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MEMORANDUM FOR PRR (Contractor/In-House Publication)

2 June 1999

FROM: PROI (TI) (STINFO)

SUBJECT: Authorization for Release of Technical Information, Control Number: AFRL-PR-ED-TP-FY99-0125
Jeff Mills, "Hydrocarbon Fuels Optimization"
Poster Session - HEDM Conference

(Public Release)

Hydrocarbon Fuels Optimization

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Introduction—

Hydrocarbon fuels, as a class, are investigated using the minimal set of parameters determinative of oxidizer-optimized specific impulse and common density-specific-impulse analogues. In this context specific families of promising candidate fuels can be systematically compared, the performance trade-offs among the relevant chemical and physical properties can be quantified, and the general characteristics of mission-tailored fuels can begin to be elucidated.

Initial Questions—

- Why do I sometimes get relatively poor theoretical performance from a high-energy hydrocarbon?
- Why is RP-1 (kerosene) such a good rocket fuel?
- What are the minimal set of fuel parameters that determine hydrocarbon I_{sp} ?
- What are the performance trade-offs among these parameters?
- How can I begin to use this information to search for the “optimum” hydrocarbon fuel and reduce “hunt and peck” methods?

General Concepts/Initial Approximations—

Characteristics of “Good” Rocket Fuels:

- High Heat of Formation
- High “Light-Atom” Content

Specific Impulse:

(Momentum Transfer from Combusted Ejecta)

For an optimal nozzle:

$$I_{sp} \equiv \frac{1}{g} \sqrt{2 (H_{chamb.} - H_{exh.})}$$

Sometimes approximated:

$$(I_{\text{sp}})_{\text{opt}} \propto \frac{1}{g} \sqrt{2 \Delta H_{\text{comb}}}.$$

or,

$$(I_{\text{sp}})_{\text{opt}} \propto \frac{1}{g} \sqrt{2 \Delta H_{\text{f}}}$$

ΔH_{comb} —stoichiometric combustion enthalpy
(per unit mass of exhaust products)
 ΔH_{f} —specific enthalpy of formation

More Realistic Model Rocket—

One-Dimensional, Adiabatic, Equilibrated, Isentropic Expansion from the Combustion Chamber

Reference Rocket Conditions for I_{sp} Optimized vs. LOX:

Sea level expansion:

$$P_{\text{chamber}} = 1000 \text{ psi}, P_{\text{exhaust}} = 14.7 \text{ psi}$$

“Vacuum” expansion:

$$P_{\text{chamber}} = 1000 \text{ psi}, \epsilon = 40$$

Sole determinative parameters of LOX-optimized specific impulse and its density variants:

$$(I_{\text{sp}})_{\text{opt}} \Leftarrow \Delta H_{\text{f}}(/g) \text{ and } r(\frac{H}{C})$$

$$(\rho_{\text{of}}^a I_{\text{sp}})_{\text{opt}} \Leftarrow \Delta H_{\text{f}}(/g), r(\frac{H}{C}), \rho_{\text{f}}, \text{ and } a$$

