



Combat Vehicle Engine Selection Methodology Based on Vehicle Integration Considerations

US Army RDECOM / TARDEC

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SUPERIOR TECHNOLOGY FOR A SUPERIOR ARMY

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Engine Selection Methodology Outline

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- ❖ Definition of the Propulsion System for Military Applications
- ❖ Advanced Integrated Propulsion System (AIPS) Power Pack
- ❖ Assessment of the Overall System Power Density Potential
 - ❖ Cooling System and Parasitic Fan Power Sizing
 - ❖ Inlet and Exhaust System Impact
 - ❖ Mission Fuel Determination
 - ❖ Propulsion System Volume Estimates
- ❖ Conclusions



Combat Vehicle Problem Power Dense Engine Not Sufficient

SUPERIOR TECHNOLOGY FOR A SUPERIOR ARMY

Need high power density of complete propulsion system *

- ❖ Engine
- ❖ Transmission including steering and brakes for tracked vehicle
- ❖ Cooling system
- ❖ Air filtration system
- ❖ Inlet and exhaust ducting
- ❖ Propulsion control system
- ❖ Accessory drive interfaces
- ❖ Batteries (for propulsion), wiring harnesses
- ❖ Fuel tanks and plumbing (sized for mission requirement)
- ❖ Final drives
- ❖ Maintenance access and clearances
- ❖ Unusable volume

* Power pack is that subset of the propulsion system that lifts or rolls out for replacement or periodic checks. Typically includes engine, transmission, air filtration, cooling and control systems.



Advanced Integrated Propulsion System (AIPS) Power Pack

SUPERIOR TECHNOLOGY FOR A SUPERIOR ARMY

- ❖ Began in 1982 to replace Abrams Main Battle Tank propulsion system

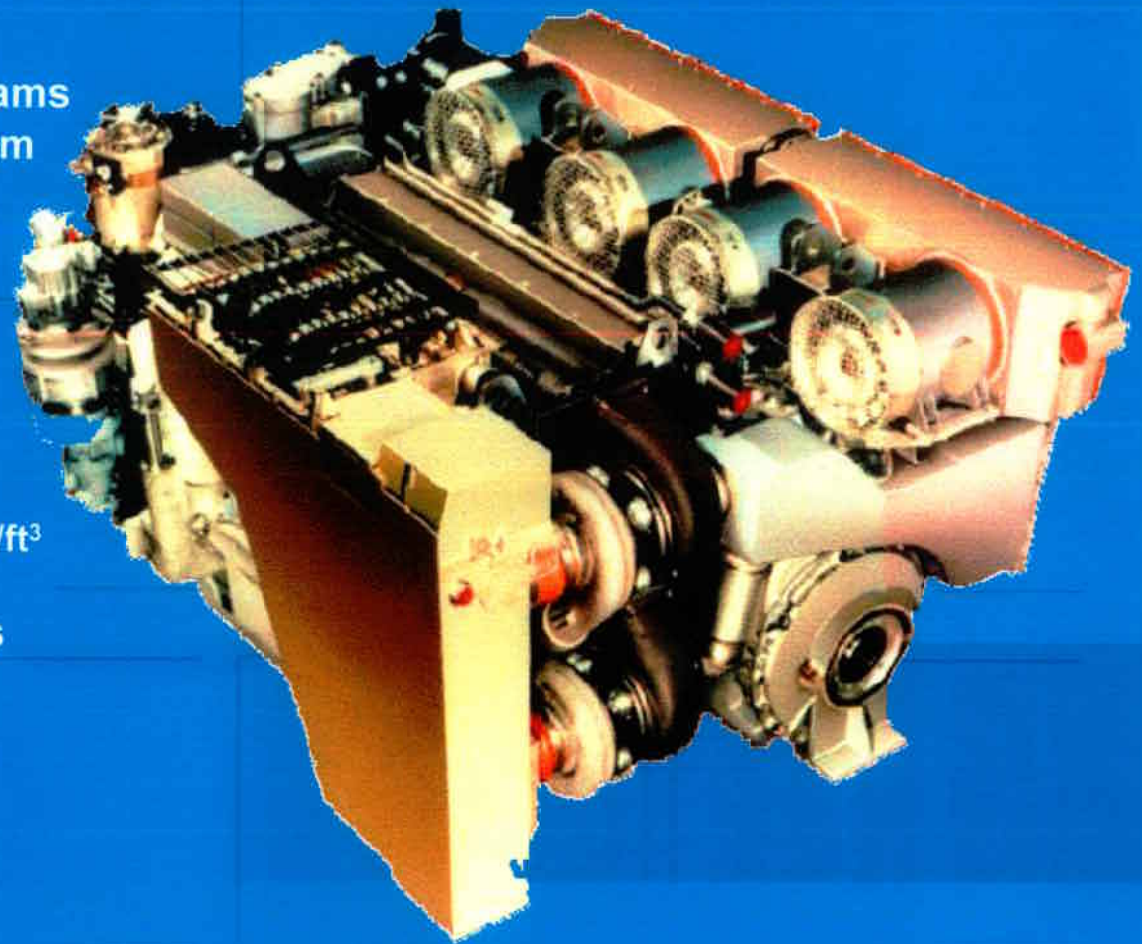
- ❖ Increased Power Density
- ❖ Improved Fuel Economy
- ❖ Improved Maintainability

