

HIGH-ORDER COMPACT SCHEMES FOR PLASMA WAVE SIMULATION*

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Abstract

A high-order finite-difference code has been developed with the ability to simulate waves in plasmas coupling Maxwell Equations with fluid species equations. Originally developed to examine the radio blackout problem, the code utilizes 6th-order finite-differences with 10th-order filters in space and 4th-order Runge-Kutta in time in order to minimize the number of grid points while retaining the fidelity of the solution. The code also uses multiple overlapping grids allows for solving complex configurations in parallel.

I. INTRODUCTION

Over the past few years the Air Force has been interested in development of high fidelity simulation capabilities to investigate the long standing problem of radio blackout during reentry. To address this issue, a high-order finite-difference code has been developed by the Air Vehicles Branch of the Air Force Research Laboratory which can simulate electromagnetic plasma waves.[1-3] The current configuration requires that the plasma be generated by another source, and the background electron density and temperature are fed into *OHMS* (Overset High-order Maxwell Solver) which solves the propagation problem.[2-3]

II. METHODOLOGY

A. Equation set

The equations for plasma propagation by the electrons can be written as:

$$\begin{aligned}\epsilon_0 \partial_t \mathbf{E}_1 + \mathbf{J}_{e,1} - \nabla \times (\mathbf{B}_1 / \mu_0) &= 0 \\ \partial_t \mathbf{B}_1 + \nabla \times \mathbf{E}_1 &= 0 \\ \partial_t n_{e,1} - \frac{\nabla \cdot \mathbf{J}_{e,1}}{e} &= 0\end{aligned}$$

$$\partial_t \mathbf{J}_{e,1} = \left(\frac{e^2 n_e}{m_e} \right) \mathbf{E}_1 + e \frac{\gamma_e R T_e \nabla n_{e,1}}{m_e} - \nu_e \mathbf{J}_{e,1}$$

where the subscript “1” indicates a perturbation of the variable. In non-dimensional form, the equations reduce to [3]:

$$\begin{aligned}\partial_t \mathbf{E} + \mathbf{J}_e - \nabla \times \mathbf{B} &= 0 \\ \partial_t \mathbf{B} + \nabla \times \mathbf{E} &= 0 \\ \partial_t \zeta_e - \nabla \cdot \mathbf{J}_e &= 0 \\ \partial_t \mathbf{J}_e &= \Omega_p^2 n_{e,0} \mathbf{E} + \beta_{th}^2 T_{e,0} \nabla \zeta_e - \nu_e \mathbf{J}_e\end{aligned}$$

where the perturbation subscript “1” has been suppressed, Ω_p is the electron plasma frequency multiplied by the time scale, ζ_e is the perturbation electron density multiplied by its charge and β_{th} is the ratio of the root-mean-squared molecular velocity of the electrons based on the reference temperature ($T_{e,ref}$) divided by the speed of light in vacuum. Thus, the code solves 10 equations needing 12 variables of storage.

B. Computational Method

The *OHMS* code uses 6th-order finite-difference modeling in generalized curvilinear coordinates:[3-5]

$$\begin{aligned}\alpha \frac{\partial F_{j-1}}{\partial \xi_i} + \frac{\partial F_j}{\partial \xi_i} + \alpha \frac{\partial F_{j+1}}{\partial \xi_i} &= \\ a \left(\frac{F_{j+1} - F_{j-1}}{2\Delta \xi_i} \right) + b \left(\frac{F_{j+1} - F_{j-1}}{4\Delta \xi_i} \right)\end{aligned}$$

for any variable F at point j being differenced in the ξ_i direction. The order of accuracy is satisfied by restricting the coefficients:

$$\begin{aligned}2^{nd} \text{ order: } (1 + 2\alpha) &= (a + b) \\ 4^{th} \text{ order: } \alpha &= (a + 4b) / 6 \\ 6^{th} \text{ order: } \alpha / 12 &= (a + 16b) / 5!\end{aligned}$$

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The resulting discrete equations for the partial derivatives are solved implicitly using a tridiagonal solver. The effective wavenumbers resulting from explicit and compact differencing is shown in figure 1. Added to the graph is the equivalently scaled spectral response of Finite Difference Time Domain (FDTD).

