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HALON FLIGHTLINE EXTINGUISHER EVALUATION: DATA SUPPORTING STANDARD DEVELOPMENT [INCLUDES NOVEMBER 2007 ADDENDUM]

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Interim Report for 1 October 2004 – 14 November 2007

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Addendum 1

Additional Evaluations Conducted 26 October to 14 November 2007

Summary

In the winter of 2002, the Halon 1211 flight line extinguisher was evaluated for fire fighting effectiveness by a standard protocol developed at the Air Force Research Laboratory for engine nacelle fires. The extinguisher was found to be 90% effective at extinguishing fires in ambient temperatures ranging from 44°F to 77°F. In 2007, extinguishing agents selected as potential replacements for Halon 1211 were evaluated under the same protocol. Results of those evaluations called into question the effectiveness of Halon 1211 at higher ambient temperatures. Five additional fires were done with the Halon 1211 flight line extinguishers in October and November 2007 at temperatures of 78°F to 85°F. All five fires were extinguished in an average time of 18 s, and required an average of 61 lbs of Halon. These results were commensurate with results from the original series of evaluations done in 2002.

Introduction

During October 2007, evaluations were conducted of agents that were being considered as potential replacement agents for Halon 1211 in U.S. Navy and Air Force flight line extinguishers under a joint U.S. Navy and U.S. Air Force (USAF) test protocol for the Environmental Security Technology Certification Program. During the evaluations, it was noted that the extinguishing agents had a low success rate putting out standard protocol rear engine fires at ambient temperatures above 76°F, and none extinguished fires when ambient temperature was above 81°F. This led to questions about the efficacy of Halon 1211 at higher ambient temperatures. The original Halon flight line extinguisher evaluations were done in the winter of 2002 at ambient temperatures from 44°F to 77°F. Consequently, five rear engine fires were done using the Halon 1211 flight line extinguisher at ambient temperatures ranging from 78°F to 85°F. This addendum is a report of the results of those five trials.

Methods, Assumptions and Procedures

The methods and procedures used to accomplish these trials were the same as those used for the original evaluations done in 2002, with the following exceptions:

- Fuel was flowed through nozzle #2 (low pressure turbine) to preheat the test nacelle, not nozzle #3 (afterburner), because experienced showed that preheating was quicker and more uniform when nozzle #2 was used.
- Nacelle temperature was measured with a hand-held infrared thermometer for this series of evaluations instead of an installed thermocouple.

- The flight line extinguisher was set on a load cell and weight was measured and recorded every tenth of a second during each trial to determine the variation of flow rate with time. For the 2002 experiments, extinguisher weight was measured only before and after each trial.

Three video cameras were used to capture visual records of the fires, and actual extinguishment times were determined from the video records, as were total times that the firefighters actually discharged Halon. Cameras were used to capture views of the tests from multiple vantage points to counteract obscuration of flames by smoke or physical obstacles, then the multiple views were observed to determine the last instant that any flames appeared in any view, which made determining how long it took to extinguish each fire accurate and verifiable. Video recordings were used in conjunction with the data measured by the load cell to distinguish the actual amount of Halon used to extinguish each fire from the total amount expended for each fire.

Halon 1211 is an ozone depleting substance. The number of trials was set at five so that enough data could be gathered to make a statistical comparison while minimizing the release of Halon 1211 for test purposes.

Extinguishment time was the chosen parameter for comparing the two data sets. Comparing the total amount of extinguishing agent used in each fire was not a good method of comparison because when fire fighters extinguish a fire, they continue to apply agent for a period of time after the fire appears to be out to insure the flames are extinguished. It is not a fixed period of time for a particular firefighter, or for fire fighters as a group; therefore, it is not a good method of comparison. Also, there wasn't any 2002 data for the amount of agent expended up to the point that each fire was extinguished to compare to the 2007 evaluations because extinguisher weight was not measured during fires in the 2002 evaluations. Therefore, comparing the agent expended in both series was not possible. In both series, video records were used to determine how long it took to extinguish each fire, therefore, time to extinguish the fires was chosen as the parameter for comparing the two data sets.

Results and Discussion

Table 1 presents a summary of the results from the 2007 trials. All five fires were extinguished. Extinguishment time given in column three was the time it took to extinguish the fires as determined from video records, and the quantity discharged to extinguishment in column four is the corresponding weight of Halon 1211 as determined from the load cell data. Firefighters as a rule apply agent for a period after the fire appears to be extinguished to deter reflash, and this data is included for information only in columns five and six, total extinguisher discharge time and total quantity of Halon discharged. Table 2 presents data from the 2002 fire trials, and it is included for convenience of comparison. Only data for the 18 fires extinguished is included in Table 2 because there were no extinguishment times for the two tests in which fires were not put out.

