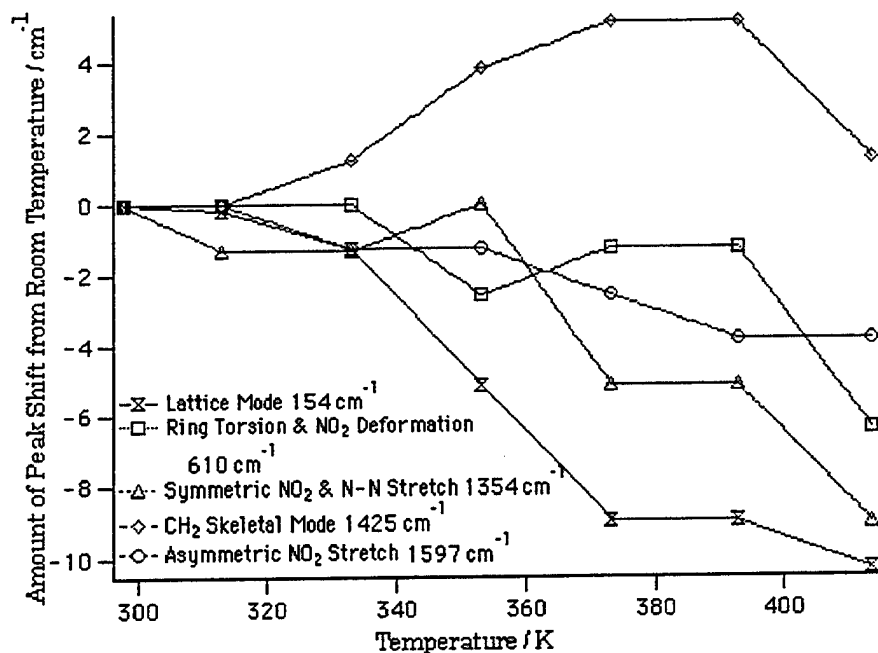


Figure 5. Raman Peak Shifts With Heating for RDX Crystal.

investigators [16]. Spectra of the four samples from 2,800 to 3,500 cm^{-1} show no significant changes in Raman shifts of individual peaks with increasing temperature and are not displayed here.

The most apparent change in the elevated temperature spectra shown in Figures 3 and 4 is a general decrease in Raman intensity as the temperature is increased. In addition to the general decrease in Raman intensity with increasing temperature, there is also a noticeable variation in relative Raman peak intensity with increasing temperature. This may be indicative of a loss of order in the crystal [13]. When the samples were allowed to return to room temperature following heating and the Raman spectrum was remeasured, overall Raman signal and relative peak intensity were similar to spectra measured prior to heating; although, for all samples measured, the final (after heating) Raman signal level was slightly less than the Raman signal level measured prior to heating.

Because reported temperatures are those of the sample holder, several measurements were made to ascertain whether the sample was being heated by the probe laser to temperatures higher than that

of the sample holder. Two different experiments were conducted to measure heating in the vicinity of the laser spot. In the first experiment, the Raman spectrum of a 0.3-mm-thick by 6.5-mm-diameter slice of unperforated XM39 propellant (76% RDX) was measured using system two (laser radiation at 785 nm as the scattering source). For this experiment, the sample was not resistively heated. The incident power was approximately 0.2 W. A K-type thermocouple, coated with heat-conducting paste, was positioned at various locations on the slice of propellant. Observed temperature increases ranged from 6 K when the thermocouple was positioned on the side opposite of the laser spot to 11 K when the thermocouple was positioned directly in the laser spot.

A second experiment used an RDX crystal placed in the heated sample holder (see Figure 3) and a Pt/Pt-10%Rh thermocouple (diameter = 0.1 mm) was placed on the surface of the crystal. A microscope cover slide was then placed on top of this assembly, pressing the thermocouple junction against the face of the crystal, and secured using a cover plate designed for this purpose. The Raman spectrum of the single crystal was then measured using 400 mW of continuous laser radiation at 1.06 μ . During these experiments, the temperature of the cell holding the RDX crystal was slowly increased. The measured temperature increases due to laser heating ranged from 6 K to 19 K. As with the experiments using visible radiation as the scattering source, the greatest temperature increase from laser heating was measured when the thermocouple was placed directly in the laser spot. Also, it was observed that as the bulk (sample holder) temperature was increased; for a given laser power, the magnitude of laser heating increased. For samples heated within 30 K of the melting point of RDX (478K), exposure to laser radiation usually caused melting/decomposition. Formulated RDX (i.e., C4, XM39) typically melted/decomposed at a lower bulk temperature than neat RDX, presumably because the formulation ingredients bring about a melting-point depression of RDX.

4. Discussion

Since the change in Raman shift frequency with temperature has been reported previously [16], the discussion section focuses on the observed changes in Raman intensity. There are a number of

possible explanations for the observed decrease of Raman signal intensity with increasing temperature. Several of these are discussed.

4.1 Laser Heating. Laser heating of RDX can result in sample sublimation, melting, and chemical decomposition. Any of these will cause less sample to be available for scattering and result in decreased Raman signal intensity. The temperature increase at the sample surface due to laser heating may be approximated by [17]:

$$T_{\text{surface}} = E_0 \alpha(1 - R)/\rho AC, \quad (1)$$

where E_0 is the incident laser energy, R is the reflectivity of RDX at the laser frequency, α is the absorption coefficient of RDX at the laser frequency, ρ is density of RDX, A is the area of the laser spot, and C is the heat capacity of RDX. For incident radiation at a wavelength of 1,064 nm (9394 cm^{-1}), α is approximately 20 cm^{-1} , and R is approximately 0.90 [18]. For a 1-s exposure, laser spot size approximately 0.5 mm in diameter, using a 400-mW incident beam, this corresponds to a temperature increase of approximately 4.5 K. Longer sample exposure, smaller laser spot size, and/or increased laser power will cause greater heating.

