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1 Attorney Docket No. 79538

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3 TELESCOPING PRESSURE BALANCED GAS

4 GENERATOR LAUNCHERS FOR UNDERWATER USE

5

6 STATEMENT OF GOVERNMENT INTEREST

7 The invention described herein may be manufactured and used
8 by or for the Government of the United States of America for
9 governmental purposes without the payment of any royalties
10 thereon or therefor.

11

12 BACKGROUND OF THE INVENTION

13 (1) Field of the Invention

14 This invention generally relates to a pressure balanced gas
15 generator launcher.

16 More particularly, the invention relates to a pressure
17 balanced gas generator launcher having telescoping elements
18 capable of extending the provision of launch forces.

19 (2) Description of the Prior Art

20 Currently, pressure balanced gas generator launchers are
21 under development. Ramming length limitations require that the
22 gas generator accelerate the projectile very rapidly in order to
23 provide the required launch speed. These relatively high
24 accelerations require that the projectile be designed to very
25 high shock standards. Thus, a problem exists in the art whereby

1 it is necessary to develop launchers providing high accelerations
2 in order to accommodate those projectiles requiring a high launch
3 speed.

4 The following patents, for example, disclose various types
5 of launchers, but do not disclose a gas generator launcher as
6 does the present invention which utilizes extended expansion from
7 a single primary cylinder housing.

8 U.S. Patent No. 3,256,777 to Choate et al.;

9 U.S. Patent No. 3,371,578 to Choate et al.;

10 U.S. Patent No. 3,745,876 to Rocha;

11 U.S. Patent No. 3,859,890 to Guthrie;

12 U.S. Patent No. 4,037,821 to Greene;

13 U.S. Patent No. 4,531,445 to Nee;

14 U.S. Patent No. 4,616,554 to Spink et al.; and

15 U.S. Patent No. 5,398,588 to Peck.

16 Specifically, Choate et al. '777 disclose a rocket launcher
17 having first and second tubular sections that are telescopingly
18 connected to enable the launcher to be extended into an elongated
19 position of use from a shortened, inoperative position with the
20 first section being the breech end and fitting within the second
21 section. During movement of the launcher sections from a
22 predetermined partly extended position into a fully extended
23 position, the firing member is moved into its cocked position,
24 the firing spring becomes operatively tensioned, and the launcher
25 sections become locked together.

1 The patent to Choate et al. '578 discloses a rocket launcher
2 of the type having first and second tubular sections
3 telescopingly connected to enable them to be extended from a
4 shortened inoperative position into an elongated operative
5 position.

6 The general objective of Choate et al. '578 is to provide
7 that type of launcher with improved safety features and to
8 provide basic constructions for those in production of launchers
9 whether the rockets to be launched therefrom are to be ignited by
10 percussion or electrically and also in the production of
11 launchers for use in firing sub-caliber practice rounds.

12 Rocha discloses a telescopic firearm including a firing tube
13 for the discharge of a projectile, a blast deflector, a flash
14 deflector for receiving said firing tube and blast deflector when
15 telescoped thereinto, covers mounted by hinges on said flash
16 deflector for closing the ends thereof, and a sight system
17 incorporated in said covers.

18 Guthrie discloses a projectile launching device in which a
19 projectile or warhead is mounted in an ejection tube in the form
20 of a piston and the ejection tube is mounted in a launching tube.
21 A propellant charge in the ejection tube supplies gas pressure to
22 the projectile to force the projectile out the ejection tube, and
23 a rocket nozzle located to the rear of the propellant charge
24 receives gases from a combustion of propellant charge to eject
25 the ejection tube from the launching tube.

1 The patent to Greene discloses a telescoping retractor
2 comprised of a two stage cylinder assembly actuated by an
3 electroexplosive power cartridge. A pair of cylinders are
4 mounted in concentric relation. A piston rod is carried in the
5 inner cylinder. The power cartridge produces the gas which
6 sequentially retracts the outer cylinder, the inner cylinder and
7 then the piston which is secured to the structure to be
8 retracted.