Table 1. Summary of Data for 2007 Rear Engine Trials

Trial No.	Air Temp (°F)	Extinguishment Time (s)	Quantity Discharged to Extinguishment (lbs)	Total Extinguisher Discharge Time (s)	Total Quantity Discharged (lbs)
1	79	34	112	41	127
2	82	17	53	25	74
3	85	11	40	15	52
4	79	10	36	18	58
5	78	18	64	26	88
average	81	18	61	25	80
standard deviation	3	10	31	10	30

Table 2. Excerpt from Data for 2002 Rear Engine Trials

Trial No.	Air Temp (°F)	Extinguishment Time (s)	Trial No.	Air Temp (°F)	Extinguishment Time (s)
1	76	18	12	44	21
3	70	12	13	47	17
4	71	9	14	53	17
5	70	10	15	55	10
6	70	11	16	56	9
7	77	29	17	59	51
8	48	10	18	67	22
10	44	11	19	59	9
11	44	13	20	59	13
		average		60	16
		standard deviation		11	10

The fire fighter who did trials two through five in 2007 was not the same fire fighter who did trial number one. The firefighter for trials two through five had completed upward of 50 fires under the same test protocol and with extinguishing agents requiring a technique of use similar to the technique for using Halon 1211 just three weeks before these tests. The fire fighter in trial one had not done a similar fire in over three months. Using two fire fighters with very different proficiencies added a source of variation that should have been anticipated and avoided in tests. In consideration of this error, data from trial number one was excluded from subsequent analysis.

In Table 2, extinguishment time for 2002 trial number 17 (51 sec) differed from the average by 3.4 standard deviations. Assuming a normal distribution, there is only a 0.04% probability of this occurrence, and so it is reasonable to exclude this data point as an outlier. Therefore, data for 2002 trial number 17 was excluded from subsequent analysis. Sample statistics for the two series with data from 2007 trial number one and 2002 trial number 17 excluded is shown in Table 3.

Table 3. Sample Statistics for Both Data Series

	Air Temp (°F)	Extinguishment Time (s)	
Series	average	average	std deviation
2002	60	14	6
2007	81	14	4

The objective was to show that Halon 1211 was as effective at extinguishing the protocol fire at ambient temperatures of 78-85°F as it was at ambient temperatures of 44-77°F. A difference of means test for small samples using the t-distribution for these two data samples, excluding data from 2007 trial number one and 2002 trial number 17, yields a probability of 93.5% that Halon 1211 is as effective at extinguishing fires at ambient temperatures of 78-85°F as it is at ambient temperatures of 44-77°F.

This type of test is an attribute test or a go/no-go test, and the data follows a binomial distribution. For the original 20 fires conducted in 2002, 18 were extinguished, and the estimate of the true probability of success (p) in a 95% confidence interval is,

$$0.70 < p < 0.97.$$

Because it has been shown that Halon 1211 is equally effective over the full range of temperatures at which tests were done, Halon 1211 was successful 23 out of 25 times it was used to extinguish the protocol fire. For this combined data, the probability of success extinguishing the protocol fire with Halon 1211 in a 95% confidence interval is,

$$0.76 < p < 0.97.$$

Conclusions

Halon 1211 is equally effective suppressing JP8 fires over a range of ambient temperatures from 44°F to 85°F, and there is no indication that its fire fighting effectiveness drops with increasing temperature up to 85°F. Using the standard USAF flight line extinguisher containing Halon 1211 on the standard test protocol fire, a proficient fire fighter should expect to extinguish better than 75% of the fires.

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HALON FLIGHTLINE EXTINGUISHER EVALUATION: DATA SUPPORTING STANDARD DEVELOPMENT

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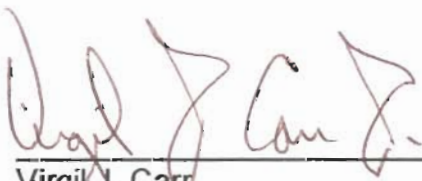
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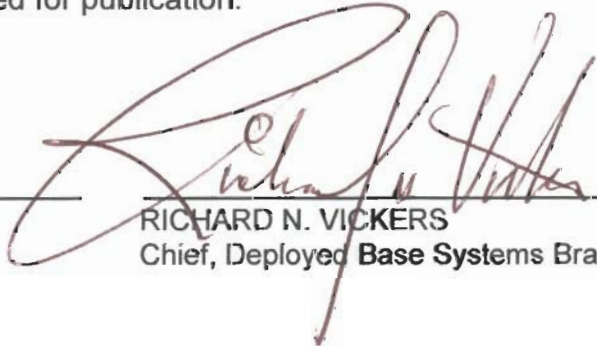
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Abstract

This evaluation of flight line fire extinguishers has been conducted to determine minimum performance criteria for assessing the capability of fire extinguishing agents to suppress two specific aircraft engine fires; three dimensional flowing fuel fires in the aircraft tail pipe and engine nacelle fires that can be fought through the access panel. This test series documented performance of 150 lb fire extinguishers containing Halon 1211 for extinguishing hidden and running fuel fires. An F100 engine nacelle mockup was used to evaluate the full extinguishment times and amount of extinguishing agent used on a series of twenty aft engine and pool fires of 100-ft² and ten access panel fires. The test series data was successfully used to develop a consistent, repeatable test procedure with pass/fail criteria for three dimensional and hidden fires through the rear engine and the access panel. The resulting document provides the performance standard for assessing alternative agents to determine suitability for use by USAF personnel in support of flight line fire extinguishers with the primary mission of rescue and fire suppression during aviation related incidents.