For the experiments reported here, in which the sample is continuously exposed to the scattering laser radiation, laser heating effects may be an important contributor to decreases in measured Raman signal intensity with increasing temperature. To determine if sample in the laser-scattering volume was removed by sublimation or melting, or by decomposition, measurements of Raman spectra of each of the samples were made after the sample had returned to room temperature. In each case studied, the Raman signal intensity returned to near preheating levels when the sample was allowed to cool. This indicates that removal of material from the scattering volume by laser heating is not entirely responsible for the observed signal decrease.

Additionally, the temperature of the sample holder for a set of experiments was compared to the temperature calculated from the Raman anti-Stokes/Stokes intensity ratio of the peak near 345 cm^{-1} for Raman spectra of a single crystal of RDX measured using radiation at a wavelength of 1064 nm

as the scattering source. The temperature of the sample can be determined from the ratio of the anti-Stokes Raman intensity, I_{as} , and the Raman Stokes intensity, I_s , as described in equation (2):

$$I_{as}/I_s = (v_L + v_i)^4/(v_L - v_i)^4 \exp(-hc\nu/kT). \quad (2)$$

Here, v_L is the laser frequency, v_i is the frequency of the vibration, h is Planck's constant, c is the speed of light, k is the Boltzman constant, and T is the temperature [19]. The results of these calculations are shown in Table 1. The sudden jump in calculated temperature above 393 K is interesting because it may suggest the bulk temperature, for this experimental setup, at which laser heating becomes important.

Table 1. Measured Temperatures and Temperatures Calculated From the Raman Anti-Stokes/Stokes Intensity Ratio From Equation (2) (Error Is Estimated at $\pm 10\%$)

Thermocouple Temperature (K)	Anti-Stokes/Stokes Intensity Ratio	Calculated Temperature	Percent Difference From Measured (%)
298.0	0.2091	267.2	-10.34
313.0	0.2354	285.3	-8.850
333.0	0.2472	293.6	-11.83
353.0	0.2641	305.5	-13.46
373.0	0.2696	309.4	-17.05
393.0	0.3763	390.6	0.6107
413.0	0.4166	424.5	2.785
433.0	0.4176	425.5	-1.732

4.2 Laser Damage. Laser radiation may cause sample surface ablation and sample cracking that can result in decreased signal intensity. To check for laser damage to the sample, single crystals of RDX were examined after several series of experiments conducted at room temperature. No visible damage (checked with an optical microscope) could be detected. Additionally, spectra of single crystals exposed to 400 mW of laser radiation at 1.06 μ for several hours at room temperature exhibited no change in Raman intensity.

4.3 Increased Rayleigh Scatter. An increase in Rayleigh scatter, brought about by temperature-induced changes in sample surface morphology (e.g., increase in the reflectivity of the sample surface) will result in decreased Raman signal intensity. An increase in the Rayleigh scatter reduces the amount of laser light available for the inelastic scattering process. No attempts were made to measure changes in Rayleigh scatter with increasing temperature.

4.4 Density Changes. Since a possible explanation for the change in frequency with increasing temperature of the Raman peak near 150 cm^{-1} is an increase in the crystal lattice spacing, a decrease in the density of the scattering material may contribute to the decrease in overall signal intensity with increasing temperature. This decrease in the number density of scattering molecules would decrease the Raman intensity. In the case of naphthalene [20], a molecular crystal, the density decreases from 1.0253 g/cm^3 at 293 K to 0.9625 g/cm^3 at 373 K or 6.125%. A similar degree of density change might be expected for RDX and RDX-containing compositions. Equation (3) describes the Raman intensity (I) as a function of laser power (E^2) and scatterer density (N). Laser power and scatterer density are functions of depth (z) within the sample.

$$I = \int_0^d K E^2(z) N(z) dz \quad (3)$$

The constant K includes the Raman cross section and geometric and instrumental factors. Figure 6 shows the Raman spectrum of naphthalene measured from room temperature to 358 K. The observed decrease in Raman signal intensity for naphthalene heated from room temperature to 358 K is larger than that predicted using equation (3).

4.5 Population Change. As temperature is increased, the population of higher vibrational states increases, thus decreasing the population of the $v = 0$ state from which most Stokes Raman scattering is generated. This would be seen experimentally by an increase in the ratio of the anti-Stokes Raman intensity (I_{as}) to the Stokes Raman intensity (I_s). From Table 1, over the temperature range studied, the I_{as}/I_s ratio increased for the 345-cm^{-1} line in the RDX crystal by a factor of 2. The Stokes signal intensity decreased continuously while the anti-Stokes Raman signal intensity initially increased, but later decreased along with the Stokes signal. If an increase in the population of higher vibrational