9 Nee discloses a projectile launcher having first and second
10 telescopingly connected sections enabling it to be extended into
11 an operative state when it is to be discharged. The firing
12 mechanism has a firing member carried by the first section which
13 extends into a housing on the second section. The rear of the
14 housing is a chamber for a pivoted trigger having a forward
15 portion of greater length than its rearward portion and
16 underlying a boot covered part. A rotary sear in the chamber has
17 first and second arcuately spaced shoulders the second of which
18 is engaged and held by the rearward trigger portion under the
19 influence of a spring. When but a short further relative
20 movement between the sections is required to fully extend the
21 launcher, the first sear shoulder engages and pulls forward the
22 firing member thus to cock the firing mechanism when the launcher
23 is fully extended, the firing spring then exerting a substantial
24 force against the first shoulder. While the difference in
25 lengths of the trigger portion offers the user a theoretical

1 mechanical advantage, the place where the effective pressure of
2 the fingers applied thereagainst through the boot is an uncertain
3 and variable factor causing inaccuracy in use due to the
4 resulting variations in the force required to pull the trigger.
5 The radial distances of the sear shoulders are such that the
6 force exerted by the second sear shoulder is so reduced that said
7 factor is minimized to an extent enabling the use of such
8 launchers to be attended with increased accuracy.

9 The patent to Spink et al. discloses an extendable rocket
10 launcher tube for a rocket wherein releasable telescoped
11 concentric inner and outer tubes house the rocket. Upon
12 ignition, the rocket releases the inner tube from the outer tube
13 and carries it forward imparting momentum thereto. The rocket is
14 released from the inner tube at a selected axial extension of the
15 inner tube. The components move at relative speeds such that
16 acceleration of the rocket is complete when the launcher tube is
17 fully extended.

18 Peck discloses a missile launching system for launching a
19 missile including an automatically telescoping launch tube and a
20 restraining system. The telescoping launch tube lengthens the
21 period of guided travel of the missile during launch, to improve
22 its accuracy. The launch tube system has an outer case and an
23 inner launch tube internally dimensioned to receive the missile
24 therein and externally dimensioned to slide within the outer case
25 from a rearward position to a forward position. The restraining

1 system holds the missile and inner launch tube at the proper
2 locations before and during firing, and includes a releasable
3 holdback for the inner launch tube at the rearward position and a
4 releasable holdback for the missile at a preselected position
5 within the inner launch tube. The inner launch tube is
6 restrained so that it may not slide past the forward position
7 relative to the outer case at the end of its travel during the
8 launching sequence.

9 It should be understood that the present invention would in
10 fact enhance the functionality of the above patents by providing
11 substantial additional extension from a single primary cylinder
12 housing by permitting staged and advancing telescoping functions
13 from both ends of the primary cylinder housing. The primary
14 housing holds plural internal cylinders which selectively expand
15 in reaction to a gas generator.

16

17 SUMMARY OF THE INVENTION

18 Therefore it is an object of this invention to provide a gas
19 generator launching apparatus.

20 Another object of this invention is to provide a gas
21 generator launching apparatus having reduced launch acceleration.

22 Still another object of this invention is to provide a gas
23 generator launching apparatus capable of providing the necessary
24 launch velocity from a compact space.

1 A still further object of the invention is to provide a gas
2 generator launching apparatus in which internal and external
3 pressures are balanced for functioning in pressurized
4 environments.

5 Yet another object of this invention is to provide a gas
6 generator launching apparatus having an extended ramming length
7 for use in an underwater environment.

8 In accordance with one aspect of this invention, there is
9 provided a launching apparatus including a launch tube having a
10 fore end and an aft end, each end having an opening formed
11 therein. A muzzle cap is fit to the fore end of the launch tube
12 and a breech mounting is formed at the aft end of the launch
13 tube. A primary plenum housing is seated in the aft end of the
14 launch tube, the primary housing having a fore end and an aft
15 end. A first telescoping cylinder is initially positioned within
16 the fore end of the primary housing and a second telescoping
17 cylinder is initially positioned within the aft end of the
18 primary housing. A first telescoping piston is positioned in the
19 first telescoping cylinder, and a second telescoping piston is
20 positioned in the second telescoping cylinder. A ram plate is
21 connected to the first telescoping piston, and the second
22 telescoping piston is connected to the breech mounting. A gas
23 generator is provided in connection with said primary housing. A
24 projectile is seated between the ram plate and the muzzle cap.
25 The first and second telescoping cylinders and the first and

1 second telescoping pistons expand in multiple stages of extension
2 from the main housing and propel the projectile.

3

4 BRIEF DESCRIPTION OF THE DRAWINGS

5 The various objects, advantages and novel features of this
6 invention will be more fully apparent from a reading of the
7 following detailed description in conjunction with the
8 accompanying drawings in which like reference numerals refer to
9 like parts, and in which:

10 FIG. 1 is a side sectional view of a launcher configuration
11 according to the present invention;

12 FIG. 2 is a side sectional view of a telescoping plenum
13 assembly schematic in pre-launch according to a preferred
14 embodiment of the present invention;

15 FIG. 3A is a schematic diagram showing a first stage of
16 launch;

17 FIG. 3B is a schematic diagram showing a second stage of
18 launch;

19 FIG. 3C is a schematic diagram showing a third stage of
20 launch;

21 FIG. 3D is a schematic diagram showing end of launch; and

22 FIG. 4 is a schematic diagram showing an alternative
23 embodiment of a launcher configuration according to the present
24 invention.

