

REPORT DOCUMENTATION PAGE				<i>Form Approved</i> OMB No. 0704-0188	
Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing this collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden to Department of Defense, Washington Headquarters Services, Directorate for Information Operations and Reports (0704-0188), 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302. Respondents should be aware that notwithstanding any other provision of law, no person shall be subject to any penalty for failing to comply with a collection of information if it does not display a currently valid OMB control number. PLEASE DO NOT RETURN YOUR FORM TO THE ABOVE ADDRESS.					
1. REPORT DATE (DD-MM-YYYY) 11/30/11		2. REPORT TYPE Final		3. DATES COVERED (From - To) 06/15/09-11/30/11	
4. TITLE AND SUBTITLE Pseudospectral Optimal Control - Hidden Properties and Flight Results				5a. CONTRACT NUMBER	
				5b. GRANT NUMBER FA9550-09-1-0454	
				5c. PROGRAM ELEMENT NUMBER	
6. AUTHOR(S) Qi Gong				5d. PROJECT NUMBER	
				5e. TASK NUMBER	
				5f. WORK UNIT NUMBER	
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) Univ. of California Santa Cruz, CA				8. PERFORMING ORGANIZATION REPORT NUMBER	
9. SPONSORING / MONITORING AGENCY NAME(S) AND ADDRESS(ES) AF OFFICE OF SCIENTIFIC RESEARCH 875 N. RANDOLPH ST. ROOM 3112 ARLINGTON VA 22203				10. SPONSOR/MONITOR'S ACRONYM(S) AFOSR	
				11. SPONSOR/MONITOR'S REPORT NUMBER(S) AFRL-OSR-VA-TR-2012-1025	
12. DISTRIBUTION / AVAILABILITY STATEMENT DISTRIBUTION A: APPROVED FOR PUBLIC RELEASE					
13. SUPPLEMENTARY NOTES					
14. ABSTRACT					
15. SUBJECT TERMS					
16. SECURITY CLASSIFICATION OF:			17. LIMITATION OF ABSTRACT	18. NUMBER OF PAGES	19a. NAME OF RESPONSIBLE PERSON
a. REPORT	b. ABSTRACT	c. THIS PAGE			19b. TELEPHONE NUMBER (include area code)

Pseudospectral Optimal Control: Hidden Properties and Flight Results

AFOSR GRANT FA9550-09-1-0454

Qi Gong

Dept. of Applied Mathematics and Statistics

Univ. of California, Santa Cruz, CA

AFOSR GRANT F1ATA0-90-4-3G001

I. Michael Ross and Wei Kang

Naval Postgraduate School, Monterey, CA, 93943

Abstract

The main goal of this joint project between the Naval Postgraduate School and the University of California, Santa Cruz, was to analyze the mathematical proprieties of pseudospectral (PS) optimal control theory. The secondary goal of our project was to exploit the new properties to develop or enhance computationally efficient control algorithms. These goals were motivated by a need to generate real-time solutions to constrained nonlinear control problems that plays a central role in low-cost operations of high-performance space systems. In this project, we developed a general framework for PS optimal control theory based on discretization over an arbitrary (i.e. non-Gaussian) grid. Our analysis on dual consistency revealed some hidden properties of PS methods; and led to a new PS scheme utilizing primal-only conditions. These new developments enhance the algorithmic efficiency that is crucial for real-time applications. During this research period, we also got an opportunity to apply our ideas to the time-optimal reorientation of a NASA spacecraft in orbit; and performed a historic command and control of the Transition Region and Coronal Explorer (TRACE) space telescope.

Introduction

High performance of space systems demands full considerations of the nonlinearity and stringent constraints. Optimal control plays a central role in designing efficient feedback control algorithms for such complicated high-performance space systems. As a means to address the challenges of future space systems, we address the problem of generating a real-time solution to the following generic constrained nonlinear optimal control problem:

$$(B) \left\{ \begin{array}{ll} \text{Minimize} & J[x(\cdot), u(\cdot)] = E(x(-1), x(1)) + \int_{-1}^1 F(x(t), u(t)) dt \\ \text{Subject to} & \dot{x}(t) = f(x(t), u(t)) \\ & e(x(-1), x(1)) = 0 \\ & h(x(t), u(t)) \leq 0 \end{array} \right.$$

where $F : \mathbb{R}^{N_x} \times \mathbb{R}^{N_u} \mapsto \mathbb{R}$, $E : \mathbb{R}^{N_x} \times \mathbb{R}^{N_x} \mapsto \mathbb{R}$, $f : \mathbb{R}^{N_x} \times \mathbb{R}^{N_u} \mapsto \mathbb{R}^{N_x}$, $e : \mathbb{R}^{N_x} \times \mathbb{R}^{N_x} \mapsto \mathbb{R}^{N_e}$, $h : \mathbb{R}^{N_x} \times \mathbb{R}^{N_u} \mapsto \mathbb{R}^{N_h}$, and N with the appropriate subscript is some given natural number.