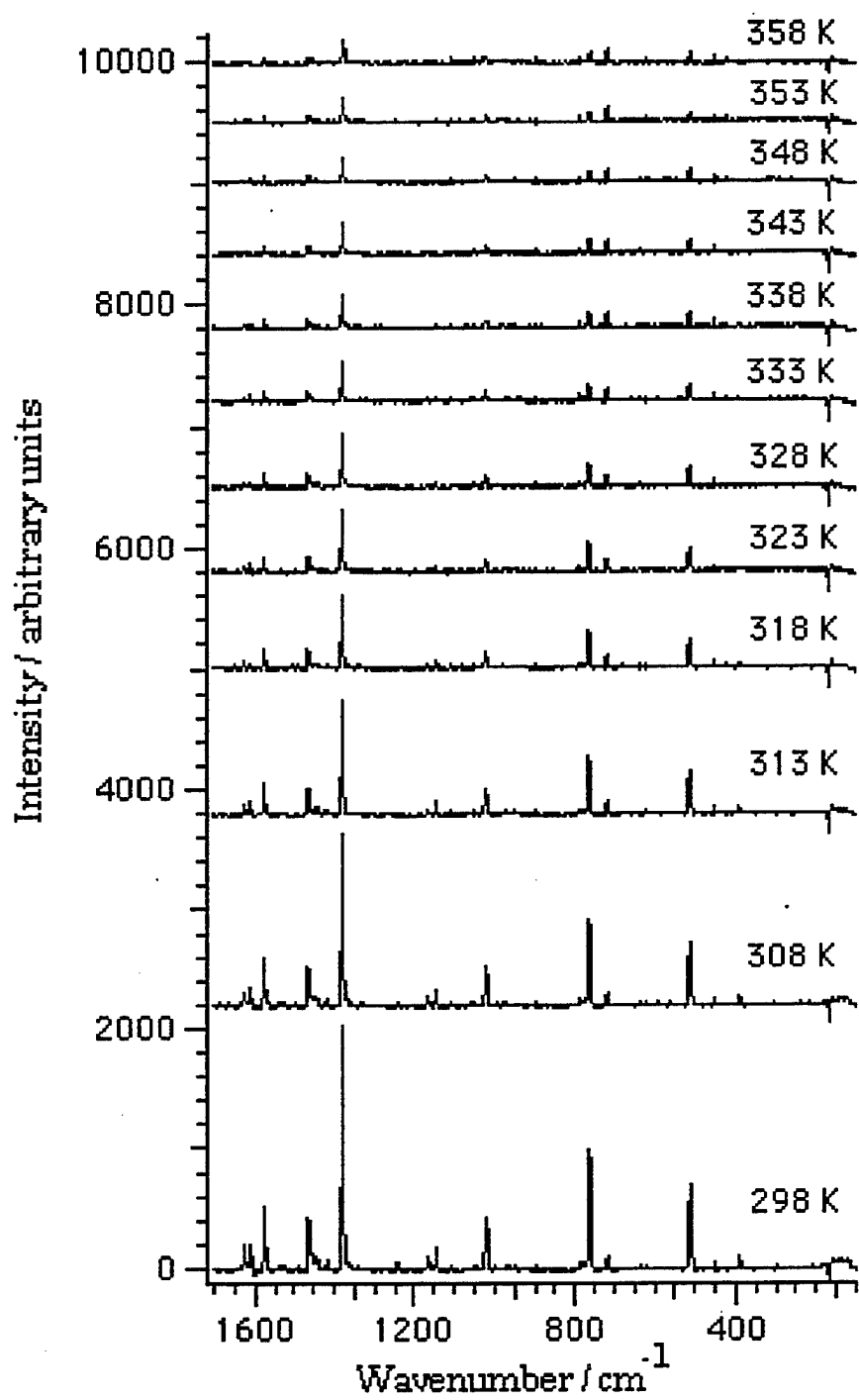


Figure 6. Raman Spectra of Naphthalene at Elevated Temperatures.

states was solely responsible for the decreased Stokes signal intensity, the anti-Stokes signal should not decrease over the temperature range studied.

4.6 Changes in Refractive Index. A change in the refractive index and transparency with increasing temperature may change the focal point of the laser radiation incident on the sample and thus move the source of the Raman signal out of focus with the entrance slit of the monochromator or the interferometer. To investigate this, at each temperature at which spectra were measured, a Raman spectrum of naphthalene was measured, then the sample position was reoptimized to obtain the maximum Raman signal, and a second Raman spectrum was collected. The results of this experiment for a range of temperatures from 313 to 373 K and only showing the Raman spectra of naphthalene after the refocusing of the sample position are shown in Figure 7. The repositioning of the sample improves, but does not restore the signal intensity to its previous level.

4.7 Premelt-Phase Transition. A premelt-phase transition may give rise to a decrease in observed Raman signal intensity with heating. Hexahydro-1,3,5-trinitroso-s-triazine (TRDX), the nitroso analog of RDX, exhibits a premelt-phase transition [21]. This type of transition may be accompanied by a decrease in the density and therefore the Raman signal. Measurements using a differential scanning calorimeter (DSC) of RDX from 300 K to 477 K did not detect such a premelt-phase transition.

4.8 Fluorescence. No measurable fluorescence signal was observed for any experiments using laser radiation at 1.06 μ as the scattering source. When laser radiation at 532 nm was used as the scattering source, a small fluorescence signal was always observed. The intensity of the fluorescence signal did not exhibit a noticeable temperature dependence.

4.9 Incandescence. For the temperature range studied in these experiments, measurable incandescence should occur only when laser heating causes the sample to combust. Since RDX melts with decomposition, there is considerable heating and gas evolution when the sample temperature reaches 477 K. Spectra of single crystals of RDX measured at the melting point using laser radiation