1 DESCRIPTION OF THE PREFERRED EMBODIMENT

2 In general, the present invention is directed to a
3 telescoping pressure-balanced gas generator launcher. The
4 primary purpose of a pressure-balanced gas generator launcher is
5 to increase the ramming length over that of the non-telescoping
6 pressure-balanced gas generator. By increasing the ramming
7 length, the necessary acceleration is reduced for projectiles
8 requiring high exit velocities.

9 Referring first to FIG. 1 of the present invention, the
10 telescoping pressure-balanced gas generator launcher is generally
11 illustrated at 10. The launcher 10 includes a fore end 12, an
12 aft/breech end 14, and an elongate housing portion 16. The fore
13 end 12 of the launcher 10 has an opening 18 formed therein.
14 Before launch, the opening 18 is blocked by a removable muzzle
15 cap 20. The muzzle cap 20 is surrounded by sealing members 22
16 thereby providing a water tight seal for an interior of the
17 elongate housing 16 of the launcher 10. The aft end 14 of the
18 housing 16 includes a partial end wall or strut, also referred to
19 as a breech mounting 24. An opening 26 allows communication of
20 environmental fluid such as seawater 34 into the aft end 14 of
21 housing 16.

22 A telescoping plenum assembly 28 is located adjacent the aft
23 end 14 of the housing 16 and is connected at one end to the
24 breech mounting 24 and at its opposite end to a ram plate 30.
25 Prior to use, the telescoping plenum assembly 28 is surrounded by

1 seawater 34. A launch projectile 32 is seated within the housing
2 16 between the ram plate 30 and the muzzle cap 20. The ram plate
3 30 is sealed against an inner surface 17 of the housing 16 with
4 sealing member 22 which prevent seawater 34 from contacting the
5 launch projectile 32.

6 Referring now to FIG. 2, the telescoping plenum assembly 28
7 includes key components of the launcher configuration by
8 incorporating staged and multiple expansions of elements therein.
9 Specifically, the telescoping plenum assembly 28 is formed as a
10 single main cylinder 38 having a first end 40 and a second end
11 42. An opening 44 is formed in the first end 40 and an opening
12 46 is formed in the second end 42 of the plenum assembly 28.

13 At least two telescoping cylinders are housed within the
14 plenum assembly 28. One of the telescoping cylinders is a first
15 inner cylinder 50 and another of the telescoping cylinders is a
16 second inner cylinder 52. The first inner cylinder 50 includes a
17 distal end 54 and a proximate end 56, such that the distal end 54
18 terminates in the opening 44 of the main cylinder 38. The second
19 inner cylinder 52 also includes a proximate end 58 and a distal
20 end 60, such that the distal end 60 terminates in the opening 46
21 of the main cylinder 38. Sealing rings 23 or similar appropriate
22 devices are provided at the distal end 54 of the first inner
23 cylinder 50 and at the distal end 60 of the second inner cylinder
24 52 to seal out the external environment of, for example, seawater
25 34.

1 The proximate end 56 of the first inner cylinder 50 and the
2 proximate end 58 of the second inner cylinder 52 terminate in a
3 flange 62, 64, respectively. Flange 62 includes an outward
4 radial portion 62a and an inner rib 62b. The outward radial
5 portion 62a is in surface contact with the inner surface 17 of
6 the housing 16 and the inner rib 62b is smaller in diameter than
7 an inner diameter of the first inner cylinder 50. A first piston
8 member having a piston head 66 and a piston shaft 68 is slidably
9 positioned within the first inner cylinder 50 such that in a
10 retracted position the piston head 66 abuts the inner rib 62b of
11 the flange 62. A distal end of the first piston shaft 68 is
12 connected to the ram plate 30.

13 Flange 64 includes an outward radial portion 64a and an
14 inner rib 64b. The outward radial portion 64a is in surface
15 contact with the inner surface 17 of the housing 16 and the inner
16 radial portion 64b is smaller in diameter than an inner diameter
17 of the second inner cylinder 52. A second piston member having a
18 piston head 70 and a piston shaft 72 is formed within the second
19 inner cylinder 52 such that the piston head 70 abuts the inner
20 rib 64b of the flange 64. A distal end of the second piston
21 shaft 72 is connected to the breech mounting 24. Each piston
22 head 66, 70 is slidably sealed by seal 25 within its respective
23 inner cylinder 50, 52. A gas generator 36 is positioned in a
24 plenum 99 within housing 16 between the first inner cylinder 50

1 and the second inner cylinder 52. Gas generator 36 can be
2 activated by any control apparatus well known in the art.

