

AFFDL-TM-73-131-FBR

FBR 8254

(Shil  
53297)

AIR FORCE FLIGHT DYNAMICS LABORATORY  
DIRECTOR OF LABORATORIES  
AIR FORCE SYSTEMS COMMAND  
WRIGHT PATTERSON AIR FORCE BASE OHIO



A USER'S MANUAL FOR THE  
SEQUENCE ACCOUNTABLE FATIGUE ANALYSIS  
COMPUTER PROGRAM

OCTOBER 1973

J. M. POTTER  
R. A. NOBLE

19990701 025

Approved for public release, distribution unlimited

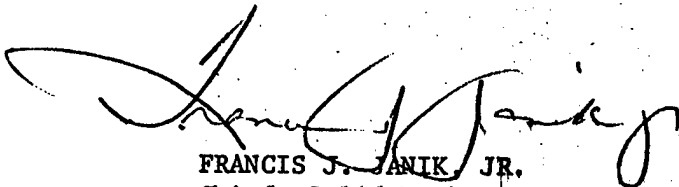
RETURN TO: AEROSPACE STRUCTURES  
INFORMATION AND ANALYSIS CENTER  
AFFDL/FBR  
WPAFB, OHIO 45433

FORWARD

This program was prepared by J. M. Potter of the Solid Mechanics Branch and R. A. Noble of the Experimental Branch, Structures Division, Air Force Flight Dynamics Laboratory. This work was conducted in-house under Project 1347, "Structural Testing of Flight Vehicles," Task 134704, "Structural Testing Criteria." This memo covers work accomplished over a time period of 1 October 1972 to 1 May 1973.

This manuscript was released by the authors in October 1973.

This Technical Memorandum has been reviewed and is approved.



FRANCIS J. JANIK JR.  
Chief, Solid Mechanics Branch  
Structures Division

## ABSTRACT

This report presents a detailed description of a computer program to calculate cumulative damage of notched structural members subjected to arbitrary spectra. The Sequence Accountable Fatigue Analysis computer program develops its sequence sensitivity by tracking residual stresses local to a notch throughout the spectrum of loads. Residual stress relaxation analysis is included to increase the generality of the results. An example spectrum and resulting cumulative damage analysis are illustrated.

## TABLE OF CONTENTS

SECTION		PAGE
I	Introduction	1
II	Program Outline	2
III	Input Data Requirements	6
IV	Output Options	9
REFERENCES		10
FIGURES		12
APPENDIX I	Program List	16
APPENDIX II	Sample Problem with Input Data Listing	37
APPENDIX III	List of Computer Program Symbols and Definitions	54
APPENDIX IV	Fatigue Life Input Data for Several Materials	60

## SYMBOLS

$\sigma_{res}$	Residual stress
$\sigma_{max}$	Maximum local stress level
$\sigma_{min}$	Minimum local stress level
$\sigma_{ys}$	Yield stress
$\sigma_{res_{EQ}}$	Equilibrium component of the residual stress
$\epsilon_t$	Local strain, total
$\epsilon_e$	Elastic component of total strain
$\epsilon_p$	Plastic component of total strain
$S_{max}$	Maximum applied stress level
$S_{min}$	Minimum applied stress level
$S_{mean}$	Mean applied stress level, $(S_{max} + S_{min})/2$
$K_t$	Elastic stress concentration Factor
D	Damage
I	Integer describing the level number
$N_{EP}$	Equilibrium period, number of cycles for the local stresses to return approximately to the equilibrium conditions following an overload
$C_1$	Residual stress relaxation constant
$E_1, E_2$	Relaxation function exponents
N	Number of cycles
E	Modulus of elasticity
$N_f$	Number of cycles of life at a given stress or strain cycling level
$\epsilon_p^0$	Strain intercept at one reversal on a log $\epsilon_p$ -log life curve
$m$	Slope of the log $\epsilon_p$ -log life curve

## SECTION I

### INTRODUCTION

Cumulative damage analyses based upon the local stress-strain behavior at a notch appear to be reasonably successful in anticipating trends in fatigue life behavior of notched specimens subjected to spectrum loading (1-6). The type of behavior that usually occurs is that peak tensile loads tend to increase the fatigue life and peak compressive loads tend to decrease the life of notched structures compared to structures experiencing load spectra not having those peaks (5,6). Local behavior analyses, such as those developed by Smith (7) and Neuber (8), help to explain this phenomenon as being a result of the tensile peak load creating a compressive residual stress at the notch and, conversely, the compressive peak creating a tensile residual stress. The change in life occurs because the residual stress state modifies the subsequent damage accumulation rates.

The Sequence Accountable Fatigue Analysis computer program was developed to incorporate the local stress-strain approach with a recent residual stress relaxation analysis (6) in order to improve the sequence sensitivity of cumulative damage analysis. This technical memorandum presents the details of the resultant computer program and an example of its use. The correlation of predictions made with this analysis to actual results of tests experiencing spectrum loading is presented by Potter (9) and Potter and Noble (10).

SECTION II  
PROGRAM OUTLINE

The Sequence Accountable Fatigue Analysis traces the stress-strain behavior local to a notch throughout an applied load spectrum and calculates the damage based on the local experience. The computer program is divided generally into the four parts or modules outlined in Fig. 1.

The basic input data for the material, specimen geometry, fatigue behavior qualities and spectrum, are developed in Module I. Module I is discussed further in Section III. Module II takes the input information and determines the local stress-strain behavior. Module III references the Range Pair Counting Method Subroutine to cycle count the local stress spectrum. Module IV determines the damage in the local stress-strain spectrum.

The basic analyses used in Modules II, III and IV are presented below.

Module II - Local Stress-Strain Behavior - The analysis used during the determination of the local stress behavior during the spectrum of loading is a combination of analyses developed by Smith (7), Neuber (8) and Petter (6). Smith's simple analysis indicated that the residual stress could be approximated by assuming that the initial stress-strain behavior was elastic upon unloading following plastic flow. Thus, the residual stress could be calculated knowing the

maximum local stress and the maximum applied stress as in Eq. 1 and in Fig. 2.

$$\sigma_{res_i} = \sigma_{max_i} - K_t S_{max_i} \quad (1)$$

The Sequence Accountable Fatigue Analysis computer program currently incorporates elastic-perfectly plastic stress-strain behavior. Therefore,  $\sigma_{max_i}$  is equal to the yield stress. For the cycles immediately following the peak stress, the residual stress determined in Eq. 1 modifies the elastic solution as shown in Eqs. 2 and 3 (provided that the following maximum applied stress is less than  $S_{max}$  and that there is no change in the residual stress due to a minimum applied stress causing reversed yielding).

$$\sigma_{max_i} = \sigma_{res_{i-1}} + K_t S_{max_i} \quad (2)$$

$$\sigma_{max_i} = \sigma_{res_{i-1}} + K_t S_{max_i} \quad (3)$$

The analysis developed by Neuber <sup>(8)</sup> has been extended to cyclic loading by Wetzel <sup>(2)</sup> and Wetzel, Morrow and Topper <sup>(3)</sup> and used by many others <sup>(1,4-6)</sup> primarily to determine local stress-strain behavior. It is used in this program only to calculate plastic strains occurring when the residual stress undergoes a step change. The plastic strain calculation routine is accessed only when the  $\sigma_{max_i}$  or  $\sigma_{min_i}$  terms in Eqs. 2 and 3 exceed tensile or compressive yield stress levels, respectively. Figure 3 illustrates the calculation of the plastic strain.

The local stress-strain behavior, according to Wetzel <sup>(2)</sup> is related to the applied load by Eq. 4

$$\sigma \cdot \epsilon = (K_t S_{max})^2 / E \quad (4)$$

The plastic strain can be found by subtracting the elastic component from the total strain.

$$\epsilon_p = \epsilon_t - \epsilon_e = (K_t S_{\max})^2 / E \cdot \sigma_{\max} - \sigma_{\max} / E$$

Therefore, the plastic strain associated with  $S_{\max_i}$  is given in Eq. 5.

$$\epsilon_{p_i} = (K_t S_{\max_i})^2 / E \sigma_{ys} - \sigma_{ys} / E \quad (5)$$

If a residual stress existed prior to this plastic strain excursion, the plastic strain associated with that prior excursion is subtracted from Eq. 5 as shown in Eq. 6.

$$\epsilon_{p_i} = (K_t S_{\max_i})^2 / E \sigma_{ys} - (\sigma_{ys} - \sigma_{res_{i-1}})^2 / E \sigma_{ys} \quad (6)$$

A similar calculation is made for plastic strains occurring during the minimum stress peak.

In the analysis developed by Potter <sup>(6)</sup> the residual stress cyclically relaxes toward zero or an equilibrium residual stress as shown in Fig. 4 according to Eq. 7.

$$\sigma_{res_{N=1,2,\dots}} = (\sigma_{res_{N=1}} - \sigma_{res_{EQ}}) \exp(N/N_{EP_i} \ln(0.1)) \quad (7)$$

The  $N_{EP}$  term, the Equilibrium Period, is dependent upon the applied stress and the Residual Stress Relaxation Constant.

$$N_{EP_i} = (C1 / (K_t S_{\max_i}^{E1} \cdot K_t S_{\text{mean}_i}^{E2})) \quad (8)$$

The Residual Stress Relaxation Constant,  $C1$ , has not yet been experimentally defined but should be a constant for a material.

### Module III - Cycle Counting Method

After the local stress and plastic strain behavior is calculated, the local stress spectrum is Range Pair Counted using a computer program developed by Tischler. (11)

#### Module IV - Damage Calculation

Damage is calculated separately for the plastic strain excursions and the elastic stress spectrum. The damage is determined from the conventional  $D = \sum \frac{n}{N}$  calculation. Damage from each of the plastic strain cycles is determined from the Coffin-Manson (12) form.

$$D_i = 1./N_{f_i} = 1./(\epsilon_{p_i}/\epsilon_f)^{1/c}$$

Damage from the elastic stress cycles is determined in a similar manner. The maximum and minimum local stress levels are sequentially compared to unnotched S-N data in a Modified Goodman Diagram format. Damage is summed, and failure of the coupon is defined as the event occurring when the summed damage equals unity.

### SECTION III

#### INPUT DATA REQUIREMENTS

In general, each spectrum analyzed will require slightly different programming in order to get the load history into a usable format for the core program. The basic program requires a certain family of information before any analytical predictions can be made. Appendix I contains a program listing for the Sequence Accountable Fatigue Analysis. The subroutine CORE which accesses the subroutines having to do with RPCM, the Range Pair Counting Method, contains the basic analysis. Subroutine SAL reads the data input and then references subroutine CORE. The subroutine SAL shown is one in which a block of cycles is repeated with optional cycles. A list of the input data cards and the resulting analysis is given in Appendix II.

The specific data requirements are given below.

1. Stress-Strain Behavior - The stress-strain behavior is presumed to be elastic - perfectly plastic with the tensile yield stress being equal to the compressive yield stress. The yield stress value used is an average of the monotonic behavior generally being above the 0.2% yield value and below the engineering ultimate strength.

2. Residual Stress Relaxation - The residual stress relaxation behavior of Eq. 7 and 8 is characterized by  $C_1$ , the Residual Stress Relaxation Constant and  $E_1$  and  $E_2$ , the relaxation equation exponents. The Residual Stress Relaxation Constant,  $C_1$ , has not yet been adequately determined. It should be a material property if the relaxation function

is correct and must be assumed. A reasonably accurate estimate of the Residual Stress Relaxation Constant for aluminum material falls in the range of  $5-20 \times 10^6$  (cycles) (Ksi)<sup>2</sup>. Further experimentation on the part of the analyst should develop a CI usable for his set of conditions until actual measurement of residual stress relaxation behavior defines the relaxation function and constants. At present E1 and E2 are considered to be equal to 1.0.

3. Specimen Geometry - The elastic  $K_t$  value (if available) is entered into the analysis. If that value is not available then an estimate from some other method may be used. In certain cases, a value may be determined from a constant amplitude fatigue test of a similar structure by fitting several values of  $K_t$  to the analysis and determining the best correlation as is done with the  $K_f$  solution. Once a stress concentration factor,  $K_t$ , is determined for a specimen, that value is not changed from test-to-test of the same coupon configuration,

4. Load Multiplier - Different spectra are presented for analysis in different manners. Some data are presented in percent of maximum stress, others in terms of nominal stress, and others in terms of bending moment. The value of the load multiplier defines the nominal stress history.

5. Cumulative Damage Analysis - The damage from the range-paired elastic stress spectrum is determined by calculating a simple  $\frac{n}{N}$  value for each level and accumulating the total. The  $N_{fi}$  value is determined from unnotched coupon S-N data in the Modified Goodman Diagram format.

The program requires the input of four second order equations describing the maximum and minimum stress levels at lives of  $10^4$ ,  $10^5$ ,  $10^6$  and  $10^7$  cycles. The coefficients of the equations are derived by least square fitting the S-N data presented in the form of Eq. 9.

$$S_{\max} = A(I)S_{\min}^2 + B(I)S_{\min} + C(I) \quad (9)$$

The A, B, and C coefficients for several typical materials are presented in Appendix IV. The S-N data shown was derived from various sources but usually from the MIL-HDBK-5A <sup>(13)</sup>. The C coefficients correspond to the maximum stress level at zero to maximum applied stress conditions on the unnotched coupons.

The damage from the plastic strain cycles is determined using the Coffin-Manson relation to calculate the  $N_{f_1}$  value. The conventional plastic strain intercept at one reversal and the  $\epsilon_p$  - life slope values are used in the analysis. Specific measured values from the literature are used when available and typical values when they are not available.

6. Analysis or Test Spectrum - The last information needed is the order and magnitude of application of the spectrum used in the test.

## SECTION IV

### OUTPUT OPTIONS

The Computer Program prints the following output in the process of the analysis.

1. Maximum and minimum applied stress and local stress response through the spectrum. Also printed out is the residual stress, equilibrium stress, applied cycles, and the equilibrium period.

2. The elastic local stress history as input into the Range Pair Subroutine and the resulting Range Paired spectrum.

3. The plastic strain occurrence during the spectrum and the damage associated with each strain reversal.

4. The accumulated damage associated with the plastic strains.

5. The Range Paired elastic stress spectrum and the damage associated with each level.

6. The accumulated damage associated with the current block of loading including the plastic strain damage and the total damage since the initiation of cycling.

At the option of the analyst, he can print out all the above items or only two. The IPRINT value controls what data is printed.

If IPRINT = 1, all six items are printed for each flight or block.

If IPRINT = 2, all items except 2. above are printed. /

If IPRINT = 3, only items 4. and 6. above are printed.

#### REFERENCES

1. Crews, J. H., Jr., "Crack Initiation at Stress Concentrations as Influenced by Prior Local Plasticity," Achievement of High Fatigue Resistance in Metals and Alloys, ASTM STP 467, 1970, pp. 37-52.
2. Wetzel, R. M., "Smooth Specimen Simulation of the Fatigue Behavior of Notches," Journal of Materials, JMLSA, Vol. 3, No. 3, Sept. 1968, p. 646.
3. Wetzel, R. M., Morrow, JoDean, and Topper, T. H., Fatigue of Notched Parts with Emphasis on Local Stresses and Strains, NADC-ST-6818, Naval Air Development Center, Johnsville, Pa., Sept. 1968.
4. Impellizzeri, L. F., "Cumulative Damage Analysis in Structural Fatigue," Effects of Environment and Complex Load History on Fatigue Life, ASTM STP 462, American Society for Testing and Materials, 1970, pp. 40-68.
5. Stadnick, S. J., Simulation of Overload Effect in Fatigue Based on Neuber's Analysis, M. S. Thesis, Dept. of Theoretical and Applied Mechanics, University of Illinois, Urbana, 1969. See also T.A.M. Report No. 325.
6. Potter, J. M., "The Effect of Load Interaction and Sequence on the Fatigue Behavior of Notched Coupons," in Cyclic Stress-Strain Behavior-Analysis, Experimentation, and Failure Prediction, ASTM STP 519, American Society for Testing and Materials, 1973, pp. 109-132.