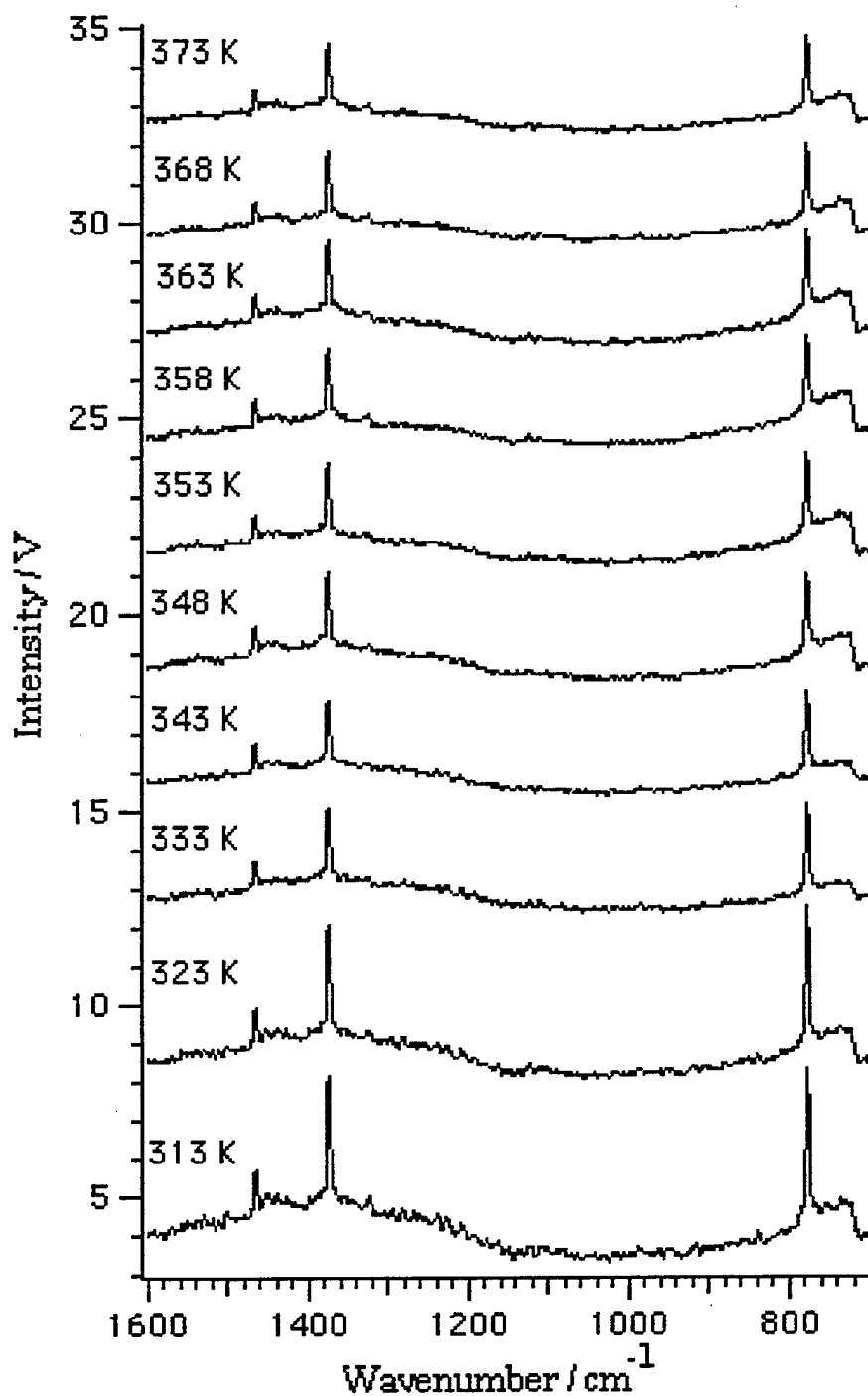


Figure 7. Raman Spectra of Naphthalene From 313 K to 373 K. Each Spectrum Shown Was Collected After the Sample Position Was Optimized After the Temperature Equilibrated at the Elevated Temperature.

at 1064 nm as the scattering source show increased background intensity beginning near a shift of 2700 cm^{-1} and increasing in the direction of greater Raman shift.

Incandescence intensity may be approximated as radiation from a blackbody. The peak in incandescence intensity near the melting point of RDX occurs at a wavelength of approximately 5,800 nm. So, when using 1,064-nm laser radiation as the scattering source, it is possible to have incandescence interferences at wavelengths corresponding to large Raman shifts typical of the C-H vibration ($\sim 3,000\text{ cm}^{-1}$). However, we saw no indication of intensity from incandescence at Raman shifts between 100 cm^{-1} and $1,600\text{ cm}^{-1}$. Potential interferences from incandescence may be almost entirely avoided by use of visible laser radiation as the scattering source.

5. Conclusions

The changes in Raman peak intensities and frequencies with increasing temperature have been examined to determine if any insight may be gained into the physical changes in RDX prior to the onset of thermal decomposition. The measured peak shifts are indicative of an increase in the lattice spacing in RDX during heating, eventually leading to loss of crystallinity (melting). These experiments have also shown a decrease in the Raman intensity as the temperature increases. It is not clear if the observed decrease in Raman intensity is due to the measuring method (laser vaporization/sublimation of sample from the scattering volume) or is caused in part by changes in the scattering cross section with increasing temperature. Results from experiments presented here indicate that sample heating by the probe laser may be an important contributor to the observed decrease in signal intensity. For RDX, sample heating is greater for laser-scattering radiation at a wavelength of 1,064 nm than for laser-scattering radiation at 514 nm. The sample heating becomes more important as the bulk sample temperature is increased. At bulk sample temperatures near the melting point of RDX, laser heating can cause the sample to undergo sublimation, melting, and/or chemical reaction. Because temperatures calculated using the Stokes/anti-Stokes ratio are in reasonable agreement with bulk temperatures, and because most of the Raman signal intensity returns when the sample is cooled, laser heating alone does not seem to be responsible for the observed decrease in Raman signal intensity with increasing temperature.

Several other factors that contribute to the decrease in observed Raman signal intensity with increasing temperature have been discussed. The decrease in Raman signal intensity with increasing temperature probably has contributions from each of these factors. In rough order of decreasing importance for experiments described in this report, the factors contributing to the decrease in Raman signal intensity with increasing sample temperature are laser sample heating, density changes, population changes, and changes in refractive index and reflectivity. To minimize problems associated with laser heating, the use of a low-power visible scattering source is recommended. Finally, to minimize sample exposure to scattering radiation for spectra of burning or elevated temperature samples, a pulsed-laser source should minimize many of the factors affecting Raman signal decrease with increasing sample temperature.

INTENTIONALLY LEFT BLANK.