3 With the arrangement of elements as described, it will be
4 understood that the first inner cylinder 50 is telescopingly
5 slidable through the opening 44 of the main cylinder 38, and the
6 shaft 68 of the first piston is telescopingly slidable through
7 the outer opening 50a of the first inner cylinder. Likewise, the
8 second inner cylinder 52 is telescopingly slidable through the
9 opening 46 of the main cylinder 38, and the shaft 72 of the
10 second piston is telescopingly slidable through the outer opening
11 52a of the second inner cylinder 52.

12 Upon complete expansion of the plenum assembly 28, the outer
13 flanged portion 62a of the first inner cylinder 50 is in contact
14 with an inner end surface 17a of the main cylinder 38, the piston
15 head 66 of the first piston is in surface contact with the outer
16 opening 50a of the first inner cylinder 50, the flanged portion
17 64 of the second inner cylinder 52 is in surface contact with the
18 inner breech end 17b of the main cylinder 38, and the piston head
19 70 of the second piston member is in surface contact with the
20 outer opening 52a of the second inner cylinder 52.

21 FIGS. 3A, 3B, 3C, and 3D illustrate the operation of the
22 launcher device 10. It should be noted that the expansion of the
23 plenum/piston stages does not necessarily occur in the order of
24 these FIGS., and may in fact occur simultaneously.

1 In FIG. 3A, when the gas generator 36 pressurizes the plenum
2 99, the first inner cylinder 50 and corresponding piston slide
3 forward inside the main cylinder 38, pushing the ram-plate 30,
4 projectile 32, and muzzle cap 20 out of the launch tube 10. In
5 FIG. 3B, the first inner cylinder 50 catches once the flange 62
6 reaches the first end 40 of the main cylinder 38, but the first
7 piston continues to slide forward until the head 56 reaches the
8 outer end 50a of the first inner cylinder 50. In FIG. 3C, the
9 main cylinder 38 begins to move forward dragged by the flange 62
10 of the first inner cylinder 50. When the aft end 42 of the main
11 cylinder 38 catches on the flange 64 of the second inner cylinder
12 52, the second inner cylinder 52 begins sliding forward. In FIG.
13 3D, the plenum 99 continues to expand (and push out the
14 projectile 32) until the outer end 52a of the second inner
15 cylinder 52 catches on the second piston head 70. Pistons and
16 cylinders are sealed against each other as shown with sealing
17 members 22, 23.

18 Since ocean pressure acts on both ends of the projectile 32
19 (through the muzzle cap 20 and the ram plate 30), the gas
20 generator 36 need only supply enough energy to overcome losses,
21 expand the volume of the plenum assembly 28, and accelerate the
22 stages of the telescoping plenum assembly 28, the ram plate 30,
23 the projectile 32, and the muzzle cap 20. The key to keeping
24 this launcher truly pressure-balanced is minimizing the volume of
25 the telescoping cylinders 50, 52. Otherwise, the gas generator

1 36 must do more work against ocean pressure in order to expand
2 the plenum assembly 28.

3 The launch tube 10 and breech mounting 14 are reusable since
4 they don't come in contact with the combustion products of the
5 gas generator 36. Depending on the nature of the combustion
6 products, the telescoping plenum 28 may also be reusable during a
7 limited lifetime.

8 FIG. 4 shows an alternate embodiment of the plenum assembly
9 28 of the current invention. This embodiment provides a main
10 cylinder 100 having a single inner cylinder 102 and piston
11 assembly 104. The main cylinder 100 is hollow and has a forward
12 end 106 and an aft end 108 with the aft end 108 being sealed and
13 joined directly to the breech mounting 24. The forward end 106
14 includes a stop ring 110 positioned thereon such that the stop
15 ring 110 has a radius reduced from that of the main cylinder 100.

16 The inner cylinder 102 is likewise hollow and includes an
17 outwardly flanged aft end 112 and an inner cylinder forward end
18 114. The inner cylinder 102 is positioned within the main
19 cylinder 100 such that the flanged aft end 112 is oriented toward
20 the main cylinder aft end 108. A gas source 116 is provided
21 inside main cylinder 100 and adjacent the flanged aft end 112 of
22 the inner cylinder 102. The flanged aft end 112 has an increased
23 external radius from the external radius of the inner cylinder
24 102. A sliding fit exists between the flanged aft end 112 and an
25 inner surface of the main cylinder 100, and an outer periphery of

1 the flanged aft end 112 is sealed against the main cylinder 100
2 by a plurality of sliding seals 118. The flanged aft end 112
3 also has a shoulder 120 extending into the hollow portion of the
4 inner cylinder and thereby reducing the inner diameter thereof.