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Development of a Performance Standard for Flight Line Fire Extinguishers

Summary

Background

No official document is in place that establishes the performance standards for flight line fire extinguishers against three dimensional (flowing fuel) fires. US Air Force flight line aircraft fire protection must employ a "clean", 3-dimensional extinguishing agent to avoid harm to jet engines and to effectively extinguish hidden fires.

Recently, a vendor approached the United States Air Force claiming that current Compressed Air Foam (CAF) technology can meet or exceed USAF fire protection requirements at a lower cost than Halon 1211, the currently used standard. The vendor also proposed use of performance standards in lieu of agent/equipment specific standards for new procurements.

Using reliable test protocol the Air Force Research Laboratory evaluates potential clean 3-D fire fighting agents. Agents are tested against flowing fuel fires in F-100 engine mockup on pavement surface. The agent must extinguish both internal engine fire and surface fire.

Test protocol is necessary to ensure that flight line fire extinguishers meet the operational requirements. While no requirement document exists, interviews with experienced Air force firefighters have defined two primary scenarios, nacelle fires fought through access panels and aft engine fires fought from the tail of the aircraft. These scenarios have been reviewed with the Air Force Fire Panel, 11Jan02, and were accepted as valid. Any specification must ensure no increased risk to maintenance personnel, flight crew, (AFI-32-7086) or aircraft and maintain the ability to avoid Depot Maintenance following fire extinguisher application (T.O.2J-F100-46-2).

Purpose

The test series is required to develop a performance standard for flight line fire extinguishers. The standard provides for evaluating alternative agents to determine suitability for use by USAF personnel in support of flight line fire extinguishers with the primary mission of rescue and fire suppression during aviation related incidents. The plan establishes criteria, procedures, safety requirements, and responsibilities for conducting research testing for characterizing the performance standards required to put out 3-dimensional engine fires without obvious damage to the system or increased risk to the aircraft crew and maintenance personnel. The data obtained from the testing provides recommendations for testing firefighting agents on jet engines.

Scope

The tests establish data for development of a Performance Standard for Flight Line Fire Extinguishers. The Performance Standard includes requirements to extinguish fires occurring within the engine nacelle through provided access ports and to extinguish

pooled or flowing fuel tail pipe fires without increasing the risk to maintenance personnel, flight crew, or aircraft and maintaining the ability to avoid Depot Maintenance following fire extinguisher application. This project provides a standard for the experimental testing of fire fighting agents on 3D simulated engine systems.

These tests evaluated the following:

- Operational adequacy for flight line fire extinguishers on engine fires including:
- 20 Aft Engine Fires with flowing fuel in the F100 engine nacelle mockup with 100 ft² static pool fires
- 10 Access Panel Fires with flowing fuel

Fire Scenarios

Two fire scenarios were used for developing a Minimum Performance Standard for Flightline Fire Extinguishers. These two specific aircraft engine fires occur frequently on the flight line. The extinguisher must effectively extinguish fuel in a flowing state (commonly called 3-dimensional or flowing fuel fires) expected in engine tail pipe fires. Also, the extinguisher must apply agent through engine access ports to penetrate and extinguish fires occurring within the engine nacelle.

Aft Engine Fires

Halon 1211 fire extinguishers regularly extinguished the tail pipe fires with pool and running fuel. Of 20 fires, 18 were fully extinguished. Time to full extinguishment ranged from 8.75 to 50.93 seconds, averaging 16.25 seconds per fire. An average of 67.6 pounds of halon was used, ranging from 45.9 to 112.6 pounds.

Access Panel Fires

Eighty percent of the access panels were completely extinguished within a range of 8.18 to 46.16 seconds, averaging 17.29 seconds per fire. The average amount of agent used was 81.8 pounds, ranging from 57 to 131.95.

Introduction

Background

No official document is in place that establishes the performance standards for flight line fire extinguishers. Flight line fire extinguishers are the first line of defense on real or suspected engine fires. US Air Force flight line aircraft fire protection must employ a "clean", 3-dimensional extinguishing agent to avoid harm to jet engines and to effectively extinguish hidden fires.

Halon 1211 has been long recognized as the best agent for this task. Past agents included carbon dioxide, which produces thermal shock, and Halon 1011, which is less effective

and has a higher toxicity than Halon 1211. Firefighting foams have not **proven** to be effective 3-D firefighting agents on aircraft engine fires and produce corrosive residues. Therefore, engines that ingest these foams require removal from the aircraft and subsequent depot corrosion inspection (reference T.O. 2J-F100-42-2, WP 012 00).

Recently, a vendor approached the United States Air Force claiming that current Compressed Air Foam (CAF) technology can meet or exceed USAF fire protection requirements at a lower cost than Halon 1211. These "clean agent" air foam fire extinguishers would replace the current Halon 1211 flight line extinguishers. The vendor also proposed use of performance standards in lieu of agent/equipment specific standards for new procurements.