6. References

1. Fell, N. F., J. W. Widder, S. V. Medlin, J. B. Morris, R. A. Pesce-Rodriguez, and K. L. McNesby. *Journal of Raman Spectroscopy*. Vol. 27, pp. 97, 1996.
2. McNesby, K. L., J. E. Wolfe, J. B. Morris, and R. A. Pesce-Rodriguez. *Journal of Raman Spectroscopy*. Vol. 25, pp. 75, 1994.
3. Akhavan, J. *Spectrochimica Acta*. Part A, vol. 47, pp. 1247, 1991.
4. Behrens, R., Jr., and S. Bulusu. *Journal of Physical Chemistry*. Vol. 96, pp. 8877, 1992.
5. Karpowicz, R. J., and T. B. Brill. *Journal of Physical Chemistry*. Vol. 88, pp. 348, 1984.
6. Botcher, T. R., and C. A. Wight. *Journal of Physical Chemistry*. Vol. 97, pp. 9149, 1993.
7. Yinon, J., R. A. Yost, and S. Bulusu. *Journal of Chromatography Part A*. Vol. 688, pp. 231, 1994.
8. Oxley, J. C., A. B. Kooh, R. Szekeres, and W. Zhang. *Journal of Physical Chemistry*. Vol. 98, pp. 7004, 1994.
9. Brill, T. B., P. E. Gongwer, and G. K. Williams. *Journal of Physical Chemistry*. Vol. 98, pp. 12242, 1994.
10. Brill, T. B. *Journal of Propulsion and Power*. Vol. 11, pp. 740, 1995.
11. Zenin, A. *Journal of Propulsion and Power*. Vol. 11, pp. 752, 1995.
12. Brill, T. B., H. Arisawa, P. J. Brush, P. E. Gongwer, and G. K. Williams. *Journal of Physical Chemistry*. Vol. 99, pp. 1384, 1995.
13. Rey-Lafon, M., C. Trinqucoste, R. Cavagnat, and M.-T. Forel. *Journal de Chimie Physique*. Vol. 68, pp. 1533, 1971.
14. Rey-Lafon, M., R. Cavagnat, C. Trinqucoste, and M.-T. Forel. *Journal de Chimie Physique*. Vol. 68, pp. 1574, 1971.
15. Rice, B., and C. Chabalowski. Private communication. U.S. Army Research Laboratory, Ignition and Combustion Branch, Aberdeen Proving Ground, MD, 1997.
16. Owens, F. J., and Z. Iqbal. *Journal of Physical Chemistry*. Vol. 74, pp. 4242, 1981.

17. Stufflebeam, J. H. 1995 JANNAF Combustion Subcommittee Meeting. Huntsville, AL, 1995.
18. Cohen, A. Private communication. U.S. Army Research Laboratory, Aberdeen Proving Ground, MD, 1997.
19. Long, D. A. *Raman Spectroscopy*. New York: McGraw-Hill, 1977.
20. Weast, R. C. (editor). *CRC Handbook of Chemistry and Physics*. 69th edition, p. C-420, Boca Raton, FL: CRC Press, 1980.
21. Mesaros, D. V., Y. Oyumi, T. B. Brill, and C. Dybowski. *Journal of Physical Chemistry*. Vol. 90, p. 1970, 1986.

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
2	DEFENSE TECHNICAL INFORMATION CENTER DTIC DDA 8725 JOHN J KINGMAN RD STE 0944 FT BELVOIR VA 22060-6218
1	HQDA DAMO FDQ DENNIS SCHMIDT 400 ARMY PENTAGON WASHINGTON DC 20310-0460
1	DPTY ASSIST SCY FOR R&T SARD TT F MILTON RM 3EA79 THE PENTAGON WASHINGTON DC 20310-0103
1	OSD OUSD(A&T)/ODDDR&E(R) R J TREW THE PENTAGON WASHINGTON DC 20301-7100
1	CECOM SP & TRRSTRM COMMCTN DIV AMSEL RD ST MC M H SOICHER FT MONMOUTH NJ 07703-5203
1	PRIN DPTY FOR TCHNLGY HQ US ARMY MATCOM AMCDG T M FISETTE 5001 EISENHOWER AVE ALEXANDRIA VA 22333-0001
1	DPTY CG FOR RDE HQ US ARMY MATCOM AMCRD BG BEAUCHAMP 5001 EISENHOWER AVE ALEXANDRIA VA 22333-0001
1	INST FOR ADVNCD TCHNLGY THE UNIV OF TEXAS AT AUSTIN PO BOX 202797 AUSTIN TX 78720-2797

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	GPS JOINT PROG OFC DIR COL J CLAY 2435 VELA WAY STE 1613 LOS ANGELES AFB CA 90245-5500
3	DARPA L STOTTS J PENNELLA B KASPAR 3701 N FAIRFAX DR ARLINGTON VA 22203-1714
1	US MILITARY ACADEMY MATH SCI CTR OF EXCELLENCE DEPT OF MATHEMATICAL SCI MDN A MAJ DON ENGEN THAYER HALL WEST POINT NY 10996-1786
1	DIRECTOR US ARMY RESEARCH LAB AMSRL CS AL TP 2800 POWDER MILL RD ADELPHI MD 20783-1145
1	DIRECTOR US ARMY RESEARCH LAB AMSRL CS AL TA 2800 POWDER MILL RD ADELPHI MD 20783-1145
3	DIRECTOR US ARMY RESEARCH LAB AMSRL CI LL 2800 POWDER MILL RD ADELPHI MD 20783-1145
	<u>ABERDEEN PROVING GROUND</u>
4	DIR USARL AMSRL CI LP (305)

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	HQDA OASA RDA DR C H CHURCH PENTAGON RM 3E486 WASHINGTON DC 20310-0103
4	CDR USARO R GHIRARDELLI D MANN R SINGLETON R SHAW PO BOX 12211 RESEARCH TRIANGLE PARK NC 27709-2211
1	DIR USARO AMXRO MCS K CLARK PO BOX 12211 RESEARCH TRIANGLE PARK NC 27709-2211
1	DIR USARO AMXRO RT IP LIB SERVICES PO BOX 12211 RESEARCH TRIANGLE PARK NC 27709-2211
2	CDR USARDEC AMSTA AR AEE B D S DOWNS PICATINNY ARSENAL NJ 07806-5000
2	CDR USARDEC AMSTA AR AEE J A LANNON PICATINNY ARSENAL NJ 07806-5000
1	CDR USARDEC AMSTA AR AEE BR L HARRIS PICATINNY ARSENAL NJ 07806-5000
2	COMMANDER US ARMY MISSILE COMMAND AMSMI RD PR E A R MAYKUT AMSMI RD PR P R BETTS REDSTONE ARSENAL AL 35809