7. Smith, C. R., "Prediction of Fatigue Failures in Aluminum Alloy Structures," Proceedings of the SESA, Vol. 12, No. 2, p. 21, 1955.
8. Neuber, H., "Theory of Stress Concentration for Shear Strained Prismatical Bodies with Arbitrary Non-Linear Stress-Strain Law," Transactions of the ASME, Journal of Applied Mechanics, Dec. 1961, p. 544.
9. Potter, J. M., An Experimental and Analytical Study of Spectrum Truncation Effects, AFFDL-TR-73-117, Air Force Flight Dynamics Laboratory, Dayton, Ohio, 1973 (AD ).
10. Potter, J. M., and Noble, R. A., Crack Initiation Life Predictions Using the Sequence Accountable Fatigue Analysis, TM-73- -FBR- , Air Force Flight Dynamics Laboratory, Dayton, Ohio (Report in Preparation.)
11. Tischler, V. A., A Computer Program for Counting Load Spectrum Cycles Based on the Range Pair Cycle Counting Method, TM-FBR-72-4, Air Force Flight Dynamics Laboratory, Dayton, Ohio, Nov. 1972.
12. Grover, H. J., Fatigue of Aircraft Structures, NAVAIR 01-1A-13, Naval Air Systems Command, Washington, D. C., 1966.
13. Metallic Materials and Elements for Aerospace Vehicle Structures, MIL-HDBK-5A, Department of Defense, Washington, D. C., Feb. 1966.

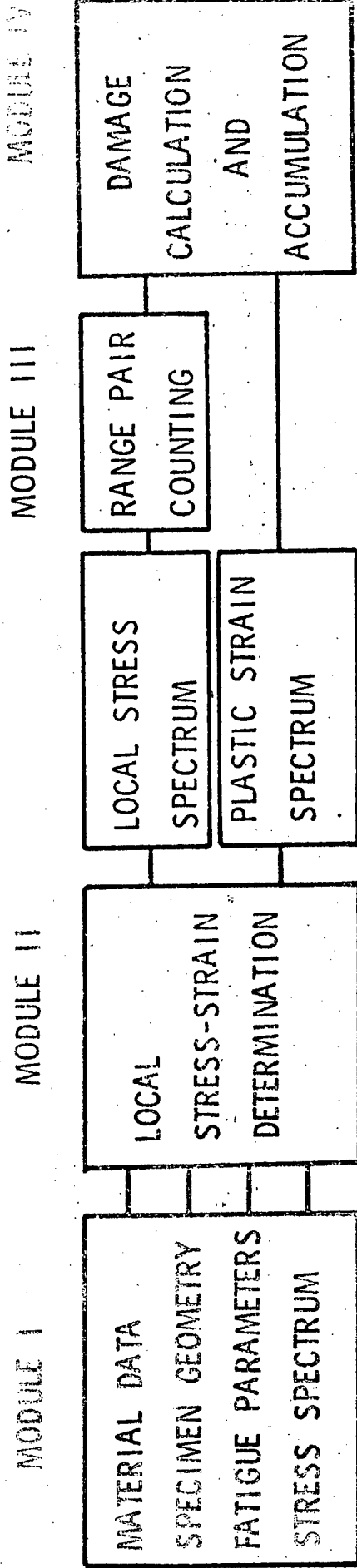


FIGURE 1. PROCEDURE USED IN THE SEQUENCE ACCOUNTABLE FATIGUE ANALYSIS

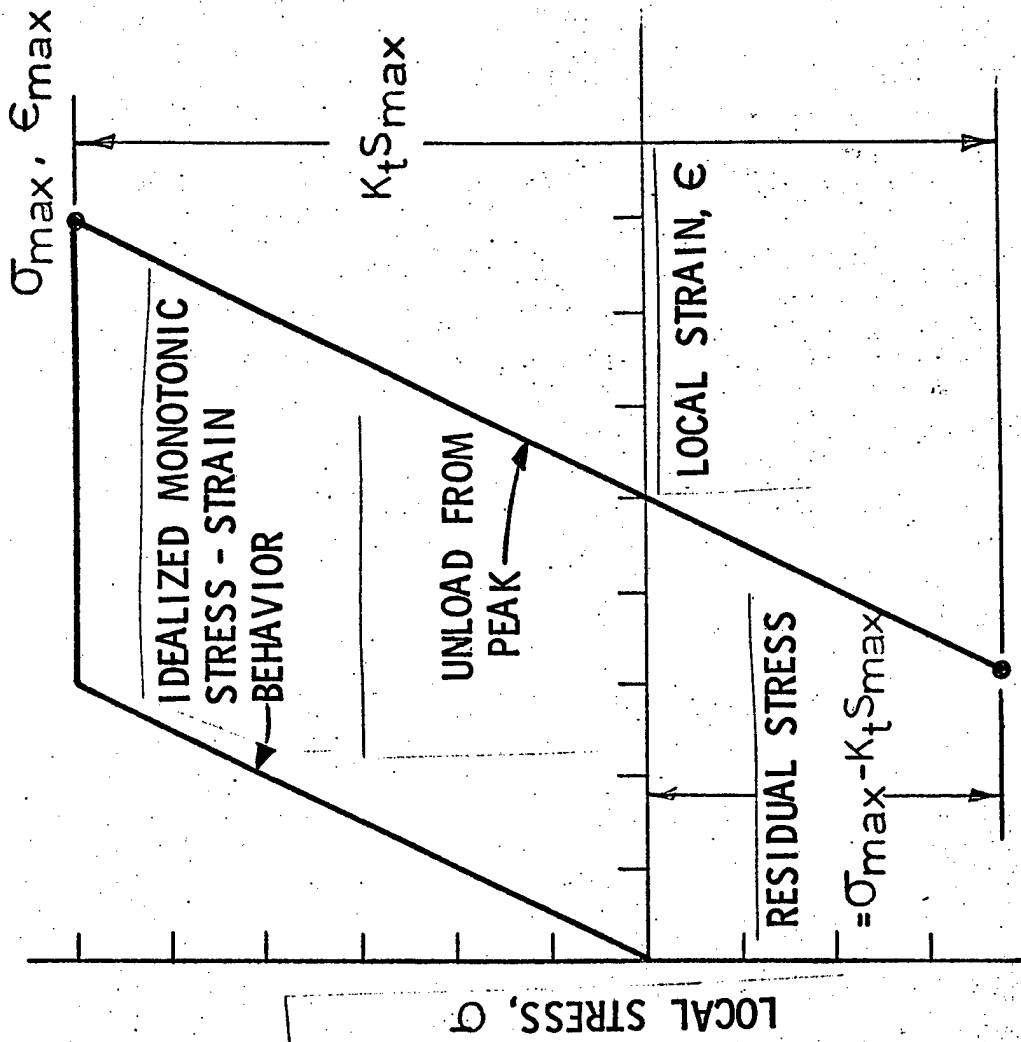


FIGURE 2. METHOD OF DETERMINING THE RESIDUAL STRESS FOLLOWING A PEAK LOAD

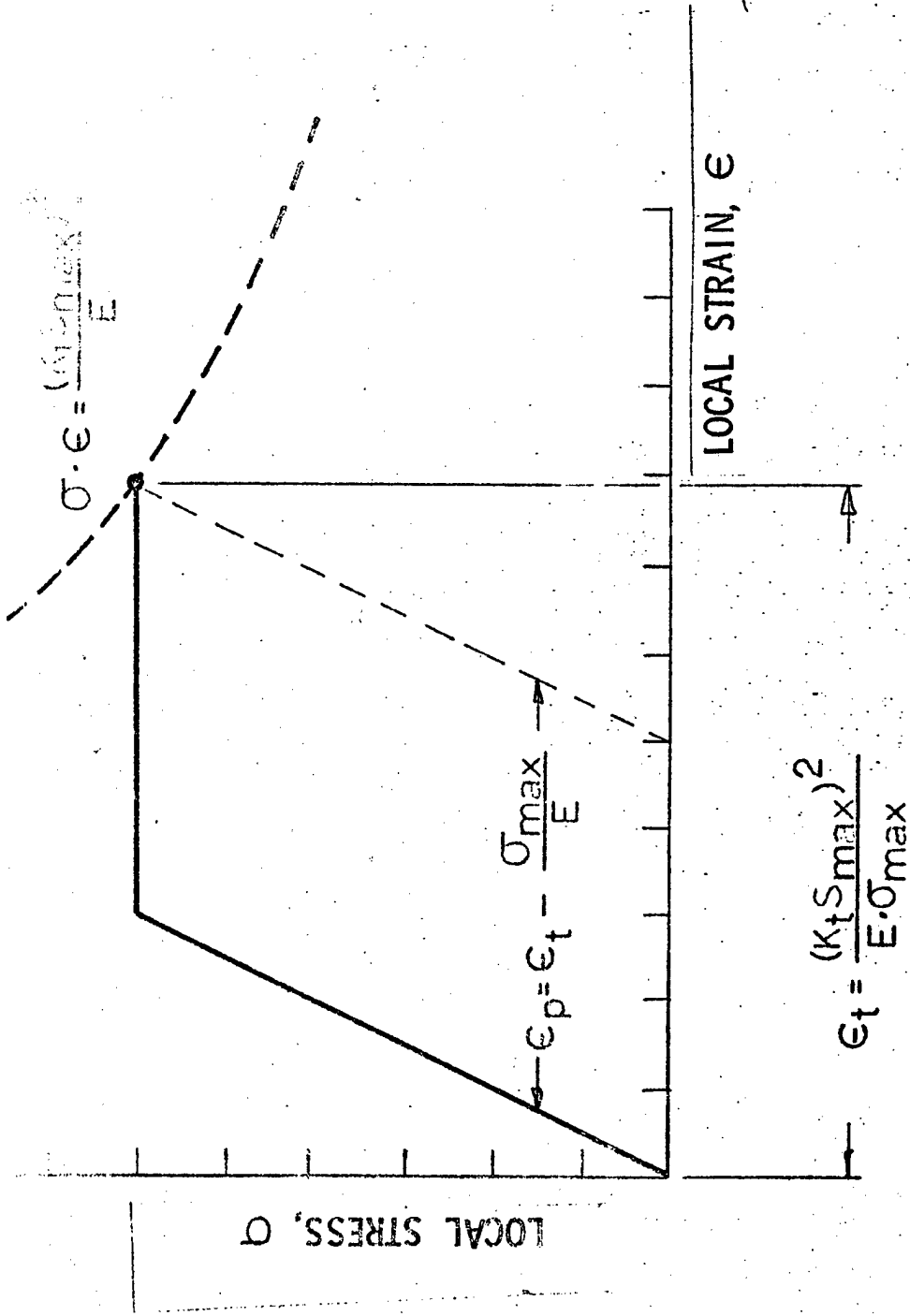
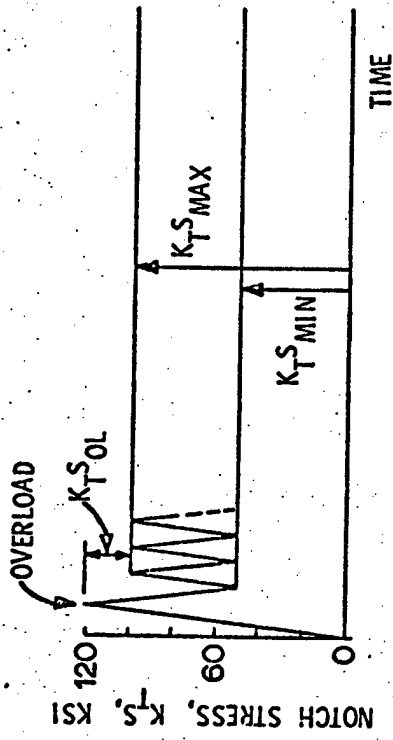
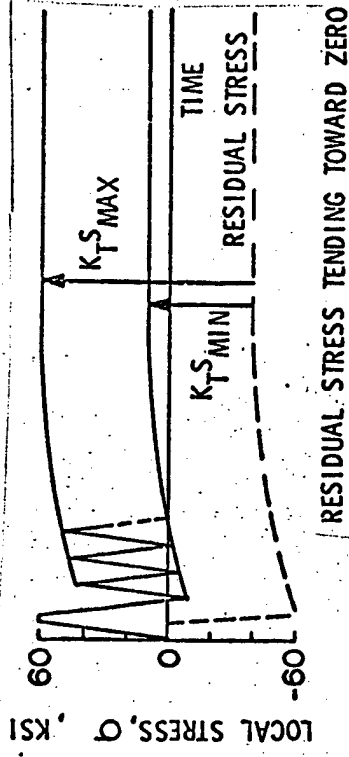


FIGURE 3. METHOD OF DETERMINING PLASTIC STRAIN LEVELS



APPLIED STRESS HISTORY



LOCAL STRESS RESPONSE

FIGURE 4. LOCAL STRESS RESPONSE FOR APPLIED CONSTANT AMPLITUDE LOADING WITH RESIDUAL STRESS RELAXATION

PROGRAM SAL(CARD) TAPES INPUT, (INPUT, TAPES OUTPUT)

\*\*\*\*\*

MODULE 1  
INPUT ROUTINE FOR THE  
SEQUENCE ACCOUNTABLE FATIGUE ANALYSIS

\*\*\*\*\*

PROGRAM DEVELOPED BY JOHN V. POTTER AND ROBERT A. NOBLE  
OF THE AIR FORCE FLIGHT DYNAMICS LABORATORY  
MARCH 1973  
ADDRESS: AFFDL/F33  
WRIGHT-PATTERSON AFB, OHIO 45433  
TELEPHONE (513) 255-5104 OR 255-6105

INPUT

DATA CARD 1. NDECK = THE NUMBER OF DATA DECKS TO BE RUN SEQUENTIALLY.  
IPRINT THE VALUE CONTROLLING THE WRITE STATEMENTS

- 1 PERMITS MAXIMUM PRINTOUT
  - 2 SUPPRESSES RAINFLOW PRINTING
  - 3 MAXIMUM SUPPRESSION OF PRINTOUT
- IRPCHE = THE VALUE CONTROLLING THE ENTRY INTO THE  
RANGE PAIR COUNTING SUBROUTINE
- 1 ENTER RANGE PAIR COUNTING SUBROUTINE
  - 2 SKIP RANGE PAIR COUNTING SUBROUTINE

FORMAT 314

EACH DATA DECK CONTAINS THE FOLLOWING CARDS --

CARD 1. TEST IDENTIFYING INFORMATION  
FORMAT 8AR

CARD 2. TM = MATERIAL TYPE  
TYS = TENSILE YIELD STRESS (KSI)  
EPS0 = LCF STRAIN INTERCEPT  
COFMAN = INVERSE OF COFFIN-MANSON SLOPE  
ELMOD = MODULUS OF ELASTICITY (KSI)  
FORMAT 2AR,3F18.5,F10.2