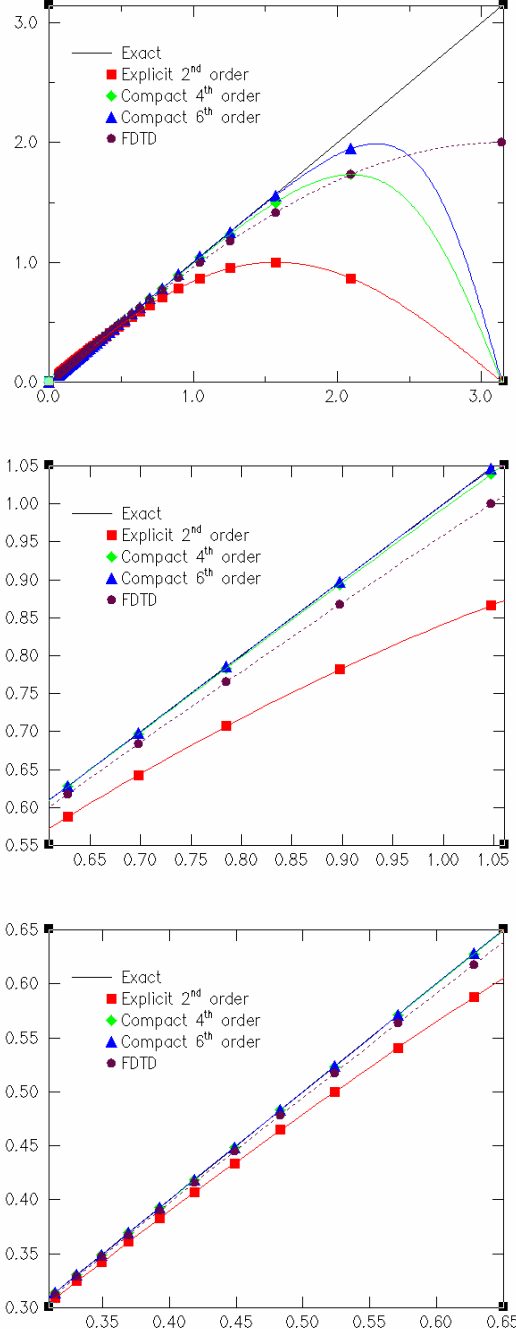


Figure 1. Reduced wavenumber for different schemes. Top: Wavenumber $k\Delta x$ vs effective wavenumber $k'\Delta x$ for $\lambda = 2\Delta$ to ∞ . Middle: 10 to 6 points per wavelength. Bottom: 20 to 10 points per wavelength.

C. Filtering

Filtering is utilized both for stability [4,5] and for the absorbing boundary conditions[6]. The filters are designed so that wavelengths $\lambda = 2\Delta$ are completely eliminated while long wavelengths are unattenuated. The basic filter equation is solved using a tridiagonal solver:

$$\alpha_f \hat{F}_{j-l} + \hat{F}_j + \alpha_f \hat{F}_{j+l} = \sum_{n=0}^N \frac{a_n}{2} (F_{j+n} + F_{j-n})$$

where for 10th order: $a_0 = (193 + 126 \alpha_f)/256$, $a_1 = (105 + 302 \alpha_f)/256$, $a_2 = -15(1 - 2 \alpha_f)/64$, $a_3 = 45(1 - 2 \alpha_f)/512$, $a_4 = -5(1 - 2 \alpha_f)/256$, $a_5 = (1 - 2 \alpha_f)/512$, and α_f varies from 0 to 0.499 (typical values are 0.3 to 0.49). The transfer function is given by:

$$T(\omega) = \frac{\sum_{n=0}^N a_n \cos(n\omega)}{1 + 2\alpha_f \cos(\omega)}$$

where $\omega = 2\pi\Delta/\lambda$. Figure 2 for different values of α_f .