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	DIRECTOR US ARMY BENET LABS AMSTA AR CCB T S SOPOK WATERVLIET NY 12189
1	OFFICE OF NAVAL RESEARCH DEPT OF THE NAVY R S MILLER CODE 432 800 N QUINCY STREET ARLINGTON VA 22217
1	COMMANDER NAVAL AIR SYSTEMS COMMAND J RAMNARACE AIR 54111C WASHINGTON DC 20360
2	CDR NSWC R BERNECKER R 13 G B WILMOT R 16 SILVER SPRING MD 20903-5000
5	COMMANDER NAVAL RESEARCH LAB M C LIN J MCDONALD E ORAN J SHNUR R J DOYLE CODE 6110 WASHINGTON DC 20375
2	COMMANDER NAVAL WEAPONS CENTER T BOGGS CODE 388 T PARR CODE 3895 CHINA LAKE CA 93555-6001
1	SUPERINTENDENT NAVAL POSTGRADUATE SCHOOL DEPT OF AERONAUTICS D W NETZER MONTEREY CA 93940
3	AL LSCF R CORLEY R GEISLER J LEVINE EDWARDS AFB CA 93523-5000

NO. OF
COPIES ORGANIZATION

1 AFOSR
J M TISHKOFF
BOLLING AIR FORCE BASE
WASHINGTON DC 20332

1 OSD SDIO IST
L CAVENY
PENTAGON
WASHINGTON DC 20301-7100

1 COMMANDANT
USAFAS
ATSF TSM CN
FORT SILL OK 73503-5600

1 UNIV OF DAYTON RSCH INSTITUTE
D CAMPBELL
AL PAP
EDWARDS AFB CA 93523

1 NASA
LANGLEY RESEARCH CTR
G B NORTHAM MS 168
LANGLEY STATION
HAMPTON VA 23365

4 NATIONAL BUREAU OF STANDARDS
US DEPT OF COMMERCE
J HASTIE
M JACOX
T KASHIWAGI
H SEMERJIAN
WASHINGTON DC 20234

2 DIR LLNL
C WESTBROOK
W TAO MS L 282
PO BOX 808
LIVERMORE CA 94550

1 DIR LANL
B NICHOLS T7 MS B284
PO BOX 1663
LOS ALAMOS NM 87545

1 NATIONAL SCIENCE FOUNDATION
A B HARVEY
WASHINGTON DC 20550

NO. OF
COPIES ORGANIZATION

2 PRINCETON COMBUSTION
RESEARCH LABORATORIES INC
N A MESSINA
M SUMMERFIELD
PRINCETON CORPORATE PLAZA
BLDG IV SUITE 119
11 DEERPARK DRIVE
MONMOUTH JUNCTION NJ 08852

3 DIR SNL
DIVISION 8354
S JOHNSTON
P MATTERN
D STEPHENSON
LIVERMORE CA 94550

1 BRIGHAM YOUNG UNIV
DEPT OF CHEMICAL ENGINEERING
M W BECKSTEAD
PROVO UT 84058

1 CALIF INSTITUTE OF TECH
JET PROPULSION LAB
L STRAND MS 125 224
4800 OAK GROVE DRIVE
PASADENA CA 91109

1 CALIF INSTITUTE OF TECH
F E CULICK MC 301 46
204 KARMAN LAB
PASADENA CA 91125

1 UNIV OF CALIF BERKLEY
CHEMISTRY DEPT
C BRADLEY MOORE
211 LEWIS HALL
BERKELEY CA 94720

1 UNIV OF CALIF SAN DIEGO
F A WILLIAMS
AMES B010
LA JOLLA CA 92093

2 UNIV OF CALIF SANTA BARBARA
QUANTUM INSTITUTE
K SCHOFIELD
M STEINBERG
SANTA BARBARA CA 93106

<u>NO. OF</u> <u>COPIES</u>	<u>ORGANIZATION</u>
1	UNIV OF COLORADO AT BOULDER ENGINEERING CENTER J DAILY CAMPUS BOX 427 BOULDER CO 80309-0427
3	UNIV OF SOUTHERN CALIF DEPT OF CHEMISTRY R BEAUDET S BENSON C WITTIG LOS ANGELES CA 90007
1	CORNEL UNIV DEPT OF CHEMISTRY T A COOL BAKER LABORATORY ITHACA NY 14853
1	UNIV OF DELAWARE CHEMISTRY DEPT T BRILL NEWARK DE 19711
1	UNIV OF FLORIDA DEPT OF CHEMISTRY J WINEFORDNER GAINESVILLE FL 32611
3	GEORGIA INSTITUTE OF TECH SCHOOL OF SEROSPACE ENGNRG E PRICE W C STRAHLE B T ZINN ATLANTA GA 30332
1	UNIV OF ILLINOIS DEPT OF MECH ENG H KRIER 144MEB 1206 W GREEN ST URGANA IL 61801
1	THE JOHNS HOPKINS UNIV CPIA T W CHRISTIAN 10630 LITTLE PATUXENT PKWY SUITE 202 COLUMBIA MD 21044-3200