Over the last decade, pseudospectral (PS) methods have emerged as one of the most efficient numerical methods for solving constrained nonlinear optimal control problems, as evident from

its current widespread use, include experimental demonstrations and flight operations. Based on different orthogonal polynomials and Gaussian grids, various PS optimal control methods can be developed. In PS methods, the grid is typically determined by the roots or the extrema of orthogonal functions. Since different orthogonal polynomials have different properties with respect to interpolation, integration, and discretization of the optimal control problem, the approximations of the states and costates based on these various grids could have different outcomes. For example, Chebyshev PS method offers rapid node computation and derivative computation via a fast Chebyshev differentiation scheme that is similar to an FFT computation. However, the non-unity weight function associated with Chebyshev polynomials creates difficulties in constructing a proper dual space. As a mean to analyze the effects of different orthogonal polynomials and grids on solving optimal control problems, we focus on developing PS methods over arbitrary grids for Problem B. Such research can provides a unified framework for the analysis of different PS methods. It also provides a way to compare performances among different PS methods and suggest guidelines for choosing the proper grids and discretization approaches for constrained optimal control problems.

Major Accomplishments of The Project

PS Method over Arbitrary Grids

Because Problem B has boundary conditions, we choose $t_0 = -1$ and $t_N = 1$ to be the boundary points. We further choose an arbitrary grid between the two end points, and denote it as $-1 < t_1 < t_2 < \dots < t_{N-1} < 1$. In Ref.[12], we developed a PS discretization of Problem B that generates a sequence of finite dimensional optimization problems given by,

$$(B^N) \left\{ \begin{array}{ll} \text{Minimize} & \bar{J}^N(\bar{x}, \bar{u}) = \sum_{k=0}^N F(\bar{x}^k, \bar{u}^k)w_k + E(\bar{x}^0, \bar{x}^N) \\ \text{Subject to} & \sum_{j=0}^N D_{kj}\bar{x}^j = f(\bar{x}^k, \bar{u}^k) \\ & e(\bar{x}^0, \bar{x}^N) = 0 \\ & h(\bar{x}^k, \bar{u}^k) \leq 0 \\ & k = 0, 1, \dots, N \end{array} \right.$$

In Problem B^N , D is a square differentiation matrix and w_k are quadrature weights defined by

$$D_{ij} = \dot{\phi}_j(t_i), \quad w_k = \int_{-1}^1 \phi_k(t) dt$$

where $\phi_j(t)$ are the Lagrange polynomials given as

$$\phi_j(t) = \frac{g_N(t)}{g'_N(t_j)(t - t_j)}, \quad g_N(t) = \prod_{j=0}^N (t - t_j).$$

Note that both w_k and D are constants that depend on the grid points only. Therefore, the performance of the PS methods is determined by the choice of the grid. This general formulation helps us to analyze the effects of grid selection. For example, the feasibility and the consistency of PS methods over arbitrary grids can be guaranteed under certain conditions. One of them is that the weights, w_k , $k = 0, 1, \dots, N$, need to be positive for all k . When a uniform grid is chosen, when $N > 11$, at least one of the weights is negative. Therefore, uniform grid is not an appropriate choice, as convergence is not guaranteed. This point is demonstrated by way of a linear optimal control problem in Ref.[12].