$\Delta H_{\text{f}}(/g)$ —specific enthalpy of formation of the fuel

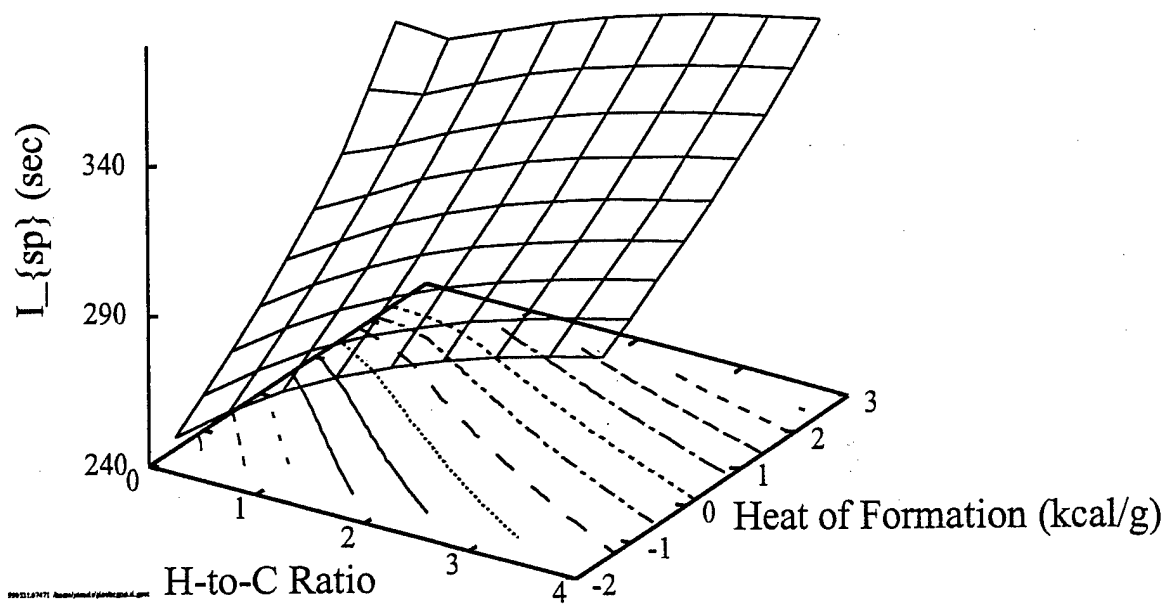
$r(\frac{H}{C})$ —atom ratio of fuel

ρ_{of} —oxidizer-fuel bulk density

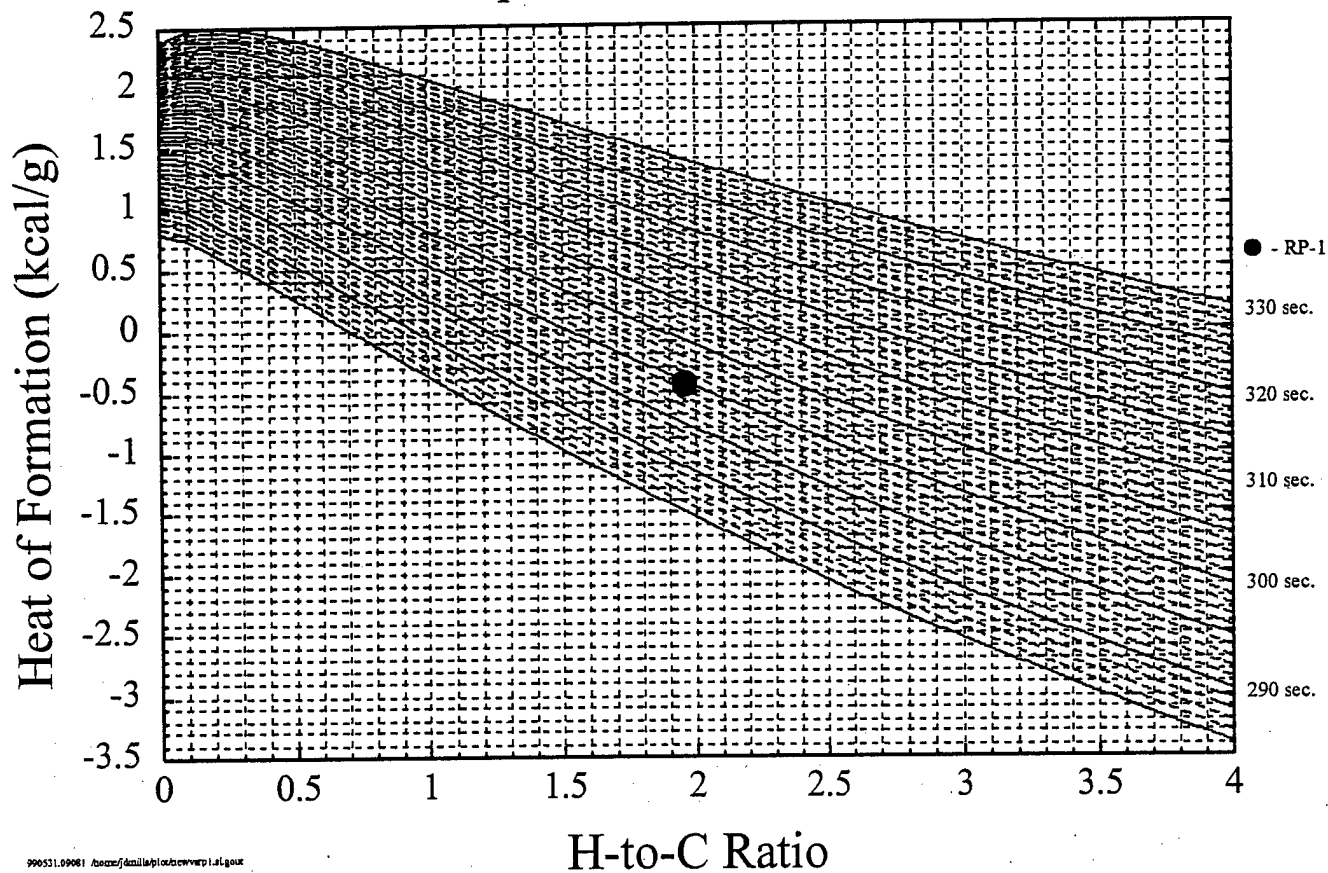
ρ_{f} —fuel density

a —density exponent (often, mission-specific constant)

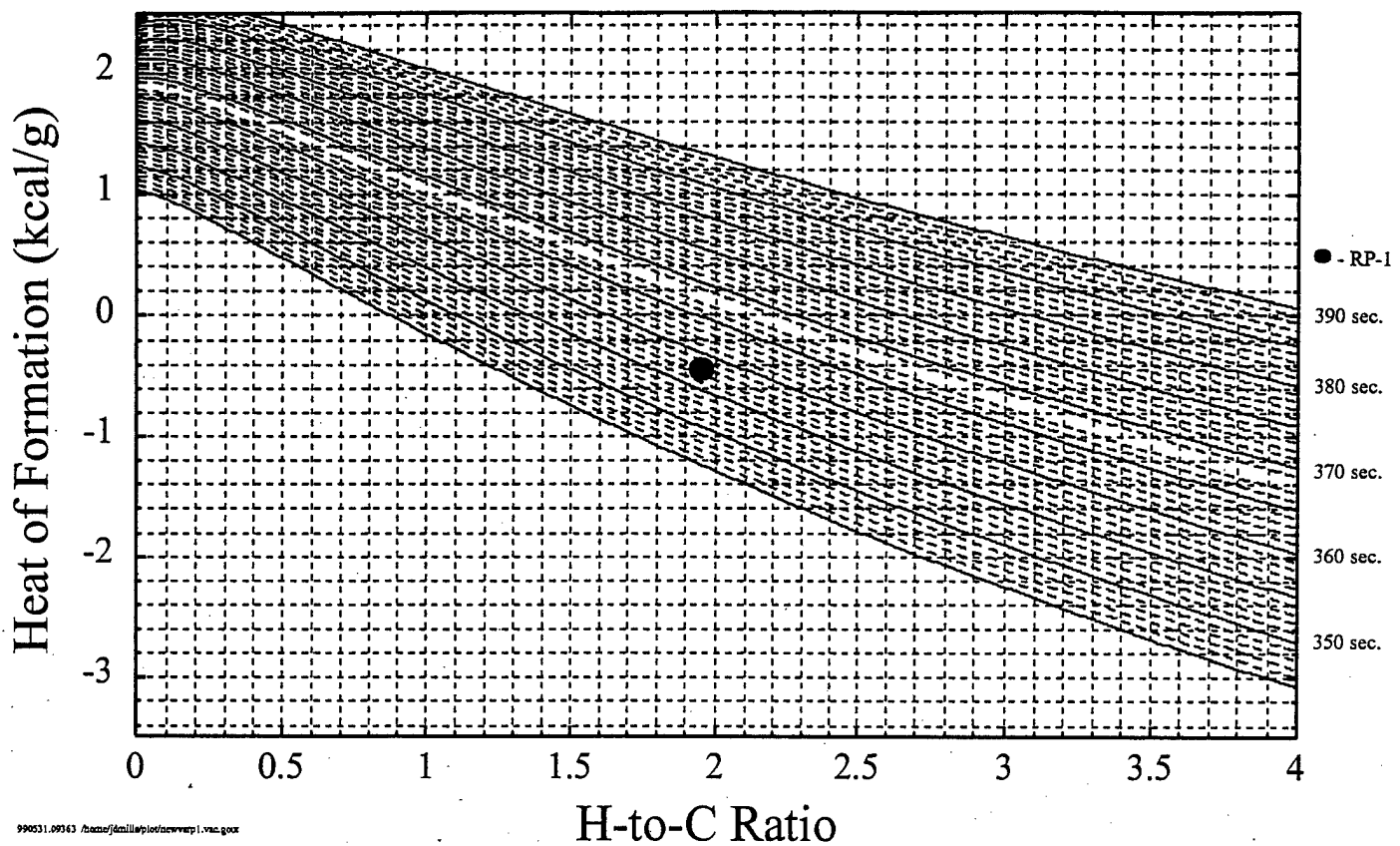
Optimum Hydrocarbon I_{sp} vs. LOX
Sea-Level Expansion (14.7 psi, $P_c=1000$ psi)



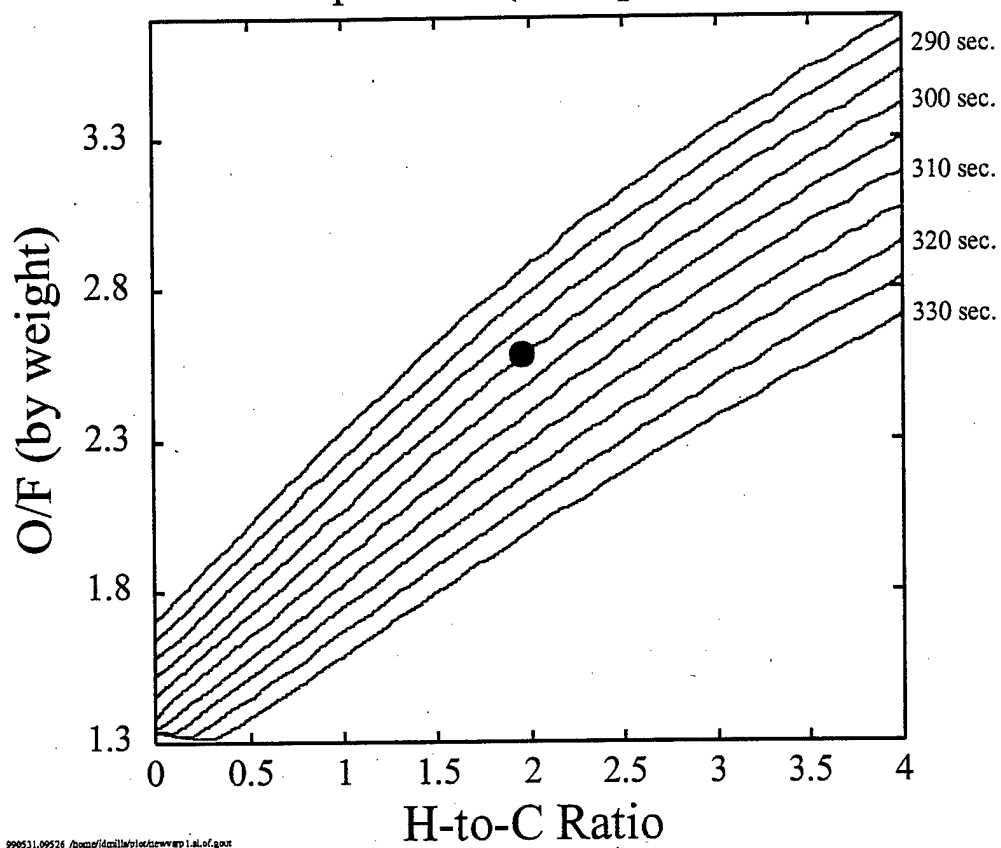
Optimum Hydrocarbon I_{sp} vs. LOX
Sea-Level Expansion (14.7 psi, P_c=1000 psi)