There is no existing test document for flight line fire suppression system performance for three dimensional (3-D) fire fighting capability. However, the Air Force Research Laboratory has test protocol to evaluate potential clean 3-D fire fighting agents. Agents are tested against flowing fuel fires in F-100 engine mockup on pavement surface. The agent must extinguish both internal engine fire and surface fire. This protocol has been used to evaluate many proposed clean agents, including C₆F₁₄ and CF₃I.

Test protocol is necessary to ensure that flight line fire extinguishers meet the operational requirements. A specification for 150lb Halon 1211 exists that requires extinguishment of 600 sq. ft. (UL 30A 240BC) pool fire, but there is no specification for extinguishing 3-D fires. While no requirement document exists, interviews with experienced Air force firefighters have defined two primary scenarios, nacelle fires fought through access panels and aft engine fires fought from the tail of the aircraft. These scenarios have been reviewed with the Air Force Fire Panel, 11Jan02, and were accepted as valid. Any specification must ensure no increased risk to maintenance personnel, flight crew, (AFI-32-7086) or aircraft and maintain the ability to avoid Depot Maintenance following fire extinguisher application (T.O.2J-F100-46-2).

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Scope

The tests establish data for development of a Performance Standard for Flight Line Fire Extinguishers. The Performance Standard includes requirements to extinguish fires occurring within the engine nacelle through provided access ports and to extinguish pooled or flowing fuel tail pipe fires without increasing the risk to maintenance

personnel, flight crew, or aircraft and maintaining the ability to avoid Depot Maintenance following fire extinguisher application. This project provides a standard for the experimental testing of fire fighting agents on 3D simulated engine systems.

These tests evaluated the following:

- Operational adequacy for flight line fire extinguishers on engine fires including:
- 20 Aft Engine Fires with flowing fuel in the F100 engine nacelle mockup with 100 ft² static pool fires
- 10 Access Panel Fires with flowing fuel

Methods, Assumptions and Procedures

Test Fixture

The F-100 engine nacelle test fixture was developed in response to the inability to determine detailed locations of engine fires. The test fixture is constructed according to the design published previouslyⁱⁱⁱ. Overall test responsibility rests with the Air Force Research Laboratory (AFRL/MLQD), Deployed Base Systems Branch Test Director.

The F100 engine nacelle mockup has baffles representative of engine nacelle obstacles and voids. Two-inch stainless steel strips alternate in two layers four inches apart. Four thermocouples are used for fixture temperature control, against the walls at the fore end, the aft end, and the center, plus one for ambient temperature in the center. Beneath the test fixture is a concave concrete test surface that is 11 feet in diameter, with center 1 foot diameter circle 3 inches lower than the rim.

Fuel is flowed through the mockup at four gallons per minute through FullJet Maximum Free Passage Spray Nozzles. There are three of these nozzles located in the test fixture.

Nozzle 3 was used to run fuel for the pre-heat phase of the test. The fuel was ignited and burned until the inside thermocouple read approximately 850 degrees. At this point the fuel flow was stopped and the fire was allowed to self extinguish. The temperature continues to increase as the metal absorbs heat. When the metal cools back down to 450 degrees, the fuel is turned on again for the actual test event.

For an aft engine running fuel pool fire, nozzles 2 and 3 run 25 gallons of fuel through the aft engine and into the concrete pan underneath the fixture, creating a pool. After this pool is complete the surface of the fuel is ignited. A 15 second pre-burn allows for the full pool and engine nacelle to become engulfed. The firefighter uses the fire extinguisher to approach and extinguish the pool fire and aft engine area.

The concentric tube design provides the hidden fire space for the Access Panel Test. After the preheat phase, Nozzle 1 is used to spray fuel into the engine nacelle. The fuel is immediately ignited and the 15 second pre-burn begins. The firefighter uses the fire extinguisher to penetrate and extinguish the engine nacelle.

Fire Extinguishers

Standard 150 lb flightline fire extinguishers containing Halon 1211 were used. This is a wheel-mounted fire extinguisher with a 150-pound capacity. It is manufactured by Amerex Corporation, part number 03496, and is suitable for use as a flight line fire extinguisher. The extinguishers were weighed immediately before and after each fire to determine the amount of agent used. Full extinguishment times were recorded. Extinguishers were serviced in between fires per T.O.13F4-4-121.

Firefighter Qualifications

All AFRL Fire Research firefighters are retired Air Force firefighters, with a minimum of 25 years of experience. All professional training in aircraft rescue and firefighting was received from the Air Force.