Dual Consistency

For Problem B^N to converge to Problem B , the discretization scheme must commute with the dualization process. That is the KKT conditions of Problem B^N must preserve the structure of the continuous necessary conditions. For dual consistency analysis, consider a distilled scalar unconstrained optimal control problem with no cost function. The costate, $\lambda(t)$, must satisfy the adjoint equation $\dot{\lambda} = -\frac{\partial f(x,u)}{\partial x}\lambda$, whose PS discretization over arbitrary grids is

$$D\bar{\lambda} = -\Lambda(\bar{x}, \bar{u}) \cdot \bar{\lambda} \quad (1)$$

where $\bar{\lambda} = [\bar{\lambda}_0, \dots, \bar{\lambda}_N]^T$ and $\Lambda(\bar{x}, \bar{u})$ is a diagonal matrix with the i th diagonal entries be $\frac{\partial f}{\partial x}(\bar{x}^i, \bar{u}^i)$. On the other hand, the discretized state equation $D\bar{x} = \bar{f}(\bar{x}, \bar{u})$ generates the KKT conditions

$$D^T \hat{\lambda} = \Lambda(\bar{x}, \bar{u}) \cdot \hat{\lambda} \quad (2)$$

where $\hat{\lambda}$ are KKT multipliers. Clearly, $\hat{\lambda}$ is not the discretization of the continuous costate, since D^T is not a differentiation operator. However, it can be shown that differentiation matrix defined on arbitrary grid can always be factorized into $D = M^{-1}S$ with M and S be square matrices defined by

$$M_{i,j} = \sum_{k=0}^N \phi_i(\tau_k^l) \phi_j(\tau_k^l) w_k^l, \quad S_{i,j} = \sum_{k=0}^N \phi_i(\tau_k^l) \dot{\phi}_j(\tau_k^l) w_k^l$$

where τ_k^l are Legendre-Gauss-Lobatto (LGL) nodes and w_k^l are LGL weights. Matrix M is always positive definite, and satisfies Sylvester-like equation

$$MD + D^T M = \Delta \quad (3)$$

with all entries of Δ be zero except for $\Delta_{00} = -1$ and $\Delta_{NN} = 1$. With the help of (3), the KKT system (2) is transferred into

$$D \cdot (M^{-1} \hat{\lambda}) = -[M^{-1} \Lambda(\bar{x}, \bar{u}) M] \cdot (M^{-1} \hat{\lambda}) + M^{-1} \Delta \cdot (M^{-1} \hat{\lambda}) \quad (4)$$

Since the discrete differentiation is applied to $M^{-1} \hat{\lambda}$, the scaled KKT multiplier, $M^{-1} \hat{\lambda}$, serves as the discretization of the continuous costate. This is the first hidden property of PS optimal control methods: *for PS methods over arbitrary grid distribution, costate is formed by the multiplication of adjoint weight matrix M and the KKT multipliers*. For standard LGL PS method, since M is a diagonal matrix formed by LGL weights, we recover the standard approach for costate computation, i.e., scale the KKT multipliers by LGL weights.

From (4), it is also clear that to maintain the structure of the vector field in the adjoint equation, we need $M^{-1} \Lambda(\bar{x}, \bar{u}) M = \Lambda(\bar{x}, \bar{u})$. For nonlinear problems, this condition holds *if and only if M is diagonal*. It is another hidden property of PS methods been discovered. From the definition of M and the property of Lagrange interpolating polynomial, M in general cannot be diagonal for arbitrary grid. It is diagonal only if the grid, t_k , coincide with LGL points, τ_k , i.e., LGL grid is adopted. It explains the superior performance of LGL PS method. In fact, the diagonal property of M is deteriorated as the grid distribution is further away from the set of LGL points. Shown in Fig.1 are the absolute value of off-diagonal entries of matrix M with LGL grid, CGL grid and uniform grid. It is clear that for CGL grid, off-diagonal entries are small implying that M is approximately diagonal. For uniform grid, M is far away from being diagonal. Since all Gaussian quadrature grids share similar distribution, such Gaussian grids maintain the diagonal property of M approximately. Our results also explain the connection between the Clenshaw-Curtis weights and the covector mapping theorem for the Chebyshev PS method developed in Ref.[2]. That is, for a CGL grid, M is not diagonal but is well-approximated by a diagonal matrix composed of the Clenshaw-Curtis weights.