5 The forward end 114 of the inner cylinder 102 extends
6 through the stop ring 110 at the forward end 106 of the main
7 cylinder 100. The inner surface of the stop ring 110 has a
8 bearing surface therein allowing sliding of the inner cylinder
9 102 out from the main cylinder 100. The forward end 114 of the
10 inner cylinder 102 has a piston support flange 122 for narrowing
11 the radius of the interior surface of the inner cylinder 102. A
12 plurality of fluid communication ports 124 are provided in the
13 piston support flange 122, thereby allowing communication between
14 the environment and the hollow interior of the inner cylinder
15 102.

16 The flanged aft end 112 and stop ring 110 bearing surface
17 prevent canting of the inner cylinder 102 with respect to the
18 main cylinder 100 by supporting the inner cylinder 102. A
19 plurality of fluid communication ports 126 are formed in the stop
20 ring 110 to allow environmental fluid to escape from the hollow
21 portion remaining between the main cylinder 100 and the inner
22 cylinder 102 when the inner cylinder slides with respect to the
23 main cylinder 100. Upon extension of the inner cylinder 102 from
24 the main cylinder 100, the increased radius of the flanged aft
25 end 112 engages with the decreased radius of the stop ring 110,

1 thereby preventing the inner cylinder 102 from leaving the main
2 cylinder 100.

3 Communication ports such as 124 and 126 are optional because
4 they introduce the risk of corrosion from seawater into the
5 piston assembly 104 and generate viscous drag as environmental
6 fluid is pushed out of the piston assembly 104. The pressure
7 generated by the gas source 116 is more than enough to overcome
8 the back pressure generated by gas trapped in between the
9 cylinders 100 and 102. The back pressure has the benefit of
10 preventing contact between the flange 112 and stop ring 110 at
11 the end of the piston stroke. Check valves could be provided in
12 communication ports 124 and 126 to prevent entry of seawater
13 therethrough.

14 Piston assembly 104 including a piston 128 and a piston
15 shaft 130 is slidably disposed in the hollow portion of the inner
16 cylinder 102. The piston 128 is sealed against an inner surface
17 of the inner cylinder 102 with a sliding seal 132. The piston
18 shaft 130 is joined to the piston 128 and extends forward through
19 the piston support flange 122. At its extreme forward end, the
20 piston shaft 130 is joined to the ram plate 30. When not
21 extended, the piston 128 is positioned against the shoulder 120
22 of the flanged aft end 112. The piston support flange 122
23 engages with the piston 128 and prevents the piston assembly 104
24 from sliding out of the hollow interior portion of the inner
25 cylinder 102.

1 Upon activation of the gas source 116, pressure is generated
2 in a plenum defined between the surface created by the inner
3 cylinder flanged aft end 112 and piston 128 and the main cylinder
4 aft end 108. Generated pressure pushes the inner cylinder 102
5 forward from the main cylinder 100 and the piston assembly 104
6 forward from the inner cylinder 102. The ram plate 30 is pushed
7 outward by contact with the piston shaft 130. Environmental
8 fluid is expelled from the region between the main cylinder 100
9 and inner cylinder 102 through fluid communication ports 126.
10 Likewise, fluid is expelled from the region between piston
11 assembly 104 and inner cylinder 102 through fluid communication
12 ports 124.

13 These telescoping pressure balanced gas generator launcher
14 designs increase the ramming length of the non-telescoping
15 pressure-balanced launcher by a factor of two, which cuts the
16 average acceleration requirement by one-half.

17 Alternatives to this design of a telescoping pressure-
18 balanced gas generator launcher include variations in the number
19 of telescoping cylinders and in the number of main cylinders
20 attachable to one another so as to multiply the ramming length.
21 Variations in cylinder/piston geometry also may occur (i.e. they
22 need not be truly "cylindrical"). Many different types of gas
23 generators could be used to pressurize the plenum.