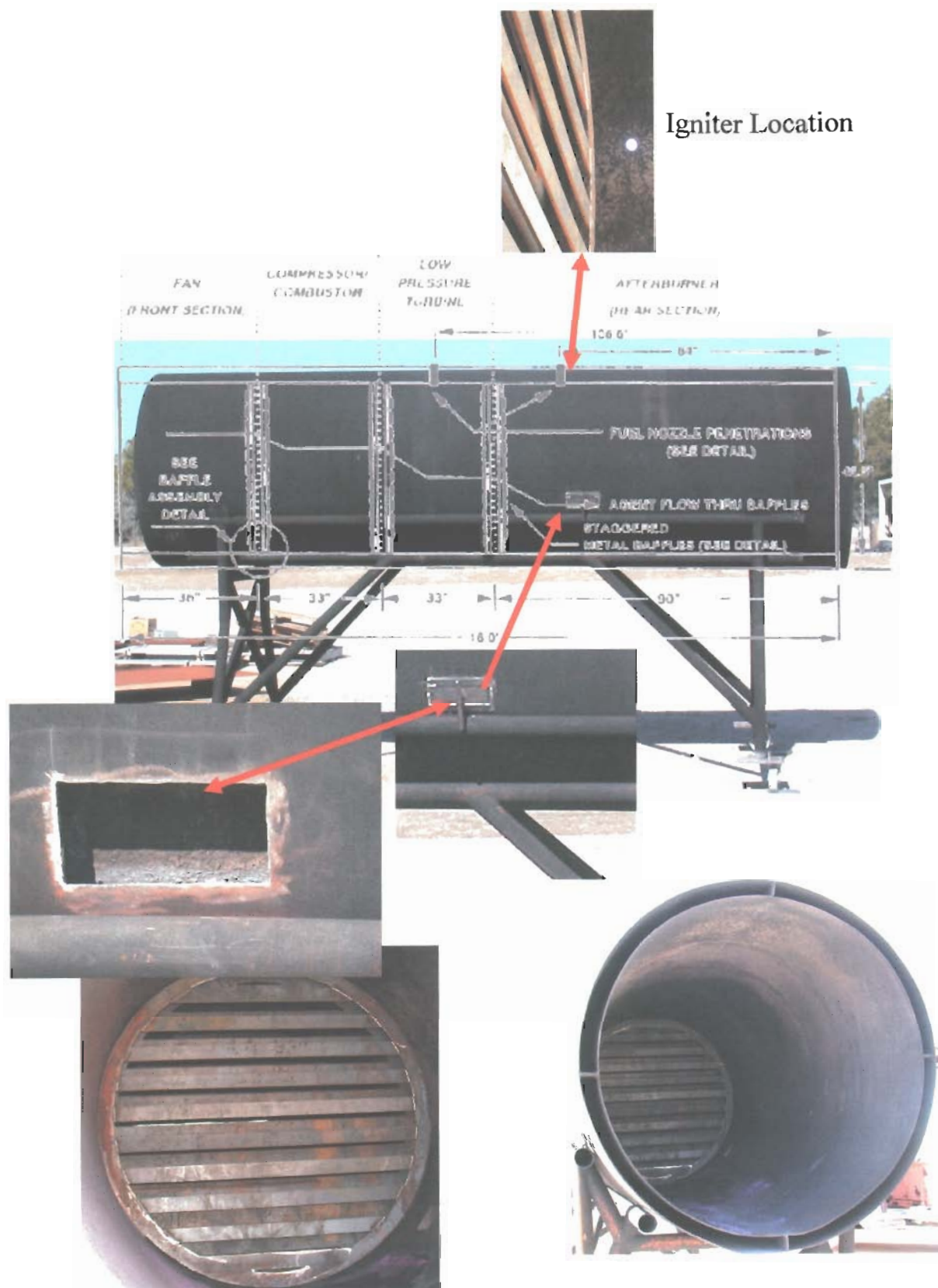


Figure 1. Engine Nacelle Mockup

Fire Scenarios

Aft Engine Fires

Description

The aft end of the test fixture was centered over the concrete pan so that the running fuel flows into the concave pool. The test fixture is lined up with the direction of the wind so that the wind travels from the back to the front of the fixture. When the test fixture reaches 450 degrees, JP-8 jet fuel is sprayed through nozzles 2 and 3 to fill the pool with 25 gallons. The extinguishers were fully serviced prior to each fire. A propane torch was used to ignite the JP-8 and a pre-burn of 15 seconds was conducted to assure full involvement of the fuel in the fire area. The firefighter was given a ten second countdown, at which time a steady attack mode was used to extinguish the fire.

Time to extinguishment was based on complete extinguishment of the pool fire and the fire inside the rear of the test fixture.

Criteria for Success

This test series did not correlate to any existing NFPA requirements; therefore, no maximum extinguishment times were established. The test was conducted to provide an estimation of current flight line fire extinguisher performance.

The series of tail pipe fires were used to gather data to develop performance parameters for flightline fire extinguishers. Data points for each test included type of fire, extinguisher number, wind speed, outside temperature, four thermocouple temperatures, amount of agent used, and full extinguishment time. This data was recorded except for the inside thermocouple reading for the first 5 tests. The data collected is adequate for determining pass-fail criteria.

Access Panel Fires

Description

With the access panel fires the aft end of the test fixture is centered so that the access panel is also located over the concrete pan. The test fixture is lined up with the direction of the wind so that the wind travels from the back to the front of the fixture. When the test fixture reaches 450 degrees, JP-8 jet fuel is sprayed through nozzle 1 into the engine nacelle. The extinguishers were fully serviced prior to each fire. A propane torch was used to ignite the JP-8 and a pre-burn of 15 seconds was conducted to assure full involvement of the fuel in the fire area. The firefighter was given a ten second countdown, at which time a steady attack mode was used to extinguish the fire.