CARD 3.....6. A(N) N=4,7 (A,B,C ARE COEFFICIENTS OF SECOND  
ORDER LEAST SQUARE FIT OF S-N DATA,  
FOR CURVE OF 10\*\*N CYCLES.)  
B(N)  
C(N)  
TITLE1 (STMAX = A(N)\*STMIN\*\*2 + B(N)\*STMIN + C(N))  
TITLE2 (TITLE1, TITLE2 IDENTIFIES THE )  
N (SOURCE OF THE S-N DATA )  
MATERIAL TYPE (PUNCHED IN COLUMN 72. FOR INFO.)  
FORMAT 3F18.5,2A8

CARD 7. C1 (CONSTANTS TO BE USED IN CALCULA- )

APPENDIX I  
PROGRAM LIST

E1 (TION OF EQUILIBRIUM PERIOD, ENEP.)  
 E2 (ENEP=CI/(KTSMAX\*\*E1\*KTSHEAN\*\*E2))  
 FORMAT 3F18.5

60 CARD 9. AKT = STRESS CONCENTRATION FACTOR USED THE FIRST  
 TIME THROUGH THE PROGRAM  
 ZKT = STRESS CONCENTRATION FACTOR USED THE LAST  
 TIME THROUGH THE PROGRAM  
 DELAKT = STEP CHANGE IN KT BETWEEN RUNS  
 FORMAT 3F18.5

65 NBLOCK = NUMBER OF BLOCKS (NO. OF TIMES TO REPEAT LIST  
 OF LOADS)

70 NLEVEL = NUMBER OF LOADS  
 NTYPE = NUMBER OF TYPES OF LOADS  
 FORMAT 3I10

75 TLL = LIMIT LOAD  
 FORMAT F18.5

CARDS 11, ..., NLEVEL + 10.  
 ISTEP(K) = K = THE KTH STEP (K=1, NLEVEL)  
 ITYPE(K) = THE IDENTIFYING TYPE OF THE KTH LOAD  
 STMIN(K) = THE KTH MINIMUM (DECIMAL FRACTION OF TLL)  
 STMAX(K) = THE KTH MAXIMUM (DECIMAL FRACTION OF TLL)  
 ENN(K) = NUMBER OF CYCLES AT THE KTH LOAD  
 FORMAT 2(I4,2X),3(F18.5,1X)

85 CARDS NLEVEL + 11, ..., NLEVEL + 10 + NBLOCK.  
 IRLCK(JJ) = JJ = THE JJTH BLOCK (JJ=1, NBLOCK)  
 NN(JJ,1) = TYPE OF LOAD INCLUDED IN JJTH BLOCK  
 NN(JJ,2) = (THERE WILL BE ONE NN VALUE)  
 NN(JJ, ) = (ON THE CARD FOR EACH DIFFERENT)  
 FORMAT 9I5 (TYPE OF LOAD INCLUDED IN THE)  
 (JJTH BLOCK)

95 COMMON/MSAL/RNCYC(200), KPMAX, IPRINT  
 DIMENSION STMIN(200), STMAX(200), ENN(200), ITYPE(200)  
 COMMON/MCOR1/NLEVEL, IRPCM, IPLOT, ELMOD, IYS, EPSD, COFMAN, C1, E2, E1,  
 \*RES(200), AKT, SUMENN, SUMNC  
 COMMON/MCOR2/STMIN(200), STMAX(200), ENN(200), A(10), B(10), C(10),  
 \*R(10), ISTEP(200), NN(200,10), IRLCK(200)  
 READ(5,4) NDECK, IPRINT, IRPCM  
 FORMAT(3I4)

100 WRITE(6,3) NDECK  
 FORMAT(1H1,14,32H DATA DECKS ARE TO BE PROCESSED.)  
 IF(IRPCM.GE.2) GO TO 6  
 WRITE(6,11)  
 GO TO 13

105 5 WRITE(6,7)  
 7 FORMAT(25H NO COUNTING METHODS USED)  
 13 CONTINUE

DO 595 LMN = 1, NDECK  
 WRITE(6,5)  
 5 FORMAT(40H SEQUENCE ACCOUNTABLE FATIGUE EVALUATION)

READ(5,10) TM1, TM2, TVS, EPSO, COFMAN, ELMON  
 FORMAT(8A4)  
 WRITE(6,8) I1, I2, I3, I4, I5, I6, I7, I8  
 FORMAT(16H) SPECTRUM PROP, 9A8)

INPUT OF DATA PECULIAR TO A MATERIAL

10 READ(5,10) TM1, TM2, TVS, EPSO, COFMAN, ELMON  
 FORMAT(2A8, 3F18.5, 5I8.2)  
 WRITE(6,12) TM1, TM2, TVS, EPSO, COFMAN, ELMON  
 FORMAT(18H) MATERIAL TYPE --, 2A8, //10H TENSILE YIELD STRESS (KSI)  
 \*, F18.5, //23H LCF STRAIN INTERCEPT =, F18.5, //31H INVERSE OF COEFF  
 \* N-MANSON SLOPE, F18.5, //14H ELASTIC MODULUS =, F18.5)  
 READ(5,14) (A(N), B(N), C(N), TITL1, TITL2, VE4, 7)  
 FORMAT(3F18.5, 2A8)  
 WRITE(6,16)  
 FORMAT(58H) COEFFICIENTS OF SECOND ORDER LEAST SQUARE FIT OF S-N DA  
 \*TA)  
 WRITE(6,18)  
 FORMAT(44H) SMAX = A(I)\*SMIN\*\*2 + B(I)\*SMIN + C(I))  
 WRITE(6,20)  
 FORMAT(5X, 5H LIFE, 10X, 5H A(I), 14X, 5H B(I), 14X, 5H C(I))  
 WRITE(6,22) (A(N), B(N), C(N), N=4, 7)  
 FORMAT(19H) 10\*\*I2, 3F18.5)  
 WRITE(6,55) TITL1, TITL2  
 FORMAT(39H) MONTCHOP COUPON S-N DATA DERIVED FROM 728H INFORMATION  
 \* SUPPLIED FROM ,2A)  
 READ(5,14) C1, E1, E2  
 WRITE(6,24)  
 FORMAT(36H) RESIDUAL STRESS RELAXATION FUNCTION)  
 WRITE(6,26)  
 FORMAT(742H) ENEP = G1/(KTSYMAX\*\*E1 \* KYSMEAN\*\*E2))  
 WRITE(6,28) C1, E1, E2  
 FORMAT(13H) WHERE G1 =, E15.8, 9H, E1 =, F10.3, 10H AND E2 =, F10.  
 \*3)

INPUT OF DATA PECULIAR TO A SEQUENCE

READ(5,65) AKT  
 FORMAT(F18.5)  
 READ(5,32) NBLOCK, JLEVEL, NTYPE  
 FORMAT(3I10)  
 WRITE(6,34) NBLOCK, JLEVEL  
 FORMAT(// 110, 23H TIMES THROUGH BLOCK OF, I10, 6H LOADS)  
 READ(5,35) TLL  
 FORMAT(F18.5)  
 WRITE(6,33) TLL  
 FORMAT(// 5X, 13H LOAD LIMIT =, F18.5)  
 READ(5,36) (ISTEP(K), ITYPE(K), RYTHIN(K), RYMAX(K), RMIN(K), KEI, JLEVEL)  
 FORMAT(I4, 2X, I4, 2X, I4, 2X, F18.5, 1X, F18.5, 1X, F18.5, 1X, 5, 1X)  
 WRITE(6,38)  
 FORMAT(// 11H STEP TYPE, 10X, 6H STMIN, 14X, 6H STMAX, 15X, 4H ENN)

```

WRITE(6,36) (ISTEP(K), ITYPE(K), RTMIN(K), RTMAX(K), RNN(K), K=1, JLEVEL)
WRITE(6,39)
39  FORMAT(/,47H 9LOCK TYPE TYPE TYPE TYPE TYPE TYPE TYPE TYPE/)
    DO 42 JJ=1,NBLOCK
170  READ(5,40) IBLOCK(JJ), (NN(JJ, KK), KK=1, NTYPE)
    WRITE(6,40) IBLOCK(JJ), (NN(JJ, KK), KK=1, NTYPE)
40  FORMAT(3I5)
42  CONTINUE
    SUMENN=0.
175  SUMNC=0.
    RES(1)=0.
    WRITE(6,8) T1, T2, T3, T4, T5, T6, T7, T8
    WRITE(6,51) (AKT)
51  FORMAT(7H AKT = , (F6.2))
    WRITE(6,55) C1
55  FORMAT(/,24H RELAXATION CONSTANT C1=, F15.2)
    IF(IRPCM,55.2) GO TO 59
    WRITE(6,11)
11  FORMAT(56H0SPECTRUM SUBJECTED TO THE RANGE-PAIR COUNTING TECHNIQUE)
185  *1
59  CONTINUE
    DO 1002 KFL=1,NBLOCK
    JJJ=1
190  DO 60 J=1, JLEVEL
    DO 70 KK=1, NTYPE
    IF(NN(KFL, KK) .EQ. 0) GO TO 60
    IF(ITYPE(J), E0, NN(KFL, KK)) GO TO 150
70  CONTINUE
150  STMIN(JJJ)=RTMIN(J) * TLL
    STMAX(JJJ)=RTMAX(J) * TLL
    ENN(JJJ)=RNN(J)
    JJJEJJJ+1
50  CONTINUE
    NLEVEL=JJJ-1
200  CALL CORE(KFL)
    IPRINT=2
1002 CONTINUE
597 CONTINUE
595 CONTINUE
596 CONTINUE
598 STOP
    END

```

```

SUBROUTINE CORE(KPL)
*****
CORE 2
CORE 3
CORE 4
CORE 5
CORE 6
CORE 7
CORE 8
STD 1
CORE 10
CORE 11
CORE 12
CORE 13
CORE 14
CORE 15
CORE 16
CORE 17
CORE 18
CORE 19
CORE 20
CORE 21
CORE 22
CORE 23
CORE 24
CORE 25
CORE 26
CORE 27
CORE 28
CORE 29
CORE 30
CORE 31
CORE 32
CORE 33
CORE 34
CORE 35
CORE 36
CORE 37
CORE 38
CORE 39
CORE 40
CORE 41
CORE 42
CORE 43
CORE 44
CORE 45
CORE 46
CORE 47
CORE 48
CORE 49
CORE 50
CORE 51
CORE 52
CORE 53
CORE 54
CORE 55
CORE 56

*****
CORE PROGRAM OF
THE SEQUENCE ACCOUNTABLE FATIGUE ANALYSIS

*****
MODULE II

*****
LOGAL STRESS AND STRAIN DETERMINATION

*****
COMMON/MDG1/SIGMAX(200),SIGMIN(200)
COMMON/MSAL/ENCCYC(200),KPMAX,IPRINT
DIMENSION PLSTRA(200),FN(200),AVSGMX(200),AVSGMN(200),EX(200),
*ENCCYC(200),ASX(200),ASH(200),ASE(200)
COMMON/MCOR1/NLEVEL,IPPCN,IPLOT,ELMOD,TYS,EPSD,COFMAN,G1,E2,E1,
*RES(200),AKT,SUMENR,SUMNC
COMMON/MCOR2/STMIN(200),STMAX(200),FHN(200),A(10),C(10),
*R(10),ISTEP(200),NN(200,10),LOCK(200)
JJ=4FL
IF(JJ.GT.3)IPRINT=3
IRAIN=1
WRITE(6,54)JJ
54 FORMAT(20H FLIGHT OR BLOCK NO.,I5)
IF(IPRINT.GE.3) GO TO 51
WRITE(6,52)
52 FORMAT(5H STMAX STMIN SIGMAX SIGMIN RES EPRES FNN
NEP)
51 CONTINUE
KPL0Y=1
DO 570 J=1,NLEVEL
149 I=J+1
PLSTRA(J)=0.
570
50
51
52
53
54
55
56
DETERMINE SEQUENCE ACCOUNTABLE RESIDUAL STRESS AND CORRESPONDING
PLASTIC STRAIN
150 ASHAX=AKT*STMAX(J)
ASMIN=AKT*STMIN(J)
ASX(KPL0Y)=ASMAX
ASN(KPL0Y)=ASMIN
ASE(KPL0Y)=ENN(J)
KPL0Y=KPL0Y+1
ASMEAN=(ASMAX+ASMIN)/2.
JA=0
170 IF(RES(I-1)+ASMIN+TYS) 190,190,200
190 JA=1
AXA=RES(I-1)+ASMIN
PLSTRA(J)=-1.*(AXA/ELMOD)*(1.+AAA/TYS)
J6=0
200
202 IF(RES(I-1)+ASMAX-TYS) 220,210,210

```

```

210 JB=-1
AAA=ASMAX
BB=IYS-RES(I-1)
IF(BB>.5E. IYS) GO TO 214
212 BRB=0.
214 PLSTRA(J)= AAA*AAA/(ELMOD*IYS)-BBB*BBB/(ELMOD*IYS)
220 IF(JA+JB) 210,250,240
230 RES(I)=IYS-ASMAX
GO TO 290
240 RES(I)=-IYS-ASHIN
GO TO 290
250 IF(JA) 260,260,270
260 RES(I)=RES(I-1)
GO TO 290
270 RES(I)=-ASMEAN
290 SIGMAX(IRAIN)=RES(I)+ASMAX
SIGMIN(IRAIN)=RES(I)-ASMIN
RNOYC(IRAIN)=ENN(J)
IF(ASHMAX.LE.IYS) GO TO 410
75 EORES=-ASMAX+IYS
GO TO 440
110 IF(ASHIN.GE.-IYS) GO TO 430
420 EORES=-ASMIN-IYS
GO TO 440
430 EORES=0.
440 DIFF=RES(I)-EORES
C
C
C
85 ABMAX=ABS(ASMAX)
ABMIN=ABS(ASMIN)
ABMEAN=ABS(ASMEAN)
IF(ABMAX.LT.1.) ABMAX=1.
IF(ABMEAN.LT.1.) ABMEAN=0.5
IF(ABMIN-ABMAX) 444,444,442
442 ABM=ABMIN
GO TO 446
444 ABME=ABMAX
446 ENEP=C1/(ABM**E1*ABMEAN**E2)
IF(IPRINT.GE.3) GO TO 3F1
WRITE(6,350) STMAX(J), STMIN(J), SIGMAX(IRAIN), SIGMIN(IRAIN),
350 *RES(I), EORES, ENN(J), ENEP, J
FORMAT(6(F7.2,1X),F6.2,1X,E15.8,5X,16)
351 CONTINUE
C
C
C
CALCULATE RESIDUAL STRESS RELAXATION
IRAIN=IRAIN+1
ABDIF=ABS(DIF)
GO TO 350
GO TO 550
360 IF(ABDIF.LT.5.) GO TO 560
370 IF(1000.*ENN(J).LT.ENEP) GO TO 560
IF(ENN(J).LE.10.) GO TO 560
110 380 NFLAG=0
CORE 57
CORE 58
CORE 59
CORE 60
CORE 61
CORE 62
CORE 63
CORE 64
CORE 65
CORE 66
CORE 67
CORE 68
CORE 69
CORE 70
CORE 71
CORE 72
CORE 73
CORE 74
CORE 75
CORE 76
CORE 77
CORE 78
CORE 79
CORE 80
CORE 81
CORE 82
CORE 83
CORE 84
CORE 85
CORE 86
CORE 87
CORE 88
CORE 89
CORE 90
CORE 91
CORE 92
CORE 93
CORE 94
CORE 95
CORE 96
CORE 97
CORE 98
CORE 99
CORE 100
CORE 101
CORE 102
CORE 103
CORE 104
CORE 105
CORE 106
CORE 107
CORE 108
CORE 109
CORE 110
CORE 111