<u>NO. OF</u> <u>COPIES</u>	<u>ORGANIZATION</u>
1	UNIV OF MICHIGAN GAS DYNAMICS LAB G M FAETH AEROSPACE ENGINEERING BLDG ANN ARBOR MI 48109-2140
1	UNIV OF MINNESOTA DEPT OF MECH ENGINEERING E FLETCHER MINNEAPOLIS MN 55455
4	PENNS STATE UNIV DEPT OF MECH ENGINEERING K KUO M MICCI S THYNELL V YANG UNIVERSITY PARK PA 16802
1	POLYTECHNIC INSTITUTE OF NY GRADUATE CENTER S LEDERMAN ROUTE 110 FARMINGDALE NY 11735
2	PRINCETON UNIVERSITY FORRESTAL CAMPUS LIBRARY K BREZINSKY I GLASSMAN PO BOX 710 PRINCETON NJ 08540
1	PURDUE UNIVERSITY SCHL OF AERONAUTICS & ASTRONAUTICS J R OSBORN GRISSOM HALL WEST LAFAYETTE IN 47906
1	PURDUE UNIVERSITY DEPT OF CHEMISTRY E GRANT WEST LAFAYETTE IN 47906
2	PURDUE UNIVERSITY SCHL OF MECH ENGNRG N M LAURENDEAU S N B MURTHY TSPC CHAFFEE HALL WEST LAFAYETTE IN 47906