Time to extinguishment was based on complete extinguishment of the fire inside the test fixture.

Criteria for Success

This test series did not correlate to any existing NFPA requirements; therefore, no maximum extinguishment times were established. The test was conducted to provide an estimation of current flight line fire extinguisher performance.

The series of tail pipe fires were used to gather data to develop performance parameters for flightline fire extinguishers. Data points for each test included type of fire, extinguisher number, wind speed, outside temperature, four thermocouple temperatures, amount of agent used, and full extinguishment time. This data was recorded except for the inside thermocouple reading for the first 3 tests. The data collected is adequate to determine pass-fail criteria.

Results and Discussion

Fire Scenarios

A combination of 20 aft engine and 10 access panel fires were conducted over eight business days. Weather conditions varied for the days of aft engine testing. Temperatures were as low as 44° and up to 77°, with winds ranging from 0-8 mph, with one test at 18.4 mph. Weather conditions for the days of access panel testing were consistent, with temperatures ranging from 66° to 73°. Wind conditions varied between 0-8.3 mph, with one test at 10.4 mph.

Time Data Sources

Extinguishment times were collected from two sources:

- From camera 1.
- From camera 2.

Differences in full extinguishment times can be attributed to position relative to the fire. An average of the time sources are considered the official times for full extinguishment.

Aft Engine Fires

90% of the fires were extinguished. All but one of these fires were extinguished within 30 seconds. Fires averaged 16.25 second extinguishment and 67.6 pounds of halon. The full extinguishment times and amounts of agent used are plotted in Figure 2.

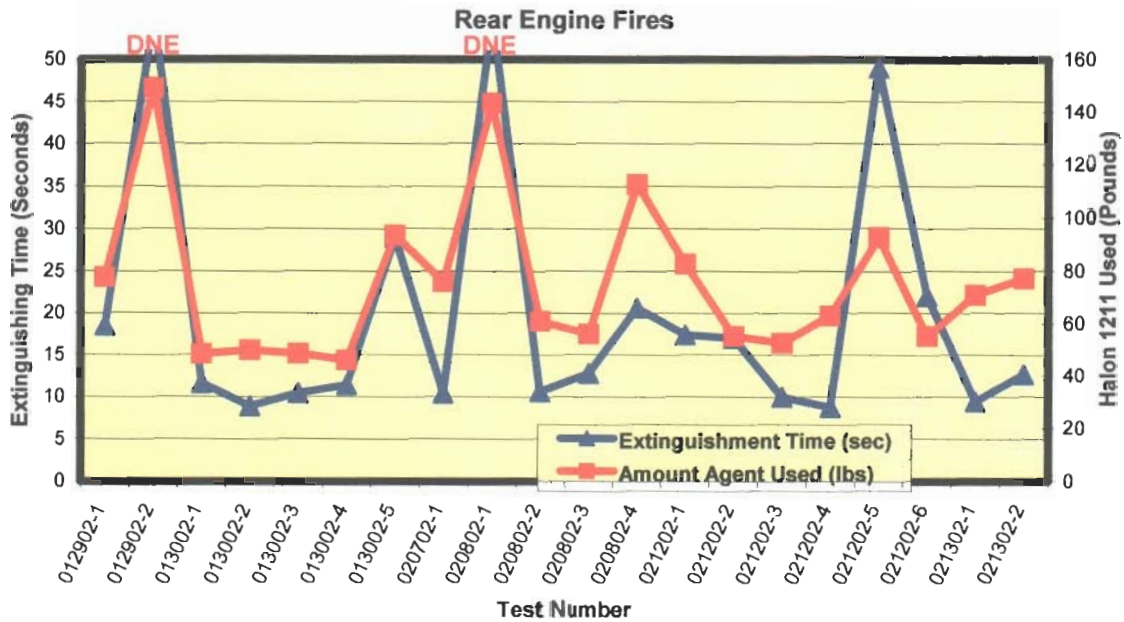


Figure 2. Summary of Extinguishment Times and Agent Used for Aft Engine Fires.

Access Panel Fires

80% of the fires were extinguished. All but one fire was extinguished within 35 seconds. Fires averaged 17.29 seconds extinguishment and 81.8 pounds of halon. The full extinguishment times and amounts of agent used are plotted in Figure 3.