```

```

112 CORE
113 CORE
114 CORE
115 CORE
116 CORE
117 CORE
118 CORE
119 CORE
120 CORE
121 CORE
122 CORE
123 CORE
124 CORE
125 CORE
126 CORE
127 CORE
128 CORE
129 CORE
130 CORE
131 CORE
132 CORE
133 CORE
134 CORE
135 CORE
136 CORE
137 CORE
138 CORE
139 CORE
140 CORE
141 CORE
142 CORE
143 CORE
144 CORE
145 CORE
146 CORE
147 CORE
148 CORE
149 CORE
150 CORE
151 CORE
152 CORE
153 CORE
154 CORE
155 CORE
156 CORE
157 STD
158 CORE
159 CORE
160 CORE
161 CORE
162 CORE
163 CORE
164 CORE
165 CORE
166 CORE

DELAG=2.10
IRAIN=IRAIN-1
DUMMY=EEN(J)
450 IF(DUMMY-ENEP) 470,460,460
460 DUMMY=DUMMY/2.
115 NFLAG=NFLAG+1
    GO TO 450
470 CYCINT=DUMMY/10.
    DO 530 <=1,10
    DECK=FLOAT(K)
    EN(K)=CYCINT*DFCK
480 IF(K=0,1) GO TO 490
    EX(K)=EXP(-2.303*EN(K-1)/ENEP)+EXP(-2.303*EN(K)/ENEP)
    GO TO 500
125 EX(K)=1.+EXP(-2.303*EN(K)/ENEP)
    CONTINUE
500 IF(NFLAG.EQ.0) GO TO 530
    NFLAG=NFLAG+10
510 DO 520 <=1,NFLAG
    EN(K)=2.*DUMMY
    EX(K)=EXP(-2.303*EN(K-1)/ENEP)+EXP(-2.303*EN(K)/ENEP)
    DUMMY=2.*DUMMY
    CONTINUE
520
530 DO 559 K=1,NFLAG
    AVSGMX(K)=ASMAX+QRES+QF*EX(K)/2.
    AVSGMH(K)=AVSGMX(K)-A*SPAX+AS*H
    SIGMIN(IRAIN)=AVSGMH(K)
    SIGMAX(IRAIN)=AVSGMX(K)
    RNCYC(IRAIN)=EH(K)
545 IF(K.EQ.1) GO TO 561
539 RNCYC(IRAIN)=RNCYC(IRAIN)-EN(K-1)
540 IF(IPRINT.GE.3) GO TO 561
    WRITE(6,550) SIGMAX(IRAIN),SIGMIN(IRAIN),EN(K),RNCYC(IRAIN)
550 FORMAT(16X,2(F7.2,1X),16X,F6.2,17X,F15.8)
    CONTINUE
145 IRAIN=IRAIN+1
    CONTINUE
559 CONTINUE
560 RES(I)=QRES+QF*EXP(-2.303*EN(J)/ENEP)
150 CONTINUE
570 RES(I)=RES(J)
569 IN=IRAIN-1
    *****
155 *****
    MODULE III
    CYCLE COUNTING TECHNIQUE
    *****
160 *****
    CALL SUBROUTINE TO RANGE PAIR COUNT SPECTRUM
    IF(IRPCH.GT.1) GO TO 591
    CALL RPSMTIN
    GO TO 592

```



```

225      220  N=4
          DO 330 M=1,4
          IF (Y.GE.TYS) GO TO 376
          CYCLE=310.**8.
          GO TO 340
          CORE 222
          CORE 223
          CORE 224
          CORE 225
          CORE 226
          CORE 227
          CORE 228
          CORE 229
          CORE 230
          CORE 231
          CORE 232
          CORE 233
          CORE 234
          CORE 235
          CORE 236
          CORE 237
          CORE 238
          CORE 239
          CORE 240
          CORE 241
          CORE 242
          CORE 243
          CORE 244
          CORE 245
          CORE 246
          CORE 247
          CORE 248
          CORE 249
          CORE 250
          CORE 251
          CORE 252
          CORE 253
          CORE 254
          CORE 255
          CORE 256
          CORE 257
          CORE 258
          CORE 259
          CORE 260
          CORE 261
          CORE 262
          CORE 263
          CORE 264
          CORE 265

230      330  CONTINUE
          IF (R(7)*R(4)) 335,338,334
          A9P7=ABS(R(7))
          A9F4=ABS(R(4))
          IF (A9P7.LE.A9F4) GO TO 335
          EXPO=4+R(4)/R(7)-R(5)
          GO TO 339
          EXP0=7+R(7)/R(6)-R(7)
          GO TO 339
          SUM2=R(4)+R(5)+R(6)+R(7)
          SUMR2=R(4)**2+R(5)**2+R(6)**2+R(7)**2
          SUMR3=R(4)**3+R(5)**3+R(6)**3+R(7)**3
          SUMR4=R(4)**4+R(5)**4+R(6)**4+R(7)**4
          SUMRN=4.**R(4)+5.**R(5)+6.**R(6)+7.**R(7)
          SUMR2N=4.**R(4)**2+5.**R(5)**2+6.**R(6)**2+7.**R(7)**2
          DEL1=4.**SUMR2*SUMR4-4.**SUMR3**2
          DEL2=SUMR*SUMR2*SUMR3-SUMR4*SUMR**2
          DEL3=SUMR*SUMR2*SUMR3-SUMR2**3
          D01=2.**SUMR2*SUMR4-22.**SUMR3**2
          D02=SUMR2**SUMR3*SUMRN-SUMR3*SUMR3*SUMRN
          D03=SUMR*SUMR3*SUMR2N-SUMR2N*SUMR2**2
          EXPO=(D01+D02+D03)/(DEL1+DEL2+DEL3)
          CYCLE=310.**EXPO
          IF (EXPO.LE.4.) CYCLES=10.**4.
          ENNYC(JKL)=ENNYC(JKL)/CYCLES
          SUMNC=SUMNC+ENNYC(JKL)
          SUMDEL=SUMDEL+ENNYC(JKL)
          IF (IPRINT.GE.3) GO TO 600
          WRITE(6,599) Y,X,RNYC(JKL),CYCLES,ENNYC(JKL)
          FORMAT(15X,2(F7.2,1X),15X,F5.0,17X,2(E15.8,1X))
          CONTINUE
          WRITE(6,593)SUMDEL
          WRITE(6,593)SUMDEL
          FORMAT(/69X,21H DAMAGE PER THIS SET=,E15.3)
          WRITE(6,575) SUMNC
          WRITE(6,575)SUMNC
          FORMAT(/69X,18H TOTAL ENN/CYC =,E15.8)
          END
  
```

ARRAY NAME	DEFINITION	DIMENSION
SIGMAX	PEAKS OF THE INPUT LOAD SPECTRUM	NPKS + KK
SI5MIN	VALLEYS OF THE INPUT LOAD SPECTRUM	NPKS + KK
RNCYC	K COUNTERS OF THE PEAKS AND VALLEYS	NPKS + KK
NSTEP	STEP NUMBERS OF THE INPUT SPECTRUM	2*NPKS
RES	RESIDUE SPECTRUM	2*NPKS
INDEX	STEP NUMBERS OF ELEMENTS IN RES	NPKS + KK
CYCLE	RANGE PAIR COUNTED CYCLES	NPKS + KK
RNECYS	K COUNTERS OF THE CYCLES OF THE	NPKS + KK
NNSTEP	UNSORTED ANALYSIS SPECTRUM	NPKS + KK
ISAVE	UNSORTED ANALYSIS SPECTRUM	99
	VALUES OF NSTEP(J) SUCH THAT RNCYC(J)	
	IS < 1.0 AND VALUES OF NSTEP(J) SUCH	
	THAT SIGMAX(J-1) = SIGMAX(J) AND	
	SIGMIN(J-1) = SIGMIN(J)	
	COMMON/MSAL/RNCYC(200),KPMAX,IPRINT	
	COMMON/MSAL/SIGMAX(200),SIGMIN(200)	
	COMMON/MSAL/NSTEP(900),LR,KMAX,KMIN,K31	
	COMMON/MSAL/RES(1400),INDEX(1400),IND1,IND2,IND3,IND4,KIND	
	COMMON/MSAL/CYCLE(900,2),RNECYC(900),NNSSTEP(900)	
	COMMON/MSAL/L,LIND	
	DIMENSION ISAVE(99),TITLE(8)	
	9999 NPUNCH = 0	
	DO 8000 I = 1, NPKS	
	8000 NSTEP(I) = I	
	IF(I.PRINT.GE.2) GO TO 103	
	WRITE(6,20) NPKS	
	20 FORMAT(1H0,60THE NUMBER OF PEAKS OR VALLEYS IN THE INPUT LOAD SPE	
	CTRUM = ,15//)	
	WRITE(6,22)	
	22 FORMAT(60X,5HSIGMA/31X,4HSTEP,13X,7HMAXIMUM,16X,7HMINIMUM,13X,	
	9HCOUNTER K2)	
	WRITE(5,25) (NSTEP(I),SIGMAX(I),SIGMIN(I),RNCYC(I), I = 1, NPKS)	
	25 FORMAT(29X,15,10X,13.6,10X,13.6,10X,F10.5)	
	103 CONTINUE	
	SORT THROUGH THE LOAD SPECTRUM - PULL OUT THOSE PEAKS AND VALLEYS	
	COUNTER K IS LESS THAN 1.0	
	J = 1	
	L = 0	

25

```

NPKN = I
RNYNO = 100
JMAX = J
60 DO 100 I = 1, NPKN
   IF (RNYC(I) .GE. 1.0) GO TO 100
   X1 = SIGMAX(I)
   X2 = SIGMIN(I)
   CALL GYGEN(X1, X2, RNYC(I), NSTEP(I))
   ISAVE(J) = I
   J = J + 1
55 CONTINUE
   JMAX = J - I
   NPKN = NPKN - JMAX
   IF (JMAX .EQ. 0) GO TO 200
   WRITE(6, 23) (ISAVE(K), K = 1, JMAX)
   IF (IPRINT .GE. 2) GO TO 101
   23 FORMAT(1H0, 32STEP NUMBERS OF THOSE PEAKS AND VALLEYS IN THE LOAD
      15 ELECTRON WHOSE COUNTER K IS LESS THAN 1.3/(1717))
   101 CONTINUE
75 DO 110 J = 1, JMAX
   I = ISAVE(J) - (J-1)
   NPKN = NPKN - J
   IF (I .EQ. NPKN) GO TO 110
   DO 115 II = I, NPKN
     SIGMAX(II) = SIGMAX(II+1)
     SIGMIN(II) = SIGMIN(II+1)
     NSTEP(II) = NSTEP(II+1)
     RNYC(II) = RNYC(II+1)
115 CONTINUE
110 CONTINUE
200 CONTINUE
90 CONTINUE
   SOFT THROUGH THE LOAD SPECTRUM DATA - COMBINE STEPS WITH IDENTICAL
   AND VALLEYS WHICH OCCUR CONSECUTIVELY
   J = 1
95 DO 300 I = 2, NPKN
   IF (SIGMAX(I) .NE. SIGMAX(I-1)) GO TO 300
   IF (SIGMIN(I) .NE. SIGMIN(I-1)) GO TO 300
   ISAVE(J) = I
   RNYC(I-1) = RNYC(I-1) + RNYC(I)
   J = J + 1
300 CONTINUE
100 IF (J .EQ. 1) GO TO 6000
   JMAX = J - 1
   DO 311 J = 1, JMAX
     NPKN = NPKN - J
     IF (I .EQ. NPKN) GO TO 311
     DO 315 II = I, NPKN
       SIGMAX(II) = SIGMAX(II+1)
       SIGMIN(II) = SIGMIN(II+1)
       NSTEP(II) = NSTEP(II+1)
       RNYC(II) = RNYC(II+1)
315 CONTINUE
110 CONTINUE
316 CONTINUE

```

301 CORE  
302 CORE  
303 CORE  
304 CORE  
305 CORE  
306 CORE  
307 CORE  
308 CORE  
309 CORE  
310 CORE  
311 CORE  
312 CORE  
313 CORE  
314 CORE  
315 CORE  
316 CORE  
317 CORE  
318 CORE  
319 CORE  
320 CORE  
321 CORE  
322 CORE  
323 CORE  
324 CORE  
325 CORE  
326 CORE  
327 CORE  
328 CORE  
329 CORE  
330 CORE  
331 CORE  
332 CORE  
333 CORE  
334 CORE  
335 CORE  
336 CORE  
337 CORE  
338 CORE  
339 CORE  
340 CORE  
341 CORE  
342 CORE  
343 CORE  
344 CORE  
345 CORE  
346 CORE  
347 CORE  
348 CORE  
349 CORE  
350 CORE  
351 CORE  
352 CORE  
353 CORE  
354 CORE  
355 CORE  
356 CORE  
357 CORE  
358 CORE  
359 CORE  
360 CORE  
361 CORE  
362 CORE  
363 CORE  
364 CORE  
365 CORE  
366 CORE  
367 CORE  
368 CORE  
369 CORE  
370 CORE  
371 CORE  
372 CORE  
373 CORE  
374 CORE  
375 CORE

311 CONTINUE

NPKSN = NPKSN - JMAS

C RANGE PAIR COUNT THE ADJUSTED LOAD SPECTRUM

115 6000 I = 1 CORE 376

KB = 0 CORE 377

L = JMAX CORE 378

KMIN = 0 CORE 379

KMAX = 0 CORE 380

LR = 0 CORE 381

382 CORE 382

383 CORE 383

384 CORE 384

385 CORE 385

386 CORE 386

387 CORE 387

388 CORE 388

389 CORE 389

390 CORE 390

391 CORE 391

392 CORE 392

393 CORE 393

394 CORE 394

395 CORE 395

396 CORE 396

397 CORE 397

398 CORE 398

399 CORE 399

400 CORE 400

401 CORE 401

402 CORE 402

403 CORE 403

404 CORE 404

405 CORE 405

406 CORE 406

407 CORE 407

408 CORE 408

409 CORE 409

410 CORE 410

411 CORE 411

412 CORE 412

413 CORE 413

414 CORE 414

415 CORE 415

416 CORE 416

417 CORE 417

418 CORE 418

419 CORE 419

420 CORE 420

421 CORE 421

422 CORE 422

423 CORE 423

424 CORE 424

425 CORE 425

426 CORE 426

427 CORE 427

428 CORE 428

429 CORE 429

430 CORE 430

1 IF (RNGYC(I) .GT. 1.0) GO TO 400

IF (KB .NE. 0) GO TO 5

X1 = SIGMAX(I)

X2 = SIGMIN(I)

IND1 = NSTEP(I)

IND2 = IND1

I = I + 1

KB = 1

GO TO 1

5 X3 = SIGMAX(I)

X4 = SIGMIN(I)

IND3 = NSTEP(I)

IND4 = IND3

KMIN = 1

KMAX = 0

K31 = 0

IF (RNGYC(I) .EQ. 1.0) GO TO 6

KEY = 1

KIND = 1

GO TO 415

6 KEY = 0

CYCN0 = AINT(RNGYC(I)+0.5)