- ❖ Power Density Comparison

- ❖ AIPS – 6 sprocket hp/ft³
- ❖ Abrams – 3.26 sprocket hp/ft³

- ❖ AIPS Evaluated Power Systems

- ❖ AIPS Turbine
- ❖ AIPS Diesel





AIPS Turbine and Diesel Concentration

SUPERIOR TECHNOLOGY FOR A SUPERIOR ARMY

- ❖ High Efficiency Components
- ❖ Dense Component Packaging
- ❖ Reduction of Parasitic Losses Throughout the System
- ❖ Engine Technologies Specific to the Diesel or Turbine types

Turbine AIPS Efforts

- ❖ Higher Turbine Inlet Temp
- ❖ High Recuperator Effectiveness
- ❖ Reduced Pressure Losses
 - ❖ Reduced Air Consumption
 - ❖ Improved Fuel Efficiency

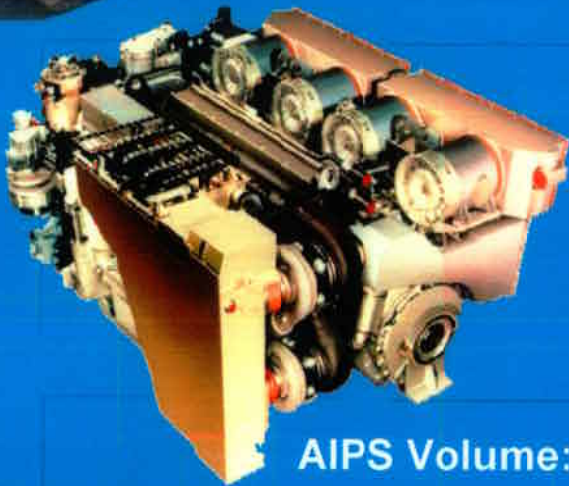
Diesel AIPS Efforts

- ❖ Improved Engine Power Density
- ❖ Improved Fuel Consumption
- ❖ Low Heat Rejection
- ❖ Higher Coolant Temp Technologies



Volumetric Comparison: AIPS and Abrams Engine

SUPERIOR TECHNOLOGY FOR A SUPERIOR ARMY



AIPS Volume: 170 ft³

transmission = 20%

fuel tanks = 23%

Exhaust = 1%

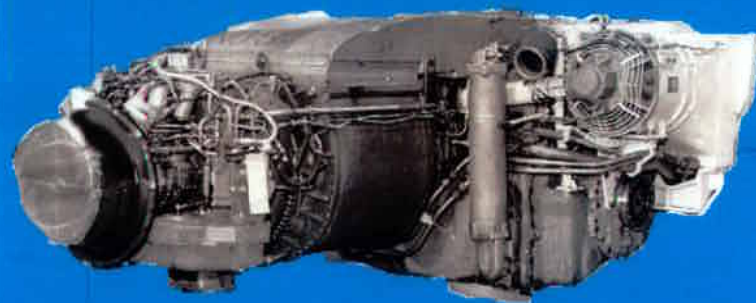
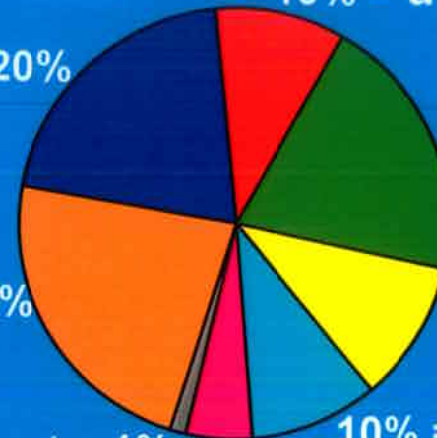
5% = air filter