NO. OF
COPIES ORGANIZATION

1 RENSSELAER POLYTECHNIC INST
DEPT OF CHEMICAL ENGRG
A FONTIJN
TROY NY 12181

1 STANFORD UNIVERSITY
DEPT OF MECH ENGRG
R HANSON
STANFORD CA 94305

1 UNIV OF TEXAS
DEPT OF CHEMISTRY
W GARDINER
AUSTIN TX 78712

1 VA POLYTECH INST AND STATE UNIV
J A SCHETZ
BLACKSBURG VA 24061

1 APPLIED COMBUSTION TECH INC
A M VARNEY
PO BOX 607885
ORLANDO FL 32860

2 APPLIED MECHANICS REVIEWS
ASME R E WHITE & A B WENZEL
345 E 47TH STREET
NEW YORK NY 10017

1 ATLANTIC RESEARCH CORP
R H W WAESCHE
7511 WELLINGTON ROAD
GAINESVILLE VA 22065

1 TEXTRON DEFENSE SYSTEMS
A PATRICK
2385 REVERE BEACH PARKWAY
EVERETT MA 02149-5900

1 BATTELLE
TWSTIAC
505 KING AVENUE
COLUMBUS OH 43201-2693

1 COHEN PROFESSIONAL SERVICES
N S COHEN
141 CHANNING STREET
REDLANDS CA 92373

NO. OF
COPIES ORGANIZATION

1 EXXON RESEARCH & ENG CO
A DEAN
ROUTE 22E
ANNADALE NJ 08801

1 GENERAL APPLIED SCIENCE LABS INC
77 RAYNOR AVENUE
RONKONKAMA NY 11779-6649

1 GEN ELEC ORDNANCE SYSTEMS
J MANDZY
100 PLASTICS AVENUE
PITTSFIELD MA 01203

1 GENERAL MOTORS RSCH LABS
PHYSICAL CHEMISTRY DEPT
T SLOANE
WARREN MI 48090-9055

2 HERCULES INC
W B WALKUP
E A YOUNT
PO BOX 210
ROCKET CENTER WV 26726

1 HERCULES INC
R V CARTWRIGHT
100 HOWARD BLVD
KENVIL NJ 07847

1 ALLIANT TECHSYSTEMS INC
MARINE SYSTEMS GROUP
D E BRODEN MS MN50 2000
600 2ND STREET NE
HOPKINS MN 55343

1 ALLIANT TECHSYSTEMS INC
R E TOMPKINS
MN 11 2720
600 2ND ST NORTH
HOPKINS MN 55343

1 IBM CORPORATION
RESEARCH DIVISION
A C TAM
5600 COTTLE ROAD
SAN JOSE CA 95193

NO. OF
COPIES ORGANIZATION

1 ITT RESEARCH INSTITUTE
R FREMALY
10 WEST 35TH STREET
CHICAGO IL 60616

1 LOCKHEED MISSILES & SPACE CO
GEORGE LO
3251 HANOVER STREET
DEPT 5235 B204 2
PALO ALTO CA 94304

1 OLIN ORDNANCE
V MCDONALD LIBRARY
PO BOX 222
ST MARKS FL 32355-0222

1 PAUL GOUGH ASSOCIATES INC
P S GOUGH
1048 SOUTH STREET
PORTSMOUTH NH 03801-5423

1 HUGHES AIRCRAFT COMPANY
T E WARD
8433 FALLBROOK AVENUE
CANOGA PARK CA 91303

1 ROCKWELL INTERNATIONAL CORP
ROCKETDYNE DIVISION
J E FLANAGAN HB02
6633 CANOGA AVENUE
CANOGA PARK CA 91304

1 SAIC
R B EDELMAN
23146 CUMORAH CREST
WOODLAND HILLS CA 91364

3 SRI INTERNATIONAL
G SMITH
D CROSLEY
D GOLDEN
333 RAVENSWOOD AVENUE
MENLO PARK CA 94025

1 STEVENS INSTITUTE OF TECH
DAVIDSON LABORATORY
R MCALEVY III
HOBOKEN NJ 07030

NO. OF
COPIES ORGANIZATION

1 SVERDRUP TECHNOLOGY INC
LERC GROUP
R J LOCKE MS SVR 2
2001 AEROSPACE PARKWAY
BROOK PARK OH 44142

1 SVERDRUP TECHNOLOGY INC
J DEUR
2001 AEROSPACE PARKWAY
BROOK PARK OH 44142

3 THIOKOL CORPORATION
ELKTON DIVISION
R BIDDLE
R WILLER
TECH LIB
PO BOX 241
ELKTON MD 21921

3 THIOKOL CORPORATION
WASATCH DIVISION
S J BENNETT
PO BOX 524
BRIGHAM CITY UT 84302

1 UNITED TECHNOLOGIES RSCH CTR
A C ECKBRETH
EAST HARTFORD CT 06108

1 UNITED TECHNOLOGIES CORP
CHEMICAL SYSTEMS DIVISION
R R MILLER
PO BOX 49028
SAN JOSE CA 95161-9028

1 UNIVERSAL PROPULSION CO
H J MCSPADDEN
25401 NORTH CENTRAL AVENUE
PHOENIX AZ 85027-7837

1 VERITAY TECHNOLOGY INC
E B FISHER
4845 MILLERSPORT HIGHWAY
EAST AMHERST NY 14051-0305

1 FREEDMAN ASSOCIATES
E FREEDMAN
2411 DIANA ROAD
BALTIMORE MD 21209-1525

NO. OF
COPIES ORGANIZATION

6 ALLIANT TECHSYSTEMS
C CANDLAND
L OSGOOD
R BECKER
J BODE
R BURETTA
M SWENSON
600 2ND ST NE
HOPKINS MN 55343

ABERDEEN PROVING GROUND

35 DIR USARL
AMSRL WM B A HORST
AMSRL WM BD
B E FORCH
G F ADAMS
W R ANDERSON
R A BEYER
S W BUNTE
C F CHABALOWSKI
S COLEMAN
A COHEN
R CUMPTON
R DANIEL
D DEVYNCK
N F FELL
J H HEIMERL
A J KOTLAR
M R MANAA
W F MCBRATNEY
K L MCNESBY
S V MEDLIN
M S MILLER
A W MIZIOLEK
S H MODIANO
J B MORRIS
J E NEWBERRY
S A NEWTON
R A PESCE-RODRIGUEZ
B M RICE
R C SAUSA
M A SCHROEDER
J A VANDERHOFF
M WENSING
A WHREN
J M WIDDER
C WILLIAMSON
AMSRL CI CA R PATEL

INTENTIONALLY LEFT BLANK.

REPORT DOCUMENTATION PAGE			Form Approved OMB No. 0704-0188	
Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project(0704-0188), Washington, DC 20503.				
1. AGENCY USE ONLY (Leave blank)	2. REPORT DATE August 1998	3. REPORT TYPE AND DATES COVERED Final, Jan 96 - Jan 97		
4. TITLE AND SUBTITLE Characterization of Raman Spectral Changes in Energetic Materials and Propellants During Heating			5. FUNDING NUMBERS 1L161102AH43	
6. AUTHOR(S) Nicholas F. Fell, Jr., John A. Vanderhoff, Rose A. Pesce-Rodriguez, and Kevin L. McNesby				
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) U.S. Army Research Laboratory ATTN: AMSRL-WM-BD Aberdeen Proving Ground, MD 21005-5066			8. PERFORMING ORGANIZATION REPORT NUMBER ARL-TR-1743	
9. SPONSORING/MONITORING AGENCY NAMES(S) AND ADDRESS(ES)			10. SPONSORING/MONITORING AGENCY REPORT NUMBER	
11. SUPPLEMENTARY NOTES				
12a. DISTRIBUTION/AVAILABILITY STATEMENT Approved for public release; distribution is unlimited.			12b. DISTRIBUTION CODE	
13. ABSTRACT (Maximum 200 words) Raman spectroscopy has been shown to be a useful tool for characterizing neat crystalline explosive samples and for identifying principle components in many propellant and explosive formulations. Herein, we report recent measurements of Raman spectra of explosives and propellant formulations during bulk heating and recent measurements of laser heating of the samples during measurement of Raman spectra. The results of these measurements are important to investigators using Raman spectroscopy to measure vibrational spectra of burning propellant samples.				
14. SUBJECT TERMS Raman, spectroscopy, explosives, heating			15. NUMBER OF PAGES 32	
			16. PRICE CODE	
17. SECURITY CLASSIFICATION OF REPORT UNCLASSIFIED	18. SECURITY CLASSIFICATION OF THIS PAGE UNCLASSIFIED	19. SECURITY CLASSIFICATION OF ABSTRACT UNCLASSIFIED	20. LIMITATION OF ABSTRACT UL	

INTENTIONALLY LEFT BLANK.

USER EVALUATION SHEET/CHANGE OF ADDRESS

This Laboratory undertakes a continuing effort to improve the quality of the reports it publishes. Your comments/answers to the items/questions below will aid us in our efforts.

1. ARL Report Number/Author ARL-TR-1743 (Fell) Date of Report August 1998

2. Date Report Received _____

3. Does this report satisfy a need? (Comment on purpose, related project, or other area of interest for which the report will be used.) _____

4. Specifically, how is the report being used? (Information source, design data, procedure, source of ideas, etc.) _____

5. Has the information in this report led to any quantitative savings as far as man-hours or dollars saved, operating costs avoided, or efficiencies achieved, etc? If so, please elaborate. _____

6. General Comments. What do you think should be changed to improve future reports? (Indicate changes to organization, technical content, format, etc.) _____

CURRENT
ADDRESS

Organization

Name

E-mail Name

Street or P.O. Box No.

City, State, Zip Code

7. If indicating a Change of Address or Address Correction, please provide the Current or Correct address above and the Old or Incorrect address below.

OLD
ADDRESS

Organization

Name

Street or P.O. Box No.

City, State, Zip Code

(Remove this sheet, fold as indicated, tape closed, and mail.)
(DO NOT STAPLE)

DEPARTMENT OF THE ARMY

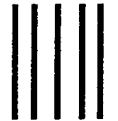
OFFICIAL BUSINESS

BUSINESS REPLY MAIL

FIRST CLASS PERMIT NO 0001,APG,MD

POSTAGE WILL BE PAID BY ADDRESSEE

DIRECTOR
US ARMY RESEARCH LABORATORY
ATTN AMSRL WM BD
ABERDEEN PROVING GROUND MD 21005-5066



NO POSTAGE
NECESSARY
IF MAILED
IN THE
UNITED STATES