CALL DECIDE(X1,X2,X3,X4,KEY,I,CYCN0,KCYGEN)

1000 GO TO (10,10,30),KCYGEN

10 K9 = 1

I = I + 1

IF (KMIN .NE. 1) GO TO 36

IF (I .LE. NPKSN) GO TO 5

RES(LR+1) = X1

RES(LR+2) = X2

INDEX(LR+1) = IND1

INDEX(LR+2) = IND2

LRMAX = LR + 2

GO TO 2000

70 IF (KMIN .NE. 1) GO TO 35

I = I + 1

IF (I .LE. NPKSN) GO TO 31

RES(LR+1) = X1

RES(LR+2) = X2

RES(LR+3) = X3

INDEX(LR+1) = IND1

INDEX(LR+2) = IND2

INDEX(LR+3) = IND3

165

27

```

LRMAX = LR + 3
GO TO 200
31 X4 = SIGMAX(I)
IND4 = NSTEP(I)
KMAX = 1
KMIN = 0
K31 = 1
32 IF (RNCYC(I) .GT. 1.0) GO TO 10
GO TO 5
40 KEY = 1
KIND = 0
GO TO 415
35 X4 = SIGMIN(I)
IND4 = NSTEP(I)
KMIN = 1
KMAX = 0
K31 = 0
GO TO 32
36 X3 = SIGMIN(I)
IND3 = NSTEP(I)
KMIN = 1
KMAX = 0
GO TO 12
400 KEY = 1
IF (X3 .NE. 0) GO TO 410
X1 = SIGMAX(I)
X2 = SIGMIN(I)
X3 = SIGMAX(I)
X4 = SIGMIN(I)
IND1 = NSTEP(I)
IND2 = IND1
IND3 = IND1
IND4 = IND1
KMIN = 1
KMAX = 0
K31 = 0
IF (RNCYC(I) .LE. 2.0) GO TO 401
RNCYC(I) = RNCYC(I) - 1.0
GO TO 402
401 RNCYC(I) = RNCYC(I) - 2.0
402 KIND = 0
GO TO 415
410 X3 = SIGMAX(I)
X4 = SIGMIN(I)
IND3 = NSTEP(I)
IND4 = IND3
KMIN = 1
KMAX = 0
K31 = 0
RNCYC(I) = RNCYC(I) - 1.0
KIND = 1
RNCYC(I) = RNCYC(I) - 1.0
K3 = 0
CYCNO = AINT(RNCYC(I)+0.5)
CALL DECIDE(X1,X2,X3,X4,KEY,I,CYCNO,KCYGEN)
GO TO 1000

```

431 CORE  
432 CORE  
433 CORE  
434 CORE  
435 CORE  
436 CORE  
437 CORE  
438 CORE  
439 CORE  
440 CORE  
441 CORE  
442 CORE  
443 CORE  
444 CORE  
445 CORE  
446 CORE  
447 CORE  
448 CORE  
449 CORE  
450 CORE  
451 CORE  
452 CORE  
453 CORE  
454 CORE  
455 CORE  
456 CORE  
457 CORE  
458 CORE  
459 CORE  
460 CORE  
461 CORE  
462 CORE  
463 CORE  
464 CORE  
465 CORE  
466 CORE  
467 CORE  
468 CORE  
469 CORE  
470 CORE  
471 CORE  
472 CORE  
473 CORE  
474 CORE  
475 CORE  
476 CORE  
477 CORE  
478 CORE  
479 CORE  
480 CORE  
481 CORE  
482 CORE  
483 CORE  
484 CORE  
485 CORE

```

2000 LMAX = 1
IF (LRMAX .LT. 4) GO TO 5000
IF (NCYNO .EQ. 0) GO TO 5000

225 C RANGE PAIR COUNT OF RESIDUE SPECTRUMS
C
NRES = NRES + 1
CALL DECRES(LRMAX,NCYNO)
GO TO 2000
230 C 5000 IF (LRMAX .LE. 1) GO TO 3000
C
COUNT THE LAST RESIDUE SPECTRUM - RANGE PAIR COUNTING WILL YIELD N
ADDITIONAL CYCLES
KK = 0
RESMAX = RES(1)
RESMIN = RES(1)
IMAX = 1
IMIN = 1
240 C DO 500 I = 2,LRMAX
IF (RES(I) .LT. RESMAX) GO TO 490
RESMAX = RES(I)
IMAX = I
GO TO 500
245 C 490 IF (RES(I) .GT. RESMIN) GO TO 500
RESMIN = RES(I)
IMIN = I
CONTINUE
250 C CALL CYCRES(RESMAX,RESMIN,1.0,INDEX(IMAX))
KK = KK + 1
510 C J = IMAX - 2
IF (J .LE. 0) GO TO 550
CALL CYCRES(RES(J),RES(J+1),1.0,INDEX(J))
KK = KK + 1
IMAX = J
255 C GO TO 510
550 C J = IMIN + 2
IF (J .GT. LRMAX) GO TO 575
CALL CYCRES(RES(J-1),RES(J),1.0,INDEX(J-1))
KK = KK + 1
IMIN = J
260 C GO TO 550
575 C KMAX = KK
LMAX = L
265 C SORT THE ANALYSIS SPECTRUM TO PRODUCE THE RANGE PAIR COUNTED SPECT
C
3000 KP = 0
DO 605 JJ = 1,NPKS
KC = 0
DO 600 I = 1,LMAX
IF (NINSTEP(I) .NE. JJ) GO TO 600
KP = KP + 1
KC = KC + 1
NSTEP(KP) = KP
270 C
275 C

```

DB

```

SUBROUTINE DDC15(L1,X2,X3,X4,X5,X6,X7,X8,X9,X10,X11,CYCNO,KCYGEN)
COMMON/DC15/SIG(X2,X3,X4,X5,X6,X7,X8,X9,X10,X11),L1,KMAX,KMIN,K31
COMMON/DC2/NSTEP(900),L2,KMAX,KMIN,K31
COMMON/DC3/RES(1470),INDEX(1400),IND1,IND2,IND3,IND4,KIND
COMMON/DC4/CYC/CYCLE(900,2),RNECYC(900),NSTEP(900)
COMMON/DC5/DEVL,LIND

```

```

C THIS SUBROUTINE DECIDES WHETHER OR NOT THE VALUES X1,X2,X3, AND X4
C FROM THE ADJUSTED LOAD SPECTRUM SATISFY THE RANGE PAIR COUNTING CO.

```

```

10 KFIRST = 0
    IF (X1 .NE. 0) GO TO 11
    IF (X3 .LE. X2) GO TO 200
    11 IF (X2 .GT. X1) GO TO 210
    IF (X2 .LT. X4 .OR. X3 .GT. X1) GO TO 500
    150 IF (X2 .GT. X3) GO TO 151
    CALL CYCGEN(X3,X2,1.0,NSTEP(I))
    GO TO 152

```

```

20 151 CALL CYCGEN(X2,X3, 1.0,NSTEP(I))
    152 X1 = X1
    X2 = X4
    IF (IND3 .NE. IND2) LIND = 1
    IND2 = IND4
    KCYGEN = 1
    IF (KEY .NE. 0) GO TO 110
    RETURN

```

```

25 210 IF (X2 .GT. X4 .OR. X3 .LT. X1) GO TO 500
    GO TO 150

```

```

30 200 X1 = X1
    X2 = X4
    IND2 = IND4
    KCYGEN = 2
    IF (KEY .EQ. 0) RETURN
    CYCNO = CYCNO - 1.0
    GO TO 110

```

```

35 C ADD X1 TO THE RESIDUE SPECTRUM
    C
    C

```

```

40 500 LR = LR + 1
    RES(LR) = X1
    INDEX(LR) = IND1
    X1 = X2
    X2 = X3
    X3 = X4

```

```

45 IND1 = IND2
    IND2 = IND3
    IND3 = IND4
    KCYGEN = 3
    IF (KEY .NE. 0) GO TO 110
    RETURN

```

```

50 110 GO TO (1150,1200,1500).KCYGEN
    1150 IF (CYCNO .GT. 1.0) GO TO 1151
    IF (CYCNO .EQ. 0.0) RETURN
    1153 CYCNO = CYCNO - 1.0
    GO TO 1152

```

```

55

```

581 CORE  
582 CORE  
583 CORE  
584 CORE  
585 CORE  
586 CORE  
587 CORE  
588 CORE  
589 CORE  
590 CORE  
591 CORE  
592 CORE  
593 CORE  
594 CORE  
595 CORE  
596 CORE  
597 CORE  
598 CORE  
599 CORE  
600 CORE  
601 CORE  
602 CORE  
603 CORE  
604 CORE  
605 CORE  
606 CORE  
607 CORE  
608 CORE  
609 CORE  
610 CORE  
611 CORE  
612 CORE  
613 CORE  
614 CORE  
615 CORE  
616 CORE  
617 CORE  
618 CORE  
619 CORE  
620 CORE  
621 CORE  
622 CORE  
623 CORE  
624 CORE  
625 CORE  
626 CORE  
627 CORE  
628 CORE  
629 CORE  
630 CORE  
631 CORE  
632 CORE  
633 CORE  
634 CORE  
635 CORE

SUBROUTINE DECIDE

```

1151 IF (LIND .EQ. 1) GO TO 1153
IF (IND3 .NE. IND4) GO TO 1153
RNECYC(L) = RNECYC(L) + CYCNO - 2.0
CYCNO = 1.0
60 1152 IF (KMAX .NE. 1) GO TO 111
X3 = SIGMIN(I)
IND3 = NSTEP(I)
IF (CYCNO .NE. 0.0) GO TO 112
KMIN = 1
KMAX = 0
KCYGEN = 3
RETURN
1200 IF (CYCNO .EQ. 0.0) RETURN
CYCNO = CYCNO - 1.0
70 X3 = SIGMAX(I)
X4 = SIGMIN(I)
KFIRST = 1
GO TO 113
111 X3 = SIGMAX(I)
X4 = SIGMIN(I)
IF (KFIRST .NE. 0) GO TO 113
CYCNO = CYCNO - 1.0
KFIRST = 1
113 IND3 = NSTEP(I)
IND4 = IND3
KMIN = 1
KMAX = 0
GO TO 11
1500 IF (KMAX .NE. 0) GO TO 1510
IF (CYCNO .EQ. 0.0) RETURN
CYCNO = CYCNO - 1.0
112 X4 = SIGMAX(I)
IND4 = NSTEP(I)
KMAX = 1
KMIN = 0
GO TO 11
1510 X4 = SIGMIN(I)
IND4 = NSTEP(I)
KMAX = 0
KMIN = 1
GO TO 10
END

```

CORE 636  
CORE 637  
CORE 638  
CORE 639  
CORE 640  
CORE 641  
CORE 642  
CORE 643  
CORE 644  
CORE 645  
CORE 646  
CORE 647  
CORE 648  
CORE 649  
CORE 650  
CORE 651  
CORE 652  
CORE 653  
CORE 654  
CORE 655  
CORE 656  
CORE 657  
CORE 658  
CORE 659  
CORE 660  
CORE 661  
CORE 662  
CORE 663  
CORE 664  
CORE 665  
CORE 666  
CORE 667  
CORE 668  
CORE 669  
CORE 670  
CORE 671  
CORE 672  
CORE 673  
CORE 674  
CORE 675  
CORE 676  
CORE 677

3

SUBROUTINE DESRES(LRMAX,NGYND)  
COMMON/COMMON/RES(1499),INDEX(1400),IND1,IND2,IND3,IND4,KIND

5 THIS SUBROUTINE DECIDES WHETHER OR NOT THE ELEMENTS OF THE RESIDUE  
SPECTRUM SATISFY THE RANGE PAIR COUNTING CONDITIONS

10 K = J  
NGYND = 0  
X1 = RES(1)  
X2 = RES(2)  
X3 = RES(3)  
X4 = RES(4)

15 IND1 = INDEX(1)  
IND2 = INDEX(2)  
IND3 = INDEX(3)  
IND4 = INDEX(4)  
J = 6

20 IF (X2 .GT. X1) GO TO 100  
IF (X2 .LT. X4 .OR. X3 .GT. X1) GO TO 300  
150 IF (X2 .GT. X3) GO TO 151  
CALL CYCRES(X3,X2,1.0,IND3)  
GO TO 152

25 151 CALL CYCRES(X2,X3,1.0,IND2)  
152 NGYND = NGYND + 1  
X1 = X1  
X2 = X4  
IND2 = IND4

30 IF (J .EQ. LRMAX) GO TO 300  
IF ((J + 1) .EQ. LRMAX) GO TO 315  
X3 = RES(J+1)  
X4 = RES(J+2)  
IND3 = INDEX(J+1)  
IND4 = INDEX(J+2)  
J = J+2

35 GO TO 10  
100 IF (X2 .GT. X4 .OR. X3 .LT. X1) GO TO 500  
GO TO 150  
500 K = K + 1

40 RES(K) = X1  
INDEX(K) = IND1  
J = J + 1  
IF (J .GT. LRMAX) GO TO 330  
X1 = X2  
X2 = X3  
X3 = X4

45 X4 = RES(J)  
IND1 = IND2  
IND2 = IND3  
IND3 = IND4  
IND4 = INDEX(J)  
GO TO 10

50 K = K + 1  
RES(K) = X1  
RES(K+1) = X2

CORE	679
CORE	679
CORE	680
CORE	681
CORE	682
CORE	683
CORE	684
CORE	685
CORE	686
CORE	687
CORE	688
CORE	689
CORE	690
CORE	691
CORE	692
CORE	693
CORE	694
CORE	695
CORE	696
CORE	697
CORE	698
CORE	699
CORE	700
CORE	701
CORE	702
CORE	703
CORE	704
CORE	705
CORE	706
CORE	707
CORE	708
CORE	709
CORE	710
CORE	711
CORE	712
CORE	713
CORE	714
CORE	715
CORE	716
CORE	717
CORE	718
CORE	719
CORE	720
CORE	721
CORE	722
CORE	723
CORE	724
CORE	725
CORE	726
CORE	727
CORE	728
CORE	729
CORE	730
CORE	731
CORE	732

```

INDEX(K) = IND1
INDEX(K+1) = IND2
LRMAX = K + 1
RETURN
60 315 K = K + 1
RES(K) = X1
RES(K+1) = X2
RES(K+2) = RES(J+1)
INDEX(K) = IND1
INDEX(K+1) = IND2
INDEX(K+2) = INDEX(J+1)
LRMAX = K + 2
RETURN
330 K = K + 1
RES(K) = X2
RES(K+1) = X3
RES(K+2) = X4
INDEX(K) = IND2
INDEX(K+1) = IND3
INDEX(K+2) = IND4
LRMAX = K + 2
RETURN
END
70
75
733 CORE
734 CORE
735 CORE
736 CORE
737 CORE
738 CORE
739 CORE
740 CORE
741 CORE
742 CORE
743 CORE
744 CORE
745 CORE
746 CORE
747 CORE
748 CORE
749 CORE
750 CORE
751 CORE
752 CORE
753 CORE
754 CORE
755 CORE

```

5 SUBROUTINE CYCRES(Y1,Y2, CYCPF,NSTEPP)  
COMMON/MSDE/L1,NO  
C THIS SUBROUTINE GENERATES CYCLES FOR THE ANALYSIS SPECTRUM FROM DA CORE 755  
C SUPPLIED BY SUBROUTINE DEGRS CORE 757  
C L = L + 1. CORE 759  
C CYCLE(L,1) = Y1 CORE 760  
C CYCLE(L,2) = Y2 CORE 761  
C RNECYC(L) = CYCPF CORE 762  
C NSTEP(L) = NSTEPP CORE 763  
C RETURN CORE 764  
C END CORE 765  
C CORE 766  
C CORE 767  
C CORE 768  
C CORE 769