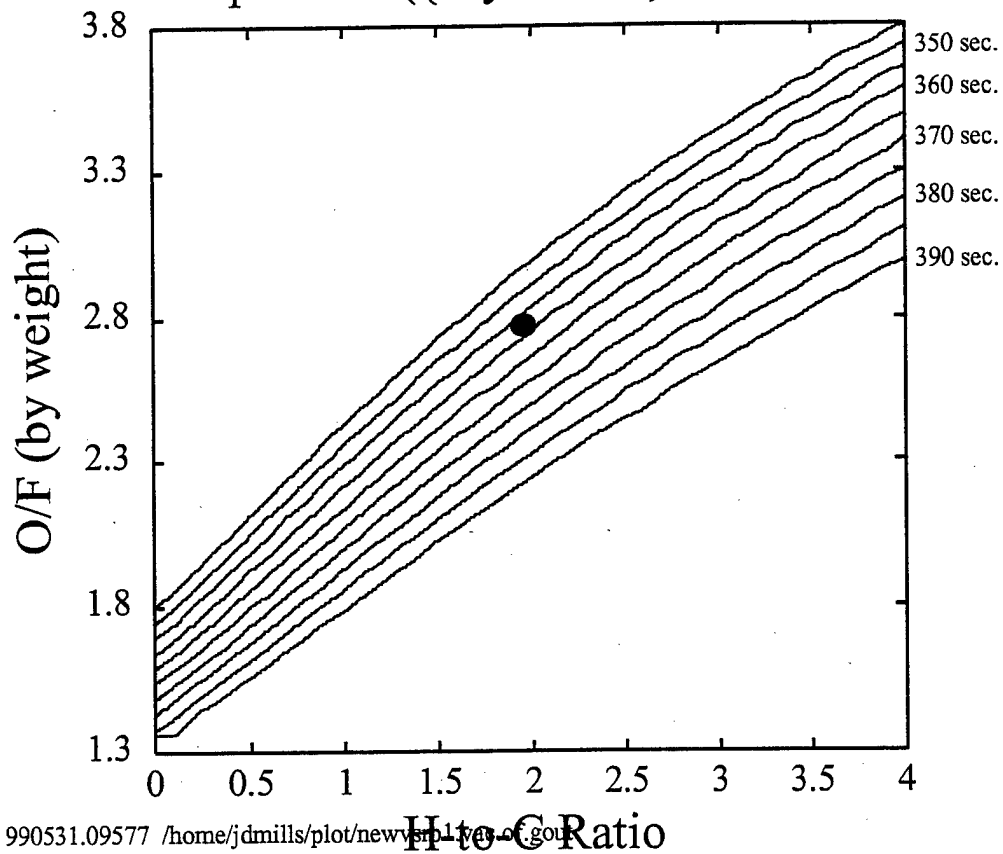
Optimum Hydrocarbon I_{sp} vs. LOX Vacuum Expansion ($\epsilon=40$, $P_c=1000$ psi)



O/F at Optimum I_{sp}
Sea-Level Expansion (14.7 psi, $P_c=1000$ psi)

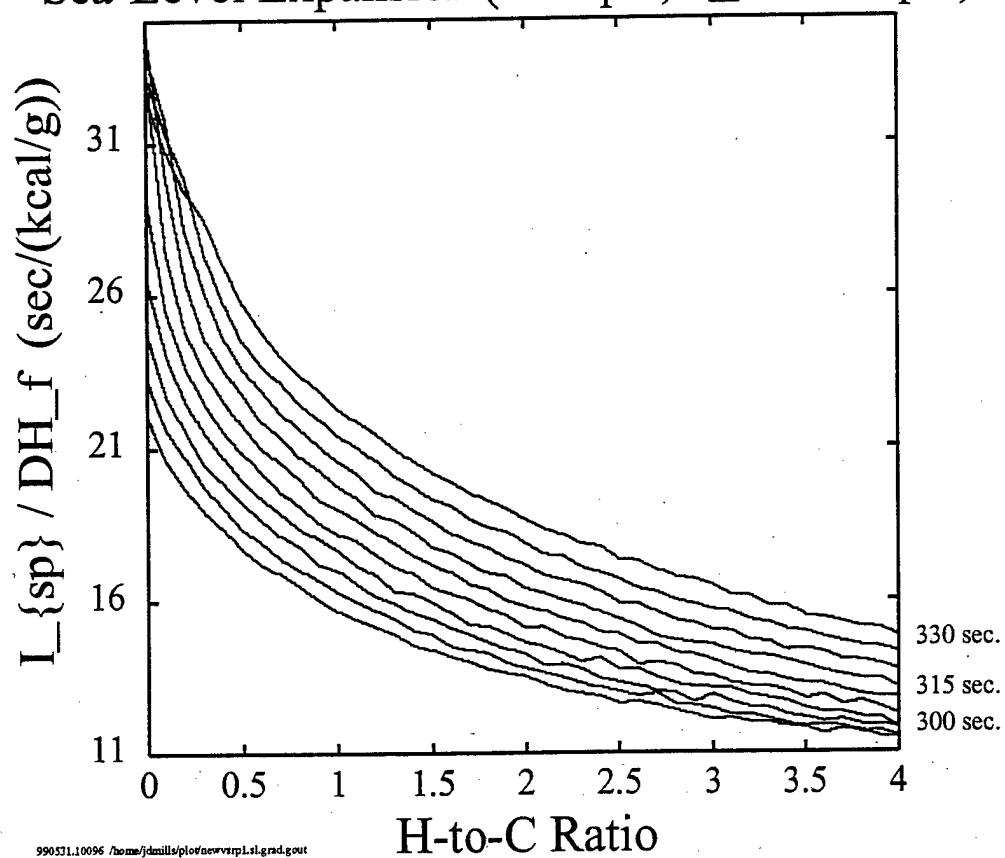


O/F at Optimum I_{sp}
Vacuum Expansion ($\epsilon=40$, $P_c=1000$ psi)

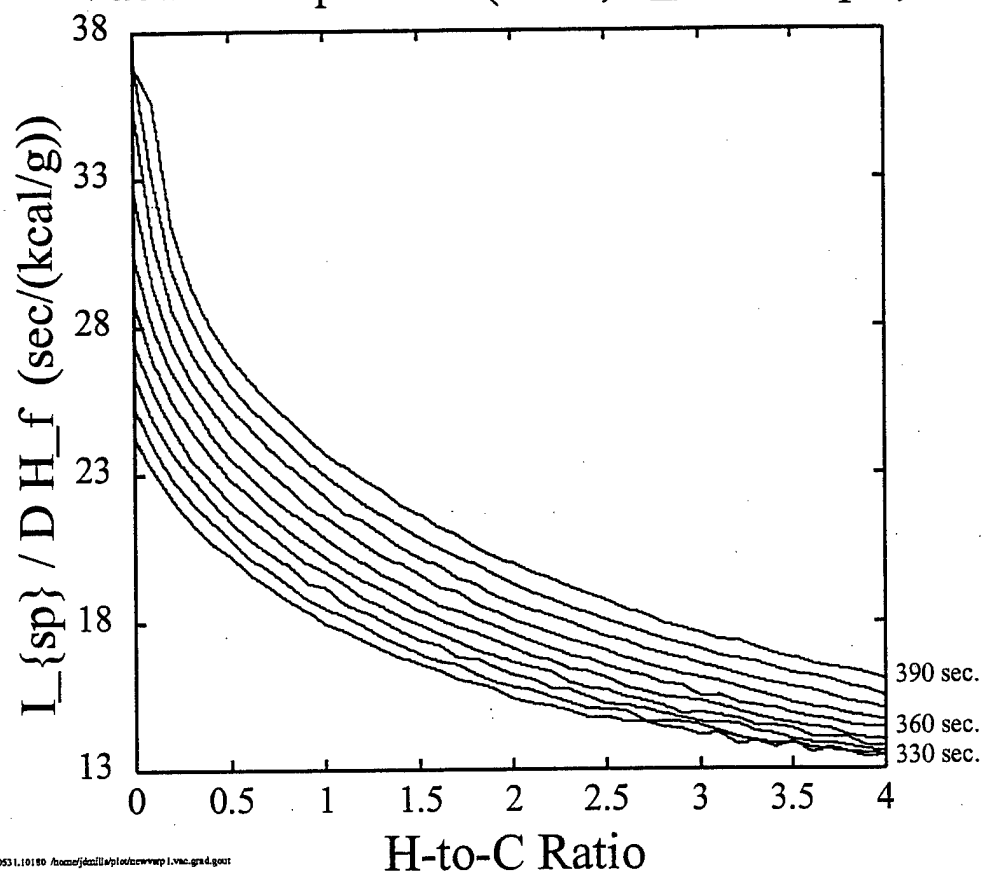


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Errors in Optimum I_{sp} with DH_f
Sea-Level Expansion (14.7 psi, $P_c=1000$ psi)