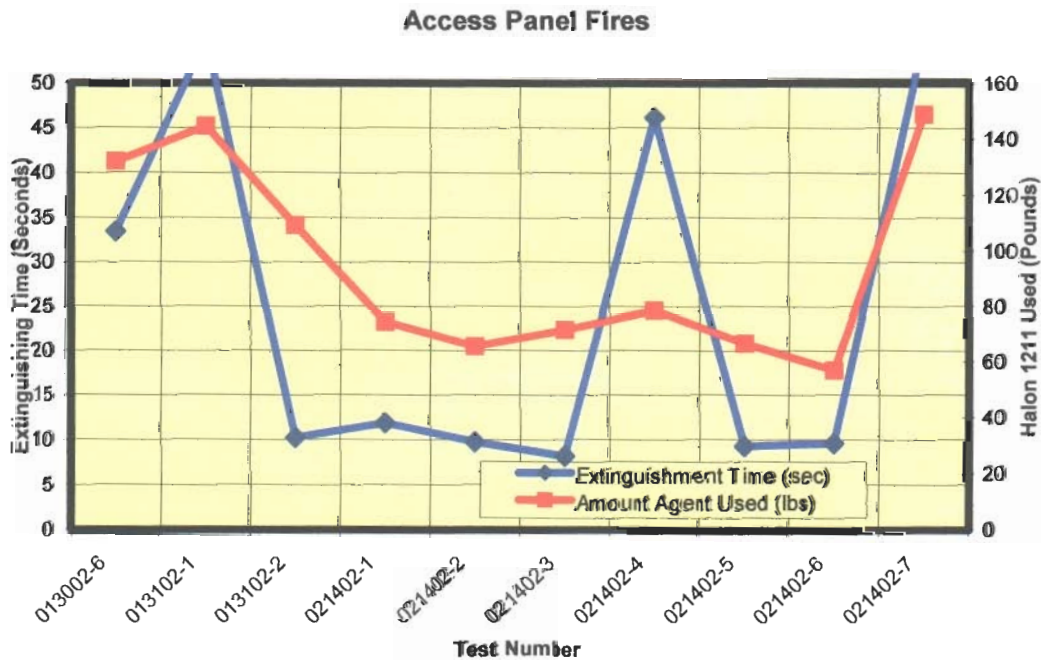


Figure 3. Extinguishment Times and Agent Used for Access Panel Fires.

Analysis

The fires (Figure 4 and 5) were regularly extinguished within 30 seconds using less than 100 lbs of Halon 1211. Two of twenty aft engine fires did not extinguish and two of ten access panel fires did not extinguish. Full extinguishment times for aft engine fires ranged from 8.75-50.93 seconds, with only one fire exceeding 30 seconds. The average time to extinguish an aft engine fire with Halon 1211 was 16.25 seconds. For access panel fires, full extinguishment times ranged from 8.18 to 46.16 seconds, with only one fire taking greater than 30 seconds. The average time to extinguish an access panel fire was 17.29 seconds.

Of the aft engine fires that were extinguished, all but one consumed less than 100 pounds of Halon 1211 agent. That one used 112.6 pounds of agent, approximately 75% of the extinguisher capacity.

Of the access panel fires that were extinguished, all but one consumed less than 110 pounds of Halon 1211 agent. That one used 131.95 pounds of agent, approximately 88% of the extinguisher capacity.