APPENDIX II

SAMPLE PROBLEM WITH INPUT DATA LISTING

1. DATA DECKS ARE TO BE PROCESSED.

SPECTRUM SUBJECTED TO THE RANGE-PAIR COUNTING TECHNIQUE  
SEQUENCE ACCOUNTABLE FATIGUE EVALUATION

SPECTRUM FROM B-1 SPECTRUM -- TRUNCATION LEVEL 270.000 CYCLES  
 MATERIAL TYPE -- 2219-1051 AL

VEHICLE YIELD STRESS (KSI) -- 55.0000

LOG STRAIN INTERCEPT = .40000

INVERSE OF COFFIN-MANSACK SLOPE -- -1.83500

ELASTIC MODULUS = 10000.00000

COEFFICIENTS OF SECOND ORDER LEAST SQUARE FIT OF S-N DATA

LIFE	A(I)	B(I)	C(I)
10** 4	-.30217	.22941	55.82462
10** 5	-.00178	.33217	48.26657
10** 6	-.00149	.46283	39.66455
10** 7	-.30243	.64199	31.70955

UNNOTCHED COUPON S-N DATA DERIVED FROM  
 INFORMATION SUPPLIED FROM NORTH AMERICAN

RESIDUAL STRESS RELAXATION FUNCTION

ENEP = C1/(KTSMAX\*\*E1 + KTSMEAN\*\*E2)  
 WHERE C1 = .25000000E+07, E1 = 1.000 AND E2 = 1.000

7 TIMES THROUGH BLOCK OF 44 LOADS

LOAD LIMIT = 26.90000

STEP	TYPE	SMIN	SMAX	ENN
1	1	-.14800	-.02900	1.00000
2	1	.58400	1.00000	1.00000
3	2	.58400	.89500	1.00000
4	1	.58400	.68400	1.00000
5	1	.45500	.59400	1.00000
6	1	.64800	.69300	1.00000
7	1	.39500	.69000	3.00000
8	1	.48800	.56300	1.00000
9	1	.23500	.59000	1.00000
10	1	.50900	.62700	2.00000
11	1	.43400	.59000	1.00000
12	1	.36400	.69000	1.00000
13	1	.13300	.36400	1.00000
14	1	.36400	.48200	6.00000
15	1	.30400	.36400	1.00000
16	1	.30700	.75600	1.00000
17	1	.31600	.52700	1.00000
18	2	.54500	.70900	1.00000
19	1	.33400	.61700	1.00000
20	1	.45100	.51200	7.00000
21	1	.12700	.54800	1.00000
22	1	.22900	.37700	48.00000
23	1	-.00600	.34000	1.00000
24	1	.13800	.24400	39.00000
25	1	.07800	.72300	1.00000
26	1	.11100	.55700	9.00000
27	1	.18700	.37700	10.00000
28	3	.52700	.99100	1.00000
29	2	-.12500	.68300	1.00000

31	1	.2200	.7470	1.00000
32	1	-.14800	-.32900	1.00000
33	1	.55700	.84600	1.00000
34	1	.50900	.88000	1.00000
35	1	.33100	.52700	1.00000
36	1	.52700	.61700	19.00000
37	1	.46400	.52700	2.00000
38	1	.48200	.81300	1.00000
39	1	.58100	.68700	1.00000
40	1	.32200	.70200	1.00000
41	1	.39500	.49000	9.00000
42	1	.43100	.59000	5.00000
43	1	.46100	.54500	29.00000
44	1	-.14800	-.02900	1.00000
	1	-.13600	-.73900	1.00000

BLOCK TYPE TYPE TYPE TYPE TYPE TYPE TYPE TYPE

1	1	-0	-0
2	1	-0	-0
3	2	1	-0
4	1	-0	-0
5	3	1	-0
6	1	-0	-0
7	1	-0	-0

SPECTRUM FROM 9-1 SPECIMEN -- TRUNCATION LEVEL 270,000 CYCLES

AKT = 4.50

RELAXATION CONSTANT C/R = 25.00000000

SPECTRUM SUBJECTED TO LIFE RANGE-PAIR COUNTING TECHNIQUE

FLIGHT OR BLOCK NO. 1

STMAX	STMIN	SIGMAX	SIGMIN	RES	EQRES	ENN	NEP
-78	3.98	-3.51	-17.92	0.00	0.00	1.00	.13025031E+05
18.40	15.71	55.50	42.89	-27.87	-27.87	1.00	.39342702E+03
15.71	12.24	42.89	27.29	-27.80	-15.69	1.00	.56235609E+03
18.64	17.43	55.20	49.55	-28.89	-28.89	1.00	.36717948E+03
19.80	10.63	50.00	18.20	-29.61	-29.61	1.00	.44621600E+03
15.14	13.13	38.54	29.66	-29.61	-13.13	3.00	.57667259E+03
13.69	6.32	32.20	6.32	-29.42	-6.61	1.00	.90105168E+03
16.37	13.69	46.54	32.25	-29.36	-29.36	2.00	.47906511E+03
13.69	11.67	32.34	23.26	-29.28	-6.61	1.00	.71090397E+03
18.56	9.79	54.32	14.96	-29.21	-28.52	1.00	.46919213E+03
9.79	3.58	14.86	-13.13	-29.21	0.00	1.00	.18861779E+04
12.97	9.79	29.18	14.91	-29.17	-3.35	6.00	.83680259E+03
9.79	8.18	15.32	8.06	-28.74	3.33	1.00	.14033309E+04
20.34	8.26	55.00	38.5	-36.51	-36.51	1.00	.42460525E+03
14.18	8.50	27.28	1.74	-36.51	-8.79	1.00	.76807282E+03
13.50	8.98	38.26	4.00	-35.43	-19.69	1.00	.58153353E+03
13.77	10.79	25.64	12.18	-35.35	-6.98	7.00	.72995133E+03
14.74	3.42	30.61	-29.35	-35.72	-11.34	1.00	.92247814E+03
10.14	6.16	9.97	-7.84	-35.66	3.31	48.00	.14935733E+04
		10.11	-7.81			48.00	
		10.37	-7.55			9.63	
		10.63	-7.29			14.43	
		10.89	-7.03			19.20	
		11.14	-6.77			24.00	
		11.40	-6.52			28.80	
		11.65	-6.27			33.60	
		11.90	-6.02			38.40	
		12.15	-5.77			43.20	
		12.40	-5.52			48.00	
9.15	-.16	8.04	-33.84	-33.12	0.00	1.00	.31047963E+04
6.56	3.50	-3.56	-17.36	-33.09	0.00	35.00	.37392025E+04
		-3.52	-17.32			3.50	
		-3.45	-17.25			7.00	
		-3.38	-17.18			10.50	
		-3.31	-17.11			14.00	
		-3.24	-17.04			17.50	
		-3.17	-16.97			21.00	
		-3.10	-16.90			24.50	
		-3.03	-16.83			28.00	
		-2.96	-16.76			31.50	
		-2.89	-16.69			35.00	
19.45	2.10	55.00	-23.08	-32.52	-32.52	1.00	.58920931E+03
14.98	2.99	34.91	-19.08	-32.52	-12.42	9.00	.91708325E+03
10.14	5.03	13.57	-9.43	-32.67	0.00	10.00	.16047965E+04
20.09	14.18	55.00	28.37	-35.42	-35.42	1.00	.35055054E+03
-.78	-3.98	-38.93	-53.34	-35.42	0.00	1.00	.13025031E+05
22.76	14.98	55.00	29.02	-47.41	-47.41	1.00	.28748314E+03
23.67	13.69	55.00	15.09	-51.52	-51.52	1.00	.27916148E+03
14.18	8.90	12.27	-11.46	-51.52	-8.79	1.00	.75464468E+03
16.60	14.18	23.29	12.40	-51.39	-19.69	19.00	.48342514E+03
		23.44	12.54			1.90	
		23.72	12.83			3.80	
		24.00	13.11			5.70	

5

CH20	13.02	7.02	1.90000000
24.56	13.66	9.50	1.90000000
24.83	13.94	11.40	1.90000000
25.11	14.21	13.30	1.90000000
25.37	14.48	15.20	1.90000000
25.64	14.75	17.10	1.90000000
25.91	15.01	19.00	1.90000000
14.18	12.48	7.52	.65336542E+03
21.79	12.97	9.98	.32605652E+03
18.48	15.63	34.83	.39170979E+03
18.88	8.66	36.76	.47468254E+03
12.38	10.63	6.23	.90043347E+03
15.87	11.59	5.14	.56645141E+03
14.66	12.40	19.56	.62236611E+03
19.75	9.58	25.90	2.90000000
20.12	9.95	5.80	2.90000000
20.50	10.33	8.70	2.90000000
20.86	10.73	11.69	2.90000000
21.23	11.06	14.50	2.90000000
21.59	11.42	17.40	2.90000000
21.94	11.78	20.30	2.90000000
22.30	12.13	23.20	2.90000000
22.65	12.48	26.10	2.90000000
22.99	12.82	29.00	2.90000000
-1.78	-3.98	-40.60	.13025831E+05
-1.05	-3.66	-41.80	.14337171E+05

THE NUMBER OF PEAKS OR VALLEYS IN THE INPUT LOAD SPECTRUM = 75

STEP	MAXIMUM	SIGMA	MINIMUM	COUNTER K
1	.351945E+01		-.179154E+02	1.00000
2	.550000E+02		.428950E+02	1.00000
3	.428950E+02		.272955E+02	1.00000
4	.550000E+02		.495527E+02	1.00000
5	.550000E+02		.182008E+02	1.00000
6	.385372E+02		.294585E+02	3.00000
7	.321966E+02		-.971139E+00	1.00000
8	.465387E+02		.322548E+02	2.00000
9	.323357E+02		.232579E+02	1.00000
10	.543191E+02		.148569E+02	1.00000
11	.148601E+02		-.131024E+02	1.00000
12	.291797E+02		.148958E+02	6.00000
13	.153186E+02		.805562E+01	1.00000
14	.550000E+02		.648595E+00	1.00000
15	.272796E+02		.173805E+01	1.00000
16	.382570E+02		.399989E+01	1.00000
17	.256130E+02		.121764E+02	7.00000
18	.306127E+02		-.203494E+02	1.00000
19	.101054E+02		-.781000E+01	4.80000
20	.103674E+02		.754800E+01	4.80000
21	.106275E+02		-.728795E+01	4.80000
22	.106856E+02		-.702978E+01	4.80000
23	.111419E+02		-.677353E+01	4.80000
24	.113962E+02		-.651917E+01	4.80000
25	.116487E+02		-.626669E+01	4.80000
26	.118993E+02		-.601667E+01	4.80000
27	.121481E+02		-.576729E+01	4.80000
28	.123950E+02		-.552036E+01	4.80000
29	.803921E+01		-.338441E+02	1.00000
30	.352359E+01		-.173203E+02	3.50000
31	-.344941E+01		-.172491E+02	3.50000
32	-.337838E+01		-.171781E+02	3.50000
33	-.331750E+01		-.171071E+02	3.50000

34	..323678E+01	..17369E+02	3.50000
35	..316621E+01	..16699E+02	3.50000
36	..309579E+01	..16895E+02	3.50000
37	..302552E+01	..16825E+02	3.50000
38	..295540E+01	..16755E+02	3.50000
39	..288544E+01	..16685E+02	3.50000
40	..281500E+01	..23072E+02	1.00000
41	..274457E+01	..19082E+02	9.00000
42	..267458E+01	..94337E+01	10.00000
43	..260400E+01	..28369E+02	1.00000
44	..253348E+01	..57333E+02	1.00000
45	..246300E+01	..20065E+02	1.00000
46	..239250E+01	..10905E+02	1.00000
47	..23224E+02	..11456E+02	1.00000
48	..225236E+02	..12542E+02	1.90000
49	..218213E+02	..12826E+02	1.90000
50	..211191E+02	..13186E+02	1.90000
51	..204174E+02	..13387E+02	1.90000
52	..197159E+02	..13664E+02	1.90000
53	..190145E+02	..13939E+02	1.90000
54	..183135E+02	..14210E+02	1.90000
55	..176127E+02	..14482E+02	1.90000
56	..169121E+02	..14741E+02	1.90000
57	..162117E+02	..15017E+02	1.90000
58	..155114E+02	..75172E+01	2.00000
59	..148112E+02	..99769E+01	1.00000
60	..141111E+02	..21937E+02	1.00000
61	..134110E+02	..92322E+01	1.00000
62	..127109E+02	..31130E+01	9.00000
63	..120108E+02	..51416E+01	5.00000
64	..113107E+02	..95783E+01	2.90000
65	..106106E+02	..99545E+01	2.90000
66	..991065E+02	..10326E+02	2.90000
67	..921064E+02	..10695E+02	2.90000
68	..851063E+02	..11059E+02	2.90000
69	..781062E+02	..11420E+02	2.90000
70	..711061E+02	..11776E+02	2.90000
71	..641060E+02	..12129E+02	2.90000
72	..571059E+02	..12478E+02	2.90000
73	..501058E+02	..12838E+02	2.90000
74	..431057E+02	..55000E+02	1.00000
75	..361056E+02	..53548E+02	1.00000

RANGE PAIR CYCLE COUNTED SPECTRUM

STEP	MAXIMUM	SIGMA	MINIMUM	COUNTER K
1	.351345E+01		.179154E+02	1.00000
2	.553000E+02		.272795E+02	1.00000
3	.553000E+02		.495527E+02	1.00000
4	.553000E+02		.294585E+02	1.00000
5	.321966E+02		.322548E+02	1.00000
6	.465387E+02		.232570E+02	1.00000
7	.465387E+02		.322548E+02	1.00000
8	.323357E+02		.971139E+00	1.00000
9	.543191E+02		.148568E+02	1.00000
10	.148601E+02		.148568E+02	1.00000
11	.291797E+02		.148958E+02	5.00000
12	.291797E+02		.805628E+01	1.00000
13	.153186E+02		.148958E+02	1.00000
14	.553000E+02		.131024E+02	1.00000
15	.272795E+02		.173800E+01	1.00000
16	.382570E+02		.648550E+00	1.00000
17	.306127E+02		.121764E+02	1.00000
18	.101354E+02		.781000E+01	5.00000
19	.103674E+02		.754800E+01	5.00000
20	.106275E+02		.728793E+01	5.00000
21	.108856E+02		.702978E+01	5.00000
22	.111419E+02		.677353E+01	5.00000
23	.113962E+02		.651917E+01	5.00000
24	.116487E+02		.626669E+01	5.00000
25	.118993E+02		.601607E+01	5.00000
26	.121481E+02		.576729E+01	5.00000
27	.123950E+02		.552036E+01	4.00000
28	.123950E+02		.203494E+02	1.00000
29	.863921E+01		.552036E+01	1.00000
30	.352059E+01		.173203E+02	4.00000
31	.344941E+01		.172491E+02	4.00000
32	.337838E+01		.171781E+02	4.00000
33	.330750E+01		.171072E+02	4.00000
34	.323678E+01		.170365E+02	4.00000
35	.316621E+01		.169659E+02	4.00000
36	.309579E+01		.168955E+02	4.00000
37	.302552E+01		.168252E+02	4.00000
38	.295540E+01		.167551E+02	4.00000
39	.288544E+01		.166851E+02	4.00000
40	.281500E+01		.190826E+02	1.00000
41	.135658E+02		.943374E+01	10.00000
42	.553000E+02		.338441E+02	1.00000
43	.553000E+02		.200165E+02	1.00000
44	.553000E+02		.533397E+02	1.00000
45	.234369E+02		.125424E+02	2.00000
46	.237213E+02		.128688E+02	2.00000
47	.240031E+02		.131086E+02	2.00000
48	.242824E+02		.133079E+02	2.00000
49	.245592E+02		.136647E+02	2.00000
50	.248335E+02		.139390E+02	2.00000
51	.251053E+02		.142108E+02	2.00000
52	.253747E+02		.146822E+02	2.00000
53	.256416E+02		.147471E+02	2.00000
54	.259062E+02		.150117E+02	1.00000
55	.259062E+02		.751720E+01	1.00000
56	.151434E+02		.150117E+02	1.00000
57	.151434E+02		.751720E+01	1.00000