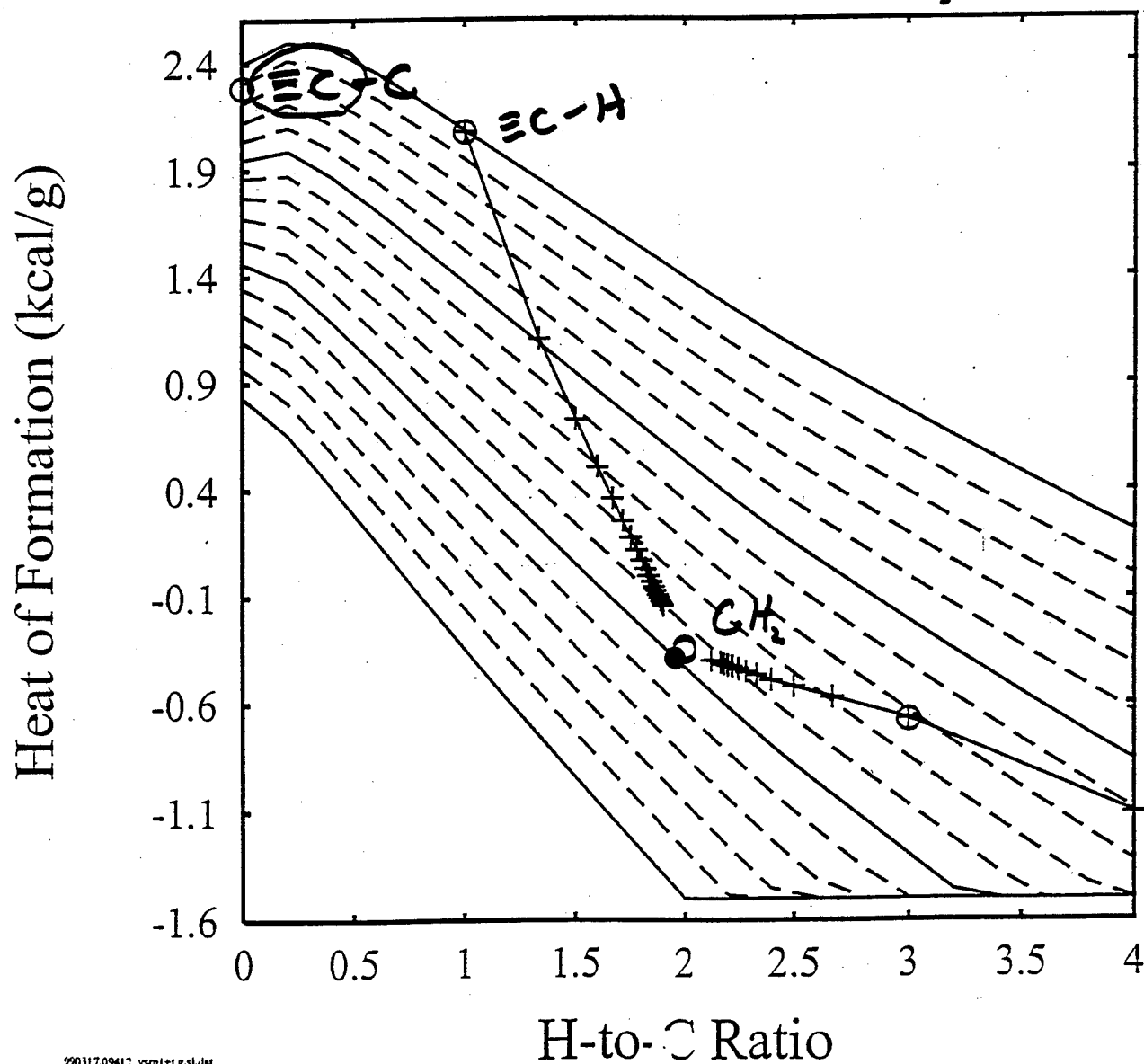


Errors in Optimum I_{sp} with DH_f
Vacuum Expansion ($e=40$, $P_c=1000$ psi)