D. Chimera

OHMS has the capability for multiple overlapping grids (Chimera) to be solved in parallel. The grid communication is given via Lagrangian interpolation.[7,8] The Chimera paradigm allows for the easy implementation of the scattered field/total field interface by defining each grid as either a scattered or total field grid. The incident fields can then be added to or subtracted from the interpolated values during the message passing.[3]

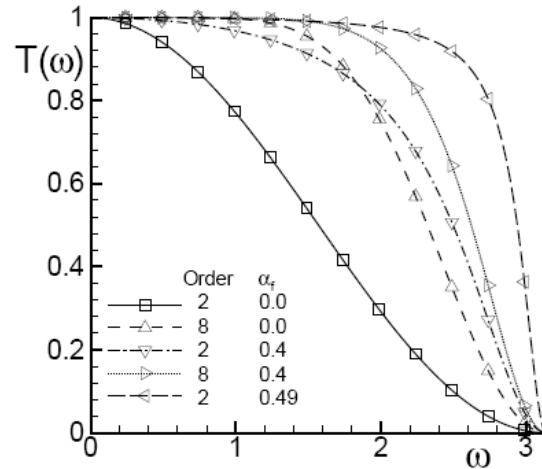


Figure 2. Transfer function for filter orders 2-8 for various values of α_f .

III. RESULTS

Propagation of a signal from an horn antenna was simulated using a background plasma generated by a non-equilibrium re-entry code developed by Josyula and Bailey.[9] For the given antenna location, the maximum electron temperature was $T = 0.1$ eV. At this temperature, no effect was seen by the addition of the electron temperature – indicating that for most re-entry problems, solving an auxiliary ordinary differential equation for the current is all that is necessary. While removing only a single equation, the number of variable storage is reduced by 2 and 25 fewer tridiagonal solves are needed per time step. To see the effect of the extra equation, the reference electron temperature was artificially raised to 100 eV and 5 keV. As seen in figure 3, there is almost no change in the electric field from 0.1 to 100 eV. At 5 keV there is a noticeable change in the field radiating from the horn. In this case, the signal frequency is below the peak plasma frequency for the electron density distribution causing the wave to be highly attenuated. Looking at the current a few grid points above the horn aperture in figure 4, there was no noticeable difference in the current for 0.1 to 100 eV reference temperature. As seen by the instantaneous E field in figure 5, the effect of the plasma temperature is greatest nearest the surface of the vehicle, and its influence drops off as the distance increases.

IV. SUMMARY

A high-order finite-difference methodology has been presented for simulating electromagnetic plasma waves in complex configurations. For most reentry problems, the effect of electron temperature will likely be confined to near the vehicle body and appears to be negligible in terms of the radio blackout problem. While not currently implemented, the addition of the $\mathbf{J}_e \times \mathbf{B}_0$ term is straight forward, requiring only three more variables of storage.

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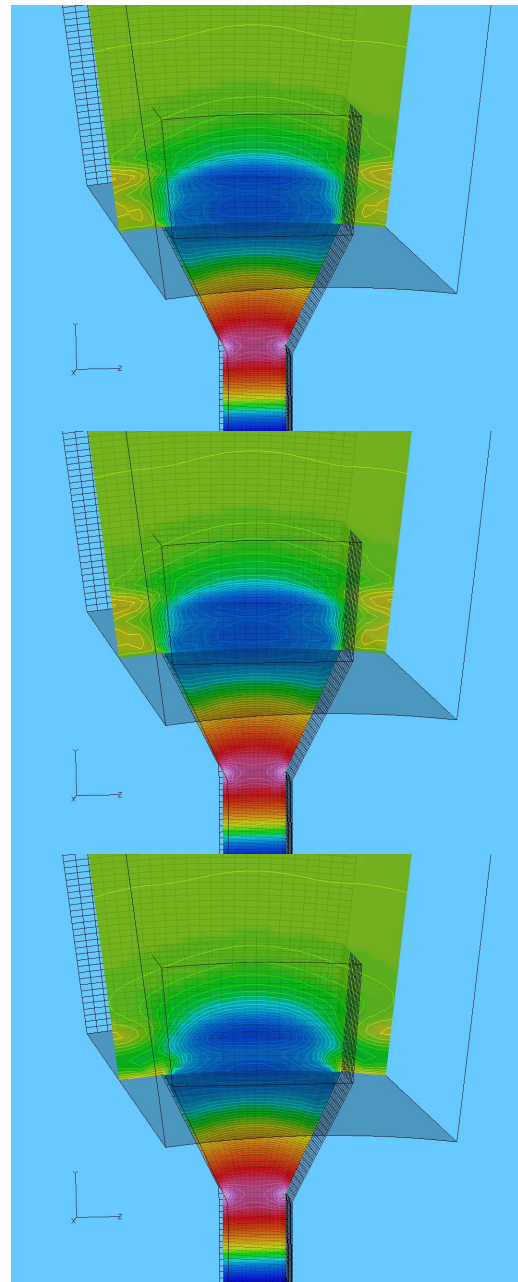