10% = cooling

11% = batt/misc

20% = engine

10% = unused space



Abrams Volume: 291 ft³

transmission = 14%

fuel tanks = 26%

3% = exhaust

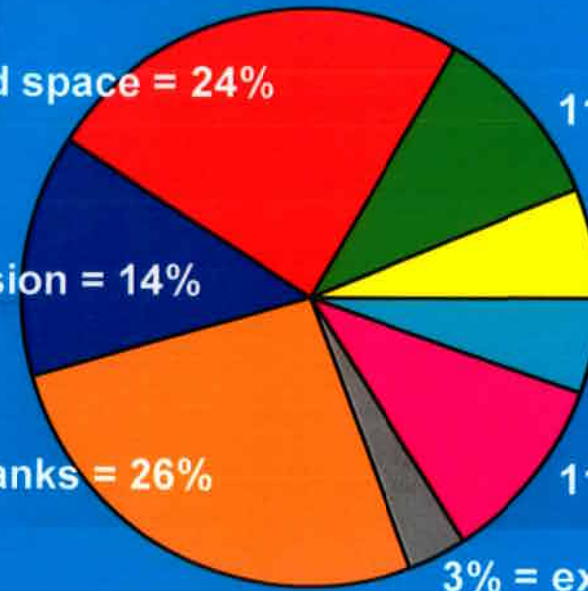
11% = air filter

5% = cooling

6% = batt/misc

11% = engine

unused space = 24%





Power Density Differences

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Primary Differences Between AIPS and Abrams Propulsion Systems

- Density of Packaging
- Air Consumption Differences
- Fuel Consumption Differences
- Heat Rejection Differences
- Parasitic Loss Differences

1 & 5 are primarily controlled by propulsion integrator and component supporters

2, 3 & 4 are primarily controlled by engine developer



AIPS vs. Marine Propulsion

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During AIPS development a new power dense diesel appears

AIPS Diesel

- ❖ 1500 rated HP @ 2600 rpm
- ❖ 12 Cylinders
- ❖ 28 Liter Displacement
- ❖ 4 Stroke / Cycle
- ❖ Single Stage VG Turbocharger
- ❖ Synthetic Oil Cooling Fluid
- ❖ 340° F Max Coolant Temp Out
- ❖ Air to Oil After Cooling
- ❖ 34 ft³ Engine Dunk Volume

Marine Diesel

- ❖ 1500 rated HP @ 1800 rpm
- ❖ 3 Cylinders
- ❖ 7 Liter Displacement
- ❖ 2 Stoke / Cycle
- ❖ 3 Stage Turbocharger
- ❖ Water / Glycol Cooling Fluid
- ❖ 230° F Max Coolant Temp Out
- ❖ Air to H₂O Jacket After Cooling
- ❖ 20 ft³ Engine Dunk Volume

New diesel smaller but system analysis shows AIPS diesel propulsion system more power dense based on cooling alone.



Manned Ground Vehicle Integration

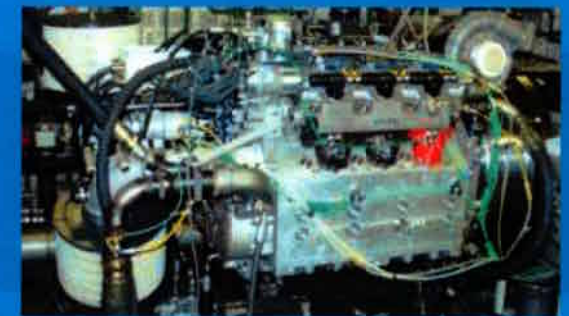
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Early 2000s, Engine Program Development

- ❖ Known need for high power density propulsion system
- ❖ Unknown platform characteristics
 - ❖ Weight?
 - ❖ Front or Rear Propulsion?
 - ❖ Tracked or Wheeled Vehicle?
 - ❖ Power Pack Shape or Size?
- ❖ Engine development only – remainder of power pack later