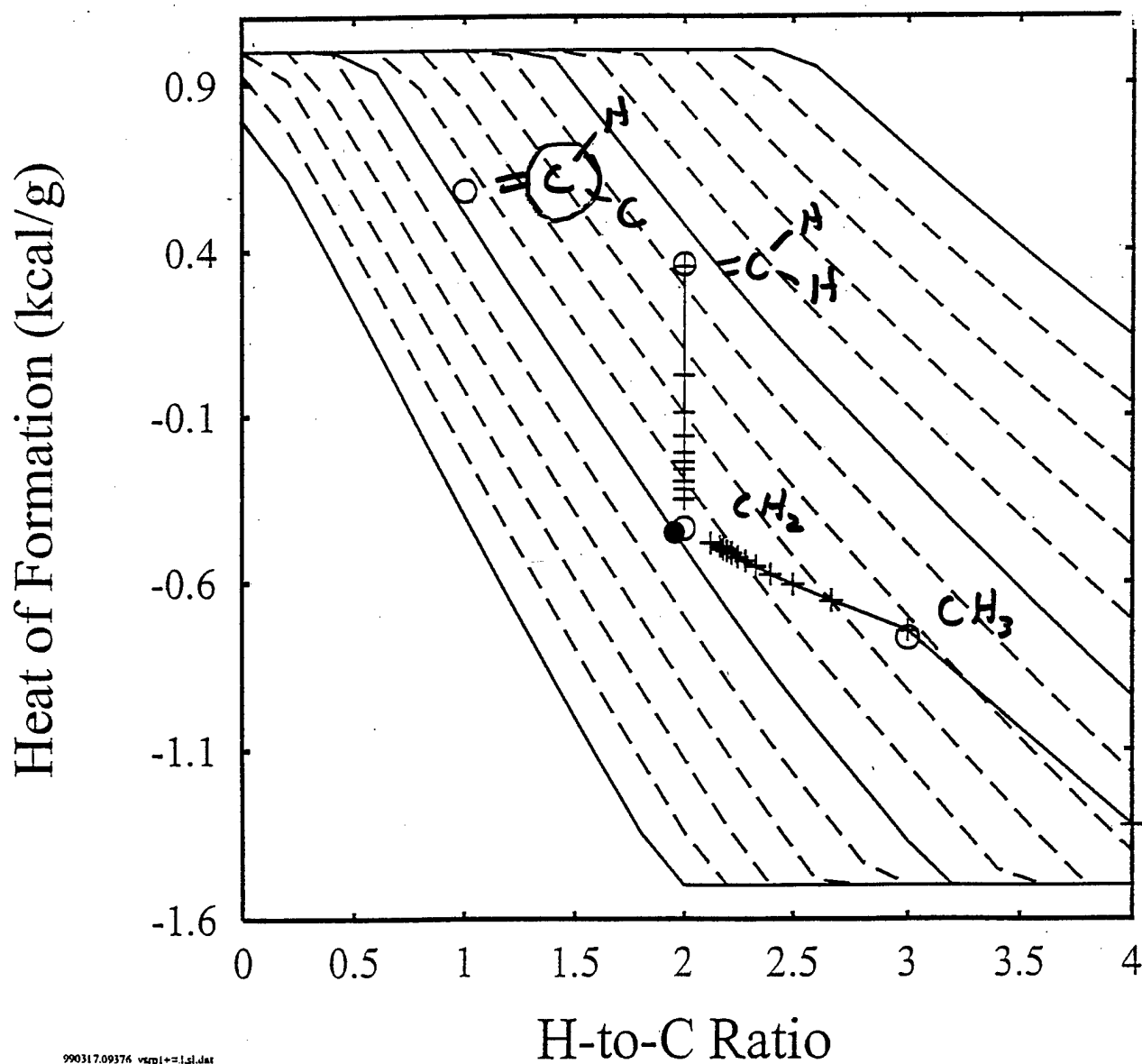


A Specific-Impulse Survey of Chemical Families—

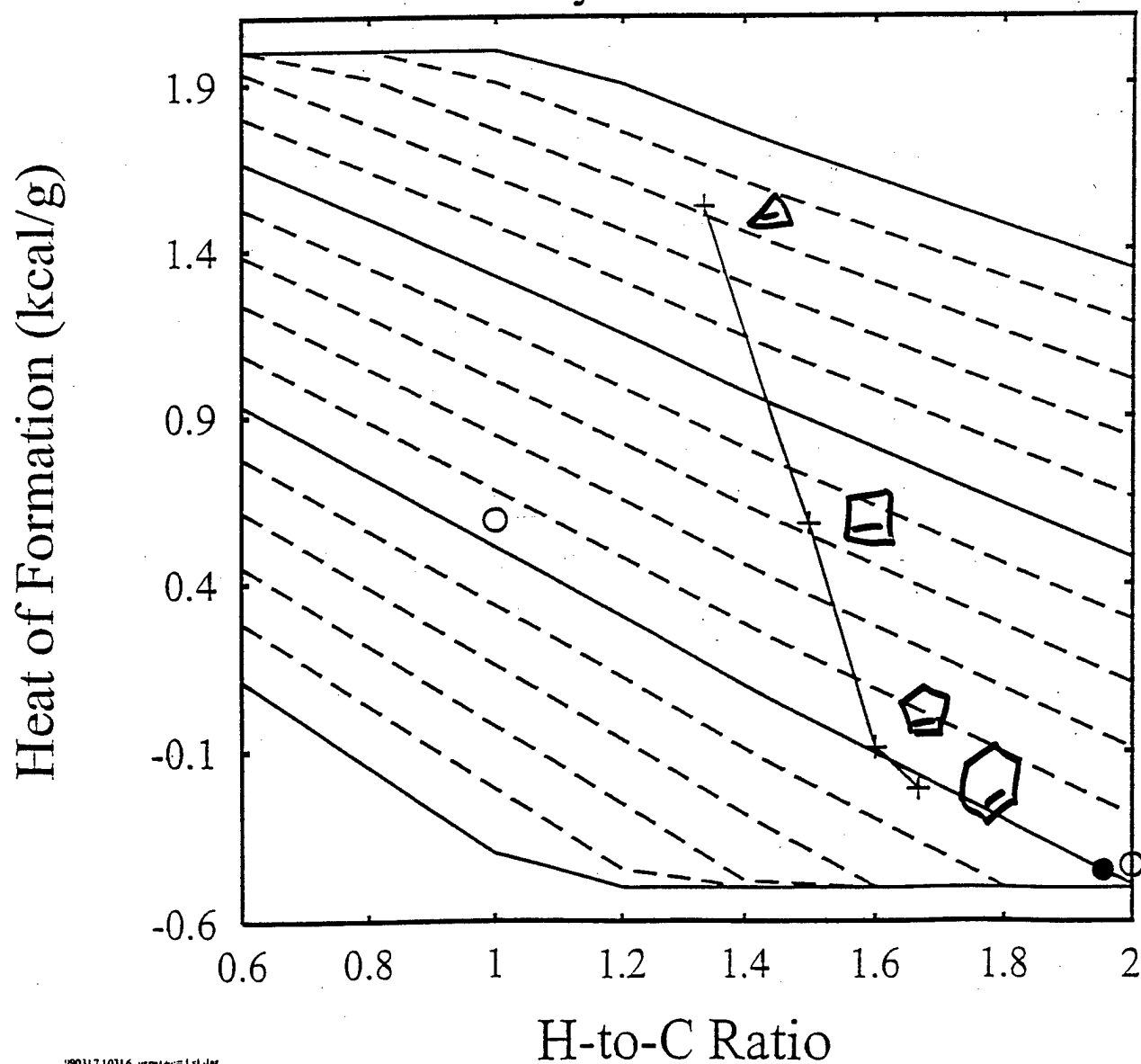
Gas, Sea-Level Hydrocarbon Performance Alkanes and Terminal Alkynes



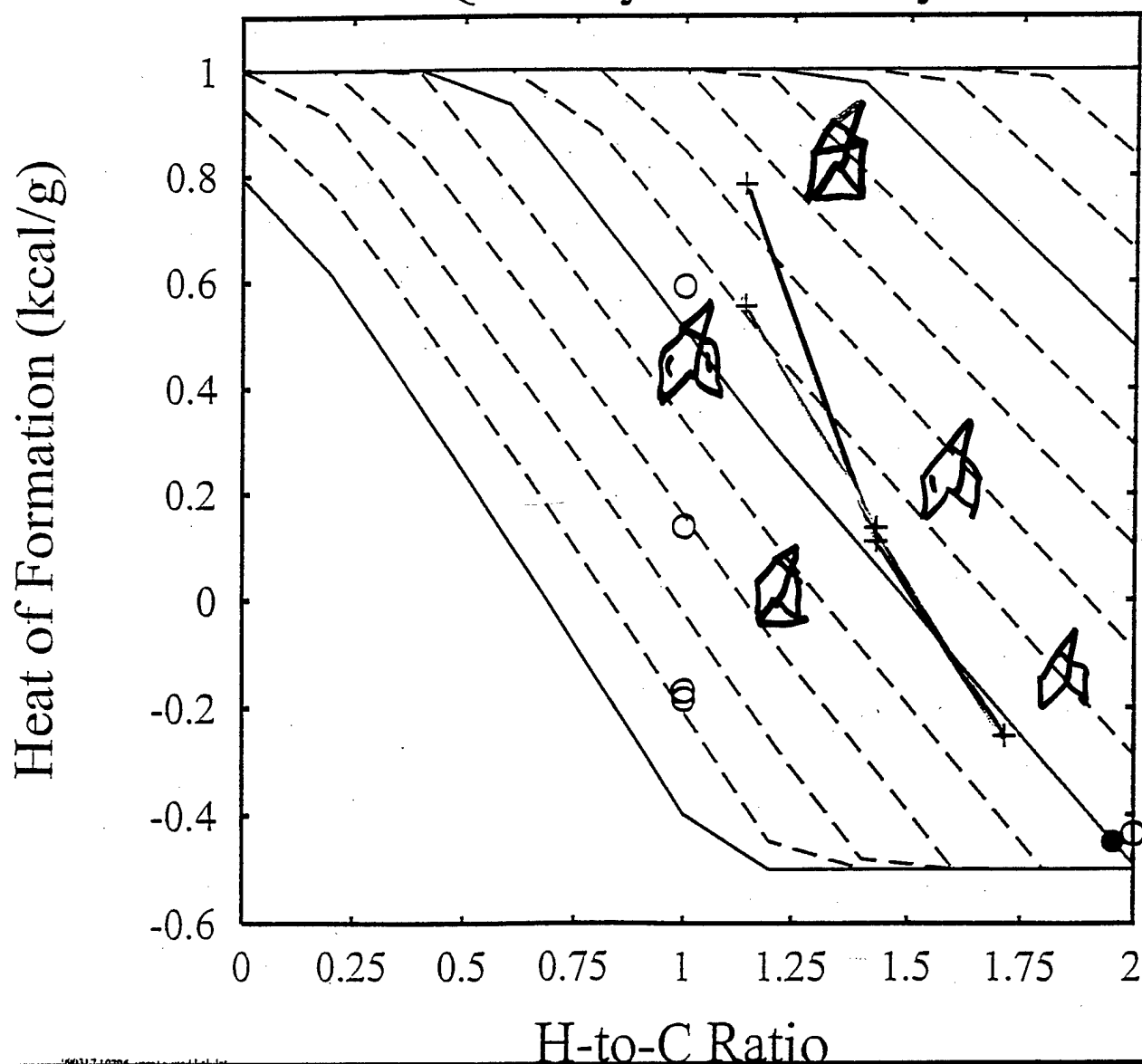
Liquid, Sea-Level Hydrocarbon Performance Alkanes and Terminal Alkenes