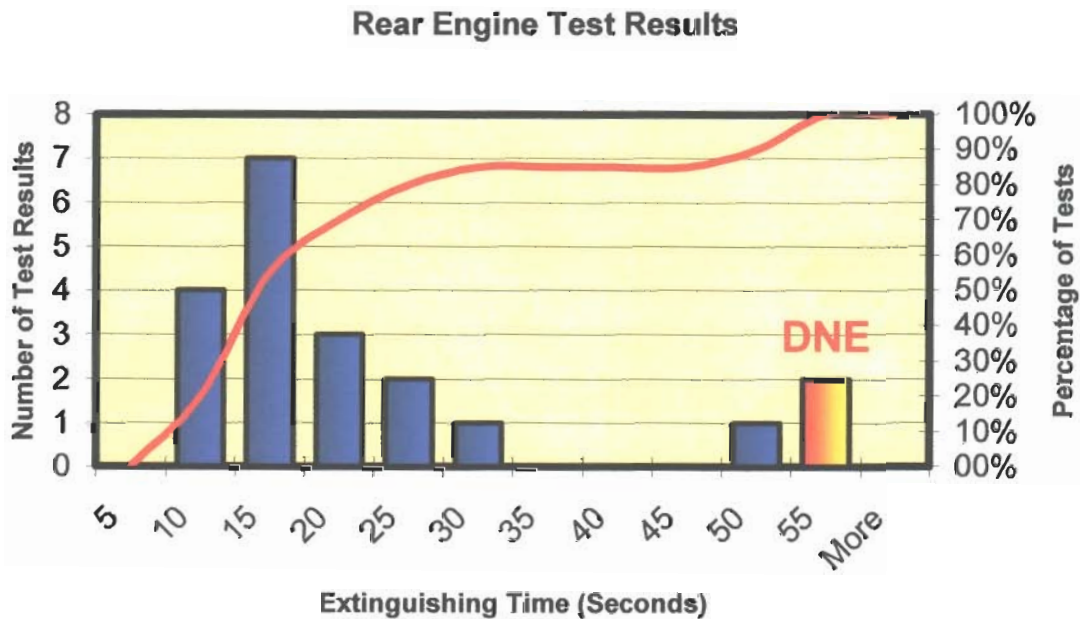


Figure 4. Histogram of Aft Engine Test Results

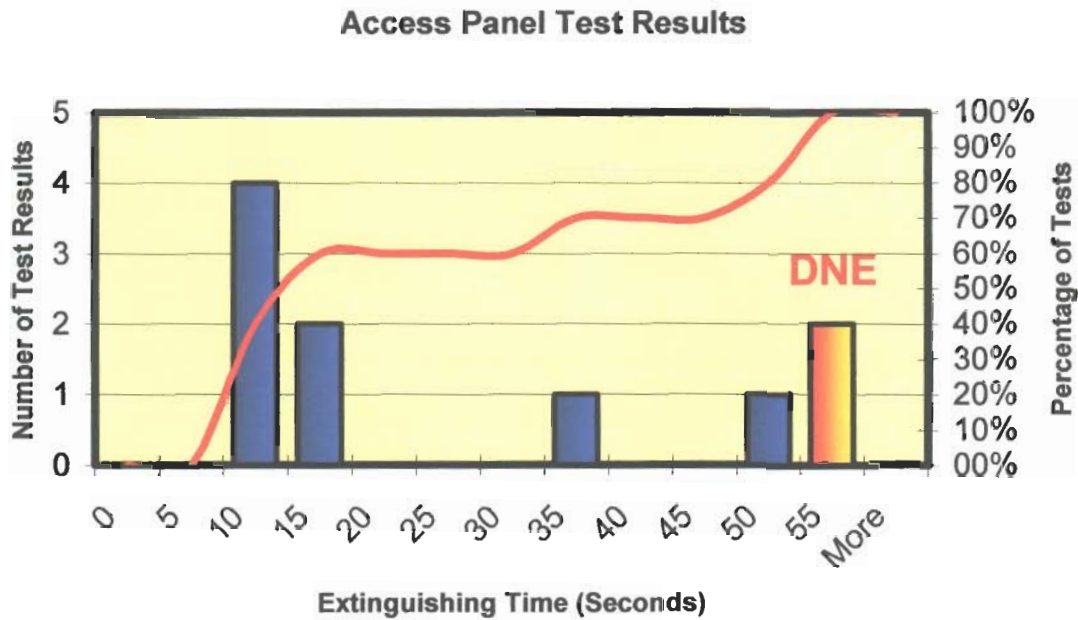


Figure 5. Summary of Access Panel Test Results

Fire Scenarios

Aft Engine Fires

The flightline fire extinguishers routinely extinguished fires in less than 30 seconds using less than 75% agent capacity.

Access Panel Fires

The flightline fire extinguishers routinely extinguished fires in less than 30 seconds using less than 75% agent capacity.

i Minimum Performance Requirement for Air Force Flightline Fire Extinguishers: Extinguishing Performance Against 3-Dimensional And Hidden Fires, May 2002, AFRL-ML-TY-TR-02-4540 (ADA402968)

ii F100 Engine Nacelle Fire Fighting Test Mockup, July 2002, AFRL-ML-TY-TR-02-4604 (ADA405232)

APPENDIX I: Check List

Check List

Daily Inspections and Initial Preparations

1. _____ Complete **base** notification of pending fire tests.
2. _____ Insure two flight line fire extinguishers are fully serviced in accordance with T.O. 13F4-4-121 and pre-positioned at the **test pan**. Log the weight of the fully serviced fire extinguisher.
3. _____ Pre-position a fully serviced P-19 fire truck or equivalent and maintain on stand-by as a back-up firefighting vehicle **near the test pan**.
4. _____ Assign a primary and secondary firefighter.
5. _____ Collect meteorological **data** prior to fuel spilling to determine whether testing can proceed. Testing will not proceed if the wind velocity is **greater than 5 knots**.
6. _____ Move engine nacelle so that the aft section is in the predetermined proper position.
7. _____ Place two (2) video cameras adjacent **to the fire**, one behind the firefighter and one at perpendicular to the engine.
8. _____ Inspect thermocouples and data acquisition system. Ensure proper placement of the thermocouples in the nacelle and that thermocouples and data acquisition are working properly.
9. _____ Inspect ignition system. Ensure proper connection of the igniter hardware and ensure isolation from ground and continuity in the ignition circuit. Test and confirm operation of the igniter.
10. _____ Open fuel valves numbers 1, 2 and 3 to **pressurize the system** and check for leaks. Tighten fuel lines as appropriate to prevent leaks.

Pre-heat Phase

11. _____ **Remove all non-essential personnel and assure all personnel involved in testing** are in their assigned stations prior to approval for lighting the fire (**as signaled by the AFRL Test Team Leader**).
12. _____ Turn on data collection equipment.
13. _____ Turn on the video cameras at the beginning of the pre-heat phase and turn off cameras at the conclusion of the pre-heat phase. Ensure the display board contains all essential test information.
14. _____ **The primary firefighter will don PPE and SCBA. A second firefighter will don PPE, serve as a backup firefighter and will be responsible for ignition of the test nacelle, if required.**
15. _____ Turn on fuel valve number 3 and continue to flow fuel from fuel valve number **3** into the aft burner chamber.