70	.496692E+02	.114084E+02	1.000000
51	.248265E+02	.21975E+02	1.000000
51	.367828E+02	.997609E+01	1.000000
61	.243885E+02	.514165E+01	4.000000
52	.243686E+02	.31136E+00	1.000000
63	.197465E+02	.957830E+01	3.000000
64	.201228E+02	.99595E+01	3.000000
65	.264951E+02	.103289E+02	3.000000
66	.298634E+02	.10992E+02	3.000000
67	.212277E+02	.110595E+02	3.000000
68	.215882E+02	.114200E+02	3.000000
69	.219448E+02	.117765E+02	3.000000
71	.222977E+02	.121295E+02	3.000000
71	.226467E+02	.124785E+02	3.000000
72	.229920E+02	.128238E+02	2.000000
73	.229920E+02	.514165E+01	1.000000
74	.4417990E+02	.535478E+02	1.000000

LOCAL STRESSES AND PLASTIC STRAINS W/RESULTING FATIGUE LIFE

STEP	FLASTIC STRAIN	MAX OR MIN	DAMAGE
2	.0696	MAX	.59085E-03
4	.0835	MAX	.236703E-05
5	.10722	MAX	.105771E-05
14	.09249	MAX	.891548E-04
22	.08123	MAX	.498322E-07
25	.08123	MAX	.242990E-04
27	.08123	MAX	.233133E-03
20	.08155	MAX	.379281E-04
30	.08153	MIN	.718916E-05

DAMAGE FROM PLASTIC STRAINS= .58426328E-03

SIGMAX	SIGMIN	RNCYC	CYCLES	ENNC/CYC
-3.51	-17.92	1.	.1000000E+10	.1000000E-08
55.00	27.28	1.	.4807856E+06	.6753172E-05
55.00	49.55	1.	.6990094E+09	.14306010E-08
55.00	-59.33	1.	.1000000E+05	.1000000E-03
32.27	29.46	1.	.4690198E+12	.21321090E-11
46.54	32.25	1.	.1153687E+09	.86678619E-08
46.54	23.26	1.	.51785591E+07	.19310406E-06
32.34	32.25	1.	.36726287E+13	.27228454E-12
54.32	-.97	1.	.14931837E+05	.66970905E-04
14.86	14.86	1.	.50066416E+12	.19973469E-11
29.18	14.90	5.	.12665797E+10	.39476395E-08
29.18	8.96	1.	.14052164E+09	.71163418E-08
15.32	14.90	1.	.42187482E+12	.23703714E-11
55.00	-13.14	1.	.1000000E+05	.1000000E-03
27.28	1.74	1.	.5300468E+08	.18866263E-07
38.26	.65	1.	.17454308E+07	.57293429E-06
30.61	12.18	1.	.27966528E+09	.35757031E-08
10.11	-7.91	5.	.1000000E+10	.5000000E-08
10.37	-7.55	5.	.1000000E+10	.5000000E-08
10.63	-7.29	5.	.1000000E+10	.5000000E-08
10.09	-7.03	5.	.1000000E+10	.5000000E-08
11.14	-6.77	5.	.56054816E+09	.89198403E-08
11.00	-6.52	5.	.5608858E+09	.89144700E-08
11.65	-6.27	5.	.56115089E+09	.89102593E-08
11.92	-6.02	5.	.56134367E+09	.89071933E-08
12.15	-5.77	5.	.56144458E+09	.89052816E-08
12.40	-5.52	4.	.561511397E+09	.71235984E-08
12.40	-21.35	1.	.27376288E+08	.36527961E-07
8.54	-5.52	1.	.1000000E+10	.1000000E-08
-3.52	-17.32	4.	.1000000E+10	.4000000E-08
-3.45	-17.25	4.	.1000000E+10	.4000000E-08
-3.38	-17.18	4.	.1000000E+10	.4000000E-08

3.51	-17.11	4.	.10000000E+10	.40000000E-08
-3.24	-17.04	4.	.10000000E+10	.40000000E-08
-3.17	-16.97	4.	.10000000E+10	.40000000E-08
-3.10	-16.90	4.	.10000000E+10	.40000000E-08
-3.03	-16.83	4.	.10000000E+10	.40000000E-08
-2.96	-16.76	4.	.10000000E+10	.40000000E-08
-2.89	-16.69	4.	.10000000E+10	.40000000E-08
55.00	-19.08	1.	.10000000E+05	.10000000E-03
13.57	-9.43	10.	.16693817E+09	.60227117E-07
55.00	-33.84	1.	.10000000E+05	.10000000E-03
55.00	20.92	1.	.63546690E+05	.15736461E-04
55.00	-53.34	1.	.10000000E+05	.10000000E-03
23.44	12.54	2.	.52786328E+10	.37888599E-05
23.72	12.83	2.	.52805280E+10	.37875000E-05
24.00	13.11	2.	.52807566E+10	.37873361E-05
24.28	13.39	2.	.52793259E+10	.37883625E-05
24.56	13.66	2.	.52762450E+10	.37905745E-05
24.83	13.94	2.	.52715248E+10	.37939687E-05
25.11	14.21	2.	.52651779E+10	.37985421E-05
25.37	14.48	2.	.52572184E+10	.38042931E-05
25.64	14.75	2.	.52476624E+10	.38112208E-05
25.91	15.01	1.	.52365271E+10	.19096626E-05
25.91	7.52	1.	.37589154E+09	.26603419E-08
15.14	15.01	1.	.48307951E+12	.20700526E-11
15.14	7.52	1.	.15515977E+11	.64449697E-10
49.68	-11.46	1.	.24152055E+05	.41404344E-04
34.83	22.00	1.	.15814258E+10	.63234078E-09
36.76	9.98	1.	.15038306E+08	.66496853E-07
24.39	5.14	4.	.31310615E+09	.12775220E-07
24.39	-3.1	1.	.77428372E+08	.12915162E-07
19.75	9.58	3.	.67337984E+10	.44551379E-05
20.12	9.95	3.	.67798754E+10	.44248601E-05
20.50	10.33	3.	.68230691E+10	.43968484E-05
20.86	10.70	3.	.68633296E+10	.43710563E-05
21.23	11.06	3.	.69036119E+10	.43474405E-05
21.59	11.42	3.	.69348755E+10	.43259609E-05
21.94	11.78	3.	.69660846E+10	.43065799E-05
22.30	12.13	3.	.69942085E+10	.42892630E-05
22.65	12.48	3.	.70192214E+10	.42739783E-05
22.99	12.82	2.	.70411021E+10	.28404644E-09
22.99	5.14	1.	.49378912E+09	.20251568E-08
-41.80	-53.54	1.	.10000000E+10	.10000000E-08

DAMAGE PER THIS SET= .16162546E-02

TOTAL ENN/CYC =, .16162546E-02

FLIGHT OR-BLOCK NO.	STMIN	SIGMAX	SIGMIN	RES	ERES	ENN	NEP
18.78	3.98	-40.58	-54.99	-37.67	0.00	1.00	.13028031E+05
18.40	15.71	45.73	33.63	-37.07	-27.81	1.00	.39342782E+03
15.71	12.24	33.68	18.37	-37.01	-15.69	1.00	.56235689E+03
18.64	17.43	46.96	41.52	-36.92	-29.89	1.00	.36717948E+03
18.80	10.63	47.74	1.94	-36.87	-29.61	1.00	.44621680E+03
15.14	13.13	31.31	22.24	-35.84	-13.15	3.00	.57667259E+03
13.69	6.32	25.06	-8.11	-35.55	-6.61	1.00	.90105168E+03
16.87	13.69	39.42	25.14	-35.48	-20.90	2.00	.47900511E+03
13.69	11.67	25.29	16.21	-35.33	-6.61	1.00	.71090997E+03
18.56	9.79	47.29	7.83	-35.23	-28.52	1.00	.46919213E+03
9.79	3.58	7.87	-23.13	-35.23	0.00	1.00	.18861779E+04
12.97	9.79	22.20	7.91	-35.15	-3.35	6.00	.83680259E+03
9.79	8.18	8.45	1.19	-35.01	0.00	1.00	.14033389E+04
20.34	8.26	59.09	.65	-35.01	-36.51	1.00	.42460525E+03
14.18	8.58	27.28	1.74	-35.01	-8.79	1.00	.76807522E+03
16.60	8.98	38.26	4.00	-35.43	-19.69	1.00	.58153350E+03
13.77	10.79	25.61	12.18	-35.35	-8.93	7.00	.72996103E+03
14.74	3.42	30.61	-20.35	-35.72	-11.33	1.00	.92247814E+03



14 .00032  
 22 .00004  
 25 .00123  
 27 .00420  
 28 .00156  
 38 .00063

.202856E-05  
 .498022E-07  
 .242990E-04  
 .233133E-03  
 .379281E-04  
 .718916E-05

DAMAGE FROM PLASTIC STRAINS= .30462764E-03

	SIGMAX	SIGMIN	RNCYC	CYCLES	ENN/CYC
1.	-46.58	-54.99	1.	.1000000E+10	.1000000E-08
1.	45.73	18.07	1.	.2409377E+07	.41504493E-06
1.	33.68	33.63	1.	.43206072E+13	.23144895E-12
1.	46.96	41.52	1.	.23661145E+11	.42263381E-10
1.	47.74	-20.13	1.	.19780666E+05	.50554416E-04
1.	25.06	22.24	1.	.27937764E+12	.35793846E-11
1.	39.42	25.14	1.	.59516846E+09	.19795377E-08
1.	39.42	16.21	1.	.24328135E+08	.40928567E-07
1.	25.29	25.14	1.	.15753982E+13	.63476016E-12
1.	47.29	-8.11	1.	.58801786E+05	.17006286E-04
1.	7.87	7.83	1.	.1000000E+10	.1000000E-08
1.	22.20	7.91	5.	.15461310E+10	.32338787E-08
1.	22.20	1.19	1.	.21206560E+09	.47155219E-08
1.	8.45	7.91	1.	.1000000E+10	.1000000E-08
1.	55.00	-53.34	1.	.1000000E+10	.1000000E-08
1.	27.28	1.74	1.	.53004668E+08	.18866583E-07
1.	38.26	.65	1.	.17454308E+07	.57293429E-06
1.	30.61	12.18	1.	.27966528E+09	.35757031E-08
1.	10.11	-7.81	5.	.1000000E+10	.5000000E-08
1.	10.37	-7.55	5.	.1000000E+10	.5000000E-08
1.	10.63	-7.29	5.	.1000000E+10	.5000000E-08
1.	10.89	-7.03	5.	.1000000E+10	.5000000E-08
1.	11.14	-6.77	5.	.1000000E+10	.5000000E-08
1.	11.40	-6.52	5.	.56054816E+09	.89198403E-08
1.	11.65	-6.27	5.	.56115089E+09	.8914700E-08
1.	11.90	-6.02	5.	.56134367E+09	.89102593E-08
1.	12.15	-5.77	5.	.56146456E+09	.89052816E-08
1.	12.40	-5.52	4.	.56151397E+09	.71235984E-08
1.	12.40	-26.35	1.	.27376289E+08	.36527961E-07
1.	8.84	-5.52	1.	.1000000E+10	.1000000E-08
1.	-3.52	-17.32	4.	.1000000E+10	.4000000E-08
1.	-3.45	-17.25	4.	.1000000E+10	.4000000E-08
1.	-3.38	-17.18	4.	.1000000E+10	.4000000E-08
1.	-3.31	-17.11	4.	.1000000E+10	.4000000E-08
1.	-3.24	-17.04	4.	.1000000E+10	.4000000E-08
1.	-3.17	-16.97	4.	.1000000E+10	.4000000E-08
1.	-3.10	-16.90	4.	.1000000E+10	.4000000E-08
1.	-3.03	-16.83	4.	.1000000E+10	.4000000E-08
1.	-2.96	-16.76	4.	.1000000E+10	.4000000E-08
1.	-2.89	-16.69	4.	.1000000E+10	.4000000E-08
1.	55.00	-19.98	1.	.1000000E+10	.4000000E-08
1.	13.57	-9.43	10.	.1000000E+05	.50227117E-07
1.	55.00	-33.84	1.	.1000000E+05	.1000000E-03
1.	55.00	26.02	1.	.1000000E+05	.1000000E-03
1.	55.00	-55.00	1.	.63546690E+05	.15736461E-04
1.	23.44	12.54	2.	.1000000E+05	.1000000E-03
1.	23.72	12.83	2.	.52786328E+10	.37888599E-09
1.	24.00	13.11	2.	.52805280E+10	.37875000E-09
1.	24.28	13.39	2.	.52807566E+10	.37873361E-09
1.	24.56	13.66	2.	.52793259E+10	.37883625E-09
1.	24.83	13.94	2.	.52762450E+10	.37805745E-09
1.	25.11	14.21	2.	.52715248E+10	.37939687E-09
1.	25.37	14.48	2.	.5252184E+10	.38042931E-09
1.	25.64	14.75	2.	.52476824E+10	.38112288E-09
1.	25.91	15.01	1.	.52365271E+10	.19096626E-09
1.	26.91	7.32	1.	.37589154E+09	.26603419E-08
1.	15.14	15.01	1.	.48307951E+12	.20700526E-11

FLIGHT OR BLOCK NO.	SIMAX	SIGMIN	SIGMAX	SIGMIN	RES	EORES	ENN	NEP	TOTAL ENN/CYC =	DAMAGE PER THIS SET=
15114	7.52									
24152	1.00									
15814	1.00									
15038	1.00									
31310	1.00									
77428	1.00									
67337	3.00									
67798	3.00									
68230	3.00									
68633	3.00									
69066	3.00									
69348	3.00									
69666	3.00									
69942	3.00									
70192	3.00									
70411	2.00									
49378	1.00									
19000	1.00									