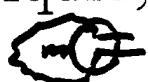
Liquid, Sea-Level Hydrocarbon Performance Cycloalkenes



Liquid, Sea-Level Hydrocarbon Performance Quadricyclane Family



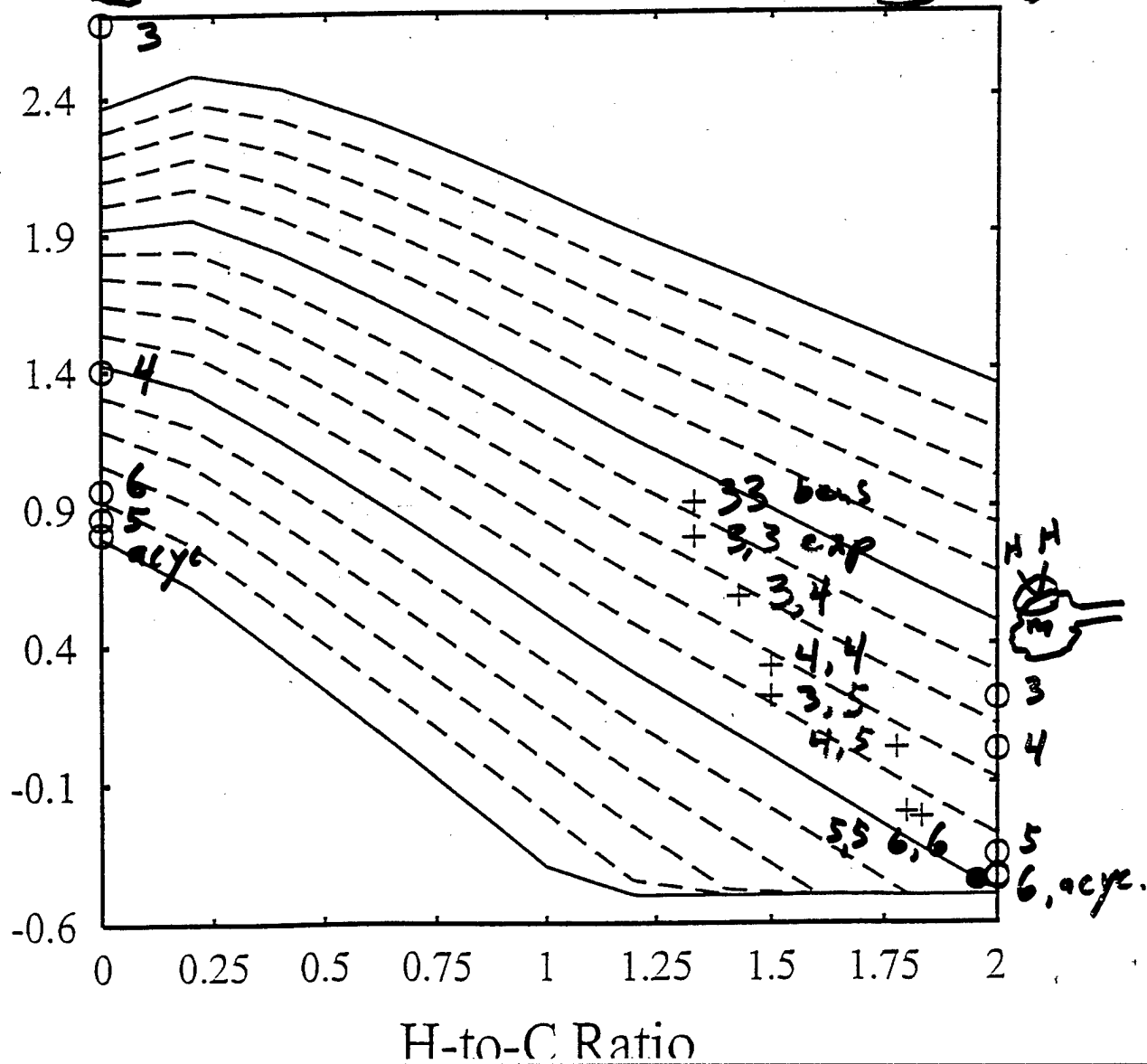
Liquid, Sea-Level Hydrocarbon Performance