1 Attorney Docket No. 79538

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TELESCOPING PRESSURE-BALANCED GAS

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GENERATOR LAUNCHERS FOR UNDERWATER USE

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ABSTRACT OF THE DISCLOSURE

7

8 A launching apparatus including a launch tube having a fore
9 end and an aft end, each end having an opening formed therein. A
10 muzzle cap is fit to the fore end of the launch tube and a breech
11 mounting is formed at the aft end of the launch tube. A primary
12 plenum housing is seated in the aft end of the launch tube, the
13 primary housing having a fore end and an aft end. A first
14 telescoping cylinder is initially positioned within the fore end
15 of the primary housing and a second telescoping cylinder is
16 initially positioned within the aft end of the primary housing.
17 A first telescoping piston is formed in the first telescoping
18 cylinder and a second telescoping piston formed in the second
19 telescoping cylinder. A ram plate is connected to the first
20 telescoping piston and the second telescoping piston is connected
21 to the breech mounting. A gas generator is provided within said
22 primary housing and a projectile is seated between the ram plate
23 and the muzzle cap. The first and second telescoping cylinders
24 and the first and second telescoping pistons expand in multiple
25 stages of extension from the main housing and propel the
projectile.

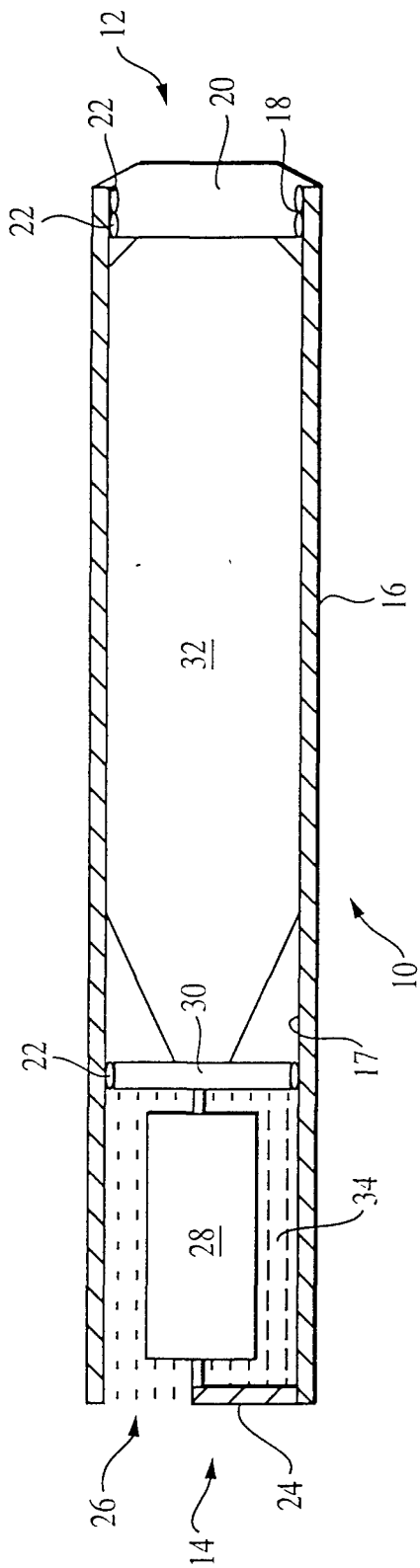


FIG. 1

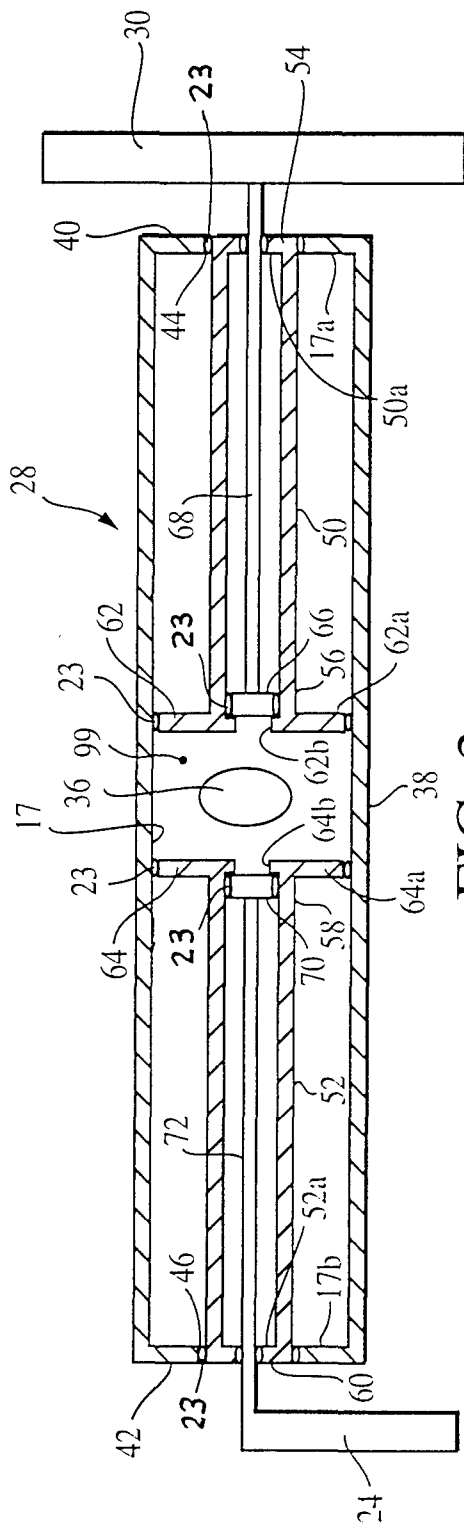


FIG. 2

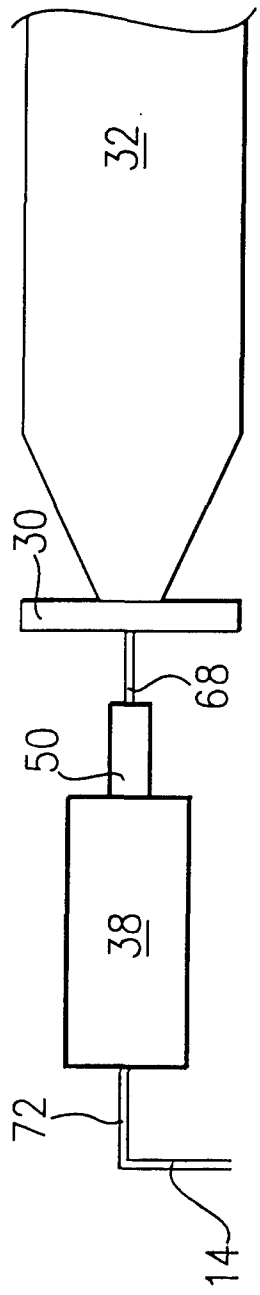


FIG. 3A

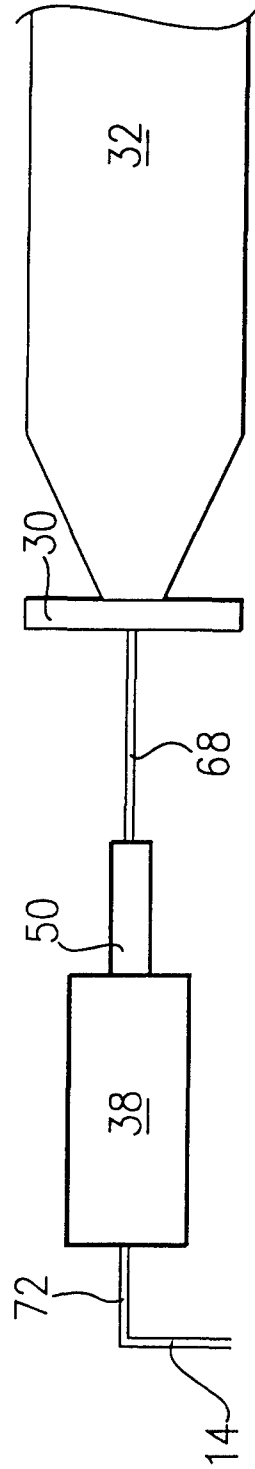


FIG. 3B

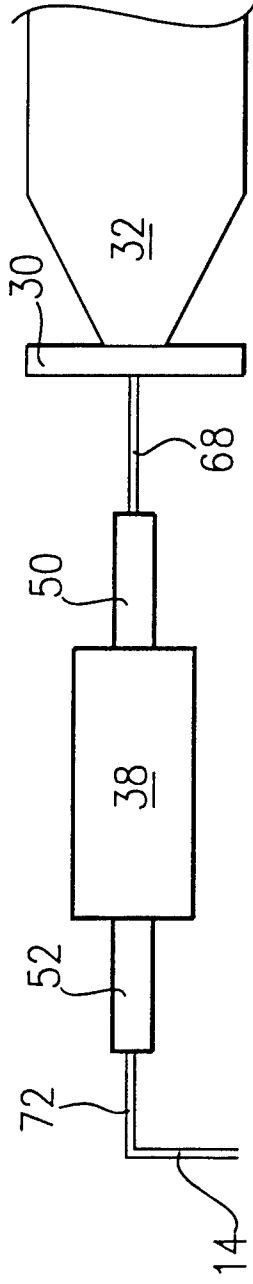


FIG. 3C

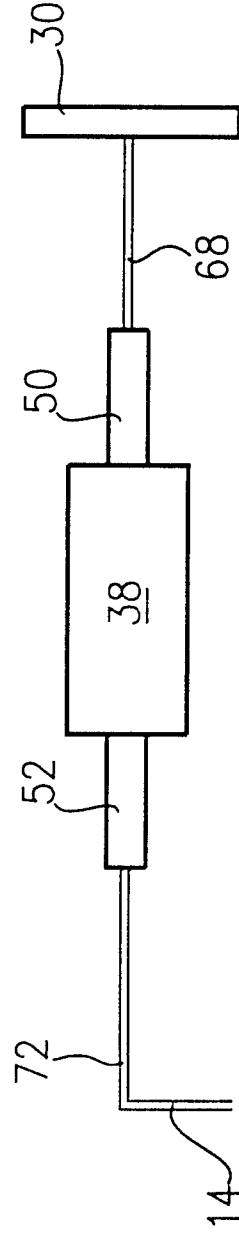


FIG. 3D

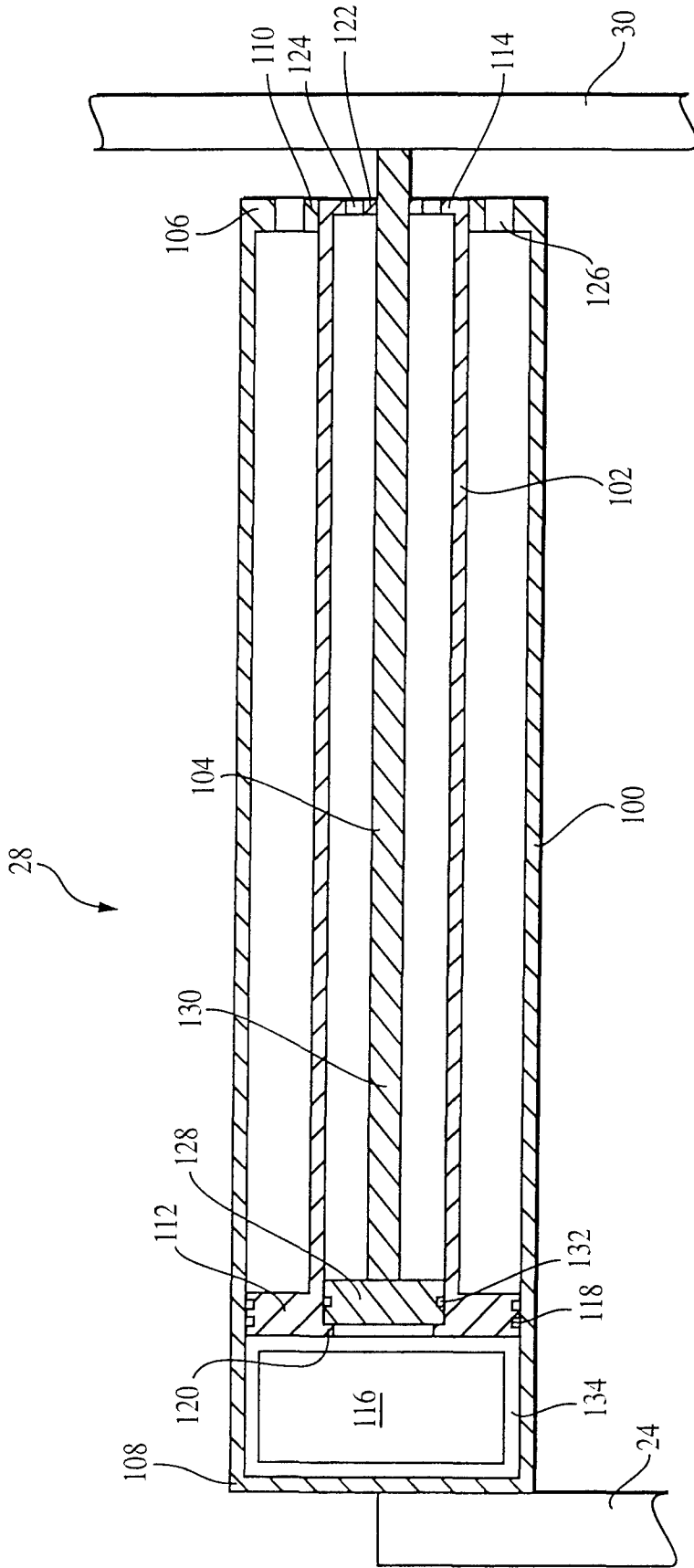


FIG. 4