Simple methodology developed for system power density potential considering only the engine development





Methodology Descriptions

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Methodology addresses:

- ❖ Certain engine characteristics
- ❖ Required cooling system impacts
- ❖ Inlet & exhaust duct impact
- ❖ Impact of required onboard fuel

Methodology doesn't address:

- ❖ Potential tight packaging
- ❖ Opportunity for synergistic parasitic reductions

Results in:

- ❖ System volume estimate
- ❖ Power density estimate





Methodology Estimates Space Claims or Volumes

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Methodology involves estimating volumes for:

1. Engine
2. Transmission
3. Cooling System
4. Air Filtration System
5. Inlet & Exhaust Ducting
6. Controls
7. Miscellaneous
8. Batteries
9. Electrical Harness
10. Fuel System for Onboard Fuel
11. Final Drive
12. Clearance & Unusable Volumes

Sum of these = Propulsion System Volume

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Methodology Estimates Net Available Power

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Methodology estimates net available power

- ❖ Engine Gross Horsepower
- ❖ Subtract Estimates For:
 - ❖ Installation Loss
 - ❖ Air Filter Scavenge Fan (if any)
 - ❖ Power Loss Due to Induction & Exhaust Restrictions
 - ❖ Transmission Power Losses
 - ❖ Final Drive Power Losses
- ❖ To Arrive at Estimated Net or Sprocket Horsepower





Methodology Applied to 28 liter AIPS 4 Stroke vs. 7 liter Marine 2 Stroke

SUPERIOR TECHNOLOGY FOR A SUPERIOR ARMY

Engine	AIPS Diesel Hot Day 6 In Core	7L 2 Stroke Hot Day 5.3 In Core	7L 2 Stroke Hot Day 7.5 In Core
Engine / Application Characteristics			
Engine [hp]	1500	1500	1500
Engine Dunk Volume [ft ³]	34	20	20
Engine Specific Heat Rejection [Btu / hp·min]	20	40	40
Engine Induction Air Flow [lbs / hr]	14400	18500	18500
BSFC at Full Power [lbs / hp·hr]	0.37	0.39	0.39
Engine Heat Rejection [Btu / min]	30000	60300	60300
Vehicle Weight [tons]	60	60	60
Cooling System Size / Parasitic (Fan) Power Sizing			
Heat Exchanger Type: Oil to Air or H ₂ O to Air	Oil to Air	H ₂ O to Air	H ₂ O to Air
Power Pack Net Horsepower [hp]	1077.4	866.51	947.92
Fan Power [hp]	103.26	366.86	265.10
Cooling System Volume [ft ³]	25.80	64.15	60.31



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Engine Induction Air Flow [lbs / hr]	14400	18500	18500
BSFC at Full Power [lbs / hp·hr]	0.37	0.39	0.39
Engine Heat Rejection [Btu / min]	30000	60300	60300
Vehicle Weight [tons]	60	60	60
Inlet and Exhaust System Impact			
Air filter System Size [ft ³]	10	12.85	12.85
Inlet and Exhaust System Volume [ft ³]	2.06	2.64	2.64
Installation Loss (Intake and Exhaust Loss) [hp]	50.00	50.00	50.00



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Vehicle Weight [tons]	60	60	60
Mission Fuel Determination @ 60 Net hp*hr / ton			
Gallons Fuel (for 60 net hp-hrs/ton) [gallons]	266.91	349.81	319.77
Weight of Fuel [lbs]	1780.29	2333.21	2132.83
Volume of Fuel [ft ³]	35.68	46.76	42.75



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Engine Heat Rejection [Btu / min]	30000	60300	60300
Vehicle Weight [tons]	60	60	60
Propulsion System Volume Estimate			
Total Propulsion System Volume [ft ³]	177.32	224.24	212.19
Sprocket Power [hp]	1055.85	849.18	928.96
Propulsion Power Density [sprocket hp / ft ³]	5.95	3.79	4.38



Conclusions

SUPERIOR TECHNOLOGY FOR A SUPERIOR ARMY

- ❖ The most power dense engine doesn't always provide the most power dense system.
- ❖ Widely different prime power systems like diesel or turbine engines, fuel cells or alternative fuel engines can be fairly compared on a system to system basis.
- ❖ Similar approach can be used to evaluate other concerns like weight or affordability.





Thank You!



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