DAMAGE PER THIS SET= .63070929E-03

TOTAL ENN/CYC = .24469639E-02

FLIGHT OR BLOCK NO.	SIMAX	SIGMIN	SIGMAX	SIGMIN	RES	EORES	ENN	NEP	TOTAL ENN/CYC =	DAMAGE PER THIS SET=
2408	-3.98	45.71	55.00	17.35	-53.34	-53.34	1.00	.13025831E+05	1	
1844	15.71	28.46	17.35	17.35	-53.34	-27.81	1.00	.39342782E+03	2	
1571	12.24	17.50	1.89	1.89	-53.34	-15.69	1.00	.56235689E+03	3	
1864	17.43	31.85	25.40	25.40	-53.34	-28.93	1.00	.36717988E+03	4	
1088	1.63	31.73	5.07	5.07	-52.89	-29.61	1.00	.44621680E+03	5	
1514	13.13	15.38	6.31	6.31	-52.77	-13.13	3.00	.37667250E+03	6	
1369	6.32	9.32	23.85	23.85	-52.29	-6.61	1.00	.91185168E+03	7	
1687	13.69	23.72	9.44	9.44	-52.38	-23.91	2.00	.47986511E+03	8	
1369	11.67	9.74	6.65	6.65	-51.83	-6.61	1.00	.71093975E+03	9	
1856	9.79	31.79	-7.67	-7.67	-51.73	-28.52	1.00	.46919233E+03	10	
979	3.58	-7.56	-35.52	-35.52	-51.62	8.33	1.00	.18861779E+04	11	
1297	9.79	6.79	-7.49	-7.49	-51.55	-3.35	6.00	.83680259E+03	12	
979	8.18	-6.73	-13.97	-13.97	-50.77	0.00	1.00	.14033389E+04	13	
2034	8.26	43.83	-13.52	-13.52	-50.68	-36.51	1.00	.42460523E+03	14	
1418	8.50	13.19	-12.35	-12.35	-50.61	-8.79	1.00	.76807252E+03	15	
1935	14.66	35.22	15.49	15.49	-50.48	-30.77	1.00	.38464145E+03	16	
1660	8.98	24.32	8.93	8.93	-50.35	-19.69	1.00	.58153350E+03	17	
1377	10.79	11.74	-1.73	-1.73	-50.24	-6.99	7.00	.72996173E+03	18	
1474	3.42	17.04	-33.92	-33.92	-49.33	-11.34	1.00	.92247814E+03	19	
1014	6.16	-3.57	-21.48	-21.48	-49.23	0.00	48.00	.14935733E+04	20	
		-3.39	-21.33	-21.33			4.80		21	
		-3.02	-20.94	-20.94			9.60		22	
		-2.66	-20.58	-20.58			14.40		23	
		-2.31	-20.22	-20.22			19.20		24	
		-1.96	-19.87	-19.87			24.00		25	
		-1.63	-19.52	-19.52			28.80		26	
		-1.26	-19.17	-19.17			33.60		27	
		-.91	-18.83	-18.83			38.40		28	
		-.57	-18.48	-18.48			43.20		29	
		-.23	-18.14	-18.14			48.00		30	
915	-16	-4.54	-46.42	-46.42	-45.69	0.00	1.00	.30047963E+04	31	
656	3.50	-16.12	-29.92	-29.92	-45.65	0.00	35.00	.37392025E+04	32	
		-16.07	-29.87	-29.87			3.50		33	
		-15.97	-29.77	-29.77			7.00		34	
		-15.88	-29.68	-29.68			11.50		35	
		-15.78	-29.58	-29.58			16.00		36	
		-15.68	-29.48	-29.48			21.50		37	
		-15.58	-29.38	-29.38			27.00		38	
		-15.49	-29.29	-29.29			32.50		39	
		-15.39	-29.19	-29.19			38.00		40	
		-15.29	-29.09	-29.09			43.50		41	

6.50000000

STEP	PLASTIC STRAIN	MAX OR MIN	DAMAGE	ENN/CYC
19.45	2.10	-29.00	0.50000000	
14.98	2.99	35.24		
10.14	5.03	31.20		
23.75	-3.23	-21.28		
20.09	14.18	60.71		
7.76	-3.98	44.22		
22.76	14.98	55.00		
23.67	13.69	55.00		
14.18	8.90	12.27		
16.60	14.18	23.29		
23.72	12.83	23.44		
24.00	13.11	24.00		
24.28	13.39	24.28		
24.56	13.66	24.56		
24.83	13.94	24.83		
25.11	14.21	25.11		
25.37	14.48	25.37		
25.64	14.75	25.64		
25.91	15.01	25.91		
14.18	12.48	15.14		
21.79	12.97	49.68		
18.48	15.63	34.83		
18.88	8.66	36.76		
12.08	10.63	6.23		
15.87	11.59	24.39		
14.66	12.40	19.56		
		19.75		
		20.12		
		20.50		
		20.86		
		21.23		
		21.59		
		21.94		
		22.30		
		22.65		
		22.99		
7.76	-3.98	40.60		
1.05	-3.66	-41.80		

LOCAL STRESSES AND PLASTIC STRAINS W/RESULTING FATIGUE LIFE

STEP	PLASTIC STRAIN	MAX OR MIN	DAMAGE	ENN/CYC
2	0.0593	MAX	0.438461E-03	
27	0.0321	MAX	0.141845E-03	
29	0.0185	MIN	0.183981E-04	
30	0.0365	MAX	0.180144E-03	
31	0.0156	MAX	0.379281E-04	
41	0.0063	MIN	0.718916E-05	

SIGMAX	SIGMIN	RNGCYC	DAMAGE FROM PLASTIC STRAINS	ENN/CYC
-40.58	-54.99	1.	0.1000000E+00	0.1000000E+00
55.00	-46.42	1.	0.1000000E+05	0.1000000E+03
29.46	17.35	1.	0.29397878E+10	0.3401606E-09
17.50	17.35	1.	0.63000870E+12	0.1587279E-11
30.85	25.43	1.	0.77590688E+11	0.12888145E-10
31.73	1.89	1.	0.14305010E+08	0.69905579E-07
9.32	6.31	1.	0.1000000E+10	0.1000000E-08
23.72	9.44	1.	0.15092544E+10	0.6625788E-09
23.72	4.66	1.	0.11630863E+09	0.84524691E-08
9.74	9.44	1.	0.1000000E+10	0.1000000E-08
31.79	-23.85	1.	0.46704942E+06	0.21411011E-05
-7.56	-7.67	1.	0.0000000E+00	0.0000000E+00

DAMAGE FROM PLASTIC STRAINS = 0.82396546E-03



FLIGHT OR BLOCK NO. 4

TOTAL ENN/CYC =, .37115261E-02

DAMAGE FROM PLASTIC STRAINS= .30462764E-03

DAMAGE PER THIS SET= .83070929E-03

FLIGHT OR BLOCK NO. 5

TOTAL ENN/CYC =, .45422354E-02

DAMAGE FROM PLASTIC STRAINS= .22665690E-02

DAMAGE PER THIS SET= .26509046E-02

FLIGHT OR BLOCK NO. 6

TOTAL ENN/CYC =, .71931400E-02

DAMAGE FROM PLASTIC STRAINS= .30462764E-03

DAMAGE PER THIS SET= .83070929E-03

FLIGHT OR BLOCK NO. 7

TOTAL ENN/CYC =, .80238493E-02

DAMAGE FROM PLASTIC STRAINS= .30462764E-03

DAMAGE PER THIS SET= .83070929E-03

TOTAL ENN/CYC =, .88545586E-02



PREPARED: 73 SEP 17

MEDIA CONVERSION LISTING

LABEL/ CARDS

PCN N102001

INPUT FILE	0	1	2	3	4	5	6	7	8	9	OUTPUT FILE
RECORD NUMBER	123456789	0123456789	0123456789	0123456789	0123456789	0123456789	0123456789	0123456789	0123456789	0123456789	RECORD NUMBER
	37	38	39	40	41	42	43	44	1	1	000048
		.482	.581	.322	.395	.431	.461	-.148	.81	1.	000049
									.687	1.	000050
									.702	1.	000051
									.449	9.	000052
									.59	5.	000053
									.545	29.	000054
									-.029	1.	000055
									-.039	1.	000056
	1	1	1	1	1	1	1	1			000057
	2	1	1	1	1	1	1	1			000058
	3	2	1	1	1	1	1	1			000059
	4	1	1	1	1	1	1	1			000060
	5	3	1	1	1	1	1	1			000061
	6	1	1	1	1	1	1	1			000062
	7	1	1	1	1	1	1	1			

51

### APPENDIX III

#### LIST OF COMPUTER PROGRAM SYMBOLS AND DEFINITIONS

- A Coefficient of the  $x^2$  term in the equation of a line on a constant life fatigue diagram where minimum stress is  $x$  and maximum stress is  $y$ . ( $R = Ax^2 + Bx + C - y$ )
- AA An assigned value of +1. or -1.
- AAA A stress used in the calculation of plastic strain.
- ABDIF The absolute value of DIF.
- ABM The absolute value of ASMAX or of ASMIN, as assigned.
- ABMAX The absolute value of ASMAX.
- ABMEAN The absolute value of ASMEAN.
- ABMIN The absolute value of ASMIN.
- ABR4 The absolute value of R(4).
- ABR7 The absolute value of R(7).
- ABS The name of a routine calling for the absolute value of a quantity.
- AKT Stress concentration factor,  $K_t$
- ASE Array of values of ENN, for the plotting subroutine.
- ASMAX The product (AKT) (STMAX)
- ASMEAN The quantity (ASMAX + ASMIN)/2
- ASMIN The product (AKT) (STMIN)
- ASN Array of values of ASMIN, for the plotting subroutine.
- ASX Array of values of ASMAX, for the plotting subroutine.
- AVSCMN Average value of SIGMIN over an interval.
- AVSCMX Average value of SIGMAX over an interval.

- B Coefficient of the x term. (See A.)
- BBB A stress used in the calculation of plastic strain.
- C The constant. (See A.)
- COFMAN Inverse of the Coffin-Manson slope.
- CYCINT The number of cycles in an interval.
- CYCLES The calculated number of cycles expected to be indicated on a constant life fatigue diagram for the applied combination of maximum and minimum stress.
- C1 The Residual Stress Relaxation Constant (See ENEP,)
- DAM Damage.
- DECK Decimal or real value of integer K after conversion.
- DEL2 A portion of a least-squares-method solution.
- DIF The difference between residual stress and equilibrium residual stress. (RES(I) - EQRES)
- DO2 A portion of a least-squares-method solution.
- DUMMY A variable used in the calculation of the number of cycles to be considered as an interval for relaxation determination.
- ELMOD The elastic modulus.
- EN The number of cycles from the beginning of the relaxation process to the end of the current interval.
- ENEP The number of cycles required for overload residual stress effect to return to within one-tenth of its original difference from equilibrium conditions.

$$( N_{ep} = C1 / (ABM)^{E1} (ABMEAN)^{E2} )$$

ENN The number of applied cycles at a load level.

ENNCYC The ratio of the number of applied cycles to the number of cycles to failure. ( ENN/CYCLES )

EPSD LCF strain intercept.

EQRES Equilibrium residual stress.

EX An exponential function depicting the relaxation of residual stress.

EXP The name of a routine calling for the exponential value of a quantity.

EXPO An exponent. The power of 10 which indicates the number of cycles to failure.

E1 } Residual stress Relaxation Exponents.  
E2 }

IFLOAT The name of a routine calling for integer-to-real conversion.

I A variable subscript.

IBLOCK The identifying number of a block the blocks being numbered consecutively from 1 to NBLOCK.

IFIX The name of a routine calling for real-to-integer conversion.

IN The number of steps input to the range pair counting subroutine.

IPRINT Value controlling the WRITE statements.

IRAIN A counter.

IRPCOM Value controlling entry into the range pair counting subroutine.

ISTEP      The identifying step number, the steps being numbered  
             from 1 to NLEVEL.

ITYPE      The identifying type number, the types being numbered  
             from 1 to NTYPE.

J            A variable subscript.

JA          Value of +1 or 0, as assigned for branch determination.

JB          Value of -1 or 0, as assigned for branch determination.

JJ          An index variable.

JJJ         An index variable.

JKL         An index variable.

K            An index variable.

KK          An index variable.

KPMAX      The number of steps output from the range pair counting  
             subroutine.

L            An index variable.

LMN         An index variable.

M            An index variable.

N            An index variable with values of N=4-7 indicating the  
             power of 10, and thus identifying a particular life cycle  
             curve.

NBLOCK     The total number of times to execute a block of loads.

NDECK      The number of data decks to be run sequentially.

NFLAG      An integer used as a counter.

NFLAG2     An integer used as a counter.

NLEVEL The total number of steps, or levels, of loads in a block.  
 NN A subscripted variable used to indicate which types of loads are experienced in which blocks.  
 NTYPE The total number of different types.  
 PLSTRA Plastic strain.  
 R Residue term in damage calculation.  
 RES Residual stress.  
 RNCYC The number of cycles for a level after exiting the range pair counting subroutine.  
 SIGMAX Maximum stress.  
 SIGMIN Minimum stress.  
 STMAX Maximum applied stress.  
 STMIN Minimum applied stress.  
 SUMDEL Summation of damage for a flight.  
 SUMENN Accumulated total of applied cycles. (Summation of ENN).  
 SUMNC Accumulated cycle ratio. (Summation of ENN/CYCLES).  
 SUMR Summation of  $R(N)$ ,  $N=4,7$ .  
 SUMRN Summation of  $nR(N)$ ,  $N=4,7$ .  $n=4$ .  
 SUMR2 Summation of  $R(N)^2$ ,  $N=4,7$ .  
 SUMR2N Summation of  $nR(N)^2$ ,  $N=4,7$ .  $n=4$ .  
 SUMR3 Summation of  $R(N)^3$ ,  $N=4,7$ .  
 SUMR4 Summation of  $R(N)^4$ ,  $N=4,7$ .  
 TITLE1, TITLE2 Identification of the source of the SN data.  
 TLL Tensile load limit.  
 TM1, TM2 Material type.

TTYS One-fifth of tensile yield stress.

TYS Tensile yield stress.

T1,T2,T3,T4,T5,T6,T7,T8 Test identifying information.

X Variable equivalent to SIGMIN.

Y Variable equivalent to SIGMAX.

APPENDIX IV

FATIGUE LIFE INPUT DATA FLR SEVERAL MATERIALS

MATERIAL	YIELD STRESS	STRAIN INTERCEPT	INVERSE OF SLOPE	LIFE, $10^I$	S-N LIFE COEFFICIENTS		
					A(I)	B(I)	C(I)
2024-T4	58.	0.4	-1.836	4	-.0020	.2091	62.6
				5	-.0032	.4366	51.4
				6	-.0035	.6207	42.2
				7	-.0042	.7003	36.1
2219-T851	55.	0.4	-1.836	4	-.0022	.2204	55.8
				5	-.0018	.3320	48.3
				6	-.0015	.4628	39.7
				7	-.0024	.6420	31.7
7075-T6	72.	0.4	-1.836	4	-.0020	.2801	71.7
				5	-.0022	.5154	56.3
				6	-.0014	.6141	44.6
				7	-.0013	.6838	38.1
RQC-100	125.	0.54	-1.493	4	0.0	.2136	98.3
				5	0.0	.2927	88.5
				6	0.0	.3669	79.1
				7	0.0	.4376	70.3
Man-Ten	55.	1.11	-1.667	4	0.0	.2257	63.5
				5	0.0	.3520	53.1
				6	0.0	.4669	43.7
				7	0.0	.5678	35.4
4340 Steel	160.	0.4	-1.836	4	-.0002	.2567	162.4
				5	-.0007	.5248	126.9
				6	-.0005	.5557	113.5
				7	-.0005	.5557	108.5
T1-6-4	158.	0.4	-1.836	4	-.0009	.2368	154.2
				5	-.0006	.4640	110.3
				6	-.0000	.4650	88.9
				7	.0001	.4752	84.2