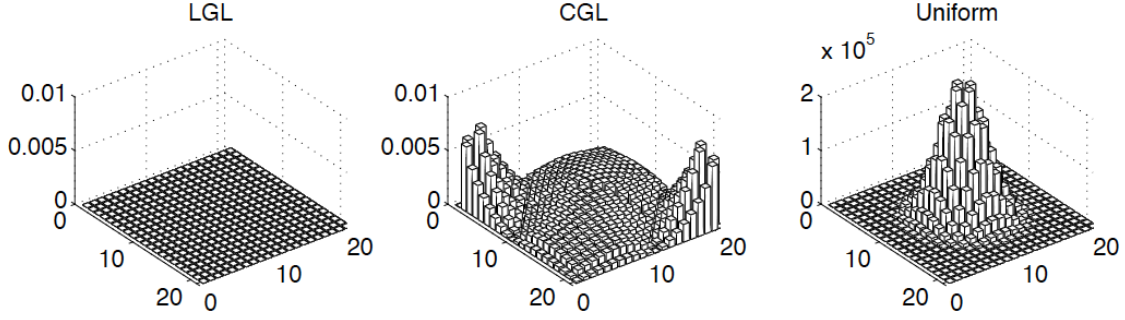


Figure 1: Magnitude of off-diagonal entries of matrix M with different types of grids ($N = 20$).

Primal-only Closure Conditions

The last term in (4) relates to the transversality conditions. For LGL PS method, utilizing the diagonal property of M , a set of primal-dual closure conditions are introduced to imposed right transversality condition. This approach was generalized to Chebyshev PS methods with M be approximated by a diagonal matrix composed of the Clenshaw-Curtis weights. For arbitrary grid PS methods, to handle the last term in (4) and the transversality conditions, we modify Problem B^N as follows: introduce new primal decision variables $\hat{x}^0$ and $\hat{x}^N$, which represent almost the same quantities as $\bar{x}^0$ and $\bar{x}^N$. In the discretization of the event cost, $E(x(-1), x(1))$, and event conditions, $e(-1, 1) = 0$, we use $\hat{x}^0$ and $\hat{x}^N$ instead of $\bar{x}^0$ and $\bar{x}^N$. Therefore, the discrete cost and the event conditions are

$$\begin{aligned}\bar{J}^N &= \sum_{k=0}^N F(\bar{x}^k, \bar{u}^k) w_k + E(\hat{x}^0, \hat{x}^N) \\ 0 &= e(\hat{x}^0, \hat{x}^N)\end{aligned}$$

The discretization of the dynamical equations is modified as

$$D\bar{x} = \bar{f}(\bar{x}, \bar{u}) + (\bar{x}^0 - \hat{x}^0)r_0 + (\bar{x}^N - \hat{x}^N)r_N$$

where $\bar{f}(\bar{x}, \bar{u}) = [f(\bar{x}^0, \bar{u}^0), \dots, f(\bar{x}^N, \bar{u}^N)]^T$. r_0 and r_N are the first and the last column of M^{-1} . It can be shown that r_0 and r_N are explicitly given by

$$\begin{aligned}r_{0,k} &= \frac{(-1)^{N+1}}{2} \dot{L}_N(t_k)(t-1) \\ r_{N,k} &= \frac{1}{2} \dot{L}_N(t_k)(t+1), \quad k = 0, 1, \dots, N\end{aligned}$$

where $L_N(t)$ is the N th order Legendre polynomial. Such modification generates a modified Problem B^N that enables dualization to commute with discretization. This new scheme facilitates enhancing the algorithmic efficiency of the underlying spectral algorithm by generating a full-rank KKT system. The proposed primal-only closure condition is easy to implement and applicable to different PS methods including Legendre and Chebyshev PS methods as special cases.

Flight Results

During this research period, we also got an opportunity to apply our ideas to the time-optimal reorientation of a NASA spacecraft in orbit. On August 2010, we led a team of engineers and

operators to perform a historic command and control of the Transition Region and Coronal Explorer (TRACE) space telescope. Prior to our flight test, no spacecraft had ever performed a minimum-time attitude maneuver. This is, in part, because of the well-known difficulties in solving the minimum-time spacecraft attitude control problem. Our techniques allow an operator to perform verification and validation of the command signals without a need to understand PS optimal control theory. Because TRACE is actuated by reaction wheels, the resulting dynamics are not only non-Eulerian but also, the control space is non-convex and state-dependent. The well-known difficulties in solving such problems were easily circumvented by our PS theory. The mathematical implications as well as the engineering results acquired from the telemetry are being analyzed now with the goal of publications in the coming years.