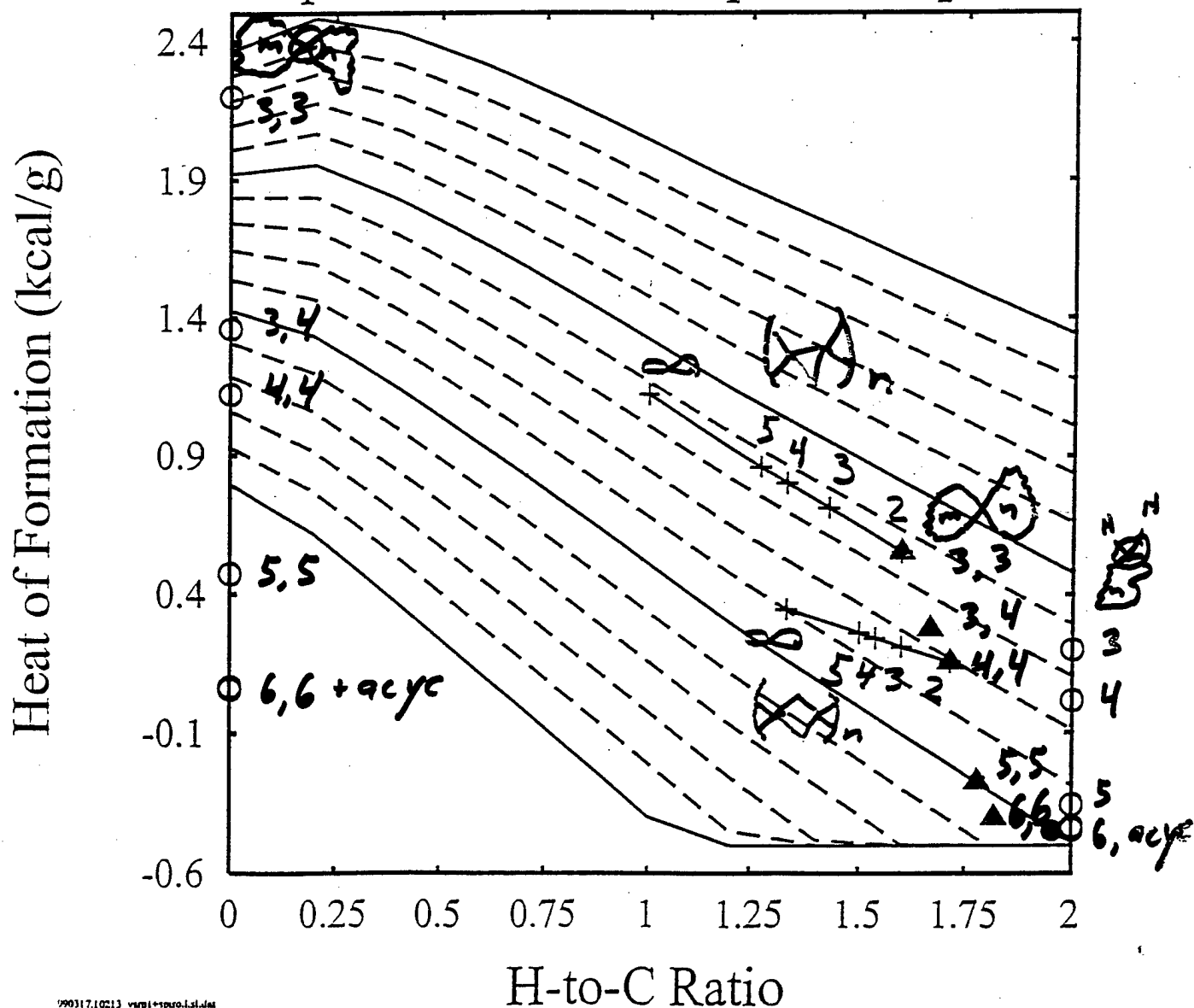
Double-Bow Molecules



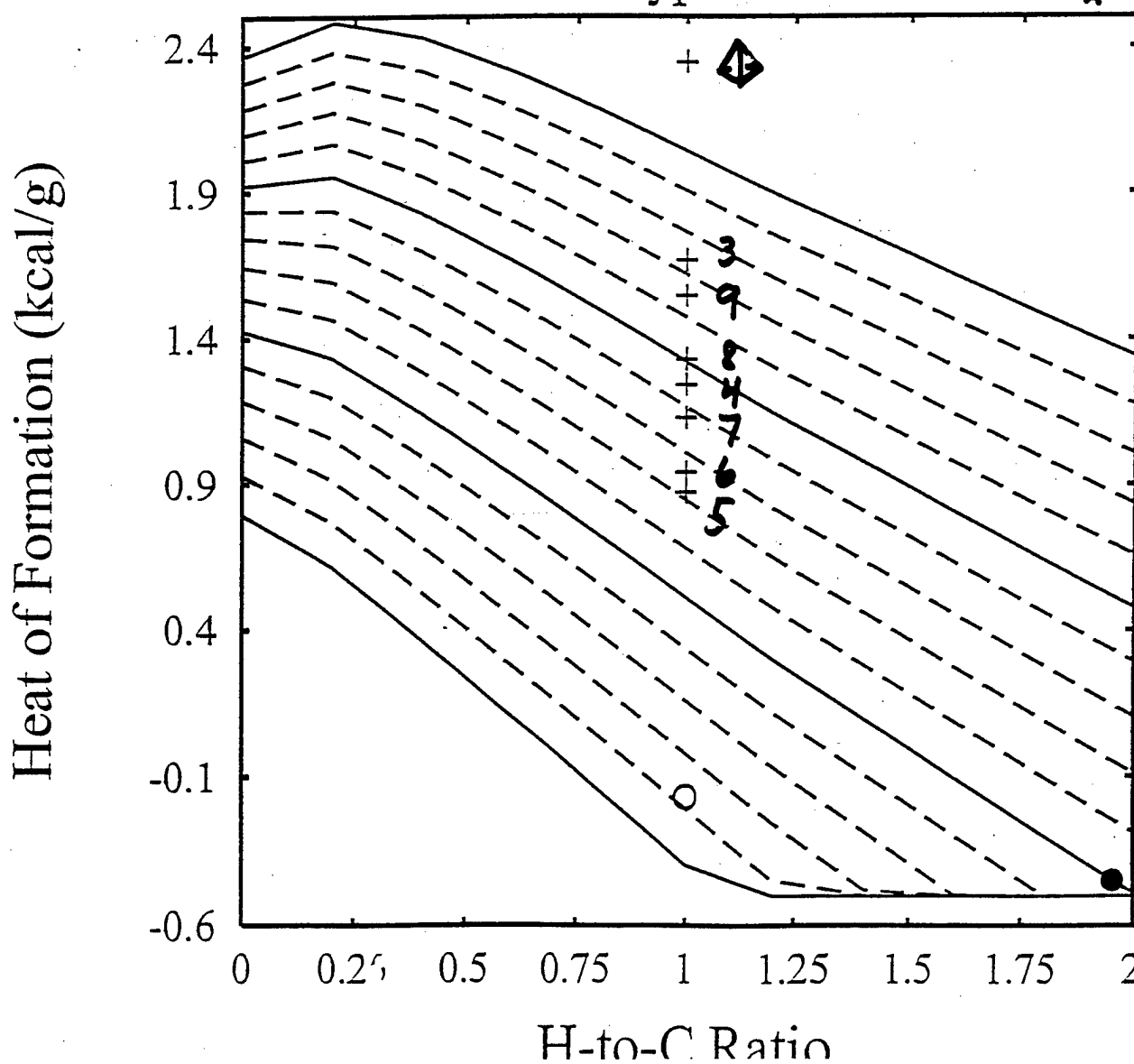
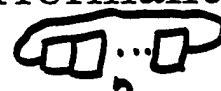
Heat of Formation (kcal/g)



Liquid, Sea-Level Hydrocarbon Performance Simple and Catenated Spiro Compounds



Liquid, Sea-Level Hydrocarbon Performance Prismane-type Molecules

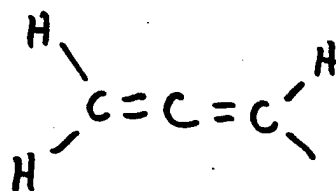


“Benson-Group” Specific Impulse—

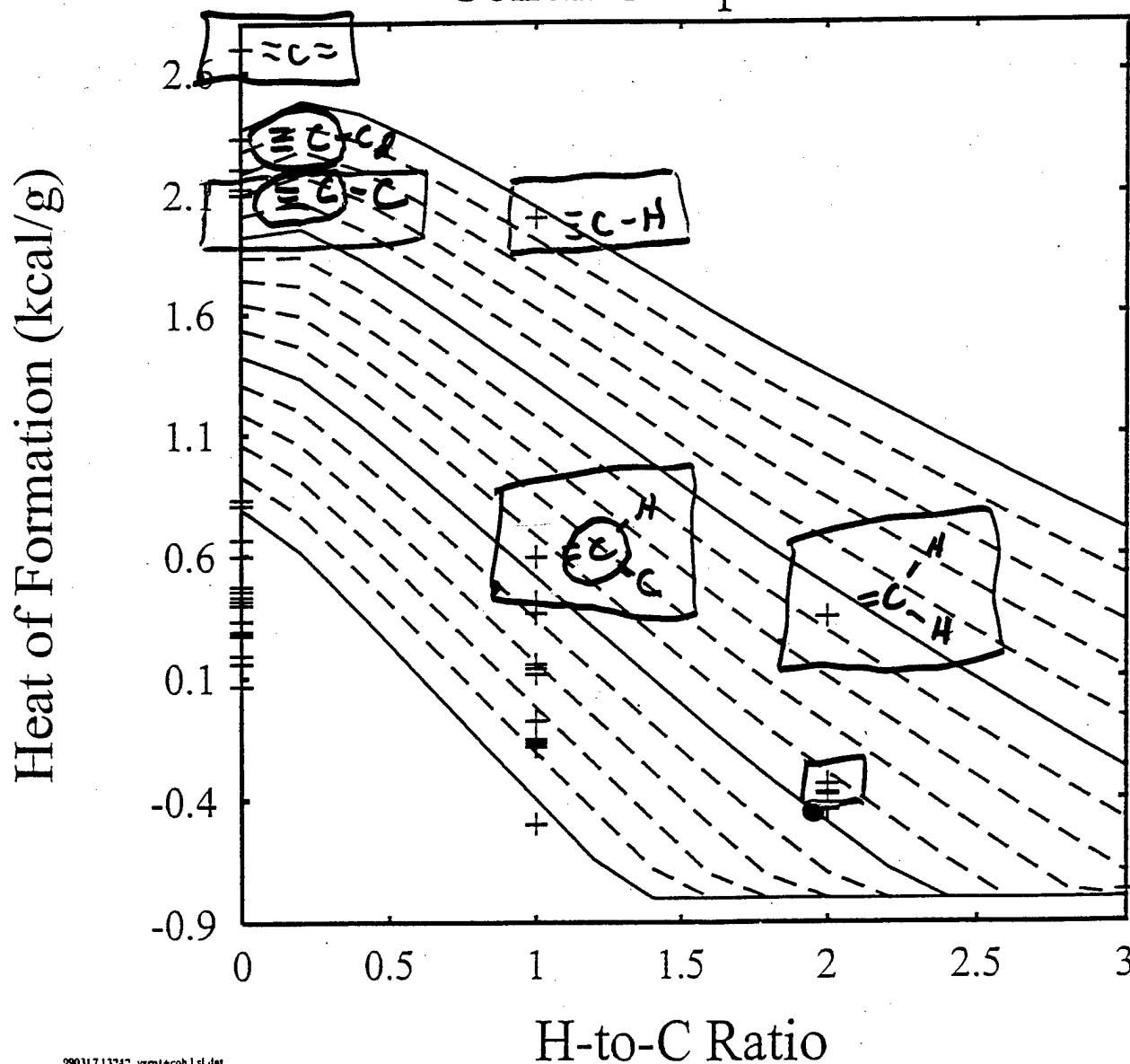
Additional Approximation: Benson-Group *Gedanken*
Rocket

$$(I_{\text{sp}})_{\text{opt}} \approx \sum_{\substack{i \\ \text{groups}}} \frac{m_i}{m_{\text{tot}}} (I_{\text{sp},i})_{\text{opt}}$$