Figure 3. Effect of the electron temperature on the wave propagation from the horn antenna. Top: $T_{e,ref} = 0.1$ eV. Center: $T_{e,ref} = 100$ eV. Bottom: $T_{e,ref} = 5.0$ keV.

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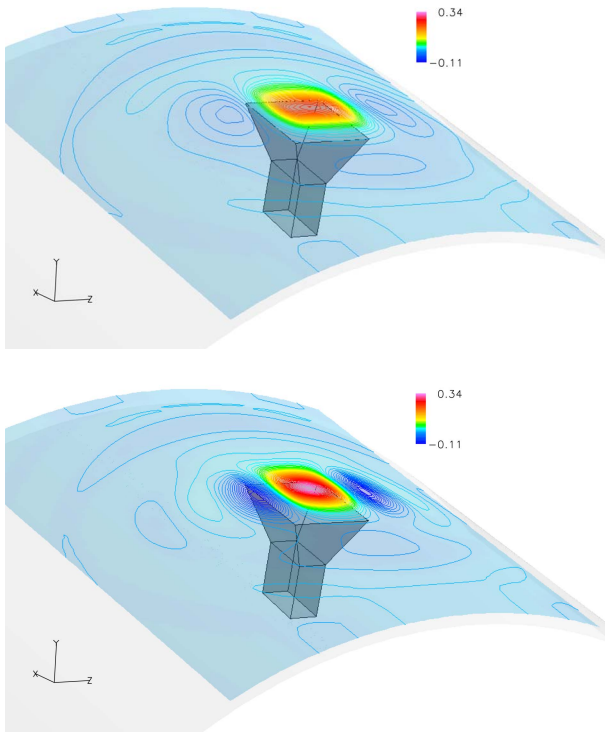


Figure 4. Contours of current above the horn antenna. Top: $T_{e,ref} = 0.1$ eV or 100 eV. Bottom: $T_{e,ref} = 5.0$ keV

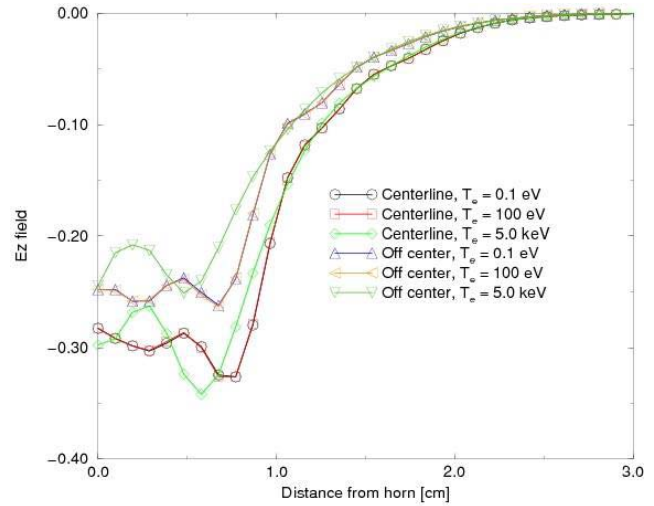


Figure 5. Effect of electron temperature on the electric field strength as it propagates from the horn.

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