Other Advancements

During this research period, we also extended our previous work on feasibility, consistency and convergence analysis. When the Legendre PS method is used for feedback linearizable systems, we showed in Ref.[3] that the rate of convergence is at least $\frac{1}{N^{2m/3-1}}$, where m is defined by the smoothness of the optimal trajectory; i.e. an appropriate Sobolev space. If the optimal control is C^∞ , then the convergence rate can be made faster than any given polynomial rate. Relative to our earlier results, we also weakened the assumptions in the theorems we proved. In this research project, we also analyzed the performance of a PS algorithm for generating real-time solution of infinite-horizon optimal control problems. We proved the closed-loop stability under box constraints on states and controls. Such results are important for real-time space control applications.

Ongoing Work and Future Research Plans

This is the final year of a three-year project. The new developed PS computational scheme paves the way to construct more efficient algorithms for solving optimal control problems, for example, multiscale PS methods for dynamical systems with different timescales. Many space applications involve both slow and fast dynamics. The nature of such systems with mixed time-constants suggests a possible efficient discretization using multi-resolution for different subsystems. The new developed arbitrary grid PS method provides flexibility in designing different grids for different subsystems. We will explore this idea, together with a covector mapping theorem for costate computation, to develop multiscale PS methods.

Acknowledgment/Disclaimer

This work was sponsored by the Air Force Office of Scientific Research, USAF, under grant/contract number F1ATA0-90-4-3G001 and FA9550-09-1-0454. The views and conclusions contained herein are those of the authors and should not be interpreted as necessarily representing the official policies or endorsements, either expressed or implied, of the Air Force Office of Scientific Research or the U.S. Government.

Personnel Supported

- I. M. Ross, Professor, Naval Postgraduate School, Monterey, CA
- W. Kang, Professor, Naval Postgraduate School, Monterey, CA
- Q. Gong, Assistant Professor, University of California, Santa Cruz, CA

- H. Yan, Postdoc, University of California, Santa Cruz, CA

Related Recent Publications

1. A. Fleming, P. Sekhavat, and I. M. Ross, Minimum time reorientation of a rigid body, *AIAA Journal of Guidance, Control and Dynamics*, Vol. 33, No. 1., pp.160-170, 2010.
2. Q. Gong, I. M. Ross and F. Fahroo, Costate estimation by a Chebyshev Pseudospectral method, *AIAA Journal of Guidance, Control and Dynamics*, Vol. 33, No. 2, pp. 623-628, 2010.
3. W. Kang, Rate of convergence for the Legendre Pseudospectral optimal control of feedback linearizable systems, *Journal of Control Theory and Application*, No. 4, pp. 391-405, 2010.
4. W. Kang, I. M. Ross, K. Phamy and Q. Gong, Autonomous observability of networked multisatellite systems, *AIAA Journal of Guidance, Control and Dynamics*, Vol. 32, No. 3., pp. 869-877, 2009.
5. Q. Gong, L. R. Lewis and I. M. Ross, Pseudospectral motion planning for autonomous vehicles, *AIAA Journal of Guidance, Control and Dynamics*, Vol. 32, No. 3, pp. 1039-1045, 2009.
6. N. Bedrossian, S. Bhatt, W. Kang, I. M. Ross, Zero-propellant maneuver guidance: rotating the international space station with computational dynamic optimization, *IEEE Control Systems Magazine*, Vol. 29, No. 5, 53-73, 2009.
7. C. D. Park, I. M. Ross, Q. Gong, and H. Yan, Necessary conditions for singular arcs for general restricted multi-body problem, *the 49th IEEE Conference on Decision and Control*, Atlanta, GA, Dec. 2010.
8. H. Yan, Q. Gong, C. D. Park, I. M. Ross and C. N. D'Souza, High-accuracy Moon to Earth trajectory optimization, *AIAA Guidance, Navigation, and Control Conference*, 2010.
9. H. Yan and Q. Gong, Feedback control for formation flying maintenance using state transition matrix, *AAS Kyle T. Alfriend Astrodynamics Symposium*, AAS 10-312, May. 2010.
10. Q. Gong, I. M. Ross and F. Fahroo, Chebyshev pseudospectral method for nonlinear constrained optimal control problems, *48th IEEE Conference on Decision and Control*, Shanghai, China, pp. 5057-5062, Dec. 2009.
11. W. Kang, A quantitative measure of observability and controllability, *48th IEEE Conference on Decision and Control*, Shanghai, China, 2009.
12. Q. Gong, I. M. Ross and F. Fahroo, Pseudospectral optimal control on arbitrary grids, *AAS Astrodynamics Specialist Conference*, AAS 09-405, August 2009.