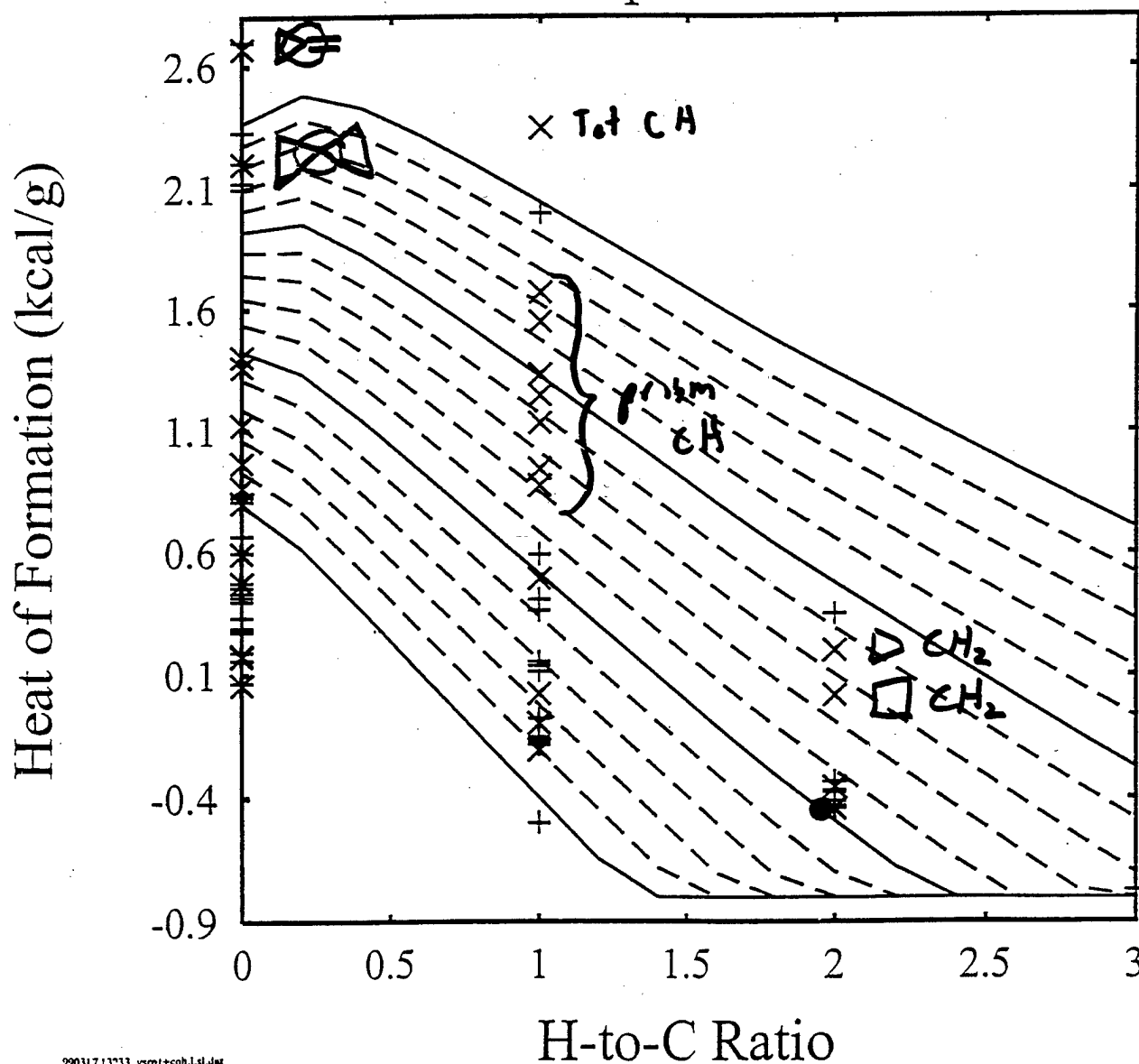
$\frac{m_i}{m_{\text{tot}}}$ —group mass fraction
 $(I_{\text{sp},i})_{\text{opt}}$ —optimized specific impulse for the chemical
group, alone



Liquid, Sea-Level Hydrocarbon Performance Cohen Group Values

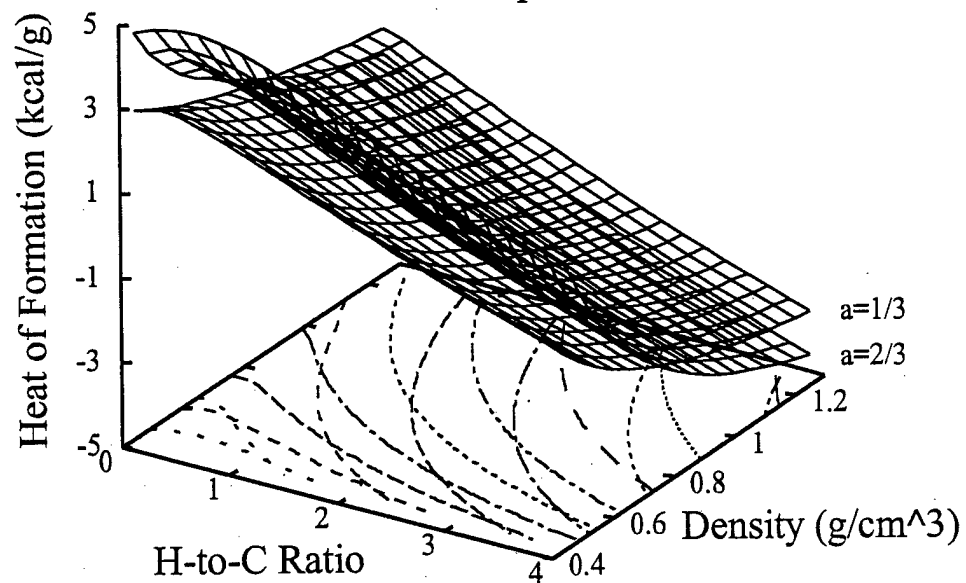


Liquid, Sea-Level Hydrocarbon Performance Cohen Group Values + Extras



Effects of Fuel Density—

Optimum Density I_{sp} vs. LOX
Sea-Level Expansion



Summary and Future Directions—

- Specific ΔH_f , $r(\frac{H}{C})$, and fuel density, alone, determine model, LOX-optimized rocket performance of hydrocarbon fuels
- Performance trade-offs among these parameters have been quantified and illustrated for ready reference
- Hydrocarbon fuels can be sought among known molecules in the parameterized performance space herein described
- The “Benson specific impulse” provides at least a semi-quantitative guide in the search for new, unknown rocket fuels

- Other chemical systems may be explored using these insights (C,H,N; doped H₂)

Acknowledgments—

These calculations were performed on local IBM RISC workstations using a modified version of the Air-Force specific-impulse rocket code written by C. Selph, R. Hall, C. Beckman, R. Acree, T. Magee and others.